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NAVAL POSTGRADUATE SCHOOL

MONTEREY, CALIFORNIA

THESIS

PASSIVE MULTISTATIC DETECTION OF MARITIME TARGETS USING OPPORTUNISTIC RADARS

by

Chong Sze Sing

March 2014

Thesis Advisor: Co-Advisor: Herschel H. Loomis David C. Jenn

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PASSIVE MULTISTATIC DETECTION OF MARITIME TARGETS USING OPPORTUNISTIC RADARS

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Submitted in partial fulfillment of the requirements for the degree of

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ABSTRACT

Passive multistatic radar (PMR) makes use of transmission from opportunistic radars to detect targets. In this thesis, a maritime scenario is created with multiple merchant vessels transmitting at S-band (3 GHz), serving as opportunistic radars and a frigate-size warship acting as the PMR receiver and to detect a low-radar cross-section (RCS) target. The simulations are carried out with actual technical parameters from open sources to best approximate practical performance. To further improve realism, the bistatic RCS simulation of the stealthy Republic of Singapore Navy Formidable-class frigate was included to validate the results.

The simulation results show that the multistatic geometry of four opportunistic transmitters at the same range from the passive receiver with 90-degree separation offers the best coverage. Passive detection of a target of up to a radial range of 30 km with detection coverage of 85% or better is possible. This range coverage is similar to that of the monostatic radar but lacks in the area of detection coverage. The simulations also demonstrated that the detection accuracy is also the highest using this same geometry. The worst-case uncertainty ellipse around the low-RCS target is less than 150 m.

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LIST OF ACRONYMS AND ABBREVIATIONS

AOA	angle-of-arrival				
ARMs	anti-radiation missiles				
AWGN	additive white Gaussian noise				
CMR	civil-marine radar				
CMS	combat management system				
COTS	commercial-off-the-shelf				
CRLB	Cramér-Rao lower bound				
DF	direction finding				
DRFM	digital radio frequency memory				
ESM	electronic support measures				
EM	electromagnetic				
EW	electronic warfare				
FEKO	field calculations for bodies with arbitrary surface (FEldberechnung für Körper mit beliebiger Oberfläche)				
FM	frequency modulated				
GPS	global positioning system				
HF	high frequency				
IFF	identification friend or foe				
IMO	International Maritime Organization				
IRST	infrared search and track				
ISAR	Inverse SAR				
LPI	low probability-of-intercept				
LOS	line-of-sight				
NMEA	National Marine Electronics Association				
MAIS	Marine Automatic Identification System				
MMSI	Maritime Mobile Service Identity				
MTI	moving target indicator				
OODA	observe-orient-decide-act				
OTH	over-the-horizon				
PBR	passive bistatic radar				

PMR	passive multistatic radar			
POA	phase-on-arrival			
PRF	pulse repetitive frequency			
RAM	radar absorbent material			
RCS	radar cross section			
RORO	roll-on roll-off			
SAR	synthetic aperture radar			
SNR	signal-to-noise ratio			
SOLAS	safety of life at sea			
SOTDMA	self-organizing time division multiple access			
ТОА	time-of-arrival			
UAV	unmanned aerial vehicle			
VHF	very high frequency			
YIG	yttrium iron garnet			

EXECUTIVE SUMMARY

Using opportunistic transmitters in the form of civil marine radar (S-band and X-band) to detect maritime targets passively is both viable and comparable in performance to monostatic radars. However, in the case of detecting low-RCS (radar cross section) targets, passive multistatic radar (PMR) outperforms monostatic radars. The highly regulated requirements of the International Marine Organization (IMO) ensure that most maritime vessels out-at-sea have the necessary information to support both passive bistatic radar (PBR) and PMR. With the world's trade routed via the sea, there is no shortage of suitable vessels possessing at least one civil marine radar (CMR) and one marine automatic identification system (MAIS) on the main shipping routes. Among all the suitable vessels, larger ships like oil tankers, container ships, etc., have their CMR installed at the highest point, which extends the radar horizon and detection range. The narrow beamwidth and high transmit power of the CMR conveniently lend themselves to exploitation by a PBR system in terms of both detection accuracy and detection range.

The unique geometry formed between the opportunistic transmitter, passive receiver and unsuspecting target provides exceptional military applications not offered by the current military technologies. It offers true passive detection (with zero transmission from the receiving warship) while at the same time supports detection coverage comparable to a monostatic radar. Passive detection means there is no minimum range detection limitation that is unavoidable in a monostatic radar. In the same vein, there is no need for 'blanking' of your electronic support measures (ESM) receiver every time the onboard radar transmits. Passive detection allows for permanent monitoring of the electromagnetic spectrum for intelligence and threat. On the vulnerability front, this removes the risk posed by anti-radiation missiles and, at the same time, enhances mission survivability, allowing the warship to continue tracking targets without a working radar or with a damaged radar.

This same unique geometry also improves the warship's probability of detecting a low-RCS ship that has been treated by shaping and radar absorbent material (RAM) as compared to a monostatic radar. The use of low frequency S-band CMR poses serious

limitations to the use of RAM treatment on a low-RCS ship. It, therefore, restricts the effectiveness of low-RCS ships in the use of shaping. However, it is shown in the bistatic RCS measurement of the model based on the RSS Formidable-class stealthy frigate that shaping alone is capable of reducing the median RCS to values not higher than 200 m². With all else being equal, with the effect of RCS on detection, the PBR system has an added advantage. The bistatic RCS simulation using FEKO, a commercial RCS modelling software, shows more incidence angles exist that demonstrate a spike in RCS returns. Simply put, there are more degrees-of-freedom created by the transmitter-target-receiver geometry that has a higher probability of directing spikes in the RCS returns toward the receiver. In the case of a monostatic radar, there are limited ways to maximize RCS returns because the shaping of a low-RCS is designed to reduce returns in the direction of the incidence angle.

The simplicity of PMR also lies in the fact that there are no expensive systems to build. Instead, all the information that the PMR system needs is already available on most warships. This information can be extracted and used to carry out tracking.

The passive detection range and coverage is related to many parameters on the transmitter, target and receiver. On the opportunistic transmitter end, the transmit power and antenna gain are the two parameters that determine the maximum detection range. However, it is the relative position of the transmitter in the PBR geometry that determines the detection coverage or detection gaps. On the receiver end, the deciding parameters are the sensitivity and bandwidth of the ESM receiver. These two parameters affect the signal-to-noise ratio and, thereby, change the maximum detection range. Similar to the opportunistic transmitter, it is the geometry of the PBR that affects the detection gaps. Collectively, the factor that is used here to quantify the geometry is the bistatic angle β formed between the transmitter and receiver (see Figure 1). Incidentally, the bistatic angle decides the bistatic RCS, which is presented to the receiver depending on the incidence angle of the transmitter's radar emission.



Figure 1. Bistatic Angle - β .

From the simulation plots presented in the thesis, it is shown that when more transmitters are spaced farther apart, they provide better overall coverage with small detection gaps. The minimum number of transmitters is four, placed 90 degrees apart, with one in each quadrant. To achieve maximum coverage with this particular combination of system parameters and transmitter constellation, the baseline distance of 18,000 m is recommended. By rotating the quadrant placement of transmitters, one could direct the detection gaps away from a specific bearing of choice to improve detection.

Using more than four transmitters in the suggested constellation does not significantly extend detection range. However, it is shown in the eight-transmitter simulation that it does improve detection coverage from 80% to 95% through the reduction of detection gaps. This translates to the ability of the PMR to track a target continuously, provided the target offers a strong bistatic return to all eight transmitters. In all likelihood, a low-RCS target would have presented at least two to four strong RCS returns to either the quadrant-based constellation using four transmitters or the two-layered quadrant-based constellation using eight transmitters. If the availability of such a constellation is not possible out-at-sea, one should position oneself to create such a constellation without jeopardizing a passive posture.

There are two dominant factors that affect detection accuracy, namely, the beamwidth of the transmitter and the direction-finding (DF) accuracy of the receiver. It is shown that the worst-case measurement error is approximately ± 120 m for a low-RCS 114 m frigate at a range of 20 km. In most cases, a lower measurement can be achieved using multiple transmitters by overlapping their individual uncertainty ellipses. It is found that the four-transmitter and eight-transmitter constellations recommended for maximum detection coverage are also optimized to reduce the target's uncertainty ellipse. It is shown that with an eight-transmitter constellation, the overlapped area by all the uncertainty ellipses has given rise to a measurement error of 20 m for the same low-RCS target at a range of 6400 m.

In summary, the use of opportunistic transmitters in the form of civil marine radar (S-band and X-band) to detect maritime targets passively is both viable and comparable in performance to monostatic radars. The PMR with four or more transmitters offers a detection range of up to 30 km with detection coverage of 85% or better. The associated worst-case detection error is approximated to be ± 120 m for a low-RCS 114 m frigate at a range of 20 km. On top of that, the PMR offers added capability to defeat low-RCS targets without the need to build extra combat systems. All information required for the PMR to work is available in most modern warships.

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I. INTRODUCTION

A. OVERVIEW

The ability to detect targets passively has been greatly sought after since the development of modern radar. The term 'passive' detection of targets should be interpreted as the ability to detect targets without deliberate or active transmission by the sensor on the target of interest. In this deadly game of cat-and-mouse, the one that gets to 'see' first strikes first. In modern day warships, new technologies and techniques like low probability-of-intercept (LPI) radars, low-radar cross-section (RCS) design, radar absorbent material (RAM) coating and ultra-sensitive electronic warfare (EW) receivers have changed the nature of the game. On the one hand, the risk of being detected by ultra-sensitive EW receivers has forced military ships to go on a strict radar transmission policy and leverage other means of sensing, such as ship-borne unmanned aerial vehicles (UAV) or passive infrared sensor and track (IRST), for target detection. On the other hand, ships equipped with capable LPI radars still have to contend with weak returns from low-RCS designed ships. For both cases, the effective detection range in most maritime missions is generally reduced to slightly beyond the visual range with the exception of over-the-horizon (OTH) radar.

A proposed solution to resolve this deadlock, which is investigated in this thesis, is the use of multiple pairs of passive bistatic radar (PBR) to detect targets using common radar transmission by opportunistic transmitters. This approach offers many advantages. Firstly, there is no radar transmission by a source eliminating the risk of being intercepted by the target's EW receiver, or worse, by anti-radiation missiles (ARMs). Detection occurs passively, thus advancing one's observe-orient-decide-act (OODA) loop while denying the target's own OODA loop. Secondly, the unique geometry formed by separating transmitter and receiver reduces the effectiveness of low-RCS design, which to date has only been optimized to reduce monostatic returns. Simulations have shown that the RCS of stealth ships changes with different bistatic angles β , and some of these angles have considerably larger RCS returns. This unique effect may support detection of targets that are not in the line-of-sight (LOS) of one's radar. In the maritime environment, the opportunistic transmitters are civil marine radars (CMR) operating in S-band (2.9 GHz – 3.1 GHz) and X-band (9.2 GHz – 9.5 GHz) installed on all seagoing vessels regulated by the International Maritime Organization (IMO) under Regulation 19 in safety of life at sea (SOLAS), Chapter V. The use of S-band transmission as a source also improves the chance of detection as most RAM treatment is frequency specific and wavelength dependent. As the wavelength increases, the thickness of the corresponding RAM material increases; however, the RAM treatment is not effective at lower frequencies. In addition, the abundance of opportunistic transmitters out-at-sea allows one to choose from different combinations of bistatic pairs to optimize detection coverage, probability of detection and range accuracy. Next, all hardware required for the PBR to work is currently available on most warships. There is no requirement for additional equipment. Lastly, it enhances the mission capability of warships by allowing target detection with a defective or damaged radar.

In Figure 1, a typical bistatic radar geometry is presented with the transmitter (Tx) and receiver (Rx) separated by the baseline distance, also known as the direct path or line-of-sight (LOS) of transmission, denoted by *L*. The distance from the transmitter to the target is denoted as R_T , while the distance from the target (Tgt) to the receiver is denoted as R_R . The angle formed between R_T and R_R is known as the bistatic angle β .



Figure 1. Bistatic geometry showing the Tx, Tgt and Rx. From [1].

1. History

The concept of bistatic radar is not new. In fact, the oldest form of radar is of the bistatic kind, which operated in the forward scattering configuration for aircraft detection in the 1930s [1]. The operation of this forward scatter radar ($\beta = 180^{\circ}$) is based on Babinet's principle that states that a target's RCS is greatly enhanced when observed by the receiver in a bistatic configuration with the transmitter, target and receiver on the same baseline, arranged in the order stated earlier [2]. Although the forward scatter radar supports extended detection, it is restricted in practical applications, serving more like a 'trip-wire'. Initially, little was known about the benefits that bistatic radar brings. Concurrently, the monostatic radar technology was gaining momentum with the invention of the magnetron during World War II. This technology gradually matured and ushered in the birth of pulsed radar, moving target indicator (MTI) radar, phased array radar and synthetic aperture radar (SAR) [3].

Thirty years later, in the 1960s, a bistatic radar exploiting a transmitter of opportunity, fitting the definition of a PBR, was mentioned in a paper by Rittenbach and Fishbein [4] and demonstrated a more advanced application of the bistatic radar. The most famous bistatic system built in the 1960s was the Sugar Tree system, a high frequency (HF) OTH bistatic receiver that exploited the Soviets' international broadcast transmitter to detect missile launches [4]. By then, the mainstream use of radar was dominated by monostatic radar, closely followed by counter-detection techniques known as EW. Concepts like LPI radars, low-RCS designs, and RAM were explored and pursued, leaving bistatic radar concepts relatively untouched.

In the 1990s, a renewed interest in bistatic radar occurred as a possible solution to defeat anti-radiation missiles and low-RCS designed platforms. With better technologies and computing capabilities, many possible applications of bistatic radars were reexamined. The most notable systems developed during the period are summarized in Table 1.

2. Bistatic Radar Research Areas

The current research interest in bistatic radar and multistatic radars can be broadly classified into three areas: (1) target detection and parameter extraction, (2) multistatic radar systems and configuration, and (3) target tracking and accuracy.

Under category (1), the forward scattering configuration of bistatic radar is combined with inverse SAR (ISAR) techniques. The bistatic radar, particularly in the forward scatter region, is used to detect crossing maritime targets cutting through the baseline formed by the transmitter and receiver. Target detection is accomplished through differentiating the target's Doppler frequency from the sea clutter. Using the rate of change of the Doppler frequency, we can approximate the target's speed and length with an error of 20% and an associated probability of detection of 80% based on the 70 maritime targets tracked [5]. This system allows for autonomous extraction of target parameters in support of target identification and classification. A separate paper went further to investigate the differences between ISAR returns from forward scatter and backscatter [6]. In that experiment, an additional transmitter, operating at a slightly different frequency, is collocated near the receiver to allow simultaneous capturing of both the forward scatter and backscatter when the ship crosses the baseline. The results confirmed that the Doppler frequency returns collected from the forward scatter and backscatter have distinctly different profiles in the Doppler versus time plot.

The forward scatter returns have a better reproduction of the ship's silhouette, most likely due to the diffraction of electromagnetic (EM) waves around the ship's superstructure. As for the case of the backscatter returns, the Doppler frequency is almost uniformly spread out across the ship structure. In summary, there is great potential in combining these two returns to enhance the capability to detect, classify and identify targets.

System	Silent Sentry [™]	TV-Based Bistatic Radar (I)	TV-Based Bistatic Radar (I)	FM Radio- based Bistatic Radar	Multistatic HDTV- based Radar
Developer	IBM, now Lockheed Martin	University College London	DERA United Kingdom	NATO	SAIC, U.S. Army
Decade	1980 - 2000	1980	1980	2000	2000
Configuration	Multistatic Rx -1, Tx – Up to 6	Bistatic	Bistatic, Near forward scatter	Bistatic	Multistatic Rx -4, Tx – 1
Transmitter Operation	FM – realtime TV – Non- realtime	TV – Non- realtime (543 MHz)	TV – Near realtime (543 MHz)	FM – realtime	HDTV - realtime
Baseline	100 km typical	12 km	150 km	50 km	10 km typical
Target	Aircraft, Missile Launch	Aircraft	Aircraft	Aircraft	Aircraft below 5000ft
Target Data	Range, Doppler bearing	Range, Bearing	Doppler, Bearing	Range, Doppler, Bearing	Range, Doppler
Measured	$R_{\rm M} = 100 \rm{km} - 150 \rm{km}$	$R_M \sim 25 km$ Occasional	$R_M \sim 160$ km	$R_M \sim 175$ km.	R _M ~ 30 km Target
Performance	2D track on Aircraft 3D track on missile launch	A/C detections, but mostly negative	Detections on high and medium altitude A/C but only 1/3 tracked	Achieved with innovative direct path excision	location via range multi- lateration Ghost excision via doppler

Table 1. Parameters of significant passive bistatic radar programsdesigned and tested for air surveillance. From [4].

A PBR system is an attractive and low-cost choice when it comes to detection of low-flying aircraft traversing a terrestrial region. Many terrestrial transmitters such as frequency modulated (FM) radio antennas, cellphone base-stations and digital radio antennas provide the transmission source for PBR to detect these aircraft. The paper by Griffiths and Baker [7] evaluates each of the opportunistic transmitters and their detection ranges. The result shows the FM radio station at Wrotham in England has a detection range of up to 27 km with a minimum signal-to-noise ratio (SNR) of 15 dB using a coherent integration period of 0.1 s. In the case of a low-power digital radio transmitter, the detection range is reduced to 9 km with the same parameters. The cellphone base

station reduced the detection range to 1.2 km, the shortest of the three. However, the extensive network of cellphone base-stations makes up for the short detection range with good detection coverage. Delving into how the transmission waveforms affect detection, the paper found that wideband signals like FM and digital radio introduced added complexity in the bistatic and multistatic ambiguity function and concluded that the narrowband signal is more suitable for PBR system. In effect, the varying bandwidth and dynamic amplitude range of the broadcast music makes detection and tracking accuracy extremely difficult to quantify.

Data fusion is the combination of information on the same target from different sources (multiple radars) to improve probability of detection and tracking accuracy. Even in network-centric warfare concepts, it is applied almost exclusively to monostatic radar sources where system parameters such as range bin, tracking errors, and SNR are known and linear. In bistatic and multistatic radar systems, these parameters change in a non-linear fashion with respect to the geometry of the transmitter, receiver and target. In an attempt to better understand whether data fusion yields a similar advantage in a multistatic netted radar system, two sets of commercial-off-the-shelf (COTS) transmitter and receiver pairs operating at 2.4 GHz were set up to form a bistatic angle of 20° with the target [8]. The results showed that SNR improved by 3 dB with data fusion of both monostatic and bistatic returns as compared to only monostatic returns. The experiment also showed that the fusion of non-coherent returns does not improve the SNR.

The greatest advantage offered by the PBR configuration lies in its ability to leverage the EM emission of opportunistic transmitters to detect targets. This allows the receiver to detect and track targets without the target being aware of the radar's presence. One of the key challenges of this approach is to ensure consistent detection coverage and good tracking accuracy. One of the solutions proposed is to derive a method to identify the suitable transmitter-receiver geometry pair to achieve good coverage. By applying the Cramer-Rao lower-bound (CRLB) to the bistatic ambiguity function, we find that the results show that regardless of the transmitter-receiver pair chosen, the range and velocity error is lowest when the target is close to and within the vicinity of the transmitter and receiver [9]. The error increases in a radially outwards manner as the target moves farther. In summary, to achieve good detection coverage and accuracy, there should be a transmitter or receiver in each sector-of-interest centered at the receiver. In one particular configuration, an array of transmitters can be chosen to form range rings around the receiver.

The ambiguity function of the radar waveform determines the accuracy of target tracking performance in both range and velocity. In the case of the bistatic radar (multistatic included), the ambiguity function incorporates many non-linear variables created by the spatial separation of the transmitter and receiver. Based on simulation, it was found that range resolution worsens with the increase of the bistatic angle. In the Doppler frequency domain, larger detection gaps and tracking errors are also created due to the geometry of transmitter-target-receiver [10]. In a later paper [11], the authors extend the investigation of the ambiguity function to cover coherent and non-coherent multistatic radars. The results agreed with the earlier findings and went further to emphasize the importance of time synchronization and positional accuracy of the transmitter and receiver to achieve good range accuracy. It also states that coherence between the transmitter and receiver improves the range resolution and accuracy.

From [12], improvement in detection and tracking accuracy can be achieved by fusion of data from multiple domains, e.g., time-of-arrival (TOA), phase-on-arrival (POA) and angle-of-arrival (AOA). This approach reduces the tracking error by overlapping the area of uncertainty of the target and taking the overlap area (smallest) of the three as the final range resolution. It also adds frequency diversity by transmitting a range of frequencies with multiple transmitters to further enhance performance against targets of different dimensions that are moving at different speeds. The results also show that further improvement in tracking accuracy can be achieved with a larger separation between transmitters.

In summary, there is a strong interest in a PBR system working with opportunistic transmitters. Experimental systems of this type are almost always used exclusively to detect aircraft in a terrestrial environment. The choice of transmitter waveform is almost always broadband, continuous transmission with or without FM chirp. In the maritime domain, the use of bistatic systems is almost always in the forward scatter mode with a

configurable transmitter. The focus lies in target parameter extraction and classification using ISAR techniques. Multistatic systems with good detection range and accuracy almost always maintain coherency between transmitter and receiver. It was also found that a narrowband transmitter waveform performs better in PBR. Good spatial placement of the transmitters by sector with sufficient distance improves detection coverage. In the case of tracking accuracy, the importance of having good transmitter and receiver position information and good time synchronization cannot be understated. One could improve range accuracy by choosing transmitters near to intended targets and keeping the bistatic angle between transmitter-receiver to between 30° and 160°.

B. THESIS OBJECTIVES

The goals of this thesis are to 1) evaluate the suitability of using passive multistatic radar to detect low-RCS marine targets using civil marine radar, 2) propose the optimal multistatic radar system configuration for best detection coverage, and 3) quantify the detection accuracy of the optimal configuration using MATLAB simulation. The simulation results are obtained using actual system parameters extracted from civil marine radar equipment manuals or open sources materials.

The proposed multistatic radar system configuration is a single receiver (warship) and multiple transmitters (merchant ships) configurations. The scenario assumes a single warship is tasked to carry out comprehensive surveillance of a sector out-at-sea to enhance maritime domain awareness. Presumably, the warship is positioned somewhere in the South China Sea to monitor ships going into and out of the Singapore Straits. The targets are assumed to have low-RCS design, operating with either LPI radars or search radars with infrequent and irregular scans. The own-force warship is equipped with a broadband EW receiver with direction-finding (DF) capability. The merchant ships and the warship are equipped with class-A maritime automatic identification system (MAIS) regulated under SOLAS, which functions like identification friend or foe (IFF) on aircraft, sending out information such as identity, global positioning system (GPS) position, course and speed at short intervals on a very high (VHF) frequency. The pictorial representation of the scenario is illustrated in Figure 2.



Coast

Figure 2. Pictorial representation of the PBR setup.

Typically, the process of target detection begins with the warship receiving all merchant ships' MAIS transmissions on its own transponder. The warship forms a tactical picture of all known targets on its combat management system (CMS) console screen. Using the onboard EW and DF receiver, each of the CMR's transmissions in both S-band and X-band can be correlated with the individual merchant ships. With these two sets of information, the EW DF receiver is now able to discern the same civic marine radar's transmission parameter coming from two different bearings due to the direct path and reflection off an unknown target. The difference in the time-of-arrival of the direct and indirect signals is used to compute the range of the unknown target from the warship. This set of range and bearing information has inherent errors due to DF error, data age and uncertainty in the positional information of the opportunistic transmitter. The error can be minimized by averaging and combining multiple range computations of the same target using another bistatic pair.

The first step is to determine the maximum possible detection range based the radar horizon and the maximum detection range using the bistatic radar equation. The latter takes into consideration system and physical parameters like antenna gains, transmission losses, frequency, bistatic RCS, antenna beamwidth, bistatic geometry of transmitter-target-receiver, receiver sensitivity for detection, etc.

Next, evaluation of the combined detection range for different bistatic geometries, based on the SNR at the receiver, is presented. Based on the coverage characteristics of individual bistatic geometry, combinations of multiple bistatic pairs at different angles and ranges are simulated to determine the optimal detection coverage. The simulations are run by varying frequency between S-band and X-band while combining them with different bistatic pairs.

Finally, the range accuracy is determined for each of the previously mentioned simulation runs. The objective is to evaluate the worst-case position uncertainty of the target and assess how choices of different bistatic pairs are affecting the results. The combination of these two results provides the basis by which the receiver can choose different combinations of bistatic pairs for optimal detection coverage and detection accuracy.

C. THESIS OUTLINE

The remaining parts of the thesis are organized as follows. The background theoretical concepts needed to understand and carry out the analysis are covered in Chapter II. Starting with Chapter III, we present simulation results in a progressive order with the detection coverage of a single bistatic pair. Different combinations of multistatic geometry are presented in Chapter IV with analysis on the optimal detection coverage. The expected detection accuracy of different multistatic geometries and possible improvements are presented in Chapter V. Finally, the conclusions drawn based on the results obtained and suggestions for future work are found in Chapter VI.

II. BACKGROUND

A. CIVIC MARINE RADAR AND MAIS

Probability of target detection by radar is determined by the SNR. A higher SNR increases the probability of detection. This is also true in the case of a PBR; the only difference is that the CMR's signal comes in many forms and may not be designed to optimize detection. Therefore, the first step is to determine the availability of the said opportunistic transmitters, their performance specifications and operational characteristics.

The purpose of CMR, as defined by the IMO, is to "assist in safe navigation and in avoiding collision by providing an indication, in relation to one's ship, of the position of other surface craft." The technology of the CMR is heavily influenced by the regulations issued by the UN-based IMO. CMR is restricted to operate in the 3 GHz range (for good range performance) and the 9 GHz range (for enhanced detection performance in rain and fog). Since 2004, IMO has encouraged the use of coherent radars in place of the simple pulsed-radar CMR for improved detection performance against heavy sea clutter conditions [13]. The New Technology Radars, as they are known, do not have restrictions on the type of waveforms that can be used. Some companies have developed LPI CMR that is marketed for its low power requirement and improved reliability [14].

