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THESIS

THE EFFECT OF CONDENSATE INUNDATION ON STEAM CONDENSATION HEAT TRANSFER IN A TUBE BUNDLE

by

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June 1985

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SECURITY CLASSIFICATION OF THIS PAGE (When Data E	Intered)	
REPORT DOCUMENTATION F	PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (end Substitle) The Effect of Condensate Inundation on Steam Condensation Heat Transfer in a Tube Bundle		s. TYPE OF REPORT & PERIOD COVERED Master's Thesis June 1985
		6. PERFORMING ORG. REPORT NUMBER
7. Author(s) Steven K. Brower		8. CONTRACT OR GRANT NUMBER(*)
9. PERFORMING ORGANIZATION NAME AND ADDRESS Naval Postgraduate School Monterey, California 93943-	5100	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
Naval Postgraduate School Monterey, California 93943-5100		June 1985
		13. NUMBER OF PAGES
14. MONITORING AGENCY NAME & ADDRESS(II different	from Controlling Office)	Unclassified 15. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report)		

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17. DISTRIBUTION STATEMENT (of the ebetract entered in Block 20, if different from Report)

18. SUPPLEMENTARY NOTES

19. KEY WORDS (Continue on reverse side if necessary and identify by block number)

Steam condensation, heat-transfer coefficient, inundation, smooth, wire-wrapped tubes.

20. ABSTRACT (Continue on reverse side If necessary and identify by block number)

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#20 - ABSTRACT - (CONTINUED)

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The Effect of Condensate Inundation on Steam Condensation Heat Transfer in a Tube Bundle

by

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Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN MECHANICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL

