

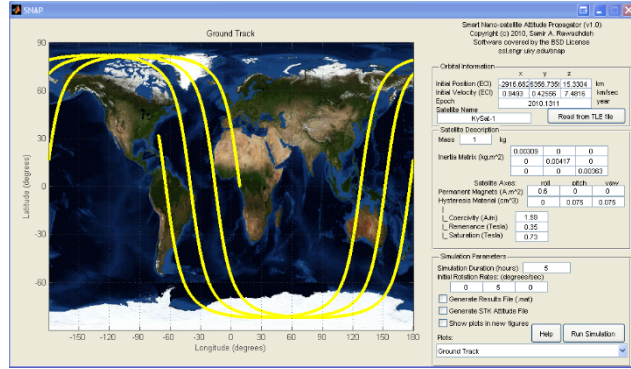


# Cubic-Centimeter Star Imager

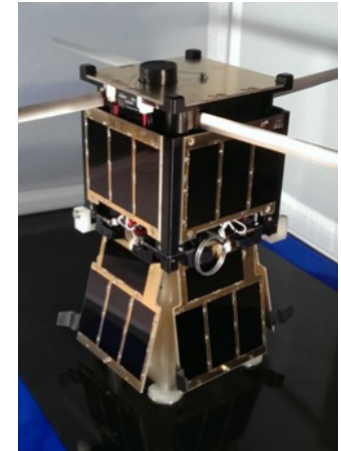
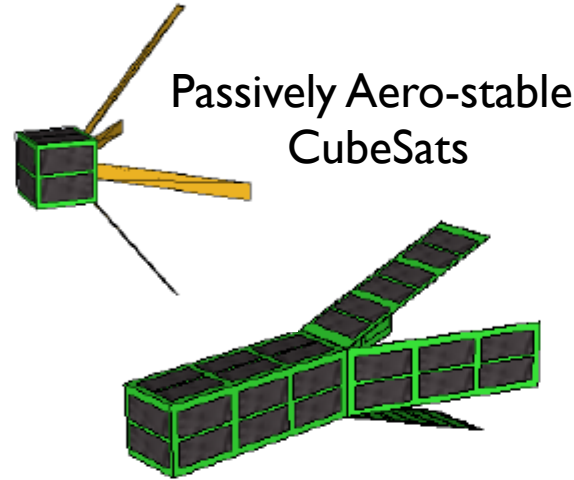
Samir A. Rawashdeh  
Assistant Professor  
Electrical and Computer Engineering  
University of Michigan - Dearborn

# Brief Personal Introduction

At Kentucky  
Space Systems Lab

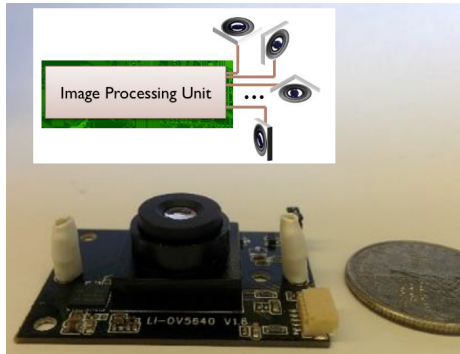


Smart Nanosatellite Attitude Propagator (SNAP)

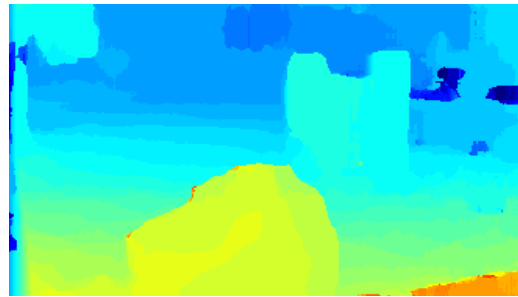


KySat-2 ADCS  
(Star Imaging)

