

Blade model comparison based on static and modal test scenarios

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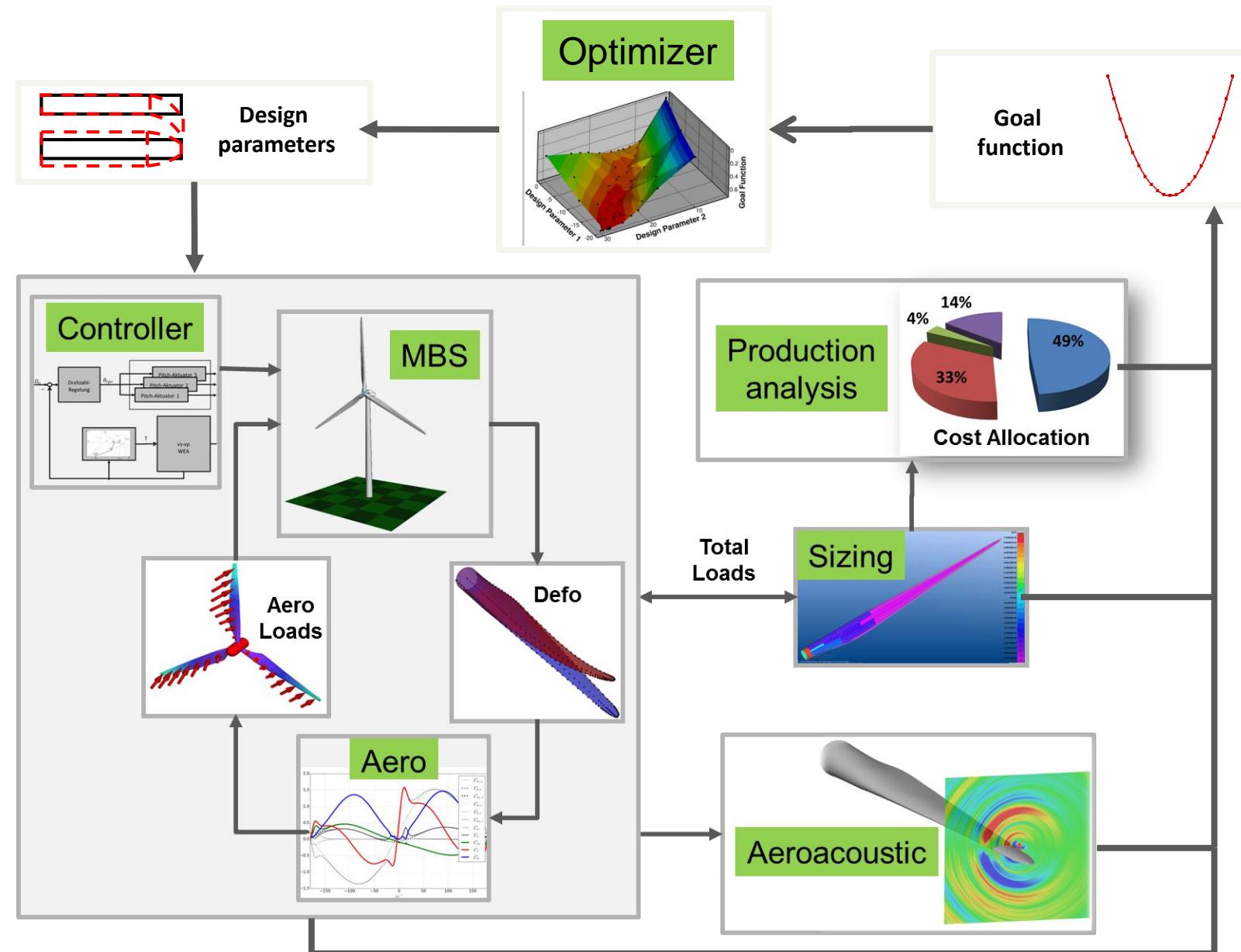
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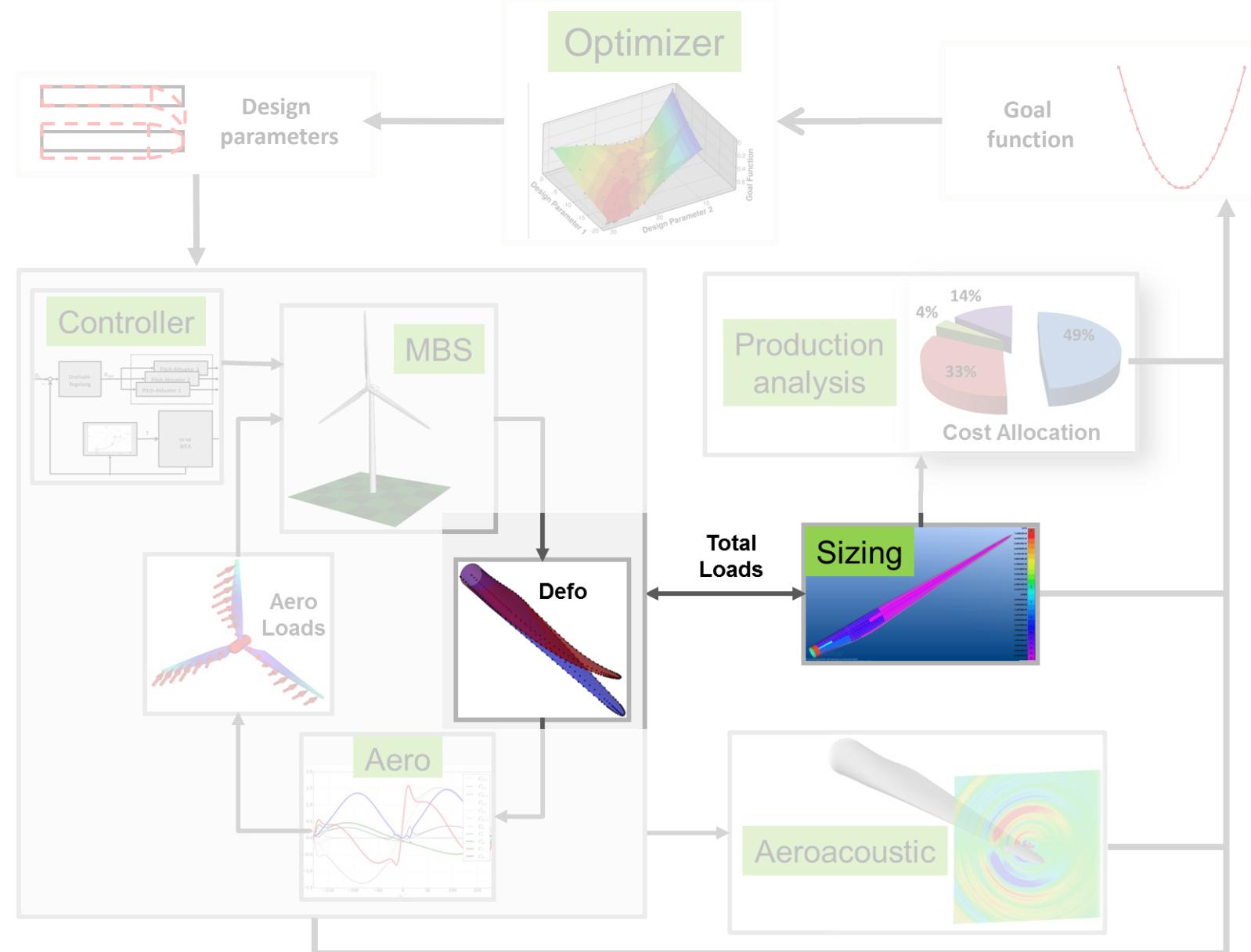
**Institut of Aeroleasticity



