

# **NAVAL POSTGRADUATE SCHOOL MONTEREY, CALIFORNIA**



## **THESIS**

**EVALUATION OF THE BOEING PAN AIR  
TECHNOLOGIES CODE (A502I) THROUGH  
PREDICTION OF SEPARATION FORCES  
ON THE GBU-24**  
by

Matthew A. LeTourneau

March, 1996

Thesis Advisor:

Max F. Platzer

**Approved for public release; distribution is unlimited.**

DUDLEY KNOX LIBRARY  
NAVAL POSTGRADUATE SCHOOL  
MONTEREY CA 93943-5101

## REPORT DOCUMENTATION PAGE

Form Approved OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instruction, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188) Washington DC 20503.

1. AGENCY USE ONLY ( <i>Leave blank</i> )	2. REPORT DATE March 1996	3. REPORT TYPE AND DATES COVERED Master's Thesis	
4. TITLE AND SUBTITLE EVALUATION OF THE BOEING PAN AIR TECHNOLOGIES CODE (A502I) THROUGH PREDICTION OF SEPARATION FORCES ON THE GBU-24		5. FUNDING NUMBERS	
6. AUTHOR(S) LeTourneau, Matthew, A			
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Naval Postgraduate School Monterey CA 93943-5000		8. PERFORMING ORGANIZATION REPORT NUMBER	
9. SPONSORING MONITORING AGENCY NAME(S) AND ADDRESS(ES)		10. SPONSORING/MONITORING AGENCY REPORT NUMBER	
11. SUPPLEMENTARY NOTES The views expressed in this thesis are those of the author and do not reflect the official policy or position of the Department of Defense or the U.S. Government.			
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited.		12b. DISTRIBUTION CODE	
13. ABSTRACT ( <i>maximum 200 words</i> ) The Boeing PAN AIR Technologies code (A502i) is investigated to explore its suitability for determination of separation forces on ordnance. To this end, A502i is first assessed by applying it to three problems for which other solutions and experimental data are available, i.e. steady flow past a rectangular, parabolic arc wing and a delta wing at both subsonic and supersonic conditions. Good agreement is found in all cases. A502i is then applied to the GBU-24's being in two configurations for a subsonic case and a supersonic case. Good agreement is found with data obtained from wind tunnel experiments for low angles of attack.			
14. SUBJECT TERMS A502i, Panel Methods, Potential Flow, Geometry Modelling, GBU-24		15. NUMBER OF PAGES 116	
		16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT Unclassified	18. SECURITY CLASSIFICATION OF THIS PAGE Unclassified	19. SECURITY CLASSIFICATION OF ABSTRACT Unclassified	20. LIMITATION OF ABSTRACT UL

NSN 7540-01-280-5500

Standard Form 298 (Rev. 2-89)

Prescribed by ANSI Std. Z39-18 298-102



**Approved for public release; distribution is unlimited.**

**EVALUATION OF THE BOEING PAN AIR TECHNOLOGIES CODE (A502I)  
THROUGH PREDICTION OF SEPARATION FORCES ON THE GBU-24**

Matthew A. LeTourneau  
Lieutenant, United States Navy  
B.S., The George Washington University, 1988

Submitted in partial fulfillment  
of the requirements for the degree of

**MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING**

from the

**NAVAL POSTGRADUATE SCHOOL  
March 1996**



## ABSTRACT

The Boeing PAN AIR Technologies code (A502i) is investigated to explore its suitability for determination of separation forces on ordnance. To this end, A502i is first assessed by applying it to three problems for which other solutions and experimental data are available, i.e. steady flow past a rectangular, parabolic arc wing and a delta wing at both subsonic and supersonic conditions. Good agreement is found in all cases. A502i is then applied to the GBU-24's being in two configurations for a subsonic case and a supersonic case. Good agreement is found with data obtained from wind tunnel experiments for low angles of attack.



## TABLE OF CONTENTS

I.	INTRODUCTION.....	1
II.	OVERVIEW OF THE A502i CODE.....	3
	A. THEORY.....	3
	B. GENERAL A502i USAGE.....	8
	1. Running A502i with an Existing Executable.....	9
	2. Creating an A502i Input File.....	10
	C. GEOMETRY MODELLING.....	12
	1. The Parabolic Arc Airfoil.....	12
	2. The Deltawing.....	13
	3. GBU-24.....	15
	4. The F-14.....	20
	5. Combination Geometries.....	20
	D. GRAPHICS VISUALIZATION.....	24
III.	DATA EXTRACTION.....	27
IV.	RESULTS OF A502i COMPUTATIONS.....	29
	A. PARABOLIC ARC AIRFOIL DISCUSSION.....	29
	B. DELTAWING DISCUSSION.....	31
	C. GBU-24 FREE-STREAM (NO CANARDS) DISCUSSION.....	35
	1. Subsonic Case ( $M=0.8$ ).....	35
	2. Supersonic Case ( $M=1.2$ ).....	35
	D. GBU-24 FREE-STREAM (WITH CANARDS) DISCUSSION.....	38
	1. Subsonic Case ( $M=0.8$ ).....	38
	2. Supersonic Case ( $M=1.2$ ).....	40
	E. F-14 DISCUSSION.....	42
	F. POST-PROCESSING DISCUSSION.....	42
V.	SUMMARY AND CONCLUSIONS.....	47
APPENDIX A.	GBU-24 OUTPUT FILE (INPUT FILE PORTION).....	49
APPENDIX B.	GBU-24 OUTPUT FILE (EDGE ANALYSIS).....	71
APPENDIX C.	GBU-24 OUPUT FILE (UNIT NORMALS).....	85

APPENDIX D. GBU-24 OUTPUT FILE (SOLUTION DATA).....	87
LIST OF REFERENCES.....	103
INITIAL DISTRIBUTION LIST.....	105

## **ACKNOWLEDGEMENT**

The author wants to thank Dr. Alex Cenko of NAWC/AD Warminster whose assistance, guidance and tutelage made this thesis possible.



## I. INTRODUCTION

In the past, ballistic trajectory determination for manual or computer predicted ordnance delivery from an aircraft was determined through measurement of separation forces on the piece of ordnance via wind tunnel or captive carry measurements. The advent of panel method codes using linearized potential theory, such as A502i, or its full potential version TranAir, allow for a cheaper and safer method of predicting separation forces. Furthermore, A502i allows for any arbitrary configuration to be modelled within the limitations of the number of panels and networks allowed and excluding transonic flow.

The purpose of this work was to determine the separation forces on a GBU-24 carried by an F-14 on stations 3 or 6 or both. It was also the purpose of this work to provide an analysis of the code itself to see if it is a viable tool for the study of flow characteristics over arbitrary wing configurations for use in the Naval Postgraduate School's (NPS) Department of Aeronautics and Astronautics. The majority of the work was conducted on the NPS computer systems. The Department of Aeronautics and Astronautics Silicon Graphics Incorporated (SGI) workstations were utilized for most of the input files as well as the execution of the code. Due to the amount of disk space required, storage of the output files took place on the NPS Computer Center's Y-MP EL98 Cray computer. The bulk of the GBU-24 data was calculated using the SGI workstations at the Naval Air Warfare Center in Warminster.

The scope of this analysis was to understand the capabilities of the A502i code. The approach was to validate A502i against existing data and linear theory. The code was run for three different geometries under assorted Mach and AOA conditions. Comparisons were made for each of the geometries.



## II. OVERVIEW OF THE A502i CODE

The A502i code is used to computationally analyze inviscid subsonic or supersonic flows about arbitrary configurations. The code differs from other panel methods in that it is a higher order panel method; that is, the singularity strengths are not constant on each panel. A502i solves the linearized potential flow boundary-value problem at subsonic and supersonic Mach numbers.

The aerodynamic solution provides surface flow properties (flow directions, pressures, Mach number), configuration forces and moments, sectional forces and moments, and pressures. Additionally, A502i calculates flow properties in the flow-field points and flow-field streamlines. Results are limited to subsonic and supersonic cases (transonic cases excluded) with attached flow. Results are not usually applicable to cases where viscous effects and separation are dominant.

## A. THEORY

The basic equations describing the flow of a viscous, compressible, heat-conducting fluid are the Navier-Stokes equations. These are:

(a) The continuity equation,

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{V}) = \frac{\partial \rho}{\partial t} + \sum_{i=1}^3 \frac{\partial (\rho V_i)}{\partial x_i} = 0 \quad (2.1)$$

where  $\nabla = (\frac{\partial}{\partial x_1}, \frac{\partial}{\partial x_2}, \frac{\partial}{\partial x_3})$  is the gradient operator with respect to the location vector  $\vec{x} = (x_1, x_2, x_3)$ , and where the conventional index notation is used instead of  $\vec{x} = (x, y, z)$ . In addition,  $t$  is time,  $\rho(\vec{x}, t)$  is the density and  $\vec{V}(\vec{x}, t)$  is the velocity vector, with components  $\vec{V} = (V_1, V_2, V_3)$ .

(b) The conservation of momentum equation,

$$\frac{\partial}{\partial t}(\rho V_j) + \sum_{i=1}^3 \frac{\partial}{\partial x_i} (\rho V_i V_j) = -\frac{\partial p}{\partial x_j} + \sum_{i=1}^3 \frac{\partial}{\partial x_i} \tau_{ji} + \rho f_j \quad (j=1,2,3) \quad (2.2)$$

where  $\tau_{ij}$  is the deviatoric portion of the stress tensor which vanishes for a frictionless fluid,  $\vec{f}(\vec{x},t)$  is an external body force per unit mass exerted on the fluid, and  $p(\vec{x},t)$  is the pressure.

(c) The conservation of energy equation

$$\begin{aligned} \frac{\partial}{\partial t} (\rho e + \frac{1}{2} \rho |\vec{V}|^2 + p) + \sum_{i=1}^3 \frac{\partial}{\partial x_i} [(\rho e + \frac{1}{2} \rho |\vec{V}|^2 + p)V_i] \\ = \frac{\partial p}{\partial t} + \sum_{i,m} \frac{\partial}{\partial x_i} (\tau_{im} V_m + k \frac{\partial T}{\partial x_i}) + \rho \sum_i f_i V_i \end{aligned} \quad (2.3)$$

where  $e(\vec{x},t)$  is the internal energy of the fluid,  $k$  is the coefficient of heat conductivity for the fluid, and  $T(\vec{x},t)$  is the temperature.

(d) The equation of state

$$f(\rho, p, T)=0 \quad (2.4)$$

where the function  $f$  depends on the type of fluid. The derivations of these equations can be found in Refs. 1,2 and 3.

The Navier-Stokes equations can be simplified by the neglect of viscosity, which is equivalent to setting the deviatoric stress tensor to zero. Combining the momentum and continuity equations yields

$$\rho \frac{dV_j}{dt} = -\frac{\partial p}{\partial x_j} + \rho f_j \quad j = 1,2,3 \quad (2.5)$$

where the convective derivative operator is defined as

$$\frac{d}{dt} = \frac{\partial}{\partial t} + \sum_i V_i \frac{\partial}{\partial x_i}$$

Equation (2.5) is referred to as Euler's equation. The continuity and energy equations can

Equation (2.5) is referred to as Euler's equation. The continuity and energy equations can be reduced to

$$\rho \frac{d}{dt} \left( \frac{1}{2} |\vec{V}|^2 \right) = -\vec{V} \cdot \nabla p + \rho \vec{V} \cdot \vec{f} \quad (2.6)$$

and the rate of increase of heat per unit mass is given by

$$q = \frac{1}{\rho} \nabla \cdot (k \nabla T) = \frac{dp}{dt} + p \frac{d}{dt} \left( \frac{1}{\rho} \right) \quad (2.7)$$

Equations 2.5, 2.6 and 2.7 can be reduced to a single equation when four further assumptions are made. First, assume isentropic flow, thus

$$q=0 \quad (2.8)$$

Second, assume irrotationality

$$\nabla \times \vec{V} = 0 \quad (2.9)$$

which allows for the introduction of the potential function [Refs 2,3]

$$\nabla \Phi = \vec{V} \quad (2.10)$$

Third, assume the existence of a freestream potential  $\Phi_\infty$ , whose gradient is the uniform velocity  $\vec{V}_\infty$  attained at points sufficiently distant from the disturbance being analyzed, and thus write

$$\phi = \Phi - \Phi_\infty \quad (2.11)$$

and

$$\vec{V} = (u, v, w) = \nabla \Phi = \nabla \Phi_\infty + \nabla \phi = \vec{V}_\infty + \nabla \phi \quad (2.12)$$

The quantities  $\phi$  and  $\vec{v}$  are called the perturbation potential and velocity [Ref 3]. Fourth, assume that

$$|\vec{v}|^2 \ll a_\infty^2 \quad (2.13)$$

everywhere, where  $a_\infty$  is the freestream speed of sound.

Based on these four assumptions, the unsteady potential equation is obtained [Refs 1,3]:

$$\begin{aligned}
 & (1 - M_\infty^2) \phi_{xx} + \phi_{yy} + \phi_{zz} - 2M_\infty^2 \phi_{xt} - M_\infty^2 \phi_{tt} \\
 & = M_\infty^2 \left[ \frac{1}{2} (\gamma - 1) (2u + 2\phi_t + |\vec{v}|^2) \nabla^2 \phi \right. \\
 & \quad + (2u - u^2) \phi_{xx} + v^2 \phi_{yy} + 2vw \phi_{yz} + w^2 \phi_{zz} \\
 & \quad \left. + 2(1+u)(v\phi_{xy} + w\phi_{xz}) + 2(uu_t + vv_t + ww_t) \right] \\
 & \tag{2.14}
 \end{aligned}$$

Assuming the flow conditions do not change with time yields the steady non-linear potential equation.

$$\begin{aligned}
 & (1 - M_\infty^2) \phi_{xx} + \phi_{yy} + \phi_{zz} \\
 & = M_\infty^2 \left[ \frac{1}{2} (\gamma - 1) (2u + |\vec{v}|^2) \nabla^2 \phi \right. \\
 & \quad + (2u + u^2) \phi_{xx} + v^2 \phi_{yy} + 2vw \phi_{yz} + w^2 \phi_{zz} \\
 & \quad \left. + 2(1+u)(v\phi_{xy} + w\phi_{xz}) \right] \\
 & \tag{2.15}
 \end{aligned}$$

When  $M_\infty = 0$ , equation (2.15) reduces to Laplace's equation,

$$\nabla^2 \phi = 0 \tag{2.16}$$

For the case of  $M_\infty \neq 0$ , the following is supposed,

$$M_\infty^2 |\vec{v}| \ll 1 - M_\infty^2 \tag{2.17}$$

$$M_\infty^2 |\vec{v}| \ll 1 \tag{2.18}$$

which are small perturbation assumptions [Refs. 1,2]. With these assumptions the steady non-linear potential equation reduces to the Prandtl-Glauert equation [Ref 1]:

$$(1 - M_\infty^2) \phi_{xx} + \phi_{yy} + \phi_{zz} = 0 \tag{2.19}$$

Through a coordinate transformation [Refs. 1,2,3], the Prandtl-Glauert equation can be re-written as:

$$s\phi_{xx} + \phi_{yy} + \phi_{zz} = 0 \quad (2.20)$$

where when  $s=1$ , it is the subsonic case and Laplace's equation applies and when  $s=-1$ , it is the supersonic case and the wave equation applies. Applying Green's third identity [Ref. 1] yields the following integral equation,

$$\phi(P) = -\frac{1}{4\pi s} \int \int \left[ \frac{\sigma}{R} - \mu \hat{n} \cdot \nabla \frac{1}{R} \right] dS \quad (2.21)$$

where  $\sigma$  represents the source strength and  $\mu$  represents the doublet strength. When supplemented with boundary conditions, it is equation (2.21) that A502i solves.

A502i solves equation (2.21) through a discretization process. The general idea of the process falls into two parts. The first is developing finite dimensional approximate representation formulas for the singularity functions, which creates a system of equations with unknown coefficients,  $\lambda_i$ . The second part involves solving the set of equations for all  $\lambda_i$ . This allows for completely determining the source and doublet functions. Then, by virtue of equation (2.21), the potential function  $\phi(P)$  is determined for all points  $P$ , solving the problem.

The features of A502i which distinguish it from predecessors are three-fold. The first is a feature known as “continuous geometry”, the second is linear source and quadratic doublet variation, the third is continuity of doublet strength.

Most panel methods approximate the configuration geometry with panels whose planform is a quadrilateral. Thus, if the panels themselves are planar, only a small class of configurations (such as cylinders and flat wings) can be described without gaps being left between panels. These gaps are generally small, except for highly twisted surfaces. The

gaps cause little numerical error in subsonic flow, but in supersonic flow, the cumulative effect of the gaps is serious [Ref. 1]. The problem is not associated with leakage of flow through the gaps, but with the doublet strength jumping abruptly from a non-zero value to zero at a panel edge which does not exactly meet the adjacent edge. In A502i, gaps are closed by means of panels which are comprised of several planar regions.

The feature of linear source and quadratic doublet variation is what makes A502i a higher order panel method. The basis function corresponding to a source parameter is locally linear, while the basis function corresponding to a doublet parameter is locally quadratic. This is what allows for A502i to find supersonic solutions. Numerical solution of the wave equation is far more sensitive to the numerical idiosyncrasies of a panel method than is the solution of Laplace's equation for subsonic flow. Experimental evidence [Ref. 1] indicates that exact surface analysis is not feasible in supersonic flow without doublet continuity, thus the potential for numerical error is greatly reduced by requiring the doublet singularity strength to be continuous across panels.

## B. GENERAL A502i USAGE

The use of the A502i code consists of generating an input file, which can be arbitrarily named, and which contains the information defining the geometry of the configuration, flow field points of interest, the flow conditions and wakes. The process of building a geometry is difficult in that A502i is particular about its input format. Simple configurations, such as a rectangular, planform wing can be modelled manually, but more complex structures require a pre-processing program, such as MACGS, where a geometry can be graphically built. MACGS will output a data file in a format that, with minor modifications, via another pre-processing program that can move the data from three columns to six columns, will be readily usable by A502i. Currently, the school does not have a copy of MACGS, but it can be acquired through McDonnell-Douglas. To complete

this thesis, MACGS was used on the SGI workstations at NAWC Warminster. Wakes also must be constructed in the same manner as the structure to be analyzed. More detailed instructions on the specifics of wakes and surface geometries can be found in Ref. [4]. Appendix A is a portion of an output file, but lines 1 thru 1120 are an exact duplicate of the input file.

## **1. Running A502i with an Existing Executable**

Assuming an A502i executable file (e.g., A502) has already been placed in a user executable directory (e.g., /usr/local/bin), the only other necessary items needed to produce a set of A502i output files is the input file and a large amount of storage space. Anything modelled with more than one thousand panels total will use more than one hundred mb of disk space. If the maximum number of panels (20,000) is used, the disk space required will be on the order of 2 gb.

To run A502i, enter after the UNIX prompt:

**A502 <input file > output file**

Prior to running the code, it is highly recommended that a Cray account be opened and linked to the department's SGI workstations. This is done by assigning the same user i.d. number to the Cray account as is assigned to the account with the department. User i.d. numbers can be changed by the computer center at the user's request. This is required due to the limited disk space available to individual accounts in the department. Once an account is opened, log on to a department workstation, change directories to an existing Cray directory, for example (after the UNIX prompt):

**cd /jedi/d1/maletour**

Transfer the input file to the Cray directory and execute the code. The screen will display what portion of the code it is performing and how long it took to perform each portion in CPU time. The code outputs numerous files in addition to the arbitrarily named output file.

The two output files of interest, in the vast majority of cases, are the arbitrarily named output file and the ft13 file. In order to run another solution all output files must be deleted or renamed prior to re-executing the code. Relevant results should not be kept on the Cray account as files on disks d1, d2 and u1 are considered temporary storage and subject to erasure after a period of time.

## 2. Creating an A502i Input File

The input file, which can be arbitrarily named, consists of two portions, the largest being the geometry data. Appendix A is a complete recreation of the input file for the GBU-24 with canards. The file begins with line 1, \$TITLE, and ends with line 1120, \$END. The line numbers are for reference only and are not part of the actual input file. The first portion consists of creating the initial conditions, i.e., the free-stream Mach number and angle of attack, the type of analysis to be performed, i.e., solution or datacheck, what types of output that are to be included in the arbitrarily named output file, and reference points to be used in calculating forces and moments. The geometry data consists of the points that bound each panel, that in turn belong to a specific group of panels that make up a network. The overall structure being modelled consists of a series of networks. A502i can run up to 150 networks and or 20,000 panels with a limit of 8,000 panels per network. Referencing Appendix A, line 28 represents the first network of the model, a canard. Line 29 represents the number of networks that will be classified under this \$POINTS statement. Line 30 indicates what kind of surface the network will be, a three-dimensional surface with flow properties to be calculated, a wake and a base are several examples. Line 31 is the number of y points and the number of x points respectively that make up the grid of that network. Line 32 is where the panel points start. Reference 4 contains detailed instructions on the options and meaning of each of the non-geometry inputs, including some capabilities not shown in Appendix A.

Two types of solutions can be run, a datacheck and a full solution. Reference 4 explains how to enter either one into the input file. The datacheck only analyzes the geometry. This can be accomplished in a matter of seconds for a simple geometry as it is only running the first several portions of the code. The full solution can take a couple of hours for a geometry of the size of 4,000 panels. The datacheck should be run once the geometry has been modelled. It will check for any panel edges that do not abut properly, and when column 4 of line 20 in Appendix A is a 1, the datacheck will list the unit normal vectors, which must be facing outward. The datacheck will also see if the wakes are attached properly. A502i is capable of giving warnings both on-screen and in the arbitrarily named output file when an edge or a wake is not modelled properly, but it only lists the unit normal vectors. The directions of the vectors must be manually checked by the user. The full solution performs the datacheck first, so the data is repeated in the arbitrarily named output file. Appendix B is a portion of the output file that contains the summary of facing surfaces. Each panel edge is looked at to see what it abuts against. Sections such as wingtips, leading edges of a flat plate or any surface that does not need a wake attached, but is unabutted to any other panel on that edge will draw probable error messages or warnings from the code. The user must ensure that the edge is not supposed to abut against anything or need a wake attached. If that is the case, the warnings may be ignored. Appendix D is the first page of the portion of the output file that lists the unit normal vectors. The three columns under zc are the x-y-z coordinates of the given panel's center. The three columns under znc are the x-y-z coordinates of the unit normal vector. In most cases, when the y coordinates are of the same sign, then the unit normal vector is pointing outward.

## C. GEOMETRY MODELLING

Five geometries needed to be modelled, each of increasing complexity. Modelling proved to be the most difficult task, in that A502i is a FORTRAN code and is very format sensitive, but the sheer number of points that need to be generated can take a lot of time and the order those points are listed in the input file is what determines whether or not the shape is correctly modelled. Of the five geometries modelled, none were done completely manually. A spreadsheet was used for generating the parabolic arc airfoil and the deltawing since those structures can be constructed out of one network, excluding wingtips and wakes, and the surface can be defined by a mathematical function. The bombs and the F-14 required the use of MACGS to be properly modelled. MACGS is indifferent as to the order that geometries are built, and often doesn't require many coordinate inputs if building a model on top of an existing IGS file. The output file from MACGS is automatically formatted and the points placed in the appropriate order for A502i to understand. Although, the order may be reversed where the unit outward normal vector is concerned. MACGS has the ability to output files in several different panel method code input formats, including PMARC. Reference 4 gives detailed instructions on how to properly order points to build a group of networks that will model a geometry. A502i uses a right-handed coordinate system that is similar to an aircraft body axes. When put in terms of a wing, the x axis is positive from leading edge to trailing edge. The z axis is positive up and the y axis is positive out the right wing.

### 1. The Parabolic Arc Airfoil

The parabolic arc airfoil is the simplest of all the geometries. The airfoil has a chord of five and a span of ten. The maximum thickness is .15. The model consists of approximately 600 panels, including the wake and wingtips. A spreadsheet was used to develop the geometry portion of the input file. Line 32 of Appendix A demonstrates the

format that the spreadsheet used. Rows consist of two points, with coordinates  $x_1, y_1, z_1$ ,  $x_2, y_2, z_2$  using a format of 6F10.0. The chord was divided into 25 points (x coordinate) from trailing edge to leading edge and then another 25 points from leading edge to trailing edge (bottom half). The span was divided into 12 points (y coordinate) from left to right. Due to the symmetry of a rectangular planform, the y coordinate was constant along the 50 x coordinates that constituted a chordwise cross-section. To attain a maximum thickness of .15 the formula,

$$z = .3 * \left( \frac{x}{c} - \frac{x^2}{c^2} \right) \quad (2.22)$$

was utilized to generate the values of the z coordinates. The wingtips simply connect the x coordinate on the top side with it's symmetrical counterpart on the bottom side. Due to a trailing edge composed of a straight line, the wake is modelled by a single panel that spans the trailing edge and has a length aft of 100. Figure 2.1 shows the panel distribution across the top surface of the parabolic arc airfoil, where the thickness is represented by the color scheme. A panel and a point are numbered to show how they were entered into the input file.

## 2. The Deltawing

The deltawing represented a step up in complexity over the parabolic arc airfoil. The chordwise cross-section is parabolic, while the spanwise cross-section is linear. The procedure for building the geometry on a spreadsheet was the same as that for the parabolic arc airfoil, only the chord length is not constant along the span. For simplicity in design, the number of panels per column of panels is constant on the deltawing, as on the parabolic arc airfoil. This means an increasing panel density in the direction of the wing tip. The wake is modelled the same as the airfoil. The right wingtip ended in a point, so no extra panelling was needed to close any gaps. The symmetry of the deltawing allowed for



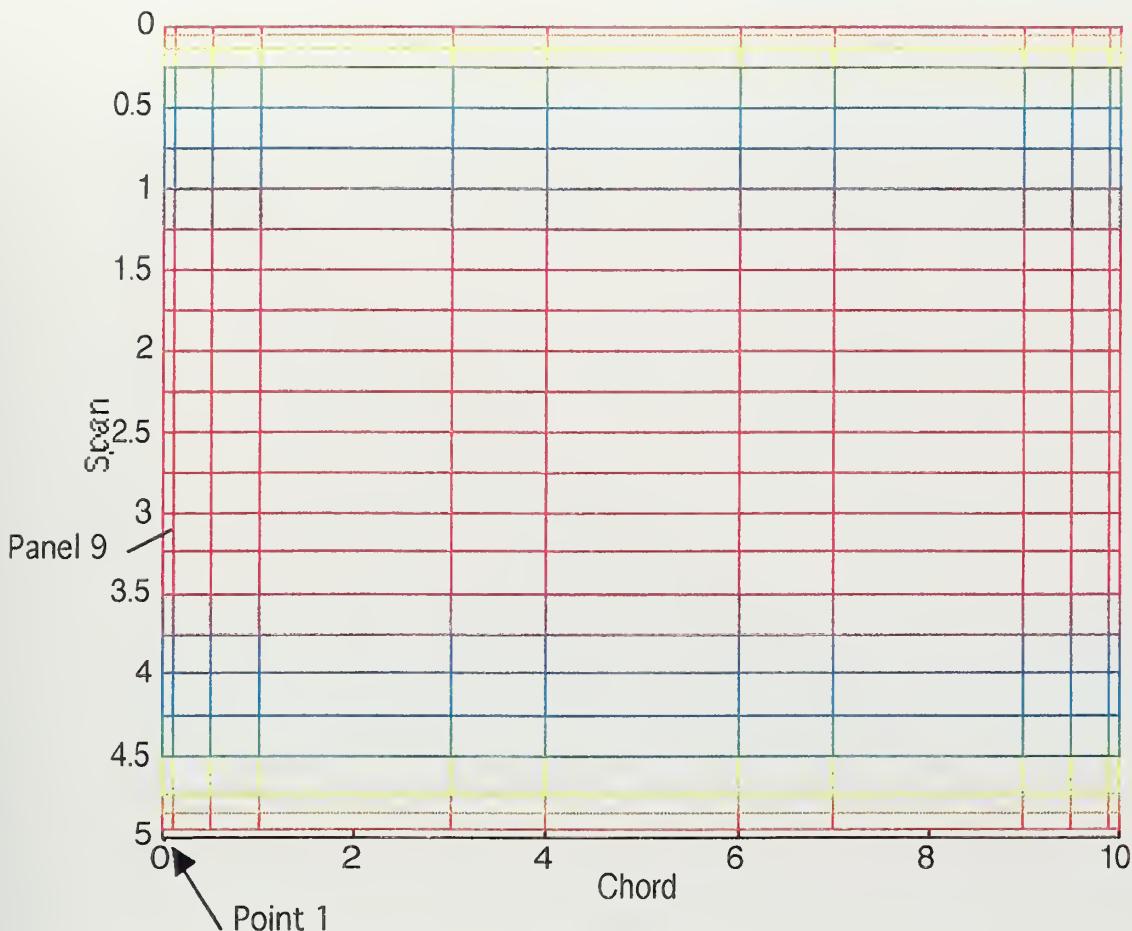


Figure 2.1 Parabolic Arc Panel Distribution

further simplification and reduction of the code's run time by only modelling from the centerline to the right tip. A502i allows the user to stipulate whether there is symmetry in the x-z plane and or the x-y plane (see line 5 and 6 of Appendix A). This means that the gap between the top and bottom panels at the center line does not need to be bridged as in the parabolic arc airfoil (symmetry could have also been used in the airfoil's case). The chord of the deltawing has a length of 90 and the semi-span has a length of 15. The maximum thickness occurs midway along the centerline and is .05. The model consists of 880 panels. Figure 2.2 shows the panel distribution along the top surface of the deltawing.



Thickness is represented by the color scheme.

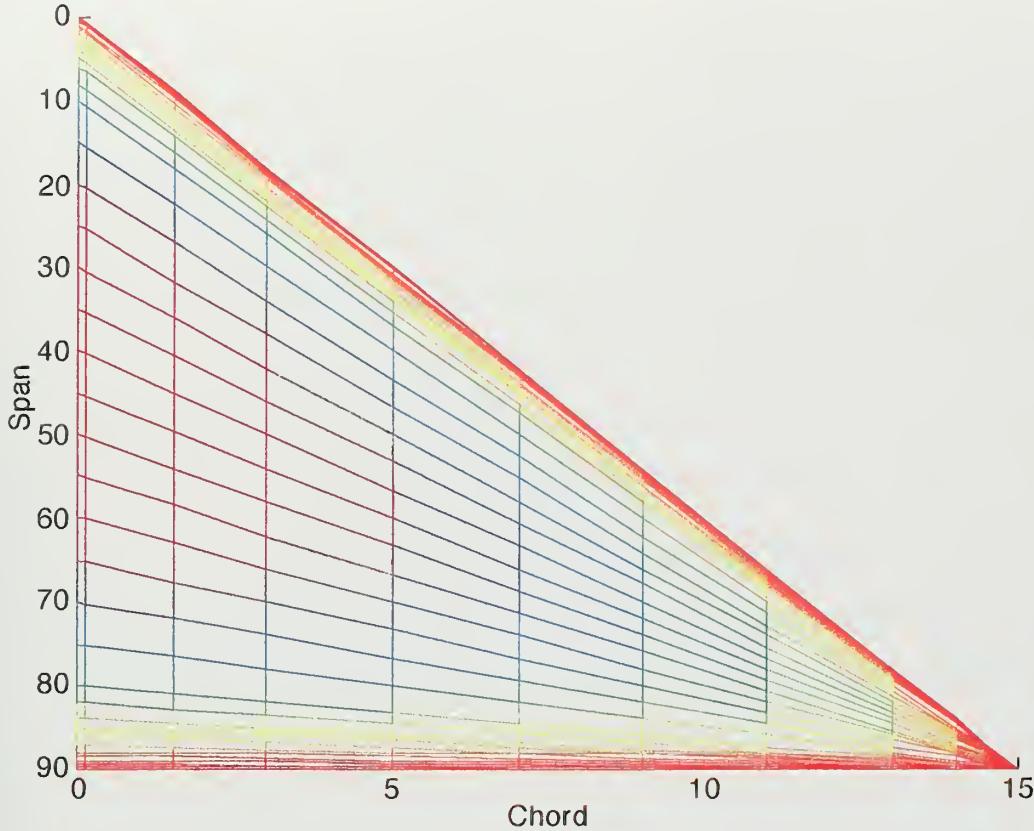


Figure 2.2 Panel Distribution of Deltawing

### 3. GBU-24

Wind tunnel experiments were run on the GBU-24 without canards attached, so it was deemed relevant to build a model with and without canards as a comparison of the code's performance. The model of the GBU-24 was too complex to build with a spreadsheet, so MACGS was used. The bomb was modelled at NAWC Warminster by superimposing a group of networks on top of an IGS file being displayed by MACGS. The complete configuration consists of approximately 1300 panels. Figures 2.3 and 2.4 are displays of the GBU-24 with canards, with Figure 2.4 including the wakes. Figure 2.5



is included to show how the GBU-24 model was assembled. Each different color represents a network.