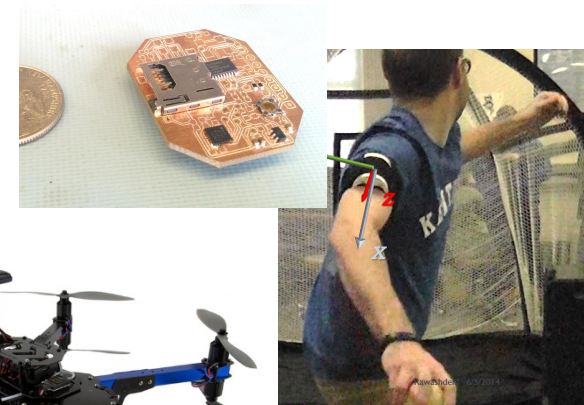
At Univ. of Michigan  
Dearborn  
(since Aug 2014)



Distributed Star Imaging for CubeSats



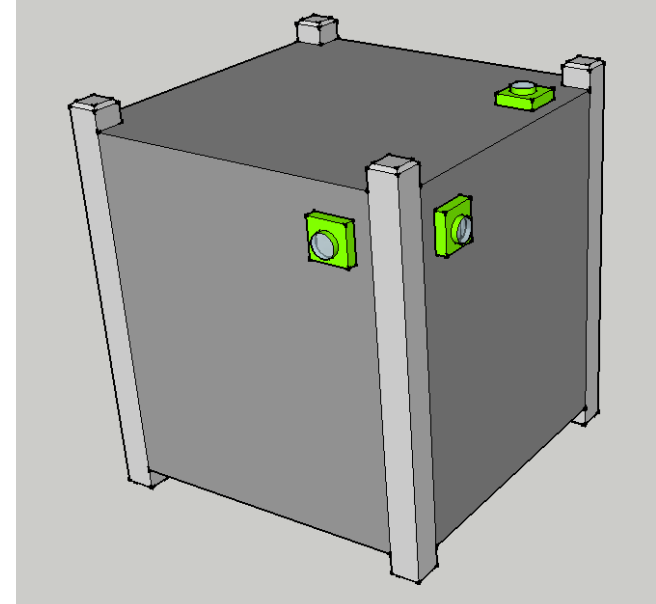
Visual Sense and Avoid for Micro Aerial Vehicles



Wearable Motion Capture for Shoulder Health

# Objective

Develop a star imager at the smart-phone scale.

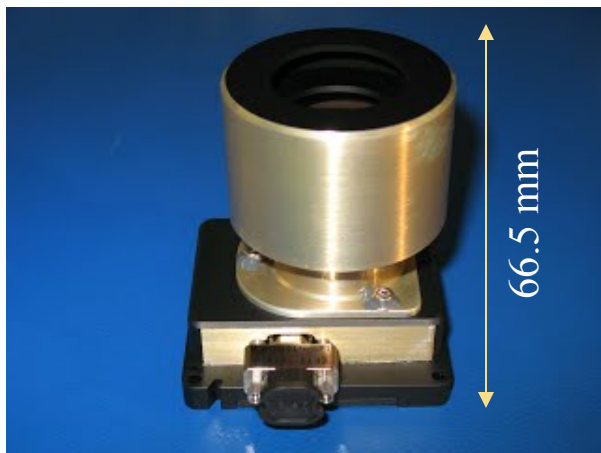
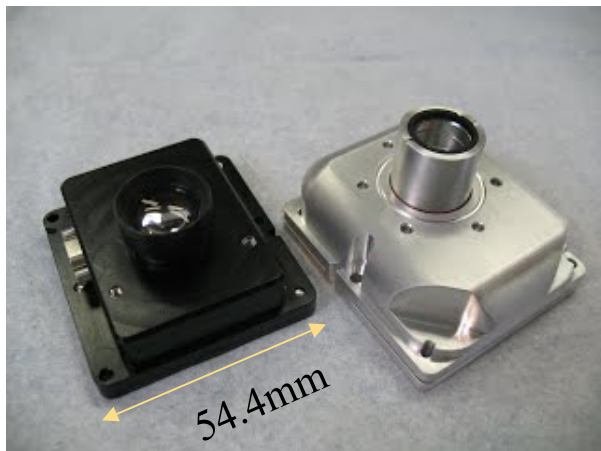


