

1 **Characterization of Flame Cut Heavy Steel – Modeling of Temperature History and Residual**  
2 **Stress Formation**

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11 **Abstract**

12 Heavy steel plates are used in demanding applications that require both high strength and hardness.  
13 An important step in the production of such components is cutting the plates with a cost-effective  
14 thermal cutting method such as flame cutting. Flame cutting is performed with a controlled flame and  
15 oxygen jet, which burns the steel and forms a cutting edge. However, the thermal cutting of heavy  
16 steel plates causes several problems. A heat-affected zone (HAZ) is generated at the cut edge due to  
17 the steep temperature gradient. Consequently, volume changes, hardness variations and  
18 microstructural changes occur in the HAZ. In addition, residual stresses are formed at the cut edge  
19 during the process. In the worst case, unsuitable flame cutting practices generate cracks at the cut  
20 edge.

21 The flame cutting of thick steel plate was modeled by using the commercial finite element software  
22 ABAQUS. The results of modeling were verified by X-ray diffraction based residual stress  
23 measurements and microstructural analysis. The model provides several outcomes, such as obtaining  
24 more information related to the formation of residual stresses and the temperature history during the  
25 flame cutting process. In addition, an extensive series of flame cut samples was designed with the  
26 assistance of the model.

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<sup>1</sup> Present address.

27 **Keywords:** flame cutting, heavy steel plate, finite element, heat-affected zone, temperature history,  
28 residual stress

## 29 **Introduction**

30 Flame cutting is a thermal cutting method generally used by steel manufacturers. It is an effective  
31 method for cutting thick wear-resistant steel plate, unlike mechanical cutting, which is both difficult  
32 and too slow for high production rates. Flame cutting is an exothermal process, which provides an  
33 advantage over other thermal cutting methods, because the heat generated from the cutting process  
34 supports the continuation of the flame cutting [1].

35 The flame cutting process consists of three steps. Firstly, the steel is heated locally to its ignition  
36 temperature by using a flame obtained from the combustion of a specific fuel gas mixed with oxygen.  
37 Secondly, the heated spot is burnt with a jet of pure oxygen, which creates a continuous chemical  
38 reaction between the oxygen and the steel. Thirdly, the oxygen jet not only burns the steel but also  
39 blows away the iron oxide that is formed during the cutting process. [2]

40 However, the flame cut edge is prone to cracking, which makes cutting of thick steel plate  
41 problematic. It has been shown [3] that an increase in both the hardness and thickness of the plate  
42 enhances the cracking tendency. Flame cutting produces a heat-affected zone (HAZ) at the cut edge  
43 of steel plate due to the generation of a steep thermal gradient during the cutting process. For  
44 example, Martín-Meizoso et al. [4] have reported that microstructural changes and hardness  
45 variations occur in the HAZ. Hardness values have been observed to be higher closer to the cut edge  
46 and decrease over a short distance from the cut edge [5]. In addition, the width of the HAZ decreases  
47 with increasing cutting speed [6]. Thomas et al. [7] found that flame cutting produces a martensitic  
48 layer on the steel edge. The thickness of the martensitic layer and the HAZ were observed to be  
49 dependent on the plate thickness and flame cutting speed.

50 The flame cutting process results in the formation of residual stresses in the cut edge of the steel. It  
51 has been reported [3] that high residual stresses in the cut edge promote crack formation. Residual  
52 stresses are formed by uneven plastic strains in the material which cause elastic strains. These  
53 elastic strains maintain the dimensional continuity in the vicinity of the plastically deformed regions [8].  
54 The elastic strains and hence the residual stresses can be either compressive or tensile. Generally,  
55 residual compressive stresses are beneficial because they reduce the probability of cracking,

56 whereas residual tensile stresses are unfavorable because they enhance it. Large thermal gradients  
57 produced by flame cutting cause residual stresses consisting both of thermal stresses and  
58 transformation stresses. Thermal stress arises from the inhomogeneous thermal expansion and  
59 contraction of the material, while transformation stress is produced by microstructural transformations  
60 and their different volumetric expansions. [9]

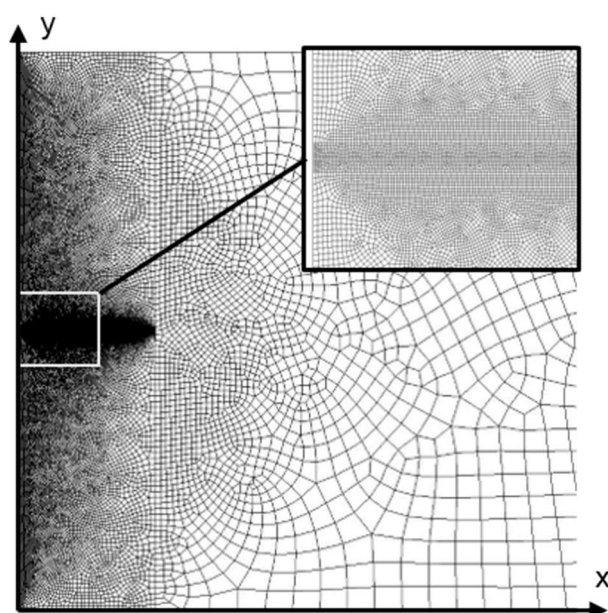
61 Several studies have been carried out to determine and model the generation of residual stresses  
62 during flame cutting. Wei et al. [10] modeled the flame cutting of 10-mm-thick steel plate and the  
63 simulation results indicated that a slower cutting speed produced a wider HAZ and more compressive  
64 stress than a faster cutting speed. However, the cutting speed did not have any notable impact on the  
65 residual tensile stress maxima. Thiébaud and Lebet [11] used the section method to measure the  
66 residual stress distribution from 60-mm-thick steel plate. The results indicated that there was a tensile  
67 stress region close to the cut edge, which decreased rapidly with increasing distance from the edge,  
68 and was partly balanced by a compressive stress region deeper in the subsurface. Thomas et al. [7]  
69 studied 25 mm and 35 mm steel plates and discovered that, at a short distance (0.1 mm) below the  
70 flame cutting edge, the stresses are compressive and deeper (>1 mm), the stresses are tensile.  
71 Lindgren et al. [3] measured and modeled the residual stresses produced by flame cutting 50-mm-  
72 thick steel plates and the simulation results indicated the formation of a low compression stress region  
73 close to the cut edge, which was followed by a high residual tensile stress region. The residual tensile  
74 stress state was lower in preheated samples compared to samples which were cut without  
75 preheating. This model was verified by using a hole drilling strain gauge method to measure the  
76 residual stresses from certain locations of the cut edge. Despite the earlier studies, the residual stress  
77 formation in thick wear-resistant steel plates during the flame cutting process remains a fairly  
78 unknown phenomenon. The effect of different cutting parameters has been studied to some extent but  
79 further information related to this topic is required.

