

A Linear Height-Resolving Wind Field Model for Tropical Cyclone Boundary Layer

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Abstract: The wind field model is one of the most important components for the tropical cyclone hazard assessment, thus the appropriate design of this element is extremely important. While solving the fully non-linear governing equations of the wind field was demonstrated to be quite challenging, the linear models showed great promise delivering a simple solution with good approximation to the wind field, and can be readily adopted for engineering applications. For instance it can be implemented in the Monte Carlo technique for rapid tropical-cyclone risk assessment. This study aims to develop a height-resolving, linear analytical model of the boundary layer winds in a moving tropical cyclone. The wind velocity is expressed as the summation of two components, namely gradient wind in the free atmosphere and frictional component near the ground surface. The gradient wind was derived straightforwardly, while the frictional component was obtained based on the scale analysis of the fully non-linear Navier-Stokes equations. The variation of wind field with respect to the angular coordinate was highlighted since its contribution to the surface wind speed and associated spatial distribution cannot be ignored in the first-order approximation. The results generated by the present model are consistent with tropical cyclone observations.

Key words: Height-resolving wind-field model, Boundary layer, Scale analysis, Tropical cyclone.

23 **1. Introduction**

24 Tropical cyclone-related natural hazards are well known for resulting in the largest contribution to
25 insured losses each year. High winds in the tropical cyclone boundary layer cause widespread
26 damage to life and property in coastal areas. This situation has become more challenging due to
27 the changing climate and increasing coastal population density. A mature tropical cyclone typically
28 consists of four dynamically distinct parts, namely a boundary layer, a region above the boundary
29 layer (almost no radial motion), an updraft region, and a quiescent eye (Carrier et al. 1971). For
30 many engineering applications, only the boundary layer is concerned. Furthermore, in the
31 consideration of wind-induced damages, the dynamics of boundary layer, where the density
32 changes could be ignored, and the thermodynamics are usually independently examined, or weakly
33 coupled.

34 The dynamics of tropical cyclone boundary layer is essentially governed by the Navier-
35 Stokes equations of incompressible flow. The solution of dynamically coupled, intensively
36 interactive pressure and velocity fields is extremely challenging. In most of wind field models for
37 engineering applications, the pressure field (steady pressure gradient) is usually prescribed, either
38 based on the gradient wind equation (e.g., Shapiro 1983) or resulting from an empirical formula
39 (e.g., Schloemer 1954; Holland 1980). The wind speed could be simulated based on a slab (depth-
40 averaged) or height-resolving model. The slab wind field model has been widely applied to storm
41 surge modeling (e.g., Hubbert et al. 1991; Kennedy et al. 2012) and hurricane damage and loss
42 estimation (e.g., Florida Hurricane Loss Projection Model and HAZUS-MH Hurricane Model)
43 (Powell et al. 2005; Vickery et al. 2006) since the pioneering work of Chow (1971), Shapiro
44 (1983), and Thompson and Cardone (1996). However, there are several inherent limitations of the
45 slab model due to the vertical averaging of dynamic quantities in the boundary layer, as

46 comprehensively discussed by several studies (e.g., Kepert 2010a; 2010b). Among a series of
47 shortcomings, considering the boundary layer height as a constant and obtaining the near surface
48 winds based on empirical-based reduction factors may have considerable impact on the simulation
49 fidelity. Recently, Khare et al. (2009) has shown the linear height-resolving model is superior to
50 the slab model of tropical cyclone boundary layer based on the observation data of over-ocean
51 surface wind field.

52 The height-resolving model was originally derived by studying the tropical cyclone
53 boundary layer as a generalized Ekman problem (Haurwitz 1935; 1936; Rosenthal 1962). Later,
54 Yoshizumi (1968) integrated the storm movement into the model to account for the left-right
55 asymmetry of tropical cyclone wind field. On the other hand, Meng (1995; 1997) obtained similar
56 height-resolving solutions of boundary layer winds by carrying out perturbation analysis of the
57 Navier-Stokes equations. A so-called equivalent roughness length was introduced in Meng's
58 model (1995; 1997) to simultaneously take the terrain roughness and topographical effects into
59 account, whereas the estimation of the equivalent roughness length for each case is not
60 straightforward. It should be noted that the abovementioned height-resolving models are actually
61 linear solutions. The state-of-the-art tropical cyclone risk assessment is essentially based on the
62 analysis framework established by Russell (1971), where the Monte Carlo sampling method is
63 utilized. This indicates a large number of simulations of the tropical cyclone boundary layer wind
64 filed are usually needed. Although nonlinear effects may not be always insignificant especially for
65 local flows (Kepert 2001; Kepert and Wang 2001), it is believed that the linear height-resolving
66 wind field model is a reasonable choice for many engineering applications (including risk
67 assessment) due to its high simulation efficiency.

68 In this study, the linear height-resolving wind field model was extracted based on the scale
69 analysis of the Navier-Stokes equations. The analytical expressions for wind speed components of
70 the tropical cyclone boundary layer were derived since they could facilitate the interpretation of
71 underlying physics. The obtained linear governing equations for boundary layer wind field are
72 different from those of previously widely-discussed models (Meng et al. 1995; Kepert 2001). For
73 example, several new terms that account for the contributions from the azimuthal variation of
74 velocity components are retained. It was demonstrated that the modification of surface wind speed
75 and its spatial distribution resulting from these new terms cannot be ignored. The new linear
76 height-resolving wind field model was validated using the observation data obtained during a
77 typhoon and a hurricane, respectively. Compared with Meng's simulation, the new proposed
78 model shows improved representation of the wind field in the tropical cyclone boundary layer.

79 2. Boundary Layer Wind Model

80 In the boundary layer of a tropical cyclone, the horizontal momentum equations are typically
81 solved with a prescribed pressure distribution and a constant air density. Thus, the governing
82 equation of the wind field of a tropical cyclone is:

$$83 \frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v} = -\frac{1}{\rho} \nabla p - f \mathbf{k} \times \mathbf{v} + \mathbf{F} \quad (1)$$

84 where \mathbf{v} = wind velocity; t = time; f = Coriolis parameter; \mathbf{k} = unit vector in the vertical direction;
85 ρ = air density; \mathbf{F} = frictional force; and p = Holland pressure expressed as:

$$86 p = p_c + \Delta p \exp\left[-\left(r_m / r\right)^B\right] \quad (2)$$

87 where p_c = central pressure; Δp = central pressure difference; r_m = radius of maximum winds; r =
88 radial distance from the tropical cyclone center; and B = Holland's radial pressure parameter. It

89 should be noted that bold character denote a vector. Eq. (1) is supplemented by the continuity
90 equation:

$$91 \quad \frac{d\rho}{dt} = -\rho \nabla \cdot \mathbf{v} \quad (3)$$

92 which becomes for an incompressible flow:

$$93 \quad \nabla \cdot \mathbf{v} = 0 \quad (4)$$

94 To solve the governing equation of motion, the wind velocity (\mathbf{v}) is expressed as the
95 summation of the gradient wind in the free atmosphere (\mathbf{v}_g) and the frictional component near the
96 ground surface (\mathbf{v}'):

$$97 \quad \mathbf{v} = \mathbf{v}_g + \mathbf{v}' \quad (5)$$

98 Consequently two separate equations can be derived:

$$99 \quad \frac{\partial \mathbf{v}_g}{\partial t} + \mathbf{v}_g \cdot \nabla \mathbf{v}_g = -\frac{1}{\rho} \nabla p - f \mathbf{k} \times \mathbf{v}_g \quad (6a)$$

$$100 \quad \frac{\partial \mathbf{v}'}{\partial t} + \mathbf{v}' \cdot \nabla \mathbf{v}' + \mathbf{v}' \cdot \nabla \mathbf{v}_g + \mathbf{v}_g \cdot \nabla \mathbf{v}' = -f \mathbf{k} \times \mathbf{v}' + \mathbf{F} \quad (6b)$$

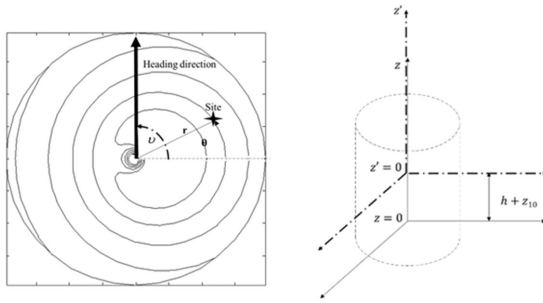
101 Similar to Meng et al. (1995), the gradient wind pattern \mathbf{v}_g is assumed to move in the free
102 atmosphere, at the translation velocity of the tropical cyclone \mathbf{c} , thus the unsteady term can be
103 expressed as: $\frac{\partial \mathbf{v}_g}{\partial t} = -\mathbf{c} \cdot \nabla \mathbf{v}_g$. On the other hand, the unsteady term $\frac{\partial \mathbf{v}'}{\partial t}$ is significantly smaller than
104 the turbulent viscosity and inertia terms in the tropical cyclone boundary layer, and hence
105 neglected.

