

1 Numerical Modeling of the Seismic Out-of-Plane Response of a Plain and 2 TRM-Strengthened Rammed Earth Subassembly

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5
6 **Abstract:** The importance of raw earth is highlighted by the millions of persons living in earthen buildings
7 around the World and by numerous historical monuments made of this material. Its widely availability led to the
8 development of a variety of building techniques, including rammed earth, which is the main focus of this study.
9 Similarly to unreinforced masonry structures, rammed earth buildings acceptably withstand gravity loads, but
10 are significantly vulnerable to earthquakes. In this regard, great attention has been put on the proposal of
11 efficient, compatible, affordable and reversible strengthening solutions. However, very limited studies address
12 either the experimental testing or modeling of the seismic response of such buildings. The current study
13 investigates the seismic out-of-plane performance of a plain and subsequently strengthened rammed earth
14 subassembly (U-shape) using an advanced finite element modeling approach calibrated based on previously
15 conducted small-scale experiments. Here, failure mechanisms, corresponding capacity and efficiency of the
16 adopted strengthening solution (low-cost textile-reinforced mortar) are evaluated by means of pushover
17 analyses. Then, the reliability of the pushover analyses is assessed by comparing its outcomes with that of the
18 incremental dynamic analyses. In general, the failure was found to be governed by overturning of the web wall
19 due to its detachment from the wing walls, while the strengthening was found to increase the capacity and delay
20 the damage development.

21 **Keywords:** Rammed earth; Nonlinear FE modeling; Out-of-plane behavior; TRM strengthening; Pushover;
22 Dynamic analysis.

23 1. Introduction

24 Earth, as the most worldwide available material, was one of the first building materials used for manmade
25 constructions. In this regard, many traditional building techniques have been developed, among which the most
26 known are adobe masonry and rammed earth. As a consequence, many historical monuments made of raw earth

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27 can be found spread worldwide. Furthermore, the low associated building costs led this material to be an
28 appropriate choice for low income societies, as well as for hardly accessible regions and isolated rural areas;
29 which caused that approximately 30-40% of the world population to live or work in earthen constructions [1],
30 although this figure may be somehow exaggerated nowadays. Furthermore, the green nature of raw earth leads it
31 to be considered as a possible choice for future sustainable constructions. Hence, it became necessary to
32 understand the behavior of earthen buildings, not only for modern design objectives, but also for the repair and
33 strengthening of existing built heritage. To achieve this goal, it is essential to recognize their weaknesses under
34 different loading conditions and circumstances. In this regard, it was previously observed that several factors
35 such as rainwater, soluble salts, and temperature oscillations can lead to damage occurrence [2]. From a
36 structural point of view, earthen constructions show an acceptable behavior under gravity loads, but they are
37 strongly endangered with respect to lateral loads, like other types of unreinforced masonry structures.
38 Furthermore, many earthen constructions are located in regions with medium or high seismic hazard. In this
39 regard, many people inhabiting earth constructions have been severely affected in recent earthquakes, as for
40 instance in Erzinkan at Turkey (1992), Bam at Iran (2003), Pisco at Peru (2007) and Concepción at Chile
41 (2010).

42 The current study is focused on rammed earth construction. This technique is well-known in all continents
43 (chineh in Iran, taipa in Portugal, tapial in Spain, pisé in France, terra battuta in Italy, stampflehm in Germany,
44 hangtu in China, and pakhsa in Uzbekistan). Earth with an adequate moisture content is placed between two
45 parallel panels (formworks which are interconnected by spacers) and is compacted (either manually or using
46 pneumatic rammers). Its main ingredients are gravel, sand, silt, clay, water, and in some cases, additives. With
47 respect to additives, rammed earth is categorized into two general classes; i.e. unstabilized rammed earth (i.e.
48 without any artificial additives) and stabilized rammed earth, which contains binders such as cement or lime.
49 Typically, the panels dislocate as layers (with an approximate thickness of 15cm) are built by initially running
50 them horizontally along the perimeter of the construction and then moving vertically along the height to the next
51 level [3].

52 Generally, rammed earth shows a weak response under compression forces and exhibits a very low tensile
53 strength. Previous studies have reported tensile strength values for rammed earth constructions in a range of
54 0.10 to 0.15 MPa, which is very low in comparison with other conventional materials [4-5]. Several
55 experimental studies (mostly uniaxial or diagonal compression tests on wallets) have been conducted to
56 characterize its properties. However, the definition of its mechanical behavior is still a fundamental challenging

57 task [2-9], as it depends on several parameters, such as particle size distribution, moisture content, compaction
 58 (rate and type), void ratio, cohesive strength of the particles, fiber content and quantity and type of additives. In
 59 other words, the mechanical performance of rammed earth is significantly influenced by hygrothermal
 60 conditions. Thus, a large scatter is observed among values reported in the literature, as exemplified in Table 1.
 61 As it is evident, the values reported for Young's modulus and shear modulus significantly vary in comparison to
 62 other parameters. Moreover, it was noticed that in spite of the layered construction procedure, the mechanical
 63 properties (especially the compressive strength and Young's modulus) do not meaningfully vary with respect to
 64 the direction of layers, namely perpendicular or parallel ones [8].

65 **Table 1.** Reported mechanical characteristics of rammed earth in different studies

Reference	ρ (kg/m ³)	E (N/mm ²)	f_c (MPa)	f_t (MPa)	f_v (MPa)	G (N/mm ²)	ν (-)
Lilley and Robinson (1995) [7]	1870-2170	-	1.8-2.0	-	-	-	-
Yamin <i>et al.</i> (2004) [5]	1920	784.8	3.24	0.15	0.36	-	-
Bui and Morel (2009) [2]	1800.0	90-105	1.0	-	-	-	-
Maniatidis <i>et al.</i> (2007) [8]	1850.0	205	3.88	-	-	-	-
Miccoli <i>et al.</i> (2014) [9]	-	4143	3.73	-	0.71	2326	0.27

66 Where ρ is bulk density, E is Young's modulus, f_c is compressive strength, f_t is tensile strength, f_v is shear strength, G is shear
 67 modulus and ν is Poisson's ratio.
 68

69
 70 As mentioned above, rammed earth buildings are very vulnerable to earthquake excitations. Widely observed
 71 failure modes in previous earthquakes include brittle failures such as falling over due to out-of-plane actions,
 72 cracks at edges and also at loading points where the load of the roof is transferred to the wall, losing
 73 connectivity due to weak connections and propagation of cracks due to close distance between openings and
 74 corners [10-11]. Several intervention solutions are proposed in the literature to mitigate vulnerability conditions
 75 of earthen structures; namely the repair of cracks with injection of compatible grouts (from physical, mechanical
 76 and chemical points of view) [12-13], strengthening with boundary wooden elements [5], strengthening with
 77 ring beams [14], and strengthening with textile reinforced mortars (TRM) [12].

78 TRM is also known as FRCM (Fiber Reinforced Cementitious Matrix) in the literature and its use has been
 79 lately gaining increasing attention for the strengthening of masonry structures [15]. This interest resulted from
 80 disadvantages observed in the strengthening of masonry resorting to FRP-based materials, namely inadequate
 81 compatibility with the substrate, poor fire/high-temperature resistance [16-17], low reversibility, brittle failure

82 and lack of vapor permeability, which are seen as serious drawbacks with respect to their application on
83 historical constructions [18].

84 Experimental studies have been recently conducted to characterize not only the material properties of TRM
85 systems, but also to evaluate the improved response of strengthened structural components. Uniaxial tensile tests
86 and single/double lap shear tests have been performed to obtain their stress-strain and textile, matrix and
87 substrate interaction behaviors, respectively [15]. Regarding the stress-strain curves, three stages are typically
88 distinguished, i.e. un-cracked, crack development and cracked. During the first stage, the behavior is linear,
89 while the crack initiation onsets the second stage, where the stiffness is reduced. In the third stage, the cracks are
90 stabilized and the load-bearing capacity increases up to failure. In the first two stages, the mechanical
91 characteristics of the mortar, textile, and their interface contribute for the behavior, while the behavior in the last
92 stage is controlled by the textile, despite the mortar matrix is still able to provide transversal load redistribution
93 [19-20]. Moreover, it is worthwhile to note that different failure modes such as shear failure of substrate,
94 interface failure between substrate and textile (so-called debonding) or interface failure between mortar and
95 reinforcement, slippage of reinforcement within the mortar and failure of reinforcement have been observed
96 depending on bond characteristics of the strengthening/substrate or mortar/textile [20-21].

97 In addition to the aforementioned small-scale tests, several experimental studies were conducted to assess the
98 efficiency of TRM solutions on the in-plane and out-of-plane response of masonry walls [20,22]. Though, they
99 were mainly applied to masonry, meaning that the efficiency of TRM strengthening on earthen constructions is
100 basically unknown, mainly rammed earth one. Despite that, it is expected that the application of this solution
101 may result in effective increase of the loading capacity or/and ductility. Additionally, the strengthening is
102 expected to promote a further redistribution of stresses that may prevent (or delay) the integrity loss of structural
103 components [20,22].

104 In addition to experimental investigations, numerical modeling is also a powerful tool to better understand the
105 behavior of earthen constructions. Generally, three different types of modeling approaches have been employed
106 to investigate the response of rammed earth walls, i.e. simplified modeling (using limit analysis) [23], finite
107 element modeling (FEM) and discrete element modeling (DEM). Due to the predominant nonlinear behavior of
108 rammed earth, predicting its response by means of analytical or linear methods would be a cumbersome task,
109 meaning that more advanced approaches should be adopted instead. In FEM modeling, two approaches are
110 typically used, namely macro- and micro-modeling. The macro-modeling approach does not consider layered

111 and anisotropic natures of rammed earth, as the material is assumed to be continuous and isotropic. In micro-
112 modeling, the rammed earth layers are discretized and the interaction occurring between layers is taken into
113 account. Due to the greater detailing of micro-modeling, most of the available studies have employed the macro-
114 modeling approach. The major reasons supporting the decision of ignoring the micro-modeling approach are the
115 lack of reliable data to define the behavior of the interfaces, the higher computational efforts and the fact that
116 macro-modeling results in a global comparable accuracy with respect to micro-modeling [24]. In turn, DEM is
117 less popular than FEM, but some researchers employed it to take into account the influence of layers in the
118 response. Though, it was concluded that the results obtained by models with or without interfaces are similar,
119 even when very low values of interface parameters are considered [25].

