

## Aberystwyth University

### *Infrared Spectrometer for ExoMars: A Mast-Mounted Instrument for the Rover*

Korablev, O. I.; Dobrolensky, Y.; Evdokimova, N.; Fedorova, A. A.; Kuzmin, V. O.; Mansevich, S. N.; Cloutis, E. A.; Carter, John; Poulet, F.; Flahaut, J.; Griffiths, A.; Gunn, Matthew; Schmitz, N.; Martín-Torres, J.; Zorzano, M. -P.; Rodionov, D. S.; Vago, J. L.; Stepanov, A. V.; Titanov, A. Y.; Vyazovetsky, N. A.

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Astrobiology Manuscript Central: <http://mc.manuscriptcentral.com/astrobiology>**INFRARED SPECTROMETER FOR EXOMARS (ISEM), A MAST-MOUNTED INSTRUMENT FOR THE ROVER**

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7 **1 INFRARED SPECTROMETER FOR EXOMARS (ISEM), A MAST-**  
8 **2 MOUNTED INSTRUMENT FOR THE ROVER**  
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50 **32 Abstract**

51 33 ISEM (Infrared Spectrometer for ExoMars) is a pencil-beam infrared  
52 34 spectrometer that will measure reflected solar radiation in the near infrared range  
53 35 for context assessment of the surface mineralogy in the vicinity of the ExoMars  
54 36 rover. The instrument will be accommodated on the mast of the rover, and will be  
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operated together with the PanCam High-Resolution Camera (HRC). ISEM will study the mineralogical and petrographic composition of the martian surface in the vicinity of the rover, and in combination with the other remote sensing instruments, aid the selection of potential targets for close-up investigations and drilling sites. Of particular scientific interest are water-bearing minerals, such as phyllosilicates, sulfates, carbonates, and minerals indicative of astrobiological potential, such as borates, nitrates, and ammonium-bearing minerals. The instrument has a  $\sim 1^\circ$  field of view and covers the spectral range between 1.15–3.30  $\mu\text{m}$  with a spectral resolution varying from 3.3 nm at 1.15  $\mu\text{m}$  to 28 nm at 3.30  $\mu\text{m}$ . ISEM's optical head is mounted on the mast, and its electronics box is located inside the rover's body. The spectrometer employs an acousto-optic tunable filter (AOTF), and a Peltier-cooled InAs detector. The mass of ISEM is 1.74 kg, including the electronics and harness. The science objectives of the experiment, the instrument design, and operational scenarios are described.

## 1 Introduction

The ExoMars rover is a mobile laboratory equipped with a drill to sample the surface of Mars to a maximum depth of 2 m, and a suite of instruments to analyze the samples. The drilling device is the only means to access near subsurface materials and introduce them to the internal analytical laboratory. As the number of samples obtained with the drill will be limited, the selection of high value sites for drilling will be crucial. The rover's mast is therefore equipped with a set of remote sensing instruments to assist the selection process by characterizing the geological and compositional properties of the surrounding terrains. It includes several cameras – a pair of navigation cameras (NavCam), and a panoramic camera (PanCam). PanCam consists of stereo multispectral wide-angle camera pair (the WACs) and a high-resolution color camera (HRC) (Coates et al, this issue) and will provide the context images used to plan travelling and sampling. To complement and enhance the capabilities of the remote sensing suit, an infrared spectrometer able to unambiguously distinguish many rocks and minerals from their spectral reflectance, will allow remote characterization and selection of potential astrobiological targets. This mast-mounted IR spectrometer was proposed during an early discussion of the new ESA-Roscosmos ExoMars configuration as a useful addition to the rover science and to help operations by characterizing from afar the mineralogical interest of targets that the rover could

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75 ISEM is a derivative of the Lunar Infrared Spectrometer (LIS) (Korablev et al,  
76 2015) being developed at the Space Research Institute (IKI) in Moscow for the  
77 Luna-25 and Luna-27 Russian landers planned for flight in 2019 and 2021,  
78 respectively (Zelenyi et al, 2014). Both the ISEM and LIS instruments have been  
79 conceived with similar spectral capabilities. The ISEM design is improved with  
80 respect to that of LIS, and modifications were also necessary to comply with the  
81 more stringent environmental conditions on the ExoMars rover. A fully  
82 operational model of ISEM is not ready at the moment of the paper submission,  
83 and the assessment of its measurement performance has been made with the  
84 available LIS prototype.

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86 ISEM is one of two Roscosmos-provided instruments for the ExoMars Rover. It is  
87 being predominantly developed at IKI, but includes contributions from the  
88 National Research Institute for Physicotechnical and Radio Engineering  
89 Measurements (VNIIFTRI) in Russia, Moscow State University, also in Russia,  
90 and the Main Astrophysical Observatory, National Academy of Sciences in  
91 Ukraine. A calibration target to be used jointly by PanCam and ISEM is being  
92 contributed by Aberystwyth University, United Kingdom. Key components, such  
93 as the AOTF and the detector, are purchased from NII Micropribor (Russia) and  
94 Teledyne (USA). The science team includes researchers from Russia (IKI and  
95 Vernadsky Institute), France, Italy, Sweden, Germany, UK, and Canada. The  
96 team shares science team members with the PanCam and MicrOmega rover  
97 instrument teams. A full list of the ISEM Science and Technical teams is given at  
98 the end of the paper.

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100 After a brief summary of the major objectives of the ExoMars mission, we  
101 describe the scientific goals of the ISEM instrument. The technical design is then  
102 detailed. Measured performances of LIS prototype and and expected  
103 performances of ISEM are presented subsequently. We conclude the paper by  
104 addressing the operational scenarios and related environmental constraints.

## 105 106 **2 Science Objectives**

### 107 **2.1 Contribution to overall rover mission science**

108 The scientific objectives of the ExoMars Program are defined as (Vago et al, this

109 issue):

- 110 1. To search for signs of past and present life on Mars;
- 111 2. To investigate the water/geochemical environment as a function of depth  
112 in the shallow subsurface;
- 113 3. To study martian atmospheric trace gases and their sources;
- 114 4. To characterize the surface environment

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116 The first two objectives are the most relevant to the ExoMars rover. The ISEM  
117 experiment will contribute to achieving each of these objectives in the following  
118 ways:

- 119 1. Many of the minerals and rocks detectable by ISEM are good indicators of  
120 past habitable conditions, and of biological processes (e.g, carbonates,  
121 oxalates, borates, nitrates, NH<sub>4</sub>-bearing minerals (Applin et al, 2015; Berg  
122 et al, 2016; Cloutis et al, 2016)) and may even contain biomolecules in  
123 detectable concentrations. These include a class of biogenic minerals  
124 which are further known to provide a substrate for, and catalyze pre-biotic  
125 reactions. Organic compounds, including Polycyclic Aromatic  
126 Hydrocarbons (PAHs) and those containing aliphatic C-H molecules, can  
127 also be distinguished by ISEM (e.g, Clark et al, 2009; Izawa et al, 2014).
- 128 2. The ISEM instrument is capable of recognizing minerals and rocks, which  
129 are indicative of the presence of water as well as geochemical  
130 environmental indicators. ISEM can also be used to analyze the drill  
131 cuttings excavated by the ExoMars drill system. The operating drill  
132 obscures ISEM's view, and observing the cuttings is only possible after  
133 the rover moves off and revisits the drilling site.
- 134 3. If present, ISEM could detect organic-bearing materials, possibly evolving  
135 trace gases, such as hydrocarbons.
- 136 4. ISEM can carry out atmosphere observations, providing information on  
137 dust and clouds, and contributing to characterization of the atmospheric  
138 humidity. ISEM is capable of identifying and monitoring surface frost, and  
139 may assess the diversity and stability of various minerals on the martian  
140 surface, e.g, by monitoring changes in the spectral properties of drill  
141 cuttings over time.

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143 In terms of where the rover will land, ESA has issued the following scientific  
144 criteria for landing site selection (Vago et al, 2015):

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8 146 For the ExoMars Rover to achieve results regarding the possible existence of  
9 147 biosignatures, the mission has to land in a **scientifically appropriate setting**:  
10 148 1. The site must be **ancient** (older than 3.6 Ga) — from Mars' early, more  
11 149 life-friendly period: the Noachian to the Noachian/Hesperian boundary;  
12 150 2. The site must show abundant morphological and mineral evidence for  
13 151 long-term, or frequently reoccurring, **aqueous activity**;  
14 152 3. The site must include numerous **sedimentary outcrops**;  
15 153 4. The outcrops must be **distributed** over the landing ellipse to ensure the  
16 154 rover can get to some of them (typical rover traverse range is a few km);  
17 155 5. The site must have **little dust** coverage.  
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23 157 ISEM addresses these criteria in the following ways:

24 158 1. The suite of samples that we have used to characterize ISEM in laboratory  
25 159 tests includes materials similar to those found in Noachian terrains (e.g,  
26 160 phyllosilicates, carbonates, etc.)  
27 161 2. ISEM will be able to identify minerals that are indicative of aqueous  
28 162 activity, such as phyllosilicates and hydrated sulfates.  
29 163 3. Rocks and minerals presumed to have formed in sedimentary  
30 164 environments are the main focus of our investigation and ISEM is  
31 165 particularly well suited to detect and characterize them.  
32 166 4. ISEM as a remote instrument is well adapted to characterize the  
33 167 stratification at the outcrop scale.  
34 168 5. Our investigation will assess the obscuring effects of dust, first of all on  
35 169 the target areas, but also observing dust accumulation on the calibration  
36 170 target, excavated drill cuttings (e.g, Rice et al, 2011).  
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## 43 172 **2.2. Synergies with other instruments**

44 173 The main goal of ISEM is to establish the mineral composition of Mars' surface  
45 174 materials remotely. ISEM, together with PanCam (Fig. 1) offer high potential for  
46 175 the remote identification and characterization of any scientifically high-value  
47 176 targets in the vicinity of the rover, including proximal and distant rocks, outcrops,  
48 177 and other geological formations. ISEM will help to establish the geological  
49 178 context of each site along the rover traverse, discriminating between various  
50 179 classes of minerals and rocks. ISEM will also be important in order to select  
51 180 promising sites for subsurface sampling.  
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8 182 Infrared reflectance spectroscopy allows the study of the composition in the  
9 183 uppermost few millimeters of a rock's surface. It allows discriminating between  
10 184 various classes of silicates, (hydr-)oxides, hydrated/hydroxylated salts and  
11 185 carbonates. As shown in Fig. 2, the 1.3° Field of View (FOV) of ISEM lies within  
12 186 the 5° FOV of the color PanCam high-resolution camera (HRC), and they both  
13 187 are within a much wider FOV (38.6°) of the Wide Angle Cameras (WACs) with  
14 188 multispectral capabilities (Coates et al, this issue). The multispectral data are  
15 189 produced using a filter wheel with 11 filter positions for each of the two WACs.  
16 190 Out of the 22 filters, six are devoted to red, green, and blue broadband color,  
17 191 duplicated in the both cameras, 12 are optimized for mineralogy in the 400-1000  
18 192 nm range, and four "solar" filters are dedicated to atmospheric studies. By  
19 193 extending the wavelength range beyond PanCam, ISEM will enable many more  
20 194 spectral features diagnostic of specific mineralogy to be detected. Together  
21 195 PanCam and ISEM provide spectrally resolved information from 0.4 to 3.3 μm.  
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24 197 The identification and mapping of the distribution of aqueous alteration products  
25 198 in the upper surface layer, combined with subsurface data from the neutron  
26 199 detector ADRON and the ground-penetrating radar WISDOM, will help to  
27 200 understand the subsurface structure and the exobiology potential at each  
28 201 prospective drilling site.  
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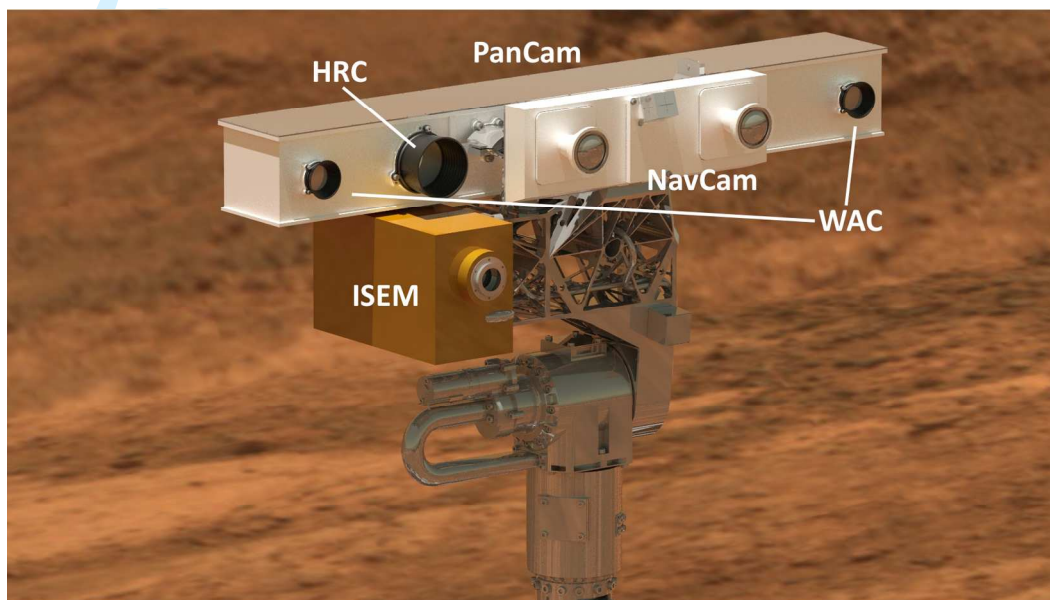
31 203 The collected drill samples will be analyzed in the rover's Analytical Laboratory  
32 204 Drawer (ALD) by several instruments. The first is an infrared hyperspectral  
33 205 microscope MicrOmega (Bibring et al, this issue). The principle of MicrOmega is  
34 206 very similar to that of ISEM. The reflectance spectroscopy is performed in the  
35 207 near-IR, but the analysis is done at the microscopic scale and the sample is  
36 208 illuminated by monochromatic light source. In contrast to MicrOmega, ISEM has  
37 209 a much wider field of view and range of detection. The distance to a target is not  
38 210 really limited, and practically may reach hundreds of meters. Thus, ISEM is better  
39 211 suited for accommodation on the rover mast, where it can be employed for target  
40 212 identification of far away objects, but also for investigating outcrop, rock, and soil  
41 213 mineralogy at close range.  
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44 215 There is no specific instrument dedicated to environmental characterization on the  
45 216 rover. Although hampered by the limited number of observation cycles, ISEM,  
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217 jointly with PanCam, will deliver information regarding atmospheric aerosol  
 218 opacity and the atmospheric gaseous composition. The data on water vapor  
 219 content and aerosol will be retrieved as a by-product of reflectance spectra, from  
 220 in-flight calibration, from the direct Sun imaging by PanCam, and from sky  
 221 observations by PanCam and ISEM.

