

Title	Interaction between terrestrial plasma sheet electrons and the lunar surface: SELENE (Kaguya) observations
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Citation	Geophysical Research Letters (2010), 37(19)
Issue Date	2010-10
URL	<a href="http://hdl.handle.net/2433/131758">http://hdl.handle.net/2433/131758</a>
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Type	Journal Article
Textversion	author

**1 Interaction between terrestrial plasma sheet**  
**2 electrons and the lunar surface: SELENE (Kaguya)**  
**3 observations**

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4 Analysis of the data obtained by SELENE (Kaguya) revealed a partial loss  
5 in the electron velocity distribution function due to the “gyro-loss effect”,  
6 namely gyrating electrons being absorbed by the lunar surface. The Moon  
7 enters the Earth’s magnetosphere for a few days around full moon, where  
8 plasma conditions are significantly different from those in the solar wind. When  
9 the magnetic field is locally parallel to the lunar surface, relatively high-energy  
10 electrons in the terrestrial plasma sheet with Larmor radii greater than SE-  
11 LENE’s orbital height strike the lunar surface and are absorbed before they  
12 can be detected. This phenomenon can be observed as an empty region in

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13 the electron distribution function, which is initially isotropic in the plasma  
14 sheet, resulting in a non-gyrotropic distribution. We observed the expected  
15 characteristic electron distributions, as well as an empty region that was con-  
16 sistent with the presence of a relatively strong electric field ( $\sim 10$  mV/m)  
17 around the Moon when it is in the plasma sheet.

## 1. Introduction

18 The Moon does not possess a global magnetic field or a thick atmosphere [*Ness et al.*,  
19 1967]. Therefore, the plasma around the Moon is ideal for investigating the interaction of  
20 charged particles with large solid bodies. The Moon enters the Earth's magnetosphere for  
21 a few days around full moon. The plasma in the magnetosphere has different properties  
22 from the solar wind, including different densities and energies, and it interacts directly  
23 with the lunar surface [*Rich et al.*, 1973; *Schubert et al.*, 1974].

24 The first measurements of the lunar plasma environment were made by Explorer 35 in  
25 the solar wind and the Earth's magnetosphere [*Lyon et al.*, 1967; *Nishida and Lyon*, 1972].  
26 Apollo 15 and 16 subsatellites observed electrons reflected from lunar crustal magnetic  
27 fields and measured the surface magnetic field intensity by electron reflectometry [*Howe*  
28 *et al.*, 1974]. Without crustal magnetic fields, almost all the electrons that strike the lu-  
29 nar surface will be absorbed, although some backscattering as well as secondary electron  
30 emission exist [*Halekas et al.*, 2009]. Electrons adiabatically reflected due to the magnetic  
31 mirror effect produce a loss cone in the upgoing electron velocity distribution function  
32 (VDF). The surface magnetic field  $B_{\text{surf}}$  is inferred by measuring both the magnetic field  
33  $B_{\text{sc}}$  and the electron loss cone (cutoff pitch angle  $\alpha_c$ ) at the spacecraft, using the relation-  
34 ship  $B_{\text{surf}} = B_{\text{sc}} / \sin^2 \alpha_c$ . This method was also used to produce a global map of the lunar  
35 crustal magnetic fields by Lunar Prospector [*Halekas et al.*, 2001; *Mitchell et al.*, 2008],  
36 Additionally these observations revealed energy-dependent loss cones, indicating reflec-  
37 tion by both electric and magnetic fields [*Halekas et al.*, 2002]. The electrostatic potential  
38 of the lunar surface varies in sunlight and shadow, and depends on the ambient plasma

39 conditions, which vary depending on whether the Moon is in the solar wind, terrestrial  
40 magnetotail lobe, or the plasma sheet [*Halekas et al.*, 2008]. The large range of lunar  
41 surface potentials implies that the electric field around the Moon is highly variable.

