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LES and RANS of air and oxy-coal combustion in a pilot-scale facility: predictions of radiative heat transfer

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Abstract

This study evaluates the use of large eddy simulation (LES) and Reynolds-averaged Navier Stokes (RANS) models for the prediction of turbulent coal combustion under air and oxyfuel environments in a pilot-scale 250 kW_{th} furnace. The furnace is part of the UKCCSRC Pilot-scale Advanced Capture Technology (PACT) facilities and was designed for detailed analysis of the combustion process. The prediction of thermal radiation is validated against experimental measurements under both air- and oxy-firing regimes. Two radiation models were evaluated during the RANS calculations, the widely used weighted sum of grey gases (WSGG) and the full-spectrum correlated k (FSCK) model, while the LES case was calculated using the FSCK radiation model. The results show that the choice in gas radiation model demonstrates only a small change in the temperature and heat flux predictions in the RANS calculations, while the LES solutions are able to achieve better agreement with measured values than the RANS predictions for both air-fired and oxyfuel coal combustion.

Keywords: Large eddy simulation, oxyfuel, computational fluid dynamics, radiation heat transfer

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1. Introduction

The international community is committed to preventing the rise of temperature attributable to anthropogenic climate forcing through the reduction of greenhouse gas (GHG) emissions. Nations have implemented targets to reduce their GHG emissions compared to baseline levels recorded in 1990, with the UK has committing to a 34% reduction in GHG emissions by 2020, which rises to an 80% reduction by 2050. The energy sector will be required to greatly curb its GHG emissions to realise these targets, however with the rising global population, and the industrialisation of developing countries, fossil fuels are still expected to be utilised.

Coal in particular is expected to remain an important global energy resource due to its widespread availability and operating flexibility, however coal-fired combustion is one of the largest global sources of CO₂ emissions [1]. It is necessary to develop carbon capture and storage (CCS) technology so that the benefits of coal-fired energy generation can be realised without violating efforts to reduce CO₂ emissions.

This study focusses on oxyfuel technology for carbon capture. The oxyfuel process for a thermal power station involves firing combustible fuel with a high-purity oxygen stream, which is often diluted with recycled flue gas to control flame temperature and heat transfer. The resulting flue gas from the oxyfuel process contains a high concentration of CO₂ that can be economically purified to a level suitable for transport and storage [2]. Oxyfuel combustion has been demonstrated at small and medium scales [3–5], and is being developed for large scale projects, such as the White Rose CCS¹ and FutureGen 2.0² projects.

Oxyfuel technology can be retrofitted to existing combustion facilities, however, with such significant changes to the combustion environment, it is important to develop an understanding of the influence that switching to oxyfuel will have over heat transfer, chemical reactions and flame stability. Furthermore, the

¹<http://www.whiteroseccs.co.uk/>

²<http://futuregenalliance.org/futuregen-2-0-project/>

29 control over the oxygen concentration in the recycled flue gas will provide an
30 additional parameter with regards to combustion efficiency and material corro-
31 sion control to optimise against the cost of the oxygen supply, as well as offering
32 further benefits with regards to fuel flexibility [6].

33 It will be beneficial in the design and optimisation of oxyfuel combustion
34 to be able to predict the influence of operating parameters on the combustion
35 performance. Under oxyfuel, the increase in the concentration of radiatively
36 participating species, namely CO_2 and H_2O , significantly modifies the transfer
37 of thermal radiation [7]. Modelling techniques, such as Computational fluid dy-
38 namics (CFD), have been used to predict air-fired combustion facilities, however
39 the novelty of the oxyfuel combustion environment poses challenges to models
40 that are often empirically defined for air-firing. Pilot-scale facilities are import-
41 ant to validate CFD models before they can be applied to larger cases as they
42 provide well controlled environments where detailed experimental measurements
43 can be performed.

44 This study presents both experimental measurements and numerical solu-
45 tions for a 250 kW down-fired combustion test facility, which is part of the UK-
46 CCSRC Pilot-scale Advanced Capture Technology (PACT) facilities, operating
47 with both air-fired and oxyfuel coal combustion. The facility was constructed
48 to offer detailed analysis of the combustion process under a range of environ-
49 ments. The measurements of the two combustion modes are used to validate
50 CFD predictions using advanced turbulence and spectral radiation treatment.

51 **2. Combustion test facility**

52 The combustion test facility that is the subject of this study is a vertical
53 down-fired cylindrical furnace, fitted with a scaled 250 kW_{th} burner provided
54 by Doosan Babcock. The burner introduces combustion gases into the furnace
55 through three registers, referred to as the primary, secondary and tertiary, which
56 is illustrated in Figure 2. A central annulus exists for preheating the furnace with
57 a natural gas flame, however this annulus was not used during the measurements.

58 The coal is transported into the furnace through the primary annulus, with the
59 majority of the combustion oxidant supplied through the secondary and tertiary
60 annuli. The three inlets are swirled with blades fitted into the burner to stabilise
61 the flame and increase the turbulent mixing of the oxidant and fuel.

62 The cylindrical furnace has an inner diameter of 0.9 m and is 4 m high
63 and is illustrated in Figure 1. The facility is comprised of eight sections that
64 are lined with a 0.1 m thick refractory. The facility was designed to allow for
65 detailed measurements and characterisation of the combustion process under
66 a wide range of operating conditions, and has numerous measurement ports
67 located down the length of the furnace. Each section is 0.5 m high, with the
68 first six sections being water cooled. The top two sections of the furnace contain
69 a number of ports for intrusive and non-intrusive flame measurements. The
70 furnace is maintained at sub-atmospheric pressure by an exhaust fan to ensure
71 safe operation. The same batch of El-Cerrejon coal was fired during the air and
72 oxyfuel combustion measurements in this study. The calorific, proximate and
73 ultimate analyses of the coal are shown in Table 1.

