

Augmented Reliability Analysis for Estimating Imprecise First Excursion Probabilities in Stochastic Linear Dynamics

Matthias G.R. Faes^{a,b,*}, Marcos A. Valdebenito^c, Xiukai Yuan^d, Pengfei Wei^{b,e}, Michael Beer^{b,f,g}

^a*KU Leuven, Department of Mechanical Engineering, Technology campus De Nayer, Jan De Nayerlaan 5,
St.-Katelijne-Waver, Belgium*

^b*Institute for Risk and Reliability, Leibniz Universität Hannover, Callinstr. 34, 30167 Hannover, Germany*

^c*Faculty of Engineering and Sciences, Universidad Adolfo Ibáñez, Av. Padre Hurtado 750, 2562340 Viña del
Mar, Chile*

^d*School of Aerospace and Engineering, Xiamen University, Xiamen 361005, P.R. China*

^e*School of Mechanics, Civil Engineering and Architecture, Northwestern Polytechnical University, Xi'an 710072,
China*

^f*Institute for Risk and Uncertainty and School of Engineering, University of Liverpool, Peach Street, Liverpool
L69 7ZF, UK*

^g*International Joint Research Center for Engineering Reliability and Stochastic Mechanics, Tongji University,
1239 Siping Road, Shanghai 200092, P.R. China*

Abstract

Imprecise probability allows quantifying the level of safety of a system taking into account the effect of both aleatory and epistemic uncertainty. The practical estimation of an imprecise probability is usually quite demanding from a numerical viewpoint, as it is necessary to propagate separately both types of uncertainty, leading in practical cases to a nested implementation in the so-called double loop approach. In view of this issue, this contribution presents an alternative approach that avoids the double loop by replacing the imprecise probability problem by an augmented, purely aleatory reliability analysis. Then, with the help of Bayes' theorem, it is possible to recover an expression for the failure probability as an explicit function of the imprecise parameters from the augmented reliability problem, which ultimately allows calculating the imprecise probability. The implementation of the proposed framework is investigated within the context of imprecise first excursion probability estimation of uncertain linear structures subject to imprecisely defined stochastic quantities and crisp stochastic loads. The associated augmented reliability problem is solved within the context of Directional Importance Sampling, leading to an improved accuracy at reduced numerical costs. The application of the proposed approach is investigated by means of two examples. The results obtained indicate that the proposed approach can be highly efficient and accurate.

Keywords: Uncertain Linear Structure, stochastic loading, Imprecise first excursion probability, Augmented reliability problem, Directional Importance Sampling

*E-mail: matthias.faes@kuleuven.be

35 *Highlights:*

- 36 • Imprecise reliability problem is replaced by an augmented reliability problem.
- 37 • Bayes' theorem allows retrieving relevant information out of augmented reliability.
- 38 • Focus on uncertain linear structures described by imprecise random properties subject to
39 stochastic loads.
- 40 • Augmented reliability problem is solved with Directional Importance Sampling.

41 **1. Introduction**

42 Numerical tools to approximate the solution of (sets of) differential equations have become in-
43 dispensable in the design of components from the micro-scale to complete structures, all subjected
44 to loads that are representative for those they encounter during their functional life. Thanks to
45 these tools, an engineer is now able to design, test and optimize designs long before a first pro-
46 totype is built. However, despite the highly detailed numerical predictions that can be obtained,
47 the results of these calculations often show a non-negligible discrepancy with the actual physical
48 behaviour of the structure. At the core of this discrepancy lies epistemic (= lack of knowledge)
49 and aleatory (= caused by inherent variation) uncertainty in the description of the model physics,
50 as well as the governing parameters. In recent years, several highly performing methods based on
51 stochastic analysis [1], Fuzzy set theory and Interval analysis [2] have been introduced in literature
52 to account for respectively aleatory and epistemic uncertainties in the model parameters. Also,
53 recent studies compared the applicability of several of these techniques in applications such as
54 geotechnical engineering [3] or inverse uncertainty quantification for stochastic dynamics [4, 5].

55 Uncertainties, be it epistemic or aleatory, are commonly encountered in the context of struc-
56 tural dynamics. This is especially true when considering natural phenomena such as earthquakes
57 or wind loads, since the corresponding dynamical loads that act on the system often cannot be
58 described in a crisp way. Similarly, when considering natural materials and/or products manufac-
59 tured using highly advanced production techniques, such uncertainties may arise. Stochastic pro-
60 cesses, see, e.g. [6, 7], provide a rigorous framework to deal with the uncertainties and space/time
61 correlations of uncertain loads and properties by resorting to the well-documented framework of

probability theory, which is highly suited to deal with aleatory uncertainties. However, in practice, the analyst is often confronted with limited, incomplete or conflicting sources of data (i.e., epistemic uncertainty). In this case, the application of a pure probabilistic framework to take this additional level of uncertainty into account is questionable since in this case, there is simply not enough information to construct an objective probabilistic uncertainty model. In this case, by introducing probability density functions to represent lack of knowledge, one automatically inserts subjective information into the analysis, which might cause further divergence from the actual physics. The framework of imprecise probabilities (see, e.g. [3]) might in this case offer a more objective framework, since, rather than assuming a specific probability measure, it incorporates credal sets of probability measures to fully represent all sources of uncertainty. In practice, this allows to explicitly consider a set of probabilistic models, each consistent with the definition of the epistemic uncertainty, and to infer bounds on the possible model behaviour [8]. For example, in the context of structural dynamics, rather than computing a crisp value of the first excursion probability given a crisp stochastic process load (e.g., following the methods proposed in [9]), an imprecise probabilistic calculation provides bounds on the first excursion probability given a set of stochastic processes that are consistent with the epistemic uncertainty. This allows both to assess the sensitivity of the model reliability to the existing epistemic uncertainty, as well as to provide an estimate of the lower bound of the reliability. Similar observations can be made when considering imprecisely defined stochastic structural properties, which is the main focus of this contribution.

In engineering practice, however, the application of the framework of imprecise probabilities to infer bounds on the probability of failure is hindered by the computational cost of propagating both sources of uncertainty (aleatory and epistemic) jointly towards the model responses. The high computational cost is a direct result of the fact that the propagation has to be conducted such that the effects of aleatory and epistemic uncertainty are kept separated [10]. Hence, double loop approaches are typically applied, where the outer loop takes care of epistemic uncertainty while the inner loop deals with aleatory uncertainty. Many recent works deal with improving the numerical efficiency of propagation schemes for imprecise probabilities. In general four groups of approaches can be considered: (1) series expansion methods, (2) surrogate modelling schemes, (3) decoupling approaches and (4) augmented space techniques. Series expansion methods rely on the approximation of the epistemic uncertain parameters via series expansion methods (see e.g., [11],

93 [12]) or orthogonal polynomial expansion schemes (see e.g., [13]), effectively enabling propagation
94 without double loop approaches. However, in case the epistemic uncertainty is comparatively large,
95 perturbation approaches are known to be inaccurate [2], a problem that is alleviated by resorting
96 to Chebyshev polynomial based schemes such as presented in [14]. Considering surrogate mod-
97 elling schemes, many efficient techniques for the propagation of imprecise probabilistic problems
98 have been proposed using sparse polynomial chaos expansion representations of the model (see
99 e.g., [15, 16]), interval predictor models [17, 18] or variants of the Sobol-Hoeffding decomposition
100 (also known as HDMR representation) of the relation between the epistemic parameters and the
101 probability of failure [19, 20], providing an efficient and accurate approximation of the problem.
102 Decoupling approaches, such as presented in recent work by some of the authors (see [21, 22, 23])
103 are proven to be extremely efficient, but their application is limited to linear models. Finally,
104 the idea of using augmented space methods was introduced by [24] in the context of probabilistic
105 failure analysis and further developed for sensitivity analysis and reliability-based optimization
106 in, e.g. [25, 26, 27, 28, 29]. In the context of imprecise probabilities, similar methods were intro-
107 duced by [19, 20, 30] and independently by [31] in a different form. Following these approaches,
108 the main idea is to propagate the epistemic and aleatory uncertainty jointly in a purely aleatory,
109 augmented space that is optimal with respect to a certain well-defined measure, in such a way
110 that both sources of uncertainty can be decoupled again at the response side.

111 This contribution is situated in the latter category: augmented space methods, and it aims at
112 solving a particular class of problems, that is, those of bounding the first excursion probability
113 of a system subjected to a Gaussian excitation, where epistemic uncertainty is present in the
114 hyper-parameters of the distributions of several uncertain model quantities (i.e., parameterized
115 p-boxes). Imprecise stochastic loads, such as considered in [21], are not considered in this con-
116 tribution. The proposed approach converts the p-box valued uncertain parameters into aleatory
117 uncertainties in an augmented space. Then, with the help of Bayes' theorem, it is possible to
118 recover an expression for the failure probability as an explicit function of the imprecise param-
119 eters from the augmented reliability problem, which ultimately allows calculating the imprecise
120 probability due to the established functional relationship. The implementation of the proposed
121 framework is investigated within the context of imprecise first excursion probability estimation of
122 uncertain linear structures where epistemic uncertainty is present in the hyper-parameters of the
123 distributions of several uncertain model quantities. The associated augmented reliability problem

is solved within the context of Directional Importance Sampling [32], leading to an improved accuracy at reduced numerical costs. Furthermore, by virtue of Directional Importance Sampling, it is possible to derive an explicit approximation of the probability of failure as a function of the interval-valued hyper-parameters of the uncertain input quantities. The paper is complementary to earlier work by the authors, presented in [21, 22, 23], in the sense that it presents a conceptually totally different framework for the propagation of imprecise probabilities towards the bounds on first excursion probabilities. Indeed, whereas the work in [21, 22, 23] focuses on the operator norm framework to decouple the double loop, the methods presented in this paper are based on solving the problem in an augmented space.

The paper is structured as follows: Section 2 introduces the general framework of augmented methods. Section 3 discusses the application of the framework to the calculation of the first excursion probability of uncertain linear dynamical systems. Section 4 provides two examples to illustrate the application, efficiency and effectivity of the proposed framework. Section 5 lists the conclusions of the work and gives some outlook for future developments.

2. General Framework

2.1. Imprecise Probability

Consider a numerical model that represents the behavior of a system. It is assumed that some input parameters of this model are uncertain. These uncertain parameters are grouped into two vectors \mathbf{z} and \mathbf{y} , whose dimensions are $n_z \times 1$ and $n_y \times 1$, respectively, and which are assumed to be independent. The uncertainty associated with these two vectors is represented in terms of probability density functions $f_{\mathbf{Z}}(\mathbf{z})$ and $f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})$, respectively, where $\boldsymbol{\theta}$ is a vector of dimension $n_\theta \times 1$ that collects distribution parameters such as mean value, standard deviation, etc. Due to issues such as lack of knowledge, imprecision, conflicting sources of information, etc., $\boldsymbol{\theta}$ cannot be identified precisely and hence, it is described as an interval valued vector, whose lower and upper bounds are $\underline{\boldsymbol{\theta}}$ and $\bar{\boldsymbol{\theta}}$, respectively. From the above discussion, it is noted that: the uncertainty associated with \mathbf{z} is aleatoric; the uncertainty associated with $\boldsymbol{\theta}$ is epistemic; and that the uncertainty associated with \mathbf{y} is both aleatoric and epistemic. In fact, the uncertainty associated with \mathbf{y} falls into the category of a parametric probability box (p-box) with specified distribution function [33].