According to IMO regulations, any vessel exceeding 300 gross metric tonnage is to be fitted with a CMR. This translates to more than 50,000 ships sailing in international waters, and many of them are fitted with two radars operating in different frequency bands to enhance navigation safety. This same regulation also applies in the carriage of MAIS but with the coverage expanded to include ships exceeding 500 gross metric tonnage not on international voyage and all passenger ships irrespective of size [15]. A good estimate of the number of ships installed with CMR and MAIS, passing through the Singapore and Malacca Straits is shown in the Figures 3 and 4. Notice that in both scenes,
there are over 500 vessels that are within the reception range of the shore-based MAIS system as captured by a free online resource, www.marinetraffic.com.



Figure 3. Snapshot of MAIS targets along Malacca Straits dated Feb. 20, 2014.



Figure 4. Snapshot of MAIS targets along Singapore Straits dated Feb. 20, 2014.

According to a report by the U.S. Energy Information Administration, over 60,000 seagoing vessels transit through the Malacca Straits (passing through the Singapore Straits) every year [16]. Assuming even distribution, one expects a daily

average of approximately 150 vessels transiting though the Singapore Straits on the Indian Ocean-South China Sea shipping route. To further narrow this down, the assumption made in the simulation is the use of frigate-class warship as own ship and the largest merchant vessels that conform to the MALACCAMAX to determine the radar horizon. MALACCAMAX is the term used in Naval Architecture to denote the largest seagoing vessel with a maximum draft of 25 m that can fit through the Malacca Straits. According to [17], the numbers of conventional cargo ship, tanker, container ships and roll-on roll-off (RORO) ships passing through the Malacca Straits in 1994 was 11620, 9688, 5244 and 1130, respectively. That translates to an average of 76 vessels on any given day.

To compute the radar horizon, two parameters are needed according to the radar horizon formula (standard atmosphere) [14]:

$$R_{\rm max} = 2.21\sqrt{h_{tx}} + 2.21\sqrt{h_{tx}} \tag{1}$$

where R_{max} is the range in nautical miles, h_{tx} is the height of the transmitter antenna in meters and h_{rx} is the height of the receiver antenna in meters.

For seagoing vessels, their dimensions are defined in length, breadth, draught and tonnage. The height above sea level or air draught, as defined in Naval Architecture terms, is seldom specified. To approximate the height of the CMR above sea level, the new PANAMAX standard states that all ships passing through the Suez Canal must not have an air draught of more than 68 m, with the restriction set by the Suez Canal bridge. It is reasonable to assume the likes of oil tankers and containers ship on international voyage are designed to conform to this requirement. For a commercial vessel, the CMR is normally installed at the highest point. On the other end, the height of the EW receiver on the frigate is approximated to be at 20 m above sea level. Therefore, R_{max} is calculated from (1) to be 52 km.

There are many international companies like the UK-based Kelvin Hughes and Japanese-based Furuno that supply the world's shipping industry with integrated bridge control systems along with CMR and MAIS. CMR such as Kelvin Hughes's MantaDigital Radar (See Appendix A for data sheet) and Furuno's FAR28x7 series (See Appendix B for data sheet) are the most commonly installed onboard larger class vessels. Out of all the technical specifications, a few parameters are needed for the computation of the SNR for detection. The specifications and comparison between the two CMRs are presented in Table 2.

Technical	MantaDigital		FAR28x7 series		
Specifications	S-Band	X-band	S-band	X-band	
Pulse Width	55 ns (Short),	70 ns (Short),	0.5 µs (Short),	0.5 µs (Short),	
	250 ns (Med),	250 ns (Med),	0.7 μs (Med),	0.7 µs (Med),	
	950 ns (Long)	900 ns (Long)	1.2 µs (Long)	1.2 µs (Long)	
Beamwidth	26°	-	25°	20°	
(Vertical)					
Beamwidth	1.9°	1.25°	1.8°	1.23°	
(Horizontal)					
Pulse	3000 Hz (Short),	3000 Hz (Short),	1000 Hz	1000 Hz (Short),	
Repetitive	750 Hz (Med),	1500 Hz (Med),	(Short),	600 Hz (Med),	
Frequency	750 Hz (Long)	750 Hz (Long)	600 Hz (Med),	600 Hz (Long)	
			600 Hz (Long)		
Receiver	< 6 dB	< 6 dB	-	-	
Noise Figure					
Antenna	28 dB	30 dB	-	-	
Gain					
Transmitter	30 kW	25 kW	30 kW	30 kW	
Power					
Frequency	3050 MHz +/- 10	9410 MHz +/- 30	3050 MHz +/-	9410 MHz +/- 30	
	MHz	MHz	30 MHz	MHz	
Scan Rate	24 rpm/ 45 rpm	24 rpm/ 45 rpm	24 rpm/ 42 rpm	21 rpm/ 26 rpm/	
				45 rpm	

Table 2. Comparison of two of most commonly installed CMR.

In reference to the technical specifications shown in Table 2, both CMRs share similar parameters, but not all parameters of the FAR-28x7 series are available in its product catalog. For the purpose of this thesis, the technical specifications of the MantaDigital Radar are used to compute the SNR and maximum detection range of the PBR.

After selecting the technical specifications of the opportunistic radar, the next step is to obtain an accurate position of all available opportunistic radars so as to calculate the baseline distance between the transmitter and EW receiver on the warship. Furthermore, the receiver needs to be able to uniquely classify each CMR transmission and associate it with the source vessel on the CMS. From the operating specifications requirement stated by the IMO, the MAIS's function is to support automated tracking and identification among vessels and port authorities. It operates autonomously and continuously to exchange ship identity and GPS locations using the VHF transceiver. There are two types of MAIS: Class A installed onboard a majority of seagoing commercial vessel, and the more economical Class B used typically by leisure and smaller crafts. The MAIS communication network uses self-organizing time division multiple access (SOTDMA) to allow for at least 4500 communication timeslots for vessels to exchange up to 27 different National Marine Electronics Association (NMEA) – 0183 messages, sorted into different level of priorities. The highest priority message is the GPS positional information accompanied by a unique nine-digit identification number assigned to each vessel known as the maritime mobile service identity (MMSI). This information allows the warship to discern targets of interest from neutrals on its CMS display. Extracted from the U.S. Coast Guard website, the key difference in specifications between Class A and Class B MAIS are summarized in Table 3.

Shipboard AIS	Class A	Class B/Self-	Class B/Carrier	
		Organizing	Sense	
Transmit Power	12.5 W / 2 W	5 W / 2 W	2 W	
(Watts)				
Primary Access	SOTDMA	SOTDMA Carrier-	TDMA	
Scheme		sense	non-competing with	
			SOTDMA units	
Position Reporting	Either every 2, 3 ¹ / ₂ , 6	Either every 5, 15 or	Every 30 s	
Rate	or 10 s based on	30 s	Every 3 min. when <	
	speed and course	based on speed Every	2 kts.	
	change. Every 3 min	$3 \min \text{when} < 2 \text{ kts}$		
	when < 3 kts			
Static Data	Every 6 min	Every 6 min	Every 6 min	
Reporting Rate				
Transmit Data	All	No Rate of Turn,	No Rate of Turn,	
		Navigation Status,	Navigation Status,	
		Destination, ETA,	Destination, ETA,	

Table 3. Comparison of specifications between Class A and Class BMAIS.

From the U.S. Coast Guard website, the worst-case GPS positional accuracy of the MAIS is 0.001 minute in both latitude and longitude with an update rate of no longer than 10 s for a Class A MAIS. That translates to a positional error of not more than 2 m all round. With reference to the MantaDigital Radar chosen earlier, the same class of vessel is equipped with a Class A MAIS. For the purpose of this thesis, the information provided by Class A MAIS is used to calculate the baseline distance and the detection accuracy of the target.

B. EW AND DF RECEIVERS

Onboard a warship, the EW receiver is responsible for the extraction of key parameters from the intercepted transmission for follow-on classification and identification typically through digital radio frequency memory (DRFM). The EW receiver is capable of measuring the intercepted transmission's center frequency, pulse width, pulse repetitive frequency (PRF), antenna scan rate, type of radar, transmission mode, etc. Integrated with the DF receiver, the angle-of-arrival can be determined and used to point to the source of transmission with the aid of a bearing line. Parameters like these are saved in the DRFM to help with future identifications.

The important parameters to quantify on the EW and DF receiver are the lowest received power that can be detected, known as sensitivity, bandwidth of the receiver, detection bandwidth and DF accuracy. Due to the classified nature of these parameters, they are seldom available in open source literature. As such, the parameters are based on systems with published specifications that best matched our scenario and as practical as reasonable.

The EW receiver specifications are based on UK-based Teledyne Defense Ltd.'s high performance receiver model QR026 (See Appendix C for data sheet). This wideband receiver operates from 0.2 GHz to 18 GHz and is expandable to frequencies up to 40 GHz. It has a sensitivity of approximately -70 dBm, a typical noise figure of 13.5 dB, and a receiver gain of 20 dB if operating with a scanning bandwidth of 500 MHz. Most EW receivers support tuning of bandwidth to enhance detection of specific frequency band and the use of a notch or Yttrium iron garnet (YIG) filter to carry out narrowband detection and classification. For our case, the bandwidth of the receiver can tuned to coincide with the frequency band of interest, thus reducing additive white gaussian noise (AWGN), thereby, improving the sensitivity and SNR.

The DF receiver specifications are based on South Africa's Poynting Defense's land-based DF Antenna model DF-A0062 (See Appendix D for data sheet), which has an effective DF range from 20 MHz to 6 GHz and U.S.-based DRS Defense Solutions shipboard DF Antenna array model ZA-4501. Both DF antenna systems work on a five-element phase interferometer concept with an azimuth DF accuracy of between 1° to 2°. For the purpose of this thesis, the average DF accuracy of 1.5° is used.

C. BISTATIC RCS

Bistatic RCS of a target is dependent on the angle formed between the direction of incidence and the direction of reception, known as the bistatic angle β . Unlike the monostatic RCS, which is a measure of the amount of EM energy re-radiated back in the direction of incidence, the bistatic RCS is the measure of the amount of EM energy returned in the direction of the receiver that is not co-located with the transmitter. Putting everything in perspective, there is only one set of 360 values in monostatic RCS

corresponding to 360 degrees in azimuth for each target. However, in the case of the bistatic RCS, each direction of incidence has its own set of RCS values for 360 degrees. So to completely map out the bistatic RCS of a target, there are 360 sets of values per target.

The definition of RCS can be stated [2] as

$$\sigma = \lim_{R \to \infty} 4\pi R^2 \frac{\left|\vec{E}_s\right|^2}{\left|\vec{E}_i\right|^2}.$$
(2)

For simple geometrical shapes, the RCS can be easily calculated using

$$\sigma = \frac{4\pi A_{ea}^2}{\lambda^2} \tag{3}$$

where A_{ea}^2 is the equivalent area in m² and λ is the wavelength in meters. In the case of a complex-shaped low-RCS warship, the approximation of the RCS has to take into consideration the 'electrical size' of the target, which is defined by the largest relative dimension of the target in terms of the incident wavelength. There are three frequency regions in which the RCS of a target can vary significantly, namely the Rayleigh region (low frequency), Mie region (resonance) and Optical region (high frequency). The three regions are classified based on the magnitude of the 'electrical size.' The graph (Figure 5) presents the effect on the monostatic RCS by varying the wavelength on the same sphere.



Figure 5. Effects on monostatic RCS of a sphere with changes to the wavelength. From [1].

Due to the profound effect of the ratio of the target length versus the input transmission wavelength, different methods of RCS prediction are used in response to the dominant scattering mechanism that is present. The RCS prediction methods most commonly used are physical optics, microwave optics, method of moments, the finite element method and the finite difference method. For our case, the low-RCS warship is chosen to be the Republic of Singapore Navy's (RSN) Formidable-class frigates. A three-dimensional model (Figure 6 and 7) of the Formidable-class frigate is created in CST Studio and ported over to FEKO, a commercial RCS modeling software, to calculate the free space monostatic and bistatic RCS at S-band and X-band using physical optics. The physical optics method is suited for use to predict the RCS for targets that are in the optical frequency region.



Figure 6. Actual photo of RSN's Formidable-class stealth frigate (top, from [18]) versus its model (side profile).

A quick comparison between the monostatic RCS and bistatic RCS shows that the RCS differences can fluctuate up to a factor of 30 dB over an aspect change of a few degrees as illustrated in Figure 8. A sample set of bistatic RCS plots are attached as Appendix E, and the values are added to the MATLAB simulation in the area of detection coverage.



Figure 7. Actual photo of RSN's Formidable-class stealth frigate (top, from [19]) versus its model (front profile).



Figure 8. Comparison of monsotatic RCS (dBsm) of the model measured at 3 GHz (left) and bistatic RCS with incidence angle at 3 degrees (right).

D. BISTATIC RADAR RANGE EQUATION

In reference [1], the bistatic radar range equation is derived. The single range in the monostatic case is now split into two range variables R_T and R_R , and the monostatic RCS is replaced with the bistatic RCS denoted by σ_B as

$$(R_T R_R)_{\max} = \left[\frac{P_T G_T G_R \lambda^2 \sigma_B F_T^2 F_R^2}{(4\pi)^3 k T_s B_n (S/N)_{\min} L_T L_R}\right]^2 = \kappa$$
(4)

where

 R_T = transmitter-to-target range,

 R_R = target-to-receiver range,

 P_T = transmitter power output,

 G_T = transmitter antenna power gain,

 G_R = receiving antenna power gain,

 λ = wavelength,

 σ_B = bistatic RCS,

 F_T = pattern propagation factor for transmitter-to-target path,

 F_{R} = pattern propagation factor for target-to-receiver path,

k = Boltzmann's constant,

 T_s = receiving system noise temperature,

 B_n = noise bandwidth of receiver's pre-detection filter, sufficient to pass all spectral components of the transmitted signal,

 $(S/N)_{min}$ = signal-to-noise power ratio required for detection,

 L_T = transmitting system losses (> 1) not included in other parameters,

 L_{R} = receiving system losses (> 1) not included in other parameters, and

 κ = bistatic maximum range product.

By grouping some of the terms listed, Eq. (4) can be presented in a more compact and generalized form for any SNR value. Defining

$$\mathbf{K} = \frac{P_T G_T G_R \lambda^2 \sigma_B F_T^2 F_R^2}{(4\pi)^3 k T_s B_n L_T L_R},$$
(5)

we can write, in general,

$$S/N = \frac{K}{R_T^2 R_R^2}.$$
 (6)

The SNR value is only sensitive to parameters associated with the CMR and EW receiver. It is not grounded in any coordinate system or locations of the transmitter, receiver and target.

From geometry (refer to Figure 1),

$$R_{R} = \frac{(R_{T} + R_{R})^{2} - L^{2}}{2(R_{T} + R_{R} + L\sin\theta_{R})}$$
(7)

where *L* is the baseline distance between the transmitter and receiver, and θ_R is the angle between North (y-axis) and the target.

In the real world, target detection begins with the measurement of $(R_T + R_R)$ using one of two methods, namely the direct method which requires a line-of-sight between the transmitter and receiver or the indirect method which requires synchronization of stable clocks between the transmitter and receiver. Given the use of opportunistic transmitters, the direct timing method is used here and

$$(R_T + R_R) = c\Delta T_n + L \tag{8}$$

where

c = speed of light, and

 ΔT_n = time interval between the reception of the transmitted pulse and reception of the target echo.

In this case, the EW receiver is the one providing this information. The EW receiver must first characterize the transmission of each CMR for quick recognition. Upon receiving the direct transmission from the CMR as verified by the DF receiver, the timer starts counting until it receives the same CMR transmission from a different bearing. This time difference is ΔT_n . The measured difference in bearing is then used to determine θ_R . With that, the detection coverage based on SNR can be plotted. The plot is

then normalized using the SNR_{min} to display the detection coverage supportable by the EW receiver. The SNR_{min} (in dB) is calculated from

$$Sensitivity = SNR_{\min} + 10\log(B_n) + NF_{rec} - 174$$
(9)

As shown in Eqs. (6) through (8), the SNR value corresponding to a specific combination of Tx-Tgt-Rx can be calculated given all the parameter values. Putting all these individual SNR contours together on a Cartesian coordinate plot, we find the detection coverage for a particular combination of PBR. Given the high availability of opportunistic transmitters out-at-sea, multiple combinations of PBR can be used for optimum coverage in real-time to allow for the highest probability of detection.

E. MEASUREMENT AND LOCATION ERROR

From [1], there are three geometry-dependent measurement and location errors in a PBR system, namely the receiver-to-target error due to measurement of $(R_T + R_R)$, baseline measurement error of L and receiver look angle error due to DF error by the DF receiver. The receiver-to-target error is the result of transmitter beamwidth spread at the target at R_T where the edge of azimuth beamwidth is strong enough to create returns back to the receiver. The baseline measurement error is dependent on the means by which the measurement of the baseline is obtained. In our case, the baseline L is computed as the difference between the differential GPS locations of the transmitter and receiver. As such, the measurement error is a function of the accuracy of the GPS position fix. The rootsum-squared (rss) of the three errors in [1] is expressed as

$$dR_{R} = \left\{ \left[\frac{\partial R_{R}}{\partial (R_{T} + R_{R})} d(R_{T} + R_{R}) \right]^{2} + \left(\frac{\partial R_{R}}{\partial L} dL \right)^{2} + \left(\frac{\partial R_{R}}{\partial \theta_{R}} d\theta_{R} \right)^{2} \right\}^{1/2}.$$
 (10)

Depending on the geometry of Tx-Tgt-Rx, the three errors are still additive but with components on different planes, as shown in Figures 9 through 11. With changing geometry, the combined error forms an ellipse with a different major and minor axis.



Figure 9. Illustration of measurement error due to variation of L.



Figure 10. Illustration of measurement error due to transmitter's θ_{AZ} .



Figure 11. Illustration of measurement errors due to receiver's DF error.

The three components of error contributing to the rss in Eq. (10) are

$$\frac{\partial R_R}{\partial (R_T + R_R)} = \frac{(R_T + R_R)^2 + L^2 + 2L(R_T + R_R)\sin\theta_R}{2(R_T + R_R + L\sin\theta_R)^2}$$
$$= \frac{1 + e^2 + 2e\sin\theta_R}{2(1 + e\sin\theta_R)^2},$$
(11)

$$\frac{\partial R_R}{\partial L} = \frac{[L^2 + (R_T + R_R)^2]\sin\theta_R + 2L(R_T + R_R)}{2(R_T + R_R + L\sin\theta_R)^2} = -\frac{(e^2 + 1)\sin\theta_R + 2e}{2(1 + e\sin\theta_R)^2},$$
(12)

and

$$\frac{\partial R_R}{\partial \theta_R} = -\frac{L[(R_T + R_R)^2 - L^2]\cos\theta_R}{2(R_T + R_R + L\sin\theta_R)^2}$$
$$= -\frac{L(1 - e^2)\cos\theta_R}{2(1 + e\sin\theta_R)^2}$$
(13)

where e = eccentricity given by $L/(R_T + R_R)$.

To begin using this PBR system in a real-world scenario, one has to first determine what CMRs (opportunistic radars) are within range and what combination of them offers the best overall detection coverage and, if not, at a minimum, optimize coverage within their operating areas.

With the warship's MAIS in the receive-only mode, all merchant ships that are within the VHF reception range are displayed on the CMS screen. Based on the returns from the MAIS, the merchant ships that operate with Class-A MAIS are selected, and their S-band and/or X-band CMRs' parameters are collected and analyzed using the EW receiver. The parameters are then saved into the DRFM and associated with the bearing relative to the warship (receiver) given by the DF receiver. The Class-A MAIS of all selected merchant ships are constantly updating their GPS positions at an update rate of no greater than 10 s. With the unique identifier provided by the MAIS, the specifications of the CMR can be determined. Putting all this information together, the crews onboard the warship have all the necessary information for Eqs. (5) and (8) to plot the detection coverage for a single PBR system. All the involved parameters tagged to their individual sources are summarized in Table 4. A further extension of this is to establish the relationship between different Tx-Rx PBR pairs and their associated detection coverage. The warship can choose multiple PBR pairs and combine their individual detection coverage plots to form a more comprehensive coverage in the area of interest. Different constellations of Tx-Rx pairs offer different advantages and disadvantages. Knowing this

in advance, along with practice, allows the warship operators to pick the best constellations from the prevailing availability of opportunistic transmitters in a minimum time.

Systems	MantaDigital (CMR)	Class-A (MAIS)	QR026 (EW Receiver)	DF-A0062 (DF Receiver)	FEKO RCS Modeling Software	Remarks
P _T	30 kW (S-band) 25 kW (X-band)	-	-	-	-	-
G_T	28 dB (S-band) 31 dB X-band)	-	-	-	-	-
G_R	-	-	2 dB	-	-	-
λ	-	-	0.0984 m (S-band) 0.0319 m (X-band)	-	-	-
$\sigma_{\scriptscriptstyle B}$	-	-	-	-	See Appendix E	-
F_T F_R	_	-	-	-	-	1 See reference [4]
T _s	-	-	3915 K (based on noise figure of 13.5 dB)	-	-	-
B_n	-	-	500 MHz	-	-	-
L_T L_R		-	-	-	-	2 See reference [4]
L	-	Based on GPS location of Tx and Rx	-	-	-	-
ΔT_n	-	-	Based on GPS location of Tx and Rx	-	-	-
θ_R	-	-	-	Bearing Line	-	-

Table 4. Parameters summary for Eqs. (5) and (8).

To plot an operationally usable detection plot, the minimum SNR with reference to the sensitivity of the EW receiver has to be determined using Eq. (9). This defines the value of SNR beyond which reception of the returns from the target is possible. Based on a sensitivity value of -70 dBm, the SNR_{min} required in our scenario is more than 3.5 dB for single pulse. To allow for unaccounted losses or unforeseen variations, an additional buffer is added to increase the SNR_{min} to 10 dB in our simulation.

In situations where signal strength of the returned signal is weak, non-coherent pulse integration across multiple pulses can improve the chance of detection by approximately \sqrt{N} dB, where N is the number of pulses used in the integration [19]. This approximation is heavily dependent on the type of receiver architecture used and is more applicable for envelope detector receiver types. The number of pulses transmitted downrange can be calculated using

$$N = \frac{\theta_{AZ}}{360(scanrate)}(PRF) \tag{14}$$

where

 θ_{AZ} = Azimuth beamwidth of the CMR in degrees, scanrate = CMR Antenna scan rate in revolution per second (rps), PRF = CMR's antenna PRF in Hz.

In the case of the S-band KH MantaDigital radar, the number of pulses that are available for integration per second is five pulses with θ_{AZ} of 1.9°, *scanrate* of 45 rpm and *PRF* of 750 Hz. This translates to a gain of 2.2 dB. Thus, by allowing for more integration time, it is possible to increase detection range. The low SNR of each pulse and the long integration time also bring with them increased measurement errors, specifically, in the TDOA measurements of the leading edge of the returned pulse. The increase in error is discussed in [20] for a monostatic radar, but the application to PBR adopting the discussed form of TDOA is not well-defined.

F. DETECTION ACCURACY

As the PBR system detects targets passively and displays them on the CMS screen, a problem of detection accuracy arises. As shown in Eqs. (13) through (15), there are measurement and location errors that form an uncertainty ellipse around the target. On top of these baseline errors, there are three sources of errors that add to the growing uncertainty ellipse. They are: (1) azimuth beam spread of the CMR with target range, (2) accuracy of the DF receiver and (3) positional errors due to data age and moving targets. These sources of errors are added in their respective planes and determine the uncertainty ellipse for each Tx-Rx pair. By knowing the passively-detected target range and azimuth with respect to the warship, one can compute the worst-case location errors in Cartesian coordinates. The location errors can be reduced by using more Tx-Rx pairs to narrow down the overlapped area.

In the next chapter, the simulation of PBR and PMR are presented. Different opportunistic transmitter placements are simulated to understand their effects on detection coverage and detection accuracy.

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III. SIMULATION RESULTS FOR DETECTION COVERAGE

In this chapter, the detection coverage of the PBR is investigated with different combinations of Tx-Rx constellations. The chapter begins with a single transmitter and receiver pair and demonstrates how a PBR system has a higher probability of detecting a low-RCS target. The simulation results also show the limited detection coverage of a single PBR system. The investigation goes on to explore a multiple PBR system and evaluates the detection performance of the combined system using different Tx-Rx constellations. All simulation results are generated using MATLAB with the source codes included in Appendix F.

A. SINGLE TX-RX PAIR

To appreciate the benefits brought about by combining multiple PBRs into a multistatic radar system, one first needs to understand the capabilities and limitations of a PBR system operating with S-band and X-band CMR.

1. Monostatic RCS with S-Band CMR

According Skolnik [14], at low grazing angle, the median RCS of a ship can be estimated using the empirical formula

$$\sigma = 52\sqrt{f}D^{3/2} \tag{15}$$

with the *f* in MHz and ship's displacement *D* in kilotons. Substituting *f* as 3050 MHz and *D* as 3.2 kilotons, the displacement of RSS Formidable-class frigate, we find the RCS of a non-RCS treated frigate to be 5137 m². A separate simulation using FEKO on the model based on RSN's Formidable-class frigate yields a more representative monostatic RCS plot (Figure 12). The monostatic RCS fluctuates from 0 dBsm to 72 dBsm dependent on the incidence angle. Using the results shown in Figure 12, we estimated the simulated median RCS as 18 dBsm (63 m²) for S-band.



Figure 12. Monostatic RCS of RSN's Formidable-class frigate simulated at 3050 MHz using FEKO.

Putting both median monostatic RCS values into simulations, we found the resulting detection coverage of a PBR with a S-band CMR at (-5000,0) as shown in Figures 13 and 14. These plots are generated using Matlab scatter plot function. Each data point is shown as a colored dot. The density of the data changes from plot to plot. Note that gaps between the individual dots on the low density plots are not areas of low SNR, which has been set at 10 dB.



Figure 13. SNR detection plot (dB) using empirical median RCS at S-band CMR with Rx at the origin and single Tx at (-5000,0).



Figure 14. SNR detection plot (dB) using simulated median RCS at S-band CMR with Rx at the origin and single Tx at (-5000,0).

