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# Stress-Intensity Factors for an Insulated Half-Plane Crack 

This paper is concerned with determining the stress-intensity factors due to disturbance of a uniform flow of heat by an insulated half-plane crack in an elastic solid. The spatial thermoelastic problem is formulated in terms of Papkovich-Neuber displacement potentials and is solved by the application of Kontorovich-Lebedev integral transform and certain singular solutions of Laplace equation in three dimensions. The analysis reveals that four distinct displacement potentials are needed to satisfy the finite displacement and inverse square root stress-singularity at the edge of the crack. Closed-form expressions are obtained for the stress-intensity factors ( $\mathrm{k}_{2}$ and $\mathrm{k}_{3}$ ) and their variations along the crack border are shown in curves.

## Introduction

When an undisturbed heat flow with constant thermal gradient is diverted round a sharp edge of an insulated crack or flaw, there is local intensification of the temperature gradient accompanied by singular thermal stress which may cause crack propagation resulting in serious damage to structural members. Since the critical value of the intensity of the local stress field can be associated with the fracture toughness of the material, it follows that by knowing the stress-intensity factors as functions of the temperature gradi. ent, material properties, and flaw size, it is possible to predict a critical temperature gradient which will not result in failure of the material. Confining attention to three-dimensional cracks or planes of discontinuities, the local intensity of the thermal stress has been determined for the so-called "penny-shaped" crack [1] ${ }^{1}$ and the more general elliptical crack [2]. In case of the elliptical geometry, both factors $k_{2}$ and $k_{3}$ (associated with the edge sliding and tearing modes of fracture [3]), are found to be operating, while for the axially symmetric crack, the local stress field is controlled by $k_{2}$ only. A summary of the solutions to these and other closely related crack problems is given in [4].

The objective of this investigation is to provide stress-intensity factor solutions to the thermoelastic problem of a uniform flow of heat disturbed by an insulated half-plane crack embedded in an elastic solid. The solid is assumed to be isotropic and homogeneous and the effects of both inertia and coupling between temperature

[^0]

Fig. 1 Semi-infinite plane crack
and strain fields are neglected. Fig. 1 shows a sketch of the crack. In terms of the coordinates indicated, the crack occupies the region $x \leq 0,|z|<\infty$ of the midplane $y=0$. The general thermal stress problem requires first finding the temperature, $T(x, y, z)$, at every point of the solid from Laplace equation in three dimensions

$$
\begin{equation*}
\nabla^{2} T(x, y, z)=0 \tag{1}
\end{equation*}
$$

and consequently the induced displacements from the Navier equations of static equilibrium

$$
\begin{equation*}
\nabla \cdot \nabla \mathbf{u}+(1-2 \nu) \nabla^{2} \mathbf{u}=2(1+\nu) \alpha \nabla T \tag{2}
\end{equation*}
$$

In equations (1) and (2), u designates the displacement vector, $v$ and $\alpha$ are, respectively, the Poisson's ratio and the coefficient of thermal expansion of the material of the solid, and $\nabla, \nabla^{2}$ are the
usual del and Laplacian operators. Once the displacements are known, the stresses can be computed in straightforward manner from the Duhamel-Neumann stress-displacement relations.

The analysis shows that four distinct harmonic functions are required to satisfy the finite displacement and inverse square root singularity conditions at the crack edge ( $r \rightarrow 0$ ). The state of the disturbed temperature field and the associated stresses throughout the solid is determined mainly in closed form. These are obtained by employing the techniques of Fourier and Kontorovich-Lebedev integral transforms and certain singular solutions of Laplace equation [5]. Explicit expressions for the stress-intensity factors are found and their variations along the crack border are displayed graphically. It should also be mentioned that the Green's functions due to concentrated normal and shear forces applied to the surfaces of the half-plane crack are treated in the works of Uflyand [5] and Kassir and Sih [6].