42 Electron reflectometry can be used when the magnetic field line passing through the  
43 observer intersects the Moon, because electrons travel along magnetic field lines. By  
44 analyzing the data obtained by SELENE (Kaguya), we found an interesting phenomenon  
45 concerning electrons in the terrestrial plasma sheet when the magnetic field line is parallel  
46 to the lunar surface; gyrating electrons are absorbed by the lunar surface, and a partial  
47 loss appears in the electron VDF due to this “gyro-loss effect”. In this paper, we refer  
48 to “empty regions” when we are describing features in observations, while “forbidden  
49 regions” when we are describing theoretical predictions.

## 2. Instrumentation

50 SELENE is a Japanese lunar orbiter that was launched on 14 September 2007 and  
51 entered a circular lunar polar orbit with an altitude of 100 km. Since SELENE is a  
52 three-axis stabilized spacecraft, one of its panels always faces the lunar surface. Magnetic  
53 field and plasma measurements were conducted by the MAGnetic field and Plasma exper-  
54 iment (MAP) onboard SELENE, which consists of the Lunar MAGnetometer (LMAG)  
55 and the Plasma energy Angle and Composition Experiment (PACE). LMAG is a triax-  
56 ial fluxgate magnetometer used to observe the magnetic field around the Moon with a  
57 sampling frequency of 32 Hz and a resolution of 0.1 nT [*Shimizu et al.*, 2008; *Takahashi*  
58 *et al.*, 2009; *Tsunakawa et al.*, 2010]. PACE was designed to perform three-dimensional  
59 plasma measurements around the Moon [*Saito et al.*, 2008]. It consists of four sensors:

60 two electron spectrum analyzers (ESA-S1 and ESA-S2), an ion mass analyzer (IMA), and  
 61 an ion energy analyzer (IEA). ESA-S1 and ESA-S2 measure the distribution function of  
 62 low-energy electrons with energies below 16 keV, while IMA and IEA measure the dis-  
 63 tribution function of low-energy ions with energies below 29 keV/q. Figure 1 shows the  
 64 satellite coordinates of SELENE. ESA-S1 and IMA are installed on the +Z panel (looking  
 65 down toward the lunar surface), while ESA-S2 and IEA are on the -Z panel (looking away  
 66 from the lunar surface). Each sensor has a hemispherical field of view.

### 3. Theoretical Predictions

67 Electrons gyrate around magnetic field lines with a Larmor radii given by  $r_L =$   
 68  $m_e v_{\perp} / eB$ , where  $m_e$  is the electron mass,  $v_{\perp}$  is the electron velocity component per-  
 69 pendicular to the magnetic field,  $e$  is the elementary charge, and  $B$  is the magnetic field  
 70 intensity. Although most electrons in the Earth's magnetosphere gyrate with a smaller  
 71 Larmor radii than the orbital height  $H$  (nominal value: 20–100 km) of SELENE, some  
 72 electrons in the plasma sheet have Larmor radii greater than or equal to  $H$  (e.g., a 1 keV  
 73 electron has a Larmor radius of 107 km in a 1 nT magnetic field). When the magnetic  
 74 field is parallel to the lunar surface, these relatively high-energy electrons strike the lunar  
 75 surface and are absorbed (Figure 2a). This can be observed as an empty region in the  
 76 electron VDF, which is isotropic in the terrestrial plasma sheet [*Machida et al.*, 1994].

Consider an electron entering the sensors with a perpendicular velocity component  $v_{\perp}$   
 and a gyrophase  $\psi$ , as shown in Figure 2b. Here, the lunar surface is assumed to be planar  
 since  $H$  is much smaller than the lunar radius, 1738 km. From the geometry, the critical

Larmor radius  $r_c$  is given by

$$r_c = \frac{H}{1 - \cos \psi}. \quad (1)$$

77 Electrons with  $r_L \geq r_c$  are absorbed by the surface and therefore cannot be observed.  
 78 At higher energies, more electrons will be absorbed, enlarging the empty region in the  
 79 electron VDF. When  $\psi = 180^\circ$ ,  $r_c$  takes a minimum value  $H/2$  and the cut-off energy  
 80 of electrons will be a minimum. On the other hand,  $r_c$  is infinite when  $\psi = 0^\circ$  and no  
 81 electrons will be cut off.