74 The operating conditions for the air and oxyfuel cases are detailed in Table 2.
75 Both cases were run with the same 200 kW thermal load with the same exit O_2
76 concentration, measured at 3.3% (dry vol.). The oxyfuel case was fired using
77 an overall 27% (vol.) O_2 concentration, with a balance of CO_2 . The O_2 and
78 CO_2 in the oxyfuel case were supplied from liquid storage tanks. The secondary
79 and tertiary gases are preheated using electrical heaters to achieve temperatures
80 that are comparable to values used for utility boilers. The oxygen concentration
81 of the primary gas, which transports the coal, was reduced in the oxyfuel test
82 case to ensure safe operation. The oxygen concentration was enriched in the
83 secondary and tertiary registers to achieve the overall 27% (vol) concentration
84 delivered to the furnace.

85 Heat transfer to the walls was measured using a Medtherm heat flux trans-
86 ducer. The measurement probe uses a Schmidt-Boelter type sensor with a ther-
87 mopile fitted at its tip. While exposed to the combustion gases, the device
88 measures the total heat transfer to the wall. The sensor was shielded from

89 convective heat transfer by applying a purge gas of N_2 from the outer circum-
90 ference of the probe tip at an inward angle to block any combustion gasses from
91 reaching the sensor.

92 Suction pyrometry was used during the air-fired campaign to measure the
93 in-flame gas temperatures. The suction pyrometer consists of a thermocouple
94 surrounded by a radiation shield. The probe draws the sample gas in at high ve-
95 locities to intensify the effect of convection and negate the temperature measure-
96 ment error associated with radiative heat exchange between the thermocouple
97 and its surroundings. Measurements were made across a single radius of the
98 furnace at a time, with the probe being reinserted for different axial locations
99 along the length of the furnace to build up a profile of the gas temperature.

100 **3. Computational modelling**

101 The combustion test facility was modelled using the commercial CFD pack-
102 age ANSYS Fluent version 15. Six cases are considered in total, three for
103 both air and oxyfuel combustion. The three cases consist of two Reynolds-
104 averaged Navier Stokes (RANS) solutions and one large eddy simulation (LES).
105 The RANS solutions are generated using two different models for the radiative
106 properties of the combustion gases; the widely used grey weighted sum of grey
107 gases (WSGG) model, and a more advanced full-spectrum correlated k (FSCK)
108 model, which has been shown to perform well under oxyfuel combustion [8, 9].
109 The LES for both the air and oxyfuel campaigns are run using the FSCK model.

110 *3.1. Turbulence*

111 RANS models are the most widely employed turbulence treatment due to
112 their relatively low computational cost. Under a steady RANS prediction, trans-
113 port equations are solved for time-averaged values to calculate the steady-state
114 condition of a system. While extremely useful in predicting flow phenomena,
115 RANS calculations require models to predict all of the scales of turbulence,
116 which can be dependent on the specific geometry and are therefore not easily
117 specified for generic flow.

118 In contrast to RANS calculations, LES solves the spatially filtered Navier
119 Stokes equations, and numerically resolves the transient flow for the large scales
120 of turbulence. The small scales of turbulence, which can often be assumed to
121 be uniform and isotropic, are modelled. While LES cases often show accurate
122 results for coal combustion [10–13], and allow for the analysis of transient phe-
123 nomena [13–15], the vastly increased resources required to resolve the transient
124 flow is often a barrier for its use.

125 In this study the Launder et al. [16] Reynolds stress model was used for the
126 RANS calculation using the pressure strain term and constants proposed by
127 [17]. Reynolds stress models have often performed well at predicting swirling,
128 confined and reacting turbulent flow, as is present in the current case [18, 19].
129 The WALE sub-grid turbulence model was used for the LES predictions with a
130 time-step of 2×10^{-4} , using a sub-grid turbulent Schmidt number of 0.4, as has
131 been used in other studies [20–22].

132 All of the cases in this study were run on a hexahedral structured mesh
133 with around three million cells. The dependency of the RANS solutions on
134 the grid size was checked using periodic meshes, with the mesh that had the
135 lowest number of cells, while still producing grid-independent solutions, was
136 used to construct the full 3D grid used in this study. The LES in this study
137 uses an implicit filter width, which is determined by the mesh cell size, and is
138 therefore sensitive to the resolution of the grid. Assuming that at least 80%
139 of the turbulent kinetic energy should be resolved to obtain an accurate LES,
140 a filter width of one twelfth of the characteristic length scale of the energy
141 containing eddies, \mathcal{L} , is required [23]. This length scale was estimated from
142 the RANS solutions, using $\mathcal{L} = k^{1.5}/\epsilon$, where k and ϵ are the turbulent kinetic
143 energy and dissipation rate respectively, and the filter width was calculated as
144 $\Delta_w = \sqrt[3]{v_{cell}}$, where v_{cell} is the cell volume. The quality of the grid for LES was
145 evaluated by analysing $\Delta_w/(\mathcal{L}/12)$, and highlighting regions where this value
146 exceeded a ratio of one. This criterion can be too relaxed, and it is often possible
147 to achieve more accurate simulations with further refinements in the grid and
148 filter width. This criterion was satisfied in the majority of the domain, however

149 the grid was not sufficiently resolved in the near burner or the near wall regions,
150 as can be shown in Figure 3. The grid was used despite this deficiency due to
151 the limitations of resources. Wengle and Werner [24] wall functions were also
152 used for the LES case and enhanced wall treatment was used for the RANS
153 calculations so that the mesh did not have to be resolved through the boundary
154 layer. Second order upwind schemes were used for the discretisation of the
155 convective terms in the RANS solutions, while a bounded central differencing
156 scheme was used for the LES, and the solution was advanced in time by the use
157 of an implicit second order scheme. Transient effects at the inlets were neglected
158 in the LES.

