

Numerical Investigation of the In-Plane Seismic Performance of Unstrengthened and TRM-Strengthened Rammed Earth Walls

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Abstract:

The large availability of raw earth around the World led to its extensive use as a building material through history. Thus, earthen materials integrate several historical monuments, but their main use was to build living and working environments for billions of people. On the other hand, past earthquakes revealed their inadequate seismic behavior, which is a matter of concern as a significant percentage of earthen buildings are located in regions with medium to high seismic hazard. Nevertheless, their seismic behavior and the development of efficient strengthening solutions are topics that are not yet sufficiently investigated in the literature. In this context, this study investigates numerically the in-plane seismic behavior of a rammed earth component by means of advanced nonlinear finite element modelling, which included performing nonlinear static (pushover) and nonlinear dynamic analyses. Moreover, the strengthening effectiveness of a low-cost textile reinforced mortar on such component was also evaluated. The strengthening was observed to increase the load and displacement capacities, to preserve the integrity for higher lateral load levels and to postpone failure without adding significant mass to the system. Furthermore, the pushover analysis was shown to predict reliably the capacities of the models with respect to the incremental dynamic analysis.

Keywords: Rammed Earth; Strengthening; Textile Reinforced Mortar; In-plane Behavior; Numerical Modelling; Pushover Analysis; Nonlinear Time-History Analysis.

1. Introduction

Different building techniques were developed since the life-style of mankind shifted from nomadic to sedentary. The development of these techniques was mainly promoted by the new materials readily available in the settling region, meaning that most of them relied in the use of raw earth, stone and timber. Among the many earth-based building techniques developed through time, adobe masonry and rammed earth are probably among the most well-known ones and widespread in the world (Houben and Guillaud 1994).

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32 Building in rammed earth consists in compacting layers of earth with adequate particle size
33 distribution and moisture inside two parallel shutters to erect walls (Miccoli et al. 2014). The strength
34 of the material is governed by binding forces due to capillary suction originated at the porous
35 structure of the material with drying (Jaquin et al. 2008) as well as by the friction and interlocking
36 capacity of the coarse particles (Silva et al. 2016 and Silva et al. 2018a). Nowadays, stabilization with
37 cement is a procedure typically used to improve the properties of rammed earth, as a consequence of
38 binding mechanisms associated to the formation of cementitious gels. Nevertheless, lime stabilization
39 has an historical relation with rammed earth, as for instance this procedure was used to build several
40 fortresses in the Iberian Peninsula centuries ago (González 1999). The compaction was traditionally
41 performed with manual rammers, while nowadays this process is facilitated by the use of mechanic
42 equipment, such as pneumatic rammers (Minke 2006).

43 Regarding the origin of rammed earth, Jaquin et al. (2008) argues that it was independently developed
44 in China and in Mediterranean region and later on was spread by the settlers of the new World. This
45 universality of rammed earth gave origin to different names according to the country, namely *Taipa* in
46 Portugal, *Tapial* in Spain, *Pisé* in France, *Terra Battuta* in Italy, *Stampflehm* in Germany, *Chineh* in
47 Iran, *Hangtu* in China and *Pakhsa* in Uzbekistan (Jaquin et al. 2008).

48 Several monuments made of earth can be found around the world. Nevertheless, the large availability
49 and low cost of the material were the main reasons why it constituted an appropriate choice for
50 sheltering societies with economic issues, as well as for hardly accessible regions and isolated rural
51 areas. The fact is that the extensive use of raw earth as building material resulted in about 33% of the
52 world population estimated to live in such environments in the nineteen eighties (Houben and
53 Guillaud 1994). Since then, this percentage has been dropping, but the recent pursue for more
54 sustainable building solutions, led to a renewed interest for this type of constructions. Hence, the
55 investigation of the structural performance of earth constructions is vital to provide tools that grant the
56 adequate assessment of the safety of new and existing constructions.

57 Several factors such as rainwater, soluble salts and temperature oscillations can lead to occurrence of
58 damage in rammed earth constructions (Parreira 2007). Furthermore, these constructions are
59 significantly vulnerable to earthquakes, as they are mainly built to withstand gravity loads. The low
60 tensile strength, lack of continuity at corners and wall connections, the occurrence of concentrated
61 roof loads, the absence of ring beams, discontinuity between roof and walls, existence of long walls,
62 absence of proper foundation, poor lintel supports, irregularity on the opening distribution and
63 existence of opening close to corners constitute the main factors contributing to high seismic
64 vulnerability of rammed earth constructions (Correia et al. 2015). In spite of such weaknesses, it was
65 observed that an important percentage of these buildings is located in regions with medium to high
66 seismic hazard (De Sensi 2003), which caused many inhabitants and historical monuments to be

67 severely affected by occurrence of earthquakes. An approximate estimation revealed that about 60%
68 of fatalities in earthquakes during the second half of the last century were attributed to failure of
69 unreinforced masonry components (Coburn 2002). For instance, the destruction of the historical
70 citadel of Arg-e-Bam by the 2003 earthquake is one of the most catastrophic cases demonstrating the
71 high seismic vulnerability of earthen constructions, namely of adobe masonry.

72 Regarding the seismic vulnerability of rammed earth structures, some numerical and experimental
73 studies have been previously conducted to assess the seismic performance of rammed earth
74 constructions. Most of the experimental studies are limited to the component level (wallets) by
75 conducting uniaxial or diagonal compression tests to characterize material properties and investigate
76 its local behavior (see Yamin et al. 2004; Miccoli et al. 2014; Miccoli et al. 2015; Bui and Morel
77 2009). With respect to the full-scale building, Bui et al. (2011) employed the frequency domain
78 decomposition procedure to extract dynamic properties of rammed earth structures from in-situ
79 dynamic identification tests. It was concluded that Eurocode 8 equation for estimating the
80 fundamental period of the building would be still valid for rammed earth buildings and that their
81 damping ratio may vary between 2.5-4.0%. In addition, Wang et al. (2016) tested on shaking table a
82 model of a typical rural rammed earth building with one story. It was observed that the failure was
83 characterized by out-of-plane rotation, cracking at the corners and at the loading points where the roof
84 load was transferred to the walls.

85 Regarding the numerical studies, three main different strategies have been employed so far, namely
86 simplified (using limit analysis), finite element (FE) and discrete element (DE) modeling. Ciancio and
87 Augarde (2013) proposed static (elastic analysis) and kinematic (ultimate strength analysis)
88 approaches to evaluate the out-of-plane wind capacity of rammed earth walls. However, the
89 simplifications introduced by such models may not be representative of real conditions and behavior.
90 Regarding FE modeling, micro- and macro-modeling approaches were used in Miccoli et al. (2014) to
91 simulate the response of rammed earth wallets tested under uniaxial and diagonal compression. In this
92 case, it was concluded that both methods showed a good agreement against experimental responses.
93 Hence, the interface between layers can be ignored and homogenous material properties can be
94 assumed in the whole rammed earth component. Allahvirdizadeh et al. (2018) used the macro-
95 modeling approach to evaluate the out-of-plane seismic performance of a rammed earth subassembly.
96 It was shown that plain walls may fail due to detachment from orthogonal walls and bend over their
97 mid-section. DE modeling is less used than FE modeling, though it was adopted in Bui et al. (2015) to
98 take into account the influence of the layers on the structural behavior of rammed earth components.
99 Similarly to FEM, it was concluded that the results obtained by models with or without interfaces
100 between layers were similar, even when very low interface parameters were considered.

101 The literature on earthen structures has been also focused on the investigation of adequate
102 strengthening solutions, which aim mainly at reducing their seismic vulnerability. In this regard, it is
103 recommended to implement repair works before applying strengthening solutions. Erosion and
104 cracking are typical damage types found in earthen walls, which can be repaired by local rebuilding
105 and injection of compatible grouts (Figueiredo et al. 2013; Silva et al. 2016 and Illampas et al. 2017).
106 With respect to strengthening, several solutions have been proposed, namely the use of boundary
107 wooden elements tying of the walls, introduction of ring beams, and application of composite-based
108 materials (Figueiredo et al. 2013; Yamin et al. 2004).

109 The strengthening of masonry with composite-based solutions has been receiving a great attention in
110 the last two decades, especially with respect to the use of solutions based on fiber reinforced polymers
111 (FRP). The popularity of FRP-based strengthening was driven by its significant efficiency in
112 increasing the shear/flexural capacity and ductility of components with a negligible increase in mass
113 (high strength and stiffness to weight ratio) and ease of application. In other words, this technique can
114 strongly improve the weak tensile strength of masonry and prevent or postpone the occurrence of
115 brittle failure. Despite that, it presents several drawbacks, such as poor fire/high-temperature
116 resistance (low glass transition temperature), lack of vapor permeability, low reversibility, high cost
117 and incompatibility with masonry substrate (Papanicolaou et al. 2008; Valluzzi et al. 2014; Michels et
118 al. 2015).

119 Most of these issues result from using organic matrices in the application process. Therefore,
120 alternative techniques have been developed in order to integrate more compatible matrices such as
121 cement- or lime-based mortars. Moreover, sheets are substituted by mesh grids to grant a good
122 embedment and bond to the support. These alternatives are known as Steel Reinforced Grout (SRG),
123 Fiber Reinforced Cementitious Matrix (FRCM) or Textile Reinforced Mortar (TRM). It is evident that
124 their effective application requires understanding their behavior, both at the level of characteristics of
125 the constituent materials and their interaction. In this regard, several experimental studies have been
126 conducted to characterize material properties of composite materials and to investigate its influence
127 on the performance of masonry components (see Papanicolaou et al. 2007; De Felice et al. 2014;
128 Ascione et al. 2015; Mordanova et al. 2016; Garofano et al. 2016; Mininno et al. 2017). However,
129 most of the research conducted so far on strengthening of masonry walls with TRM is addressed to
130 brick masonry rather than rammed earth.

131 In this context, the current study presents a numerical investigation on the in-plane behavior of an
132 unstrengthened and TRM-strengthened rammed earth wall by means of an advanced nonlinear finite
133 element model. The outcomes of this study will be used to design an experimental program on an
134 identical model, but serve firstly to provide a better understanding on the in-plane shear behavior of

135 rammed earth walls subjected to dynamic loading and on the strengthening efficiency of the TRM
136 technique.

137

138 **2. Model Definition**

139 In order to reliably assess the in-plane seismic performance of rammed earth walls and also evaluate
140 the strengthening effectiveness of TRM on enhancing their behavior, it is essential to consider a
141 representative geometry and construct valid numerical models. For this purpose, an unstrengthened
142 and a TRM-strengthened models were considered for numerical analysis. This section addresses the
143 main aspects regarding the definition of such models, namely in terms of geometry, nonlinear material
144 models and meshing considerations. Furthermore, it should be noted that the models were
145 implemented and computed using DIANA 10.1 software (DIANA FEA BV 2017).

146

147 **2.1 Geometry**

148 An I-shaped geometry was idealized to investigate the in-plane seismic performance of the rammed
149 earth walls. The web wall transfers lateral loads, while the wing walls are only necessary for stability
150 objectives during the experimental program, which is planned to be designed in near future based on
151 outcomes of the current study. Therefore, the wing walls are required to avoid changing the desired
152 failure mode of the model, which is discussed in detail in the following sections.

