

TEACHING COMPOSITE MATERIALS USING TECHNOLOGY

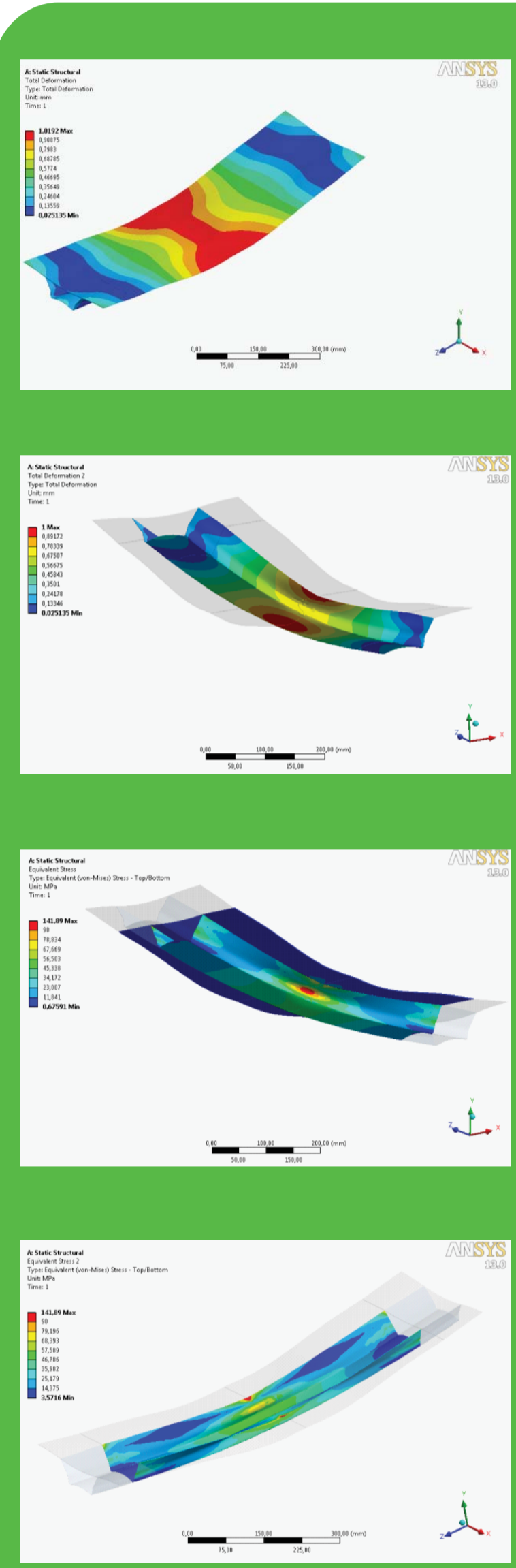
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COMPUTER SIMULATION