With the minimum SNR set at 10 dB and the EW receiver bandwidth of 500 MHz over an integration time of 1 s, the radial detection coverage under such conditions extends to approximately 7500 m from the receiver with near 100% coverage. From Figure 14, the detection coverage with the use of simulated median RCS shows a drop in detection coverage range to that of 1500 m from the receiver. The observable gaps in the plot of Figure 14 are the result of plotting resolution and not actually detection gaps. The detection void between the transmitter and receiver is due to the limitation offered by the PBR geometry. As the target approaches the baseline formed by the transmitter and receiver for the TDOA calculation expressed in Eq. (8), the difference between the direct path *L* and the reflected path ($R_T + R_R$) converges to zero. All equations expressed in this thesis break down both in theory and in practice since the target is now directly between the transmitter and receiver line-of-sight. In such cases, the PBR functions as a forward scatter radar instead of a PBR.

2. Monostatic RCS with X-Band CMR

Using the Eq. (15), we can show the monostatic RCS of a typical frigate at Xband (9410 MHz) to be 28,875 m². A separate simulation using FEKO on the model based on RSN's Formidable-class frigate yields a more representative monostatic RCS plot (see Figure 15). The monostatic RCS fluctuates from -10 dBsm to 80 dBsm dependent on the incidence angle. Using the results shown in Figure 15, we find the estimated simulated median RCS to be 15 dBsm (32 m²) for X-band.



Figure 15. Monostatic RCS of RSN's Formidable-class frigate's model simulated at 9410 MHz using FEKO.

For the purpose of comparison, we put both median monostatic RCS values (median from formula and median from FEKO) into the simulation. The resulting detection coverage of a PBR with an X-band CMR at (-5000,0) is shown in Figures 16 and 17. The magnitude of the SNR behaves in the same way for all PBR systems with high SNR around the vicinity of the transmitter and receiver and reduces in a linearly radial manner. This simulation is consistent with findings by [10].



Figure 16. SNR detection plot (dB) using empirical median monstatic RCS from Eq. (15) at X-band CMR with Rx at the origin and single Tx at (-5000,0).



Figure 17. SNR detection plot (dB) using simulated median monostatic RCS from FEKO at X-band with Rx at the origin and single Tx at (-5000,0).

The low FEKO simulated monostatic RCS has significantly reduced the detection coverage to be restricted to regions around the transmitter and receiver. This indicates that without using a PBR, the ability to detect a low-RCS target monostatically is severely limited.

3. Bistatic RCS with S-Band CMR

With the inclusion of bistatic RCS in the simulation, the detection coverage varies with the change in target aspect with respect to the transmitter and receiver. The effect of the bistatic RCS is most evident with changes made to bistatic angle β by changing the course and heading of the target. Many possible combinations of β with different bistatic geometry of Tx-Tgt-Rx exist. The most representative are those with the target's general direction headed towards the receiver and the other with the target's broadside directed towards the receiver on a crossing course. The former presents one of the smallest bistatic RCS returns, while the latter presents a larger bistatic RCS return. With the ship forward and aft line along the 0° and 180°, it can be seen from Figure 18 that the bistatic returns are mostly small with a narrow peak occurring at the negative of the incidence angle but always higher than the direct monostatic returns.

The RCS data was generated in 1° increments. When used in the PBR simulations, the incidence and observation angles were rounded to the nearest degree in order to obtain the RCS.



Figure 18. Bistatic RCS of RSN's Formidable-class frigate's model simulated at 3050 MHz using FEKO.

A comparison of Figures 17 and 19 shows clearly the difference in detection coverage when a different aspect of the low-RCS target is presented to the transmitter and receiver. The detection range increases from 3000 m to that of beyond 20,000 m with a change of target aspect. It is, therefore, important to understand a single PBR system is also limited in its ability to ensure consistent detection coverage and low detection gaps. The resultant detection coverage is dependent on the Tx-Tgt-Rx geometry and the aspect of the target.



Figure 19. SNR detection plot (dB) using bistatic RCS at S-band having reflections off the forward aspect of the target ship by Tx1 (-5000,0).



Figure 20. SNR detection plot (dB) using bistatic RCS at S-band having reflections off the broadside of the target ship by Tx1 (-5000,0).

Due to the large comparative difference in the ship's dimensions with respect to the wavelength of the transmission, a higher resolution plot of the target RCS reveals more RCS fluctuations between the current sampling intervals of one degree.

4. Bistatic RCS with X-Band CMR

Similar conclusions can be drawn from the detection coverage plot, as shown in Figures 21 and 22, with an X-band CMR. In general, the X-band CMR as an opportunistic transmitter has a smaller detection coverage as compared to the S-band CMR. The detection gaps are also more pronounced. In a real-world scenario, the use of RAM becomes more practicable at small wavelengths. It follows that a low-RCS target is likely to have even lower RCS returns at higher frequencies, reducing the usefulness of X-band CMR as an opportunistic transmitter.



Figure 21. SNR detection plot (dB) using bistatic RCS at X-band having reflections off the forward aspect of the target ship by Tx1 (-5000,0).



Figure 22. SNR detection plot (dB) using bistatic RCS at X-band having reflections off the broadside of the target ship by Tx1 (-5000,0).

B. TWO TX-RX PAIR

Equipped with the knowledge of how a single PBR system is limited in its ability to provide consistent detection coverage, it is important to explore passive multistatic radar (PMR) and how a different transmitter constellation affects the detection coverage. The three parameters for transmitter constellation selection are the quantity of transmitters and their position relative to the receiver.

1. Bistatic RCS with Two S-Band CMRs

In this two-transmitter PMR system setup, the two transmitters (Tx1 and Tx2) are 90° degrees apart and share the same baseline distance of 5000 m. Although the subsequent detection plots bear semblance to the detection plots of a single Tx (Figure 23), it is important to note that detection gaps have been reduced in all cases (Figure 24), and the two-transmitter PMR system is now more tolerant of target RCS changes due to aspect. From Figures 25 and 26, the detection plots show the detection coverage of the Tx-Tgt-Rx geometry has at least one broadside reflection return from the low-RCS target.



Figure 23. SNR detection plot (dB) with 2Tx at S-band (90° apart) having reflections off the forward aspect of the target ship.



Figure 24. SNR detection plot (dB) with 2Tx at S-band having reflections off the broadside of the target ship.



Figure 25. SNR detection plot (dB) at S-band with Tx1 with reflections off the forward aspect of the target ship and Tx2 off the broadside of the target ship.



Figure 26. SNR detection plot (dB) at S-band with Tx1 with reflections off the broadside of the target ship and Tx2 off the forward aspect of the target ship.

2. Bistatic RCS with X-Band CMR

Extending the simulation using X-band CMR, we show the detection plots in Figures 27 through 30. They are consistent with that using S-band CMR as the opportunistic transmitter.



Figure 27. SNR detection plot (dB) with 2Tx at X-band (90° apart) with reflections off the forward aspect of the target ship.

From Figures 29 and 30, it can be observed that there two distinct shades of blue denoting the detection area about 5000 m further away from the receiver. The dark shades denote regions where the SNR is above the detection threshold of 10 dB but less than 40 dB. This dark shade of blue exists only between ranges above 5000 m and 12,000 m, while there are repeated patterns of coverage and non-coverage when the target range increases.

To explain this phenomenon, recall the variables in Eqs. (5) and (6) that affect the SNR. The obvious factor is the combined value of $R_T^2 R_R^2$ that is non-linear in nature acting as the denominator in Eq. (6). With β constant, the change in $R_T^2 R_R^2$ affects the SNR. The more asymmetric the R_T and R_R are, the larger the SNR. This explains why the SNR increases at the ends of the constant β -contours; however, if the bistatic RCS returns corresponding to a β are small, there are detection gaps for that value of β . In summary, there are two variables involved in determining the SNR, namely, the bistatic RCS returns which are sensitive to β and $R_T^2 R_R^2$, where the latter loses its dominance as the target moves further away from the receiver.



Figure 28. SNR detection plot (dB) with 2Tx at X-band with reflections off the broadside of the target ship.



Figure 29. SNR detection plot (dB) at X-band with Tx1 with reflections off the forward aspect of the target ship and Tx2 off the broadside of the target ship.



Figure 30. SNR detection plot (dB) at X-band with Tx1 with reflections off the broadside of the target ship and Tx2 off the forward aspect of the target ship.

3. Combination and Constellation

The approach taken to evaluate a transmitter constellation is to compare the detection coverage with a monostatic radar. That requirement is to achieve all-round coverage with a minimum number of detection gaps. A series of different constellations are explored to establish a decision matrix to choose the optimal number of opportunistic transmitters to maximize coverage and minimize gaps. To simplify the simulation without sacrificing validity, all subsequent detection plots are shown with strong RCS returns using S-band CMR. This is fair as the low-RCS target presents a different aspect to each transmitter, and at least one of these combinations invokes a strong RCS return back to the receiver.

As shown in Figures 31 through 36, both detection coverage and range can be improved by choosing an opportunistic transmitter that has a larger baseline distance from the receiver. The upper limit of the choice of baseline distance lies in the transmitter's power and the receiver's sensitivity. Complemented by another transmitter 90° or 180° apart, the detection gap is significantly reduced. It is also shown that detection gaps in a particular bearing can be reduced by a transmitter that lies orthogonally to it.



Figure 31. SNR detection plot with Tx1 at (-5000,0) and Tx2 at (5000,0).

Comparing Figures 31 and 32, we see that detection gaps to the right of the receiver are reduced with Tx2 at 90° as compared to 180° . The converse is also true for detection gaps that are reduced forward of the receiver if both Tx1 and Tx2 lie on the same baseline as the receiver.



Figure 32. SNR detection plot with Tx1 at (-5000,0) and Tx2 at (0,8000).

It is noteworthy that the detection gaps can be classified into two types, interval gaps and through gaps. The interval gaps represent detection gaps that appear in a lateral manner and create loss in tracking as the target moves towards or away from the receiver. The through gaps are detection gaps that allow the target to evade detection and reach the receiver. Refer to Figure 32 for a depiction of how the interval gaps are detection gaps that form concentric ovals connecting the transmitter and receiver. The through gaps are detection with the transmitter (Tx1) on the left or the receiver on the right.



Figure 33. SNR detection plot with Tx1 at (-5000,0) and Tx2 at (0,10000).



Figure 34. SNR detection plot with Tx1 at (-8000,0) and Tx2 at (0,10000).


Figure 35. SNR detection plot with Tx1 at (-10000,0) and Tx2 at (0,10000).



Figure 36. SNR detection plot with Tx1 at (-10000,0) and Tx2 at (10000,0).

To minimize detection gaps as a result of the bistatic angle formed between the Tx-Tgt-Rx, the optimal transmitter constellation should have transmitters at every quadrant and of varying baseline distance. This is consistent with the findings by [10].

C. TRANSMITTER CONSTELLATIONS

How different constellations formed using 4Tx-Rx and 8Tx-Rx affects detection coverage is explored in this section.

1. Four Tx-Rx Pair

In this simulation, four transmitters are placed 10000 m away from the receiver, each occupying a quadrant at 90° apart (Figure 37). It is worth noting that 100% detection coverage is achieved within a 20 km radius of the receiver. The second detection tier between 20 km to 28 km has about 60% detection coverage. An increase in the baseline distance from 10 km to 15 km (Figure 38) also saw an improvement in detection range in the second detection tier with the reduction of through gaps. The improvement in detection coverage in the second tier increased from 60% to approximately 80%.



Figure 37. SNR detection coverage plot with 4Tx (0°, 90°, 180° and 270°) at 10000 m from the receiver.



Figure 38. SNR detection coverage plot with 4Tx (0°, 90°, 180° and 270°) at 15000 m from the receiver.



Figure 39. SNR detection coverage plot with 4Tx (45°, 135°, 225° and 315°) at 15000 m from the receiver.

As shown in Figure 39, the 4Tx constellation placement with transmitters at 45°, 135°, 225° and 315°, respectively, has modified the second tier detection coverage between 20 km and 28 km. By rotating the transmitter constellation by 45°, the location of through detection gaps in the second tier has also rotated. Essentially, the use of a different constellation of the four transmitters allows the receiver to direct optimal detection coverage in its quadrant of choice.

2. Eight Tx-Rx Pair

The use of eight transmitters offers the best all-round detection coverage with near 95% coverage up to a range of 30 km. From Figures 40 and 41, one can see the detection coverage by eight transmitters placed at 15000 m and 18000 m away from the receiver, respectively.



Figure 40. SNR detection coverage plot with 8Tx (0°, 45°, 90°, 135°, 180°, 225°, 270° and 315°) at 15000 m from the receiver.



Figure 41. SNR detection coverage plot with 8Tx (0°, 45°, 90°, 135°, 180°, 225°, 270° and 315°) at 18000 m from the receiver.

In this chapter it was shown how the placement of the opportunistic transmitters affects detection range and detection coverage. In the next chapter the behavior of the measurement errors and how to minimize errors through the use of opportunistic transmitter placement is investigated.

IV. SIMULATION RESULTS FOR MEASUREMENT ERRORS

Although an eight-transmitter constellation offers detection coverage nearly identical to that of a monostatic radar, there are measurement errors that have to factor into consideration. A target detection with an unacceptably large error is not useful. Depending on the geometry of the PBR, we found that the worst-case measurement error varies across range and bistatic angle. As shown in Eqs. (11) through (13), the RCS of the target does not have a role to play in measurement error; measurement error calculations come after target detection with a SNR of more than 10 dB. Despite the varying nature of the measurement errors, it is possible to use a careful choice of transmitter constellation to minimize the errors.

A. UNCERTAINTY ELLIPSE

As explained earlier with the aid of Figure 9, every target position with a Tx-Rx geometry pair has its own measurement error. The measurement errors typically span three different axes and form an uncertainty ellipse around the target position. It should be interpreted as the target's actual location is anywhere within the ellipse. If the target is also tracked by another Tx-Rx pair, the resultant uncertainty ellipse is likely to have the dominant measurement error on a different axis. In such cases, the uncertainty of the target's position is further reduced to the overlapped area. This improvement extends to multiple pairs of Tx-Rx on the same target.

To be commensurate with the detection plots shown in Chapter III, the uncertainty ellipse of a different target location is plotted using the same transmitter constellations. The MATLAB source code for the following plots is included as Appendix G.

1. Two Tx-Rx Pairs

In this case, the target is 20.6 km away from the receiver with two transmitters 90° apart, as shown in Figure 42. The inputs to this simulation plot assume transmitter positioning update by the MAIS at 5 s and a net displacement of the target at 10 knots.

The uncertainty ellipse around the target (-5000, 20000) is the overlap area between the two uncertainty ellipses formed. At this extended distance, the target's measurement error area is approximately 120 m all round. Comparing with the target length of 114 m, we assess the error as reasonable.



Figure 42. Uncertainty ellipse of Target (-5000,20000) with 2Tx at 90° apart.



Figure 43. Uncertainty ellipse of Target (5000,20000) with 2Tx at 90° apart.

A change of target position to the right of the Tx-Tx constellation shows a slightly larger uncertainty ellipse due to geometry (Figure 43). As the target approaches the Tx-Rx geometry, one can see the uncertainty ellipse reduces with an all-round error of 40 m as shown in Figure 42. In certain geometries (Figure 45), the uncertainty ellipse is the same from two different transmitter locations.



Figure 44. Uncertainty ellipse of Target (5000,5000) with 2Tx at 90° apart.



Figure 45. Uncertainty ellipse of Target (-2000,10000) with 2Tx at 180° apart.

2. Four Tx-Rx Pairs

In a four-transmitter constellation, the effect on the uncertainty ellipse is similar (Figure 46). The major factors affecting the size of the uncertainty ellipse are the beamwidth of the transmitter and DF accuracy of the receiver. As the target approaches the transmitter or the receiver, the dominant factor changes. The important observation here is that the movement of the transmitter, receiver and target has very little influence

in determining the uncertainty ellipse. With the target at (4500,4500) and using the Sband CMR with a beamwidth of 1.8°, one can see from Figure 47 that the shape of the uncertainty ellipse has changed. Extending this further, we see the uncertainty ellipse of the same constellation with the transmitter changed to X-band CMR with a beamwidth of 0.95° as illustrated in Figure 48. For Figure 49, the transmitter, receiver and target speeds have been reduced to zero from 10 knots. It can be seen that the data age and target speed have negligible effects on the measurement error in this setup.



Figure 46. Uncertainty ellipse of Target (8000,13000) with 4Tx at 90° apart from each other.



Figure 47. Uncertainty ellipse of Target (4500,4500) with 4Tx at 90° apart from each other using S-band CMR.



Figure 48. Uncertainty ellipse of Target (4500,4500) with 4Tx at 90° apart from each other using X-band CMR.



Figure 49. Uncertainty ellipse of Target (4500,4500) with 4Tx at 90° apart from each other using X-band CMR and zero net speed for transmitter, receiver and target.

3. Eight Tx-Rx Pairs

One can expect similar results from an eight-transmitter constellation when compared to the four-transmitter constellation; however, it is worthwhile to note that the higher the number of transmitters and the more they are spatially diversified, the higher the number of target positions that the PMR can achieve and the smaller the uncertainty ellipse that results. Therefore, a higher accuracy of detection can be obtained. In Figure 50, an eight-transmitter constellation is set up similar to that shown in Chapter III. Letting the target position be (4500,4500), it can be seen that a very small measurement error of approximately 20 m is achievable with only four transmitters out of the eight transmitters used. This should be interpreted as the PMR system now has eight data sets to compare and choose from, so the best sets can be used to represent the uncertainty ellipse. To achieve the same low measurement error, one needs to actively seek transmitter spatial diversity if one wishes to focus on obtaining high accuracy on one target.



Figure 50. Uncertainty ellipse of Target (4500,4500) with 8Tx at 45° apart from each other using S-band CMR while the target is moving at the speed of 10 knots.

V. CONCLUSION

A. SUMMARY OF THESIS FINDINGS

Using opportunistic transmitters in the form of civil marine radar (S-band and Xband) to detect maritime targets passively is both viable and has comparable performance to monostatic radars. However, in the case of detecting low-RCS targets, the PMR outperforms the former. The highly regulated requirements instilled by the IMO have ensured that most of maritime vessels out-at-sea have the necessary information to support a PBR and PMR. With the world's trade routed via the sea, there is no shortage of suitable vessels possessing at least one CMR and one MAIS on the main shipping routes. Among all the suitable vessels, larger ships like oil tankers, container ships, etc., have their CMR installed at the highest point, which extends the radar horizon and detection range. The narrow beamwidth and high transmit power of the CMR conveniently lend themselves to exploitation by a PBR system in terms of both detection accuracy and detection range.

The unique geometry formed among the opportunistic transmitter, passive receiver and unsuspecting target provides exceptional military applications not achieved by current military technologies. It offers true passive detection (with zero transmission from the receiving warship) while at the same time supports detection coverage comparable to a monostatic radar. Passive detection means there is no minimum range detection that is unavoidable in a monostatic radar. In the same vein, there is no need for 'blanking' of the ESM receiver every time the onboard radar transmits. It allows for continuous monitoring of the EW spectrum for intelligence and threat. On the vulnerability front, this has removed the risk posed by ARM and, at the same time, has enhanced mission survivability, allowing the warship to continue with tracking information without a working radar or with a damaged radar.

This same unique geometry also improves the warship's probability of detecting a low-RCS ship treated both in shaping and RAM as compared to a monostatic radar. The use of low frequency S-band CMR poses serious limitations to the use of RAM treatment

on a low-RCS ship. It, therefore, restricts the effectiveness of low-RCS ships in the use of shaping. It was shown in the bistatic RCS measurement of the model based on the RSN Formidable-class stealthy frigate that shaping alone is capable of reducing the median RCS to values not higher than 200 m². With all else being equal with the effect of RCS on detection, the PBR system has an added advantage. The bistatic RCS simulation using FEKO shows that more incidence angles exist that demonstrate a spike in RCS returns. Simply put, there are more degrees-of-freedom created by the transmitter-target-receiver geometry that has a higher probability of directing spikes in the RCS returns toward the receiver. In the case of a monostatic radar, there are limited ways to maximize RCS returns because the shaping of a low-RCS is designed to reduce returns in the direction of the incidence angle.

The simplicity of PMR also lies in the fact that there are no expensive systems to build. Instead, all the information that the PMR system needs is already available on most warships. This information can be extracted and then used to carry out tracking.

1. Conclusion for Passive Detection Coverage

The passive detection range and coverage are related to many parameters on the transmitter, target and receiver. On the opportunistic transmitter end, the transmit power and antenna gain are the two parameters that determine the maximum detection range. However, it is the relative position of the transmitter in the PBR geometry that determines the detection coverage or detection gaps. On the receiver end, the deciding parameters are the sensitivity and bandwidth of the ESM receiver. These two parameters affect the SNR and, thereby, change maximum detection range. Similar to the opportunistic transmitter, it is the geometry of the PBR that affects the detection gaps. Collectively, the factor that is used here to quantify the geometry is the bistatic angle β , defined earlier in this thesis. Incidentally, the bistatic angle determines the bistatic RCS of what is presented to the receiver depending on the incidence angle of the transmitter's radar emission.

From the simulation plots presented in Chapter III, it was shown that when more transmitters are spaced further apart, they provide better overall coverage with small detection gaps. The minimum number of transmitters is four, placed 90 degrees apart, with one in each quadrant. To achieve maximum coverage with this particular combination of system parameters and transmitter constellation, a baseline distance of 18000 m is recommended. By rotating the quadrant placement of transmitters, one can direct the detection gaps away from a specific bearing of choice to improve detection.

Using more than four transmitters in the suggested constellation does not significantly extend the detection range. As shown in the eight-transmitter simulation, it does improve detection coverage from approximately 80% to 95% through the reduction of detection gaps. This translates to the ability of the PMR to track a target continuously provided the target offers a strong bistatic return to all eight transmitters. In all likelihood, a low-RCS target would have presented at least two to four strong RCS returns to either the quadrant-based constellation using four transmitters or the two-layered quadrant-based constellation using eight transmitters. If the availability of such a constellation is not possible out at sea, one should position oneself to create such a constellation without jeopardizing a passive posture.

2. Conclusion for Detection Accuracy

There are two dominant factors that affect detection accuracy, namely the beamwidth of the transmitter and the DF accuracy of the receiver. It was shown in multiple figures that the worst-case measurement error comes up to approximately ± 120 m for a low-RCS 114 m frigate at a range of 20 km. In most cases, a lower measurement can be achieved using multiple transmitters by overlapping their individual uncertainty ellipses. It was found that the four-transmitter and eight-transmitter constellations recommended for maximum detection coverage are also optimized to reduce the target's uncertainty ellipse. It was shown that with an eight-transmitter constellation, the area overlapped by all the uncertainty ellipses have given rise to a measurement error of 20 m for the same low-RCS target at a range of 6400 m.

B. FUTURE WORK

Improvements are needed in the simulation model. Higher resolution RCS data should be provided. This is possible by calling FEKO directly from the Matlab simulation with the exact incidence and observation angles. Also, the effect of multipath from the sea surface should be included.

A natural extension of this research is to look into tracking algorithms for such a PMR given that there is only range and no Doppler information. There are a couple of proposed tracking methods that might be suitable for the PMR system. There are possibilities for extracting target Doppler information from opportunistic targets using the coherent-on-receive method [21]. Such receivers support a pseudo-coherent reception of radar signals to extract Doppler information. With both range and Doppler information in a pseudo-coherent form, the improved PMR system can generate ambiguity function for better detection accuracy and also adopt typical tracking methods used in monostatic radars.

With detection and tracking assured, it is possible to take a step further into classification and identification of targets. From [6], forward scatter and backscatter based ISAR are currently being investigated for their ability to detect target outline. It might be interesting to consider adding another dimension to this by using bistatic ISAR. In combination, the three ISAR plots might offer added information about the target to help in classification and identification.