June 1985

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NOMENCLATURE

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Outside heat-transfer area of one tube (m<sup>2</sup>)
A
         Mean Vapor flow area (m<sup>2</sup>)
A_{F}
         Inside heat-transfer area of one tube (m<sup>2</sup>)
Ai
В
         Coefficient defined in equation (2.8)
         Correction factor (\mu_{C}/\mu_{C})^{0.14}
C_{\mathsf{f}}
Cf,c
         Calculated correction factor
C_{i}
         Sieder-Tate coefficient
         Specific heat of water (kJ/kg·K)
Cpc
         Average flow dimension (m)
D_{\mathbf{F}}
D;
         Inner diameter of the tube
         Outer diameter of the tube
D
         Temperature difference (°C)
DT
         Dimensionless quantity defined in equation (2.8)
\mathbf{F}
         Acceleration of gravity (9.81 m/s<sup>2</sup>)
g
h<sub>fg</sub>
         Latent heat of vaporization (kJ/kg)
         Inside heat-transfer coefficient (W/m<sup>2</sup>K)
h;
         Local, outside heat-transfer coefficient for the Nth tube (W/m^2K)
h_N
         Heat-transfer<sub>2</sub>coefficient calculated from the Nusselt equation (W/m<sup>2</sup>K)
h
Nu
         Outside heat-transfer coefficient (W/m2K)
h
         Outside heat-transfer coefficient for the first
h<sub>1</sub>
         tube (W/m^2K)
         Thermal conductivity of the cooling water (W/mK)
k
         Thermal conductivity of the condensate film (W/mK)
kf
```

k_m Thermal conductivity of Titanium (W/mK) L Condensing length (m) Logarithmic Mean Temperature Difference (K) LMTD Mass flow rate of cooling water (kg/s) m Exponent defined in equation (2.8) n The number of tubes in a column or the tube number N of a given tube Water-side Nusselt number Nu $P_{T.}$ Longitudinal pitch of the tube bundle Pr Prandtl number Transverse pitch of the tube bundle $P_{\mathbf{T}}$ Heat flux based on the outside area (W/m²) q 0 Heat-transfer rate (W) Water-side Reynolds number Re Two-phase Reynolds number $(\rho_f V_y D_0/\mu_f)$ $Re_{2\phi}$ Wall thermal resistance based on the outside area Rw (m^2K/W) Inundation exponent defined in equation (5.1) S Average cooling water bulk temperature (°C) Th Tci Cooling water inlet temperature (°C) Cooling water outlet temperature (°C) T Tcon Condensate film temperature (°C) T_{ϵ} Average condensate film temperature (°C) Calculated condensate film temperature (°C) Tf.c Saturation temperature of steam (°C) Tsat Tvap Vapor temperature (°C) Tw Wall temperature (°C)

- U_O Outside heat-transfer coefficient (m²K/W)

 V_C Cooling water velocity (m/s)

 V_V Vapor velocity (m/s)
- Y Dimensionless quantity defined in equation (4.21)

GREEK SYMBOLS

μ _c	Dynamic viscosity of the cooling water (N·s/m ²)
$\mu_{ t f}$	Dynamic viscosity of the condensate film $(N \cdot s/m^2)$
$\mu_{\mathbf{w}}$	Dynamic viscosity of water (N·s/m ²)
ρ _c	Density of the cooling water (kg/m ³)
ρf	Density of the condensate film (kg/m ³)
ρ	Density of the vapor (kg/m^3)

I. INTRODUCTION

A. HISTORICAL BACKGROUND

Considerable interest has been generated in reducing the size and the weight of propulsion systems for naval applications. Advances in condenser design could do much to reduce the size and the weight of the propulsion plant. Measures to raise the condensing-side heat-transfer coefficient (of condenser tubes) is one way to achieve this reduction in condenser size. This reduction, however, comes at a price. This is usually due to an increase in the pumping power or due to an increase in the initial cost of the tubes. For naval applications, where the size of a vessel may depend upon the size of the condenser (in a submarine, for example), this reduction in the size is justified, even at the greater cost.

Search [Ref. 1], at the Naval Postgraduate School, investigated the present condenser design process, and examined the potential benefits that might occur if heat-transfer enhancement was established in the condenser. He concluded that reductions of as much as forty percent in the size and weight of condensers are possible. This is dependent, of course, on the heat-transfer-enhancement technique utilized. Much further research work at the Naval Postgraduate School has been directed toward these heat-transfer-enhancement techniques.

Beck [Ref. 2], Pence [Ref. 3], Reilly [Ref. 4], Fenner [Ref. 5], and Ciftci [Ref. 6] conducted research employing a single-tube test condenser. Their research concluded that the overall heat-transfer coefficient of enhanced tubes may be as much as twice that for smooth tubes of similr geometry. In a separate report, Marto, Reilly and Fenner [Ref. 7] reported that most of the increase in the overall heat-transfer coefficient was on the cooling-water side and was due to an increase of the turbulence and the swirl, as well as to an increase in the inside surface area. Only a small increase occurred on the steam side.

Present-day steam condensers utilizing smooth tubes are limited in their thermal efficiency, due primarily to the large thermal resistances occurring on the tube side of the condenser. It is possible, however, by utilizing enhanced tubes, to increase the inside heat-transfer coefficient by 100 percent or more. The corresponding increase in the outside heat-transfer coefficient, however, is less than 50 percent. In studying ways to further increase the outside heat-transfer coefficient, Webb [Ref. 8] reported that conduction across the condensate film is the primary thermal resistance in film condensation. This thermal resistance is usually larger than the thermal resistance of the tube wall, that attributed to fouling or that due to noncondensable gases. It is possible to reduce this thermal resistance by utilizing a geometry that reduces the film thickness. This

reduction of the thermal resistance would mean a corresponding increase in the outside heat-transfer coefficient.

For large tube bundles, condensate inundation is present and must be considered when attempting to increase the overall heat-transfer coefficient. Thomas [Ref. 9] wrapped wire around smooth tubes in a helical manner and tested the condensation of ammonia in a large tube bundle. Increases of as much as 200 percent in the oustide heat-transfer coefficient over that predicted by Nusselt [Ref. 10] were measured. This increase was attributed to the effect of surface tension drawing the condensate to the wire and acting as a condensate run-off channel. In a paper by Cunningham [Ref. 11], "roped" tubes were considered and again the increase in the outside heat-transfer coefficient was attributed to condensate drainage, while an increase in the inside heat-transfer coefficient was attributed to the increase in the inside convective coefficient, due to the increased turbulence. Kanakis [Ref. 12] tested both smooth and roped tubes, with and without a wire wrap in an in-line tube bundle simulating up to 30 tubes. Adding the wire wrap on the smooth tube increased the average, outside heattransfer coefficient for the 30 tube bundle by 50 percent, while adding the wire warp to the roped tube increased the average outside heat-transfer coefficient for the 30 tube bundle by more than 35 percent.

For condenser tubes, the increase in the outside heattransfer coefficient may also be accomplished by promoting

dropwise condensation. Tanasawa [Ref. 13] reviewed dropwise condensation and discussed the methods for promoting these dropwise conditions. He concluded that the use of organic polymers (such as Teflon) was the most promising of all the dropwise-promoting techniques. Investigations of organic coatings by Brown [Ref. 14] have shown that enhancements of up to 180 percent are possible. Of primary concern, however, is that these coatings have very low conductivities and must therefore be ultra-thin. Another concern is that these coatings must be strongly adherent and sufficiently tough to withstand the conditions of their assembly and their use. Holden [Ref. 15] investigated numerous dropwise-promoting coatings. His tests were based not only on the ability of the coating to promote dropwise condensation, but also on the ability of the coating to sustain this dropwise performance over an extended period of time. In addition, the coatings that were able to sustain this dropwise performance were evaluated for their heat-transfer performance. His results indicated that the outside heat-transfer coefficient for a single tube could be increased by a factor of from five to eight through the use of polymer coatings.

Since the use of a wire wrap can increase the heattransfer performance of a tube bundle, an important question arises: Is there some optimal wire diameter and wire pitch combination? In addition, during dropwise conditions what is the effect of condensate inundation upon heat transfer in a tube bundle?

B. OBJECTIVES

The objectives of this thesis were therefore to:

- 1. Conduct steam-condensation tests to determine the steam-side heat-transfer coefficient for 16-mm-o.d. smooth titanium tubes,
- 2. Confirm the heat-transfer-performance measurements of Kanakis [Ref. 12] on 16-mm-o.d. smooth titanium tubes with a 1.6-mm-o.d. titanium wire wrapped at a pitch of 7.6 mm,
- 3. Conduct heat-transfer-performance measurements to determine the effect of wire pitch and wire diameter on the steam-side heat-transfer coefficient with inundation, and
- 4. Conduct heat-transfer-performance measurements to determine the effect of inundation on a dropwise coated tube and compare this performance with the performance of a smooth tube.

II. THEORETICAL AND EMPIRICAL BACKGROUND

The basis for the analysis of film condensation on a horizontal tube was set forth by Nusselt in 1916. His analysis was, however, for laminar film condensation on a single horizontal tube. Nusselt's analysis yielded the well-known relationship for the heat-transfer coefficient:

$$h_{Nu} = 0.725 \left[\frac{k_f^3 \rho_f^2 h_{fg} g}{D_0 \mu_f (T_{sat}^{-T} w)} \right]^{1/4}$$
 (2.1)

This relationship is subject to the following restrictions, as stated by Nobbs [Ref. 16]:

- 1. The wall temperature is constant,
- 2. The flow is laminar in the condensate film,
- 3. The film thickness is small compared to normal tube diameters,
- 4. All fluid properties are constant within the condensate film,
- 5. Heat transfer in the film is by conduction,
- 6. All forces due to hydrostatic pressure, surface tension, inertia, and vapor/liquid interfacial shear are negligible when compared to viscous and gravitational forces, and
- 7. The vapor/liquid interface and the surrounding steam are at the saturation temperature.

Jakob [Ref. 17] extended the Nusselt analysis to film condensation on a vertical in-line column of horizontal tubes by assuming that all the condensate from a tube drains

as a laminar sheet onto the tube below it. This is depicted in Figure 2.1a. In this idealized situation, the average coefficient for a vertical column of N tubes was predicted to be:

$$\overline{h}_{N} = 0.725 \left[\frac{k_{f}^{3} \rho_{f}^{2} h_{fg} g}{D_{o}^{N} \mu_{f} (T_{sat}^{-T}w)} \right]^{1/4}$$
 (2.2)

Combining equations (2.1) and (2.2) yields the Nusselt idealized theory for the average coefficient compared to the coefficient of the top tube:

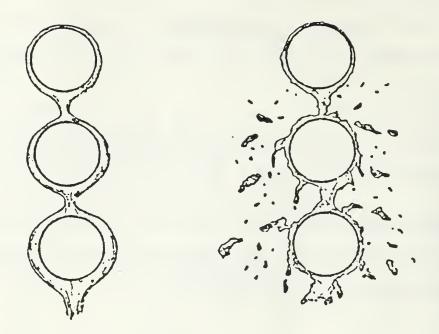
$$\overline{h_N}/h_1 = N^{-1/4} \tag{2.3}$$

In terms of the local heat-transfer coefficient for the Nth tube, this result becomes:

$$h_N/h_1 = N^{3/4} - (N-1)^{3/4}$$
 (2.4)

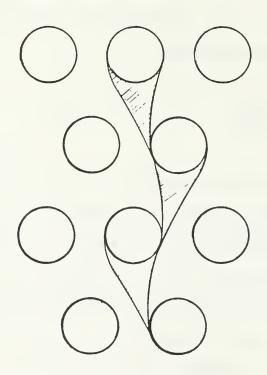
Realizing that condensate does not flow in a laminar sheet but by discrete droplets, Kern [Ref. 18] proposed a less-conservative relationship to account for the ripples and turbulence introduced into the condensate film. This is depicted in Figure 2.1b. The Kern relationship is:

$$\overline{h_N}/h_1 = N^{-1/6} \tag{2.5}$$



a. Nusselt's Idealized Laminar Flow Model

b. Kern's More Realistic Flow Model



c. Eissenberg's Side-Drainage Flow Model

Figure 2.1 Representations of Condensate Flow

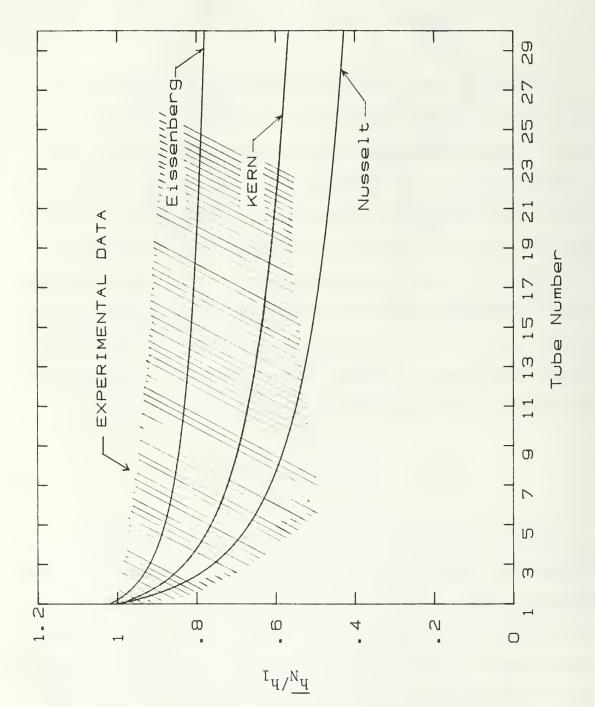
or in terms of the local heat-transfer coefficient for the Nth tube:

$$h_N/h_1 = N^{5/6} - (N-1)^{5/6}$$
 (2.6)

Eissenberg [Ref. 19] did extensive experimentation to investigate the effects of steam velocity, condensate inundation and noncondensable gases on the condensation heattransfer coefficient. He theorized that condensate does not always drain only onto tubes aligned vertically, but can be diverted sideways, especially in staggered tube bundles. The condensate draining sideways strikes the lower tubes on their sides rather than on their tops. This is depicted in Figure 2.1c. Since more heat is transferred from the top of a tube than from its bottom, the net effect of inundation is less severe. He predicted the following relationship:

$$\overline{h_N}/h_1 = 0.60 + 0.42 \text{ N}^{-1/4}$$
 (2.7)

Extensive experimental research into the effect of condensate inundation has been conducted. However, the data exhibit a substantial amount of scatter as shown in Figure 2.2 by the cross-hatched area. Berman [Ref. 20] conducted a compilation of film condensation data on bundles of horizontal tubes and identified the following variables as important in causing the scatter:



Theories and Data for Condensate Inundation Figure 2.2

- 1. Bundle geometry (in-line or staggered),
- 2. Tube spacing,
- 3. Type of condensing fluid,
- 4. Operating pressure,
- 5. Heat flux, and
- 6. Local vapor velocity.

Marto and Wanniarachchi [Ref. 21] noted that other factors can have an effect on the data such as noncondensable gases, the direction of the vapor flow, partial dropwise conditions, an insufficient abount of steam reaching the lower tubes in the bundle and difficulty in the measurement of the local condensing coefficient.

Very small amounts of noncondensable gases can result in significant reductions in the condensation heat-transfer rate. Summaries of the phenomenon have been made by Chisholm [Ref. 22], and Webb and Wanniarchchi [Ref. 23]. The consensus among them is that noncondensable gases impose an added thermal resistance, since the vapor molecules must diffuse through a gas layer prior to reaching the condensing surface. These noncondensable gases can also lead to regions where the tubes are inoperative in a condensing role. Noncondensable gases have an adverse effect on the condenser performance, and must be taken into account when designing a tube bundle.

The motion of vapor within a condenser affects the film condensation process because of its effect on the surface

shear between the vapor and the film, and the resulting effect on vapor separation. Although the results may be different depending upon the orientation of the steam flow, the general effect of an increase in vapor velocity is a corresponding increase in the condensing heat-transfer coefficient. There exist a number of both theoretical and empirical equations representing vapor-shear effects on the condensate heat-transfer coefficient [Refs. 24, 25 and 26].

It is clear that the measurements that were made during this thesis had both inundation and vapor-shear effects.

Marto [Ref. 27] stated that these two effects are difficult to separate from one another. During this study, however, an attempt was made to separate these two effects. For this purpose, data were taken on the top tube of the bundle with vapor velocity as a variable, and the data were correlated using an expression similar to a correlation suggested by Fujii [Ref. 28].

$$Nu/Re_{2\phi}^{1/2} = BF^{n}$$
 (2.8)

This correlation is based on numerous data for both staggered and in-line tube bundles. To represent the test-condenser tube bundle, the constants B and n in equation (2.8) were computed using a least-squares technique.

III. EXPERIMENTAL APPARATUS

The test facility was designed and built originally by Morrison [Ref. 29], and modified by Noftz [Ref. 30] and Kanakis [Ref. 12] to simulate an active tube column of up to 30 in-line tubes. A detailed description of all the components is contained in Ref. 30 and further modifications are described in Ref. 12. The descriptions in this report are brief, with particular focus given to the modifications undertaken by the author.

A. STEAM SUPPLY

The steam supply system is shown in Figure 3.1. House steam flows through a supply valve (MS-3), into a steam separator, and is throttled down by another valve (MS-4) to operating conditions. The steam then passes through an orifice and into the test condenser diffuser before entering the test condenser where it flows through a simulated in-line tube bundle.

B. TEST CONDENSER

The dimensions of the test condenser shown in Figure 3.2 were unchanged from Noftz's original design. The condenser consisted of five active tubes, twelve dummy tubes and one perforated tube. The tubes were positioned in the test condenser with a pitch-to-diameter ratio of 1.5. The active

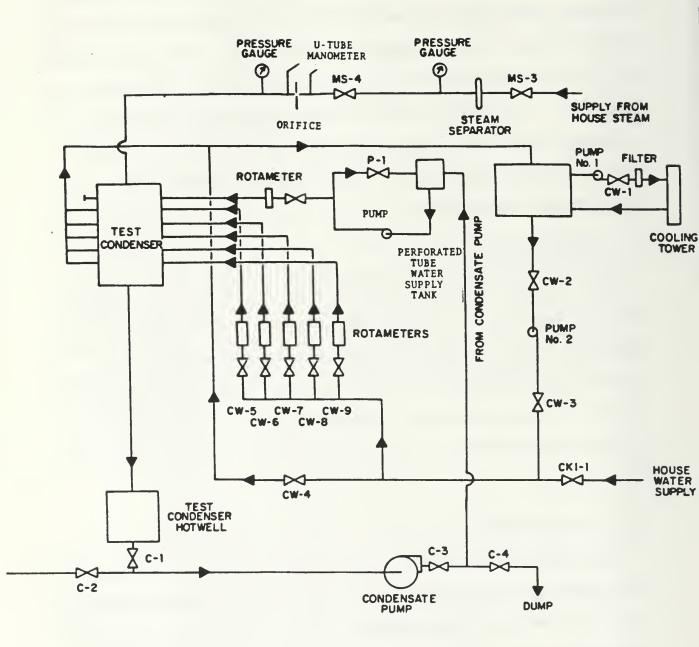
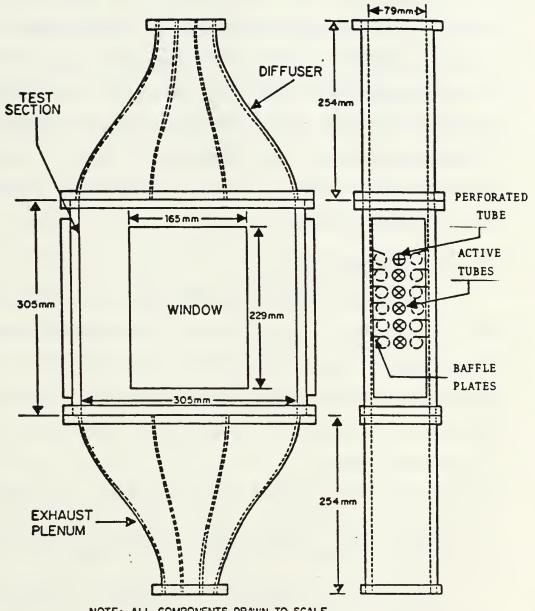


Figure 3.1 Schematic Diagram of Test Apparatus



NOTE: ALL COMPONENTS DRAWN TO SCALE

Figure 3.2 Sketch of Test Condenser

condensing length of each tube was 305 mm. A double-walled glass viewing window was provided on the front of the test condenser to allow for visual observation of the condensation process. Although the viewing window was designed to have heated air flowing between the glass walls, this feature was not utilized since no fogging of the glass was observed. A modification of the test condenser was made to minimize the steam flow along the test condenser side walls. The modification consisted of the addition of twelve baffle plates, placed as illustrated in Figure 3.2.

C. TEST-CONDENSER TUBES

Ten different types of active tubes were tested in this set of experiments. The tubes were manufactured by Wolverine Division of Universal Oil Products, and were made of titanium. All were of 16 mm o.d. and had a minimum wall thickness of 0.89 mm. The ten types of tubes are as listed below:

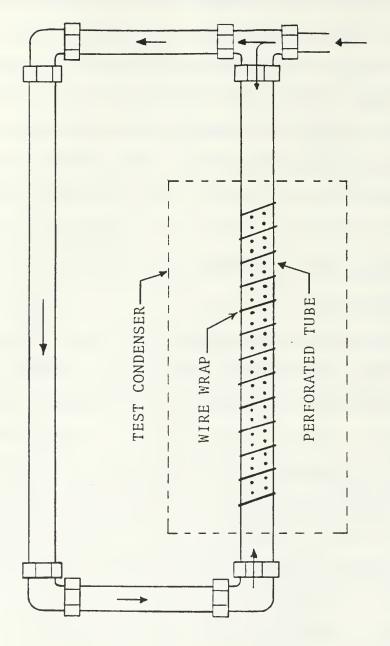
- 1. Smooth tubes,
- Smooth tubes wrapped with 1.6-mm-o.d. titanium wire at three different pitches (16 mm, 7.6 mm and 4 mm),
- 3. Smooth tubes wrapped with 1.0-mm-o.d. titanium wire at three different pitches (8 mm, 6 mm and 4 mm),
- 4. Smooth tubes wrapped with 0.5-mm-o.d. titanium wire at two different pitches (4 mm and 2 mm), and
- 5. Smooth tubes with a dropwise coating.

The actual wrapping of the wire around the smooth tubes was performed by the author of this thesis. A detailed description of the procedure is included in Chapter III.

The dropwise-coated tubes were sent to General Magnaplate Corporation where a dropwise coating, commercially referred to as "Nedox," was applied. First an electrodeposited nickel-cobalt substrate was applied to the tube surface, and then this substrate was infused with a microthin Teflon layer. A heat-treating cycle was employed to ensure thorough infusion of the substrate with the Teflon. This coating is highly non-wetting, has a low coefficient of friction and is heat-resistant. The specifics of the "Nedox" process are proprietory, but the manufacturer claims control of the surface thickness to 0.0025 mm. For this thesis the requested surface thickness was 0.0075 mm.

D. PERFORATED TUBE

Each set of five active tubes had its own corresponding perforated tube. This perforated tube was located above the uppermost active tube; see Figure 3.2. Figure 3.3 shows a typical pattern for the perforations of a tube, and is an illustration of the supply-water tubing at the test condenser. This particular arrangement of the tubing was used in an attempt to provide an even flow from the entire length of the perforated tube. A heated tube was used to provide the water supply for the perforated tube. A separate rotameter was provided in order to control the rate of flow of condensate through the perforated tube.



Perforated-tube water supply flow shown by arrows

Figure 3.3 Plan View of a Perforated Tube

E. CONDENSATE SYSTEM

The condensate system remained unchanged from Kanakis' modifications. Figure 3.1 shows the system as designed by Noftz and as modified by Kanakis. As steam condenses in the test condenser, it runs into and collects in the test condenser hotwell. The hotwell is a calibrated cylinder equipped with a sight glass, providing a visual measure of the condensate being produced. Measurement of the condensate collection rate provided the flow rate to be supplied to the perforated tube.

F. COOLING WATER SYSTEM

Although the cooling water system remained unchanged from Noftz original design, the five rotameters were recalibrated after Kanakis' data were taken. Each active tube had its own rotameter and regulating valve, and it was possible to control the coolant velocity through the active tubes up to values of 5 m/s. All data for this thesis, however, were taken at a constant coolant velocity of 1.56 m/s. The inlet temperature of the cooling water was maintained at a nearly constant temperature by the use of a large supply tank and a separate cooling tower to dissipate excess heat to the atmosphere. In order to provide a bulk temperature at the outlet of each active tube, a mixing chamber was installed just upstream of the temperature monitoring point.

G. INSTRUMENTATION

1. Flow Rates

Rotameters were used to measure the flow rate of the cooling water to each active tube and to the perforated tube.

2. Temperatures

Stainless-steel-sheathed copper-constantan thermocouples were used as the primary temperature monitoring devices. Table 1 lists the locations monitored by each of the thermocouples. Calibration, as performed by Kanakis, was utilized for all data taken.

The perforated-tube water supply was not maintainable at a constant temperature, nor was it quickly adjustable to a desired temperature. The system, as designed, contained a Gulton Industries, West 20 temperature controller. Although the manufacturer's accuracy was listed as 1.25 degrees F, the system response was so poor that the temperature of the perforated-tube supply water varied by as much as 10 degrees C from the beginning of the data run to the completion of the data run.

3. Pressure

A Bourdon-tube pressure gauge was used to measure the house steam supply pressure, while a compound pressure gauge was used to measure the steam pressure downstream of the orifice. A pressure transducer was used to measure the pressure in the test condenser, and a U-tube mercury manometer was fitted to measure the pressure drop across the orifice.