153 Furthermore, the definition of the geometry of the model demands satisfying observable conditions
154 (be compatible with real rammed earth buildings) and limitations of the experimental facilities.
155 Hence, the outcomes of previous surveys on rammed earth dwellings located in Alentejo region
156 (southern Portugal) were taken into account (Correia 2007 and Dominguez 2015). This region
157 presents an expressive number of rammed earth dwellings (see Fig. 1), thus the statistical analysis of
158 the different in-plane components identified from the aforementioned survey is expected to provide
159 valid dimensions for the model.

160

161 (a) (b)
162 **Fig. 1.** Rammed earth constructions in Portugal: (a) Alentejo region (in red); (b) examples of typical dwellings
(Silva et al. 2018b)

163

164 The average height and length of the surveyed buildings are presented in Fig. 2. As it can be seen, the
165 average height and length values are 2.20 m and 3.75 m, respectively. Moreover, the thickness of the
166 walls was in all cases of about 0.5 m, which led to consider this same value in this study. Considering

167 the observed values (and also the limitations of the testing facilities), two geometries were defined as
168 illustrated in Fig. 3. It is worthwhile to note that all dimensions of the considered models are
169 identical, excepting the length of their wing walls. It is expected that such difference affects the
170 failure mode of the web walls. It should be noted that the wing walls are very important in the
171 experimental setup due to stability concerns. The considered geometries result in components with a
172 weight of approximately 134 kN and 160 kN (assuming density equal to 2000 kg/m³) for the model
173 with 50cm and 80cm wing walls, respectively. The final component is aimed to be tested on a shaking
174 table, which allows for a maximum mass of about 21 tons and plan dimensions of about 5.6 × 4.6 m².

175

176 **Fig. 2.** Average length and height of the rammed earth walls identified in the surveyed rammed earth dwellings

177

178 (a) (b)
Fig. 3. Considered in-plane models: (a) 50 cm long wing walls (b) 80 cm long wing walls

179

180

181 2.2 Material Properties

182 Conducting advanced FE analyses requires assigning representative properties to the materials
183 considered to contribute to the structural behavior, as well as to the several levels of interaction
184 between them (see Fig. 4). In order to balance reliability with computational power requirements, the
185 current study adopted the macro-modeling approach, meaning that the rammed earth and the
186 strengthening composite system (mesh and mortar) were assumed with homogenized properties. As
187 previously discussed, ignoring the influence of the interfaces between rammed earth layers is not
188 expected to significantly affect the obtained outcomes, despite being preferential surfaces for cracking
189 development and failure. The simplification assumed for the strengthening composite prevents the
190 simulation of the sliding failure mode of the mesh within the mortar, which can be a non-negligible
191 aspect when longitudinal and transversal yarns are not welded at the nodes. Furthermore, the
192 connection between the rammed earth and the mortar was assumed as perfectly bonded, meaning that
193 the model is not able to simulate debonding failure. These failure modes are expected to affect the
194 local behavior of the strengthening; nevertheless the total absence of reliable experimental data on
195 bond behavior of TRM-strengthened rammed earth justifies the assumed simplifications.

196

197

198 **Fig. 4.** Detailed view of the materials and interaction levels contributing for the structural behavior of the
199 unstrengthened and strengthened models

200 **2.2.1 Rammed Earth**

201 From a general point of view, rammed earth presents a fragile response under tension due to low
202 tensile strength values, meaning that its seismic response is expected to be controlled by cracking
203 mechanisms. In this regard, the use of smeared cracking models is expected to result in adequate
204 simulation of the mechanical behavior of rammed earth (see Silva et al. 2014; Librici 2016 and
205 Allahvirdizadeh 2017). Thus, the total strain rotating crack model (TSRCM) implemented in DIANA
206 10.1 (DIANA FEA BV 2017) was used to simulate the rammed earth material of the models. In this
207 model the crack initiates when the principal tensile stress reaches the tensile strength of the material
208 and its direction rotates according to the direction of the principal tensile strain. Furthermore, in the
209 post-peak region, the tensile strength degrades following the predefined softening rule (in this study
210 an exponential curve is adopted). Furthermore, it should be noted that the unloading and reloading of
211 the TSRCM (hysteretic behavior) is simulated by a secant approach, meaning that
212 unloading/reloading is processed to/from the origin, respectively.

213 Identifying the nonlinear mechanical properties of rammed earth is still a fundamental challenge
214 within the investigation of this type of structures, as there are different parameters affecting them,
215 such as particle size distribution, moisture content, compaction (rate and type), void ratio, cohesive
216 strength of particles, fiber content, and quantity and type of additions. Thus, the values available in
217 the literature present high scattering (see Liley and Robinson 1995; Yamin et al. 2004; Parreira 2007;
218 Maniatidis et al. 2007; Bui and Morel 2009; Miccoli et al. 2014).

219 Previous studies revealed the expressive nonlinear behavior of rammed earth under compression,
220 which initiates at very low stress levels (Silva 2013). Such behavior led conventional parabolic
221 relationships typically used to simulate concrete and masonry to be deemed as inadequate for rammed
222 earth, as they result in excessively rigid behaviors that do not portray adequately the nonlinear
223 behavior of the rammed earth. Adopting a multi-linear relationship extracted from average of results
224 of uniaxial tests was shown to lead acceptable outcomes instead (Miccoli et al. 2015; Librici 2016).
225 Thus, the current numerical investigation adopted a previously calibrated multi-linear stress-strain
226 relationship in compression (Silva et al. 2014), portrayed in Fig. 5. The experimental results used to
227 obtain this relationship were obtained from compression tests on rammed earth cylindrical specimens,
228 which were made of soil collected from Alentejo region (Silva et al. 2016). It should be noted that due
229 to lack of results in the post-peak phase, its development was idealized by assuming a linear trend of
230 the experimental data obtained.

231

232

Fig. 5. Adopted compressive behavior of the rammed earth material

233

234 Regarding the tensile behavior, an exponential relationship was taken into account. This relationship
235 is defined by the tensile strength (f_t) and mode-I tensile fracture energy (G_f^I). These parameters were
236 assumed with basis on the calibrated model presented in Silva et al. (2014), from which the values
237 considered for the tensile strength and mode-I tensile fracture energy were 0.05 MPa and 0.074
238 N/mm, respectively. The crack bandwidth was assumed as the square root of the element area (A) to
239 make the numerical outcomes independent from the size of the element.

240 Finally, the density adopted for the rammed earth was of 2000 kg/m³, while a Poisson's ratio of 0.27
241 was assumed considering the calibrated model presented in Silva et al. (2014).

242

243 **2.2.2 TRM strengthening**

244 One of the objectives of the current study is to investigate the TRM strengthening effect on rammed
245 earth walls subjected to in-plane loading. The implementation of this type of strengthening on
246 rammed earth requires adopting a compatible solution, which is being investigated in the framework
247 of the project SafEarth (Barroso 2017; Oliveira et al. 2017 and Sadeghi et al. 2017). This type of
248 strengthening also aims to be affordable in order to facilitate its dissemination, meaning that low cost
249 meshes are being proposed to integrate this composite material. Thus, the selected solution is
250 hereinafter called as low-cost textile reinforced mortar (LC-TRM) and it consists of a low-cost glass
251 fiber mesh embedded in an earth-based mortar, whose characterization of materials and composite
252 behavior is detailed elsewhere (Barroso 2017). In brief, the solution adopted in this study presents the
253 highest values of tensile strength and stiffness among the solutions characterized in the
254 aforementioned study.

255 The outcomes of uniaxial tensile tests on mesh-mortar coupons (Barroso 2017) were used to define
256 the tensile behavior of the adopted LC-TRM strengthening, by averaging the experimental response
257 curves. In compression, the contribution of the mesh was ignored and the average response curve of
258 mortar specimens tested under compression was adopted (Barroso 2017) to simulate the behavior of
259 the LC-TRM. Both tensile and compressive behaviors were simulated using multi-linear relationships,
260 as illustrated in Fig. 6. The tensile behavior is characterized by a trilinear relationship that simulates
261 the three stages typically observed in TRM, namely uncracked, crack development and cracked (see
262 Ascione et al. 2015). The lack of experimental data in the post-peak phase of the mortar tested in
263 compression led also to idealize a linear trend to complete the curve. The TSRCM (DIANA FEA BV
264 2017) was also used to simulate the material behavior of the selected LC-TRM composite. The
265 adopted mechanical properties of the LC-TRM are presented in Fig. 6. The bulk density and
266 Poisson's ratio of the LC-TRM were considered as 1810 kg/m³ and 0.27, respectively.

267 To the knowledge of authors, there is no experimental study available on the performance of TRM-
268 strengthened rammed earth walls subjected to lateral loads. In spite of that, the adopted modelling
269 approach was satisfactorily used in previous studies to predict the seismic performance of TRM-
270 strengthened masonry panels (Basili et al. 2016).

271

272 **Fig. 6.** Adopted stress-strain behavior of the LC-TRM strengthening

273

274 **2.3 Meshing**

275 The modeling of the unstrengthened structural component was evaluated initially by means of two
276 meshing strategies, namely by considering solid and shell elements. Shell elements are widely used in
277 the modeling of masonry structures with the advantage of requiring lower computational demand.
278 However, the considerable thickness of rammed earth walls in comparison to the other dimensions
279 creates doubts on the reliability of shell elements. This concern was evaluated by comparing the use
280 of both element types. It is worth mentioning that the shell models were prepared considering the mid-
281 section planes of each wall, as presented schematically in Fig. 7. This strategy evidently presents
282 limitations, namely with regard to the simulation of the connection between walls (assumed as infinite
283 rigid) and of the correct length of the wing walls (higher lengths are assumed). Furthermore, the
284 overlapping thickness of the walls leads to a wrong consideration of the real self-weight value and
285 mass distribution, and thus of the inertial forces. These limitations are expected to have influence on
286 the response of the models.

287 **Fig. 7.** Schematic view of the shell models

288

289 Generally, three types of elements were adopted for meshing the unstrengthened and strengthened
290 models. For the rammed earth, 20 nodes iso-parametric brick elements (designated by CHX60) were
291 used in the solid strategy, while 8 nodes quadrilateral curved shell elements (denoted as CQ40S) were
292 used for the shell strategy and for meshing the TRM strengthening. Moreover, 8+8 nodes quadrilateral
293 rigid interface elements were adopted for the interface between strengthening and wall (called as
294 CQ48I). These elements are shown in Fig. 8. It should be noted that the default integration scheme
295 $3 \times 3 \times 3$ was used for the solid elements, while the 2×2 scheme was used for the shell ones, where the
296 integration along the thickness considered 7 layers.