## Temperature Field

The disturbed temperature field at every point of the solid is obtained from a solution of equation (1) in the region $y \geqslant 0$ subject to the conditions

$$
\begin{gather*}
\frac{\partial T}{\partial y}=-Q(x, z), y=0, \theta=\pi  \tag{3a}\\
T=0, y=0, \theta=0 \tag{3b}
\end{gather*}
$$

in which $Q(x, z)$ is the specified temperature gradient. The desired expression for the temperature is conveniently constructed by employing a Fourier cosine transform in the variable $z$ and a Kon-torovich-Lebedev transform in the variable $r$. Toward this end, let

$$
T=\left(\frac{2}{\pi}\right)^{1 / 2} \int_{0}^{\infty} \int_{0}^{\infty} A(s, t) \frac{\sinh (\theta t)}{t \cosh (\pi t)} K_{i t}(s r) \cos (s z) d s d t
$$

where $K_{i t}(s r)$ is the modified Bessel function of the second kind of imaginary order [7] and $A(s, t)$ is an arbitrary function such that equation (3b) is satisfied. Inserting (4) into (3a) and carrying out the appropriate inversions, $A(s, t)$ is determined as

$$
\begin{equation*}
A(s, t)=\frac{2^{3 / 2}}{\pi^{5 / 2}} t \sinh (\pi t) \int_{0}^{\infty} Q^{*}(r, s) K_{i t}(s r) d r \tag{5a}
\end{equation*}
$$

where

$$
\begin{equation*}
Q^{*}(r, s)=\int_{0}^{\infty} Q(r, z) \cos (s z) d z \tag{5b}
\end{equation*}
$$

In order to establish the Green's function for the temperature field, suppose that there is a temperature gradient of magnitude $Q_{0} \delta(x+a) \delta(z)$ at the point $r=a, \theta=\pi, z=0$ of the crack surface, then equations (5) yield

$$
\begin{equation*}
A(s, t)=\frac{\sqrt{2}}{\pi^{5 / 2}} Q_{0} t \sinh (\pi t) K_{i t}(a s) \tag{6}
\end{equation*}
$$

Inserting equation (6) into (4) and performing the resulting integrals [4], the temperature field is obtained

$$
\begin{equation*}
T(r, \theta, z)=\frac{Q_{0}}{\pi^{2} \rho} \tan ^{-1}\left(\frac{\sqrt{2 a(r-x)}}{\rho}\right) \tag{7}
\end{equation*}
$$

where $\rho$ represents the distance of any point in $y \geqslant 0$ to the point $r$ $=a, \theta=\pi, z=0$, i.e.,

$$
\begin{equation*}
\rho=\left[(x+a)^{2}+y^{2}+z^{2}\right]^{1 / 2} \tag{8}
\end{equation*}
$$

Having obtained the temperature field, the induced displacements and stresses in the solid can be found. This will be done in the next section.

## Thermal Stresses

The thermal displacements and stresses are governed by equations (2) where the temperature is known from equations (4) and (5). Denoting the projections of the displacement vector in the directions of the cylindrical coordinates by ( $u_{r}, u_{\theta}, u_{z}$ ), the Papko-
vich-Neuber potential representation of equations (2) gives the displacement field

$$
\begin{gather*}
u_{r}(r, \theta, z)=4(1-\nu)\left(f_{1} \cos \theta+f_{2} \sin \theta\right)-\frac{\partial F}{\partial r}+\sin \theta \frac{\partial}{\partial r}(r \Omega)  \tag{9a}\\
u_{\theta}(r, \theta, z)=4(1-\nu)\left(f_{2} \cos \theta-f_{1} \sin \theta\right)-\frac{1}{r} \frac{\partial F}{\partial \theta}+\frac{\partial}{\partial \theta}(\sin \theta \Omega)  \tag{9b}\\
u_{z}(r, \theta, z)=4(1-\nu) f_{3}-\frac{\partial F}{\partial z}+r \sin \theta \frac{\partial \Omega}{\partial \dot{z}} \tag{9c}
\end{gather*}
$$

in which the following abbreviation has been introduced:

$$
\begin{equation*}
F=f_{0}+(r \cos \theta) f_{1}+(r \sin \theta) f_{2}+z f_{3} \tag{10}
\end{equation*}
$$

and $f_{n}(r, \theta, z), n=0,1,2,3$, are space harmonic functions. The potential $\Omega(r, \theta, z)$ can be determined from a knowledge of the temperature through the relation

$$
\begin{equation*}
\frac{\partial \Omega}{\partial y}=\frac{(1+\nu) \alpha}{2(1-\nu)} T(r, \theta, z) \tag{11}
\end{equation*}
$$