Design Process

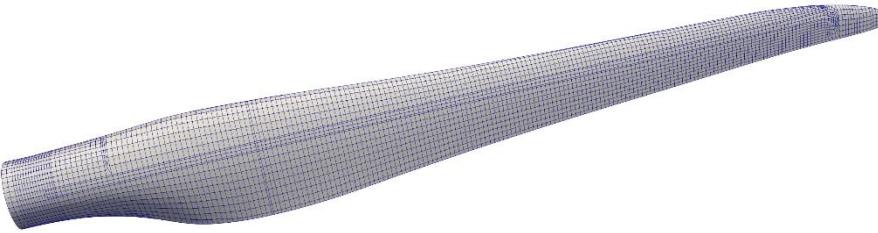


Design Process



Reference rotorblade

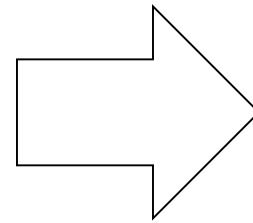
- Size: 20m
- Modelled with quadratic finite shell and solid elements
- Validated against an experimental data set



Reference model: <http://doi.org/10.5281/zenodo.3628356>

Reference rotorblade

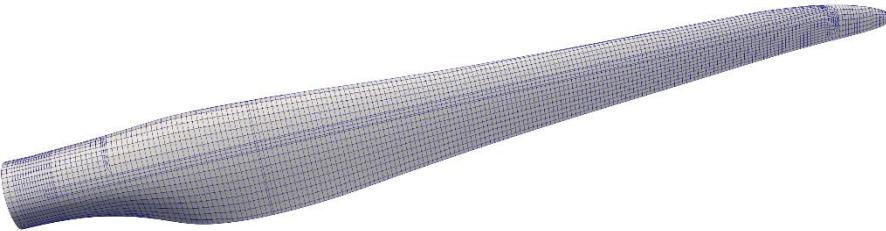
- Size: 20m
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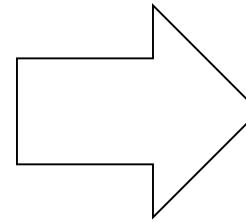
CPACS - xml schema



Reference model: <http://doi.org/10.5281/zenodo.3628356>

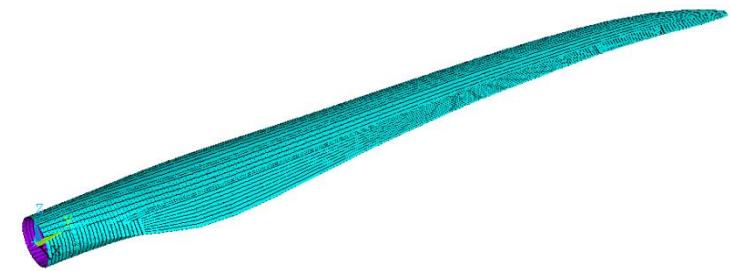
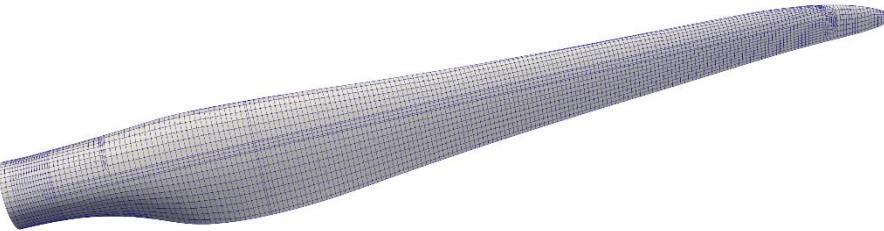
Reference rotorblade