- ▶ Major challenges:
  - ▶ Physically small sensor: less **light** sensitivity
  - ▶ Small Lens: small aperture (less **light** enters the sensor)
  - ▶ Small Lens: typically short focal length (wide **field of view**)

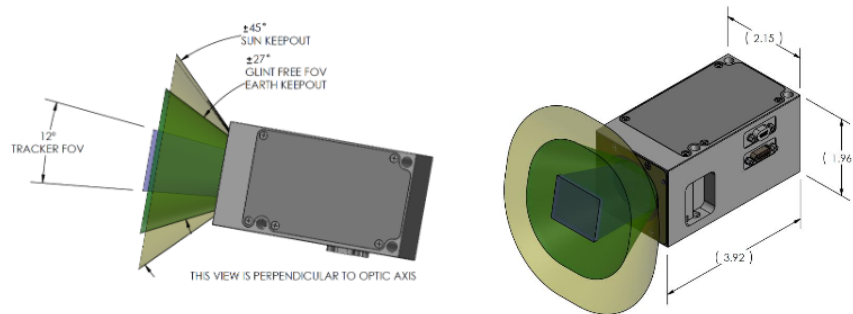
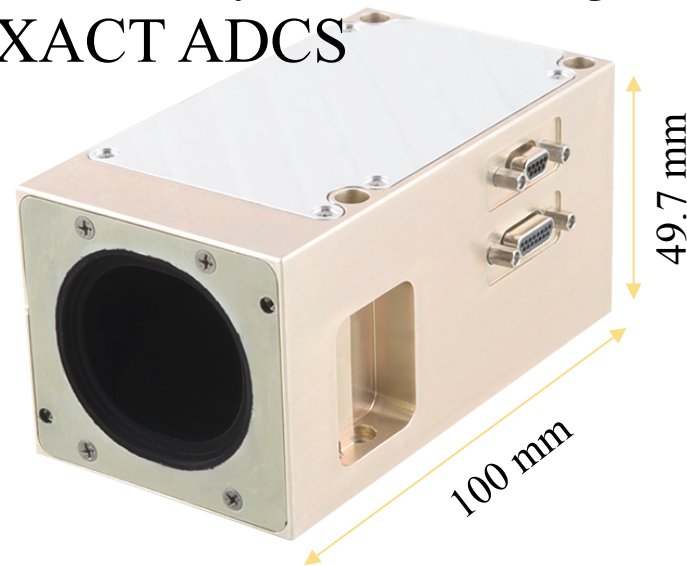
# Introduction: Off-The-Shelf Star Trackers