The corresponding stress field is readily obtained from equations (9) and (10) and the usual Duhamel-Neumann stress-displacement relations in linear thermoelasticity [8]. In particular, the stresses associated with the $\theta$-plane are found as

$$
\begin{align*}
\frac{\sigma_{\theta}}{2 \mu}= & -(1-2 \nu)\left(\cos \theta \frac{\partial f_{1}}{\partial r}+\sin \theta \frac{\partial f_{2}}{\partial r}\right) \\
& +2(1-\nu) \frac{1}{r}\left(\cos \theta \frac{\partial f_{2}}{\partial \theta}-\sin \theta \frac{\partial f_{1}}{\partial \theta}\right)+2 \nu \frac{\partial f_{3}}{\partial z}+\frac{\partial^{2} f_{0}}{\partial r^{2}}+\frac{\partial^{2} f_{0}}{\partial z^{2}} \\
& -\frac{1}{r}\left(\cos \theta \frac{\partial^{2} f_{1}}{\partial \theta^{2}}+\sin \theta \frac{\partial^{2} f_{2}}{\partial \partial^{2}}\right)+z\left(\frac{\partial^{2} f_{3}}{\partial r^{2}}+\frac{\partial^{2} f_{3}}{\partial z^{2}}\right) \\
& -\sin \theta\left(\frac{\partial \Omega}{\partial r}-\frac{1}{r} \frac{\partial^{2} \Omega}{\partial \theta^{2}}\right) \tag{12a}
\end{align*}
$$

In equations (12), $\mu$ designates the shear modulus of the solid and the function $G(r, \theta, z)$ is defined by means of the relation

$$
\begin{align*}
& G=(1-2 \nu)\left(f_{2} \cos \theta-f_{1} \sin \theta\right)-\frac{1}{r} \frac{\partial f_{0}}{\partial r}-\cos \theta \frac{\partial f_{1}}{\partial \theta} \\
&-\sin \theta \frac{\partial f_{2}}{\partial \theta}-\frac{z}{r} \frac{\partial f_{3}}{\partial \theta} \tag{13}
\end{align*}
$$

By virtue of the fact that the induced deformation is skew-symmetric with respect to the variable $\theta$, the quantities $u_{r}, u_{z}, \sigma_{\theta}$ are odd in $\theta$ while $u_{\theta,}, \tau_{\theta r}$, and $\tau_{\theta z}$ are even in the same variable. This circumstance suggests that the problem can be formulated for the upper half space $y \geqslant 0$ with appropriate boundary conditions prescribed in the regions $\theta=0$ and $\theta=\pi$. In view of these observations, the continuity of the solid outside the crack region implies

$$
\begin{align*}
& u_{r}(r, 0, z)=0  \tag{14a}\\
& u_{z}(r, 0, z)=0  \tag{14b}\\
& \sigma_{0}(r, 0, z)=0 \tag{14c}
\end{align*}
$$

On the other hand, as the crack surface is assumed to be free from mechanical loading, the following conditions must be satisfied in the region $\theta=\pi$

$$
\begin{align*}
\sigma_{\theta}(r, \pi, z) & =0  \tag{15a}\\
\tau_{\theta r}(r, \pi, z) & =0  \tag{15b}\\
\tau_{\theta z}(r, \pi, z) & =0 \tag{15c}
\end{align*}
$$

These conditions need to be accompanied by the regularity re-
quirements at infinity, namely, the vanishing of displacements and stresses in that vicinity. Furthermore, near the crack border $(r \rightarrow$ 0 ) the displacements must be finite and the stresses are expected to have the usual square root singularity $r^{-1 / 2}$.

Making use of equations ( $9 a$ ) , ( $9 c$ ), and (12a) in conjunction with (10), it is readily confirmed that the boundary conditions given in equations (14) are satisfied by

$$
\begin{align*}
& f_{0}(r, 0, z)=0  \tag{16a}\\
& f_{1}(r, 0, z)=0  \tag{16b}\\
& f_{3}(r, 0, z)=0  \tag{16c}\\
& \frac{\partial f_{2}}{\partial \theta}(r, 0, z)=0 \tag{16d}
\end{align*}
$$