120 It is worthwhile to note that previous studies obtained relatively accurate outcomes under in-plane loading in
121 comparison to the experimental results. On the other hand, predicting the out-of-plane response may lead to
122 significantly diverse outcomes and still is a challenging task [26].

123 The current study investigates the out-of-plane seismic response of a rammed earth component and assesses the
124 strengthening efficiency of a selected TRM solution. In this regard, the FEM is adopted to simulate the
125 structural behavior in both cases. The numerical models were defined with basis on experimental results
126 available in the literature and were subjected to mass proportional pushover analyses to assess their loading
127 capacity and corresponding failure modes. Additionally, incremental dynamic analyses were performed to
128 evaluate the reliability of pushover analyses for rammed earth structural components.

129 This study was carried out within the framework of an ongoing research project on the seismic behavior of
130 rammed earth, where experimental models, similar to those simulated here numerically, will be tested on the
131 shaking table. Given the novelty of the shaking table testing of plain and strengthened rammed earth
132 components, this numerical work has the twofold purpose of gaining insight into the out-of-plane seismic
133 behavior of plain and TRM-strengthened rammed earth and guiding the design of the experimental campaign.

134 **2. Rammed Earth Structural Subassembly**

135 Most of the previously conducted studies investigating the behavior of rammed earth constructions are limited to
136 small components (wallets). It is evident that in spite of providing valuable information at the material level,
137 they do not provide a general understanding of possible building failure mechanisms. Furthermore, the referred
138 specimens were typically subjected to static loading conditions and the influence of other key parameters such

139 as connections or perpendicular walls were neglected. In this regard, testing structural components on shaking
140 table is a more appropriate approach to clarify probable damage under earthquake excitations and also to assess
141 the applicability of strengthening solutions in improving the seismic behavior. Due to the costly and time-
142 consuming process of testing components on shaking table, an efficient and accurate experimental test design
143 requires the analysis of possible structural responses by means of advanced numerical analyses.

144 The geometry of the subassembly should be representative of real constructions as much as possible, satisfying
145 also geometry and weight limitations of the shaking table, and must be designed in such a way as to fail
146 according to the expected behavior (i.e. out-of-plane as aimed in the current study).

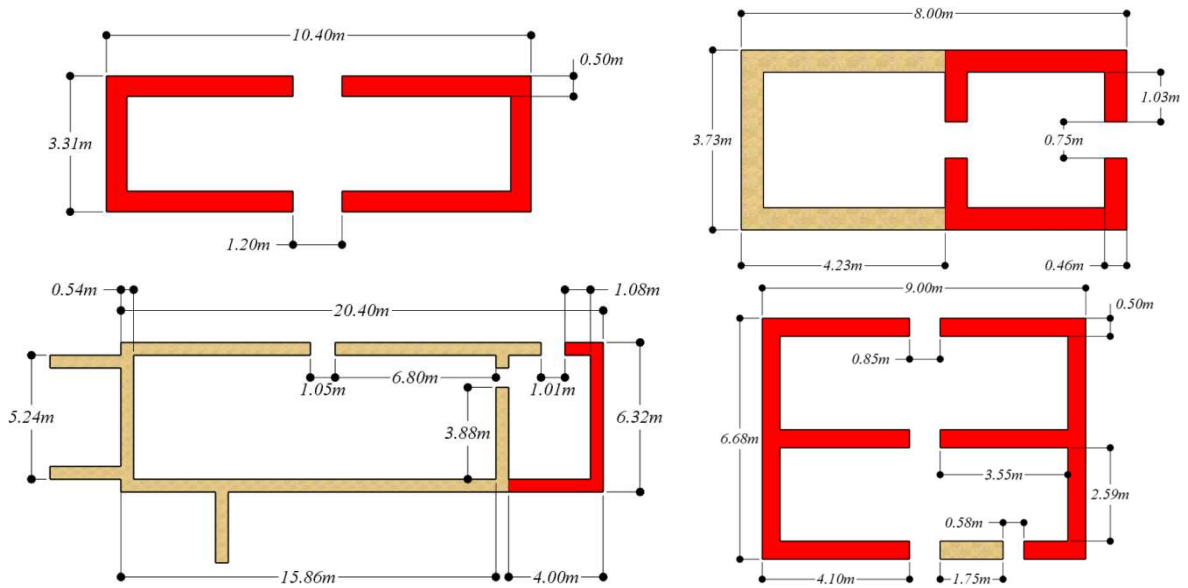
147 Alentejo region in southern Portugal has an extensive number of monuments and buildings built in rammed
148 earth. Some of these typical vernacular buildings in this region are shown in Fig. 1. Thus, a survey of eleven
149 representative rammed earth buildings located in this region was taken into account to define the geometry of
150 the model. They were all single story buildings constructed before the 1950s. It was observed that in all cases
151 the thickness of walls is of about 0.5m [27]; hence the same dimension is adopted in the current study. The
152 height, length and longitudinal to transversal length ratio found in the sample are around $2.2 \pm 0.3\text{m}$, $3.7 \pm 1.5\text{m}$
153 and 2.2 ± 1.0 , respectively.



154 **Fig. 1.** Typical vernacular rammed earth buildings in southern Portugal

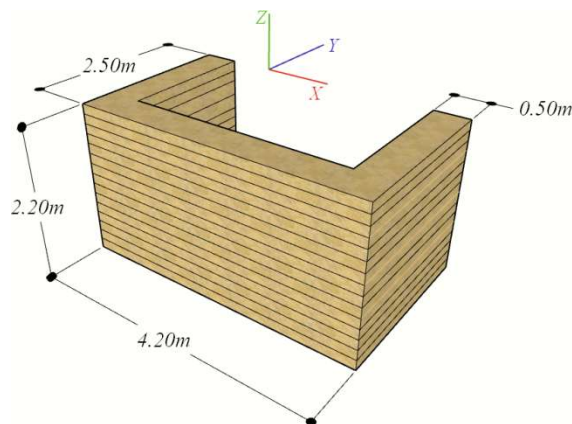
155 Regarding the roof typology of rammed earth buildings, they were constructed by lightweight timber shed or
156 gable roofs, which normally support on the walls. Nevertheless, the roof system does not impose a significant
157 load to the walls in this constructional system. Also, the low in-plane stiffness of roofs does not allow these
158 structures to be considered as box-behavior.

159 On the other hand, some samples of typical plan view of the surveyed vernacular rammed earth buildings are
160 shown in Fig. 2. As it can be seen, the walls are generally supported by perpendicular walls (which are
161 highlighted in red). Thus, it seems that a U-shape component could be a representative geometry for typical
162 rammed earth walls.



163 **Fig. 2.** Plan samples of surveyed typical vernacular rammed earth buildings in Alentejo region, southern
 164 Portugal [27]

165 In addition to the desirable representative conditions, the outcomes of the current study are aimed to be used in
 166 designing a full-scale shaking table test in near future. Therefore, it was essential to consider a model with
 167 geometry within practical dimension ranges (obtained from the surveyed buildings), which is supported by
 168 transversal walls and satisfies maximum size and weight limitations of the shaking table. The shaking table to
 169 test the rammed earth model is the one from the Portuguese national civil engineering laboratory (LNEC), where
 170 the maximum weight of models is limited to 21 tons. Taking into account both representativeness and technical
 171 conditions, the geometry of the U-shape subassembly presented in Fig.3 has been fixed with a constant wall
 172 thickness of 0.5 m. It is noted that the expected weight of the model is 18 tons (for a rammed earth density of
 173 about 2000 kg/m^3 [28]).



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 175 **Fig.3.** Geometry of the out-of-plane model

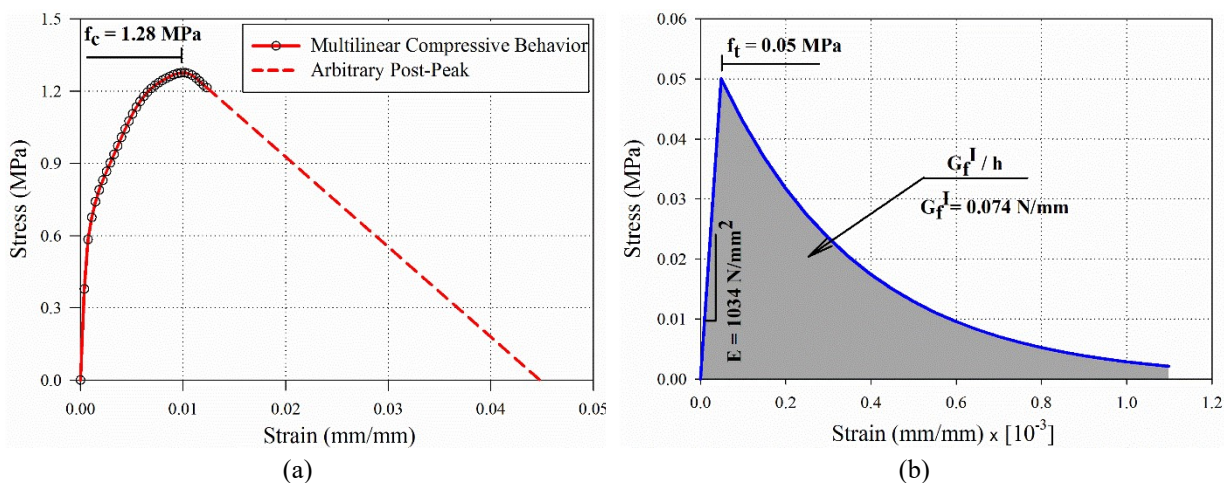
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177 **3. Mechanical Characteristics and Modeling**

178 FEM models were prepared and computed using the software DIANA 10.2 [29]. To perform the nonlinear finite
 179 element modeling of rammed earth subassembly, the definition of proper material properties for rammed earth,
 180 strengthening (including textile and mortar) and their interfaces (between substrate/strengthening and
 181 mortar/mesh grid) was required. The following sections present the adopted constitutive laws and employed
 182 modeling approach.