159 *3.2. Radiation heat transfer*

160 Calculating radiative heat transfer is challenging due to the spatial, angular
161 and spectral variation in the radiative intensity field. Due to the dominance of
162 thermal radiation at combustion temperatures, separate models that account
163 for the spectral variation in radiative transfer are compared in this study; the
164 grey WSGG method, which is provided by default in Fluent, and the FSCK
165 model, which has been implemented with user-defined functions. This study
166 uses the finite volume method implemented in Fluent (discrete ordinates) to
167 solve radiative transfer in spatial and angular dimensions, due to its superiority
168 in calculating incident radiation at the boundary of the domain [25]. The model
169 is used with a 3×3 angular discretisation for each octant of the solid angle,
170 resulting in 72 ordinates for each control volume. A 4×4 discretisation was
171 tested for the RANS calculation with the grey WSGG model, and the maximum
172 variations in the temperature and incident radiation predictions were less than
173 2.5% across both the air-fired and oxyfuel cases. The internal emissivity of the
174 refractory-lined walls were assumed to be grey, as mandated by the use of the
175 global models, and were set to a constant value of 0.8.

176 The grey WSGG method calculates an effective gas absorption coefficient
177 based on the weighted sum of emissivity from fictitious grey gases. The absorp-
178 tion coefficients and the weights of the grey gases are fitted to values of emissiv-

179 ity, which are often calculated from band models or high-resolution spectral
180 databases. The values that are built into Fluent are based on the calculations
181 by Smith et al. [26], which were generated for air-fired combustion, and therefore
182 should not be applied to oxyfuel combustion.

183 Unlike the grey WSGG method, the FSCK method is not restricted to any
184 specific environment. The FSCK method calculates radiative intensity based
185 on a reordered absorption coefficient against a normalised spectral dimension
186 [27]. Through this manipulation of the radiation transfer equation (RTE), it is
187 possible to accurately calculate radiative transfer with a small number of discret-
188 isations in the spectral dimension. In this study a five-point Gauss quadrature
189 was used to calculate radiative heat transfer for the FSCK model, as it has been
190 shown to perform well for oxyfuel conditions [9]. While a five-band quadrature
191 is small compared to line-by-line and band models, it still requires a significant
192 increase in the memory and CPU-time requirements of the calculation over the
193 more widely-used grey WSGG method. The FSCK implementation considered
194 gas absorption and emission from CO_2 , H_2O and CO , with the k-distributions
195 themselves being calculated from the narrow-band k-distributions from Cai and
196 Modest [28], using the mixing scheme by Modest and Riazzi [29]. Further details
197 of the FSCK implementation and validation can be found in Clements et al. [9].

198 Turbulent fluctuations in temperature and gas concentrations can differ sig-
199 nificantly from statistically averaged or spatially filtered values. Due to the
200 fourth power relationship between temperature and radiative emission, these
201 turbulent structures significantly increases the amount of radiation emitted from
202 participating gases, as well as also increasing the gas absorptivity through the
203 absorption coefficient's dependence on local thermodynamic properties [30]. The
204 accuracy of the turbulence prediction can significantly influence the calculation
205 of radiative intensity that, through the energy equation, will also modify the
206 fluid dynamics, which is known as turbulence-radiation interaction (TRI). While
207 LES resolves some of the TRI, it is unclear whether it is necessary to further
208 resolve TRI at sub-grid scales [31, 32]. This study did not utilise a sub-grid
209 model for radiation.

210 In addition to the gas participation, fuel, char, soot and ash particles all
211 contribute to radiative transfer. Due to exothermic reactions on the particle
212 surface, char particles are often over 200 K hotter than the surrounding gas
213 [33], and therefore contribute significantly to the emission of radiation. Fly ash
214 can also have a significant effect on radiation emission and scattering [34]. While
215 the sensitivity of the results to the particle radiation properties is acknowledged,
216 this study only used typical spectrally constant values for the particle absorption
217 efficiency of 0.9, and a low particle scattering efficiency of 0.01 to compensate
218 for the use of an isotropic scattering phase function. Grey particle emission in
219 the FSCK model was included for each band by scaling the radiative source
220 by the emissivity weight function evaluated at the particle temperature, while
221 grey particle absorption was added to the local k-distribution values. Due to
222 their high emissivity, coal-derived soot particles were also accounted for using
223 the model by Brown and Fletcher [35]. Soot radiative properties were treated
224 with the default treatment for the WSGG model in Fluent, but non-grey soot
225 participation was included in the FSCK model using the correlations by Chang
226 and Charalampopoulos [36] to calculate the soot absorption coefficient at the
227 narrow-band centres when constructing the full-spectrum k-distributions.

228 *3.3. Particle combustion*

229 Coal particles are tracked within a Lagrangian frame, and are coupled to the
230 Eulerian fluid phase through appropriate source terms. Turbulent dispersion
231 of the particles in the RANS cases were modelled using the discrete random
232 walk model that is available in Fluent, which tracks the same physical particle
233 numerous times while stochastically perturbing the particle's velocity based on
234 the local turbulent kinetic energy of the fluid domain. Unsteady particles were
235 tracked with the fluid in the LES without any stochastic variations, with the
236 assumption that the sub-grid scales did not influence the particle motion. Unlike
237 in the fluid phase, particle temperatures are not averaged during the tracking,
238 and peaks in temperatures will be correctly accounted for in the particle emission
239 terms for both the RANS and LES cases.

240 The combustion of a coal particle is modelled as a series of contiguous steps;
 241 inert heating, drying, devolatilisation, heterogeneous char combustion and inert
 242 heating/cooling of resultant ash particles. The process of devolatilisation and
 243 char combustion is expected to differ between air and oxyfuel combustion [37],
 244 however, in the absence of any empirically derived rates for the precise com-
 245 bustion conditions being modelled in this study, the same combustion model
 246 parameters were used for both the air and oxyfuel cases.

247 Coal volatiles are modelled as an empirically defined species, derived from
 248 the proximate and ultimate analysis of the coal, assuming that the volatile yield
 249 at high temperatures is 1.57 times greater than the value measured in the prox-
 250 imate analysis. The volatile evolution from the coal is modelled using a single
 251 Arrhenius rate. Char combustion is modelled using the intrinsic model [38],
 252 with the char combustion products being treated as CO. The parameters for
 253 the devolatilisation and char combustion models were obtained from Pranzitelli
 254 et al. [39].