TABLE 1
Thermocouple Monitoring Locations

LOCATION	CHANNEL
T _{ci} #1	000
T _{ci} #3	001
T _{ci} #5	002
T _{co} #1	003
T _{co} #1	004
T _{co} #1	005
T _{co} #2	006
T _{CO} #2	007
T _{co} #3	008
T _{co} #3	009
T _{CO} #4	010
T _{CO} #4	011
T _{CO} #4	012
T _{co} #5	013
T _{co} #5	014
T _{sat}	015
T _{sat}	016
Tcon	017
T _{vap}	018

4. Data Collection and Display

A Hewlett-Packard model 3054A Automatic, DataAcquisition System, with a HP 2671G printer was used to
record and display the thermocouple and transducer readings.
The pressure transducer was assigned to channel 19 of the
data-acquisition system, with the thermocouples assigned to
the channels as indicated in Table 1.

IV. EXPERIMENTAL PROCEDURES

A. CALIBRATION

1. Rotameters

Calibration was performed on the five rotameters that supplied the cooling water to the active tubes. This calibration was performed using a weighing tank and a stop-watch, and a least-squares-fit calibration line was generated for each of the rotameters.

2. Steam Orifice

To calibrate the steam orifice the following steps were taken:

- 1. Set a steam flow rate,
- 2. Adjust the cooling water flow rate to condense all the incoming steam,
- 3. Measure the pressure drop across the orifice,
- 4. Measure the condensate collection rate,
- 5. Repeat steps 1 through 4 for different steam flow rates, and
- 6. Develop a least-squares-fit straight line for the steam mass flow rate vs. pressure drop.

B. PREPARATION OF CONDENSER TUBES

Prior to the winding of the titanium wire, the outside surface of the tubes was buffed using steel wool unitl all evidence of surface oxidation was removed. If any oxidation was evident on the inner surface of the tube, a test-tube

brush was used to apply a 50-percent sulfuric-acid solution to the inner surface of the tube. The tube was then thoroughly rinsed using tap water.

The procedure followed in the wrapping of the wire onto the smooth tubes was the same for every wire diameter and pitch combination tested. The wire was first welded to the tube and then, under a constant tension of 22 Newtons, the wire was guided (using a metal plate to control the spacing between the wire wraps) to the proper pitch as the tube was manually turned. Upon completion of the wrapping, the free end of the wire was then welded to the tube to yield a helically-wrapped surface 305 mm in length. After the wire wrapping of the tubes was completed and just prior to the installation of the tubes in the test condenser, the tubes were cleaned inside and out using a brush and a biodegradable detergent. A thorough rinsing was agin undertaken, and the tubes were checked to ensure that they displayed proper wetting characteristics. If any non-wetting areas were evident, a 50-percent solution of sodium hydroxide mixed with an equal amount of ethyl alcohol and heated to about 80 degrees C, was brushed onto the tube surface. After rinsing with tap water, the tube was ready for installation in the test condenser. Upon completion of the installation of the tubes in the test condenser, steam was introduced into the test condenser. Again the tubes were visually examined. If there were any areas that displayed dropwise condensation, the tubes were

washed from above using the perforated-tube water supply. This was sufficient in all cases to restore filmwise condensation.

C. SYSTEM OPERATION

The system was operated in accordance with the operating instructions outlined in Appendix A of Ref. 17. The system was deemed to be operating at a steady-state condition when the cooling water inlet temperature was steady. The cooling water inlet temperature was monitored using the HP-3054A Automatic Data Acquisition System, and was considered steady when two consecutive readings (at five-minute intervals) varied by less than 0.1 degrees C.

One hour was usually required from system start-up to the steady-state condition. Upon reaching the steady-state condition, a data run was begun. Any changes in the cooling water flow or changes in the perforated-tube water flow were accompanied by a five-minute wait for system restabilization, prior to taking any further data. The duration of each data set was approximately one minute, and a completed data run consisted of 30 separate data sets. During each data set, the condensation collection rate was measured. For each inundation condition, five consecutive data sets were taken and the average values were computed. After each set of five consecutive data sets, the average condensate collection rate was used to determine the rate of flow of the water into the perforated tube which was to be used for the next set of

data (the average amount of condensate collected during a data set was introduced into the perforated tube for the next inundation condition). For example, to simulate tubes 6 through 10, the perforated-tube water flow rate was set equal to the condensate collection rate for tubes 1 through 5. A data run was considered completed upon the simulation of inundation conditions through the 30th tube.

Thermocouple readings and pressure-transducer readings were taken automatically by the data-acquisition system, while the settings of the rotameters and the test condenser initial and final hotwell levels were entered into the computer using the keyboard. Initial and final hotwell levels were taken for a time interval of one minute.

The following conditions were the operating conditions for all data taken for this thesis:

- 1. The coolant velocity was 1.55 \pm 0.05 m/s,
- 2. The coolant inlet temperature was 24 degrees C. (± 3 degrees C. depending on the day of the run, but constant during a given run), and
- 3. The saturation temperature was 100.5 ± 0.5 degrees C.

D. DATA-REDUCTION PROGRAM

A computer program was utilized to process and plot all raw data. The program was written in BASIC language and was run on an HP-9826 computer. A listing of this program is included as Appendix C.

E. HEAT-TRANSFER-COEFFICIENT CALCULATION

1. Inside Heat-Transfer Coefficient

The inside heat-transfer coefficient was computed using the Sieder-Tate equation described in Holman [Ref. 31]:

Nu =
$$h_i D_i/k_c = C_i Re^{0.8} Pr^{1/3} (\mu_c/\mu_w)^{0.14}$$
 (4.1)

where the coefficient C_i was computed using a modified Wilson plot [Ref. 12]. Using this technique, Kanakis [Ref. 12] obtained a Sieder-Tate coefficient of 0.029 ± 0.001. The method Kanakis used, however, did not include the variation of the condensate film properties as a function of heat flux. A slightly different, modified Wilson plot, as described by Nobbs [Ref. 16], that does account for this variation in the film properties, was used in this thesis. Appendix D gives a description of this method. Reprocessing Kanakis' data with this method, the Sieder-Tate coefficient was determined to be 0.028 ± 0.001.

2. Outside Heat-Transfer Coefficient

The heat-transfer rate to the cooling water can be computed using an energy balance, and can be expressed in terms of an overall heat-transfer coefficient by:

$$Q = \underset{C}{\overset{\bullet}{m}} C \underset{C}{\overset{\bullet}{p}} C (T_{CO} - T_{Ci}) = U_{O} \underset{O}{\overset{\bullet}{A}} LMTD$$
 (4.2)

where

$$LMTD = \frac{\frac{T_{co} - T_{ci}}{\ln((T_{sat} - T_{ci})/(T_{sat} - T_{co}))}}{(4.2a)}$$

Solving for the overall heat-transfer coefficient gives:

$$U_{o} = \stackrel{\circ}{m}_{c} \stackrel{\circ}{c}_{pc} / A_{o} \ln ((T_{sat} - T_{ci}) / (T_{sat} - T_{co}))$$
 (4.3)

The overall thermal resistance can be written as the sum of the internal, wall and external resistances:

$$\frac{1}{U_0 A_0} = \frac{1}{h_i A_i} + \frac{R_w}{A_0} + \frac{1}{h_0 A_0}$$
 (4.4)

where

$$R_{W} = D_{O} \ln(D_{O}/D_{i})/2 k_{m}$$
 (4.4a)

or from this result, the outside heat-transfer coefficient may be written:

$$h_{o} = \frac{1}{1/U_{o} - D_{o}/D_{i} h_{i} - R_{w}}$$
 (4.4b)

F. DATA-REDUCTION TECHNIQUE FOR THE OUTSIDE HEAT-TRANSFER COEFFICIENT

A listing of the computer program used in the calculation of the outside heat-transfer coefficient can be found in Appendix C. The steps required in the technique are outlined below:

1. Calculate the average bulk cooling water temperature:

$$T_b = (T_{CO} + T_{Ci})/2$$
 (4.5)

2. Calculate the cooling water velocity:

$$V_{C} = \dot{m}_{C}/A_{i} \rho_{C} \tag{4.6}$$

3. Calculate the cooling water Reynolds number:

$$Re = \rho_C V_C D_i / \mu_C \qquad (4.7)$$

- Calculate the heat transferred to the cooling water using the left-hand side of equation (4.2).
- 5. Calculate the heat flux:

$$q = Q/(\pi D_{Q} L)$$
 (4.8)

- 6. Since the film temperature was not a known quantity, an iterative scheme was employed to calcualte the film temperature, as indicated below:
 - a) Assume a film temperature (say $T_f = T_{sat}$),
 - b) Calculate the Nusselt coefficient,

$$h_{Nu} = 0.651 \left[\frac{k_f^3 \rho_f^2 h_{fg} g}{\mu_f D_o q} \right]^{1/3}$$
 (4.9)

c) Evaluate T_{f,c}, and

$$T_{f,C} = T_{sat} - \frac{q}{2 h_{Nu}}$$
 (4.10)

- d) Repeat steps a) through c) until $T_{f,c}$ is approximately equal to T_{f} .
- 7. Calculate the inside heat-transfer coefficient:

$$h_i = k_c C_i Re^{0.8} Pr^{1/3} C_f/D_i$$
 (4.11)

where
$$C_f$$
 is the correction factor $(\mu_C/\mu_W)^{0.14}$. (4.11a)

- 8. Since the correction factor is evaluated at the inner wall temperature, an iterative scheme was employed to calculate the wall temperature as indicated below:
 - a) Assume a value for C_{f} (e.g., 1.0),
 - b) Calculate h; using equation (4.11),
 - c) Calculate a water-side temperature drop,

$$DT = q D_{i}/h_{i} D_{O}$$
 (4.12)

d) Calculate a new correction factor, and

$$C_{f,C} = (\mu_C/\mu_W)^{0.14}$$
 (4.13)

- e) Repeat steps a) through d) until C_f,c is approximately equal to C_f.
- 9. Calculate the log-mean-temperature difference using equation (4.2a).
- 10. Calculate the overall heat-transfer coefficient:

$$U_{O} = q/LMTD \tag{4.14}$$

- 11. Calculate the outside heat-transfer coefficient using equation (4.4b).
- 12. Calculate the normalized, local, outside heat-transfer coefficient.

$$h_N/h_1$$

13. Calculate the normalized, average, outside heattransfer coefficient.

$$\overline{h_N}/h_1$$

NOTE: In the above formulation, the thermal resistances due to noncondensable gases and any fouling were assumed to be negligible.

- G. DATA-REDUCTION TECHNIQUE FOR THE VAPOR-SHEAR CORRELATION

 For the calculation of the vapor-shear correlation,

 measurements on only the first active tube in the tube

 bundle were taken.
 - 1. Calculate the inside heat-transfer coefficient as outlined above.
 - Calculate the outside heat-transfer coefficient as outlined above.
 - 3. Calculate a temperature correction:

$$DT = q/h \tag{4.15}$$

4. Assume a film temperature:

$$T_{f} = T_{sat} - DT/2 \tag{4.16}$$

5. Calculate the steam velocity:

$$V_{V} = m_{V} V_{V}/A_{f}$$
 (4.17)

where

$$A_{F} = 2 L(P_{T} P_{L} - \pi D_{O}^{2}/4)/P_{L}$$
 (4.17a)

 ${\rm A_F}$ is the Mean Vapor Flow Area defined by Nobbs [Ref. 16] and depicted in Figure 4.1.

6. Calculate the two-phase Reynolds number:

$$Re_{2\phi} = \rho_f V_v D_O/\mu_f \qquad (4.18)$$

7. Calculate the dimensionless quantity F:

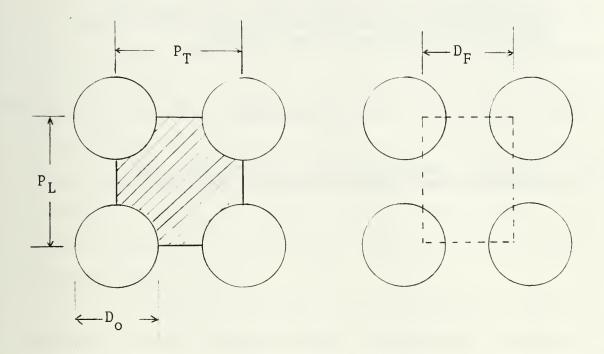
$$F = D_o \mu_f h_{fg} g/V_v^2 k_f DT$$
 (4.19)

8. Calculate the Nusselt number:

$$Nu = h_0 D_0/k_f \tag{4.20}$$

9. Calculate Y:

$$Y = Nu/Re_{2\phi}^{1/2}$$
 (4.21)



To calculate the mean flow area the following steps are required:

- 1) Calculate the cross-hatched area above, $AREA = P_{T} P_{L} - \pi D_{o}^{2} / 4$
- 2) Divide by longitudinal pitch to obtain an average flow dimension ${\rm D_F} = {\rm AREA} \; / \; {\rm P_L}$
- Multiply by the condenser length and the number of flow paths (2 flow paths in the case of this thesis).

$$A_F = 2 L (P_T P_L - \pi D_0^2 / 4) / P_L$$

Figure 4.1 Mean Vapor Flow Area

- 10. Repeat steps 1 through 9 at different steam mass flow rates into the test condenser. For this thesis 10 different steam mass flow rates were considered, and three measurements taken at each mass flow rate.
- 11. Starting with an assumption that the data would be of a form similar to that postulated by Fujii [Ref. 28], an expression was assumed:

$$Y = B F^{n}$$
; F as defined in equation (4.19) (4.22)

12. Perform a least-squares-fit of the data and determine the constant B and the exponent n.

V. RESULTS AND DISCUSSION

A. RESULTS BEFORE AND AFTER TEST CONDENSER MODIFICATION
Initially, data were taken with the test condenser used
by Kanakis [Ref. 12], with no modifications. As can be seen
from Figure 5.1, the nondimensionalized heat-transfer coefficient exhibits a saw-toothed variation over the inundation
range considered. The particular wire pitch and wire diameter
combination chosen for Figure 5.1 was representative of the
pattern apparent in the data sets taken prior to the testcondenser modification.

It was determined that the original design of the test condenser allowed too much steam to bypass the five active tubes, thereby allowing the steam to flow between the walls of the test condenser and the two rows of dummy tubes. Since steam was bypassing the active tubes, inadequate steam was flowing in the vicinity of the active tubes. This led to artificially low heat-transfer readings in the lower tubes of the tube bundle. To minimize this problem, baffle plates were installed on each side of the test condenser as described previously (Figure 3.2).

After the placement of the baffle plates, the previously tested tube sets were retested. As can be seen in Figure 5.1, the saw-toothed variation of the curve is much less pronounced and appears to be nearly corrected. The net effect of the removal of the saw-toothed variation by the

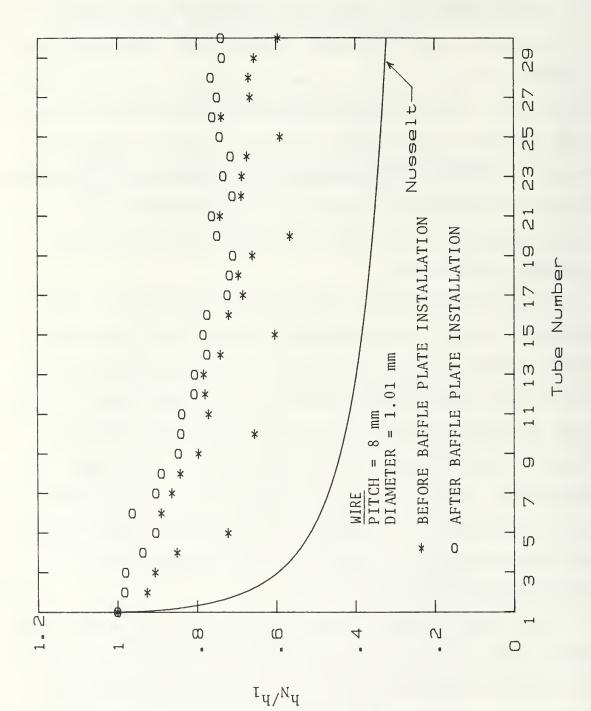


Figure 5.1 Data With and Without Baffles

installation of the baffle plates is an increase in the measured heat-transfer coefficient. This, in turn, means that the effect of inundation is not actually as great as had been earlier suspected. All data represented in this report will, hereafter, be with the baffles in place.

B. VAPOR-SHEAR CORRELATION

As discussed previously, the vapor shear correlation was assumed to be of the form postulated by Fujii [Ref. 28]—see equation (4.8). Figure 5.2 is the graph of Y vs. F for the first active smooth titanium tube. For this tube, the value of B was determined to be 0.834 and the value of n was determined to be 0.166. These results fall about 15 percent below the Fujii correlation [Ref. 28] (B = 0.96 and n = 0.20). There are several possible explanations for this difference.

- 1. Both the value of Y and the value of F are partly based on the steam velocity, with a higher value of steam velocity resulting in a lower value for both Y and F. During this thesis, the steam velocity was only varied from 1.4 to 2.1 m/s. The corresponding range of F was therefore from 3.6 to 1.5, while the corresponding range of Y was from 1.1 to 0.8. Since the range of both F and Y were considerably more extensive for the calculation of the Fujii correlation, this is a possible explanation for the discrepancy.
- 2. For the Fujii correlation, both in-line and staggered tube bundles were considered. This could also be a possible reason for the discrepancy.
- 3. The direction of the steam flow considered for the determination of the Fujii correlation included vertically downward flow, vertically upward flow and also horizontal flow. This difference in flow directions is another possible explanation for the discrepancy between the correlation determined in this report and the Fujii correlation.

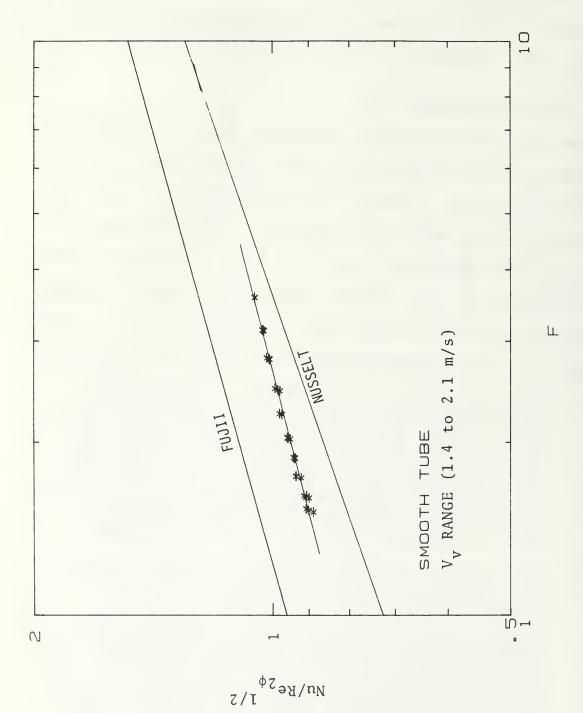


Figure 5.2 Vapor-Shear Correlation

4. As mentioned previously, baffle plates were installed in order to reduce the steam bypassing the active tubes. Even though the baffles reduced the amount of steam bypassing the active tubes, this is no guarantee that the steam still able to bypass the active tubes could not have had some influence on the vapor-shear correlation calculated in this thesis.

Since Reference 28 does not specifically say how the flow area used in his correlation was determined, this is another possible area for discrepancy. In addition, Fujii [Ref. 28] does not show which tube in the bundle was used for the formulation of his correlation. The actual location would have an important effect on the vapor velocity and could be another possible area for discrepancies to develop.

Vapor-shear correlations were determined for all of the tube sets tested for this report. Table 2 is a summary of the constants determined for the vapor-shear correlations.

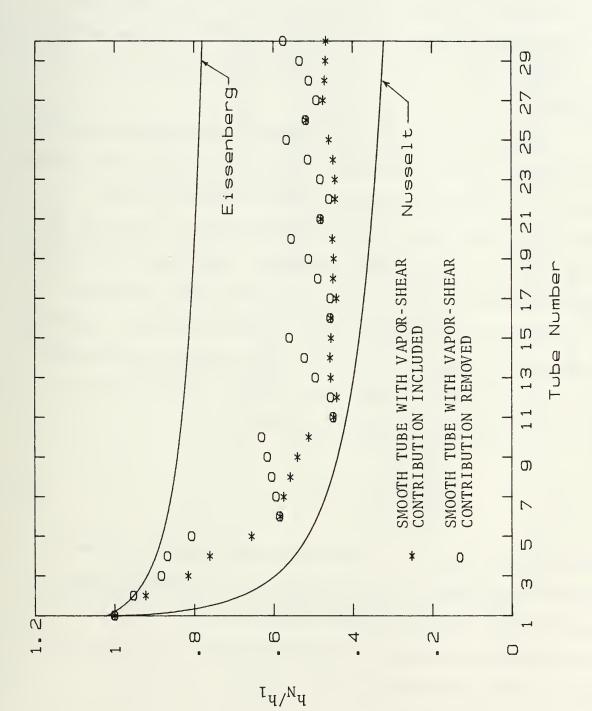
C. THE EFFECT OF INUNDATION ON A SMOOTH TUBE BUNDLE

1. With Vapor-Shear Contribution Present

Figure 5.3 shows the variation of the normalized, local heat-transfer coefficient for up to 30 tubes. The value for the normalized, local heat-transfer coefficient for the 30th tube is approximately 45% above the value predicted by Nusselt theory, but very near the value predicted by the Kern relationship. As discussed in Chapter II, the Nusselt theory for a tube bundle is based on idealized laminar flow from tubes above to tubes below. The idealizations of Nusselt result in a conservative estimate of the heat-transfer coefficient, so the fact that the data are 45% above the

TABLE 2
Vapor-Shear Correlations

TUBE SET	WIRE DIAMETER (mm)	PITCH (mm)	<u>B</u>	<u>n</u>
1	Smooth tube		0.834	0.185
2a	1.58	16	0.835	0.178
b	1.58	7.6	0.837	0.189
С	1.58	4	0.725	0.192
3a	1.01	8	0.916	0.161
b	1.01	6	0.871	0.173
С	1.01	4	0.876	0.220
4 a	0.50	4	1.036	0.200
Ъ	0.50	2	0.915	0.192
5	Dropwise-co	ated tube	0.796	0.166



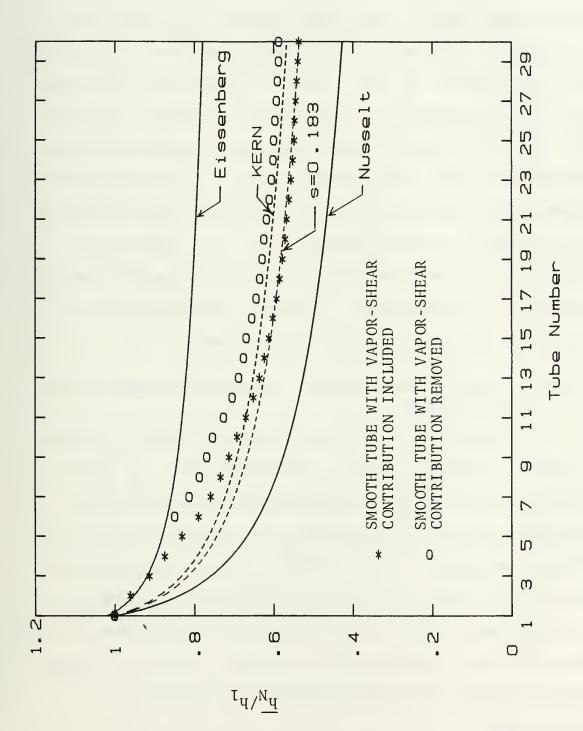
(Normalized, Local Heat-Transfer Coefficient) Inundation Effect on a Smooth Tube Bundle 5.3 Figure

Nusselt prediction is not totally unexpected. Since the Kern relationship is a more realistic view of the condensate action in the tube bundle, one would expect the data to be much nearer to the Kern relationship—as it is. The close agreement with the Kern relationship also was considered to be an indication that the test apparatus were operating properly.

Figure 5.4 shows the normalized, average heat-transfer coefficient for up to 30 tubes. The curve generated by these data is much smoother than the curve generated by the normalized, local heat-transfer coefficients. The value for the normalized, average heat-transfer coefficient of the 40th tube lies about 26% above the value predicted by Nusselt, but lies only 5% below the value predicted by Kern. Assuming a generalized form of the average heat-transfer coefficient for N tubes, divided by the heat-transfer coefficient for the first tube in the bundle, yields:

$$\overline{h_N}/h_1 = N^{-S}$$
 (5.1)

Referring to Figure 5.4, the dashed line roughly following the data represents a least-squares-fit exponential curve for the data, and the exponent derived is s=0.183. This exponent and the exponent suggested by Kern (s=0.167) are in reasonable agreement.



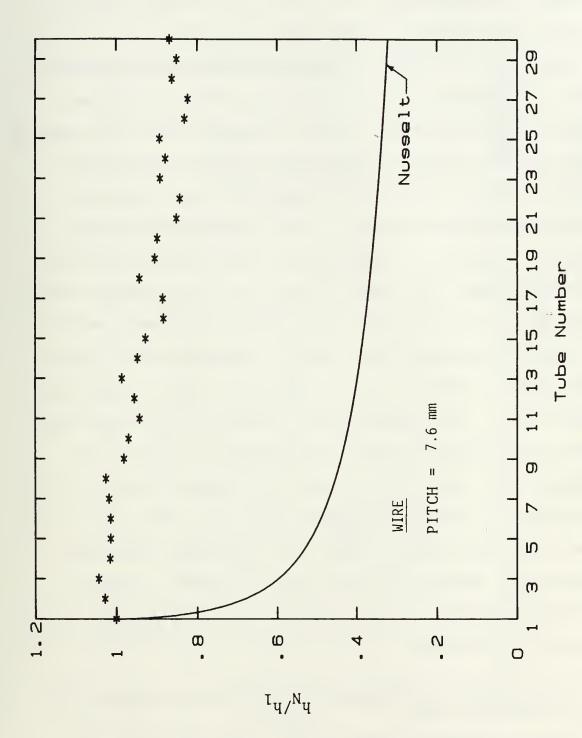
(Normalized, Average Heat-Transfer Coefficient) Inundation Effect on a Smooth Tube Bundle 5.4 Figure

2. With Vapor-Shear Contribution Removed

local heat-transfer coefficient for up to 30 tubes with the contribution due to vapor shear removed. The vapor-shear correlation used was the one presented earlier in this report (B = 0.834 and n = 0.185). As can be seen from the graph, the trend in the data has now become a reversed saw-tooth. This is an indication of overcorrecting for the effect of vapor shear. This unacceptable trend in the data shows that mathematical separation of the vapor-shear contribution is not possible as stated by Marto [Ref. 27]. Therefore, throughout the remainder of this thesis the vapor-shear contribution is included, and no further attempt is made to remove the contribution due to vapor shear.

D. THE EFFECT OF INUNDATION ON SMOOTH TUBES WRAPPED WITH 1.6-MM-O.D. WIRE

Even though three different pitches (16 mm, 7.6 mm and 4 mm) were tested at this wire diameter, only the graph for the optimum pitch will be presented for the normalized, local heat-transfer coefficient for a bundle of up to 30 tubes. The graph for the pitch of 7.6 mm is presented in Figure 5.5. The value for the normalized, local heat-transfer coefficient for the 30th tube is approximately 170% above the value predicted by the Nusselt theory and approximately 83% above the value predicted by the Kern relationship.



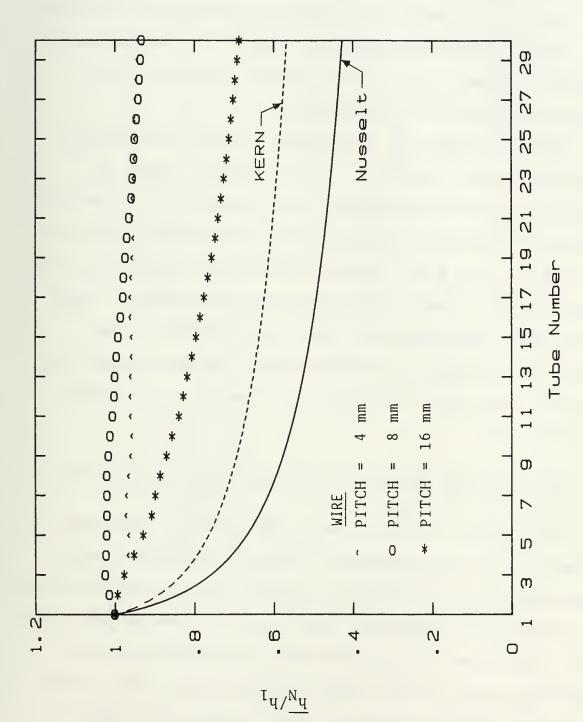
Effect of 1.6-mm-o.d. Wire on Inundation
(Normalized, Local Heat-Transfer Coefficient) Figure 5.5

Figure 5.6 shows the normalized, average heat-transfer coefficient for a bundle of up to 30 tubes. All three pitches are represented, with the values of the normalized, average heat-transfer coefficient of the 30th tube for the pitches of 16, 7.6 and 4 mm approximately 45%, 118%, and 120%, respectively, above the value predicted by Nusselt. As mentioned previously, the wire wrapping makes two separate contributions towards the increase in the heat-transfer coefficient—it thins the condensate film and acts as a condensate drainage channel. When compared with Figure 5.3, Figure 5.5 shows that condensate inundation has less effect on wire-wrapped tubes than it does on a set of smooth tubes and that a wire pitch of 7.6 mm gives the optimal results.

The optimum wire pitch or wire diameter is not simply based on the largest value of $h_{\rm N}/h_{\rm l}$, but also on the actual value of $h_{\rm l}$. For example, both 4 mm wire pitch and 7.6 mm wire pitch resulted in about the same $h_{\rm N}/h_{\rm l}$ value (see Figure 5.6). But the $h_{\rm l}$ on the tube wrapped with the 7.6 mm wire pitch is about 10 percent greater than the tube with the 4.0 mm wire pitch. Thus, the 7.6 mm wire pitch gave the optimal results for the 1.6-mm-o.d. wire. A more thorough discussion of the point is provided later in this chapter (Section H).

E. THE EFFECT OF INUNDATION ON SMOOTH TUBES WRAPPED WITH 1.0-MM-O.D. WIRE

Three different pitches (8 mm, 6 mm and 4 mm) were tested at this wire diameter. Again, only the graph for the



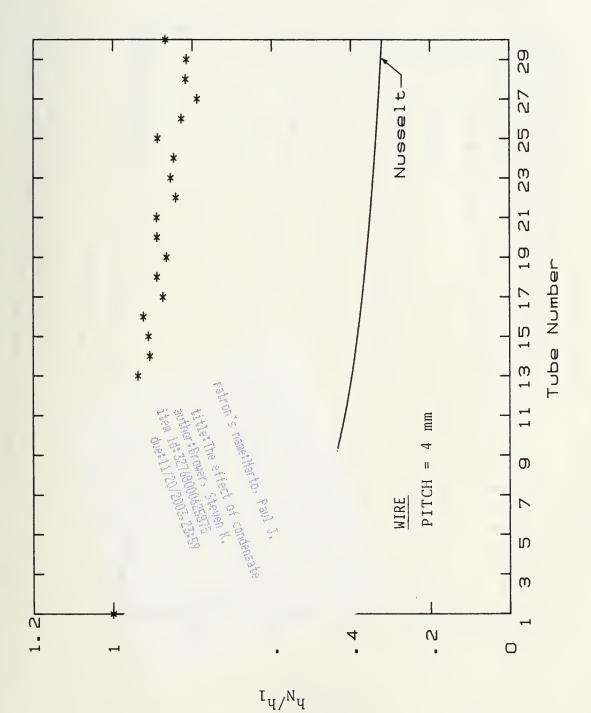
(Normalized, Average Heat-Transfer Coefficient) Effect of 1.6-mm-o.d. Wire on Inundation 5.6 Figure

