

CFD ANALYSIS OF THE NEEDLE TIP ANGLE IN PELTON INJECTOR ON JET QUALITY FOR THE POWER GENERATION

Daniel G. Taborda

*Department of Mechatronics Engineering¹
danielgiraldo209012@correo.itm.edu.co*

Jorge Sierra-Del Rio

*Department of Mechatronics Engineering¹
jorgesierra@itm.edu.co*

Juan Diego Perez-Alvarez

*Department of Mechatronics Engineering¹
juanperez238421@correo.itm.edu.co*

Arley Cardona-Vargas

*Department of Mechatronics Engineering¹
arleycardona5670@correo.itm.edu.co*

Daniel Sanin Villa

*Department of Mechanic Engineering
Pontifical Bolivarian University
Cq. 1 #70-01, Medellin, Antioquia, Colombia, 54
daniel.sanin@upb.edu.co*

*¹Metropolitan Technological Institute
Cl. 75 #75-101, Medellin, Antioquia, Colombia, 54*

Abstract

Fossil fuels are energy sources that supply a large part of the world's energy generation. However, they produce greenhouse gases such as carbon dioxide (CO₂), nitrogen oxide (NO_x) and particulates that increase global warming. For this reason, other forms of renewable energy such as hydropower have begun to be implemented through turbomachinery such as Pelton turbines, which significantly reduce these emissions since they are highly efficient turbines based on the use of natural resources (water). Pelton turbines are based mainly on three components for their operation, which are the Pelton injector, the bucket and the wheel. The injector is an important component in the energy transformation of Pelton turbines. Although to analyze its behavior, it is possible to use fluid dynamics (CFD) software to predict the trajectory of the flow through a solid or free surface. The objective of this work is to analyze by means of computational fluid dynamics (CFD) the incidence of the length and the needle tip angle of a Pelton turbine injector on the generated power. For this, an ANSYS 2020R2 computational fluid analysis software was used to study how the variation of the injector needle tip angle influences through the volume of fluid (VOF) method, starting from the generation of a commercial model with a tip angle of 60° and two (2) geometries of 55° and 75° respectively. Numerical results show a better performance for the 75° angle of 96 % and lower for the 55° and 60° with 94.1 % and 95.5 % respectively, whereby steeper angles achieve higher performances. In summary, the present study pretends to increase the power generation, in the face of phenomena occurred in the energy transfer. Although the performance of the injector in each angle configuration must be tested in practice.

Keywords: computational fluids dynamics, volume of fluid method, injector efficiency.

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1. Introduction

Nowadays, fossil fuels supply 84.3 % of the world's total electricity generation. However, the combustion and generation of pollutants emitted into the atmosphere such as carbon dioxide (CO₂) and nitrogen oxides (NO_x) increase the production of greenhouse gases (GHG) and, consequently, global warming. For this reason, the use of hydropower has been increasing in recent years as a renewable energy source, where it is estimated that today they cover approximately 6 % of the

world's total electricity generation [1]. Here, hydraulic impulse turbines such as Pelton turbines have become one of the most used renewable energy sources since their invention by [2] in 1880, due to they take advantage of the kinetic energy and potential of high head and relatively low flow rates from an injector that is responsible for accelerating and regulating the flow of water towards the turbine buckets, transforming the kinetic energy into rotational mechanical energy [3].

The main factors affecting the performance of Pelton turbines are found in the casing, the injector, the runner and the operating conditions. Among them, the injector plays an important role in the conversion of fluid pressure energy into kinetic energy. For this reason, in recent years, Pelton turbine efficiency gains have been studied with the support of experimental and/or numerical methodologies, modifying both geometrical and operational parameters in order to estimate the effects of the injector on energy transfer. However, the analysis of Pelton-type turbines is complex and requires sophisticated equipment implementing computational techniques such as the VOF (volume of fluid) method [4] and the SPH (Smoothed Particle Hydrodynamics) [5] or experimental observation and measurement techniques of water flow such as the LDA (Laser Doppler Anemometry) method [6, 7] and PIV (Particle Image Velocity) devices [8]. For example, [9, 10] adopted the VOF model configurations to analyze the jet characteristics in the face of head changes. The results of their investigation revealed that the secondary flows induced by Dean vortices upstream of the injector caused deformations downstream of the injector affecting the turbine performance. Then, [11] numerically and experimentally analyzed the free jet surface of a Pelton turbine, and confirmed that the asymmetric presence in the jet profile is due to the secondary flows influenced by the Dean vortices generated at the bend of the elbow. Later, [12–14], compared the effects on jet quality in industrial Pelton turbines by flow visualization and performance tests, confirming that jet dispersion is the principal cause of the decrease in Pelton turbine performance. Then, [15–17] analyze by numerical studies the effect of jet quality on the efficiency of a Pelton turbine injector for three configurations of 80/55, 110/70 and 180/90 in nozzle angle and needle tip angle respectively, finding efficiencies close to 98 % at steeper needle tip angles. [18–21] analyzed the jet flow from the injector opening percentage, showing increases in efficiencies close to 4 % for openings between 60 % and 90 %. Then, [22, 23] numerically compared turbine injector performance under varying needle and nozzle angles, concluding better efficiencies at steeper angles. Finally, [24, 25] compared the changes in turbine performance with changing geometrical parameters in the injector, confirming that the cause of jet dispersion is mainly due to the injector geometry.

In general, it is possible to improve efficiency by varying the geometrical parameters and percentage of a Pelton turbine injector, especially the length and the needle tip angle. However, improving efficiency involves analyzing factors that affect the quality of the jets, whether in dispersion, deformation or rotation. Although nowadays there are numerous studies to guarantee an injector with optimum characteristics, there is still a lack of studies to further increase the turbine efficiency. For this reason, this study focuses only on the CFD analysis of the variation in the needle tip angle using VOF simulation methods. Thus, the objective of the study is to analyze by CFD simulation the incidence of the length and the needle tip angle of a Pelton turbine injector on the generated power.