The response of the system under analysis becomes uncertain due to the uncertainty in its in-

154 put parameter vectors \mathbf{z} and \mathbf{y} . Such response is monitored in terms of a performance function
 155 g , which assumes a value equal or smaller than zero whenever an unacceptable system behavior
 156 occurs. In other words, unacceptable behavior occurs whenever $g(\mathbf{z}, \mathbf{y}) \leq 0$. It is important to
 157 note that the performance function does not depend on $\boldsymbol{\theta}$: recall that this is a vector collect-
 158 ing distribution parameters which do not affect the physical behavior of the system but just the
 159 probabilistic description associated with \mathbf{y} . The chances that an unacceptable behavior occur are
 160 measured in terms of a probability of failure p_F , which is defined as:

$$p_F(\boldsymbol{\theta}) = \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{z} \in \Omega_z} I_F(\mathbf{y}, \mathbf{z}) f_{\mathbf{Z}}(\mathbf{z}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) d\mathbf{z} d\mathbf{y}, \quad (1)$$

161 where Ω_y and Ω_z are the sets containing all values that \mathbf{y} and \mathbf{z} may assume, respectively; and
 162 $I_F(\mathbf{y}, \mathbf{z})$ is the indicator function, which is equal to one in case $g(\mathbf{y}, \mathbf{z}) \leq 0$ and zero, otherwise.
 163 It is noted from the equation above that the failure probability p_F does depend on $\boldsymbol{\theta}$. This is
 164 evident, as $\boldsymbol{\theta}$ affects the probabilistic description of the problem. Furthermore, as the uncertainty
 165 associated with $\boldsymbol{\theta}$ is described in terms of an interval, the failure probability itself becomes interval
 166 valued as well. In other words, due to the imprecision associated with the vector of distribution
 167 parameters, the failure probability is actually imprecise. Its lower and upper bounds (denoted as
 168 \underline{p}_F and \bar{p}_F , respectively) can be determined by solving the following two optimization problems.

$$\underline{p}_F = \min_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} (p_F(\boldsymbol{\theta})), \quad (2)$$

$$\bar{p}_F = \max_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} (p_F(\boldsymbol{\theta})). \quad (3)$$

169 The structure of eqs. (2) and (3) indicates that for determining the bounds of the imprecise
 170 failure probability, it is necessary to solve two optimization problems with respect to the epistemic
 171 uncertainty associated with $\boldsymbol{\theta}$. In turn, for a given crisp value of $\boldsymbol{\theta}$, it is necessary to propagate
 172 the aleatoric uncertainty associated with \mathbf{z} and \mathbf{y} in order to calculate the failure probability
 173 as shown in eq. (1). Such structure reveals the challenge associated with the calculation of an
 174 imprecise probability, as it is necessary to propagate both epistemic and aleatoric uncertainty at
 175 different levels. In other words, failure probabilities must be calculated for different crisp values
 176 of $\boldsymbol{\theta}$ as required by the optimization. In turn, as the calculation of failure probabilities demands
 177 performing repeated system analyses for different realizations of the uncertain input parameters
 178 \mathbf{z} and \mathbf{y} , the numerical cost associated with the estimation of an imprecise failure probability

becomes insurmountable, even for simple problems. For the sake of completeness, it should be noted that surrogate modelling schemes, such as e.g., presented in [15] or [19], can alleviate the computational cost of propagating imprecise probabilities drastically. Indeed, in some cases, the application of appropriate surrogate modelling schemes might even render the application of brute-force double loop approaches tractable at acceptable levels of accuracy. Nonetheless, when applying surrogate modelling schemes, it is advantageous to apply more efficient calculation methods to further increase the numerical efficiency. For a thorough discussion on the applicability of decoupling approaches and surrogate modelling schemes, the reader is referred to [34].

2.2. Augmented Reliability Problem

A possible means for avoiding the double loop problem associated with the calculation of imprecise probabilities consists of formulating an augmented reliability problem. This augmented problem was originally devised within the context of sensitivity analysis and reliability-based optimization in [24, 25, 27, 28] and was later applied to calculation of imprecise probabilities arising in problems where the performance function is affected by interval valued parameters [30]. In the following, a novel application of the augmented reliability approach is proposed, where the focus is on estimating imprecise probabilities in case imprecision affects the distribution parameters of some random variables of a reliability problem. This novel application can be seen as an extension of the approach proposed in [35] for sensitivity analysis in structural reliability.

The augmented reliability problem is defined such that an instrumental probability density function $f_{\Theta}(\boldsymbol{\theta})$ is associated with the distribution parameter vector $\boldsymbol{\theta}$. It is emphasized that this probability distribution is only an artifact, which has no physical meaning, as the uncertainty associated with $\boldsymbol{\theta}$ is epistemic (and not aleatoric). In principle, this instrumental probability distribution could be chosen arbitrarily. In this sense, note that the distribution does not have to be bounded between the lower and upper bounds of the associated intervals. In fact, any distribution that is capable of characterizing the effect of the epistemic uncertain parameters from a physical standpoint is admissible. In this contribution, a uniform distribution is selected, such that $f_{\Theta}(\boldsymbol{\theta}) \sim \mathcal{U}(\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}})$. Thus, the failure probability p_F^A associated with the augmented reliability problem is:

$$p_F^A = \int_{\boldsymbol{\theta} \in \Omega_{\theta}} \int_{\mathbf{y} \in \Omega_{\mathbf{y}}} \int_{\mathbf{z} \in \Omega_{\mathbf{z}}} I_F(\mathbf{y}, \mathbf{z}) f_{\mathbf{Z}}(\mathbf{z}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\Theta}(\boldsymbol{\theta}) d\mathbf{z} d\mathbf{y} d\boldsymbol{\theta}, \quad (4)$$

207 where $\Omega_\theta = [\underline{\theta}, \bar{\theta}]$. This augmented failure probability can be calculated using any suitable
 208 reliability method.

209 Let F represent the failure set, that is $F = \{(\mathbf{z}, \mathbf{y}) : g(\mathbf{z}, \mathbf{y}) \leq 0\}$. Applying Bayes' theorem, it is
 210 noted that [24, 25, 30]:

$$P(F|\boldsymbol{\theta}) = p_F(\boldsymbol{\theta}) = \frac{f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|F)p_F^A}{f_{\boldsymbol{\theta}}(\boldsymbol{\theta})}, \quad (5)$$

211 where $P(\cdot)$ denotes probability of occurrence of the argument and $f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|F)$ is the (instrumental)
 212 probability distribution associated with $\boldsymbol{\theta}$ conditioned on failure. Eq. (5) provides an expression
 213 for calculating the failure probability as a function of the imprecise distribution parameters. In
 214 this expression, $f_{\boldsymbol{\theta}}(\boldsymbol{\theta})$ is known (as it is selected arbitrarily) and p_F^A is estimated by means of a
 215 reliability method. Hence, the only term which remains to be evaluated is $f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|F)$. In order to
 216 calculate this term, recall the definition of a marginal distribution, that is:

$$f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|F) = \int_{\mathbf{y} \in \Omega_{\mathbf{y}}} \int_{\mathbf{z} \in \Omega_{\mathbf{z}}} f_{\boldsymbol{\theta}, \mathbf{Y}, \mathbf{Z}}(\boldsymbol{\theta}, \mathbf{y}, \mathbf{z}|F) d\mathbf{z} d\mathbf{y}, \quad (6)$$

217 with $f_{\boldsymbol{\theta}, \mathbf{Y}, \mathbf{Z}}(\boldsymbol{\theta}, \mathbf{y}, \mathbf{z}|F)$ the joint distribution of \mathbf{y} , \mathbf{z} and $\boldsymbol{\theta}$ conditional on the failure set. Applying
 218 Bayes' theorem, it is found that:

$$f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|F) = \int_{\mathbf{y} \in \Omega_{\mathbf{y}}} \int_{\mathbf{z} \in \Omega_{\mathbf{z}}} f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|\mathbf{y}, \mathbf{z}, F) f_{\mathbf{Y}, \mathbf{Z}}(\mathbf{y}, \mathbf{z}|F) d\mathbf{z} d\mathbf{y}. \quad (7)$$

219 Now, focusing on the conditional probability distribution $f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|\mathbf{y}, \mathbf{z}, F)$ and recalling that \mathbf{y} and
 220 \mathbf{z} are independent between them, it is possible to deduce the following:

$$\begin{aligned} f_{\boldsymbol{\theta}}(\boldsymbol{\theta}|\mathbf{y}, \mathbf{z}, F) &= I_F(\mathbf{y}, \mathbf{z}) \frac{f_{\boldsymbol{\theta}, \mathbf{Y}, \mathbf{Z}}(\boldsymbol{\theta}, \mathbf{y}, \mathbf{z})}{f_{\mathbf{Y}, \mathbf{Z}}(\mathbf{y}, \mathbf{z})}, \\ &= I_F(\mathbf{y}, \mathbf{z}) \frac{f_{\boldsymbol{\theta}, \mathbf{Y}}(\boldsymbol{\theta}, \mathbf{y})}{f_{\mathbf{Y}}(\mathbf{y})}, \\ &= I_F(\mathbf{y}, \mathbf{z}) \frac{f_{\boldsymbol{\theta}}(\boldsymbol{\theta}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\int_{\boldsymbol{\theta} \in [\underline{\theta}, \bar{\theta}]} f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\boldsymbol{\theta}}(\boldsymbol{\theta}) d\boldsymbol{\theta}}. \end{aligned} \quad (8)$$

221 The last equation can be further simplified recalling that $f_{\boldsymbol{\theta}}(\boldsymbol{\theta})$ is a constant, as it is selected
 222 as a uniform distribution. Taking into account this fact and defining $\Delta(\mathbf{y}) = \int_{\boldsymbol{\theta} \in [\underline{\theta}, \bar{\theta}]} f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) d\boldsymbol{\theta}$,

eq. (8) reduces to:

223

$$\begin{aligned}
 f_{\Theta}(\boldsymbol{\theta}|\mathbf{y}, \mathbf{z}, F) &= I_F(\mathbf{y}, \mathbf{z}) \frac{f_{\Theta}(\boldsymbol{\theta}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\int_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\Theta}(\boldsymbol{\theta}) d\boldsymbol{\theta}}, \\
 &= I_F(\mathbf{y}, \mathbf{z}) \frac{f_{\Theta}(\boldsymbol{\theta}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{f_{\Theta}(\boldsymbol{\theta}) \int_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) d\boldsymbol{\theta}}, \\
 &= I_F(\mathbf{y}, \mathbf{z}) \frac{f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\Delta(\mathbf{y})}.
 \end{aligned} \tag{9}$$

Inserting eq. (9) into (7) leads to the following expression for $f_{\Theta}(\boldsymbol{\theta}|F)$:

224

$$f_{\Theta}(\boldsymbol{\theta}|F) = \int_{\mathbf{y} \in \Omega_{\mathbf{y}}} \int_{\mathbf{z} \in \Omega_{\mathbf{z}}} \frac{f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\Delta(\mathbf{y})} f_{\mathbf{Y}, \mathbf{Z}}(\mathbf{y}, \mathbf{z}|F) dz d\mathbf{y}, \tag{10}$$

where the indicator function $I_F(\mathbf{y}, \mathbf{z})$ has been omitted, as it assumes a value equal to one given the presence of the probability density $f_{\mathbf{Y}, \mathbf{Z}}(\mathbf{y}, \mathbf{z}|F)$, which ensures that integration is carried out exclusively over the failure domain.

The expression in eq. (10) is quite convenient from a practical viewpoint, as it can be evaluated by means of simulation. This is due to the following: the terms $f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})$ and $\Delta(\mathbf{y})$ are both known either in closed form or numerically. Moreover, samples distributed according to $f_{\mathbf{Y}, \mathbf{Z}}(\mathbf{y}, \mathbf{z}|F)$ can be obtained as a byproduct of a reliability analysis associated with the calculation of the augmented failure probability in eq. (4), see e.g. [24]. Thus, by combining eqs. (5) and (10), it is possible to construct an explicit expression for estimating the failure probability as a function of the vector of distribution parameters $\boldsymbol{\theta}$. This expression can then be used to solve the optimization problems in eqs. (2) and (3) at reduced numerical efforts, as it is no longer required to solve a nested problem.

As a summary, it can be stated that the approach based on an augmented reliability problem can be quite convenient from a numerical viewpoint. This is due to the fact that the calculation of the bounds for the imprecise probability reduces to:

- A single reliability analysis, carried out considering the augmented problem.
- Solution of two optimization problems considering the explicit approximation of the failure probability as a function of $\boldsymbol{\theta}$

243 3. Application to the Calculation of First Excursion Probabilities of Uncertain Linear 244 Dynamical Systems

245 3.1. General Remarks

246 The framework for determining imprecise probabilities as described in Section 2.2 is quite general
247 in the sense that no assumptions have been introduced with respect to the type of system in-
248 volved. In principle, the approach can cope with, e.g. linear or nonlinear systems under static or
249 dynamic loading. Nonetheless, the choice of a specific reliability method for solving the associated
250 augmented failure probability is heavily influenced by the type of system. For the particular case
251 of this contribution, the focus is on the analysis of uncertain linear structural systems subject
252 to imprecise stochastic loading. In such case, Directional Importance Sampling can offer a con-
253 venient means for calculating failure probabilities, as shown in [32, 36]. Thus, the rest of this
254 Section is organized as follows. Section 3.2 describes the type of stochastic load considered here,
255 which corresponds to a Gaussian process. Then, Section 3.3 describes the characterization of the
256 dynamic response of a linear system, where the structural matrices (that is, mass, damping and
257 stiffness) may be affected by the input parameter \mathbf{y} . Then, Sections 3.4 and 3.5 address the solu-
258 tion of the augmented reliability problem in order to estimate the augmented failure probability
259 (p_F^A , see eq. (4)) as well as the probability distribution of the vector of distribution parameters
260 conditioned on failure ($f_{\Theta}(\boldsymbol{\theta}|F)$, see eq. (10)), respectively. Section 3.6 formulates the expression
261 for estimating the failure probability as a function of the vector of distribution parameters (that
262 is, $p_F(\boldsymbol{\theta})$) while Section 3.7 provides a summary.