## Sinclair Interplanetary



## Blue Canyon Technologies XACT ADCS

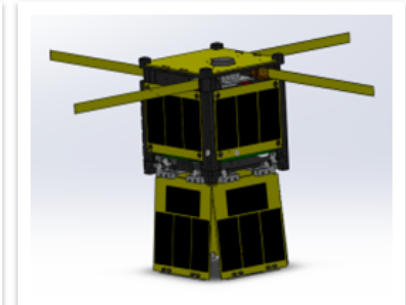
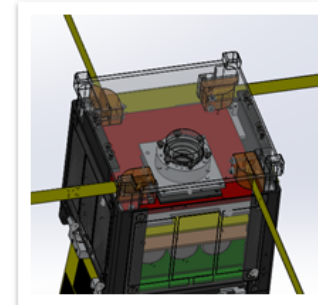
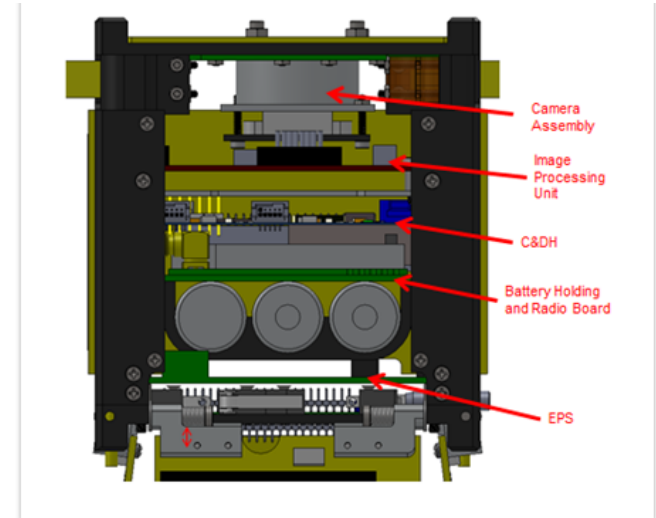
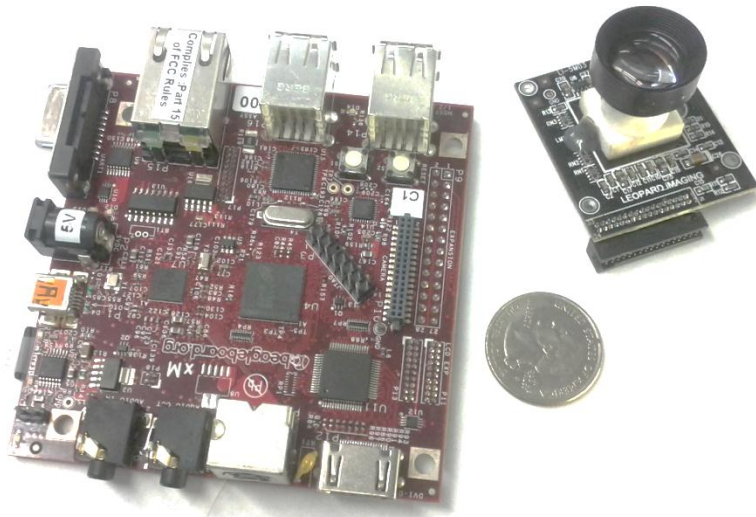


Mechanical Layout (Inches)