106

107

108 **2.1 Gradient wind velocity**

109 The cylindrical coordinate system (r, θ, z) whose origin is located at the center of the tropical
 110 cyclone is used to solve the governing equations. Fig. 1 illustrates the necessary parameters needed
 111 for the present study, where ν = approach angle (counterclockwise positive from the East); θ =
 112 azimuthal angle (counterclockwise positive from the East); h = mean height of the roughness
 113 elements; $z_{10} = 10m$ height above the mean height of roughness elements; and z' = new vertical
 114 coordinate used as the base of the computation scheme where $z' = 0$ is located at $h + z_{10}$.



115
116 **Fig. 1.** Cylindrical coordinate system

117 Accordingly, Eq. (6a) could be expressed as:

118
$$\left(v_{rg} - c \cdot \cos(\theta - \nu)\right) \frac{\partial v_{rg}}{\partial r} + \frac{v_{\theta g} + c \cdot \sin(\theta - \nu)}{r} \frac{\partial v_{rg}}{\partial \theta} - \frac{v_{\theta g}^2}{r} - \frac{v_{\theta g} \cdot c \cdot \sin(\theta - \nu)}{r} = -\frac{1}{\rho} \frac{\partial p}{\partial r} + f v_{\theta g} \quad (7a)$$

119
$$\left(v_{rg} - c \cdot \cos(\theta - \nu)\right) \frac{\partial v_{\theta g}}{\partial r} + \frac{v_{\theta g} + c \cdot \sin(\theta - \nu)}{r} \frac{\partial v_{\theta g}}{\partial \theta} + \frac{v_{\theta g} v_{rg}}{r} + \frac{v_{rg} \cdot c \cdot \sin(\theta - \nu)}{r} = -f v_{rg} \quad (7b)$$

120 The radial velocity v_{rg} could be derived based on the continuity equation, which, due to its
 121 insignificant effect, is usually disregarded as suggested by Meng et al. (1995). Hence, $v_{\theta g}$ could
 122 be derived from Eq. (7a) as:

$$123 \quad v_{\theta g} = \frac{(-c \sin(\theta - \nu) - fr)}{2} + \left[\frac{(-c \sin(\theta - \nu) - fr)^2}{4} + \frac{r}{\rho} \frac{\partial p}{\partial r} \right]^{1/2} \quad (8)$$

124 The solution of gradient wind velocity is straightforward and was discussed in detail by several
125 researchers (e.g. Georgiou 1986; Meng et al. 1995).

126 2.2 Frictional wind velocity

127 The absolute angular velocity and vertical component of absolute vorticity of the gradient wind
128 are introduced first and given respectively by the following formulas (Meng et al. 1995):

$$129 \quad \xi_g = \frac{2v_{\theta g}}{r} + f \quad (9a)$$

$$130 \quad \xi_{ag} = \frac{\partial v_{\theta g}}{\partial r} + \frac{v_{\theta g}}{r} + f \quad (9b)$$

131 As a result, Eq. (6b) in the cylindrical coordinate becomes:

$$132 \quad u' \frac{\partial u'}{\partial r} + \frac{v_{\theta g} + v'}{r} \frac{\partial u'}{\partial \theta} + w \frac{\partial u'}{\partial z} - \frac{v'^2}{r} - \xi_g v' = K \left[\nabla^2 u - \frac{1}{r^2} \left(u + 2 \frac{\partial v}{\partial \theta} \right) \right] \quad (10a)$$

$$133 \quad u' \frac{\partial v'}{\partial r} + \frac{v_{\theta g} + v'}{r} \frac{\partial v'}{\partial \theta} + w \frac{\partial v'}{\partial z} + \frac{u'v'}{r} + \xi_{ag} u' + \frac{v'}{r} \frac{\partial v_{\theta g}}{\partial \theta} = K \left[\nabla^2 v - \frac{1}{r^2} \left(v - 2 \frac{\partial u}{\partial \theta} \right) \right] \quad (10b)$$

134 where (u, v, w) = velocity vector; and u', v' are the frictional components of the wind velocity. The
135 right hand side of Eqs. (10a) and (10b) represents the radial and azimuthal frictional force
136 components, respectively. The turbulent diffusivity K is assumed to be constant in this study.

137 The continuity equation can be expressed in the cylindrical coordinates as:

$$138 \quad \frac{1}{r} \frac{\partial(ru)}{\partial r} + \frac{1}{r} \frac{\partial v}{\partial \theta} + \frac{\partial w}{\partial z} = 0 \quad (11)$$

139 Then the vertical component of the wind velocity can be derived:

140
$$w = -\frac{1}{r} \frac{\partial}{\partial r} \left(r \int_0^z u dz \right) - \frac{1}{r} \frac{\partial}{\partial \theta} \left(\int_0^z v dz \right) \quad (12)$$

141 Solving Eqs. (10a) and (10b) analytically is extremely difficult. In this study, they will be first
 142 simplified using the scale analysis approach.

143 **2.3 Scale analysis**

144 In this section the same notation used by Vogl and Smith (2009) will be adopted for denoting the
 145 scales of various quantities in Eqs. (10a) and (10b), as showed in Table 1:

146 **Table 1.** Scales of various quantities

| Quantity | u' | v' | $v_{\theta g}$ | v | w | r | z | ξ_g | ξ_{ag} | Change in p |
|----------|------|------|----------------|-----|-----|-----|-----|---------|------------|---------------|
| Scale | U' | V' | V_g | V | W | R | Z | Ξ | Λ | Δp |

147
 148 Six non-dimensional parameters were introduced in Vogl and Smith (2009), namely Reynolds
 149 number $Re = VZ/K$; local Rossby numbers $Ro_\Lambda = V/(R\Lambda)$ and $Ro_\Xi = V/(R\Xi)$; Swirl
 150 parameters $S_{u'} = U'/V$ and $S_{v'} = V'/V$; and $A = Z/R$ that is the aspect ratio of the boundary-layer
 151 depth to the radial scale. It should be noted that based on the continuity equation namely Eq. (11),
 152 the following result is obtained $U'/R \sim W/Z$.

153 The scale analysis for Eqs. (10a) and (10b) is presented in Tables 2 and 3, respectively. The non-
 154 dimensional form is obtained by multiplying Eqs. (10a) and (10b) by $1/V'\Xi$ and $1/U'\Lambda$,
 155 respectively. It is worth mentioning that the terms $2\frac{K}{r^2} \frac{\partial u}{\partial \theta}$ and $-2\frac{K}{r^2} \frac{\partial v}{\partial \theta}$ are not included in

156 Tables 2 and 3 since they have the same scale order as $K\left(\nabla_h^2 u - \frac{u}{r^2}\right)$ and $K\left(\nabla_h^2 v - \frac{v}{r^2}\right)$,
 157 respectively.

158

Table 2. Scale analysis of Eq. (10a)

| Quantity | Scale | Normalized form |
|--|----------------------------|---------------------------------------|
| $u' \frac{\partial u'}{\partial r}$ | U'^2 / R | $S_{U'}^2 S_{V'}^{-1} Ro_{\Xi}$ |
| $\frac{v'_{\theta g} + v'}{r} \frac{\partial u'}{\partial \theta}$ | $\frac{(V'_g + V')}{R} U'$ | $S_{V'}^{-1} S_{U'} Ro_{\Xi}$ |
| $w' \frac{\partial u'}{\partial z}$ | $W U' / Z$ | $S_{U'}^2 S_{V'}^{-1} Ro_{\Xi}$ |
| $-\frac{v'^2}{r}$ | V'^2 / R | $S_{V'} Ro_{\Xi}$ |
| $-\xi'_g v'$ | $\Xi V'$ | 1 |
| $K \left(\nabla_h^2 u - \frac{u}{r^2} \right)$ | $K U / R^2$ | $A (R_e S_{V'})^{-1} S_{U'} Ro_{\Xi}$ |
| $K \frac{\partial^2 u'}{\partial z^2}$ | $K U' / Z^2$ | $(A R_e S_{V'})^{-1} S_{U'} Ro_{\Xi}$ |