APPENDIX A. SPECIFICATION SHEET OF MANTADIGITAL RADAR BY KELVIN HUGHES



Transceivers

Transceiver Types

The Mantadigital[™] range includes both magnetron and solid state Radar transceivers. In both cases the latest lowprofile anniennas are used to reduce wind loading and increase in-service life. Upmast and downmast options are available in both X and S Band for the magnetron transceivers. In all cases high reliability and performance have been the design criteria. In all systems a transmission

Low-Profile Technology

Using the latest technology all Kelvin Hughes antennas use the polyrod lans principle. For both conventional and SharpEye¹⁴⁴ systems this results in an antenna with reduced vertical dimension which is significantly lighter that a conventional unit. This has the effect of reducing motor and gearbox loading and results in an increase in service life,

	Transceiver Characteristics												
	Х-Ва	S-Band Systems											
Transceiver	10kW Upmast	25kW Upmast	25kW Downmast	SharpEye TH	30kW Upmast	30kW Downmast	SharpEye™						
Peak Power	10kW	25kW	25kW	170W	30kW	30kW	170W						
Frequency	9410 +/=30MHz	9410 +/-30MHz	9410 +/-30MHz	9220 -A3 9480MHz	3050MHz +/- 10MHz	3050MHz +/- 10MHz	2900- 3100MHz						
Pulse Length/PRF [Short]	55ns/3000Hz	70ns/3000Hz	70ns/3000Hz	N/A	55ns/3000Hz	70ns/3000Hz	N/A						
Pulse Length/PRF (Med)	230ns/1500Hz	250ns/1500Hz	260ns/1500Hz	N/A	250ns/750Hz	260ns/750Hz	N/A						
Pulse Length/PRF (Long)	600ns/750Hz	900ns/750Hz	900ns/750Hz	N/A	950ns/750Hz	900ns/750Hz	N/A						
Receiver Noise Figure	<6dB	<6dB	<6dB	N/A	<6dB	<ódB	N/A						
Receiver F Frequency	60MHz	60MHz	60MHz	N/A	60MHz	60MHz	N/A						
Receiver Bandwidth	6MHz / 22MHz	6MHz/22MHz	6MHz/22MHz	20MHz	6MHz/22MHz	6MHz/22MHz	20MHz						
Minimum Range	<40m	<40m	<40m	<40m	<40m	<40m	<40m						
Range Resolution	<2m	<2m	<2m	<15m	<2m	<2m	<15m						
Compass Safe Dist.	3.0m /1.8m	3.0m / 1,8m	3.0m/1.8m & 1.4m/0.8m	3.0m / 1.8m	3.0m/1.8m	3_0m/1.8m & 1_4m/0.8m	3.0m / 1,8m						

Antenna Characterístics										
	X-Ba	nd Systems	-	S=Band Systems						
Antenna Length	1.3m	1.9m	2.5m	3,7m						
Beamwidth (H)	1.8°	1_25*	0.95*	1.9*						
Beamwidth (V)	25*			26*						
Antenna Gain	28dB	30dB	31dB	28dB						
Polarisation	Horizontal	Horizontal	Horizontal	Horizontal						
Sidelobes (+/= 10°)	< -26dB	< -26dB	< =26dB	< -30dB						
Sidelobes [->10*]	< -34dB	< -34dB	< -34dB	< -35dB						
Rotation Rate	24,4	45 rpm option ava	ilable	24, 45 rpm option available						

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KELVIN HUGHES marine systems



25kW Upmast X-Band and 30kW Upmast 5 - Band Transvelv

APPENDIX B. SPECIFICATION SHEET OF FAR28X7 SERIES BY FURUNO



Antenna Radiators **Target Tracking** 1 Type Slotted waveguide array 2. Beamwidth and sidelobe attenuation 1. Acquisition Auto or manual acquisition: 100 targets in 0.2-24 (32) nm 2. Tracking Auto tracking on all acquired targets X-Band S-Band XN-12AF XN-20AF XN-24AF 41 651 81 SN-36AF 12.ft Radiator Type Guard zone (Target Acquisition Area) 0,5 nm width sector, within 3-6 nm, desired bearing. Length Beamwidth(H) 1.9 0.95 18 123 nm width sector or polygon, desired range and bearing Beamwidth(W) 20 20 20 25 4. Past positions Sidelobe (visio 10) -24 dB -28 dB -28 dB -24 dB 5 or 10 past positions on all targets Side obe joutside to -30 dB •30 dB -32 d -32 dB 5. Collision warning CPA Limit: 0.1-20 nm, TCPA Limit: 1-60 minutes 3. Rotation Trial maneuver Dynamic or static, with selected delay time. X-Band Rotation 24 rpm 42 rpm Gear Box S-Band RSB-096/103 RSB-097 AIS FUNCTIONS (Data input from AIS is required) 1. Number of Targets Botation 21/26 rpm 45 rpm 1,000 targets max. 2. Past Position Plot Intervals Gear Box RSB-098/099/104/10 RSB-100/101/102 **RF Transceiver** OFF, 30 s, 1-60 minutes 1. Frequency 9410 MHz 30 MHz X-band: POWER SUPPLY (specify when ordering) S-band: 3050 MHz 30 MHz 1. Processor Unit FAR-2817 2. Output power FAB-0817 FAB-0827 FAB-0827W FAB-08375W Output Power 12 kW 25 kW 25 kW 30 kW 30 kW Transceiver RTR-078 RTR-081 RTR-080 RTR-082 100-115 VAC: 2.6 A (3.0 A for HSC), 220-230 VAC: 1.6 A (1.7 A for HSC), 1 ø, 50/60 Hz FAB-2827 100-115 VAC: 3.0 A (3.4 A for HSC), 220-230 VAC: 1.8 A (1.9 A 3, Pulselength/PRR for HSC), 1 s, 50/60 Hz Range scale (nm) Pulselength (s) 0.125, 0.25 0.07 0.5 0.07, 0.15 PRR (Hz) FAR-2827W 3000 100-115 VAC: 3.2 A, 220-230 VAC: 1.6 A, 1 ø, 50/60 Hz 3000 FAR 2837S/2837SW 0.75, 1.5 0.07, 0.15, 0.3 3000, 1500 100 115 VAC: 3.0 A, 220 230 VAC: 1.5 A, 1 a, 50/60 Hz 0 15, 0 3, 0 5, 0 7 0 3, 0 5, 0 7, 1 2 3000, 1500, 1000 3 2. Display Unit 1500, 1000, 600 100-230 VAC, 0.9 A, 1 s, 50/60 Hz 12 24 0.5.0.7.1.2 1000, 600 440 VAC, 1 s, 50/60 Hz with optional transformer RU 1803 48,96 600 3. Antenna Unit FAR-2837S/2837SW: 4. LF. 5. Bandwidth 60 MHz, Logarithmic Short pulse: 40 MHz 200 VAC, 3.0 A, 3a, 50 Hz; 220 VAC, 3.0 A (3.5 A for HSC*), 3a, 60 Hz; 380 VAC, 1.5 A, 3a, 50 Hz; 440 VAC, 1.5 A (1,7 A for Middle pulse: 10 MHz Long pulse: 3 MHz HSC*), 3s, 60 Hz for FAR-2837S only RADAR DISPLAY 110 VAC, 3ø, 60 Hz with RU-5693; 220 VAC, 3ø, 50 Hz with RU-6522; 440 VAC, 3ø, 50 Hz with RU-5466-1 Display 23.1° color LCD (UXGA 1600 x 1280 pixels), 470 (H) x 353 (V) mm. Effective display diameter: 341 mm EQUIPMENT LIST Echo Color: Yellow, green or white in 32 levels 2. Range scales and ring intervals (nm) Standar 1 Display Unit MU-231 Range 125, 25, 5, 75, 15, 3, 6, 12, 24, 48, 96 2. Processor Unit RPU-013 Ring 025, 05, 1, 25, 25, 5, 1, 2, 4, 8, 16 Full-keyboard Control Unit RCU-014 Trackball Control Unit RCU-015 3. Minimum range (Specify when ordering) 4. Antenna Unit with cable (15/30/40/50 m) 22 m 4, Range discrimination Power Supply Unit PSU-007 for FAR-2837S Standard Spare Parts and Installation Materials 26 m 5, Range ring accuracy Option Within ±1 % of the current range scale or 10 m, whichever the 1. Performance Monitor PM-31 for X-band, creater PM-51 for S-band 6. Presentation modes Remote Control Unit RCU-016 Head-Up, Cursor-gyro, North-Up, Course-Up, True Motion (sea or ground stabilization) 3. Gyro Interface GC-10 4 DVI-Analog RGB Conversion Kit OP03-180 (SXGA output) 7. Heading information GPS compass SC-50/110 is a recommended heading sensor as 5. RGB Connector DSUB-BNC-1 (for VDR) Memory Card Interface Unit CU-200 6. a backup for a gyrocompass. Confirm if your Administrations 7 Transformer RU-1803/5466-1/5693/6522 permit its use. Rectifier RU-3424/1746B 8. Parallel index lines 8, 9. Junction Box RJB-001 1, 2, 3 or 6 lines (menu selectable) 10, Antenna Cable RW-9600 9 Radar map 11. Hand Grip FP03-09840 20.000 points to create coastines, own ship safety contour, isolated underwater dangers, buoys, traffic routing systems, 12. Bracket EP03-09820 13. Switching Hub HUB-100 prohibited areas and fairways as required by MO.

APPENDIX C. SPECIFICATION SHEET OF QR026 EW RECEIVER BY TELEYDNE DEFENSE



A Teledyne Technologies Company



0.5 to 18GHz High Performance Receiver Technical Data 0R026

Features

Full Band Coverage (options to >40GHz) Fast Tune Speed High Spurious Free Dynamic Range Low Colouration Simultaneous ELINT & ESM operation Frequency Selective RF Protection Compact, Expandable Architecture Low power consumption Optional 1GHz IF

Applications

Electronic Intelligence (ELINT) Electronic Support Measures (ESM) Defensive Aids Suites (ECM Set-on) Combined ELINT / ESM Sensors Ground, Airborne & Naval Environments

Product Description

The QRO28 high performance receiver provides a new approach for ESM / ELINT applications incorporating patenced features offering wide band synthesised capability from 500MHz to 18 GHz, with frequency extensions available up to 40 GHz. The QRO26 builds on the concept developed for the QRD10 receiver family and uses standard, interchangeable modules for maximum system flexibility.

The QRO26 offers high pre-selection selectivity, but without the drawbacks of slow tune speeds and high DC power requirements associated with conventional YIG based receivers. Tune speeds of <125µsec are achievable over the full tuning range. Amplitude and group delay characteristics are excellent over the full 500MHz IF bandwidth, with even higher performance over the middle 80% (400MHz) bandwidth. Performance over the central 100MHz is exceptional blecter than 1ns group delay, offering the lowest colouration for high quality ELINT signal collection. User control is via front panel Ethernet 10 / 100 Base T.

The temperature compensated unit exhibits low noise figure, high IP3 performance and consumes <80 Watts in single channel configuration.

The modular architecture of the receiver allows it to be configured as a single, dual, triple or quad channel unit, with phase & amplitude matched performance across all channels. The receiver provides IF outputs at 960 and 160MHz, with user selectable IF bandwidths available. A 1GHz IF output is available as an option. The receiver can be supplied as an integrated rack or as cards for integration in a customer system.



0.5 to 18GHz High Performance Receiver Technical Data QR026

Electrical Specification

Parameter	Specification
•RF Input Frequency Range:	0.5-19GHz
Optional extensions to > 40GHz	
*Maximum Input CW	+15dBm
*LO Re-radiacion	-90dBm Max
*Gain 20dB nom. (960MHz IF)	
28d8 nom. (160MHz IF)	
*Gain Variation (SOOMHz IF BW)	+/-3dB max for any cune frequency and any RF
signal in the IF bandwidth	
*Gain Variation (any 100MHz of IF BW)	+/-1.25dB max
*Group Delay Variation (SOOMHz IF BW)	Bns over the central 80% BW; 10ns over the full
bandwich; 3ns typ, <1ns in central 100N	Hz
*Input Accenuator	20dB (single scep)
*IF Accenuacon	15dB Min, 1dB scep size
*VSWR (RF In/IF out.)	2.5:1
*Noise Figure	16dB max (Typically 13.5dB)
*Input P1dB	-12dBm
*Input IP3	-2dBm (Typically OdBm)
*Single Tone SFDR	60dB min (500MHz BW)
In-band 2 Tone SFDR	50dB min (500MHz BW)
 Incernally Generated Spurious 	-70dBm max equiv. Input power
*Tune Speed	125µs max – Input cune command to Valid IF Output
*Tune Accuracy	2kHz
•Tune Resolution	1MHz (1kHz capable)
*Phase Noise	
1kHz	-88dBc/Hz
10kHz	-90dBc/Hz
100kHz	-95dBc/Hz
1MHz	-95dBc/Hz
10MHz	-115dBc/Hz
*Size	19 In rack x 2U high (single channel)
 Total Power Consumption: 	<80W in full rack configuration
 Input Power Supply 	94 - 253 VAC, 47 - 63 Hz
•Welghc:	< 12kg
 Operating Temperature Range 	-10°C to +60°C
*Options	 Frequency Excension to beyond 40GHz
	 Single, Dual or Quad Tracked Channels
	 Digital IFM(s) with hi-POI PDW outputs
	 Alternative Speed / Accuracy Oscillators
	 Alternative IF Outputs (3GHz, 1GHz, 70MHz)
	 Alternative rack configurations or individual cards in
	cransic frame for incergration in customer rack
	Alternative user control interfaces

APPENDIX D. SPECIFICATION SHEET FOR DF-A0062 DF RECEIVER BY POYNTING DEFENSE



VERSION: 2.0

Direction Finding and Monitoring Antenna

20 – 6000 MHz

Product Code: DF-A0062

SPECIFICATIONS:

Product codes: DF-A0062

	_

PRODUCT DESCRIPTION:

This direction finding antenna covers a frequency range of 20 MHz to 6 GHz. A separate, but fully integrated active monitoring antenna gives high sensitivity omnidirectional coverage on the same axis as the DF antenna, without interfering with the DF array.

The full-size elements on all bands give excellent DF sensitivity. Ultimate angular resolution for strong signals is well under 1° for most of the frequency range. Dipole elements provide good cross-polarisation rejection, and fair performance for signals arriving from up to 15° above or below the horizon.

The integrated monitoring antenna is in two bands, mounted above the Band C & D DF antenna module housed in the radome. Each band of the monitoring array is amplified at the top of the cable, and passively combined to give continuous coverage over the frequency range 20 MHz to 6 GHz, with a single output connector.

This DF antenna is designed to be usable with either a 5- or 2-channel phase-sensitive receiver and correlative algorithm. Characterisation of the antenna can be performed on request.

Related products: DF-A0064, DF-A0057-03, DF-A0038 and DF-A0098

Electrical - I	DF:							
DF frequency	y range	Band A: 20 – 300 MHz; Band B: 300 – 1000 MHz; Band C: 1000 – 3000 MHz; Band D: 3000 – 6000 MHz						
Nominal Inpu	t Impedance	50 Q						
Antenna type	2	5-element DF Interferometer (From 5 to 2-channel receiver compatible)						
Polarisation		Vertical						
Output cable	8	RG 400 cables						
DF connecto	rs	20 x TNC male						
Monitoring co	onnector	1 x N male						
Electrical - r	nonitoring:							
Frequency ra	ange	20 – 6000 MHz						
Nominal H-p	ane beamwidth	360*						
Nominal E-pl	ane beamwidth	60*						
Typical VSW	R'	2:1						
Polarisation		Vertical						
Connector ty	pe	N-type male						
Nominal Imp	edance	50 Ω						
Input voltage	(via coax)"	13 - 24 V DC						
Input current		< 150 mA						
Power consu	mption (nom)*	< 2.25 W						
OP1dB (typ.)		> 11 dBm						
OIP2 (typ.)*		> 31 dBm						
OIP3 (typ.)*		> 21 dBm						
Sensitivity	20 – 100 MHz	-20 dBµV/m						
(typ.) (S/N =	100 – 1000 MHz	-30 dBµV/m						
0 dB, BW -	1 – 3 GHz	-35 dBµV/m						
1 Hz)*	3 – 6 GHz	-40 dBµV/m						
Mechanical:								
Cross-section	nal wind area	1.50 m ² (including antenna switch)						
Maximum wit	nd speed	160 km/h (without ice load)						
Assembled h	eight	3.60 m						
Assembled d	lameter (max)	2.70 m						
Shipping dim	ensions	2.20 m x 0.90 m x 0.60 m						
weight of an	ienna	60 Kg						
including sh	ipping container	135 Kg						

DF with monitoring up to 6 GHz

'in active mode

ELECTRICAL FEATURES:

- Full-size DF
- Wideband DF
- 5-element interferometer
- Optimised for 2-channel receivers
- High sensitivity omni antenna integrated
- Tactical rapid deploy/stow design with integrated dust caps

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APPENDIX E. SAMPLE OF BISTATIC RCS RESULTS OF MODEL OF RSN'S FORMIDABLE-CLASS FRIGATE

Reflected]	Incidence	Angle in ^o)			
Angle in $^{\circ}$	1	2	3	4	5	6	7	8	9	10
1	31.53	31.53	21.74	21.74	14.80	14.80	15.80	15.80	14.62	14.62
2	28.31	28.31	23.84	23.84	14.42	14.42	12.99	12.99	16.08	16.08
3	21.23	21.23	18.88	18.88	13.18	13.18	15.45	15.45	5.49	5.49
4	14.15	14.15	14.28	14.28	18.14	18.14	8.34	8.34	12.75	12.75
5	19.32	19.32	13.93	13.93	13.90	13.90	17.18	17.18	15.36	15.36
6	12.99	12.99	13.15	13.15	8.37	8.37	12.13	12.13	13.46	13.46
7	13.99	13.99	14.61	14.61	16.79	16.79	9.87	9.87	14.51	14.51
8	1.22	1.22	16.13	16.13	12.77	12.77	13.42	13.42	15.77	15.77
9	11.07	11.07	16.83	16.83	13.94	13.94	14.12	14.12	17.87	17.87
10	15.99	15.99	12.30	12.30	16.34	16.34	13.45	13.45	15.53	15.53
11	17.23	17.23	7.59	7.59	13.60	13.60	15.86	15.86	14.21	14.21
12	12.32	12.32	17.34	17.34	16.82	16.82	16.59	16.59	16.57	16.57
13	10.85	10.85	16.89	16.89	18.07	18.07	13.87	13.87	16.67	16.67
14	16.77	16.77	16.29	16.29	13.15	13.15	17.67	17.67	14.66	14.66
15	11.04	11.04	15.80	15.80	15.05	15.05	13.57	13.57	17.75	17.75
16	14.73	14.73	17.79	17.79	17.10	17.10	13.81	13.81	17.36	17.36
17	15.81	15.81	13.67	13.67	13.97	13.97	19.43	19.43	15.44	15.44
18	16.89	16.89	14.36	14.36	11.99	11.99	17.78	17.78	14.46	14.46
19	15.63	15.63	12.89	12.89	16.94	16.94	15.27	15.27	14.25	14.25
20	14.52	14.52	17.17	17.17	16.02	16.02	14.63	14.63	13.51	13.51
21	15.27	15.27	18.76	18.76	13.62	13.62	15.95	15.95	15.05	15.05
22	16.27	16.27	18.62	18.62	16.30	16.30	15.81	15.81	16.12	16.12
23	18.98	18.98	15.81	15.81	16.94	16.94	14.33	14.33	16.22	16.22
24	18.12	18.12	14.14	14.14	15.03	15.03	16.72	16.72	16.61	16.61
25	11.94	11.94	16.35	16.35	16.15	16.15	16.83	16.83	17.41	17.41
26	13.82	13.82	13.11	13.11	15.65	15.65	16.98	16.98	14.61	14.61
27	12.48	12.48	17.23	17.23	16.91	16.91	15.62	15.62	15.69	15.69
28	14.67	14.67	17.02	17.02	17.39	17.39	15.53	15.53	16.67	16.67
29	15.20	15.20	14.33	14.33	15.00	15.00	16.40	16.40	14.78	14.78
30	14.46	14.46	13.59	13.59	14.24	14.24	16.29	16.29	13.05	13.05
31	14.06	14.06	14.35	14.35	14.48	14.48	15.07	15.07	12.30	12.30
32	14.78	14.78	15.32	15.32	12.75	12.75	14.47	14.47	12.93	12.93
33	15.81	15.81	13.42	13.42	12.58	12.58	15.02	15.02	13.34	13.34
34	14.25	14.25	13.67	13.67	15.18	15.18	14.85	14.85	14.72	14.72
35	11.97	11.97	16.41	16.41	15.84	15.84	13.54	13.54	15.30	15.30
36	12.99	12.99	14.34	14.34	12.07	12.07	12.87	12.87	13.00	13.00
37	15.63	15.63	13.65	13.65	13.51	13.51	15.26	15.26	14.73	14.73
38	10.62	10.62	13.39	13.39	14.44	14.44	11.10	11.10	13.19	13.19
39	14.63	14.63	13.80	13.80	11.52	11.52	14.83	14.83	13.27	13.27
40	31.53	31.53	13.27	13.27	12.87	12.87	12.94	12.94	12.26	12.26

The following table has a sample of the bistatic RCS values in dBsm for the frequency of 3050 MHz and the model shown in Figures 6 and 7 as generated by FEKO.

Reflected					Incidence	Angle in	0			
Angle in $^{\circ}$	1	2	3	4	5	6	7	8	9	10
41	12.48	12.48	13.67	13.67	14.13	14.13	12.54	12.54	13.98	13.98
42	13.82	13.82	11.47	11.47	12.01	12.01	12.92	12.92	11.17	11.17
43	12.42	12.42	11.36	11.36	11.82	11.82	10.54	10.54	9.76	9.76
44	12.05	12.05	11.97	11.97	10.94	10.94	9.80	9.80	12.15	12.15
45	10.62	10.62	10.74	10.74	10.97	10.97	9.83	9.83	12.38	12.38
46	8.32	8.32	10.23	10.23	9.42	9.42	8.00	8.00	12.18	12.18
47	8.83	8.83	9.27	9.27	10.35	10.35	9.87	9.87	13.12	13.12
48	6.56	6.56	9.50	9.50	9.87	9.87	9.81	9.81	12.45	12.45
49	9.03	9.03	8.98	8.98	10.05	10.05	9.70	9.70	12.84	12.84
50	5.50	5.50	7.42	7.42	10.01	10.01	10.28	10.28	8.79	8.79
51	11.05	11.05	8.01	8.01	6.30	6.30	12.03	12.03	10.82	10.82
52	11.11	11.11	7.88	7.88	10.53	10.53	12.83	12.83	11.14	11.14
53	9.15	9.15	10.77	10.77	11.34	11.34	9.10	9.10	8.88	8.88
54	10.08	10.08	4.64	4.64	9.60	9.60	9.29	9.29	6.05	6.05
55	11.36	11.36	8.85	8.85	11.01	11.01	7.35	7.35	8.42	8.42
56	8.92	8.92	9.03	9.03	8.91	8.91	5.63	5.63	2.61	2.61
57	12.28	12.28	8.08	8.08	5.95	5.95	9.55	9.55	-0.73	-0.73
58	7.43	7.43	6.07	6.07	5.29	5.29	1.92	1.92	0.25	0.25
59	9.18	9.18	2.70	2.70	8.86	8.86	3.55	3.55	4.72	4.72
60	9.73	9.73	6.47	6.47	3.76	3.76	3.31	3.31	4.69	4.69
61	7.93	7.93	4.70	4.70	4.22	4.22	1.20	1.20	6.80	6.80
62	10.79	10.79	-2.42	-2.42	0.33	0.33	5.65	5.65	8.59	8.59
63	-4.50	-4.50	1.43	1.43	3.83	3.83	4.60	4.60	8.44	8.44
64	5.53	5.53	-0.31	-0.31	1.01	1.01	6.21	6.21	10.88	10.88
65	8.79	8.79	-5.83	-5.83	-1.06	-1.06	9.59	9.59	11.51	11.51
66	-2.80	-2.80	-9.00	-9.00	3.12	3.12	9.93	9.93	9.51	9.51
67	-2.95	-2.95	1.10	1.10	4.20	4.20	3.95	3.95	2.61	2.61
68	4.57	4.57	4.19	4.19	5.63	5.63	-2.17	-2.17	7.21	7.21
69	3.70	3.70	-0.07	-0.07	3.57	3.57	-1.55	-1.55	9.12	9.12
70	5.05	5.05	2.70	2.70	1.94	1.94	7.00	7.00	7.37	7.37
71	2.02	2.02	4.56	4.56	3.37	3.37	7.15	7.15	7.13	7.13
72	6.98	6.98	8.71	8.71	7.03	7.03	4.61	4.61	5.78	5.78
73	7.77	7.77	11.86	11.86	11.88	11.88	-1.92	-1.92	6.40	6.40
74	9.04	9.04	13.71	13.71	13.84	13.84	11.81	11.81	13.24	13.24
75	6.22	6.22	13.94	13.94	10.40	10.40	14.65	14.65	7.29	7.29
76	10.65	10.65	13.19	13.19	11.79	11.79	8.42	8.42	17.98	17.98
77	14.29	14.29	9.69	9.69	16.86	16.86	17.66	17.66	25.03	25.03
78	12.52	12.52	16.92	16.92	11.15	11.15	15.77	15.77	25.23	25.23
79	11.89	11.89	14.69	14.69	18.01	18.01	23.58	23.58	26.26	26.26
80	13.07	13.07	15.53	15.53	15.69	15.69	26.55	26.55	27.71	27.71
81	14.91	14.91	12.82	12.82	23.97	23.97	27.51	27.51	27.83	27.83
82	12.79	12.79	21.71	21.71	26.64	26.64	27.94	27.94	26.41	26.41
83	13.91	13.91	23.55	23.55	25.76	25.76	27.35	27.35	27.19	27.19
84	13.82	13.82	23.12	23.12	27.96	27.96	25.91	25.91	23.07	23.07
85	14.90	14.90	27.52	27.52	26.76	26.76	27.81	27.81	22.61	22.61
86	17.52	17.52	23.78	23.78	30.06	30.06	23.22	23.22	9.02	9.02
87	8.39	8.39	30.03	30.03	24.66	24.66	25.11	25.11	15.40	15.40
88	15.74	15.74	24.02	24.02	24.29	24.29	8.33	8.33	11.54	11.54
89	14.62	14.62	27.59	27.59	21.46	21.46	17.36	17.36	9.80	9.80
90	19.12	19.12	19.23	19.23	22.59	22.59	15.77	15.77	14.32	14.32