255 3.4. Homogeneous combustion

256 The gas-phase combustion of volatile matter and CO released from char
 257 combustion was modelled using the eddy-dissipation model [40], which assumes
 258 that the rate of combustion is only limited by the turbulent mixing of reactants.
 259 The eddy dissipation model calculates the net production rate of a species due to
 260 a reaction r as the minimum of the reactant dissipation, $R_{R,r}$, and the dissipation
 261 of product species, $R_{P,r}$, which are calculated as

$$R_{R,r} = \nu_{r,i} M_{w,i} A \rho \frac{1}{\tau} \min \left(\frac{Y_R}{\nu_{i,R} M_{w,R}} \right) \quad (1)$$

$$R_{P,r} = \nu_{r,i} M_{w,i} A B \rho \frac{1}{\tau} \frac{\sum Y_P}{\sum_j^{N_r} \nu_{r,j} M_{w,j}} \quad (2)$$

262 Where R and P denote reactant and product species respectively, $\nu_{r,i}$ is the
 263 stoichiometric coefficient of species i in reaction r , $M_{w,i}$ is the molecular weight
 264 of species i , ρ is the gas density, τ is the eddy mixing time scale, N_r is the number

265 of reactions, Y denotes mass fraction and A and B are constants. The eddy
266 mixing time scale is taken as k/ϵ in the RANS calculations, and is calculated
267 as the reciprocal of the strain rate for the LES cases. A two-step reaction
268 mechanism was used, where volatile gas species are first oxidised to CO, H₂O,
269 N₂ and SO₂, and the CO is further oxidised to CO₂. The mixing rate parameters
270 for the eddy-dissipation model were taken from values recommended for swirling,
271 confined coal flames [41], using the same values for the RANS and LES cases,
272 where A is set to 0.5 and 0.7 for volatile and CO combustion respectively, and
273 B is set to 0.5.

274 4. Results and discussion

275 All of the CFD calculations were run using 64 CPU cores, and took 2 days, 3
276 days and 30 days to complete for the RANS calculation with the WSGG model,
277 RANS calculation with the FSCK model, and the LES cases respectively. Each
278 LES case was run to compute four seconds of simulation time before statistics
279 were initialised, to account for the residence time of the gas within the measured
280 domain, and were run for a further one second while gathering time-averaged
281 temperature, heat flux and exit gas composition data, until statistical conver-
282 gence. The LES cases contained roughly eight million particles when the domain
283 was filled.

284 Figures of the temperature distribution for the RANS cases, as well as in-
285 stantaneous and time-averaged LES results, can be seen in Figures 4 and 5
286 for the air and oxyfuel case respectively. The instantaneous temperature dis-
287 tributions reveal the resolution of turbulent structures with regions of higher
288 temperatures compared to the mean flow field. The time-averaged LES results
289 show a much smoother temperature distribution than the RANS predictions,
290 with a narrower flame that is rooted inside the quarl.

291 Figure 6 plots the radial distribution of temperature near to the burner for
292 the air-fired case, comparing the CFD results to suction pyrometry measure-
293 ments. The plots also compare the predictions against a RANS case without

294 calculating radiative heat transfer, which shows that radiation is responsible for
295 over 400 K difference in the gas temperature, however, as can be seen in the
296 temperature distributions as well (Figures 4 and 5), there is very little difference
297 in the temperature predictions between the two radiation models for the RANS
298 cases. In all three cases, the predicted temperature shows the most deviation
299 from the measured data close to the burner, around 0.15 m from the centre of
300 the furnace. The CFD calculations predict a low rate of mixing between the in-
301 let streams and the combustion gases, resulting in a significant under-prediction
302 of the temperature near the burner. The time-averaged velocity predictions for
303 the air-fired case in the near-burner region, shown in Figure 7, show that the
304 RANS predictions are very similar, however the LES shows greater variation
305 across the radial direction in the external recirculation zone. A similar trend
306 is also visible in the distribution of participating species, Figure 8, where the
307 LES calculation produces much smoother profiles, while the RANS predictions
308 are similar to each other. The LES calculation shows a much smoother vari-
309 ation in the temperature profile, with a higher minimum value, however, there
310 is still a deviation from the experimental data. This near-burner region of the
311 burner has been identified as being likely to be under-resolved, which is caused
312 by the high velocities of the oxidiser streams. It is expected that reducing the
313 cell size in this region, and therefore resolving the smaller length scales of tur-
314 bulence, will improve the predictions of turbulent mixing, however the RANS
315 simulations, which have been tested for grid dependency, will remain the same.
316 Further downstream of the burner, past 575 mm from the exit of the quarl, the
317 temperature measurements and predictions show a reasonably uniform profile,
318 with the LES and RANS calculations producing similar temperatures.

319 The temperature profiles for the oxy-27 case, Figure 9, show similar trends
320 to the air-fired case; the RANS predictions are very similar in their temperature
321 distribution, with the LES case predicting a much smoother profile. As with
322 the air-fired results, the RANS calculations without accounting for radiative
323 transfer increases the gas temperature predictions by roughly 400 K, further
324 demonstrating the importance of considering radiation. Additionally, the oxy-

325 fuel case also shows similar predictions between the LES and RANS results when
326 the temperature profile becomes more uniform at 575 mm from the quartz exit.
327 The time-averaged predictions of velocity in the oxyfuel case, Figure 10, show
328 similar trends to the air-fired case, however the recirculation in the centre of the
329 furnace is predicted to be stronger in the LES case, which draws in a greater
330 concentration of CO_2 close to the burner, which can be seen in Figure 11.

331 Comparisons between predicted and measured values of surface incident ra-
332 diation are shown in Figures 12 and 13 for air and oxy-27 respectively. Under
333 air-fired conditions the two RANS cases over-predict the surface incident radi-
334 ation, with the FSCK model providing a small improvement over the WSGG
335 predictions. The LES results for surface incident radiation, although show-
336 ing a similar trend to the RANS predictions, are significantly lower than the
337 RANS results, and are much closer to the experimental measurements. These
338 results agree with other LES and RANS by Edge et al. [14], which showed an
339 over-prediction in surface incident radiation for RANS results, but a very good
340 agreement with LES calculations for a similar Doosan Babcock triple-staged
341 low- NO_x burner. The combination of these findings suggest that the improved
342 treatment of flow turbulence with this burner design provide significantly better
343 predictions with regards to the calculation of surface incident radiation.