263 3.2. Stochastic loading

264 The uncertain load acting over the structure is denoted as $p(t, \mathbf{z})$, where t stands for time. Its
265 duration is T and is represented at discrete time instants $t_k = (k - 1)\Delta t$, $k = 1, \dots, n_T$, where
266 Δt is the time step and n_T is the total number of time instants considered. The load process
267 has zero mean and it is assumed as a crisp Gaussian process (that is, it does not depend on
268 \mathbf{y}). Such an assumption implies that the probability density $f_{\mathbf{Z}}(\mathbf{z})$ associated with \mathbf{z} is actually
269 Gaussian. The covariance matrix of the discretized process is denoted as $\mathbf{\Gamma}$, whose dimension is
270 $n_T \times n_T$. Following the above assumptions, the stochastic load is represented by means of the
271 Karhunen-Loève expansion (see, e.g. [37]):

$$\mathbf{p}(\mathbf{z}) = \mathbf{\Psi}\mathbf{\Lambda}^{1/2}\mathbf{z}, \quad (11)$$

where $\mathbf{p}(\mathbf{z})$ is a vector containing a realization of the uncertain load of dimension $n_T \times 1$, where its k -th component p_k represents the load at time t_k ; $\mathbf{\Lambda}$ is a matrix whose diagonal contains the largest n_{KL} eigenvalues of the covariance matrix $\mathbf{\Gamma}$; $\mathbf{\Psi}$ is a matrix of dimension $n_T \times n_{KL}$ containing the n_{KL} eigenvectors of the covariance matrix $\mathbf{\Gamma}$ associated with the aforementioned eigenvalues; n_{KL} is the number of terms retained for the Karhunen-Loève expansion ($n_{KL} \leq n_T$, see, e.g. [37]); and \mathbf{z} is a realization of a standard Gaussian random variable vector \mathbf{Z} of dimension $n_{KL} \times 1$. It is easily verified that due to the above formulation, $n_z = n_{KL}$.

3.3. Imprecise Structural Response

This contribution considers systems that are modeled as a linear structures with classical damping subject to dynamic loading. The associated equation of motion is [38]:

$$\mathbf{M}(\mathbf{y})\ddot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) + \mathbf{C}(\mathbf{y})\dot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) + \mathbf{K}(\mathbf{y})\mathbf{x}(t, \mathbf{y}, \mathbf{z}) = \boldsymbol{\rho}(\mathbf{y})p(t, \mathbf{z}), \quad (12)$$

where \mathbf{M} , \mathbf{C} and \mathbf{K} are the mass, damping and stiffness matrices of dimension $n_D \times n_D$ each; n_D is the number of degrees-of-freedom of the structural model; $\boldsymbol{\rho}$ is a vector of dimension $n_D \times 1$ coupling the stochastic load $p(t, \mathbf{z})$ with the degrees-of-freedom of the structure; and where $\ddot{\mathbf{x}}$, $\dot{\mathbf{x}}$ and \mathbf{x} denote the acceleration, velocity and displacement vectors, each of dimension $n_D \times 1$. It is noted that the system's matrices as well as the coupling vector $\boldsymbol{\rho}$ are affected by \mathbf{y} , whose uncertainty is described by a p-box model, while the loading is affected by aleatoric uncertainty. Hence, the uncertainty associated with the response of the structure becomes a p-box as well. It is assumed that there are n_R responses of interest of the structural system that are of interest due to, e.g. practical design reasons. These responses are denoted as $\eta_i(t, \mathbf{y}, \mathbf{z})$, $i = 1, \dots, n_R$. Due to the linearity of the structure, these responses are calculated by means of a convolution integral. In fact, as shown in Appendix A, the i -th response of interest at the k -th time instant is given by the expression:

$$\eta_i(t_k, \mathbf{y}, \mathbf{z}) = \mathbf{a}_{i_k}(\mathbf{y})\mathbf{z}, \quad (13)$$

where $\mathbf{a}_{i_k}(\mathbf{y})$ is a row vector of dimension $1 \times n_{KL}$ which is associated with the discrete time representation of the convolution integral (see Appendix A for details). Furthermore, the i -th

297 response of interest at each of the n_T time instants of analysis is equal to:

$$\boldsymbol{\eta}_i(\mathbf{y}, \mathbf{z}) = \mathbf{A}_i(\mathbf{y})\mathbf{z}, \quad i = 1, \dots, n_R, \quad (14)$$

298 where $\boldsymbol{\eta}_i$ is a vector of dimension $n_T \times 1$ containing the i -th response along the duration T of the
 299 stochastic loading; and \mathbf{A}_i is a matrix of dimension $n_T \times n_{KL}$ whose k -th row contains the vector
 300 $\mathbf{a}_{i_k}(\mathbf{y})$.

301 The structural system undergoes an unacceptable behavior whenever the absolute value of any of
 302 its responses exceeds corresponding prescribed thresholds b_i , $i = 1, \dots, n_R$ at any time within the
 303 duration of the imprecise stochastic load. The occurrence of such unacceptable behavior can be
 304 expressed in mathematical terms with the help of the normalized response function $\xi(\mathbf{y}, \mathbf{z})$ [39],
 305 which is equal to:

$$\xi(\mathbf{y}, \mathbf{z}) = \|\mathbf{A}(\mathbf{y})\mathbf{z}\|_\infty, \quad (15)$$

306 where $\|\cdot\|_\infty$ represents infinity norm; and \mathbf{A} is a matrix of dimension $(n_R n_T) \times n_{KL}$ that allows
 307 calculating all responses of interest throughout the duration T of the load and which is calculated
 308 as shown below, assuming that $b_i > 0$, $i = 1, \dots, n_R$. The matrix \mathbf{A} is specifically given as:

$$\mathbf{A}(\mathbf{y}) = \begin{bmatrix} b_1^{-1} \mathbf{A}_1(\mathbf{y}) \\ \vdots \\ b_{n_R}^{-1} \mathbf{A}_{n_R}(\mathbf{y}) \end{bmatrix}. \quad (16)$$

309 Matrix \mathbf{A} as shown in eq. (16) provides a means for calculating the responses of interest in a
 310 dimensionless, normalized fashion. This is a consequence of multiplying by the inverse of the cor-
 311 responding threshold level. Furthermore, the infinity norm introduced in eq. (15) allows retrieving
 312 the maximum value of these dimensionless responses. Hence, whenever $\xi \geq 1$, one or more of the
 313 n_R responses of interest of the structure exceed their prescribed thresholds at one or more time
 314 instants within the duration of the stochastic load. Thus, the performance function associated
 315 with the failure probability in eq. (1) and its augmented counterpart in eq. (4) is equal to:

$$\begin{aligned} g(\mathbf{y}, \mathbf{z}) &= 1 - \xi(\mathbf{y}, \mathbf{z}), \\ &= 1 - \|\mathbf{A}(\mathbf{y})\mathbf{z}\|_\infty. \end{aligned} \quad (17)$$

3.4. Application of Directional Importance Sampling for Calculating Augmented Failure Probability 316
317

The structure of the performance function in eq. (17) reveals that the limit state function $g(\mathbf{y}, \mathbf{z}) = 0$ is actually a piecewise linear function with respect to \mathbf{z} for a fixed value of \mathbf{y} [9, 40, 41]. In fact, linearity is due to the functional form of the i -th response with respect to \mathbf{z} as shown in eq. (14) while the piecewise characteristic stems out of the infinity norm present in eq. (15). Figure 1 shows an schematic representation of such issue. For simplicity, it has been assumed that $n_y = 1$ and $n_z = n_{KL} = 2$. The abscissa and ordinate contain coordinates z_1 and z_2 while coordinate y is orthogonal to the plane of the Figure. The limit state function is represented schematically for two different realizations y and $y + \Delta y$ with solid and dashed line, respectively. It is noted that both limit state functions are different, as they correspond to two different values of y . However, they are both piecewise linear with respect to \mathbf{z} .

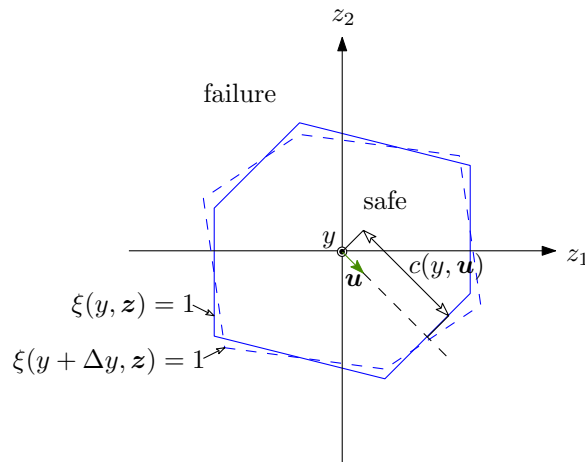


Figure 1: Schematic representation of limit state surface.

The fact that the limit state function is piecewise linear with respect to \mathbf{z} for a fixed value of \mathbf{y} opens the way to solve the augmented reliability problem in eq. (4) by means of Directional Importance Sampling. Such an approach has been shown to be particularly effective for addressing this type of limit state functions [32, 36]. Therefore, the application of this simulation method is investigated in the following for the problem at hand.

As a preliminary step before applying Directional Importance Sampling, it is necessary to express vector \mathbf{z} in polar form, that is $\mathbf{z} = r\mathbf{u}$, where $\mathbf{u} = \mathbf{z}/\|\mathbf{z}\|$ is the unit vector pointing towards \mathbf{z} , $r = \|\mathbf{z}\|$ is the radius and $\|\cdot\|$ denotes Euclidean norm. Thus, the augmented failure probability

336 in eq. (4) is expressed as:

$$p_F^A = \int_{\boldsymbol{\theta} \in \Omega_\theta} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_U} \int_0^\infty I_F(\mathbf{y}, r\mathbf{u}) 2r f_{R^2}(r^2) f_U(\mathbf{u}) f_Y(\mathbf{y}|\boldsymbol{\theta}) f_\Theta(\boldsymbol{\theta}) dr d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}, \quad (18)$$

337 where Ω_U denotes the unit hypersphere in $(n_{KL} - 1)$ dimensions; $f_U(\mathbf{u})$ is the uniform probability
 338 density over the $(n_{KL} - 1)$ -dimensional hypersphere; and $f_{R^2}(r^2)$ is the Chi-square distribution of
 339 n_{KL} degrees-of-freedom. The factor $2r$ in eq. (18) appears due to the rule of change of variables
 340 for probability distributions [42]. In this sense, note that r^2 follows a Chi-square distribution of
 341 n_{KL} degrees-of-freedom as it is the sum of the squares of n_{KL} standard Gaussian variables. Hence,
 342 the probability density function associated with r is equal to $f_R(r) = 2r f_{R^2}(r^2)$.
 343 The integral associated with r in eq. (18) can be solved in closed form taking advantage of the
 344 linearity of the responses of interest with respect to \mathbf{z} . In fact, let c be the scalar that solves the
 345 equation $g(\mathbf{y}, c\mathbf{u}) = 0$ for fixed values of \mathbf{y} and \mathbf{u} . Then, c can be obtained in explicit form out of
 346 eq. (17), as shown below.

$$c(\mathbf{y}, \mathbf{u}) = \min_{i=1, \dots, n_\eta} \left(\min_{k=1, \dots, n_T} (c_{i,k}(\mathbf{y}, \mathbf{u})) \right), \quad (19)$$

347 where $c_{i,k}(\mathbf{y}, \mathbf{u})$ is defined as:

$$c_{i,k}(\mathbf{y}, \mathbf{u}) = \frac{b_i}{|\mathbf{a}_{i,k}(\mathbf{y})\mathbf{u}|}, \quad (20)$$

348 and where $\mathbf{a}_{i,k}(\mathbf{y})$ represents the k -th row of matrix $\mathbf{A}_i(\mathbf{y})$ (see Appendix A) and $|\cdot|$ denotes
 349 absolute value. Note that $c(\mathbf{y}, \mathbf{u})$ in eq. (19) represents the Euclidean distance between the origin
 350 of the standard normal space and the limit state function along direction \mathbf{u} . Figure 1 sketches
 351 distance $c(\mathbf{y}, \mathbf{u})$ for a given unit vector \mathbf{u} .