optimum pitch will be presented for the normalized, local heat-transfer coefficient for a bundle of up to 30 tubes. The graph for a pitch of 4 mm is presented as Figure 5.7. The value for the normalized, local heat-transfer coefficient for the 30th tube is approximately 151% above the value predicted by the Nusselt theory and 71% above the value predicted by the Kern relationship.

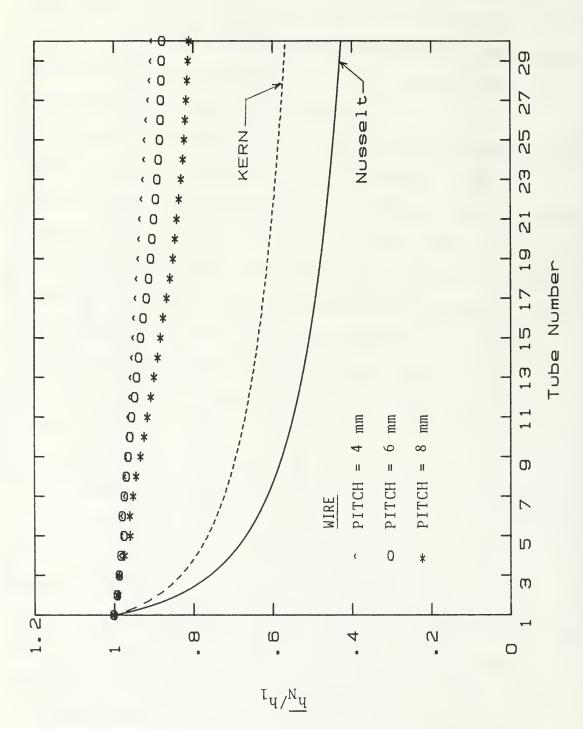
Figure 5.8 shows the normalized, average heat-transfer coefficient for a bundle of up to 30 tubes. Again, all three pitches are represented. The values of the normalized, average heat-transfer coefficient of the 30th tube (for the pitches 8, 6 and 4 mm) are approximately 90%, 105% and 117%, respectively, above the value predicted by Nusselt. These graphs again show that this tube configuration is less affected by condensate inundation than the smooth-tube set, and that the optimal wire pitch is 4 mm with this smaller diameter wire.

F. THE EFFECT OF INUNDATION ON SMOOTH TUBES WRAPPED WITH 0.5-MM-O.D. WIRE

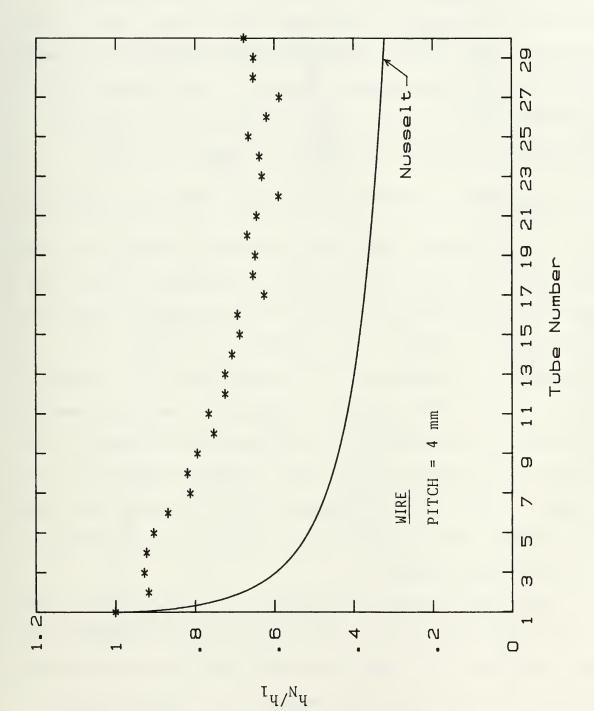
Two pitches (4 mm and 2 mm) were tested at this wire diameter. For the normalized, local heat-transfer coefficient for a bundle of up to 30 tubes, only the optimum pitch of 4 mm will be presented. The graph is presented as Figure 5.9. The value for the normalized, local heat-transfer coefficient for the 30th tube is approximately 101% above the value predicted by Nusselt and 37% above the value predicted by the Kern relationship.



Effect of 1.0-mm-o.d. Wire on Inundation
(Normalized, Local Heat-Transfer Coefficient) Figure 5.7



(Normalized, Average Heat-Transfer Coefficient) Effect of 1.0-mm-o.d. Wire on Inundation Figure 5.8



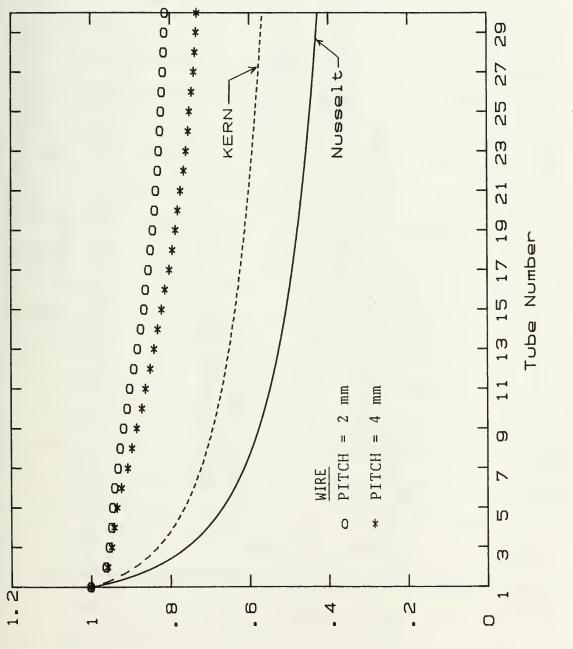
(Normalized, Local Heat-Transfer Coefficient) Effect of 0.5-mm-o.d. Wire on Inundation Figure 5.9

Figure 5.10 shows the normalized, average heat-transfer coefficient for a bundle of up to 30 tubes. Both of the pitches are represented. The values of the normalized, average heat-transfer coefficient for the 30th tube (for the pitches 4 and 2 mm) are approximately 71% and 80%, respectively, above the value predicted by Nusselt. These graphs also indicate that this tube configuration is not affected by condensate inundation as much as the smooth-tube set.

G. THE EFFECT OF INUNDATION ON SMOOTH TUBES WITH A DROPWISE COATING

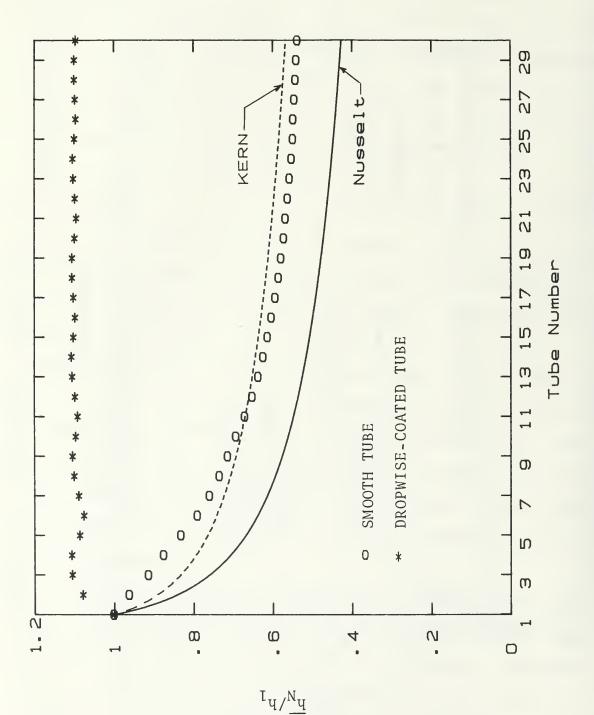
A set of smooth tubes with a dropwise coating was tested primarily to observe the effect of inundation. Figure 5.11 shows the normalized, average heat-transfer coefficient for a bundle of 30 tubes. Also included in the figure for comparison are the data for the smooth tube. As is immediately evident from the graph, condensate inundation does not decrease the performance of the dropwise-coated tubes. The ratio of the average outside heat-transfer coefficient of N tubes, divided by the heat-transfer coefficient of the first tube can be seen to rise at first and then remain steady at a value of approximately 1.1.

Since the coating is non-wetting, no large droplets form; rather, the droplets are small when they first form and roll down the side of the tube, as soon as they reach a moderate size. As these droplets are making their descent down the side of the tube, they collect additional droplets until they



(Normalized, Average Heat-Transfer Coefficient) Effect of 0.5-mm-o.d. Wire on Inundation Figure 5.10

^Тч/^Nч



Effect of Inundation on a Dropwise Tube Bundle Figure 5.11

reach a point where they are able to fall off the tube. As these droplets strike the tubes below, they again collect condensate and act as a sweeper of the condensate across the surface of the tubes below.

H. SUMMARY

Table 3 represents a summary of the results obtained from the data sets presented in this thesis. The column $h_1/h_{N_{11}}$ is the measure of the local heat-transfer coefficient of the first tube compared to the Nusselt prediction. can be considered a measure of the effectiveness of the wire-and-pitch combination to thin the condensate film between successive wire wraps. Refer to Figure 5.12 for clarification of this point. The column $\overline{h_{30}}/h_1$ is the measure of the average heat-transfer coefficient for 30 tubes compared to the coefficient for the first tube. can be considered to be a measure of the effectiveness of the wire-and-pitch combination at handling the condensate drainage. The column $\overline{h_{30}}/h_{\mathrm{Nu}}$ is the measure of the average heat-transfer coefficient for 30 tubes compared to the heattransfer coefficient predicted by the Nusselt theory for a single tube. It can be viewed as a measure of the overall performance of the 30 tube bundle. The last column (labelled s) is the exponent defined in equation (5.1), and is calculated using the least-squares fit for each data run.

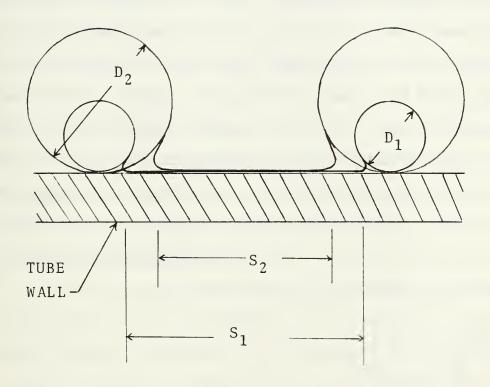
As can be seen from the data, it appears that the larger the wire, the better it aids in condensate drainage.

TABLE 3

Data Summary

* v	0.183	0.097	0.012	0.017	0.055	0.034	0.024	0.082	0.061	-0.072
h30/hNu	0.640	0.791	0.981	0.897	0.989	1.093	1.151	1.101	1.050	1.064
h30/h1	0.537	0.687	0.932	0.941	0.810	0.877	0.904	0.732	0.811	1.097
h ₁ /h _{Nu}	1.193	1.151	1.053	0.953	1.221	1.247	1.273	1.504	1.294	0.970
h ₁ (kW/m ² .K)	12.65	12.32	11.36	10.41	12.94	13.28	13.38	15.82	13.90	10.64
(mm) h ₁										ibe
WIRE PITCH (mm)	þe	16.0	7.6	4.0	8.0	0.9	4.0	4.0	2.0	coated tube
WIRE DIAMETER (mm)	Smooth tul	1.58	1.58	1.58	1.01	1.01	1.01	0.50	0.50	Dropwise-c
TUBE	1	2a	p	υ	3a	p	υ	4a	p	2

Determined by curve fitting the data for tubes 11 through 30. NOTE: *



 S_1 = Exposed surface for diameter 1.

 S_2 = Exposed surface for diameter 2.

Figure 5.12 Effect of Wire Diameter on Condensation

Conversely, the smaller the wire the higher the heat-transfer performance of the first tube. These last two observations are based on a comparison at a 4 mm pitch but with three different wire diameters. The explanation for this is that, for a given pitch, as the wire diameter of the wrap is increased, less and less tube surface area remains exposed to the steam flow. In addition to this, the condensate film thickness may also increase as the wire diameter is increased. Thus, for a given wire pitch, the heat-transfer coefficient of the first tube will increase with a decrease in the wire diameter. On the other hand, the amount of condensate collected around the wire increases with increasing wire diameter (see Figure 5.12). While the surface-tension forces tend to retain liquid at the wire, gravitational forces tend to draw liquid along the wire. As the wire diameter increases, the gravitational forces increase at a faster rate than the surface-tension forces. Therefore, condensate drainage increases with increasing wire diameter.

I. OBSERVATIONS

1. Smooth Tubes

During the data runs, filmwise condensation was observed with no visible indication of dropwise condensation. When the inundation rate was increased, more drops formed along the bottom of the tubes, but the droplet size did not increase noticeably. However, at the higher inundation

conditions there was a noticeable amount of splashing occurring in the test condenser.

2. Wire-Wrapped Tubes

As the steam was condensing on the tubes, the condensate was observed to be drawn to the wire wrap, leaving a thin film between the wires (film thickness was dependent upon the particular wire diameter and pitch). The condensate then ran down the wire and formed droplets on the bottom of the tube. The droplet size was different for each configuration of wire diameter and pitch. For the larger diameter wire, rivulets formed and there was a solid condensate "bridge" from the tubes above to the tubes just below them. As the inundation rate increased, these bridges became more pronounced. It was also noticed that the tendency to splash increased as the inundation rate increased, and was quite noticeable for all of the tube sets tested.

3. <u>Dropwise-Coated Tubes</u>

As was mentioned previously, the condensate droplets from the upper tubes in the bundle act as sweepers of the condensate across the surface of the tubes below them in the tube bundle. During the data run, the condensate collecting on the upper tubes and then falling onto the tubes below was sweeping the lower tubes at such a rate that the droplets forming on the lower tubes never reached a size larger than approximately 2 mm.

VI. CONCLUSIONS

- 1. The average, outside heat-transfer coefficient for 30 smooth tubes in a vertical column was 64% of the Nusselt coefficient for the first tube in the tube bundle.
- With wire wrapping, the average, outside heat-transfer performance of the tube bundle is considerably improved. Significant improvements in the outside heat-transfer performance are also possible by utilizing dropwise coatings.
- 3. The optimum pitch for the 1.6-mm-o.d. wire wrapping occurred at 7.6 mm. With this wire arrangement, the average, outside heat-transfer coefficient for the 30 tubes was 92% of the Nusselt coefficient for the first tube in the tube bundle.
- 4. The best pitch tested for the 1.0-mm-o.d. wire wrapping occurred at 4 mm. With this wire arrangement, the average, outside heat-transfer coefficient for the 30 tubes was 115% of the Nusselt coefficient for the first tube in the tube bundle.
- 5. The best pitch tested for the 0.5-mm-o.d. wire wrapping occurred at 4 mm. In this case, the average, outside heat-transfer coefficient for the 30 tubes was 110% of the Nusselt coefficient for the first tube in the tube bundle.
- 6. In simulating a bundle of 30 tubes coated with a dropwise promoter, the normalized, average, outside heat-transfer coefficient increased by approximately 10% over the value for the top tube in the bundle.

VII. RECOMMENDATIONS

A. TEST APPARATUS MODIFICATIONS

The modifications listed below should be considered prior to continued use of the test apparatus:

- 1. Redesign the test condenser to reduce the possibility of the steam bypassing the active test condenser tubes.
- 2. Install a larger heating system in the perforatedtube water supply tank or modify the existing heating system to increase the slow response of the heating system.

The modifications listed below would be beneficial to the operation of the test apparatus but are not necessary for proper operation:

- 1. Redesign the existing controls so that the test apparatus is controllable from a single panel.
- 2. Redesign the test condenser hotwell to provide a more accurate measure of the condensate collection rate.

B. ADDITIONAL TESTS TO CONDUCT

The following additional tests would be important in the continuation of this investigation:

- 1. Conduct tests with additional wire pitches for the 1.0-mm-diameter wire to determine the optimum wire pitch.
- 2. Conduct tests using other dropwise coatings.
- 3. Conduct tests with the commercially-available finned tubes.
- 4. Conduct additional tests to determine the effect of vapor shear on the outside heat-transfer coefficient.

APPENDIX A

SAMPLE CALCULATIONS

A. A sample calculation is performed in this section to illustrate the solution procedure used in the data-reduction program presented in Appendix C. The calculations are limited to the first tube in the bundle only. All thermophysical properties were determined from the Tables in Reference 32.

EXPERIMENTAL CONDITIONS:

Pressure Condition	Atmospheric			
Inundation Condition	5 tubes			
Inlet Temperature of Cooling Water	24.10 °C			
Outlet Temperature of Cooling Water	28.32 °C			
Saturation Temperature	100.65 °C			
Cooling Water Rotameter Setting	21.7%			

1. Determination of Average Bulk Temperature

$$T_b = (T_{ci} + T_{co})/2$$

$$T_b = (24.10 + 28.32)/2$$

$$T_b = 26.21 °C$$

2. Thermophysical Properties (evaluated at T_b)

$$P_{r} = 5.83$$

$$\rho_{C} = 997 \text{ kg/m}^{3}$$

$$\mu_{C} = 855 \times 10^{-6} \text{ N·s/m}^2$$

$$C_{pc} = 4.179 \text{ kJ/kg} \cdot \text{K}$$

$$k_{C} = 618 \times 10^{-3} \text{ W/m} \cdot \text{K}$$

$$\dot{m}_{C} = 0.243 \text{ kg/s}$$

4. Determination of Cooling Water Velocity

$$V_{c} = \dot{m}_{c}/(\rho_{c} \cdot A_{i})$$

$$V_{C} = (0.243)/(997) \cdot (1.56 \times 10^{-4})$$

$$V_{C} = 1.56 \text{ m/s}$$

5. Determination of Reynolds Number

$$Re = (\rho_C V_C D_i)/\mu_C$$

Re =
$$(997)(1.56)(.0141)/855 \times 10^{-6}$$

$$Re = 25,649$$

6. Determination of Heat Transfer

$$Q = \dot{m}_{C} (T_{CO} - T_{Ci}) C_{pc}$$

$$Q = 0.243(28.32 - 24.1)4179$$

$$Q = 4285 W$$

7. Determination of Heat Flux

$$q = Q/(\pi D_O L)$$

$$q = 4285/\pi (0.015875) (0.305)$$

$$q = 281,700 \text{ W/m}^2$$

8. Determination of Nusselt Coefficient

$$h_{Nu} = 0.651 \left[\frac{k_f^3 \rho_f^2 h_{fg} (9.81)}{\mu_f D_o q} \right] 1/3$$

Assume:

$$T_f = T_{sat}$$

$$T_f = 100.65 \, ^{\circ}C$$

$$\rho_f = 957.8 \text{ kg/m}^3$$

$$h_{fg} = 2255 J/kg$$

$$k_f = 680.52 \text{ W/m} \cdot \text{K}$$

$$\mu_{f} = 277.6 \times 10^{-6} \text{ N·s/m}^2$$

$$h_{Nu} = 0.651 \left[\frac{(680.52 \times 10^{-3})^3 (957.8)^2 (2255 \times 10^3) (9.81)}{(277.6 \times 10^{-6}) (0.015875) (281,700)} \right]^{1/3}$$

$$h_{Nu} = 11,243.6 \text{ W/m}^2 \cdot \text{K}$$

9. Determination of Tf,c

$$T_{f,c} = T_{sat} - q/(2 \cdot h_{Nu})$$

$$T_{f,C} = 100.65 - 281,700/2(11,243.6)$$

$$T_{f,C} = 88.12 \, ^{\circ}C$$

10. Thermophysical Properties

$$k_f = 674.6 \times 10^{-3} \text{ W/m} \cdot \text{K}$$

$$\rho_f = 966 \text{ kg/m}^3$$

$$\mu_f = 320.3 \times 10^{-6} \text{ N·s/m}^2$$

$$h_{fg} = 2288.3 \times 10^3 \text{ J/kg}$$

11. Determination of Nusselt Coefficient

$$h_{Nu} = 0.651 \left[\frac{k_f^3 \rho_f^2 h_{fg} 9.81}{\mu_f D_o q} \right] 1/3$$

$$h_{Nu} = 0.651 \left[\frac{(674.6 \times 10^{-3})^3 (966)^2 (2288.3 \times 10^3) (9.81)}{(320.3 \times 10^{-6}) (0.015875) (281,700)} \right]^{1/3}$$

$$h_{Nu} = 10,739.6 \text{ W/m}^2 \text{K}$$

12. Determination of T_f, c

$$T_{f.c} = T_{sat} - q/(2 \cdot h_{Nu})$$

$$T_{f,c} = 100.65 - 281,700/(2)(10,739.6)$$
 $T_{f,c} = 87.53 °C$

13. Determination of Logarithmic Mean Temperature

LMTD =
$$(T_{CO} - T_{Ci})/\ln(\frac{T_{sat} - T_{Ci}}{T_{sat} - T_{CO}})$$

LMTD = $(28.32 - 24.10)/\ln(\frac{100.65 - 24.10}{100.65 - 28.32})$
LMTD = 74.42 °C

14. Determination of Overall Heat-Transfer Coefficient

$$U_{O} = q/LMTD$$

$$U_{O} = 281,700/74.42$$

$$U_{O} = 3785.3 \text{ W/m}^{2} \cdot \text{K}$$

15. Determination of Inside Heat-Transfer Coefficient

Assume:

$$C_{f} = 1.1$$
 $C_{i} = 0.028$
 $h_{i} = \frac{k_{C}}{D_{i}} C_{i} Re^{0.8} Pr^{0.333} C_{f}$

$$h_{i} = \frac{0.618}{0.0141}(0.028)(25,649)^{0.8}(5.83)^{0.333}(1.1)$$

$$h_{i} = 8181 \text{ W/m}^{2}\text{K}$$

16. Determination of Inner Wall Temperature

$$T_W = T_b + q D_O/(h_i D_i)$$

$$T_W = 26.21 + (281,700) (0.015875)/(8181) (0.0141)$$

$$T_W = 64.98 °C$$

17. Determination of μ_{W} at the Average Wall Temperature

$$\mu_{\rm w} = 433 \times 10^{-6} \, \, \text{N·s/m}^2$$

18. Determination of Correction Factor

$$C_{f,c} = (\mu_{c}/\mu_{w})^{0.14}$$
 $C_{f,c} = 1.1$

19. Determination of Outisde Heat-Transfer Coefficient

$$h_{o} = \frac{1}{\frac{1}{U_{o}} - \frac{D_{o}}{D_{i} h_{i}} - R_{w}}$$

$$h_{o} = \frac{\frac{1}{1}{3785.3} - \frac{0.015875}{(0.0141)(8181)} - 0.000042925}$$

$$h_{o} = 11,957 \text{ W/m}^{2}\text{K}$$

APPENDIX B

UNCERTAINTY ANALYSIS

The uncertainty analysis performed for this thesis is based on the development described in Kline and McClintock [Ref. 33]. Kanakis [Ref. 12] employed this development and derived the expressions for uncertainty in the variables of this thesis also. In this thesis, the results presented by Kanakis [Ref. 12] in Appendix C will be utilized, but where necessary, corrections will be pointed out.

A. UNCERTAINTY IN THE COOLING WATER VELOCITY

$$\frac{\delta V_{C}}{V_{C}} = \left[\left(\frac{\delta \hat{m}_{C}}{\hat{m}_{C}} \right)^{2} + \left(\frac{\delta \rho_{C}}{\rho_{C}} \right)^{2} + \left(\frac{\delta A_{i}}{A_{i}} \right)^{2} \right]^{1/2}$$

with the following uncertainties assigned:

$$\delta \dot{m}_{C} = \pm 0.01 \text{ kg/s}$$

$$\delta \rho_{C} = \pm 3 \text{ kg/m}^{3}$$

$$\delta A_{i} = \pm 0.00000022 \text{ m}^2$$

B. UNCERTAINTY IN THE REYNOLDS NUMBER

$$\frac{\delta \operatorname{Re}}{\operatorname{Re}} = \left[\left(\frac{\delta \rho_{\mathbf{C}}}{\rho_{\mathbf{C}}} \right)^{2} + \left(\frac{\delta V_{\mathbf{C}}}{V_{\mathbf{C}}} \right)^{2} + \left(\frac{\delta D_{\mathbf{i}}}{D_{\mathbf{i}}} \right)^{2} + \left(\frac{\delta \mu_{\mathbf{C}}}{\mu_{\mathbf{C}}} \right)^{2} \right]^{1/2}$$

with the following uncertainties assigned:

$$\delta \rho_{\rm C} = \pm 3 \text{ kg/m}^3$$

 δV_{C} as calculated in Section A

$$\delta D_{i} = \pm 0.0001 \text{ m}$$

$$\delta \mu_{c} = \pm 10 \times 10^{-6} \text{ N·s/m}^2$$

C. UNCERTAINTY IN HEAT TRANSFER

$$\frac{\delta Q}{Q} = \left[\left(\frac{\delta \dot{m}_{c}}{\dot{m}_{c}} \right)^{2} + \left(\frac{\delta T_{co}}{T_{co} - T_{ci}} \right)^{2} + \left(\frac{\delta T_{ci}}{T_{co} - T_{ci}} \right)^{2} + \left(\frac{\delta C_{pc}}{C_{pc}} \right)^{2} \right]^{1/2}$$

with the following uncertainties assigned:

$$\delta \dot{m}_{c} = \pm 0.01 \text{ kg/s}$$

$$\delta T_{CO} = \pm 0.025 \, ^{\circ}C$$

$$\delta T_{Ci} = \pm 0.025 \, ^{\circ}C$$

$$\delta C_{pc} = \pm 2 \text{ J/kg} \cdot \text{C}$$

D. UNCERTAINTY IN THE HEAT FLUX

$$\frac{\delta q}{q} = \left[\left(\frac{\delta Q}{Q} \right)^2 + \left(\frac{\delta D_O}{D_O} \right)^2 + \left(\frac{\delta L}{L} \right)^2 \right]^{1/2}$$

with the following uncertainties assigned:

 δQ as calculated in Section C

$$\delta D_{O} = \pm 0.0001 \text{ m}$$

$$\delta L = \pm 0.001 \text{ m}$$

E. UNCERTAINTY IN h

$$\frac{\delta h_{Nu}}{h_{Nu}} = \left[\left(\frac{\delta k_{f}}{k_{f}} \right)^{2} + \left(\frac{2}{3} \frac{\delta \rho_{f}}{\rho_{f}} \right)^{2} + \left(\frac{1}{3} \frac{\delta h_{fg}}{h_{fg}} \right)^{2} + \left(\frac{1}{3} \frac{\delta q}{q} \right)^{2} + \left(\frac{1}{3} \frac{\delta \mu_{f}}{\mu_{f}} \right)^{2} + \left(\frac{1}{3} \frac{\delta \rho_{o}}{\rho_{o}} \right)^{2} + \left(\frac{1}{3} \frac{\delta q}{q} \right)^{2} \right]^{1/2}$$

with the following uncertainties assigned:

$$\delta k_f = \pm 0.0005 \text{ W/m} \cdot \text{K}$$

$$\delta \rho_f = \pm 3 \text{ kg/m}^3$$

$$\delta h_{fg} = \pm 0.12 \text{ J/kg}$$

$$\delta g = \pm 0.0005 \text{ m/s}$$

$$\delta\mu_f = \pm 4.9 \times 10^{-6} \text{ N·s/m}^2$$

$$\delta D_{O} = \pm 0.0001 \text{ m}$$

 δq as calculated in Section D

F. UNCERTAINTY IN Tf.c

$$\frac{\delta T_{f,c}}{T_{f,c}} = \left[\left(\frac{\delta T_{sat}}{T_{sat}} \right)^2 + \left(\frac{\delta q}{q} \right)^2 + \left(\frac{\delta h_{Nu}}{h_{Nu}} \right)^2 \right]^{1/2}$$

with the following uncertainties assigned:

$$\delta T_{sat} = \pm 0.025 \, ^{\circ}C$$

 δq as calculated in Section D

 $\delta h_{_{{
m N}_{11}}}$ as calculated in Section E

G. UNCERTAINTY IN LOGARITHMIC-MEAN-TEMPERATURE DIFFERENCE

$$\frac{\delta \text{LMTD}}{\text{LMTD}} = \left[\left(\frac{\delta T_{\text{sat}} (T_{\text{co}} - T_{\text{ci}})}{(T_{\text{sat}} - T_{\text{ci}}) (T_{\text{sat}} - T_{\text{co}}) \ln (\frac{T_{\text{sat}} - T_{\text{ci}}}{T_{\text{sat}} - T_{\text{co}}})} \right)^{2} \right]$$

$$+ \left(\frac{\delta T_{co}}{(T_{sat} - T_{co}) \ln (\frac{T_{sat} - T_{ci}}{T_{sat} - T_{co}})} \right)^{2}$$

+
$$\left(\frac{\delta T_{ci}}{(T_{sat}^{-T}_{ci}) \ln (\frac{T_{sat}^{-T}_{ci}}{T_{sat}^{-T}_{co}})}\right)^{2}$$

with the following uncertainties assigned:

$$\delta T_{sat} = \pm 0.025 \, ^{\circ}C$$

$$\delta T_{CO} = \pm 0.025 \, ^{\circ}C$$

$$\delta T_{Ci} = \pm 0.025 \, ^{\circ}C$$

H. UNCERTAINTY IN OVERALL HEAT-TRANSFER COEFFICIENT

$$\frac{\delta U_{o}}{U_{o}} = \left[\left(\frac{\delta q}{q} \right)^{2} + \left(\frac{\delta LMTD}{LMTD} \right)^{2} \right]^{1/2}$$

with the following uncertainties assigned:

 δq as calculated in Section D

 δ LMTD as calculated in Section G

I. UNCERTAINTY IN INSIDE HEAT-TRANSFER COEFFICIENT

$$\frac{\delta h_{i}}{h_{i}} = \left[\left(\frac{\delta k_{c}}{k_{c}} \right)^{2} + \left(\frac{\delta D_{i}}{D_{i}} \right)^{2} + \left(\frac{0.8 \, \delta \, Re}{Re} \right)^{2} + \left(\frac{0.333 \, \delta Pr}{Pr} \right)^{2} + \left(\frac{\delta C_{i}}{C_{i}} \right)^{2} + \left(\frac{0.14 \delta \left(\mu_{c} / \mu_{w} \right)}{\mu_{c} / \mu_{w}} \right)^{2} \right]^{1/2}$$

with the following uncertainties assigned:

$$\delta k_c = \pm 0.0005 \text{ W/m} \cdot \text{K}$$

$$\delta D_i = \pm 0.0001 \text{ m}$$

 δ Re as calculated in Section B

$$\delta P_r = \pm 0.063$$

$$\delta C_{i} = \pm 0.0001$$

$$\delta (\mu_C/\mu_W) = \pm 4.9 \times 10^{-6} \text{ N·s/m}^2$$

J. UNCERTAINTY IN TEMPERATURE DIFFERENCE

$$\frac{\delta DT}{DT} = \left[\left(\frac{\delta q}{q} \right)^2 + \left(\frac{\delta h_i}{h_i} \right)^2 + \left(\frac{\delta D_o}{D_o} \right)^2 + \left(\frac{\delta D_i}{D_i} \right)^2 \right]^{1/2}$$

with the following uncertainties assigned:

 δq as calculated in Section D

 δh_i as calculated in Section I

$$\delta D_{O} = \pm 0.0001 \text{ m}$$

$$\delta D_i = \pm 0.0001 \text{ m}$$

K. UNCERTAINTY IN OUTSIDE HEAT-TRANSFER COEFFICIENT

$$\frac{\delta h_{o}}{h_{o}} = \left[\left(\frac{\delta U_{o}}{U_{o}^{2} (\frac{1}{U_{o}} - R_{w} - \frac{D_{o}}{D_{i} h_{i}})} \right)^{2} + \left(\frac{\delta R_{w}}{\frac{1}{U_{o}} - R_{w} - \frac{D_{o}}{D_{i} h_{i}}} \right)^{2} + \left(\frac{\frac{\delta D_{o}}{D_{i} h_{i}} (\frac{\delta h_{i}}{h_{i}})}{\frac{1}{U_{o}} - R_{w} - \frac{D_{o}}{D_{i} h_{i}}} \right)^{2} \right] \frac{1}{2}$$

with the following uncertainties assigned:

 δU_{Ω} as calculated in Section H

$$\delta R_{tr} = \pm 0.00001 \text{ m}^2 \cdot \text{K/W}$$

 $\delta h_{\, \dot{1}} \,$ as calculated in Section I

For the data presented in Appendix A, the following uncertainties are the result:

$$\frac{\text{quantity}}{\text{V}_{\text{C}}} = 1.56 \pm 0.064 \text{ m/s}$$

$$\text{Re} = 25,649 \pm 1.12$$

quantity uncertainty

 $Q = 4258 \pm 169 w$

 $q = 281,700 \pm 12,010 \text{ W/m}^2$

 $h_{Nu} = 11,243.6 \pm 163.4 \text{ W/m}^2 \cdot \text{K}$

 $T_{f,C} = 88.12 \pm 4.0 \, ^{\circ}C$

LMTD = 74.42 ± 0.625 °C

 $U_{o} = 3785.3 \pm 164.48 \text{ W/m}^2 \cdot \text{K}$

 $h_i = 8181 \pm 292.6 \text{ W/m}^2 \cdot \text{K}$

 $DT = 38.77 \pm 2.19 \, ^{\circ}C$

 $h_0 = 11,957 \pm 2287 \text{ W/m}^2 \cdot \text{K}$

APPENDIX C

DATA REDUCTION AND PLOTTING PROGRAMS