344 The reduced prediction of incident radiation in the LES cases, despite the
345 simulation resolving highly-emitting hot eddies, may be related to the temperat-
346 ure predictions, specifically by analysing the temperature distributions shown in
347 Figures 4 and 5. The figures illustrate that the RANS calculations predict peaks
348 in gas temperature close to the furnace wall, while the LES calculations show
349 higher temperature peaks near the centre of the furnace, which is highlighted
350 by the instantaneous LES results. The location of the peak temperatures in
351 the centre of the domain effectively increases the path-length from the radiation
352 source to the wall significantly, influencing the heat flux at the wall surface.
353 Since the verification of radiative heat transfer prediction is generally measured
354 at the wall, this highlights the importance of predicting the correct flame shape
355 and transient effects of the flame. The sensitivity of the spectral radiation mod-

356 els for the LES calculations should be considered in a further study, to determine
357 whether this increase in radiative path influences the impact over the choice of
358 radiation model.

359 The LES calculations of air-fired combustion fail to predict the peak in sur-
360 face incident radiation that was measured in the experiments near 0.7 m from
361 the quarl exit. It is believed that this region, where there is a significant number
362 of combusting char particles, will be sensitive to the correct treatment of particle
363 radiative properties, and this discrepancy could be explained by the simplistic
364 approach used for these values. The measurements of the oxy-27 combustion
365 case do not demonstrate the same peak as the air-fired case, and the LES pre-
366 dictions show a much closer agreement to the measurements. The agreement
367 between the WSGG and the FSCK models in both cases, despite the signifi-
368 cant reduction in the temperature prediction from the case when radiation is
369 neglected, suggest that the influence of spectrally constant radiative quantities,
370 such as the particle and refractory wall properties, dominate how radiation is
371 transferred in these cases. In future work it will be important to understand
372 the sensitivity of calculations to more-realistic non-grey radiative properties,
373 and how this influences the predictions from different spectral radiation models.

374 5. Summary and conclusions

375 This study compared the influence of a gas radiation model between two
376 RANS cases and a LES case for both air-fired and oxyfuel coal combustion with
377 measurements at a 250 kW pilot-scale facility. The LES results show greater
378 agreement with experimentally measured values than the RANS predictions
379 in the cases studied. The LES predicts greater turbulent mixing of the inlet
380 streams near the burner, which is an important region of practical interest of
381 burner performance with regards to pollutant formation. The RANS calcula-
382 tions using different spectral radiation models demonstrated similar predictions
383 for the two cases that were studied. Further work may investigate the influence
384 of using non-grey radiative properties with a spectral radiation model, such as

385 the FSCK model, over using the more widely adopted grey radiative properties.
386 The LES results showed greater agreement with experimental measurements
387 for surface incident radiation, predicting lower values than the RANS calcula-
388 tions. While the computational demands for LES are high, roughly ten times
389 that of a RANS on the same computational grid, there is a noticeable increase in
390 the agreement with experimental measurements in the solution, especially in the
391 near burner region, even though the turbulence is under-resolved in this region,
392 and further work should investigate whether an improved resolution will produce
393 greater agreement with measured values. LES predictions are promising, and
394 with further improvement in computational power it will be possible to further
395 resolve the turbulence in similar combustion cases, which should improve the
396 accuracy of predictions, as well as providing other beneficial comparisons with
397 physical phenomena, such as analysis of flame dynamics or statistics on length-
398 and time scales.

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403 tion Software (SRCS) that was used to generate the full-spectrum k-distribution
404 parameters for this work.

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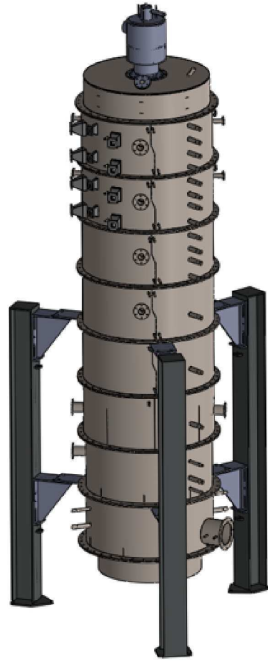


Figure 1: CAD image of the combustion test facility.

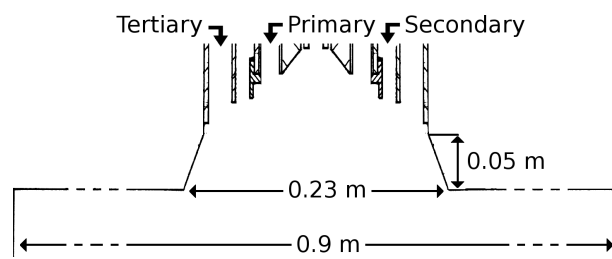


Figure 2: Sketch of the near burner region of the combustion rig.

El-Cerrejon coal properties	
<i>Calorific values (MJ/kg)</i>	
GCV	30.79
NCV	29.57
<i>Proximate analysis (AR, wt. %)</i>	
Fixed carbon	54.92
Volatiles	37.84
Ash	1.43
Moisture	5.81
<i>Ultimate analysis (DAF, wt. %)</i>	
C	79.31
H	5.43
N	2.67
S	0.40
O (by diff.)	12.19

Table 1: Details of the El-Cerrejon coal that was fired during the experimental measurements. The proximate analysis is reported ‘as received’ (AR), and the ultimate analysis is reported on a dry ash-free (DAF) basis. Oxygen content is calculated by difference.