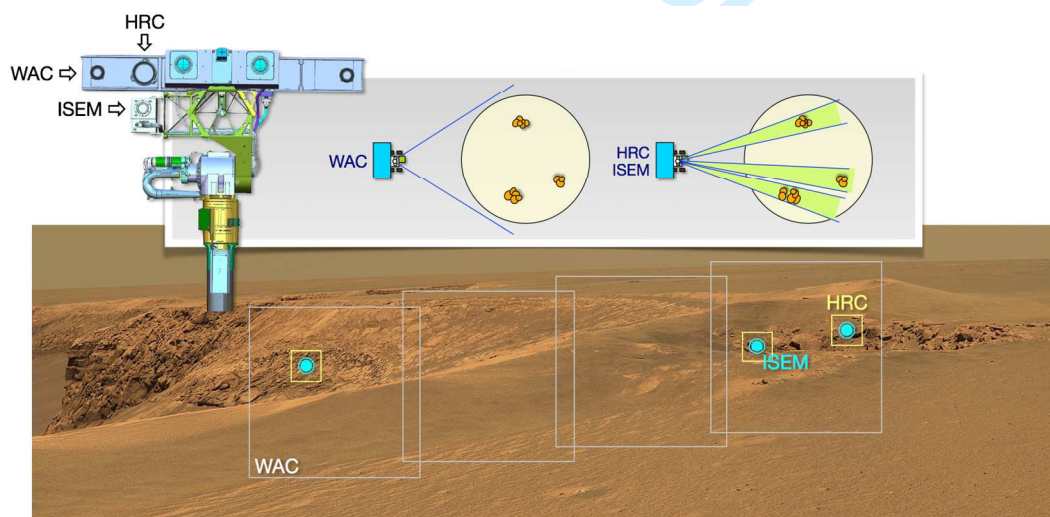
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224 Figure 1. Schematic view of the ExoMars Rover mast instruments: PanCam,  
 225 navigation cameras, and ISEM.

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7 228 Figure 2. Schematic representation of possible ISEM and PanCam joint  
8 229 observation scenario showing a sequence of ISEM measurements acquired  
9 230 together with WAC and HRC PanCam frames.  
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### 12 233 **2.3 The method**

13 234 Near-IR spectroscopic observations of the Mars surface have not been performed  
14 235 from any surface platform to date. Spectroscopy was employed on the Mars  
15 236 Exploration Rovers (MER) in the thermal IR range (the radiation emitted by the  
16 237 surface, 5-29  $\mu\text{m}$ ) with the Mini-TES instrument (Christensen et al, 2003,  
17 238 2004a,b). These remote observations proved very useful for selecting targets for  
18 239 *in situ* analyses by the Alpha Proton X-Ray Spectrometer (APXS) and Mossbauer  
19 240 instruments on Spirit and Opportunity (e.g, Squyres et al, 2004). They also  
20 241 allowed the identification of carbonates in Gusev Crater (Morris et al, 2010), and  
21 242 undertook a number of atmospheric investigations (Smith et al, 2006).  
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23 244 Conversely, the near-IR spectral range (i.e. the infrared solar reflected radiation)  
24 245 is being widely used in orbital observations, as it allows for significantly better  
25 246 characterization of aqueous minerals than does the thermal infrared. To illustrate  
26 247 this point, the TES instrument on Mars Global Surveyor (MGS) in the wavelength  
27 248 range from 6 to 50  $\mu\text{m}$  has mapped the distribution of both mafic and anhydrous  
28 249 high-silica minerals (Bandfield et al, 2000; Christensen et al, 2001), but could not  
29 250 unambiguously detect clay/clay-like minerals and salts, including carbonates. In  
30 251 the near-IR, the OMEGA hyperspectral instrument on Mars Express (MEx)  
31 252 measured the martian surface reflectance from 0.5 to 5.2  $\mu\text{m}$ , and was able to  
32 253 recognize a number of different phyllosilicates and sulfates (Bibring et al, 2006).  
33 254 Building on OMEGA's success, the higher spatial resolution of the CRISM  
34 255 instrument on Mars Reconnaissance Orbiter (MRO), which measures reflectance  
35 256 from 0.362 to 3.92  $\mu\text{m}$ , permitted the first detections of carbonates and serpentine  
36 257 (Ehlmann et al, 2008, 2010). One shortcoming of the orbital observations is the  
37 258 limited surface resolution (300-500 m per pixel for OMEGA/MEx and  $\sim$ 20 m for  
38 259 CRISM/MRO), hindering the detection of small-scale exposures. In addition, such  
39 260 instruments suffer from limited detection sensitivity owing to the combination of  
40 261 the subtle nature of aqueous mineral spectroscopic features and to the short dwell  
41 262 time of orbiting spacecraft. The wide variety of minerals on Mars detected by the  
42 263 higher spatial resolution CRISM instrument compared to the lower resolution  
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7 264 OMEGA instrument demonstrates the greater mineralogical diversity that can be  
8 265 identified as spatial resolution improves. Therefore the close-up near-IR  
9 266 capability of ISEM on the ExoMars rover offers a very high diagnostic potential.  
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11 268 The small size and low mass of the ISEM instrument allows it to be  
12 269 accommodated more easily on a rover platform. A similar AOTF-based near IR  
13 270 spectrometer has been selected to enhance the capabilities of the successor to the  
14 271 ChemCam instrument on the Curiosity Rover (Fouchet et al, 2015). The new  
15 272 SuperCam instrument planned for the NASA 2020 rover utilizes a combination of  
16 273 the Laser-Induced Breakdown Spectrometer (LIBS), Raman, and near-IR  
17 274 reflectance spectroscopy, offering complementary elemental and mineralogical  
18 275 analysis techniques.  
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20 277 Spectral range: An optimal spectral range for a near-IR mineralogical  
21 278 characterization is 0.9-4.0  $\mu\text{m}$  to encompass a broad absorption centered in the  
22 279 0.95-1.1  $\mu\text{m}$  region characteristic for pyroxene or olivine, and 3.4-3.9  $\mu\text{m}$  for  
23 280 carbonate overtones. The spectral range of the prototype LIS spectrometer is a  
24 281 compromise of science requirements and technical limitations. Its longwave  
25 282 bound of 3.3  $\mu\text{m}$  allows for detection of hydration features in the 3  $\mu\text{m}$  region  
26 283 that can be used to discriminate different types of phyllosilicates and other  
27 284 water/hydroxyl-bearing materials (e.g, Clark et al, 1990). The broad absorption in  
28 285 the 3  $\mu\text{m}$  region has contributions from the long wavelength wing of a hydroxyl  
29 286 (OH) fundamental stretch usually centered near 2.7-2.8  $\mu\text{m}$ , H<sub>2</sub>O fundamental  
30 287 stretches, and the first overtone of the H<sub>2</sub>O bending fundamental (Clark et al,  
31 288 1990). On Mars, hydration is strong enough that it can be detected at shorter  
32 289 wavelengths. Clays and other hydrated minerals can be detected and  
33 290 discriminated from overtones and combinations absorption features, at  $\sim$ 1.4  $\mu\text{m}$ ,  
34 291  $\sim$ 1.9  $\mu\text{m}$ , and 2.2-2.3  $\mu\text{m}$ . Detection of carbonates by ISEM will be possible using  
35 292 correlated 2.3- and 2.5- $\mu\text{m}$  region bands, similar to CRISM/MRO (Ehlmann et al,  
36 293 2008). Beyond 3.3-3.4  $\mu\text{m}$  the signal is complicated by the thermal radiation of  
37 294 the surface (and from the instrument itself, detectors with longer-wavelength  
38 295 bound are much less sensitive because of thermal background). On the short  
39 296 wavelength bound, the sensitivity of detectors optimized for 3- $\mu\text{m}$  range falls  
40 297 abruptly below 1  $\mu\text{m}$ , hampering mafic silicate identification and characterization.  
41 298 However mafic silicates can be recognized and discriminated by the shape of the  
42 299 long wavelength wing of the 1- $\mu\text{m}$  ferrous iron absorption band and the  
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7 300 wavelength position of the 2- $\mu\text{m}$  ferrous iron absorption (see the following  
8 301 section for the details).

9 302

10 303 Taking into account the advantages of the heritage design, we choose using the  
11 304 same 1.15-3.3  $\mu\text{m}$  range for ISEM on ExoMars as for the lunar instrument.

12 305

13 306 Spectral resolution. The spectral features of Mars surface materials will generally  
14 307 include a mixture of several minerals, and are usually broad ( $>20$  nm full width at  
15 308 half maximum), so that there is no stringent requirement on the spectral resolution  
16 309 for orbital instruments (see Table 1). With increasing spatial resolution, individual  
17 310 minerals may become more apparent, with deeper characteristic absorption  
18 311 features. Pure minerals can likely be observed only at the scale of individual  
19 312 grains, e.g. with the microscope-spectrometer MicrOmega. Given that the ISEM  
20 313 FOV will typically encompass an area of few  $\text{cm}^2$ , a mixture of minerals will  
21 314 likely be present, therefore a spectral resolution requirement of  $25 \text{ cm}^{-1}$  was  
22 315 chosen that corresponds to 3.3 nm at 1.15  $\mu\text{m}$ , 16 nm at 2.5  $\mu\text{m}$ , and 28 nm at 3.3  
23 316  $\mu\text{m}$ .

24 317

25 318 In mineralogical studies, the best approach is the acquisition of the full available  
26 319 spectral range. Nyquist sampling (two measured points per spectral resolution) of  
27 320 the full 1.15-3.30  $\mu\text{m}$  ( $3030\text{-}8696 \text{ cm}^{-1}$ ) range with a spectral resolution of  $25 \text{ cm}^{-1}$   
28 321 results in 453 (512 with margin) spectral points. With a one-second exposure for  
29 322 each spectral point, a complete spectrum is therefore measured in 8.5 minutes.  
30 323 Customable or random wavelength access by the AOTF (see below) allows  
31 324 oversampling, selecting different portions of the spectrum, and focusing on the  
32 325 most interesting or diagnostic spectral intervals.

33 326

34 327 The color and IR reflectance of the Mars surface are what one would expect of  
35 328 iron-bearing mineral species that are either primary or formed by hydrous or  
36 329 anhydrous chemical weathering. The infrared reflectance varies from 5% to 35%,  
37 330 with typical values being 15-20% for low albedo regions and 25-30% for high-  
38 331 albedo regions (Erard 2001). A characteristic feature of Mars spectra is a deep  
39 332 absorption feature starting beyond the 2.7- $\mu\text{m}$   $\text{CO}_2$  saturated atmospheric band.  
40 333 Its depth reflects the degree of hydration of the martian surface. An increase in  
41 334 measured reflectance toward longer wavelengths signifies an increasing  
42 335 contribution from emitted thermal radiation. Observations from the ExoMars

336 rover mast will be carried out using a broad range of phase angles, preferably at  
 337 high Sun. There is no Sun avoidance requirement for ISEM.

338

339 Table 1. Near-IR spectrometers used and planned to study Mars surface

Instrument	Spectral range, $\mu\text{m}$	Spectral resolution, nm	Surface resolution	Ref.
ISM/Phobos 2	0.8 -3.1	50	20x30 km	Bibring et al, 1990
OMEGA/MEx	0.35-1 1-2.5 2.5-5.1	7 14 20	0.3-5 km	Bibring et al, 2004
CRISM/MRO	0.362-3.92	6.6/pix	18 m	Murchie et al, 2007
ISEM/ExoMars	1.1-3.3	3.3 at 1.15 $\mu\text{m}$ 16 at 2.5 $\mu\text{m}$ 28 at 3.3 $\mu\text{m}$	3-10 cm	This study
MicrOmega/ExoMars	0.5-3.65	2 at 1.0 $\mu\text{m}$ 25 at 3.6 $\mu\text{m}$	20 $\mu\text{m}$	Bibring et al, this issue
SuperCam/2020 Rover	1.3-2.6	5 at 1.3 $\mu\text{m}$ , 20 at 2.6 $\mu\text{m}$	1.3-7 mm	Fouchet et al, 2015

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341

#### 342 2.4 Potentially detectable mineral groups

343 The greatest share of what we know about martian mineralogy has been gleaned  
 344 from orbital near- and thermal-infrared spectroscopy measurements. The infrared  
 345 instruments have been successful in identifying igneous minerals, while the near-  
 346 IR instruments were more important for detecting ancient hydrated minerals, such  
 347 as phyllosilicates, as well as carbonates and sulfates, which could be used to  
 348 develop a timeline of changing environmental conditions on Mars and evidence of  
 349 previous more clement epochs on Mars. The capabilities of ISEM can be gauged  
 350 against these detections to assess its expected science return, as well as  
 351 expectations for detecting other mineral species of high scientific value, all in the  
 352 context of the proposed ExoMars landing sites.

353

354 1. Phyllosilicates: The identification of hydrated phyllosilicates on the martian  
 355 surface is one of the discoveries driving *in situ* exploration of Mars.