```
1000! FILE NAME: DRP3
                   May 9, 1984
1005! DATE:
1010!
1015
       REFE
1020
       PRINTER IS 1
      PRINT USING "6X.""Select Uption:"""
PRINT USING "6X.""O Taking data or re-processing previous data"""
PRINT USING "6X.""1 Plotting previous data"""
PRINT USING "6X.""2 Computing exponent for experimental data"""
PRINT USING "6X.""3 Plotting on LOG-LOG"""
PRINT USING "6X.""4 Labelling"""
1025
1030
1035
1040
1045
1050
1055
       INPUT IOP
       IF Iop=0 THEN CALL Main
IF Iop=1 THEN CALL Plot
1060
1065
       IF IOP=2 THEN CALL EXPO
IF IOP=3 THEN CALL LPlot
1070
1075
       IF Iop=4 THEN CALL Label
1080
       END
1085
1090
       SUB Main
1095
       CBM /C1/ C(7)
       DIM Tc1(2), Tco(4,2). T1(4). Mft(4). Vw(4). Ho(4)
1100
1105
       DIM To(4).Ts(1),Tb(4),R3(4),R4(4),S3(4),S4(4),Fc(4),Mfi(4),Mfs(4)
1110!
1115!
       ASSIGN COEFFICIENTS FOR THE 8-TH ORDER
1120! POLYNOMIAL FOR TYPE-T (COPPER-CONSTANTAN)
1125! THERMOCOUPLES
       DATA 0.10086091.25727.94369.-767345.8295,78025595.81
1130
       DATA -9247486589,6.97688E+11,-2.66192E+13,3.94078E+14
1135
1140
       READ C(*)
1145!
1150! ASSIGN CONSTANTS FOR ROTAMETER CALIBRATION
1155! LINES (kg/min)
1160!
1165
       DATA -1.256E-4.5.866E-3.2.456E-4.-3.352E-3.-5.878E-4
1170
       DATA 1.121E-2.1.237E-2,1.221E-2.1.233E-2,1.232E-2
1175
       READ Mf1(*), Mfs(*)
1180!
1185! ASSIGN SIEDER-TATE COEFFICIENT AND EXPONENT
1190! FOR REYNOLDS NUMBER
1195 Ex=.8
1200!
1205! ASSIGN GEOMETRIC VARIABLES
1210
       D_1 = .0141
                       ! Inner diameter (m)
1215
       Do=.015875
                         Outer diameter (m)
1228
       Ktm=21.9
                       ! Thermal conductivity of titanium (W/m-K)
1225
       L=.305
                       ! Condensing length (m)
1230
1235!
       Pt=Do+1.5
                       ! Transverse tube pitch-to-diameter ratio
1240! COMPUTE THE MEAN STEAM FLOW AREA IN THE TEST CONDENSER (m 2)
       Af=(Pt^2-PI*Do 2/4)/Pt*2*L
1245
1250!
1255! COMPUTE INSIDE AREA AND WALL RESISTANCE
1260
      A1=PI+D1'2/4
1265
       Rw = Do * LOG(Do/D_1)/(2*Ktm)
1270!
1275
       PRINTER IS 701
1280
       CLEAR 709
1285
       BEEP
1290
       INPUT "ENTER MONTH, DATE, AND TIME (MM:DD:HH:MM:SS)",Time$
```