Several features of the geometry are relevant to point out. Two of them are modifications made to the geometry that differ from the actual dimensions of the bomb. Dr. Alex Cenko of NAWC Warminster has extensive experience with modelling stores in A502i. The modifications were made on his knowledge of how to get the most accurate results from the code when modelling stores. The first is to model the fins and canards as flat plates, i.e., no thickness, which A502i allows you to do through a single numerical change in the input code for each network that represents a flat plate (see line 30 of Appendix A). The fins and canards are extremely thin when compared to the rest of the bomb, and to add a third dimension to the geometry complicates the construction of the fin or canard for several reasons. The leading and trailing edges must be sharp and the surface the fin or canard attaches to would have to be modified to abut properly with two edges instead of one. Experience has shown that the simpler version yields accurate predictions. A502i is an inviscid code, so it cannot take into account separation effects on its own. The GBU-24 does not have a flat base. In reality, it is more bullet nosed in shape. However, at the speeds with which the bomb is being analyzed, separation does occur near the trailing edge of the bomb. Experience has shown that truncating the end into a flat base and designating it a separated flow region through an appropriate input (see line 702, column 1, Appendix A) yields better results than attempting to model the bomb to exact physical dimensions. The last feature to point out are the wakes, as seen in Figure 2.4. A502i has a limitation in that the wakes must be modelled by the user, and they have the same abutment requirements as physical surfaces. Regardless of angle of attack, the wakes remain stationary with respect to the body to which they are attached. At higher angles of attack, the wakes are no longer close to paralleling the free-stream velocity. Remodelling the wakes is nearly an impossible task. The fin and base wakes would not be too difficult



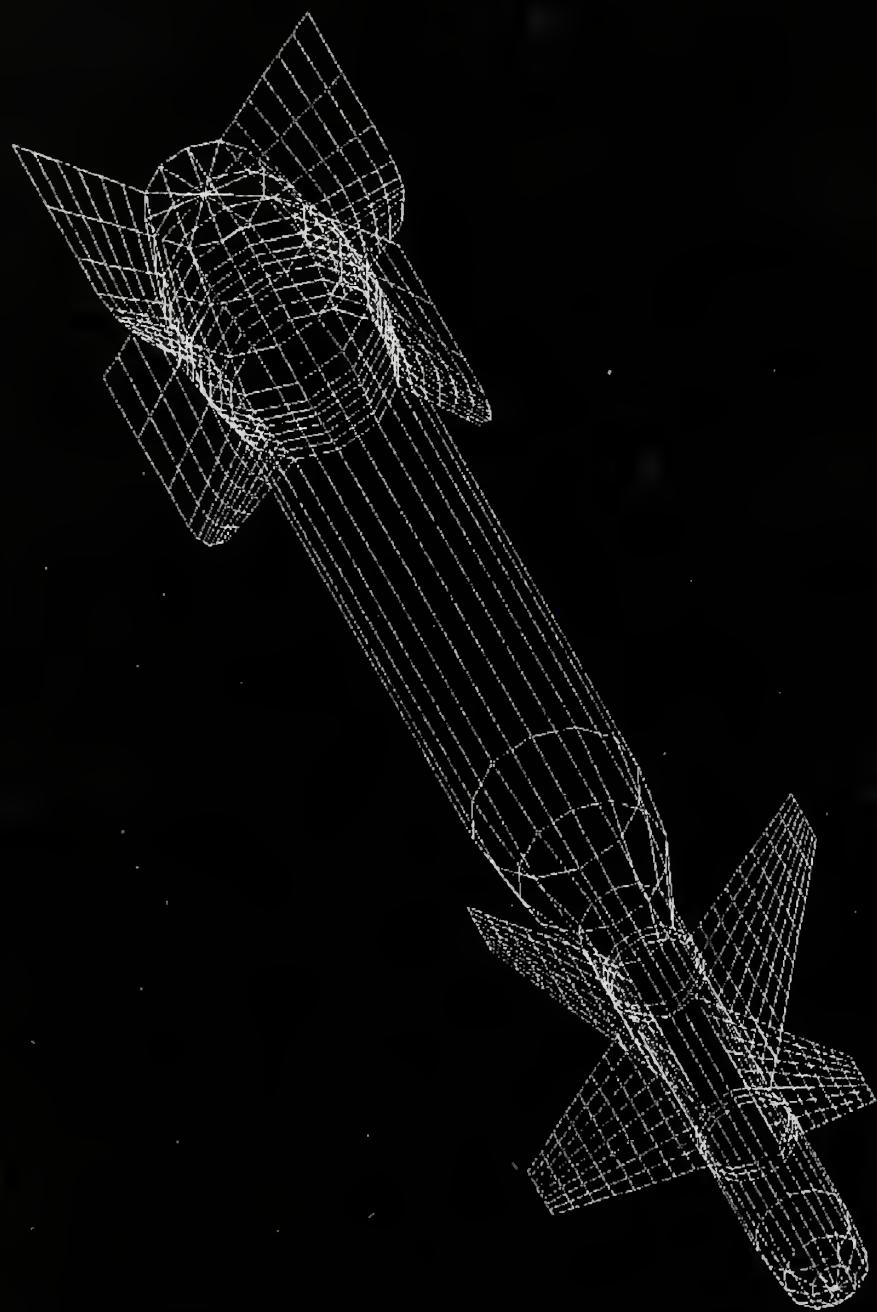


Figure 2.3 GBU-24 Geometry (Wakes not Shown)

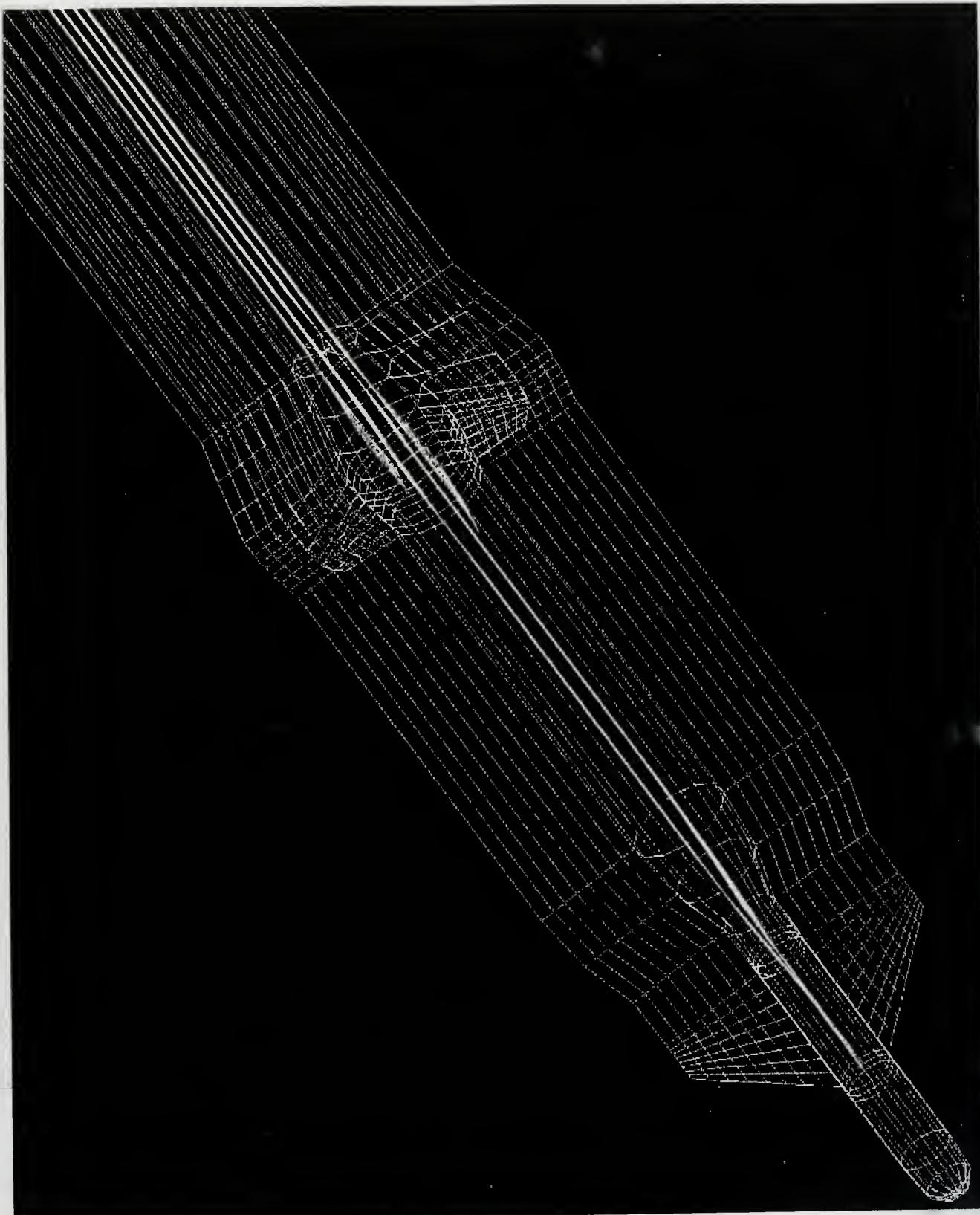


Figure 2.4 GBU-24 Geometry (Wakes Shown)

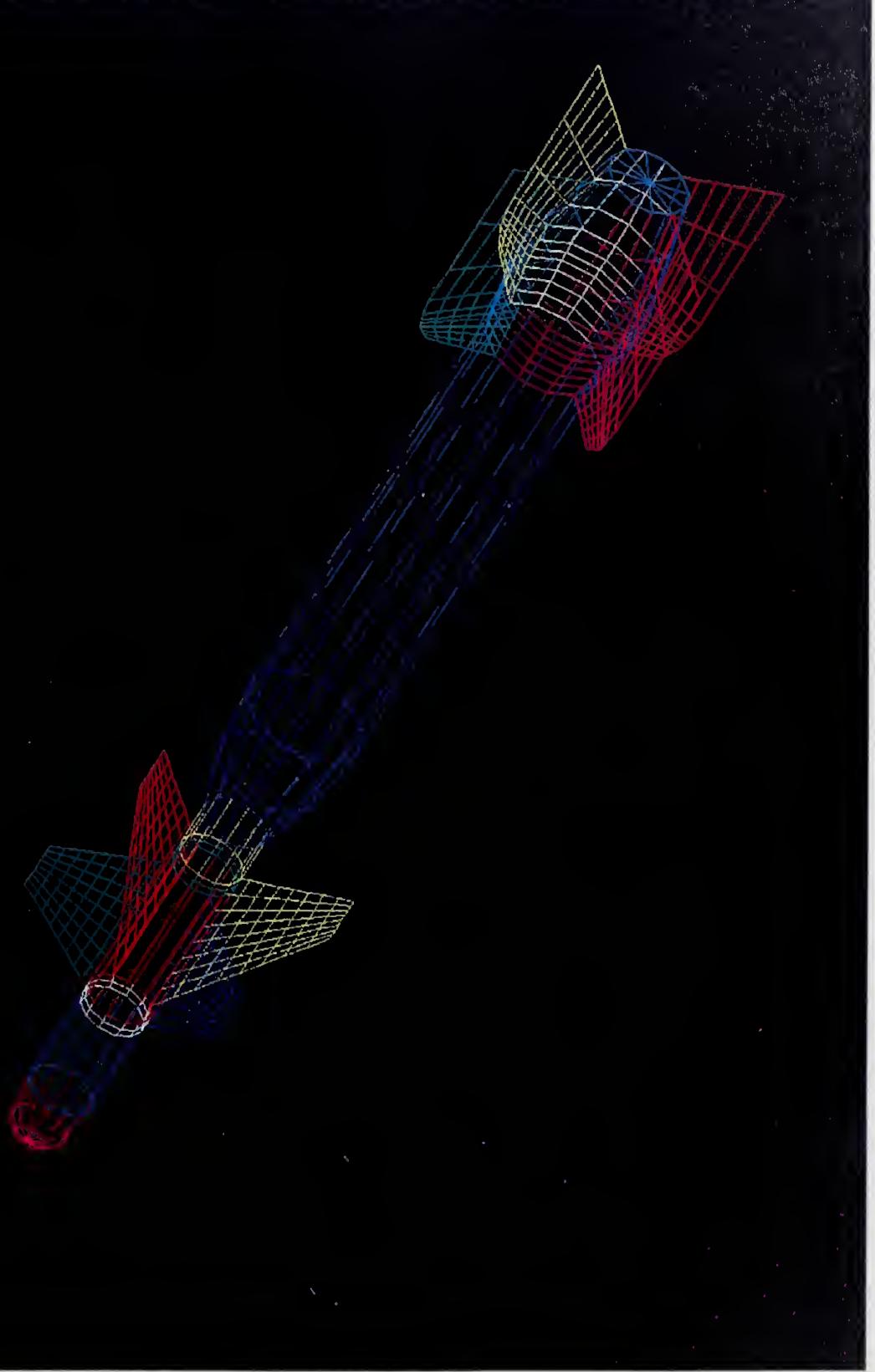


Figure 2.5 GBU-24 Geometry (Network Modelling)



because they do not abut against any physical surfaces except for the surfaces they are trailing from. The canard wakes must abut properly against the bomb's body all the way to the base. Modification of those wakes would entail modification of the entire body, or in a possible simplification, letting the wakes remain attached along the body until the base of the bomb and then shifting them relative to the free-stream.

#### **4. The F-14**

The F-14 geometry was modelled in the same fashion as the GBU-24. The geometry consists of approximately 1500 panels. While that may seem fairly coarse for such a complex structure, experience shows that it is all that is required to get accurate predictions. The primary area of interest is the underside of the fuselage forward and between the two nacelles. Higher panel density on the top half is not required. Figures 2.6 and 2.7 display the F-14 geometry without and with wakes shown. Several omissions are made to the model as having a trivial effect on the analysis or no effect at all. Phoenix rails and bomb racks are not modelled along with the chin pod because they are deemed insignificant to achieve reasonably accurate predictions over small angles of attack. External tanks were not considered, but could be modelled much in the same way as the bomb and inserted into the input file to see what effects the drop tanks have on separation forces. The vertical tails and horizontal stabilizers were deemed irrelevant to the prediction of the separation forces and were left out. This reduces the number of panels and networks, which also reduces the amount of time it takes to run a solution.

#### **5. Combination Geometries**

The F-14 and GBU-24 were modelled separately, but were combined together as shown in Figure 2.8. The first step to accomplish this was using the FORTRAN code NAVSEP which, among many of its functions will translate coordinates to relocate items in the flow-field. Once accomplished, the GBU-24 file was pasted into the F-14 input file.



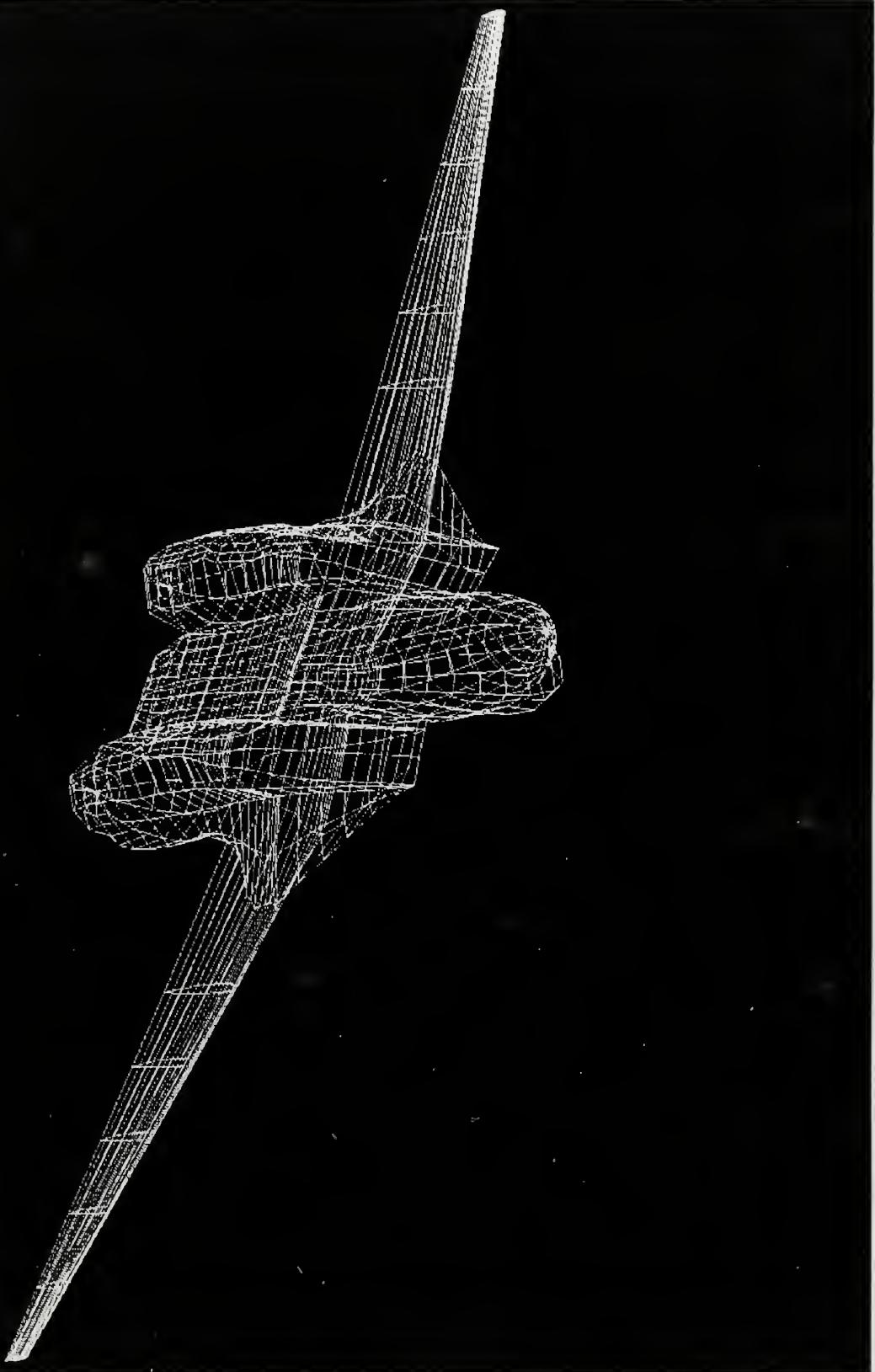


Figure 2.6 F-14 Geometry (Wakes not Shown)

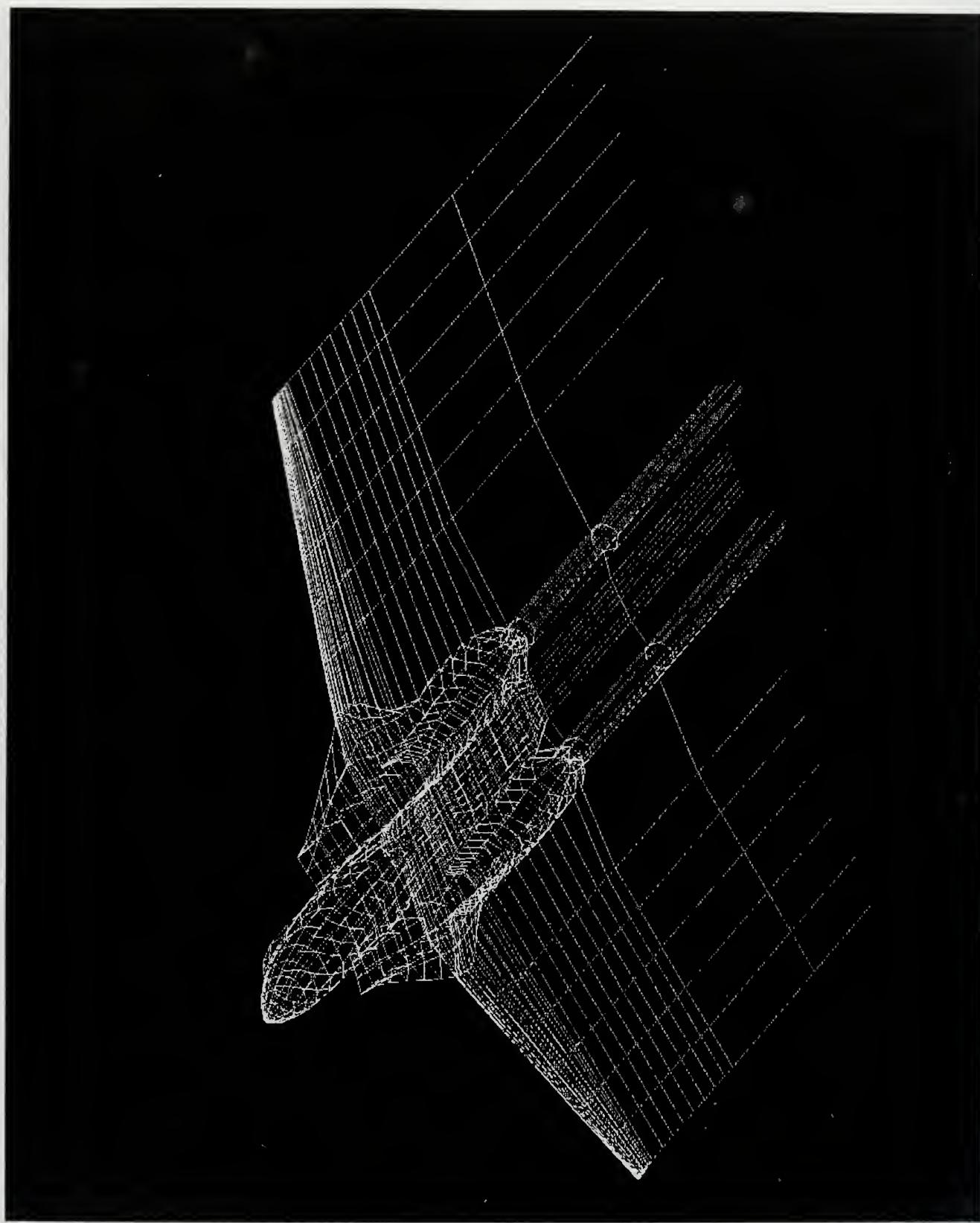


Figure 2.7 F-14 Geometry (Wakes Shown)

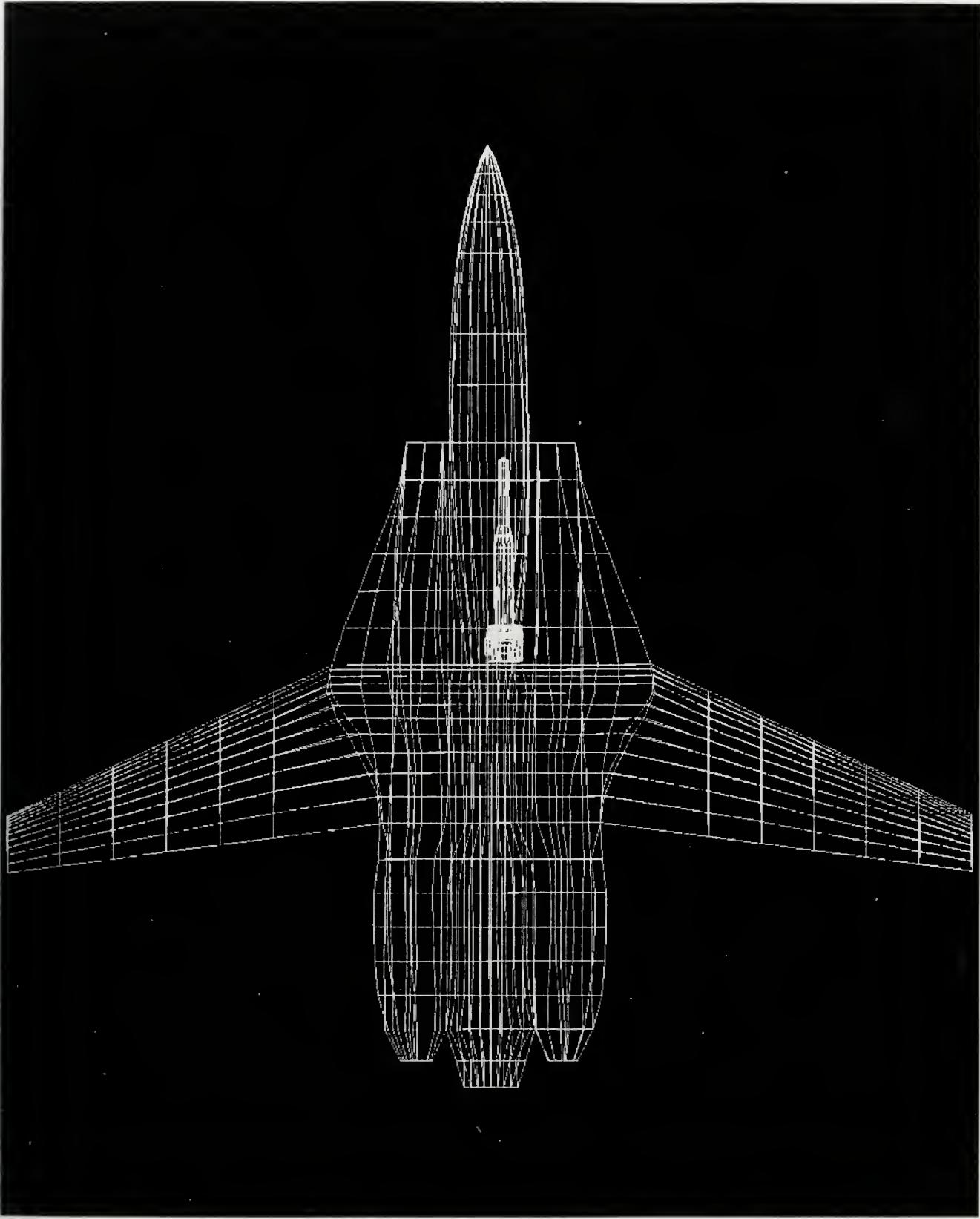


Figure 2.8 F-14 and GBU-24 on Station 3 Combined Geometries

## **D. GRAPHICS VISUALIZATION**

One of the pre/post-processing codes that came with A502i is called RAID. Currently an executable exists in the department's computer system that can be accessed through typing after the UNIX prompt,

**raid**

Raid is a basic graphics program that can read A502i and TRANAIR input files and display geometries and flow properties from solutions. After accessing raid, it will ask what type of input file it is being asked to visualize. It can handle five other modes of input besides A502i/TRANAIR [Ref. 10]. It will then ask for the name of the input file. A prompt will follow asking about object definition matrices which needs to be answered by,

**EACH**

The next prompt will ask if the panels are going to be shaded by a Cp value. Cp is used generically in that Cp can be displayed or Mach or any of the relevant 49 surface flow properties [Ref. 4]. This is only used after a solution has been run, data has been extracted and a colorscale for contour plots has been determined. If only the geometry is to be displayed, then hit carriage return for all the next questions until a pink window appears with a menu in the lower left corner of the window. An anomaly of the program is that if you want to display the wakes, then wake display must be deselected. To view the geometry select view on the menu. All selections with the mouse in RAID are made with the center mouse button. A new window appears with a menu bar at the bottom and left and the geometry in the center. From there, rotation, translation, scaling, axes, reflections and other manipulation of the geometry is possible. Figures 2.3 through 2.8 are examples of geometries displayed on RAID.

When presentation of solutions (i.e., Cp or Mach contours) are desired, the use of another post-processing program is required to generate a colormap file. The program is

called crebar. An executable currently exists on the department's system. Type in,

**crebar**

after the UNIX prompt. The program will ask straightforward questions. Number values associated with colors available can be found in Ref. 10. The color file can be saved under any name, but must lie in the same directory as the input and solution files. The first line of the color file will list four numbers. The first number is the number of colors assigned to the colormap (248 maximum). Occasionally, the color bar displayed in RAID when using a colormap will disappear when certain menu items are selected. To prevent this, change the last three numbers to read 6, 1, -1. Plotting outputs from RAID requires saving the file in a format, such as RGB, that a printer will recognize. It is possible to change the text color and background color, the default is black, to avoid excessive use of black ink in hard copies.



### **III. DATA EXTRACTION**

When a solution has run to completion, there are two files of interest, the arbitrarily named output file and the ft13 file. The arbitrarily named output file contains results for everything that A502i solves for. The ft13 file contains only the 49 surface flow properties on each panel. Appendix D is the solution portion of an arbitrarily named output file for the first network.

For purposes of displaying flow properties on RAID, it is necessary to utilize the ft13 file. A post-processing code called RAIDCONV is used to extract the specific information. To access RAIDCONV, type

**raidconv**

after the UNIX prompt. An executable currently exists in the department's system. The ft13 file must be in the same directory as RAIDCONV is accessed in. RAIDCONV will prompt the user for which kind of panel method is being used (A502i is one of three choices). The next prompt will ask for the name of the ft13 file. The last prompt will ask for the flow property that is to be extracted. A file called ft13.cp will be created. It can be renamed for purposes of multiple flow properties extraction. Abbreviations for the 49 flow properties can be found at the bottom of page 1 of Appendix D. The two primary flow properties are

**LMACHU** for local Mach number

**CP2ND** for second order pressure coefficient

CP2ND is the default setting for RAIDCONV. Once the ft13.cp file is created, RAID can be used as previously discussed to display the flow properties. An anomaly of RAIDCONV is that it does not recognize kt=20 type wakes, used where wakes from a wing abut against a body [Ref. 4]. To assist in extracting all the data, the kt=20 wakes

should be placed at the end of the input file. In general, it is good practice to place all wakes at the end of the input file when using A502i.

The arbitrarily named output file duplicates the data found in the ft13 file and includes moments and forces. A502i will sum up the moments and forces on each network and for all networks so far [App. D]. The moments are computed based on the coordinates entered into the input file [Ref. 4 and App. A].

## IV. RESULTS OF A502i COMPUTATIONS

### A. PARABOLIC ARC AIRFOIL DISCUSSION

This simple geometry was analyzed primarily to evaluate A502i's capabilities by a comparison to known linear theory. To this end, the geometry discussed in section II-C and shown in Figure 2.1 was run by A502i at a Mach of 0.3 and a Mach of 1.5 at an angle of attack of zero. Two of the 49 flow properties that A502i computes [Ref. 4] for each panel are linear  $C_p$  and second order  $C_p$ , given by

$$CPLIN = -2u_c \quad (4.1)$$

$$CP2ND = -2u_c - [(1-M_\infty^2)u_c^2 + v_c^2 + w_c^2] \quad (4.2)$$

Where  $u_c$ ,  $v_c$  and  $w_c$  are the compressible components of the perturbation velocity. Figure 4.1 plots the linear theory, A502i linear and second order results for the subsonic case, while Figure 4.2 represents the supersonic solution.