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7 356 Phyllosilicates are commonly detected in ancient, Noachian-aged martian terrains  
8 357 (Poulet et al, 2005). Various types of phyllosilicates have been observed,  
9 358 including Fe/Mg and Al-rich smectites, micas, vermiculites, kaolinite, chlorite,  
10 359 and serpentine (Ehlmann et al, 2009; Murchie et al, 2009; Carter et al, 2013). The  
11 360 presence of these minerals likely indicates that conditions in the past were  
12 361 favorable for the presence of liquid water at or near the surface. They are  
13 362 considered good reaction templates for organic molecules and excellent for  
14 363 biosignature preservation (Bishop et al, 2013, and references therein).  
15 364 Phyllosilicates have been detected from orbit within the ellipse of two out of the  
16 365 three candidate landing sites for the ExoMars rover, Oxia Planum and Mawrth  
17 366 Vallis (Vago et al, this issue). They have not been detected in Aram Dorsum yet;  
18 367 even if they are there, it may be difficult to see them from orbit. ISEM will have  
19 368 the capabilities to detect sharp features between 1.35 and 2.6  $\mu\text{m}$  in phyllosilicate  
20 369 reflectance spectra, which are related to combinations and overtones of OH-M  
21 370 stretching and bending bound to various cations (M) as well as the H-O-H stretch  
22 371 and bend from water bound in interlayer regions or adsorbed on mineral surfaces  
23 372 (Clark et al, 1990) (Figures 3a, b, and c).  
24 373

25 374 2. Carbonates have been discovered on Mars at a number of locations from orbit  
26 375 (Ehlmann et al, 2008, Wray et al, 2016), by surface rovers (Morris et al, 2010),  
27 376 and in martian meteorites (e.g, McKay et al, 1996). The precise species are not  
28 377 always known, but various Mg-Fe-Ca carbonates provide the best match for  
29 378 observed spectral features. Carbonates are important minerals for identification as  
30 379 they often indicate the presence of habitable (circum-neutral) environments. They  
31 380 are also widely implicated in explaining the loss of a substantial fraction of the  
32 381 presumed early Mars dense carbon dioxide atmosphere. Carbonate detection and  
33 382 characterization is possible in a number of wavelength regions. The 2.2-2.6  $\mu\text{m}$   
34 383 region is one of the best because it is not affected by thermal emission, and  
35 384 absorption bands in this region vary in their positions for different carbonate  
36 385 species (Figure 3d). ISEM may also be able to identify Fe-bearing from Fe-free  
37 386 carbonates on the basis of the presence or absence of a ferrous iron absorption  
38 387 band near 1.2  $\mu\text{m}$ .  
39 388

40 389 3. Sulfates: A number of sulfate minerals have been observed on Mars, both from  
41 390 orbit and in situ, including poorly ( $\leq 2 \text{ H}_2\text{O}$ ) hydrated species (gypsum, kieserite,  
42 391 bassanite), and polyhydrated varieties (Squyres et al, 2004; Gendrin et al, 2005;  
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7 392 Wang et al, 2006; Flahaut et al, 2014; Nachon et al, 2014) as well as hydroxylated  
8 393 species (copiapite, jarosite, alunite and the dehydrated form of amarantite  
9 394  $\text{FeSO}_4(\text{OH})$ ). These sulfates are all water/hydroxyl-bearing and several are also  
10 395 indicative of circum-neutral, perhaps habitable environments. Sulfates also have  
11 396 the potential to preserve microfossils (e.g, Allwood et al, 2013) and concentrate  
12 397 organic compounds (Noe Dobrea et al, 2016). Specific sulfates can be powerful  
13 398 indicators of environmental conditions on Mars, past and present (e.g, Leftwich et  
14 399 al, 2013). Sulfate discrimination is possible using a number of wavelength  
15 400 intervals in the ISEM range. For example, gypsum is characterized by a uniquely-  
16 401 shaped absorption band in the 1.4- $\mu\text{m}$  region, jarosite and alunite by absorption  
17 402 bands in the 1.8- $\mu\text{m}$  region, and other sulfates by S-O associated absorption bands  
18 403 in the 2.0-2.5  $\mu\text{m}$  region (Cloutis et al, 2006) (Figures 3e and f).  
19 404

20 405 4. Silica: Silica, in a variety of forms, is present in a number of terrains on Mars  
21 406 (Bandfield et al, 2004; Smith and Bandfield, 2012; Smith et al, 2013, Carter et al,  
22 407 2013). Differences in reflectance spectra can be used to distinguish different  
23 408 forms of silica, some of which are associated with habitability (Rice et al, 2013).  
24 409 The region most useful for distinguishing different forms of silica is located near  
25 410 2.2  $\mu\text{m}$  and is due to Si-OH overtones. The nature and abundance of these bonds  
26 411 varies among different types of silica (Rice et al, 2013). This diagnostic feature  
27 412 can be clearly seen in the reflectance spectra of two different grain sizes of quartz  
28 413 (Figure 3g).  
29 414

30 415 5. Igneous minerals. Petrologic investigations of martian rocks have been  
31 416 accomplished by mineralogical, geochemical, and textural analyses by remote  
32 417 sensing observations, in situ investigations, and laboratory analyses of martian  
33 418 meteorites. Igneous rocks are found in numerous settings; NIR spectroscopy from  
34 419 orbiting spacecraft has been an effective mineralogic tool to identify and to map  
35 420 at a global scale various rock-forming minerals such as olivines, pyroxenes and  
36 421 iron-bearing plagioclases (Poulet et al. 2009; Ody et al. 2013; Carter and Poulet  
37 422 2013). (Figure 3h and i).  
38 423

39 424 6. Ferrous oxides/hydroxides. These minerals are the widespread weathering  
40 425 products of primary iron-bearing materials (Ody et al, 2012). In the ISEM spectral  
41 426 range, they manifest themselves mainly in the shape of the continuum.  
42 427 Hydroxides exhibit shallow absorption features, including near 1.41 and 1.93  $\mu\text{m}$ ,  
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7 428 and several species have been reported from NIR orbital investigations. Some  
8 429 iron-bearing species, such as goethite, exhibit Fe-OH absorption bands longward  
9 430 of the same band in Fe-free minerals (Beck et al, 2011) (Figure 3j).

10 431

11 432 Several other potential candidates that are amenable to detection and analysis by  
12 433 ISEM, and are of high scientific value, are described below.

13 434

14 435 Organic compounds –PAHs. Infrared spectroscopy is potentially sensitive to the  
15 436 presence of organic compounds, which have already been identified *in situ*,  
16 437 though in trace quantities, within Mars samples (Freissinet et al, 2015). PAHs are  
17 438 generally a stable form of organic molecules, and if organic compounds are or  
18 439 were present on the surface of Mars, they have likely transformed to PAHs  
19 440 (Anders et al, 1996). PAHs are also the dominant form of organic material in  
20 441 Archaean terrestrial rocks (Marshall et al, 2007; and references therein). Their  
21 442 reflectance spectra exhibit absorption features that are largely associated with a  
22 443 number of functional groups that may be present, particularly aliphatic C-H  
23 444 molecules (Izawa et al, 2014). They can exhibit absorption bands near 1.69  $\mu\text{m}$   
24 445 due to CH overtones, 1.50  $\mu\text{m}$  due to N-H stretching overtones, and numerous  
25 446 other overtone and combination bands beyond 2.1  $\mu\text{m}$ . A shoulder of the  
26 447 fundamental 3.2-3.35  $\mu\text{m}$  aromatic band could be also detectable in the spectra,  
27 448 though complicated by a deep 3- $\mu\text{m}$  absorption, characteristic for Mars  
28 449 reflectance in general, see below.

29 450

30 451 Perchlorates and chlorides. Perchlorates are strongly suspected to be present on  
31 452 Mars on the basis of analytical results from the Phoenix lander in the polar region  
32 453 (Hecht et al, 2009; Cull et al, 2010) and from the MSL Curiosity rover in the near-  
33 454 equatorial region (Farley et al, 2016). Perchlorates have been linked both to the  
34 455 destruction of organic compounds, and to liquid water. They can absorb  
35 456 atmospheric water and allow for the existence of stable liquid water brines  
36 457 (Martin-Torres et al, 2015). Their spectral properties have been the focus of  
37 458 multiple studies (Bishop et al, 2014; Hanley et al, 2015) since the presence of  
38 459 perchlorates could have strong implications for the (non) preservation of  
39 460 biosignatures. Reflectance spectra are characterized by possible Cl-O-H<sub>2</sub>O-  
40 461 associated absorption bands near 1.35, 1.75, and 2.15  $\mu\text{m}$ . The 3-micron signature  
41 462 of water is present in IR spectra of perchlorate samples in the form of hydrate, ice  
42 463 or liquid brine (Zorzano et al, 2009). Spectral features in a more suitable NIR

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7 464 range have possibly been detected from orbit (Ojha et al, 2015), the efforts to  
8 465 confirm this are under way. While anhydrous chloride salts are mostly featureless  
9 466 in the wavelength range of ISEM, chlorinated species have been detected in  
10 467 abundance at Mars (Osterloo et al, 2010), and their hydrated forms would be  
11 468 detectable in the NIR (Hanley et al, 2011).  
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13 469  
14 470 Oxalates. Oxalates are carbon-bearing minerals that new evidence suggests may  
15 471 be present on Mars in addition to, or instead of, carbonates (Applin et al, 2015).  
16 472 Terrestrial oxalates are typically formed in biological processes (Applin et al,  
17 473 2016, and references therein). Their presence on Mars would suggest past  
18 474 habitability and possibly biological processes. Oxalates have some similarities  
19 475 with carbonates in terms of multiple C-O associated absorption bands, but their  
20 476 wavelength positions and shapes differ from carbonates and can be used to  
21 477 discriminate different oxalate species. Major oxalate absorption bands are present  
22 478 in the 2.2-2.5 and 3.2-3.3  $\mu\text{m}$  regions.  
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24 479  
25 480 Water ice. The presence of water ice at the landing site during daytime  
26 481 observations is unlikely, but cannot be excluded (Carrozzo et al, 2009). Seasonal  
27 482 frost, if present at the landing site, will be readily identifiable by ISEM, allowing  
28 483 study of its season deposition cycle and dependence on local conditions.  
29 484 Unambiguous signatures of water ice are present in the ISEM wavelength range  
30 485 with diagnostic absorption bands at 1.25, 1.5, 2.0 and 3.0  $\mu\text{m}$ .  
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32 486  
33 487 Nitrates. Nitrates may have been detected *in situ* on Mars (Navarro-Gonzalez et al,  
34 488 2013). They are important minerals because they may indicate the operation of  
35 489 biological processes and/or can serve as a bioavailable source of nitrogen. Their  
36 490 spectral reflectance properties have recently been studied (Cloutis et al, 2016).  
37 491 The  $\text{NO}_3$  molecule gives rise to multiple absorption bands that can be detected  
38 492 down to as low as 1.8  $\mu\text{m}$ .  
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40 493  
41 494 Phosphates. Phosphates are important as a potential indicator of biological  
42 495 processes. They are a likely possibility to explain phosphorus, which has been  
43 496 detected on Mars (Blake et al, 2013), but its form is not yet determined. Their  
44 497 spectral properties are only partially known (Lane et al, 2011). They share some  
45 498 similarities with some other mineral groups, including ferrous iron-associated  
46 499 absorption bands in the 1- $\mu\text{m}$  region and weak P-O absorption bands in the 2-2.5  
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6 500  $\mu\text{m}$  region. Robust phosphate detection by ISEM may not be feasible.  
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8 501

9 Borates. Borates are another important mineral group that can be associated with  
10 habitability and biological processes (Stephenson et al, 2013). Boron cannot be  
11 detected with an APXS instrument (Cloutis et al, 2016), but it has been detected  
12 504 detected with an APXS instrument (Cloutis et al, 2016), but it has been detected  
13 505 and quantified in clays from martian meteorite MIL090030 (Stephenson et al,  
14 506 2013). Borate deposits on Earth are often associated with enclosed evaporitic  
15 507 deposits and might be present in association with martian chlorides and sulfate  
16 508 salts. Borates exhibit B-O associated absorption bands in the 1.55, and 2.15-2.25  
17 509  $\mu\text{m}$  regions, enabling their detection using multiple wavelength intervals and  
20 510 absorption features measurable by ISEM. Absorption band positions and shapes  
21 511 vary between different borates.  
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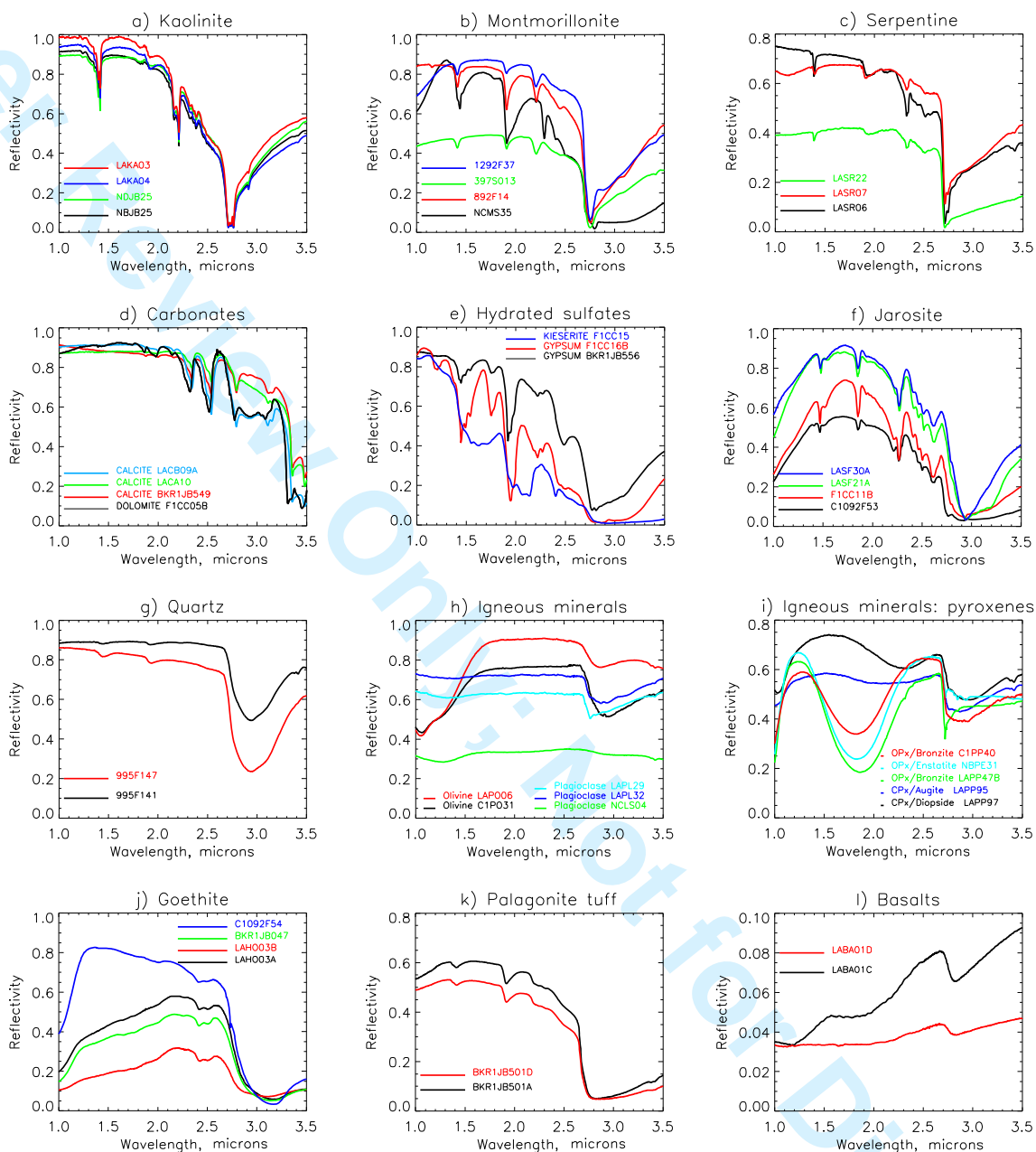
23 512