80 The aim of this study was to develop a model, which provides an effective tool for investigating the  
81 flame cutting process of a thick wear-resistant steel plate. In addition, modeling creates an opportunity  
82 to obtain information about the steel plate during the flame cutting process, which is almost  
83 impossible to obtain experimentally. The present model enables us to systematically study the effect  
84 of different flame cutting parameters, such as various flame cutting speeds, cutting preheated plate

85 and cutting steel plate of different plate thicknesses. In addition, with the assistance of the model we  
86 can design the flame cut parameters to be used in a comprehensive test series for future studies.

### 87 **Material modeling and Experimental procedure**

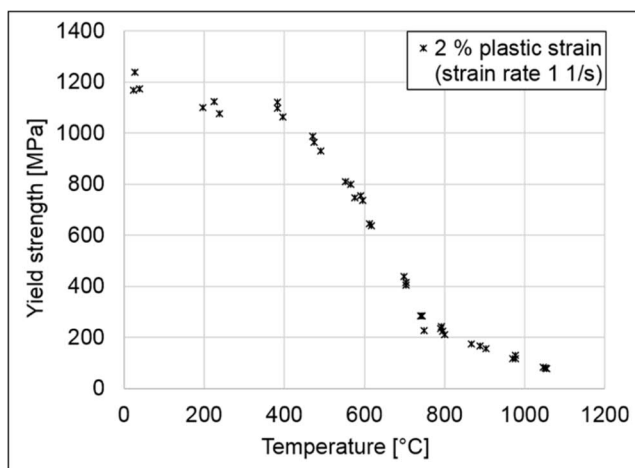
88 The modeling of the thick steel plate flame cutting process was carried out by using the commercial  
89 finite element software ABAQUS. The purpose of the modeling work was to simulate the behavior of a  
90 previously studied [12] low-alloyed wear-resistant steel, the composition of which is given in Table 1.  
91 In the preliminary study [12], the residual stress profiles were measured from some flame-cut thick  
92 wear-resistant steel plates. The modeled part here was a rectangular shape steel plate modelled as a  
93 two-dimensional plane strain section: the thickness (y-direction) was set to 40 millimeters and the  
94 width (x-direction) was defined as long enough that the body could be considered semi-infinite. The  
95 model was constructed with a mesh with over 33 000 four-node bilinear thermo-mechanically coupled  
96 elements (Fig.1). The mesh was designed to be denser (element size of 0.04 mm) in the middle  
97 section in order to ensure accurate results from the area of interest, which is the most critical for crack  
98 formation.



99  
100 **Fig. 1.** Finite element mesh with a zoomed view from the middle section.

101 During the flame cutting process, every material point within the solid has its individual temperature  
102 history (i.e. maximal temperature, heating and cooling rate, etc.), which affects the material  
103 properties. For example, material properties during cooling depend on the maximum temperature

104 attained during heating. Accurate simulation of these history effects would require a very large  
 105 number of experimental tests and a large number of material parameters in the model. Therefore, in  
 106 the model presented here a simplification is made, i.e. most of the thermal and mechanical properties  
 107 are assumed to be directly temperature-dependent without any history effects. The only exception is  
 108 the thermal expansion coefficients, which have different values depending on the maximum  
 109 temperature and whether the part is heating up or cooling down. As explained below, thermal  
 110 expansion coefficients are used to model the effect of phase transformations on a specific material  
 111 volume. Therefore, a history-dependent approach is needed for these material properties. In order to  
 112 acquire the temperature-dependent yield strength properties of the material, a series of uniaxial  
 113 compression tests was conducted in various temperatures using a Gleeble 3800 thermo-mechanical  
 114 simulator. The strain rate in the test was set to 1 1/s to correspond to the actual cutting process and  
 115 the specimen was heated to the target temperature at a heating rate of 250 °C/s. Loading was applied  
 116 for 0.5-1.0 seconds after reaching the target temperature, thus minimizing excess temperature effects  
 117 like tempering.



118  
 119 **Fig. 2.** The results of the uniaxial compression tests using Gleeble: yield strength at 2 % plastic strain  
 120 as a function of temperature.

121 Fig. 2 shows that yield strength is a highly temperature-dependent material property and there is a  
 122 significant drop in the yield strength values after the temperature rises above 400 °C. Acquiring  
 123 correct values for yield strength as a function of temperature is important, since the yield limit decides  
 124 whether the material reacts elastically or elastoplastically, which has a major impact on the stress  
 125 distribution inside the steel plate. It should be noted that the yield strength values were measured for

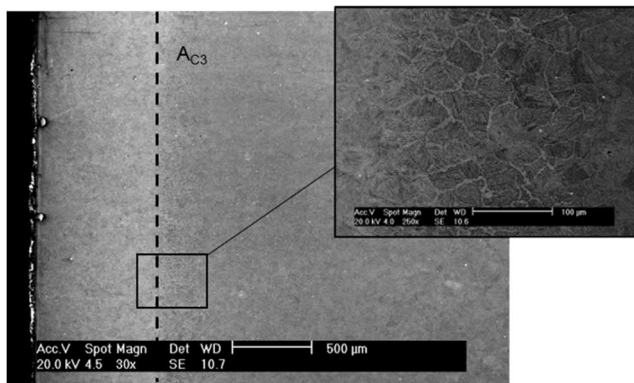
126 the heating stage only and assumed to adequately represent the material behavior also during the  
 127 cooling stage.

128 In order to work correctly, the model requires phase transformation temperatures for both austenite  
 129 ( $A_{c3}$  and  $A_{c1}$ ) and martensite ( $M_s$  and  $M_f$ ) transformations. Both the austenite start temperature ( $A_{c1}$ )  
 130 and the martensite start temperature ( $M_s$ ) were obtained using the Andrews equations [13]:

$$131 \quad A_{c1}(\text{°C}) = 723 - 10,7Mn - 16.9Ni + 29.1Si + 16.9Cr + 290As + 6.38W \quad (1)$$

$$132 \quad M_s(\text{°C}) = 539 - 423C - 30.4Mn - 17.7Ni - 12.1Cr - 7.5Mo \quad (2)$$

133 where the chemical symbols denote the weight percentage of the element in question. However, a  
 134 different approach was needed for the full austenitizing temperature ( $A_{c3}$ ), due to the rapid heating  
 135 characteristic of flame cutting. The  $A_{c3}$  was set to 1077 °C after comparing the temperature distribution  
 136 obtained from the model with microstructures observed from SEM micrographs, such as Fig. 3.



137  
 138 **Fig. 3.** SEM micrograph from the cut edge of a 300 mm/min flame cut sample.

139 In Fig. 3, the microstructural regions formed during the 300 mm/min flame cutting process can be  
 140 seen. The fully martensitic region extends 0.8 mm from the flame cut edge of the sample. The  $M_f$   
 141 temperature was estimated to be 234 °C. According to Steven and Haynes [14], the  $M_f$  temperature  
 142 can be approximated as  $215 \pm 15$  °C below the  $M_s$  temperature. The phase transformation  
 143 temperatures implemented in the model are shown in Table 2.