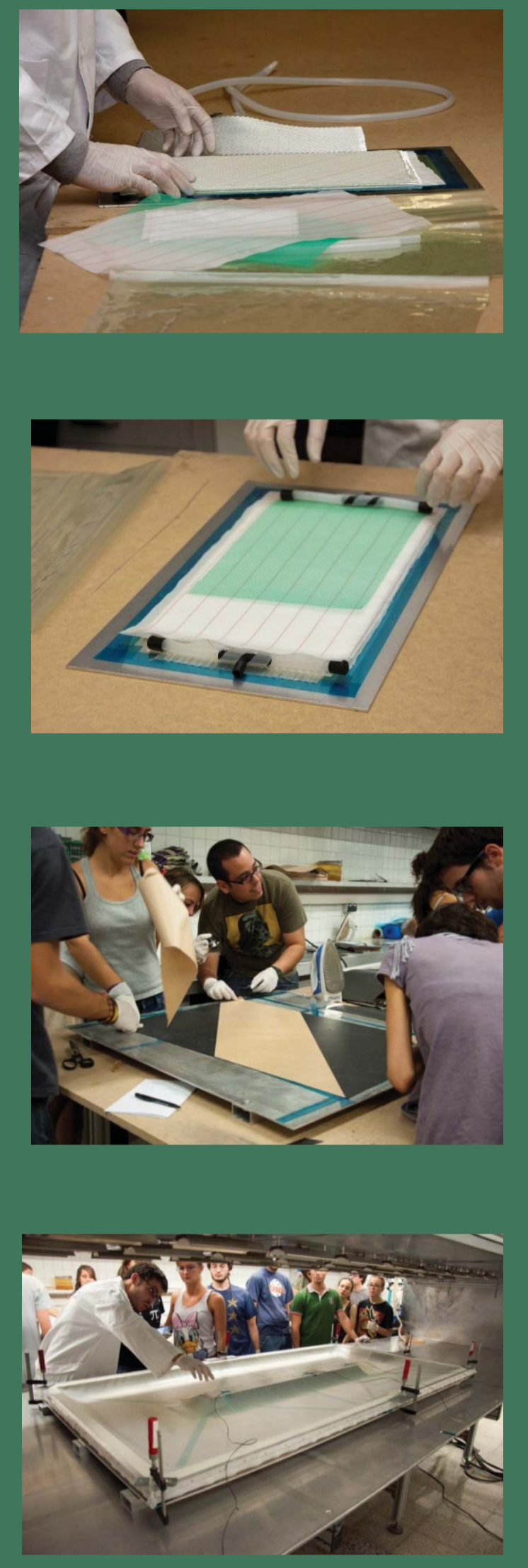


Modelling of stringers of an airplane wing using finite elements (type shell). Analytical calculation of the elastic properties of the composite with micromodelling. Fiber orientation produces orthotropic material properties.

Boundary conditions according real mechanical test.

Manufacturing of a multilayer of glass fibers and epoxy resin. Vacuum infusion process.

Layering and moulding for manufacturing a stringer with pre-pregs. Longitudinal reinforcement.



MATERIAL DESIGN

SIMULATION

ENGINEER
EXPLAIN IT

RAW
MATERIALS

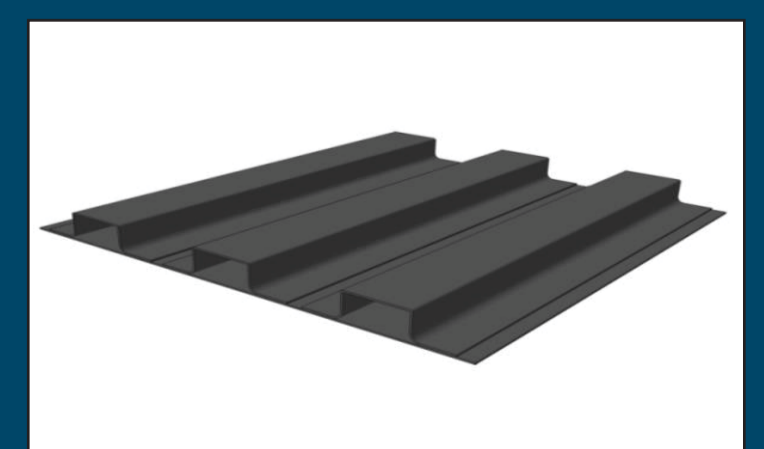
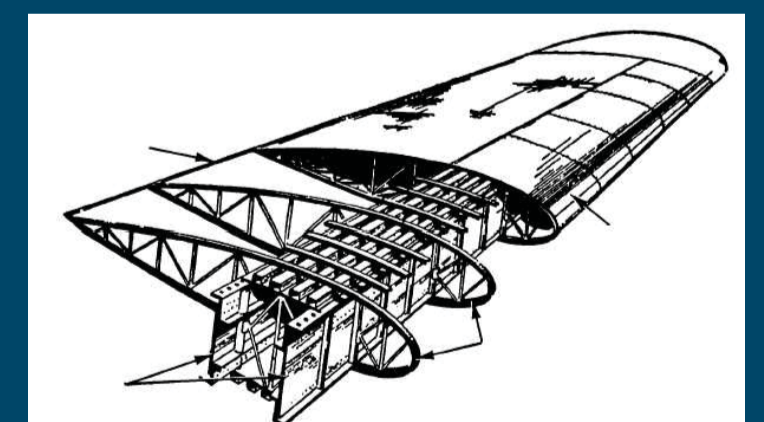
MANUFACTURING