159
160**Table 3.** Scale analysis of Eq. (10b)

| Quantity | Scale | Normalized form |
|--|----------------------------|---|
| $u' \frac{\partial v'}{\partial r}$ | $U V' / R$ | $S_{V'} Ro_{\Lambda}$ |
| $\frac{v'_{\theta g} + v'}{r} \frac{\partial v'}{\partial \theta}$ | $\frac{(V'_g + V') V'}{R}$ | $S_{V'} S_{U'}^{-1} Ro_{\Lambda}$ |
| $w' \frac{\partial v'}{\partial z}$ | $W V' / Z$ | $S_{V'} Ro_{\Lambda}$ |
| $\frac{u' v'}{r}$ | $U V' / R$ | $S_{V'} Ro_{\Lambda}$ |
| $\xi'_{\theta g} u'$ | $\Lambda U'$ | 1 |
| $\frac{v'}{r} \frac{\partial v'_{\theta g}}{\partial \theta}$ | $\frac{V V'}{R}$ | $S_{V'} S_{U'}^{-1} Ro_{\Lambda}$ |
| $K \left(\nabla_h^2 v - \frac{v}{r^2} \right)$ | $K V / R^2$ | $A (R_e S_{U'})^{-1} Ro_{\Lambda}$ |
| $K \frac{\partial^2 v'}{\partial z^2}$ | $K V' / Z^2$ | $(A R_e S_{U'})^{-1} S_{V'} Ro_{\Lambda}$ |

161

162 Typical values for the boundary layer such as the boundary layer height and the turbulent
 163 diffusivity K were considered to assess the magnitude of each term. As a result, a new set of
 164 equations could be obtained:

165
$$u' \frac{\partial u'}{\partial r} + \frac{v_{\theta g} + v'}{r} \frac{\partial u'}{\partial \theta} + w \frac{\partial u'}{\partial z} - \frac{v'^2}{r} - \xi_g v' = K \frac{\partial^2 u'}{\partial z^2} \quad (13a)$$

166
$$u' \frac{\partial v'}{\partial r} + \frac{v_{\theta g} + v'}{r} \frac{\partial v'}{\partial \theta} + w \frac{\partial v'}{\partial z} + \frac{u'v'}{r} + \xi_{ag} u' + \frac{v'}{r} \frac{\partial v_{\theta g}}{\partial \theta} = K \frac{\partial^2 v'}{\partial z^2} \quad (13b)$$

167 Solving the above nonlinear equations is demonstrated quite expensive. For engineering purposes,
168 the nonlinear equations could be simplified by linearization techniques.

169 **3. Linear height-resolving model**

170 **3.1 Governing equations**

171 Equations (13a) and (13b) could be linearized as:

172
$$\frac{v_{\theta g}}{r} \frac{\partial u'}{\partial \theta} - \xi_g v' = K \frac{\partial^2 u'}{\partial z^2} \quad (14a)$$

173
$$\frac{v_{\theta g}}{r} \frac{\partial v'}{\partial \theta} + \xi_{ag} u' + \frac{v'}{r} \frac{\partial v_{\theta g}}{\partial \theta} = K \frac{\partial^2 v'}{\partial z^2} \quad (14b)$$

174 To solve the above-mentioned governing equations, the boundary conditions at the upper
175 atmosphere (15a) and above the ground surface (15b) need to be respectively employed:

176
$$v'|_{z \rightarrow \infty} = 0 \quad (15a)$$

177
$$\rho K \left. \frac{\partial v'}{\partial z} \right|_{z'=0} = \rho C_d |v_s| v_s \quad (15b)$$

178 where v' = frictional component of the wind velocity; v_s = total wind velocity near the ground
179 surface; and C_d = drag coefficient.

180 New variables are introduced to simplify the solution namely: $\alpha = \frac{1}{2K} \xi_g$; $\beta = \frac{1}{2K} \xi_{ag}$;

181 $\gamma = \frac{1}{2K} \frac{v_{\theta g}}{r}$; and $\phi = \frac{1}{2Kr} \frac{\partial v_{\theta g}}{\partial \theta}$. In addition, the variable ω used in Kepert (2001) is employed:

182
$$\omega = \sqrt{\frac{\beta}{\alpha}} u' + i v' \quad (16)$$

183 It should be noted that the equations to be solved in this study are different from those in Kepert
 184 (2001). Specifically, the equations have an additional term $\frac{v'}{r} \frac{\partial v_{\theta g}}{\partial \theta}$ since the gradient wind velocity
 185 is considered to be dependent not only on the radial coordinate but the azimuthal one as well.
 186 Furthermore, the translational velocity is integrated into the gradient wind velocity based on the
 187 assumption that the wind pattern v_g moves at the translation velocity of the tropical cyclone c ,
 188 while Kepert (2001) assumed a symmetric case of the gradient wind. Finally, the absolute angular
 189 velocity ξ_g and vertical component of absolute vorticity of the gradient wind ξ_{ag} are exhibiting
 190 not only a radial variation but and azimuthal one as well. Accordingly, Eqs. (14a) and (14b) could
 191 be expressed as:

192
$$2\gamma \frac{\partial \omega}{\partial \theta} + 2i\sqrt{\alpha\beta}\omega - \frac{\partial^2 \omega}{\partial z^2} + 2\phi \text{Im}(\omega) = 0 \quad (17)$$

193 $\text{Im}(\omega)$ can be written as $i(-\omega + \omega^*)/2$ (where * indicates a complex conjugate), then Eq. (17)
 194 becomes:

195
$$2\gamma \frac{\partial \omega}{\partial \theta} + i(2\sqrt{\alpha\beta} - \phi)\omega + \phi i \omega^* - \frac{\partial^2 \omega}{\partial z^2} = 0 \quad (18)$$

196 The corresponding boundary condition expressed by Eq. (15b) could be decomposed using the
 197 cylindrical coordinates into:

198
$$K \frac{\partial u'}{\partial z} = C_d \sqrt{(u' + v_{rg})^2 + (v_{\theta g} + v')^2} (v_{rg} + u') \quad (19a)$$

199
$$K \frac{\partial v'}{\partial z} = C_d \sqrt{(u' + v_{rg})^2 + (v_{\theta g} + v')^2} (v_{\theta g} + v') \quad (19b)$$

200 Eqs. (19a) and (19b) can be linearized considering $u', v' \ll v_{\theta g}$, hence, the final system to be solved

201 becomes:

$$202 \quad 2\gamma \frac{\partial \omega}{\partial \theta} + i(2\sqrt{\alpha\beta} - \phi)\omega + \phi i \omega^* - \frac{\partial^2 \omega}{\partial z^2} = 0 \quad (20a)$$

$$203 \quad K.Re(\omega'(0)) = C_d v_{\theta g} (v_{rg} + Re(\omega(0))) \quad (20b)$$

$$204 \quad K.Im(\omega'(0)) = C_d v_{\theta g} (v_{\theta g} + 2Im(\omega(0))) \quad (20c)$$

205 3.2 Analytical solutions

206 To solve system (20), ω is expanded as Fourier series in azimuth, i.e., $\omega(\theta, z') = \sum_{k=-\infty}^{\infty} a_k(z') e^{ik\theta}$

207 where $a_k(z')$ is a complex coefficient (Kepert 2001). Then Eq. (20a) becomes:

$$208 \quad \sum_{k=-\infty}^{\infty} \left[i(2\gamma k + 2\sqrt{\alpha\beta} - \phi) a_k(z') - a_k''(z') + i\phi a_{-k}^*(z') \right] e^{ik\theta} = 0 \quad (21)$$

209 The complex coefficient a_k can be expressed as $a_k(z') = A_k \exp(q_k z')$, and hence the following
210 equation could be obtained:

$$211 \quad A_k q_k^2 = i \left[2\gamma k + 2\sqrt{\alpha\beta} - \phi \right] A_k + i\phi A_{-k}^* \quad (22)$$

212 Since the wavenumbers higher than unity are not excited for a linear model, only the cases of
213 $k = -1, k = 0$, and $k = 1$ need to be considered.