24 513 Ammonium-bearing minerals. The presence of the ammonium ( $\text{NH}_4$ ) molecule in  
25 514 various minerals leads to strong absorption bands in the 1.6, 2.0-2.2 and 3.1  $\mu\text{m}$   
26 515 regions. Band shapes and positions can be used to discriminate different  
27 516 ammonium-bearing minerals (Berg et al, 2016). Ammonium-bearing minerals are  
28 517 of astrobiological importance for a number of reasons. Ammonium is often of  
29 518 biological origin, has high thermal stability, and can withstand some level of  
30 519 metamorphism (Boyd, 2001). Ammonium-bearing minerals on Mars have  
31 520 possibly been detected from spectroscopic observations made from orbit (Sefton-  
32 521 Nash et al, 2012), and a nitrogen cycle on Mars has been suggested to operate  
33 522 (Manning et al, 2008). Low concentrations of ammonium in soil have been  
34 523 reported for the Phoenix landing site (Quinn et al, 2011).  
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36 525

37 526 One of the limitations of spectrally characterizing Mars-relevant minerals with  
38 527 ISEM under ambient terrestrial conditions is that these mineral spectra display a  
39 528 nearly ubiquitous broad and frequently deep absorption feature in the 3- $\mu\text{m}$  region.  
40 529 This is normally attributable to water that may be present in various forms,  
41 530 including adsorbed, fluid inclusions, impurities or accessory phases, or due to  
42 531 incipient alteration. This feature is normally seen even in nominally anhydrous  
43 532 minerals, such as olivine and pyroxene (see Figure 3h and i; more examples of  
44 533 Mars background material spectra are presented in Figure 3 j, k, and l). When  
45 534 such minerals are exposed to Mars-like surface conditions, the 3- $\mu\text{m}$  feature  
46 535 commonly shows a reduction in both depth and width, but rarely disappears  
47 completely (Cloutis et al, 2007; Cloutis et al, 2008). It may be possible to use  
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7 536 characteristics of the 3- $\mu$ m absorption feature to constrain or determine water  
8 537 content (Milliken and Mustard, 2005), but in multicomponent targets, such  
9 538 determinations will be more difficult. Small amounts of water in the near  
10 539 subsurface and on the surface of Mars also seems to be ubiquitous, displaying  
11 540 seasonal variations (Milliken et al, 2007). Collectively these results suggest that  
12 541 the 3- $\mu$ m region may not be best suited for mineralogical determinations using  
13 542 ISEM. However, as discussed above, other wavelength regions accessible to  
14 543 ISEM have significant diagnostic potential for a wide range of minerals, as  
15 544 demonstrated effusively from terrestrial and Mars orbital studies.  
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Figure 3. Example spectra of minerals and rocks in the ISEM spectral range, which may be expected at the ExoMars landing sites (from CRISM spectral library, [http://pds-geosciences.wustl.edu/missions/mro/spectral\\_library.htm](http://pds-geosciences.wustl.edu/missions/mro/spectral_library.htm); sample labels from this library). The upper half of the figure (a-f) shows the spectra of minerals the most significant for habitability, while the spectra shown

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7 554 in the lower half (g-i) are more relevant to background minerals and dust. The  
8 555 spectral resolution of CRISM is close to that of ISEM in the center of the range;  
9 556 ISEM resolution is better at shorter wavelengths, and coarser at 3  $\mu\text{m}$  (see Table  
10 557 1).  
11 558

## 13 559 **2.5 Atmospheric studies (aerosol, gaseous content)**

14 560 There is no dedicated instrument, such as e.g. meteorological station on the  
15 561 ExoMars rover to characterize the environmental conditions. Therefore any  
16 562 information available by other means is of particular value. Spectrally resolved  
17 563 data in the visible-NIR and in the thermal IR ranges have been used to assess the  
18 564 atmospheric state from Mars Pathfinder and Mars Exploration Rovers. In the  
19 565 spectral range of ISEM there are absorption bands of atmospheric  $\text{CO}_2$  (at 1.43,  
20 566 1.6, 2.0, and 2.7  $\mu\text{m}$ ) and  $\text{H}_2\text{O}$  (at 1.38 and 2.56  $\mu\text{m}$ ), and it is well suited, in  
21 567 particular its short-wave sub-range to characterize the main mode of the martian  
22 568 dust (Fedorova et al, 2014). The condensation clouds in the equatorial region  
23 569 consist predominantly of water ice (Vincendon et al, 2011), with multiple  
24 570 absorption bands within the ISEM spectral range (see section 2.4). Carbon  
25 571 dioxide clouds if present at high altitudes may be detected by distinct features of  
26 572 solid  $\text{CO}_2$  near 1.2, 1.4, 1.5, 2.0, 2.7, and 3.0  $\mu\text{m}$ .  
27 573

28 574 The gaseous absorptions will be measured by ISEM as a by-product of every  
29 575 surface measurement. Depth of the  $\text{CO}_2$  absorptions can be used to determine the  
30 576 surface pressure, and the  $\text{H}_2\text{O}$  absorptions quantify the total column water  
31 577 contents in the atmosphere above the site. To disentangle the possible overlapping  
32 578 atmospheric and mineral features in the reflected surface spectra, preferentially  
33 579 the observations of the *in situ* calibration target will be interpreted. The estimate  
34 580 of accuracy to retrieve the atmospheric pressure by ISEM is given by OMEGA  
35 581 study of the 2- $\mu\text{m}$   $\text{CO}_2$  absorption from the orbit: 7-10 Pa (~1%) 1-sigma. (Forget  
36 582 et al, 2007). With the optical path being half of the OMEGA case, our retrieval  
37 583 accuracy will be comparable.  
38 584

39 585 More precise measurements of surface pressure, minor gases, and better  
40 586 characterization of the atmospheric dust and cloud situation by ISEM will be  
41 587 possible using dedicated atmospheric observations. They will be coordinated with  
42 588 those by PanCam. The PanCam atmospheric study (Coates et al, this issue) will  
43 589 include direct solar observations just prior to sunset (to take advantage of the  
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7 590 maximal atmospheric path lengths) and cross sky brightness measurements. Such  
8 591 observations, e.g. performed with Mars Pathfinder camera (Titov et al, 1999;  
9 592 Markiewicz et al, 1999) or MERs (Lemmon et al, 2004; Smith et al, 2006) can be  
10 593 significantly strengthened by extending the spectral range to the near-IR.  
11 594

12 595 Direct solar observations by ISEM are not so far in the baseline. Their  
13 596 implementation would allow for high-accuracy pressure and water retrievals,  
14 597 better optical characterization of dust, direct and localized detection of H<sub>2</sub>O ice  
15 598 and, if present, the CO<sub>2</sub> ice clouds. Observing the Sun at different zenith angles  
16 599 helps putting constraints on the vertical distribution of water vapor. However  
17 600 ISEM is optimized for weak reflected light, and measuring the direct Sun signal  
18 601 may not be feasible. We successfully tested this possibility with the lunar  
19 602 prototype, but the ISEM aperture is larger, and a solar-blind filter within the  
20 603 foreoptics might become necessary. Such a filter would cause a several percent  
21 604 signal loss for the baseline ISEM observations. A final assessment of the direct  
22 605 Sun mode will be done during the characterization of the flight model.  
23 606

24 607 The cross-sky brightness measurements by ISEM are the best suited for  
25 608 characterizing the aerosol component of the Mars atmosphere. Because of quasi-  
26 609 permanent aerosol loading, the brightness of the martian sky at low zenith angles  
27 610 is not significantly lower than that from light scattered from the surface.  
28 611 Measurements at a range of phase angles will allow to extract the size distribution,  
29 612 optical properties, and even to assess the shape of aerosols suspended in the  
30 613 atmosphere. Water vapor absorption at 2.56  $\mu\text{m}$  is much stronger than that at 0.94  
31 614  $\mu\text{m}$  to be observed by WAC in solar filters, and the ISEM data might put  
32 615 additional constraints on the vertical distribution of water in the boundary layer of  
33 616 atmosphere.  
34 617

35 618 The variety of atmospheric measurements by ISEM calibrated using an *in situ*  
36 619 calibration target will allow the refinement of atmospheric scattering models and  
37 620 therefore refinement of the calibration of orbital spectral measurements of the  
38 621 martian surface.  
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### 40 623 **3 Instrument Description**

#### 41 624 **3.1 Instrument concept**

42 625 The measurement principle of ISEM is based on the use of an AOTF. The core  
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7 626 element of an AOTF is a birefringent crystal (typically of paratellurite,  $\text{TeO}_2$ , due  
8 627 to the combination of acoustic and optical properties) with a welded  
9 628 piezotransducer. The radio frequency (RF) applied to the transducer generates an  
10 629 acoustic field in the crystal, implementing acousto-optic interactions in Bragg's  
11 630 regime. The spectral selectivity of the acousto-optic diffraction allows the  
12 631 filtering of light. The diffraction occurs for a single wavelength, and there are no  
13 632 diffraction orders. The applied RF controls the tuning of the AOTF. AOTFs are  
14 633 technologically mature and widely used for spectral analysis. The robust design,  
15 634 small dimensions and mass, coupled to the absence of moving parts in an AOTF-  
16 635 based spectrometer, makes them popular for space applications. So far AOTF-  
17 636 based spectrometers have been used in space science: (i) to study the atmospheric  
18 637 composition of Mars and Venus (Korablev et al, 2006, 2012); (ii) on the Moon  
19 638 within the Chang'e-3 VNIS spectrometer (0.45-2.4  $\mu\text{m}$ ) mounted on the Yutu  
20 639 rover (He et al, 2014); (iii) for isolation of echelle-spectrometer diffraction orders  
21 640 in high-resolution instruments (Nevejans et al, 2006; Korablev et al, 2011, 2014;  
22 641 Neefs et al, 2015); and (iv) to illuminate the sample of an IR microscope with  
23 642 monochromatic light (Pilorget and Bibring 2013).  
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31 644 Wider application of the AOTFs in remote sensing is hampered by the inherent  
32 645 requirement to sequentially scan the spectrum. On an orbital mission with a short  
33 646 dwell time, this scanning interferes with the spacecraft or line of sight motion,  
34 647 complicating the analysis. Even for a pencil-beam device, different parts of the  
35 648 acquired spectrum would correspond to different observed areas. Conversely,  
36 649 observations from a static point, such as a planetary lander, or a rover, which  
37 650 remains immobile during the measurement, are well suited for AOTF-based  
38 651 instruments.  
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43 653 As described in the Introduction, ISEM is a close equivalent of LIS being  
44 654 developed for two Russian lunar landers. The LIS development is more advanced  
45 655 with respect to that of ISEM — by about two years. LIS benefited from the  
46 656 experience gained from the pencil-beam spectrometer design of the Mars Express  
47 657 and Venus Express instruments, and from the early developments of MicrOmega.  
48 658 SPICAM-IR AOTF spectrometer with the spectral range 0.9-1.7  $\mu\text{m}$  (Korablev et  
49 659 al, 2006) has been operating in Mars orbit since 2004. A similar SPICAV-IR  
50 660 instrument employing a double-range AOTF operated in Venus orbit from 2006  
51 661 to 2014 (Korablev et al, 2012). A double-range AOTF for the spectral range of  
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662 0.7-4.1  $\mu\text{m}$  serving as a prototype to the LIS and ISEM's AOTFs was developed  
 663 for the Phobos Grunt mission (Leroi et al, 2009).

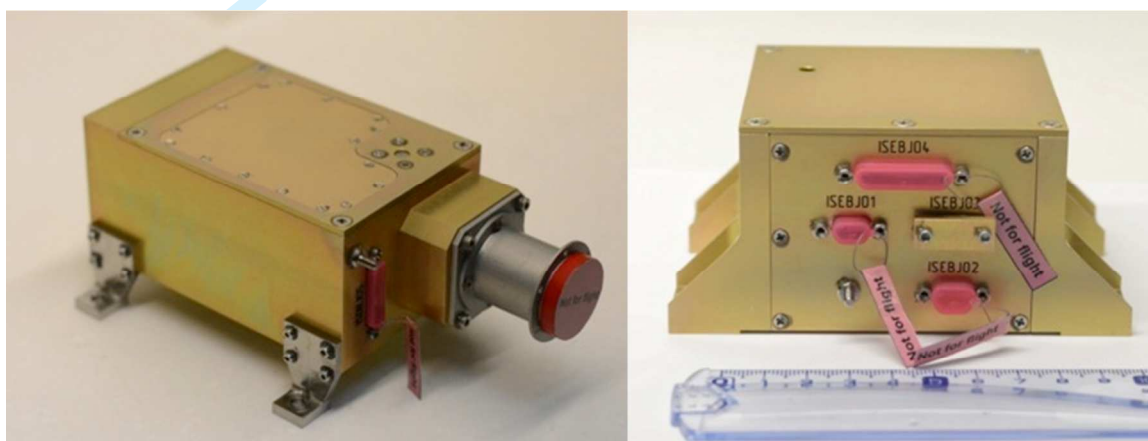
664

665 In order to reduce the influence of the extreme temperature conditions at the  
 666 rover's mast on the electronics, the instrument is implemented as two separate  
 667 boxes, a mast-mounted Optical Box (OB) and the Electronics Box (EB) (Fig. 4).

