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## L-DART: A Penetrator Mission for Lunar Permanently Shadowed Regions

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**L-DART: a Penetrator Mission for Lunar Permanently Shadowed Regions.** S. J. Barber<sup>1</sup>, H. M. Sargeant<sup>1</sup>, S. Sheridan<sup>1</sup>, I. P. Wright<sup>1</sup>, A. Ballard<sup>2</sup>, P. D. Church<sup>2</sup>, P. Gould<sup>2</sup>, M. D. Gupta<sup>2</sup>, S. Hussain<sup>2</sup>, A. Griffiths<sup>3</sup>, G. H. Jones<sup>3</sup>, U. Derz<sup>4</sup>, M. Perkinson<sup>4</sup>. <sup>1</sup>School of Physical Sciences, The Open University, Milton Keynes, MK7 6AA, UK (simeon.barber@open.ac.uk), <sup>2</sup>QinetiQ Limited, UK, <sup>3</sup>University College London, Mullard Space Science Laboratory, UK, <sup>4</sup>Airbus Defence and Space Limited, UK,

**Introduction:** Lunar Direct Analysis of Resource Traps (L-DART) will address many current knowledge gaps concerning lunar volatiles and permanently shadowed regions (PSRs), providing in-situ ground truth data to calibrate numerous existing remote datasets.

It builds on UK expertise in developing and testing penetrator system concepts for the Moon and Europa (e.g. MOONLITE [1]). Following release of a Penetrator Descent Module in lunar orbit (Figure 1), its Penetrator Delivery System performs de-orbit and orientation before releasing the instrumented Penetrator to penetrate a few meters into target lunar surface at ~300 m/s. The penetrator itself serves as the sampling tool and an on-board mass spectrometer analyses in-situ the volatiles released both in the impact and in the subsequent thermal soak from lander to surrounding regolith. A pair of 3-axis accelerometers measure regolith structure during the landing event and constrain penetrator final location. Temperature sensors enable regolith thermal properties to be deduced. Pre-and post impact imagery is obtained for context. Science is complete and data relayed to Earth within 1-2 hours, minimizing system mass and lifetime requirements.

Possible landing sites include Cabeus (for comparison with LCROSS) or Shoemaker which exhibits excess hydrogen, or areas indicated by LRO to exhibit putative surface frost. Alternatively, L-DART could target the hypothesised ancient (paleo) south pole [2] and hence potentially ancient volatiles.

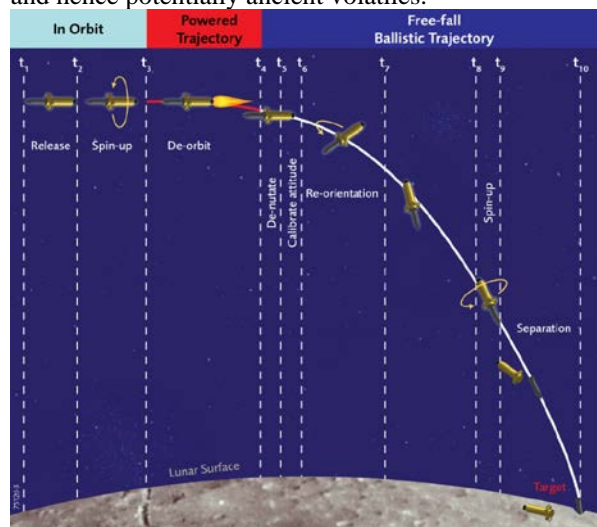


Figure 1: Typical events during penetrator delivery (credit Airbus Defence and Space)

### Lunar polar volatiles and knowledge gaps:

Today is a fascinating time in the history of lunar science and exploration. After the lull that followed the Apollo and Luna era, many spacefaring nations are preparing new lunar missions. Many of these missions target the lunar poles, which have been shown to be very different to the largely equatorial areas studied in detail via Apollo and its returned samples. Orbital missions have revealed an extraordinary near-polar thermal environment, with the Moon's low angle of tilt meaning that topography dominates illumination conditions and temperatures in permanently shaded cratered regions are among the lowest of any solar system body. These areas act as cold traps where a range of volatiles are thermally stable on geological timescales [3] offering a scientific treasured trove of time-integrated volatile inventory. Yet nearby raised areas offer near-constant illumination (e.g. [4]) leading to the possibility of abundant solar electricity generation. These potentially abundant supplies of volatile resources including water, together with a source of solar power, could enable sustained exploration and even a permanent human presence.