Linear theory for parabolic arc airfoils is outlined in Refs. 2 and 3. The equation representing the subsonic case is given by:

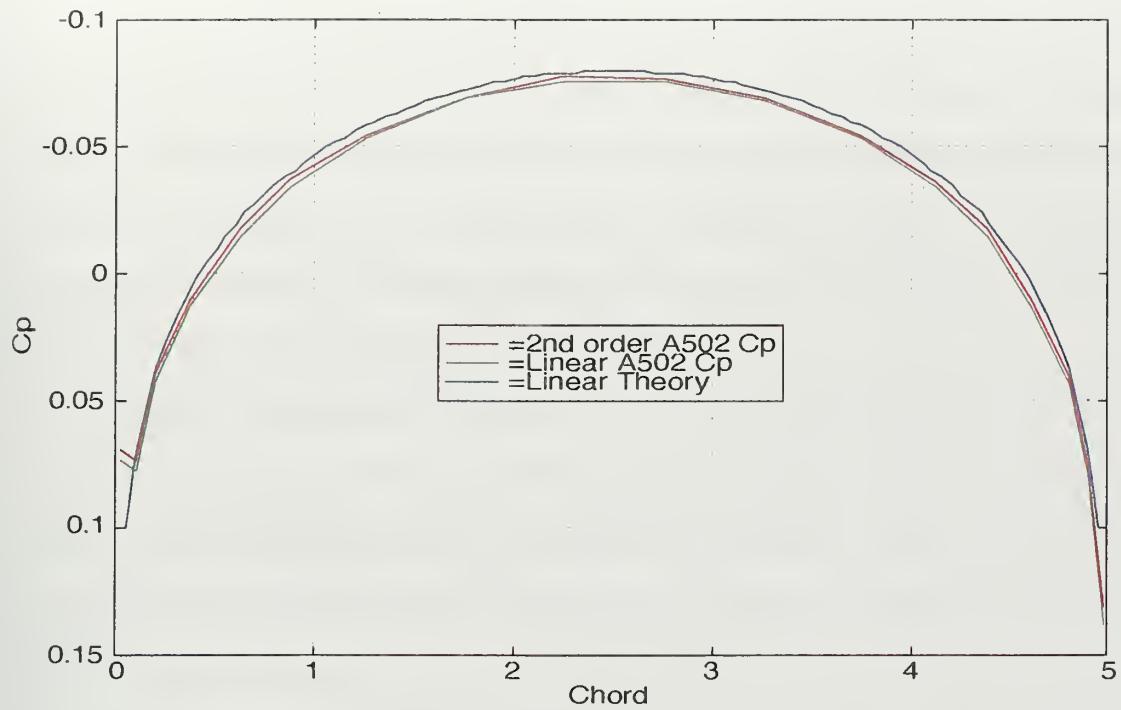
$$Cp(x) = \frac{-8 * \tau_{max}}{\pi * chord * \sqrt{1 - M_\infty^2}} * (1 - (.5 - x) * \ln \left| \frac{1 - x}{x} \right|) \quad \text{where } 0 < x < 1 \quad (4.3)$$

Equation 4.3 includes a Prandtl-Glauert compressibility correction. The equation representing the supersonic case is given by:

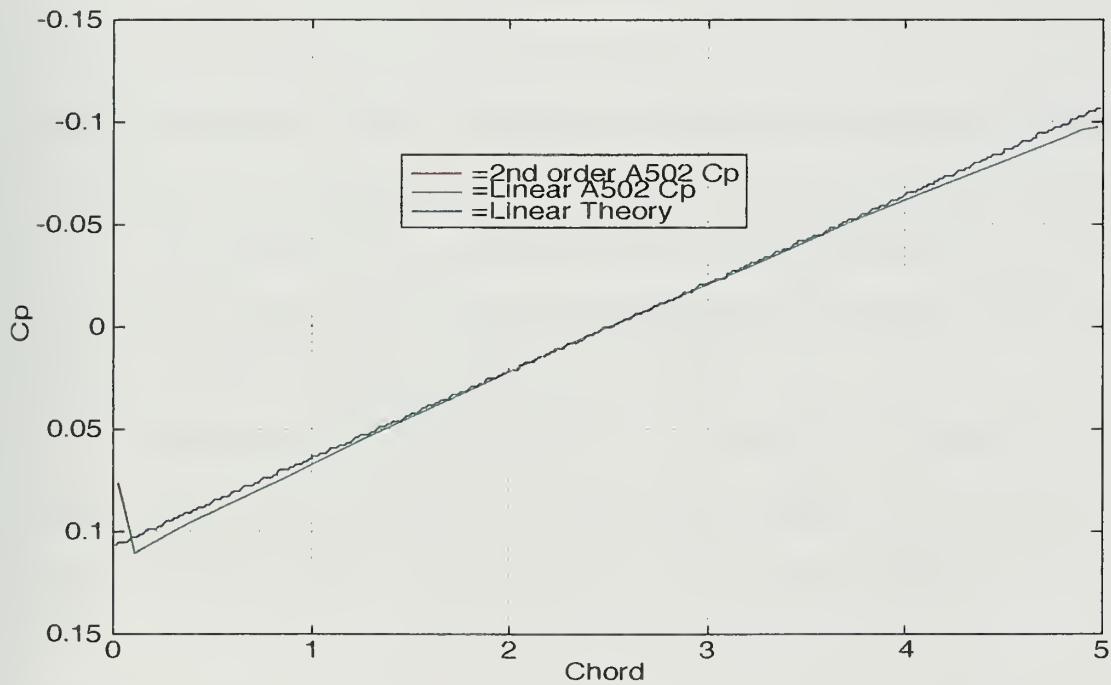
$$Cp(x) = \frac{2\theta}{\sqrt{M_\infty^2 - 1}} \quad \text{where } 0 < x < chord \quad (4.4)$$

and  $\theta = \tau_{max} * (\frac{1}{chord} - \frac{2x}{chord^2})$





**Figure 4.1 Cp Comparison of a Parabolic Arc Airfoil at Mach = 0.3.**



**Figure 4.2 Cp Comparison of a Parabolic Arc Airfoil at Mach = 1.5.**



Figure 4.1 shows very good agreement with the linear theory curve. There is a small but noticeable difference between A502i's linear results and second order results, with the second order results being more accurate, as expected. The maximum difference between the linear theory curve and the second order A502i curve amounts to a value of 2.5% right at mid-chord. The gap between the two curves from .25 chord to .75 chord were the result of thin panel density in that region.

Figure 4.2 shows excellent agreement with the linear theory curve. The A502i values of  $C_p$  for the linear and second order analysis are virtually identical. There are small deviations from linear theory near the leading and trailing edges, but this is expected due to numerical error associated with the discontinuity A502i would encounter right on the leading or trailing edges.

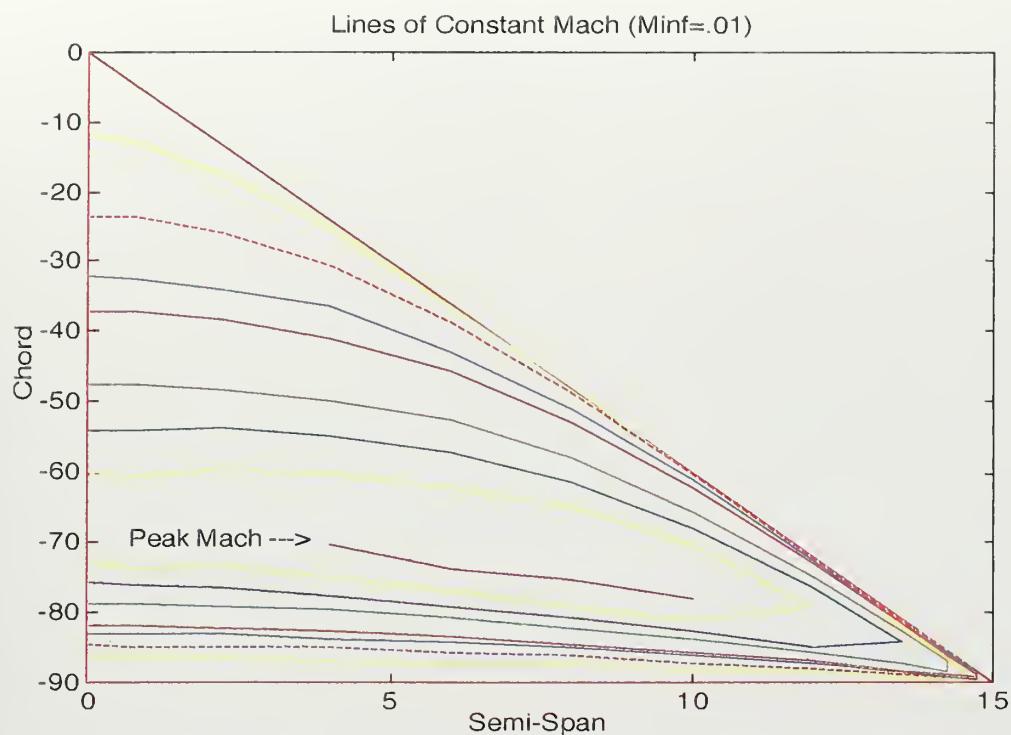
## B. DELTAWING DISCUSSION

Reference 9 provides data on Mach distribution, using approximated linear theory, over a deltawing of the configuration discussed in section II-D. This simple geometry provided another test of A502i's capabilities. Figures 4.3 and 4.5 show the A502i results for the subsonic and supersonic case, while Figures 4.4 and 4.6 reflect the results from Ref. 9. For both cases, good agreement is found with the linear theory, with A502i's subsonic analysis being physically more accurate than the approximate linear theory, while A502i's supersonic analysis is not as physically accurate.

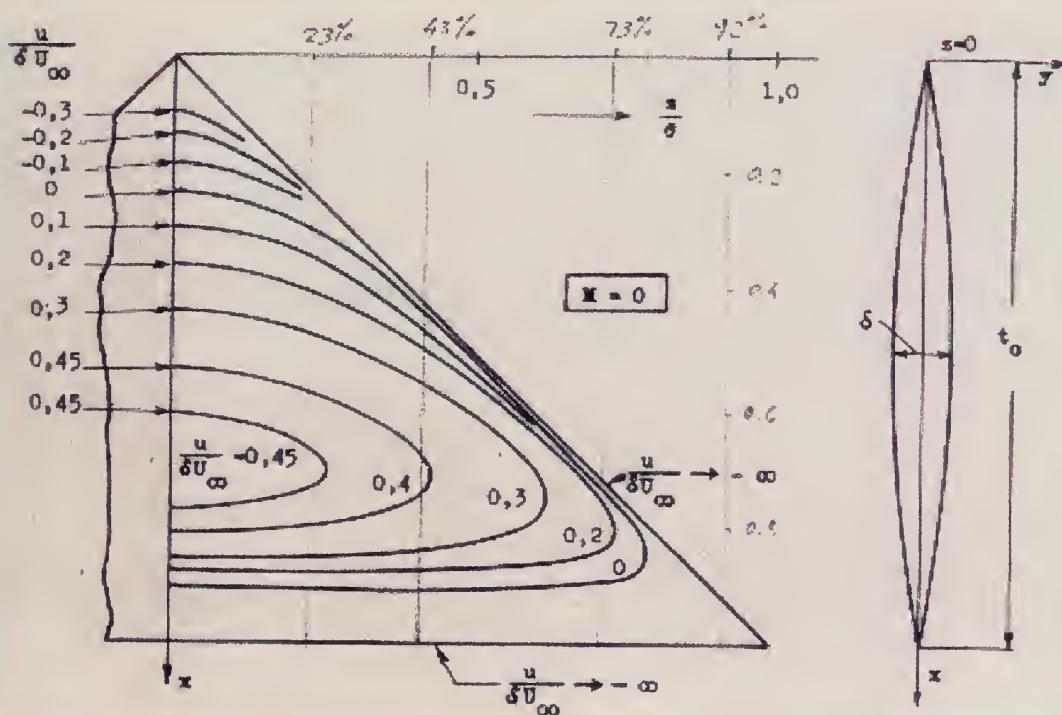
A comparison of Figures 4.3 and 4.4 reveals several points of interest. The Mach contour representing the free-stream value is given by the dashed line. All lines outside the dashed line represent areas where the Mach value is less than free-stream, and inside the dashed line is where the Mach value is more than free-stream. The location of where the free-stream Mach contour, in Figure 4.3, intersects the centerline agrees very well with Figure 4.4. However, Figure 4.4 does not have the contour extending all the way to the

tip. This is a physical limitation of the approximate theory used in Figure 4.4 and A502i is giving a more realistic solution. Figure 4.4 suggests that the peak Mach value occurs at approximately two-thirds chord along the centerline. The A502i results show the peak Mach contour occurring out midway along the semi-span. Those Mach values are less than 1% larger than the yellow Mach contour surrounding it, and can be attributed to how the panel density increases with movement towards the wingtip. A502i performed very well for this subsonic case.

A comparison of Figures 4.5 and 4.6 shows that A502i did not perform as well as in the subsonic case. Again, the contour representing the free-stream value of  $\text{Mach} = 1.414$  is given by the dashed line. All lines forward are below free-stream and all lines aft are above free-stream. Figure 4.6 shows the intersection of the free-stream Mach contour on the centerline occurring at approximately 39% chord, which is in excellent agreement with A502i's result. Figure 4.6 shows the peak Mach value occurring at the trailing edge on the centerline. This makes more physical sense than the results that A502i yielded. The maximum thickness of the deltawing occurs along the centerline, allowing for greater expansion. The discrepancy may be attributable to panel density and accumulation of numerical error. A close study of the A502i results reveals some discontinuities along the column of panels out at the wing-tip which would have adversely affected the solution and caused errors to propagate along the semi-span.



**Figure 4.3 A502i Mach Contour Plot ( $M_{\infty}=.01$ )**



**Figure 4.4 Approximate Linear Theory [Ref. 9]  $M_{\infty}=0$**



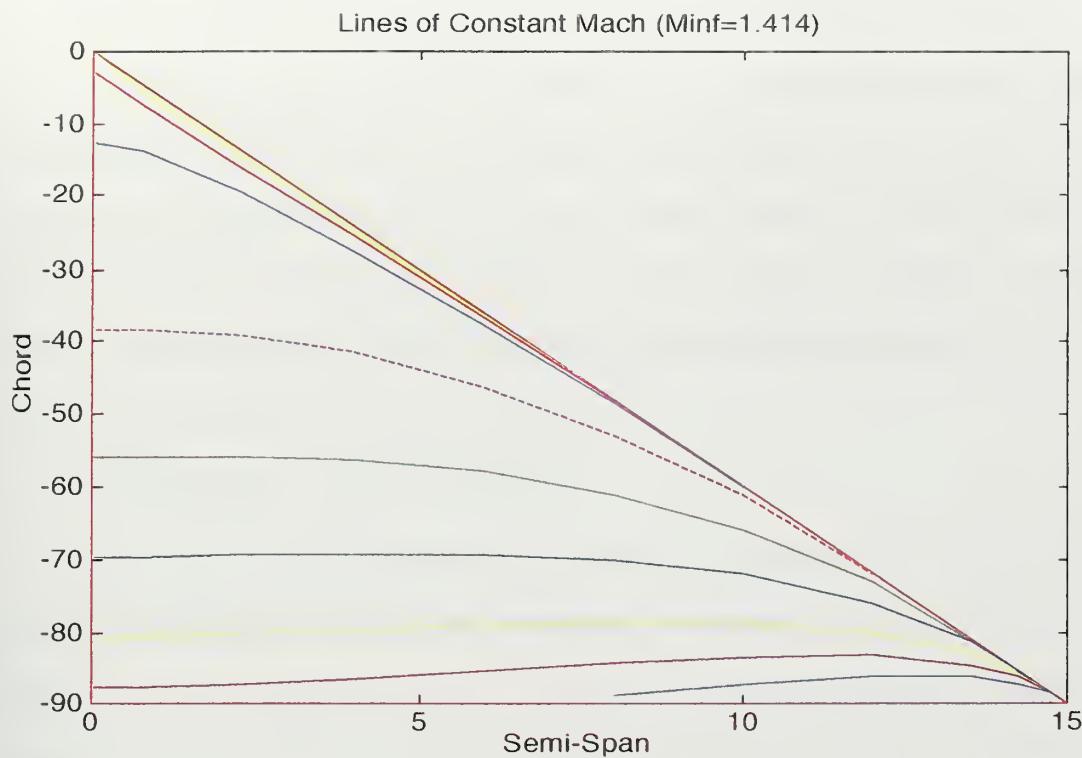


Figure 4.5 A502i Mach Contour Plot ( $M_{\infty}=1.414$ )

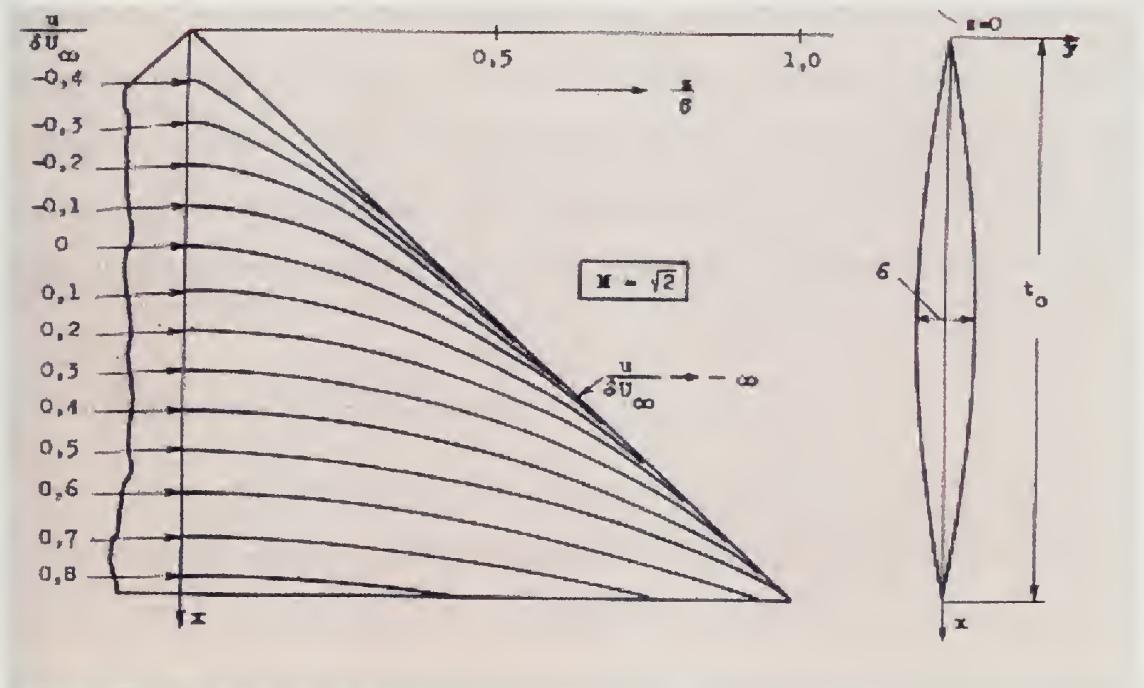


Figure 4.6 Approximate Linear Theory [Ref. 9]  $M_{\infty}=1.414$



## C. GBU-24 FREE-STREAM (NO CANARDS) DISCUSSION

A free-stream measurement of separation forces on the GBU-24, without canards, was conducted in a wind tunnel for various Mach numbers from .8 to 1.2 [Ref. 7]. Since A502i uses linear potential theory, the model of the GBU-24, without canards, was evaluated at both Mach .8 and 1.2, avoiding the transonic regime, to ascertain the accuracy of the code with the given geometry. Normal forces and pitching moments for both cases are plotted and compared to the wind tunnel data.

### 1. Subsonic Case ( $M_\infty=0.8$ )

The GBU-24 model, without the canards, was run for angles of attack varying from -10 to +10 degrees in two degree increments. Values much higher than that ran into wake modelling problems as the wake's angle relative to the free-stream was getting large enough that results would become questionable, and remodelling the wake was too difficult for such a complex geometry. The results of the A502i analysis are displayed in Figures 4.7 and 4.8. For angles of attack between -4 and +4 degrees, A502i does a good job of predicting the separation forces. The pitching moment, which happens to be unstable without the canards, is approximately linear over the -4 to +4 degree range and is the limiting factor to the models accuracy. The normal force is approximately linear over a wider range, and A502i does a good job of predicting the normal forces from -6 to +6 degrees.

### 2. Supersonic Case ( $M_\infty=1.2$ )

The results for the subsonic case demonstrated that the effective range of angle of attack that A502i needed to explore was from -6 to +6 degrees. Figures 4.9 and 4.10 plot the comparison of wind tunnel data versus A502i results for pitching moment and normal force. The results for the supersonic case are slightly better than that of the subsonic case. The actual pitching moment of the GBU-24 is approximately linear over a wider angle of



GBU-24 Free-Stream ( $M=0.8$ )  
Canards Off  
Pitching Moment

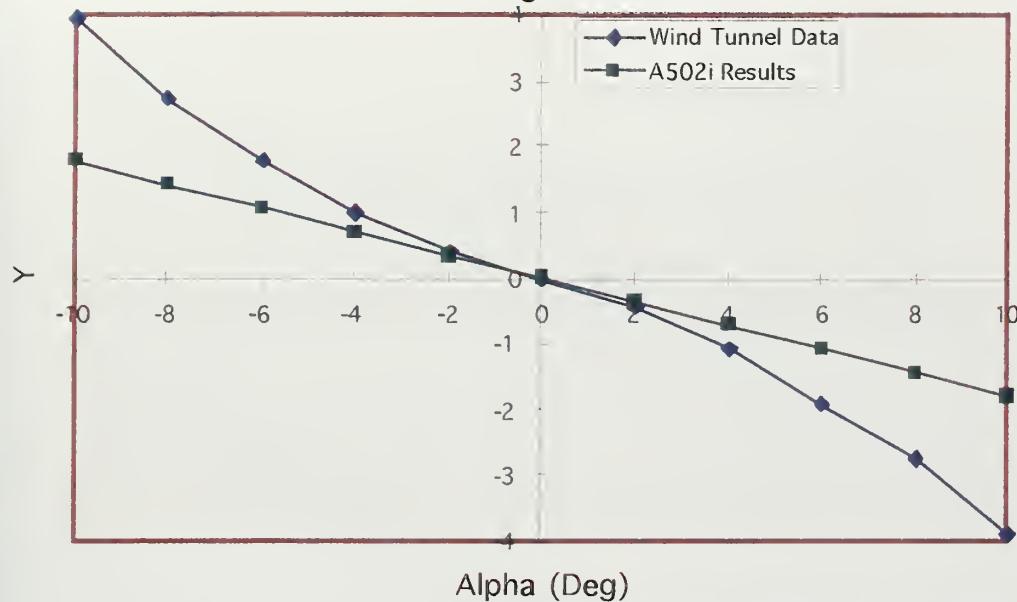


Figure 4.7 Comparison of Pitching Moments

GBU-24 Free-Stream ( $M=0.8$ )  
Canards Off  
Normal Force

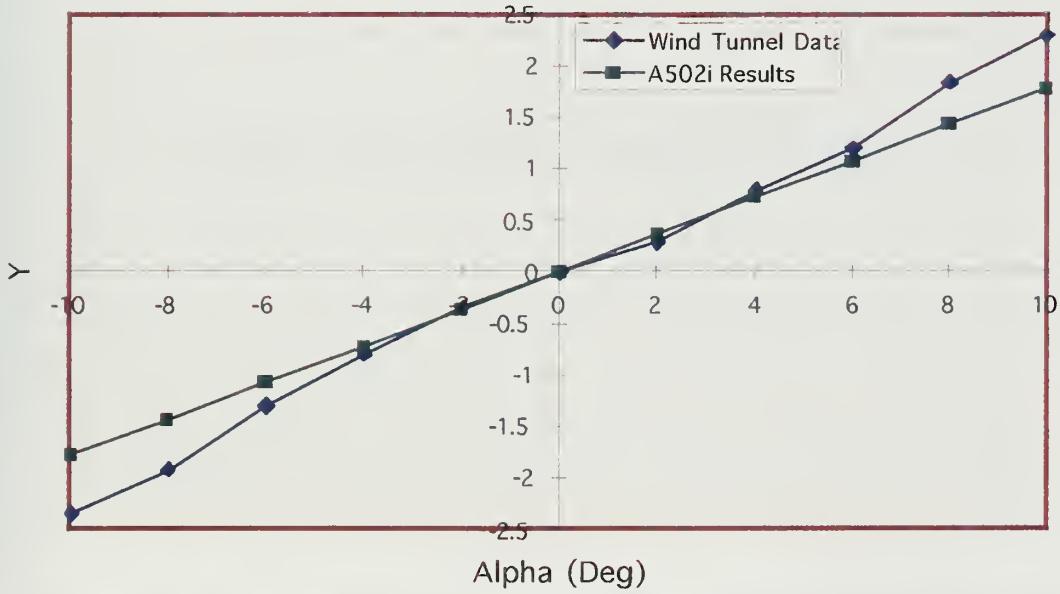


Figure 4.8 Comparison of Normal Forces



attack region, but fluctuations in the data at -6 and +6 degrees means that the model is still only viable from -4 to +4 degrees. The normal force line is nearly linear from -10 to +10 degrees and extrapolating the A502i results out to 10 degrees would still yield good predictions.

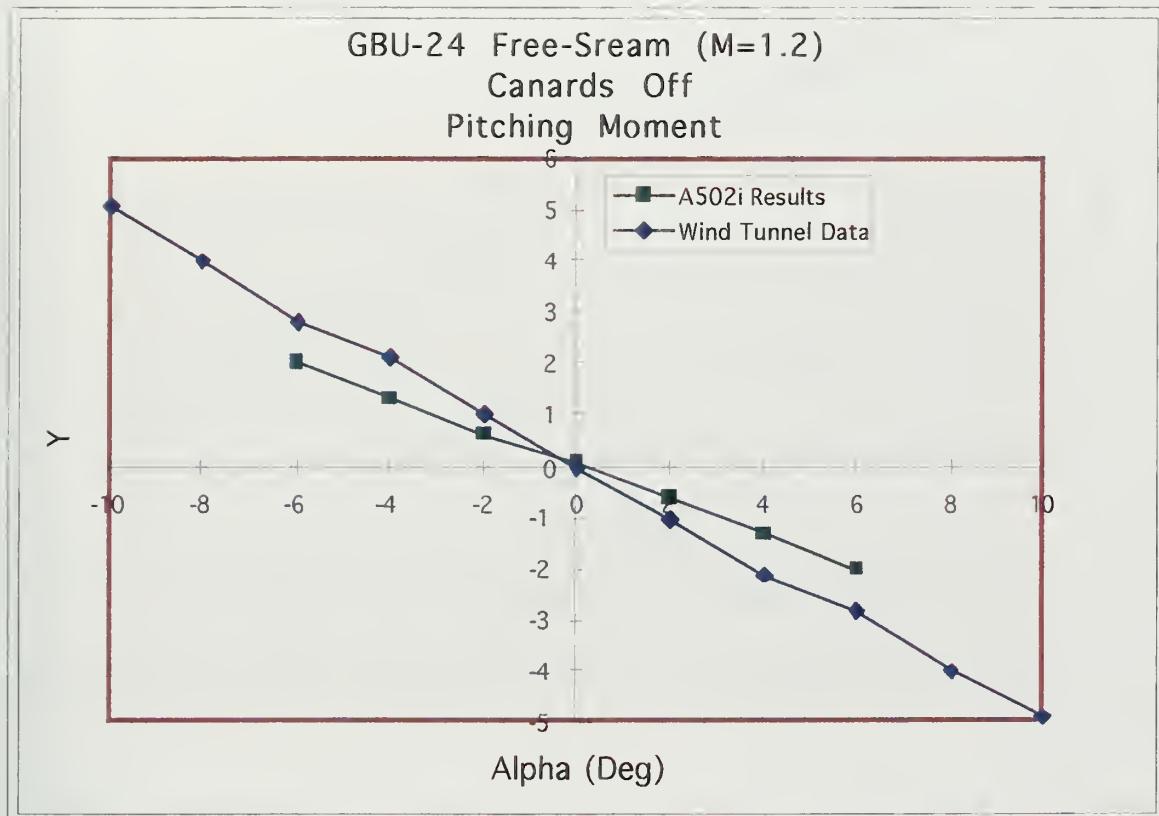


Figure 4.9 Comparison of Pitching Moments



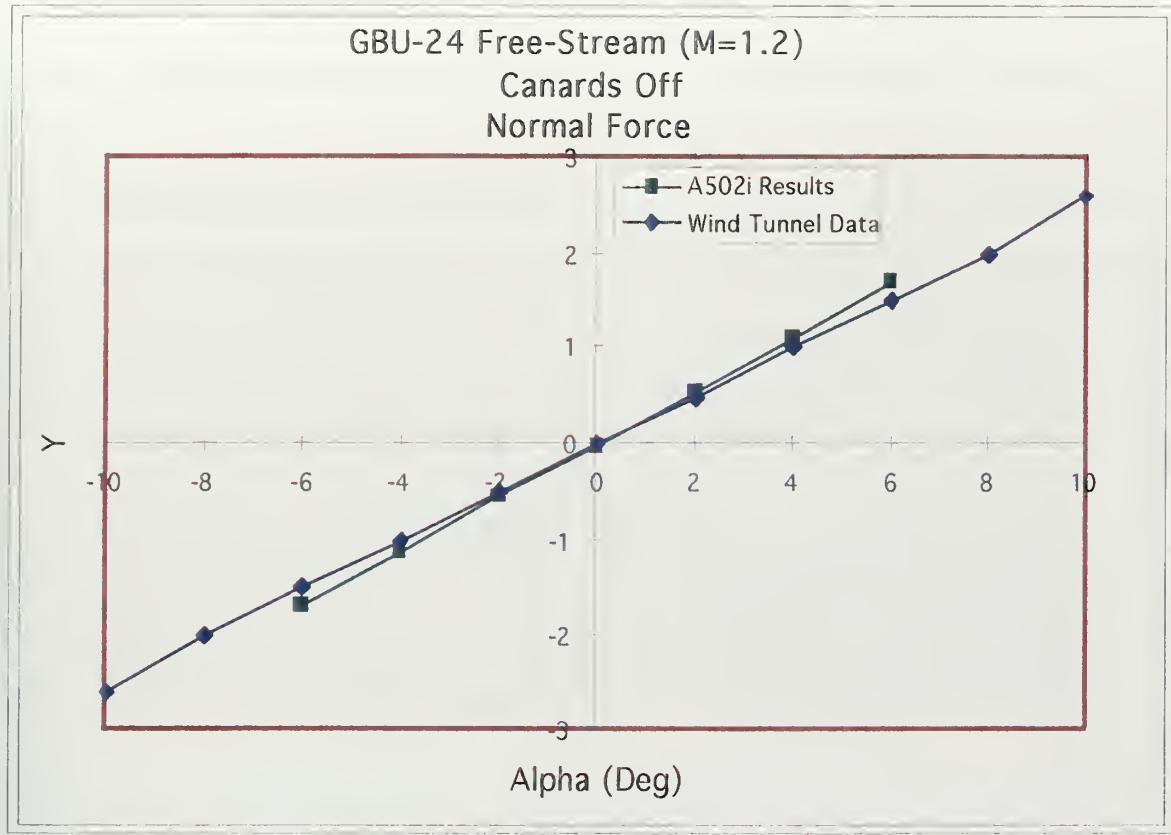


Figure 4.10 Comparison of Normal Forces

#### D. GBU-24 FREE-STREAM (WITH CANARDS) DISCUSSION

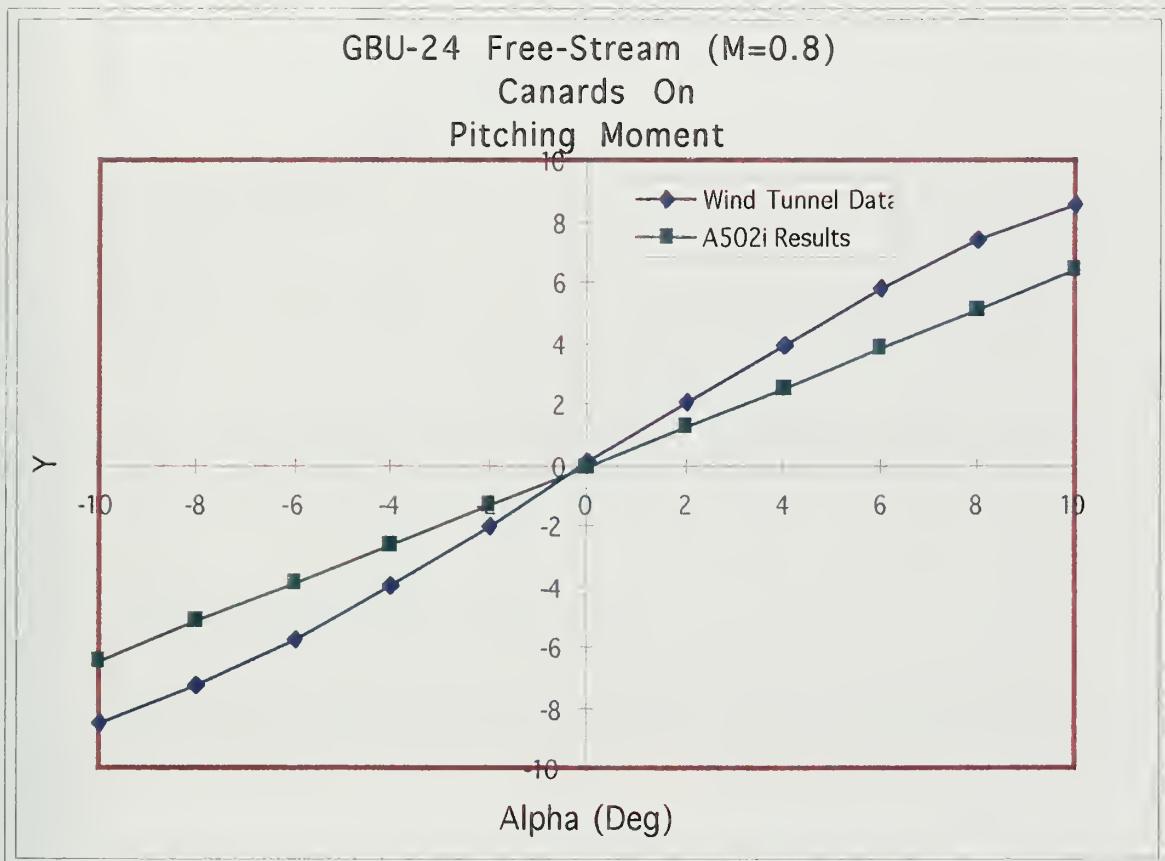
As in the case with no canards, free-stream measurements of the separation forces on GBU-24 were taken from Mach .8 to 1.2 in a wind tunnel [Ref. 7]. Again, due to the limitations of linear theory inherent in the code, an analysis was done for Mach numbers of .8 and 1.2 to minimize transonic effects. Even with the more complex geometry, A502i does an accurate job of predicting the separation forces over the range of angles of attack that are approximately linear.