With reference to the region $\theta=\pi$, equations (15) when inserted in the appropriate stress expressions yield the following relations:

$$
\begin{align*}
(1-2 \nu) \frac{\partial f_{1}}{\partial r}-\frac{2(1-\nu)}{r} \frac{\partial f_{2}}{\partial \theta}+ & 2 \nu \frac{\partial f_{3}}{\partial z}+\frac{\partial^{2} f_{0}}{\partial r^{2}}+\frac{\partial^{2} f_{0}}{\partial z^{2}} \\
& +\frac{1}{r} \frac{\partial^{2} f_{1}}{\partial \theta^{2}}+z\left(\frac{\partial^{2} f_{3}}{\partial r^{2}}+\frac{\partial^{2} f_{3}}{\partial z^{2}}\right)=0  \tag{17a}\\
& \frac{\partial G}{\partial r}-\frac{2(1-\nu) \partial f_{1}}{r}=\frac{\partial \Omega}{\partial r}  \tag{17b}\\
& \frac{\partial G}{\partial z}+ \tag{17c}
\end{align*}
$$

in which $G$ can be found from

$$
\begin{equation*}
G=-(1-2 \nu) f_{2}-\frac{1 \partial f_{0}}{r \partial r}+\frac{\partial f_{1}}{\partial \theta}-\frac{z}{r} \frac{\partial f_{3}}{\partial \theta}, \theta=\pi \tag{18}
\end{equation*}
$$

Further simplification may be achieved by setting

$$
\begin{equation*}
G=0, \theta=\pi \tag{19}
\end{equation*}
$$

and as a consequence of (19) it follows that

$$
\begin{equation*}
\frac{\partial G}{\partial r}=\frac{\partial G}{\partial z}=0, \theta=\pi \tag{20}
\end{equation*}
$$

By invoking the identities

$$
\left.\begin{array}{l}
\frac{\partial}{\partial x}=-\frac{\partial}{\partial r^{\prime}}, \frac{\partial}{\partial y}=-\frac{1}{r} \frac{\partial}{\partial \theta}  \tag{21}\\
\frac{\partial^{2}}{\partial y^{2}}=-\left(\frac{\partial^{2}}{\partial r^{2}}+\frac{\partial^{2}}{\partial z^{2}}\right)
\end{array}\right\} \theta=\pi
$$

in connection with equations (2), the relations (17) are transformed into

$$
\begin{align*}
2(1-\nu) \frac{\partial f_{2}}{\partial y}-(1-2 \nu) \frac{\partial f_{1}}{\partial x}+2 \nu \frac{\partial f_{3}}{\partial z}-\frac{\partial^{2} f_{0}}{\partial y^{2}}- & -\frac{\partial^{2} f_{1}}{\partial \theta \partial y} \\
& -z \frac{\partial^{2} f_{3}}{\partial y^{2}}=0, \theta=\pi  \tag{22a}\\
\frac{\partial f_{1}}{\partial y}= & -\frac{1}{2(1-\nu)} \frac{\partial \Omega}{\partial x}, \theta=\pi  \tag{22b}\\
\frac{\partial f_{3}}{\partial y}= & -\frac{1}{2(1-\nu)} \frac{\partial \Omega}{\partial z}, \theta=\pi \tag{22c}
\end{align*}
$$

In a similar manner, equations (18) and (19) render

$$
\begin{equation*}
(1-2 \nu) f_{2}-\frac{\partial f_{0}}{\partial y}-\frac{\partial f_{1}}{\partial \theta}-z \frac{\partial f_{3}}{\partial y}=0, \theta=\pi \tag{23}
\end{equation*}
$$

The mixed relations ( $16 b$ ), ( $16 c$ ), (22b), and (22c) provide the necessary information for the evaluation of $f_{1}$ and $f_{3}$. In these relations the derivatives of the thermoelastic potential are determined from a knowledge of the temperature in the solid already treated in the previous section. In particular, when a constant heat flux is present at the point ( $a, \pi, 0$ ) equations (7) and (11) yield

$$
\begin{equation*}
\frac{\partial \Omega}{\partial x}+i \frac{\partial \Omega}{\partial z}=\frac{(1+\nu) \alpha Q_{0}}{2(1-\nu) \pi^{2}(x+a-i z)}\left[\left(\frac{2 a}{x-a-i z}\right)^{1 / 2}\right. \tag{24}
\end{equation*}
$$

$$
\begin{equation*}
\left.\times \ln \left(\frac{\sqrt{r+x}+\sqrt{x-a-i z}}{\sqrt{r+a+i z}}\right)-\frac{y}{\rho} \tan ^{-1}\left(\frac{\sqrt{2 a(r-x)}}{\rho}\right)\right] \tag{24}
\end{equation*}
$$