183 **3.1 Rammed earth material**

184 The material behavior of the rammed earth was simulated by means of the total strain rotating crack model
 185 (TSRCM) implemented in DIANA 10.2 [29]. Furthermore, the marked nonlinear behavior of rammed earth in
 186 compression, highlighted in the literature, led authors to adopt a multi-linear relationship, as recommended in
 187 [14,24]. The adopted relationship is depicted in Fig. 4a and was defined by averaging the experimental stress-
 188 strain curves obtained from compression tests on cylindrical rammed earth specimens reported in [30]. It is
 189 worthwhile to note that due to lack of reliable experimental data regarding the post-peak branch, the expected
 190 behavior (shown by dashed line) was estimated by continuing the curve with the same slope of the experimental
 191 data. The behavior in tension was assumed to follow an exponential relationship and the respective parameters
 192 are calibrated values from a previous numerical study [28] on the simulation of the shear behavior of rammed
 193 earth wallets tested under diagonal compression (see also [30]). The adopted relationship in tension and
 194 respective parameters are depicted in Fig. 4b.



195 **Fig. 4.** Adopted stress-strain relationships for rammed earth: (a) multi-linear relationship in compression (b)
 196 exponential relationship in tension (E is Young's modulus, f_t is the tensile strength, G_f^I is the mode-I tensile
 197 fracture energy and f_c is the compressive strength)

198 In order to make the numerical outcomes independent from the size of the element, the crack bandwidth (h) was
199 assumed to be square root of the element area (A). Note that the aforementioned experimental data resulted from
200 specimens manufactured with soil collected from the same region where the geometrical survey was conducted
201 (Alentejo region, southern Portugal).

202 **3.2 TRM composite material**

203 The selection of the TRM strengthening solution to be applied to the plain rammed earth model resulted from a
204 recent research work proposing and characterizing different low-cost textile reinforced mortars (LC-TRM)
205 solutions [31-32]. It should be noted that the fundamental concept of the proposed strengthening solution
206 consists in using compatible (from the physical, chemical and mechanical points of view), affordable and readily
207 available materials in order to generalize its use. The aforementioned studies evaluated several low-cost
208 reinforcing meshes available locally, among which a glass fiber mesh (denoted hereafter as G1) and a nylon
209 fiber mesh (denoted hereafter as G2) were selected to integrate the strengthening solution. With respect to the
210 other meshes evaluated, G1 presents higher strength, low deformation capacity and linear behavior up to peak
211 load followed by a fragile post-peak; whereas G2 has much less stiffness and strength, with a clear hardening
212 region, again ending at a brittle failure [32]. The proposed LC-TRM solution consists additionally of an earth
213 based mortar prepared with the same soil used in the construction of the rammed earth specimens tested in [30].

214 Regarding the modeling of the material behavior of the LC-TRM strengthening, a similar approach to that of
215 rammed earth was assumed, namely using the TSRCM. This material model requires knowing the stress-strain
216 behavior of the composite material in tension and compression. The behavior in tension was assumed with basis
217 on the composite tensile behavior (considering both mortar and mesh together) obtained from direct tensile tests
218 [31], as depicted in Fig.5 (shown as solid lines). The aforementioned tests were conducted on coupon specimens
219 with dimensions of $100 \times 400 \text{ mm}^2$, based on the procedure of ASTM D6637. The specimens were composed of
220 two low cost meshes available locally and an earth-based mortar. The tests were conducted by applying a tensile
221 load under monotonic displacement control and by measuring the axial deformations [31]. It should be noted
222 that within this modeling approach a perfect bond hypothesis between the mesh and mortar is assumed. The
223 numerical behavior in tension consisted of a typical tri-linear relationship obtained by averaging of the
224 experimental data (see dashed lines in Fig.5).

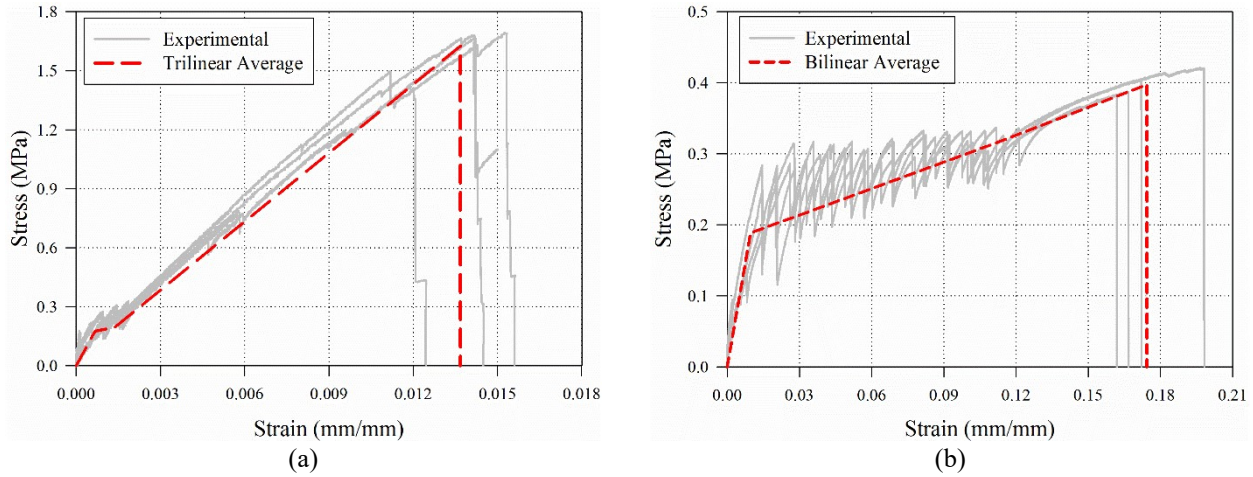


Fig.5. Tri-linear numerical curves based on the experimental uniaxial tensile stress-strain curves of TRM composite specimens: (a) G1 mesh (b) G2 mesh

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227 The behavior of the LC-TRM in compression is mainly governed by the behavior of the mortar, meaning that
228 the contribution of the mesh can be disregarded. Thus, the experimental stress-strain curves of mortar cylinders
229 tested under uniaxial compression were used to define the numerical behavior in compression (see [31]). These
230 curves present also an expressive nonlinear behavior, which led the multi-linear relationship to be assumed for
231 this study by averaging experimental data (see Fig.6a). Here, an estimated post-peak descending branch was
232 also proposed (shown as dashed line) to take into account the stress degradation of the TRM composite in
233 compression. Finally, the complete composite stress-strain behaviors are presented in Fig.6b and c. Considering
234 all, the adopted material properties in the current article are reported in Table 2.

235

Table 2. Adopted material properties

Material	E (N/mm ²)	ν (-)	ρ (kg/m ³)	f_c (MPa)	f_t (MPa)	G_f^I (N/mm)	ϵ_{ut} (-)	Comment
Rammed Earth	1034	0.27	2000	1.28	0.05	0.074	-	Multilinear compressive and exponential tensile behavior
LC-TRM (G1)	3431	0.27	1810	1.30	1.62	-	0.0137	Multilinear compressive and trilinear tensile behavior
LC-TRM (G2)	3431	0.27	1810	1.30	0.40	-	0.1744	Multilinear compressive and bilinear tensile behavior

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237
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Where E is Young's modulus, ν is Poisson's ratio, ρ is bulk density, f_c is compressive strength, f_t is tensile strength, G_f^I is mode-I tensile fracture energy and ϵ_{ut} is ultimate tensile strain.

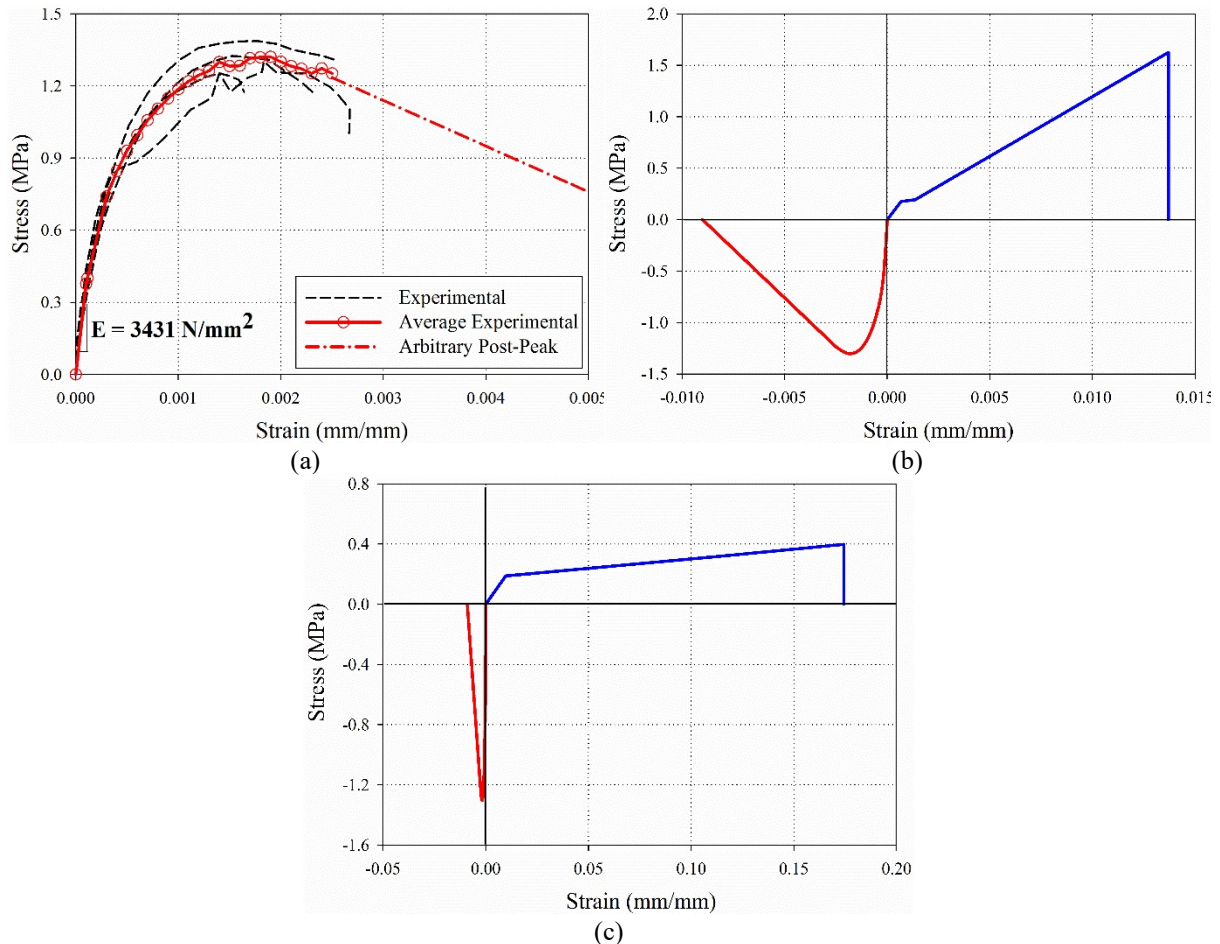


Fig.6. Stress-strain behavior of the LC-TRM strengthening: (a) behavior of mortar in compression; (b) full behavior with G1 mesh (c) full behavior with G2 mesh