- Size: 20m
- Modelled with quadratic finite shell and solid elements
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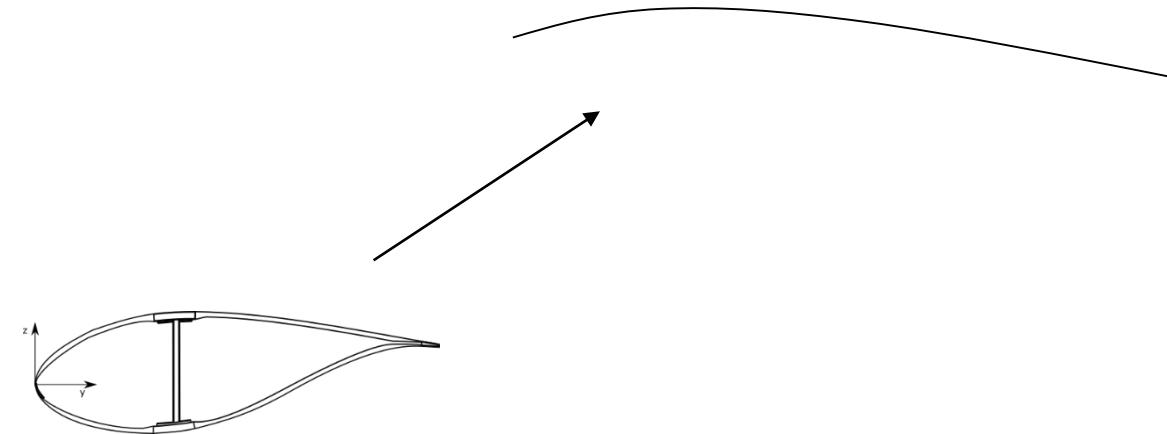
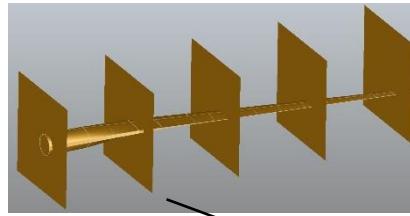
CPACS - xml schema



DELiS model based on CPACS

Reference model: <http://doi.org/10.5281/zenodo.3628356>

Beam model



<https://docs.abdbeam.org/en/latest/>

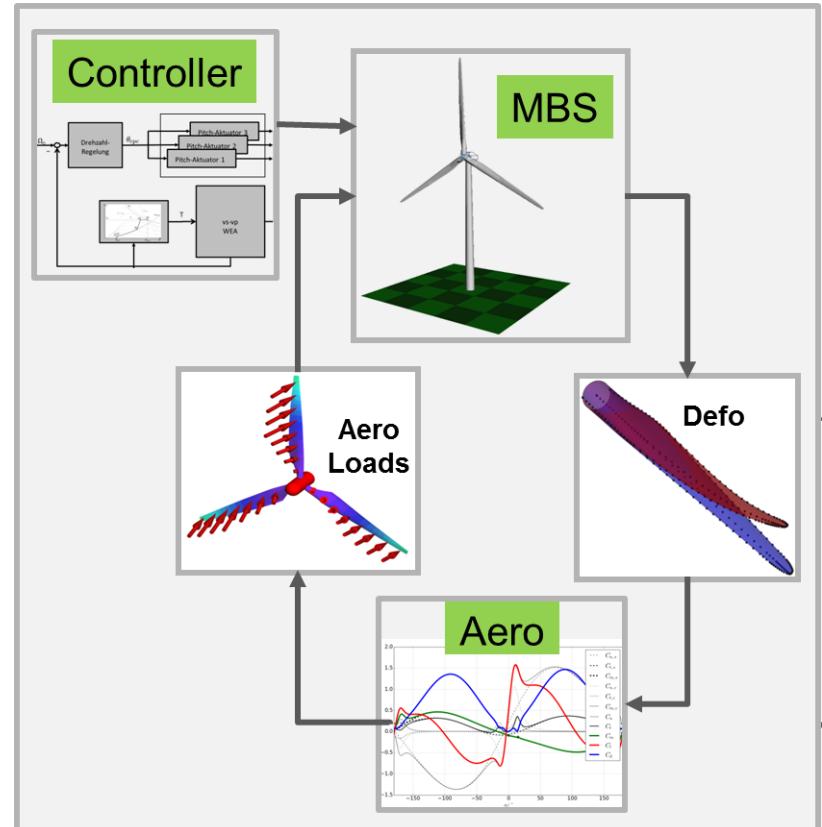
Beam model

Ansys

$$\begin{bmatrix} m & 0 & 0 & 0 & mG_z & -mG_y \\ 0 & m & 0 & -mG_z & 0 & mG_x \\ 0 & 0 & m & mG_y & -mG_x & 0 \\ 0 & -mG_z & mG_y & J_{oxz} & J_{oxy} & J_{oxz} \\ mG_z & 0 & -mG_x & J_{oxy} & J_{oyy} & J_{oyz} \\ -mG_y & mG_x & 0 & J_{oxz} & J_{oyz} & J_{ozz} \end{bmatrix}$$