668 The thermally stabilized EB is mounted within the rover's body.

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672 Figure 4. The ISEM instrument, the photographs of the optical and electronics  
 673 box structural and thermal models.

674

675

676 Table 1. ISEM main characteristics and resources

Parameter	Value
Spectral range	1.15-3.3 $\mu\text{m}$
Spectral resolution	better than $25 \text{ cm}^{-1}$ 3.3 nm at 1.15 $\mu\text{m}$ , 16 nm at 2.5 $\mu\text{m}$ , 28 nm at 3.3 $\mu\text{m}$
FOV	1.3°
Temperature range, operational	-45°C...+30°C (Optical box, OB) -40°C...+50°C (Electronics box, EB)
Temperature range, non-operational	-60°...+60°C (OB and EB) -130°...+60°C for OB pending final confirmation
AOTF	
Material	TeO <sub>2</sub>
Spectral range	1.15-3.39 $\mu\text{m}$

Parameter	Value
Effectiveness	>50 % (in polarized light)
Aperture	Ø 5 mm, 5°×5°;
Mean RF power	5 W
RF frequency range	23-82 MHz
Detector	InAs photodiode, Ø1 mm, 1-3.45 µm Teledyne Judson Technologies J12TE3-66D-R01M, 3-stage Peltier cooler
ADC	16-bit
Number of points per spectral range	variable, by default 1024 for one observation
Data volume	variable, by default 20 Kbit for one observation
Data/command interface	RS-422
Powers supply voltage	28 V
Power consumption, W	
Peak	14
Average	11.5
Standby	9
Dimensions	160×80×96 mm OB 116×84×55 mm EB
Mass, overall	1.740 kg
OB	0.690 kg
EB	0.560 kg
Calibration target (ISEM part)	0.014 kg
Harness	0.476 kg including 20% margin (TAS-I data)

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### 680 3.1 The Optical Box

681 Though most of the electronics are in the EB, some electronics such as the

682 detector's preamplifier, and the RF conditioning electronics remain in the OB.

683 Neither the weak photodiode current, nor the power RF can be transmitted via the

684 5-m harness. The block-diagram of ISEM is presented in Fig. 5.

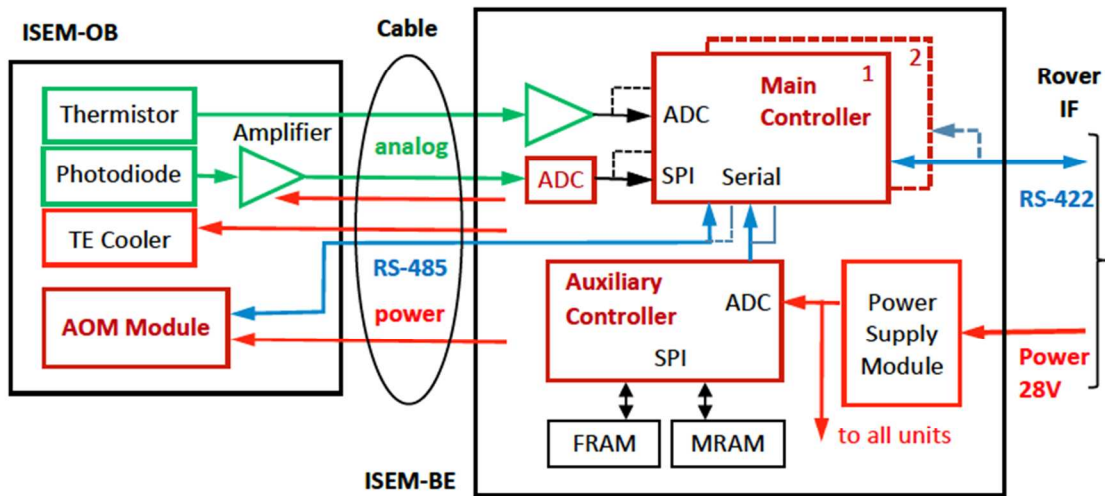
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686 The OB contains all the optical elements, the AOTF with associated electronics

687 (ultrasound frequency synthesizer and amplifier boards), the photo detector and

688 the photo detector board (Fig. 6).

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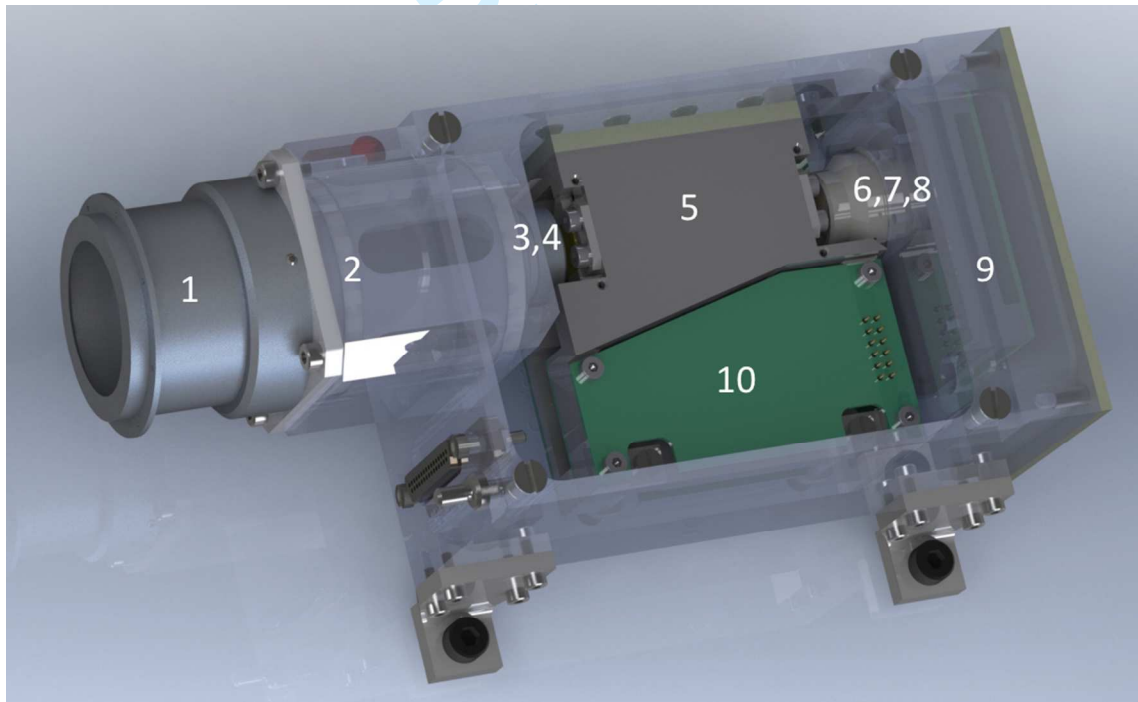


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Figure 5. The block diagram of the ISEM control electronics.



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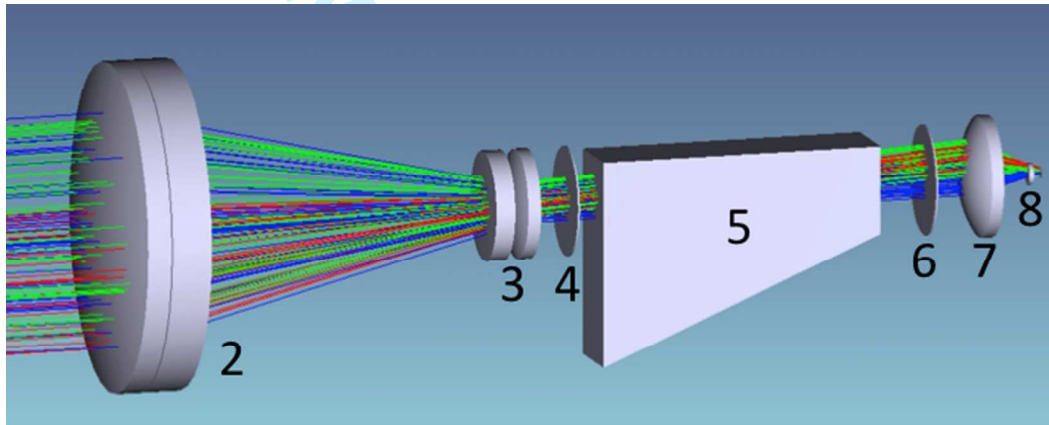
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Figure 6. The 3D model open view of the ISEM Optical Box. The main elements are visible: 1- baffle, 2- entry optics; 3, 7- AOTF collimating optics; 4, 6- polarizers; 5- AOTF crystal; 8- detector; 9- detector's preamplifier; 10- AOTF RF

697 proximity electronics.

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699 The spectrometer is built following a standard layout for an AOTF spectrometer,  
700 with the AOTF in the path of a quasi-parallel beam. The optical scheme is based  
701 on a Galileo system with remote pupil built using CaF<sub>2</sub> and ZnSe lenses and it is  
702 presented in Fig. 7. The image is transferred through the optical system, and the  
703 field-of-view (FOV) of 1.3° is formed on the detector sensitive area. The  
704 achromatic lens entry telescope (1) has the aperture of 25 mm. The AOTF crystal  
705 (5) is placed in a quasi-parallel beam between collimating lenses (3, 7), and a pair  
706 of polarizers (4, 6); the output collimating lens (8) serves also as a focusing optic  
707 for the detector (8).  
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709

710 Figure 7. ISEM optical scheme. The numbering is the same as in Fig. 6. 2-  
711 foreoptics; 3, 7- AOTF collimating optics; 4, 6- polarizers; 5- AOTF crystal; 8-  
712 sensitive area of the detector.

713

714 A wide-angle AOTF from NII Micropribor in Zelenograd, Russia is manufactured  
715 on the base of a tellurium dioxide crystal. The ultrasound frequency range of 23-  
716 82 MHz provides a spectral range from 1.15  $\mu\text{m}$  to 3.3  $\mu\text{m}$  with two  
717 piezotransducers. The transducers operate in the sub-bands of 23-42 MHz and 42-  
718 82 MHz. The crystal cut-off angle is 12.5° in the (110) crystallographic plane.  
719 The anisotropic Bragg diffraction regime is used. The incident optical radiation  
720 has ordinary polarization and the diffracted optical beam has the extraordinary  
721 polarization. The angle between the passed and diffracted optical beams is 6° at  
722 the output of the AO crystal. A pair of polarizers with crossed polarizing planes is  
723 used to filter out the non-desired zero diffraction order.

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8 725 The AOTF and its electronics are assembled in a single functional unit, which  
9 726 includes the acousto-optic cell, the polarizers, a proximity RF matching board, an  
10 727 RF synthesizer, the driver of the AO crystal (the RF power amplifier), and a  
11 728 dedicated internal microcontroller, which communicates with the Main Controller  
12 729 (MC) in the EB and controls all the functions of the acousto-optic module  
13 730 (AOM). Commands define magnitude and frequency of the RF signal applied, as  
14 731 well as the RF driver ON/OFF states. Within the RF range of 23-82 MHz, the  
15 732 minimum step of frequency sweeping is 10 kHz (5900 frequency points). The RF  
16 733 amplitude is may be set at one of 16 even levels. The AOM transmits back to the  
17 734 MC a few housekeeping parameters, such as measured RF voltage, AO crystal  
18 735 temperature, etc. The MC and the AOM are connected via an RS-485 interface  
19 736 running at a speed of 115.2 Kbit/s.

20 737

21 738 During the measurement, the RF level is being alternated between ON and OFF  
22 739 states, with the cadence being defined by integration time. The measured signal is  
23 740 then processed as the AC allowing to remove offsets caused by the detector's dark  
24 741 current and stray light, and improving the dynamic range of the instrument.

25 742

26 743 The detector is a single-pixel InAs thermo-electrically cooled photodiode. A  
27 744 detector module J12TE3-66D-R01M from Teledyne Judson Technologies is used.  
28 745 The built-in three-stage thermo-electric Peltier cooler maintains a detector  
29 746 temperature about 90°C below that of the hot side. For the ISEM OB operating in  
30 747 the range from -10° to +30°C, it results in the detector's temperature ranging  
31 748 between -100° and -60°C, the corresponding detector's shunt resistance is  
32 749 therefore 60-400 kOhms. The temperature of the sensitive area is monitored by a  
33 750 built-in thermistor.

34 751

35 752 The detector's photocurrent is amplified with a two-stage circuit. The first stage is  
36 753 a trans-impedance amplifier, AC-coupled (1 s time constant) to the second stage.  
37 754 The second stage has a gain of 83 and a time constant of 80 μs and is based on an  
38 755 ADA4610 (Analog Devices) operational amplifier characterized by low current  
39 756 and voltage noise (50 fA Hz<sup>-1/2</sup>, 7.5 nV Hz<sup>-1/2</sup>). Its output signal is transmitted via  
40 757 the harness to the Electronic Box.