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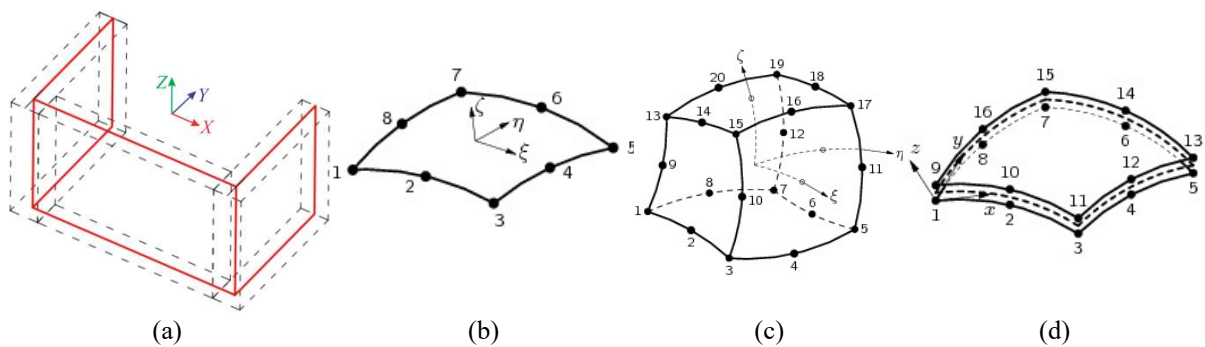
241 3.3 Numerical modeling approach

242 The FEM 3D modeling of structural walls typically follows two main approaches, namely by considering shell
243 or solid elements, being the first less compute-intensive and the second more accurate in accounting for three-
244 dimensional effects. Here, both approaches were used in a first phase to compare their outcomes and concluded
245 about their accuracy. The shell model was prepared by considering the mid-section planes of each wall
246 (schematically depicted in Fig.7a). As a first consideration, this approach is shown to introduce modeling
247 incoherencies. For instance, the length of the wing cantilevers is not properly modeled, as they present higher
248 length than in the reality. Furthermore, the overlapping thicknesses of the walls lead to misleading
249 considerations of the self-weight and mass distribution, and thus of the inertial forces. The implications of these
250 issues in the modeling are discussed later.

251 The shell model was discretized by means of 8-node quadrilateral curved shell elements CQ40S (see Fig.7b),
252 while 20-node iso-parametric brick elements CHX60 (Fig.7c) were used in the case of the solid model.
253 Moreover, the integration scheme of the shell elements consisted of 2×2 with 7 integration layers, while the

254 default integration $3 \times 3 \times 3$ was used for the solid elements. Regarding the boundary conditions, the subassembly
 255 is considered as totally fixed at the base.

256 Furthermore, the shell element CQ40S was also used to discretize the TRM strengthening, which is connected to
 257 the rammed earth by means of 16-node quadrilateral interface elements CQ48I. Note that the interfaces between
 258 strengthening and rammed earth were assumed as rigid due to the absence of experimental data addressing this
 259 behavior. However, it is believed that such simplification does not have a significant influence on general
 260 behavior of the rammed earth component as the use of anchorage devices connecting TRM and substrate
 261 prevents (or postpones) the debonding between them.



262 **Fig.7.** Definition of the out-of-plane model: (a) schematic view of the shell model (b) CQ40S shell element [29]
 263 (c) CHX60 brick element [29] (d) CQ48I interface element [29]

264

265 Mesh sensitivity analysis was conducted to verify the proper mesh size. Hence, three mesh sizes including
 266 25mm (benchmark), 50mm and 100mm were taken into account. Subsequently, the models were subjected to
 267 their self-weight and pushed by a lateral mass-proportional load equal to self-weight. Comparing both lateral
 268 displacements and base shear values with those of the benchmark model revealed that the model with mesh size
 269 equal to 100mm leads to a maximum 1% error. Hence, this mesh size was adopted for further investigations.
 270 Additionally, the resulted vertical reactions were compared with calculated self-weight of the models to ensure
 271 the validity of the models. More details regarding the employed numerical modeling approach can be found in
 272 [33].

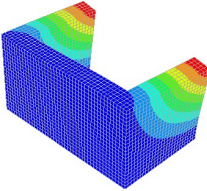
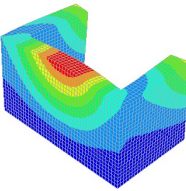
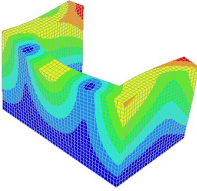
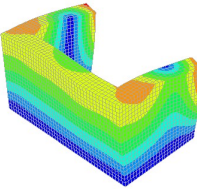
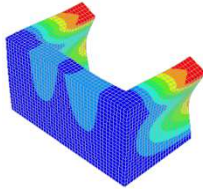
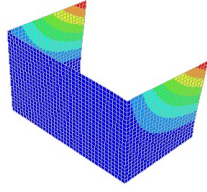
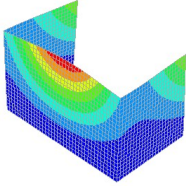
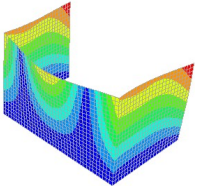
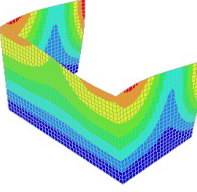
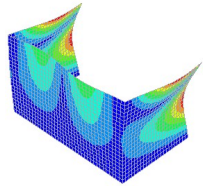
273 It is worthwhile to mention that the current approach has previously being used successfully to model behavior
 274 of plain rammed earth wallets. The obtained outcomes presented a good agreement from both capacity (load and
 275 displacement) prediction and damage aspects [13,24,28].

276

277 **4. Modal Analyses**

278 An eigenvalue analysis was conducted for all models (plain/strengthened and solid/shell) by considering the 20
 279 first modes of vibration, which cover most of the modal mass participation in the dynamic behavior. Among
 280 them, those with the highest contributions for the plain models are reported in Table 3. In general, mode shapes
 281 are very similar, but the shell discretization leads to a more flexible model (higher periods). This situation
 282 results from the definition of the geometry of the shell model by means of the mid-section of the component,
 283 which leads to an increased effective length of the walls (web and wings). Furthermore, the obtained effective
 284 modal mass of the shell model is different from that of the solid model.

285 **Table 3.** Dynamic properties of the plain model (solid and shell) obtained from modal analysis

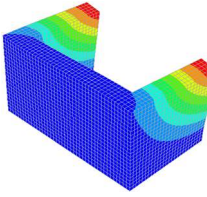
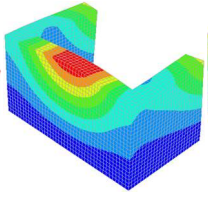
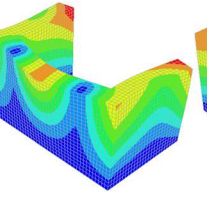
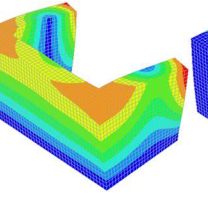
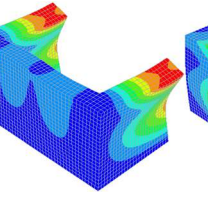
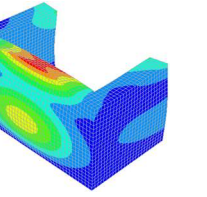
Mode Model	Mode 2	Mode 3	Mode 5	Mode 6	Mode 9
Solid Element					
	$T_2 = 0.052 \text{ sec}$ $CEM_X = 32.1 \%$ $CEM_Y = 4.4 \%$	$T_3 = 0.039 \text{ sec}$ $CEM_X = 32.1 \%$ $CEM_Y = 44.4 \%$	$T_5 = 0.024 \text{ sec}$ $CEM_X = 36.5 \%$ $CEM_Y = 65.8 \%$	$T_6 = 0.023 \text{ sec}$ $CEM_X = 65.3 \%$ $CEM_Y = 65.8 \%$	$T_9 = 0.015 \text{ sec}$ $CEM_X = 78.1 \%$ $CEM_Y = 66.7 \%$
Shell Element					
	$T_2 = 0.055 \text{ sec}$ $CEM_X = 33.4 \%$ $CEM_Y = 5.1 \%$	$T_3 = 0.042 \text{ sec}$ $CEM_X = 33.4 \%$ $CEM_Y = 42.6 \%$	$T_5 = 0.025 \text{ sec}$ $CEM_X = 37.5 \%$ $CEM_Y = 65.6 \%$	$T_6 = 0.024 \text{ sec}$ $CEM_X = 63.5 \%$ $CEM_Y = 65.6 \%$	$T_9 = 0.015 \text{ sec}$ $CEM_X = 78.0 \%$ $CEM_Y = 66.4 \%$

286 T_i is the corresponding period of mode i and CEM is the cumulative effective modal mass.

287

288 The same approach was followed for the strengthened model; though, only outcomes of the solid model are
 289 presented here for the sake of brevity. The implemented strengthening solution does not significantly increase
 290 the mass of the models, while a slight stiffness increment is observed, as the mode shapes, periods, and modal
 291 mass participation changed slightly (Table 4). The period values decreased in the strengthened model, while a
 292 slight increase in the cumulative effective modal mass was observed. This increment can be due to an improved
 293 integrity of the model provided by the adopted strengthening, meaning that some local modes may have been
 294 mitigated in the strengthened model.