144 The thermal and phase transformation induced (austenite and martensite) volume changes were  
 145 entered into the model as subroutines. The austenite phase fraction ( $f_a$ ) was calculated using a  
 146 modified Avrami function called Weibull's cumulative distribution function [15]:

$$147 \quad f_a = f_{a_{final}} \left( 1 - \exp \left\{ A \left( \frac{T - A_{c1}}{A_{c3} - A_{c1}} \right)^B \right\} \right), \quad (2)$$

148 where  $f_{a_{final}}$  is the phase fraction at the end of the transformation, and A and B are material-  
 149 dependent constants, set to -6 and 2, respectively [15]. The effect of austenite formation in the  
 150 thermal axial expansion ( $\Delta L/L$ ) was calculated using the following equation:

$$151 \quad \frac{\Delta L}{L} = (f_a \alpha_a + (1 - f_a) \alpha_s) \Delta T, \quad (3)$$

152 where  $\alpha_s$  is the thermal expansion coefficient of the parent steel and  $\alpha_a$  is the austenite thermal  
 153 expansion coefficient, set to  $13 \times 10^{-6}$  1/K and  $20 \times 10^{-6}$  1/K, respectively. The thermal expansion  
 154 coefficient of the martensite ( $\alpha_m$ ) was the same as that of the parent steel. The martensitic phase  
 155 fraction ( $f_m$ ) was calculated by using the equation derived from Koistinen and Marburger [16]:

$$156 \quad f_m = 1 - \exp\{\beta(M_s - T)\}, \quad (4)$$

157 where a value of -0.04 was used for  $\beta$ , which was selected so that 50 per cent of the martensite  
 158 transformation happens almost instantly. The axial expansion changes caused by the martensitic  
 159 phase transformation were introduced to the model via thermal expansion subroutines, as shown in  
 160 the following equation:

$$161 \quad \frac{\Delta L}{L} = \left\{ f_a \left( (1 - f_m) \alpha_a + (f_m \alpha_m) + \left( \frac{\Delta x_m}{x_m} / (M_f - M_s) \right) \right) + (1 - f_a) \alpha_s \right\} \Delta T, \quad (5)$$

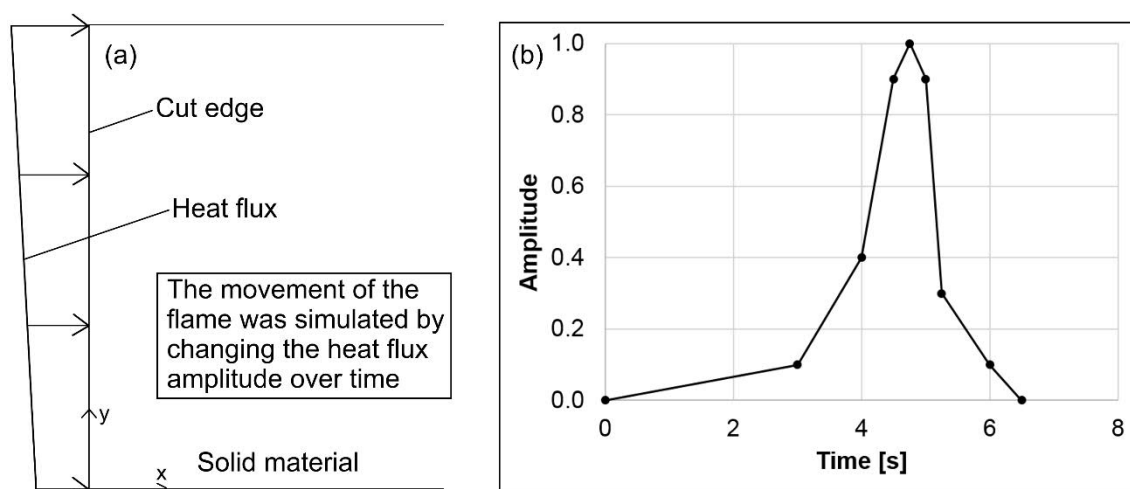
162 where the  $\Delta x_m/x_m$  is the axial expansion of the martensitic phase and was set to 0.75 per cent, which  
 163 was evaluated to correspond to the real situation. The axial expansion can be converted to a volume  
 164 expansion using the equation:

$$165 \quad \frac{\Delta V}{V} = \left( 1 + \frac{\Delta L}{L} \right)^3 - 1, \quad (6)$$

166 For simplification, the martensitic transformation was considered to be an isotropic volume expansion,  
 167 which may not fully correlate with the actual martensitic transformation process.

168 The simulation of the flame is one step in the modeling of the flame cutting of a steel plate. From a  
 169 modeling perspective, flame cutting is an extremely complex process with a large set of variables,  
 170 which it is difficult to verify. However, the main purpose of this model was to study what occurs inside

171 the steel when it is subjected to a large amount of heat, rather than the perfect modeling of the flame.  
 172 Consequently, some simplifications had to be made. Therefore, the flame was created as a time-  
 173 dependent heat flux, which simulated the movement of the flame. In the three dimensional preliminary  
 174 simulations the flame was modeled as a moving line heat flux on the surface (the cutting surface) of  
 175 the plate. Based on the results of these preliminary studies the heat source was modelled in the  
 176 actual two-dimensional simulations as a heat flux boundary condition (Fig. 4(a)) on the edge on the  
 177 element mesh (the left edge of the model in Fig.1). The movement of the flame was simulated by  
 178 changing the amplitude of the flux with respect to time. Similar method has previously been used by  
 179 Lindgren et al. [3]. To represent the real flame cutting process, the heat flux applied to the part was  
 180 not totally uniform, thus the upper (flame) side of the part was subjected to more heat since the flame  
 181 has a greater impact there. In addition, we used a time-dependent amplitude distribution of the heat  
 182 flux in our model to resemble a moving flame. The amplitude distribution for the 150 mm/min cutting  
 183 speed heat flux (Fig. 4(b)) was created by studying the data obtained from a simulation based on a  
 184 three-dimensional flame model.



185  
 186 **Fig. 4.** (a) Heat distribution along the cut edge ( $y$ -axis is the cutting depth). (b) The time-dependence  
 187 of the heat flux amplitude for 150 mm/min cutting speed. It should be noted that the  $y$ -axis presents  
 188 the magnitude of the heat flux relative to the maximum value.

189 In Fig. 4(b), the time frame between 0 and 4.5 seconds simulates the heat transfer which occurs  
 190 through conduction before the flame arrives. The period between 4.5 and 5 seconds simulates the  
 191 moment when the flame is connected to the observed position. The time frame between 5 and 6.5  
 192 seconds represents the heat transfer to the observed position after the flame has passed. A similar



193 heat flux amplitude was used for other cutting speeds, although the periods were divided according to  
 194 how fast the process was compared to the 150 mm/min cutting speed.