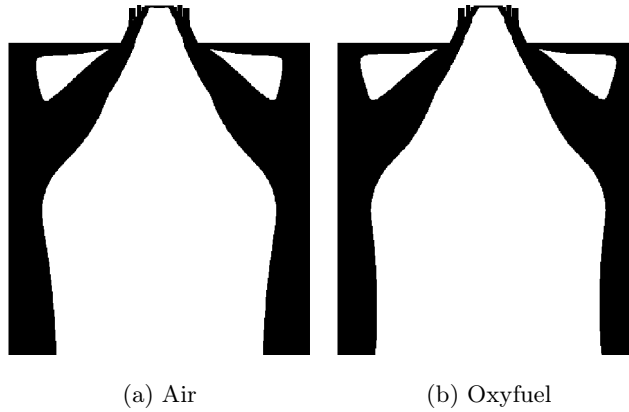


Figure 3: Grid resolution criteria for the two cases in the near-burner region. Shaded areas indicate regions in which the cell size is too coarse to resolve 80% of the turbulent scales, as predicted by the RANS solutions.

	Air	Oxy-27
<i>Mass flow rate (kg/hr)</i>		
Fuel	24.4	24.4
Primary	55.7	59.8
Secondary	95.9	102.4
Tertiary	129.2	129.2
<i>Inlet gas temperature (K)</i>		
Primary	293	294
Secondary	524	517
Tertiary	524	517
<i>Oxygen concentration (vol. %)</i>		
Primary	20.95	17.95
Secondary	20.95	29.24
Tertiary	20.95	29.24
<i>Approximate furnace pressure (Pa)</i>		
	-100	-130

Table 2: Inlet flow rates and gas compositions that were used for the CFD calculations. The balance of the gas compositions in the oxyfuel case was made up of CO₂. The furnace pressure is relative to ambient pressure.

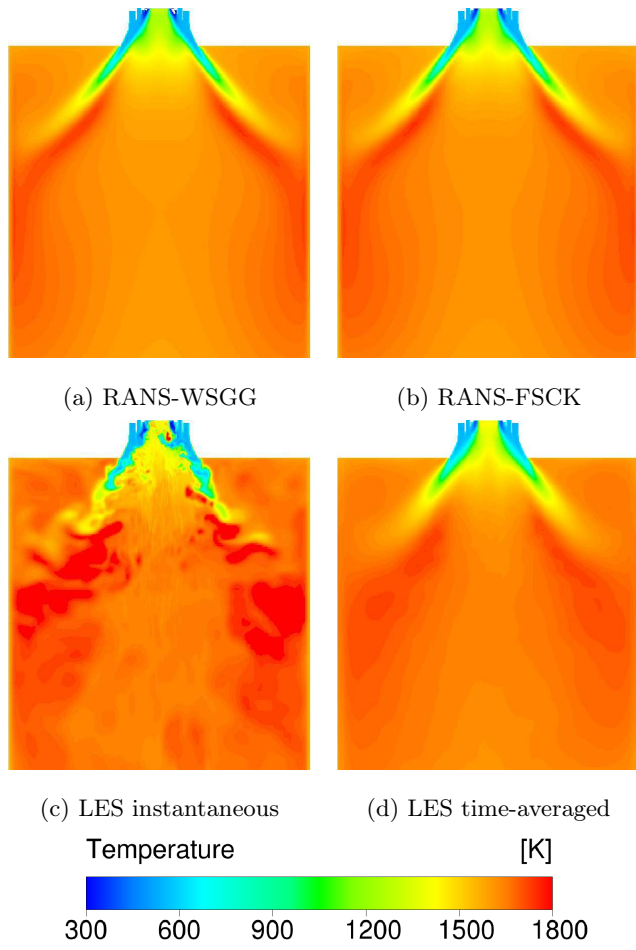


Figure 4: Temperature distributions for the air-fired case.

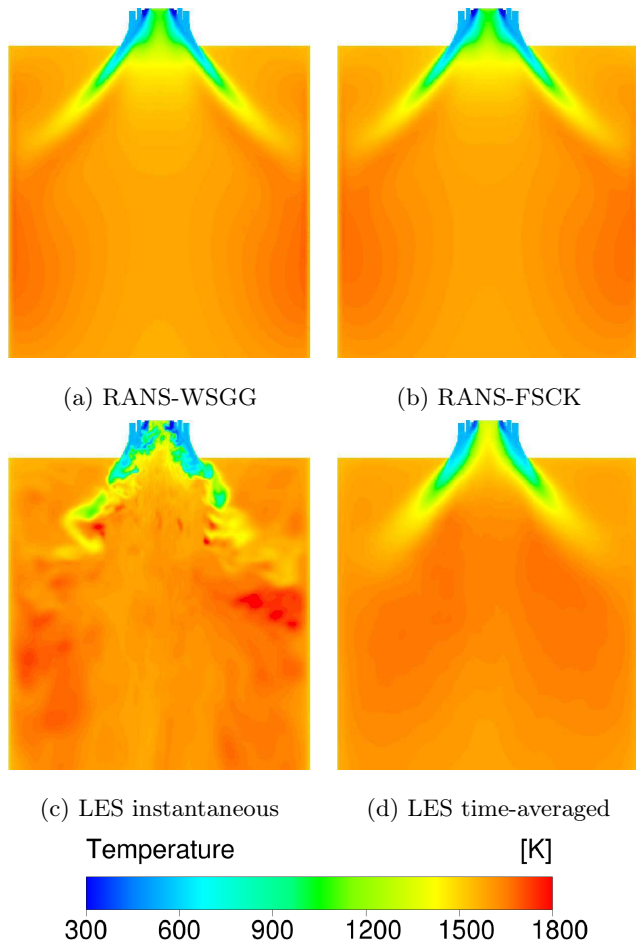


Figure 5: Temperature distributions for the oxyfuel case.