ABD beam

$$\begin{bmatrix} EA & -z_cEA & y_cEA & 0 & 0 & 0 \\ -z_cEA & EI_{yy} + z_c^2EA & -EI_{yz} - y_cz_cEA & 0 & 0 & 0 \\ y_cEA & -EI_{yz} - y_cz_cEA & EI_{zz} + y_c^2EA & 0 & 0 & 0 \\ 0 & 0 & 0 & GJ + y_s^2GAsc_z + z_s^2GAsc_y & 0 & 0 \\ 0 & 0 & 0 & 0 & GAsc_z & 0 \\ 0 & 0 & 0 & 0 & 0 & GAsc_y \end{bmatrix}$$



<https://docs.abdbeam.org/en/latest/>

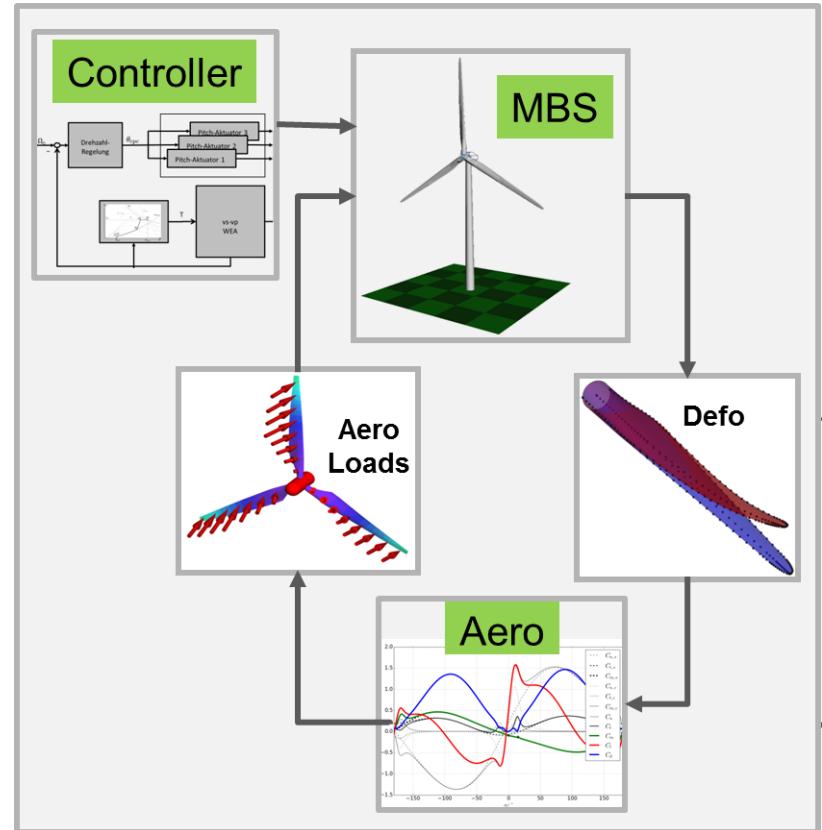
Beam model

Ansys

$$\begin{bmatrix} m & 0 & 0 & 0 & mG_z & -mG_y \\ 0 & m & 0 & -mG_z & 0 & mG_x \\ 0 & 0 & m & mG_y & -mG_x & 0 \\ 0 & -mG_z & mG_y & J_{oxz} & J_{oxy} & J_{oxz} \\ mG_z & 0 & -mG_x & J_{oxy} & J_{oyy} & J_{oyz} \\ -mG_y & mG_x & 0 & J_{oxz} & J_{oyz} & J_{ozz} \end{bmatrix}$$

ABD beam

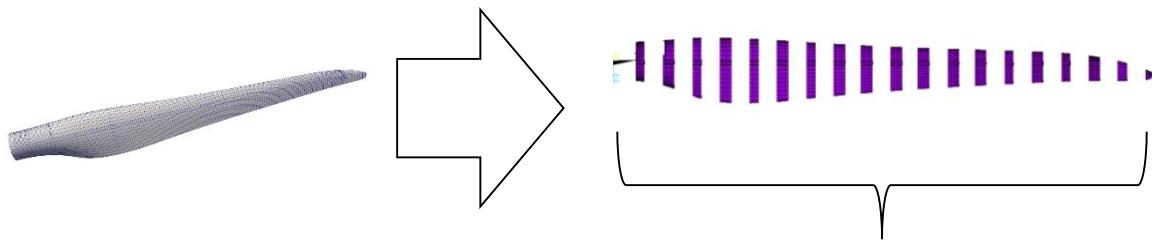
$$\begin{bmatrix} EA & -z_cEA & y_cEA & 0 & 0 & 0 \\ -z_cEA & EI_{yy} + z_c^2EA & -EI_{yz} - y_c z_c EA & 0 & 0 & 0 \\ y_cEA & -EI_{yz} - y_c z_c EA & EI_{zz} + y_c^2EA & 0 & 0 & 0 \\ 0 & 0 & 0 & GJ + y_s^2GA_{scz} + z_s^2GA_{scy} & 0 & 0 \\ 0 & 0 & 0 & 0 & GA_{scz} & 0 \\ 0 & 0 & 0 & 0 & 0 & GA_{scy} \end{bmatrix}$$