297

298 **Fig. 8.** Types of elements employed in the preparation of the models: (a) CHX60 (b) CQ40S (c) CQ48I
299 (DIANA FEA BV 2017)

300 Proper selection of the meshing size is necessary to obtain accurate results and relatively adequate
301 computational times. In this regard, three meshing sizes were tested in the models, namely 25mm
302 (over-meshed), 50mm and 100mm. The accuracy of the selected meshing sizes was evaluated in the
303 unstrengthened model by comparing the outcomes under both self-weight and in-plane pushing. It
304 was observed that the model with the meshing size equal to 100mm results in less than 1% error (both
305 in terms of base shear and displacements) with respect to the over-meshed model. Thus, the 100mm
306 mesh size was used in the subsequent numerical investigation.

307 The models were validated by comparing the obtained reactions under gravity load (self-weight of the
308 walls) with the weight computed with basis on the geometry and density of the rammed earth. The
309 solid models accurately predicted the wall's self-weight, while the shell models result in an error of
310 about 7% due to the previously referred geometric limitations. The influence of the error introduced
311 by the shell modeling approach on the dynamic properties of the rammed earth wall is also evidenced
312 in Fig. 9, which presents the frequency ratios between the shell and solid models for the six first
313 corresponding modes; it should be noted that the natural frequencies depend on the assembled mass
314 and stiffness matrices of the models. The frequency ratios are clearly shown to be smaller than 1 due
315 to the higher mass of the shell model, which results in lower frequency values. Higher modes seem to
316 be more affected.

317

318 **Fig. 9.** Frequency ratios between the shell and solid models for the six first corresponding modes

319

320 **3. Pushover Analyses**

321 This section, presents the results of conventional mass-proportioned nonlinear static analyses (so-
322 called pushover) performed on all considered models. First a sensitivity analysis on the material
323 properties adopted for rammed earth was performed to evaluate their influence on the in-plane
324 behavior. Then, the results of the considered models are discussed with respect to the loading
325 capacity, displacement capacity and failure modes. It should be noted that pushover analysis is widely
326 employed to assess seismic capacities both in research and practice. In spite of a simplified approach
327 with respect to the dynamic nonlinear analyses, pushover was previously shown to reliably predict the
328 average of the responses; however the predicted damage patterns may differ from reality
329 (Allahvirdizadeh and Gholipour 2017).

330 It is worthwhile mentioning that the models were monotonically pushed only in the positive
331 longitudinal (in-plane) direction (+X), since their symmetric geometry leads to similar mechanical
332 results when monotonically pushed in the negative direction (-X); see Fig. 3 for directions.

333

334 **3.1 Sensitivity Analysis**

335 The sensitivity analysis was conducted by considering lower and upper values for the mechanical
336 properties of the rammed earth, in addition to the adopted reference values. These values are reported
337 in Table 1. It is worthwhile to note that the reference values correspond to the previously discussed
338 values in material characterization (see sections 2.2.1 and 2.2.2); while the lower and upper values
339 were obtained by calculating half and double of those reference values, respectively. These wide
340 ranges of values were considered instead of narrow ranges resulting from the lower and upper bounds
341 shown in Fig. 5 to better distinguish the most sensitive parameters. Nevertheless, the adopted ranges
342 are still within the values reported in the literature (Miccoli et al. 2014). In the case of Young's
343 modulus of the multi-linear compression, only the initial slope of the curve was adjusted to obtain
344 desired values without changing compressive strength or idealized post-peak branch. Similarly, the
345 multi-linear curve was scaled with identical initial and post-peak slopes to obtain aimed compressive
346 strength values.

347 For sake of brevity, the outcomes are only presented for the unstrengthened solid model with 50cm
348 long wing walls. The results of the sensitivity analyses are presented in Fig. 10, in terms of the
349 pushover curves (representing the normalized base shear to the weight of the wall as a function of the
350 displacement at the top mid-section of the right wing wall). The compressive strength and the
351 Poisson's ratio seem to have negligible influence on the behavior. The tensile fracture energy also
352 seems to present negligible influence on the loading capacity, though it seems to control the
353 deformation capacity in the post-peak phase. The loading capacity is not significantly affected by
354 changing the Young's modulus, which controls the stiffness of the models, meaning that the variation
355 of this parameter changes significantly the deformation behavior of the model. Among all tested
356 parameters, the tensile strength seems to be the parameter affecting mostly the in-plane behavior of
357 the model. As it is clear, the tensile strength controls both load and displacement capacities. For
358 instance, doubling or halving the tensile strength, results in about 50% increase or decrease in lateral
359 load capacity of the component, respectively. Furthermore, the in-plane shear failure of the rammed
360 earth component is demonstrated to be mainly governed by cracking damage.

361

362 **Table 1.** Parameters considered in the sensitivity analysis

363
364

(a) (b)
(c) (d)
(e)

365 **Fig. 10.** Pushover curves obtained from the sensitivity analyses of the unstrengthened solid model with 50cm
366 wing walls: (a) Compressive strength (b) Poisson's ratio (c) Young modulus (d) Tensile strength (e) Tensile
367 fracture energy

368 3.2 Unstrengthened Models

369 In addition to the evaluation of the in-plane behavior of rammed earth components, the pushover
370 analyses of the unstrengthened models allowed to conclude about the modeling approach (i.e. shell or
371 solid) showing the best compromise between accuracy of results and computational effort.

372 The pushover curves of the models are portrayed in Fig. 11, which the lateral displacement of three
373 nodes, namely on top of the left and right wings, and on top of the middle section of the web were
374 considered. In all cases, the right wing (the wing which was leaned on during the push) controls the
375 behavior. Regarding the meshing approach, the lateral displacements in the shell models are greater
376 than those of the solid ones. Nevertheless, a minor increase in peak capacity is observed from the shell
377 to the solid models.

378 The point of damage initiation of the models is also highlighted in the curves, which corresponds to
379 the onset of the cracks' opening. As it can be seen, this state occurs for very low values of the
380 imposed lateral loading, evidencing the great influence of the nonlinear behavior of the rammed earth
381 on the structural behavior.

382

(a) (b)
(c) (d)

383 **Fig. 11.** Pushover curves of the unstrengthened models: (a) Shell model with 80 cm wings (b) Solid model with
384 80 cm wings (c) Shell model with 50 cm wings (d) Solid model with 50 cm wings

385

386 The models with 50 cm wings achieved higher load and displacement capacities than those of the
387 models with 80 cm wings. It is also true that the damage initiation occurs earlier for the models with
388 80 cm wings. This difference in behavior is explained by the influence of the out-of-plane bending of
389 wings on the response, where the higher their length the earlier is the damage initiation due to tension
390 cracking. Thus, a response of the models governed by the in-plane behavior of the web wall is very
391 unlikely to be the dominant failure mode in the models with 80 cm wings, since these walls induce
392 high bending stresses. These aspects are later discussed by investigating developed strains/stresses.
393 Furthermore, it should be noted that experimental models with similar geometry to that of the
394 numerical models are planned to be experimentally tested in near future, and that due to stability
395 concerns during the tests the wings cannot be eliminated. Therefore, it is of utmost interest to find the
396 dimensions that satisfy not only the experimental concerns, but also represent the desired lateral
397 behavior of the rammed earth walls.

398 With respect to the failure modes evidenced by the models, when the response is considered by the
399 left wing and the mid web nodes, it is possible to observe that an apparent unloading occurs after
400 reaching the peak load. This situation can be explained by the possible detachment between the right

401 wing and the web wall. Such detachment increases displacements on the right wing, whereas the left
402 wing and the web unload. It is clear that the sway of the right wing cannot be interpreted entirely as
403 ductility of the model.

404 The contour maps of the total lateral displacements in X-direction (in-plane) at the peak capacity of
405 the models are shown in Fig. 12. As it can be seen, the shell models experienced higher lateral
406 displacements at the right wing. This behavior is a consequence of the disregarded thickness of the
407 web, where the supporting effect is not simulated in its full extension, meaning that the wings are
408 considered with a longer effective length and are more easily bended. Thus, it can be stated that the
409 thickness disregarded of the shell models may lead to the prediction of unreliable failure mechanisms
410 and capacities.

411

(a) (b) (c) (d)
412 **Fig. 12.** Total lateral displacements at the peak capacity of the unstrengthened models: (a) shell model with
413 80cm long wings (b) solid model with 80cm long wings (c) shell model with 50cm long wings (d) solid model
414 with 50cm long wings

415

416 To assess the load paths through the models and highlight the regions with damage concentration, the
417 principal tensile strains were also analyzed. The respective contours are presented in Fig. 13. The
418 connection of the web and the right wing is the region with the highest values of tensile strains,
419 indicating that this region is more likely to control the response of the in-plane models and to
420 concentrate the cracking process. The difference between solid and shell models is evident, namely
421 with respect to the distribution of damage in the web of the shell model near the right wing. On the
422 other hand, no diagonal cracks are detected in the model with 80 cm wings, showing the absence of
423 the shear failure of the web. In the case of the model with 50 cm wings, the formation of diagonal
424 cracks is evident, even though not in its full extension, meaning that this model is more representative
425 of the expected behavior for the experimental models. Given the above discussions, only the solid
426 model with 50cm wing length will be considered in the subsequent numerical investigation.

427

(a) (b) (c) (d)
428 **Fig. 13.** Principal tensile strains at the peak capacity of the unstrengthened models: (a) shell model with 80cm
429 long wings (b) solid model with 80cm long wings (c) shell model with 50cm long wings (d) solid model with
430 50cm long wings

431

432 Furthermore, the influence of transversal (wing) walls on the observed failure mode is addressed by
433 conducting pushover analysis on a model without wing walls. By considering the control node on the
434 middle section of the web wall the obtained pushover curve is shown in Fig. 14a. As it can be seen,

435 the existence of transversal walls has a considerable influence on the in-plane load capacity of the
436 rammed earth wall (10% reduction); although, their effect on the displacement capacity is much more
437 evident. The occurred failure modes were investigated by comparing the principal tensile strains of
438 both cases, as presented in Fig. 14b and c. As it is evident, the dominant failure mode is changed
439 from detachment of wing walls, in the component with wing walls, to sliding/rocking in the model
440 without wing walls.

441

442 (a) (b) (c)
443 **Fig. 14.** Influence of wing walls on performance of the rammed earth component: (a) pushover curve (b)
444 principal tensile strains of the unstrengthened rammed earth wall with 50cm long wing walls (c) principal tensile
445 strains of the unstrengthened rammed earth wall without wing walls

445

446 3.3 Strengthened Models

447

448 The LC-TRM strengthening was applied continuously on all vertical surfaces of the model, which
449 corresponds to the situation that is expected to grant the maximum improvement capacity. The
450 pushover curves of the strengthened model are compared with those of the unstrengthened one in Fig.
451 15. The strengthening slightly increased the lateral stiffness and increased considerably the loading
452 and displacement capacities of the component. Despite that, the right wing still sways, meaning that
453 the failure mode did not changed from the unstrengthened model to the strengthened one. By
454 considering the control node on the right wing, the lateral displacement and load capacities of the
455 strengthened model increased approximately 90% and 21%, respectively. Nevertheless, the
456 detachment of the right wing from the web makes the displacement of the control node on the middle
457 section of the web a global indicator of the displacement capacity improvement introduced by the
458 strengthening. With respect to this control node, a 57% increase in the lateral displacement was
459 observed. Regarding the damage initiation point, also highlighted in Fig. 15, no difference was
460 detected with respected to the unstrengthened model. As previously discussed, this point corresponds
461 to localized damage occurrence, thus the onset of the damage in the strengthened model is identical to
462 that of the unstrengthened model.