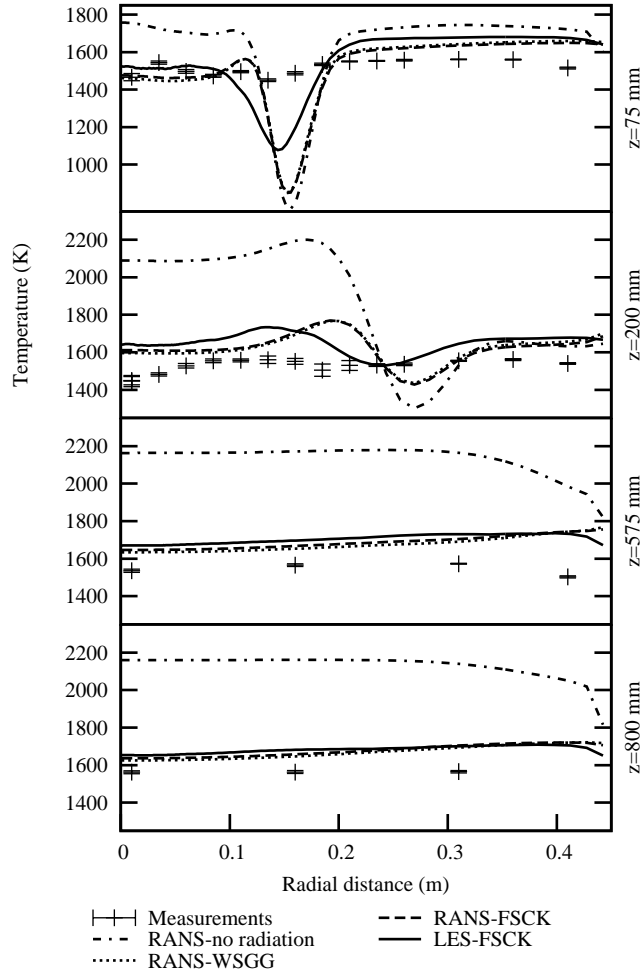


Figure 6: Radial temperature plots for the air-fired CFD cases alongside suction pyrometry measurements. In the figure, z represents the distance from the quarl exit.

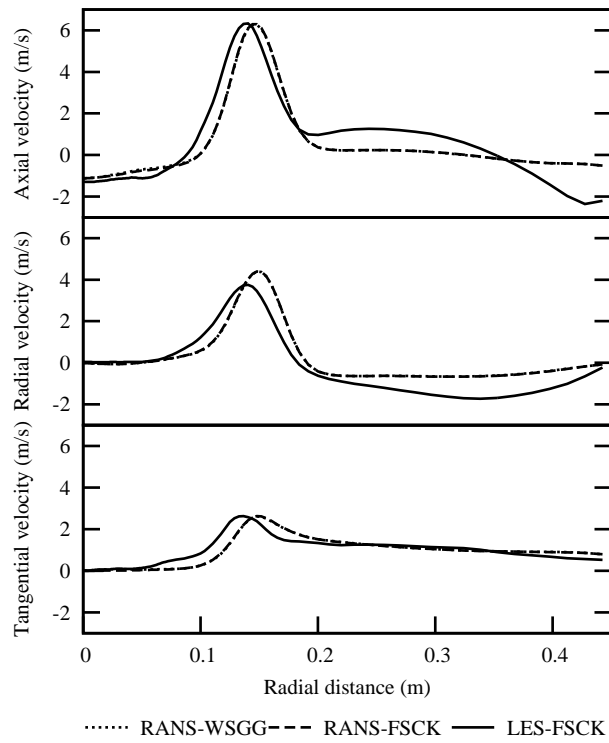
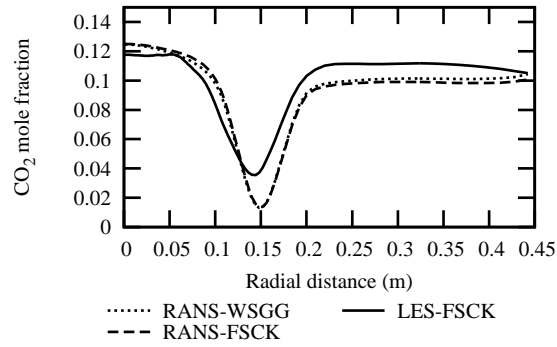
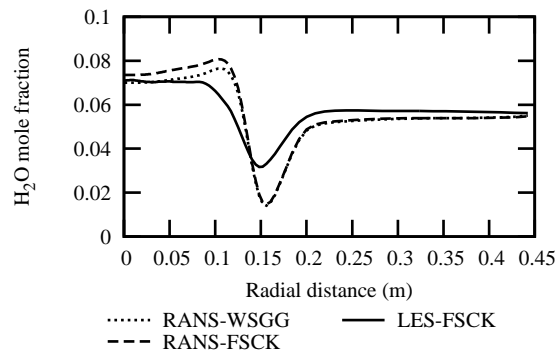


Figure 7: Plots of the time-averaged axial, radial and tangential velocities in the near burner region for the air-fired case ($z=75$ mm).



(a) CO₂ mole fraction



(b) H₂O mole fraction

Figure 8: Distribution of the participating species close to the burner ($z=75$ mm) for the air-fired case.

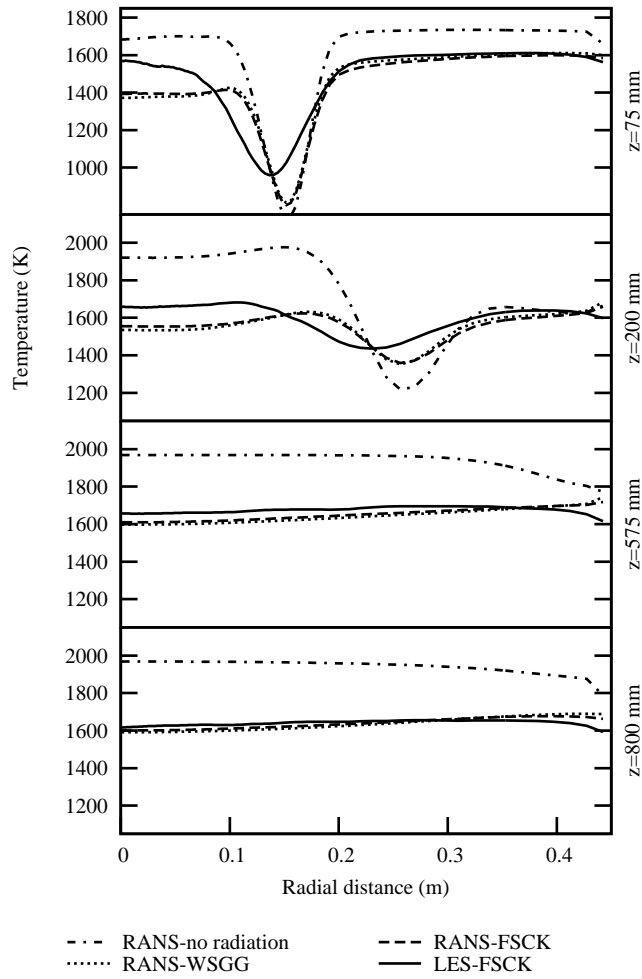


Figure 9: Radial temperature plots for the oxy-27 CFD cases. In the figure, z represents the distance from the quarl exit.