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42 759 **3.2 The Electronics Box**

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7 760 The EB is mounted inside the rover at the rear balcony, and it is thermally  
8 761 stabilized. It includes the ADC, the main and auxiliary controllers, power  
9 762 conditioning (power supply unit, PSU), and the interface and bridge boards,  
10 763 which support RS-422 communication between the ISEM and the Rover data and  
11 764 command system.  
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13 765  
14 766 The two ISEM blocks are connected via a 5-m harness running from the inside of  
15 767 the Rover to the mast. It is fabricated by Thales Alenia Space-Italy (TAS-I). The  
16 768 cable includes the analog signals from the detector and the thermal sensor, the  
17 769 power supply lines and a digital RS-485 connection to control the AOFT RF  
20 770 synthesizer and the power amplifier.  
21

22 771  
23 772 The Main Controller (MC) located in the EB commands the operation of all  
24 773 modules of the instrument. It uses a MSP430FR5739 Texas Instruments circuit  
25 774 running at 19.68 MHz. The controller is equipped with 1 Kbyte of RAM and 16  
26 775 Kbyte of ferroelectric memory (FRAM) used for program and data storage.  
27 776 Compared to commonly used FLASH memory, the FRAM has better radiation  
28 777 immunity. A 10-bit internal ADC of the Main controller is used to digitize the  
29 778 signal from the detector's thermistor.  
30 779

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32 780 The main ADC serves to digitize the signal from the detector, amplified in the OB.  
33 781 A 16-bit ADC (ADS8320, Texas Instruments) is used. The full-scale range of the  
34 782 ADC is 2.9 V and the peak-to-peak noise is 3 LSB (22  $\mu$ V RMS). The Main  
35 783 Controller receives the ADC output data via a serial interface operating at 2  
36 784 Mbit/s. The preamplifier and the ADC contribute little to the total noise in the  
37 785 signal path, thus the Johnson noise of the detector controls the limit of signal  
38 786 detection.  
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41 788 The Auxiliary controller, based on C8051F121 Silicon Laboratories circuit  
42 789 operating at 12.25 MHz, incorporates a 12-bit ADC, and serves to monitor power  
43 790 supply voltages and to dispatch the operation of different memory devices. In all,  
44 791 ISEM employs two built-in FLASH memories within the microcontroller, and  
45 792 two external chips of ferroelectric (FRAM) and magneto-resistive (MRAM)  
46 793 memory.  
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7 795 The two microcontrollers hold four identical copies of their firmware in program  
8 796 memories. At the start of operation, the controllers check the copies, repair  
9 797 damaged ones and run a validation program. In order to increase its reliability,  
10 798 ISEM contains two redundant main controller units. After turning on, first the  
11 799 MC1 is powered. Its sequence, if performed correctly, commands the PSU to keep  
12 800 power at MC1. If not commanded in 5 seconds, the PSU automatically powers on  
13 801 the MC2, and so on.  
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17 803 The rover Onboard Computer (OBC) controls ISEM operations via RS-422  
18 804 interface. There are two (nominal and redundant) RS-422 links. Communication  
19 805 via only one of the links is available at a given time. The baud rate is 112.179  
20 806  $\pm 1\%$  Kbit/s.  
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### 23 808 **3.3 The Calibration Target**

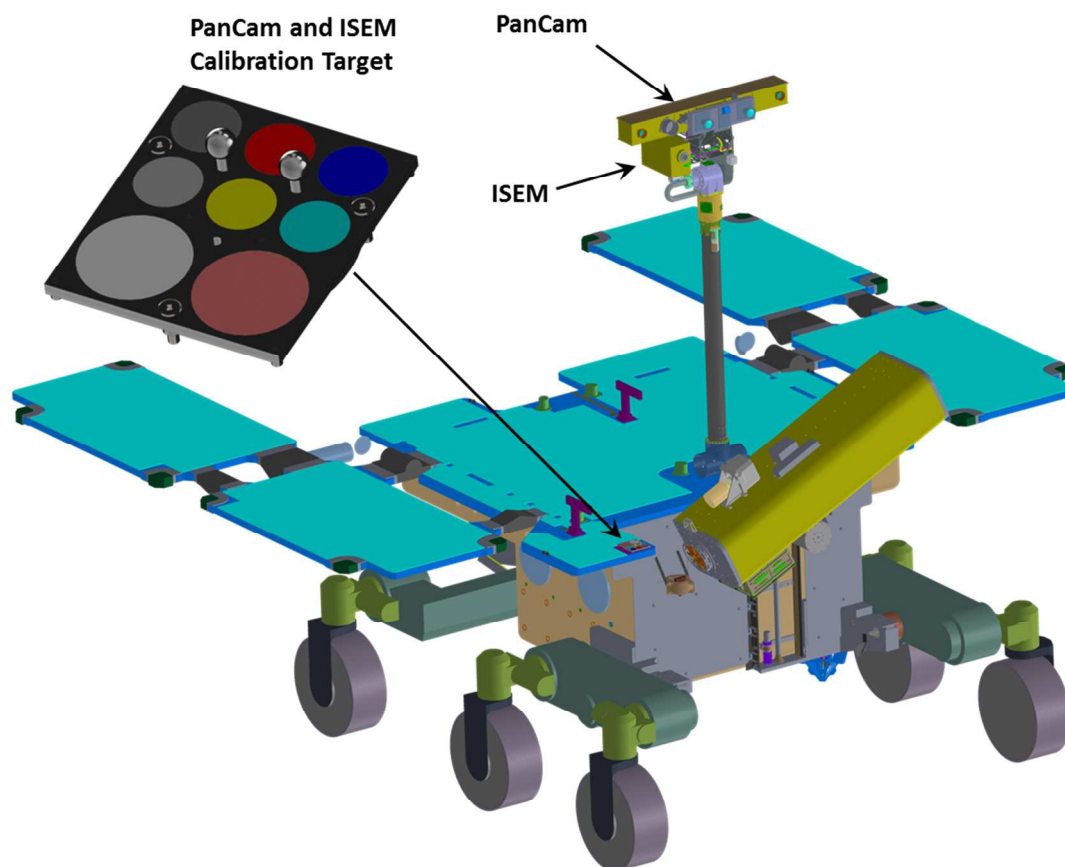
24  
25 809 In order to determine the incident solar illumination spectrum for deriving I/F data,  
26 810 and to verify the in-flight performance and stability of the instrument, ISEM will  
27 811 observe the radiometric calibration target prior to each measurement. The target is  
28 812 used for the in-flight radiometric calibration of both PanCam and ISEM, and will  
29 813 be located on the front deck of the rover as shown in Fig. 8. At this location it will  
30 814 be interrogated by ISEM from a distance of 1.1 m at an emittance angle of 24°.

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32 815 The calibration target occupies an area of  $67 \times 76 \text{ mm}^2$  and has a mass of 40 g.  
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36 817 The target includes 8 stained glass diffuse reflectance calibration patches with  
37 818 different spectral reflectance properties. Two of these calibration patches will be  
38 819 used by ISEM - the “white” which has a reflectance near 100% in the 0.4-3.0  $\mu\text{m}$   
39 820 spectral range, and the multiband patch, which has distinct spectral features. The  
40 821 white patch is manufactured from Pyroceram provided by the Vavilov State  
41 822 Optical Institute in St. Petersburg and the multiband patch is manufactured from  
42 823 WCT-2065 — a rare earth doped glass manufactured by Schott and supplied by  
43 824 Avian Technologies in the USA. The calibration patches will be calibrated for  
44 825 absolute total hemispherical reflectance and Bidirectional Reflectance  
45 826 Distribution Function (BRDF) and all measurement will be traceable to  
46 827 photometric standards.  
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50 829 Dust deposition on the radiometric calibration target during the ExoMars mission  
51 830 will be accounted for in the data processing by developing a model of the  
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7 831 calibration target and dust system, building on the results of previous missions,  
8 832 measurements of settling rates on the rover panels, solar arrays, etc. (Kinch et al.  
9 833 2007, 2015), and from PanCam calibration results.  
10 834



40 835  
41 836 Figure 8. PanCam and ISEM calibration target and its location on the rover.  
42 837 Larger circles ( $\text{\O}30$  mm) are for both ISEM and PanCam. The smaller circles  
43 838 ( $\text{\O}18$  mm) and the shadow posts will be used by PanCam only.  
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#### 47 841 **4 Measurement Scenario**

##### 48 842 **4.1 The Experiment Cycle**

49 843 The Experiment Cycle (EC) consists of one spectrum measurement by ISEM. It is  
50 844 explained in Figure 9.

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7 846 In order to eliminate the dark signal of the detector and any possible stray light  
8 847 signal, the instrument measures photocurrents when the AOTF RF is OFF (the  
9 848 dark signal) and when it is ON (the full signal). A subtraction of the two values  
10 849 gives the true signal corresponding to the spectrally filtered radiation on the  
11 850 detector.

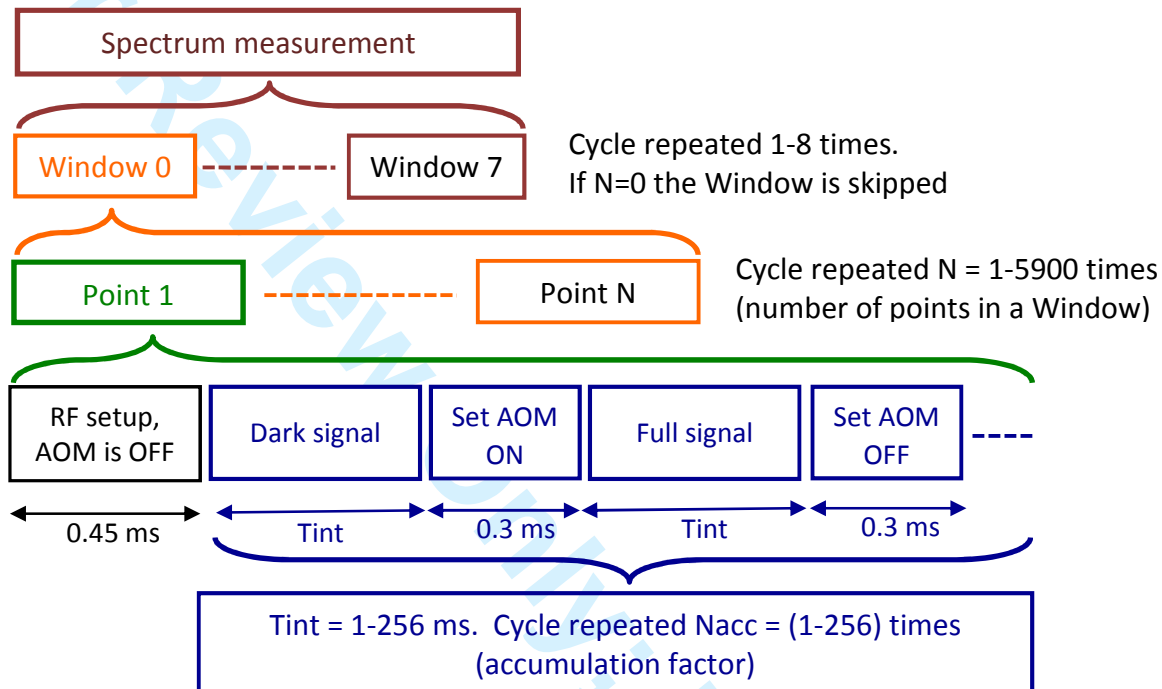
12 851  
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14 852 During both the ON and OFF states, the main ADC continuously samples signals  
15 853 at a 20 kHz rate. The MC stacks all the measured values and makes the ON–OFF  
16 854 subtraction. This elementary measurement can be repeated up to 256 times  
17 855 (“accumulation factor” parameter), effectively increasing the integration time. All  
18 856 resulting values are stacked as well. Each elementary measurement includes a  
19 857 dead time of ~0.6 ms needed to turn the AOTF ON or OFF. These overheads are  
20 858 mostly the result of remote commanding of the acousto-optic module in the OB,  
21 859 but also include two 0.12-ms transients in the AOTF. An additional dead time of  
22 860 0.45 ms is needed to change the RF. The integration time, variable between 1 and  
23 861 256 ms and defining the ON/OFF timing, and the number of repeated ON/OFF  
24 862 cycles for a given RF value (the accumulation factor) are set by a command.  
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32 864 As discussed above, the measurement time is a critical factor. To achieve  
33 865 flexibility in spectral sampling, up to 8 “windows” can be defined for one  
34 866 measurement of a spectrum. Every window can be placed anywhere in the AOM  
35 867 frequency range and it has its own number of spectrum points, frequency steps,  
36 868 and accumulation factors.  
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39 869  
40 870 During an elementary measurement the 16-bit ADC readings add up to a 24-bit  
41 871 value, and further accumulation, up to 256 times, results in a 32-bit value for each  
42 872 spectrum point. Only 16 bits out of it are used, three options being (i) to transfer  
43 873 merely the 16 LSBs, or (ii) 16 bits starting from a certain bit in order to remove  
44 874 noise, or (iii) to normalize within one spectrum so that the maximum fits exactly  
45 875 within 16 bits. In the latter case the risk is that an errant spike distorts the entire  
46 876 spectrum.  
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49 877  
50 878 The exchange of commands between ISEM and the OBC proceeds as follows.  
51 879 The OBC transmits a command (TC) and Rover Elapsed Time (RET). The TC  
52 880 defines ISEM operation parameters and triggers the spectrum measurement. The  
53 881 RET command synchronizes ISEM’s internal time. In the course of operation, the  
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882 ISEM sends scientific and housekeeping packets to the OBC and these data are  
 883 stored into the OBC memory.  
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887 Figure 9. A diagram showing timing and breakdown of one spectrum  
 888 measurement.

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#### 890 4.2 Operations on the surface

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892 ISEM operations over the course of the Reference Surface Mission (RSM) are  
 893 intimately connected those of the mast and PanCam. All operations are to be  
 894 guided by the mast's pointing. Three basic operating regimes have been  
 895 identified:

896

897 1) Operations to support sampling.