(Cont.)
The character of equations (16b), (16c), (22b), (22c), and (24) suggests that the functions $f_{1}$ and $f_{3}$ can be expressed by ${ }^{2}$

$$
\begin{align*}
& f_{1}(r, \theta, z)=- \frac{1}{2(1-\nu)}\left[y \frac{\partial \Omega}{\partial x}-(x+a) \frac{\partial \Omega}{\partial y}\right] \\
&+C_{1} \frac{\sin (\theta / 2)}{\sqrt{r}} \operatorname{Re}[g(\zeta)]  \tag{25}\\
& f_{3}(r, \theta, z)=-\frac{1}{2(1-\nu)}\left[y \frac{\partial \Omega}{\partial z}-z \frac{\partial \Omega}{\partial y}\right] \tag{26}
\end{align*}
$$

where $C_{1}$ is a constant introduced for convenience and Re designates the real part of an analytic function of the variable $\zeta$ defined by

$$
\begin{equation*}
\zeta=r+a+i z, i=\sqrt{-1} \tag{27}
\end{equation*}
$$

In equation (25), the term associated with $C_{1}$ is a singular solution of Laplace equation which satisfies conditions (16b) and (22b) and conform to the usual regularity requirement at infinity. It has been introduced to insure finiteness of displacements at the crack bor$\operatorname{der}(r \rightarrow 0)$.

The next step in the analysis is to derive another set of relations which govern the remaining potentials $f_{0}$ and $f_{2}$. In order to achieve this, it is expedient to add the term $y\left(\partial f_{3} / \partial z\right)$ (which vanishes in the region $\theta=\pi$ ) to both sides of equations (23), i.e.,

$$
\begin{equation*}
(1-2 \nu) f_{2}-\frac{\partial f_{0}}{\partial y}-\frac{\partial f_{1}}{\partial \theta}+y \frac{\partial f_{3}}{\partial z}-z \frac{\partial f_{3}}{\partial y}=0, \theta=\pi \tag{28}
\end{equation*}
$$

Moreover, utilizing equations (16) and the identities

$$
\begin{equation*}
\frac{\partial}{\partial y}=\frac{1}{r} \frac{\partial}{\partial \theta^{\prime}}, \frac{\partial^{2}}{\partial y^{2}}=-\left(\frac{\partial^{2}}{\partial r^{2}}+\frac{\partial^{2}}{\partial z^{2}}\right), \theta=0 \tag{29}
\end{equation*}
$$

it is not difficult to verify that
$\frac{\partial}{\partial y}\left[(1-2 \nu) f_{2}-\frac{\partial f_{0}}{\partial y}-\frac{\partial f_{1}}{\partial \theta}+y \frac{\partial f_{3}}{\partial z}-z \frac{\partial f_{3}}{\partial y}\right]=0, \quad \theta=0$
A glance at the quantity inside the bracket in equation (30) reveals that it is harmonic. It follows that equations (28) and (30) suggest the following relation involving the singular solution introduced in (25)
$(1-2 \nu) f_{2}-\frac{\partial f_{0}}{\partial y}-\frac{\partial f_{1}}{\partial \theta}+y \frac{\partial f_{3}}{\partial z}-z \frac{\partial f_{3}}{\partial y}=C_{2} \frac{\cos (\theta / 2)}{\sqrt{r}} \operatorname{Re}[g(\zeta)]$
A second relation between $f_{0}$ and $f_{2}$ may be derived from equations (16), (22a), and (29). The mathematical manipulation necessary for obtaining the second relation is identical to that used by Kassir and Sih [6] in analyzing the problem of the considered crack under concentrated shear load parallel to the crack edge. Without going into the details, it can be shown that