Modal reduction



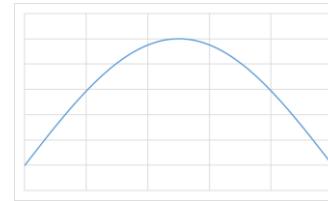
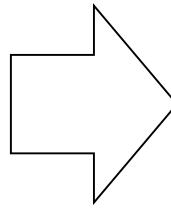
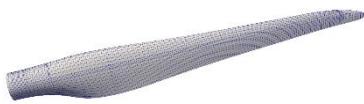
Modal reduction



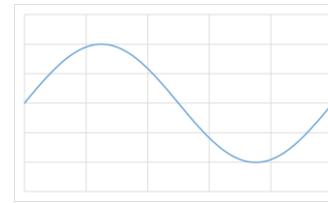
- Condensation onto 20 nodes at aerodynamic centers
- Inclusion of 10 Component-Mode-Synthesis (CMS) Modes



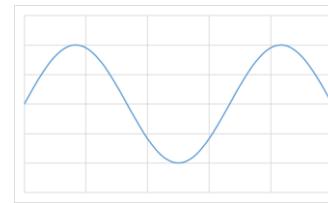
Modal reduction



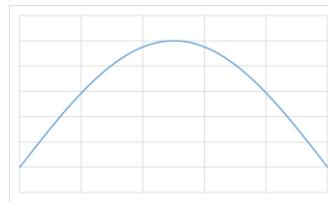
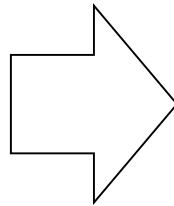
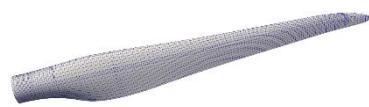
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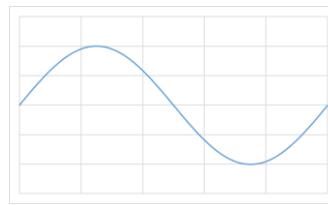
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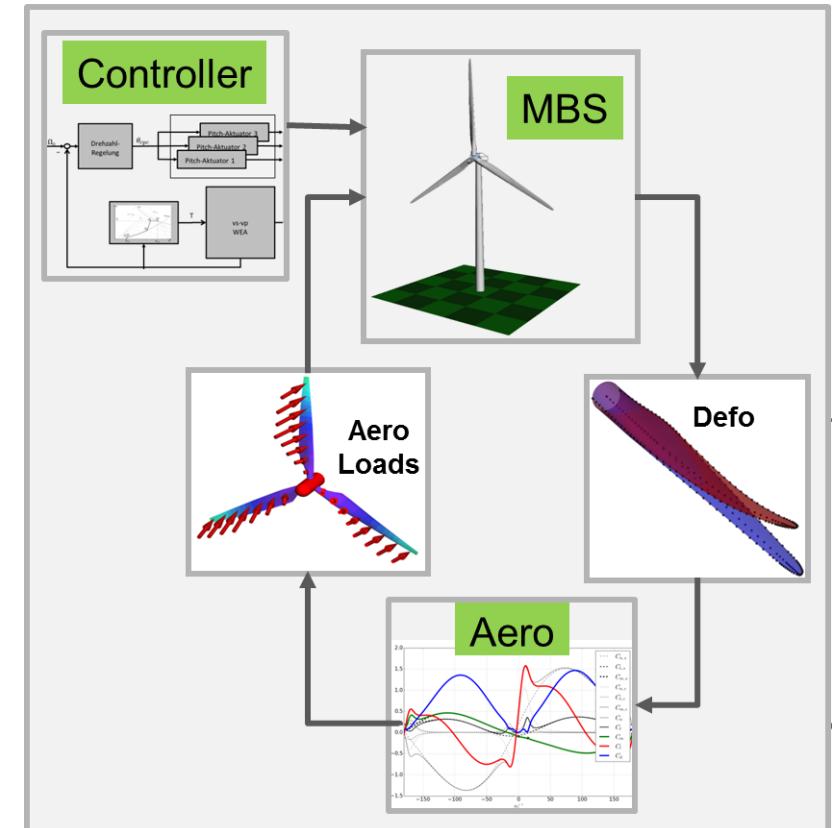
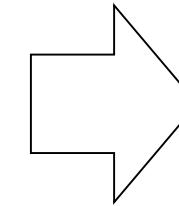
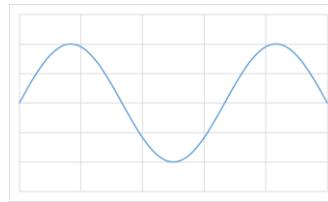
Modal reduction