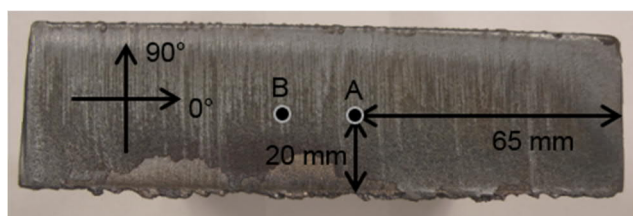
195 The heat input for the flame was determined by iterating the magnitude of the heat flux. The heat  
 196 input (maximum amplitude with reference to Fig. 4(b)) for thermal analysis was  $1.8 \times 10^7 \text{ W/m}^2$  and it  
 197 was selected so that the surface temperature at the center of the plate (i.e.,  $x=0$  and  $y=0.5 \times \text{thickness}$   
 198 in Fig. 1) reached the melting point of  $1520 \text{ }^\circ\text{C}$ . The heat input for the stress analysis of 150 mm/min  
 199 and 300 mm/min cutting speeds was set to  $1.65 \times 10^7 \text{ W/m}^2$  and  $2.37 \times 10^7 \text{ W/m}^2$ , respectively. The  
 200 heat flux for stress analysis was selected so that the maximum temperature at the above-mentioned  
 201 location (surface of the center of the plate) was just below the melting point. This was necessary in  
 202 order to avoid the removal of elements or setting them to zero, which would have an undesired effect  
 203 on the analysis of the stress curves in the surface region. Since the heat flux represents the net heat,  
 204 the heat losses of the flame are ignored. In addition, the model was used to study the effect of  
 205 preheating, as it has been observed to lower tensile stress maximum values in residual stress  
 206 measurements. Preheating was simulated by setting the modeled part for different predefined  
 207 temperature fields and the heat flux magnitudes were adjusted so that the surface elements would not  
 208 exceed the melting temperature.

209 The results of the model were verified by residual stress measurements done with an XStress 3000  
 210 X-ray diffractometer (manufactured by Stresstech Oy) and the measurement method used is called  
 211 the modified Chi method [17]. This method calculates, using Bragg's law, the interplanar lattice  
 212 spacing of the ferrite [211] plane from the  $156^\circ$  Bragg diffraction angle. The lattice plane spacing  
 213 changes from a stress-free value to some new value depending on the magnitude of the residual  
 214 stress. With this method, the lattice spacing  $d$  of the sample is measured at different  $\psi$  tilts, where the  
 215  $\psi$  angle is the angle between the normal of the sample and the normal of the diffracting plane. The  
 216 measured values provide a slope containing a plot of lattice spacing  $d$  as a function of  $\sin^2 \psi$ . This  
 217 slope with elastic constants can be used to calculate the residual stress from the measured location.  
 218 Residual stresses are calculated using the following equation [18]:

$$219 \quad \sigma = \left( \frac{E}{(1+\nu)} \right) m, \quad (7)$$

220 where the  $\sigma$  is the residual stress in the measured direction, E is the Young's modulus,  $\nu$  is the  
 221 Poisson's ratio and m is the slope obtained from the lattice spacing d vs.  $\sin^2 \psi$  curve. The  
 222 parameters used are listed in Table 3.

223 Residual stresses, used for model verification, were measured from 40-mm-thick samples, which  
 224 were flame cut using cutting speeds of 150 mm/min, 300 mm/min and 300 mm/min with preheating at  
 225 200 °C. Samples were measured from two locations (A and B) and in two perpendicular measurement  
 226 directions: the flame cut direction (0°) and the thickness direction (90°). These selected directions are  
 227 the most critical orientations for crack formation. The measurement locations and directions are  
 228 shown in Fig. 5.

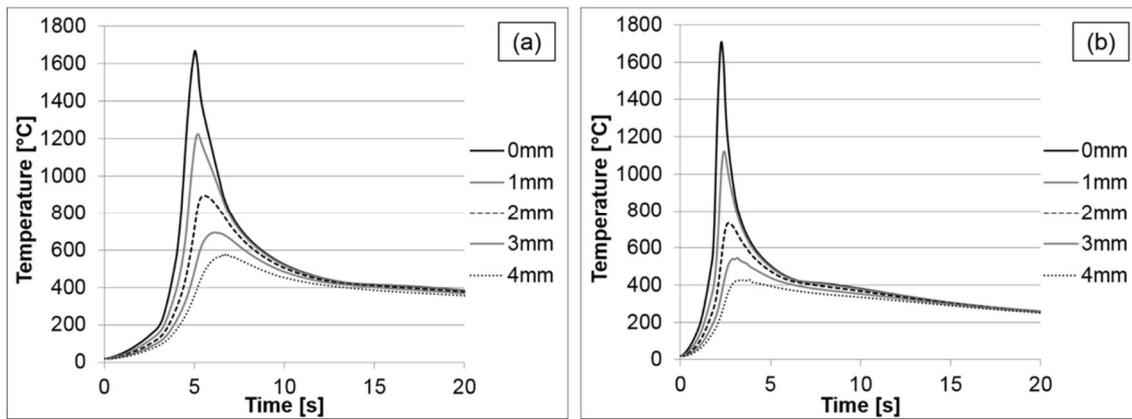


229  
 230 **Fig. 5.** Residual stress measurement locations of X-ray diffraction method for a flame cut sample.

231 Between the residual stress measurements, material layers were removed from the measurement  
 232 location by electrochemical polishing. The polishing was done using Struers A2 electrolyte (a mixture  
 233 of 60% perchloric acid, 65-85% ethanol, 10-15% 2-butoxyethanol and 5-15% water) and material  
 234 removal was verified with a dial indicator. Residual stress measurement, combined with the layer  
 235 removal method, provides residual stress depth profiles. The polished material depth was  
 236 approximately 100-200  $\mu\text{m}$  between each measurement. The measurement results were analyzed  
 237 with XTronic software and residual stress profiles were plotted from the analyzed results.

## 238 Results and discussion

239 The model provided valuable information related to the temperature history of the part during the  
 240 flame cutting process. Fig. 6 shows the modelled temperature profiles from different distances from  
 241 the flame cut edge calculated at cutting speeds of 150 mm/min and 300 mm/min, respectively.