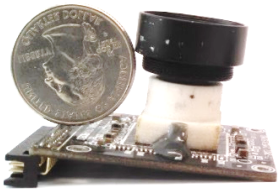


# Star Imager on KySat-2

- 5 Megapixel Sensor (MT9P031)
- BeagleBoard-xM Linux single-board computer



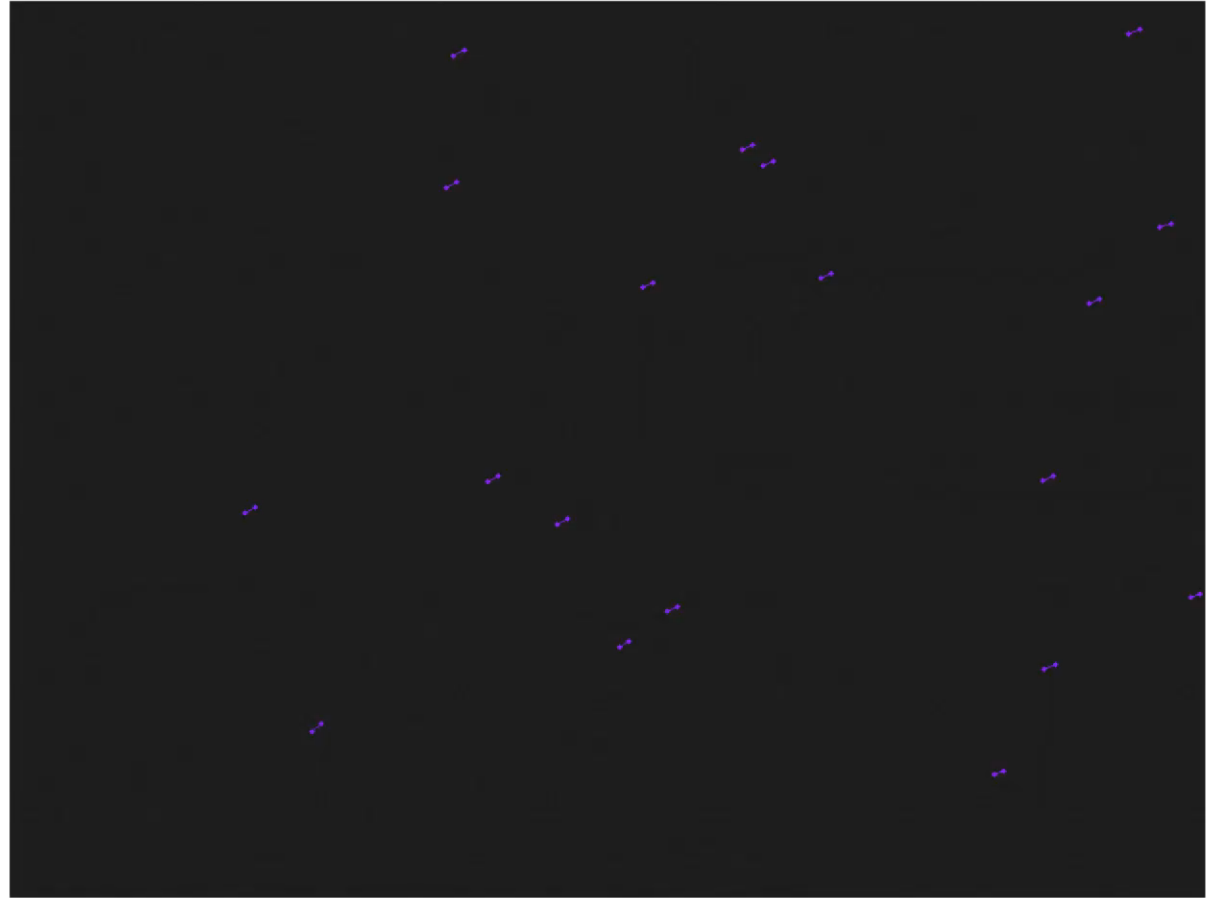
*Rawashdeh, Samir, "VISUAL ATTITUDE PROPAGATION FOR SMALL SATELLITES" (2013). PhD Dissertations--Electrical and Computer Engineering, University of Kentucky*

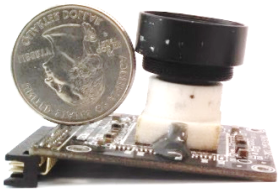


# Stellar Gyroscope - Panning

Essentially: For two images of a star field, produce rotation estimates (three degree-of-freedom / quaternion).

- ▶ Estimation bias: within  $0.005^\circ$
- ▶ Standard deviation: below  $0.02^\circ$
- ▶ Dimmest Star: magnitude 5.7