214 Substituting ω expression into Eqs. (20b) and (20c) leads to:

$$215 \quad Re \left\{ \sqrt{\frac{\alpha}{\beta}} \sum_{k=-\infty}^{\infty} A_k \left[\frac{C_d v_{\theta g}}{K} - q_k \right] e^{ik\theta} \right\} = 0 \quad (23a)$$

$$216 \quad \text{Im} \left\{ \sum_{k=-\infty}^{\infty} A_k \left[\frac{2C_d v_{\theta g}}{K} - q_k \right] e^{ik\theta} + i \frac{C_d v_{\theta g}^2}{K} \right\} = 0 \quad (23b)$$

217 After a series of manipulations, one could obtain the frictional velocity components u' and v' (see
 218 detailed derivation in Appendix A). Fig. 2 presents a step-by-step procedure in the development
 219 of solutions for the proposed wind field model, where X_1 , X_2 , X_3 , X_4 and η are five parameters
 220 defined, respectively, as follows:

$$221 \quad X_1 = \left[q_0 + \frac{fr}{K} C_d - \frac{2\eta C_d}{K} - \frac{c^2 C_d^2}{4K^2 (q_1 - q_{-1}^*)} + \frac{c^2 C_d^2}{4K^2 (q_1^* - q_{-1})} \right] \quad (24a)$$

$$222 \quad X_2 = \left[-q_0^* - \frac{fr}{K} C_d + \frac{2\eta C_d}{K} - \frac{c^2 C_d^2}{4K^2 (q_1 - q_{-1}^*)} + \frac{c^2 C_d^2}{4K^2 (q_1^* - q_{-1})} \right] \quad (24b)$$

$$223 \quad X_3 = -2 \frac{iC_d}{K} \left(\eta - \frac{fr}{2} \right)^2 \quad (24c)$$

$$224 \quad X_4 = - \left[-q_0 - \frac{fr}{2K} C_d + \frac{\eta C_d}{K} \right] / \left[-q_0^* - \frac{fr}{2K} C_d + \frac{\eta C_d}{K} \right] \quad (24d)$$

$$225 \quad \eta = \left[\frac{(-c \sin(\theta - \nu) - fr)^2}{4} + \frac{r}{\rho} \frac{\partial p}{\partial r} \right]^{-1/2} \quad (24e)$$

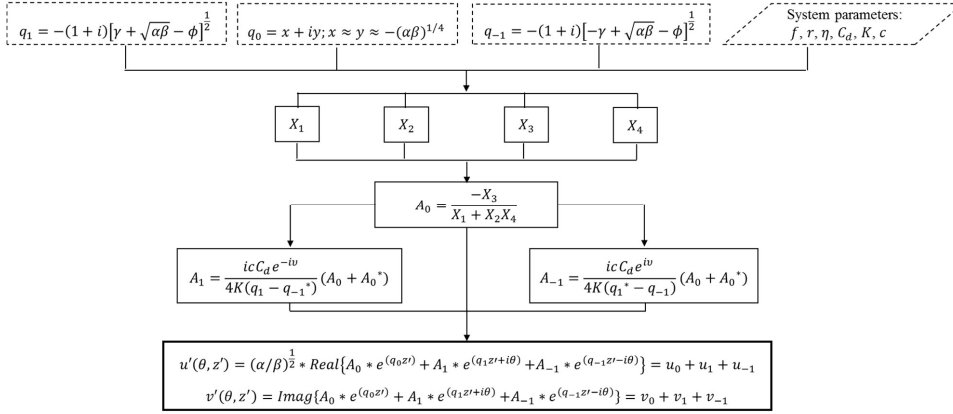


Fig. 2. u' and v' components of wind field model

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235 4. Improved Representation

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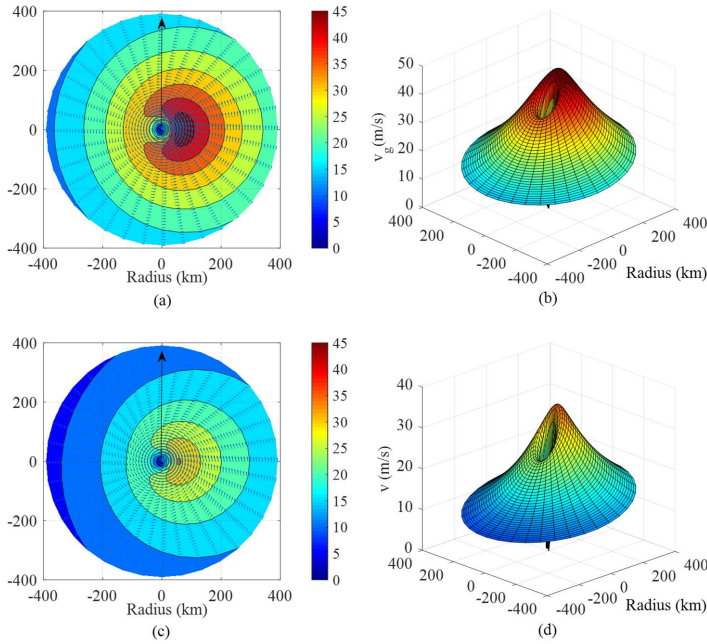
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241 4.1 Spatial Distribution of the tropical cyclone wind field

242 To investigate the tropical cyclone spatial distribution, a case study is analyzed using the following
243 parameters: $\Delta p = 60 \text{ hpa}$; $r_m = 80 \text{ km}$; $c = 15 \text{ m/s}$; $\nu = 90^\circ \text{C}$; Latitude $\psi = 32.8^\circ$; Longitude
244 $\lambda = 129.7^\circ$; $B = 1$; $z_0 = 0.1 \text{ m}$; $K = 100 \text{ m}^2/\text{s}$; and $\rho = 1.2 \text{ N s}^2/\text{m}^4$.

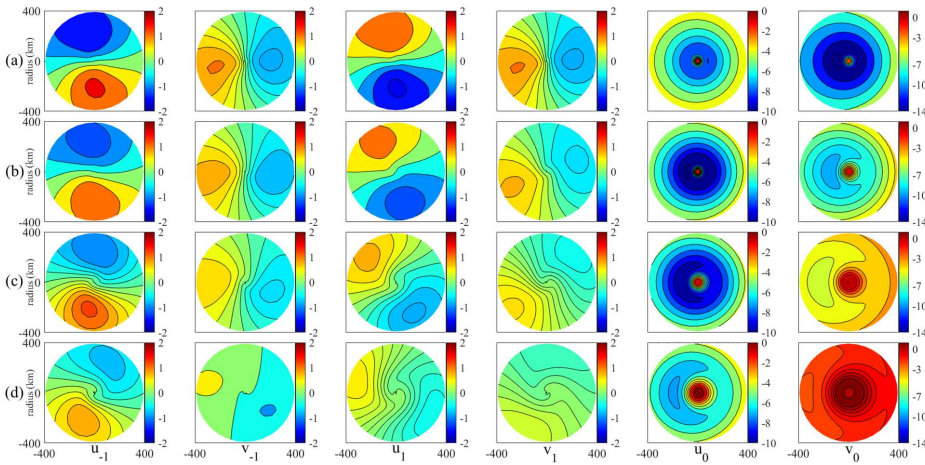
245 Figures (3a) and (3b) depict the contours of the gradient wind velocity and its three-
246 dimensional shaded surface, respectively. It can be concluded that the maximum wind speed is
247 located at r_m and the difference of the maximum wind velocity between the right-hand side and
248 left-hand side is almost equal to the translational hurricane velocity c . Figures (3c) and (3d)
249 illustrate the contours of the surface wind velocity at the height around 10 m and its three-
250 dimensional shaded surface, respectively. The maximum wind speed is located on the right rear
251 quadrant for this specific case. The exact location of the maximum wind speed depends on a
252 number of factors such as the translation of the storm and the surface friction. Recently, Li and
253 Hong (2014) inspected the location of the maximum wind speed based on the H*Wind data where
254 totally 489 snapshots from 45 hurricanes, occurred during 2002 to 2013 were utilized. The H*Wind
255 was essentially developed by the Hurricane Research Division of the National Oceanic and
256 Atmospheric Administration (NOAA). It integrates several sources of wind data such as aircraft
257 reconnaissance, buoys, surface and remote sensing observations (Powell et al. 1998). To ensure
258 quality control, collected data are post-processed. The H*Wind database provides maximum 1-
259 min sustained surface wind speed corresponding to a marine or open terrain over land exposures.
260 It is found that the maximum wind speed for 84% of the snapshots was located on the right side of
261 the storm motion. More specifically, 58% of these snapshots present the maximum wind speed at
262 the right rear quadrant and 42% at the right front quadrant.