352 From the above discussion, it is seen that $I_F(\mathbf{y}, r\mathbf{u}) = 1$ for $r \in [c(\mathbf{y}, \mathbf{u}), \infty)$. Thus, the integral
 353 associated with r in eq. (18) can be solved in closed form, leading to the following expression:

$$\begin{aligned} p_F^A &= \int_{\boldsymbol{\theta} \in \Omega_\theta} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_U} \int_0^\infty I_F(\mathbf{y}, r\mathbf{u}) 2r f_{R^2}(r^2) f_U(\mathbf{u}) f_Y(\mathbf{y}|\boldsymbol{\theta}) f_\Theta(\boldsymbol{\theta}) dr d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}, \\ &= \int_{\boldsymbol{\theta} \in \Omega_\theta} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_U} \int_{c(\mathbf{y}, \mathbf{u})}^\infty 2r f_{R^2}(r^2) f_U(\mathbf{u}) f_Y(\mathbf{y}|\boldsymbol{\theta}) f_\Theta(\boldsymbol{\theta}) dr d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}, \\ &= \int_{\boldsymbol{\theta} \in \Omega_\theta} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_U} (1 - F_{R^2}(c(\mathbf{y}, \mathbf{u})^2)) f_U(\mathbf{u}) f_Y(\mathbf{y}|\boldsymbol{\theta}) f_\Theta(\boldsymbol{\theta}) d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}, \end{aligned} \quad (21)$$

where $F_{R^2}(r^2)$ denotes the Chi-square cumulative density function of n_{KL} degrees-of-freedom. 354
 Eq. (21) provides a mean for estimating the augmented failure probability in a directional fashion 355
 with respect to \mathbf{u} by means of simulation [43, 44]. However, the associated estimator may possess 356
 a large variability, which is undesirable from a numerical viewpoint. This issue can be addressed 357
 by introducing an Importance Sampling density (ISD) function associated with \mathbf{u} that takes 358
 advantage of the geometry of the limit state function, that is, piecewise linear for a fixed value 359
 \mathbf{y} , as shown in Figure 1. For this purpose, consider the so-called elementary failure event $F_{i,k}$ 360
 [9], where the i -th response of interest exceeds its prescribed threshold level b_i at time t_k for a 361
 given value \mathbf{y} . Recalling the linearity of the response with respect to \mathbf{z} (see eq. (13)), the polar 362
 representation $\mathbf{z} = r\mathbf{u}$ and the definition of quantity $c_{i,k}(\mathbf{y}, \mathbf{u})$ in eq. (20), the elementary failure 363
 event is defined in mathematical terms as: 364

$$\begin{aligned}
 F_{i,k}(\mathbf{y}) &= \{\mathbf{z} \in \Omega_z : |\mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{z}| \geq b_i\}, \\
 &= \{\mathbf{u} \in \Omega_u \wedge r \in [0, \infty) : |\mathbf{a}_{i,k}(\mathbf{y})^T (r\mathbf{u})| \geq b_i\}, \\
 &= \{\mathbf{u} \in \Omega_u \wedge r \in [0, \infty) : r \geq c_{i,k}(\mathbf{y}, \mathbf{u})\}.
 \end{aligned} \tag{22}$$

According to Bayes' theorem, the probability density associated with \mathbf{u} conditioned on the occur- 365
 rence of the elementary failure event $F_{i,k}(\mathbf{y})$ is the following [9, 32]. 366

$$f_{\mathcal{U}}(\mathbf{u}|F_{i,k}(\mathbf{y})) = \frac{f_{\mathcal{U}}(\mathbf{u})P(F_{i,k}(\mathbf{y})|\mathbf{u})}{P(F_{i,k}(\mathbf{y}))}. \tag{23}$$

From the above equation, it is noted that $f_{\mathcal{U}}(\mathbf{u})$ is available in closed form (see, e.g. [45]). The 367
 term $P(F_{i,k}(\mathbf{y})|\mathbf{u})$ measures the probability of occurrence of the elementary failure event for a 368
 fixed value of \mathbf{u} . Given the definition of the elementary failure event in eq. (22) and recalling that 369
 r^2 follows a Chi-square distribution of n_{KL} -degrees-of-freedom, this probability can be determined 370
 in closed form, that is: 371

$$P(F_{i,k}(\mathbf{y})|\mathbf{u}) = P(r \geq c_{i,k}(\mathbf{y}, \mathbf{u})) = 1 - F_{R^2}(c_{i,k}(\mathbf{y}, \mathbf{u})^2). \tag{24}$$

Moreover, the denominator in eq. (23) can also be determined in closed form recalling that 372
 $\mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{z}$ follows a Gaussian distribution with zero mean and variance $\mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{a}_{i,k}(\mathbf{y})$ for a fixed 373

374 value \mathbf{y} . Thus, it is noted that:

$$P(F_{i,k}(\mathbf{y})) = P(|\mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{z}| \geq b_i) = 2F_Z(-\beta_{i,k}(\mathbf{y})), \quad (25)$$

375 where $F_Z(\cdot)$ is the standard Gaussian cumulative distribution and $\beta_{i,k}(\mathbf{y})$ is the so-called reliability index, which is equal to $\beta_{i,k}(\mathbf{y}) = b_i / \sqrt{\mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{a}_{i,k}(\mathbf{y})}$ [9, 40]. As the terms $P(F_{i,k}(\mathbf{y})|\mathbf{u})$ and $P(F_{i,k}(\mathbf{y}))$ can be calculated in closed form as shown in eqs. (24) and (25), the conditional probability density function $f_{\mathcal{U}}(\mathbf{u}|F_{i,k}(\mathbf{y}))$ in eq. (23) is available as well. A salient characteristic of this density function is that it assigns no weight to direction vectors \mathbf{u} which do not lead to the occurrence of the elementary failure event. Thus, the ISD function is chosen as a weighted summation of the density functions $f_{\mathcal{U}}(\mathbf{u}|F_{i,k}(\mathbf{y}))$ associated with each elementary failure event (that is, $F_{i,k}$, $i = 1, \dots, n_\eta$, $k = 1, \dots, n_T$) [9, 36, 32]. The weight associated with each density function is denoted as $w_{i,k}$, $i = 1, \dots, n_\eta$, $j = 1, \dots, n_T$, and is selected proportional to $P(F_{i,k}(\mathbf{y}))$, as done customarily in Importance Sampling [9, 46]. Hence:

$$w_{i,k}(\mathbf{y}) = \frac{2F_Z(-\beta_{i,k}(\mathbf{y}))}{\hat{p}_F(\mathbf{y})}, \quad (26)$$

385 where $\hat{p}_F(\mathbf{y}) = \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} 2F_Z(-\beta_{i,k}(\mathbf{y}))$. Inserting eqs. (24) and (25) into eq. (23) and taking the summation of all conditional density functions weighted by the expression in eq. (26), the ISD function becomes the following [32, 36].

$$\begin{aligned} f_{\mathcal{U}}^{\text{IS}}(\mathbf{u}|\mathbf{y}) &= \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} w_{i,k}(\mathbf{y}) f_{\mathcal{U}}(\mathbf{u}|F_{i,k}(\mathbf{y})), \\ &= \frac{f_{\mathcal{U}}(\mathbf{u})}{\hat{p}_F(\mathbf{y})} \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}, \mathbf{u})^2)). \end{aligned} \quad (27)$$

388 It is emphasized that the ISD function as shown in the above equation is valid for a given value of \mathbf{y} . Now, this ISD function can be considered for estimating the augmented failure probability

defined in eq. (21), leading to:

390

$$\begin{aligned}
p_F^A &= \int_{\boldsymbol{\theta} \in \Omega_{\boldsymbol{\theta}}} \int_{\mathbf{y} \in \Omega_{\mathbf{y}}} \int_{\mathbf{u} \in \Omega_U} (1 - F_{R^2}(c(\mathbf{y}, \mathbf{u})^2)) \frac{f_U(\mathbf{u})}{f_U^{\text{IS}}(\mathbf{u}|\mathbf{y})} f_U^{\text{IS}}(\mathbf{u}|\mathbf{y}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}) d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta} \\
&= \int_{\boldsymbol{\theta} \in \Omega_{\boldsymbol{\theta}}} \int_{\mathbf{y} \in \Omega_{\mathbf{y}}} \int_{\mathbf{u} \in \Omega_U} \frac{\hat{p}_F(\mathbf{y}) (1 - F_{R^2}(c(\mathbf{y}, \mathbf{u})^2))}{\sum_{i=1}^{n_{\eta}} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}, \mathbf{u})^2))} f_U^{\text{IS}}(\mathbf{u}|\mathbf{y}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}) d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}, \\
&\approx \tilde{p}_F^A = \frac{1}{N} \sum_{j=1}^N \frac{\hat{p}_F(\mathbf{y}^{(j)}) (1 - F_{R^2}(c(\mathbf{y}^{(j)}, \mathbf{u}^{(j)})^2))}{\sum_{i=1}^{n_{\eta}} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}^{(j)}, \mathbf{u}^{(j)})^2))}, \\
&\quad \boldsymbol{\theta}^{(j)} \sim f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}), \mathbf{y}^{(j)} \sim f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}^{(j)}), \mathbf{u}^{(j)} \sim f_U^{\text{IS}}(\mathbf{u}|\mathbf{y}^{(j)}), j = 1, \dots, N,
\end{aligned} \tag{28}$$

where $\tilde{(\cdot)}$ denotes an estimator of a quantity. From the above equation, it is important to note 391
that samples of $\boldsymbol{\theta}$ are generated such that they follow the instrumental uniform probability density 392
 $f_{\boldsymbol{\Theta}}(\boldsymbol{\theta})$; then, samples of \mathbf{y} are generated conditional on those samples of $\boldsymbol{\theta}$; and finally, samples of 393
 \mathbf{u} are generated conditional on the samples of \mathbf{y} . For details on how samples of \mathbf{u} are generated, 394
it is referred to Appendix B. 395

3.5. Application of Directional Importance Sampling for Calculating Conditional Probability Den- 396 sity Function $f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F)$ 397

The next challenge for applying the formulation presented in Section 2 is the calculation of 398
the conditional probability density function $f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F)$ in eq. (10). The objective is estimating 399
this density function by means of simulation with samples of \mathbf{y} and \mathbf{z} distributed according to 400
 $f_{\mathbf{Y},\mathbf{Z}}(\mathbf{y}, \mathbf{z}|F)$. In this sense, the samples used for estimating the augmented failure probability in 401
eq. (28) could be used for that purpose. Nonetheless, these samples cannot be used directly, as 402
they are distributed according to the ISD function, which is not necessarily equal to $f_{\mathbf{Y},\mathbf{Z}}(\mathbf{y}, \mathbf{z}|F)$. 403
Therefore, special care must be taken for using those samples, as described in the following. 404
According to the definition of a marginal distribution and Bayes' theorem, $f_{\mathbf{Y},\mathbf{Z}}(\mathbf{y}, \mathbf{z}|F)$ is equal 405
to: 406

$$\begin{aligned}
f_{\mathbf{Y},\mathbf{Z}}(\mathbf{y}, \mathbf{z}|F) &= \int_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} f_{\boldsymbol{\Theta},\mathbf{Y},\mathbf{Z}}(\boldsymbol{\theta}, \mathbf{y}, \mathbf{z}|F) d\boldsymbol{\theta} \\
&= \frac{1}{p_F^A} \int_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} I_F(\mathbf{y}, \mathbf{z}) f_{\mathbf{Z}}(\mathbf{z}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}) d\boldsymbol{\theta}.
\end{aligned} \tag{29}$$

407 Recalling the definition of polar coordinates $\mathbf{z} = r\mathbf{u}$, the latter expression is further simplified to:

$$f_{\mathbf{Y},\mathbf{Z}}(\mathbf{y}, \mathbf{z}|F) = f_{\mathbf{Y},R,U}(\mathbf{y}, r, \mathbf{u}|F) = \frac{1}{p_F^A} \int_{\boldsymbol{\theta} \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} I_F(\mathbf{y}, r\mathbf{u}) 2r f_{R^2}(r^2) f_U(\mathbf{u}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}) f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}) d\boldsymbol{\theta}. \quad (30)$$