2. Methodology

2.1. Geometric generation

This work was developed for two injectors with needle tip angle of 55°/130° and 75°/130°, starting from the model provided by the Techydro trademark, which considers a needle tip angle of 60°/130°. The geometrical parameters in the design of the injector were obtained through the methodology established in **Fig. 1** where the site conditions were initially determined, specifically the head (1), the generated power (2) and the velocity of the jet (3). Then, the jet diameter (d_0) was determined by (4) and the nozzle diameter (d_1) by (5) from the site conditions corresponding to the flow and head of 0.021 m³/s and 35 m respectively. Finally, it was validated by (6) if the calculated nozzle diameter is between the maximum and minimum nozzle diameter allowed.

Once the diameter of the nozzle was determined and using the parameters established in **Table 1** and **Fig. 2** according to the study carried out by [26], the dimensions corresponding to the geometry of the injector were determined.

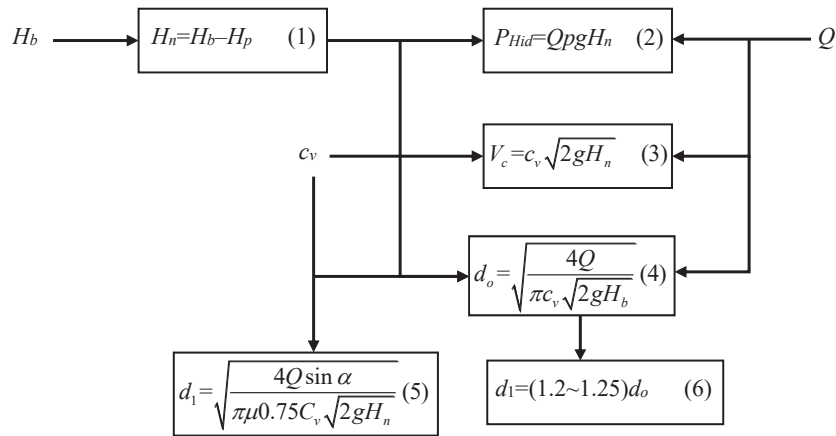


Fig. 1. Nozzle diameter and jet calculation process

Table 1
Obtaining the dimensions of the injector [26]

Nomenclature	Correlation
<i>C</i>	$0.63d_1$
<i>X</i>	$0.503d_1$
<i>S</i>	$1.35d_1$
<i>d</i>	$0.63d_1$
<i>l</i>	$3.17d_1$
<i>r</i>	$0.705d_1$
<i>R</i>	$2.2d_1$
<i>h</i>	$0.6d_1$
<i>e</i>	$0.05d_1$
<i>f</i>	$1.13d_1$

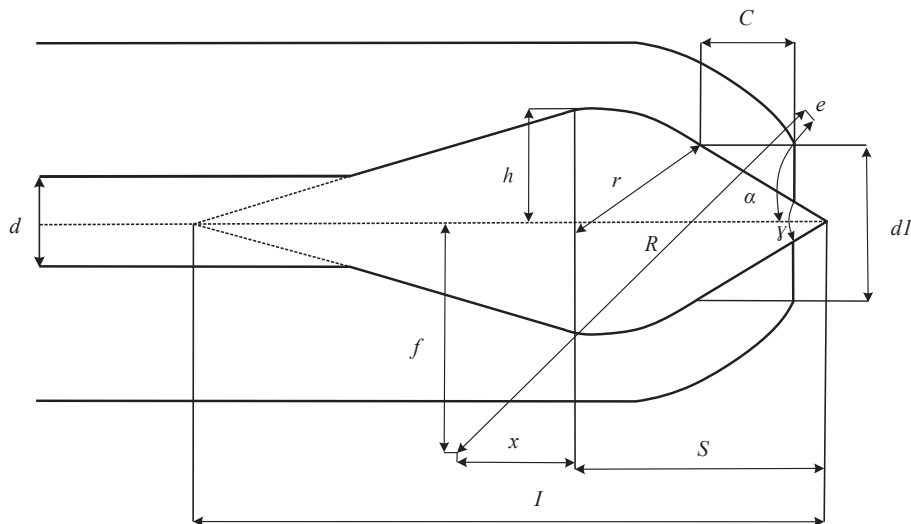


Fig. 2. Dimensions of the injector [26]

Table 1 and Fig. 2 show the nomenclature for obtaining the injector geometry and model. In addition, the dimensions correlate with the injector nozzle diameter in order to obtain the injector model for each selected needle tip angle (55°, 60° and 75°).

2. 2. Control volume and boundary conditions

Based on the geometries obtained in the design stage and using the SPACECLAIM® module of the ANSYS WORKBENCH program, control volumes were obtained corresponding to the volume of water flow contained inside the injector and the volume of air flow, as shown schematically in Fig. 3.

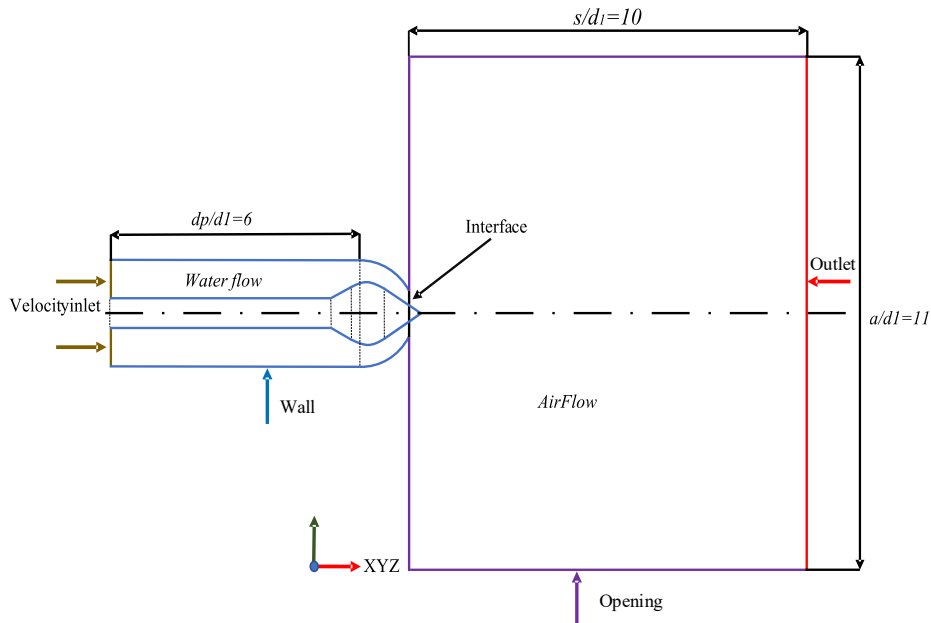


Fig. 3. Control volume and boundary conditions of the injector

For the configuration of the numerical model, the boundary conditions representing the physics of the problem were established. For this, a water-air interaction interface, a normal speed to the surface of 4,779 m/s at the input, and an atmospheric boundary condition (0 kPa) in the outlet was configured. Then, in order to establish and capture adequately the air currents induced by the action of the water jet, opening conditions were defined at the outlet of the nozzle. Finally, stationary walls were configured with a roughness of 0.03 and the boundary conditions for a volume fraction for gradients from 0 (air) to 1 (water) for the boundary conditions.