263
 264 **Fig. 3.** Spatial distribution of the wind speed: (a) Contour of the gradient wind speed; (b) three-dimensional shaded
 265 surface of the gradient wind speed; (c) Contour of the surface wind speed at z around 10m; (d) three-dimensional
 266 shaded surface of the surface wind speed

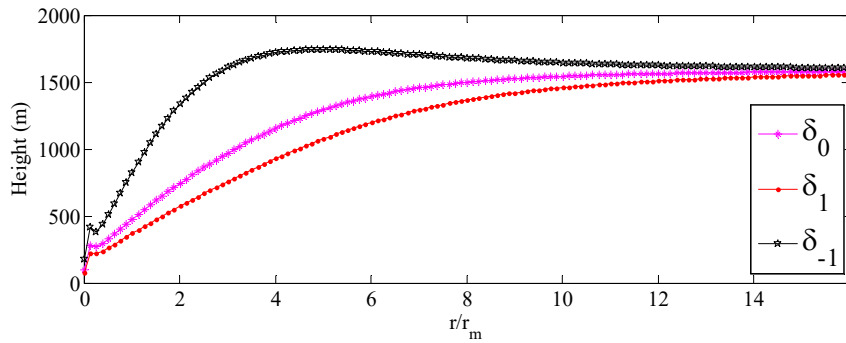
267 Figure (4) shows the u_0 , v_0 , u_1 , v_1 , u_{-1} , and v_{-1} terms in the frictional component of the
 268 boundary layer wind velocity using the same tropical cyclone data as in Fig. (3). Several
 269 conclusions related to the behavior of u_0 , v_0 , u_1 , v_1 , u_{-1} , and v_{-1} terms, can be drawn. First, it is
 270 obvious that all frictional velocity components decay with height. Second, while u_1 and v_1 rotate
 271 counterclockwise u_{-1} and v_{-1} rotate clockwise. This behavior can be attributed to the signs of the
 272 imaginary parts related to z' and θ in the complex exponential argument (i.e., $e^{(q_1 z' + i\theta)}$ and
 273 $e^{(q_{-1} z' - i\theta)}$) where opposite signs results in a counterclockwise rotation and vice-versa. Finally, due to
 274 the asymmetry of the tropical cyclone wind field, the proposed model allows the vertical length

275 scale for the depth of the boundary layer to exhibit not only radial variation but azimuthal one as
 276 well, which is not the case for Kepert's model (2001) where it varies only radially. The depth scale
 277 related to (u_0, v_0) , (u_1, v_1) and (u_{-1}, v_{-1}) are $\delta_0 = 1/(\alpha\beta)^{1/4}$, $\delta_1 = 1/[\gamma + \sqrt{\alpha\beta} - \phi]^{1/2}$ and
 278 $\delta_{-1} = 1/[-\gamma + \sqrt{\alpha\beta} - \phi]^{1/2}$, respectively. For instance, δ_0 is a function of α and β which are
 279 proportional to the absolute angular velocity and vertical component of absolute vorticity of the
 280 gradient wind, respectively ($\alpha = \frac{1}{2K}\xi_g$ and $\beta = \frac{1}{2K}\xi_{ag}$). These two variables are varying
 281 radially and azimuthally due to the asymmetric structure of a moving tropical cyclone, therefore
 282 δ_0 exhibits an asymmetric distribution. Since the rotational rate of each frictional component with
 283 height is dependent on the corresponding depth scale, it can be concluded from the comparison of
 284 δ_1 and δ_{-1} that (u_1, v_1) rotates faster than (u_{-1}, v_{-1}) , as indicated in the simulation results of Fig. 4.



286 **Fig. 4.** Frictional component of the boundary layer wind velocity u_0 , v_0 , u_1 , v_1 , u_{-1} , and v_{-1} at $z'=10m$
 287 (a); $z'=200m$ (b); $z'=500m$ (c); and $z'=1000m$ (d)

288 Figure 5 depicts the radial variation of the three depth scales δ_0 , δ_1 , δ_{-1} . It can be
 289 concluded that for larger radii all depth scales reduce to the classical Ekman height scale (namely
 290 $\delta = \sqrt{2K/f}$), where the Rossby number that measures the magnitude of the acceleration
 291 compared to the Coriolis force is sufficiently small ($Ro \ll 1$) ($Ro \ll 1$). On the other hand, the
 292 boundary layer depth decreases toward the center of the tropical cyclone (large Rossby numbers)
 293 due to the effects of the storm rotation that results in predominated inertial and centrifugal forces,
 294 as highlighted by several researchers (e.g., Rosenthal 1962; Kepert 2001). The results shown in
 295 Fig. 5 are consistent with several numerical simulations and observations conducted by several
 296 researchers (e.g., Kepert 2001; Kepert and Wang 2001) where the boundary layer depth varies
 297 from a few hundred meters near from the inner core of the tropical cyclone to 1-3 km at larger
 298 radii. Table 4 depicts the azimuthal variation of the depth scale for all frictional components.
 299 Clearly, all depth scales depend on the azimuthal coordinate and thus should be taken into
 300 consideration to enhance the simulation of the real behavior of a moving tropical cyclone.



301
 302 Fig. 5. Radial variation of the depth scale of the boundary layer
 303

304 **Table 4.** Boundary layer depth scale at $r = r_m$ (using the same tropical cyclone data of section 4.1)

| $\theta(^{\circ})$ | $\delta_0(m)$ | $\delta_1(m)$ | $\delta_{-1}(m)$ |
|--------------------|---------------|---------------|------------------|
| 0 | 477.2 | 369.7 | 826.0 |
| 30 | 482.3 | 367.4 | 769.2 |
| 60 | 496.4 | 374.1 | 749.0 |
| 90 | 516.2 | 388.1 | 761.5 |
| 120 | 536.3 | 406.0 | 800.6 |
| 150 | 551.3 | 423.1 | 858.9 |
| 180 | 556.7 | 434.6 | 929.1 |
| 210 | 551.3 | 437.4 | 1002.7 |
| 240 | 536.3 | 430.6 | 1062.0 |
| 270 | 516.2 | 415.9 | 1073.7 |
| 300 | 496.4 | 397.4 | 1015.0 |
| 330 | 482.3 | 380.4 | 916.0 |
| 360 | 477.2 | 369.7 | 826.0 |

305

306 **4.2 Stationary tropical cyclone**

307 For a stationary tropical cyclone, the translational velocity $c = 0$, and hence $A_1 = A_{-1} = 0$.

308 Accordingly, the frictional wind velocity components are degenerated into

309 $u'(\theta, z') = (\alpha/\beta)^{\frac{1}{2}} * Real\{A_0 * \exp(q_0 z')\}$ and $v'(\theta, z') = Imag\{A_0 * \exp(q_0 z')\}$ with

310 $q_0 = -(1+i)(\alpha\beta)^{1/4}$. It can be demonstrated in the stationary case that:

311 $X_1 = -(\alpha\beta)^{\frac{1}{4}}(1+i) - \frac{2C_D}{K} v_g$ (25a)

312 $X_2 = (\alpha\beta)^{\frac{1}{4}}(1-i) + \frac{2C_D}{K} v_g$ (25b)

313 $X_3 = -\frac{2iC_D}{K} v_g^2$ (25c)

314 $X_4 = -\left[(\alpha\beta)^{\frac{1}{4}}(1+i) + \frac{C_D}{K} v_g \right] / \left[(\alpha\beta)^{\frac{1}{4}}(1-i) + \frac{C_D}{K} v_g \right]$ (25d)

315 It should be noted that $v_{\theta g}$ was replaced by v_g since the gradient wind is symmetric for a stationary
 316 tropical cyclone. In Kepert (2001) a parameter X is defined as $C_D v_g / K(\alpha\beta)^{1/4}$. Then A_0 of a
 317 stationary tropical cyclone could be expressed in terms of X as:

$$318 \quad A_0 = -\frac{X[1+i(1+X)]v_g}{2X^2+3X+2} \quad (26)$$

319 The expression is exactly the same with that of Kepert (2001). This indicates that the specific case
 320 of the stationary tropical cyclone in the proposed model provides the same solution in Kepert
 321 (2001). This observation is expected since the gradient wind velocity for the case of a stationary
 322 tropical cyclone is symmetric, which was an essential assumption in Kepert (2001).

323 4.3 Comparison with Meng's model

324 The wind field model in Meng et al. (1995) is a special case of the present study. More specifically,
 325 the contribution associated with u_1 , u_{-1} , v_1 , v_{-1} and ϕ was disregarded by Meng et al. (1995).
 326 Accordingly, suppose $u_1 = u_{-1} = v_1 = v_{-1} = \phi = 0$, one has (see Appendix B for detailed derivation):

$$327 \quad u(\theta, z') = -(\alpha/\beta)^{1/2} e^{-\lambda z'} \left[\frac{\chi v_{\theta g}}{1+(1+\chi)^2} \cos(\lambda z') + \frac{\chi(\chi+1)v_{\theta g}}{1+(1+\chi)^2} \sin(\lambda z') \right] \quad (27a)$$

$$328 \quad v(\theta, z') = e^{-\lambda z'} \left[\frac{\chi(\chi+1)v_{\theta g}}{1+(1+\chi)^2} \cos(\lambda z') + \frac{\chi v_{\theta g}}{1+(1+\chi)^2} \sin(\lambda z') \right] \quad (27b)$$

$$329 \quad \text{where } \chi = \frac{C_d}{K\lambda} |v_s| = \frac{C_d}{K\lambda} \sqrt{v_{\theta s}^2 + v_{rs}^2}$$

330 The solution developed by Meng et al. (1995) has similar form:

$$331 \quad u = -\xi e^{-\lambda z'} * [D_2 \cos(\lambda z') - D_1 \sin(\lambda z')] \quad (28a)$$

332 $v = e^{(-\lambda z')} * [D_1 \cos(\lambda z') + D_2 \sin(\lambda z')]$ (28b)

333 where $D_1 = -\chi(\chi + 1)v_{\theta g} / (1 + (1 + \chi)^2)$; $D_2 = \chi v_{\theta g} / (1 + (1 + \chi)^2)$; and $\xi = (\beta/\alpha)^{\frac{1}{2}}$. Consequently, it
334 is shown that Meng's model (1995) can be derived from the proposed model under the above-
335 mentioned assumptions. By comparison between Eqs. (27a) and (28a), it is noted that there is an
336 error in Meng's original model, where the coefficient ξ should be replaced by $1/\xi$. This
337 modification of the wind speed due to this error may not be negligible, as will be illustrated in the
338 validation example.