$$
\begin{align*}
2(1-\nu) f_{2}-\frac{\partial f_{0}}{\partial y}=\frac{\partial f_{1}}{\partial \theta}+ & z \frac{\partial f_{3}}{\partial y}-y \frac{\partial f_{3}}{\partial z} \\
& +(1-2 \nu) \int_{\infty}^{y}\left(\frac{\partial f_{1}}{\partial x}+\frac{\partial f_{3}}{\partial z}\right) d y, y \geqslant 0 \tag{32}
\end{align*}
$$

exists throughout the entire region $y \geqslant 0$ (see equation (22) of [6]). Solving equations (31) and (32) simultaneously and utilizing equations (25) and (26) result in

$$
f_{2}=\left[(1-2 v) C_{1}-C_{2}\right] \frac{\cos (\theta / 2)}{\sqrt{r}} \operatorname{Re}[g(\zeta)]
$$

[^1]\[

$$
\begin{aligned}
& -(1-2 \nu) C_{1} \int_{\infty}^{y} \frac{\sin (\theta / 2)}{\sqrt{r}} \operatorname{Re}\left[g^{\prime}(\zeta)\right] d y \\
& \quad+\frac{(1-2 \nu)}{2(1-\nu)}\left[\Omega+(x+a) \frac{\partial \Omega}{\partial x}+y \frac{\partial \Omega}{\partial y}+z \frac{\partial \Omega}{\partial z}\right]
\end{aligned}
$$
\]

(33a) (Cont.)
and

$$
\begin{align*}
& \frac{\partial f_{0}}{\partial y}=\left[\left(\frac{1}{2}-4 \nu\right.\right.\left.\left.+4 \nu^{2}\right) C_{1}-2(1-\nu) C_{2}\right] \frac{\cos (\theta / 2)}{\sqrt{r}} \operatorname{Re}[g(\zeta)] \\
&-(1-2 \nu)^{2} C_{1} \int_{\infty}^{y} \frac{\sin (\theta / 2)}{\sqrt{r}} \operatorname{Re}\left[g^{\prime}(\zeta)\right] d y+\frac{1}{2(1-\nu)} \\
& \times\left\{[ ( 1 - 2 \nu ) ^ { 2 } + y \frac { \partial } { \partial y } ] \left[\Omega+(x+a) \frac{\partial \Omega}{\partial x}\right.\right. \\
&\left.+y \frac{\partial \Omega}{\partial y}+z \frac{\partial \Omega}{\partial z}\right]+x\left[\frac{\partial \Omega}{\partial x}+y \frac{\partial^{2} \Omega}{\partial x \partial y}\right. \\
&\left.\left.-(x+a) \frac{\partial^{2} \Omega}{\partial y^{2}}\right]+z\left[\frac{\partial \Omega}{\partial z}+y \frac{\partial^{2} \Omega}{\partial y \partial z}-z \frac{\partial^{2} \Omega}{\partial y^{2}}\right]\right\} \tag{33b}
\end{align*}
$$

where $g^{\prime}(\zeta)=(d / d \zeta) g(\zeta)$ and $\Omega$ is given in (24).
The remaining conditions to be satisfied are equations ( $16 a$ ) and the requirement that the displacements are finite at $r=0$. These conditions are utilized to determine the function $g(\zeta)$ as well as the constants $C_{1}$ and $C_{2}$ in equations (25) and (33). However, instead of applying the condition in equation (16a), it is easier to subject (33b) to the equivalent condition

$$
\begin{equation*}
\frac{\partial f_{0}}{\partial z}=0, \theta=0 \tag{34}
\end{equation*}
$$

Differentiating ( $33 b$ ) with respect to $z$ and then integrating in the variable $y$ between the limits $y$ and $\infty$, making use of equation (24) and the results

$$
\begin{gather*}
\int_{\infty}^{y} \frac{\cos (\theta / 2)}{\sqrt{r}} g^{\prime}(\zeta) d y=\frac{1}{\sqrt{2}} \int_{\infty}^{\zeta} \frac{g^{\prime}(t) d t}{\sqrt{\zeta-\zeta_{0}}}  \tag{35a}\\
\int_{\infty}^{y} d y \int_{\infty}^{y} \frac{\sin (\theta / 2)}{\sqrt{r}} g^{\prime \prime}(\zeta) d y=\frac{1}{2 \sqrt{2}}\left\{\int_{\infty}^{\zeta} \frac{g^{\prime}(t) d t}{\sqrt{t-\zeta_{0}}}\right. \\
\left.-2 y \int_{\infty}^{\zeta} g^{\prime}(t) \frac{d}{d t}\left(t-\zeta_{0}+2 x\right)^{-1 / 2} d t\right\} \tag{35b}
\end{gather*}
$$