Reflected					Incidence	Angle in	0			
Angle in $^{\circ}$	1	2	3	4	5	6	7	8	9	10
91	16.62	16.62	25.39	25.39	13.42	13.42	10.42	10.42	8.56	8.56
92	18.30	18.30	9.82	9.82	17.38	17.38	15.04	15.04	15.41	15.41
93	15.41	15.41	20.56	20.56	15.46	15.46	11.52	11.52	11.15	11.15
94	11.73	11.73	16.71	16.71	9.27	9.27	15.02	15.02	10.33	10.33
95	16.52	16.52	13.76	13.76	17.46	17.46	9.30	9.30	10.94	10.94
96	14.32	14.32	14.59	14.59	5.16	5.16	5.55	5.55	5.32	5.32
97	15.66	15.66	7.23	7.23	14.88	14.88	12.43	12.43	9.45	9.45
98	13.95	13.95	14.52	14.52	7.30	7.30	3.56	3.56	5.88	5.88
99	12.01	12.01	11.58	11.58	10.66	10.66	11.37	11.37	8.88	8.88
100	10.62	10.62	13.40	13.40	12.05	12.05	5.37	5.37	8.61	8.61
101	14.92	14.92	13.40	13.40	6.51	6.51	11.84	11.84	6.93	6.93
102	14.96	14.96	14.10	14.10	15.28	15.28	11.12	11.12	11.23	11.23
103	12.17	12.17	12.73	12.73	12.21	12.21	8.70	8.70	7.53	7.53
104	12.97	12.97	13.29	13.29	12.69	12.69	9.81	9.81	10.52	10.52
105	13.42	13.42	16.79	16.79	15.06	15.06	11.21	11.21	12.71	12.71
106	13.95	13.95	17.22	17.22	14.69	14.69	10.98	10.98	12.60	12.60
107	15.82	15.82	16.20	16.20	15.79	15.79	12.48	12.48	1.45	1.45
108	15.09	15.09	15.21	15.21	13.74	13.74	13.41	13.41	1.69	1.69
109	14.15	14.15	12.99	12.99	17.34	17.34	15.05	15.05	6.45	6.45
110	15.33	15.33	15.45	15.45	16.72	16.72	13.82	13.82	3.92	3.92
111	13.86	13.86	11.15	11.15	16.90	16.90	17.18	17.18	9.99	9.99
112	15.05	15.05	9.16	9.16	15.70	15.70	17.03	17.03	10.42	10.42
113	10.32	10.32	15.59	15.59	15.63	15.63	19.51	19.51	12.11	12.11
114	14.42	14.42	11.99	11.99	13.17	13.17	17.19	17.19	12.73	12.73
115	6.27	6.27	8.41	8.41	18.87	18.87	12.32	12.32	14.04	14.04
116	9.23	9.23	16.92	16.92	22.61	22.61	18.42	18.42	15.87	15.87
117	8.77	8.77	14.12	14.12	17.76	17.76	19.40	19.40	14.24	14.24
118	2.77	2.77	18.31	18.31	16.56	16.56	12.94	12.94	12.21	12.21
119	12.65	12.65	21.25	21.25	16.73	16.73	17.56	17.56	12.78	12.78
120	11.05	11.05	14.23	14.23	13.43	13.43	4.92	4.92	18.77	18.77
121	16.08	16.08	12.15	12.15	7.02	7.02	13.07	13.07	15.50	15.50
122	6.78	6.78	11.65	11.65	16.07	16.07	14.29	14.29	9.83	9.83
123	12.52	12.52	11.31	11.31	18.08	18.08	8.83	8.83	13.96	13.96
124	6.04	6.04	5.84	5.84	14.04	14.04	11.82	11.82	14.17	14.17
125	12.44	12.44	12.77	12.77	-0.15	-0.15	10.66	10.66	18.89	18.89
126	10.14	10.14	12.44	12.44	6.81	6.81	18.86	18.86	15.68	15.68
127	12.01	12.01	8.19	8.19	15.67	15.67	18.58	18.58	6.97	6.97
128	11.83	11.83	4.41	4.41	18.31	18.31	10.31	10.31	2.33	2.33
129	12.46	12.46	14.00	14.00	20.59	20.59	2.99	2.99	9.92	9.92
130	7.08	7.08	15.85	15.85	14.25	14.25	8.25	8.25	8.40	8.40
131	18.90	18.90	21.26	21.26	3.25	3.25	5.48	5.48	4.78	4.78
132	12.28	12.28	9.29	9.29	-6.25	-6.25	9.75	9.75	9.14	9.14
133	20.86	20.86	15.74	15.74	8.93	8.93	9.08	9.08	24.62	24.62
134	20.86	20.86	10.76	10.76	13.29	13.29	3.87	3.87	10.17	10.17
135	16.61	16.61	11.66	11.66	14.95	14.95	23.81	23.81	20.53	20.53
136	15.03	15.03	13.87	13.87	1.56	1.56	18.97	18.97	18.77	18.77
137	15.72	15.72	20.99	20.99	19.68	19.68	17.07	17.07	15.61	15.61
138	17.95	17.95	16.87	16.87	14.76	14.76	15.03	15.03	16.33	16.33
139	18.31	18.31	25.54	25.54	18.51	18.51	17.20	17.20	11.68	11.68
140	17.37	17.37	20.89	20.89	15.90	15.90	16.10	16.10	4.16	4.16

Reflected					Incidence	Angle in	0			
Angle in °	1	2	3	4	5	6	7	8	9	10
141	25.17	25.17	19.12	19.12	13.23	13.23	14.39	14.39	13.48	13.48
142	19.37	19.37	15.16	15.16	18.71	18.71	17.20	17.20	15.28	15.28
143	19.92	19.92	20.13	20.13	16.87	16.87	15.18	15.18	8.47	8.47
144	9.21	9.21	19.72	19.72	0.53	0.53	17.46	17.46	15.95	15.95
145	16.35	16.35	17.15	17.15	15.94	15.94	10.64	10.64	16.38	16.38
146	14.20	14.20	10.74	10.74	12.02	12.02	17.08	17.08	2.94	2.94
147	16.75	16.75	4.29	4.29	16.85	16.85	13.11	13.11	20.67	20.67
148	15.82	15.82	14.01	14.01	13.23	13.23	11.49	11.49	15.14	15.14
149	18.16	18.16	14.15	14.15	12.81	12.81	7.26	7.26	9.03	9.03
150	11.65	11.65	17.56	17.56	12.06	12.06	13.12	13.12	13.29	13.29
151	18.53	18.53	20.68	20.68	12.32	12.32	13.00	13.00	15.54	15.54
152	18.15	18.15	15.65	15.65	16.95	16.95	14.03	14.03	7.85	7.85
153	18.71	18.71	9.39	9.39	13.31	13.31	16.05	16.05	16.40	16.40
154	15.74	15.74	12.92	12.92	16.24	16.24	13.13	13.13	12.76	12.76
155	19.69	19.69	9.29	9.29	15.34	15.34	17.32	17.32	15.99	15.99
156	5.95	5.95	18.21	18.21	20.14	20.14	13.00	13.00	13.18	13.18
157	16.99	16.99	16.96	16.96	21.29	21.29	10.72	10.72	18.52	18.52
158	15.73	15.73	16.43	16.43	18.04	18.04	20.36	20.36	22.92	22.92
159	18.17	18.17	18.27	18.27	15.87	15.87	23.21	23.21	19.38	19.38
160	14.31	14.31	22.57	22.57	21.28	21.28	22.04	22.04	25.05	25.05
161	9.92	9.92	16.67	16.67	16.61	16.61	23.11	23.11	15.39	15.39
162	20.86	20.86	17.48	17.48	17.35	17.35	16.15	16.15	28.07	28.07
163	18.82	18.82	22.26	22.26	10.20	10.20	16.73	16.73	29.82	29.82
164	16.47	16.47	21.29	21.29	20.95	20.95	11.73	11.73	25.46	25.46
165	9.39	9.39	7.17	7.17	23.36	23.36	10.15	10.15	28.05	28.05
166	22.88	22.88	23.93	23.93	22.85	22.85	18.72	18.72	31.73	31.73
167	22.27	22.27	14.01	14.01	20.75	20.75	25.96	25.96	31.84	31.84
168	16.65	16.65	23.70	23.70	16.55	16.55	11.50	11.50	35.03	35.03
169	29.95	29.95	20.54	20.54	14.62	14.62	26.24	26.24	37.27	37.27
170	30.67	30.67	22.16	22.16	22.02	22.02	26.95	26.95	34.22	34.22
171	24.17	24.17	23.35	23.35	19.97	19.97	27.68	27.68	36.87	36.87
172	26.07	26.07	24.93	24.93	29.18	29.18	34.31	34.31	49.15	49.15
173	32.45	32.45	27.50	27.50	35.31	35.31	34.74	34.74	39.48	39.48
174	34.28	34.28	26.45	26.45	35.58	35.58	48.46	48.46	36.34	36.34
175	28.64	28.64	27.86	27.86	39.40	39.40	36.93	36.93	39.02	39.02
176	38.80	38.80	37.34	37.34	44.49	44.49	33.15	33.15	30.89	30.89
177	37.77	37.77	39.45	39.45	43.99	43.99	24.56	24.56	29.71	29.71
178	45.24	45.24	47.41	47.41	41.08	41.08	33.65	33.65	33.35	33.35
179	61.50	61.50	50.60	50.60	43.77	43.77	31.57	31.57	34.65	34.65
180	78.86	78.86	49.50	49.50	43.67	43.67	37.55	37.55	27.72	27.72
181	61.64	61.64	61.46	61.46	47.32	47.32	41.96	41.96	36.18	36.18
182	46.01	46.01	79.67	79.67	48.33	48.33	41.59	41.59	31.70	31.70
183	37.48	37.48	57.93	57.93	54.60	54.60	32.26	32.26	39.70	39.70
184	37.65	37.65	48.32	48.32	81.19	81.19	50.41	50.41	44.34	44.34
185	34.87	34.87	40.98	40.98	49.67	49.67	49.87	49.87	44.81	44.81
186	35.43	35.43	42.10	42.10	50.11	50.11	82.18	82.18	51.02	51.02
18/	29.38	29.38	34.66	34.66	45.76	45.76	49.93	49.93	51.23	51.23
188	21.31	21.31	20.33	20.33	41.99	41.99	41.30	41.30	83.06 40.20	83.06 40.20
109	14.33	14.33	24.00	24.00	22.01	22.01	42.41	42.41	49.20	49.20
190	10.22	10.22	32.23	32.23	33.01	55.01	J4.8Z	J4.8Z	49.12	49.12

Reflected					Incidence	Angle in	0			
Angle in °	1	2	3	4	5	6	7	8	9	10
191	24.67	24.67	23.26	23.26	23.02	23.02	36.05	36.05	42.32	42.32
192	17.57	17.57	26.96	26.96	19.98	19.98	33.80	33.80	43.83	43.83
193	19.73	19.73	18.55	18.55	26.95	26.95	27.81	27.81	39.86	39.86
194	15.90	15.90	24.28	24.28	24.10	24.10	30.81	30.81	25.96	25.96
195	22.61	22.61	22.61	22.61	16.25	16.25	29.32	29.32	26.83	26.83
196	11.71	11.71	20.60	20.60	21.07	21.07	30.66	30.66	34.41	34.41
197	16.66	16.66	2.21	2.21	22.57	22.57	31.86	31.86	31.80	31.80
198	19.92	19.92	17.74	17.74	18.11	18.11	29.69	29.69	22.77	22.77
199	7.48	7.48	14.07	14.07	16.59	16.59	23.17	23.17	29.77	29.77
200	8.04	8.04	20.62	20.62	12.40	12.40	11.70	11.70	26.32	26.32
201	16.45	16.45	9.21	9.21	8.05	8.05	16.75	16.75	16.01	16.01
202	17.43	17.43	10.46	10.46	17.62	17.62	22.39	22.39	21.75	21.75
203	14.45	14.45	16.91	16.91	16.50	16.50	20.13	20.13	22.61	22.61
204	3.37	3.37	18.38	18.38	22.88	22.88	19.21	19.21	14.36	14.36
205	17.14	17.14	22.68	22.68	24.40	24.40	15.20	15.20	8.64	8.64
206	18.44	18.44	21.96	21.96	18.99	18.99	19.07	19.07	17.71	17.71
207	17.47	17.47	11.83	11.83	10.29	10.29	22.84	22.84	18.80	18.80
208	19.00	19.00	21.17	21.17	17.39	17.39	14.10	14.10	23.08	23.08
209	15.51	15.51	8.20	8.20	10.79	10.79	18.46	18.46	18.48	18.48
210	13.66	13.66	20.60	20.60	22.07	22.07	23.80	23.80	27.02	27.02
211	19.13	19.13	8.35	8.35	16.75	16.75	20.60	20.60	18.58	18.58
212	16.43	16.43	12.76	12.76	15.63	15.63	27.42	27.42	24.24	24.24
213	16.80	16.80	16.04	16.04	15.93	15.93	22.22	22.22	17.35	17.35
214	13.54	13.54	18.88	18.88	18.29	18.29	14.52	14.52	14.30	14.30
215	15.00	15.00	17.92	17.92	24.64	24.64	2.38	2.38	13.60	13.60
216	12.31	12.31	22.16	22.16	15.44	15.44	8.73	8.73	13.77	13.77
217	17.63	17.63	19.36	19.36	13.22	13.22	4.40	4.40	13.54	13.54
218	17.13	17.13	16.26	16.26	2.58	2.58	7.52	7.52	10.42	10.42
219	25.13	25.13	12.47	12.47	-2.71	-2.71	14.27	14.27	18.56	18.56
220	11.46	11.46	12.77	12.77	-0.74	-0.74	17.41	17.41	18.56	18.56
221	13.62	13.62	4.67	4.67	11.39	11.39	18.06	18.06	18.52	18.52
222	14.73	14.73	5.61	5.61	14.56	14.56	19.44	19.44	13.22	13.22
223	14.01	14.01	13.29	13.29	19.75	19.75	19.19	19.19	13.59	13.59
224	12.99	12.99	19.66	19.66	20.67	20.67	17.67	17.67	14.49	14.49
225	12.31	12.31	20.20	20.20	17.90	17.90	10.82	10.82	6.52	6.52
226	18.75	18.75	19.30	19.30	14.68	14.68	5.17	5.17	14.82	14.82
227	20.83	20.83	3.06	3.06	15.17	15.17	2.18	2.18	12.77	12.77
228	15.47	15.47	11.32	11.32	17.92	17.92	1.76	1.76	14.52	14.52
229	17.75	17.75	14.50	14.50	16.50	16.50	11.15	11.15	11.96	11.96
230	2.76	2.76	12.22	12.22	10.17	10.17	11.95	11.95	12.22	12.22
231	12.11	12.11	8.65	8.65	-1.00	-1.00	7.61	7.61	15.03	15.03
232	10.97	10.97	13.05	13.05	7.87	7.87	1.35	1.35	15.80	15.80
233	14.41	14.41	1.19	1.19	8.89	8.89	11.83	11.83	13.59	13.59
234	2.83	2.83	4.93	4.93	1.01	1.01	12.37	12.37	10.72	10.72
235	8.00	8.00	4.25	4.25	4.50	4.50	5.05	5.05	14.45	14.45
236	3.85	3.85	8.27	8.27	6.18	6.18	7.26	7.26	10.84	10.84
237	3.61	3.61	-3.69	-3.69	-2.64	-2.64	-2.80	-2.80	8.14	8.14
238	1.33	1.33	4.27	4.27	8.60	8.60	3.87	3.87	8.86	8.86
239	-9.57	-9.57	-0.57	-0.57	-0.70	-0.70	8.61	8.61	5.38	5.38
240	0.19	0.19	9.39	9.39	6.38	6.38	7.85	7.85	4.06	4.06

Reflected					Incidence	Angle in	0			
Angle in °	1	2	3	4	5	6	7	8	9	10
241	2.96	2.96	4.61	4.61	-5.00	-5.00	6.40	6.40	-4.28	-4.28
242	-8.96	-8.96	8.27	8.27	5.12	5.12	11.02	11.02	-1.57	-1.57
243	3.56	3.56	6.51	6.51	6.81	6.81	8.94	8.94	6.10	6.10
244	7.16	7.16	4.97	4.97	7.35	7.35	10.16	10.16	2.75	2.75
245	3.54	3.54	9.32	9.32	8.24	8.24	11.69	11.69	8.49	8.49
246	8.96	8.96	8.77	8.77	9.89	9.89	9.92	9.92	7.71	7.71
247	12.29	12.29	3.82	3.82	12.23	12.23	8.38	8.38	13.11	13.11
248	12.29	12.29	6.66	6.66	13.03	13.03	3.44	3.44	3.69	3.69
249	11.86	11.86	9.73	9.73	11.40	11.40	11.95	11.95	13.05	13.05
250	11.44	11.44	10.68	10.68	13.84	13.84	10.24	10.24	13.55	13.55
251	9.53	9.53	13.26	13.26	11.94	11.94	0.63	0.63	13.23	13.23
252	10.77	10.77	9.76	9.76	5.61	5.61	4.81	4.81	13.52	13.52
253	12.57	12.57	3.04	3.04	0.01	0.01	12.44	12.44	11.17	11.17
254	12.24	12.24	7.27	7.27	0.01	0.01	14.19	14.19	8.13	8.13
255	11.28	11.28	11.12	11.12	6.35	6.35	10.74	10.74	11.85	11.85
256	9.95	9.95	10.86	10.86	11.12	11.12	11.17	11.17	9.41	9.41
257	8.75	8.75	9.12	9.12	13.11	13.11	-1.54	-1.54	13.79	13.79
258	11.58	11.58	12.08	12.08	11.88	11.88	9.50	9.50	11.91	11.91
259	11.37	11.37	8.47	8.47	10.04	10.04	9.99	9.99	19.07	19.07
260	5.68	5.68	11.20	11.20	6.83	6.83	10.66	10.66	18.75	18.75
261	10.29	10.29	12.08	12.08	5.38	5.38	12.67	12.67	13.82	13.82
262	14.52	14.52	10.31	10.31	7.33	7.33	14.87	14.87	8.35	8.35
263	9.93	9.93	12.57	12.57	8.39	8.39	15.47	15.47	20.23	20.23
264	13.86	13.86	8.91	8.91	14.77	14.77	15.69	15.69	17.03	17.03
265	15.81	15.81	15.20	15.20	16.57	16.57	19.34	19.34	16.31	16.31
266	9.23	9.23	16.34	16.34	14.99	14.99	15.98	15.98	15.59	15.59
267	16.71	16.71	10.81	10.81	16.32	16.32	12.48	12.48	13.48	13.48
268	18.03	18.03	6.84	6.84	14.29	14.29	9.46	9.46	12.34	12.34
269	21.80	21.80	9.97	9.97	13.20	13.20	7.81	7.81	12.78	12.78
270	19.58	19.58	13.79	13.79	11.30	11.30	11.61	11.61	14.50	14.50
271	16.63	16.63	12.93	12.93	7.47	7.47	7.77	7.77	15.45	15.45
272	16.64	16.64	9.37	9.37	9.23	9.23	14.63	14.63	12.82	12.82
273	9.34	9.34	10.89	10.89	7.38	7.38	13.91	13.91	11.58	11.58
274	19.72	19.72	2.42	2.42	15.11	15.11	11.94	11.94	10.95	10.95
275	14.22	14.22	8.66	8.66	13.63	13.63	10.44	10.44	10.95	10.95
276	14.68	14.68	13.08	13.08	10.97	10.97	12.84	12.84	12.36	12.36
277	12.37	12.37	14.22	14.22	9.57	9.57	12.26	12.26	10.68	10.68
278	13.01	13.01	10.64	10.64	7.30	7.30	13.37	13.37	8.52	8.52
279	13.78	13.78	10.23	10.23	11.62	11.62	6.83	6.83	12.23	12.23
280	14.87	14.87	6.94	6.94	11.51	11.51	5.48	5.48	9.36	9.36
281	12.52	12.52	13.53	13.53	13.09	13.09	7.14	7.14	10.95	10.95
282	15.11	15.11	11.54	11.54	8.54	8.54	9.93	9.93	5.50	5.50
283	15.01	15.01	12.27	12.27	7.25	7.25	4.99	4.99	11.24	11.24
284	12.69	12.69	11.46	11.46	9.88	9.88	7.29	7.29	10.45	10.45
285	12.04	12.04	9.92	9.92	6.50	6.50	9.64	9.64	10.68	10.68
286	12.95	12.95	8.86	8.86	8.28	8.28	7.82	7.82	12.98	12.98
287	11.67	11.67	8.52	8.52	8.52	8.52	11.03	11.03	10.50	10.50
288	10.55	10.55	8.28	8.28	8.40	8.40	10.74	10.74	12.14	12.14
289	8.64	8.64	9.51	9.51	7.42	7.42	11.05	11.05	6.96	6.96
290	9.96	9.96	8.87	8.87	9.22	9.22	9.10	9.10	6.27	6.27

Reflected					Incidence	Angle in	0			
Angle in °	1	2	3	4	5	6	7	8	9	10
291	8.08	8.08	10.18	10.18	11.69	11.69	8.81	8.81	6.23	6.23
292	9.47	9.47	10.47	10.47	10.41	10.41	11.00	11.00	8.35	8.35
293	8.51	8.51	11.43	11.43	9.25	9.25	8.42	8.42	8.85	8.85
294	9.18	9.18	8.90	8.90	10.61	10.61	4.49	4.49	5.06	5.06
295	12.32	12.32	8.85	8.85	8.55	8.55	6.65	6.65	0.56	0.56
296	8.52	8.52	11.78	11.78	5.17	5.17	5.64	5.64	1.62	1.62
297	5.55	5.55	10.23	10.23	8.19	8.19	3.00	3.00	5.50	5.50
298	11.90	11.90	7.46	7.46	4.35	4.35	4.08	4.08	-8.89	-8.89
299	7.79	7.79	10.47	10.47	6.92	6.92	6.78	6.78	8.63	8.63
300	8.57	8.57	4.41	4.41	3.42	3.42	4.05	4.05	-0.04	-0.04
301	7.91	7.91	8.78	8.78	7.63	7.63	7.66	7.66	10.54	10.54
302	2.38	2.38	5.85	5.85	7.28	7.28	9.81	9.81	7.09	7.09
303	7.77	7.77	2.38	2.38	6.90	6.90	11.37	11.37	9.10	9.10
304	2.71	2.71	5.41	5.41	7.97	7.97	6.92	6.92	7.06	7.06
305	4.58	4.58	6.62	6.62	9.33	9.33	11.00	11.00	11.62	11.62
306	3.84	3.84	7.78	7.78	11.16	11.16	6.97	6.97	11.13	11.13
307	3.41	3.41	7.08	7.08	8.27	8.27	12.84	12.84	12.98	12.98
308	6.57	6.57	10.16	10.16	11.24	11.24	12.66	12.66	11.77	11.77
309	8.73	8.73	8.99	8.99	11.13	11.13	14.50	14.50	13.75	13.75
310	1.50	1.50	11.96	11.96	14.41	14.41	12.70	12.70	15.07	15.07
311	9.35	9.35	10.15	10.15	11.87	11.87	14.82	14.82	14.94	14.94
312	4.39	4.39	11.87	11.87	14.20	14.20	15.19	15.19	18.15	18.15
313	9.07	9.07	13.44	13.44	12.91	12.91	15.97	15.97	12.02	12.02
314	6.52	6.52	15.09	15.09	15.07	15.07	11.27	11.27	16.30	16.30
315	10.51	10.51	13.50	13.50	8.85	8.85	17.31	17.31	12.86	12.86
316	12.71	12.71	12.99	12.99	16.83	16.83	14.16	14.16	14.09	14.09
317	13.43	13.43	15.21	15.21	17.61	17.61	12.15	12.15	14.97	14.97
318	13.79	13.79	14.59	14.59	14.50	14.50	15.23	15.23	16.49	16.49
319	13.06	13.06	14.55	14.55	15.40	15.40	14.73	14.73	16.36	16.36
320	11.07	11.07	10.16	10.16	12.97	12.97	13.65	13.65	14.61	14.61
321	11.06	11.06	13.40	13.40	14.52	14.52	16.75	16.75	15.26	15.26
322	9.14	9.14	14.25	14.25	15.83	15.83	15.09	15.09	14.85	14.85
323	15.06	15.06	13.41	13.41	14.76	14.76	12.63	12.63	15.99	15.99
324	11.54	11.54	12.98	12.98	14.67	14.67	14.95	14.95	15.41	15.41
325	12.49	12.49	13.09	13.09	12.37	12.37	14.62	14.62	14.81	14.81
326	13.56	13.56	16.38	16.38	15.29	15.29	11.18	11.18	17.04	17.04
327	14.74	14.74	15.45	15.45	15.58	15.58	16.13	16.13	15.64	15.64
328	14.17	14.17	11.87	11.87	14.51	14.51	14.00	14.00	17.37	17.37
329	11.48	11.48	13.29	13.29	13.16	13.16	18.12	18.12	16.84	16.84
330	11.77	11.77	14.88	14.88	16.60	16.60	13.19	13.19	15.59	15.59
331	13.22	13.22	16.18	16.18	16.45	16.45	15.34	15.34	17.36	17.36
332	15.03	15.03	11.67	11.67	14.88	14.88	14.30	14.30	19.35	19.35
333	10.79	10.79	12.57	12.57	9.40	9.40	18.89	18.89	11.99	11.99
334	10.66	10.66	18.71	18.71	17.38	17.38	17.11	17.11	14.10	14.10
335	17.69	17.69	21.58	21.58	16.66	16.66	17.24	17.24	13.33	13.33
336	20.71	20.71	17.27	17.27	19.21	19.21	10.41	10.41	9.35	9.35
337	24.64	24.64	17.59	17.59	11.52	11.52	8.86	8.86	17.63	17.63
338	20.11	20.11	19.68	19.68	14.41	14.41	14.94	14.94	20.71	20.71
339	18.73	18.73	14.60	14.60	15.13	15.13	15.33	15.33	18.52	18.52
340	13.35	13.35	12.40	12.40	17.67	17.67	13.11	13.11	19.48	19.48

Reflected		Incidence Angle in °								
Angle in °	1	2	3	4	5	6	7	8	9	10
341	13.75	13.75	14.04	14.04	16.69	16.69	10.31	10.31	13.62	13.62
342	16.64	16.64	14.48	14.48	13.24	13.24	13.56	13.56	14.19	14.19
343	16.88	16.88	18.26	18.26	18.09	18.09	17.68	17.68	24.58	24.58
344	16.87	16.87	11.88	11.88	20.61	20.61	13.88	13.88	19.38	19.38
345	8.19	8.19	14.83	14.83	19.87	19.87	25.56	25.56	16.03	16.03
346	17.94	17.94	24.24	24.24	13.24	13.24	21.75	21.75	16.67	16.67
347	17.19	17.19	7.26	7.26	24.64	24.64	5.61	5.61	19.10	19.10
348	24.78	24.78	20.85	20.85	17.46	17.46	14.20	14.20	22.48	22.48
349	18.67	18.67	25.25	25.25	17.22	17.22	17.89	17.89	24.02	24.02
350	20.67	20.67	9.57	9.57	16.23	16.23	15.73	15.73	29.56	29.56
351	24.58	24.58	10.26	10.26	17.52	17.52	24.12	24.12	37.39	37.39
352	11.74	11.74	9.75	9.75	16.80	16.80	33.99	33.99	64.51	64.51
353	15.73	15.73	21.41	21.41	25.53	25.53	36.90	36.90	38.93	38.93
354	15.63	15.63	18.43	18.43	36.39	36.39	64.58	64.58	31.57	31.57
355	18.67	18.67	26.36	26.36	38.28	38.28	31.76	31.76	22.49	22.49
356	14.94	14.94	33.68	33.68	64.43	64.43	28.69	28.69	11.89	11.89
357	26.43	26.43	37.85	37.85	35.23	35.23	22.89	22.89	19.38	19.38
358	28.28	28.28	64.52	64.52	31.68	31.68	15.21	15.21	6.34	6.34
359	37.85	37.85	35.69	35.69	23.48	23.48	14.92	14.92	11.42	11.42
360	64.43	64.43	28.41	28.41	13.76	13.76	12.60	12.60	4.39	4.39