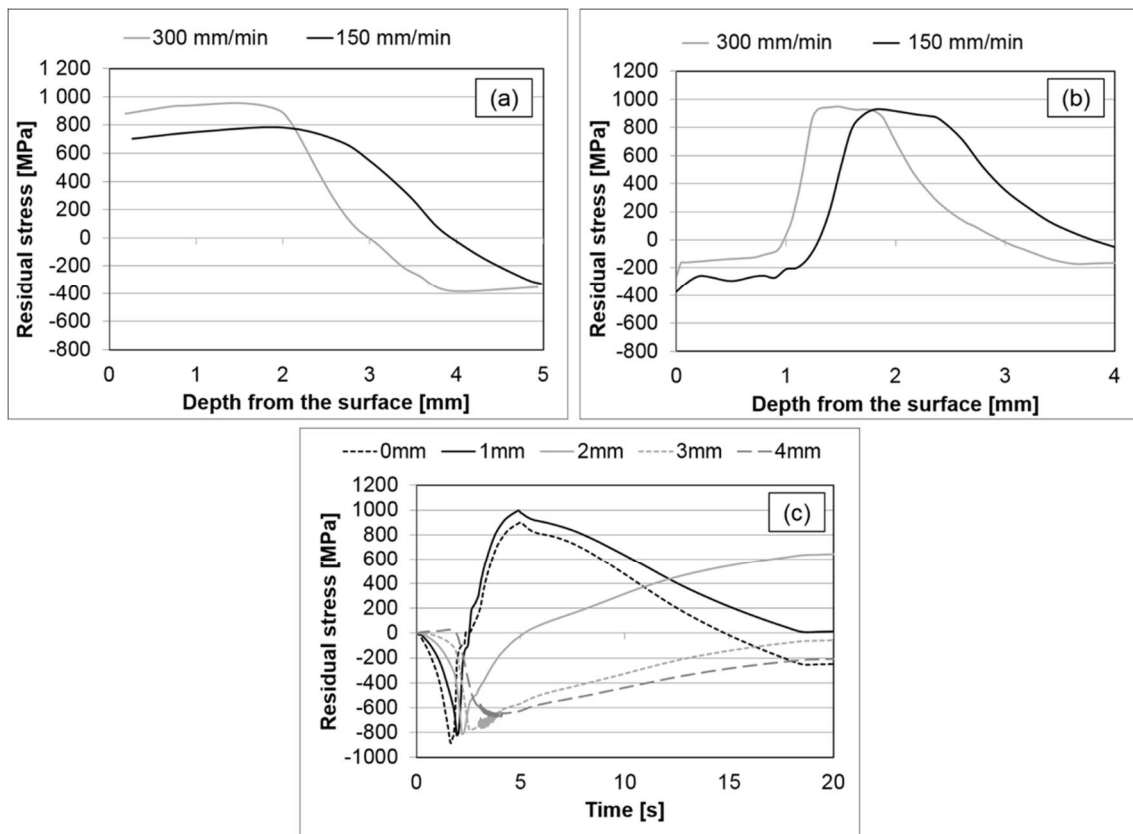


242

243 **Fig. 6.** Temperature curves from different distances from the flame cut edge at (a) 150 mm/min and  
 244 (b) 300 mm/min cutting speeds.

245 Fig. 6 shows that the slower cutting speed creates more heat at the cut edge of the plate. In contrast,  
 246 the faster cutting speed produces steeper thermal gradients compared to the slower cutting speed.  
 247 With the slower cutting speed, the part has more time to heat up and the material has more time to  
 248 adapt to the cutting situation. The shapes of the curves correspond to the experimental results of  
 249 Thiébaud et al. [2].

250 One of the main purposes of the model was to reveal information on what occurs to the steel part  
 251 during the flame cutting process. The uneven temperature distribution in the cut edge creates differing  
 252 thermal expansion (and contraction) and consequently different residual thermal stresses. Fig. 7(a)  
 253 shows the modeled thermal stress profiles (thickness direction) produced during the flame cutting  
 254 process at cutting speeds of 150 mm/min and 300 mm/min. It should be noted that in general the  
 255 maximum possible stress at a given temperature is limited by the current yield strength and fracture  
 256 stress of the material, but in the simulations the plasticity of the material was taken into consideration.  
 257 Therefore, the simulation results can be considered to indicate the best-case scenario, i.e. in reality,  
 258 material fracture might take place at stress levels below those represented by the current simulations.



259

260 **Fig. 7.** Simulated residual thermal stress profiles (thickness direction) of 150 mm/min and 300  
 261 mm/min flame cutting speeds (a) without phase transformations and (b) with phase transformations.  
 262 (c) Modelled residual stress formation during flame cutting with 300 mm/min cutting speed.

263 The residual thermal stress state during the flame cutting process is very difficult to determine  
 264 experimentally; therefore, modeling is essentially the only tool capable of providing such information.  
 265 In addition, simulations allow us to separate the effects of pure thermal expansion from the effects of  
 266 phase transformations. This is illustrated in Fig. 7(a). Without the phase transformation, the residual  
 267 thermal stresses are due to the uneven thermal expansion and contraction that occurs during cutting.  
 268 The high temperature causes volume expansion in the cut edge, which is constrained by the cold  
 269 surroundings, thus creating a compressive stress near the cut edge. Consequently, as the  
 270 temperature increases and simultaneously the yield limit is lowered, the compressive stress exceeds  
 271 the yield limit and produces a plastically deformed (compressed) region near the cut edge. During  
 272 cooling, the contraction of the compressed region is restrained by the region without plastic  
 273 deformation. For this reason, residual tensile stress is generated in the deformed region near the cut  
 274 edge. The residual tensile stress is then balanced by the residual compressive stress deeper inside