In the case of an electric field component perpendicular to the magnetic field as indicated in Figure 2c, electrons will drift toward the lunar surface. If we take the guiding center rest frame (quantities are indicated by  $'$ ), the lunar surface effectively approaches the spacecraft with the  $\mathbf{E} \times \mathbf{B}$  drift velocity  $v_{E \times B}$ . Since the time in which an electron gyrates from a point nearest the lunar surface to SELENE (the red solid arc in Figure 2c)  $t_{\psi'}$  is described as  $t_{\psi'} = r'_L \psi' / v'_\perp = m_e \psi' / eB$  (with  $\psi'$  in radians), the critical Larmor radius will be modified as follows:

$$r'_c = \frac{H'}{1 - \cos \psi'} = \frac{H + v_{E \times B} t_{\psi'}}{1 - \cos \psi'} = \frac{H + \frac{E_\perp}{B} \frac{m_e}{eB} \psi'}{1 - \cos \psi'}. \quad (2)$$

82 Electrons that satisfy  $r'_L \geq r'_c$  strike the lunar surface and are absorbed. If  $E_\perp > 0$ , the  
 83 forbidden region will be smaller than the previous case because  $r'_c > r_c$ .

#### 4. Observations

84 Table 1 shows the location of the Moon and SELENE, as well as the ambient plasma  
 85 conditions for two events discussed below. Figure 3 shows an electron angular distribution  
 86 observed during Event 1. The red lines show the forbidden regions derived from equation



87 (1). At this time,  $B$  was 5.4 nT and relatively high-energy electrons ( $\geq 1$  keV) were  
88 detected. Therefore, the Moon was thought to be located in the plasma sheet or in the  
89 plasma sheet boundary layer, and SELENE was located on the dayside of the Moon at  
90 (Lat.  $33^\circ\text{N}$ , Lon.  $0^\circ$ ) where strong magnetic anomalies do not exist. During this time  
91 interval, the magnetic field was nearly stable and parallel to the lunar surface and it  
92 had an azimuthal angle (in the satellite coordinates X-Y plane) of about  $222^\circ$ . Empty  
93 regions in the electron distribution appeared at an azimuthal angle of around  $312^\circ$ , where  
94  $\psi = 180^\circ$  and  $r_c$  is a minimum. As expected, high-energy empty regions were larger  
95 than low-energy empty regions. These empty regions seem to correspond to theoretically  
96 derived forbidden regions.

97 Figure 4 shows another example observed during Event 2, when SELENE was located  
98 at (Lat.  $28^\circ\text{S}$ , Lon.  $92^\circ\text{W}$ ) where strong magnetic anomalies do not exist. At this time,  
99 the Moon was in the central plasma sheet. The magnetic field line was nearly stable and  
100 parallel to the lunar surface (although it was slightly inclined in this example) and it had  
101 an azimuthal angle (in the satellite coordinates X-Y plane) of about  $238^\circ$ . Empty regions  
102 also appeared around  $328^\circ$  in Figure 4, but the forbidden regions derived from equation  
103 (1) (solid red lines) are larger than the observed empty regions. By assuming that the  
104 perpendicular electric field is 10 mV/m in equation (2), we can fit the forbidden regions  
105 (broken red lines).