##### 1. Subsonic Case ( $M_\infty=0.8$ )

The GBU-24 model, with canards, was run in two degree increments of angle of attack from -10 to +10. The wake modelling limitation, as well as reviewing the data from the wind tunnel measurements [Ref. 7] showed the non-linearity of the separation forces at

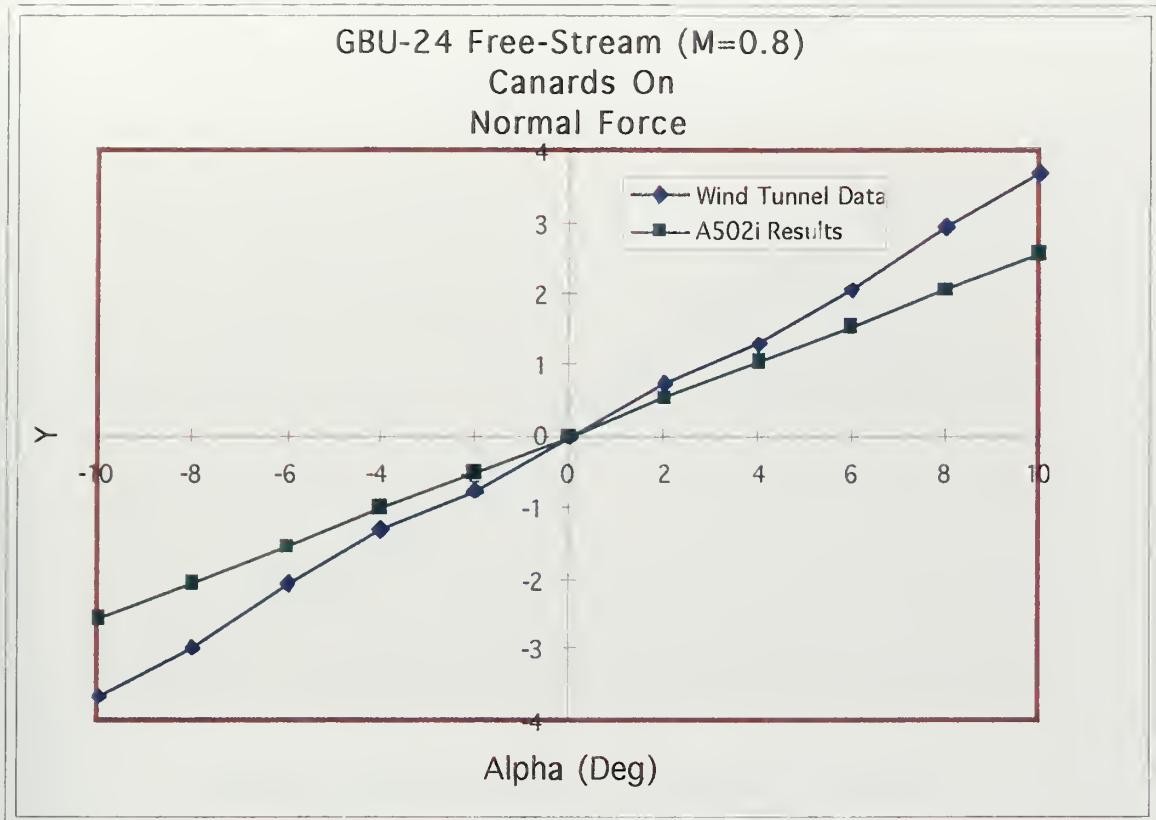


the higher values of angle of attack, precluded any attempts to predict forces beyond the aforementioned angle of attack interval. The results of the A502i analysis are displayed in Figures 4.11 and 4.12. The addition of the canards makes the pitching moment stable, but linear over a smaller region than without the canards. A502i gave accurate results from -2 to +2 degrees angle of attack when predicting pitching moment. The prediction of normal forces fared better, showing accurate results from -3 to +3 degrees angle of attack.



**Figure 4.11 Comparison of Pitching Moments**





**Figure 4.12 Comparison of Normal Forces**

## 2. Supersonic Case ( $M_\infty=1.2$ )

As in the case with no canards, the region of accuracy, with the model used, was assumed to be less than + or - 10 degrees angle of attack. Cases were run from -6 to +6 degrees angle of attack in two degree increments. A comparison of A502i results with wind tunnel data is shown in Figures 4.13 and 4.14. For both the pitching moment and the normal force, A502i does a much better job of prediction than when subsonic. The wind tunnel data is nearly linear in both pitch moment and normal force from -8 to +8 degrees angle of attack. Extrapolating the A502i data out to + or - 8 degrees angle of attack, shows excellent agreement with the wind tunnel data.



GBU-24 Free-Stream ( $M=1.2$ )  
Canards On  
Pitching Moment

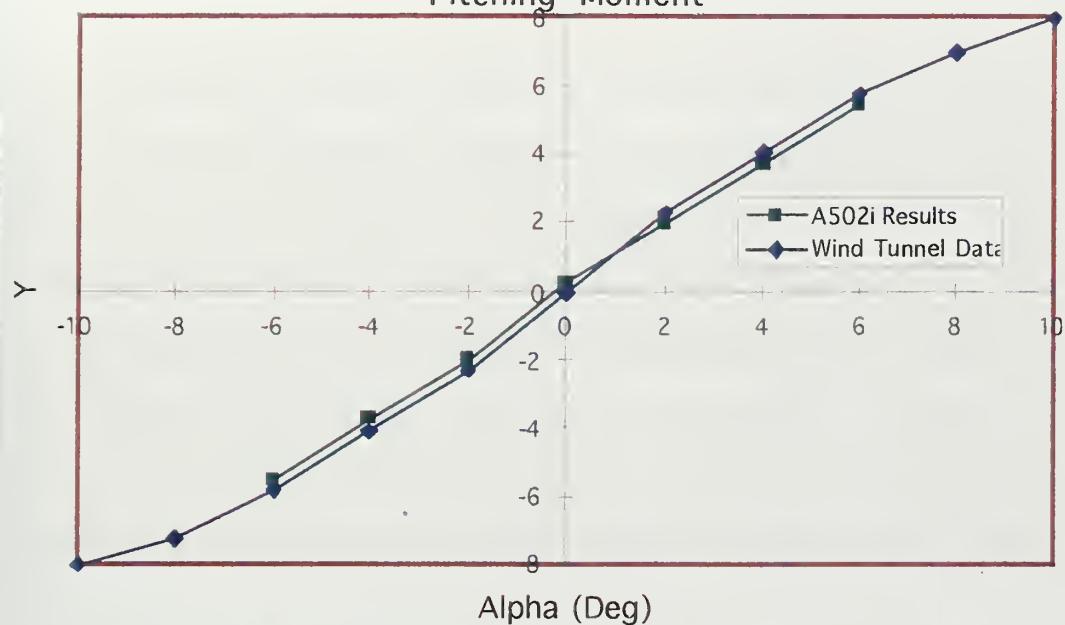


Figure 4.13 Comparison of Pitching Moments

GBU-24 Free-Stream ( $M=1.2$ )  
Canards On  
Normal Force

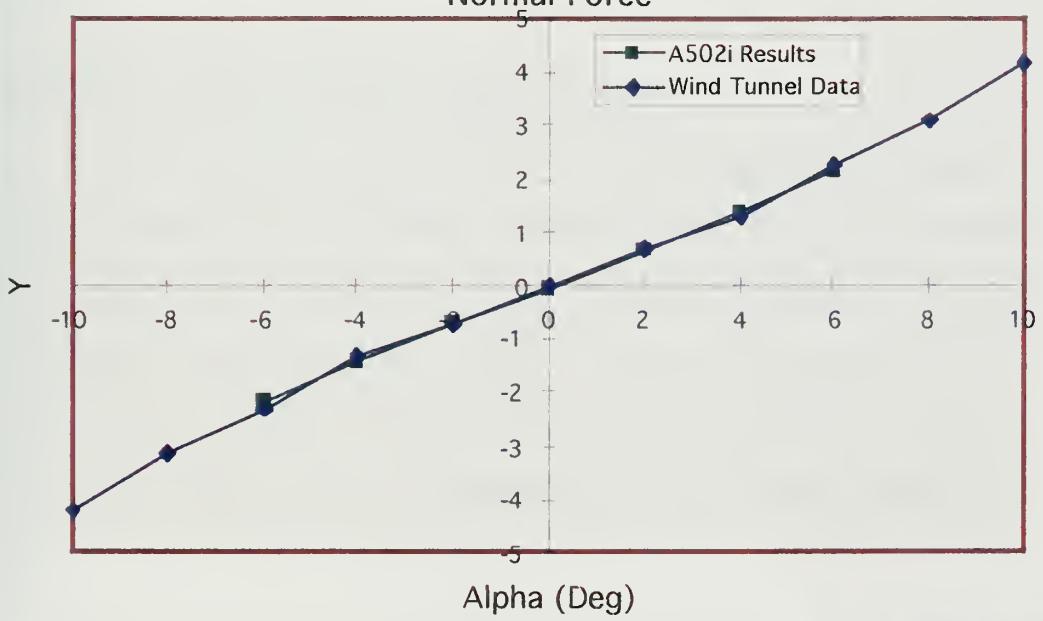


Figure 4.14 Comparison of Normal Forces



## **E. F-14 DISCUSSION**

The GBU-24, without canards was located at station 3 of the F-14 via the NAVSEP code. In the process of running a solution, the combined geometries were found to have a total of 150 networks (A502i's maximum). The combined geometry had about 4,000 panels, far short of the 20,000 panel maximum, so there was room for more detailed modelling, but there was not a chance to insert the GBU-24, with canards, into the F-14's flow field and analyze the forces on the bomb. The geometry with the canardless bomb ran to a solution that appeared to be valid, so there is a high degree of confidence that if the number of networks could be reduced to allow the GBU-24, with canards, to be inserted into the F-14 flow field, the code would yield accurate predictions at small angles of attack on the forward stations. To reduce the total number of networks by combining existing networks would have required a large time investment and the use of MACGS, which the department currently does not possess, the two reasons why it was not done. Figure 4.15 shows a Mach distribution of the solution of the canardless bomb and F-14 at Mach = 0.8 and 0 degrees angle of attack

## **F. POST-PROCESSING DISCUSSION**

The Mach values for the subsonic case of the GBU-24, with canards, at 4 degrees angle of attack, were extracted from the ft13 file. These values, used in conjunction with RAID are shown in Figures 4.16 and 4.17. The color distribution over the nose in Figure 4.16 indicates that the bomb is at an angle of attack, and scanning the rest of the model showed no discontinuous solutions, which is generally represented in A502i by a Mach value of 0 or 1,000. The visual representation is a quick way of telling if A502i ran an accurate solution. The only other way is to individually check the Mach or Cp values of each panel in the ft13 file or the arbitrarily named output file. The other point of interest in Figure 4.16 is the lack of panel density along the mid-section of the bomb. The goal, in the



case of stores separation prediction, is to have as simple a model as possible that still gives accurate predictions. The fewer the number of panels, the shorter the run time. The fact that A502i is a higher order panel method allows the luxury of using fewer panels. Figure 4.17 highlights the approach used to take into account separation effects as discussed on page 16.



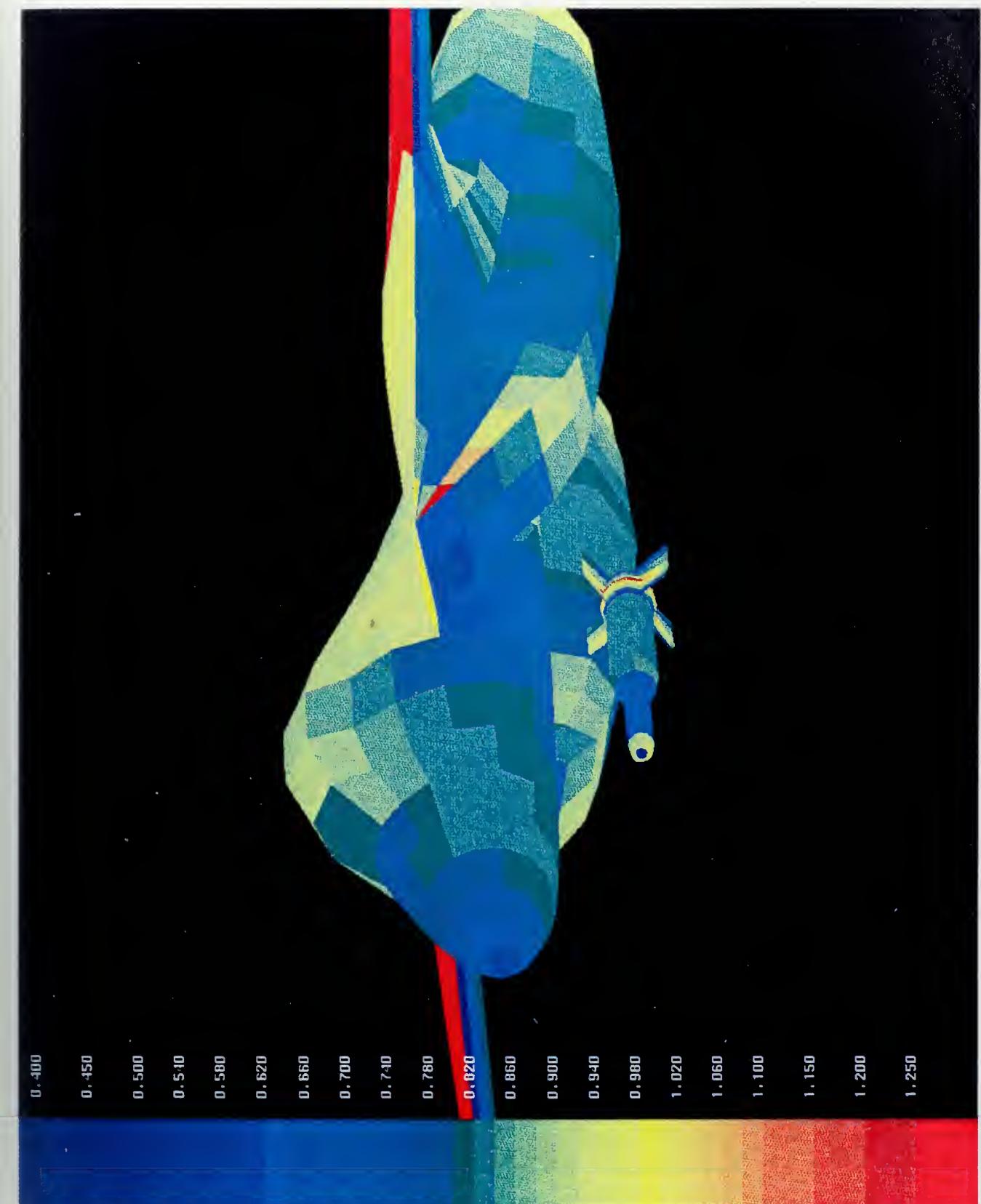


Figure 4.15 Mach Distribution over GBU-24 and F-14 ( $M_{\infty}=0.8$ ,  $\alpha=0^\circ$ )



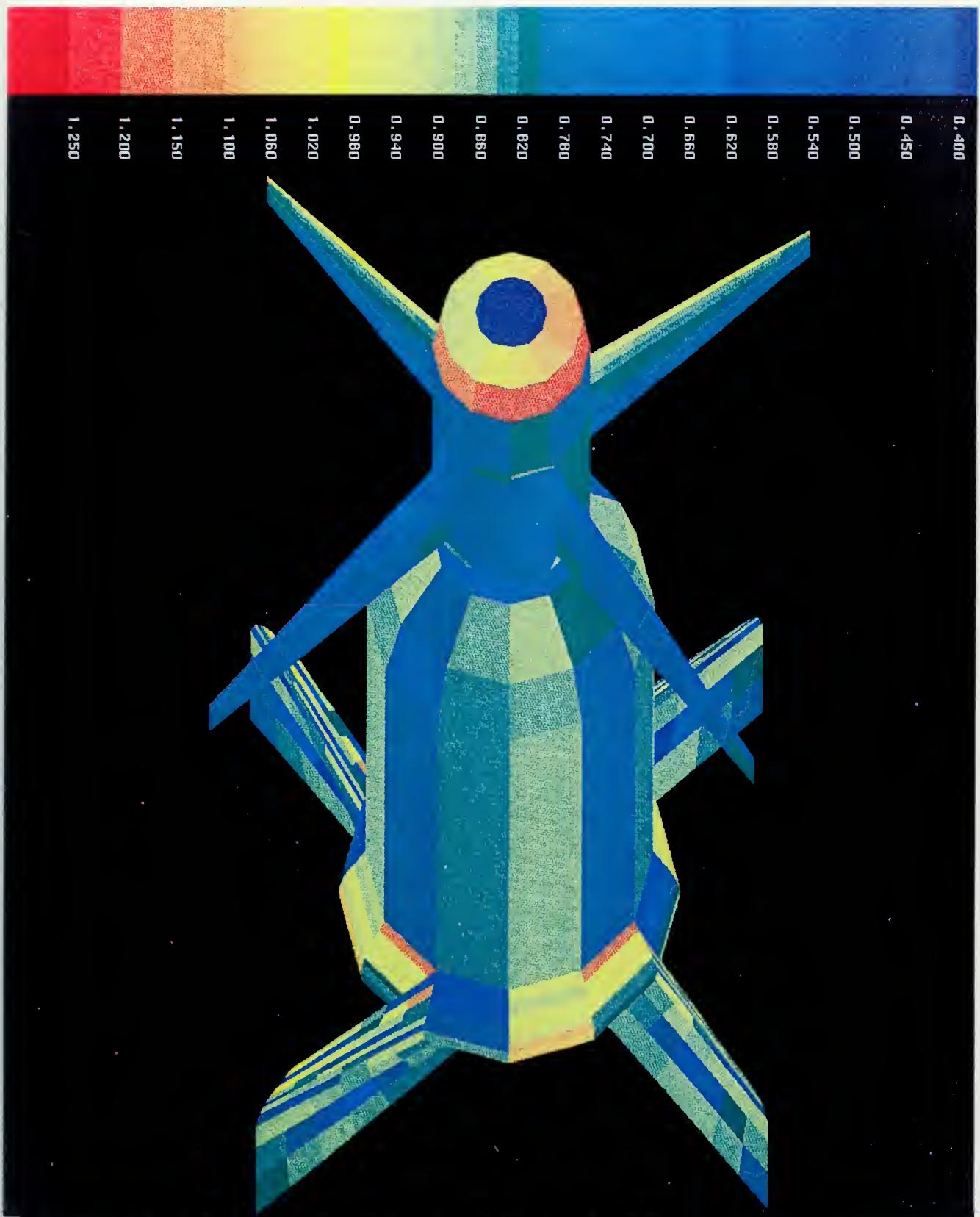


Figure 4.16 Mach Distribution over GBU-24 ( $M_{\infty}=0.8$ ,  $\alpha=4^\circ$ )





Figure 4.17 Mach Distribution over GBU-24 ( $M_\infty=0.8$ ,  $\alpha=4^\circ$ )



## V. SUMMARY AND CONCLUSIONS

The main goal of this analysis is to determine the accuracy of A502i on both simple geometries and complex geometries. To accomplish this, A502i is compared with results available from linear theory and wind tunnel experiments. This allows conclusions to be drawn on the capabilities as well as the limitations of A502i.

In general, A502i can accurately predict flow properties, forces and moments on simple and complex geometries at low angles of attack. The predictions are valid over a wide Mach range, from 0 up to and including 0.8 and from 1.2 and above. The supersonic solutions are available due to A502i's higher order capabilities.

The limitations of A502i are consistent with most panel methods. A502i cannot predict flow dominated by viscous, separated or transonic effects. It cannot predict flow with different total pressures, such as flow properties inside a jet plume or a propeller slipstream swirl. The biggest shortcoming of A502i is its inability to handle unsteady cases and automatically determine wake shapes.

Experience or knowledge of the flow properties around the geometry being tested is important in building an accurate model. An accurate model may not be physically accurate. Flight test results revealed a yawing moment on the GBU-24 that was not discovered in wind tunnel experiments when the bomb was carried on an aft station. The yawing moment may be caused by the fact that the canards are not fixed, but spring-damped. A502i predictions would not be accurate without inputting a moment to simulate the deflection of the canards, since the canards are fixed by the geometry. A502i, while a powerful tool in terms of cost savings and time, cannot completely substitute for wind tunnel experiments and flight tests, as constructing a complex geometry to exact physical specifications will probably not yield accurate predictions.



## APPENDIX A. GBU-24 OUPUT FILE (INPUT FILE PORTION)

Printed by maletour from osprey  
Page 1

Mar 7 1996 13:51

Results

```
***** dynamic memory management initialization *****
max no. levels      15   max no. arrays      200   maximum scratch storage    900000
addr (maplev)          0   addr (maplws)          0   addr (scratch storage)     1   total storage provided    900000
***** wopen call on unit 1 blocks: 10 status: 0 *****
***** wopen call on unit 2 blocks: 10 status: 0 *****
***** wopen call on unit 3 blocks: 10 status: 0 *****
***** a502 - pan-air technology program *****
***** Potential flow about arbitrary configurations *****
***** Version id = ht2 (12 feb 92) boeing ver 100 *****
***** 07-Mar-9 *****
***** GBU-24 FREESTREAM M=0 .8 *****
***** $SOLUTION *****
***** 1 0 *b*input-da *****
***** - list of a502 input data cards - *****
1 $TITLE
2 GBU-24 FREESTREAM M=0 .8
3 $SOLUTION
4 $SYMMETRY
5 =MISYMM  MJSYMM
6 0.
7 $MACH NUMBER
8 =AMACH
9 .8
10 $CASES
11 =NCASE
12 1.
13 $ANGLES-OF-ATTACK
14 =ALPC
15 2.
16 =ALPHA (1)
```

17 2.	18 \$PRINTOUT OPTIONS	19 =ISINGS IGEOMP	20 0.	21 =IPPAIC NEXDGN	22 0.	23 SREFENCES FOR ACCUMULATED FORCES AND MOMENTS	24 =XREF YREF ZREF	25 100 716 0.0	26 =SREF BREF CREF	27 165.1248 1.0	28 \$POINTS NETWORK = ZCAN4
29 1.0											
30 2.0											
31 9.0	9.0										
32 45.236	3.138	-2.324	42.548	3.138	-2.324						
33 39.859	3.138	-2.324	37.171	3.138	-2.324						
34 34.483	3.138	-2.324	31.795	3.138	-2.324						
35 29.107	3.138	-2.324	26.418	3.138	-2.324						
36 23.730	3.138	-2.324									
37 45.183	4.642	-3.583	42.732	4.642	-3.582						
38 40.281	4.642	-3.582	37.830	4.642	-3.582						
39 35.380	4.642	-3.581	32.929	4.642	-3.581						
40 30.478	4.642	-3.581	28.027	4.642	-3.580						
41 25.576	4.642	-3.580									
42 45.130	6.145	-4.841	42.917	6.145	-4.840						
43 40.703	6.146	-4.840	38.489	6.146	-4.839						
44 36.276	6.146	-4.839	34.063	6.146	-4.838						
45 31.849	6.145	-4.837	29.635	6.146	-4.837						
46 27.422	6.145	-4.836									
47 45.077	7.650	-6.100	43.101	7.650	-6.099						
48 41.125	7.650	-6.098	39.149	7.650	-6.097						
49 37.173	7.650	-6.096	35.196	7.650	-6.095						
50 33.220	7.650	-6.094	31.244	7.649	-6.093						
51 29.258	7.649	-6.092									
52 45.025	9.155	-7.358	43.286	9.155	-7.357						
53 41.547	9.154	-7.355	39.808	9.154	-7.354						
54 38.059	9.154	-7.353	36.330	9.154	-7.352						
55 34.591	9.153	-7.350	32.853	9.153	-7.349						
56 31.114	9.153	-7.348									
57 44.972	10.659	-8.616	43.470	10.659	-8.615						
58 41.969	10.659	-8.613	40.467	10.658	-8.612						
59 38.956	10.658	-8.610	37.464	10.658	-8.608						
60 35.953	10.658	-8.607	34.461	10.657	-8.605						
61 32.959	10.657	-8.603									
62 44.919	12.163	-9.875	43.655	12.163	-9.873						
63 42.390	12.163	-9.871	41.126	12.162	-9.869						
64 39.862	12.162	-9.867	38.598	12.162	-9.865						
65 37.334	12.161	-9.863	36.069	12.161	-9.861						
66 34.805	12.161	-9.859									
67 44.866	13.668	-11.133	43.839	13.667	-11.131						
68 42.812	13.667	-11.129	41.785	13.667	-11.127						
69 40.759	13.666	-11.124	39.732	13.666	-11.122						
70 38.705	13.665	-11.120	37.678	13.665	-11.117						
71 36.551	13.665	-11.115									
72 44.813	15.172	-12.392	44.024	15.172	-12.389						
73 43.234	15.171	-12.387	42.445	15.171	-12.384						

results									
Mar 7 1996 13:51									
74	41.655	15.170	-12.381	40.866	15.170	-12.379			
75	40.076	15.169	-12.376	39.287	15.169	-12.374			
76	38.497	15.168	-12.371	ZCAN3					
77	\$POINTS NETWORK =								
78	1.0								
79	2.0								
80	9.0	9.0							
81	45.236	-2.324	-3.138	42.548	-2.324	-3.138			
82	39.859	-2.324	-3.138	37.171	-2.324	-3.138			
83	34.483	-2.324	-3.138	31.795	-2.324	-3.138			
84	29.107	-2.324	-3.138	26.418	-2.324	-3.138			
85	23.730	-2.324	-3.138						
86	45.183	-3.583	-4.642	42.732	-3.582	-4.642			
87	40.281	-3.582	-4.642	32.929	-3.581	-4.642			
88	35.380	-3.581	-4.642	28.027	-3.580	-4.642			
89	30.478	-3.581	-4.642						
90	25.576	-3.580	-4.642						
91	45.130	-4.841	-6.146	42.917	-4.840	-6.146			
92	40.703	-4.840	-6.146	38.489	-4.839	-6.146			
93	36.276	-4.839	-6.146	34.063	-4.838	-6.146			
94	31.849	-4.837	-6.146	29.635	-4.837	-6.146			
95	27.422	-4.836	-6.145						
96	45.077	-6.100	-7.650	43.101	-6.099	-7.650			
97	41.125	-6.098	-7.650	39.149	-6.097	-7.650			
98	37.173	-6.096	-7.650	35.196	-6.095	-7.650			
99	33.220	-6.094	-7.650	31.244	-6.093	-7.649			
100	29.268	-6.092	-7.649						
101	45.025	-7.358	-9.155	43.286	-7.357	-9.155			
102	41.547	-7.355	-9.154	39.808	-7.354	-9.154			
103	38.069	-7.353	-9.154	36.330	-7.352	-9.154			
104	34.591	-7.350	-9.153	32.853	-7.349	-9.153			
105	31.114	-7.348	-9.153						
106	44.972	-8.616	-10.659	43.470	-8.615	-10.659			
107	41.969	-8.613	-10.659	40.467	-8.612	-10.658			
108	38.966	-8.610	-10.658	37.464	-8.608	-10.658			
109	35.963	-8.607	-10.658	34.461	-8.605	-10.657			
110	32.959	-8.603	-10.657						
111	44.919	-9.875	-12.163	43.655	-9.873	-12.163			
112	42.390	-9.871	-12.163	41.126	-9.869	-12.162			
113	39.862	-9.867	-12.162	38.598	-9.865	-12.162			
114	37.334	-9.863	-12.161	36.069	-9.861	-12.161			
115	34.805	-9.859	-12.161						
116	44.866	-11.133	-13.668	43.839	-11.131	-13.667			
117	42.812	-11.129	-13.667	41.785	-11.127	-13.667			
118	40.759	-11.124	-13.666	39.732	-11.122	-13.666			
119	38.705	-11.120	-13.665	37.678	-11.117	-13.665			
120	36.651	-11.115	-13.665						
121	44.813	-12.392	-15.172	44.024	-12.389	-15.172			
122	43.234	-12.387	-15.171	42.445	-12.384	-15.171			
123	41.655	-12.381	-15.170	40.866	-12.379	-15.170			
124	40.076	-12.376	-15.169	39.287	-12.374	-15.169			
125	38.497	-12.371	-15.168						
126	\$POINTS NETWORK =			ZCAN1					
127	1.0								
128	2.0								
129	9.0	9.0							
130	45.236	2.324	3.138	42.548	2.324	3.138			

Mar 7 1996 13:51	results
131	39.859
132	34.483
133	29.107
134	23.730
135	45.183
136	40.281
137	35.380
138	30.478
139	25.576
140	45.130
141	40.703
142	36.276
143	31.849
144	27.422
145	45.077
146	41.125
147	37.173
148	33.220
149	29.268
150	45.025
151	41.547
152	38.069
153	34.591
154	31.114
155	44.972
156	41.969
157	38.966
158	35.963
159	32.959
160	44.919
161	42.390
162	39.862
163	37.334
164	34.805
165	44.866
166	42.812
167	40.759
168	38.705
169	36.651
170	44.813
171	43.234
172	41.655
173	40.076
174	38.497
175	\$POINTS_NETWORK =
176	1.0
177	2.0
178	9.0
179	45.236
180	39.859
181	34.483
182	29.107
183	23.730
184	45.183
185	40.281
186	35.380
187	30.478
177	2.324
178	3.138
179	2.324
180	3.138
181	2.324
182	2.324
183	2.324
184	2.324
185	2.324
186	2.324
187	2.324
177	31.138
178	31.138
179	31.138
180	31.138
181	31.138
182	31.138
183	31.138
184	31.138
185	31.138
186	31.138
187	31.138
177	26.418
178	26.418
179	26.418
180	26.418
181	26.418
182	26.418
183	26.418
184	26.418
185	26.418
186	26.418
187	26.418
177	42.917
178	38.489
179	34.063
180	31.244
181	29.635
182	28.027
183	28.027
184	28.027
185	28.027
186	28.027
187	28.027
177	4.840
178	4.839
179	4.838
180	4.837
181	4.837
182	4.837
183	4.837
184	4.837
185	4.837
186	4.837
187	4.837
177	6.146
178	6.146
179	6.146
180	6.146
181	6.146
182	6.146
183	6.146
184	6.146
185	6.146
186	6.146
187	6.146
177	6.099
178	6.099
179	6.099
180	6.099
181	6.099
182	6.099
183	6.099
184	6.099
185	6.099
186	6.099
187	6.099
177	6.097
178	6.097
179	6.097
180	6.097
181	6.097
182	6.097
183	6.097
184	6.097
185	6.097
186	6.097
187	6.097
177	7.650
178	7.650
179	7.650
180	7.650
181	7.650
182	7.650
183	7.650
184	7.650
185	7.650
186	7.650
187	7.650
177	7.352
178	7.352
179	7.352
180	7.352
181	7.352
182	7.352
183	7.352
184	7.352
185	7.352
186	7.352
187	7.352
177	9.154
178	9.154
179	9.154
180	9.154
181	9.154
182	9.154
183	9.154
184	9.154
185	9.154
186	9.154
187	9.154
177	10.659
178	10.659
179	10.659
180	10.659
181	10.659
182	10.659
183	10.659
184	10.659
185	10.659
186	10.659
187	10.659
177	10.657
178	10.657
179	10.657
180	10.657
181	10.657
182	10.657
183	10.657
184	10.657
185	10.657
186	10.657
187	10.657
177	10.655
178	10.655
179	10.655
180	10.655
181	10.655
182	10.655
183	10.655
184	10.655
185	10.655
186	10.655
187	10.655
177	10.653
178	10.653
179	10.653
180	10.653
181	10.653
182	10.653
183	10.653
184	10.653
185	10.653
186	10.653
187	10.653
177	10.651
178	10.651
179	10.651
180	10.651
181	10.651
182	10.651
183	10.651
184	10.651
185	10.651
186	10.651
187	10.651
177	12.163
178	12.163
179	12.163
180	12.163
181	12.163
182	12.163
183	12.163
184	12.163
185	12.163
186	12.163
187	12.163
177	12.162
178	12.162
179	12.162
180	12.162
181	12.162
182	12.162
183	12.162
184	12.162
185	12.162
186	12.162
187	12.162
177	12.161
178	12.161
179	12.161
180	12.161
181	12.161
182	12.161
183	12.161
184	12.161
185	12.161
186	12.161
187	12.161
177	12.160
178	12.160
179	12.160
180	12.160
181	12.160
182	12.160
183	12.160
184	12.160
185	12.160
186	12.160
187	12.160
177	12.159
178	12.159
179	12.159
180	12.159
181	12.159
182	12.159
183	12.159
184	12.159
185	12.159
186	12.159
187	12.159
177	12.158
178	12.158
179	12.158
180	12.158
181	12.158
182	12.158
183	12.158
184	12.158
185	12.158
186	12.158
187	12.158
177	12.157
178	12.157
179	12.157
180	12.157
181	12.157
182	12.157
183	12.157
184	12.157
185	12.157
186	12.157
187	12.157
177	12.156
178	12.156
179	12.156
180	12.156
181	12.156
182	12.156
183	12.156
184	12.156
185	12.156
186	12.156
187	12.156
177	12.155
178	12.155
179	12.155
180	12.155
181	12.155
182	12.155
183	12.155
184	12.155
185	12.155
186	12.155
187	12.155
177	12.154
178	12.154
179	12.154
180	12.154
181	12.154
182	12.154
183	12.154
184	12.154
185	12.154
186	12.154
187	12.154
177	12.153
178	12.153
179	12.153
180	12.153
181	12.153
182	12.153
183	12.153
184	12.153
185	12.153
186	12.153
187	12.153
177	12.152
178	12.152
179	12.152
180	12.152
181	12.152
182	12.152
183	12.152
184	12.152
185	12.152
186	12.152
187	12.152
177	12.151
178	12.151
179	12.151
180	12.151
181	12.151
182	12.151
183	12.151
184	12.151
185	12.151
186	12.151
187	12.151
177	12.150
178	12.150
179	12.150
180	12.150
181	12.150
182	12.150
183	12.150
184	12.150
185	12.150
186	12.150
187	12.150
177	12.149
178	12.149
179	12.149
180	12.149
181	12.149
182	12.149
183	12.149
184	12.149
185	12.149
186	12.149
187	12.149
177	12.148
178	12.148
179	12.148
180	12.148
181	12.148
182	12.148
183	12.148
184	12.148
185	12.148
186	12.148
187	12.148
177	12.147
178	12.147
179	12.147
180	12.147
181	12.147
182	12.147
183	12.147
184	12.147
185	12.147
186	12.147
187	12.147
177	12.146
178	12.146
179	12.146
180	12.146
181	12.146
182	12.146
183	12.146
184	12.146
185	12.146
186	12.146
187	12.146
177	12.145
178	12.145
179	12.145
180	12.145
181	12.145
182	12.145
183	12.145
184	12.145
185	12.145
186	12.145
187	12.145
177	12.144
178	12.144
179	12.144
180	12.144
181	12.144
182	12.144
183	12.144
184	12.144
185	12.144
186	12.144
187	12.144
177	12.143
178	12.143
179	12.143
180	12.143
181	12.143
182	12.143
183	12.143
184	12.143
185	12.143
186	12.143
187	12.143
177	12.142
178	12.142
179	12.142
180	12.142
181	12.142
182	12.142
183	12.142
184	12.142
185	12.142
186	12.142
187	12.142
177	12.141
178	12.141
179	12.141
180	12.141
181	12.141
182	12.141
183	12.141
184	12.141
185	12.141
186	12.141
187	12.141
177	12.140
178	12.140
179	12.140
180	12.140
181	12.140
182	12.140
183	12.140
184	12.140
185	12.140
186	12.140
187	12.140
177	12.139
178	12.139
179	12.139
180	12.139
181	12.139
182	12.139
183	12.139
184	12.139
185	12.139
186	12.139
187	12.139
177	12.138
178	12.138
179	12.138
180	12.138
181	12.138
182	12.138
183	12.138
184	12.138
185	12.138
186	12.138
187	12.138
177	12.137
178	12.137
179	12.137
180	12.137
181	12.137
182	12.137
183	12.137
184	12.137
185	12.137
186	12.137
187	12.137
177	12.13