295 **Table 4.** Dynamic properties of the strengthened solid element model obtained from modal analysis

Mode 2	Mode 3	Mode 5	Mode 6	Mode 9	Mode 11
					
$T_2 = 0.046$ sec	$T_3 = 0.035$ sec	$T_5 = 0.023$ sec	$T_6 = 0.022$ sec	$T_9 = 0.013$ sec	$T_{11} = 0.012$ sec
$CEM_X = 33.86$ %	$CEM_X = 33.86$ %	$CEM_X = 37.52$ %	$CEM_X = 66.99$ %	$CEM_X = 78.81$ %	$CEM_X = 78.81$ %
$CEM_Y = 4.5$ %	$CEM_Y = 47.70$ %	$CEM_Y = 66.26$ %	$CEM_Y = 66.26$ %	$CEM_Y = 67.11$ %	$CEM_Y = 76.92$ %

296

297 **5. Pushover Analyses**

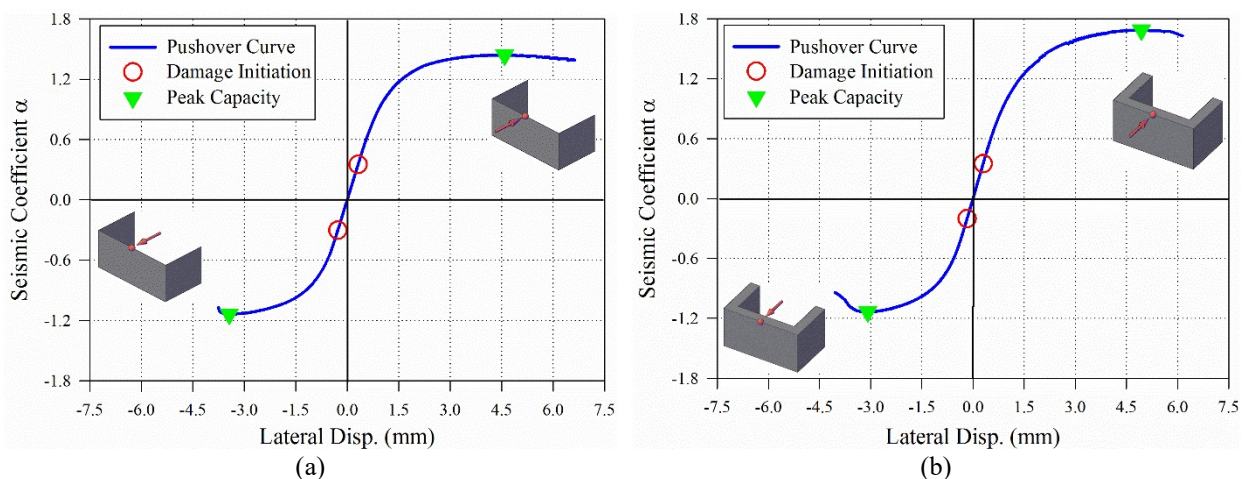
298 As referred previously, the main objective of the current study is to investigate the seismic out-of-plane response
 299 of a representative rammed earth subassembly to be later tested on shaking table, aiming at predicting its
 300 possible failure modes and assessing the performance of the LC-TRM strengthening solution adopted. It is clear
 301 that performing nonlinear time-history analyses would result in more detailed information, but also at a high
 302 computational effort. On the other hand, nonlinear static analyses (so-called pushover) may lead to an
 303 acceptable simulation of the lateral response at lower computational effort, though the predicted damage pattern
 304 may differ significantly from the reality and the reliability of the estimated maximum lateral displacements can
 305 go beyond given acceptable limits [34-35]. In spite of such drawbacks, pushover analyses can provide a
 306 preliminary and general overview of the behavior, whereby they are widely used in the literature. Hence, this
 307 section presents the outcomes of the conducted pushover analyses. In this regard, a mass-proportional lateral
 308 load pattern was applied to push the shell and solid models. However, pushover analyses of the strengthened
 309 model were performed for the case of solid model only, to be justified later. The pushing was performed in the
 310 y-direction to induce the out-of-plane behavior of the web. It is worthwhile noting that due to the un-symmetric
 311 geometry of the models, analyses were performed for both positive (inside) and negative (outside) directions.
 312 Furthermore, the results are presented in two stages, i.e. at damage initiation stage (crack opening) and at the
 313 peak capacity.

314 **5.1 Plain model**

315 The pushover curves representing the seismic coefficient (normalized base shear at each step of analysis to the
 316 weight of the models) versus the lateral displacement of the control node (located at the top of the web's mid-
 317 section) are presented in Fig.8. The pushing direction and considered control node are also shown in Fig.8.

318 Pushing in the negative direction results in lower load and displacement capacities than those obtained for the
 319 models pushed in positive direction. This behavior is explained by the less effective supporting contribution of
 320 the wings in the former direction, which also explains the earlier damage initiation. For instance, in the case of
 321 the solid model, the damage onset in the negative direction occurs for a base shear ratio of approximately 0.2,
 322 whereas in the case of the positive direction this value is of about 0.4. Thus, it can be concluded that the
 323 negative direction is the direction limiting the response of the plain rammed earth model. As previously
 324 mentioned, the total strain rotating crack model is adopted in the current article to identify damage (crack)
 325 initiation and propagation, which follows a smeared cracking approach. Furthermore, the crack direction rotates
 326 according to the direction of the principal strains. Within this concept, the crack initiates when the principal
 327 tensile stress reaches the tensile strength of the material. Then, the tensile strength degradation follows the
 328 predefined softening rule [29].

329 With regard to the post-peak behavior, the models pushed in the negative direction experience a sudden drop
 330 immediately after the peak, while pushing it in the positive direction results in a smooth degradation of the
 331 capacity. In other words, the brittle response of the subassembly when pushed outside the wings results from the
 332 overturning of the web due to loss of connection with the wings. When the models are pushed towards the
 333 wings, the connections are compressed, meaning that the wings are able to counteract the overturning
 334 movement.

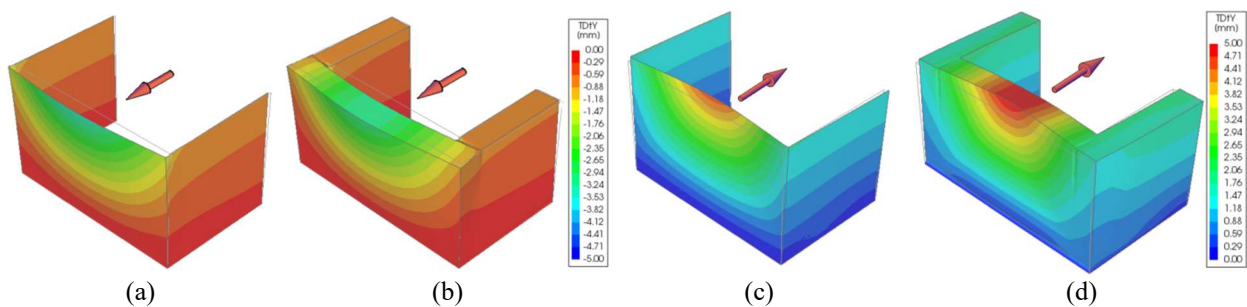


335 **Fig.8.** Pushover curves of the plain models: (a) shell model (b) solid model

336

337 The failure mechanisms of the models were investigated by means of the lateral displacement and principal
 338 tensile strain contours at the peak capacity of the models. The contour maps of the experienced lateral
 339 displacements at the peak capacity of the both solid and shell models pushed in positive and negative directions

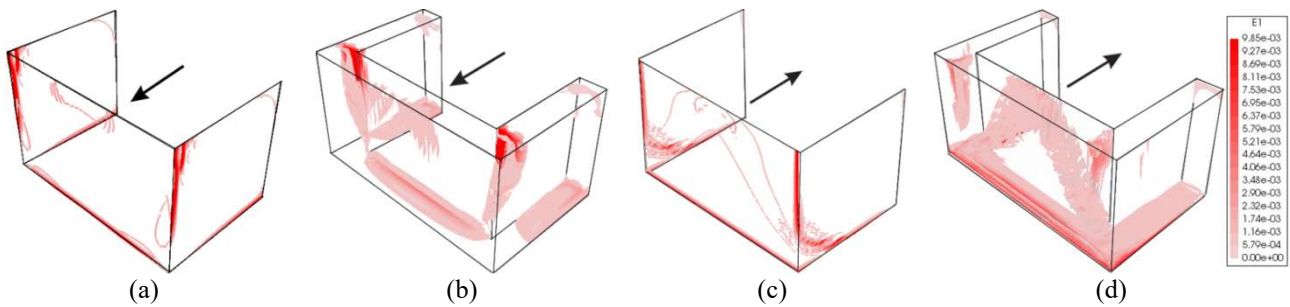
340 are presented in Fig.9. In both cases, the middle of the web experienced the highest lateral displacements, as
341 expected. It should be noted that in the solid model a portion of the wings collaborates in the out-of-plane
342 response of the wall, while in the shell model this contribution seems incipient, as the thickness of the walls is
343 disregarded. The absence of this contribution seems to be a major aspect explaining the different pushover
344 capacities exhibited by the shell and solid models. Furthermore, the displacement contour maps reveal different
345 contribution levels of the wing walls in the models considering different modeling approaches and pushing
346 directions.



347 **Fig.9.** Total lateral displacements at the peak capacity: (a) shell model pushed in the negative direction (b) solid
348 model pushed in the negative direction (c) shell model pushed in the positive direction (d) solid model pushed in
349 the positive direction

350 A more detailed insight on damage detection was achieved by investigating the principal tensile strains at the
351 peak capacity of the models, as presented in Fig. 10. The highest values of the principal tensile strains
352 concentrate around the connection between web and wing walls, at the mid-span section of the web and also at
353 the base of the wall. This pattern can be interpreted as the tendency of the web to detach from the supporting
354 wing walls, bending of its mid-span section and overturning of the wall along the base. As it is evident for the
355 models pushed in the negative direction, a small mid-span section bends and the discontinuity in the tensile
356 strains at the base may be due to loss of integrity, which results in three parts of the model to individually
357 overturn. On the other hand, the large bending mid-span section of the web when the models pushed in the
358 positive direction and the high strain values along the height of the connection between web and wing walls
359 reveals the supporting function of the wing walls, which results in the aforementioned greater displacement and
360 strength capacities. Moreover, the strain contour map at the base of the model pushed in positive direction
361 reveals that the integrity of the wall is probably preserved.

362 Considering the resulted capacities and damage states, the solid modeling approach seems to lead to more
363 accurate outcomes of the out-of-plane behavior, whereby the subsequent investigation was only performed for
364 the solid model.