Reflected					Incidence	Angle in	0			
Angle in $^{\circ}$	11	12	13	14	15	16	17	18	19	20
1	17.59	17.59	8.33	8.33	16.04	16.04	14.31	14.31	15.99	15.99
2	11.82	11.82	17.41	17.41	16.51	16.51	17.35	17.35	15.04	15.04
3	15.12	15.12	15.12	15.12	18.11	18.11	14.44	14.44	17.95	17.95
4	16.28	16.28	17.05	17.05	13.37	13.37	16.77	16.77	12.33	12.33
5	14.11	14.11	14.97	14.97	14.80	14.80	13.27	13.27	18.12	18.12
6	13.52	13.52	16.50	16.50	17.74	17.74	13.87	13.87	18.00	18.00
7	14.74	14.74	17.82	17.82	16.02	16.02	19.83	19.83	14.92	14.92
8	15.55	15.55	16.59	16.59	14.73	14.73	17.64	17.64	14.95	14.95
9	13.56	13.56	15.96	15.96	18.93	18.93	16.79	16.79	14.96	14.96
10	11.98	11.98	16.00	16.00	15.42	15.42	16.47	16.47	15.38	15.38
11	15.01	15.01	19.29	19.29	16.55	16.55	15.72	15.72	16.61	16.61
12	15.96	15.96	14.08	14.08	13.21	13.21	15.19	15.19	15.61	15.61
13	17.36	17.36	16.82	16.82	15.48	15.48	16.83	16.83	13.94	13.94
14	15.36	15.36	13.22	13.22	16.37	16.37	13.99	13.99	16.42	16.42
15	15.97	15.97	17.20	17.20	17.13	17.13	17.01	17.01	16.08	16.08
16	15.82	15.82	14.84	14.84	14.15	14.15	15.54	15.54	13.94	13.94
17	16.97	16.97	15.64	15.64	16.46	16.46	15.10	15.10	16.35	16.35
18	15.01	15.01	15.83	15.83	16.87	16.87	13.95	13.95	15.80	15.80
19	15.61	15.61	16.93	16.93	16.48	16.48	14.85	14.85	15.30	15.30
20	16.24	16.24	17.10	17.10	16.41	16.41	14.97	14.97	15.73	15.73
21	17.37	17.37	16.26	16.26	15.61	15.61	14.10	14.10	16.24	16.24
22	18.11	18.11	15.86	15.86	13.99	13.99	15.28	15.28	15.51	15.51
23	16.54	16.54	16.87	16.87	13.83	13.83	16.41	16.41	13.33	13.33
24	15.18	15.18	14.79	14.79	14.77	14.77	13.76	13.76	13.92	13.92
25	15.65	15.65	14.14	14.14	15.09	15.09	14.93	14.93	14.14	14.14
26	15.22	15.22	15.31	15.31	14.45	14.45	12.94	12.94	15.46	15.46
27	13.97	13.97	15.60	15.60	15.00	15.00	15.45	15.45	14.57	14.57
28	15.89	15.89	13.44	13.44	12.30	12.30	14.88	14.88	12.06	12.06
29	15.77	15.77	13.48	13.48	14.58	14.58	13.86	13.86	14.32	14.32
30	14.37	14.37	14.82	14.82	15.82	15.82	13.85	13.85	15.93	15.93
31	14.74	14.74	15.39	15.39	14.84	14.84	14.31	14.31	15.58	15.58
32	15.02	15.02	14.54	14.54	15.21	15.21	14.67	14.67	15.29	15.29
33	15.40	15.40	14.29	14.29	14.64	14.64	14.12	14.12	15.19	15.19
34	14.76	14.76	14.08	14.08	12.64	12.64	13.99	13.99	12.59	12.59
35	13.10	13.10	13.65	13.65	13.26	13.26	13.81	13.81	10.40	10.40
36	14.14	14.14	14.93	14.93	13.89	13.89	14.16	14.16	12.97	12.97
37	14.15	14.15	11.84	11.84	11.93	11.93	10.17	10.17	9.57	9.57
38	13.00	13.00	13.84	13.84	12.51	12.51	13.58	13.58	9.09	9.09
39	13.10	13.10	11.50	11.50	13.57	13.57	8.38	8.38	9.99	9.99
40	10.40	10.40	11.79	11.79	12.22	12.22	9.67	9.67	8.79	8.79
41	12.89	12.89	10.54	10.54	10.86	10.86	9.60	9.60	6.82	6.82
42	11.09	11.09	12.60	12.60	12.70	12.70	8.19	8.19	9.10	9.10
43	10.44	10.44	13.29	13.29	12.55	12.55	9.77	9.77	7.81	7.81
44	10.17	10.17	12.23	12.23	12.05	12.05	9.13	9.13	6.05	6.05
45	11.39	11.39	12.88	12.88	11.55	11.55	9.63	9.63	9.80	9.80
46	13.87	13.87	12.32	12.32	9.57	9.57	11.15	11.15	8.22	8.22
47	11.96	11.96	10.67	10.67	5.61	5.61	4.05	4.05	2.83	2.83
48	12.61	12.61	12.07	12.07	0.90	0.90	11.01	11.01	0.70	0.70
49	9.5/	9.5/	8.64	8.64	2.11	2.11	5.61	5.61	1.58	/.38
50	10.28	10.28	9.12	9.12	1.28	1.28	4.42	4.42	5.30	5.50

Reflected	Incidence Angle in °									
Angle in °	11	12	13	14	15	16	17	18	19	20
51	8.81	8.81	-0.67	-0.67	2.82	2.82	4.40	4.40	4.83	4.83
52	5.03	5.03	2.79	2.79	2.24	2.24	8.24	8.24	5.27	5.27
53	5.82	5.82	-3.64	-3.64	4.26	4.26	2.06	2.06	8.64	8.64
54	0.31	0.31	0.77	0.77	4.68	4.68	4.35	4.35	7.13	7.13
55	-0.42	-0.42	1.28	1.28	5.07	5.07	8.13	8.13	11.17	11.17
56	0.78	0.78	1.59	1.59	10.21	10.21	7.65	7.65	13.48	13.48
57	5.18	5.18	4.87	4.87	9.62	9.62	12.55	12.55	12.75	12.75
58	3.92	3.92	11.13	11.13	12.28	12.28	14.32	14.32	14.44	14.44
59	7.24	7.24	9.87	9.87	12.88	12.88	14.69	14.69	15.56	15.56
60	7.43	7.43	12.89	12.89	15.62	15.62	15.48	15.48	17.71	17.71
61	9.83	9.83	13.96	13.96	13.84	13.84	16.77	16.77	16.75	16.75
62	11.79	11.79	12.50	12.50	14.31	14.31	17.36	17.36	17.83	17.83
63	11.00	11.00	14.68	14.68	16.53	16.53	15.29	15.29	18.11	18.11
64	11.25	11.25	15.24	15.24	15.64	15.64	18.12	18.12	16.54	16.54
65	9.80	9.80	15.86	15.86	15.96	15.96	17.84	17.84	14.34	14.34
66	12.59	12.59	11.63	11.63	18.09	18.09	16.31	16.31	15.39	15.39
67	9.35	9.35	12.30	12.30	17.85	17.85	14.72	14.72	19.30	19.30
68	8.74	8.74	12.33	12.33	14.26	14.26	17.71	17.71	23.06	23.06
69	12.56	12.56	11.20	11.20	7.09	7.09	19.49	19.49	25.86	25.86
70	13.83	13.83	12.59	12.59	16.21	16.21	22.44	22.44	26.23	26.23
71	10.27	10.27	16.50	16.50	23.71	23.71	26.23	26.23	29.53	29.53
72	7.76	7.76	16.88	16.88	27.90	27.90	30.05	30.05	30.17	30.17
73	15.41	15.41	18.38	18.38	28.33	28.33	31.16	31.16	29.14	29.14
74	15.39	15.39	26.00	26.00	25.83	25.83	28.80	28.80	19.09	19.09
75	18.98	18.98	29.17	29.17	28.08	28.08	22.18	22.18	16.63	16.63
76	26.82	26.82	26.84	26.84	29.72	29.72	25.42	25.42	20.33	20.33
77	27.36	27.36	26.50	26.50	25.49	25.49	24.10	24.10	14.15	14.15
78	26.57	26.57	29.60	29.60	20.42	20.42	8.67	8.67	16.44	16.44
79	29.61	29.61	25.07	25.07	24.00	24.00	16.74	16.74	12.74	12.74
80	28.11	28.11	22.41	22.41	18.01	18.01	10.34	10.34	11.95	11.95
81	26.16	26.16	23.21	23.21	13.21	13.21	15.57	15.57	9.81	9.81
82	24.69	24.69	12.97	12.97	8.77	8.77	11.95	11.95	6.43	6.43
83	19.12	19.12	13.01	13.01	15.79	15.79	8.88	8.88	8.37	8.37
84	16.34	16.34	2.29	2.29	11.86	11.86	10.66	10.66	4.80	4.80
85	10.77	10.77	14.62	14.62	11.53	11.53	13.58	13.58	10.74	10.74
86	10.64	10.64	8.20	8.20	8.35	8.35	8.85	8.85	2.29	2.29
87	5.73	5.73	15.99	15.99	12.98	12.98	13.58	13.58	10.30	10.30
88	15.33	15.33	11.34	11.34	12.06	12.06	10.48	10.48	11.93	11.93
89	8.87	8.87	12.20	12.20	14.74	14.74	10.46	10.46	9.67	9.67
90	15.98	15.98	12.75	12.75	10.10	10.10	11.48	11.48	6.59	6.59
91	13.26	13.26	10.41	10.41	7.80	7.80	8.73	8.73	0.91	0.91
92	9.80	9.80	14.98	14.98	10.63	10.63	6.41	6.41	4.38	4.38
93	11.87	11.87	8.53	8.53	10.93	10.93	10.57	10.57	10.18	10.18
94	11.37	11.37	2.33	2.33	8.64	8.64	8.81	8.81	4.86	4.86
95	8.96	8.96	5.60	5.60	6.25	6.25	7.27	7.27	12.71	12.71
96	5.03	5.03	4.74	4.74	6.89	6.89	10.01	10.01	7.65	7.65
97	8.06	8.06	9.02	9.02	7.69	7.69	15.85	15.85	8.54	8.54
98	2.30	2.30	7.93	7.93	11.04	11.04	14.10	14.10	11.99	11.99
99	9.46	9.46	8.97	8.97	4.29	4.29	15.07	15.07	10.01	10.01
100	8.12	8.12	11.81	11.81	14.60	14.60	13.22	13.22	10.06	10.06

Reflected	Incidence Angle in °										
Angle in $^{\circ}$	11	12	13	14	15	16	17	18	19	20	
101	10.68	10.68	4.44	4.44	9.91	9.91	14.41	14.41	10.51	10.51	
102	7.26	7.26	13.24	13.24	13.05	13.05	12.18	12.18	12.41	12.41	
103	13.60	13.60	10.05	10.05	11.40	11.40	11.65	11.65	11.46	11.46	
104	14.30	14.30	11.65	11.65	12.71	12.71	14.10	14.10	9.89	9.89	
105	8.95	8.95	11.43	11.43	12.47	12.47	10.70	10.70	8.26	8.26	
106	4.47	4.47	10.15	10.15	13.28	13.28	9.30	9.30	10.99	10.99	
107	6.38	6.38	12.15	12.15	11.81	11.81	12.14	12.14	9.87	9.87	
108	7.64	7.64	10.50	10.50	12.22	12.22	12.93	12.93	15.91	15.91	
109	-2.55	-2.55	10.25	10.25	12.97	12.97	10.96	10.96	14.80	14.80	
110	-1.28	-1.28	8.55	8.55	12.03	12.03	14.63	14.63	3.91	3.91	
111	4.63	4.63	5.51	5.51	10.56	10.56	15.03	15.03	8.76	8.76	
112	1.06	1.06	5.99	5.99	11.24	11.24	7.39	7.39	9.28	9.28	
113	2.90	2.90	3.35	3.35	13.32	13.32	8.88	8.88	10.99	10.99	
114	7.26	7.26	11.48	11.48	13.18	13.18	12.55	12.55	21.19	21.19	
115	10.30	10.30	12.28	12.28	7.29	7.29	-1.36	-1.36	21.23	21.23	
116	6.53	6.53	9.76	9.76	8.97	8.97	15.38	15.38	13.13	13.13	
117	16.10	16.10	-1.59	-1.59	-6.16	-6.16	16.31	16.31	4.87	4.87	
118	13.78	13.78	8.08	8.08	17.84	17.84	11.11	11.11	8.48	8.48	
119	10.90	10.90	2.39	2.39	18.99	18.99	12.25	12.25	8.68	8.68	
120	15.62	15.62	19.00	19.00	3.99	3.99	9.62	9.62	8.51	8.51	
121	14.13	14.13	18.04	18.04	-3.29	-3.29	7.63	7.63	9.46	9.46	
122	17.68	17.68	10.38	10.38	5.94	5.94	10.90	10.90	11.62	11.62	
123	21.62	21.62	12.74	12.74	2.75	2.75	10.60	10.60	24.58	24.58	
124	10.10	10.10	3.16	3.16	11.19	11.19	10.52	10.52	12.99	12.99	
125	14.20	14.20	11.31	11.31	6.89	6.89	24.70	24.70	11.18	11.18	
126	10.80	10.80	11.91	11.91	8.59	8.59	6.02	6.02	16.74	16.74	
127	7.36	7.36	9.80	9.80	25.48	25.48	13.89	13.89	9.60	9.60	
128	7.68	7.68	8.76	8.76	11.18	11.18	17.92	17.92	18.73	18.73	
129	6.78	6.78	22.86	22.86	13.99	13.99	18.28	18.28	8.53	8.53	
130	9.00	9.00	6.44	6.44	8.74	8.74	15.49	15.49	11.10	11.10	
131	23.50	23.50	10.69	10.69	10.95	10.95	10.00	10.00	16.72	16.72	
132	15.13	15.13	12.46	12.46	12.76	12.76	12.76	12.76	16.54	16.54	
133	17.40	17.40	13.77	13.77	9.26	9.26	14.86	14.86	19.61	19.61	
134	18.21	18.21	16.74	16.74	5.08	5.08	13.61	13.61	15.50	15.50	
135	1.45	1.45	14.24	14.24	11.09	11.09	9.14	9.14	13.37	13.37	
136	18.06	18.06	11.50	11.50	7.42	7.42	18.17	18.17	9.98	9.98	
137	8.48	8.48	1.70	1.70	9.42	9.42	9.47	9.47	14.11	14.11	
138	3.52	3.52	13.33	13.33	16.23	16.23	11.10	11.10	16.03	16.03	
139	7.95	7.95	8.36	8.36	12.11	12.11	10.50	10.50	16.52	16.52	
140	15.69	15.69	22.12	22.12	9.07	9.07	11.21	11.21	9.73	9.73	
141	15.90	15.90	16.04	16.04	4.56	4.56	8.77	8.77	14.84	14.84	
142	20.79	20.79	16.65	16.65	12.86	12.86	12.12	12.12	14.92	14.92	
143	7.87	7.87	6.29	6.29	12.45	12.45	11.67	11.67	11.66	11.66	
144	9.72	9.72	13.41	13.41	9.43	9.43	18.09	18.09	16.16	16.16	
145	15.70	15.70	11.12	11.12	17.31	17.31	12.11	12.11	12.86	12.86	
146	10.03	10.03	16.41	16.41	7.30	7.30	12.32	12.32	15.34	15.34	
147	9.46	9.46	11.70	11.70	18.18	18.18	16.90	16.90	21.60	21.60	
148	8.73	8.73	12.57	12.57	18.92	18.92	18.78	18.78	20.02	20.02	
149	11.96	11.96	14.32	14.32	18.05	18.05	15.39	15.39	15.90	15.90	
150	7.73	7.73	15.66	15.66	20.87	20.87	16.15	16.15	21.16	21.16	

Reflected	Incidence Angle in °										
Angle in °	11	12	13	14	15	16	17	18	19	20	
151	9.66	9.66	14.74	14.74	16.43	16.43	17.13	17.13	15.55	15.55	
152	10.69	10.69	11.68	11.68	16.60	16.60	15.64	15.64	13.44	13.44	
153	18.65	18.65	18.04	18.04	18.21	18.21	17.20	17.20	21.33	21.33	
154	18.47	18.47	11.43	11.43	17.83	17.83	11.98	11.98	23.36	23.36	
155	20.81	20.81	8.87	8.87	8.40	8.40	14.01	14.01	27.01	27.01	
156	16.17	16.17	14.74	14.74	19.41	19.41	29.56	29.56	26.58	26.58	
157	14.37	14.37	11.35	11.35	24.99	24.99	32.88	32.88	32.84	32.84	
158	21.56	21.56	19.58	19.58	26.64	26.64	32.65	32.65	36.11	36.11	
159	24.22	24.22	22.44	22.44	30.23	30.23	29.60	29.60	39.23	39.23	
160	18.62	18.62	26.54	26.54	31.02	31.02	29.36	29.36	38.50	38.50	
161	19.23	19.23	24.40	24.40	33.88	33.88	34.25	34.25	41.75	41.75	
162	27.19	27.19	21.68	21.68	31.56	31.56	35.21	35.21	54.47	54.47	
163	30.19	30.19	28.26	28.26	34.43	34.43	40.25	40.25	43.28	43.28	
164	28.93	28.93	33.02	33.02	30.40	30.40	52.28	52.28	36.20	36.20	
165	33.03	33.03	34.80	34.80	36.93	36.93	40.23	40.23	28.58	28.58	
166	28.18	28.18	37.36	37.36	46.81	46.81	36.30	36.30	30.53	30.53	
167	34.73	34.73	37.38	37.38	39.51	39.51	32.44	32.44	34.91	34.91	
168	40.28	40.28	47.95	47.95	41.66	41.66	21.85	21.85	35.35	35.35	
169	39.35	39.35	40.01	40.01	36.37	36.37	32.66	32.66	25.63	25.63	
170	50.69	50.69	40.69	40.69	28.49	28.49	31.44	31.44	31.03	31.03	
171	37.33	37.33	37.37	37.37	35.07	35.07	32.70	32.70	26.56	26.56	
172	37.94	37.94	35.89	35.89	32.91	32.91	27.00	27.00	32.41	32.41	
173	37.27	37.27	36.60	36.60	34.32	34.32	31.23	31.23	31.82	31.82	
174	34.86	34.86	28.22	28.22	30.51	30.51	32.62	32.62	17.80	17.80	
175	32.68	32.68	32.99	32.99	33.10	33.10	34.89	34.89	32.87	32.87	
176	31.38	31.38	29.22	29.22	23.01	23.01	31.49	31.49	22.49	22.49	
177	33.01	33.01	26.73	26.73	35.09	35.09	32.50	32.50	32.50	32.50	
178	22.70	22.70	33.06	33.06	35.15	35.15	11.79	11.79	32.61	32.61	
179	31.53	31.53	37.49	37.49	32.11	32.11	30.74	30.74	23.67	23.67	
180	29.61	29.61	35.13	35.13	30.06	30.06	34.77	34.77	27.86	27.86	
181	32.42	32.42	31.90	31.90	27.43	27.43	37.66	37.66	26.77	26.77	
182	33.26	33.26	31.96	31.96	32.34	32.34	39.75	39.75	28.89	28.89	
183	28.02	28.02	38.76	38.76	38.59	38.59	38.83	38.83	29.97	29.97	
184	42.72	42.72	41.91	41.91	41.66	41.66	26.19	26.19	35.39	35.39	
185	45.36	45.36	35.09	35.09	37.95	37.95	38.96	38.96	37.70	37.70	
186	39.30	39.30	40.98	40.98	38.72	38.72	41.36	41.36	30.89	30.89	
187	40.59	40.59	39.56	39.56	41.12	41.12	38.74	38.74	38.75	38.75	
188	50.11	50.11	46.48	46.48	38.01	38.01	41.69	41.69	27.07	27.07	
189	50.90	50.90	37.73	37.73	42.80	42.80	43.06	43.06	42.86	42.86	
190	83.95	83.95	37.84	37.84	36.81	36.81	41.47	41.47	41.85	41.85	
191	55.81	55.81	58.30	58.30	45.54	45.54	39.89	39.89	41.62	41.62	
192	40.31	40.31	84.69	84.69	52.59	52.59	45.87	45.87	43.66	43.66	
193	44.33	44.33	60.30	60.30	60.51	60.51	43.02	43.02	44.69	44.69	
194	32.20	32.20	51.64	51.64	85.36	85.36	44.65	44.65	42.19	42.19	
195	33.14	33.14	41.08	41.08	55.51	55.51	50.19	50.19	39.39	39.39	
196	39.00	39.00	44.49	44.49	46.14	46.14	85.92	85.92	51.24	51.24	
197	35.15	35.15	43.09	43.09	35.30	35.30	57.27	57.27	60.06	60.06	
198	33.91	33.91	41.61	41.61	37.11	37.11	49.93	49.93	86.58	86.58	
199	28.76	28.76	37.96	37.96	41.21	41.21	43.63	43.63	60.17	60.17	
200	24.61	24.61	18.10	18.10	28.91	28.91	46.88	46.88	40.60	40.60	

