

Feasibility Study for a Calibration Satellite

To expand beyond LiteBIRD full success scientific goals

Mission



LCS Mission

Lissajous orbit like Planck and Herschel.

Formation flying of LCS and LB for periodic calibration.

Orbit maintenance:

 Manoeuvres required periodically. L2 presents low force gradients.



Concurrent Design Analysis

LCS has been pre-designed in IDR-UPM's Concurrent **Design Facility (CDF)**

It has been designed as a piggy-back of the LB minimizing the impact on its architecture. The concept of LCS can be extended to a microsatellite capable of reaching L2 autonomously by modifying mainly the propulsion and communication subsystems.

- **Main Specifications:**
- 3 years mission
- Size: 6U Cubesat

- Very low ΔV (~1 m/s per year).

Control strategy, formation, orbit and tolerance define a calibration: ΔV of 9 m/s for the entire mission.



Calibration

30°

15°

0°

-15°

Calibration strategy analysis

LCS must be inside LB's FoV during calibration and move away at finishing.

- 4 impulses for each calibration manoeuvre.

LiteBird scanning law and LCS relative position define:

- Total access time.
- Number of viewed LB instrument sensors.

The best option to place the LCS is close to the anti-Sun axis.

- The percentage of viewed sensors is around 14% but it reaches the 70% in that region for one day even when the LCS is at the same position.
- Average access time is uniform and around 30 s, regardless of the LCS position.



- Minimum distance between LCS and LB of 240 m;
- Calibration distance = 270 ± 3 m;
- Distance error determination < 13.5 cm;
- LB direction location error < 10'; LB orientation detection error < 1';
- LCS pointing error < 3';
- Power enough for 1 calibration per month;
- Minimum impact on LB (CubeSat deployer and 3 patch antennas).

Critical Technologies:

- Attitude control and determination: Sun sensors (Fine Sun Sensor) and star trackers (CubeSpace CubeStar).
- Metrology (relative position determination) and communications: RF
- ranging e.g. "Swift RelNav" by Tethers Unlimited.
- Propulsion: Cold gas thrusters (small and precise impulses with low consumption) e.g. GOMSPACE NanoProp.

Thermal Control:

Battery

Payload

Computer

Radio

- A preliminary LCS design implemented in the ESATAN® software shows a maximum temperature of 39 °C in the solar panels. - In the covered bandwidth by the telescope, this corresponds to a noise signal of 70 fW/pixel that is ten times below the LCS calibration signal at those frequencies.

Overall features:

- Total mass: 7.07 kg (dry mass) + 0.08 kg (propellant).
- Power: 2 orientable arrays of 6U-cubesat solar panels with a triplejunction technology by SpectroLabs.
- Total power generation: 57.2 W.

Estimated Characteristics:

- Volume: 10 cm x 10 cm x 20 cm
- Weight: 1.3 kg
- Power consumption: 50 W



- **Payload: Calibration Source**
- -30° -45 -60° -75°

Total access time according to the position of LCS relative to LB throughout 90 minutes.

75°

60°

45°

mean median fixed (from Yuto)

Hardware resources at IFCA

 Altamira supercomputer (RES) - 158 nodes 32 cores, 64GB RAM, Infiniband connection + 4 GPU nodes



- Development, fabrication, integration and operation of CMB experiments over more than 30 years (some of them in-house, others in collaboration with other institutes)
- Most recent ones: COSMOSOMAS (10-17 GHz, 1998-2008), VSA (33 GHz, 1999-2008),



- In order to assure a 1% of the budget assigned to systematics, the requirements are not far from the expected errors coming from EB = 0 at the spectra level.
- among frequencies are below 60%, even the most damaging case (in the sense If correlation of polarization angle error) is below the budget.
- Expected contribution to the component separation process using Internal Template Fitting and Neural Networks

- 7.4M CPU hours since May 2018
- **Grid** resources
- 4M (56M normalized) CPU hours in 2018
- OpenStack Cloud: http://portal.cloud.ifca.es/
 - Up to 7700 cores available Largest node: 48 cores, ~1TB RAM GPU nodes: 1 NVIDIA TitanX (donated by NVIDIA grant), 1 NVIDIA 1080ti, 10 NVIDIA 1080ti 90 nodes with In niband connection 1st overall resource provider (size) in EGI.eu Federated Cloud
 - 1st overall contributor (CPU-hours) in EGI.eu Federated Cloud 6M CPU hours during 2018
- Renewed **storage** system (2018)
- Cloud: Up to 300TB (Ceph) via object storage (S3-like) or block storage
- GPFS: 1PB GPFS storage + 1PB tape

Interest in DPC activities

The **funding expected** from the **National Research Program for Space** and **IFCA**, in relation to the **DPC**, can be used for:

- Computing (~ 500 cores)
- Storage (0.5 PB)
- 2 FTE



- and QOIJOTE (10-40 GHZ, 2012-)
- Future experiments: GroundBIRD (150-220 GHz), LSPE- STRIP (43, 90 GHz), TMS (10-20 GHz)
- Programming and planning observational programs
- Basic data processing and data reduction (calibration, characterization of noise properties, characterization of systematics, data flagging).
- Characterization of foregrounds (mainly at low frequencies, synchrotron and AME) both in intensity and in polarisation, and component separation
- Extraction of power spectra and cosmological-parameter estimation
- Statistical tools and techniques: MCMC, cross-correlations



Current and future QUIJOTE maps will be essential to clean higher-frequency maps (including LiteBIRD) from synchrotron emission:

- Synchrotron contributes to r=0.01 @ 200GHz in 70% of sky (Planck 2015 results X)
- Preliminary QUIJOTE results $\Box < \beta_{c} >_{D} = -3.00 \pm 0.05$, with dust-sych $\rho \sim 0.20 \pm 0.06$, with high variability in the sky



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