463

464 **Fig. 15.** Pushover curves of the strengthened model

465

466 The contour maps of displacements were also investigated to understand the failure mechanism of the
467 strengthened model (see Fig. 16). By comparing the experienced lateral displacements of the
468 unstrengthened and strengthened models at the load factor equal to the peak capacity of the

469 unstrengthened model, it is observed that the strengthened model presents lower deformations, due to
470 increase in the lateral stiffness and a probable better stress distribution capacity. On the other hand,
471 the contour of the strengthened model at its peak capacity shows important deformations at the right
472 wing and in the region of its connection with the web. Therefore, it can be concluded that the failure
473 mechanism of the strengthened model is also governed by detachment of the right wing. Furthermore,
474 the adopted LC-TRM strengthening solution is shown to be efficient on postponing this failure mode.
475

(a) (b) (c)
476 **Fig. 16.** Total lateral displacements of the strengthened model in comparison to the plain one: (a)
477 unstrengthened model at its peak capacity (b) strengthened model at the peak capacity of the unstrengthened
478 model (c) strengthened model at its peak capacity

479

480 Additionally, the applied strengthening solution was expected to increase integrity of the wall by
481 promoting the redistribution stresses and decreasing stress concentration in the most vulnerable
482 regions, as evidenced in the contour maps of the principal tensile strains presented in Fig. 17. A
483 diagonal strut (shear crack) was observed to form at the web of the unstrengthened model at its peak
484 capacity, while this type of damage did not occur in the strengthened model at this point. This
485 situation is due to the increased capacity promoted by the LC-TRM composite and by its contribution
486 in transferring the tensile stresses. At this stage, the detachment between the right wing and the web is
487 completely prevented. Only a small damage in the toe of the left wing was observed, evidencing the
488 tendency of the wall to overturn. It can be also seen that the strengthened model experiences smaller
489 strains in this region in comparison with the unstrengthened model. The principal tensile strains at the
490 peak capacity of the strengthened model show an important detachment of the right wing, despite a
491 portion of the web following the wing. From the kinematic point of view, this added portion means
492 that a greater load is required to cause the right wing to detach from the wall and overturn. Moreover,
493 a diagonal shear crack was observed in the web, whose development is much more expressive than
494 that evidenced in unstrengthened one. This developed diagonal shear crack illustrates the mechanical
495 efficiency of the adopted strengthening solution in improving the in-plane shear behavior of the
496 rammed earth component.

497 On the other hand, it is also important to investigate the damage state of the strengthening. In this
498 regard, the contour of the principal tensile strains at the peak capacity of the strengthened model is
499 presented in Fig. 17d. It clearly shows the working mode of the strengthening solution. In other
500 words, the efficient strengthening technique should mostly work in regions likely to fail without
501 reinforcement, namely at the connection of the right wing with the web and at the diagonal of the
502 web. Thus, it is comprehensible that considerable tensile strains developed at the strengthening
503 adjacent to the right wing, which were responsible to postpone the detachment.

504

505 (a) (b) (c) (d)
506 **Fig. 17.** Principal tensile strains of the strengthened model in comparison to the unstrengthened one: (a)
507 unstrengthened model at its peak capacity (b) strengthened model at the peak capacity of the unstrengthened
508 model (c) strengthened model at its peak capacity (d) LC-TRM strengthening at peak capacity of the
strengthened model

509

510 **4. Influence of the Damage on Dynamic Behavior (Modal Analysis)**

511

512 The initiation and development of cracks (damage) under monotonically increasing lateral loads cause
513 the stiffness of a structure to decrease. As a consequence, the dynamic properties (i.e. frequencies and
514 mode shapes) of the damaged models change as well. Such changes can be employed to evaluate and
515 monitor in a simple way the damage evolution during the pushover analyses. To this purpose,
516 stepwise modal analyses were conducted on both unstrengthened and strengthened models during the
517 pushover analyses presented above.

518 The initial (undamaged) mode shapes, periods and cumulative effective mass participation (*CEM*) of
519 four highest contributing modes in each principal direction (X and Y) of both models are reported in
520 Table 2. As it can be seen, the introduction of the strengthening did not change the mode shapes of the
521 component, since it introduced minor influence on the mass and stiffness. However, a slight increase
522 on the *CEM* and a slight reduction in periods can be distinguished.

523 The damage development in the models was evaluated by normalizing the frequency values obtained
524 from different lateral loading levels (imposed during the pushover analyses) to the initial values. It is
525 worthwhile to note that the frequencies at each imposed lateral displacement were obtained by
526 running a modal analysis at the corresponding step considering the updated stiffness matrix. This
527 frequency ratio was only determined up to the loading capacity of the models and is plotted as
528 function of the displacement at the middle node on the top of the web wall (see Fig. 18). In general,
529 the frequencies of the unstrengthened model present an exponential decrease with increasing
530 displacement, though this reduction is smoother in the case of the strengthened model. At the loading
531 capacity of the unstrengthened model, the frequency decreased about 11% (average of considered
532 modes), while in the case of the strengthened model the decrease was of about 6% for the same
533 corresponding displacement level. This lower decrease of the frequency ratio of the strengthened
534 model with respect to the unstrengthened one means that the LC-TRM strengthening is able to reduce
535 the level of damage of the component for equivalent levels of deformation.

536

537 **Table 2.** Initial (undamaged) dynamic properties of the unstrengthened and strengthened models
538

539 (a) (b)
540 (c) (d)
Fig. 18. Damage evolution based on the frequency ratio of the highest participating modes: (a) Mode 1 (b)
541 Mode 4 (c) Mode 8 (d) Mode 10

542 **5. Nonlinear Dynamic Analyses**

543 Employing nonlinear static analyses (pushover) may lead to acceptable results of the dynamic
544 response of existing structures (Allahvirdizadeh et al. 2017); however, the predicted damage can
545 significantly differ from the more robust nonlinear dynamic analyses (Allahvirdizadeh and Gholipour
546 2017). Thus, the applicability and reliability of the pushover analyses in assessing the seismic
547 performance of rammed earth components was evaluated by comparison with the results from
548 nonlinear time-history analyses. In this regard, incremental dynamic analyses, so-called IDA
549 (Vamvatsikos and Cornell 2002) were conducted by applying a series of linearly scaled ground
550 motion records.

551 The outcomes obtained from IDA can be interpreted from two perspectives, namely force-based or
552 displacement-based. In the former, the maximum experienced lateral force (i.e. the intensity of the
553 applied ground motion record) and corresponding displacement are extracted from each nonlinear
554 dynamic analysis, while the latter seeks for the maximum experienced lateral displacement and
555 corresponding lateral force. In general, a reliable pushover prediction should lie down within the
556 boundaries defined by the aforementioned perspectives. Thus, an identical approach is here presented
557 with respect to both unstrengthened and strengthened models.

558 The outcomes from dynamic analyses depend on the applied ground motion record, meaning that it
559 should be properly defined. The source of that record can be either instrumental (recorded from
560 previously occurred earthquakes) or synthetically generated. Each of these methods can induce a level
561 of uncertainty to the obtained outcomes, though this topic is beyond the scope of this study (for details
562 see Watson-Lamprey 2007; Haselton et al. 2009; Wang 2011; Allahvirdizadeh et al. 2013).

563 Regarding the IDA performed on the models, an artificial generated ground motion record was
564 adopted. The ground motion was generated taking into account the seismicity conditions of Odemira
565 (Alentejo region, southern Portugal) for the near-field earthquake, as established in the Portuguese
566 national annex of Eurocode 8 (IPQ 2010). Simqke-gr software (Simqke_gr 2012), was used to
567 generate a ground motion record compatible with the design spectrum. Subsequently, a baseline
568 correction and a filtering of the frequencies outside of the range 0.1-20 Hz were performed by means
569 of the SeismoSignal software (Seismosoft 2016). The spectrum of the generated record is compared

570 with the design spectra in Fig. 19. The modes with the highest mass contribution (see Table 2) are
571 also presented in Fig. 19 and reveal the sensitivity of the rammed earth component to earthquakes
572 with high frequencies.

573

574 **Fig. 19.** Generated ground motion record in comparison with the design spectrum (near-field earthquake of
575 Odemira region)

576

577 In addition to the ground motion record, it is vital to define a proper damping ratio of the system to
578 take into account the energy dissipation. In this regard, the Rayleigh viscous damping approach was
579 adopted (Chopra 2012). It should be noted that there is no general consensus about the damping ratio
580 value in rammed earth constructions, particularly when running non-linear dynamic analyses. Hence,
581 a 3% damping ratio was considered.

582 The IDA was performed by linearly scaling the generated ground motion a series of times until
583 numerical instability started to be observed. The resulting scaled ground motions were applied to the
584 models in the longitudinal direction (X direction in Fig. 3). Then, the hysteretic curves of each
585 analysis, representing the normalized base shear (load factor) as function of the experienced lateral
586 displacement, were used to extract the envelop curves (Fig. 20a). Finally, the points of maximum
587 experienced force and displacements at both positive and negative directions were extracted to plot
588 the force- and displacement-based IDA curves. The resulting IDA curves of the unstrengthened and
589 strengthened models are presented respectively in Fig. 20 b and c, where they are also compared with
590 the corresponding pushover curve. In general, the pushover analysis seems to accurately predict both
591 the load and displacement capacities of the models with respect to the IDA.

592

(a)

(b)

(c)

593 **Fig. 20.** Outcomes of the nonlinear dynamic analyses: (a) example of hysteretic curve envelop (b)
594 displacement- and force-based IDA curves of the plain model (c) displacement- and force-based IDA curves of
595 the strengthened model

596

597 Regarding the damage observed in the IDA, Fig. 21 present the contour maps of the maximum values
598 of the principal tensile strains experienced by the models when subjected to the ground motion with
599 the highest intensity. Again, the applied LC-TRM is shown not to change the failure mode, which is
600 composed of shear cracking in the web and detachment of wing walls. Furthermore, the comparison
601 of these contour maps with those presented in Fig. 17 reveal that, in general, the damage predicted by
602 the pushover analyses agrees with that of the nonlinear dynamic analysis. Nevertheless, the damage

603 observed in the web wall due to sway of wing walls in the unstrengthened model is not correctly
604 portrayed by pushover.

605

606 **Fig. 21.** Maximum values of the principal tensile strains of the models observed for the highest intensity ground
607 motion: (a) unstrengthened model (b) strengthened model

608

609 **6. Conclusions**

610 The current study investigated the in-plane seismic performance of rammed earth walls by means of
611 advanced nonlinear finite element modeling. The main remarks are highlighted as follows:

612 - The conducted sensitivity analyses have shown that parameters other than tensile strength have
613 minor influence on the load and displacement capacities of the numerical models. It was noticed
614 that doubling or halving the tensile strength results in 50% increase or reduction of the load
615 capacity, respectively.