```
1295
      OUTPUT 709:"TD": Time%
1300
       Sgy=0
1305
       Sgf=0
       Sgyf=0
1310
       Sgf2=0
1315
1320
       BEEP
1325
       INPUT "VAPOR SHEAR DATA (1=Y.0=N)?".Ivd
1330
       IF Ivd=0 THEN
1335
       INPUT "ENTER OUTPUT MODE (1=SHORT, 0=LONG)", Jop
1340
       END IF
1345
       BEEP
1350
1355
       INPUT "ENTER THE INPUT MODE (1=3054A.2=FILE)".Im
       IF Im=2 THEN
1360
       BEEP
1365
       INPUT "ENTER THE NAME OF THE EXISTING DATA FILE", Olddata$ PRINT USING "10X,""This analysis is for data file "",10A"
1370
1375
                                                                       '.10A":Dlddata$
1380
       BEEP
       INPUT "ENTER 1 IF PROCESSING 1983 DATA".Idt ASSIGN @File2 TO Olddata$
1385
1390
1395
       END IF
1400
       IF Im=1 THEN
1405
       BEEP
       INPUT "GIVE A NAME FOR THE DATA FILE TO BE CREATED".Newdata$ CREATE BDAT Newdata$,25
1410
1415
       ASSIGN @File1 TO Newdata$
1420
1425
       END IF
1430
       BEEP
       INPUT "GIVE A NAME FOR THE OUTPUT FILE", File out$
1435
       BEEP
1440
1445
       INPUT "ENTER TUBE TYPE (0=PLAIN.1=ROPED)".Itt
1450
       IF Itt=0 THEN
       PRINT USING "10X.""Tube type Ci=.028
                                                         : Plain """
1455
1460
1465
       END IF
1470
       IF Itt=1 THEN
                                                         : Roped """
1475
       PRINT USING "10X,""Tube type
1480
       C_{1} = .057
       END IF
1485
1490
       PRINT USING "10X.""Sieder-Tate constant = "".Z.4D":Ci
1495
       BEEP
       INPUT "ENTER THE INUNDATION CONDITION (1=5 TUBES, 2=30 TUBES)".Mi