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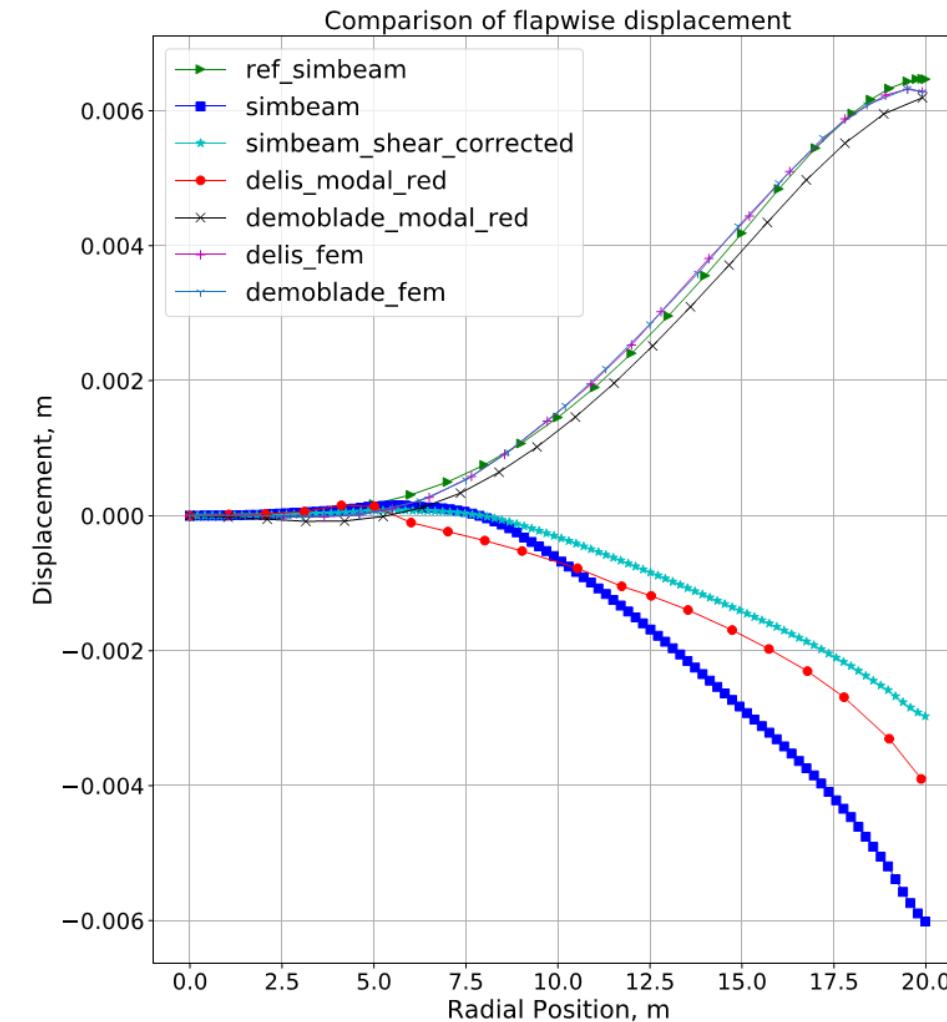
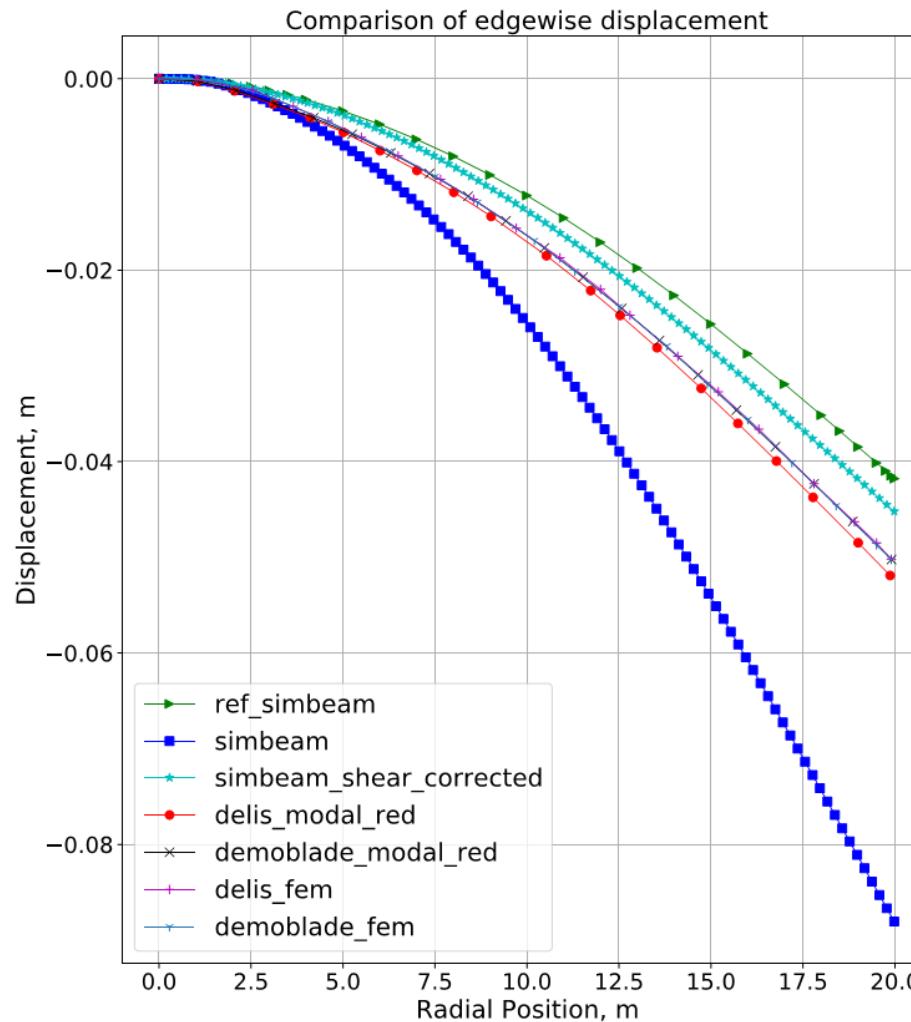
Models

| Name in Plots | Model | Method |
|-------------------------|--|---------------------|
| Ref_simbeam | Reference loads model | MBS beam |
| Simbeam | DELiS model as beam model | MBS beam |
| Simbeam_shear_corrected | DELiS model as beam model + shear correction from ref_simbeam | MBS beam |
| Delis_modal_red | DELiS model | MBS modal reduction |
| Demoblade_modal_red | Reference model | MBS modal reduction |
| Delis_fem | DELiS model | FEM |
| Demoblade_fem | Reference model | FEM |