COMPOSITE
MATERIALS
FROM RAW TO MARKET

MECHANICAL
TEST

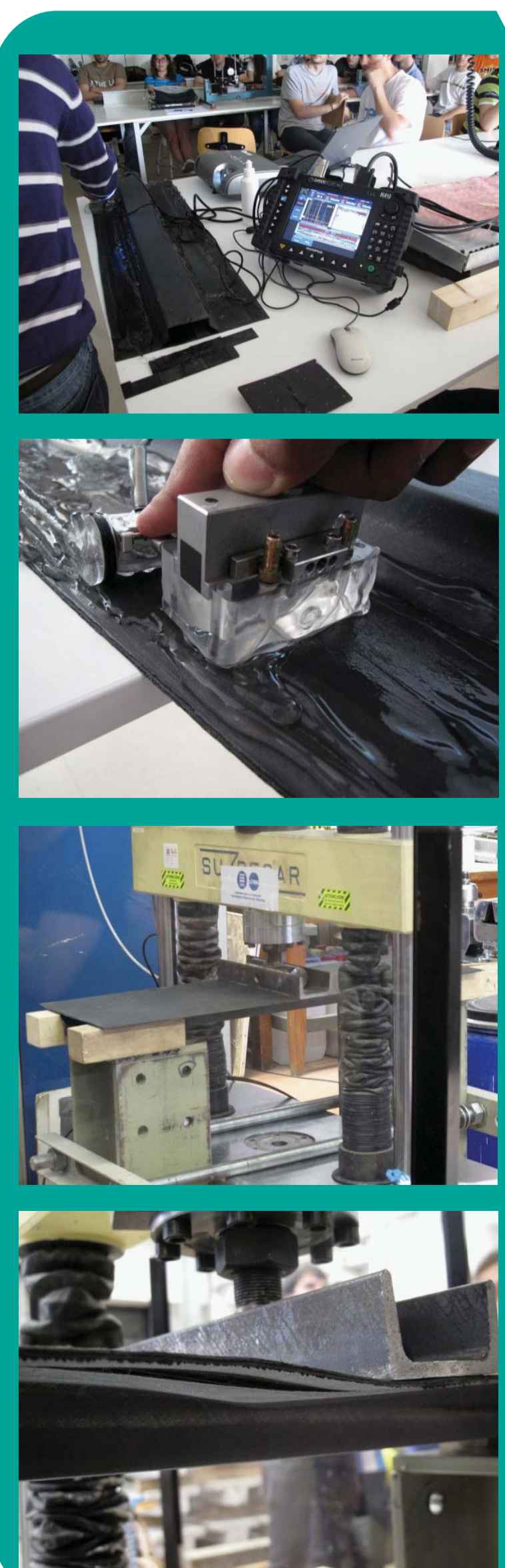
MOLDING
AND
CURING

ULTRASOUND
NDT



STRUCTURAL DESIGN

EXPERIMENTAL TESTING



The experimental work performs two activities: firstly, a non-destructive inspection with ultrasounds to identify defects and manufacturing damages; secondly, a mechanical destructive three-points bending test.

Non destructive test NDT performs a phase-array 3D scan of the stringer.

Use Mic-Mac software to calculate the structural performance of a stringer of an aircraft component part with composite laminates. Evaluate the stacking sequence, ply properties and failure criteria.

Material	Thickness	Modulus	Poisson's Ratio	Strength	Failure Mode
Carbon Fiber	0.125	140000	0.28	3500	Fiber Break
Epoxy Resin	0.005	3000	0.35	100	Matrix Cracking
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