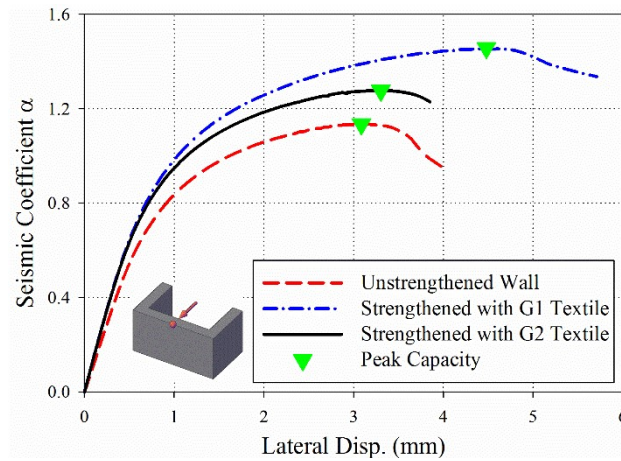
365 **Fig. 10.** Principal tensile strains at the peak capacity of the models: (a) shell model pushed in the negative
366 direction (b) solid model pushed in the negative direction (c) shell model pushed in the positive direction (b)
367 solid model pushed in the positive direction

368 5.2 TRM-strengthened model

369 This subsection aims at evaluating the influence of applying LC-TRM on the seismic out-of-plane performance
370 of the rammed earth component. Following common practical applications, the strengthening is applied on both
371 sides (inner and outer) of the whole model (both web and wing walls).

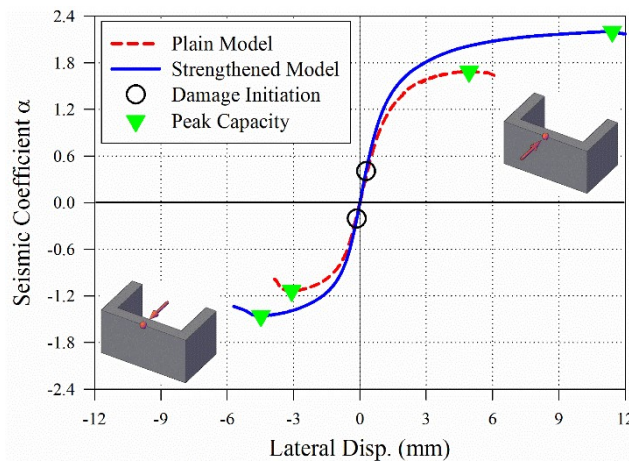
372 Considering that both G1 and G2 meshes have a relatively similar cost, around 0.8 Euro/m² [31], the rational
373 selection between these two meshes is related to the best structural performance. In this regard, both meshes are
374 examined numerically by pushing the solid model in the negative out-of-plane direction (see Fig.7) to assess
375 their strengthening effectiveness. The corresponding pushover curves are shown in Fig.11. As it can be seen, the
376 strengthening with the G2 LC-TRM results in a 7.0% and 13% increase in displacement and load capacities,
377 respectively; while for the G1 composite, the increases are about 45% and 29%, for displacement and strength,
378 respectively. In conclusion, using the G1 LC-TRM composite seems more reasonable, whereby the subsequent
379 investigation is conducted using this particular solution.

380 Thus, the pushover curve of the strengthened model (with G1 mesh) in comparison with that of the plain one is
381 presented in Fig.12. As it can be observed, the applied strengthening slightly increases the pre-peak stiffness of
382 the model by controlling the cracking process, though it has no meaningful influence on the onset of damage,
383 since it tends to initiate in the rammed earth. The most highlighted influence of the strengthening can be
384 observed in the lateral displacement and load capacities, which in the case of the negative direction leads to
385 increase of 45% and 29%, respectively. As previously discussed, the response in the positive direction is less
386 critical than that in the negative one, but 131% and 31% improvement can be observed for displacement and
387 load capacities, respectively.



388
389 **Fig.11.** Pushover analysis of the strengthened solid models in the negative out-of-plane direction

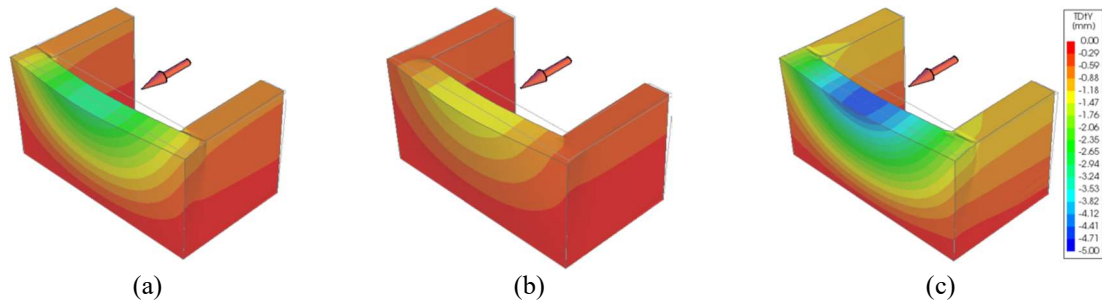
390



391
392 **Fig.12.** Pushover curves of the plain and strengthened models

393

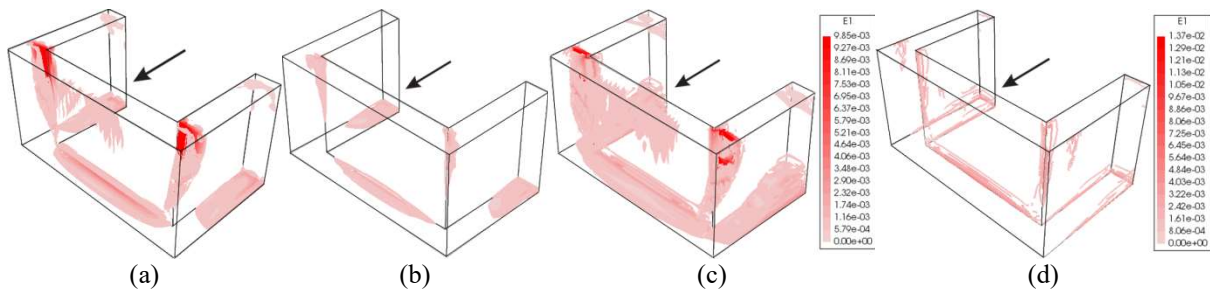
394 The contours of lateral displacements of both plain and strengthened models pushed in the negative direction are
 395 presented in Fig.13. For the sake of brevity, only this critical direction is here discussed. By comparing the
 396 experienced lateral displacements of the strengthened model with those of the plain one at the peak capacity of
 397 the plain model (see Fig.13a and b), it is possible to distinguish a significant reduction especially in the mid-
 398 span section (mid-span bending) of the web. This was expected due to previously mentioned increase in the
 399 lateral stiffness of the wall. On the other hand, the displacement contour of the strengthened model at its peak
 400 capacity exhibits considerable improvements with respect to the unstrengthened case. For instance, a larger
 401 section of the strengthened web tends to deform, meaning that a greater resistance against out-of-plane actions is
 402 achieved. Furthermore, a higher contribution of the wing walls is evident for the strengthened model.



403 **Fig.13.** Total lateral displacements of the models pushed in the negative direction: (a) plain model at its peak
404 capacity (b) strengthened model at the peak capacity of the plain model (c) strengthened model at its peak
405 capacity
406

407 The contours of the principal tensile strains in the rammed earth for both plain and strengthened models are
408 presented in Fig.14, which additionally presents the individual strain contours of the LC-TRM strengthening. A
409 considerable reduction in the principal tensile strain levels was observed for the strengthened model at a lateral
410 load equal to the peak capacity of the plain one (see Fig.14a and b). As it can be seen, the detachment of the web
411 from the wing walls and the bending of the web's midsection are delayed. Furthermore, the tensile strains at the
412 base of the wall are decreased. This situation can be interpreted as an improvement of the integrity and lateral
413 stiffness of the wall due to the application of the strengthening. In other words, the employed strengthening
414 solution enables the wall to redistribute the stresses and decreases its tendency to overturn. The contour of the
415 principal tensile strains of the strengthened model at its peak capacity is presented in Fig.14c, which shows that
416 the final failure mechanism is similar to that of the plain model, while apparently a larger midsection of the web
417 is bending. As previously stated, this larger section means that a higher lateral load is required to initiate the
418 collapse mechanism. Moreover, the high strain values concentrated at the base demonstrates the efficiency of
419 the applied strengthening in preserving the integrity of the model.

420 Regarding the damage in the strengthening composite material at the peak capacity of the strengthened model
421 (see Fig.14d), it is clear that strengthening contributes to the stress transferring in regions of the rammed earth
422 that are prone to fail, i.e. at the connections of the web with wing walls and also at the base. Thus, the efficiency
423 of the TRM-strengthening in enhancing the out-of-plane response of the rammed earth components is
424 numerically demonstrated.