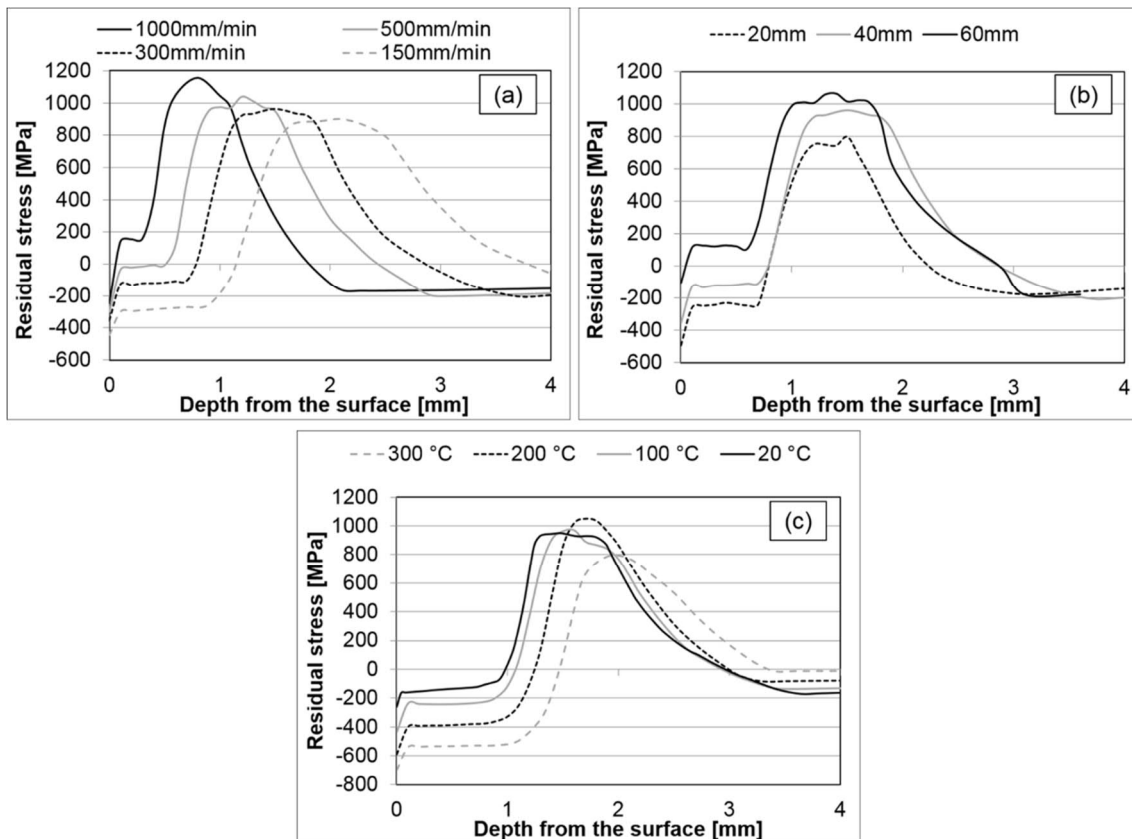
275 the part. As we can see from Figs. 6(a), 6(b) and 7(a), the faster cutting speed produces steeper  
276 thermal gradients in the cut edge, consequently creating higher thermal stresses in the cut edge  
277 compared to the slower cutting speed. Therefore, rapid and significant temperature changes should  
278 be avoided in the flame cut edge.

279 In the actual case, martensitic phase transformation is also involved in the steel structure during the  
280 flame cutting process. Fig. 7(b) shows the modeled residual stress profiles produced by 150 mm/min  
281 and 300 mm/min cutting speeds, which also takes into account the phase transformation. The effect  
282 of the volume expansion caused by martensitic transformation on the formation of the residual stress  
283 profiles is clearly seen. The martensitic transformation relieves the residual thermal tensile stresses  
284 near the flame cut edge and produces a residual compressive stress. The shapes of both residual  
285 stress curves are quite similar, but the residual compressive stress area at the surface is larger at the  
286 cutting speed of 150 mm/min than at 300 mm/min. The reason for this is the higher heat input caused  
287 by the slower cutting speed. Therefore, the phase transformation regions are larger and more  
288 elements experience the expansion effect due to martensitic transformation. Fig. 7(a) and 7(b) show  
289 that a cutting speed producing lower thermal stress also produces more residual compressive stress  
290 during martensitic transformation. This result also indicates that steep thermal gradients should be  
291 avoided during the flame cutting process.

292 Fig. 7(c) summarizes the whole chain of events that takes place during the flame cutting process. At  
293 first, the heat from the flame produces compressive stress near the surface. The stress changes to  
294 tensile once the heat is no longer applied and the part begins to cool down after 3 seconds. The  
295 tensile stress peak value is highest at 5 seconds, when the part has reached the  $M_s$  temperature and  
296 martensite starts to form, leading to volume expansion and hence to a change in the local stress state  
297 from tensile into compression. After the martensite transformation, the stresses gradually set into the  
298 final state (Fig. 7(b)) as the temperature decreases towards room temperature. The above-mentioned  
299 effect of martensite nucleation can be seen from the 0 mm and 1 mm stress curves, which are located  
300 in the phase transformation regions. However, the 2 mm curve is not located in the phase  
301 transformation region, and therefore the tensile stress continuously increases until the part reaches  
302 room temperature. Similar changes take place deeper in the material, but the resulting residual stress  
303 levels are at a lower level. It is noteworthy that, as Fig. 7(c) shows, the stress state near the surface

304 changes during the cutting process (around 2.5 seconds in Figure 7c) from compressive stress to  
 305 high tensile stress before changing back to compressive. This rapid change in the stress state takes  
 306 place just prior to the martensitic transformation and might create potential sites for crack formation  
 307 during the cutting process.

308 The model was used to predict residual stress formation with different flame cutting parameters and  
 309 various plate thicknesses. Fig. 8 shows the residual stress curves for (a) different cutting speeds, (b)  
 310 different plate thicknesses and (c) cutting at different preheating temperatures. The flame cutting  
 311 speed of 300 mm/min was used in Fig. 8(b) and 8(c). The presented residual stress profiles are in the  
 312 thickness direction of the modeled plate.



313

314 **Fig. 8.** Simulation of the residual stresses (thickness direction) for different flame cutting processes:

315 (a) cutting speed, (b) plate thickness, (c) preheating temperatures.

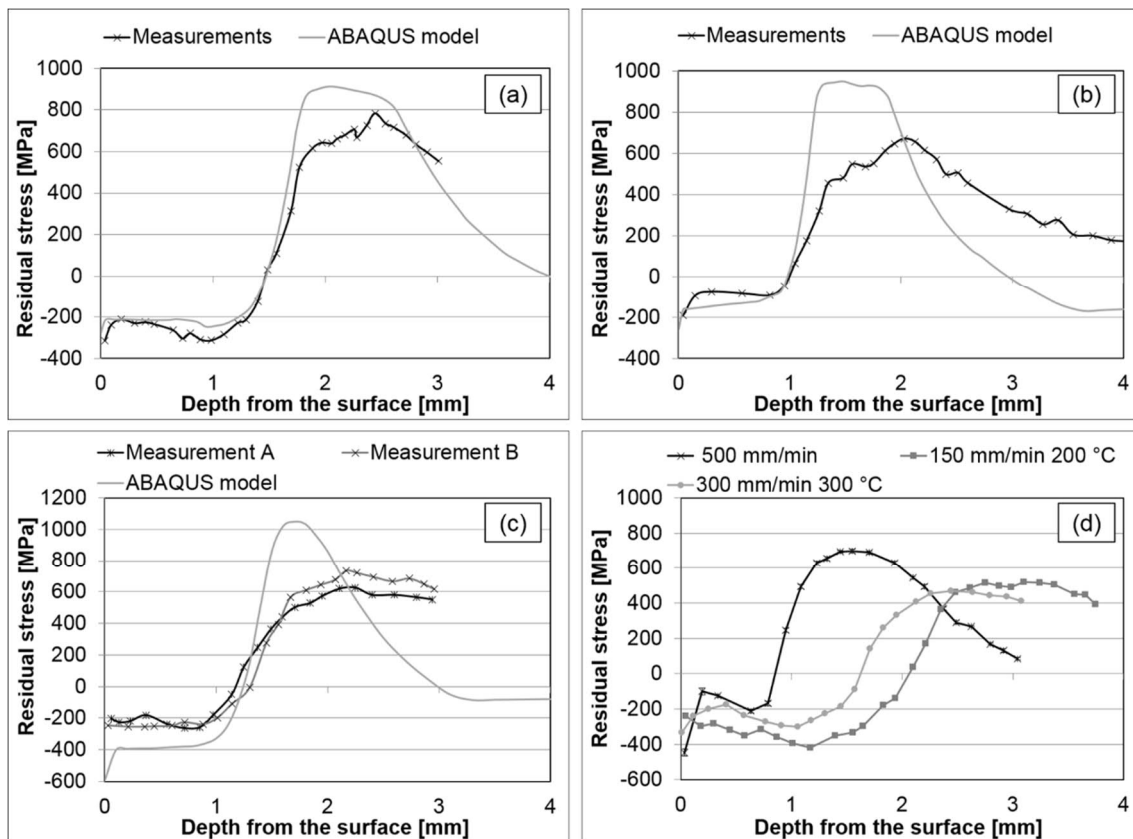
316 Fig. 8(a) shows that there is a linear development in the residual stress profiles according to the  
 317 cutting speed. The thermal shock effect from the flame is greater at faster cutting speeds; as a result,  
 318 greater residual stress values are produced closer to the cut surface. The volume expansion of  
 319 martensitic transformation is not enough to produce a residual compressive stress region on the