2. 3. Mesh generation

The control volume discretization process was performed in the commercial ANSYS V20R1 software in the ICEM module. There, a structured mesh with hexahedral elements was implemented through the association of blocks and curves that compose the geometry of the control volume, as shown in Fig. 4, a.

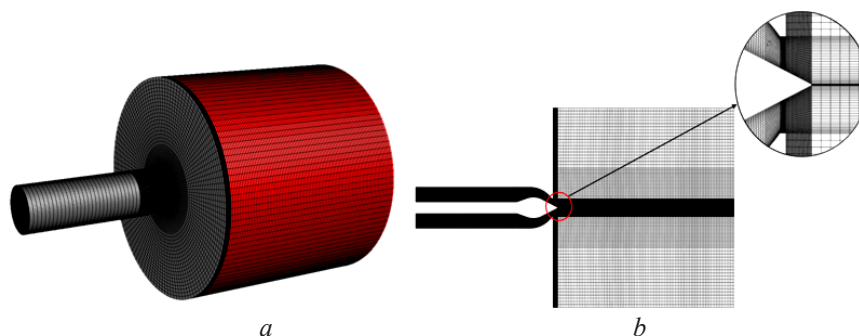


Fig. 4. Distribution mesh:
a – Isometric distribution mesh; b – Distribution plane XY, refinement of the mesh

Additionally, inflation was performed close to the walls of the needle and nozzle from a minimum required wall in order to adequately capture the hydrodynamic profile of the fluid as it interacts with the injector needle as shown in **Fig. 4, b**. For this, it was calculated the value of y_1 (height of the first cell), using a line tool as «Pointwise» [27]. Here, it was specified the velocity, density, dynamic viscosity and the diameter of the injector pipe for a $30 < y^+ < 80$, obtaining a resolution close to the wall of 4 mm [15].

Table 2 presents the results obtained in the mesh metrics for each configured case of the injector in which it is validated that the quality of the mesh meets the necessary criteria for the simulation in ANSYS FLUENT, according to the parameters shown in the ANSYS working guides [28].

Table 2

Meshing metrics for injector needle geometries with 55°, 60°, 75° tip angle

Needle tip angle	Mesh	Recommend by [28]			
		Quality [>0.15]	Aspect relation [<25]	Angle (Deg) [$>18^\circ$]	Determinant $2 \times 2 \times 2$ [>15]
55°	Mesh 1	0.206	20	35.37	0.206
	Mesh 2	0.352	20	35.37	0.352
	Mesh 3	0.321	20	35.37	0.321
60°	Mesh 1	0.215	20	35.37	0.215
	Mesh 2	0.285	13	35.37	0.285
	Mesh 3	0.329	17	35.37	0.329
75°	Mesh 1	0.219	20	35.37	0.219
	Mesh 2	0.278	18	35.37	0.278
	Mesh 3	0.342	20	35.37	0.342

In order to decrease the numerical error associated with the control volume discretization process, a mesh independence analysis was performed in response to the maximum water jet velocity as shown in **Table 3** and **Fig. 5**.

Furthermore, it is possible to infer that the variable of interest (velocity) does not suffer significant changes higher than 2 %, that is, the simulations with 2.06×10^6 (55° needle), 3.14×10^6 (60° Techydro needle) and 2.73×10^6 (75° needle) are suitable to follow the simulation process, without consuming too many computational resources.

Table 3

Mesh Independence

Needle tip angle	Simulations	Elements number	Nodes number	Maximum velocity [m/s]	%Error
55°	Simulation 1	1.776.445	1.738.012	33,47	0 %
	Simulation 2	2.068.591	2.027.692	34,28	2 %
	Simulation 3	2.643.893	2.596.412	34,73	4 %
60°	Simulation 1	1.927.341	1.891.248	34.95	0 %
	Simulation 2	2.449.921	2.408.608	35.48	2 %
	Simulation 3	3.148.241	3.099.968	36.34	3 %
75°	Simulation 1	2.003.911	1.966.648	46.6	0 %
	Simulation 2	2.739.351	2.695.128	48.1	2 %
	Simulation 3	3.521.191	3.470.008	45.5	4 %