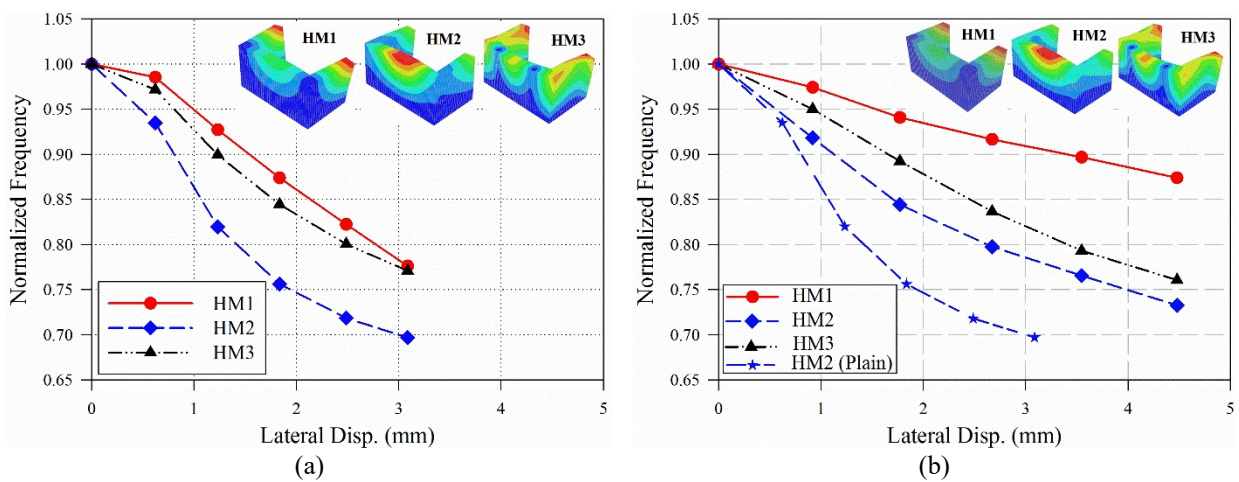
425 **Fig.14.** Principal tensile strains of the models pushed in the negative direction: (a) plain model at its peak
426 capacity (b) strengthened model at the peak capacity of the plain model (c) strengthened model at its peak
427 capacity (d) LC-TRM strengthening at peak capacity of the strengthened model
428

429 *5.3 Influence of damage on dynamic properties*

430 During the pushover analyses, the damage initiates and develops leading to a progressive reduction of the
431 stiffness. In this regard, eigenvalue analyses were conducted at selected steps of the analyses of the models
432 (plain and strengthened) pushed in the negative direction, starting with the initial undamaged condition up to the
433 peak capacity. Hence, the changes in frequencies are considered as an indicator of damage state in the walls.
434 The detailed analysis of the results confirmed the progressive reduction in frequencies of the models with
435 damage progression and demonstrated that the damage also changes the mode shapes and modal mass
436 contributions of the modes. Therefore and for the sake of simplicity, the three modes with the highest effective
437 modal mass contribution in the undamaged condition were selected for comparison. Furthermore, the mode
438 shapes of these selected modes were also considered in each evaluated step to find the modes more compatible
439 with the original ones. As the orders of the modes are not necessarily identical in all considered steps, they are
440 called hereafter as high participating modes (HM). The frequencies of each HM were normalized to the initial
441 frequency value (corresponding to the undamaged state) as a function of the corresponding displacement at the
442 control node, as represented in Fig.15. It can be observed that the greatest frequency reduction generally belongs
443 to the mode with the highest contribution (HM2 in this case).

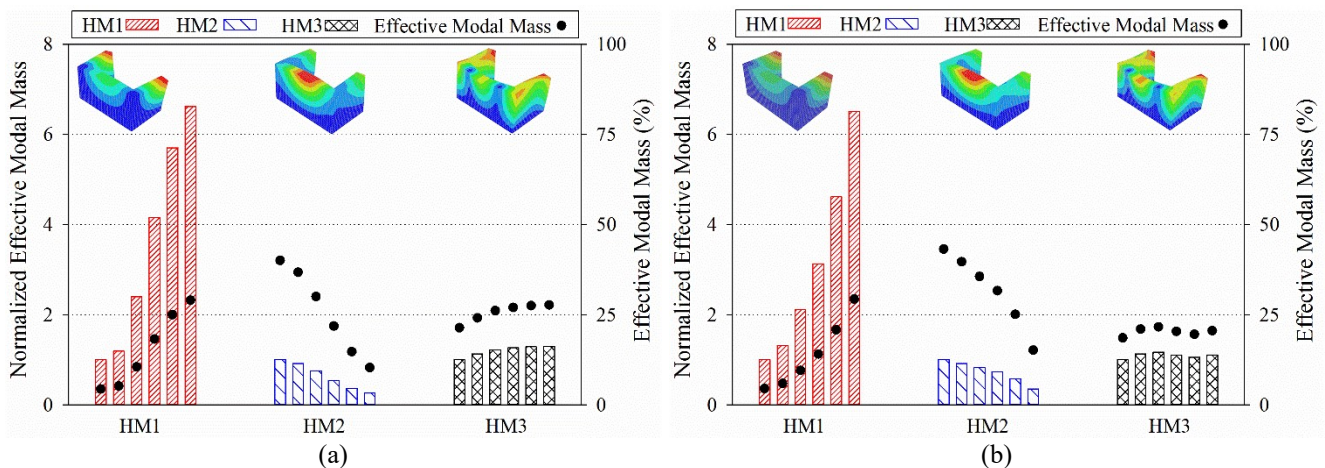
444 The changes in frequencies of the strengthened models are presented in Fig.15b, which also shows a reduction
445 of the frequencies with damage development, though a smaller reduction can be interpreted as the efficiency of
446 the strengthening in limiting damage. The maximum frequency drop in the plain model is of about 30%, while
447 the corresponding value for the strengthened model is about 22%. It should be noted that these reduction values
448 do not correspond to identical lateral displacement values. In other words, the strengthened model experienced
449 less damage at higher lateral displacement values, which clears its efficiency in damage reduction.

450 The analysis of the effective modal mass of the HMs can also clarify the damage pattern evolution of the models
 451 during the pushover analyses and its influence on the changes in stiffness. In this regard, the effective modal
 452 mass in each step is normalized to the corresponding value at the initial undamaged state. As it is clear from
 453 Fig.16, the applied strengthening does not considerably change the modal characteristics of the model, meaning
 454 that the development of the damage pattern is similar in both plain and strengthened models. Furthermore, the
 455 contribution of HM2 is shown to drastically decrease in both models in favor of the increase in contribution of
 456 HM1, while the effective model mass of HM3 shows minor variations. This behavior results from the influence
 457 of the progressive detachment of the web from the wing walls on the dynamic behavior of the models.



458 **Fig.15.** Reduction in frequencies of the models during the pushover analyses in the negative direction: (a) plain
 459 model (b) strengthened model

460



461 **Fig.16.** Changes in the effective modal masses of the models during the pushover analyses and pushed in the
 462 negative direction: (a) plain model (b) strengthened model
 463

464

465 **6. Dynamic Analyses**

466 In addition to the pushover analyses, incremental dynamic analyses (IDA) [36] were also performed to evaluate
467 the seismic capacity of the plain and strengthened out-of-plane models. Furthermore, performing the IDA
468 allowed evaluating the accuracy of the pushover analysis in predicting the seismic behavior of the models. This
469 section starts by presenting the employed ground motion input and damping conditions. Then, the main
470 outcomes of the IDA are discussed and compared with those from pushover analyses.

471

472 *6.1 Seismic input and time-history procedure*

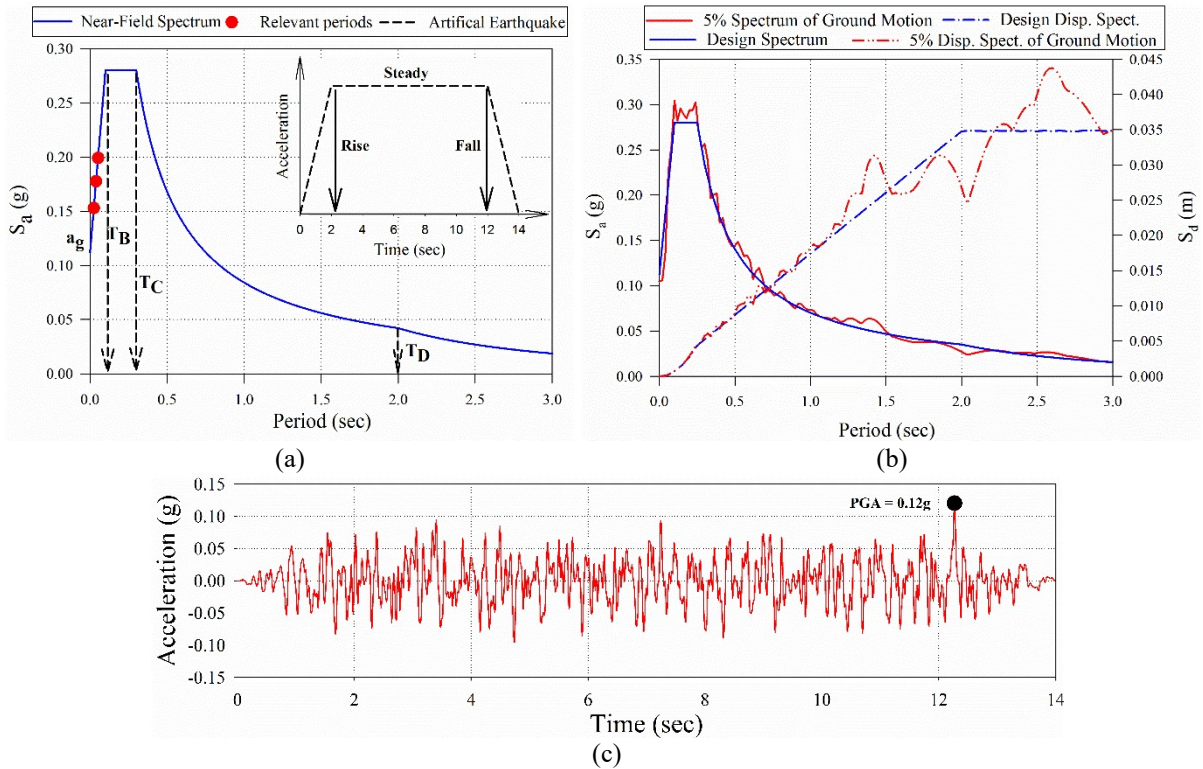
473 The reliability of time-history analyses is a function of the proper ground motion input selection, which should
474 be compatible with the seismicity characteristics of the hosting region. It can be selected from previously
475 recorded seismic events or it can be generated artificially; though each approach has its own drawbacks. For
476 instance, selecting ground motion records from previous events may not exactly satisfy the seismological
477 conditions of the site, requiring scaling of the ground motion. In this regard, a variety of methods are proposed
478 in the literature, nevertheless it should be highlighted that diversity in the outcome should be expected by
479 following this procedure [37-38]. On the other hand, artificial ground motion records may not precisely
480 represent the frequency and energy contents of a real earthquake. The discussion of these drawbacks is beyond
481 the scope of the current article; though it should be noted that both approaches are valid options to proceed with
482 the dynamic analyses. In this case, it was preferred to use an artificially generated ground motion.

483 The municipality of Odemira in Alentejo region, Portugal was selected as the site location of the rammed earth
484 subassembly, being worthwhile to mention that this region presents an important rammed earth built heritage
485 and moderate seismic hazard [39]. Fig.17a presents the corresponding design spectrum and the characteristics of
486 the artificial ground motion record derived from the Portuguese national annex of Eurocode 8 [40] for the near-
487 field scenario. Moreover, Fig.17a also situates the modes with the highest modal mass participation of the plain
488 model within the considered spectrum. As it can be seen, all modes are in the initial branch of the spectrum,
489 which shows the sensitivity of the model to earthquakes with high-frequency content.