320 surface at faster cutting speeds. These results also indicate that rapid heating should be prevented  
321 during flame cutting.

322 Fig. 8(b) shows the effect of plate thickness on the formation of residual stresses at a cutting speed of  
323 300 mm/min. The vertical deformation is not as restricted in thinner plates as it is in thicker plates.  
324 Consequently, the cutting of thicker plates causes higher residual tensile stresses in the cut edge. In  
325 addition, the residual compressive stress decreases near the cut edge as the plate thickness  
326 increases. Therefore, due to the residual stress state produced by flame cutting, thicker plates are  
327 more prone to cracking than thinner plates.

328 As can be seen from Fig. 8(c), preheating lowers the residual stresses produced during flame cutting.  
329 Lindgren et al. [3] discovered similar effects with preheating compared to flame cutting without  
330 preheating. Preheating of the sample increases the residual compressive stresses and decreases the  
331 residual tensile stresses. However, present studies indicate that preheating not only increases the  
332 compressive stress but also expands the compressive stress region deeper in the subsurface. This is  
333 because the part is at a uniform preheating temperature; therefore, material deeper in the plate  
334 reaches the phase transformation temperatures when the heat is applied to the cut edge. Higher  
335 preheating temperatures decrease the effect of thermal shock by lowering the temperature  
336 differences inside the part, and consequently sufficiently high preheating temperatures decrease the  
337 residual tensile stresses.

338 Fig. 9 shows the modeled residual stress curve (thickness direction) (a) of 150 mm/min, (b) 300  
339 mm/min and (c) 300 mm/min with 200 °C preheating, compared to experimentally measured data  
340 from a similar sample.



341

342 **Fig. 9.** Comparison of modeled data with experimentally obtained data from (a) 150 mm/min, (b) 300  
 343 mm/min flame cutting speeds and (c) 300 mm/min cutting speed with 200 °C preheating. (d)  
 344 Experimental measurements from samples which were flame cut using a cutting speed of 500  
 345 mm/min, 150 mm/min cutting speed with 200 °C preheating and 300 mm/min cutting speed with 300  
 346 °C preheating.

347 Fig. 9(a) and 9(b) show that the compressive stress region is quite similar in both profiles; however,  
 348 the residual tensile stress peaks are different. The tensile stress maximum is different in the model,  
 349 which is to be expected, since the behavior of the material in the simulation is not totally equivalent to  
 350 the behavior of steel in reality. The differences between the modelled and experimentally obtained  
 351 residual stress curves of 300 mm/min (Fig. 9(b)) cutting speed are larger than for the 150 mm/min  
 352 cutting speed (Fig. 9(a)). In the experimentally measured 300 mm/min flame cut sample, the tensile  
 353 stress region is distributed to a larger area, which indicates that the material behavior and the heat  
 354 load in the model are not fully accurate. Fig. 9(c) shows that preheating has a notable impact on the  
 355 residual compressive stress region in both the modeled and measured profiles. However, there are  
 356 differences, which can be explained by the slightly different cutting conditions in the actual flame  
 357 cutting compared to the model. In addition, in the actual flame cutting, the tempering of martensite



358 also occurs, which is not taken into account in the model. In the preheated sample there is more  
359 tempering during the flame cutting compared to that without preheating, which may also be a reason  
360 for the difference between measured and modeled data. In addition, it has been noted [19] that  
361 diffraction based residual stress measuring method not only measures the type I residual stresses  
362 (macrostresses over large distances) but the type II residual stresses (microstresses over grain scale)  
363 can be also superimposed in the results. The purpose of this work was to study only the generation of  
364 long range (type I residual stresses) stresses during flame cutting by finite element simulations. This  
365 difference might also cause the deviation between modeled and experimental results.

366 Perfect modeling of the flame cutting process is a challenging task and there are many variables.  
367 Consequently, there are some simplifications in the model, which might have an effect on the results.  
368 It should be noted that the cracking was not taken into account in the modeling. In addition, two-  
369 dimensional modeling involves some restrictions. For example, in actual flame cutting, the moving  
370 flame not only heats one side of the part but it also simultaneously heats the part from both the cut  
371 and approach directions. Furthermore, in the actual cutting process the solid part is still intact in front  
372 of the cutting flame, which may have an effect on the formation of residual stresses. To summarize  
373 the differences of the model and experimental results (Fig. 9(a), 9(b) and 9(c)), the main discrepancy  
374 is seen at depths of 1.5 mm and deeper. Based on Fig. 6, the material temperature in this depth  
375 reaches the two-phase region and the resulting final microstructure is therefore a mixture of  
376 transformed and non-transformed material. From simulation point of view, this is the most demanding  
377 region because of the following reasons: 1) The actual microstructure in this region is very sensitive to  
378 the temperature history, thus highlighting any uncertainties in the simulated temperature field. 2) As  
379 noted previously, the simulations involved some simplifications in terms of material behavior, the most  
380 notable of which was the use of one yield strength – temperature – curve (Fig. 2) for all metallurgical  
381 states of the material. This means that the plastic-deformation behavior in the above mentioned two-  
382 phase region is probably oversimplified. In addition, the yield strength data obtained from the Gleeble  
383 experiments is from the heating stage only and at the maximum temperature, which is lower than in  
384 the actual flame cutting process. 3) Some phenomena, such as tempering of the martensite, were left  
385 outside the scope of the simulations, which most likely influences the results deeper in the material  
386 (the temperature deeper in the material is high enough for a long enough time so that some tempering  
387 may take place). The tempering of steel during flame cutting has an effect on the volume of the

388 tempered region. Therefore, the tempering also has an effect on the formation of residual stresses. In  
389 this respect, the correspondence between the simulations and the measurements is considered good.