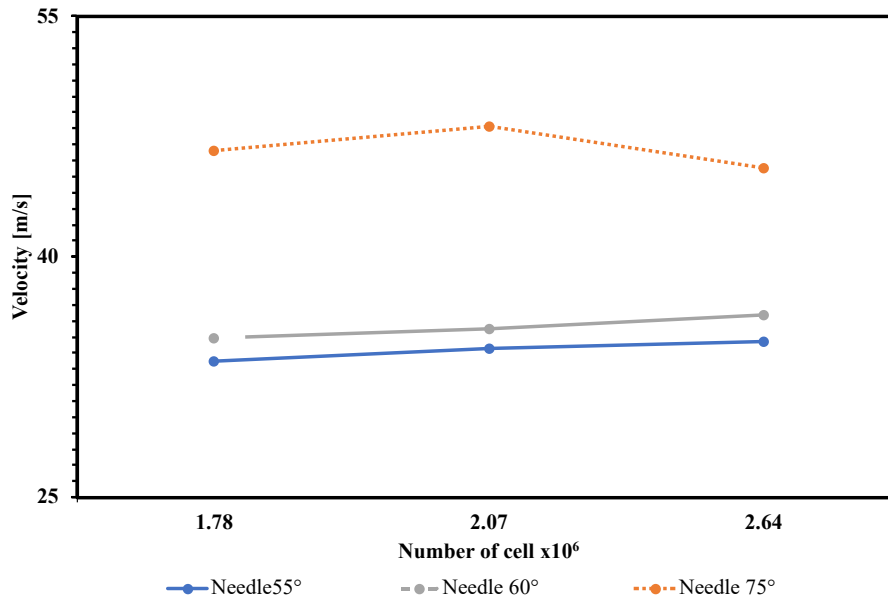


Fig. 5. Mesh Independence distribution

2. 4. Numerical Method

A computational analysis was performed using ANSYS FLUENT V2020R1 from the approximations established by the Navier Stokes equations and the continuity, momentum and energy equations. The flow of the water jet has multiphase characteristics in a Pelton turbine which allows predicting the interaction between two immiscible fluids on a free surface, in which the position of the phase-to-phase interface is of interest for transient monitoring. Therefore, a multiphase VOF (volume of fluid) model was configure water and air at 25 °C and a constant surface tension of 0.072 N/m. In addition, the governing equations throughout the domain which depend on the volume fractions of all phases across the properties ρ and μ as shown below [4, 29]:

$$\rho = \sum \alpha_i \rho_i, \quad (7)$$

$$\mu = \sum \alpha_i \mu_i, \quad (8)$$

$$\frac{\partial \rho}{\partial t} + \frac{\partial \rho \mu_i}{\partial x_i} = 0, \quad (9)$$

$$\frac{\partial \alpha_w}{\partial t} - u_i \frac{\partial \alpha_w}{\partial x_i} = 0, \quad (10)$$

$$\frac{\partial}{\partial t}(\rho \mu_i) + \frac{\partial}{\partial x_i}(\rho \mu_i \mu_j) = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial \mu_i}{\partial x_j} + \frac{\partial \mu_j}{\partial x_i} - \frac{2}{3} \delta_{ij} \frac{\partial \mu_l}{\partial x_l} \right) \right] + \frac{\partial}{\partial x_j} (-\rho \overline{\mu'_i \mu'_j}). \quad (11)$$

The $k-\varepsilon$ turbulence model was selected for the study due to its good ability to correctly capture the viscous effects on the injector needle tip boundary layer and the free surface of the water jet, with low computational cost. The equations that govern the model are shown below [30, 31]:

$$\frac{\partial(\rho \overline{\mu k})}{\partial x} + \frac{\partial(\rho \overline{v k})}{\partial y} = \frac{\partial}{\partial x} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x} \right] + \frac{\partial}{\partial y} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial y} \right] + P_k + G_k - \rho \varepsilon - Y_M, \quad (12)$$

$$\frac{\partial(\rho \overline{\mu \varepsilon})}{\partial x} + \frac{\partial(\rho \overline{v \varepsilon})}{\partial y} = \frac{\partial}{\partial x} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x} \right] + \frac{\partial}{\partial y} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial y} \right] + \rho C_1 - \rho C_2 \varepsilon \frac{\varepsilon^2}{k + \sqrt{v \varepsilon}} + C_{1\varepsilon} \frac{\varepsilon}{k} C_{3\varepsilon} G_b. \quad (13)$$

Subsequently, the simulation was performed in a transient state in order to observe the behavior of the jet surface through time. Then, a total simulation time of 100 s was configured assuming a Courant-Friedrichs-Lewy (CFL) of 0.5 and an adaptive time step according to the cell size as shown in **Table 4**.

$$CFL = \frac{V \Delta t}{\Delta x} < 1. \quad (14)$$

This was calculated by (14), where Δx is the distance from the cell in [m], Δt is the time step in [s] and $[V]$ is the velocity of the fluid in [m/s] where it required a simulation time of approximately 36 hours for each case and an approximate duration of 324 hours, for the entire study. For this, it used a computer that has an Intel Xeon Es-2680 operating system with 16 logical processors and a RAM of 256 Gb. Finally, as a stop criterion for each simulation case, a minimum residual value of 4×10^{-4} was established and a fluid velocity monitor at the nozzle outlet was monitored and controlled in order to verify a stable response.

Table 4

Time step Adaptive configuration in each case

Needle tip angle	CFL	Cell's size [m·10 ⁻⁵]	Time Step [s·10 ⁻⁶]
55°	0.5	4.11	4.30
60°	0.5	3.47	3.63
75°	0.5	3.64	7.62

3. Results

3.1. Geometry results

As a result of the design and modeling process of the injectors, three CAD models were obtained in commercial software ANSYS V20R1 in the «SPACECLAIM» module for 100 % opening, one of which corresponds to the commercial model of the Techydro brand with a 60° needle tip angle and two from the design equations described in the methodology.

Fig. 6 show the needle tip angle variation where the black color with continue line indicate the needle tip angle for 55°, the blue color with dash line for 60° and red color with round dot for 75°.

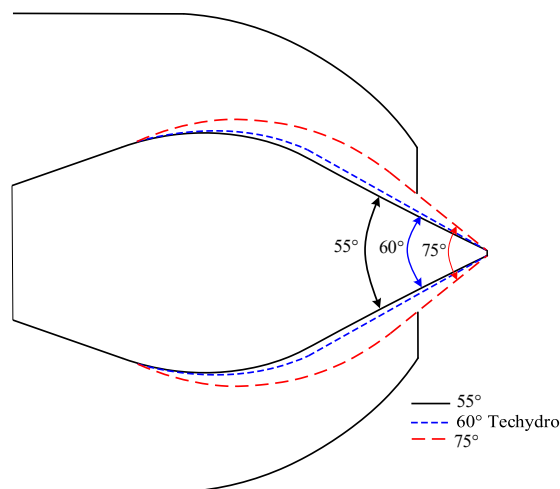


Fig. 6. Needle tip angle variation

3.2. Discussion of simulation results

The results are shown in a longitudinal reference plane and 3 transverse reference planes located in a normalized distance (s/d_1), of 0.29, 0.74, 1.23, 5.61 and 10 at the nozzle exit where «(s)» corresponds to the distance between the nozzle and the reference plane and d_1 to the nozzle diameter (**Fig. 7**).