Mar 7 1996 13:51	results
188 25.576	-4.642
189 45.130	-6.146
190 40.703	-6.146
191 36.276	-6.146
192 31.849	-6.146
193 27.422	-6.145
194 45.077	-7.650
195 41.125	-7.650
196 37.173	-7.650
197 33.220	-7.650
198 29.288	-7.649
199 45.025	-9.155
200 41.547	-9.154
201 38.063	-9.154
202 34.591	-9.153
203 31.114	-9.153
204 44.972	-10.659
205 41.969	-10.659
206 38.966	-10.658
207 35.963	-10.658
208 32.959	-10.657
209 44.919	-12.163
210 42.390	-12.163
211 39.862	-12.162
212 37.334	-12.161
213 34.805	-12.161
214 44.866	-13.668
215 42.812	-13.667
216 40.759	-13.666
217 38.705	-13.665
218 36.651	-13.665
219 44.813	-15.172
220 43.234	-15.171
221 41.655	-15.170
222 40.076	-15.169
223 38.497	-15.168
224 38.497	12.371
225 1.0	ZNOSF
226 1.0	
227 4.0	13.0
228 8.499	0.000
229 0.492	0.000
230 8.499	1.995
231 0.482	0.924
232 8.499	3.463
233 0.482	1.604
234 8.499	3.999
235 0.482	1.853
236 8.499	3.463
237 0.492	1.604
238 8.499	2.007
239 0.482	0.930
240 8.499	0.000
241 0.482	0.000
242 8.499	-2.007
243 0.482	-0.930
244 8.499	-3.463
226 1.0	
227 4.0	
228 8.499	
229 0.492	
230 8.499	
231 0.482	
232 8.499	
233 0.482	
234 8.499	
235 0.482	
236 8.499	
237 0.492	
238 8.499	
239 0.482	
240 8.499	
241 0.482	
242 8.499	
243 0.482	
244 8.499	

			results		
245	0.482	-1.604	-0.930	0.000	0.000
246	8.499	-3.699	0.000	2.498	-3.565
247	0.482	-1.853	0.000	0.000	0.000
248	8.499	-3.463	2.007	2.498	-3.086
249	0.482	-1.604	0.930	0.000	0.000
250	8.499	-1.995	3.464	2.498	-1.778
251	0.482	-0.924	1.605	0.000	3.088
252	8.499	0.000	3.999	2.498	0.000
253	0.482	0.000	1.853	0.000	0.000
254 \$POINTS NETWORK = ZNOSAFT					
255 1.0					
256 1.0					
257 13.0	2.0				
258 8.499	0.000	3.999	8.499	1.995	3.465
259 8.499	3.463	2.007	8.499	3.999	0.000
260 8.499	3.463	-2.007	8.499	2.007	-3.463
261 8.499	0.000	-4.000	8.499	-2.007	-3.463
262 8.499	-3.463	-2.007	8.499	-3.999	0.000
263 8.499	-3.463	2.007	8.499	-1.995	3.464
264 8.499	0.000	3.999	8.499	1.995	3.465
265 21.730	0.000	3.999	21.730	1.995	3.465
266 21.730	3.463	2.007	21.730	3.999	0.000
267 21.730	3.463	-2.007	21.730	2.007	-3.463
268 21.730	0.000	-4.000	21.730	-2.007	-3.463
269 21.730	-3.463	-2.007	21.730	-3.999	0.000
270 21.730	-3.463	2.007	21.730	-1.995	3.465
271 21.730	0.000	3.999	21.730	1.995	3.465
272 \$POINTS NETWORK = ZCANFWDLI					
273 1.0					
274 1.0					
275 3.0	13.0				
276 23.730	-3.138	2.324	22.730	-3.300	2.165
277 21.730	-3.463	2.007	22.730	-1.995	3.465
278 23.730	-1.995	3.465	22.730	0.000	4.000
279 21.730	-1.995	3.465	22.730	0.000	4.000
280 23.730	0.000	4.000	22.730	0.000	4.000
281 21.730	0.000	3.999	23.730	2.159	3.301
282 23.730	2.324	3.138	22.730	3.301	-2.166
283 21.730	1.995	3.465	22.730	3.463	2.007
284 23.730	3.463	2.007	22.730	3.463	2.007
285 21.730	3.463	2.007	22.730	4.000	0.000
286 23.730	4.000	0.000	22.730	4.000	0.000
287 21.730	3.999	0.000	22.730	3.301	-2.166
288 23.730	3.138	-2.324	22.730	3.301	-2.166
289 21.730	3.463	-2.007	22.730	-3.463	-2.007
290 23.730	2.007	-3.464	22.730	-3.463	-2.007
291 21.730	2.007	-3.463	22.730	0.000	-4.000
292 23.730	0.000	-4.000	22.730	0.000	-4.000
293 21.730	0.000	-4.000	22.730	0.000	0.000
294 23.730	-2.324	-3.138	22.730	-2.165	-3.301
295 21.730	-2.007	-3.463	22.730	-3.463	-2.007
296 23.730	-3.463	-2.007	22.730	-3.463	-2.007
297 21.730	-3.463	-2.007	22.730	-3.463	-2.007
298 23.730	-4.000	0.000	22.730	-4.000	0.000
299 21.730	-3.999	0.000	22.730	0.000	0.000
300 23.730	-3.138	2.324	22.730	-3.300	2.165
301 21.730	-3.463	2.007	22.730	0.000	0.000

## Results

SPOINTS NETWORK = BODCAN1									
302	1.	0.	0.	0.	0.	0.	0.	0.	1.
303	1.	2.	2.	2.324	3.138	23.730	0.000	4.000	
304	1.	0.	0.	0.	0.	0.	0.	0.	
305	4.	23.730	23.730	-1.995	3.465	23.730	-3.138	2.324	
306		23.730	23.730	-2.007	2.007	23.730	-2.324	-3.138	
307		23.730	23.730	-3.463	-2.007	23.730	-4.000	0.000	
308		2.324	2.324	-3.138	2.324	45.236	0.000	4.000	
309	45.236	-1.995	3.465	45.236	45.236	45.236	-3.138	2.324	
310	SPOINTS NETWORK =	BODCAN2							
311	1.								
312	1.	0.	0.	0.	0.	0.	0.	0.	
313	4.	2.	2.	2.	2.	2.	1.	1.	
314	23.730	-3.138	2.324	2.324	2.324	23.730	-4.000	0.000	
315	23.730	-3.463	-2.007	-2.007	-2.007	23.730	-2.324	-3.138	
316	45.236	-3.138	2.324	2.324	2.324	45.236	-4.000	0.000	
317	45.236	-3.463	-2.007	-2.007	-2.007	45.236	-2.324	-3.138	
318	SPOINTS NETWORK =	BODCAN3							
319	1.								
320	1.	0.	0.	0.	0.	0.	0.	0.	
321	4.	2.	2.	2.	2.	2.	1.	1.	
322	23.730	-2.324	-3.138	-3.138	-3.138	23.730	0.000	-4.000	
323	23.730	2.007	-3.464	-3.464	-3.464	23.730	3.138	-2.324	
324	45.236	-2.324	-3.138	-3.138	-3.138	45.236	4.000	0.000	
325	45.236	2.007	-3.464	-3.464	-3.464	45.236	3.138	-2.324	
326	SPOINTS NETWORK =	BODCAN4							
327	1.								
328	1.	0.	0.	0.	0.	0.	0.	0.	
329	4.	2.	2.	2.	2.	2.	1.	1.	
330	23.730	3.138	-2.324	-2.324	-2.324	23.730	4.000	0.000	
331	23.730	3.463	2.007	2.007	2.007	23.730	2.324	3.138	
332	45.236	3.138	-2.324	-2.324	-2.324	45.236	4.000	0.000	
333	45.236	3.463	2.007	2.007	2.007	45.236	2.324	3.138	
334	SPOINTS NETWORK = ZCANAFIT1								
335	1.0								
336	1.0	0.	0.	0.	0.	0.	0.	0.	
337	3.0	13.0	0.	0.	0.	0.	0.	0.	
338	47.236	-3.463	2.007	2.007	2.007	46.236	-3.300	2.165	
339	45.236	-3.138	2.324	2.324	2.324	46.236	-4.000	0.000	
340	47.236	-4.000	0.000	0.000	0.000	45.236	-4.000	0.000	
341	45.236	-4.000	0.000	0.000	0.000	45.236	-4.000	0.000	
342	47.236	-3.463	-2.007	-2.007	-2.007	46.236	-3.463	-2.007	
343	45.236	-3.463	-2.007	-2.007	-2.007	46.236	-4.000	0.000	
344	47.236	-2.007	-3.164	-3.164	-3.164	46.236	-2.166	-3.301	
345	45.236	-2.324	-3.138	-3.138	-3.138	46.236	-2.324	-3.138	
346	47.236	0.000	-4.000	-4.000	-4.000	46.236	0.000	-4.000	
347	45.236	0.000	-4.000	-4.000	-4.000	46.236	0.000	-4.000	
348	47.236	2.007	-3.464	-3.464	-3.464	46.236	2.007	-3.464	
349	45.236	2.007	-3.464	-3.464	-3.464	46.236	2.007	-3.464	
350	47.236	3.464	-2.007	-2.007	-2.007	46.236	3.301	-2.166	
351	45.236	3.138	-2.324	-2.324	-2.324	46.236	3.301	-2.166	
352	47.236	4.000	0.000	0.000	0.000	46.236	4.000	0.000	
353	45.236	4.000	0.000	0.000	0.000	46.236	3.463	2.007	
354	47.236	3.463	2.007	2.007	2.007	46.236	3.463	2.007	
355	45.236	3.463	2.007	2.007	2.007	46.236	3.463	2.007	
356	47.236	1.995	3.465	3.465	3.465	46.236	2.159	3.301	
357	45.236	2.324	3.138	3.138	3.138	46.236	0.000	0.000	
358	47.236	0.000	4.000	4.000	4.000	46.236	0.000	4.000	

Mar 7 1996 13:51	Results	Printed by malotour from osprey
359	45.236	0.000
360	47.236	-1.995
361	45.236	-1.995
362	47.236	-3.463
363	45.236	-3.138
364 \$POINTS NETWORK = ZCANAF2	2.324	
365 1.0		
366 1.0	0.	0.
367	13.0	2.0
368	53.318	0.000
369	53.318	3.463
370	53.318	3.464
371	53.318	0.000
372	53.318	-3.463
373	53.318	-3.463
374	53.318	0.000
375	47.236	0.000
376	47.236	3.463
377	47.236	3.464
378	47.236	0.000
379	47.236	-3.463
380	47.236	-3.463
381	47.236	0.000
382 \$POINTS NETWORK = ZBOD2	4.000	
383 1.0		
384 1.0		
385	4.0	13.0
386	132.459	-3.616
387	61.785	-3.167
388	132.459	0.000
389	61.785	0.000
390	132.459	3.616
391	61.785	3.166
392	132.459	6.278
393	61.785	5.497
394	132.459	7.251
395	61.785	6.349
396	132.459	6.279
397	61.785	5.498
398	132.459	3.639
399	61.785	3.186
400	132.459	0.001
401	61.785	0.001
402	132.459	-6.638
403	61.785	-3.185
404	132.459	-6.278
405	61.785	-5.497
406	132.459	-7.250
407	61.785	-6.349
408	132.459	3.638
409	61.785	-5.497
410	132.459	-3.616
411	61.785	-3.167
412 \$POINTS NETWORK = ZBODA4	5.500	
413 1.0		
414 1.0		
415 1.0	4.0	11.0

## results

Mar 7 1996 13:51

416	132.459	-6.278	-3.639	132.459	-7.250	-0.000
417	132.459	-6.278	3.638	132.459	-3.616	6.281
418	134.175	-6.549	-3.796	134.175	-7.564	-0.000
419	134.175	-6.549	3.795	134.175	-3.773	6.552
420	135.865	-6.816	-3.951	135.865	-7.872	0.000
421	135.865	-6.817	3.950	135.865	-3.927	6.820
422	137.500	-7.075	-4.100	137.500	-8.171	0.000
423	137.500	-7.075	4.100	137.500	-4.076	7.078
424	138.947	-7.239	-4.196	138.947	-8.361	0.000
425	138.947	-7.240	4.195	138.947	-4.170	7.243
426	140.100	-7.341	-4.254	140.100	-8.479	0.000
427	140.100	-7.341	4.254	140.100	-4.229	7.344
428	141.551	-7.469	-4.328	141.551	-8.626	0.000
429	141.551	-7.469	4.328	141.551	-4.302	7.472
430	143.975	-7.420	-4.300	143.975	-8.569	0.000
431	143.975	-7.420	4.299	143.975	-4.274	7.423
432	148.235	-6.650	-3.854	148.235	-7.680	0.000
433	148.235	-6.649	3.853	148.235	-3.830	6.652
434	153.934	-5.619	-3.257	153.934	-6.489	0.000
435	153.934	-5.619	3.256	153.934	-3.237	5.621
436	160.353	-4.457	-2.584	160.353	-5.148	-0.000
437	160.353	-4.457	2.584	160.353	-2.568	4.459
438	SPOINTS NETWORK = ZBODA3					
439	1.0					
440	1.0					
441	4.0	11.0				
442	132.459	3.639	-6.278	132.459	0.000	-7.251
443	132.459	-3.638	-6.279	132.459	-6.278	-3.639
444	134.175	3.796	-6.550	134.175	-6.000	-7.564
445	134.175	-3.795	-6.550	134.175	-6.549	-3.796
446	135.865	3.951	-6.817	135.865	0.000	-7.873
447	135.865	-3.950	-6.817	135.865	-6.816	-3.951
448	137.500	4.100	-7.076	137.500	0.000	-8.172
449	137.500	-4.100	-7.076	137.500	-7.075	-4.100
450	138.947	4.195	-7.240	138.947	0.000	-8.361
451	138.947	-4.195	-7.240	138.947	-7.239	-4.196
452	140.100	4.254	-7.341	140.100	0.000	-8.479
453	140.100	-4.254	-7.341	140.100	-7.341	-4.254
454	141.551	4.328	-7.469	141.551	0.000	-8.626
455	141.551	-4.328	-7.469	141.551	-7.469	-4.328
456	143.975	4.300	-7.420	143.975	0.000	-8.569
457	143.975	-4.300	-7.420	143.975	-7.420	-4.300
458	148.235	3.853	-6.650	148.235	0.000	-7.680
459	148.235	-3.853	-6.650	148.235	-6.650	-3.854
460	153.934	3.256	-5.619	153.934	0.000	-6.489
461	153.934	-3.256	-5.619	153.934	-5.619	-3.257
462	160.353	2.584	-4.458	160.353	0.000	-5.148
463	160.353	-2.583	-4.458	160.353	-4.457	-2.584
464	SPOINTS NETWORK = ZBODA2					
465	1.0					
466	1.0					
467	4.0	11.0				
468	132.459	6.278	3.638	132.459	7.251	0.000
469	132.459	6.279	-3.638	132.459	3.639	-6.278
470	134.175	6.550	3.795	134.175	7.564	0.000
471	134.175	6.550	-3.796	134.175	3.796	-6.550
472	135.865	6.917	3.950	135.865	7.873	0.000

Mar 7 1996 13:51		results	
473	135.865	6.817	-3.950
474	137.500	7.075	4.110
475	137.500	7.075	-4.100
476	138.947	7.240	4.195
477	138.947	7.239	-4.196
478	140.100	7.341	4.254
479	140.100	7.341	-4.254
480	141.551	7.469	4.328
481	141.551	7.469	-4.328
482	143.975	7.420	4.300
483	143.975	7.420	-4.300
484	148.235	6.649	3.853
485	148.235	6.649	-3.853
486	153.934	5.618	3.256
487	153.934	5.619	-3.256
488	160.353	4.457	2.583
489	160.353	4.457	-2.583
490	SPOINTS NETWORK = ZFIN2		
491	1.0		
492	2.0		
493	11.0	7.0	
494	132.459	6.278	3.638
495	135.865	6.817	3.950
496	138.948	7.240	4.195
497	141.551	7.469	4.328
498	148.235	6.649	3.853
499	160.353	4.457	2.583
500	132.480	7.119	4.856
501	135.495	7.695	5.246
502	138.272	8.147	5.552
503	140.676	8.382	5.708
504	147.477	7.746	5.351
505	160.341	5.903	4.283
506	132.502	7.960	6.074
507	135.128	8.572	6.540
508	137.598	9.055	6.907
509	139.801	9.294	7.086
510	146.713	8.835	6.844
511	160.328	7.348	5.983
512	132.523	8.801	7.291
513	134.763	9.448	7.833
514	136.926	9.962	8.261
515	138.929	10.205	8.462
516	145.945	9.917	8.333
517	160.316	8.794	7.682
518	132.544	9.642	8.508
519	134.402	10.324	9.123
520	136.255	10.869	9.615
521	138.059	11.115	9.835
522	145.175	10.994	9.819
523	160.303	10.239	9.382
524	132.566	10.484	9.726
525	134.047	11.198	10.413
526	135.587	11.775	10.968
527	137.190	12.023	11.206
528	144.403	12.065	11.301
529	160.291	11.685	11.081

results										
Mar 7 1996 13:51										
530	132.587	11.325	10.943	133.137	11.695	11.318				
531	133.701	12.073	11.702	134.288	12.467	12.102				
532	134.924	12.681	12.320	135.424	12.769	12.410				
533	136.323	12.929	12.574	138.428	13.131	12.781				
534	143.632	13.131	12.781	151.331	13.130	12.781				
535	160.278	13.130	12.781							
536	SPOINTS NETWORK = ZFIN4									
537	1.0									
538	2.0									
539	11.0	7.0								
540	132.459	-6.278	-3.639	134.175	-6.549	-3.796				
541	135.865	-6.816	-3.951	137.500	-7.075	-4.100				
542	138.947	-7.239	-4.196	140.100	-7.341	-4.254				
543	141.551	-7.469	-4.328	144.975	-7.420	-4.300				
544	148.235	-6.650	-3.854	153.934	-5.619	-3.257				
545	160.353	-4.457	-2.584							
546	132.480	-7.119	-4.856	133.996	-7.408	-5.052				
547	135.495	-7.695	-5.246	136.960	-7.975	-5.436				
548	138.272	-8.147	-5.552	139.314	-8.249	-5.619				
549	140.676	-8.382	-5.708	145.055	-8.379	-5.718				
550	147.477	-7.746	-5.351	153.507	-6.882	-4.850				
551	160.341	-5.903	-4.283							
552	132.502	-7.960	-6.074	133.818	-8.267	-6.307				
553	135.128	-8.572	-6.540	136.422	-8.874	-6.770				
554	137.598	-9.055	-6.907	138.530	-9.156	-6.983				
555	139.801	-9.294	-7.086	142.132	-9.335	-7.134				
556	146.713	-8.835	-6.844	153.075	-8.140	-6.441				
557	160.328	-7.348	-5.983							
558	132.523	-8.801	-7.291	133.642	-9.125	-7.562				
559	134.763	-9.448	-7.833	135.885	-9.773	-8.104				
560	136.926	-9.962	-8.261	137.749	-10.062	-8.344				
561	138.929	-10.205	-8.462	141.207	-10.288	-8.548				
562	145.945	-9.917	-8.333	152.641	-9.394	-8.030				
563	160.316	-8.794	-7.682							
564	132.544	-9.642	-8.508	133.469	-9.982	-8.815				
565	134.402	-10.324	-9.123	135.350	-10.671	-9.437				
566	136.255	-10.869	-9.615	136.970	-10.966	-9.702				
567	138.059	-11.115	-9.835	140.281	-11.238	-9.960				
568	145.175	-10.994	-9.819	152.205	-10.643	-9.616				
569	160.303	-10.239	-9.382							
570	132.566	-10.484	-9.726	133.301	-10.838	-10.166				
571	134.047	-11.198	-10.413	134.817	-11.569	-10.770				
572	135.587	-11.775	-10.968	136.195	-11.869	-11.058				
573	137.190	-12.023	-11.206	139.354	-12.185	-11.771				
574	144.403	-12.065	-11.301	151.768	-11.888	-11.199				
575	160.291	-11.685	-11.081							
576	132.587	-11.325	-10.943	133.137	-11.695	-11.318				
577	133.701	-12.073	-11.702	134.288	-12.467	-12.102				
578	134.924	-12.681	-12.320	135.424	-12.769	-12.410				
579	136.323	-12.929	-12.574	138.428	-13.131	-12.781				
580	143.632	-13.131	-12.781	151.331	-13.130	-12.781				
581	160.278	-13.130	-12.781							
582	SPOINTS NETWORK = ZFIN3									
583	1.0									
584	2.0									
585	11.0	7.0								
586	132.459	3.639	-6.278	134.175	3.796	-6.550				



results										Page 13
Mar 7 1996 13:51										
644	141.551	-4.302	7.472	141.551	0.000	8.525				
645	141.551	4.302	7.472	141.551	7.469	4.322				
646	143.975	-4.274	7.423	143.975	0.000	8.568				
647	143.975	4.274	7.423	143.975	7.420	4.300				
648	148.235	-3.830	6.652	148.235	0.000	7.679				
649	148.235	3.830	6.652	148.235	6.649	3.853				
650	153.934	-3.237	5.621	153.934	0.000	6.488				
651	153.934	3.236	5.621	153.934	5.618	3.256				
652	160.353	-2.568	4.459	160.353	0.000	5.147				
653	160.353	2.568	4.460	160.353	4.457	2.583				
654	SPOINTS NETWORK = ZFINI									
655	1.0									
656	2.0									
657	11.0	7.0								
658	132.459	-3.616	6.281	134.176	-3.773	6.552				
659	135.865	-3.927	6.820	137.500	-4.076	7.078				
660	138.948	-4.170	7.243	140.101	-4.229	7.344				
661	141.551	-4.302	7.472	143.975	-4.274	7.423				
662	148.235	-3.830	6.652	153.934	-3.237	5.621				
663	160.353	-2.568	4.459	160.353	0.000	5.147				
664	132.480	-4.856	7.119	133.996	-5.052	7.408				
665	135.495	-5.246	7.695	136.960	-5.436	7.975				
666	138.272	-5.552	8.147	139.314	-5.619	8.249				
667	140.676	-5.708	8.382	143.055	-5.718	8.379				
668	147.477	-5.351	7.746	153.507	-4.850	6.882				
669	160.341	-4.283	5.903							
670	132.502	-6.074	7.960	133.818	-6.307	8.267				
671	135.128	-6.540	8.572	136.422	-6.770	8.874				
672	137.598	-6.907	9.075	138.530	-6.983	9.156				
673	139.801	-7.086	9.294	142.132	-7.134	9.335				
674	146.713	-6.844	8.835	153.075	-6.841	8.140				
675	160.328	-5.983	7.348							
676	132.523	-7.291	8.801	133.642	-7.562	9.125				
677	134.763	-7.833	9.448	135.885	-8.104	9.773				
678	136.926	-8.261	9.962	137.749	-8.344	10.062				
679	138.929	-8.462	10.205	141.207	-8.548	10.288				
680	145.945	-8.333	9.917	152.641	-8.030	9.394				
681	160.316	-7.682	8.794							
682	132.544	-8.508	9.642	133.469	-8.815	9.982				
683	134.402	-9.123	10.324	135.350	-9.437	10.671				
684	136.255	-9.615	10.869	136.970	-9.702	10.966				
685	138.059	-9.835	11.115	140.281	-9.960	11.238				
686	145.175	-9.819	10.994	152.205	-9.616	10.633				
687	160.303	-9.382	10.239							
688	132.566	-9.726	10.484	133.301	-10.066	10.838				
689	134.047	-10.413	11.198	134.817	-10.770	11.569				
690	135.587	-10.968	11.775	136.195	-11.058	11.869				
691	137.190	-11.206	12.023	139.354	-11.371	12.185				
692	144.403	-11.301	12.065	151.768	-11.199	11.888				
693	160.291	-11.081	11.685							
694	132.587	-10.943	11.325	133.137	-11.318	11.695				
695	133.701	-11.702	12.073	134.288	-12.102	12.467				
696	134.924	-12.320	12.681	135.424	-12.410	12.769				
697	136.323	-12.574	12.929	138.428	-12.781	13.131				
698	143.632	-12.781	13.131	151.331	-12.781	13.130				
699	160.278	-12.781	13.130							
700	SPOINTS NETWORK = ZEND									

RESULTS							
Mar 7 1996 13:51		Page 14					
701	1.0						
702	5.0						
703		13.0		0.	2.0	4.459	160.353
704		160.353		-2.568	2.568	4.459	160.353
705		160.353		5.148	4.459	160.353	4.457
706		160.353		2.584	-4.458	160.353	0.000
707		160.353		-2.583	-4.458	160.353	-5.148
708		160.353		-5.148	-0.001	160.353	-2.584
709		160.353		-2.568	4.459	160.353	-4.457
710		160.353		0.000	0.000	160.350	0.000
711		160.350		0.000	0.000	160.350	0.000
712		160.350		0.000	0.000	160.350	0.000
713		160.350		0.000	0.000	160.350	0.000
714		160.350		0.000	0.000	160.350	0.000
715		160.350		0.000	0.000	160.350	0.000
716		160.350		0.000	0.000	160.350	0.000
717		160.350		0.000	0.000	160.350	0.000
718	\$POINTS	NETWORK	=	ZFINNWK2			
719	1.0						
720	18.0		0.	0.	0.	0.	1.
721		7.0		2.0	2.583	160.341	5.903
722		160.353		4.457	5.983	160.316	8.794
723		160.328		7.348	9.382	160.291	11.685
724		160.303		10.239	12.781	11.081	11.081
725		160.278		13.130	12.781	11.081	11.081
726		1160.353		4.457	2.583	1160.341	5.903
727		1160.328		7.348	5.983	1160.316	8.794
728		1160.303		10.239	9.382	1160.291	11.685
729		1160.278		13.130	12.781	11.081	11.081
730	\$POINTS	NETWORK	=	ZFINNWK1			
731	1.0						
732		18.0		0.	0.	0.	1.
733		7.0		2.0	4.459	160.341	-4.283
734		160.353		-2.568	4.459	160.316	5.903
735		160.328		-5.983	7.348	160.316	-7.682
736		160.303		-9.382	10.239	160.291	11.685
737		160.278		-12.781	13.130	11.081	11.081
738		1160.353		-2.568	4.459	1160.341	-4.283
739		1160.328		-5.983	7.348	1160.316	5.903
740		1160.303		-9.382	10.239	1160.291	-7.682
741		1160.278		-12.781	13.130	11.081	11.081
742	\$POINTS	NETWORK	=	ZFINNWK4			
743	1.0						
744		18.0		0.	0.	0.	1.
745		7.0		2.0	4.457	-2.584	160.341
746		160.353		-4.457	-2.584	160.341	-5.903
747		160.328		-7.348	-5.983	160.316	-8.794
748		160.303		-10.239	-9.382	160.291	-11.685
749		160.278		-13.130	-12.781	11.081	-11.081
750		1160.353		-4.457	-2.584	1160.341	-5.903
751		1160.328		-7.348	-5.983	1160.316	-8.794
752		1160.303		-10.239	-9.382	1160.291	-11.685
753		1160.278		-13.130	-12.781	11.081	-11.081
754	\$POINTS	NETWORK	=	ZFINNWK3			
755	1.0						
756		18.0		0.	0.	0.	1.
757		7.0		2.0	0.	0.	1.