IF Mi=2 THEN PRINT " Inundation condition: 30 TUBES"

IF Mi=1 THEN PRINT " Inundation condition: 5 TUBES"
1500
1505
1510
1515
       IF Ivd=0 THEN
1520
       BEEP
1525
       INPUT "WANT TO INCLUDE VAPOR SHEAR (1=Y.0=N)?".Ivs
1530
       IF Ivs=1 THEN
1535
       BEEP
1540
       INPUT "ENTER B-VALUE FOR VAPOR SHEAR CORRELATION".B
1545
       BEEP
1550
       INPUT "ENTER EXPONENT FOR VAPOR-SHEAR CORRELATION". Nvs
       END IF
1555
1560
       END IF
       CREATE BDAT File_out$.6
ASSIGN @File3 TO File_out$
1565
1570
1575
        Ja=0
1580
       Nrun=0
1585
       FOR I=0 TO 4
1590
       S3(I)=0.
1595
       S4(I)=0.
```

```
1600
       NEXT I
       IF Im=1 THEN
BEEP
1605
1610
       INPUT "ENTER MANDMETER READINGS (HLW.HLM.HRW.HRM)".Hlw.Hlm.Hrw.Hrm
1615
1620
       OUTPUT @File1;Hlw,Hlm,Hrw,Hrm
       ELSE
1625
1630
1635
       IF Idt=0 THEN ENTER 9File2:Hlw.Hlm.Hrw.Hrm
       END IF
       Dp=Hlm-Hrm+(Hlw-Hlm-Hrw+Hrm)/13.5
1640
1645
       Mdot=FNMdot(Dp)
1650
       Nrun=Nrun+1
       OUTPUT 709:"TD"
ENTER 709:Time$
1655
1660
       IF Ivd=0 AND Jop=0 THEN PRINT " "
1665
       IF Nrun=1 THEN
1570
1675
       PRINT USING "10X,""Month, date, and time: "".15A"; Time$
1680
       PRINT
       END IF
1685
1690
       IF Im=2 THEN Rdf
1695
       BEEP
       INPUT "ENTER FLOW METER READINGS (AS PERCENTAGES)", Fm1, Fm2
1700
       DISP "START COLLECTING CONDENSATE"
1705
       BEEP
1710
       WAIT 20
1715
       OUTPUT 709: "AR AFO AL19"
OUTPUT 722: "F1 R1 T1 Z1 FL1"
1720
1725
1730!
1735!
       READ INLET WATER TEMPERATURES
1740!
       FOR I=0 TO 2
OUTPUT 709:"AS SA"
ENTER 722:Tc1(I)
1745
1750
1755
1760
       CALL Tysy(Tc1(I))
1765
       Tc_{I}(I) = FNTemp(Tc_{I}(I), I)
1770
       NEXT I
1775!
       READ OUTLET WATER TEMPERATURES
1780!
1785!
1790
       I_1 = 2
1795
       FOR I=0 TO 4
       IF I=0 OR I=3 THEN
1800
1805
       Iu=2
       ELSE
1810
1815
       Iu=1
       END IF
FOR J=0 TO IU
I1=I1+1
1820
1825
1830
       DUTPUT 709: "AS SA"
1835
1840
       ENTER 722:Tco(I.J)
       CALL Tvsv(Tco(I,J))
Tco(I,J)=FNTemp(Tco(I,J),I1)
1845
1850
       NEXT J
1855
       NEXT I
1860
1865!
1870! READ STEAM TEMPERATURES
1875
1880
       FOR I=15 TO 16
       OUTPUT 709: "AS SA"
1885
       ENTER 722: Ts(I-15)
CALL Tysy(Ts(I-15))
1890
1895
1900
       Ts(I-15)=FNTemp(Ts(I-15),I)
```

```
1905
      NEXT I
1910!
1915!
      READ CONDENSATE TEMPERATURE
1920!
1925
      OUTPUT 709: "AS SA"
1930
      ENTER 722: Tcon
1935
      CALL Ivsv(Icon)
      Tcon=FNTemp(Tcon, 17)
1940
1945!
1950!
      READ VAPOR TEMPERATURE
1955!
1960
      OUTPUT 709: "AS SA"
1965
      ENTER 722:Tv
1970
      CALL Tusu(Tu)
      Tv=FNTemp(Tv.18)
1975
1980!
1985!
      READ VAPOR PRESSURE
1990!
1995
      OUTPUT 709: "AS SA"
2000
      ENTER 722:P volts
2005!
2010! COMPUTE AVERAGE WATER TEMPERATURES AT INLET
2015!
2020
      T_1(0) = T_{C1}(0)
2025
      T_1(1) = (T_{C_1}(0) + T_{C_1}(1)) * .5
      T_1(2) = T_{C1}(1)
2030
2035
      Ti(3)=(Tc_1(1)+Tc_1(2))*.5
2040
2045!
      T_1(4) = T_{C1}(2)
2050!
      COMPUTE AVERAGE WATER TEMPERATURES AT OUTLET
2055!
2060
      FOR I=0 TO 4
       IF I=0 OR I=3 THEN
2065
2070
      To(I) = (Tco(I,0) + Tco(I,1) + Tco(I,2)) + .3333
2075
      ELSE
2080
2085
      To(I)=(Tco(I.0)+Tco(I.1))*.5
END IF
NEXT I
2090
2095
       T_{sa}=(T_{s}(0)+T_{s}(1))*.5
2100
      Pvap=FNPvsv(P_volts)
2105
       Tsat=FNTvsp(Pvap*133.322)
2110
       Dsup=Iv-Isat
2115!
2120! READ INFORMATION FOR CONDENSATE FLOW RATE
2125!
2130
2135
2140
       IF Ivd=0 THEN
      BEEP
      INPUT "ENTER INITIAL AND FINAL LEVELS IN HOT WELL 1".H1.H2
2145
      Dh=H2-H1
2150
      IF Nrum MOD 5=1 THEN Msum=0
2155
       Mf1=540.4836*Dh
2160
      Md1=Mf1*FNRhow(Tsat-10)*1.0E-6/60
2165
       Msum=Msum+Mf1
2170
      IF Mi=2 AND Nrun<>30 AND Nrum MDD 5=0 THEN
2175
2180
       Mave=Msum/5
      Set=(Mave*FNRhow(Tsat-10)/10 6+.03238)/.042132
      END IF
2185
2190!
2195 Rdf: !
2200!
      IF Ivd=( AND Jop=0 THEN
2205
```

```
PRINT USING "10X,""Run number = "".DD":Nrun
PRINT " Tube # : 1
2215
                                                                                    3
                                                                                                           5"
                                                                                                4
2220
2225
2230
2235
       END IF
        IF Ivd=1 AND Nrun=1 THEN
                                                                                             F .....
        PRINT USING "10X.""Data #
                                                ٧v
                                                         Retp
                                                                        Nuc
2240
2245
2250
        END IF
        IF Im=2 THEN
IF Nrun MOD 5=1 AND Mi=2 AND Nrun>5 THEN ENTER @File2:Fpt
        ENTER @File2:Ti(*),To(*).Tsa.Tcon.Tv.Pvap.Tsat.Dsup.Fm1.Fm2
2255
2260
        ENTER @File2:H1,H2
2265
2270
2275
        END IF
        IF Ivd=0 AND Jop=0 THEN
PRINT USING "10X,""Inlet temp (Deg C):"".5(DDD.DD.2X)";T<sub>1</sub>(*)
PRINT USING "10X,""Outlet temp (Deg C):"".5(DDD.DD.2X)";T<sub>0</sub>(*)
PRINT USING "10X,""Saturation temperature = "".3D.DD."" (Deg C)""";Tsat
PRINT USING "10X,""Degree of superheat = "".3D.DD."" (Deg C)""";Dsup
PRINT USING "10X,""Static pressure = "".3D.DD."" (mm Hg)""";Pvap
2280
2285
2290
2295
2300
        END IF
2305!
2310! CALCULATE AVERAGE BULK TEMPERATURES
2315!
2320
        IF Ivd=0 THEN Nx=4
2325
        IF Ivd=1 THEN Nx=0
2330
        FOR I=0 TO Nx
2335
        Tb(I)=(Ti(I)+To(I))*.5
2340
        NEXT I
2345!
2350
        IF Mi=1 OR (Mi=2 AND Nrum(6) THEN As!=0.
2355
        IF M1=2 AND Nrun>5 THEN S1=As1
2360
        S1=As1
2365
        FOR J=0 TO Nx
        IF J=0 THEN Cwf=Fm1
IF J=1 THEN Cwf=Fm2
2370
2375
2380
        Mf=Mf1(J)+Mfs(J)*Cwf
2385
        Tx = Tb(J)
2390
        V_{\omega}(J) = Mf/(FNRhow(Tx) + A_1)
2395!
2400! CALCULATE INSIDE AND OUTSIDE COEFFICIENTS
2405!
2410
        Rew=FNRhow(Tx)*Vw(J)*D1/FNMuw(Tx)
2415
        Cf=1.
2420
        Q=Mf*FNCpw(Tx)*(To(J)-T_1(J))
2425
        IF Nrun=1 AND J=0 THEN Q1=Q
2430
        Qp=Q/(PI*Do*L)
2435
         IF (Mi=1 DR (Mi=2 AND Nrun<6)) AND J=0 THEN
2440
         If 11m=Tsat
2445
        Kf=FNKw(Tfilm)
        Rhof=FNRhow(Tfilm)
2450
2455
        Hfg=FNHfg(Tsat) *1000
        Muf=FNMuw(Tfilm)
 2460
        Hnu=.651*Kf*(Rhof'2*Hfg*9.799/(Muf*Do*Op)) .3333
2465
2470
         Ifilmc=Tsat-Qp/Hnu*.5
2475
         IF ABS((Tfilmc-Tfilm)/Tfilmc)>.01 THEN
2480
         Tf:lm=Tf:lmc
         GDTO 2445
 2485
 2490
        END IF
        IF Ivd=0 AND Jop=0 THEN
PRINT USING "10X,""Nusselt coefficient for first tube = "".5D.D."" (W/m 2.
 2495
2500 PRI
2505 END IF
```

```
2510
       END IF
2515
2520
       IF J=0 AND Ivd=0 AND Jop=0 THEN
IF M1=1 AND Nrun=1 THEN Ho1=0.
PRINT " Tube Vw
2525
RR"
                                             Heat flux
                                                             Cond coef
                                                                              R1
                                                                                        R2
2530
       PRINT "
                                    (m/s)
                                               (W/m 2)
                                                             (H/m 2.K)"
2535
2540
       END IF
       IF Nrum=1 AND Ivs=1 AND Ivd=0 THEN
2545
       Vg=FNVvst(Tsat)
2550
       Vv=Mdot*Vg/Af
2555
2560
       Dt=(Tsat-Tf:1m) +2
       Muf=FNMuw(Tfilm)
2565
2570
       Hfg=FNHfg(Tsat) *1000
       Kf=FNKw(Tfilm)
       Rhof=FNRhow(Tfilm)
2575
2580
2585
2590
       Retp=Rhof*Vv*Do/Muf
       Ff=9.799*Do*Muf*Hfg/(Vv'2*Kf*Dt)
IF Ivd=0 THEN
2595
       Nuc=B*Ff'Nvs*Retp'.5
2600
       Hoc=Nuc+Kf/Do
2605
       Tfilmc=Tsat-Qp/Hoc*.5
       IF ABS(Tfilm-Tfilmc)>.1 THEN
2610
2615
       Tfilm=Tfilmc
       GOTO 2555
2620
2625
2630
       END IF
       END IF
2635
2640
       Muw=FNMuw(Tx)
2645
       Hi=FNKw(Tx)/Di*Ci*Rew Ex*(FNPrw(Tx))^.3333*Cf
2650
2655
       Dt = Qp/H_1 * Do/D_1
       Cfc=(Muw/(FNMuw(Tx+Dt)))^.14
2660
       IF ABS((Cf-Cfc)/Cfc)>.01 THEN
2665
       Cf = (Cf + Cfc) * .5
       GOTO 2645
2670
2575
       END IF
2680
       Lmtd=(To(J)-1;(J))/LOG((Tsat-T;(J))/(Tsat-To(J)))
2685
2590
       IF Nrun=1 AND Ivs=1 AND Ivd=0 THEN
       Fc(J)=Hoc/Hnu*(Q/Q1) .3333
       Mdot=Mdot-Q/Hfg
2695
       END IF
2700
2705
       Uo=Qp/Lmtd
2710
       Ho(J)=1/(1/Uo-Do/(D1*H1)-Rw)
2715
2720
2725
2730
       IF Ivd=1 THEN
       Dt=Qp/Ho(0)
       Tf:lm=Tsat-Dt/2
       Vg=FNVvst(Tsat)
2735
       Rhof=FNRhow(Tfilm)
2740
       Vv=Mdot*Vg/Af
2745
       Muf=FNMuw(Tfilm)
2750
       Kf=FNKw(Tfilm)
2755
       Hfg=FNHfg(Tsat) + 1000
2760
       Retp=Rhof+Vv*Do/Muf
2765
2770
       Ff=9.799*Do*Muf*Hfg/(Vv'2*Kf*Dt)
       Nuc=Ho(0) *Do/Kf
2775
2780
       Y=LOG(Nuc/Retp'.5)
       F=LOG(Ff)
2785
       Sgy=Sgy+Y
       Saf=Saf+F
Sayf=Sayf+Y*F
2790
2795
2800
       Saf2=Saf2+F 2
2805
       END IF
```