$\Omega+(x+a) \frac{\partial \Omega}{\partial x}+y \frac{\partial \Omega}{\partial y}+z \frac{\partial \Omega}{\partial z}=-\frac{(1+\nu) \alpha Q_{0} \sqrt{a}}{2 \sqrt{2}(1-\nu) \pi^{2}}$

$$
\begin{equation*}
\times \operatorname{Re}\left\{\frac{1}{\sqrt{2 x-\zeta_{0}}} \ln \left|\frac{\sqrt{r+x}-\sqrt{2 x-\zeta_{0}}}{\sqrt{r+x}+\sqrt{2 x-\zeta_{0}}}\right|\right\} \tag{35c}
\end{equation*}
$$

it is found that condition (34) results in

$$
\begin{equation*}
\left(\nu C_{1}+C_{2}\right) \int_{\zeta_{0}}^{\infty} \frac{g^{\prime}(t) d t}{\sqrt{t-\zeta_{0}}}=-\frac{\nu(1+\nu) \alpha Q_{0} \sqrt{a}}{4(1-\nu) \pi \sqrt{\zeta_{0}}} \tag{36}
\end{equation*}
$$

where the variable $\zeta_{0}$ is defined as

$$
\begin{equation*}
\zeta_{0}=[\zeta]_{\theta=0}=x+\alpha+i z \tag{37}
\end{equation*}
$$

Equation (36) is a standard integral equation of Abel type whose solution is [9]

$$
\begin{equation*}
g(\zeta)=\ln \zeta \tag{38}
\end{equation*}
$$

provided that

$$
\begin{equation*}
\nu C_{1}+C_{2}=-\frac{\nu(1+\nu) \alpha Q_{0} \sqrt{a}}{4(1-\nu)^{2} \pi^{2}} \tag{39}
\end{equation*}
$$

The final step in the analysis is to relate the constants $C_{1}$ and $C_{2}$. This may be done by imposing the regularity condition on the displacements at the crack edge. Inserting the appropriate values of the potentials in equations (9), making use of (24) and (38), expanding asymtotically for small $r$ and retaining the lowest order terms, it is found that the displacements near the crack edge assume the form


Fig. 2 Variations of $k_{2}$ and $k_{3}$ along crack border

$$
\begin{gather*}
u_{r}=\frac{\sin \frac{\theta}{2}}{4 \sqrt{r}} \ln \left(a^{2}+z^{2}\right)\left[2(1-\nu) C_{1}-C_{2}\right][3-4 \nu \\
 \tag{40a}\\
\quad+(7-8 \nu) \cos \theta]+0\left(r^{0}\right) \\
u_{i \prime}=-\frac{\cos \frac{\theta}{4 \sqrt{r}}}{4 \sqrt{2}} \ln \left(a^{2}+z^{2}\right)\left[2(1-\nu) C_{1}-C_{2}\right]  \tag{40b}\\
\times[3-4 \nu-(5-8 \nu) \cos \theta]+0\left(r^{0}\right)  \tag{40c}\\
u_{z}=0\left(r^{0}\right)
\end{gather*}
$$

Now, the finite displacements requirement at the crack edge gives

$$
\begin{equation*}
2(1-\nu) C_{1}-C_{2}=0 \tag{41}
\end{equation*}
$$

and it follows from equation (39) that

$$
\begin{align*}
C_{1} & =-\frac{\nu(1+\nu) \alpha Q_{0} \sqrt{a}}{4(2-\nu)(1-\nu)^{2} \pi^{2}}  \tag{42a}\\
C_{2} & =-\frac{\nu(1+\nu) \alpha Q_{0} \sqrt{\alpha}}{2(2-\nu)(1-\nu) \pi^{2}} \tag{42b}
\end{align*}
$$

This, basically, completes the solution of the problem. The induced displacements and stresses in the solid may be readily computed from equations (9)-(12) when use is made of equations (24)-(26), (33), (38), and (42). These quantities serve as the Green's functions for any distribution of heat flux applied to arbitrary regions of the crack surface.