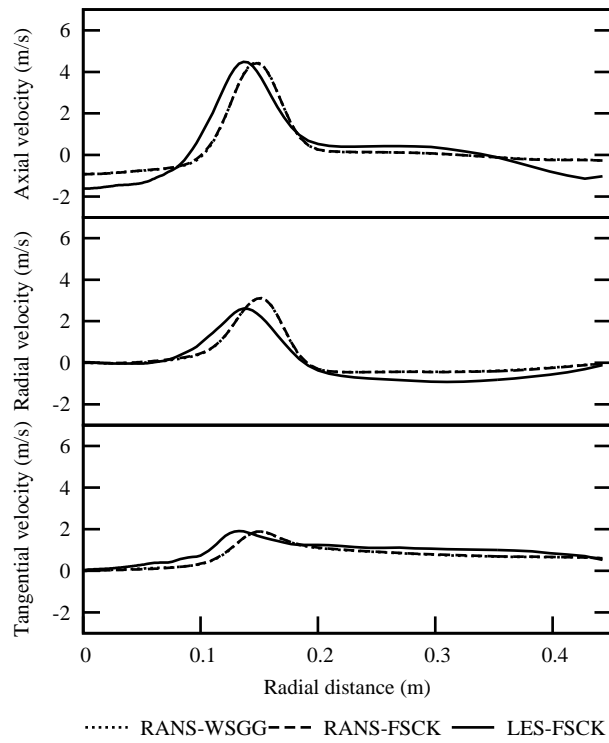
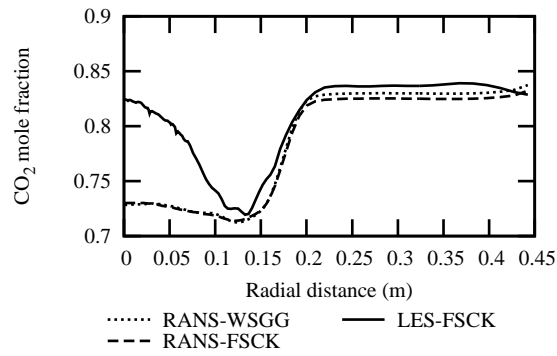
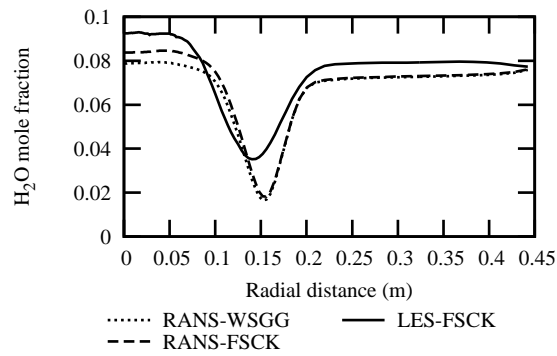


Figure 10: Plots of the time-averaged axial, radial and tangential velocities in the near burner region for the oxyfuel case ($z=75$ mm).



(a) CO₂ mole fraction



(b) H₂O mole fraction

Figure 11: Distribution of the participating species close to the burner ($z=75$ mm) for the oxyfuel case.

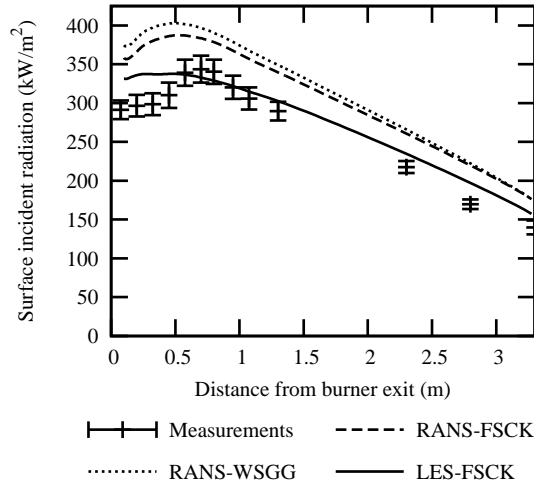


Figure 12: Surface incident radiation for the air-fired CFD cases alongside experimental measurements. Measurements were taken down the height of the furnace. Points represent the time-averaged mean measurement value, with error-bars representing one standard deviation of the values and a 3% error margin quoted from the probe manufacturer.

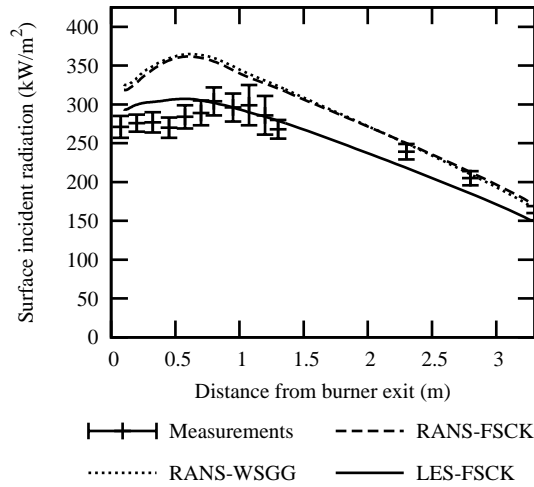


Figure 13: Surface incident radiation for the oxy-27 CFD cases alongside experimental measurements. Measurements were taken down the height of the furnace. Points represent the time-averaged mean measurement value, with error-bars representing one standard deviation of the values and a 3% error margin quoted from the probe manufacturer.