616 - Damage (cracking) at both unstrengthened and strengthened models initiates at very low lateral
617 load levels, though due to its local occurrence, the behavior of the wall seems to remain elastic
618 up to higher load levels.

619 - The comparison between models based on solid elements and those based on shell elements
620 revealed that the latter experience higher lateral displacements due to disregarding of the
621 thickness of the walls. Furthermore, the shell based models were shown to not allow a correct
622 prediction of damage. Thus, the use of solid elements is recommended in the modeling of thick
623 rammed earth walls, like the ones from typical Portuguese dwellings.

624 - The models with short wing walls achieved higher load and displacement capacities. Moreover,
625 failure due to shear cracking of the web wall is more likely to occur in this component.

626 - The sections in the unstrengthened model deemed as the most critical are the connections
627 between web and wing walls, despite the observation of some diagonal cracks in the web wall.
628 Thus, detachment of the wing walls is the most likely failure mode of the unstrengthened
629 rammed earth component when subjected to in-plane loading.

630 - The LC-TRM strengthening increased the loading and displacement capacities of the
631 unstrengthened model in about 21% and 56%, respectively.

632 - The LC-TRM strengthening does not change the failure mode of the rammed earth component;
633 nevertheless it postpones failure by assuring a better stress distribution in the critical sections.

634 - The decrease of the frequency ratio of the highest contributing modes was used as a damage
635 indicator of the pushover analyses and it allowed observing that the LC-TRM strengthening
636 decrease this indicator from 11% to 6%.

- 637 - The comparison of the outcomes of IDA with those of pushover analyses revealed that pushover
638 analysis can reliably predict both the in-plane loading and displacement capacities of the rammed
639 earth models.
- 640 - The damage evidenced from IDA and pushover analyses portrayed identical failure modes,
641 nevertheless the damage distribution is not properly identical due to the dynamic nature of the
642 loading in the IDA.

643

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649

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788 **Fig. 1.** Rammed earth constructions in Portugal: (a) Alentejo region (in red); (b) examples of typical dwellings
789 (Silva et al. 2018b)

790 **Fig. 2.** Average length and height of the rammed earth walls identified in the surveyed rammed earth dwellings

791 **Fig. 3.** Considered in-plane models: (a) 50 cm long wing walls (b) 80 cm long wing walls

792 **Fig. 4.** Detailed view of the materials and interaction levels contributing for the structural behavior of the
793 unstrengthened and strengthened models

794 **Fig. 5.** Adopted compressive behavior of the rammed earth material

795 **Fig. 6.** Adopted stress-strain behavior of the LC-TRM strengthening

796 **Fig. 7.** Schematic view of the shell models

797 **Fig. 8.** Types of elements employed in the preparation of the models: (a) CHX60 (b) CQ40S (c) CQ48I
798 (DIANA FEA BV 2017)

799 **Fig. 9.** Frequency ratios between the shell and solid models for the six first corresponding modes

800 **Fig. 10.** Pushover curves obtained from the sensitivity analyses of the unstrengthened solid model with 50cm
801 wing walls: (a) Compressive strength (b) Poisson's ratio (c) Young modulus (d) Tensile strength (e) Tensile
802 fracture energy

803 **Fig. 11.** Pushover curves of the unstrengthened models: (a) Shell model with 80 cm wings (b) Solid model with
804 80 cm wings (c) Shell model with 50 cm wings (d) Solid model with 50 cm wings

805 **Fig. 12.** Total lateral displacements at the peak capacity of the unstrengthened models: (a) shell model with
806 80cm long wings (b) solid model with 80cm long wings (c) shell model with 50cm long wings (d) solid model
807 with 50cm long wings

808 **Fig. 13.** Principal tensile strains at the peak capacity of the unstrengthened models: (a) shell model with 80cm
809 long wings (b) solid model with 80cm long wings (c) shell model with 50cm long wings (d) solid model with
810 50cm long wings

811 **Fig. 14.** Influence of wing walls on performance of the rammed earth component: (a) pushover curve (b)
812 principal tensile strains of the unstrengthened rammed earth wall with 50cm long wing walls (c) principal tensile
813 strains of the unstrengthened rammed earth wall without wing walls

814 **Fig. 15.** Pushover curves of the strengthened model

815 **Fig. 16.** Total lateral displacements of the strengthened model in comparison to the plain one: (a)
816 unstrengthened model at its peak capacity (b) strengthened model at the peak capacity of the unstrengthened
817 model (c) strengthened model at its peak capacity

818 **Fig. 17.** Principal tensile strains of the strengthened model in comparison to the unstrengthened one: (a)
819 unstrengthened model at its peak capacity (b) strengthened model at the peak capacity of the unstrengthened

820 model (c) strengthened model at its peak capacity (d) LC-TRM strengthening at peak capacity of the
821 strengthened model

822 **Fig. 18.** Damage evolution based on the frequency ratio of the highest participating modes: (a) Mode 1 (b)
823 Mode 4 (c) Mode 8 (d) Mode 10

824 **Fig. 19.** Generated ground motion record in comparison with the design spectrum (near-field earthquake of
825 Odemira region)

826 **Fig. 20.** Outcomes of the nonlinear dynamic analyses: (a) example of hysteretic curve envelop (b) displacement-
827 and force-based IDA curves of the plain model (c) displacement- and force-based IDA curves of the
828 strengthened model

829 **Fig. 21.** Maximum values of the principal tensile strains of the models observed for the highest intensity ground
830 motion: (a) unstrengthened model (b) strengthened model

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850 **Table 1.** Parameters considered in the sensitivity analysis

851 **Table 2.** Initial (undamaged) dynamic properties of the unstrengthened and strengthened models

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(a)



(b)

876 **Fig. 1.** Rammed earth constructions in Portugal: (a) Alentejo region (in red); (b) examples of typical dwellings
877 (Silva et al. 2018b)

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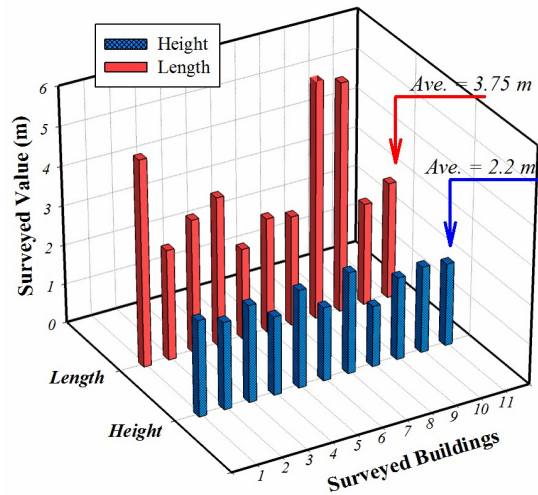
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891 **Fig. 2.** Average length and height of the rammed earth walls identified in the surveyed rammed earth dwellings

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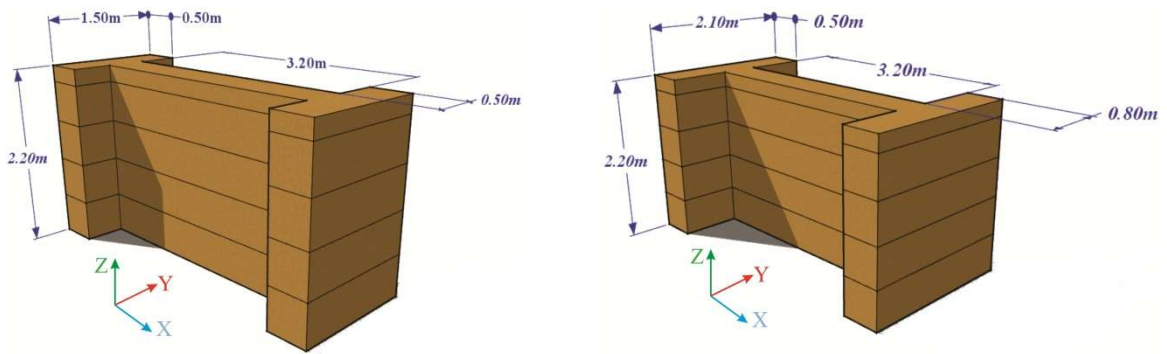
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(a)

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Fig. 3. Considered in-plane models: (a) 50 cm long wing walls (b) 80 cm long wing walls

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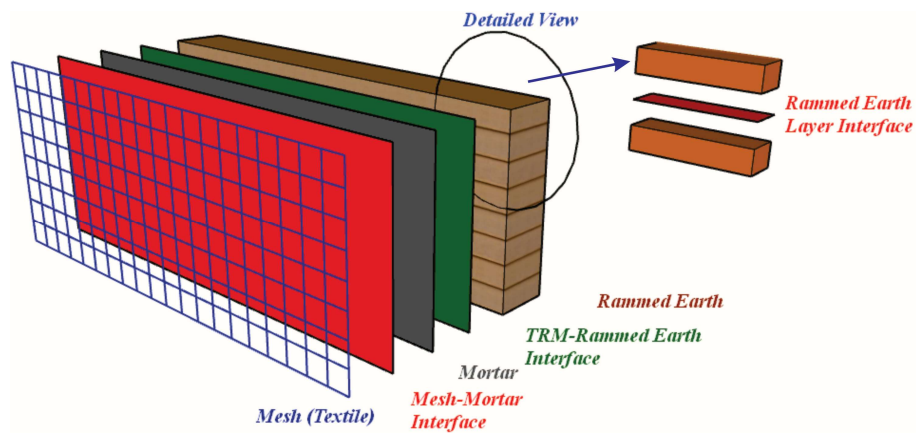
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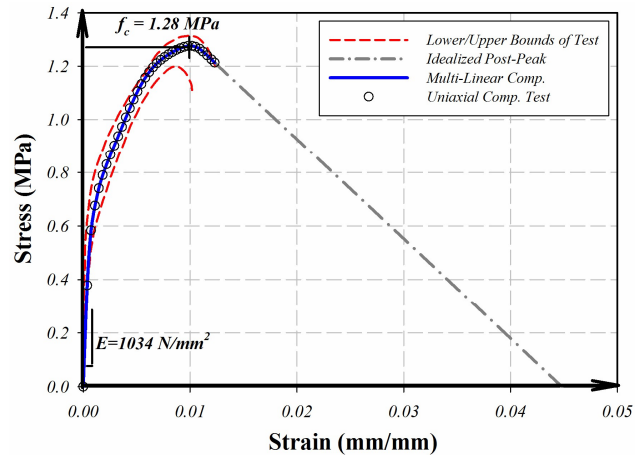
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Fig. 4. Detailed view of the materials and interaction levels contributing for the structural behavior of the unstrengthened and strengthened models



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Fig. 5. Adopted compressive behavior of the rammed earth material