## Stress-Intensity Factors

The shear stresses across the surface $\theta=0$ are computed from equations (12b) and (12c) as

$$
\begin{align*}
\tau_{0 r}(r, 0, z) & =\frac{E \alpha Q_{0} \sqrt{a}(x+a)}{(2-\nu) \pi^{2} \sqrt{x}\left[(x+a)^{2}+z^{2}\right]}+\ldots .  \tag{43a}\\
\tau_{\theta z}(r, 0, z) & =\frac{E \alpha Q_{0} \sqrt{a} z}{(2-\nu) \pi^{2} \sqrt{x}\left[(x+a)^{2}+z^{2}\right]}+\ldots \tag{43b}
\end{align*}
$$

where $E$ is Young's modulus of the material, $E=2 \mu(1+\nu)$, and the nonsingular terms have been neglected. Equations (43) may be expressed in the standard form

$$
\begin{align*}
& \tau_{\theta r}(r, 0, z)=\frac{k_{2}}{\sqrt{2 x}}+0\left(x^{0}\right)  \tag{44a}\\
& \tau_{\theta z}(r, 0, z)=\frac{k_{3}}{\sqrt{2 x}}+0\left(x^{0}\right) \tag{44b}
\end{align*}
$$

in which the stress-intensity factors, $k_{2}$ and $k_{3}$, associated with the edge-sliding and tearing modes of crack extension, respectively, are given in terms of the nondimensional parameter $z_{0}=z / a$ by way of the relations

$$
\begin{align*}
& k_{2}=\frac{\sqrt{2} E \alpha Q_{0}}{(2-\nu) \pi^{2} \sqrt{a}} \frac{1}{1+z_{0}^{2}}  \tag{45a}\\
& k_{3}=\frac{\sqrt{2} E \alpha Q_{0}}{(2-\nu) \pi^{2} \sqrt{a} 1+z_{0}^{2}} \tag{45b}
\end{align*}
$$

Fig. 2 shows the variation of equations (45) with $z_{0}$.

The formulas in equations (45) can be used to generate results for any distribution of heat flux applied to arbitrary regions of the crack surface. As an example, suppose that the heat flux is applied to a rectangular region, $-a \leq x \leq 0, y=0,|z| \leq b$, of the crack surface, Fig. 1, then double integrations performed on equations (45) yield the stress-intensity factors
in which the functions $H_{1}$ and $H_{2}$ are given by

$$
\begin{align*}
& H_{1}(s)=s^{3 / 2} \tan ^{-1} \frac{1}{s}+2 s^{1 / 2}-\frac{1}{\sqrt{2}}[ \tan ^{-1}\left(\frac{\sqrt{2 s}}{1-s}\right) \\
&\left.+\ln \left(\frac{1+\sqrt{2 s}+s}{\sqrt{1+s^{2}}}\right)\right]  \tag{47a}\\
& H_{2}(s)=2 s^{3 / 2}-\frac{3}{\sqrt{2}} \tan ^{-1}\left(\frac{\sqrt{2 s}}{1-s}\right)
\end{aligned} \quad \begin{aligned}
& +\frac{3}{\sqrt{2}} \ln \left(\frac{1+\sqrt{2 s}+s}{\sqrt{1+s^{2}}}\right)
\end{align*}
$$

and the signum function, sgn $(z)$, is $+1,0$, or -1 depending on whether the argument, $z$, is positive, zero, or negative, respectively. Other solutions can be generated in a similar manner.

## Conclusion

The linear thermoelastic problem of a uniform heat flow disturbed by an insulated semi-infinite plane crack embedded in a three-dimensional elastic solid has been formulated and solved. The method of analysis involves the application of the technique of Fourier-Kontorovich-Lebedev Integral transforms and certain singular solutions of Laplace equation. The Green's functions for the distribution of the temperature field as well as the induced displacements and stresses in the solid are derived mainly in closed form. These results are useful in examining theories of brittle fracture for crack propagation caused by passage of steady-state heat in solids.

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[^0]:    ${ }^{1}$ Numbers in brackets designate References at end of paper.
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[^1]:    ${ }^{2}$ The results in equations (25) and (26) can also be reached by applications of Fourier and Kontorovich-Lebedev integral transforms to equations (16b), (16c), (22b), and (22c).