Mar 7 1996 13:51	results										Page 15
758 160.353	2.584	-4.458	160.341	4.283	-5.903						
759 160.328	5.983	-7.348	160.316	7.682	-8.794						
760 160.303	9.382	-10.239	160.291	11.081	-11.685						
761 160.278	12.181	-13.130									
762 1160.353	2.584	-4.458	1160.341	4.283	-5.903						
763 1160.328	5.983	-7.348	1160.316	7.682	-8.794						
764 1160.303	9.382	-10.239	1160.291	11.081	-11.685						
765 1160.278	12.781	-13.130									
766 SPOINTS NETWORK = ZCNBDMWKI											
767 1.0											
768 18.0	1.	0.	0.	0.	1.						
769 9.0	16.0										
770 45.236	-3.138	2.324	45.183	-4.642	3.583						
771 45.130	-6.146	4.841	45.077	-7.650	6.100						
772 45.025	-9.155	7.358	44.972	-10.659	8.616						
773 44.919	-12.163	9.875	44.866	-13.668	11.133						
774 44.813	-15.172	12.392									
775 53.318	-4.463	2.007	53.265	-4.967	3.265						
776 53.212	-6.471	4.523	53.160	-7.976	5.782						
777 53.107	-9.480	7.040	53.054	-10.984	8.299						
778 53.001	-12.489	9.557	52.948	-13.933	10.816						
779 52.896	-15.497	12.074									
780 61.785	-5.497	3.185	61.732	-7.001	4.444						
781 61.679	-8.505	5.702	61.627	-10.010	6.960						
782 61.574	-11.514	8.219	61.521	-13.018	9.477						
783 61.468	-14.523	10.736	61.415	-16.027	11.994						
784 61.362	-17.531	13.253									
785 71.254	-6.278	3.638	71.201	-7.782	4.896						
786 71.148	-9.286	6.155	71.056	-10.790	7.413						
787 71.043	-12.295	8.672	70.990	-13.799	9.930						
788 70.937	-15.303	11.189	70.884	-16.898	12.447						
789 70.831	-18.312	13.705									
790 132.459	-6.278	3.638	132.406	-7.782	4.896						
791 132.353	-9.286	6.155	132.301	-10.791	7.413						
792 132.248	-12.295	8.672	132.155	-13.799	9.930						
793 132.142	-15.304	11.189	132.089	-16.898	12.447						
794 132.036	-18.312	13.705									
795 134.175	-6.549	3.795	134.122	-8.054	5.054						
796 134.070	-9.558	6.312	134.017	-11.062	7.570						
797 133.964	-12.566	8.829	133.911	-14.071	10.087						
798 133.858	-15.575	11.346	133.806	-17.079	12.604						
799 133.753	-18.583	13.863									
800 135.865	-6.817	3.950	135.812	-8.321	5.208						
801 135.759	-9.825	6.667	135.676	-11.329	7.725						
802 135.653	-12.834	8.984	135.600	-14.338	10.242						
803 135.548	-15.842	11.501	135.495	-17.347	12.759						
804 135.442	-18.851	14.018									
805 137.500	-7.075	4.100	137.447	-8.580	5.358						
806 137.394	-10.084	6.617	137.341	-11.588	7.875						
807 137.288	-13.092	9.134	137.236	-14.597	10.392						
808 137.183	-16.101	11.651	137.130	-17.605	12.909						
809 137.077	-19.109	14.168									
810 138.447	-7.240	4.195	138.395	-8.744	5.454						
811 138.842	-10.248	6.712	138.789	-11.752	7.971						
812 138.736	-13.257	9.229	138.683	-14.761	10.488						
813 138.631	-16.265	11.746	138.578	-17.770	13.004						
814 138.525	-19.274	14.263									

Mat	7	1996	13:51	results
815	140	100	-7.341	4.254
816	139	995	-10.350	6.771
817	139	889	-13.358	9.288
818	139	784	-16.367	11.805
819	139	678	-19.375	14.322
820	141	551	-7.469	4.328
821	141	445	-10.477	6.845
822	141	340	-13.486	9.362
823	141	234	-16.494	11.879
824	141	129	-19.503	14.396
825	143	975	-7.420	4.299
826	143	870	-10.428	6.816
827	143	764	-13.437	9.333
828	143	658	-16.445	11.850
829	143	553	-19.454	14.367
830	148	235	-6.649	3.853
831	148	129	-9.658	6.370
832	148	023	-12.667	8.887
833	147	918	-15.675	11.404
834	147	812	-18.684	13.921
835	153	934	-5.619	3.256
836	153	828	-8.627	5.773
837	152	723	-11.636	8.290
838	153	617	-14.644	10.806
839	153	512	-17.653	13.323
840	160	353	-4.457	2.583
841	160	248	-7.466	5.100
842	160	142	-10.475	7.617
843	160	036	-13.483	10.134
844	159	931	-16.492	12.651
845	1160	353	-4.457	2.583
846	1160	248	-7.466	5.100
847	1160	142	-10.475	7.617
848	1160	036	-13.483	10.134
849	1159	931	-16.492	12.651
850	1159	828	-19.503	14.396
851	1159	723	-22.492	17.000
852	1159	617	-25.486	18.000
853	1159	512	-28.480	19.000
854	1159	400	-31.474	20.000
855	1159	296	-34.468	21.000
856	1159	191	-37.462	22.000
857	1159	91	-40.456	23.000
858	1159	81	-43.450	24.000
859	1159	71	-46.444	25.000
860	1159	61	-49.438	26.000
861	1159	51	-52.432	27.000
862	1159	41	-55.426	28.000
863	1159	31	-58.420	29.000
864	1159	21	-61.414	30.000
865	1159	11	-64.408	31.000
866	1159	1	-67.402	32.000
867	1159	-9	-70.396	33.000
868	1159	-19	-73.390	34.000
869	1159	-49	-76.384	35.000
870	1159	-79	-79.378	36.000
871	1159	-109	-82.372	37.000
872	1159	-139	-85.366	38.000
873	1159	-169	-88.360	39.000
874	1159	-199	-91.354	40.000
875	1159	-229	-94.348	41.000
876	1159	-259	-97.342	42.000
877	1159	-289	-100.336	43.000
878	1159	-319	-103.330	44.000
879	1159	-349	-106.324	45.000
880	1159	-379	-109.318	46.000
881	1159	-409	-112.312	47.000
882	1159	-439	-115.306	48.000
883	1159	-469	-118.300	49.000
884	1159	-499	-121.294	50.000
885	1159	-529	-124.288	51.000
886	1159	-559	-127.282	52.000
887	1159	-589	-130.276	53.000
888	1159	-619	-133.270	54.000
889	1159	-649	-136.264	55.000
890	1159	-679	-139.258	56.000
891	1159	-709	-142.252	57.000
892	1159	-739	-145.246	58.000
893	1159	-769	-148.240	59.000
894	1159	-799	-151.234	60.000
895	1159	-829	-154.228	61.000
896	1159	-859	-157.222	62.000
897	1159	-889	-160.216	63.000
898	1159	-919	-163.210	64.000
899	1159	-949	-166.204	65.000
900	1159	-979	-169.198	66.000
901	1159	-1009	-172.192	67.000
902	1159	-1039	-175.186	68.000
903	1159	-1069	-178.180	69.000
904	1159	-1099	-181.174	70.000
905	1159	-1129	-184.168	71.000
906	1159	-1159	-187.162	72.000
907	1159	-1189	-190.156	73.000
908	1159	-1219	-193.150	74.000
909	1159	-1249	-196.144	75.000
910	1159	-1279	-199.138	76.000
911	1159	-1309	-202.132	77.000
912	1159	-1339	-205.126	78.000
913	1159	-1369	-208.120	79.000
914	1159	-1399	-211.114	80.000
915	1159	-1429	-214.108	81.000
916	1159	-1459	-217.102	82.000
917	1159	-1489	-220.096	83.000
918	1159	-1519	-223.090	84.000
919	1159	-1549	-226.084	85.000
920	1159	-1579	-229.078	86.000
921	1159	-1609	-232.072	87.000
922	1159	-1639	-235.066	88.000
923	1159	-1669	-238.060	89.000
924	1159	-1699	-241.054	90.000
925	1159	-1729	-244.048	91.000
926	1159	-1759	-247.042	92.000
927	1159	-1789	-250.036	93.000
928	1159	-1819	-253.030	94.000
929	1159	-1849	-256.024	95.000
930	1159	-1879	-259.018	96.000
931	1159	-1909	-262.012	97.000
932	1159	-1939	-265.006	98.000
933	1159	-1969	-268.000	99.000
934	1159	-1999	-271.994	100.000
935	1159	-2029	-274.988	101.000
936	1159	-2059	-277.982	102.000
937	1159	-2089	-280.976	103.000
938	1159	-2119	-283.970	104.000
939	1159	-2149	-286.964	105.000
940	1159	-2179	-289.958	106.000
941	1159	-2209	-292.952	107.000
942	1159	-2239	-295.946	108.000
943	1159	-2269	-298.940	109.000
944	1159	-2299	-301.934	110.000
945	1159	-2329	-304.928	111.000
946	1159	-2359	-307.922	112.000
947	1159	-2389	-310.916	113.000
948	1159	-2419	-313.910	114.000
949	1159	-2449	-316.904	115.000
950	1159	-2479	-319.898	116.000
951	1159	-2509	-322.892	117.000
952	1159	-2539	-325.886	118.000
953	1159	-2569	-328.880	119.000
954	1159	-2599	-331.874	120.000
955	1159	-2629	-334.868	121.000
956	1159	-2659	-337.862	122.000
957	1159	-2689	-340.856	123.000
958	1159	-2719	-343.850	124.000
959	1159	-2749	-346.844	125.000
960	1159	-2779	-349.838	126.000
961	1159	-2809	-352.832	127.000
962	1159	-2839	-355.826	128.000
963	1159	-2869	-358.820	129.000
964	1159	-2899	-361.814	130.000
965	1159	-2929	-364.808	131.000
966	1159	-2959	-367.802	132.000
967	1159	-2989	-370.796	133.000
968	1159	-3019	-373.790	134.000
969	1159	-3049	-376.784	135.000
970	1159	-3079	-379.778	136.000
971	1159	-3109	-382.772	137.000
972	1159	-3139	-385.766	138.000
973	1159	-3169	-388.760	139.000
974	1159	-3199	-391.754	140.000
975	1159	-3229	-394.748	141.000
976	1159	-3259	-397.742	142.000
977	1159	-3289	-400.736	143.000
978	1159	-3319	-403.730	144.000
979	1159	-3349	-406.724	145.000
980	1159	-3379	-409.718	146.000
981	1159	-3409	-412.712	147.000
982	1159	-3439	-415.706	148.000
983	1159	-3469	-418.700	149.000
984	1159	-3499	-421.694	150.000
985	1159	-3529	-424.688	151.000
986	1159	-3559	-427.682	152.000
987	1159	-3589	-430.676	153.000
988	1159	-3619	-433.670	154.000
989	1159	-3649	-436.664	155.000
990	1159	-3679	-439.658	156.000
991	1159	-3709	-442.652	157.000
992	1159	-3739	-445.646	158.000
993	1159	-3769	-448.640	159.000
994	1159	-3799	-451.634	160.000
995	1159	-3829	-454.628	161.000
996	1159	-3859	-457.622	162.000
997	1159	-3889	-460.616	163.000
998	1159	-3919	-463.610	164.000
999	1159	-3949	-466.604	165.000
1000	1159	-3979	-469.598	166.000
1001	1159	-4009	-472.592	167.000
1002	1159	-4039	-475.586	168.000
1003	1159	-4069	-478.580	169.000
1004	1159	-4099	-481.574	170.000
1005	1159	-4129	-484.568	171.000
1006	1159	-4159	-487.562	172.000
1007	1159	-4189	-490.556	173.000
1008	1159	-4219	-493.550	174.000
1009	1159	-4249	-496.544	175.000
1010	1159	-4279	-499.538	176.000
1011	1159	-4309	-502.532	177.000
1012	1159	-4339	-505.526	178.000
1013	1159	-4369	-508.520	179.000
1014	1159	-4399	-511.514	180.000
1015	1159	-4429	-514.508	181.000
1016	1159	-4459	-517.502	182.000
1017	1159	-4489	-520.496	183.000
1018	1159	-4519	-523.490	184.000
1019	1159	-4549	-526.484	185.000
1020	1159	-4579	-529.478	186.000
1021	1159	-4609	-532.472	187.000
1022	1159	-4639	-535.466	188.000
1023	1159	-4669	-538.460	189.000
1024	1159	-4699	-541.454	190.000
1025	1159	-4729	-544.448	191.000
1026	1159	-4759	-547.442	192.000
1027	1159	-4789	-550.436	193.000
1028	1159	-4819	-553.430	194.000
1029	1159	-4849	-556.424	195.000
1030	1159	-4879	-559.418	196.000
1031	1159	-4909	-562.412	197.000
1032	1159	-4939	-565.406	198.000
1033	1159	-4969	-568.400	199.000
1034	1159	-4999	-571.394	200.000
1035	1159	-5029	-574.388	201.000
1036	1159	-5059	-577.382	202.000
1037	1159	-5089	-580.376	203.000
1038	1159	-5119	-583.370	204.000
1039	1159	-5149	-586.364	205.000
1040	1159	-5179	-589.358	206.000
1041	1159	-5209	-592.352	207.000
1042	1159	-5239	-595.346	208.000
1043	1159	-5269	-598.340	209.000

Mar 7 1986 13:51	results	Page 17
872	70.937	-11.189
873	70.931	-13.705
874	132.459	-6.638
875	132.353	-6.155
876	132.248	-8.672
877	132.142	-11.189
878	132.036	-13.705
879	134.175	-3.795
880	134.070	-6.312
881	133.964	-8.829
882	133.858	-11.346
883	133.753	-13.863
884	135.865	-3.950
885	135.759	-6.467
886	135.653	-8.884
887	135.548	-11.501
888	135.442	-14.018
889	137.500	-4.400
890	137.394	-6.917
891	137.288	-9.134
892	137.183	-11.631
893	137.077	-14.168
894	138.947	-4.195
895	138.842	-6.712
896	138.736	-9.229
897	138.631	-11.746
898	138.525	-14.263
899	140.100	-4.254
900	139.995	-6.771
901	139.889	-9.288
902	139.784	-11.805
903	139.678	-14.322
904	141.551	-4.328
905	141.445	-6.845
906	141.340	-9.362
907	141.234	-11.879
908	141.129	-14.396
909	143.975	-4.299
910	143.870	-6.816
911	143.764	-9.333
912	143.658	-11.450
913	143.553	-14.367
914	148.235	-3.853
915	148.129	-6.370
916	148.023	-8.887
917	147.918	-11.404
918	147.812	-13.921
919	153.934	-3.256
920	153.828	-5.773
921	153.723	-8.290
922	153.617	-10.806
923	153.512	-13.323
924	160.353	-2.383
925	160.248	-5.100
926	160.142	-7.617
927	160.036	-10.134
928	159.931	-12.651
		-16.492
		-11.392
		-14.987

Mar 7 1996 13:51	results
929	1160.353
930	1160.248
931	-5.100
932	-7.466
933	-7.617
934	-10.475
935	-11.036
936	-11.134
937	-13.483
938	-15.651
939	-16.492
940	-17.651
941	-18.875
942	-19.875
943	-20.007
944	-20.392
945	-20.702
946	-20.948
947	-21.074
948	-21.163
949	-21.481
950	-21.679
951	-21.813
952	-21.931
953	-22.254
954	-22.486
955	-22.679
956	-22.937
957	-23.123
958	-23.253
959	-23.286
960	-23.295
961	-23.304
962	-23.303
963	-23.312
964	-23.459
965	-23.533
966	-23.748
967	-23.753
968	-23.865
969	-23.825
970	-23.834
971	-23.548
972	-23.542
973	-23.700
974	-23.739
975	-23.788
976	-23.787
977	-23.777
978	-23.847
979	-23.842
980	-23.876
981	-23.831
982	-23.825
983	-23.820
984	-23.995
985	-23.989
986	-24.000
987	-24.341
988	-24.350
989	-24.771
990	-24.942
991	-24.942
992	-25.000
993	-25.492
994	-25.500
995	-25.500
996	-25.500
997	-25.500
998	-25.500
999	-25.500

results									
Mar 7 1996 13:51	986	139.784	16.367	-11.805	139.731	17.871	-13.063		
987	139.678	19.375	-14.322						
988	141.551	7.469	-4.328	141.498	8.973	-5.586			
989	141.445	10.477	-6.845	141.393	11.981	-8.103			
990	141.340	13.486	-9.362	141.287	14.990	-10.620			
991	141.234	16.494	-11.879	141.181	17.999	-13.137			
992	141.129	19.503	-14.396						
993	143.975	7.420	-4.299	143.923	8.924	-5.558			
994	143.870	10.428	-6.816	143.817	11.933	-8.075			
995	143.764	13.437	-9.333	143.711	14.941	-10.592			
996	143.658	16.445	-11.850	143.606	17.950	-13.109			
997	143.553	19.454	-14.367						
998	148.235	6.649	-3.853	148.182	8.154	-5.111			
999	148.129	9.658	-6.370	148.076	8.154	-5.111			
1000	148.023	12.667	-8.887	147.971	11.162	-7.628			
1001	147.918	15.675	-11.404	147.865	11.162	-7.628			
1002	147.812	18.684	-13.921						
1003	153.934	5.619	-3.256	153.881	7.123	-4.514			
1004	153.828	8.627	-5.773	153.776	10.131	-7.311			
1005	153.723	11.636	-8.290	153.670	13.140	-9.548			
1006	153.617	14.644	-10.806	153.564	16.149	-12.065			
1007	153.512	17.653	-13.323						
1008	160.353	4.457	-2.583	160.301	5.962	-3.841			
1009	160.248	7.466	-5.100	160.195	8.970	-6.358			
1010	160.142	10.475	-7.617	160.089	11.979	-8.875			
1011	160.036	13.483	-10.134	159.984	14.987	-11.392			
1012	159.921	16.492	-12.651						
1013	1160.353	4.457	-2.583	1160.301	5.962	-3.841			
1014	1160.248	7.466	-5.100	1160.195	8.970	-6.358			
1015	1160.142	10.475	-7.617	1160.089	11.979	-8.875			
1016	1160.036	13.483	-10.134	1159.984	14.987	-11.392			
1017	1159.931	16.492	-12.651						
1018	\$POINTS NETWORK = ZCNBDWK4								
1019	1.0								
1020	18.0	1.	0.	0.	0.	1.			
1021	9.0	16.0							
1022	45.236	2.324	3.138	45.183	3.583	4.642			
1023	45.130	4.841	6.146	45.077	6.100	7.650			
1024	45.025	7.358	9.155	44.972	8.616	10.659			
1025	44.919	9.275	12.163	44.866	11.133	13.668			
1026	44.813	12.392	15.172						
1027	53.318	2.007	3.463	53.265	4.967				
1028	53.212	4.523	6.471	53.160	5.782	7.976			
1029	53.107	7.040	9.480	53.054	8.299	10.884			
1030	53.001	9.557	12.489	52.948	10.816	13.993			
1031	52.896	12.074	15.497						
1032	61.785	3.638	6.278	71.201	4.896	7.782			
1033	61.679	5.702	8.505	61.627	6.960	10.010			
1034	61.574	8.219	11.514	61.521	9.477	13.018			
1035	61.468	10.736	14.523	61.415	11.994	16.027			
1036	61.362	13.253	17.531						
1037	71.254	3.638	6.278						
1038	71.148	6.155	9.286	71.096	7.413	10.790			
1039	71.043	8.672	12.295	70.990	9.930	13.799			
1040	70.937	11.189	15.303	70.884	12.447	16.808			
1041	70.831	13.705	18.312						
1042	132.459	3.638	6.278	132.406	4.896	7.782			

Mat	7/1996	13:51	results
1043	132.353	6.155	9.286
1044	132.248	8.672	12.295
1045	132.142	11.189	11.705
1046	132.036	11.705	18.312
1047	131.975	6.175	6.549
1048	134.070	6.312	9.558
1049	133.964	8.829	12.566
1050	133.858	11.575	133.911
1051	133.753	13.863	133.806
1052	135.865	3.950	6.817
1053	135.759	6.467	9.825
1054	135.653	8.984	12.834
1055	135.548	11.501	15.842
1056	135.442	14.018	18.851
1057	137.500	4.100	7.075
1058	137.394	6.617	10.084
1059	137.288	9.134	13.092
1060	137.183	11.651	16.101
1061	137.077	13.168	19.109
1062	138.947	4.195	7.240
1063	138.842	6.712	10.248
1064	138.736	9.229	13.257
1065	138.631	11.746	16.265
1066	138.525	14.263	19.274
1067	140.100	4.254	7.341
1068	139.995	6.771	10.350
1069	139.889	9.288	13.358
1070	139.784	11.805	16.367
1071	139.678	14.322	19.375
1072	141.551	4.328	7.469
1073	141.445	6.845	10.477
1074	141.340	9.362	13.486
1075	141.234	11.879	16.494
1076	141.129	14.396	19.503
1077	143.975	4.299	7.420
1078	143.870	6.816	10.428
1079	143.764	9.333	13.437
1080	143.658	11.850	16.445
1081	143.553	14.367	19.454
1082	148.235	3.853	6.649
1083	148.129	6.370	9.658
1084	148.023	8.887	12.667
1085	147.918	11.404	15.675
1086	147.812	13.921	18.684
1087	153.934	3.256	5.619
1088	153.828	5.773	8.627
1089	153.723	8.290	11.636
1090	153.617	10.806	14.644
1091	153.512	13.323	17.653
1092	160.353	2.583	4.457
1093	160.248	5.100	7.466
1094	160.142	7.617	10.475
1095	160.036	10.134	13.483
1096	159.931	12.651	16.492
1097	160.353	2.583	4.457
1098	160.248	5.100	7.466
1099	160.142	7.617	10.475
		11.60.089	8.875

Mar 7 1996 13:51

## results

Page 21						
1100	1160.036	10.134	13.483	1159.984	11.392	14.987
1101	1159.931	12.651	16.492			
1102	SPOINTS NETWORK = ZBODWAK					
1103	1.0					
1104	18.	1.	1.		1.	
1105	13.0	2.0	2.0			
1106	160.353	-2.568	4.459	160.353	0.000	5.148
1107	160.353	2.568	4.460	160.353	4.457	2.583
1108	160.353	5.148	0.000	160.353	4.457	-2.583
1109	160.353	2.584	-4.458	160.353	0.000	-5.148
1110	160.353	-2.583	-4.458	160.353	-4.457	-2.584
1111	160.353	-5.148	-0.001	160.353	-4.457	2.583
1112	160.353	-2.568	4.459	1160.353	0.000	5.148
1113	1160.353	-2.568	4.459	1160.353	4.457	2.583
1114	1160.353	2.568	4.460	1160.353	4.457	-2.583
1115	1160.353	5.148	0.000	1160.353	4.457	-2.583
1116	1160.353	2.584	-4.458	1160.353	0.000	-5.148
1117	1160.353	-2.583	-4.458	1160.353	-4.457	-2.584
1118	1160.353	-5.148	-0.001	1160.353	-4.457	2.583
1119	1160.353	-2.568	4.459			
1120	SEND					

1

record of input processing

```

STIT
$SYM
SNAC
SCAS
SANG
SPRI
SREF
SPOI kn,kt
network # being processed 1 2 9.0000 9.0000 0.0000 0.0000 0.0000
$POI kn,kt
network # being processed 2 2 9.0000 9.0000 0.0000 0.0000 0.0000
$POI kn,kt
network # being processed 3 2 9.0000 9.0000 0.0000 0.0000 0.0000
$POI kn,kt
network # being processed 4 2 9.0000 9.0000 0.0000 0.0000 0.0000
$POI kn,kt
network # being processed 5 1 4.0000 13.0000 0.0000 0.0000 0.0000
$POI kn,kt
network # being processed 6 1 13.0000 2.0000 0.0000 0.0000 0.0000

```



## APPENDIX B. GBU-24 OUPUT FILE (EDGE ANALYSIS)

Mar 7 1986 13:51		results										Page 58	
abutment	nw-ident	nd	knot-edge	nw-ident	nd	knot-edge	nw-ident	nd	knot-edge	nw-ident	nd	knot-edge	
1	12	1.1-	1.1-	18	30.1-	1.1-	12	1.1+	30.1+	1.1-	12	1.1+	
2	12	1.2-	1.2-	12	1.2-	1.2-	12	1.2+	1.2-	1.2-	12	1.2+	
3	12	1.3-	1.3-	12	1.3-	1.3-	12	1.3+	1.3-	1.3-	12	1.3+	
4	12	10.2-	10.2-	12	11.4+	11.4+	12	11.4-	11.4-	11.4-	12	11.4-	
5	12	1.4-	1.4-	12	2.1-	2.1-	12	10.2+	10.2+	10.2+	12	10.2+	
6	12	2.2-	2.2-	18	29.1-	29.1-	12	2.1+	2.1+	2.1+	12	2.1+	
7	12	2.3-	2.3-	12	3.2-	3.2-	12	2.2+	2.2+	2.2+	12	2.2+	
8	12	9.2-	9.2-	12	11.2-	11.2-	12	10.4-	10.4-	10.4-	12	10.4-	
9	12	10.4+	10.4+	12	12	12	12	2.4+	2.4+	2.4+	12	2.4+	
10	12	2.4-	2.4-	12	3.1-	3.1-	12	3.1+	3.1+	3.1+	12	3.1+	
11	12	3.2-	3.2-	12	3.3-	3.3-	12	3.2+	3.2+	3.2+	12	3.2+	
12	12	11.2-	11.2-	12	8.4+	8.4+	12	8.4-	8.4-	8.4-	12	8.4-	
13	12	8.4+	8.4+	12	3.4-	3.4-	12	3.4+	3.4+	3.4+	12	3.4+	
14	12	4.1-	4.1-	18	28.1-	28.1-	12	4.1+	28.1+	28.1+	12	4.1+	
15	12	4.2-	4.2-	12	4.3-	4.3-	12	4.2+	4.2+	4.2+	12	4.2+	
16	12	4.3-	4.3-	12	8.2-	8.2-	12	4.3+	4.3+	4.3+	12	4.3+	
17	12	9.4+	9.4+	12	4.4-	4.4-	12	9.4-	9.4-	9.4-	12	9.4-	
18	12	6.4+	6.4+	12	5.1-	5.1-	12	6.4+	6.4+	6.4+	12	6.4+	
19	12	5.1-	5.1-	12	5.2+	5.2+	12	5.2+	5.2+	5.2+	12	5.2+	
20	12	5.2+	5.2+	12	6.2-	6.2-	12	6.2+	6.2+	6.2+	12	6.2+	
21	12	6.2-	6.2-	12	7.3+	7.3+	12	7.3-	7.3-	7.3-	12	7.3-	
22	12	7.3+	7.3+	12	6.2-	6.2-	12	6.2+	6.2+	6.2+	12	6.2+	
23	12	6.2-	6.2-	12	8.1+	8.1+	12	6.1+	6.1+	6.1+	12	6.1+	
24	12	7.1-	7.1-	12	7.1-	7.1-	12	8.1-	8.1-	8.1-	12	8.1-	
25	12	7.1-	7.1-	12	9.1+	9.1+	12	7.1+	7.1+	7.1+	12	7.1+	
26	12	7.4+	7.4+	12	7.1+	7.1+	12	7.2+	7.2+	7.2+	12	7.2+	
27	12	7.2-	7.2-	12	11.1+	11.1+	12	7.1+	7.1+	7.1+	12	7.1+	
28	12	12.2+	12.2+	12	7.1-	7.1-	12	10.1-	10.1-	10.1-	12	10.1-	
29	12	9.3-	9.3-	12	12.2+	12.2+	12	9.3+	9.3+	9.3+	12	9.3+	
30	12	10.3-	10.3-	12	10.3-	10.3-	12	10.3+	10.3+	10.3+	12	10.3+	

Mar 7 1996 13:51	results	Page
31	12	12.3+
	12	12.1-
32	12	12.3-
	12	13.3+
32	12	12.4-
	12	13.3-
33	12	13.3+
	12	12.4-
33	12	12.4-
	12	13.3-
34	12	13.1-
	12	14.3-
34	12	14.3+
	12	13.1+
35	12	13.1-
	12	14.3-
35	12	14.3+
	12	13.1+
36	12	13.4+
	12	13.2+
36	12	13.2-
	12	13.4-
37	12	14.1-
	12	21.4-
37	12	21.4+
	12	14.1+
38	12	14.1-
	12	17.4-
38	12	17.4+
	12	14.1+
39	12	14.1-
	12	16.4-
40	12	16.4+
	12	14.1+
40	12	14.1-
	12	15.4-
41	12	15.4+
	12	14.1+
41	12	14.4+
	12	14.2+
42	12	14.2-
	12	14.4-
42	12	16.3+
	12	19.4-
42	12	19.4+
	12	12
43	12	15.1-
	12	16.3-
43	12	23.4+
	18	32.1+
43	18	32.1-
	12	15.2+
44	12	15.2-
	12	23.4-
44	12	21.1+
	12	22.4+
45	12	22.4-
	12	12
45	12	15.3-
	12	21.1-
45	12	17.3+
	12	20.4-
45	12	20.4+
	12	16.1+
46	12	16.1-
	12	17.3-
46	12	23.4+
	18	32.1+
46	18	32.1-
	12	16.2+
47	12	16.2-
	12	23.4-
47	12	21.3+
	12	18.4-
47	12	18.4+
	12	12
48	12	17.1-
	12	21.3-
48	12	23.4+
	18	32.1+
48	18	32.1-
	12	17.2+
49	12	17.2-
	12	23.4-
49	12	18.1-
	12	18.1+
49	12	18.2-
	12	18.2+
49	12	18.3-
	18	24.1-
49	18	24.1+
	12	18.3+
50	12	19.1-
	12	19.1+
50	12	19.2-
	12	19.2+
50	12	19.3-
	18	26.1-
51	12	18.1+
	12	19.3+
51	12	20.1-
	12	20.1+
51	12	20.2-
	12	20.2+
52	12	20.3-
	12	27.1-
52	12	27.1+
	12	20.3+
52	12	23.4+
	18	32.1-
53	12	32.1-
	12	21.2+

results									
region	nw-id	nw-name	dblt-type	surfaces associated with various regions of the configuration	material	r/ctr	r/ctr	r/ctr	r/ctr
1	1		analysis	upper	air	75.546633	0.341333	-0.739267	
1	2		analysis	lower	air				
2	2		analysis	upper	air				
3	3		analysis	lower	air				
3	4		analysis	upper	air				
4	4		analysis	lower	air				
5	5		analysis	upper	air				
6	6		analysis	upper	air				
7	7		analysis	upper	air				
8	8		analysis	upper	air				
9	9		analysis	upper	air				
10	10		analysis	upper	air				
11	11		analysis	upper	air				
12	12		analysis	upper	air				
13	13		analysis	upper	air				
14	14		analysis	upper	air				
15	15		analysis	upper	air				
59	12	21.2-	12	22.1-	12	22.1+	probable error: unabutted free edge		
60	12	22.1-	12	22.2-	12	22.2+	probable error: unabutted free edge		
61	12	22.2-	12	22.3-	18	25.1-			
62	12	25.1+	18	25.1+	12	22.1+			
63	12	23.1-	12	23.1-	12	23.3-			
64	18	24.2-	18	24.3-	18	24.2+			
65	18	24.4-	18	24.4-	18	24.4+			
66	18	25.2-	18	25.3-	18	25.3+			
67	18	25.3-	18	32.2+	18	32.1+			
68	18	32.4-	18	32.4-	18	25.4+			
69	18	25.4-	18	26.2-	18	32.2-			
70	18	26.3-	18	26.3-	18	26.3+			
71	18	26.4-	18	27.2-	18	27.2+			
72	18	27.2-	18	27.3-	18	27.3+			
73	18	27.3-	18	27.4-	18	27.4+			
74	18	27.4-	18	28.2-	18	28.2+			
75	18	28.2-	18	28.3-	18	28.3+			
76	18	28.3-	18	28.4-	18	28.4+			
77	18	28.4-	18	29.2-	18	29.2+			
78	18	29.2-	18	29.3-	18	29.3+			
79	18	29.3-	18	29.4-	18	29.4+			
80	18	29.4-	18	30.2-	18	30.2+			
81	18	30.2-	18	30.3-	18	30.3+			
82	18	30.3-	18	30.4-	18	30.4+			
83	18	30.4-	18	31.2-	18	31.2+			
84	18	31.2-	18	31.3-	18	31.3+			
85	18	31.3-	18	31.4-	18	31.4+			
86	18	31.4-	18	32.3-	18	32.3+			
87	18								

nw	edge	point	row	col	fine grid location			
1.	6	11	11	2	21	3	air	upper
2.	7	2	1	4	1	7	air	upper
3.	7	1	7	1	1	13	air	lower
3	23	analysis	upper	air	160.353000	2.584000	-4.458000	
32	nt=18 wake		upper	air				
0*b*extra-cp								
0 *****	summary of extra control points	*****						

## results

4.	7	1	10	1	10	1	19
5.	7	3	11	3	3	5	5
6.	12	2	4	4	3	7	5
7.	12	2	7	3	13	5	5
8.	12	2	10	10	3	19	5
9.	12	4	3	11	1	21	1
10.	13	1	12	1	12	1	23
11.	13	3	3	2	11	3	21
12.	14	1	4	1	4	1	7
13.	14	1	7	1	7	1	13
14.	14	1	10	1	10	1	19
15.	14	3	12	4	2	7	3
16.	23	4	4	10	1	19	1
17.	23	4	7	7	1	13	1
18.	23	4	10	4	1	7	1
19.	32	1	4	1	4	1	7
20.	32	1	7	1	7	1	13
21.	32	1	10	1	10	1	19

0 \*e\*extra-cp  
0 \*b\*extra-vp

## 0 \*\*\*\*\* summary of extra v-parameter points \*\*\*\*\*

nw	edge	point	row	col	fine grid location
1.	6	2	11	2	21
2.	7	1	4	1	3
3.	7	1	7	1	7
4.	7	1	10	1	13
5.	7	3	11	3	5
6.	12	2	4	4	5
7.	12	2	7	3	7
8.	12	2	10	10	5
9.	12	4	3	11	1
10.	13	1	12	1	21
11.	13	3	3	2	1
12.	14	1	4	1	23
13.	14	1	7	1	3
14.	14	1	10	1	1
15.	14	3	12	4	1
16.	23	4	4	10	1
17.	23	4	7	1	19
18.	23	4	10	4	1
19.	32	1	4	1	7
20.	32	1	7	1	13
21.	32	1	10	1	19

0 \*e\*extra-vp  
0 \*b\*abutment  
1

## abutment summary

abutment #	1
nw-edge	dblt edge
nw/id	type type matching kutta-f1
1.1	12 4
30.1	18 5
abutment #	mu-match
2	doublet strength matched to zero along this abutment

starts at ai # 1 ends at ai # 2  
corresponding edge points ( minus (-) indicates point moved by \$seat  
 1 2 3 4 5 6 7 8 9  
 1 2 3 4 5 6 7 8 9  
 \*\*\* warning \*\*

```

*          dblt edge      starts at ai # 2      ends at ai # 3      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponds edge points ( minus (-) ) indicates point moved by $seat
1.2     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 3      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 3      ends at ai # 4      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
1.3     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 4      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 1      ends at ai # 4      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
1.4     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 5      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 5      ends at ai # 6      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
2.1     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 6      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 5      ends at ai # 6      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
2.2     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 7      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 7      ends at ai # 8      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
2.3     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 8      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 7      ends at ai # 8      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
2.4     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 9      doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 9      ends at ai # 10     indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
3.1     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 10     doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 11      ends at ai # 12     indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
3.2     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 11     doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 12      ends at ai # 9      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
3.3     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 12     doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 1      ends at ai # 2      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
3.4     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 13     doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 1      ends at ai # 2      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
8.4     12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 14     doublet strength matched to zero along this abutment      *** warning **

*          dblt edge      starts at ai # 1      ends at ai # 2      indicates point moved by $seat
nw_edge nw/ld type type matching kutta-fl      corresponding edge points ( minus (-) ) indicates point moved by $seat
11.2    12   4      mu-match      1    2    3    4    5    6    7    8    9
0abutment # 15     doublet strength matched to zero along this abutment      *** warning **