The search for lunar polar water is therefore a highly active research topic. Although a variety of instruments on orbiters have indicated the presence of hydrogen or water at or near the surface, consensus is lacking on the interpretation of orbital data. The most direct evidence for lunar polar volatiles comes from the Lunar Crater Observation and Sensing Satellite (LCROSS) mission launched together with Lunar Reconnaissance Orbiter (LRO) in June, 2009 [5]. This culminated in the impact of the spacecraft's spent upper rocket stage into the permanently shadowed Cabeus crater near the lunar south pole. The impact was observed by a suite of instruments on the LCROSS shepherding spacecraft, and led to the deduction that the concentration of water ice at the impact site was  $5.6 \pm 2.9\%$  by mass [6]. In addition to water, a number of other volatile species were detected by instruments on both the shepherding spacecraft and LRO (including near-infrared, visible and ultraviolet spectrometers; [6], [7]). Many of these were consistent with calculations on thermal stability [8], [9].

But still L-CROSS leaves questions unanswered. The amount of material excavated in the impact is uncertain (at least in part due to uncertainties in mechanical properties of the landing site regolith) and this

propagates into uncertainties in the inferred concentrations of volatiles in the regolith. There is some suggestion of a delayed flash implying porosity in the target material; the sub-surface thermal environment was not measured directly. And because the observations on chemical composition of the plume were made remotely, these are subject to chemical processing that may occur within the plume hence measurements may not be fully representative of the original source materials. Notably, not only do current uncertainties relate to fundamental scientific questions, but they also involve many of the physical parameter inputs demanded by engineers seeking to design future missions.

Arguably, the only consensus concerning lunar polar volatiles is that new landed missions are necessary, carrying suites of instruments to extract and analyze volatiles and characterize the source environment.

One such example is ESA's PROSPECT package [10]. At its heart is an ion trap mass spectrometer weighing only ~500 g yet capable of detecting a wide range of volatiles with high sensitivity, and with heritage of successful analogous measurements on the Rosetta comet lander [11], [12]. But to receive samples it requires to be serviced by a 25 kg drill, and hence is only suitable for deployment on a large static platform such as Luna-27 or perhaps a 250 kg-class rover. Smaller rovers in the ~25 kg class may also house such compact analysis instruments but direct regolith sampling depth may be limited to the upper ~10 cm [13].

Penetrators mission such as L-DART may therefore present an ideal or even unique means through which to directly sample and analyze at depths analogous to those probed by remote sensing (i.e. the upper ~2 m), in lunar regions such as PSRs inaccessible to conventional spacecraft. Moreover, by using miniaturized and ruggedized hardware such as mass spectrometers similar to those planned for landers such as Luna-27 [10], L-DART can provide data that are valuable ground truth for the interpretation of other data-sets.

#### Science Objectives and Instrumentation:

The science objectives of L-DART are summarized in its Science Traceability Matrix (to be presented). This table also demonstrates how a focused suite of complementary and high TRL instruments (accelerometers/datalogger, mass spectrometer, temperature sensor) on the penetrator can acquire a unique dataset on lunar polar volatiles and can greatly enhance existing and future remote datasets by placing strong constraints on the environmental conditions at the landing site (thermal, geotechnical) in the wider geological context (local topography, geology, illumination) provided by the descent imager. In addition to its scientific merit the descent imager will provide data of high

value for public engagement and outreach, as will the post-landing imager (if implemented).

**Current Status:** The L-DART consortium is currently conducting a payload definition exercise to refine the reference payload. This will enable the updating of key payload parameters such as mass and volume to reflect recent developments in technology readiness level. In addition, preliminary results may be presented on a study by The Open University and QinetiQ (funded by UK Space Agency) aimed at increasing the understanding of the attenuation of the penetrator's UHF communications signal by any regolith overlying the onboard antenna post-landing.

The outcomes of the above activities will generate outputs ready for input to a Phase A type study in which the system design feasibility can be assessed and realistic estimates of the all-up penetrator system mass can be generated for input to more detailed mission design and cost estimating activities.

**Conclusions:** L-DART targets key scientific and exploration knowledge gaps concerning lunar polar volatiles, by making in situ measurements of volatiles and regolith thermal and geotechnical properties supported by contextual imaging. Many technological aspects have been matured in the last decade, and international partners are now sought to exploit this significant potential and make the mission a reality.

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