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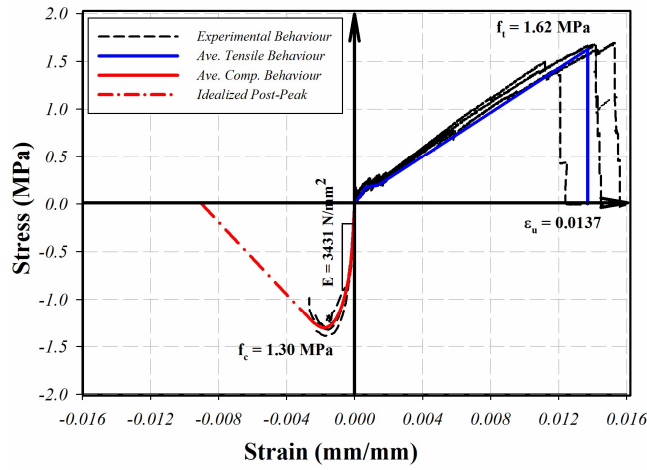
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Fig. 6. Adopted stress-strain behavior of the LC-TRM strengthening

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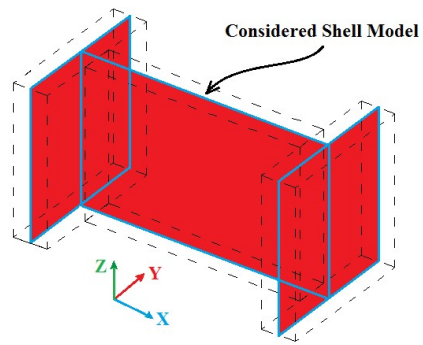
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Fig. 7. Schematic view of the shell models

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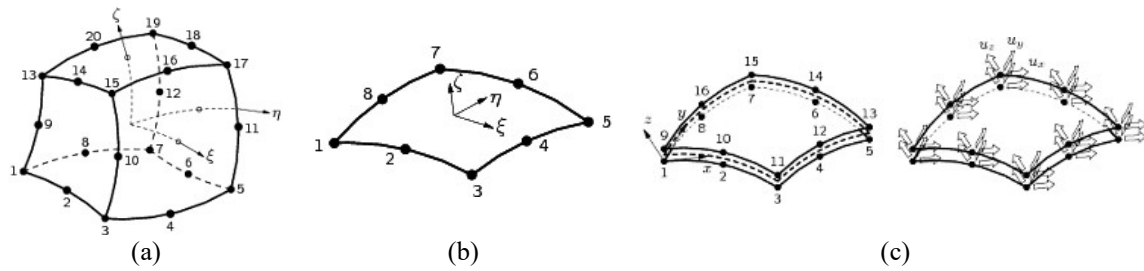
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997 **Fig. 8.** Types of elements employed in the preparation of the models: (a) CHX60 (b) CQ40S (c) CQ48I
998 (DIANA FEA BV 2017)

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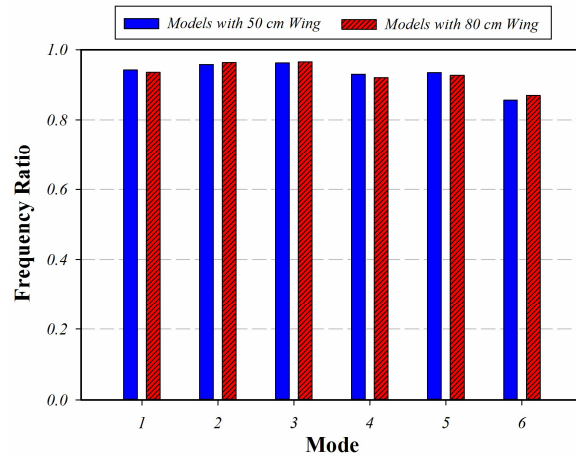
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Fig. 9. Frequency ratios between the shell and solid models for the six first corresponding modes

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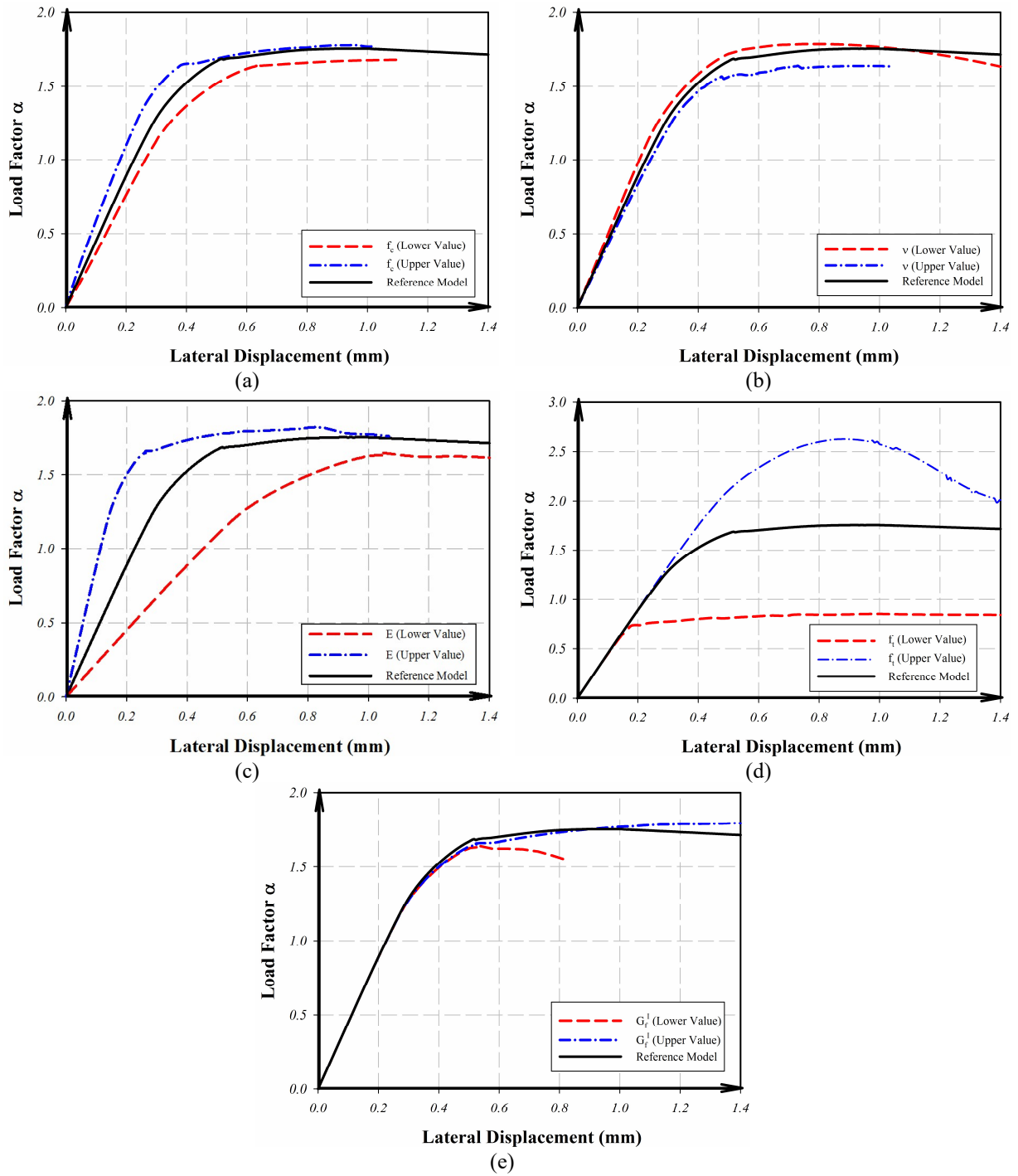


Fig. 10. Pushover curves obtained from the sensitivity analyses of the unstrengthened solid model with 50cm wing walls: (a) Compressive strength (b) Poisson's ratio (c) Young modulus (d) Tensile strength (e) Tensile fracture energy

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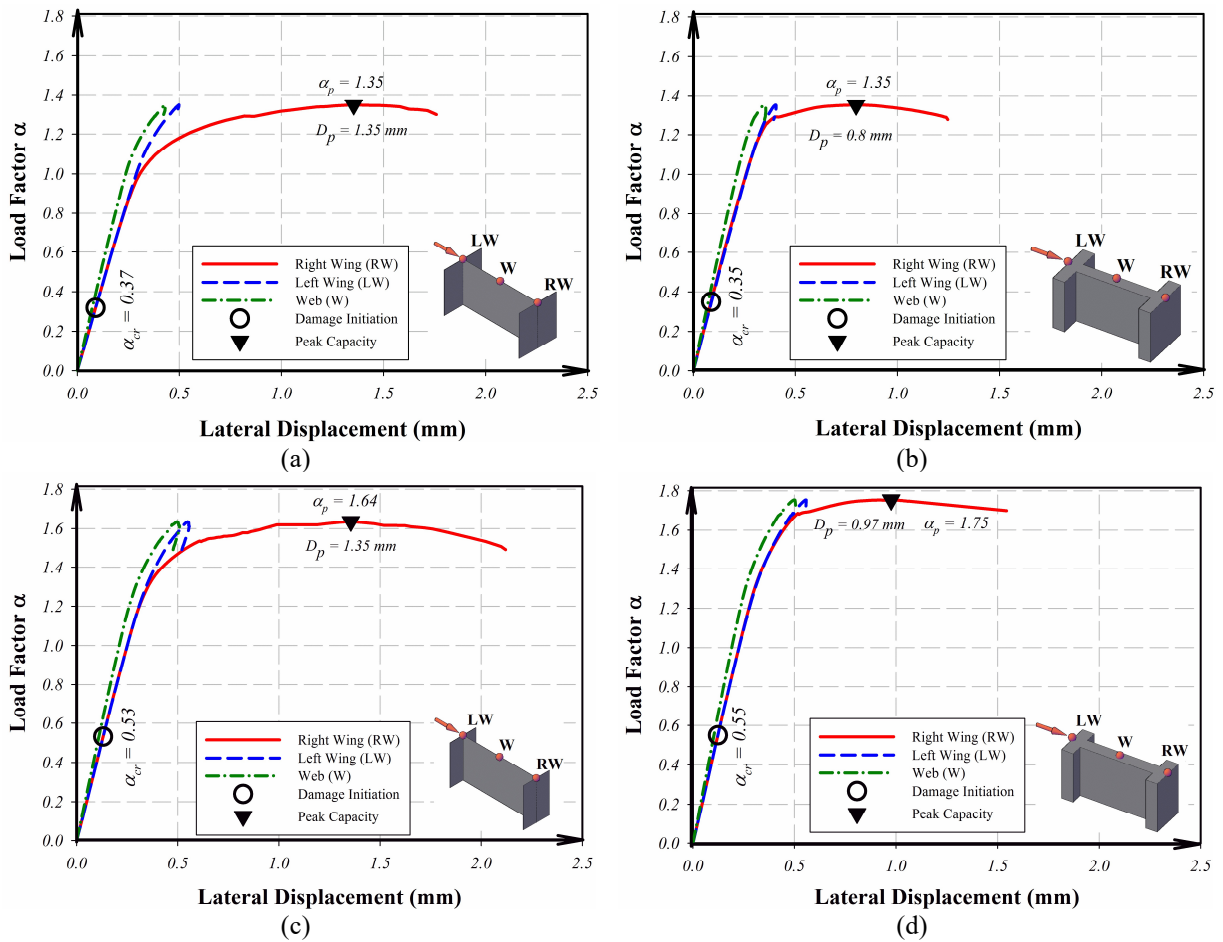