```

Mar 7 1996 13:51

results												
Oabutment # 13	dblt edge	type type	matching	kutta-f1	starts at ai # 13	ends at ai # 14	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 4.1	edge type	12 4			1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 28.1	edge type	18 5	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 14	doublet strength	matched to zero along this abutment			1 2 3	4 5 6	7 8 9	*** warning **				
*												
Oabutment # 15	dblt edge	type type	matching	kutta-f1	starts at ai # 14	ends at ai # 15	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 4.2	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 15	doublet strength	matched to zero along this abutment			1 2 3	4 5 6	7 8 9	*** warning **				
*												
Oabutment # 16	dblt edge	type type	matching	kutta-f1	starts at ai # 15	ends at ai # 16	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 4.4	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 8.2	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 9.4	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 17	dblt edge	type type	matching	kutta-f1	starts at ai # 13	ends at ai # 16	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 5.1	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 6.4	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 18	dblt edge	type type	matching	kutta-f1	starts at ai # 17	ends at ai # 17	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 5.2	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 5.4	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 19	dblt edge	type type	matching	kutta-f1	starts at ai # 18	ends at ai # 17	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 6.1	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 6.3	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 20	dblt edge	type type	matching	kutta-f1	starts at ai # 19	ends at ai # 17	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 6.2	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 7.3	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 21	dblt edge	type type	matching	kutta-f1	starts at ai # 19	ends at ai # 20	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 6.2	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 7.3	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 22	dblt edge	type type	matching	kutta-f1	starts at ai # 12	ends at ai # 16	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 7.1	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
nw.edge nw/idx 8.1	edge type	12 4	mu-match		1 2 3	4 5 6	7 8 9					
Oabutment # 23	dblt edge	type type	matching	kutta-f1	starts at ai # 4	ends at ai # 12	corresponding edge points ( minus (-) indicates point moved by \$eat					
nw.edge nw/idx 7.1	edge type	12 4	mu-match		4 5 6	7						
nw.edge nw/idx 11.1	edge type	12 4	mu-match		4 3 2	1						
Oabutment # 24												

results						
nb-edge	nw/ld	dblt edge	starts at ai # 8	ends at ai # 4	corresponding edge points ( minus (-) indicates point moved by \$seat	
7.1	12	type type	matching	kutta-fl	7 8 9 10	
10.1	12	4	mu-match		4 3 2 1	
0abutment # 25		dblt edge	starts at ai # 16	ends at ai # 8	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl	10 11 12 13	
7.1	12	4	mu-match		4 3 2 1	
9.1	12	4				
0abutment # 26		dblt edge	starts at ai # 20	ends at ai # 16	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
7.2	12	4	mu-match			
7.4	12	4				
0abutment # 27		dblt edge	starts at ai # 23	ends at ai # 24	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
8.3	12	4	mu-match			
12.2	12	4				
0abutment # 28		dblt edge	starts at ai # 9	ends at ai # 13	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
9.3	12	4	mu-match			
12.2	12	4				
0abutment # 29		dblt edge	starts at ai # 13	ends at ai # 5	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
10.3	12	4	mu-match			
12.2	12	4				
0abutment # 30		dblt edge	starts at ai # 5	ends at ai # 1	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
11.3	12	4	mu-match			
12.2	12	4				
0abutment # 31		dblt edge	starts at ai # 1	ends at ai # 9	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
12.1	12	4	mu-match			
12.3	12	4				
0abutment # 32		dblt edge	starts at ai # 22	ends at ai # 21	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
12.4	12	4	mu-match			
13.3	12	4				
0abutment # 33		dblt edge	starts at ai # 21	ends at ai # 22	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
12.4	12	4	mu-match			
13.3	12	4				
0abutment # 34		dblt edge	starts at ai # 24	ends at ai # 23	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		
13.1	12	4	mu-match			
14.3	12	4				
0abutment # 35		dblt edge	starts at ai # 23	ends at ai # 24	corresponding edge points ( minus (-) indicates point moved by \$seat	
nb-edge	nw/ld	type type	matching	kutta-fl		

results											Page 66
Mar 7 1996 13:51	13.1	12	4	mu-match		12	13				
0abutment #	14.3	12	4	mu-match		13	12				
nw.edge nw/id	36	dblt edge	type type	matching	kutta-fil	starts at ai #	22	ends at ai #	23		
13.2	12	4	mu-match			1	2	corresponding edge points ( minus (-) indicates point moved by \$seat			
13.4	12	4				2	1				
0abutment #	37	dblt edge	type type	matching	kutta-fil	starts at ai #	26	ends at ai #	25		
nw.edge nw/id						1	2	corresponding edge points ( minus (-) indicates point moved by \$seat			
14.1	12	4	mu-match			4	3				
21.4	12	4				4	3				
0abutment #	38	dblt edge	type type	matching	kutta-fil	starts at ai #	27	ends at ai #	26		
nw.edge nw/id						4	5	corresponding edge points ( minus (-) indicates point moved by \$seat			
14.1	12	4	mu-match			6	7				
17.4	12	4				4	3				
0abutment #	39	dblt edge	type type	matching	kutta-fil	starts at ai #	28	ends at ai #	27		
nw.edge nw/id						7	8	corresponding edge points ( minus (-) indicates point moved by \$seat			
14.1	12	4	mu-match			9	10				
16.4	12	4				4	-3				
0abutment #	40	dblt edge	type type	matching	kutta-fil	starts at ai #	25	ends at ai #	28		
nw.edge nw/id						10	-11	corresponding edge points ( minus (-) indicates point moved by \$seat			
14.1	12	4	mu-match			12	13				
15.4	12	4				4	-3				
0abutment #	41	dblt edge	type type	matching	kutta-fil	starts at ai #	24	ends at ai #	25		
nw.edge nw/id						1	2	corresponding edge points ( minus (-) indicates point moved by \$seat			
14.2	12	4	mu-match			3	4				
14.4	12	4				4	3				
0abutment #	42	dblt edge	type type	matching	kutta-fil	starts at ai #	29	ends at ai #	28		
nw.edge nw/id						1	2	corresponding edge points ( minus (-) indicates point moved by \$seat			
15.1	12	4	mu-match			3	4				
16.3	12	4				11	10				
19.4	12	4				9	8				
0abutment #	43	dblt edge	type type	matching	kutta-fil	starts at ai #	29	ends at ai #	25		
nw.edge nw/id						11	10	corresponding edge points ( minus (-) indicates point moved by \$seat			
15.2	12	4	mu-match			9	8				
23.4	12	4				7	6				
32.1	18	5	mu-match			5	4				
0abutment #	44	dblt edge	type type	matching	kutta-fil	starts at ai #	30	ends at ai #	25		
nw.edge nw/id						1	2	corresponding edge points ( minus (-) indicates point moved by \$seat			
15.3	12	4	mu-match			3	4				
21.1	12	4				11	10				
22.4	12	4				9	8				
0abutment #	45	dblt edge	type type	matching	kutta-fil	starts at ai #	31	ends at ai #	27		
nw.edge nw/id						1	2	corresponding edge points ( minus (-) indicates point moved by \$seat			
16.1	12	4	mu-match			3	4				
17.3	12	4				11	10				
20.4	12	4				9	8				
0abutment #	46					7	6				

results									
0abutment #	47	dblt edge	starts at ai # 31	ends at ai # 29					
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat				
16.2	12	4			1 2 3 4				
23.4	12	4			7 6 5 4				
32.1	18	5	mu-match		7 7 9 10				
0abutment #	48	dblt edge			starts at ai # 32	ends at ai # 26			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
17.1	12	4			1 1 -2 3	7 8 9 10	11		
18.4	12	4			10 9 8 7	5 4 3 2	1		
32.1	18	5	mu-match		11 -10 9 8	-7 5 4 3	2		
0abutment #	49	dblt strength	matched to zero along this abutment					*** warning	**
*		dblt edge			starts at ai # 32	ends at ai # 31			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
18.1	12	4	mu-match		1 2 3 4	5 6 7	10 11		
0abutment #	50	dblt strength	matched to zero along this abutment					*** warning	**
*		dblt edge			starts at ai # 33	ends at ai # 34			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
18.2	12	4	mu-match		1 2 3 4	5 6 7	10 11		
0abutment #	51	dblt edge			starts at ai # 32	ends at ai # 34			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
18.3	12	4	vor-mtch		1 2 3 4	5 6 7	10 11		
24.1	18	5	mu-match	vor-mtch	7 6 5 4	3 2 1			
0abutment #	52	dblt strength	matched to zero along this abutment					*** warning	**
*		dblt edge			starts at ai # 28	ends at ai # 35			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
19.1	12	4	mu-match		1 2 3 4	5 6 7	10 11		
0abutment #	53	dblt strength	matched to zero along this abutment					*** warning	**
*		dblt edge			starts at ai # 35	ends at ai # 36			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
19.2	12	4	mu-match		1 2 3 4	5 6 7	10 11		
0abutment #	54	dblt edge			starts at ai # 29	ends at ai # 36			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
19.3	12	4	vor-mtch		1 2 3 4	5 6 7	10 11		
25.1	18	5	mu-match	vor-mtch	7 6 5 4	3 2 1			
0abutment #	55	dblt strength	matched to zero along this abutment					*** warning	**
*		dblt edge			starts at ai # 27	ends at ai # 37			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
20.1	12	4	mu-match		1 2 3 4	5 6 7	10 11		
0abutment #	56	dblt strength	matched to zero along this abutment					*** warning	**
*		dblt edge			starts at ai # 37	ends at ai # 38			
nw edge	nw/ id	type type	matching	kutta-fl	corresponding edge points ( minus (-) indicates point moved by \$seat	corresponding edge points ( minus (-) indicates point moved by \$seat			
20.2	12	4	mu-match		1 2 3 4	5 6 7	10 11		

Mar 7 1996 13:51

## results

0abutment #	57	dblt edge type type matching vor-match mu-match	kutta-f1	starts at ai # 31 corresponding edge points ( minus (-) indicates point moved by \$eat	38
nw.edge nw/id	20.3	12 4	1 2 3	5 6 7	
27.1	18 5	7 6 5	4 3 2 1		
0abutment #	58	dblt edge type type matching vor-match mu-match	kutta-f1	starts at ai # 30 corresponding edge points ( minus (-) indicates point moved by \$eat	32
nw.edge nw/id	21.2	12 4	1 -2 3 4		
23.4	12 4	13 -12 11 10			
32.1	18 5	1 -2 3 4			
0abutment #	59	doublet strength matched to zero along this abutment			*** warning ***
*					
nw.edge nw/id	22.1	dblt edge type type matching kutta-f1 mu-match		starts at ai # 25 corresponding edge points ( minus (-) indicates point moved by \$eat	39
0abutment #	60	doublet strength matched to zero along this abutment		1 2 3 4 5 6 7	
*					*** warning ***
nw.edge nw/id	22.2	dblt edge type type matching kutta-f1 mu-match		starts at ai # 39 corresponding edge points ( minus (-) indicates point moved by \$eat	40
0abutment #	61	12 4	1 2 3 4 5 6 7		
nw.edge nw/id	22.3	dblt edge type type matching kutta-f1 vor-match mu-match		starts at ai # 30 corresponding edge points ( minus (-) indicates point moved by \$eat	40
25.1	18 5	7 6 5 4 3 2 1			
0abutment #	62	dblt edge type type matching kutta-f1 mu-match		starts at ai # 41 corresponding edge points ( minus (-) indicates point moved by \$eat	30
nw.edge nw/id	23.1	12 4	1 2		
23.3	12 4	2 1			
0abutment #	63	wake side edge left unabutted			*** warning ***
*					
nw.edge nw/id	24.2	dblt edge type type matching kutta-f1		starts at ai # 34 corresponding edge points ( minus (-) indicates point moved by \$eat	42
0abutment #	64	18 2	1 2		
*		wake trailing edge unabutted.		wake filaments will be added	*** gentle reminder ***
nw.edge nw/id	24.3	dblt edge type type matching kutta-f1		starts at ai # 42 corresponding edge points ( minus (-) indicates point moved by \$eat	43
0abutment #	65	18 2	1 2 3 4 5 6 7		
*		wake side edge left unabutted			
nw.edge nw/id	24.4	dblt edge type type matching kutta-f1		starts at ai # 43 corresponding edge points ( minus (-) indicates point moved by \$eat	32
0abutment #	66	18 2	1 2		
*		wake side edge left unabutted			
nw.edge nw/id	25.2	dblt edge type type matching kutta-f1		starts at ai # 40 corresponding edge points ( minus (-) indicates point moved by \$eat	44
0abutment #	67	18 2	1 2		
*		wake trailing edge unabutted.		wake filaments will be added	*** gentle reminder ***
nw.edge nw/id	25.3	dblt edge type type matching kutta-f1		starts at ai # 44 corresponding edge points ( minus (-) indicates point moved by \$eat	45
0abutment #	68	18 2	1 2 3 4 5 6 7		

	starts at ai #	ends at ai #	corresponding edge points ( minus (-) indicates point moved by \$seat
nw_edge nw_id type type matching kutta-fl	45	30	
25.4 18 2	1	2	
32.2 18 2	2	1	
32.4 18 2	1	2	
*abutment # 69 wake side edge left unabutted			*** warning **
nw_edge nw_id type type matching kutta-fl	36	46	
26.2 18 2	1	2	
*abutment # 70 wake trailing edge unabutted.	wake filaments will be added		*** gentle reminder **
nw_edge nw_id type type matching kutta-fl	46	47	*** gentle reminder **
26.3 18 2	1	2	
*abutment # 71 wake side edge left unabutted	wake filaments will be added		*** warning **
nw_edge nw_id type type matching kutta-fl	47	48	*** warning **
26.4 18 2	1	2	
*abutment # 72 wake side edge left unabutted	wake filaments will be added		*** warning **
nw_edge nw_id type type matching kutta-fl	48	49	*** gentle reminder **
27.2 18 2	1	2	
*abutment # 73 wake trailing edge unabutted.	wake filaments will be added		*** warning **
nw_edge nw_id type type matching kutta-fl	49	50	*** warning **
27.3 18 2	1	2	
*abutment # 74 wake side edge left unabutted	wake filaments will be added		*** warning **
nw_edge nw_id type type matching kutta-fl	49	51	*** warning **
27.4 18 2	1	2	
*abutment # 75 wake side edge left unabutted	wake filaments will be added		*** warning **
nw_edge nw_id type type matching kutta-fl	50	51	*** warning **
28.2 18 2	1	2	
*abutment # 76 wake trailing edge unabutted.	wake filaments will be added		*** gentle reminder **
nw_edge nw_id type type matching kutta-fl	50	51	*** warning **
28.3 18 2	1	2	
*abutment # 77 wake side edge left unabutted	wake filaments will be added		*** warning **
nw_edge nw_id type type matching kutta-fl	51	52	*** warning **
28.4 18 2	1	2	
*abutment # 78 wake side edge left unabutted	wake filaments will be added		*** gentle reminder **

results

Mar 7 1996 13:51

```

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 52 ends at ai # 53
29.3 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake side edge left unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** warning

* * * warning

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 53 ends at ai # 54
29.4 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake side edge left unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** gentle reminder
* * * gentle reminder

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 53 ends at ai # 55
30.2 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake trailing edge unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** warning
* * * warning

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 54 ends at ai # 55
30.3 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake side edge left unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** warning
* * * warning

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 55 ends at ai # 56
30.4 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake side edge left unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** warning
* * * warning

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 55 ends at ai # 56
31.2 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake trailing edge unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** gentle reminder
* * * gentle reminder

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 56 ends at ai # 57
31.3 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake side edge left unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** warning
* * * warning

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 57 ends at ai # 59
31.4 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake trailing edge unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** gentle reminder
* * * gentle reminder

nw.edge nw/!d dblt edge type type matching kutta-f1 starts at ai # 45 ends at ai # 45
32.3 18 2 1 2 3 4 5 6 7 8 indicates point moved by $seat
wake side edge left unabotted corresponding edge points ( minus (-) ) 10 11 12 13 14 15 16 *** warning
* * * warning

0*a*abutment 0*b*nwedmov 1 nw.edge dblt edge type type matching kutta-f1 movement of network edge points
0.50E-03 ** .lt. eps ** tolerance = 0.1911E-02
1 orig X 0.000000 0.000500 0.000000 0.000000
Y 8.499000 2.498000 0.482000 0.000000
Z 0.000000 0.000000 0.000000 0.000000
moved X 3.990000 3.565000 1.853000 0.000000
Y 8.499000 2.498000 0.482000 0.000000
Z 0.000000 0.000000 0.000000 0.000000

```



results												Page 76
summary of networks for trefftz plane analysis												
nwltrf/in	9	24	25	26	27	28	29	30	31	32		
nw	network-id											
nwltrf/out	-----	9	24	25	26	27	28	29	30	31	32	
1.												
24												
25												
26												
27												
28												
29												
30												
31												
32												
t/abtiddn	2.099325											
abtcal/anal	0.0000034											
0*elbgecab												
0 control points for network :	1											
jc jc/naive												
zc												
zc												
u	1	1	45.090395	3.246673	-2.414971	0.000000	0.641888	0.7666799	1	1.1	1	1
000	2	2	43.834818	3.2332000	-2.402644	0.000149	0.641734	0.7665227	1	1.8	2	2
375	3	3	41.158462	3.2332000	-2.402625	0.000000	0.641588	0.767049	1	2.8	3	4
125	4	4	38.481511	3.2332000	-2.402625	0.000000	0.641588	0.767049	1	3.8	4	4
875	5	5	35.8904544	3.2332000	-2.402580	0.000149	0.641392	0.767213	1	4.8	5	8
625	6	6	33.127182	3.2332000	-2.402582	0.000000	0.641288	0.767301	1	5.8	6	10
375	7	7	30.449590	3.2332000	-2.402562	0.000000	0.641288	0.767301	1	6.8	7	12
125	8	8	27.771476	3.2332000	-2.402515	0.000149	0.641050	0.767499	1	7.8	8	14
875	9	9	25.093237	3.2330462	-2.401316	0.000000	0.640987	0.767552	1	8.8	9	16
625	10	10	24.013415	3.231929	-2.402441	0.000000	0.640987	0.767552	1	8.4	10	17
000	11	11	45.085496	3.858495	-2.927114	0.000149	0.641734	0.766927	1	1.5	11	1
625	12	12	43.930127	3.885736	-2.949683	0.000149	0.641734	0.766927	1	1.5	12	2
000	13	13	41.359743	3.885756	-2.94950	0.000000	0.641588	0.767049	1	2.5	13	4
000	14	14	38.789418	3.885803	-2.949489	0.000000	0.641588	0.767049	1	3.5	14	6
000	15	15	36.219655	3.885870	-2.949238	0.000149	0.641392	0.767213	1	4.5	15	8



## results

Page 144

```

pic counts
no influence          0      0      0      0      0      0      0      0
monopole far field   140654  614894  0      0      0      0      0      0
dipole far field     68550   341612  0      0      0      0      0      0
quadrupole far field 35875   148773  0      0      0      0      0      0
one sub-panel intermediate field
two sub-panel intermediate field
eight sub-panel near field

```

```

ncalg= 0 influence coefficient generation i/o count
n6chg: 4 nwdg= 0 ncalt= 0 nwdt= 0
1. 0 16313 112589 16313

logical flags for cp/2 iteration:
F = bkptn, print flag for solver statistics
wopen call on unit 19 blocks: 100 status:
0 *****
* condition indicators
* uniform solution 0.25779E-12 *
* *****
1 0*b*solution

```

simultaneous solution number 1

```

mach number = 0.80000 angle of attack = 2.00000 sideslip angle = 0.00000 freestream speed = 1.00000
compressibility factor = 0.60000 compressibility angle of attack = 2.00000 compressibility angle of sideslip = 0.0000
freestream velocity = ( 0.99939, 0.00000, 0.03490) compressibility direction = ( 0.99939, 0.00000,
1 network id: index: 1 source type = 0 doublet type = 12 number rows = 8 number columns = 8
jc ip x y z d0 dx dy dz aux any a
nz lmachu wxu wyu wzu pheu vxu vyu vzu cplnu cp2ndu cpi
snu lmachl wxl wy1 wzl phel vxl vy1 vz1 cplnl cp2ndl cpi
snl wnu wnl pwnu vt1 pvtu pvtl cp1nd cp2ndd cpi
snd .0234 1 43.9301 3.8857 -2.9497 0.4701 -0.0365 0.1604 -0.1352 0.0000 0.0000 0.0196 0
0.7580 0.9818 0.0789 -0.0662 -0.6205 0.9443 0.0789 -0.0675 0.1172 0.1009 0.0997 0

```

result

results												Page 146	
Mar 7 1996 13:51	j_c	i_p	x	y	z	d0	dx	dy	dz	s0	aux	any	a
.1318	0	0.0000	-0.0269	-0.0269	1.0143	0.9479	0.0150	0.0617	-0.1339	-0.1329	-0.1320	-0	
.19	8	25.9399	3.8863	-2.9489	0.2091	0.1259	-0.0564	0.0500	0.0000	0.0000	0.0195	0	
.0234	0	0.8428	1.0159	-0.0363	0.0303	-0.3655	1.0449	-0.0363	0.0313	-0.0908	-0.0921	-0.0929	-0
.0928	0	0.7307	0.9716	0.0201	-0.0168	-0.5745	0.9190	0.0201	-0.0186	0.1644	0.1614	0.1589	0
.1585	0	0.0000	-0.0268	-0.0268	1.0460	0.9194	0.0522	0.0945	-0.2551	-0.2535	-0.2518	-0	
.2513													
.22	9	43.9956	5.3899	-4.2081	0.6121	-0.0103	0.0309	-0.0261	0.0000	0.0000	0.0178	0	
.0212	0	0.7641	0.9858	0.0114	-0.0099	-0.5134	0.9589	0.0114	-0.0109	0.0841	0.0820	0.0813	0
.0812	0	0.7731	0.9890	-0.0195	0.0159	-1.1255	0.9693	-0.0195	0.0152	0.0616	0.0609	0.0605	0
.0605	0	0.0000	-0.0271	-0.0271	0.9591	0.9696	0.0555	0.0302	0.0225	0.0211	0.0208	0.0208	0
.0207													
.23	10	41.6628	5.3899	-4.2076	0.6418	-0.0113	0.0155	-0.0132	0.0000	0.0000	0.0178	0	
.0212	0	0.7708	0.9885	0.0025	-0.0021	-0.4271	0.9668	0.0025	-0.0028	0.0678	0.0664	0.0660	0
.0660	0	0.7807	0.9923	-0.0130	0.0109	-1.0689	0.9781	-0.0130	0.0104	0.0442	0.0435	0.0433	0
.0433	0	0.0000	-0.0268	-0.0268	0.9668	0.9783	0.0418	0.0220	0.0236	0.0230	0.0227	0.0227	0
.0227													
1	network id:		index:	1	source type = 0	doublet type = 12			number rows = 8	number columns = 8			
nz	j_c	i_p	x	y	z	d0	dx	dy	dz	s0	aux	any	a
lmachu	wxu	wyu	wzu	ph <u>eu</u>	v <u>xu</u>	v <u>yu</u>	v <u>zu</u>	cplinu	cpsinu	cp2ndu	cp1		
snu	wxl	wyl	wzl	ph <u>el</u>	v <u>xl</u>	v <u>yl</u>	v <u>zl</u>	cplinl	cpsil	cp2ndl	cp1		
snl	wnl	pwnu	pwl	v <u>tu</u>	pvtu	pvtu	pvtu	cplind	cpsind	cp2ndd	cp1		
snd													
.24	11	39.3297	5.3900	-4.2074	0.6593	-0.0025	0.0030	-0.0026	0.0000	0.0000	0.0178	0	
.0212	0	0.7787	0.9917	-0.0043	0.0034	-0.3620	0.9760	-0.0043	0.0028	0.0490	0.0480	0.0478	0
.0478	0	0.7810	0.9925	-0.0073	0.0059	-1.0213	0.9785	-0.0073	0.0054	0.0438	0.0429	0.0427	0
.0427	0	0.0000	-0.0269	-0.0269	0.9760	0.9786	0.0291	0.0247	0.0053	0.0052	0.0051	0.0051	0
.0051													
.25	12	36.9972	5.3901	-4.2069	0.6518	0.0096	-0.0072	0.0063	0.0000	0.0000	0.0178	0	
.0212	0	0.7863	0.9946	-0.0095	0.0077	-0.3185	0.9844	-0.0055	0.0073	0.0318	0.0310	0.0309	0
.0309	0	0.7778	0.9913	-0.0022	0.0017	-0.9703	0.9749	-0.0022	0.0011	0.0514	0.0503	0.0500	0
.0500	0	0.0000	-0.0269	-0.0269	0.9845	0.9749	0.0182	0.0116	-0.0196	-0.0193	-0.0192	-0.0192	-0



results												Page 148
.0191	0.7814	0.9926	-0.0109	0.0086	-0.3496	0.9789	-0.0109	0.0082	0.0428	0.0420	0.0418	0
.0418	0.7779	0.9913	-0.0031	0.0021	-1.0010	0.9750	-0.0031	0.0016	0.0511	0.0500	0.0498	0
.0498	0.0000	0.0000	-0.0271	-0.0271	0.9790	0.9750	0.0223	0.0309	-0.0083	-0.0081	-0.0080	-0
.0080												
.35	20	37.7751	6.8943	-5.4646	0.6334	0.0137	-0.0133	0.0115	0.0000	0.0000	0.0159	0
.0191	0.7881	0.9952	-0.0140	0.0115	-0.3148	0.9864	-0.0140	0.0112	0.0275	0.0268	0.0267	0
.0267	0.7759	0.9906	-0.0007	0.0004	-0.9483	0.9728	-0.0007	-0.0003	0.0557	0.0545	0.0542	0
.0542	0.0000	0.0000	-0.0270	-0.0270	0.9866	0.3728	0.0137	0.0345	-0.0281	-0.0277	-0.0275	-0
.0275												
1	network id:	index:	1	source type = 0	doublet type = 12	number rows = 8	number columns = 8	any	any	any	any	a
jC	1P	x	y	z	d0	dx	dy	dz	s0	anx	any	
nz												
lmachu	wxu	wyu	wzu	ph <u>eu</u>	v <u>xu</u>	v <u>yu</u>	v <u>zu</u>	cplinu	cplinu	cp2ndu	cp2ndu	cp1
snu												
lmachl	wxl	wyl	wzl	ph <u>el</u>	v <u>xl</u>	v <u>yl</u>	v <u>zl</u>	cplinl	cplinl	cp2ndl	cp2ndl	cp1
snl												
wnu	wnl	pnwu	pwnl	vtu	vtl	pvtu	pvtl	cplind	cplind	cp2ndd	cp2ndd	cp1
snd												
.36	21	35.6799	6.8943	-5.4639	0.5927	0.0260	-0.0197	0.0170	0.0000	0.0000	0.0159	0
.0191	0.7952	0.9980	-0.0179	0.0144	-0.2957	0.9943	-0.0179	0.0143	0.0117	0.0109	0.0109	0
.0109	0.7720	0.9890	0.0018	-0.0020	-0.8884	0.9683	0.0018	-0.0027	0.0648	0.0635	0.0631	0
.0631	0.0000	0.0000	-0.0271	-0.0271	0.9945	0.9683	0.0051	0.0402	-0.0532	-0.0526	-0.0522	-0
.0522												
.37	22	33.5845	6.8944	-5.4630	0.5225	0.0419	-0.0284	0.0247	0.0000	0.0000	0.0159	0
.0191	0.8034	1.0011	-0.0229	0.0186	-0.2942	1.0033	-0.0229	0.0187	-0.066	-0.0074	-0.0074	-0
.0074	0.7661	0.9866	0.0055	-0.0051	-0.8167	0.9614	0.0055	-0.0060	0.0788	0.0772	0.0766	0
.0766	0.0000	0.0000	-0.0271	-0.0271	1.0037	0.9614	0.0081	0.0487	-0.0854	-0.0846	-0.0840	-0
.0839												
.38	23	31.4892	6.8943	-5.4624	0.4118	0.0690	-0.0454	0.0395	0.0000	0.0000	0.0159	0
.0191	0.8171	1.0062	-0.0320	0.0267	-0.3148	1.0177	-0.0320	0.0271	-0.0360	-0.0371	-0.0373	-0
.0372	0.7554	0.9822	0.0133	-0.0112	-0.7266	0.9487	0.0133	-0.0124	0.1046	0.1023	0.1013	0
.1012	0.0000	0.0000	-0.0268	-0.0268	1.0186	0.9489	0.0268	0.0649	-0.1406	-0.1394	-0.1386	-0
.1384												
.39	24	29.3942	6.8939	-5.4617	0.2250	0.1566	-0.1066	0.0926	0.0000	0.0000	0.0159	0
.0191												

Mar 7 1996 13:51							results					
.1332	0. 8624	1. 0217	-0. 0639	0. 0531	-0. 3754	1. 0626	-0. 0639	0. 0545	-0. 1277	-0. 1320	-0. 1335	-0
.1810	0. 7214	0. 9673	0. 0427	-0. 0359	-0. 6004	0. 9059	0. 0427	-0. 0380	0. 1919	0. 1852	0. 1819	0
.3142	0. 0000	0. 0000	-0. 0270	-0. 0270	1. 0659	0. 9077	0. 0883	0. 1227	-0. 3195	-0. 3172	-0. 3154	-0
.42	25	44. 1269	8. 3988	-6. 7254	0. 6354	0. 0002	-0. 0095	0. 0080	0. 0000	0. 0000	0. 0142	0
.0169	0. 7726	0. 9890	-0. 0100	0. 0078	-0. 4399	0. 9689	-0. 0100	0. 0071	0. 0628	0. 0620	0. 0616	0
.0616	0. 7725	0. 9891	-0. 0005	-0. 0002	-1. 0752	0. 9688	-0. 0005	-0. 0009	0. 0637	0. 0625	0. 0621	0
.0621	0. 0000	0. 0000	-0. 0272	-0. 0272	0. 9690	0. 9688	0. 0321	0. 0380	-0. 0009	-0. 0005	-0. 0005	-0
.0005												
.43	26	42. 2688	8. 3986	-6. 7242	0. 6339	0. 0024	-0. 0117	0. 0098	0. 0000	0. 0000	0. 0141	0
.0169	0. 7777	0. 9910	-0. 0125	0. 0099	-0. 3885	0. 9747	-0. 0125	0. 0093	0. 0512	0. 0504	0. 0502	0
.0501	0. 7756	0. 9904	-0. 0008	0. 0001	-1. 0224	0. 9723	-0. 0008	-0. 0005	0. 0566	0. 0554	0. 0551	0
.0551	0. 0000	0. 0000	-0. 0272	-0. 0272	0. 9748	0. 9723	0. 0256	0. 0349	-0. 0054	-0. 0050	-0. 0049	-0
.0049												
.44	27	40. 4108	8. 3984	-6. 7230	0. 6254	0. 0075	-0. 0148	0. 0126	0. 0000	0. 0000	0. 0141	0
.0169	0. 7831	0. 9932	-0. 0150	0. 0120	-0. 3482	0. 9808	-0. 0150	0. 0116	0. 0388	0. 0381	0. 0380	0
.0379	0. 7764	0. 9908	-0. 0002	-0. 0004	-0. 9737	0. 9733	-0. 0002	-0. 0010	0. 0547	0. 0535	0. 0533	0
.0532	0. 0000	0. 0000	-0. 0272	-0. 0272	0. 9810	0. 9733	0. 0189	0. 0347	-0. 0159	-0. 0154	-0. 0153	-0
.0153												
.45	28	38. 5529	8. 3985	-6. 7221	0. 6045	0. 0161	-0. 0183	0. 0156	0. 0000	0. 0000	0. 0141	0
.0169	0. 7892	0. 9956	-0. 0174	0. 0140	-0. 3198	0. 9876	-0. 0174	0. 0137	0. 0251	0. 0244	0. 0243	0
.0243	0. 7748	0. 9901	0. 0009	-0. 0013	-0. 9243	0. 9715	0. 0009	-0. 0019	0. 0584	0. 0571	0. 0568	0
.0567	0. 0000	0. 0000	-0. 0272	-0. 0272	0. 9878	0. 9715	0. 0118	0. 0370	-0. 0332	-0. 0327	-0. 0324	-0
.0324												
.46	29	36. 6947	8. 3986	-6. 7211	0. 5641	0. 0281	-0. 0236	0. 0203	0. 0000	0. 0000	0. 0141	0
.0169	0. 7962	0. 9983	-0. 0206	0. 0167	-0. 3050	0. 9954	-0. 0206	0. 0166	0. 0093	0. 0086	0. 0086	0
.0086	0. 7711	0. 9887	0. 0029	-0. 0030	-0. 8690	0. 9672	0. 0029	-0. 0037	0. 0670	0. 0656	0. 0652	0
.0651	0. 0000	0. 0000	-0. 0272	-0. 0272	0. 9957	0. 9672	0. 0057	0. 0420	-0. 0576	-0. 0570	-0. 0566	-0
.0565												
.47	30	34. 8366	8. 3984	-6. 7200	0. 4978	0. 0443	-0. 0322	0. 0279	0. 0000	0. 0000	0. 0141	0
.0169	0. 8049	1. 0016	-0. 0256	0. 0208	-0. 3059	1. 0048	-0. 0256	0. 0209	-0. 0098	-0. 0106	-0. 0106	-0
.0106	0. 7653	0. 9863	0. 0066	-0. 0061	-0. 8037	0. 9604	0. 0066	-0. 0070	0. 0808	0. 0791	0. 0785	0