Reflected	Incidence Angle in °										
Angle in °	11	12	13	14	15	16	17	18	19	20	
201	26.93	26.93	28.86	28.86	33.44	33.44	37.48	37.48	46.33	46.33	
202	19.69	19.69	23.90	23.90	28.74	28.74	34.07	34.07	37.44	37.44	
203	21.25	21.25	19.67	19.67	32.42	32.42	27.21	27.21	29.36	29.36	
204	18.66	18.66	19.73	19.73	22.56	22.56	30.18	30.18	33.55	33.55	
205	10.44	10.44	22.40	22.40	24.29	24.29	23.99	23.99	33.37	33.37	
206	24.83	24.83	30.02	30.02	23.95	23.95	10.09	10.09	30.32	30.32	
207	22.80	22.80	25.21	25.21	16.82	16.82	18.88	18.88	31.67	31.67	
208	25.08	25.08	25.10	25.10	20.90	20.90	22.73	22.73	17.32	17.32	
209	25.88	25.88	18.85	18.85	16.69	16.69	17.23	17.23	27.69	27.69	
210	4.64	4.64	5.57	5.57	5.99	5.99	27.32	27.32	19.74	19.74	
211	23.53	23.53	16.59	16.59	15.15	15.15	14.88	14.88	19.19	19.19	
212	19.98	19.98	18.79	18.79	12.65	12.65	14.87	14.87	23.79	23.79	
213	20.63	20.63	21.88	21.88	22.70	22.70	21.13	21.13	14.20	14.20	
214	19.24	19.24	23.21	23.21	18.60	18.60	22.92	22.92	12.72	12.72	
215	22.14	22.14	25.15	25.15	22.06	22.06	21.86	21.86	9.77	9.77	
216	22.93	22.93	20.15	20.15	24.08	24.08	13.52	13.52	18.52	18.52	
217	23.33	23.33	22.29	22.29	24.36	24.36	20.28	20.28	23.44	23.44	
218	22.53	22.53	21.73	21.73	18.82	18.82	19.74	19.74	17.58	17.58	
219	17.18	17.18	20.49	20.49	20.58	20.58	20.87	20.87	21.98	21.98	
220	15.55	15.55	18.00	18.00	19.70	19.70	18.15	18.15	21.78	21.78	
221	15.22	15.22	12.67	12.67	14.45	14.45	13.46	13.46	10.40	10.40	
222	10.03	10.03	16.70	16.70	7.56	7.56	16.08	16.08	16.38	16.38	
223	17.08	17.08	9.66	9.66	8.92	8.92	12.97	12.97	15.80	15.80	
224	13.40	13.40	5.77	5.77	9.89	9.89	15.47	15.47	16.04	16.04	
225	-2.65	-2.65	8.37	8.37	13.50	13.50	15.52	15.52	14.49	14.49	
226	10.79	10.79	8.98	8.98	7.73	7.73	9.05	9.05	6.64	6.64	
227	13.99	13.99	-2.79	-2.79	5.46	5.46	15.66	15.66	15.59	15.59	
228	15.07	15.07	8.83	8.83	8.97	8.97	8.75	8.75	6.41	6.41	
229	13.74	13.74	12.52	12.52	12.31	12.31	-1.11	-1.11	-0.51	-0.51	
230	13.68	13.68	10.72	10.72	13.56	13.56	-1.10	-1.10	4.49	4.49	
231	11.18	11.18	11.23	11.23	16.52	16.52	8.99	8.99	-9.12	-9.12	
232	13.41	13.41	15.49	15.49	14.13	14.13	14.54	14.54	4.20	4.20	
233	15.07	15.07	15.89	15.89	8.99	8.99	10.30	10.30	-6.03	-6.03	
234	12.57	12.57	15.55	15.55	15.78	15.78	11.79	11.79	10.53	10.53	
235	10.26	10.26	14.60	14.60	15.43	15.43	14.90	14.90	9.30	9.30	
236	14.67	14.67	15.61	15.61	15.45	15.45	13.36	13.36	17.97	17.97	
237	11.07	11.07	13.10	13.10	12.80	12.80	12.45	12.45	13.73	13.73	
238	9.84	9.84	12.74	12.74	13.75	13.75	14.33	14.33	9.68	9.68	
239	8.08	8.08	13.99	13.99	15.70	15.70	18.14	18.14	17.22	17.22	
240	11.31	11.31	10.63	10.63	15.28	15.28	14.58	14.58	16.39	16.39	
241	8.86	8.86	11.34	11.34	11.11	11.11	14.97	14.97	13.57	13.57	
242	6.55	6.55	12.75	12.75	5.86	5.86	14.02	14.02	12.58	12.58	
243	6.43	6.43	3.37	3.37	9.60	9.60	10.88	10.88	15.98	15.98	
244	-4.93	-4.93	10.53	10.53	8.47	8.47	14.51	14.51	14.03	14.03	
245	-3.66	-3.66	3.64	3.64	5.07	5.07	14.81	14.81	13.13	13.13	
246	14.16	14.16	1.68	1.68	2.73	2.73	11.55	11.55	14.45	14.45	
247	10.29	10.29	12.28	12.28	3.31	3.31	8.65	8.65	9.94	9.94	
248	14.57	14.57	1.15	1.15	-1.60	-1.60	9.87	9.87	-1.54	-1.54	
249	12.25	12.25	7.31	7.31	2.54	2.54	9.24	9.24	6.46	6.46	
250	11.37	11.37	9.89	9.89	8.39	8.39	5.66	5.66	12.62	12.62	
Reflected	Incidence Angle in °										
------------	----------------------	-------	-------	-------	-------	-------	-------	-------	-------	-------	
Angle in °	11	12	13	14	15	16	17	18	19	20	
251	10.31	10.31	9.78	9.78	7.74	7.74	13.26	13.26	16.25	16.25	
252	12.99	12.99	12.40	12.40	13.07	13.07	13.30	13.30	12.78	12.78	
253	15.30	15.30	14.23	14.23	17.40	17.40	9.52	9.52	5.17	5.17	
254	13.25	13.25	17.39	17.39	16.33	16.33	11.89	11.89	11.95	11.95	
255	14.33	14.33	18.77	18.77	0.44	0.44	15.89	15.89	6.25	6.25	
256	16.44	16.44	18.34	18.34	15.91	15.91	15.92	15.92	3.09	3.09	
257	14.52	14.52	12.42	12.42	14.04	14.04	14.59	14.59	6.11	6.11	
258	13.31	13.31	17.21	17.21	16.87	16.87	4.95	4.95	8.93	8.93	
259	17.82	17.82	14.52	14.52	8.33	8.33	8.42	8.42	11.43	11.43	
260	4.80	4.80	22.90	22.90	13.68	13.68	11.47	11.47	13.76	13.76	
261	12.45	12.45	12.99	12.99	5.33	5.33	12.02	12.02	13.92	13.92	
262	20.96	20.96	10.61	10.61	4.07	4.07	14.77	14.77	10.42	10.42	
263	16.69	16.69	15.10	15.10	12.84	12.84	15.70	15.70	11.08	11.08	
264	14.92	14.92	10.23	10.23	14.19	14.19	10.38	10.38	14.61	14.61	
265	12.07	12.07	8.86	8.86	12.99	12.99	11.49	11.49	14.72	14.72	
266	15.69	15.69	11.87	11.87	10.99	10.99	8.46	8.46	15.20	15.20	
267	10.63	10.63	9.18	9.18	0.19	0.19	11.83	11.83	15.57	15.57	
268	15.12	15.12	11.73	11.73	11.47	11.47	11.60	11.60	11.15	11.15	
269	11.96	11.96	7.24	7.24	6.87	6.87	13.39	13.39	10.46	10.46	
270	13.29	13.29	11.01	11.01	15.50	15.50	11.33	11.33	12.30	12.30	
271	8.78	8.78	3.39	3.39	7.87	7.87	8.12	8.12	13.84	13.84	
272	9.09	9.09	12.56	12.56	10.11	10.11	10.54	10.54	7.37	7.37	
273	9.57	9.57	11.24	11.24	12.77	12.77	11.49	11.49	7.51	7.51	
274	12.56	12.56	10.99	10.99	14.92	14.92	11.53	11.53	11.00	11.00	
275	8.32	8.32	10.60	10.60	13.09	13.09	10.69	10.69	6.70	6.70	
276	9.35	9.35	15.38	15.38	15.00	15.00	13.69	13.69	8.87	8.87	
277	10.55	10.55	12.08	12.08	12.31	12.31	10.09	10.09	8.15	8.15	
278	10.08	10.08	14.64	14.64	12.48	12.48	13.17	13.17	-0.71	-0.71	
279	12.23	12.23	11.88	11.88	14.98	14.98	9.96	9.96	2.47	2.47	
280	11.63	11.63	13.50	13.50	13.82	13.82	10.73	10.73	-1.04	-1.04	
281	13.21	13.21	12.22	12.22	14.38	14.38	6.90	6.90	6.84	6.84	
282	13.29	13.29	15.44	15.44	13.09	13.09	11.52	11.52	6.10	6.10	
283	14.36	14.36	14.15	14.15	9.71	9.71	8.31	8.31	9.90	9.90	
284	12.15	12.15	11.79	11.79	12.76	12.76	8.26	8.26	10.02	10.02	
285	11.12	11.12	13.90	13.90	10.54	10.54	11.37	11.37	7.42	7.42	
286	10.75	10.75	8.38	8.38	8.82	8.82	11.70	11.70	11.18	11.18	
287	12.72	12.72	8.62	8.62	10.85	10.85	10.44	10.44	11.48	11.48	
288	7.37	7.37	8.92	8.92	11.45	11.45	10.17	10.17	12.44	12.44	
289	8.52	8.52	9.80	9.80	11.49	11.49	10.36	10.36	12.48	12.48	
290	8.79	8.79	9.09	9.09	11.72	11.72	12.72	12.72	12.69	12.69	
291	8.96	8.96	9.09	9.09	12.86	12.86	12.54	12.54	11.54	11.54	
292	1.99	1.99	10.58	10.58	11.79	11.79	11.22	11.22	12.62	12.62	
293	6.17	6.17	8.94	8.94	10.42	10.42	9.44	9.44	13.15	13.15	
294	8.14	8.14	4.72	4.72	10.41	10.41	7.51	7.51	10.39	10.39	
295	6.14	6.14	6.44	6.44	7.96	7.96	11.16	11.16	10.47	10.47	
296	-7.47	-7.47	9.03	9.03	8.80	8.80	9.67	9.67	/.59	/.59	
297	6.43	6.43	-/.54	-1.54	0.69	6.69	6.85	6.85	10.68	10.68	
298	2.71	2.71	7.44	7.44	0.81	0.81	9.90	9.90	9.32	9.32	
299	4.45	4.45	-2.89	-2.89	5.70	5.70	9.92	9.92	12.73	12.73	
300	4.58	4.58	6.40	6.40	5.18	5.18	10.46	10.46	9.52	9.52	

Reflected	Incidence Angle in °									
Angle in $^{\circ}$	11	12	13	14	15	16	17	18	19	20
301	5.28	5.28	3.16	3.16	3.72	3.72	6.93	6.93	13.80	13.80
302	11.06	11.06	2.06	2.06	6.87	6.87	6.29	6.29	14.89	14.89
303	4.20	4.20	11.00	11.00	7.69	7.69	5.36	5.36	7.11	7.11
304	9.83	9.83	8.22	8.22	8.10	8.10	10.56	10.56	5.37	5.37
305	5.66	5.66	12.15	12.15	5.38	5.38	5.85	5.85	6.80	6.80
306	12.01	12.01	11.42	11.42	14.62	14.62	12.52	12.52	6.65	6.65
307	8.73	8.73	15.30	15.30	14.25	14.25	11.42	11.42	4.77	4.77
308	12.64	12.64	12.85	12.85	8.95	8.95	14.04	14.04	7.49	7.49
309	11.39	11.39	12.33	12.33	14.35	14.35	14.33	14.33	11.74	11.74
310	15.84	15.84	12.96	12.96	13.52	13.52	10.67	10.67	11.35	11.35
311	16.01	16.01	14.72	14.72	14.64	14.64	14.54	14.54	9.99	9.99
312	15.71	15.71	13.32	13.32	15.24	15.24	11.39	11.39	12.73	12.73
313	14.79	14.79	15.11	15.11	14.77	14.77	15.30	15.30	9.21	9.21
314	15.76	15.76	12.58	12.58	14.48	14.48	9.43	9.43	12.43	12.43
315	15.47	15.47	15.98	15.98	16.22	16.22	13.57	13.57	13.69	13.69
316	15.38	15.38	16.44	16.44	16.12	16.12	11.68	11.68	12.00	12.00
317	16.39	16.39	16.18	16.18	14.41	14.41	15.28	15.28	8.91	8.91
318	15.74	15.74	17.17	17.17	12.88	12.88	15.91	15.91	14.66	14.66
319	14.80	14.80	16.57	16.57	13.37	13.37	18.22	18.22	14.56	14.56
320	14.49	14.49	14.68	14.68	18.03	18.03	16.68	16.68	11.48	11.48
321	12.15	12.15	14.73	14.73	17.25	17.25	15.19	15.19	20.47	20.47
322	16.53	16.53	11.70	11.70	16.04	16.04	15.50	15.50	18.82	18.82
323	11.53	11.53	16.60	16.60	15.89	15.89	17.53	17.53	10.17	10.17
324	16.03	16.03	15.62	15.62	15.76	15.76	10.68	10.68	17.50	17.50
325	15.49	15.49	19.04	19.04	17.31	17.31	17.10	17.10	17.22	17.22
326	17.44	17.44	16.16	16.16	15.51	15.51	17.72	17.72	6.89	6.89
327	14.21	14.21	16.31	16.31	12.60	12.60	13.94	13.94	12.10	12.10
328	16.88	16.88	13.12	13.12	13.63	13.63	11.55	11.55	14.74	14.74
329	15.42	15.42	14.45	14.45	14.39	14.39	13.38	13.38	18.99	18.99
330	20.60	20.60	11.23	11.23	10.33	10.33	14.09	14.09	20.14	20.14
331	16.77	16.77	9.38	9.38	9.49	9.49	11.32	11.32	15.75	15.75
332	11.26	11.26	14.12	14.12	15.32	15.32	13.34	13.34	12.46	12.46
333	15.53	15.53	14.92	14.92	20.78	20.78	12.35	12.35	27.08	27.08
334	15.18	15.18	16.15	16.15	19.10	19.10	17.38	17.38	17.29	17.29
335	13.71	13.71	13.88	13.88	19.25	19.25	25.12	25.12	19.70	19.70
336	14.23	14.23	17.99	17.99	19.77	19.77	19.80	19.80	14.22	14.22
337	18.78	18.78	18.19	18.19	21.37	21.37	20.67	20.67	18.90	18.90
338	21.58	21.58	14.56	14.56	11.90	11.90	3.30	3.30	22.46	22.46
339	18.42	18.42	22.81	22.81	16.50	16.50	11.40	11.40	24.86	24.86
340	23.22	23.22	21.33	21.33	14.43	14.43	14.33	14.33	33.24	33.24
341	26.51	26.51	21.11	21.11	14.27	14.27	24.97	24.97	40.15	40.15
342	17.94	17.94	21.27	21.27	22.16	22.16	37.32	37.32	63.55	63.55
343	18.66	18.66	19.63	19.63	24.57	24.57	43.28	43.28	41.32	41.32
344	9.58	9.58	21.59	21.59	36.95	36.95	64.09	64.09	30.42	30.42
345	18.34	18.34	22.16	22.16	41.54	41.54	36.20	36.20	19.29	19.29
346	1/.46	1/.46	33.81	33.81	63./1	63./1	27.93	27.93	12.51	12.51
347	25.26	25.26	54.04	54.04	35.23	35.23	19.79	19.79	17.51	1/.51
348	51.43	51.43	03.96	03.96	51.46	51.46	10.00	10.00	18.0/	18.67
250	41.01	41.01	33.31	33.31	23.28	23.28	19.20	19.20	10.38	10.38
330	04.33	04.33	28.28	28.28	10.52	10.52	13.37	13.37	10.78	10.78

Reflected	Incidence Angle in °									
Angle in °	11	12	13	14	15	16	17	18	19	20
351	30.45	30.45	19.82	19.82	18.92	18.92	13.74	13.74	0.97	0.97
352	28.76	28.76	22.20	22.20	15.98	15.98	15.62	15.62	14.58	14.58
353	20.71	20.71	20.03	20.03	9.53	9.53	7.99	7.99	9.54	9.54
354	17.81	17.81	15.04	15.04	19.14	19.14	14.55	14.55	16.48	16.48
355	19.76	19.76	13.00	13.00	9.55	9.55	16.33	16.33	10.34	10.34
356	13.23	13.23	14.32	14.32	12.71	12.71	4.00	4.00	14.58	14.58
357	13.80	13.80	17.12	17.12	17.32	17.32	13.32	13.32	18.57	18.57
358	3.69	3.69	14.62	14.62	15.89	15.89	12.94	12.94	15.69	15.69
359	12.05	12.05	17.37	17.37	13.46	13.46	16.63	16.63	16.40	16.40
360	16.56	16.56	12.73	12.73	17.76	17.76	15.90	15.90	18.21	18.21

APPENDIX F. MATLAB SOURCE CODE FOR DETECTION COVERAGE PLOT

```
% Chong Sze Sing (March 2014)
% Bistatic RCS Look-up Table
load('3GHz full.mat')
biRCS3 = num;
load('9GHz full.mat')
biRCS9 = num;
sen=10; % sensitivity
% For Tx 1
Xcoord1 = [];
Ycoord1 = [];
SNR1 = [];
SNRdBm1 = [];
Err thetaR1 = [];
Err time1 = [];
Err L1 = [];
Tx1=45;
L1=18000;
%For Tx2
Xcoord2 = [];
Ycoord2 = [];
SNR2 = [];
SNRdBm2 = [];
Err thetaR2 = [];
Err_time2 = [];
Err_L2 = [];
Tx2=135;
L2=18000;
%For Tx3
Xcoord3 = [];
Ycoord3 = [];
SNR3 = [];
SNRdBm3 = [];
Err thetaR3 = [];
Err time3 = [];
Err L3 = [];
Tx3=225;
L3=18000;
%For Tx4
Xcoord4 = [];
Ycoord4 = [];
SNR4 = [];
SNRdBm4 = [];
Err thetaR4 = [];
Err time4 = [];
Err L4 = [];
Tx4=315;
L4=18000;
```

```
%For Tx5
Xcoord5 = [];
Ycoord5 = [];
SNR5 = [];
SNRdBm5 = [];
Err thetaR5 = [];
Err time5 = [];
Err L5 = [];
Tx5=0;
L5=18000;
%For Tx6
Xcoord6 = [];
Ycoord6 = [];
SNR6 = [];
SNRdBm6 = [];
Err thetaR6 = [];
Err time6 = [];
Err L6 = [];
Tx6=90;
L6=18000;
%For Tx7
Xcoord7 = [];
Ycoord7 = [];
SNR7 = [];
SNRdBm7 = [];
Err thetaR7 = [];
Err time7 = [];
Err L7 = [];
Tx7=180;
L7=18000;
%For Tx8
Xcoord8 = [];
Ycoord8 = [];
SNR8 = [];
SNRdBm8 = [];
Err thetaR8 = [];
Err_time8 = [];
Err_L8 = [];
Tx8=270;
L8=18000;
% Bi-static Radar Equation
% Constants
k = 1.38e-23; % Boltzmann
c = 3e8;
               % Speed of light
% Variables
% Parameters are based Furuno Marine Radar FAR-28x7 series in both X-band
and S-band
% Bi-static Radar Equation
% Constants
k = 1.38e-23; % Boltzmann
c = 3e8;
               % Speed of light
```

```
Select = 1; %chose which radar to use
% Variables
% Parameters are based Furuno Marine Radar FAR-28x7 series in both X-band
% and S-band
%S-band Radar (1)
BeamAz S = 1.8; % in degrees
BeamEL S = 25;
               % in degrees
Scanrate_S = 45; % in rpm for range of 24 nmi or higher
Pt S = 30e3;
               % Tx peak power output
Gt S = 10^(28/10); % Tx Antenna Gain given as 28dB
          % Rx Antenna Gain
Gr S = 2;
f S = 3050e6;
              % frequency
lambda S = c/f S; % wavelength
Radar - Chp 6.5.5.6
Fr S = 1;
                % Prop Factor for Rx-Tgt path
Ts S = 290+290*(13.5-1); % System Noise Temp of Rx of NF=6dB using
formula Te=(NF-1)290 and Ts=Te+290K
Bn S = 500e6; % Noise Bandwidth of ESM Rx Wideband Channalised (500
MHz)
Lt S = 2;
                % Ref to See Advances in Bistatic Radar - Chp 6.5.5.6
Lr S = 2;
% this is calculation of pulse integration sqrt(N)
PRF S = 2000;
TOT S = BeamAz S/(Scanrate S*360);
N S = TOT S * PRF S;
%X-band Radar (2)
                % in degrees
BeamAz X = 0.95;
BeamEL X = 25;
               % in degrees
Scanrate X = 45; % in rpm for range of 24 nmi or higher
Pt X = 25e3;
             % Tx peak power output
Gt X = 10<sup>(31/10)</sup>; % Tx Antenna Gain is given as 31 dB
Gr X = 2; % Rx Antenna Gain
              % frequency
f X = 9410e6;
lambda X = c/f X; % wavelength
Ft X = 1;
               % Prop Factor for Tx-Tqt path. See Advances in Bistatic
Radar - Chp 6.5.5.6
Fr X = 1; % Prop Factor for Rx-Tqt path
Ts X = 290+290*(6-1); % System Noise Temp of Rx of NF=6dB using
formula Te=(NF-1)290 and Ts=Te+290K
Bn X = 500e6; % Noise Bandwidth of ESM Rx Wideband Channalised (500
MHz)
Lt X = 2;
               % Ref to See Advances in Bistatic Radar - Chp 6.5.5.6
Lr X = 2;
% this is calculation of pulse integration sqrt(N)
PRF X = 1500;
TOT X = BeamAz X/(Scanrate_X*360);
N X = TOT X*PRF X;
```

```
if Select ==1
   BeamEL = BeamEL_S; % in degrees
   BeamAz = BeamAz S; % in degrees
   Scanrate = Scanrate S; % in rpm for range of 24 nmi or higher
   Pt = Pt S;
                % Tx peak power output
   Gt = Gt S;
              % Tx Antenna Gain is given as 31 dB
   Gr = Gr_S; % Rx Antenna Gain
f = f_S; % frequency
   lambda = lambda_S; % wavelength
   Ft = Ft S; % Prop Factor for Tx-Tgt path. See Advances in
Bistatic Radar - Chp 6.5.5.6
   Fr = Fr S;
                    % Prop Factor for Rx-Tgt path
   Ts = Ts S; % System Noise Temp of Rx of NF=6dB using formula
Te=(NF-1)290 and Ts=Te+290K
   Bn = Bn S; % Noise Bandwidth of ESM Rx Wideband Channalised (500
MHz)
   Lt = Lt S;
                    % Ref to See Advances in Bistatic Radar - Chp
6.5.5.6
   Lr = Lr S;
% this is calculation of pulse integration sqrt(N)
   PRF = PRF S;
   TOT = TOT S;
   N = N S;
else
   BeamEL = BeamEL X; % in degrees
   BeamAz = BeamAz X; % in degrees
   Scanrate = Scanrate X; % in rpm for range of 24 nmi or higher
   Pt = Pt X; % Tx peak power output
   Gt = Gt X; % Tx Antenna Gain is given as 31 dB
   Gr = Gr X;
                 % Rx Antenna Gain
   f = f X;
              % frequency
   lambda = lambda X; % wavelength
   Ft = Ft X;
                   % Prop Factor for Tx-Tgt path. See Advances in
Bistatic Radar - Chp 6.5.5.6
   Fr = Fr X;
                % Prop Factor for Rx-Tgt path
   Ts = Ts X; % System Noise Temp of Rx of NF=6dB using formula
Te=(NF-1)290 and Ts=Te+290K
   Bn = Bn X; % Noise Bandwidth of ESM Rx Wideband Channalised (500
MHz)
   Lt = Lt X;
                    % Ref to See Advances in Bistatic Radar - Chp
6.5.5.6
   Lr = Lr X;
% this is calculation of pulse integration sqrt(N)
   PRF = PRF X;
   TOT = TOT X;
   N = N X;
end
```

```
% We only take in three measurments from the ESM - L, Rt+Rr and thetaR
for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L1^2) / (2* (CombinR+L1*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L1^2-2*RrAct*L1*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L1^2) / (2*RrAct*RtAct)));
           % Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Beta1=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1+1,90+Beta1)/10);
           %RCS=10^ (biRCS3 (Beta1+1, Beta1+1) /10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3) * k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                         % Bi-static Radar Constant
           e=L1/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2 * RtAct^2);
           SNRdBm = 10*log(SNR) + sqrt(N);
           Xcoord = RrAct*sin((thetaR+Tx1)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx1)*pi/180);
                      % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tgt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamAz/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            \% Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1) *sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600));% Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            % the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err thetaR = Err factor theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord1 = [Xcoord1, Xcoord];
            Ycoord1 = [Ycoord1, Ycoord];
            SNR1 = [SNR1, SNR];
            SNRdBm1 = [SNRdBm1, SNRdBm];
            Err time1 = [Err time1,Err time];
            Err L1 = [Err L1,Err L];
            Err thetaR1 = [Err thetaR1,Err thetaR];
            else
            end
        end
    end
    Err Total1=zeros(size(Err time1));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time1)
    Err Total1(aa) =
max(abs(Err time1(aa)), abs(Err L1(aa))+abs(Err thetaR1(aa)));
 end
 for aa=1:1:length(Err time1)
    Err Total1(aa) = max(abs(Err Total1(aa)), abs(Err thetaR1(aa)));
 end
```

```
for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L2^2)/(2*(CombinR+L2*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L2^2-2*RrAct*L2*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L2^2)/(2*RrAct*RtAct)));
           % Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Betal=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1+1,90+Beta1)/10);
           %RCS=10^ (biRCS3 (Beta1+1, Beta1+1) /10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3)*k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                          % Bi-static Radar Constant
           e=L2/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2*RtAct^2);
           SNRdBm = 10*log(SNR) + sqrt(N);
           Xcoord = RrAct*sin((thetaR+Tx2)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx2)*pi/180);
           % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tgt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamAz/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 de
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            \% Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1) *sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            \% the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err_thetaR = Err_factor_theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord2 = [Xcoord2, Xcoord];
            Ycoord2 = [Ycoord2, Ycoord];
            SNR2 = [SNR2, SNR];
            SNRdBm2 = [SNRdBm2, SNRdBm];
            Err time2 = [Err time2,Err time];
            Err_L2 = [Err_L2,Err_L];
            Err thetaR2 = [Err thetaR2,Err thetaR];
            else
            end
        end
      end
   Err Total2=zeros(size(Err time2));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time2)
   Err Total2(aa) = max(abs(Err time2(aa)), abs(Err L2(aa)));
 end
 for aa=1:1:length(Err time2)
   Err Total2(aa) = max(abs(Err Total2(aa)), abs(Err thetaR2(aa)));
 end
```

```
for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L3^2)/(2*(CombinR+L3*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L3^2-2*RrAct*L3*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L3^2)/(2*RrAct*RtAct)));
           % Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Betal=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1+1,90+Beta1)/10);
           %RCS=10^ (biRCS3 (Beta1+1, Beta1+1) /10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3)*k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                          % Bi-static Radar Constant
           e=L3/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2*RtAct^2);
           SNRdBm = 10*log(SNR) + sqrt(N);
           Xcoord = RrAct*sin((thetaR+Tx3)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx3)*pi/180);
           % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tgt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamAz/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            \% Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1) *sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            % the error will be coming from the DF error from the ESM
system
            % which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err thetaR = Err factor theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord3 = [Xcoord3, Xcoord];
            Ycoord3 = [Ycoord3, Ycoord];
            SNR3 = [SNR3, SNR];
            SNRdBm3 = [SNRdBm3, SNRdBm];
            Err_time3 = [Err_time3,Err_time];
            Err L3 = [Err L3,Err L];
            Err thetaR3 = [Err thetaR3,Err thetaR];
            else
            end
        end
      end
    Err Total3=zeros(size(Err time3));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time3)
    Err Total3(aa) = max(abs(Err time3(aa)), abs(Err L3(aa)));
 end
 for aa=1:1:length(Err time3)
    Err Total3(aa) = max(abs(Err Total3(aa)), abs(Err thetaR3(aa)));
 end
```

```
for CombinR=18500:200:45000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L4^2)/(2*(CombinR+L4*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L4^2-2*RrAct*L4*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L4^2) / (2*RrAct*RtAct)));
            % Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Beta1=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1+1,90+Beta1)/10);
           %RCS=10^ (biRCS3 (Beta1+1, Beta1+1) /10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3)*k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                         % Bi-static Radar Constant
           e=L4/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2*RtAct^2);
           SNRdBm = 10*log(SNR) + sqrt(N);
           Xcoord = RrAct*sin((thetaR+Tx4)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx4)*pi/180);
           % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           \ensuremath{\$} beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tgt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamAz/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            \% Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1)*sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            % the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err thetaR = Err factor theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord4 = [Xcoord4, Xcoord];
            Ycoord4 = [Ycoord4, Ycoord];
            SNR4 = [SNR4, SNR];
            SNRdBm4 = [SNRdBm4, SNRdBm];
            Err time4 = [Err time4,Err time];
            Err L4 = [Err L4, Err L];
            Err thetaR4 = [Err thetaR4,Err thetaR];
            else
            end
        end
      end
   Err Total4=zeros(size(Err time4));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time4)
   Err Total4(aa) = max(abs(Err time4(aa)), abs(Err L4(aa)));
 end
 for aa=1:1:length(Err time4)
   Err Total4(aa) = max(abs(Err Total4(aa)), abs(Err thetaR4(aa)));
 end
```

```
for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L5^2)/(2*(CombinR+L5*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L5^2-2*RrAct*L5*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L5^2)/(2*RrAct*RtAct)));
           % Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Beta1=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1,90+Beta1)/10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3)*k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                         % Bi-static Radar Constant
           e=L5/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2*RtAct^2);
           SNRdBm = 10 * log(SNR) + 30;
           Xcoord = RrAct*sin((thetaR+Tx5)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx5)*pi/180);
           % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tgt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamEL/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            % Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1)*sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            \% the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            \% I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err_thetaR = Err_factor_theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord5 = [Xcoord5, Xcoord];
            Ycoord5 = [Ycoord5, Ycoord];
            SNR5 = [SNR5, SNR];
            SNRdBm5 = [SNRdBm5, SNRdBm];
            Err time5 = [Err time5,Err time];
            Err_L5 = [Err_L5, Err_L];
            Err thetaR5 = [Err thetaR5,Err thetaR];
            else
            end
        end
      end
   Err Total5=zeros(size(Err time5));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time5)
   Err Total5(aa) = max(abs(Err time5(aa)), abs(Err L5(aa)));
 end
 for aa=1:1:length(Err time5)
   Err Total5(aa) = max(abs(Err Total5(aa)), abs(Err thetaR5(aa)));
 end
```

```
8
  for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L6^2)/(2*(CombinR+L6*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L6^2-2*RrAct*L6*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L6^2) / (2*RrAct*RtAct)));
           \% Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Beta1=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1,90+Beta1)/10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3)*k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                          % Bi-static Radar Constant
           e=L6/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2*RtAct^2);
           SNRdBm = 10 \times \log(SNR) + 30;
           Xcoord = RrAct*sin((thetaR+Tx6)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx6)*pi/180);
           % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tqt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamEL/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            % Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1) *sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            % the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err thetaR = Err factor theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord6 = [Xcoord6, Xcoord];
            Ycoord6 = [Ycoord6, Ycoord];
            SNR6 = [SNR6, SNR];
            SNRdBm6 = [SNRdBm6, SNRdBm];
            Err time6 = [Err time6,Err time];
            Err L6 = [Err L6, Err L];
            Err thetaR6 = [Err thetaR6,Err thetaR];
            else
            end
             end
      end
   Err Total6=zeros(size(Err time6));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time6)
    Err Total6(aa) = max(abs(Err time6(aa)), abs(Err L6(aa)));
 end
 for aa=1:1:length(Err time6)
    Err Total6(aa) = max(abs(Err Total6(aa)), abs(Err thetaR6(aa)));
 end
```

```
for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L7^2)/(2*(CombinR+L7*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L7^2-2*RrAct*L7*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L7^2)/(2*RrAct*RtAct)));
           % Assuming the low RCS ship is heading south towards the Rx
           % position. Then, Beta/2 is incident angle on one side and
           % scattered angle on the other side.
           Beta1=round(Beta/2);
           % Organise the RCS values by
           RCS=10^ (biRCS3 (90-Beta1, 90+Beta1) /10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3) * k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                          % Bi-static Radar Constant
           e=L7/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2 * RtAct^2);
           SNRdBm = 10 * log(SNR) + 30;
           Xcoord = RrAct*sin((thetaR+Tx7)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx7)*pi/180);
           % Error to measurement of Rt + Rr
           % Antenna scan rate of 45 rpm or 270 deg/s
           % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tgt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err time=Err factor time*(BeamEL/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            \% Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1)*sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            % the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err thetaR = Err factor theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord7 = [Xcoord7, Xcoord];
            Ycoord7 = [Ycoord7, Ycoord];
            SNR7 = [SNR7, SNR];
            SNRdBm7 = [SNRdBm7, SNRdBm];
            Err time7 = [Err time7,Err time];
            Err L7 = [Err L7, Err L];
            Err thetaR7 = [Err thetaR7,Err thetaR];
            else
            end
        end
      end
    Err Total7=zeros(size(Err time7));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time7)
    Err Total7(aa) = max(abs(Err time7(aa)), abs(Err L7(aa)));
 end
 for aa=1:1:length(Err time7)
    Err Total7(aa) = max(abs(Err Total7(aa)), abs(Err thetaR7(aa)));
 end
```

```
for CombinR=18500:200:50000
       for thetaR=0:0.5:360
           RrAct=((CombinR)^2 -
L8^2)/(2*(CombinR+L8*sin(thetaR*pi/180)));
           RtAct=sqrt(RrAct^2+L8^2-2*RrAct*L8*cos(pi/2+thetaR*pi/180));
           Beta=round((180/pi)*acos((RrAct^2+RtAct^2-
L8^2) / (2*RrAct*RtAct)));
            % Assuming the low RCS ship is heading south towards the Rx
            % position. Then, Beta/2 is incident angle on one side and
           \% scattered angle on the other side.
           Beta1=round(Beta/2);
            % Organise the RCS values by
           RCS=10^ (biRCS3(90-Beta1,90+Beta1)/10);
           K1 = Pt*Gt*Gr*(lambda^2)*RCS*(Ft^2)*(Fr^2);
           K2 = ((4*pi)^3) * k*Ts*Bn*Lt*Lr;
           K = K1/K2;
                           % Bi-static Radar Constant
           e=L8/(RrAct+RtAct); % eccentricity
           SNR = K/(RrAct^2*RtAct^2);
           SNRdBm = 10 \times \log(SNR) + 30;
           Xcoord = RrAct*sin((thetaR+Tx8)*pi/180);
           Ycoord = RrAct*cos((thetaR+Tx8)*pi/180);
           % Error to measurement of Rt + Rr
            % Antenna scan rate of 45 rpm or 270 deg/s
            % With horizontal beamwidth of 1.8 deg, assuming that the
edge of the
           % beamwidth creates enough SNR to be received by the receiver
as
           % reflections from the tqt
           % From eqn 3.18
           % rms error
           Err factor time =
(1+e^2+2*e*sin(thetaR*pi/180))/(2*(1+e*sin(thetaR*pi/180))^2);
           Err_time=Err_factor_time*(BeamEL/2*pi/180)*RtAct; % r*theta
corresponding to half of the beamwidth of 1.8 deg
```

```
%Err time3 = (0.55/Bn)*1/sqrt(Bn*SNR1)*c; % Ref to Alogrithms for
Ambiguity Function Processing (Eqn - 11b)
            %Err time1 = Err time3 + Err time1;
            \% Error to measurement of L
            % For a Class A Marine AIS, positional acciracy is up to
0.0001 mins where
            % 1 deg equals to 60nmi
            % Update rate is between 2 to 10 seconds
            % Assumming a opening speed of Tx at 12 knots and Rx at 12
knots
            Err factor L = (-
(e^2+1) *sin(thetaR*pi/180)+2*e)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err L = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net
opening speed of 10 knots + 5s update rate
            % Error to measurement of thetaR
            % Error due to thetaR
            % the error will be coming from the DF error from the ESM
system
            \% which is up to +/-1.5 deg
            % I modified this equation as I have factor in the L in my
error so I
            % normalised it with L so as not to double count
            Err factor theta = (1 -
e^2) *cos(thetaR*pi/180)/(2*(1+e*sin(thetaR*pi/180))^2);
            Err thetaR = Err factor theta*(RtAct*1.5*pi/180);
            if SNRdBm >= sen
            Xcoord8 = [Xcoord8, Xcoord];
            Ycoord8 = [Ycoord8, Ycoord];
            SNR8 = [SNR8, SNR];
            SNRdBm8 = [SNRdBm8, SNRdBm];
            Err time8 = [Err time8,Err time];
            Err L8 = [Err L8, Err L];
            Err thetaR8 = [Err thetaR8,Err thetaR];
            else
            end
        end
      end
   Err Total8=zeros(size(Err time8));
 % Find th emax of the three source of error and use that for that
 % coordinate position
 for aa=1:1:length(Err time8)
    Err Total8(aa) = max(abs(Err time8(aa)), abs(Err L8(aa)));
 end
 for aa=1:1:length(Err time8)
    Err Total68(aa) = max(abs(Err Total8(aa)), abs(Err thetaR8(aa)));
 end
```

```
X=horzcat(Xcoord1,Xcoord2,Xcoord3,Xcoord4,Xcoord5,Xcoord6,Xcoord7,Xcoord8
);
Y=horzcat (Ycoord1, Ycoord2, Ycoord3, Ycoord4, Ycoord5, Ycoord6, Ycoord7, Ycoord8
);
Z=horzcat (SNRdBm1, SNRdBm2, SNRdBm3, SNRdBm4, SNRdBm5, SNRdBm6, SNRdBm7, SNRdBm8
);
XX=horzcat(Err thetaR1,Err thetaR2,Err thetaR3,Err thetaR4,Err thetaR5,Er
r thetaR6,Err thetaR7,Err thetaR8);
YY=horzcat(Err time1,Err time2,Err time3,Err time4,Err time5,Err time6,Er
r time7,Err time8);
ZZ=horzcat(Err L1,Err L2,Err L3,Err L4,Err L5,Err L6,Err L7,Err L8);
AA=horzcat(Err Total1,Err Total2,Err Total3,Err Total4,Err Total5,Err Tot
al6,Err Total7,Err Total8);
    figure(1)
    load seamount
    scatter(X,Y,4,Z,'fill','o')
    colorbar('EastOutside')
    %title('Detection Coverage in Polar Plot(4 Tx)')
    xlabel('Distance in m', 'fontsize',14)
    ylabel('Distance in m', 'fontsize',14)
    hold on
    plot(0,0,'k*')
    text(40,0,'Rx','FontSize',15)
    plot(-12730,12730,'k*')
    text(-12730,12930,'Tx1','FontSize',15)
    plot(12730,12730,'k*')
    text(12730,12930,'Tx2','FontSize',15)
    plot(12730,-12730,'k*')
    text(12730,-12930,'Tx3','FontSize',15)
    plot(-12730,-12730,'k*')
    text(-12730, -12930, 'Tx4', 'FontSize', 15)
    plot(0,-18000,'k*')
    text(200,-18000,'Tx5','FontSize',15)
    plot(0,18000,'k*')
    text(200,18000,'Tx6','FontSize',15)
    plot(18000,0,'k*')
    text(18000,200,'Tx7','FontSize',15)
    plot(-18000,0,'k*')
    text(-18000,200,'Tx8','FontSize',15)
    hold off
```