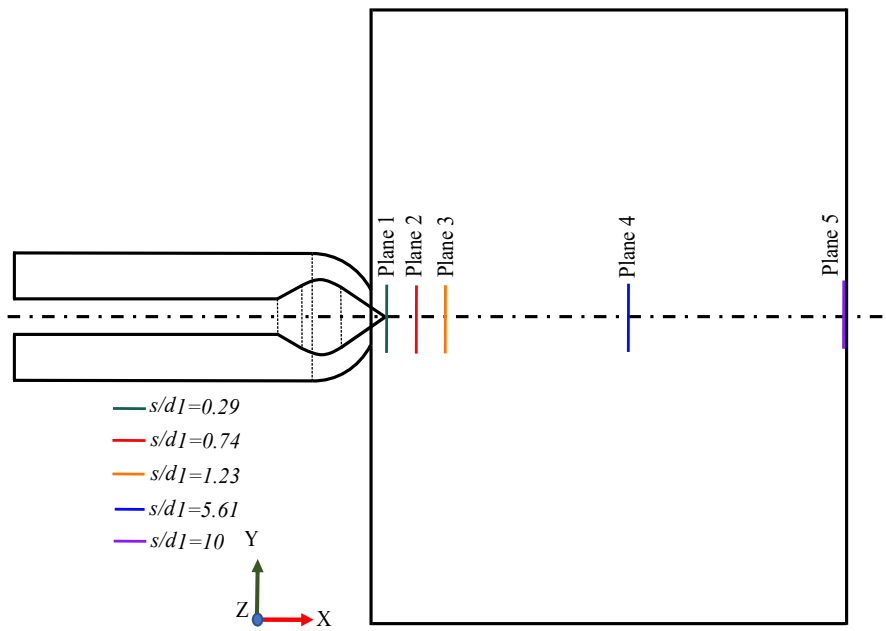


Fig. 7. Transversal and longitudinal reference planes

Additionally, it is possible to see the plane in **Fig. 7** with colored lines indicating the position in which the results were taken where the red color indicated a normalized distance same to 0.29, green line same to 0.74, orange line same to 1.23, blue line same to 5.61 and finally purple line same to 10.

3. 2. 1. Jet characteristics

Fig. 8 presents the volumetric fraction contours from the planes described in **Fig. 7** for each design (55° , 60° Techydro and 75°).

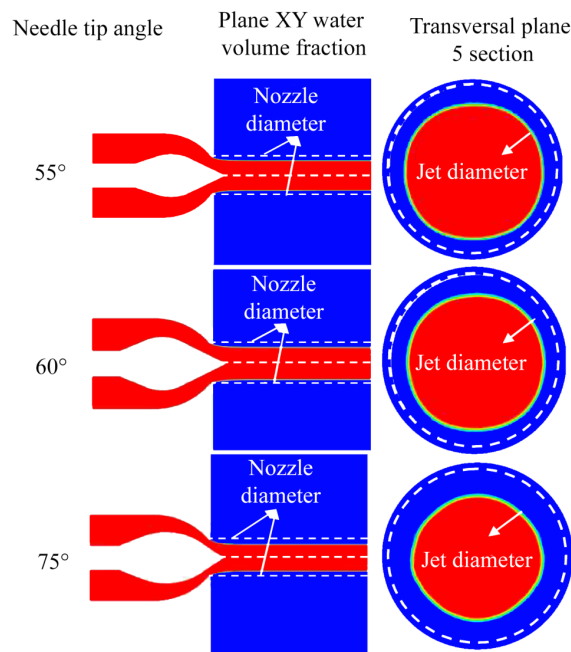


Fig. 8. Water volume fraction for each case

The volume fraction of water allowed the prediction and monitoring of the water-air jet interface where a value of $\alpha_w = 1$ represents the volume fraction of a cell occupied by water

while $\alpha_w=0$ indicates a cell occupied by air (10). Furthermore, white dotted lines corresponding to the nozzle diameter and 0 % eccentricity described by [32] are presented, observing a minimum jet contraction and transversal deformations as a result of secondary currents on its surface [33], resulting in standardized jet diameters (d_{oCFD}/d_{oTH}) of 0.94, 0.85 and 0.76 for the needle tip angles of 55°, 60° and 75° respectively. This means that the smaller the jet diameter, the higher the velocity and efficiency.

Fig. 9 presents the profile of the water fluid volume with respect to the diameter of the jet for each case of injector (55°, 60° and 75°). The Fig. shows slight changes in the jet diameter as the normalized distance $s/d1$ specified in planes 3, 4 and 5 of **Fig. 5** is increased. Here, a symmetrical dispersion of the jet along its trajectory can be visualized, especially for the proposed needle angles (55° and 75°). This is caused by the conservation of the mass and loss of energy in the jet, resulting in an increase in the jet's cross-sectional area and a conservation of the flow along its trajectory. However, in the commercial injector a jet dispersion is observed once it exceeds 0.5 of the fluid volumes for the three normalized distances mentioned due to the deformations that occur on the surface of the jet during its trajectory to the turbine buckets.