## 5. Discussion

106 By analyzing the data obtained by SELENE, we discovered characteristic electron VDFs  
107 produced by the interaction between terrestrial plasma sheet electrons and the lunar

108 surface. We compared theoretically derived forbidden regions with the observed empty  
109 regions in electron VDFs and found that such forbidden regions do exist. Interestingly,  
110 these electron VDFs are asymmetric relative to the magnetic field line; in other words,  
111 they are “non-gyrotropic”. Such non-gyrotropic VDFs are very rare in space plasmas,  
112 especially for electrons in a steady state. However, these VDFs commonly exist at low  
113 altitudes around the Moon. We note that field aligned upward-going electron beams  
114 and energy-dependent loss cones observed when the magnetic field line intersects the  
115 lunar surface [*Halekas et al.*, 2002] are “gyrotropic” VDFs (symmetric with respect to the  
116 magnetic field line).

117 The empty regions in Figure 4 are more consistent with the forbidden regions in the  
118 presence of a perpendicular electric field than those when no perpendicular electric field  
119 is present. This finding suggests that a relatively strong electric field ( $\sim 10$  mV/m) exists  
120 around the Moon in the plasma sheet, although other explanations are also possible. For  
121 example, the plasma can be diffused in phase space to form a smaller empty region due  
122 to scattering by unstable waves [*Kennel and Petschek*, 1966].

123 An electric field of 10 mV/m is quite strong in the Earth’s magnetotail near lunar orbit  
124 (*McCoy et al.* [1975] reported typical value of 0.15 mV/m and up to 2 mV/m). When  
125  $B = 2.2$  nT and  $E_{\perp} = 10$  mV/m, we can get  $v_{E \times B} = 4.5 \times 10^3$  km/s. However, the bulk  
126 flow obtained from the ion observation in Event 2 was 410 km/s (Table 1). Therefore,  
127 ions did not execute  $\mathbf{E} \times \mathbf{B}$  drift and the scale length of the region characterized by the  
128 strong electric field was less than the diameter of the ion gyromotion (i.e. twice of the ion  
129 Larmor radius).

130 In Event 2, SELENE was located near the terminator of the Moon (solar zenith angle  
131  $106^\circ$ ) and assumed  $E_\perp$  is  $(-6.3, -2.0, 7.5)$  mV/m in SSE coordinates. This electric field  
132 has  $-x$  component, which is directed from the sunlit side to the night side. Therefore, the  
133 potential difference between the two sides (the electrostatic potential is higher in the sunlit  
134 side than the night side due to photoelectron emission) may generate this electric field.  
135 However, this idea has to be considered carefully since a Debye length around the Moon  
136 ( $< 1$  km) is much smaller than SELENE's orbital height  $H$  and the surface potential can  
137 be shielded within a few Debye lengths. [Farrell *et al.*, 2007].

## 6. Conclusions

138 A partial loss in the electron VDF due to the “gyro-loss effect” was discovered. Electron  
139 VDFs produced by this effect are “non-gyrotropic” VDFs which are very rare in space  
140 plasmas. The phenomena discussed above are not limited to the case of terrestrial elec-  
141 trons and the Moon, but general and fundamental processes when a plasma interacts with  
142 a solid surface. The electron VDFs suggest a relatively strong electric field is sometimes  
143 present in the near-lunar plasma environment. This study can be used as a technique to  
144 measure electric fields in the vicinity of the Moon.

145 **Acknowledgments.** The authors wish to express their sincere thanks to the team  
146 members of MAP-PACE and MAP-LMAG for their great support in processing and an-  
147 alyzing the MAP data. The authors also wish to express their gratitude to the system  
148 members of the SELENE project. SELENE-MAP-PACE sensors were manufactured by  
149 Mitaka Kohki Co. Ltd., Meisei Elec. Co., Hamamatsu Photonics K.K., and Kyocera Co.

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**Table 1.** Moon and SELENE locations and ambient plasma conditions for two events.