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APPENDIX G. MATLAB SOURCE CODE FOR UNCERTAINTY ELLIPSE

% Chong Sze Sing (March 2014) % 8 Tx Uncertainty Ellipse TargetX=4500; TargetY=4500; % Initializattion %For Tx 1 Tx1=45; L1=18000; %For Tx2 Tx2=135; L2=18000; %For Tx3 Tx3=225; L3=18000; %For Tx4 Tx4=315; L4=18000; %For Tx5 Tx5=0; L5=18000; %For Tx6 Tx6=90; L6=18000; %For Tx7 Tx7=179; L7=18000; %For Tx8 Tx8=270; L8=18000; % Variables % Parameters are based Furuno Marine Radar FAR-21x7 series in both Xband % and S-band % Bi-static Radar Equation % Constants k = 1.38e-23; % Boltzmann c = 3e8; % Speed of light

```
Select = 1; %chose which radar to use
%S-band Radar (1)
BeamAz S = 1.8;
                 % in degrees
                % in degrees
BeamEL S = 25;
Scanrate S = 45; % in rpm for range of 24 nmi or higher
Pt S = 25e3;
                % Tx peak power output
Gt S = 10<sup>(28/10)</sup>; % Tx Antenna Gain given as 28dB
                % Rx Antenna Gain
Gr S = 2;
f S = 3050e6; % frequency
lambda S = c/f S;
                    % wavelength
Ft S = 1;
              % Prop Factor for Tx-Tqt path. See Advances in Bistatic
Radar - Chp 6.5.5.6
                % Prop Factor for Rx-Tgt path
Fr S = 1;
Ts S = 290+290*(6-1); % System Noise Temp of Rx of NF=6dB using
formula Te=(NF-1)290 and Ts=Te+290K
                % Noise Bandwidth of ESM Rx Wideband Channalised (500
Bn S = 100e6;
MHz)
Lt S = 2;
                 % Ref to See Advances in Bistatic Radar - Chp 6.5.5.6
Lr S = 2;
% this is calculation of pulse integration sqrt(N)
PRF S = 2000;
TOT S = BeamEL S/(Scanrate S*60);
N S = TOT S*PRF S;
%X-band Radar (2)
BeamAz X = 0.95; % in degrees
BeamEL X = 25; % in degrees
Scanrate X = 45; % in rpm for range of 24 nmi or higher
Pt X = 25e3;
                   % Tx peak power output
Gt X = 10^(31/10); % Tx Antenna Gain is given as 31 dB
Gr X = 2;
            % Rx Antenna Gain
; % frequency
f X = 9410e6;
lambda X = c/f X;
                    % wavelength
                 % Prop Factor for Tx-Tgt path. See Advances in Bistatic
Ft X = 1;
Radar - Chp 6.5.5.6
Fr X = 1;
                 % Prop Factor for Rx-Tgt path
Ts X = 290+290*(6-1); % System Noise Temp of Rx of NF=6dB using
formula Te=(NF-1)290 and Ts=Te+290K
Bn X = 500e6;
                % Noise Bandwidth of ESM Rx Wideband Channalised (500
MHz)
                 % Ref to See Advances in Bistatic Radar - Chp 6.5.5.6
Lt X = 2;
Lr X = 2;
% this is calculation of pulse integration sqrt(N)
PRF X = 1500;
TOT X = BeamEL X/(Scanrate X*360);
N X = TOT X*PRF X;
```

```
if Select ==1
   BeamEL = BeamEL_S; % in degrees
   BeamAz = BeamAz S; % in degrees
   Scanrate = Scanrate S; % in rpm for range of 24 nmi or higher
   Pt = Pt S; % Tx peak power output
   Gt = Gt S; % Tx Antenna Gain is given as 31 dB
   Gr = Gr S;
                % Rx Antenna Gain
   f = f S; % frequency
    lambda = lambda S; % wavelength
   Ft = Ft S; % Prop Factor for Tx-Tgt path. See Advances in
Bistatic Radar - Chp 6.5.5.6
   Fr = Fr_S; % Prop Factor for Rx-Tgt path
Ts = Ts_S; % System Noise Temp of Rx of NF=6dB using formula
Te=(NF-1)290 and Ts=Te+290K
                 % Noise Bandwidth of ESM Rx Wideband Channalised (500
   Bn = Bn S;
MHz)
   Lt = Lt S;
                      % Ref to See Advances in Bistatic Radar - Chp
6.5.5.6
   Lr = Lr S;
% this is calculation of pulse integration sqrt(N)
   PRF = PRF S;
   TOT = TOT S;
   N = N S;
else
   BeamEL = BeamEL_X; % in degrees
BeamAz = BeamAz_X; % in degrees
   Scanrate = Scanrate X; % in rpm for range of 24 nmi or higher
                % Tx peak power output
   Pt = Pt X;
   Gt = Gt X; % Tx Antenna Gain is given as 31 dB
    Gr = Gr X;
                     % Rx Antenna Gain
    f = f X; % frequency
    lambda = lambda_X; % wavelength
   Ft = Ft X; % Prop Factor for Tx-Tgt path. See Advances in
Bistatic Radar - Chp 6.5.5.6
   Fr = Fr X;
                  % Prop Factor for Rx-Tgt path
   Ts = Ts X; % System Noise Temp of Rx of NF=6dB using formula
Te=(NF-1)290 and Ts=Te+290K
   Bn = Bn X; % Noise Bandwidth of ESM Rx Wideband Channalised (500
MHz)
                      % Ref to See Advances in Bistatic Radar - Chp
   Lt = Lt X;
6.5.5.6
   Lr = Lr X;
% this is calculation of pulse integration sqrt(N)
   PRF = PRF X;
   TOT = TOT X;
   N = N X;
end
```

```
Rr=sqrt(TargetX^2+TargetY^2);
theta=atan(TargetX/TargetY); % Angle made with True North into
clockwise direction
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L1^2+2*Rr*L1*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
e=L1/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L1^2)/(2*Rr*Rt)); % I am using beta to
add to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time1=Err factor time* (BeamAz/2*pi/180)*Rt; % r*theta corresponding
to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2+1})*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err L1 = Err factor L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^2) \cos(\text{theta})/(2*(1+e^{\sin(\text{theta})})^2);
Err thetaR1 = Err factor theta*(Rt*1.5*pi/180);
% Decide which one is the major and minor axis
if Err time1 > Err thetaR1
    Err Max1=Err time1;
   Err Min1=Err thetaR1;
   % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
        angle1=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2
        angle1=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle1=(-theta+pi/4);
    else
        angle1=pi/2+(-theta+pi/4);
    end
```

```
else
    Err Max1=Err thetaR1;
    Err Min1=Err time1;
     % this handles the four different quadrant and if err theta
dominates
    if theta>=0 && theta<=pi/4
        angle1=-theta;
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle1=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle1=-theta;
    else
        angle1=pi/2-theta;
    end
end
% Decide what angle to put for each Tx contribuion
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L2^2+2*Rr*L2*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
e=L2/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L2^2)/(2*Rr*Rt)); % I am using beta to
add to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time2=Err factor time* (BeamAz/2*pi/180)*Rt; % r*theta corresponding
to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2+1})*sin(theta)+2*e)/(2*(1+e*sin(theta))^2));
Err_L2 = Err_factor_L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^2) \cos(\text{theta}) / (2 \cdot (1+e \cdot \sin(\text{theta}))^2);
Err_thetaR2 = Err_factor_theta*(Rt*1.5*pi/180);
```

```
% Decide which one is the major and minor axis
if Err_time2 > Err_thetaR2
   Err Max2=Err time2;
    Err Min2=Err thetaR2;
    % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
        angle2=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle2=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle2=(-theta+pi/4);
    else
        angle2=pi/2+(-theta+pi/4);
    end
else
    Err Max2=Err thetaR2;
   Err Min2=Err time2;
    % this handles the four different quadrant and if err theta
dominates
   if theta>=0 && theta<=pi/4
        angle2=-theta;
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle2=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle2=-theta;
    else
        angle2=pi/2-theta;
    end
end
% this function will always draw the majoraxis on the horizontal plane
and
% minoraxis in vertical plane
```

```
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L3^2+2*Rr*L3*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
e=L3/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L3^2)/(2*Rr*Rt)); % I am using beta to
add to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time3=Err factor time*(BeamAz/2*pi/180)*Rt; % r*theta corresponding
to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2}+1)*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err_L3 = Err_factor_L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^2)*\cos(\text{theta})/(2*(1+e*\sin(\text{theta}))^2);
Err thetaR3 = Err factor theta* (Rt*1.5*pi/180);
% Decide which one is the major and minor axis
if Err time3 > Err thetaR3
    Err Max3=Err time3;
    Err Min3=Err thetaR3;
    % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
       angle3=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle3=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle3=(-theta+pi/4);
    else
        angle3=pi/2+(-theta+pi/4);
    end
```

```
else
    Err Max3=Err thetaR3;
    Err Min3=Err time3;
    % this handles the four different quadrant and if err theta
dominates
    if theta>=0 && theta<=pi/4
        angle3=-theta;
    elseif theta>pi/4 && theta<=pi/2
        angle3=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle3=-theta;
    else
        angle3=pi/2-theta;
    end
end
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L4^2+2*Rr*L4*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
e=L4/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L4^2)/(2*Rr*Rt)); % I am using beta to add
to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time4=Err factor time*(BeamAz/2*pi/180)*Rt; % r*theta corresponding
to \overline{h} alf of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2}+1)*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err L4 = Err factor L*((0.0001*1824)+(10*1824*5/3600)); % Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^2) \cos(\text{theta})/(2*(1+e^{\sin(\text{theta})})^2);
Err thetaR4 = Err factor theta*(Rt*1.5*pi/180);
```

```
\% Decide which one is the major and minor axis
if Err time4 > Err thetaR4
   Err Max4=Err time4;
    Err Min4=Err thetaR4;
    % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
       angle4=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2
        angle4=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle4=(-theta+pi/4);
    else
       angle4=pi/2+(-theta+pi/4);
    end
else
   Err Max4=Err thetaR4;
   Err Min4=Err time4;
    % this handles the four different quadrant and if err theta
dominates
    if theta>=0 && theta<=pi/4
        angle4=-theta;
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle4=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle4=-theta;
    else
       angle4=pi/2-theta;
    end
end
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L5^2+2*Rr*L5*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
```

```
e=L5/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L5^2)/(2*Rr*Rt)); % I am using beta to
add to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time5=Err factor time*(BeamAz/2*pi/180)*Rt; % r*theta
corresponding to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2}+1)*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err L5 = Err factor L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^{2}) \cos(theta) / (2 + (1+e \sin(theta))^{2});
Err thetaR5 = Err factor theta*(Rt*1.5*pi/180);
% Decide which one is the major and minor axis
if Err time5 > Err thetaR5
    Err Max5=Err time5;
    Err Min5=Err thetaR5;
    % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
        angle5=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2
        angle5=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle5=(-theta+pi/4);
    else
        angle5=pi/2+(-theta+pi/4);
    end
else
    Err Max5=Err thetaR5;
    Err Min5=Err time5;
    % this handles the four different quadrant and if err theta
dominates
    if theta>=0 && theta<=pi/4
        angle5=-theta;
    elseif theta>pi/4 && theta<=pi/2
        angle5=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle5=-theta;
    else
        angle5=pi/2-theta;
    end
end
```

```
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L6^2+2*Rr*L6*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
e=L6/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L6^2)/(2*Rr*Rt)); % I am using beta to
add to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time6=Err factor time*(BeamAz/2*pi/180)*Rt; % r*theta
corresponding to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2}+1)*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err L6 = Err factor L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^2) \cos(\text{theta})/(2*(1+e\sin(\text{theta}))^2);
Err thetaR6 = Err factor theta*(Rt*1.5*pi/180);
% Decide which one is the major and minor axis
if Err time6 > Err thetaR6
   Err Max6=Err time6;
    Err Min6=Err thetaR6;
   % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
       angle6=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle6=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle6=(-theta+pi/4);
    else
        angle6=pi/2+(-theta+pi/4);
    end
```
```
else
    Err Max6=Err thetaR6;
    Err Min6=Err time6;
    % this handles the four different quadrant and if err theta
dominates
   if theta>=0 && theta<=pi/4
        angle6=-theta;
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle6=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle6=-theta;
    else
        angle6=pi/2-theta;
    end
end
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L7^2+2*Rr*L7*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
e=L7/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L7^2)/(2*Rr*Rt)); % I am using beta to
add to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time7=Err factor time*(BeamAz/2*pi/180)*Rt; % r*theta
corresponding to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2+1})*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err L7 = Err factor L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^{2}) \cos(\text{theta}) / (2 + (1+e \sin(\text{theta}))^{2});
Err thetaR7 = Err factor theta*(Rt*1.5*pi/180);
```

```
% Decide which one is the major and minor axis
if Err_time7 > Err_thetaR7
   Err Max7=Err time7;
   Err Min7=Err thetaR7;
   % this handles the four different quadrant and if err time
dominates
   if theta>=0 && theta<=pi/4
        angle7=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2</pre>
        angle7=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
       angle7=(-theta+pi/4);
    else
       angle7=pi/2+(-theta+pi/4);
    end
else
   Err Max7=Err thetaR7;
    Err Min7=Err time7;
    % this handles the four different quadrant and if err theta
dominates
   if theta>=0 && theta<=pi/4
       angle7=-theta;
    elseif theta>pi/4 && theta<=pi/2
        angle7=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
       angle7=-theta;
    else
       angle7=pi/2-theta;
    end
end
% Eqn 5.1
syms a
syms A positive
A=solve(a^2-2*Rr*a-L8^2+2*Rr*L8*sin(theta));
RrRt=double(A);
Rt=abs(RrRt(1))-Rr;
```

```
e=L8/(Rr+Rt); % eccentricity
beta=(180/pi)*acos((Rr^2+Rt^2-L8^2)/(2*Rr*Rt)); % I am using beta to add
to the angle of
Err factor time = (1+e^2+2*e*sin(theta))/(2*(1+e*sin(theta)^2));
Err time8=Err factor time* (BeamAz/2*pi/180)*Rt; % r*theta corresponding
to half of the beamwidth of 1.8 deg
Err factor L = abs((-(e^{2}+1)*sin(theta)+2*e)/(2*(1+e*sin(theta))^{2}));
Err L8 = Err factor L*((0.0001*1824)+(10*1824*5/3600));% Net opening
speed of 10 knots + 5s update rate
Err factor theta = (1-e^2) \cos(\text{theta}) / (2 \cdot (1+e \cdot \sin(\text{theta}))^2);
Err thetaR8 = Err factor theta*(Rt*1.5*pi/180);
\% Decide which one is the major and minor axis
if Err time8 > Err thetaR8
   Err Max8=Err time8;
   Err Min8=Err thetaR8;
    % this handles the four different quadrant and if err time dominates
    if theta>=0 && theta<=pi/4
        angle8=(-theta-pi/4);
    elseif theta>pi/4 && theta<=pi/2
        angle8=pi/2+(-theta-pi/4);
    elseif theta<0 && theta>=-pi/4
        angle8=(-theta+pi/4);
    else
        angle8=pi/2+(-theta+pi/4);
    end
else
    Err Max8=Err thetaR8;
    Err Min8=Err time8;
    % this handles the four different quadrant and if err theta
dominates
   if theta>=0 && theta<=pi/4
        angle8=-theta;
    elseif theta>pi/4 && theta<=pi/2
        angle8=pi/2-theta;
    elseif theta<0 && theta>=-pi/4
        angle8=-theta;
    else
        angle8=pi/2-theta;
    end
end
```



```
pdeellip(TargetX, TargetY, Err_Max1, Err_Min1, angle1-(Tx1*pi/180), 'Tx1');
pdeellip(TargetX, TargetY, Err_Max2, Err_Min2, angle2-(Tx2*pi/180), 'Tx2');
pdeellip(TargetX, TargetY, Err_Max3, Err_Min3, angle3-(Tx3*pi/180), 'Tx3');
pdeellip(TargetX, TargetY, Err_Max4, Err_Min4, angle4-(Tx4*pi/180), 'Tx4');
pdeellip(TargetX, TargetY, Err_Max5, Err_Min5, angle5-(Tx5*pi/180), 'Tx5');
pdeellip(TargetX, TargetY, Err_Max6, Err_Min6, angle6-(Tx6*pi/180), 'Tx6');
pdeellip(TargetX, TargetY, Err_Max7, Err_Min7, angle7-(Tx7*pi/180), 'Tx7');
pdeellip(TargetX, TargetY, Err_Max8, Err_Min8, angle8-(Tx8*pi/180), 'Tx8');
```

```
hold on
axis square
grid on
plot(0,0,'k*')
text(40,0,'Rx','FontSize',14)
plot(-12730,12730,'k*')
text(-12730,12930,'Tx1','FontSize',15)
plot(12730,12730,'k*')
text(12730,12930,'Tx2','FontSize',15)
plot(12730,-12730,'k*')
text(12730,-12930,'Tx3','FontSize',15)
plot(-12730, -12730, 'k*')
text(-12730, -12930, 'Tx4', 'FontSize', 15)
plot(0,-18000,'k*')
text(200,-18000,'Tx5','FontSize',15)
plot(0,18000,'k*')
text(200,18000,'Tx6','FontSize',15)
plot(18000,0,'k*')
text(18000,200,'Tx7','FontSize',15)
plot(-18000,0,'k*')
text(-18000,200,'Tx8','FontSize',15)
plot(4500,4500,'k*')
text(4600,4500,'Target','FontSize',15)
hold off
```

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