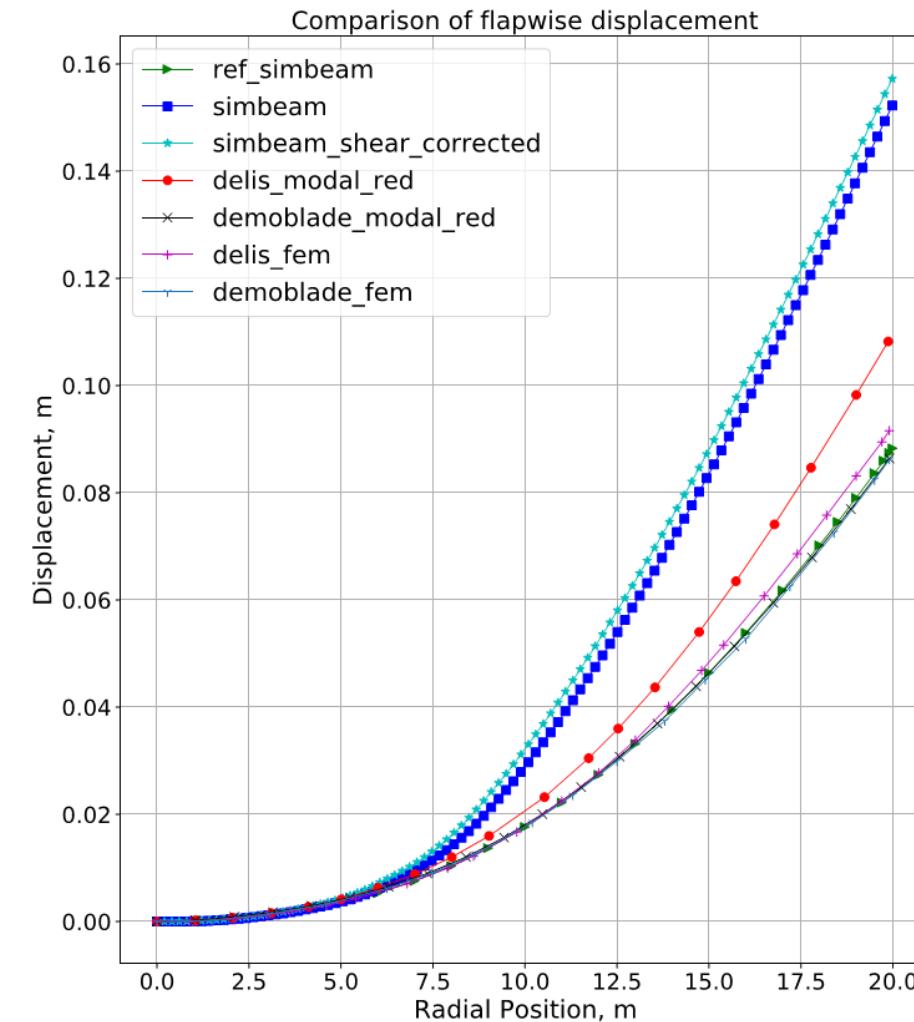
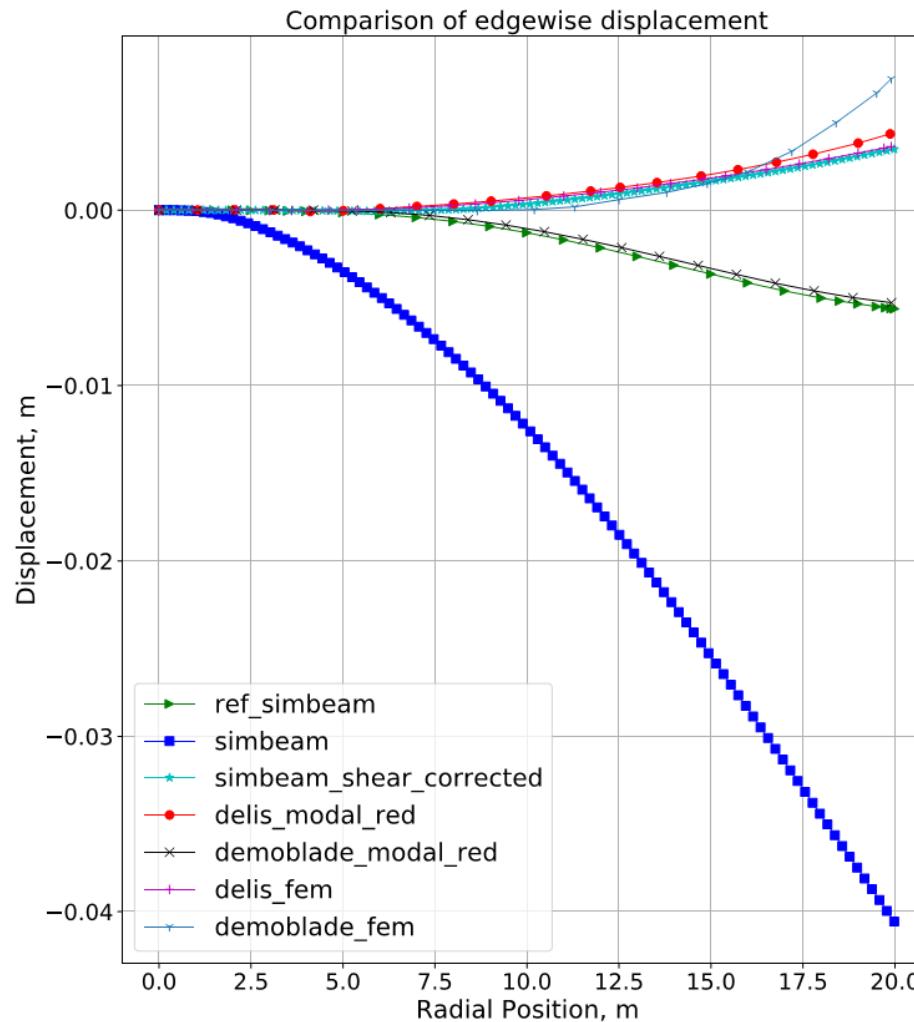
Edgewise 1g