490 The Simqke-gr software [41], developed at the University of Brescia, was employed to generate an artificial
491 ground motion record compatible with the considered elastic design spectrum. This process is controlled
492 considering the acceleration and displacement design spectrums, for which an acceptable agreement should be

493 obtained (see Fig.17b). The SeismoSignal software [42] was then used to perform a baseline correction by
 494 filtering the frequencies below 0.1 Hz and above 20 Hz. In conclusion, the accelerogram shown in Fig.17c was
 495 applied at the base of the model in the y-direction.

496



497 **Fig.17.** Artificially generated ground motion: (a) design spectrum for near field (type2) earthquakes in Odemira
 498 (southern Portugal) and corresponding artificial earthquake characteristics (b) acceleration and displacement
 499 response spectra of the generated ground motion record (c) earthquake input

500

501 Additionally, it is crucial to define a proper damping of the system to take into account the energy dissipation. In
 502 this regard, the Rayleigh damping approach is here adopted. This approach requires selecting the principal
 503 modes and assigning them a damping coefficient. These natural frequencies should be chosen in a way that the
 504 constructed damping matrix correctly characterizes the dissipative behavior of the rammed earth model in the
 505 desired frequency range. In this regard, the modes with significant mass participation were selected. Although,
 506 there is no general consensus about the damping ratio value in rammed earth constructions, in-situ dynamic
 507 identification tests [25] resulted in damping ratios of the studied rammed earth buildings in the range of 2.5-
 508 4.0%. Hence, a 3% viscous damping ratio was considered in the current study.

509

510 **6.2 Incremental dynamic analysis versus pushover analysis**

511 The plain and strengthened models were subjected to IDA by scaling up (step by step starting from 1.0 up to the
512 failure of the subassembly [36]) the artificially generated ground motion record with the main purpose of
513 extracting the corresponding envelope of the hysteretic behavior (see example in Fig.18a), which relates the
514 seismic coefficient to the lateral displacement of the control node at top of the web wall's mid-section (same
515 node considered in the pushover analyses). The extracted envelopes of the plain and strengthened models for
516 identical scale factors are presented in Fig.18b, where they are compared with the respective pushover capacity
517 curve. It is clear that the strengthened model experienced lateral displacements substantially lower than those of
518 the plain one. Furthermore, the area of the hysteretic curves' envelope, related to the dissipated energy,
519 evidences an important enhancement with respect to the energy dissipation capacity. Regarding the comparison
520 with the pushover curves, it seems that they lead to misleading estimation of the load and displacement
521 capacities, as it is later discussed.

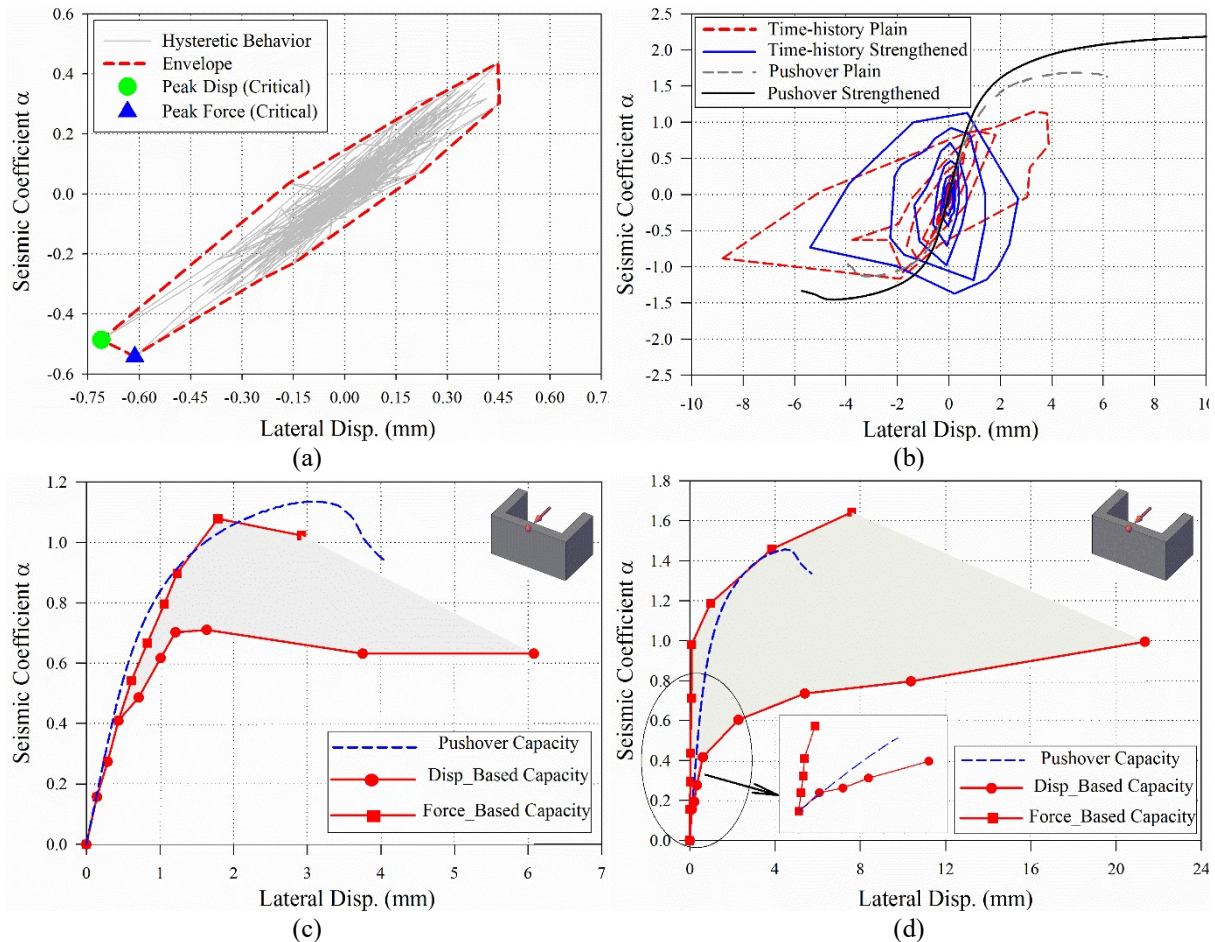
522 The results of the IDA were further investigated using two approaches, namely based on the peak experienced
523 displacements (displacement-based) and on the maximum induced base shear forces (force-based). The
524 objective is to evaluate the reliability of the pushover analyses in predicting the load and displacement capacities
525 of the rammed earth subassembly under study. The results are only presented for the critical out-of-plane
526 direction in Fig.18c and d, respectively for the plain and strengthened models. As it can be observed, the
527 pushover curves have a relatively good agreement with the force-based IDA curves, in both models, but they
528 cannot accurately predict the ultimate displacement capacity. Regarding the pushover curve of the plain model,
529 it falls outside the capacity range (limited by force- and displacement-based IDA curves), particularly at peak
530 load. In turn, the pushover curve of the strengthened model falls within the corresponding capacity range. The
531 differences observed between pushover and time-history behaviors, discussed above, can be due to a deficiency
532 of the pushover analysis in predicting damage evolution, which leads the stiffness loss to be clearly different in
533 the two analysis strategies.

534

535

536

537



538 **Fig.18.** Results of the IDA: (a) example of the hysteretic behavior curve for the plain model (scale factor of 4.0)
 539 (b) hysteretic curves' envelopes of the plain and strengthened models (c) displacement- and force-based IDA
 540 curves of the plain model (d) displacement- and force-based IDA curves of the strengthened model

541

542 7. Conclusions

543 The primary objective of the current study was to assess the out-of-plane seismic behavior of a representative
 544 rammed earth subassembly and evaluate the mechanical efficiency of a low-cost TRM strengthening solution,
 545 which was achieved by conducting a series of advanced nonlinear static and dynamic analyses. Furthermore, the
 546 outcomes of this research serve to support the design of a near future shaking table test and to provide an insight
 547 into advanced approaches for predicting the seismic behavior of rammed earth structures. The outcomes of the
 548 mass-proportional pushover analyses showed that the critical direction of the plain model corresponds to
 549 pushing it outside the wing walls and that the out-of-plane failure mechanism consists on the detachment of the
 550 web wall from the wing (transversal) walls, bending of the mid-span section of the web and overturning of each
 551 wall of the component. Furthermore, damage (cracking) was found to initiate at very low lateral load levels due
 552 to the relevant nonlinear behavior of rammed earth.

553 Regarding the modeling of the strengthening composite system, the clear lack of experimental evidence on the
554 bond between substrate/mortar and mortar/mesh led to assume, at this stage, a perfect bond behavior. Despite
555 the limitations inherent to this simplification in reproducing the behavior of the strengthening system, it allowed
556 for a first and minimally reliable insight into the expected behavior of the strengthened subassembly. In this
557 regard, the TRM strengthening was found to slightly increase the pre-peak lateral stiffness of the model
558 subjected to pushover analyses, though it did not promote a delay on the damage initiation. The clear influence
559 of the strengthening system was visible through the increase of about 45% and 29% in terms of displacement
560 and load capacities in the critical direction, respectively.

561 Conducting modal analyses at different steps of the pushover curves allowed evaluating the damage
562 development of the models. The main conclusion is that the strengthening led to a lower maximum drop of the
563 higher contributing modes at failure of the models (30% and 20% for the plain and strengthened models,
564 respectively), despite the strengthened one achieving higher lateral displacements. Thus, the strengthening
565 solution is found to delay the global damage development in the rammed earth component.

566 Finally, the incremental dynamic analyses carried out allowed concluding about the reliability of the pushover
567 approach in predicting the seismic behavior of rammed earth structures. In general, the pushover curves of the
568 models pushed in the critical direction were found to present good agreement with the force-based IDA curves.
569 On the other hand, the displacement-based IDA curves show that the rammed earth subassembly can achieve
570 greater displacement levels than those predicted by the pushover analyses.

571 **Acknowledgments**

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574 projects POCI-01-0145-FEDER-016737 (PTDC/ECM-EST/2777/2014) and POCI-01-0145-FEDER-007633.
575 The support from grant SFRH/BPD/97082/2013 is also acknowledged.

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