390 In addition to the modeled results, Fig. 9(d) shows experimentally obtained residual stress profiles  
391 from samples which were flame cut using the following parameters: 500 mm/min cutting speed, 150  
392 mm/min cutting speed with 200 °C preheating and 300 mm/min cutting speed with 300 °C preheating.  
393 It clearly shows how the residual stress state can be affected by different cutting parameters. The 500  
394 mm/min cutting speed produces a high residual tensile stress peak but only a small amount of  
395 residual compressive stress near the cut edge. In contrast, the 300 mm/min cutting speed with 300 °C  
396 preheating produced significantly more residual compressive stress near the cut edge compared to  
397 cutting without preheating. The 150 mm/min cutting speed with 200 °C preheating also produced a  
398 similar kind of compressive stress region near the cut surface, although the compressive stress region  
399 extends deeper from the cut edge than the previous preheated sample. In addition, both preheated  
400 samples have a much lower tensile stress peak compared to the sample that was cut without  
401 preheating. These experimentally measured results confirm the predictions of the model: a slower  
402 cutting speed produces more residual compressive stress, lowers the residual tensile stress peak and  
403 preheating also has a similar effect on the residual stress state. In addition, the experimental  
404 measurements show that the widest compressive residual stress region and a significantly lower  
405 tensile stress peak can be produced by combining both a slow cutting speed and preheating. These  
406 results also confirm that the developed model gives accurate trend lines for evaluating residual stress  
407 formation during flame cutting.

## 408 **Conclusions**

409 Flame cutting of thick wear-resistant steel plates can be very problematic. It creates high residual  
410 stresses and may cause cracking of the steel plate. Consequently, a model was developed to  
411 investigate the problem and to study the flame cutting process. The model was created using the  
412 finite element software ABAQUS and the input material was made to behave as similarly to the  
413 studied steel as possible. The model takes into account the volume changes caused by thermal  
414 expansion (contraction) and phase transformation (austenite and martensite). A variety of simulations  
415 were computed using the model: flame cutting temperature histories, cutting at different cutting  
416 speeds, cutting different plate thicknesses and cutting using different preheating temperatures. The

417 model enabled the study of residual stress formation during the flame cutting process, which would be  
418 extremely difficult or impossible to do experimentally. The results showed that the faster cutting speed  
419 produces steeper thermal gradients in the cut edge than the slower cutting speed and consequently  
420 higher residual tensile stresses. In addition, the slower cutting speed produced more residual  
421 compressive stress near the cut edge than the faster cutting speed. Therefore, rapid and large  
422 temperature variations during flame cutting should be avoided. In addition, the residual stresses vary  
423 quickly from compressive stress to tensile stress during the cutting process depending on the time  
424 and depth, which might create potential sites for crack formation. The results of the model also  
425 showed that varying the process parameters have an effect on the residual stress formation. The  
426 plate thickness also has an effect on the residual stress formation during the cutting process. Flame  
427 cutting of thinner plates created lower tensile stress maxima and more residual compressive stress  
428 than thicker plates. Therefore, the cracking tendency of thick plates is higher than thinner plates. The  
429 results also showed that preheating was an effective way to influence the residual stress formation  
430 during the cutting process. Flame cutting with preheating reduced the residual tensile stress and  
431 produced more compressive stress near the cut edge than cutting without preheating. In addition, the  
432 experimentally measured results confirmed the predictions of the model, as the slower cutting speed  
433 and preheating produced a wider residual compressive region and lowered tensile stresses.  
434 Additionally, the experimental results showed that combining both a slow cutting speed and  
435 preheating produced even more compressive stress and a significantly lower tensile stress peak. To  
436 conclude, the model produced valuable information about the flame cutting process and formation of  
437 residual stresses. In addition, the results of the model can be used as a basis for a new flame cut test  
438 series for future studies to reveal the comprehensive effect of microstructural features on residual  
439 stress and crack formation.

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473 **Fig. 1.** Finite element mesh with a zoomed view from the middle section.

474 **Fig. 2.** The results of the uniaxial compression tests using Gleeble: yield strength at 2 % plastic strain  
475 as a function of temperature.

476 **Fig. 3.** SEM micrograph from the cut edge of a 300 mm/min flame cut sample.

477 **Fig. 4.** (a) Heat distribution along the cut edge (y-axis is the cutting depth). (b) The time-dependence  
478 of the heat flux amplitude for 150 mm/min cutting speed. It should be noted that the y-axis presents  
479 the magnitude of the heat flux relative to the maximum value.

480 **Fig. 5.** Residual stress measurement locations of X-ray diffraction method for a flame cut sample.

481 **Fig. 6.** Temperature curves from different distances from the flame cut edge at (a) 150 mm/min and  
482 (b) 300 mm/min cutting speeds.

483 **Fig. 7.** Simulated residual thermal stress profiles (thickness direction) of 150 mm/min and 300  
484 mm/min flame cutting speeds (a) without phase transformations and (b) with phase transformations.  
485 (c) Modelled residual stress formation during flame cutting with 300 mm/min cutting speed.

486 **Fig. 8.** Simulation of the residual stresses (thickness direction) for different flame cutting processes:  
487 (a) cutting speed, (b) plate thickness, (c) preheating temperatures.

488 **Fig. 9.** Comparison of modeled data with experimentally obtained data from (a) 150 mm/min, (b) 300  
489 mm/min flame cutting speeds and (c) 300 mm/min cutting speed with 200 °C preheating. (d)  
490 Experimental measurements from samples which were flame cut using a cutting speed of 500  
491 mm/min, 150 mm/min cutting speed with 200 °C preheating and 300 mm/min cutting speed with 300  
492 °C preheating.

493 **Table 1.** Chemical composition approximation of studied steel.

Amount of elements [Wt%]			
C	Cr	Mn	Si
0.130	0.890	0.970	0.620
Mo	Al	Ni	B
0.270	0.08	0.06	0.001
balanced with Fe			

494

495

496 **Table 2.** Phase transformation temperatures implemented in the model.

Temperature	Value [°C]	Value [K]
$A_{c3}$	1077	1350
$A_{c1}$	745	1018
$M_s$	440	713
$M_f$	234	507

497

498 **Table 3.** Measurement parameters used for XStress 3000 equipment.

Parameters:			
$\phi$ rotations (measurement directions)	0° and 90°	Modulus of elasticity	211 GPa
Collimator	3 mm	Poisson's ratio	0.3
$\psi$ tilt angles in one direction (side / side)	6 / 6	Voltage	30 kV
Maximum tilt angle	40°	Current	6.7 mA
$\psi$ oscillation	5°	Radiation	CrK $\alpha$

499