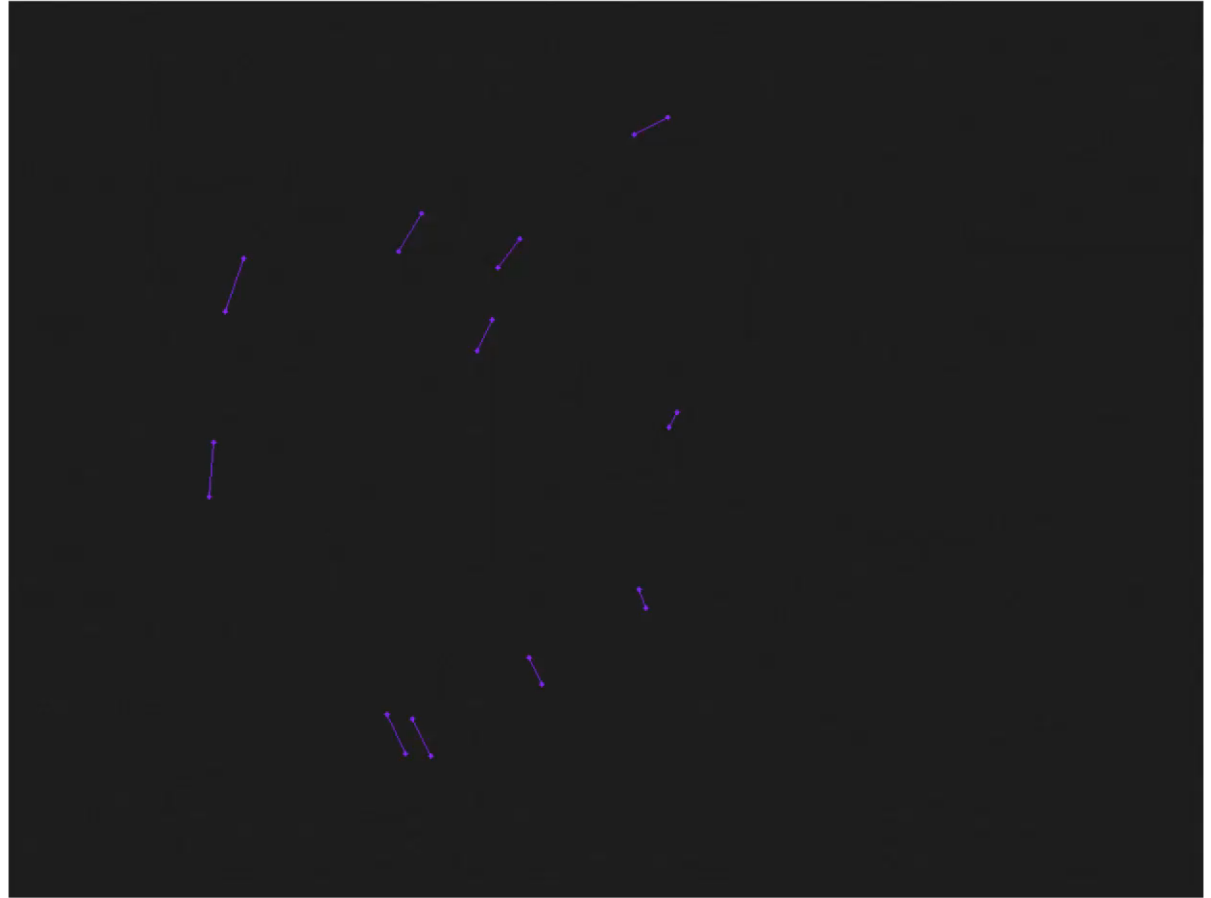




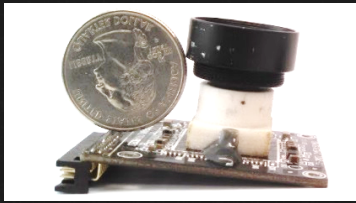
# Stellar Gyroscope - Rotating

Essentially: For two images of a star field, produce rotation estimates (three degree-of-freedom / quaternion).