898 ISEM will work in parallel with PanCam, as shown in Fig. 2. First the WAC  
 899 observes an area panoramically, and then the HRC is pointed to the most  
 900 interesting spots. The spot must be illuminated by direct sunlight. At least one

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7 901 ISEM spectrum is planned be acquired together with every HRC image, and  
8 902 ISEM measurements may be also conducted separately. The ISEM data will be  
9 903 used to characterize the mineral composition. Sometimes ISEM will be used  
10 904 together with the WAC “geology” filters; however, full analysis with the PanCam  
11 905 WAC geological filters is slower and produces much more data so this capability  
12 906 cannot be used regularly.  
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16 907  
17 908 2) Operations to assess the geology value of targets

18 909 A number of geologically interesting scenes may be distant from the planned  
19 910 rover track or inaccessible for rover sampling, such as crater rims, outcrops, etc.  
20 911 Remote characterization is the only possibility in this case. ISEM will be operated  
21 912 in the same manner as in the first case. There is no theoretical limit on the  
22 913 sounding range provided the illumination conditions (including the phase angle)  
23 914 are acceptable. For instance, illuminated hill slopes may be observed from dozens  
24 915 of km. A practical limit is determined by the ISEM’s FOV, a ~1-meter spot is  
25 916 observed from 44-meter distance. This mode could be used for rover-orbiters  
26 917 coordinated measurements and validations. For a favorable scene ISEM may  
27 918 provide cross-validation over the area covering multiple CRISM pixels.  
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32 919  
33 920 3) Dedicated environmental operations

34 921 ISEM data may be used to characterize atmospheric humidity and aerosol loading.  
35 922 Some information will be obtained as a by-product of geological measurements  
36 923 intermitted with observations of the calibration target. The information about the  
37 924 gaseous atmospheric absorption will be assured by dense spectral sampling of the  
38 925 appropriate spectral intervals. Aerosols are best characterized by cross sky  
39 926 observations. PanCam WAC plans direct Sun observations and cross sky viewing  
40 927 atmospheric campaigns. As discussed in Section 2.5, the ISEM capability to  
41 928 observe the direct Sun is not confirmed. At the same time, following preliminary  
42 929 laboratory characterization we do not expect any operational constraints placed by  
43 930 ISEM on PanCam solar imaging.  
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49 932 In the case of cross sky imaging the ISEM line of sight will be oriented above the  
50 933 horizon, and a special ISEM data acquisition sequence will be implemented. The  
51 934 HRC measurements are not needed, and ISEM will observe in parallel with WAC.  
52 935 The elevation and phase angle for these measurements needs to be carefully  
53 936 planned, targeting as much as possible range of phase angles for WAC and  
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7 937 detectable scattered signal in the IR for ISEM.

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9 **4.2 Resources required**

10 ISEM's data volume is relatively small. One spectrum is only two kilobytes in  
11 size (see Table 1 and Section 3). The factor most affecting the rover operations is  
12 the measurement time. Fixed surface platforms are better suited for lengthy  
13 operations, generally allowing significant accumulation of measurements.  
14 However, sequential acquisition of spectra effectively increases the recording  
15 time, and, as a result, required measurement times become an important factor for  
16 the rover operation cycle.  
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21 The allocated duration of measurements is from 2.5 min (minimum case) to 8.0  
22 minutes (optimal case) for a single target. During the whole sequence, ISEM will  
23 consume ~12W, requiring an energy of 0.5-4.4 Watts per hour. As described in  
24 section 3.3 the length of the spectrum, or the number of points in the spectrum,  
25 can be selected. For a "geology" target we will always measure across the full  
26 spectral range of ISEM, which may be divided into 1-8 intervals with different  
27 sampling. The total number of points in one spectrum will vary between 256 and  
28 1024. Within the spectral range, some portions of the spectrum will be measured  
29 more accurately, where e.g. a narrow mineral absorption feature or important  
30 atmospheric absorption are expected, while other portions of smooth continuous  
31 character can be undersampled.  
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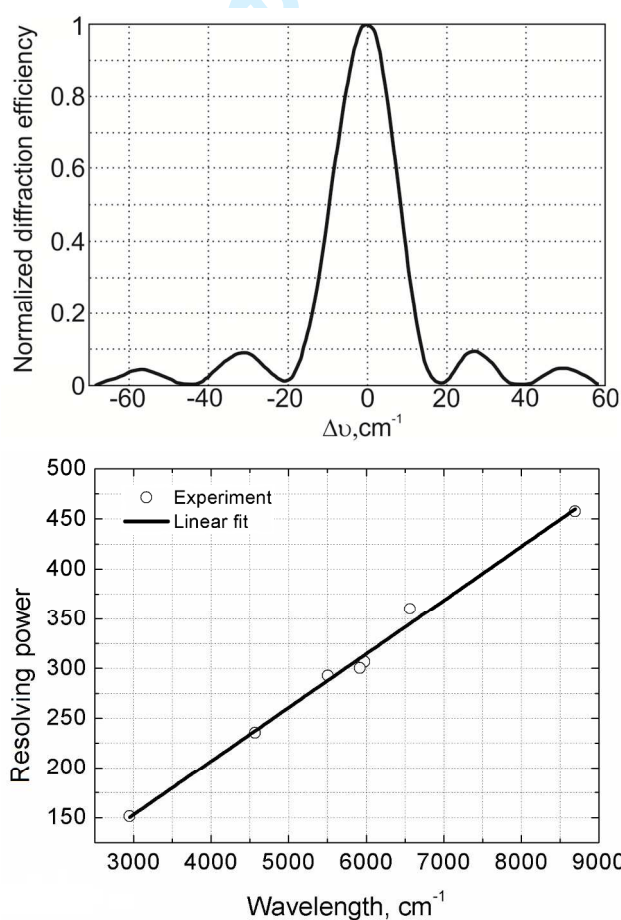
38 **4.3 Measurement performance, examples, comparison with state of the art**

39 As the full working prototype of ISEM has not been built yet, the performance of  
40 the instrument was verified using the closest prototype, the qualification model  
41 (QM) of LIS for the Luna 25 lander. The ISEM hardware involves some  
42 improvements with respect to LIS, including better optical throughput (and more  
43 sensitive detector featuring the three-stage Peltier cooler. The spectral properties  
44 of the AOTF remain the same.  
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48 967

49 Spectral range and spectral resolution. The bounds of the spectral range were  
50 verified using He-Ne 1.152  $\mu\text{m}$  and 3.39  $\mu\text{m}$  laser lines. The theoretical shape of  
51 an AOTF spectral instrument function is close to  $\text{sinc}^2x$ . Side lobes of this  
52 function modify the measured spectrum, but this could be readily accounted for  
53 once the bandpass is properly characterized. More dangerous are distant side  
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lobes, which manifest themselves as a sort of stray light and may reduce the apparent depth of spectral features. More discussion on this aspect of AOTF characterization may be found in Korablev et al. (2013). The measured band pass transmission function of the ISEM prototype is shown in Fig. 10a. The evolution of the resolving power  $\lambda/\Delta\lambda$  through the spectral range measured at multiple wavelengths using light sources with sharp emission features (different He-Ne laser, single-mode laser diodes, low-pressure gas-filled lamps) is shown in Fig. 10b.



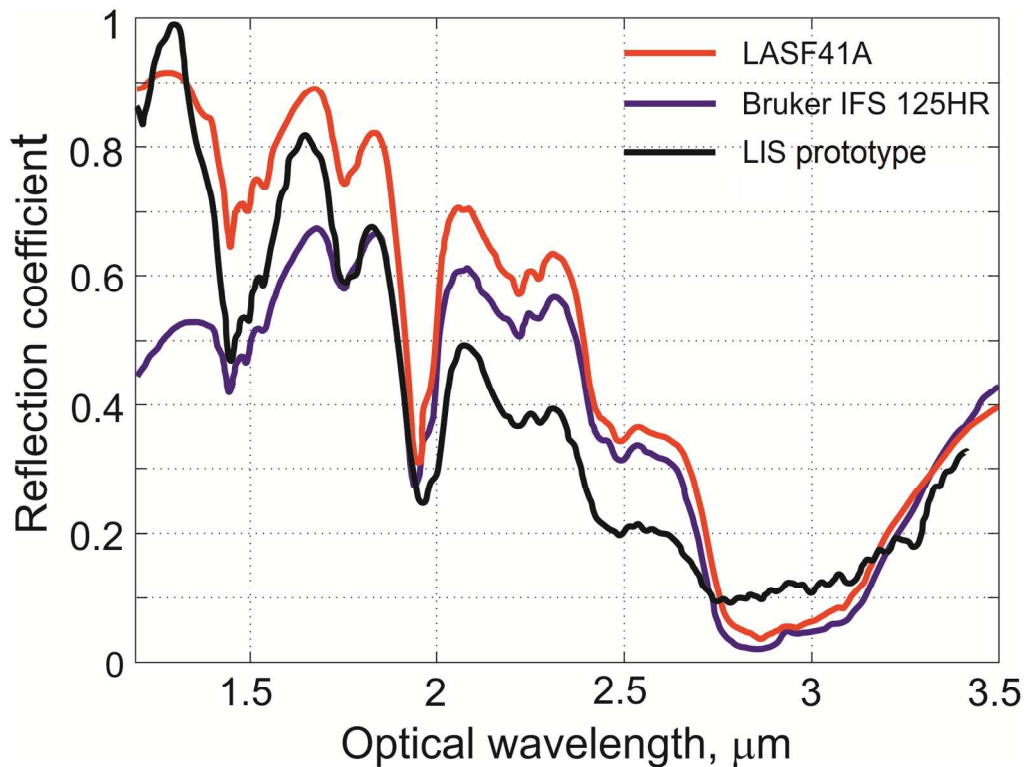
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985 Figure 10. (a) Normalized transmission function of the instrument measured at  
 986 1.523  $\mu\text{m}$  laser line. The FWHM is 19  $\text{cm}^{-1}$  (2.7 nm). (b) The resolving power  
 987 measures through the full spectral range.

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7 989 A simulation measurement of a mineral spectrum of a Mars analogue by LIS  
8 990 prototype is presented in Fig. 11. Crystalline gypsum was manually ground and  
9 991 sieved to obtain ~1-mm grains, and the resulting sample was illuminated with a  
10 992 150-W halogen lamp from a distance of 18 cm. The full time to measure this  
11 993 spectrum was 30 min, and the effective exposure to measure one spectral point  
12 994 was 1 s in the long-wave AOTF sub-range, and 0.6 s in the short-wave sub-range.  
13 995 The spectrum shown is normalized by a spectrum of a Lambertian screen (metal  
14 996 surface covered by special reflective coating). The differences between the library  
15 997 and measured spectra might be attributed to different origin of the samples; the  
16 998 difference between the FTS and ISEM spectra are likely due to different  
17 999 observation geometry.



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47 1001 Figure 11. A reflectance spectrum of crystalline gypsum measured by the closest  
48 1002 equivalent of ISEM: LIS qualification prototype, and a laboratory Fourier-  
49 1003 spectrometer (Bruker), compared to a CRISM library spectrum (LASF41A).  
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51 1005 Signal-to-noise ratio estimation. The signal-to-noise ratio (S/N) of ISEM and its  
52 1006 LIS prototype is limited by the detector background noise (see section 3.1), so we

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7 1007 may estimate the S/N for ISEM based on LIS prototype testing. With 3 ms  
8 1008 integration time repeated 6 times per point LIS provides the S/N~10 in the center  
9 1009 of the spectral range and 3-5 on the edges. This estimation using the reflected  
10 1010 sunlight and laboratory light sources involves many uncertainties, including  
11 1011 transparency of the terrestrial atmosphere, scaling to the Sun intensity at Mars  
12 1012 distance, and extrapolation. In turn there remains a margin on the duration of the  
13 1013 measurement. The real time to measure one spectral point is not 18 ms, but  
14 1014  $(3 \cdot 2 + 0.6) \cdot 6 + 0.45 = 40.05$  ms, because the light measurement is alternated with  
15 1015 dark measurement, and there are some extra delays in the system (see section 4.1.)  
16 1016 With 3 ms integration time the full spectrum is measured in 41 s, while the  
17 1017 measurements at the surface of Mars may last 2.5-8 min (see section 4.2) that  
18 1018 translates in sensitivity gain of 2-3.  
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24 1020 The improvements in ISEM with respect to LIS are i) the deeply cooled detector  
25 1021 with a peak sensitivity attaining a factor of 10 better, and ii) the larger throughput  
26 1022 of the entry optics and the AOTF: The LIS f-number is F:6; the improved ISEM  
27 1023 throughput is ~F:1.7, but vignetting in the AOTF results in effective throughput of  
28 1024 ~F:2. The resulting gain factor is therefore ~90. The detector's peak sensitivity is  
29 1025 at 3.3  $\mu\text{m}$  in both cases but the curve for the LIS detector is somewhat more flat.  
30 1026 The AOTF characteristics may vary as well. Using more conservative factor of 30  
31 1027 the resulting S/N of ISEM can be estimated as ~300 in the center of the range, and  
32 1028 100-150 at the edges.  
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38 1030 Estimation of detection capabilities. Starting from the S/N estimation above we  
39 1031 may estimate minimum detectable abundance for minerals discussed in section  
40 1032 2.4 (see Fig. 1). Precaution factors in the measurement time (2-3), and in  
41 1033 sensitivity (~3) allow sufficient margin to account for, e.g, unfavorable phase  
42 1034 angle, or other factors. We used spectra of different minerals of interest (Fig. 1 a-  
43 1035 f) , on the background spectra of "bright" and "dark" regions of Mars (Erard  
44 1036 1997) to simulate the reflectance of an admixture to the bulk spectrum of the  
45 1037 martian surface. The preliminary sensitivity estimates to detect the fraction of the  
46 1038 selected mineral are presented in Table 3. Given relatively high ISEM S/N ratio  
47 1039 deduced above, these estimates based on the relative depth of individual spectral  
48 1040 features are very conservative. A better sensitivity analysis is planned in the  
49 1041 future, involving analysis of martian mineral analogues, and multicomponent  
50 1042 analysis.  
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1044 Table. 3. Preliminary estimation of the ISEM sensitivity to detect different  
 1045 minerals and mineral groups from CRISM spectral library (Fig. 1) on the  
 1046 background of bright and dark region Mars spectra (Erard, 1997). Mineral  
 1047 features used to estimate the sensitivity are listed in the last column.