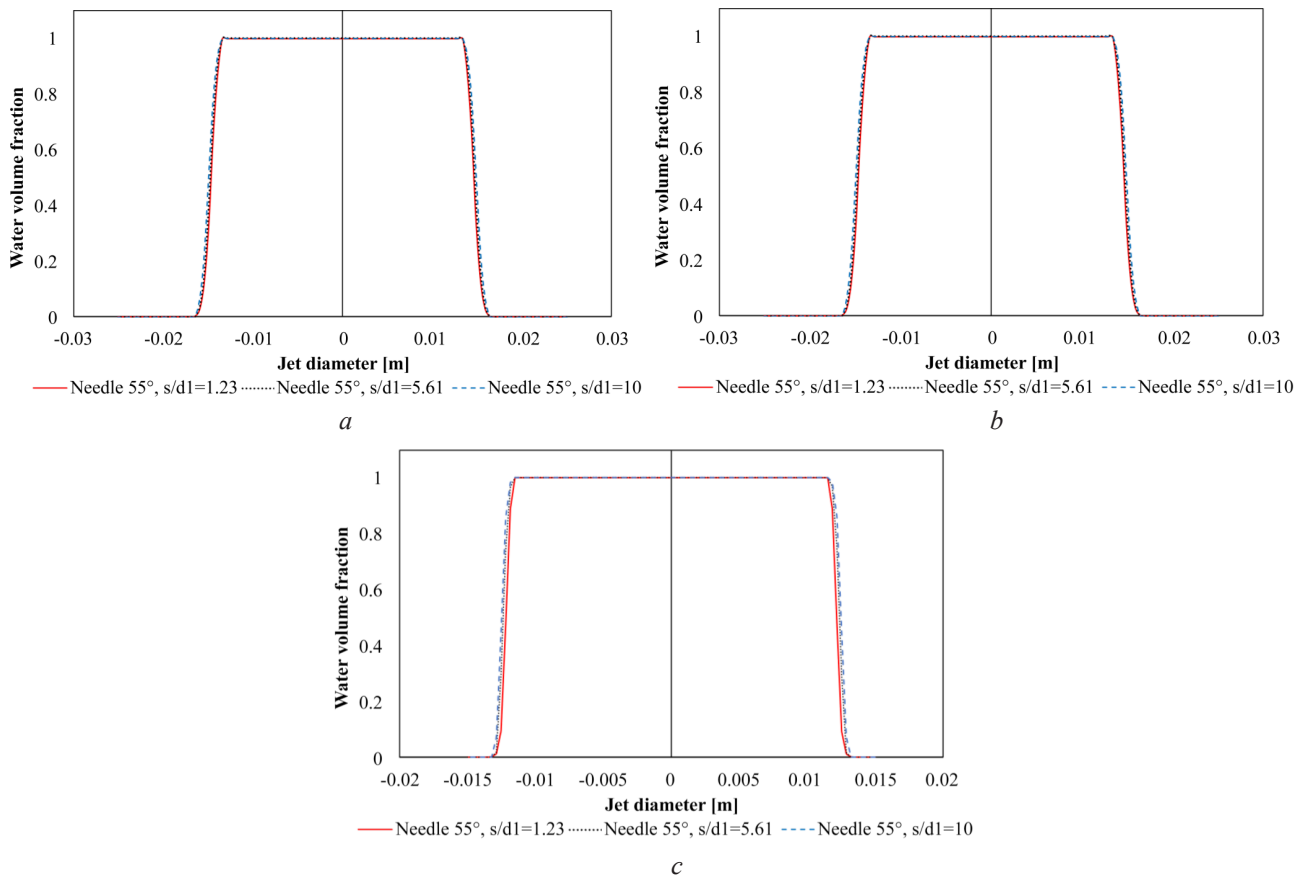


Fig. 9. Jet diameter for water volume fraction: *a* – Needle 55°; *b* – Needle 60°; *c* – Needle 75°

Fig. 10 shows the velocity contours for the injector of *a* – 55°, *b* – 60° and *c* – 75°. It is possible to observe maximum velocity in the jet of 33.5 m/s, 35.5 m/s and 48.1 m/s for 55°, 60° and 75° needle tip angles respectively, which also decrease at the edge of the jet and at the core of the jet. This indicates, obtaining higher velocities for slightly more pronounced angles as observed in the study conducted by [15]. There, the increasing the angle of the needle and nozzle reduces the length of fluid interaction on them, reduces losses and increases the uniformity of the water jet which more beneficial for the performance of the turbine. Additionally, the cases show that the water jet is not totally uniform in its core because it has not reached its maximum velocity.

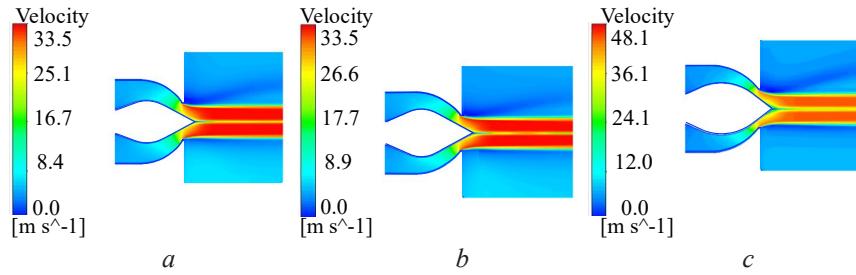


Fig. 10. Velocity contour distribution: *a* – Needle 55°; *b* – Needle 60°; *c* – Needle 75°

Fig. 11 shows the velocity profile in different positions of the water jet for the cases $s/d = 0.29$; 0.74 and 1.23. The curves present lower velocities in the core of the jet, especially for the needle tip angle of 60°/130° where the velocity in the core is almost zero for a normalized distance of 0.29. This indicates that the distribution of the jet is not completely uniform in the core because the curvature presented has not reached its maximum value of velocity, and therefore it is still in the process of acceleration as presented in the experimental study conducted by [7]. Furthermore, according to the study provided by [16]. They indicated that the steeper angles in the injector needle present greater efficiency and better conditions in the operation of the turbine.

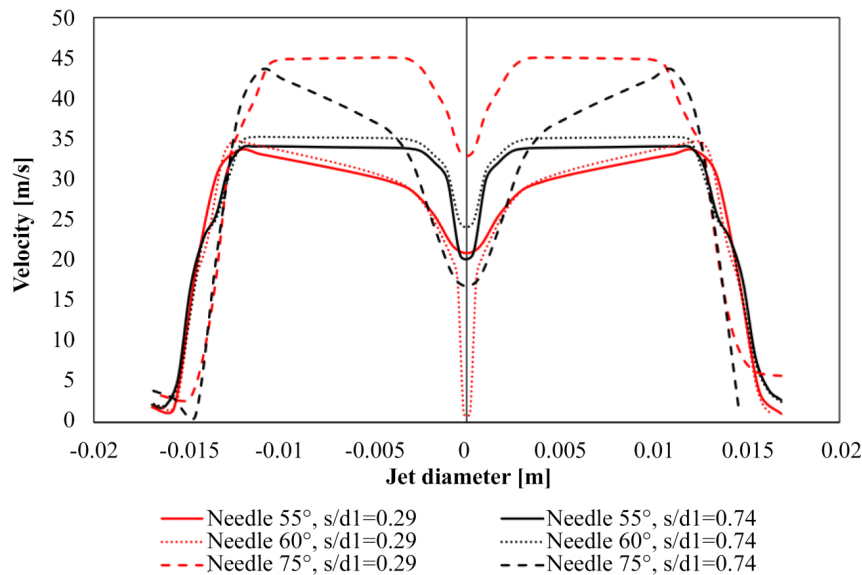


Fig. 11. Injector velocity profile for 55°, 60° and 75°

Fig. 12 shows the velocity profile in different positions of the water jet for the cases s/d_1 of 1.23, 5.61 and 10 respect to Y axis. It is possible to perceive a reduction of the velocity in the core of the water jet, especially for a normalized longitudinal distance (s/d_1) of 1.23 for 55°, 60° and 75°. This indicates that the jet has not yet achieved its maximum velocity and is therefore in the process of accelerating. In addition, a non-symmetrical divergence of the jet velocity profile can be observed as s/d_1 increases, because the effect of secondary currents increases the velocity of the fluid at those points, generating bucket imbalances that affect the power of the turbine.