339 5. Validation

340 To validate the new tropical cyclone wind field model, wind records from hurricane Fran (1996)
341 and typhoon Maemi T0314 (2003) are used.

342 5.1 Hurricane Fran

343 Fran was one of the worst hurricanes ever to be recorded in North Carolina. It ~~has~~ reached
344 hurricane status on ~~29th~~ August ~~29th, 1996~~ and was registered as a Category 3 storm. Hurricane
345 Fran made landfall on September ~~5th, 1996~~ on the North Carolina coast with an estimated central
346 pressure of 954 hpa ~~and a maximum sustained surface winds of 50 m/s~~. It created flooding in
347 the Carolinas, Virginia, West Virginia and Pennsylvania. Severe damage due to strong wind were
348 also recorded. Fran weakened to a tropical storm status over central North Carolina. Figure 6
349 depicts the location of the anemometer with respect to the storm track.

Mis en forme : Exposant

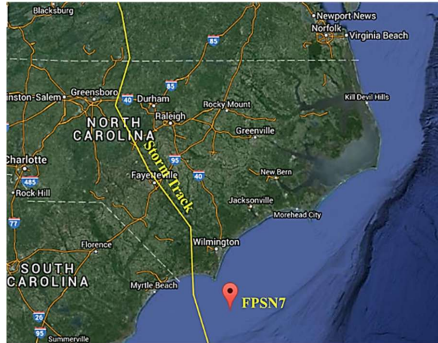


Fig. 6. Track of hurricane Fran and anemometer location

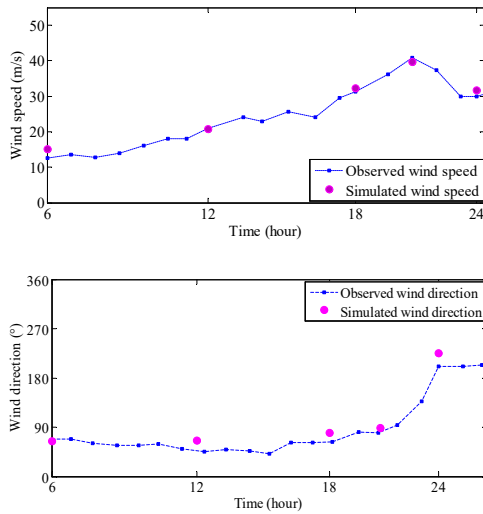
5.1.1 Hurricane Parameters

The necessary parameters for the simulation were recorded by the marine station FPSN7 from September 5th to September 6th. The station ID is 41013, located at (N33.44°,W77.74°). The National Hurricane Center's North Atlantic hurricane database (HURDAT) (Jarvinen et al. 1984) is the primary source of data. Typically, the parameters needed for the simulation are: ν approach angle; c translation velocity of the hurricane; p_c central pressure; Δp central pressure difference; r_m radius of maximum winds; ψ latitude; and λ longitude. The parameter r_m can be estimated using several sources in the literature (e.g., Powell et al. 1991; 1998). This information is supplemented by the H*Wind snapshots. There are several methods available in the literature to estimate the parameter B (e.g., Vickery et al. 2000a; Holland 2008). For hurricane Fran the value used was $B = 0.95$, which is the same value calculated by Vickery et al. (2000b). The period from September 5th to September 6th has known very strong winds from hurricane Fran especially in the eyewall region. Actually, the central pressure reached 952 hpa at 0600 UTC September 5th in which the hurricane center was located at (N29.8°,W76.7°) and slightly increased to reach

366 954 hpa at 0000 UTC September 6th where the hurricane center was located at (N33.7°,W78.0°)
367 with a maximum sustained surface winds of 55 m/s.

368 5.1.2 Hurricane Simulation

369 The observed wind speeds and directions have been compared with those obtained by the proposed
370 wind field model. It should be noted that all parameters are obtained from the 6-hour interval track
371 information provided by the HURDAT database. The results generated by the present model are
372 consistent with hurricane observations as depicted in the Fig. 7.



373

374

375 Fig. 7. Observed and simulated wind speeds (top) and directions (bottom) of Hurricane Fran

376 5.2 Typhoon Maemi

377 Typhoon Maemi T0314 had devastating impacts on Japan and South Korea. It was born as tropical
378 depression near Guam on September 5th, 2003. Later, it was evolved into a typhoon on ~~the~~
379 September 7th of September and then intensified to reach a Category 5 typhoon storm reaching
380 with a central pressure of 910 hpa . Maemi made landfall on September 12, ~~2003~~ on the south coast

381 of the Korean peninsula and lasted approximately 6 hours in the South Korea leading to extensive
 382 damage from wind, torrential rainfall and flooding. Wind records from typhoon Maemi will be
 383 employed to highlight the difference between the results generated by the present model and
 384 Meng's model.

385 5.2.1 Typhoon Parameters

386 The observation point is located at the observatory of Miyako Island in Okinawa prefecture
 387 (N24.8°, E125.3°). All the other necessary parameters for the typhoon simulation are obtained
 388 from Yoshida et al. (2008), and listed in Table 5 for completeness:

389 **Table 5.** Typhoon Maemi parameters, September 2003 (Yoshida et al., 2008)

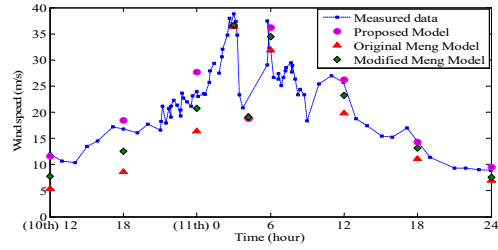
| Day/h | $\psi(^{\circ})$ | $\lambda(^{\circ})$ | $\nu(^{\circ})$ | $c(m/s)$ | $p_c(hPa)$ | $\Delta p(hPa)$ | $r_m(km)$ |
|-------|------------------|---------------------|-----------------|----------|------------|-----------------|-----------|
| 10 12 | 23.7 | 126.9 | 143.8 | 3.29 | 920 | 88 | 23.2 |
| 10 18 | 24.2 | 126.3 | 118.6 | 2.99 | 910 | 99 | 29.5 |
| 11 00 | 24.6 | 125.7 | 75.6 | 3.29 | 910 | 100 | 34.7 |
| 11 06 | 24.8 | 125.3 | 76.2 | 2.88 | 910 | 101 | 31.0 |
| 11 12 | 25.7 | 125.3 | 78.5 | 3.70 | 923 | 89 | 40.7 |
| 11 18 | 27.0 | 125.5 | 75.1 | 5.79 | 930 | 82 | 42.9 |
| 12 00 | 28.7 | 125.9 | 71.6 | 7.25 | 935 | 77 | 43.3 |

390

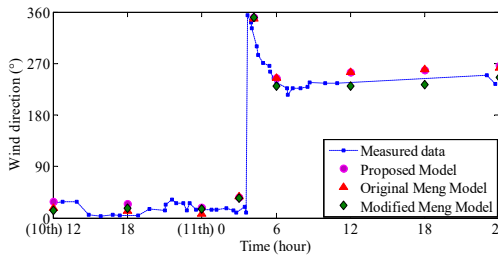
391 5.2.2 Typhoon Simulation

392 The observed wind speeds and directions have been compared with those obtained by the proposed
 393 wind field model. In addition, simulation results of the original and modified Meng's models are
 394 presented. As shown in Fig. 8, the results generated by the present model are consistent with the
 395 typhoon observations.

396



397



398

Fig. 8. Observed and simulated wind speeds (top) and directions (bottom) of typhoon Maemi

399

To further inspect the results, a comparison of wind speeds and directions between results

400

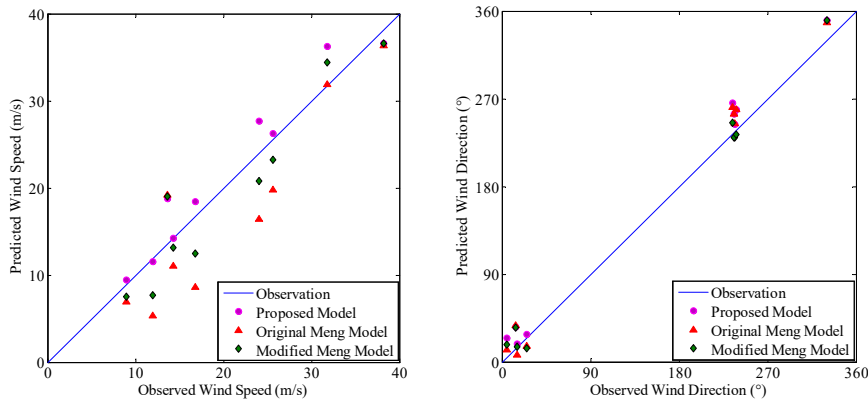
obtained from the observed data and using the present and Meng's models are presented in Fig. 9.