Fig. 11. Pushover curves of the unstrengthened models: (a) Shell model with 80 cm wings (b) Solid model with 80 cm wings (c) Shell model with 50 cm wings (d) Solid model with 50 cm wings

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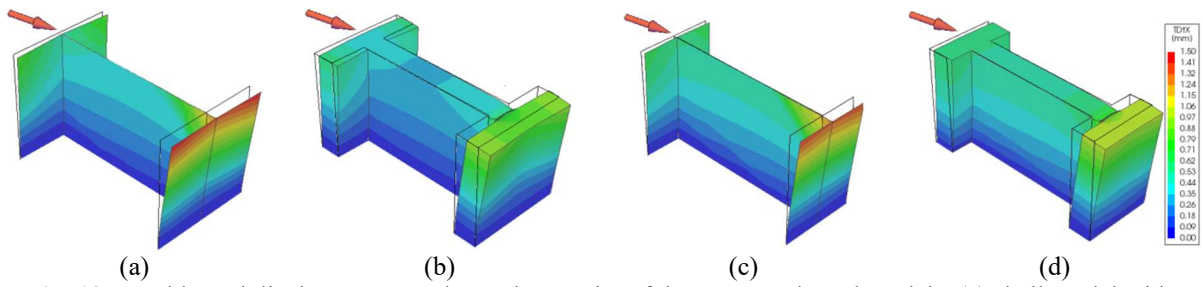
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Fig. 12. Total lateral displacements at the peak capacity of the unstrengthened models: (a) shell model with 80cm long wings (b) solid model with 80cm long wings (c) shell model with 50cm long wings (d) solid model with 50cm long wings

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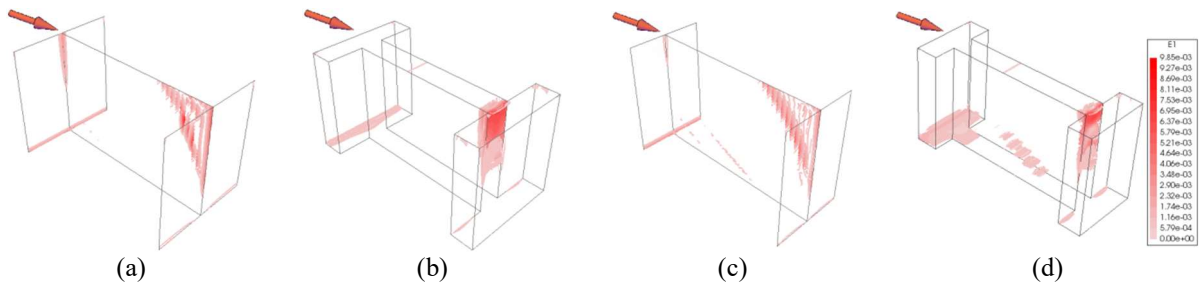
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1072 **Fig. 13.** Principal tensile strains at the peak capacity of the unstrengthened models: (a) shell model with 80cm
1073 long wings (b) solid model with 80cm long wings (c) shell model with 50cm long wings (d) solid model with
1074 50cm long wings

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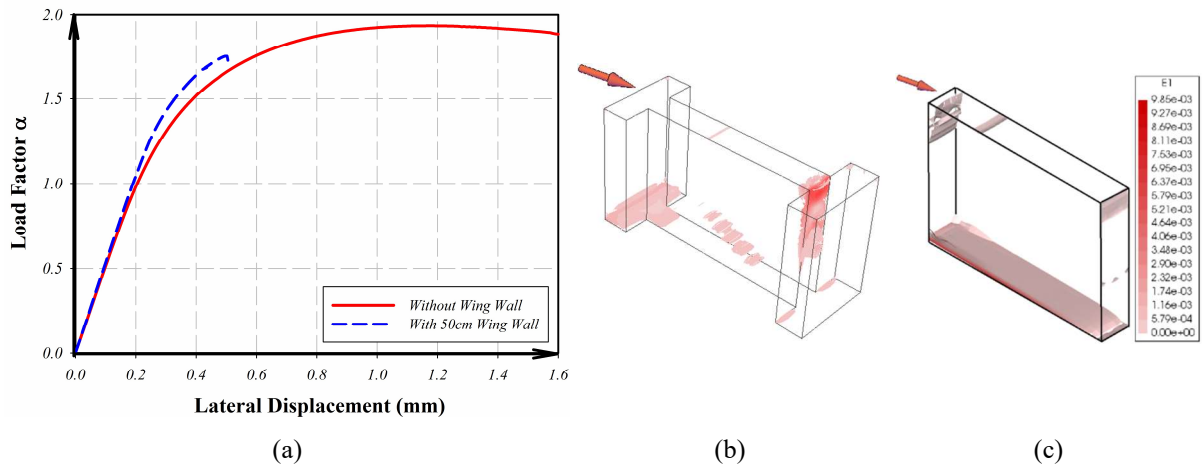
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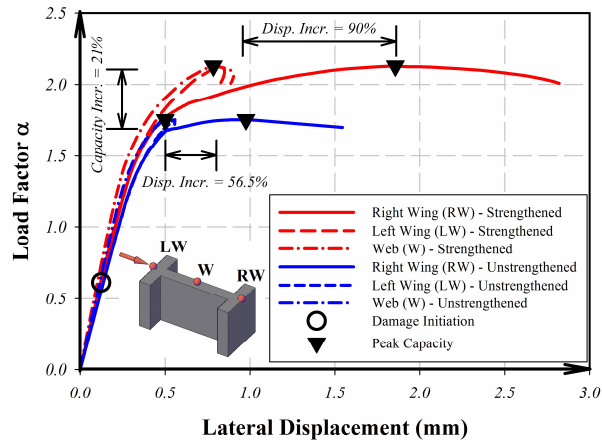
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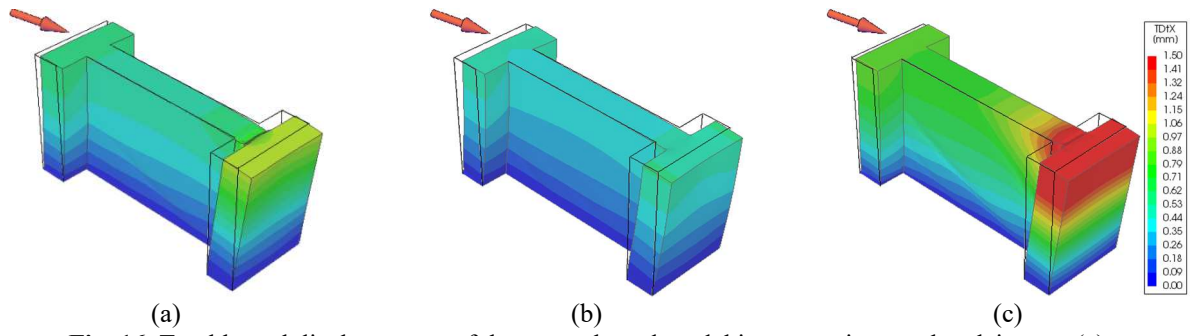
1090 **Fig. 14.** Influence of wing walls on performance of the rammed earth component: (a) pushover curve (b)
1091 principal tensile strains of the unstrengthened rammed earth wall with 50cm long wing walls (c) principal
1092 strains of the unstrengthened rammed earth wall without wing walls

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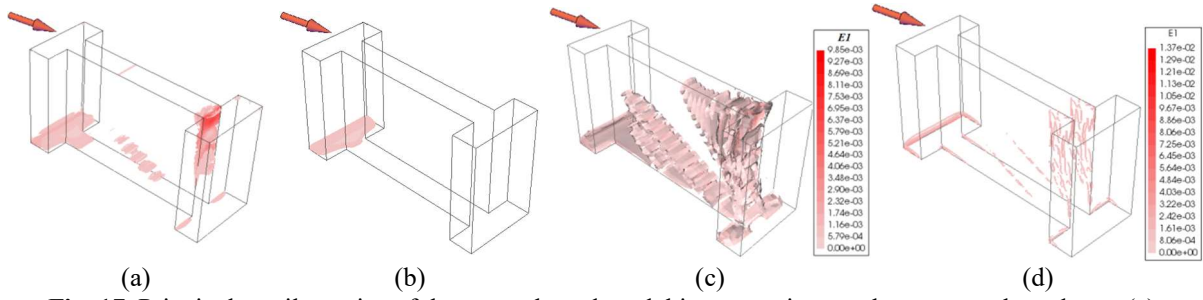
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Fig. 15. Pushover curves of the strengthened model



1129 **Fig. 16.** Total lateral displacements of the strengthened model in comparison to the plain one: (a)
1130 unstrengthened model at its peak capacity (b) strengthened model at the peak capacity of the unstrengthened
1131 model (c) strengthened model at its peak capacity