In contrast, **Fig. 12, 13** show a symmetrical increase in velocity at the edge of the jet, generating a balance that positively affects the impact of the jet on the bucket with respect to the Z -axis. Additionally, it is possible to observe a symmetry in the jet core for all three cases, obtaining jets more uniform in Z axis. On the other hand, as in **Fig. 12**, velocities of approximately 6 m/s are observed at the edge of the jet as a result of the increased velocity of the secondary currents in the

jet. For this reason, although the more pronounced angles provide more uniform velocities, they also provide higher currents on its surface, causing deformations and dispersions that can affect the quality and efficiency of the turbine, as presented in the study by [34].

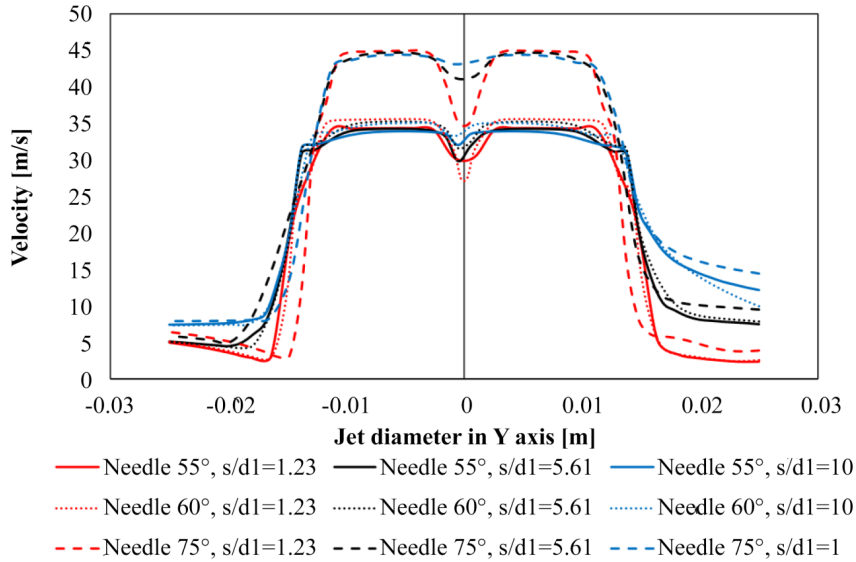


Fig. 12. Injector velocity profile for 55°, 60° Techydro and 75°

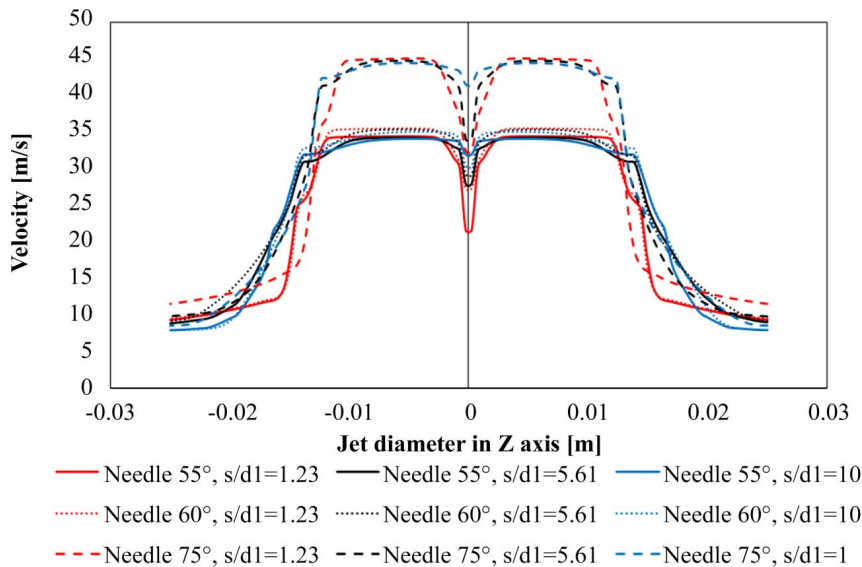


Fig. 13. Injector velocity profile in Z axis for 55°, 60° Techydro and 75°

3. 2. 2. Secondary currents

The Secondary currents are formed mainly by the interaction of the flow with respect to the elements inside the injector, producing small Dean's vortices that move towards the free surface of the jet, significantly affecting the quality of the jet around deformations and dispersions on its surface. Because of this, secondary currents have been the subject of study in recent years in order to improve the quality of the jets in the performance of the turbine. This is the case of [7] and [32] who by means of experimental methods of visualization and measurement LDA (Laser Doppler Anemometry) or computational CFD methods observed the dispersions and deformations of the jets produced by the secondary currents and the release of water drops in their trajectory.

Fig. 14 shows the secondary currents traced in several transversal sections at normalized distances of s/d_1 of 1.23, 5.61 and 10 for each case (55° , 60° Techydro and 75°), which location is shown in **Fig. 7**.