408 Note that eq. (30) provides an expression for the conditional probability distribution of $f_{\mathbf{Y},\mathbf{Z}}(\mathbf{y}, \mathbf{z}|F)$
 409 as a function of the polar coordinates, which has been considered for the implementation of Direc-
 410 tional Importance Sampling in Section 3.4. Now, this expression is inserted into eq. (10) for the
 411 calculation of $f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F)$. However, special consideration should be paid to one detail: in eq. (10), $\boldsymbol{\theta}$
 412 refers to an independent variable, that is, $\boldsymbol{\theta}$ may assume any value within $[\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]$. On the contrary,
 413 in eq. (30), $\boldsymbol{\theta}$ corresponds to an integration variable over the domain $[\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]$. Hence, and in order
 414 to avoid confusion, the variable $\boldsymbol{\theta}$ appearing in eq. (30) is written in the following as $\boldsymbol{\theta}'$. Having
 415 taken this special consideration, the expression for $f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F)$ becomes:

$$f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F) = \frac{1}{p_F^A} \int_{\boldsymbol{\theta}' \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_u} \int_0^\infty \frac{f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\Delta(\mathbf{y})} I_F(\mathbf{y}, r\mathbf{u}) 2r f_{R^2}(r^2) f_U(\mathbf{u}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}') f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}') d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}'. \quad (31)$$

416 This last equation can be simplified by recalling that $I_F(\mathbf{y}, r\mathbf{u}) = 1$ for $r \in [c(\mathbf{y}, \mathbf{u}), \infty)$ and
 417 that for its evaluation, it is possible to introduce the ISD function defined in eq. (27). Thus, the
 418 expression for $f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F)$ is as follows.

$$f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}|F) = \frac{1}{p_F^A} \int_{\boldsymbol{\theta}' \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_u} \frac{f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\Delta(\mathbf{y})} \frac{\hat{p}_F(\mathbf{y}) (1 - F_{R^2}(c(\mathbf{y}, \mathbf{u})^2))}{\sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}, \mathbf{u})^2))} f_U^{\text{IS}}(\mathbf{u}|\mathbf{y}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}') f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}') d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}'. \quad (32)$$

419 3.6. Explicit Approximation of the Failure Probability as a Function of the Distribution Parameter 420 Vector

421 Considering all previous deductions, it is possible to determine the following expression for the
 422 failure probability as a function of the distribution parameter vector by inserting eq. (32) into
 423 eq. (5).

$$p_F(\boldsymbol{\theta}) = \frac{1}{f_{\boldsymbol{\Theta}}(\boldsymbol{\theta})} \int_{\boldsymbol{\theta}' \in [\underline{\boldsymbol{\theta}}, \bar{\boldsymbol{\theta}}]} \int_{\mathbf{y} \in \Omega_y} \int_{\mathbf{u} \in \Omega_u} \frac{f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta})}{\Delta(\mathbf{y})} \frac{\hat{p}_F(\mathbf{y}) (1 - F_{R^2}(c(\mathbf{y}, \mathbf{u})^2))}{\sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}, \mathbf{u})^2))} f_U^{\text{IS}}(\mathbf{u}|\mathbf{y}) f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}') f_{\boldsymbol{\Theta}}(\boldsymbol{\theta}') d\mathbf{u} d\mathbf{y} d\boldsymbol{\theta}'. \quad (33)$$

It is noted that the above expression can be estimated by using samples of \mathbf{u} , \mathbf{y} and $\boldsymbol{\theta}'$ generated when conducting augmented reliability analysis with Directional Importance Sampling, as performed in Section 3.4. This estimator is equal to:

$$p_F(\boldsymbol{\theta}) \approx \tilde{p}_F(\boldsymbol{\theta}) = \frac{1}{N} \sum_{j=1}^N \frac{f_{\mathbf{Y}}(\mathbf{y}^{(j)}|\boldsymbol{\theta})}{\Delta(\mathbf{y}^{(j)})} \frac{\hat{p}_F(\mathbf{y}^{(j)}) (1 - F_{R^2}(c(\mathbf{y}^{(j)}, \mathbf{u}^{(j)})^2))}{\sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}^{(j)}, \mathbf{u}^{(j)})^2))},$$

$$\boldsymbol{\theta}'^{(j)} \sim f_{\boldsymbol{\theta}'}(\boldsymbol{\theta}'), \mathbf{y}^{(j)} \sim f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}'^{(j)}), \mathbf{u}^{(j)} \sim f_U^{\text{IS}}(\mathbf{u}|\mathbf{y}^{(j)}), j = 1, \dots, N. \quad (34)$$

As $p_F(\boldsymbol{\theta})$ is estimated with independent, identically distributed samples, it is straightforward to estimate the standard deviation of this estimator, which is equal to:

$$\sigma_{\tilde{p}_F(\boldsymbol{\theta})} = \sqrt{\frac{1}{N(N-1)} \sum_{j=1}^N \left(\frac{f_{\mathbf{Y}}(\mathbf{y}^{(j)}|\boldsymbol{\theta})}{\Delta(\mathbf{y}^{(j)})} \frac{\hat{p}_F(\mathbf{y}^{(j)}) (1 - F_{R^2}(c(\mathbf{y}^{(j)}, \mathbf{u}^{(j)})^2))}{\sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} (1 - F_{R^2}(c_{i,k}(\mathbf{y}^{(j)}, \mathbf{u}^{(j)})^2))} - \tilde{p}_F(\boldsymbol{\theta}) \right)^2}. \quad (35)$$

3.7. Summary

As a summary of the material presented in this Section, it is seen that it is possible to formulate an explicit approximation of the failure probability as a function of the vector of distribution parameters $\boldsymbol{\theta}$, as shown in eq. (34). This explicit approximation is constructed by carrying out a single run of Directional Importance Sampling for solving the associated augmented reliability problem. Then, this explicit approximation can be coupled with any appropriate optimization algorithm in order to estimate the bounds of the imprecise probability. In addition, it is possible to estimate the standard deviation of the probability estimator by means of eq. (35). This is most useful from a practical viewpoint, as it is possible to assert whether or not the estimator of the failure probability $p_F(\boldsymbol{\theta})$ is sufficiently accurate and take proper action if necessary: for example, increasing the number of samples in order to improve the accuracy.

As a side remark, it should be noted that the formulation of Directional Importance Sampling as presented here considers an ISD function related with the direction vector \mathbf{u} only. Nonetheless, references in the literature suggest that for the class of problems considered here, that is, first excursion probability of uncertain linear systems subject to imprecise dynamic load, it is also feasible to establish an ISD function associated with the uncertain input parameters \mathbf{y} , as discussed in [27, 47, 48, 41]. Such an approach has the potential to improve even further accuracy when

446 solving the augmented reliability problem. In spite of this issue, the possibility of introducing this
 447 additional ISD function is not further discussed here, in order to focus more on the application
 448 of the augmented reliability problem and its connection with the estimation of imprecise first
 449 excursion probabilities.

450 4. Examples

451 4.1. General Remarks

452 This Section illustrates the application of the augmented reliability approach for estimating
 453 the bounds of first excursion probabilities. A test example involving a single-degree-of-freedom
 454 oscillator and an application example comprising the finite element model of a composite wing are
 455 considered. For both cases, aleatoric and epistemic uncertainties affect the structural parameters
 456 while the loading is modeled as purely aleatoric. In this sense, it should be noted that the scheme
 457 proposed in Section 3 can consider epistemic uncertainty on the loading as well. Studies on
 458 the effect of imprecise stochastic loads on the bounds on the first excursion probability of linear
 459 structures can be found in previous work of the authors [21, 22, 23]. These studies acknowledge
 460 the importance of including imprecision in the definition of stochastic loads modelled by advanced
 461 auto-correlation methods. However, this possibility is not explored further in order to simplify
 462 the presentation of the performance of the method and focus on the application of the augmented
 463 reliability framework for calculating imprecise probabilities.

464 4.2. Test Example 1: Single-degree-of-freedom shear beam model subject to stochastic ground ac- 465 celeration

466 The first example involves the single-story shear beam model depicted in Figure 2. This model
 467 is subjected to a stochastic ground acceleration and its stiffness is characterized as a p-box. The
 468 objective is estimating the bounds for the first excursion probability.

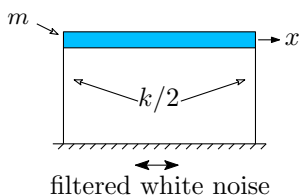


Figure 2: Example 1 – Single story shear beam model subject to filtered white noise excitation

469 The mass of the shear beam model is equal to $m = 10^4$ [kg] while its (classical) damping ratio
 470 is $d = 5\%$. The lateral stiffness k of the model is characterized by means of a lognormal p-box,

such that the expected value is represented by the interval $\mu_k \in [0.7 \times 10^6, 1.3 \times 10^6]$ [MN/m] 471
and the standard deviation is $\sigma_k = 10^5$ [MN/m]. The stochastic ground acceleration is modeled 472
as a white noise process of duration $T = 20$ [s] and spectral density $S = 5 \times 10^{-3}$ [m²/s³], 473
which is represented at discrete time steps of $\Delta t = 0.01$ [s]. This discrete white noise is modulated 474
considering the Shinozuka-Sato envelope, with parameters $c_1 = 0.14$ and $c_2 = 0.16$ [7]. In addition, 475
the modulated white noise is filtered considering the Clough-Penzien model (see, e.g. [49]), with 476
circular frequencies $\omega_{g,1} = 6\pi$ [rad/s] and $\omega_{g,2} = 0.6\pi$ [rad/s] and damping ratios $d_{g,1} = d_{g,2} =$ 477
60% for the primary and secondary filters, respectively. This stochastic ground acceleration is 478
represented by means of the Karhunen-Loève expansion, considering all $n_{KL} = n_z = 2001$ terms. 479
The responses of interest of the shear beam model are its relative displacement with respect to 480
the ground as well as its absolute acceleration. Failure occurs whenever any of these responses 481
exceeds the prescribed thresholds $b_1 = 0.07$ [m] and $b_2 = 7.5$ [m/s²] within the duration of the 482
acceleration. 483

In order to estimate the bounds of the first excursion probability, the augmented reliability problem 484
is solved first with Directional Importance Sampling. For this purpose, a total of $N = 2000$ samples 485
are considered. The results obtained for the estimate \tilde{p}_F^A as well as its coefficient of variation $\delta_{\tilde{p}_F^A}$ 486
are shown in Figure 3. It is noted from this figure that the estimate of the augmented failure 487
probability stabilizes after about 500 samples and that a coefficient of variation of less than 10% 488
is attained at about 2000 samples. These results suggest that the level of precision of the estimator 489
is adequate. 490

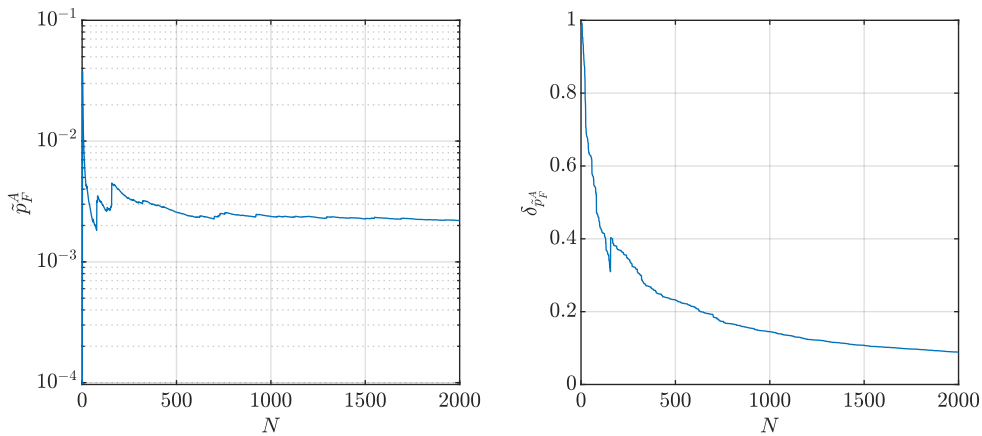


Figure 3: Example 1 – Evolution of the estimator of the augmented failure probability \tilde{p}_F^A with respect to the number of samples.