Mar 7 1996 13:51

## results

results												Page 150
network id:	index:	1	source type = 0	doublet type = 12	number rows = 8	s0	dz	vzu	cplinu	cpslnu	anx	any
jc ip	x	y	z	d0	dx	dy	dz					a
nz lmachu	wxu	wyu	wzu	pheu	vxu	vyu	vzu	cplinu	cpslnu	cp2ndu	cp1	
snu lmachl	wxl	wyl	wzl	phel	vxl	vyl	vzl	cplinl	cpslnl	cp2ndl	cp1	
snl wnu	wnl	pnu	pwl	vtu	vtl	pvtu	pvtl	cplind	cpslnd	cp2nnd	cp1	
48 31	32.9791	8.3980	-6.7188	0.3938	0.0734	-0.0501	0.0435	0.0000	0.0000	0.0141	0	
.0169 0	8197	1.0071	-0.0352	0.0291	-0.3278	1.0203	-0.0352	0.0295	-0.0415	-0.0428	-0.0429	
.0429 0	7540	0.9816	0.0149	-0.0128	-0.7216	0.9469	0.0149	-0.0140	0.1082	0.1058	0.1047	
.1046 0	0.0000	0.0000	-0.0270	-0.0270	1.0214	0.9472	0.0315	0.0677	-0.1497	-0.1486	-0.1477	
.1475 49	32	31.1216	8.3979	-6.7179	0.2164	0.1693	-0.1168	0.1013	0.0000	0.0000	0.0141	
.0169 0	8697	1.0241	-0.0695	0.0574	-0.3883	1.0693	-0.0695	0.0590	-0.1415	-0.1468	-0.1486	
.1481 0	7171	0.9653	0.0473	-0.0400	-0.6047	0.9000	0.0473	-0.0423	0.2040	0.1963	0.1926	
.1915 0	0.0000	0.0000	-0.0272	-0.0272	1.0732	0.9023	0.0981	0.1312	-0.3455	-0.3431	-0.3411	
.3396 52	33	44.1925	9.9036	-7.9837	0.5979	0.0014	-0.0196	0.0165	0.0000	0.0000	0.0123	
.0148 0	7754	0.9900	-0.0154	0.0123	-0.4289	0.9720	-0.0154	0.0116	0.0564	0.0557	0.0554	
.0554 0	7741	0.9899	0.0042	-0.0041	-1.0269	0.9705	0.0042	-0.0048	0.0604	0.0589	0.0586	
.0585 0	0.0000	0.0000	-0.0272	-0.0272	0.9722	0.9706	0.0276	0.0407	-0.0040	-0.0032	-0.0032	
.0032 53	34	42.5718	9.9034	-7.9822	0.5938	0.0046	-0.0206	0.0173	0.0000	0.0000	0.0123	
.0148 0	7797	0.9918	-0.0174	0.0136	-0.3883	0.9769	-0.0174	0.0131	0.0465	0.0457	0.0455	
.0455 0	7756	0.9905	0.0032	-0.0036	-0.9821	0.9723	0.0032	-0.0043	0.0568	0.0553	0.0550	
.0550 0	0.0000	0.0000	-0.0275	-0.0275	0.9772	0.9724	0.0225	0.0386	-0.0103	-0.0096	-0.0095	
.0095 54	35	40.9511	9.9029	-7.9807	0.5827	0.0091	-0.0221	0.0187	0.0000	0.0000	0.0123	
.0148 0	7842	0.9935	-0.0189	0.0154	-0.3562	0.9820	-0.0189	0.0150	0.0362	0.0355	0.0354	
.0354 0	7761	0.9907	0.0033	-0.0031	-0.9389	0.9729	0.0033	-0.0037	0.0557	0.0543	0.0540	

results									
.0186	0.0000	0.0000	-0.0271	-0.0271	0.9822	0.9729	0.0175	0.0380	-0.0195
55	36	39.3304	9.9027	-7.9795	0.5632	0.0167	-0.0239	0.0204	0.0000
.0148	0.7896	0.9957	-0.0208	0.0164	-0.3323	0.9879	-0.0208	0.0162	0.0242
.0234	0.7747	0.9901	0.0031	-0.0035	-0.8956	0.9713	0.0031	-0.0042	0.0589
.0571	0.0000	0.0000	-0.0275	0.9883	0.9713	0.0121	0.0393	-0.0347	-0.0340
.0337									-0.0338
56	37	37.7098	9.9028	-7.9781	0.5269	0.0293	-0.0281	0.0241	0.0000
.0148	0.7969	0.9985	-0.0235	0.0187	-0.3200	0.9960	-0.0235	0.0186	0.0080
.0071	0.7707	0.9885	0.0046	-0.0047	-0.8469	0.9667	0.0046	-0.0055	0.0682
.0662	0.0000	0.0000	-0.0275	0.9964	0.9667	0.0083	0.0440	-0.0603	-0.0595
.0591									-0.0591
57	38	36.0892	9.9027	-7.9767	0.4661	0.0467	-0.0362	0.0313	0.0000
.0148	0.8062	1.0021	-0.0281	0.0228	-0.3224	1.0060	-0.0281	0.0229	-0.0124
.0134	0.7644	0.9859	0.0082	-0.0075	-0.7885	0.9593	0.0082	-0.0084	0.0831
.0805	0.0000	0.0000	-0.0273	1.0067	0.9594	0.0154	0.0525	-0.0955	-0.0946
.0939									-0.0940
58	39	34.4689	9.9023	-7.9753	0.3703	0.0782	-0.0549	0.0476	0.0000
.0148	0.8222	1.0079	-0.0380	0.0311	-0.3443	1.0229	-0.0380	0.0316	-0.0467
.0484	0.7521	0.9808	0.0169	-0.0148	-0.7146	0.9447	0.0169	-0.0160	0.1128
.1088	0.0000	0.0000	-0.0273	-0.0273	1.0241	0.9450	0.0359	0.0712	-0.1595
.1571									-0.1583
59	40	32.8484	9.9021	-7.9739	0.2042	0.1828	-0.1270	0.1100	0.0000
.0148	0.8773	1.0265	-0.0749	0.0615	-0.4030	1.0762	-0.0749	0.0632	-0.1555
.1634	0.7122	0.9630	0.0522	-0.0444	-0.6072	0.8934	0.0522	-0.0468	0.2175
.2032	0.0000	0.0000	-0.0275	1.0807	0.8962	0.1078	0.1406	-0.3731	-0.3705
.3666									-0.3684
1									-0
Network id:									
index: 1 source type = 0 doublet type = 12 number rows = 8 number columns = 8									
jc	ip	x	y	z	d0	dx	dy	dz	s0
nz	lmachu	wxu	wyu	wzu	pheu	vku	vyu	vzu	cplinu
snu	lmachl	wxl	wyl	wzl	phel	vxl	vyl	vzl	cplinl
snl	wnu	wnl	pwnu	pwnl	vtu	vtl	pvtu	pvtl	cplind
									cplnd
									cpi

Results											
snd	41	44.2579	11.4079	-9.2422	0.5332	0.0019	-0.0316	0.0265	0.0000	0.0105	0
.0126	62	0.7779	0.9909	-0.0220	0.0173	-0.4324	0.9746	-0.0220	0.0168	0.0508	0.0497
.0497	0	0.7761	0.9908	0.0096	-0.0091	-0.9657	0.9728	0.0096	-0.0097	0.0564	0.0541
.0540	0	0.0000	0.0000	-0.0276	-0.0276	0.9750	0.9729	0.0253	0.0447	-0.0056	-0.0043
.0043											
.0126	63	42	42.8745	11.4079	-9.2404	0.5290	0.0052	-0.0318	0.0267	0.0000	0.0105
.0418	0	0.7814	0.9923	-0.0235	0.0182	-0.4007	0.9786	-0.0235	0.0177	0.0428	0.0418
.0527	0	0.7767	0.9910	0.0083	-0.0084	-0.9297	0.9734	0.0083	-0.0090	0.0550	0.0528
.0109	0	0.0000	0.0000	-0.0279	-0.0279	0.9790	0.9735	0.0219	0.0432	-0.0122	-0.0110
.0180	64	43	41.4910	11.4075	-9.2387	0.5189	0.0087	-0.0320	0.0269	0.0000	0.0105
.0126	0	0.7848	0.9936	-0.0240	0.0196	-0.3750	0.9824	-0.0240	0.0192	0.0350	0.0341
.0341	0	0.7769	0.9911	0.0080	-0.0071	-0.8939	0.9738	0.0080	-0.0077	0.0542	0.0521
.0521	0	0.0000	0.0000	-0.0271	-0.0271	0.9829	0.9738	0.0189	0.0421	-0.0192	-0.0180
.0180	65	44	40.1079	11.4070	-9.2370	0.5042	0.0150	-0.0326	0.0276	0.0000	0.0105
.0126	0	0.7893	0.9954	-0.0255	0.0199	-0.3543	0.9874	-0.0255	0.0196	0.0250	0.0241
.0241	0	0.7757	0.9906	0.0071	-0.0073	-0.8584	0.9724	0.0071	-0.0080	0.0570	0.0549
.0549	0	0.0000	0.0000	-0.0279	-0.0279	0.9879	0.9724	0.0154	0.0426	-0.0320	-0.0311
.0308											
.0126	66	45	38.7248	11.4071	-9.2351	0.4755	0.0279	-0.0352	0.0300	0.0000	0.0105
.0080	0	0.7965	0.9982	-0.0275	0.0215	-0.3428	0.9954	-0.0275	0.0214	0.0090	0.0080
.07714	0	0.7714	0.9888	0.0077	-0.0079	-0.8183	0.9674	0.0077	-0.0086	0.0663	0.0647
.0646	0	0.0000	0.0000	-0.0279	-0.0279	0.9960	0.9675	0.0131	0.0465	-0.0579	-0.0566
.0566											
.0126	67	46	37.3418	11.4069	-9.2334	0.4248	0.0469	-0.0415	0.0357	0.0000	0.0105
.0080	0	0.8064	1.0020	-0.0309	0.0250	-0.3443	1.0061	-0.0309	0.0251	-0.0127	-0.0138
.07644	0	0.9859	0.0106	-0.0096	-0.7691	0.9593	0.0106	-0.0106	0.0834	0.0814	0.0807
.0806	0	0.0000	0.0000	-0.0274	-0.0274	1.0069	0.9594	0.0187	0.0548	-0.0962	-0.0952
.0944											
.0180	68	47	35.9585	11.4065	-9.2317	0.3406	0.0821	-0.0598	0.0517	0.0000	0.0105

results												
Mat 7 1996 13:51	1	0.0086	-0.0408	0.0328	-0.3642	1.0248	-0.0408	0.0334	-0.0507	-0.0524	-0.0526	-0
.0126	0.8242	1.0086	-0.0408	0.0328	-0.3642	1.0248	-0.0408	0.0334	-0.0507	-0.0524	-0.0526	-0
.0526	0.7506	0.9801	0.0190	-0.0170	-0.7048	0.9427	0.0190	-0.0183	0.1170	0.1140	0.1127	0
.1125	0.0000	0.0000	-0.0277	-0.0277	1.0262	0.9431	0.0396	0.0747	-0.1677	-0.1664	-0.1654	-0
.1651												
.69	48	34.5750	11.4064	-9.2298	0.1888	0.1978	-0.1381	0.1195	0.0000	0.0000	0.0105	0
.0126	0.8855	1.0291	-0.0806	0.0657	-0.4193	1.0836	-0.0806	0.0676	-0.1707	-0.1780	-0.1807	-0
.1799	0.7067	0.9604	0.0575	-0.0493	-0.6081	0.8859	0.0575	-0.0519	0.2330	0.2228	0.2179	0
.2164	0.0000	0.0000	-0.0279	-0.0279	1.0887	0.8892	0.1182	0.1511	-0.4036	-0.4009	-0.3986	-0
.3962												
.72	49	44.3232	12.9125	-10.5007	0.4285	0.0014	-0.0546	0.0457	0.0000	0.0000	0.0087	0
.0104	0.7803	0.9914	-0.0341	0.0272	-0.4570	0.9768	-0.0341	0.0267	0.0458	0.0446	0.0444	0
.0444	0.7788	0.9919	0.0204	-0.0185	-0.8854	0.9753	0.0204	-0.0190	0.0519	0.0486	0.0484	0
.0483	0.0000	0.0000	-0.0278	-0.0278	0.9777	0.9757	0.0308	0.0557	-0.0060	-0.0040	-0.0040	-0
.0039												
.73	50	43.1771	12.9122	-10.4987	0.4256	0.0040	-0.0545	0.0457	0.0000	0.0000	0.0087	0
.0104	0.7828	0.9924	-0.0352	0.0277	-0.4325	0.9795	-0.0352	0.0272	0.0403	0.0390	0.0389	0
.0388	0.7789	0.9920	0.0193	-0.0179	-0.8581	0.9755	0.0193	-0.0184	0.0515	0.0484	0.0481	0
.0481	0.0000	0.0000	-0.0281	-0.0281	0.9805	0.9759	0.0297	0.0545	-0.0112	-0.0093	-0.0093	-0
.0092												
1	network id:	index:	1	source type = 0	doublet type = 12	number rows = 8	number columns = 8					
jC	iP	x	y	z	d0	dx	dy	dz	s0	anx	any	a
nz	lmachu	wxu	wyu	wzu	pheu	vxu	vyy	vzu	cplinu	cplslnu	cplndu	cpi
snu	lmachl	wxl	wyl	wzl	phel	vxl	vyl	vzl	cplnl	cplslnl	cplndl	cpi
snl	wnu	wnl	pwmu	pwnl	vtu	vtl	pvtu	pvtl	cplind	cplndl	cplndd	cpi
snd												
74	51	42.0309	12.9120	-10.4967	0.4194	0.0065	-0.0537	0.0450	0.0000	0.0000	0.0087	0
.0104	0.7852	0.9934	-0.0354	0.0282	-0.4115	0.9822	-0.0354	0.0278	0.0348	0.0335	0.0334	0
.0333	0.7790	0.9920	0.0183	-0.0167	-0.8309	0.9757	0.0183	-0.0172	0.0509	0.0480	0.0477	0
.0477	0.0000	0.0000	-0.0278	-0.0278	0.9833	0.9760	0.0284	0.0530	-0.0162	-0.0145	-0.0144	-0
.0144												
75	52	40.8853	12.9116	-10.4945	0.4102	0.0110	-0.0534	0.0448	0.0000	0.0000	0.0087	0

results										Page 154
Mar 7 1998 13:51										
.0261	0.7885	0.9947	-0.0361	0.0283	-0.3936	0.9859	-0.0361	0.0280	0.0275	0.0262
.0495	0.7782	0.9917	0.0172	-0.0162	-0.8038	0.9748	0.0172	-0.0168	0.0527	0.0498
0.0000	0.0000	-0.0282	-0.0282	0.9869	0.9751	0.0269	0.0534	-0.0252	-0.0236	-0.0234
.0234										-0
.0104	76	53	39.7398	12.9114	-10.4924	0.3929	0.0211	-0.0536	0.0452	0.0000
.0135	0.7941	0.9969	-0.0368	0.0290	-0.3813	0.9921	-0.0368	0.0288	0.0150	0.0136
.0570	0.7749	0.9903	0.0168	-0.0157	-0.7742	0.9711	0.0168	-0.0164	0.0602	0.0574
0.0000	0.0000	-0.0281	-0.0281	0.9932	0.9713	0.0254	0.0538	-0.0452	-0.0438	-0.0435
.0435										-0
.0104	77	54	38.5940	12.9110	-10.4904	0.3601	0.0390	-0.0560	0.0477	0.0000
.0068	0.8033	1.0005	-0.0382	0.0309	-0.3780	1.0022	-0.0382	0.0309	-0.0053	-0.0068
.0725	0.7680	0.9875	0.0178	-0.0159	-0.7382	0.9632	0.0178	-0.0167	0.0760	0.0732
.0793	0.0000	-0.0000	-0.0276	-0.0276	1.0034	0.9635	0.0269	0.0591	-0.0813	-0.0800
.0104	78	55	37.4480	12.9106	-10.4883	0.2983	0.0783	-0.0700	0.0601	0.0000
.0493	0.8227	1.0078	-0.0463	0.0364	-0.3905	1.0228	-0.0463	0.0369	-0.0470	-0.0491
.07523	0.9809	0.0238	-0.0219	-0.6887	0.9445	0.0238	-0.0232	0.1138	0.1101	0.1089
.1087	0.0000	-0.0000	-0.0285	-0.0285	1.0245	0.9451	0.0437	0.0780	-0.1608	-0.1592
.1580										-0.1583
.0104	79	56	36.3021	12.9107	-10.4861	0.1687	0.2123	-0.1512	0.1306	0.0000
.1961	0.8937	1.0315	-0.0871	0.0708	-0.4378	1.0907	-0.0871	0.0728	-0.1852	-0.1940
.2289	0.7016	0.9578	0.0641	-0.0550	-0.6066	0.8784	0.0641	-0.0578	0.2483	0.2364
.4250	0.0000	0.0000	-0.0282	-0.0282	1.0966	0.8826	0.1291	0.1625	-0.4335	-0.4304
.0083	82	57	44.3885	14.4174	-11.7593	0.2395	-0.0004	-0.1363	0.1141	0.0000
.0348	0.7852	0.9910	-0.0761	0.0613	-0.5250	0.9777	-0.0761	0.0609	0.0415	0.0350
.0371	0.7847	0.9937	0.0602	-0.0526	-0.7644	0.9781	0.0602	-0.0532	0.0487	0.0375
.0024	0.0000	0.0000	-0.0285	-0.0285	0.9826	0.9814	0.0781	0.1049	-0.0072	-0.0025
.0304	83	58	43.4799	14.4169	-11.7570	0.2381	0.0018	-0.1356	0.1135	0.0000
.07872	0.9918	-0.0760	0.0618	-0.5067	0.9799	-0.0760	0.0614	0.0371	0.0306	0.0305
0.7846	0.9936	0.0596	-0.0515	-0.7448	0.9780	0.0596	-0.0521	0.0487	0.0378	0.0375

Mar 7 1995 13:51

Results

Results											
<b>network id:</b>											
<b>index:</b>											
<b>source type = 0</b>											
<b>doublet type = 12</b>											
<b>number rows = 8</b>											
<b>number columns = 8</b>											
<i>nz</i>	<i>jc</i>	<i>ip</i>	<i>x</i>	<i>y</i>	<i>z</i>	<i>d0</i>	<i>dx</i>	<i>dy</i>	<i>dz</i>	<i>s0</i>	<i>anx</i>
1	0.070	0.0000	0.0000	-0.0281	-0.0281	0.9848	0.9812	0.0778	0.1037	-0.0116	-0.0072
2	0.074	0.0000	0.0000	-0.0281	-0.0281	0.9848	0.9812	0.0778	0.1037	-0.0116	-0.0072
3	0.083	0.7923	0.9939	-0.0752	0.0605	-0.4605	0.9858	-0.0752	0.0503	0.0254	0.0191
4	0.090	0.7841	0.9936	0.0550	-0.0480	-0.6884	0.9780	0.0550	-0.0486	0.0485	0.0386
5	0.083	0.0000	0.0000	-0.0285	-0.0285	0.9905	0.9808	0.0752	0.0980	-0.0231	-0.0195
6	0.083	0.7955	0.9954	-0.0731	0.0591	-0.4498	0.9898	-0.0731	0.0589	0.0176	0.0116
7	0.016	0.7817	0.9927	0.0523	-0.0454	-0.6683	0.9758	0.0523	-0.0460	0.0529	0.0437
8	0.033	0.0000	0.0000	-0.0282	-0.0282	0.9942	0.9783	0.0722	0.0950	-0.0353	-0.0321
9	0.017	0.0000	0.0000	-0.0282	-0.0282	0.9942	0.9783	0.0722	0.0950	-0.0353	-0.0318
10	0.083	0.8084	1.0005	-0.0738	0.0587	-0.4450	1.0040	-0.0738	0.0589	-0.0109	-0.0170
11	0.016	0.7706	0.9881	0.0513	-0.0454	-0.6439	0.9631	0.0513	-0.0463	0.0783	0.0693
12	0.0685	0.0000	0.0000	-0.0289	-0.0289	0.9863	0.9816	0.0775	0.1021	-0.0137	-0.0095
13	0.0093	0.0000	0.0000	-0.0289	-0.0289	0.9863	0.9816	0.0775	0.1021	-0.0137	-0.0094
14	0.084	59	42.5713	14.4167	-11.7548	0.2366	0.0029	-0.1345	0.1125	0.0000	0.0000
15	0.0273	0.7886	0.9923	-0.0766	0.0612	-0.4894	0.9815	-0.0766	0.0608	0.0340	0.0275
16	0.0366	0.7849	0.9938	0.0580	-0.0512	-0.7260	0.9785	0.0580	-0.0517	0.0477	0.0370
17	0.0135	0.0000	0.0000	-0.0286	-0.0286	0.9883	0.9815	0.0768	0.1004	-0.0175	-0.0137
18	0.085	60	41.6630	14.4162	-11.7521	0.2332	0.0049	-0.1330	0.1112	0.0000	0.0000
19	0.0083	0.7903	0.9931	-0.0762	0.0612	-0.4741	0.9834	-0.0762	0.0609	0.0301	0.0235
20	0.0234	0.7848	0.9938	0.0568	-0.0498	-0.7072	0.9786	0.0568	-0.0503	0.0476	0.0372
21	0.0369	0.0000	0.0000	-0.0286	-0.0286	0.9883	0.9815	0.0768	0.1004	-0.0175	-0.0137
22	0.0135	0.0000	0.0000	-0.0286	-0.0286	0.9883	0.9815	0.0768	0.1004	-0.0175	-0.0137
23	0.085	61	40.7547	14.4158	-11.7497	0.2278	0.0077	-0.1302	0.1088	0.0000	0.0000
24	0.0083	0.7923	0.9939	-0.0752	0.0605	-0.4605	0.9858	-0.0752	0.0603	0.0254	0.0191
25	0.0190	0.7841	0.9936	0.0550	-0.0480	-0.6884	0.9780	0.0550	-0.0486	0.0485	0.0386
26	0.0383	0.0000	0.0000	-0.0285	-0.0285	0.9905	0.9808	0.0752	0.0980	-0.0231	-0.0195
27	0.0193	0.0000	0.0000	-0.0285	-0.0285	0.9905	0.9808	0.0752	0.0980	-0.0231	-0.0194
28	0.086	61	40.7547	14.4158	-11.7497	0.2278	0.0077	-0.1302	0.1088	0.0000	0.0000
29	0.0083	0.7923	0.9939	-0.0752	0.0605	-0.4605	0.9858	-0.0752	0.0603	0.0254	0.0190
30	0.0190	0.7841	0.9936	0.0550	-0.0480	-0.6884	0.9780	0.0550	-0.0486	0.0485	0.0386
31	0.0383	0.0000	0.0000	-0.0285	-0.0285	0.9905	0.9808	0.0752	0.0980	-0.0231	-0.0194
32	0.0193	0.0000	0.0000	-0.0285	-0.0285	0.9905	0.9808	0.0752	0.0980	-0.0231	-0.0194
33	0.087	62	39.8463	14.4154	-11.7475	0.2185	0.0140	-0.1254	0.1050	0.0000	0.0000
34	0.0083	0.7955	0.9954	-0.0731	0.0591	-0.4498	0.9898	-0.0731	0.0589	0.0176	0.0116
35	0.0116	0.7817	0.9927	0.0523	-0.0454	-0.6683	0.9758	0.0523	-0.0460	0.0529	0.0437
36	0.0433	0.0000	0.0000	-0.0282	-0.0282	0.9942	0.9783	0.0722	0.0950	-0.0353	-0.0321
37	0.0317	0.0000	0.0000	-0.0282	-0.0282	0.9942	0.9783	0.0722	0.0950	-0.0353	-0.0318
38	0.088	63	38.9378	14.4150	-11.7451	0.1989	0.0410	-0.1251	0.1052	0.0000	0.0000
39	0.0083	0.8084	1.0005	-0.0738	0.0587	-0.4450	1.0040	-0.0738	0.0589	-0.0109	-0.0170
40	0.0169	0.7706	0.9881	0.0513	-0.0454	-0.6439	0.9631	0.0513	-0.0463	0.0783	0.0693
41	0.0685	0.0000	0.0000	-0.0289	-0.0289	0.9863	0.9816	0.0775	0.1021	-0.0137	-0.0094

results Mar 7 1996 13:51

		results							
		area	fx	fy	fz	mx	my	mz	
totals for column	5	area	fx	fy	fz	mx	my	mz	
	5	0.0000	0.00073	0.00088	0.01453	0.00560	-0.0046		
	8	0.0001	0.00845	0.01012	0.16733	0.04459	-0.0372		
	2	0.0001	0.00917	0.01100	0.18186	0.05018	-0.0418		
totals for column	6	area	fx	fy	fz	mx	my	mz	
	4	0.0000	0.00094	0.00114	0.02161	0.00619	-0.0051		
	2	0.0001	0.00726	0.00870	0.16598	0.03775	-0.0315		
	6	0.0001	0.00820	0.00984	0.18759	0.04394	-0.0366		
totals for column	7	area	fx	fy	fz	mx	my	mz	
	6	0.0000	0.00085	0.00103	0.02206	0.00525	-0.0043		
	6	0.0001	0.00580	0.00695	0.14997	0.02966	-0.0247		
	2	0.0001	0.00665	0.00737	0.17204	0.03491	-0.0291		
totals for column	8	area	fx	fy	fz	mx	my	mz	
	4	0.0000	0.00037	0.00045	0.01058	0.00233	-0.0019		
	9	0.0001	0.00352	0.00422	0.10168	0.01772	-0.0147		
	3	0.0001	0.00389	0.00467	0.11226	0.02005	-0.0167		
totals for network	area	fx	fy	fz	mx	my	mz		
	9	0.0000	-0.00237	-0.00279	0.02527	0.00427	-0.0033		
	9	0.0005	0.06546	0.07837	1.07455	0.35320	-0.2950		

Mar 7 1996 13:51 results

3	2	218.09473	0.00006	0.06308	0.07559	1.09982	0.35747	-0.2984
totals for all networks so far								
9		218.09473	0.00000	-0.00237	-0.00279	0.02527	0.00427	-0.0033
3		218.09473	0.00005	0.06546	0.07837	1.07455	0.35320	-0.2950
2		218.09473	0.00006	0.06308	0.07559	1.09982	0.35747	-0.2984
0*e*for-mom								
1	network id:	index:	2	source type = 0	doublet type = 12	number rows = 8	number columns = 8	
j_c	ip	x	y	z	d0	dx	dy	dz
nz	wxu	wyu	wzu	pheu	vxu	vyy	vzu	cpliu
lmachu				phel	vxl	vyl	vzl	cplnl
snu	wxl	wyl	wzl			pvtu	pvtl	cplnd
lmachl								cplndl
snl	wnl	pnwl	vnwl					cplndd
snd								cpli
.112	65	43.9301	-2.9497	-3.8857	-0.3830	0.0239	0.1109	0.1332
.0196	0.7810	0.9695	0.0571	0.0684	-1.0960	0.9741	0.0571	0.0679
.0436	0.7609	0.9839	-0.0538	-0.0641	-0.7130	0.9501	-0.0538	-0.0653
.0914	0.0000	0.0000	0.0222	0.0222	0.9781	0.9539	0.0669	0.1218
.0477								-0.0571
.113	66	41.3597	-2.9495	-3.8858	-0.4581	0.0263	0.0643	0.0775
.0196	0.7883	0.9942	0.0345	0.0412	-1.0532	0.9853	0.0345	0.0409
.0265	0.7655	0.9864	-0.0298	-0.0356	-0.5950	0.9590	-0.0298	-0.0366
.0791	0.0000	0.0000	0.0224	0.0224	0.9867	0.9601	0.0302	0.0842
.0526								-0.0579
.114	67	38.7894	-2.9495	-3.8858	-0.5064	0.0122	0.0308	0.0371
.0196	0.7849	0.9936	0.0181	0.0216	-1.0143	0.9826	0.0181	0.0212
.0339	0.7743	0.9901	-0.0127	-0.0152	-0.5079	0.9705	-0.0127	-0.0159
.0583	0.0000	0.0000	0.0224	0.0224	0.9830	0.9707	0.0168	0.0553
.0245								-0.0269
.115	68	36.2197	-2.9493	-3.8859	-0.5195	-0.0020	0.0076	0.0091
.0196	0.7800	0.9921	0.0066	0.0081	-0.9653	0.9774	0.0066	0.0076
								0.0459
								0.0451
								0.0449
								0



## LIST OF REFERENCES

1. Epton, M., Magnus, A., "PAN AIR - A Computer Program for Predicting Subsonic or Supersonic Linear Potential Flows About Arbitrary Configurations Using a Higher Order Panel Method", Volume I - Theory Document (Version 3.0), NASA Contractor Report 3251, Ames Research Center, Moffett Field, CA., March 1992.
2. Bertin, J. J., and Smith, M. L., Aerodynamics for Engineers, 2nd ed., Prentice-Hall, 1989.
3. Anderson, J. D., Fundamentals of Aerodynamics, 2nd ed., McGraw-Hill, 1991.
4. Saaris, G. R., "A502i User's Manual-PAN AIR Technology Program for Solving Problems of Potential Flow about Arbitrary Configurations", Boeing Document Number D6-54703, Boeing, February 1992.
5. Cenko, A., "Determination of Correct AIWS Carriage Loads", Report Number NAWCADAWR-92095-60, Naval Air Warfare Center Aircraft Division, Warminster, PA., June 1992.
6. Cenko, A., Tinoco, E., Dyer, R., DeJongh, J., "PAN AIR Applications to Weapons Carriage and Separation", AIAA Paper 80-0187, 18th Aerospace Sciences Meeting, Pasadena, CA, January 1980.
7. Cenko, A., Talbot, M., Piranian, A., "Analysis of the F-14/GBU-24 and Wind Tunnel Test", Air Vehicle and Crew Systems Technology Department (Code 6053), Naval Air Warfare Center Aircraft Division, Warminster, PA., December 1995.
8. Johnson, F., Samant, S., Bieterman, M., Melvin, R., Young, D., Bussoletti, J. Hilmes, C., "TranAir: A Full-Potential, Solution-Adaptive, Rectangular Grid Code for Predicting Subsonic, Transonic, and Supersonic Flows about Arbitrary Configurations", Theory Document, NASA Contractor Report 4348, Ames Research Center, Moffett Field, CA., December 1992.
9. Keube, F., "Low Apsect Ratio Wings with Small Thickness at Zero Lift in Subsonic and Supersonic Flow", KTH-Aero TN21, Royal Institute of Technology, Stockholm, Sweden, June 1952.
10. Hermstad, D., RAID User's Guide, NASA Ames Division Federal Systems Group, Sterling Software, Inc., Palo Alto, CA., March 1991.



## INITIAL DISTRIBUTION LIST

1. Defense Technical Information Center..... 2  
8725 John J. Kingman Rd., STE 0944  
Ft. Belvoir, VA 22060-6218
2. Dudley Knox Library..... 2  
Naval Postgraduate School  
Monterey, CA 93943-5100
3. Chairman..... 1  
Department of Aeronautics and Astronautics, Code AA  
Naval Postgraduate School  
Monterey, CA 93943-5000
4. Dr. Max F. Platzer..... 5  
Department of Aeronautics and Astronautics, Code AA/PL  
Naval Postgraduate School  
Monterey, CA 93943-5000
5. Dr. Kevin Jones..... 1  
Department of Aeronautics and Astronautics, Code AA/Jo  
Naval Postgraduate School  
Monterey, CA 93943-5000
6. Dr. Ismail Tuncer..... 1  
Department of Aeronautics and Astronautics, Code AA/Tu  
Naval Postgraduate School  
Monterey, CA 93943-5000
7. LT Matthew LeTourneau..... 1  
Department of Aeronautics and Astronautics, Code 31  
Naval Postgraduate School  
Monterey, CA 93943-5000



GOLLEY KNOX LIBRARY  
CIVIL POSTGRADUATE SCHOOL  
MONTEREY CA 93943-5101

DUDLEY KNOX LIBRARY



3 2768 00322908 9