	Event 1	Event 2
Time period	2009/05/06/22:22:11-22:22:27	2008/01/21/15:00:23-15:01:11
Moon location in GSE coordinates	$(-53, 27, -5)R_E$	$(-57, 12, 3)R_E$
SELENE location		
- Orbital height	51 km	98 km
- Latitude and longitude in selenographic coordinates	$(33^\circ\text{N}, 0^\circ)$	$(28^\circ\text{S}, 92^\circ\text{W})$
- Solar zenith angle	$43^\circ$	$106^\circ$
Magnetic field intensity	5.4 nT	2.2 nT
Density	$0.15 \text{ cm}^{-3}$	$0.07 \text{ cm}^{-3}$
Ion temperature	476 eV	1.86 keV
Electron temperature	361 eV	435 eV
Bulk flow	–	410 km/s

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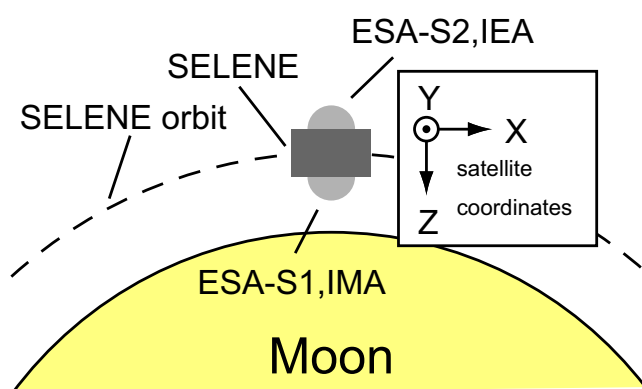
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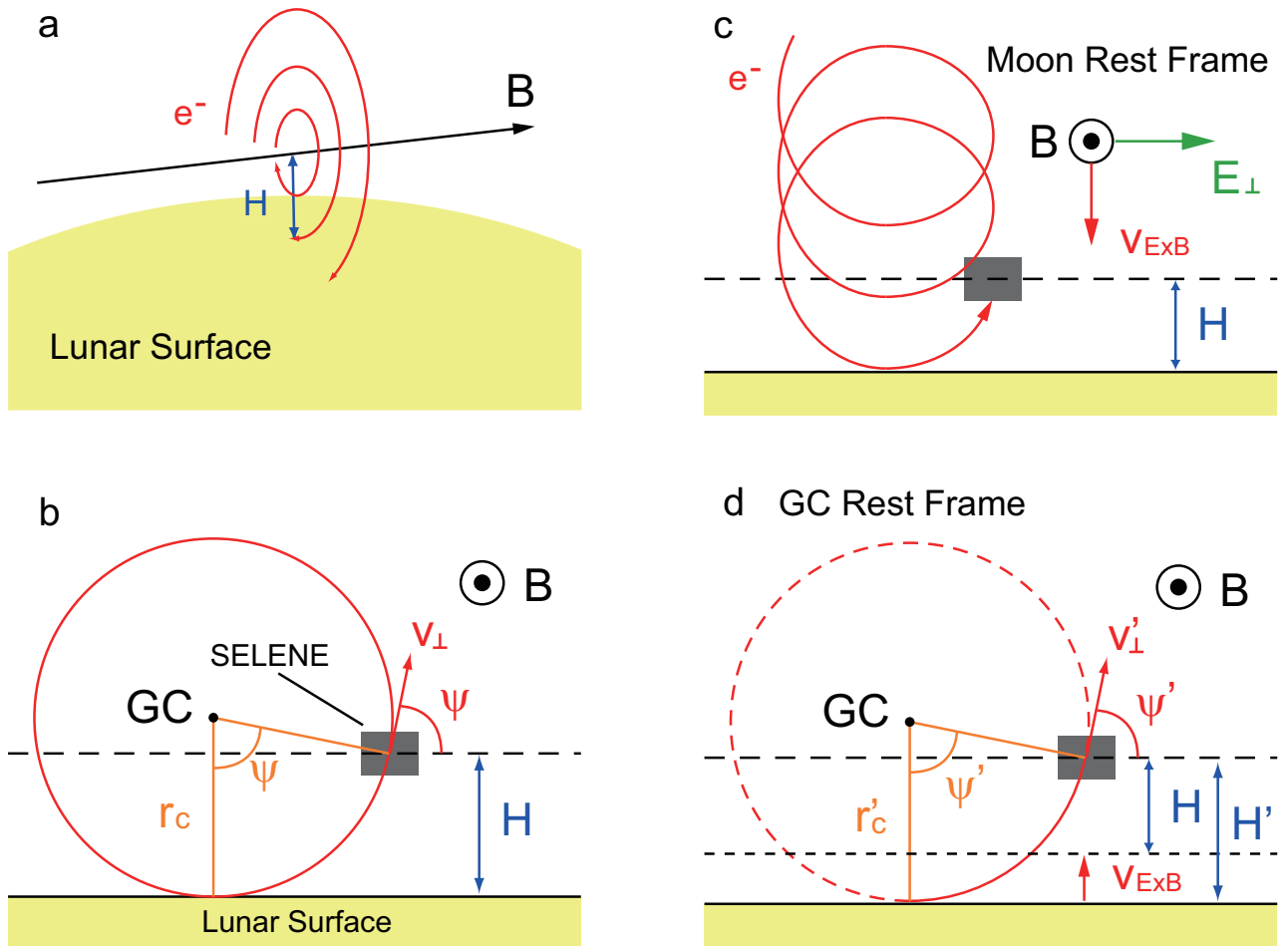
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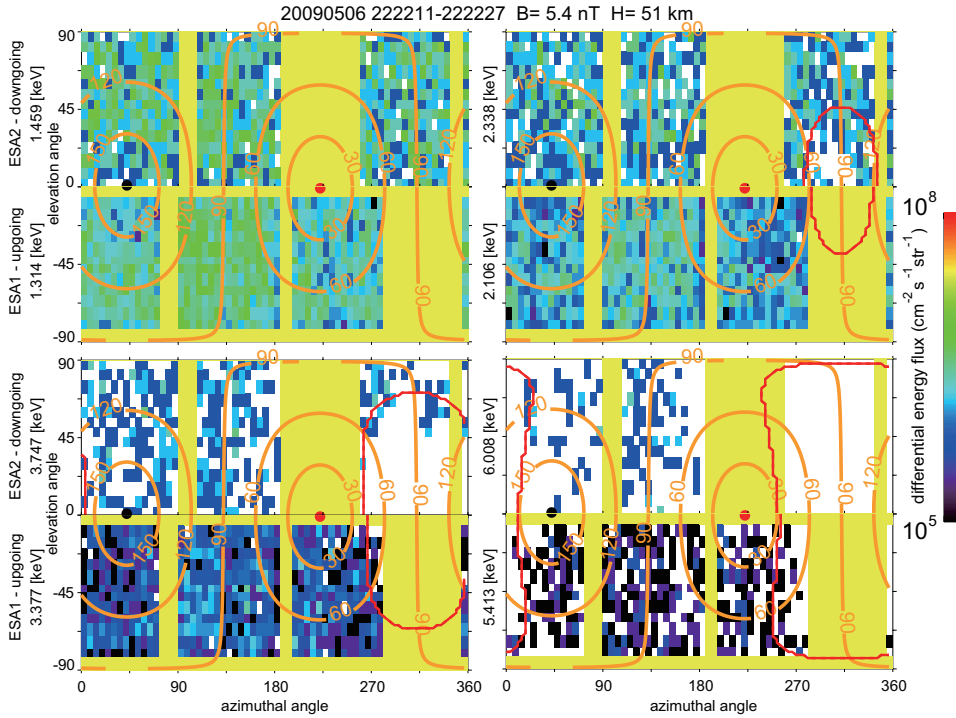


**Figure 1.** Satellite coordinates of SELENE. +Z is directed toward the lunar surface; +X or -X is the direction of travel; and Y completes the orthogonal set.