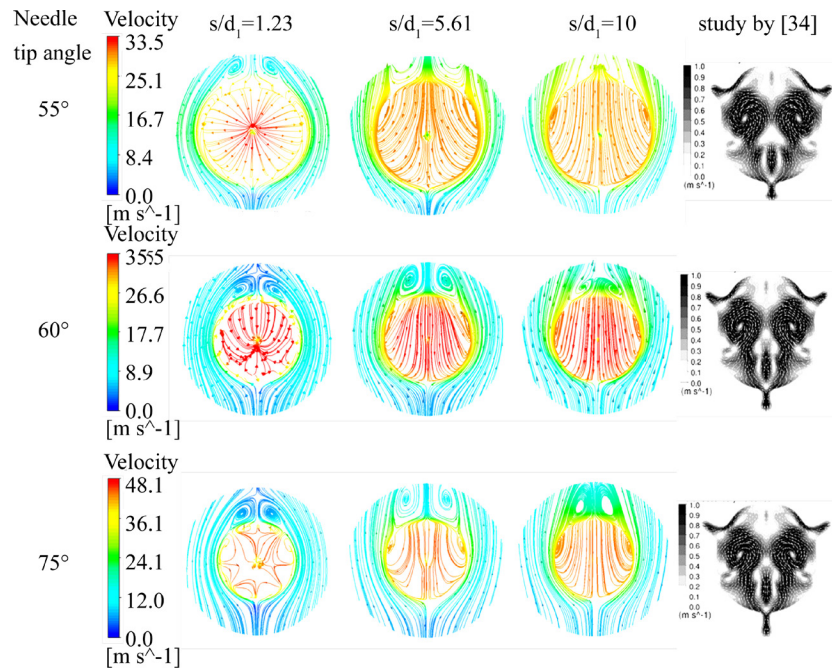


Fig. 14. Secondary current in the jet

The results show some disturbances caused by the Dean vortices due to the union of two currents that create stagnation zones at the top of the water jet that affect its quality. Consequently, since the jet is a free surface and does not have a retaining wall, the vortices increase the release of water drops that cause the dispersion and deformation of the jet, as shown in **Fig. 8, 9**. The first showed deformations in the jet transversal section for the three cases (55° , 60° Techydro and 75°) in the needle tip angle due to secondary currents escaping from its surface. The second showed that the velocity profiles are not completely uniform, and therefore have slight displacements in the core and the edge of the jet. On the other hand, these vortices cause the deflection of the jet and small irregularities that are observed in the upper part of the jet, where they grow significantly, especially for the 75° angle (**Fig. 14**) where a complete detachment of the drop can produce. In contrast to the study carried out by [34], it can be seen that the more pronounced angles present higher secondary currents, resulting in the formation of a sequence of drops that increase the dispersion and deformation of the jet.

3. 2. 3. Injector efficiency

Fig. 15 shows the efficiency of the injector (55° , 60° Techydro and 75°) from the volumetric flow rate normalized Q/Q_{TH} as a function of (s/d_1) .

Recent studies have estimated that the presence of more pronounced angles tends to increase the velocity of the water jet and consequently its efficiency [16]. Thus, results were compared with respect to two references in the literature [22, 32] for 100 % injector needle opening in relation to each configured case. The first shows an efficiency of 95.8 % to 100 % of opening, presenting a difference below 1 % and 1.7 % for the 55° and 60° injector respectively, and above 0.21 % for the 75° injector. The second shows an efficiency of 96.2 %, establishing a difference below 2.18 %, 1.03 % and 0.2 % for 55° , 60° and 75° respectively. Additionally, it is possible to observe an efficiency of 95.17 % for the 60° commercial injector which presents over 1.15 % with respect to the proposed 55° injector and an efficiency below 0.83 % with respect to the 75° injector.

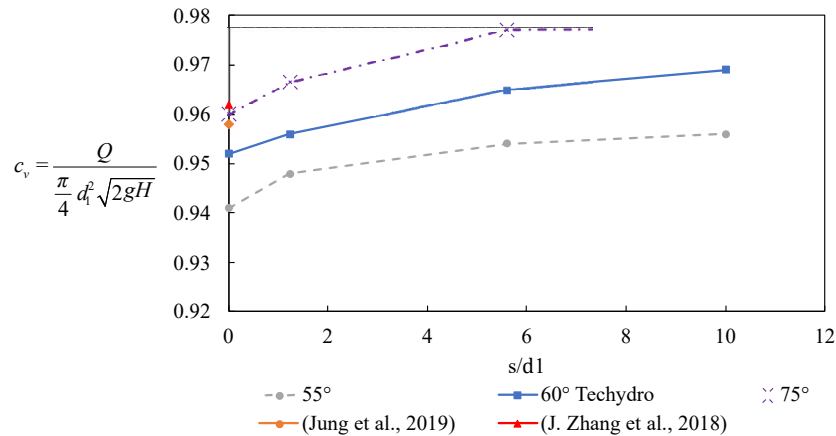


Fig. 15. Volumetric flow efficiency of the injector as a function of the normalized longitudinal distance of the nozzle diameter

On the other hand, it can be observed that in the three curves (55°, 60° and 75°) the efficiency increases in proportion to the increase in s/d_1 . However, between the normalized distance s/d_1 of 5.61 and 10 the increase in efficiency is not very significant for the injector with needle tip angle of 55°, 60° and 75°, where its efficiency only increases by 0.2 %, 0.4 % and 0.02 % respectively. This indicates that the area of the jet increases and the speed of the jet decreases gradually, obtaining a dispersion effect in the jet, as observed in **Fig. 9**.

4. Conclusions

The study presented and analyzed by means of CFD computational simulations the variation of the needle tip angle on the jet characteristics and the efficiency of the injector. This is based on a methodology developed through the generation of a commercial Techydro® geometric model with 60° angle and two geometries designed with 55° and 75° angles at the needle tip. Also, they were analyzed using the multiphase VOF (volume of fluid) method and a turbulence model $k-\varepsilon$, obtaining computational results that affect the quality of the free surface of the water jet and the efficiency of the injector described in studies conducted by authors in the literature. Consequently, the discretization of the control volume influences the velocity simulation results according to the number of elements and nodes set. The efficiency and quality of the jets have a direct relationship with the variation in the injector needle tip angles. These angle configurations present Dean vortices that cause an increase in the deformation and dispersion of the water jet. the influence of the needle tip angle showed significant variations in efficiency, obtaining results of 96 % for a 75°/110° nozzle outlet configuration, showing a variation of 0.83 % and -1.23 % in efficiency respectively to the commercial needle tip angle geometry of 55°/110°, respectively.

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