Edgewise 1g load, 0° pitch, 90° azimuth

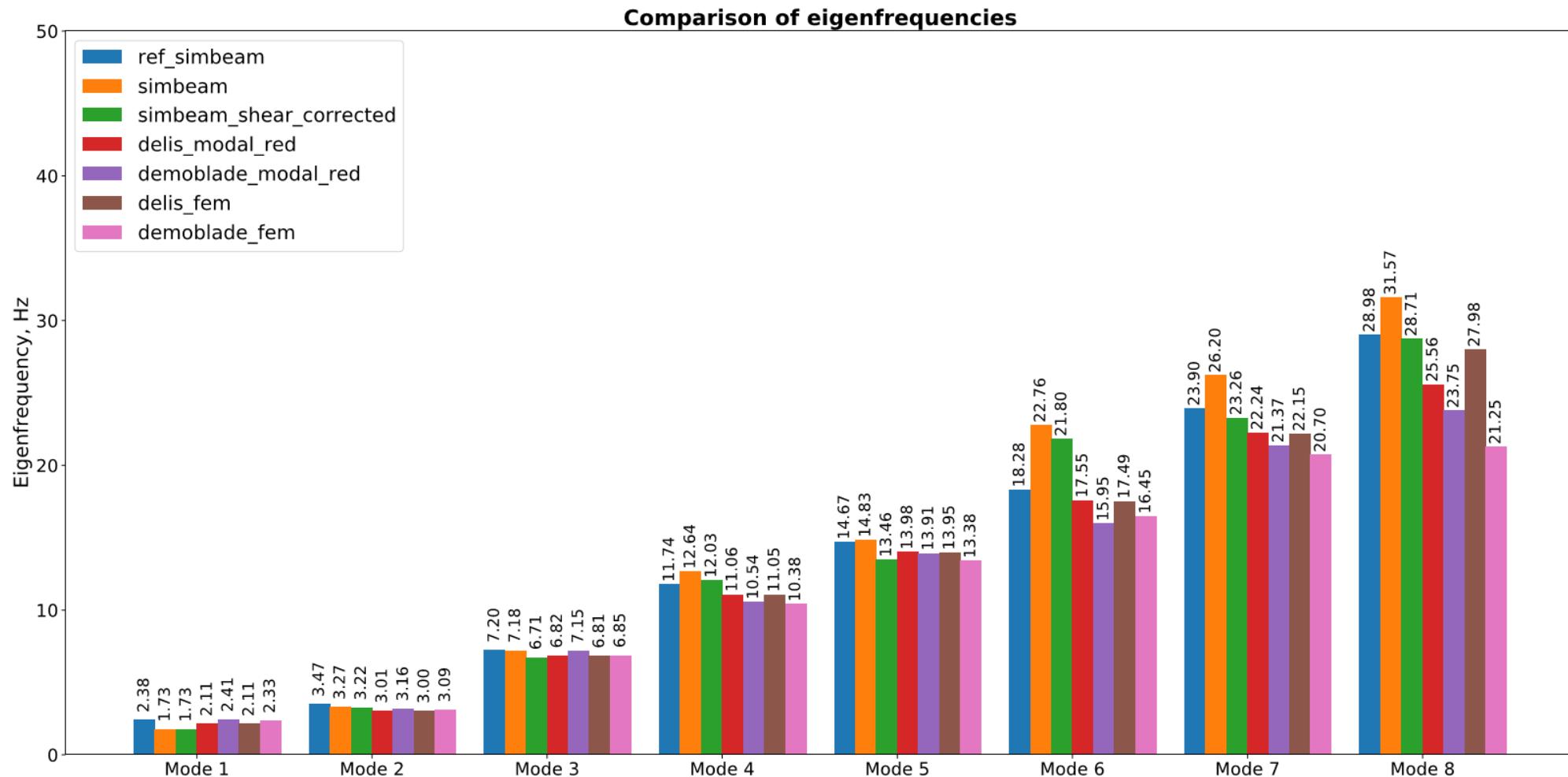


Flapwise 1g

Flapwise 1g load, 90° pitch, 90° azimuth



Eigenfrequencies



Conclusion

- Modal reduction is a good way of modelling and allows to obtain accurate beam models
 - It circumvents the struggle of finding appropriate shear correction factors
 - ABD python package needs more tests for wind turbine blades
-
- Determine shear correction factors and beam parameters with optimization or fitting



Thank you!

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Web: leichtbau.dlr.de

Reference model can be found here

Christian Willberg. (2021). Smartblades 2 finite element
reference wind turbine blade model

<http://doi.org/10.5281/zenodo.3628356>