In a next step, the first excursion probability is estimated as a function of the interval variable 491

492 μ_k applying eq. (34), using the samples generated when estimating the augmented failure proba-
 493 bility. This probability $\tilde{p}_F(\mu_k)$ is plotted in Figure 4 with blue, solid line. In addition, the blue
 494 shaded area in that figure shows the standard deviation associated with the estimate of the failure
 495 probability, that is $\tilde{p}_F(\mu_k) \pm \sigma_{\tilde{p}_F(\mu_k)}$. For comparison and validation purposes, the first excursion
 496 probability is also calculated for ten different crisp values of μ_k using Directional Importance Sam-
 497 pling. In other words, ten different simulations of Directional Importance Sampling are carried
 498 out, each of them considering $N = 1000$ samples. Hence, the total number of samples involved in
 499 such an approach is equal to $N_T = 10N = 10^4$. The probability estimates along the grid of crisp
 500 values of μ_k are shown with red x marks; in addition, the standard deviation of these probability
 501 estimates is marked with a bar. It is seen that there is an excellent agreement between the two
 502 approaches, suggesting that the framework provided by augmented reliability is appropriate for
 503 estimating the first excursion probability as a function of the imprecise distribution parameter.
 504 Furthermore, it should be noted that the relation between the failure probability and the interval-
 505 valued μ_k is non-monotonic. In this sense, it is quite remarkable that the proposed approach can
 506 produce this type of approximations, as no particular assumptions have to be introduced to arrive
 507 to such result. The non-monotonic behavior can be explained as follows. For small crisp values
 508 of μ_k , the shear beam model is quite flexible and failure due to displacements is likely. As the
 509 crisp value of μ_k increases, the shear beam possesses more stiffness and it is capable of controlling
 510 displacements better, which decreases the failure probability. However, as the crisp value of μ_k
 511 continues increasing, the shear beam becomes too stiff, thus increasing the absolute acceleration
 512 and in turn, the failure probability.

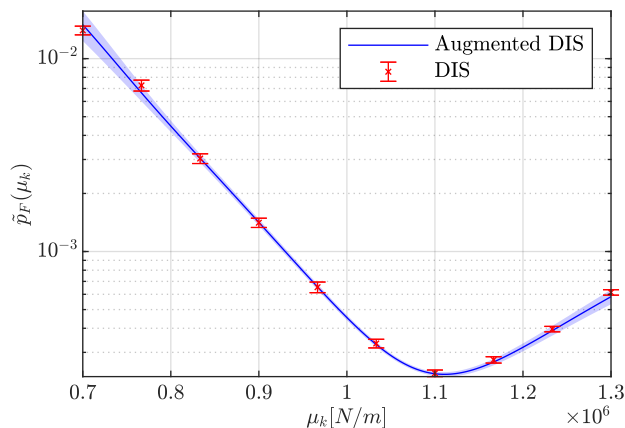


Figure 4: Example 1 – Failure probability as a function of the interval parameter μ_k . DIS: Directional Importance Sampling

Finally, the bounds on the imprecise failure probability are determined by minimizing/maximizing $\tilde{p}_F(\mu_k)$. The numerical cost associated with this step is negligible, as the failure probability as a function of μ_k is available in closed form. It is found that $p_F(\mu_k) \in [2.3 \times 10^{-4}, 1.5 \times 10^{-2}]$, revealing that the uncertainty on μ_k has a major impact in the first excursion probability.

4.3. Example 2: Composite wing

4.3.1. Case introduction and physical description

The second example comprises of a Finite Element of a fictitious aircraft wing that is produced in a laminated composite. The structure has a total length of 30 [m]. This wing is produced using a multi-layer laminar composite material, with deterministic orthotropic ply material properties $E_1 = 231$ [GPa], $E_2 = 77$ [GPa], $\nu_{12} = 0.31$ and $G_{12} = G_{23} = G_{13} = 42.7$ [GPa]. The wing consists of a composite outer shell (top, leading edge, bottom, trailing edge), as well as two vertical stiffening ribs in the centre for structural stiffness. A total of 15 different composite lay-ups are present throughout the structure, which are summarized in Table 1. The dynamic behavior of the structure is modeled using a Finite Element model containing 621 nodes, 606 bi-linear shell elements, 573 rigid connections, 10 concentrated masses and 132 rod elements. The finite element model of this structure is shown in Figure 5. A single evaluation of this model takes approx. 30 [s] on a server equipped with two AMD EPYC 7601 CPUs running at 2.65 [GHz] and 512 [GB] of RAM.

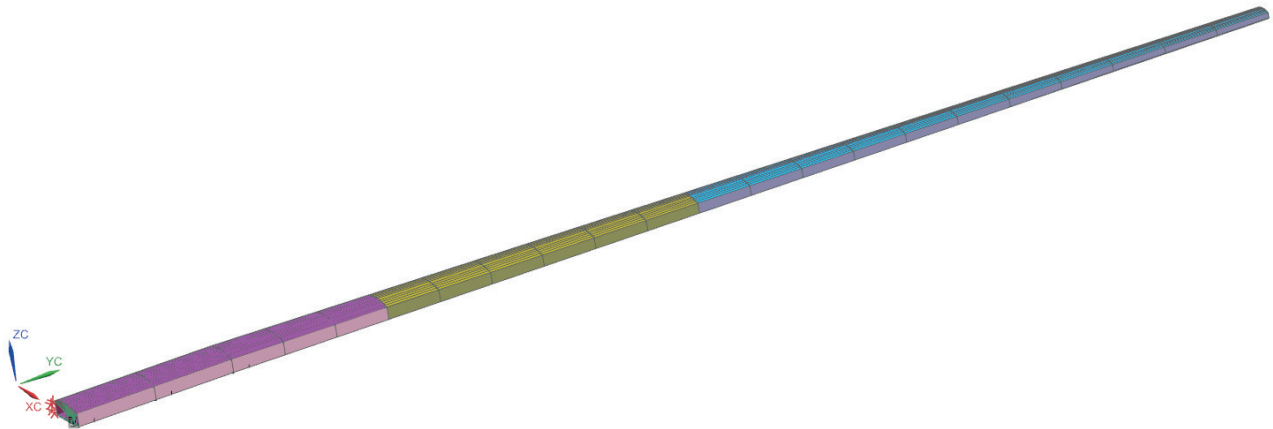


Figure 5: Finite element model of the composite blade (taken from [50])

In this case, the equation of motion of the wing can be represented as in Eq. (12), where n_D of the unconstrained model is $n_D = 3726$.

Table 1: Composite lay-up structure of the wing. Left means at $y = 0$ in Figure 5 and the leading edge is depicted at the back side of figure 5. The + and - signs in front of 45° denote an alternating layering sequence of laminae oriented at $+45^\circ$ and -45° .

Nr	Location	Lay-up (symmetrical)	thickness per layer [mm]
1	top and bottom left	+ - 45°	7.5 [mm]
2	leading edge left	+ - 45°	7.5 [mm]
3	front-middle edge left	+ - + - + - $+45^\circ$	7.5 [mm]
4	back-middle vertical left	+ - + - + - $+45^\circ$	7.5 [mm]
5	trailing edge left	+ - + - + - $+45^\circ$	7.5 [mm]
6	top and bottom middle	+ - 45°	7.5 [mm]
7	leading edge middle	+ - 45°	7.5 [mm]
8	front-middle edge middle	+ - + - + - $+45^\circ$	7.5 [mm]
9	back-middle vertical middle	+ - + - + - $+45^\circ$	7.5 [mm]
10	trailing edge middle	+ - + - + - $+45^\circ$	7.5 [mm]
11	top and bottom right	+ - 45°	7.5 [mm]
12	leading edge right	+ - 45°	7.5 [mm]
13	front-middle edge right	+ - + - + - $+45^\circ$	7.5 [mm]
14	back-middle vertical right	+ - + - + - $+45^\circ$	7.5 [mm]
15	trailing edge right	+ - + - + - $+45^\circ$	7.5 [mm]

533 In this case study, we consider the case where the wing is subjected to a turbulence wind load,
534 which is described as:

$$p(t, \boldsymbol{\xi}) = p_0(t) + p'(t, \boldsymbol{\xi}), \quad (36)$$

535 where $p_0(t)$ is the mean load of the turbulence load and $p'(t, \boldsymbol{\xi})$ is a zero-mean Gaussian stochastic
536 process governed by the transverse Dryden spectrum [51]:

$$S_{p'p'}(\omega) = \frac{c\sigma_{p'}^2}{2\pi} \frac{1 + 3c^2\omega^2}{(1 + c^2\omega^2)^2}, \quad (37)$$

537 where $c = L/v$ is a scale for the turbulence and σ^2 is the variance, with L the turbulence length
538 scale and v the true air speed. Applying the well-known Wiener-Kinchin transformation, following
539 auto-correlation model is obtained:

$$\Gamma_{p,p'}(\tau) = (1 - 0.5\frac{\tau}{L}) \exp(-\tau/L), \quad (38)$$

540 with τ a positive time-lag factor, which can be used directly in combination with the Karhunen-
541 Loève expansion, as explained in Section 3.2. In this case study, failure is conceived as the first
542 passage of the wing displacement over a threshold value of $b = 7$ [mm] under a unit variance

load (i.e., $\sigma_p^2 = 1$). It is important to note that in this case study, $p_0(t)$ is considered to be zero, which is permissible due to linearity in the FE model. In all proceeding computations, the effect of mode-crossover and -veering, which may occur during the calculation of the impulse response functions, is accounted for by tracking the numerically computed mode shapes via the modal assurance criterion.

4.4. Uncertainty propagation

The presented approach is applied to this model considering the thickness t of the 1st, 5th and 6th laminae as being modelled by a p-box, whereas the other 12 thickness values are considered to have a thickness of 7.5 [mm]. This selection is made based on the observation that the thickness of these three laminae has the largest influence on the wing tip displacement (see also [22]). Specifically, these thickness values are modelled as a lognormal distribution with interval-valued mean $\mu_{t1} = \mu_{t5} = \mu_{t6} = [0.006, 0.009]$ [m] and standard deviation $\sigma_t = 0.00075$ [m]. This imprecision in the mean of the distributions can come for instance from limited experimental data set sizes, or a lack of precision in the employed measurement equipment such that only the bounds of one deterministic measurement are known [34]. The aleatory (random) part of the uncertainty is assumed to come from variations during the manufacturing of the laminae. The bounds on the probability of failure are estimated by solving the augmented reliability problem using Directional Importance Sampling with a sample size of $N = 2000$ samples.

Then, the first excursion probability is estimated as a function of the interval variables $\boldsymbol{\mu}_t = [\mu_{t1}, \mu_{t5}, \mu_{t6}]$ applying eq. (34), using the samples generated when estimating the failure probability. This probability $\tilde{p}_F(\boldsymbol{\mu}_t)$ is plotted in Figure 6 with blue, solid line, assuming that $\boldsymbol{\mu}_t = [\mu_{t1}, \mu_{t5}, \mu_{t6}]$. Please note that during the analysis to determine the bounds on P_f , the intervals are considered to be fully independent, and all samples in the augmented space are drawn from the corresponding uniform distributions independently. It is only for the sake of visualisation in Figure 6 that fully dependent samples are generated within the interval bounds. In addition, the blue shaded area in that figure shows the standard deviation associated with the estimate of the failure probability, that is $\tilde{p}_F(\boldsymbol{\mu}_t) \pm \sigma_{\tilde{p}_F(\boldsymbol{\mu}_t)}$. It is important to notice that this blue line is obtained in a single run of Directional Importance Sampling. In addition, for the purpose of validation, the first excursion probability is also determined using 10 runs of Directional Importance Sampling for several crisp values of $\mu_{t1} = \mu_{t5} = \mu_{t6}$, evenly distributed throughout $\boldsymbol{\mu}_t^I$. Each of these 10 runs requires a full Directional Importance Sampling propagation involving

574 1000 deterministic model analyses. In Figure 6, the corresponding first excursion probabilities are
 575 indicated as red crosses, and their standard deviation is indicated using error bars.

576 As can be noted, also in the case of a realistic Finite Element model, the method gives an
 577 excellent agreement with a more conventional approach, with a reduced cost of a factor 5. Finally
 578 the bounds on the probability of failure can be determined by minimizing/maximizing over $\tilde{p}_F(\boldsymbol{\mu}_t)$,
 579 which are found to be $\tilde{p}_F(\boldsymbol{\mu}_t) \in [5.62 \times 10^{-4}, 7.67 \times 10^{-2}]$. From Figure 6, it can be noted that
 580 concerning the upper bound of $\tilde{p}_F(\boldsymbol{\mu}_t)$, a slight deviation exists with respect to the result obtained
 581 for that particular value of $\boldsymbol{\mu}_t$ using conventional DIS, which is caused by the relatively big size
 582 of the interval in combination with the more complicated dynamical behavior of the wing.