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1149 **Fig. 17.** Principal tensile strains of the strengthened model in comparison to the unstrengthened one: (a)
1150 unstrengthened model at its peak capacity (b) strengthened model at the peak capacity of the unstrengthened
1151 model (c) strengthened model at its peak capacity (d) LC-TRM strengthening at peak capacity of the
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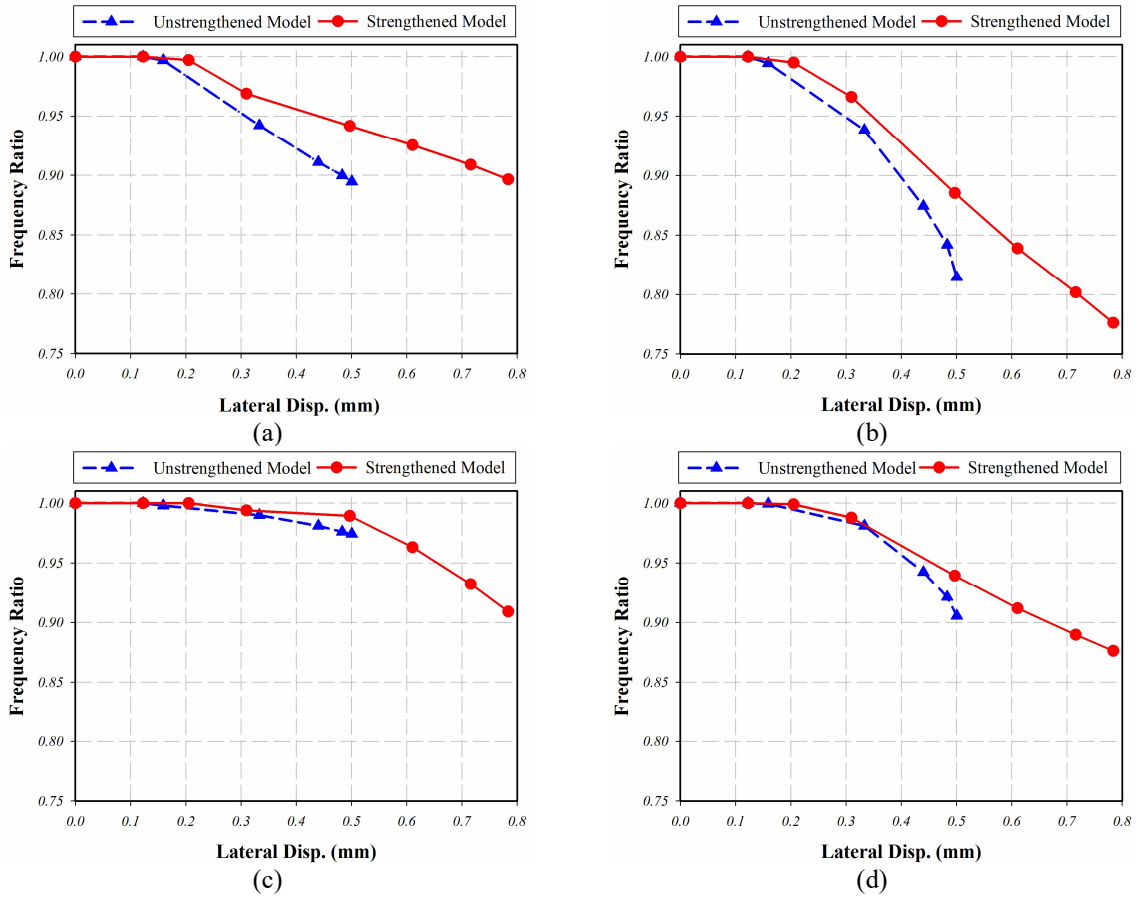
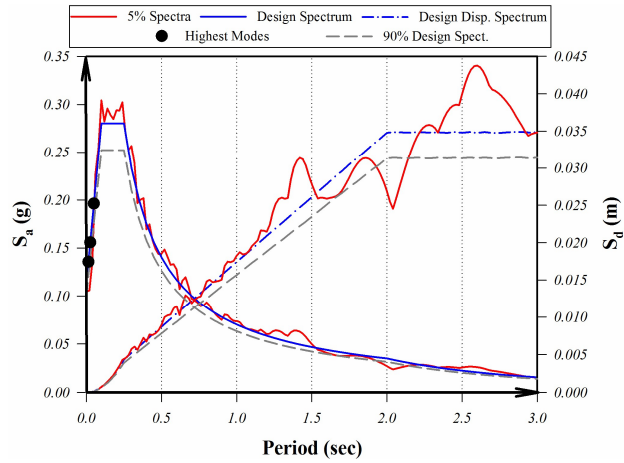


Fig. 18. Damage evolution based on the frequency ratio of the highest participating modes: (a) Mode 1 (b) Mode 4 (c) Mode 8 (d) Mode 10

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Fig. 19. Generated ground motion record in comparison with the design spectrum (near-field earthquake of Odemira region)

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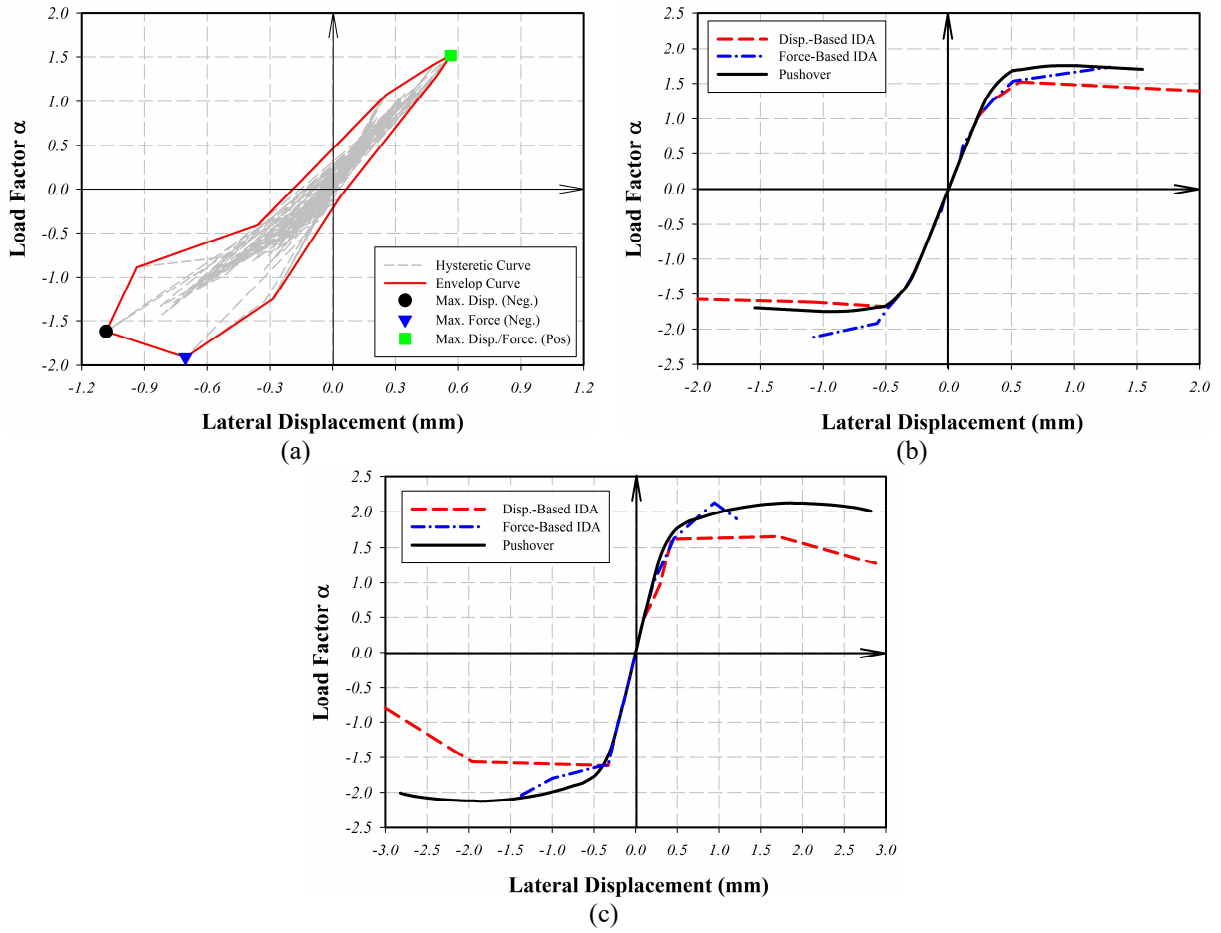
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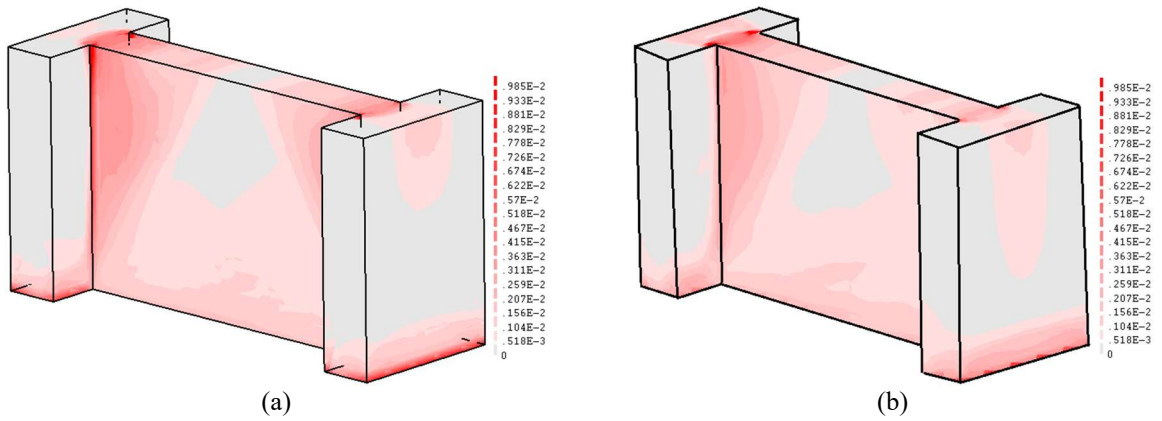
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1205 **Fig. 20.** Outcomes of the nonlinear dynamic analyses: (a) example of hysteretic curve envelop (b) displacement-
1206 and force-based IDA curves of the plain model (c) displacement- and force-based IDA curves of the
1207 strengthened model

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1220 **Fig. 21.** Maximum values of the principal tensile strains of the models observed for the highest intensity ground
1221 motion: (a) unstrengthened model (b) strengthened model

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Table 1. Parameters considered in the sensitivity analysis

Parameter	Reference Value	Lower Value	Upper Value
Compressive Strength	$f_c = 1.28$ MPa	$0.5f_c = 0.64$ MPa	$2.0f_c = 2.56$ MPa
Poisson's Ratio	$\nu_{ref} = 0.27$	$\nu_{lower} = 0.1$	$\nu_{upper} = 0.4$
Young Modulus	$E = 1034$ MPa	$0.5E = 517$ MPa	$2.0E = 2068$ MPa
Tensile Strength	$f_t = 0.05$ MPa	$0.5f_t = 0.025$ MPa	$2.0f_t = 0.1$ MPa
Tensile Fracture Energy	$G_f^I = 0.074$ N/mm	$0.5G_f^I = 0.037$ N/mm	$2.0G_f^I = 0.148$ N/mm

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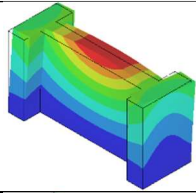
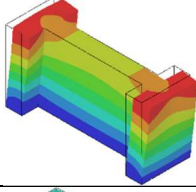
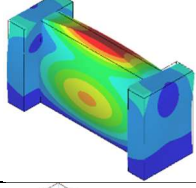
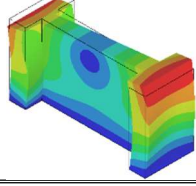
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Table 2. Initial (undamaged) dynamic properties of the unstrengthened and strengthened models

Mode	Mode Shape	Unstrengthened Model	Strengthened Model
1		$f_1 = 19.9 \text{ Hz}$ $CEM_x = 0.0\%$ $CEM_y = 55.8\%$	$f_1 = 22.1 \text{ Hz}$ $CEM_x = 0.0\%$ $CEM_y = 57.5\%$
4		$f_4 = 38.3 \text{ Hz}$ $CEM_x = 72.2\%$ $CEM_y = 62.5\%$	$f_4 = 40.3 \text{ Hz}$ $CEM_x = 72.3\%$ $CEM_y = 63.0\%$
8		$f_8 = 73.0 \text{ Hz}$ $CEM_x = 72.2\%$ $CEM_y = 77.6\%$	$f_8 = 80.0 \text{ Hz}$ $CEM_x = 72.3\%$ $CEM_y = 79.3\%$
10		$f_{10} = 84.7 \text{ Hz}$ $CEM_x = 78.6\%$ $CEM_y = 77.6\%$	$f_{10} = 89.2 \text{ Hz}$ $CEM_x = 78.3\%$ $CEM_y = 79.3\%$

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