Mineral or mineral group	Bright region	Dark region	Comment
Kaolinite (Fig. 1a)	<5%	5-10%	1.4, 2.2, 2.4 $\mu\text{m}$
Montmorillonite (Fig. 1b)	5-10%	10-15%	1.4, 1.9, 2.2-2.3 $\mu\text{m}$
Serpentine (Fig. 1c)	5-10%	5-10%	1.4, 2, 2.35 $\mu\text{m}$
Carbonates (Fig. 1d)	5-10%	~10%	2.3, 2.5, 3.4 $\mu\text{m}$
Gypsum (Fig. 1e)	~5%	5-10%	1.4-1.6, 1.8, 2.2 $\mu\text{m}$
Kieserite (Fig. 1e)	<5%	~10%	Broad structure 1.4-2.4 $\mu\text{m}$
Jarosite (Fig. 1f)	<5%	5-10%	1.48, 1.75, 2.25 etc. $\mu\text{m}$

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1049 The issue of detection capabilities can be divided into two broad categories: (1)  
 1050 characterization of a target in the presence of windblown dust; and (2) detection  
 1051 of a phase or phases of interest in a multicomponent assemblages. There are a  
 1052 number of studies that have examined detection limits for various components  
 1053 that are relevant to this issue, and these are supplemented by ongoing studies by  
 1054 our team that are designed to specifically address this issue. The results of these  
 1055 studies are summarized in Table 4 below. Some selected examples are shown in  
 1056 Figures 12-14.

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1058 Detection limits depend on a wide variety of factors, beside end member  
 1059 abundances. These include factors such as relative grain size, the availability of  
 1060 diagnostic absorption bands, the relative intensity of diagnostic absorption bands,  
 1061 and physical properties such as whether phases of interest are physically mixed,  
 1062 or whether an obscuring phase, such as dust cover is present.

1063

1064 We have found that, at least for the variety of mixtures examined, detection limits  
 1065 are on the order of 10-25 wt.%. For some phases, such as certain organic  
 1066 compounds, detection limits are much lower (e.g, 1 wt.% adenine is detectable in  
 1067 the presence of 99 wt.% nontronite (an iron-bearing clay) or hematite (an iron  
 1068 oxide) (Figures 13 and 14)). Further extensive tests of mineral mixtures and



1069 detection limits are ongoing and planned.

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1071 Table 4. Summary of some studies concerning detection limits and discrimination

1072 of different phases relevant to the capabilities of ISEM

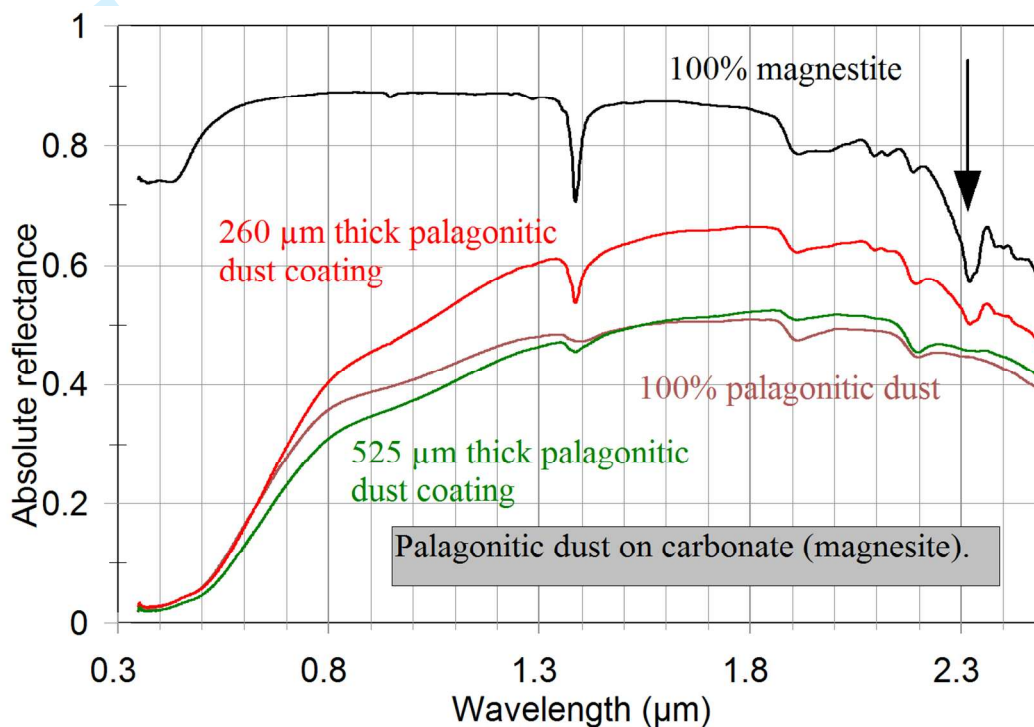
Mixture type	Detection limit	Characteristic spectral features	Source
Orthopyroxene + clinopyroxene	~15% of either phase	2 $\mu\text{m}$ region	Cloutis and Gaffey (1991)
Olivine + orthopyroxene	~20% olivine, 5% pyroxene	2 $\mu\text{m}$ region	Cloutis et al. (1986)
Orthopyroxene + clinopyroxene	~15% of either phase	2 $\mu\text{m}$ region	Cloutis and Gaffey (1991)
Pyroxene + palagonitic dust (spectral equivalent of Mars dust) or hematite	~10% pyroxene	2 $\mu\text{m}$ region	Cloutis and Bell (2004)
Carbonate + basalt	~20% carbonate	2.3-2.6 $\mu\text{m}$ region	Palomba et al. (2009) and this study
Carbonate + palagonitic dust (Fig. 12)	~15% carbonate	2.3-2.6 $\mu\text{m}$ region	Palomba et al. (2009) and this study
Palagonitic dust on carbonate (Fig. 12)	<~500 $\mu\text{m}$ thick palagonite coating required	2.3, 2.5 $\mu\text{m}$ regions	This study
Palagonitic dust on basalt	<~250 $\mu\text{m}$ thick palagonite coating required		This study
Kaolinite + illite	~20% of either phase	2.2-2.5 $\mu\text{m}$ region	This study
Gypsum + basalt	~5% gypsum	1.4, 1.9 $\mu\text{m}$ regions	This study
Gypsum + palagonitic dust	~5% gypsum	1.4, 1.9 $\mu\text{m}$ regions	This study
Pyroxene + volcanic glass	~15% pyroxene	2 $\mu\text{m}$ region	This study
Palagonitic dust + carbonate	~20% carbonate	2.3, 2.5 $\mu\text{m}$ regions	This study
Nontronite + adenine (Fig. 13)	~1% adenine	1.65, 2.2 $\mu\text{m}$ regions	This study
Hematite + adenine	~1% adenine	1.65, 2.2 $\mu\text{m}$	This study

(Fig. 14)		regions	
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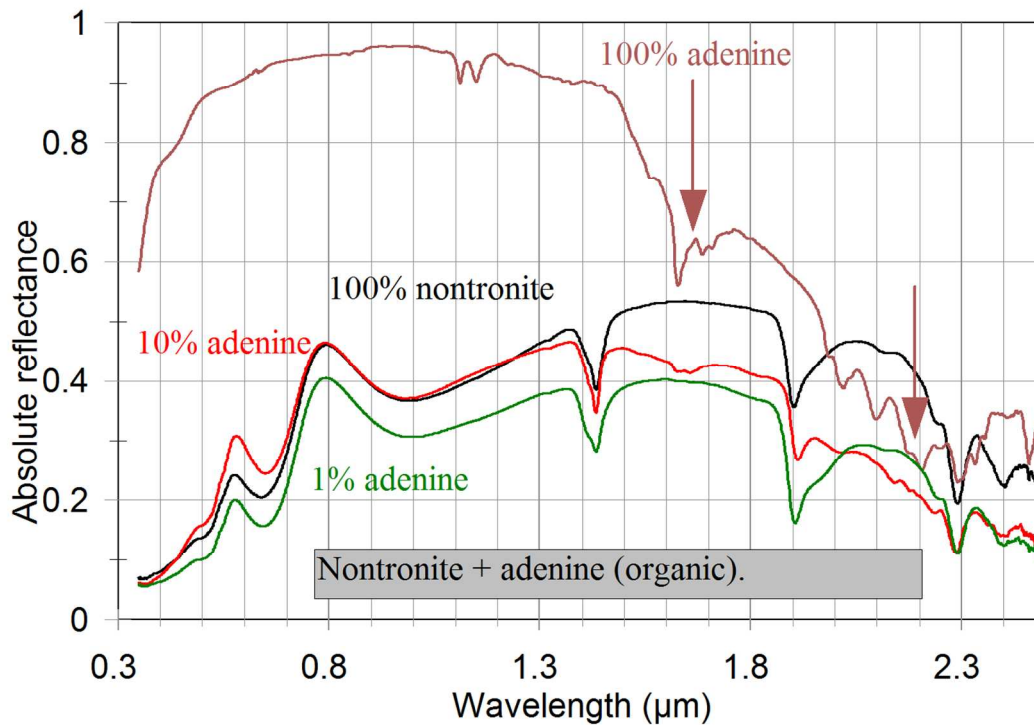
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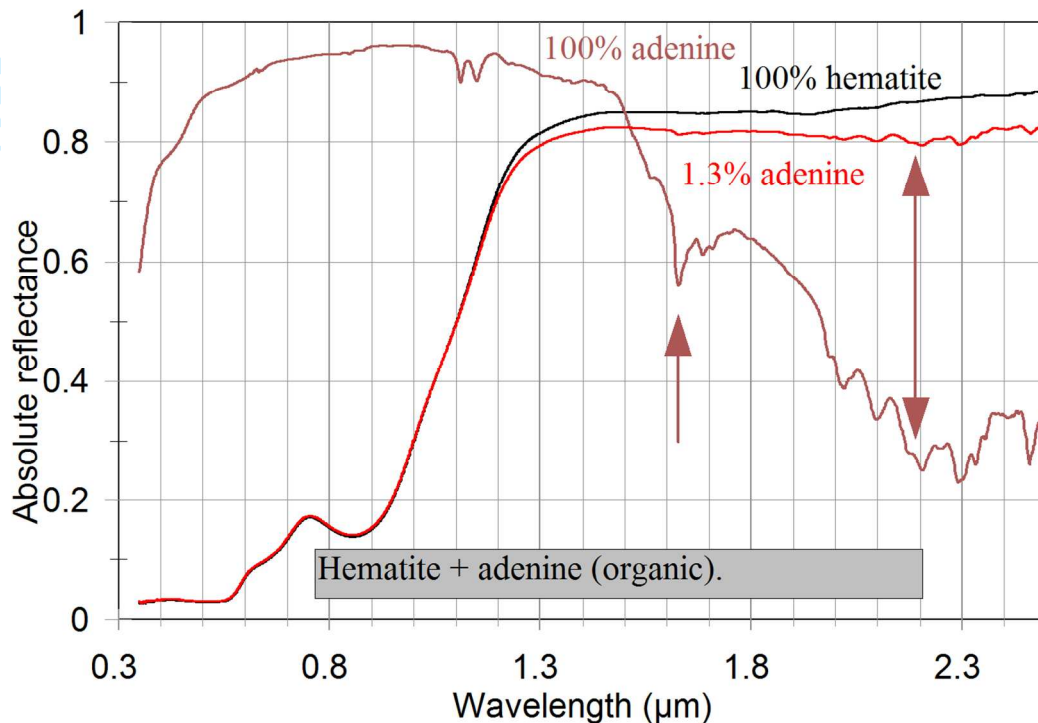
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**Figure 12.** Reflectance spectra of palagonitic dust sprinkled on a powdered carbonate (magnesite). The most diagnostic absorption band, near 2.33 μm is indicated by an arrow. Note that the band is clearly present with a 260 μm palagonitic dust coating. It is still present, but barely discernible for a 525 μm thick dust coating.



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**Figure 13.** Reflectance spectra of intimate mixtures of nontronite (an Fe-rich clay) and adenine (a component of DNA). The brown arrows indicate spectral regions where adenine absorption bands are still detectable when only 1 wt.% of adenine is present in the mixtures.



**Figure 14.** Reflectance spectra of intimate mixtures of hematite (an Fe-oxide) and adenine (a component of DNA). The brown arrows indicate spectral regions where adenine absorption bands are still detectable when only 1.3 wt.% of adenine is present in the mixtures.

#### 4.4. Environmental requirements and characterization

The placement of the ISEM optical head on the Rover mast with no possibility of thermal control imposes stringent requirements on survival temperature. The AOTF is a critical device involving bonding of highly anisotropic materials, such as  $\text{TeO}_2$  and  $\text{LiNbO}_3$ . Anisotropic thermal expansion makes the AOTF devices vulnerable to large temperature excursion, in particular at low temperatures. AOTF technology has demonstrated survival and even operation down to  $-130^\circ\text{C}$  and below (Leroi et al, 2009; Mantsevich et al, 2015). Several non-operational thermal cycles attaining  $-130^\circ\text{C}$  lower bound were also performed on two dedicated ISEM acousto-optic components; more testing to fully validate the technology is to be completed.

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7 1109 An important aspect of the AOTF spectrometer characterization is the thermal  
8 1110 calibration, because the optical heads are intended for operations across a  
9 1111 relatively wide temperature range. Some results of thermal characterization of the  
10 1112 acousto-optic module are described in Mantsevich et al. (2015). The operation of  
11 1113 the AO filters was tested in the range of  $-50^{\circ}$  to  $+40^{\circ}\text{C}$ . The temperature affects  
12 1114 the elastic properties of  $\text{TeO}_2$ , changing the ultrasound velocity and birefringence.  
13 1115 The variation of the  $\text{TeO}_2$  refraction coefficient as a function of temperature is  
14 1116 negligible. In turn, the change of the slow acoustic wave velocity causes a  
15 1117 noticeable shift of the dispersion curve, comparable with the AO filter pass band  
16 1118 in the operating temperature range. The characterization of the flight instrument  
17 1119 will therefore include calibration of the dispersion curve within the operational  
18 1120 temperature range.  
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## 1122 **5. Conclusions**

1123 Sampling from beneath the martian surface to reach and to analyze material  
1124 unaltered or minimally affected by cosmic radiation is a significant feature and  
1125 perhaps the strongest advantage of the ExoMars rover. The selection of the  
1126 sampling sites and, in more general terms, the remote reconnaissance and studies  
1127 of the landing site area, are a major part of the Rover's mission. ISEM offers a  
1128 method of mineralogical characterization using IR reflectance spectroscopy, a  
1129 technique well proven in orbital studies, and expected to be even more valuable  
1130 and precise at the local scale.  
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1132 In the coming year we aim to deliver field-compatible hardware and to participate  
1133 in tests at terrestrial analogue sites and in other joint validations of instrument  
1134 performance with the ExoMars team.  
1135

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