583 Finally, as a word of caution, numerical experience indicates that the performance of the
 584 method degrades rapidly with the number of uncertain epistemic parameters. For instance, if
 585 one considers all 15 thickness values of this model to be uncertain, with the same uncertainty
 586 model as prescribed above, and using the same augmented DIS estimator, the bounds on the
 587 probability of failure are found to be $[1.04 \times 10^{-4}, 6.00 \times 10^{-3}]$, as compared to the correct interval
 588 $[2.3 \times 10^{-3}, 1.43 \times 10^{-2}]$ which is obtained via a double-loop implementation. This performance
 589 drop is explained by the comparatively high number of samples that is required to fully explore the
 590 functional behaviour of a response with respect to a high-dimensional input space, and hence, to
 591 correctly establish the functional relationship between $\boldsymbol{\mu}_t$ and $\tilde{p}_F(\boldsymbol{\mu}_t)$. This is a well-documented
 592 shortcoming of this class of methods (see e.g., [29]), and remains an open research question.

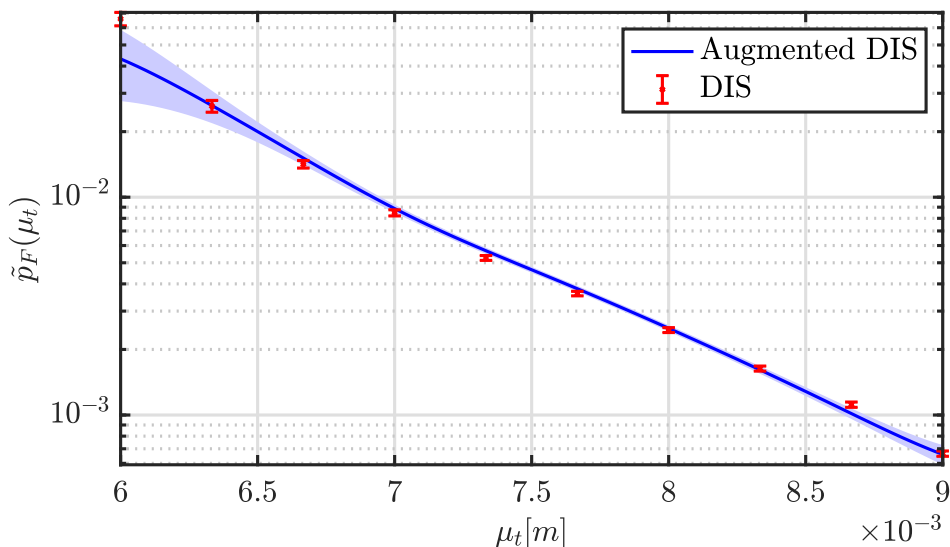


Figure 6: Estimated $\tilde{p}_F(\boldsymbol{\mu}_t)$ as a function of the mean value $\boldsymbol{\mu}_t$ of the thickness values t , as computed by Augmented DIS (blue) and individual runs of DIS for selected values of $\boldsymbol{\mu}_t$ (red).

5. Conclusions and Outlook

593

This paper presents an efficient augmented space method to propagate parametric p-boxes towards the bounds on the first excursion probability of a structure subjected to Gaussian excitation. The method establishes a functional relationship between the interval-valued hyper-parameters of the p-box valued uncertain input quantities and the first excursion probability by representing the problem in an augmented space. Then, by virtue of Bayes' theorem, an expression for the failure probability as an explicit expression of the imprecise parameters from the augmented reliability problem can be recovered, which ultimately allows calculating the imprecise probability by means of Directional Importance Sampling. Following conclusions can be made concerning the proposed approach:

594

595

596

597

598

599

600

601

602

- The method allows for a highly efficient and accurate calculation of the bounds on the probability of failure for both a small-scale academic case, as for a realistic finite element model
- In case the dimension of the input epistemic space is small, the approach is highly accurate, however, numerical experience suggests that the accuracy of the calculated bounds degrades quickly when the dimension of the input epistemic space increases. This is caused by the comparatively high number of samples that is required to fully establish the required functional relationship between the probability of failure and the interval-valued hyper-parameters of the uncertain input quantities. This is consistent with conclusions drawn by other researchers in the area of, e.g. optimal design under uncertain conditions (see, e.g. [29, 27]). Further research however has to be performed in order to define a precise (preferably a priori) criterion in which the augmented reliability problem does not bring substantial advantages compared to double loop approaches.

603

604

605

606

607

608

609

610

611

612

613

614

615

As a last, more practical, remark, it can be noted that in both case studies the bounds of the probability of failure span several orders of magnitude. As an analyst, such observation shows that additional data should be gathered to reduce the epistemic uncertainty on the definition of the p-boxes that represent the uncertainty on the input parameters to shrink the bounds on the estimation of the probability of failure. Alternatively, in case no further data can be gathered, one should consider the upper bound of the analysis and use this to base decisions on, as this is the most conservative estimate given the epistemic uncertainty.

616

617

618

619

620

621

622

623 A future challenge concerning the application of the framework proposed here involves includ-
624 ing epistemic uncertainty in the characterization of stochastic load. Such epistemic uncertainty
625 may have a large impact on the failure probability, as discussed in [21]. In addition, the applica-
626 tion of the framework to problems involving structural non linearities should be explored as well.
627 For that purpose, it would be necessary to consider simulation techniques which are more general
628 than Directional Importance Sampling, such as Subset Simulation. Other path for future develop-
629 ment consists of considering instrumental probability distributions (associated with the epistemic
630 parameters) different from uniform distributions, which may eventually bring advantages for de-
631 ducing the functional relationship between epistemic parameters and the failure probability. As
632 a final comment, it can be noted that the principles laid out in this paper also can be applied to
633 combinations of parametric p-boxes and ‘regular’ random variables. Such extension will also be
634 pursued in future work.

635 6. Acknowledgments

636 Matthias Faes gratefully acknowledges the financial support of the Research Foundation Flan-
637 ders (FWO) under grant number 12P3519N. Matthias Faes and Pengfei Wei acknowledge the sup-
638 port of the Alexander von Humboldt foundation. Marcos Valdebenito acknowledges the support of
639 ANID (National Agency for Research and Development, Chile) under its program FONDECYT,
640 grant number 1180271.

641 Appendix A. Calculation of Structural Response

642 In view of the linearity of the structural system, the responses of interest are calculated by
643 means of a convolution, that is:

$$\eta_i(t, \mathbf{y}, \mathbf{z}) = \int_0^t h_i(t - \tau, \mathbf{y}) p(\tau, \mathbf{z}) dt, \quad i = 1, \dots, n_R \quad (\text{A.1})$$

644 where $h_i(t, \mathbf{y})$ is the unit impulse response function associated with the i -th response. For the
645 case where the i -th response of interest is expressed as a linear combination of the displacement
646 vector, the corresponding unit impulse response function is equal to:

$$h_i(t, \mathbf{y}) = \sum_{v=1}^{n_D} \frac{\mathbf{q}_i^T \boldsymbol{\phi}_v(\mathbf{y}) \boldsymbol{\phi}_v(\mathbf{y})^T \boldsymbol{\rho}(\mathbf{y})}{\boldsymbol{\phi}_v(\mathbf{y})^T \mathbf{M}(\mathbf{y}) \boldsymbol{\phi}_v(\mathbf{y})} \frac{e^{-\zeta_v(\mathbf{y}) \omega_v(\mathbf{y}) t}}{\omega_{d,v}(\mathbf{y})} \sin(\omega_{d,v}(\mathbf{y}) t), \quad i = 1, \dots, n_R \quad (\text{A.2})$$

where ω_v , $v = 1, \dots, n_D$ are the natural frequencies; ζ_v , $v = 1, \dots, n_D$ are the damping ratios; $\omega_{d,v} = \omega_v \sqrt{1 - \zeta_v^2}$, $v = 1, \dots, n_D$ are the damped frequencies; ϕ_v , $v = 1, \dots, n_D$ are the eigenvectors associated with the eigenproblem of the undamped equation of motion; and \mathbf{q}_i is a vector such that $\eta_i = \mathbf{q}_i^T \mathbf{x}$, where $(\cdot)^T$ denotes transpose. Note that while the above equation applies to a unit impulse response function associated with displacement, similar expressions can be deduced for other quantities such as, e.g. accelerations. In view of the discrete time representation of the imprecise stochastic loading as shown in eq. (11), the convolution integral in eq. (A.1) reduces to the following expression.

$$\begin{aligned}
\eta_i(t_k, \mathbf{y}, \mathbf{z}) &= \sum_{l_1=1}^k \epsilon_{l_1} h_i(t_k - t_{l_1}, \mathbf{y}) p(t_{l_1}, \mathbf{z}) \Delta t \\
&= \sum_{l_1=1}^k \epsilon_{l_1} h_i(t_k - t_{l_1}, \mathbf{y}) \left(\sum_{l_2=1}^{n_{KL}} \psi_{l_1, l_2} \sqrt{\lambda_{l_2}} z_{l_2} \right) \Delta t \\
&= \mathbf{a}_{i_k}(\mathbf{y}) \mathbf{z}, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T
\end{aligned} \tag{A.3}$$

where ψ_{l_1, l_2} is the (l_1, l_2) -th element of matrix Ψ ; ϵ_{l_1} is a coefficient depending on the quadrature scheme used to approximate the convolution integral; and \mathbf{a}_{i_k} is a row vector of dimension $1 \times n_{KL}$ defined as:

$$\begin{aligned}
\mathbf{a}_{i_k}(\mathbf{y}) &= \left[\sum_{l_1=1}^k \Delta t \epsilon_{l_1} h_i(\mathbf{y}, t_k - t_{l_1}) \psi_{l_1, 1} \sqrt{\lambda_1}, \dots, \sum_{l_1=1}^k \Delta t \epsilon_{l_1} h_i(\mathbf{y}, t_k - t_{l_1}) \psi_{l_1, n_{KL}} \sqrt{\lambda_{n_{KL}}} \right] \\
i_k &= (i - 1)n_T + k, \quad i = 1, \dots, n_R, \quad k = 1, \dots, n_T
\end{aligned} \tag{A.4}$$

Each of the rows of matrix \mathbf{A}_i introduced in eq. (14) contains the row vectors \mathbf{a}_{i_k} , that is:

$$\mathbf{A}_i(\mathbf{y}) = \begin{bmatrix} \mathbf{a}_{i_1}(\mathbf{y}) \\ \vdots \\ \mathbf{a}_{i_{n_T}}(\mathbf{y}) \end{bmatrix}, \quad i = 1, \dots, n_R \tag{A.5}$$

The coefficients ϵ_{l_1} are chosen following the trapezoidal rule for integration (see, e.g. [52]), yielding $\epsilon_{l_1} = 1/2$ in case $l_1 = 1$ or $l_1 = k$; otherwise, $\epsilon_{l_1} = 1$.

661 **Appendix B. Sample generation of Direction Vector**

662 Samples $\mathbf{u}^{(j)}$, $j = 1, \dots, N$ required for evaluating the augmented failure probability in eq. (28)
 663 are generated by means of the following procedure [9, 32, 45].

- 664 1. Set $j = 1$.
- 665 2. Sample $\boldsymbol{\theta}^{(j)}$ from $f_{\Theta}(\boldsymbol{\theta})$.
- 666 3. Sample $\mathbf{y}^{(j)}$ from $f_{\mathbf{Y}}(\mathbf{y}|\boldsymbol{\theta}^{(j)})$.
- 667 4. Draw a pair of indices (I, K) from the set $\Omega = \{(i, k) : i \in \{1, \dots, n_{\eta}\}, k \in \{1, \dots, n_T\}\}$
 668 with probability proportional to the weights $w_{i,k}(\mathbf{y}^{(j)})$, $i = 1, \dots, n_{\eta}$, $k = 1, \dots, n_T$.
- 669 5. Generate a sample \mathbf{z} of the random variable vector \mathbf{Z} and two realizations κ_1 and κ_2 following
 670 a uniform distribution between 0 and 1.
- 671 6. Calculate $\alpha = -F_Z^{-1}((1 - \kappa_1)F_Z(-\beta_{I,K}(\mathbf{y}^{(j)})))$, where $F_Z^{-1}(\cdot)$ is the inverse cumulative
 672 standard Gaussian distribution.
- 673 7. Compute $\mathbf{a}_{IK}^* = \mathbf{a}_{IK}(\mathbf{y}^{(j)}) / \|\mathbf{a}_{IK}(\mathbf{y}^{(j)})\|$. See eq. (A.4) for the calculation of $\mathbf{a}_{IK}(\mathbf{y}^{(j)})$.
- 674 8. Define \mathbf{z}^* as:

$$\mathbf{z}^* = \begin{cases} \mathbf{z} + (\alpha - \mathbf{z}^T \mathbf{a}_{I,K}^*) \mathbf{a}_{I,K}^* & \text{if } \kappa_2 \leq 1/2 \\ -\mathbf{z} - (\alpha - \mathbf{z}^T \mathbf{a}_{I,K}^*) \mathbf{a}_{I,K}^* & \text{otherwise} \end{cases} \quad (\text{B.1})$$
- 675 9. Calculate the sought sample as $\mathbf{u}^{(j)} = \mathbf{z}^* / \|\mathbf{z}^*\|$.
- 676 10. In case $j = N$, stop the procedure. Otherwise, return to step 2 with $j = j + 1$.