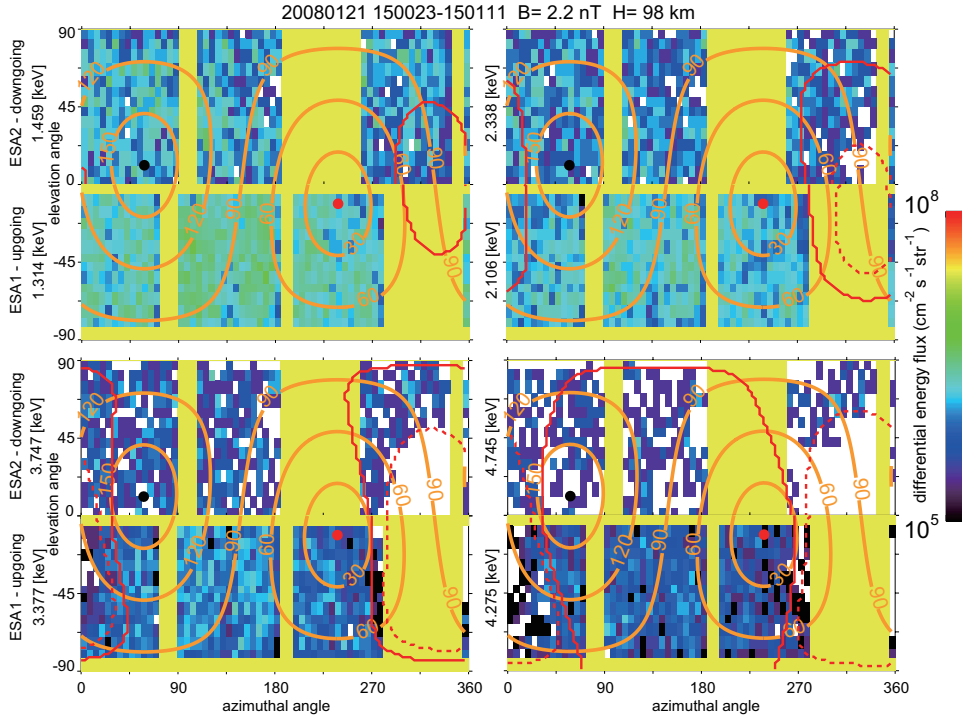


**Figure 2.** Schematic illustration of (a) interaction between electrons and the lunar surface, (b) the critical Larmor radius  $r_c$ , gyrophase  $\psi$ , perpendicular velocity  $v_{\perp}$ , and SELENE's orbital height  $H$ , and the case of an electric field perpendicular to the magnetic field in (c) the Moon's rest frame and (d) in the guiding center (GC) rest frame. GC indicates the guiding center of the electron.





**Figure 3.** Electron angular distribution for different energies in satellite coordinates obtained during 22:22:11-22:22:27 UT (16 sec) on 6 May 2009, when the Moon was in the Earth’s magnetotail and the magnetic field was nearly parallel to the lunar surface. The top left panel shows the distributions of electrons with energies of 1.459 (ESA-S2) and 1.314 (ESA-S1) keV, the top right panel shows those with energies of 2.338 (ESA-S2) and 2.106 (ESA-S1) keV. The bottom left panel shows the distributions of electrons with energies of 3.747 (ESA-S2) and 3.377 (ESA-S1) keV, and the bottom right panel shows those with energies of 6.008 (ESA-S2) and 5.413 (ESA-S1) keV. Angles with little or no sensitivity are indicated by yellow. Note that ESA-S1 and ESA-S2 have different sensitivities. The red and black circles respectively indicate the magnetic field direction and the opposite direction obtained from LMAG data. The orange contours indicate the pitch angles. The red lines indicate the theoretically derived forbidden regions assuming that there is no perpendicular electric field.



**Figure 4.** Electron angular distribution obtained during 15:00:23-15:01:11 UT (48 sec averaged) on 21 January 2008, in the same format as Figure 3. The broken red lines indicate the theoretically derived forbidden regions assuming a perpendicular electric field of 10 mV/m, whereas the solid red lines indicate regions with no electric field.