```
if lvd=0 THEN
IF Ivs=1 THEN Ho(J)=Ho(J)/Fc(J)
2810
2815
2820
      Rr=Uo/Ho(J)
       S1=S1+Ho(J)
2825
      IF Nrun MOD 5=1 THEN
2830
      IF Mi=1 OR (Mi=2 AND Nrun=31) THEN Ja=0
2835
      IF Mi=2 AND 5<Nrun AND Nrun<30 THEN Ja=Nrun-1 IF Mi=2 AND 35<Nrun THEN Ja=Nrun-1
2840
2845
2850
2855
      END IF
      IF Mi=1 DR (30<Nrun AND Nrun<36 AND Mi=1) DR Nrun<6 THEN
2860
      R1 = Ho(J)/Ho(0)
      R2=S1/((J+1+Ja) +Ho(0))
2865
      ELSE
2870
2875
      R1=Ho(J)/Ho1
      R2=$1/((J+1+Ja)*Ho1)
2880
      END IF
2885
2890 :
2895! PRINT RESULTS
2900!
2905
       IF Jop=0 THEN
2910
      PRINT USING "11X,DD.4X,DD.DD.2X,2(D.5DE,2X),3(Z.4D,2X)";J+1+Ja,Vw(J).Qp.Ho
(J),R1,R2,Rr
2915
       IF Im=1 AND J=4 THEN
2920
      BEEP
       INPUT "OK TO ACCEPT THIS DATA SET (1=Y,0=N)?", Dks
2925
       IF Ors >1 THEN
2930
2935
       Nrun=Nrun-1
2940
       GOTO 1650
2945
       END IF
2950
      END IF
2955
      END IF
2960!
2965
       IF Mi=2 AND Nrun<6 AND J=0 THEN
      Ho1=Ho1+Ho(0)/5
2970
2975
      END IF
2980
       FOR K=0 TO 4
       IF K=J THEN S3(K)=S3(K)+R1
IF K=J THEN S4(K)=S4(K)+R2
2985
2990
2995
       NEXT K
3000
       END IF
3005
       NEXT J
       IF Ivd=0 THEN
IF Nrun MOD 5=0 THEN
3010
3015
       FOR K=0 TO 4
3020
3025
       R3(K)=S3(K)/5
3030
       R4(K) = S4(K)/5
       S3(K) = 0.
3035
3040
       S4(K)=0.
3045
       NEXT K
3050
      IF Mi=2 AND Nrun MOD 5=0 AND Nrun<>30 THEN As1=Nrun*R4(4)*Ho1
3055!
3060! PRINT AVERAGE RATIOS
3055!
 3070
       IF Jop=0 OR (Jop=1 AND Nrun<6) THEN
3075
       PRINT
       PRINT "
                                                 R4"
3080
                                      R3
                           Tube #
3085
       END IF
       FOR J=1 TO 5
PRINT USING "12X.DD.2(4X.Z.4D)":J+Ja.R3(J-1),R4(J-1)
3090
3095
3100
       OUTPUT @File3:J+Ja.R3(J-1).R4(J-1)
       NEXT J
3105
```

```
3110 END IF
      IF Nrun MOD 5=0 AND M1=2 AND Nrun<>30 AND Im=1 THEN
3115
       BEEP
3120
3125 PRINT USING "10X,""Set porous-tube flowmeter reading to "",3D.D,"" PERCENT
3130 END IF
3135
      END IF
3140
      IF Ivd=1 THEN
3145
       Ev=EXP(Y)
       PRINT USING "12X,DD.5X,Z.DD.2X,Z.3DE.2X,3D.DD.2X.2(Z.3D,2X)";Nrun,Vv,Retp,
3150
Nuc , Ey , Ff
      IF Im=1 THEN
3155
       BEEP
3160
       INPUT "OK TO ACCEPT THIS DATA SET (1=Y,0=N)?".Oks
3165
3170
       IF Oks <> 1 THEN
3175
       Nrun=Nrun-1
3180
       GOTO 1605
3185
       END IF
3190
       END IF
3195
       OUTPUT @File3;Ff.Ey
3200
       END IF
3205!
3210
       IF Nrun MOD 5=1 AND M1=2 AND Nrun>5 AND Im=1 THEN
       DUTPUT @File1;Set
Mpt=-8.361613+10.076742*Set
3215
3220
3225
       END IF
3230
       IF Im=1 THEN
       OUTPUT @File1; Ti(*), To(*), Tsa. Tcon, Tv. Pvap, Tsat. Dsup, Fm1. Fm2
3235
3240
       DUTPUT @File1:H1.H2
3245
       BEEP
      INPUT "WILL THERE BE ANOTHER RUN (1=Y.0=N)?", Go_on IF Go_on=1 THEN IF Ivd=0 THEN 1650 IF Ivd=1 THEN 1605
3250
3255
3260
3265
      END IF
3270
3275
       ELSE
      IF M1=2 AND Nrun<30 THEN
IF Ivd=0 THEN 1650
IF Ivd=1 THEN 1605
3280
3285
3290
3295
      END IF
      IF Mi=1 AND Nrun<10 THEN
3300
      IF Ivd=0 THEN 1650
IF Ivd=1 THEN 1605
3305
3310
3315
      END IF
3320
      END IF
       IF Im=1 THEN PRINT USING "10X.DD."" Data runs were stored in file "",10A";
3325
Nrun.Newdatas
      PRINT
3330
3335
      PRINT USING "10X.""Plot data are stored in file "",14A";File out$
      IF Ivd=1 THEN
3340
      Nvs=(Nrun*Sgyf-Sgy*Sgf)/(Nrun*Sgf2-Sgf 2)
3345
       B=EXP((Sgy-Nvs*Sgf)/Nrun)
3350
3355
       PRINT
       PRINT USING "10X.""Vapor-Shear Correlation:"""
PRINT USING "12X.""B = "".Z.3D":B
PRINT USING "12X.""n = "".Z.3D":Nvs
3360
3365
3370
3375
       END IF
3380
       ASSIGN @File1 TO .
       ASSIGN @File2 TO * ASSIGN @File3 TO *
3385
 3390
3395
       SUBEND
```

```
3400!
3405! THIS SUROUTINE CONVERTES THERMOCOUPLE VOLTAGE INTO TEMPERATURE
3410!
3415
      SUB Tysy(T)
3420
      COM /C1/ C(7)
3425
      Sum=0.
      FOR I=0 TO 7
3430
3435
      Sum=Sum+C(I)*T I
      NEXT I
3440
      T=Sum
3445
3450
      SUBEND
3455!
3460! THIS FUNCTION CALCULATES PRANDTL NUMBER OF WATER IN THE
      RANGE 15 TO 45 DEG C
3465!
3470!
3475
      DEF FNPrw(T)
      Y=10 (1.09976605-T*(1.3749326E-2-T*(3.968875E-5-3.45026E-7*T)))
3480
      RETURN Y
3485
3490
      FNEND
3495!
3500!
      THIS FUNCTION CALCULATES THERMAL CONDUCTIVITY OF WATER
3505! IN THE RANGE OF 15 TO 105 DEG C
3510!
      DEF ENKW(T)
3515
3520
      Y=.5625894+T*(2.2964546E-3-T*(1.509766E-5-4.0581652E-8*T))
3525
      RETURN Y
3530
      FNEND
3535!
3540!
      THIS FUNCTION CALCULATES SPECIFIC HEAT OF WATER
3545!
      IN THE RANGE 15 TO 45 DEG C
3550!
3555
      DEF FNCpu(T)
3560
      Y=(4.21120858-T*(2.26826E-3-T*(4.42361E-5+2.71428E-7*T)))*1000
3565
      RETURN Y
3570
      FNEND
3575!
3580! THIS FUNCTION CALCULATES DENSITY OF WATER IN THE
3585!
      RANGE 15 TO 105 DEG C
3590!
3595
      DEF FNRhow(T)
3600
      Ro=999.52946+T*(.01269-T*(5.482513E-3-T*1.234147E-5))
      RETURN Ro
3605
3610
      ENEND
3615!
3620!
      THIS FUNCTION APPLIES CORRECTIONS TO THERMOCOUPLE READINGS
3625!
3630
      DEF FNTemp(I.I)
3635
      DIM A(14), B(14)
      DATA 0.640533.0.573054.0.593101.0.57298.0.56228.0.567384.0.569577
3640
3645
      DATA 0.553951.0.552008.0.566955.0.520998.0.522661.0.531008.0.560788.0.5524
05
3650
      DATA 11.8744.8.63163.9.39412.8.570246.8.299436.8.36677.8.04507.7.459766
      DATA 7.498928.7.9408.5.87072.5.391556.6.13399.6.48586.6.326224
3655
3660
      READ A(*).B(*)
3665
      IF IK15 THEN
      \bar{I} = T - (A(I) - B(I) + .001 + T)
3670
3675
      ELSE
      T=T-.5
3680
3685
      END IF
3590
      RETURN T
3695
      FNEND
```

```
3700!
3705!
      THIS FUNCTION COMPUTES THE SPECIFIC VOLUME OF STEAM
3710!
      DEF FNVvst(Tt)
3715
3720
3725
      P=FNPvst(Tt)
      T=Tt+273.15
3730
      X=1500/T
3735
      F1=1/(1+T*1.E-4)
3740
      F2=(1-EXP(-X))^2.5*EXP(X)/X.5
      B*.0015*F1-.000942*F2-.0004882*X
K*2*P/(461.52*T)
3745
3750
3755
      V=(1+(1+2*B*K)^{-}.5)/K
      RETURN V
3760
      FNEND
3765
3770!
3775!
      THIS FUNCTION CONVERTS THE VOLTAGE READING OF THE PRESSURE
3780! TRANSDUCER INTO PRESSURE IN MM HG
3785!
3790
3795
      DEF FNPvsv(V)
       Y=1.1103462+163.36413*V
      RETURN Y
3800
3805
      FNEND
3810!
3815!
      THIS FUNCTION CALCULATES THE SATURATION TEMPERATURE OF STEAM AS A FUNCTION
3820! OF PRESSURE
3825!
3830
      DEF FNTVsp(P)
      Tu=110
3835
3840
       T1 = 80
3845
      Ta=(Tu+T1)*.5
3850
      Pc=FNPvst(Ta)
      IF ABS((P-Pc)/P)>.0001 THEN
3855
      IF Pc<P THEN T1=Ta
IF Pc>P THEN Tu=Ta
3860
3865
       GOTO 3845
3870
3875
       END IF
       RETURN Ta
3880
3885
       FNEND
3890!
3895!
      THIS FUNCTION COMPUTES THE VISCOSITY OF WATER
3900!
3905
      DEF FNMuw(T)
3910
      A=247.8/(T+133.15)
       Mu=2.4E-5+10^A
3915
3920
       RETURN Mu
3925
       FNEND
3930!
3935!
      THIS FUNCTION COMPUTES THE LATENT HEAT OF VAPORIZATION
3940!
3945
       DEF FNHfg(T)
       Hfg=2497.7389-T*(2.2074+T*(1.7079E-3-2.8593E-6*T))
RETURN Hfg
3950
3955
3960
       FNEND
3965!
3970! THIS FUNCTION COMPUTES THE SATURATION PRESSURE
3975!
       DEF FNPvst(Tsteam)
3980
3985
       DIM K(8)
       DATA -7.691234564.-26.08023696.-168.1706546.64.23285504.-118.9646225
DATA 4.16711732.20.9750676.1E9.6
3990
3995
```

```
4000 KEAD K(*)
4005
      T=(Tsteam+273.15)/647.3
      Sum=0
FOR N=0 TO 4
4010
4015
      Sum=Sum+K(N)*(1-T)^(N+1)
NEXT N
4020
4025
      Br=Sum/(T*(1+K(5)*(1-T)+K(6)*(1-T)^2))-(1-T)/(K(7)*(1-T)^2+K(8))
4030
4035
      Pr=EXP(Br)
4040
      P=22120000*Pr
4045
4050
      RETURN P
      FNEND
4055!
4060!
      THIS FUNCTION CALCULATES THE STEAM MASS FLOW RATE
4065!
4070
      DEF FNMdot(Dp)
4075
      Mdot=4.3183E-3+5.6621E-4*Dp
4080
      RETURN Mdot
4085
      FNEND
```

APPENDIX D

SEIDER-TATE CONSTANT CALCULATION

The following solution procedure used in calculating the Sieder-Tate constant, by using a modified Wilson plot, is presented in this appendix:

1. The overall heat-transfer coefficient is defined

as:
$$\frac{1}{U_0} = \frac{1}{h_0} + R_w + \frac{D_0}{D_i h_i}$$
 (D.1)

2. The Nusselt number is defined as:

Nu =
$$h_i D_i / k_c$$

or Nu = $C_i Re^{0.8} Pr^{1/3} (\mu_c/\mu_w)^{0.14} (D.2)$
or Nu = $C_i \Omega$; where (D.2a)
 $\Omega = Re^{0.8} Pr^{1/3} (\mu_c/\mu_w)^{0.14}$

3. By using the Nusselt equation:

$$h_{o} = 0.725 \left[\frac{k_{f}^{3} \rho_{f}^{2} h_{fg} g}{\mu_{f}^{D} o (T_{sat} - T_{w})} \right]^{1/4}$$
 (D.3)

and
$$q = h_o (T_{sat} - T_w)$$
 (D.4)

and upon combining equations (D.3) and (D.4)

$$h_{o} = 0.651 \left[\frac{k_{f}^{3} \rho_{f}^{2} h_{fg} g}{\mu_{f}^{D} \rho_{o}^{q}} \right]^{1/3}$$
 (D.5)

or
$$h_0 = 0.651 v^{1/3}$$
; where (D.5a)
 $v = \frac{k_f^3 \rho_f^2 h_{fg} g}{\mu_f D_0 q}$

4. Substituting equations (D.2a) and (D.5a) into equation (D.1), and rearranging gives:

$$\left(\frac{1}{U_0} - R_W\right) v^{1/3} = \frac{1}{C_i} \frac{D_0 v^{1/3}}{k_c \Omega} + \frac{1}{0.0651}$$
 (D.6)

or Z = m W + 1/0.0651; where

$$Z = (\frac{1}{U_0} - R_W) v^{1/3}$$
,

 $m = 1 / C_{i}$ and

$$W = D_o v^{1/3} / k_c \Omega$$

5. By plotting Z vs. W and computing the slope of the least-squares-fit line, and then taking the reciprocal of this slope, the resulting value is the Sieder-Tate constant.

NOTE: T_{w} and T_{f} must be determined iteratively, as described in Chapter IV, section E.

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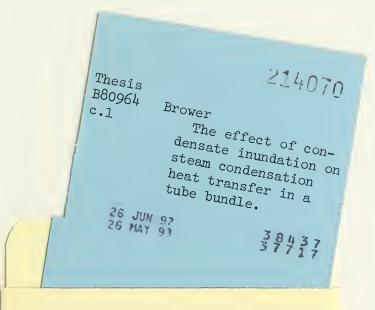
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