677 **References**

- 678 [1] G. Schuëller, Computational Methods in Stochastic Dynamics, Springer Netherlands, 2011,
 679 Ch. Model Reduction and Uncertainties in Structural Dynamics, pp. 1–24.
- 680 [2] M. Faes, D. Moens, Recent Trends in the Modeling and Quantification of Non-probabilistic
 681 Uncertainty, Arch. Comput. Methods Eng. (feb 2019).
- 682 [3] M. Beer, Y. Zhang, S. T. Quek, K. K. Phoon, Reliability analysis with scarce information:
 683 Comparing alternative approaches in a geotechnical engineering context, Struct. Saf. 41 (2013)
 684 1–10.
- 685 [4] M. Broggi, M. Faes, E. Patelli, Y. Govers, D. Moens, M. Beer, Comparison of Bayesian and
 686 interval uncertainty quantification: Application to the AIRMOD test structure, in: 2017

- IEEE Symposium Series on Computational Intelligence, SSCI 2017 - Proceedings, Vol. 2018- 687
Janua, 2018, pp. 1–8. 688
- [5] M. Faes, M. Broggi, E. Patelli, Y. Govers, J. Mottershead, M. Beer, D. Moens, A multivariate 689
interval approach for inverse uncertainty quantification with limited experimental data, *Mech.* 690
Syst. Sig. Process. 118 (2019) 534–548. 691
- [6] G. Deodatis, Non-stationary stochastic vector processes: seismic ground motion applications, 692
Probab. Eng. Mech. 11 (3) (1996) 149 – 167. 693
- [7] M. Shinozuka, Y. Sato, Simulation of nonstationary random process, *Journal of the Engi-* 694
neering Mechanics Division 93 (1) (1967) 11–40. 695
- [8] S. Ferson, V. Kreinovich, L. Ginzburg, D. S. Myers, K. Sentz, Constructing Probability Boxes 696
and Dempster-Shafer Structures, Tech. Rep. January, Technical report, Sandia National Lab- 697
oratories (2003). 698
- [9] S. Au, J. Beck, First excursion probabilities for linear systems by very efficient importance 699
sampling, *Probab. Eng. Mech.* 16 (3) (2001) 193–207. 700
- [10] D. Moens, D. Vandepitte, An interval finite element approach for the calculation of envelope 701
frequency response functions, *Int. J. Numer. Methods Eng.* 61 (14) (2004) 2480–2507. 702
- [11] W. Gao, D. Wu, C. Song, F. Tin-Loi, X. Li, Hybrid probabilistic interval analysis of bar 703
structures with uncertainty using a mixed perturbation monte-carlo method, *Finite Elem.* 704
Anal. Des. 47 (7) (2011) 643–652. 705
- [12] B. Xia, D. Yu, J. Liu, Hybrid uncertain analysis for structural-acoustic problem with random 706
and interval parameters, *J. Sound Vib.* 332 (11) (2013) 2701–2720. 707
- [13] S. Yin, D. Yu, Z. Luo, B. Xia, Unified polynomial expansion for interval and random response 708
analysis of uncertain structure-acoustic system with arbitrary probability distribution, *Com-* 709
put. Methods Appl. Mech. Eng. 336 (2018) 260 – 285. 710
- [14] J. Wu, Z. Luo, Y. Zhang, N. Zhang, L. Chen, Interval uncertain method for multibody 711
mechanical systems using Chebyshev inclusion functions, *Int. J. Numer. Methods Eng.* 95 (7) 712
(2013) 608–630. 713

- 714 [15] R. Schöbi, B. Sudret, Structural reliability analysis for p-boxes using multi-level meta-models,
715 Probab. Eng. Mech. 48 (2017) 27–38.
- 716 [16] R. Schöbi, B. Sudret, Global sensitivity analysis in the context of imprecise probabilities (p-
717 boxes) using sparse polynomial chaos expansions, Rel. Eng. Syst. Saf. 187 (2019) 129 – 141,
718 sensitivity Analysis of Model Output.
- 719 [17] M. Faes, J. Sadeghi, M. Broggi, M. de Angelis, E. Patelli, M. Beer, D. Moens, On the Robust
720 Estimation of Small Failure Probabilities for Strong Nonlinear Models, ASCE-ASME J Risk
721 and Uncert in Engrg Sys Part B Mech Engrg 5 (4) (dec 2019).
- 722 [18] J. Sadeghi, M. de Angelis, E. Patelli, Robust propagation of probability boxes by interval
723 predictor models, Struct. Saf. 82 (2020) 101889.
- 724 [19] P. Wei, J. Song, S. Bi, M. Broggi, M. Beer, Z. Lu, Z. Yue, Non-intrusive stochastic analysis
725 with parameterized imprecise probability models: I. performance estimation, Mech. Syst. Sig.
726 Process. 124 (2019) 349 – 368.
- 727 [20] P. Wei, J. Song, S. Bi, M. Broggi, M. Beer, Z. Lu, Z. Yue, Non-intrusive stochastic analysis
728 with parameterized imprecise probability models: II. reliability and rare events analysis,
729 Mech. Syst. Sig. Process. 126 (2019) 227 – 247.
- 730 [21] M. G. Faes, M. A. Valdebenito, D. Moens, M. Beer, Bounding the first excursion probability
731 of linear structures subjected to imprecise stochastic loading, Computers & Structures 239
732 (2020) 106320.
- 733 [22] M. G. R. Faes, M. A. Valdebenito, Fully Decoupled Reliability-Based Design Optimization
734 of Structural Systems Subject to Uncertain Loads, Computer Methods in Applied Mechanics
735 and Engineering (2020).
- 736 [23] M. G. R. Faes, M. A. Valdebenito, D. Moens, M. Beer, Operator norm theory as an effi-
737 cient tool to propagate hybrid uncertainties and calculate imprecise probabilities, Mechanical
738 Systems and Signal Processing (2021).
- 739 [24] S. Au, Probabilistic failure analysis by importance sampling Markov chain simulation, J. Eng.
740 Mech. 130 (3) (2004) 303–311.

- [25] J. Ching, Y. Hsieh, Local estimation of failure probability function and its confidence interval with maximum entropy principle, *Probab. Eng. Mech.* 22 (1) (2007) 39–49.
- [26] J. Ching, Y. Hsieh, Approximate reliability-based optimization using a three-step approach based on subset simulation, *J. Eng. Mech.* 133 (4) (2007) 481–493.
- [27] H. Jensen, M. Valdebenito, G. Schuëller, An efficient reliability-based optimization scheme for uncertain linear systems subject to general Gaussian excitation, *Comput. Methods Appl. Mech. Eng.* 198 (1) (2008) 72–87.
- [28] A. Taflanidis, J. Beck, An efficient framework for optimal robust stochastic system design using stochastic simulation, *Comput. Methods Appl. Mech. Eng.* 198 (1) (2008) 88–101.
- [29] P. Koutsourelakis, Design of complex systems in the presence of large uncertainties: A statistical approach, *Comput. Methods Appl. Mech. Eng.* 197 (49-50) (2008) 4092–4103.
- [30] J. Song, P. Wei, M. Valdebenito, S. Bi, M. Broggi, M. Beer, Z. Lei, Generalization of non-intrusive imprecise stochastic simulation for mixed uncertain variables, *Mech. Syst. Sig. Process.* 134 (2019) 106316.
- [31] J. Zhang, M. D. Shields, On the quantification and efficient propagation of imprecise probabilities resulting from small datasets, *Mech. Syst. Sig. Process.* 98 (June) (2018) 465–483.
- [32] M. A. Misraji, M. A. Valdebenito, H. A. Jensen, C. F. Mayorga, Application of directional importance sampling for estimation of first excursion probabilities of linear structural systems subject to stochastic Gaussian loading, *Mechanical Systems and Signal Processing* 139 (2020) 106621. doi:10.1016/j.ymssp.2020.106621.
URL <https://linkinghub.elsevier.com/retrieve/pii/S0888327020300078>
- [33] M. Beer, S. Ferson, V. Kreinovich, Imprecise probabilities in engineering analyses, *Mech. Syst. Sig. Process.* 37 (1-2) (2013) 4–29.
- [34] M. G. R. Faes, M. Daub, S. Marelli, E. Patelli, M. Beer, Engineering analysis with probability boxes : a review on computational methods (2021).
- [35] X. Yuan, S. Liu, M. Valdebenito, J. Gu, M. Beer, Efficient framework for failure probability function estimation in augmented space, *Structural Safety* (2021, submitted).

- 768 [36] O. Ditlevsen, P. Bjerager, R. Olesen, A. Hasofer, Directional simulation in Gaussian processes,
769 Probab. Eng. Mech. 3 (4) (1988) 207 – 217.
- 770 [37] G. Stefanou, The stochastic finite element method: Past, present and future, Comput. Meth-
771 ods Appl. Mech. Eng. 198 (9-12) (2009) 1031–1051.
- 772 [38] A. Chopra, Dynamics of structures: theory and applications to earthquake engineering, Pren-
773 tice Hall, 1995.
- 774 [39] S. Au, J. Beck, Estimation of small failure probabilities in high dimensions by subset simu-
775 lation, Probab. Eng. Mech. 16 (4) (2001) 263–277.
- 776 [40] A. Der Kiureghian, The geometry of random vibrations and solutions by FORM and SORM,
777 Probab. Eng. Mech. 15 (1) (2000) 81–90.
- 778 [41] M. Valdebenito, H. Jensen, A. Labarca, Estimation of first excursion probabilities for uncer-
779 tain stochastic linear systems subject to gaussian load, Computers & Structures 138 (2014)
780 36–48.
- 781 [42] A. Ang, W. Tang, Probability Concepts in Engineering: Emphasis on Applications to Civil
782 and Environmental Engineering, Wiley, 2007.
- 783 [43] I. Deák, Three digit accurate multiple normal probabilities, Numerische Mathematik 35 (4)
784 (1980) 369–380.
- 785 [44] P. Bjerager, Probability integration by directional simulation, J. Eng. Mech. 114 (8) (1988)
786 1285–1297.
- 787 [45] L. Katafygiotis, S. Cheung, Domain decomposition method for calculating the failure proba-
788 bility of linear dynamic systems subjected to Gaussian stochastic loads, J. Eng. Mech. 132 (5)
789 (2006) 475–486.
- 790 [46] G. Schuëller, R. Stix, A critical appraisal of methods to determine failure probabilities, Struct.
791 Saf. 4 (4) (1987) 293–309.
- 792 [47] H. Jensen, M. Valdebenito, Reliability analysis of linear dynamical systems using approximate
793 representations of performance functions, Struct. Saf. 29 (3) (2007) 222–237.

- [48] H. Pradlwarter, G. Schuëller, Uncertain linear structural systems in dynamics: Efficient stochastic reliability assessment, *Computers & Structures* 88 (1-2) (2010) 74–86. 794
795
- [49] A. Zerva, *Spatial Variation of Seismic Ground Motions – Modeling and Engineering Applications*, CRC Press, 2009. 796
797
- [50] M. Faes, D. Moens, Multivariate dependent interval finite element analysis via convex hull pair constructions and the extended transformation method, *Comput. Methods Appl. Mech. Eng.* 347 (2019) 85 – 102. 798
799
800
- [51] L. J. Howell, Y. Lin, Response of flight vehicles to nonstationary atmospheric turbulence, *AIAA Journal* 9 (11) (1971) 2201–2207. doi:10.2514/3.50026. 801
802
- [52] W. Gautschi, *Numerical Analysis*, 2nd Edition, Birkhäuser Boston, 2012. 803