401

The corresponding correlation coefficients between the observations and various simulations for

402

wind speed and direction are calculated and presented in Table 6.



403

404

Fig. 9. Comparison of wind speeds (left) and directions (right)

405

406

Table 6. Correlation Coefficients of wind speed and direction

| Model | Wind speed | Wind direction |
|-------------------------|------------|----------------|
| Proposed Model | 0.973 | 0.998 |
| Meng's model (original) | 0.916 | 0.997 |
| Meng's model (modified) | 0.953 | 0.995 |

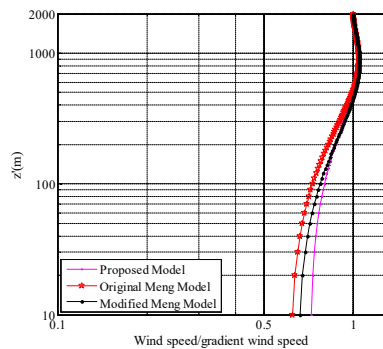
407

408 The results demonstrate that the wind velocities predicted by various models all match reasonably
409 well with the measured data, whereas the proposed model is superior to Meng's model.

410

411 5.3 Vertical wind speed profile

412 Vertical wind speed profiles given by the proposed model, original and modified Meng's models,
413 are plotted in Fig. 10 using the same tropical data of section 4.1. The location of the presented
414 wind speed profiles is selected at a distance equal to radius of maximum wind from the tropical
415 cyclone center and at zero degree from the East. The simulated wind profiles are normalized by
416 the gradient wind speed.



417

418

Fig. 10. Vertical profile of wind speed at $r = r_m$

419 As it can be remarked from Fig. 10, the surface wind speed is underestimated by Meng's
420 model compared to the proposed one. Same conclusion could be also obtained from the simulation
421 results in Fig. 8. Another interesting feature in the vertical wind speed profile of tropical cyclones
422 is the presence of a super-gradient-wind region, where the tangential winds are larger than the
423 gradient wind. A possible mechanism of this region was discussed by Kepert and Wang (2001),
424 where the super-gradient winds are attributed to the strong inward advection of angular
425 momentum. It is important to take the super-gradient wind region into account in the engineering
426 applications such as the wind design of high-rise buildings to ensure the target safety and
427 performance levels of civil infrastructures. As a result, the power or logarithmic law profiles and
428 hence the use of reduction factors to obtain the surface winds may not be appropriate for many
429 structures in the coastal regions.

430

431 **6. Concluding Remarks**

432 A linear height-resolving analytical model for the boundary layer winds of a translating tropical
433 cyclone has been developed and validated in this study. The construction of the new model started
434 from the Navier-Stokes equations coupled with the decomposition method. The obtained dynamic
435 system was then simplified based on the scale analysis. Finally, a simple height-resolving solution
436 of the wind field was analytically obtained by linearization techniques with the imposed boundary
437 conditions at upper atmosphere and above ground surface. The general solution for a stationary
438 tropical cyclone was found to be consistent with the one discussed by Kepert (2001). Also, it is
439 demonstrated that Meng et al. (1995) model is a special case of the present solution for the wind
440 field in tropical cyclones. Furthermore, it was demonstrated that the vertical length scale for the
441 depth of the boundary layer related to each component of the frictional wind velocity exhibited not
442 only radial variation but azimuthal one as well, which is conform to the asymmetric structure of a

443 moving tropical cyclone. The present model shows great promise in giving a first-order
 444 approximation of the boundary layer wind field of a tropical cyclone. Due to its simplicity and
 445 computational efficiency, the proposed wind field model could be easily implemented in the risk
 446 assessments of engineering applications.

447

448

APPENDIX A

449 To solve Eqs. (23a) and (23b), the gradient wind which depends on the radial and azimuthal
 450 coordinates, is expanded with respect to θ . Accordingly, $v_{\theta g}$ could be expressed as:

$$451 \quad v_{\theta g} = \frac{(-c \sin(\theta - \nu) - fr)}{2} + \left[\frac{(-c \sin(\theta - \nu) - fr)^2}{4} + \frac{r}{\rho} \frac{\partial p}{\partial r} \right]^{1/2} = \tau + \eta \quad (A.1)$$

452 It is shown that the η term is much less sensitive to the azimuthal coordinate compared to the term
 453 τ , as numerically verified in Table A.1.

454 **Table A.1** Comparison of η and τ (using the same tropical cyclone data of section 4.1)

| $\theta(^{\circ})$ | $r = r_m$ | | $r = 2r_m$ | |
|--------------------|-------------|-------------|-------------|-------------|
| | $\tau(m/s)$ | $\eta(m/s)$ | $\tau(m/s)$ | $\eta(m/s)$ |
| 0 | 4.34 | 43.11 | 1.18 | 38.96 |
| 30 | 3.33 | 43.01 | 0.17 | 38.94 |
| 60 | 0.59 | 42.89 | -2.57 | 39.02 |
| 90 | -3.16 | 43.00 | -6.32 | 39.35 |
| 120 | -6.91 | 43.44 | -10.07 | 40.22 |
| 180 | -10.66 | 44.19 | -13.82 | 41.32 |
| 270 | -3.16 | 43.00 | -6.32 | 39.45 |

455

456 Substituting Eq. (A.1) into Eqs. (23a) and (23b), where η is treated azimuthal coordinate
 457 independent, yields:

$$458 \quad Re \left\{ \sqrt{\frac{\alpha}{\beta}} \sum_{k=-\infty}^{\infty} A_k \left[\frac{C_d}{K} \left(\frac{ic}{4} (e^{i(\theta-\nu)} - e^{-i(\theta-\nu)}) - \frac{fr}{2} + \eta \right) - q_k \right] e^{ik\theta} \right\} = 0 \quad (A.2)$$

$$459 \quad \text{Im} \left\{ \sum_{k=-\infty}^{\infty} A_k \left[\frac{2C_d}{K} \left(\frac{ic}{4} (e^{i(\theta-\nu)} - e^{-i(\theta-\nu)}) - \frac{fr}{2} + \eta \right) - q_k \right] e^{ik\theta} + i \frac{C_d}{K} \left(\frac{ic}{4} (e^{i(\theta-\nu)} - e^{-i(\theta-\nu)}) - \frac{fr}{2} + \eta \right)^2 \right\} = 0 \quad (\text{A.3})$$

460 After several algebraic manipulations, one can obtain:

$$461 \quad \left[-q_0 - \frac{fr}{2K} C_d + \frac{\eta C_d}{K} \right] A_0 + \left[-q_0^* - \frac{fr}{2K} C_d + \frac{\eta C_d}{K} \right] A_0^* = 0 \quad (\text{A.4})$$

$$462 \quad A_1 + A_{-1}^* = 0 \quad (\text{A.5})$$

$$463 \quad \frac{iC_d c}{4K} e^{-i\nu} A_0 + \frac{iC_d c}{4K} e^{-i\nu} A_0^* + (-q_1 + q_{-1}^*) A_1 = 0 \quad (\text{A.6})$$

$$464 \quad \left[q_0 + \frac{fr}{K} C_d - \frac{2\eta C_d}{K} \right] A_0 + \left[-q_0^* - \frac{fr}{K} C_d + \frac{2\eta C_d}{K} \right] A_0^* + \frac{icC_d}{K} e^{i\nu} A_1 - \frac{icC_d}{K} e^{-i\nu} A_{-1} - 2 \frac{iC_d}{K} \left(\eta - \frac{fr}{2} \right)^2 = 0 \quad (\text{A.7})$$

465 Based on Eq. (A.5), q_1 and q_{-1} could be derived from Eq. (22), hence:

$$466 \quad q_1 = -(1+i) \left[\gamma + \sqrt{\alpha\beta} - \phi \right]^{\frac{1}{2}} \quad (\text{A.8})$$

$$467 \quad q_{-1} = -(1+i) \left[-\gamma + \sqrt{\alpha\beta} - \phi \right]^{\frac{1}{2}} \quad (\text{A.9})$$

468 A_1 and A_{-1} expressions could be derived from Eqs. (A.5) and (A.6), hence:

$$469 \quad A_1 = \frac{icC_d e^{-i\nu}}{4K (q_1 - q_{-1}^*)} (A_0 + A_0^*) \quad (\text{A.10})$$

$$470 \quad A_{-1} = -A_1^* = \frac{icC_d e^{i\nu}}{4K (q_1^* - q_{-1})} (A_0 + A_0^*) \quad (\text{A.11})$$

471 Substituting Eqs. (A.10) and (A.11) into Eq. (A.7) gives:

$$472 \quad X_1 A_0 + X_2 A_0^* + X_3 = 0 \quad (\text{A.12})$$

473 where X_1 , X_2 , and X_3 are expressed as follows:

$$474 \quad X_1 = \left[q_0 + \frac{fr}{K} C_d - \frac{2\eta C_d}{K} - \frac{c^2 C_d^2}{4K^2 (q_1 - q_{-1}^*)} + \frac{c^2 C_d^2}{4K^2 (q_1^* - q_{-1})} \right] \quad (\text{A.13})$$

$$475 \quad X_2 = \left[-q_0^* - \frac{fr}{K} C_d + \frac{2\eta C_d}{K} - \frac{c^2 C_d^2}{4K^2 (q_1 - q_{-1}^*)} + \frac{c^2 C_d^2}{4K^2 (q_1^* - q_{-1})} \right] \quad (\text{A.14})$$

$$476 \quad X_3 = -2 \frac{i C_d}{K} \left(\eta - \frac{fr}{2} \right)^2 \quad (\text{A.15})$$

477 On the other hand Eq. (A.4) could be expressed as:

$$478 \quad A_0^* = X_4 A_0 \quad (\text{A.16})$$

479 where X_4 is:

$$480 \quad X_4 = - \left[-q_0 - \frac{fr}{2K} C_d + \frac{\eta C_d}{K} \right] / \left[-q_0^* - \frac{fr}{2K} C_d + \frac{\eta C_d}{K} \right] \quad (\text{A.17})$$

481 As a result, the following equation could be obtained:

$$482 \quad q_0^2 \left[\left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} - q_0^* \right] = i \left[2\sqrt{\alpha\beta} - \phi \right] \left[\left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} - q_0^* \right] - i\phi \left[\left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} - q_0 \right] \quad (\text{A.18})$$

483 To determine q_0 , it could be expressed in the complex form as $q_0 = x + iy$ where $x < 0$ to be
 484 consistent with the boundary condition of Eq. (15a). Hence, the following system of equations is
 485 obtained:

$$486 \quad -x^3 + \left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} x^2 - xy^2 - \left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} y^2 + 2\sqrt{\alpha\beta} y = 0 \quad (\text{A.19})$$

$$487 \quad -x^2 y + 2 \left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} xy + (2\sqrt{\alpha\beta} - 2\phi) x - y^3 + (-2\sqrt{\alpha\beta} + 2\phi) \left(\eta - \frac{fr}{2} \right) \frac{C_d}{K} = 0 \quad (\text{A.20})$$

488 Consider a stationary tropical cyclone where $\phi = 0$, one has:

$$489 \quad x = y \approx -(\alpha\beta)^{1/4} \quad (\text{A.21})$$

490 Substituting the above-mentioned approximation into Eqs. (A.19) and (A.20) results in a required
 491 condition of $\phi(\alpha\beta)^{1/4} + \phi\left(\eta - \frac{fr}{2}\right)\frac{C_d}{K} = 0$. Using the data of section 4.1 corresponding to a typical
 492 tropical cyclone, the value of $\phi(\alpha\beta)^{1/4} + \phi\left(\eta - \frac{fr}{2}\right)\frac{C_d}{K}$ is approximately 10^{-9} . Hence, it is
 493 suggested that $x = y \approx -(\alpha\beta)^{1/4}$ could be adopted for any typical tropical cyclone.

494 On the other hand A_0 can be determined from Eqs. (A.12) and (A.16):

$$495 \quad A_0 = \frac{-X_3}{X_1 + X_2 X_4} \quad (\text{A.22})$$

496 Hence the frictional components of the wind velocity are:

$$497 \quad u'(\theta, z') = \left(\frac{\alpha}{\beta}\right)^{\frac{1}{2}} * \text{Real}(\omega) = \left(\frac{\alpha}{\beta}\right)^{\frac{1}{2}} * \text{Real}\{A_0 * e^{(q_0 z')} + A_1 * e^{(q_1 z' + i\theta)} + A_{-1} * e^{(q_{-1} z' - i\theta)}\} = u_0 + u_1 + u_{-1} \quad (\text{A.23})$$

$$498 \quad v'(\theta, z') = \text{Imag}(\omega) = \text{Imag}\{A_0 * e^{(q_0 z')} + A_1 * e^{(q_1 z' + i\theta)} + A_{-1} * e^{(q_{-1} z' - i\theta)}\} = v_0 + v_1 + v_{-1} \quad (\text{A.24})$$

499

500

APPENDIX B

501 To obtain the solution of the proposed model in the case $u_1 = u_{-1} = v_1 = v_{-1} = \phi = 0$, the A_0 can be
 502 expressed in the complex form as $a_1 + ia_2$. In addition, the expression of q_0 could be derived from
 503 Eq. (22):

$$504 \quad q_0 = \left[2i\sqrt{\alpha\beta}\right]^{1/2} = -\lambda(1+i) \quad (\text{B.1})$$

505 where $\lambda = (\alpha\beta)^{1/4}$. As a result, the frictional components of the wind velocity can be explicitly
 506 given as:

507 $u'(\theta, z') = \left(\frac{\alpha}{\beta}\right)^{\frac{1}{2}} * Real\{A_0 * \exp(q_0 z')\} = \left(\frac{\alpha}{\beta}\right)^{\frac{1}{2}} e^{-\lambda z'} * [a_1 \cos(-\lambda z') - a_2 \sin(-\lambda z')] \quad (B.2)$

508 $v'(\theta, z') = Imag\{A_0 * \exp(q_0 z')\} = e^{-\lambda z'} * [a_2 \cos(-\lambda z') + a_1 \sin(-\lambda z')] \quad (B.3)$

509 According to the relationship $A_0 q_0 = \frac{\partial \omega}{\partial z} \Big|_{z=0}$, A_0 can be determined as:

510 $A_0 = \frac{1}{q_0} \frac{\partial \omega}{\partial z} \Big|_{z=0} = \frac{1}{q_0} \left[\sqrt{\frac{\beta}{\alpha}} \frac{\partial u'}{\partial z} \Big|_{z=0} + i \frac{\partial v'}{\partial z} \Big|_{z=0} \right] \quad (B.4)$

511 The ground surface wind velocity components can be denoted as: $v_{rs} = u'_0 = \sqrt{\alpha/\beta} a_1$ and

512 $v_{\theta s} = v'_0 + v_{\theta g} = a_2 + v_{\theta g}$ where $a_1 + ia_2 = \omega(z=0) = \sqrt{\beta/\alpha} u'_0 + i v'_0$. Based on Eq. (15b) and

513 substituting the expressions of $\frac{\partial u'}{\partial z} \Big|_{z=0}$ and $\frac{\partial v'}{\partial z} \Big|_{z=0}$ into Eq. (B.4), results in:

514 $a_1 - a_2 = -\sqrt{\frac{\beta}{\alpha}} \chi v_{rs} = -\chi a_1 \quad (B.5)$

515 $a_1 + a_2 = -\chi v_{\theta s} = -\chi (a_2 + v_{\theta g}) \quad (B.6)$

516 where $\chi = \frac{C_d}{K\lambda} |v_s| = \frac{C_d}{K\lambda} \sqrt{v_{\theta s}^2 + v_{rs}^2}$. Solving Eqs. (B.5) and (B.6) leads to the following results:

517 $a_1 = -\frac{\chi v_{\theta g}}{1 + (1 + \chi)^2} \quad (B.7)$

518 $a_2 = -\frac{\chi(\chi + 1)v_{\theta g}}{1 + (1 + \chi)^2} \quad (B.8)$

519 Hence, one obtains the frictional components of the wind velocity as:

520 $u'(\theta, z') = -\left(\frac{\alpha}{\beta}\right)^{\frac{1}{2}} e^{-\lambda z'} * \left[\frac{\chi v_{\theta g}}{1 + (1 + \chi)^2} \cos(\lambda z') + \frac{\chi(\chi + 1)v_{\theta g}}{1 + (1 + \chi)^2} \sin(\lambda z') \right] \quad (B.9)$

521
$$v'(\theta, z') = e^{-\lambda z'} * \left[-\frac{\chi(\chi+1)v_{\theta g}}{1+(1+\chi)^2} \cos(\lambda z') + \frac{\chi v_{\theta g}}{1+(1+\chi)^2} \sin(\lambda z') \right] \quad (\text{B.10})$$

522

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526

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