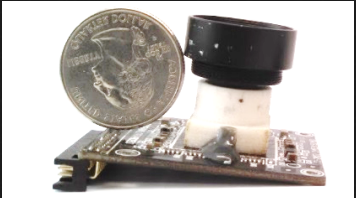
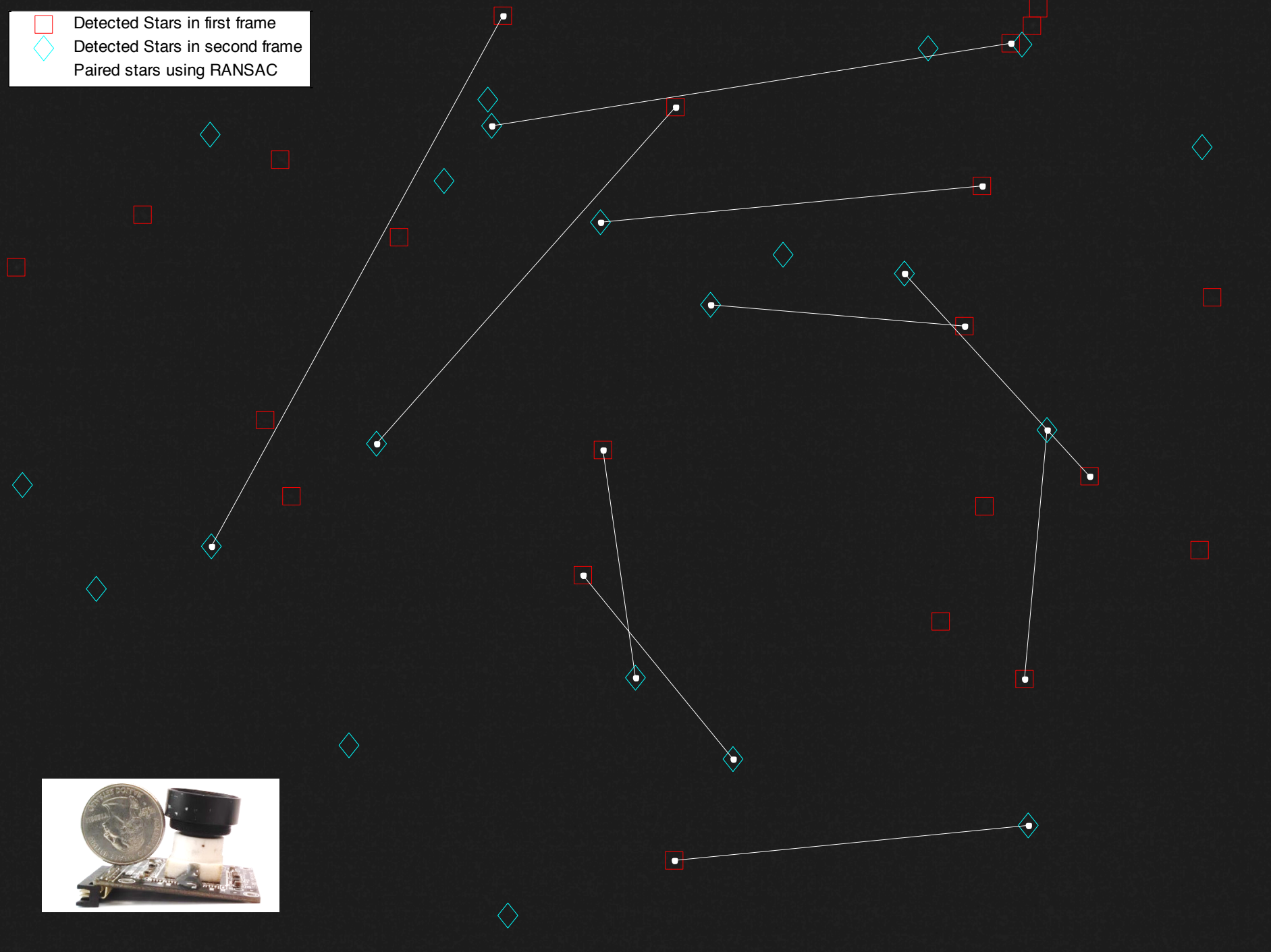
- ▶ Estimation bias: within  $0.005^\circ$
- ▶ Standard deviation: below  $0.02^\circ$
- ▶ Dimmest Star: magnitude 5.7



Detected Stars in first frame  
Detected Stars in second frame



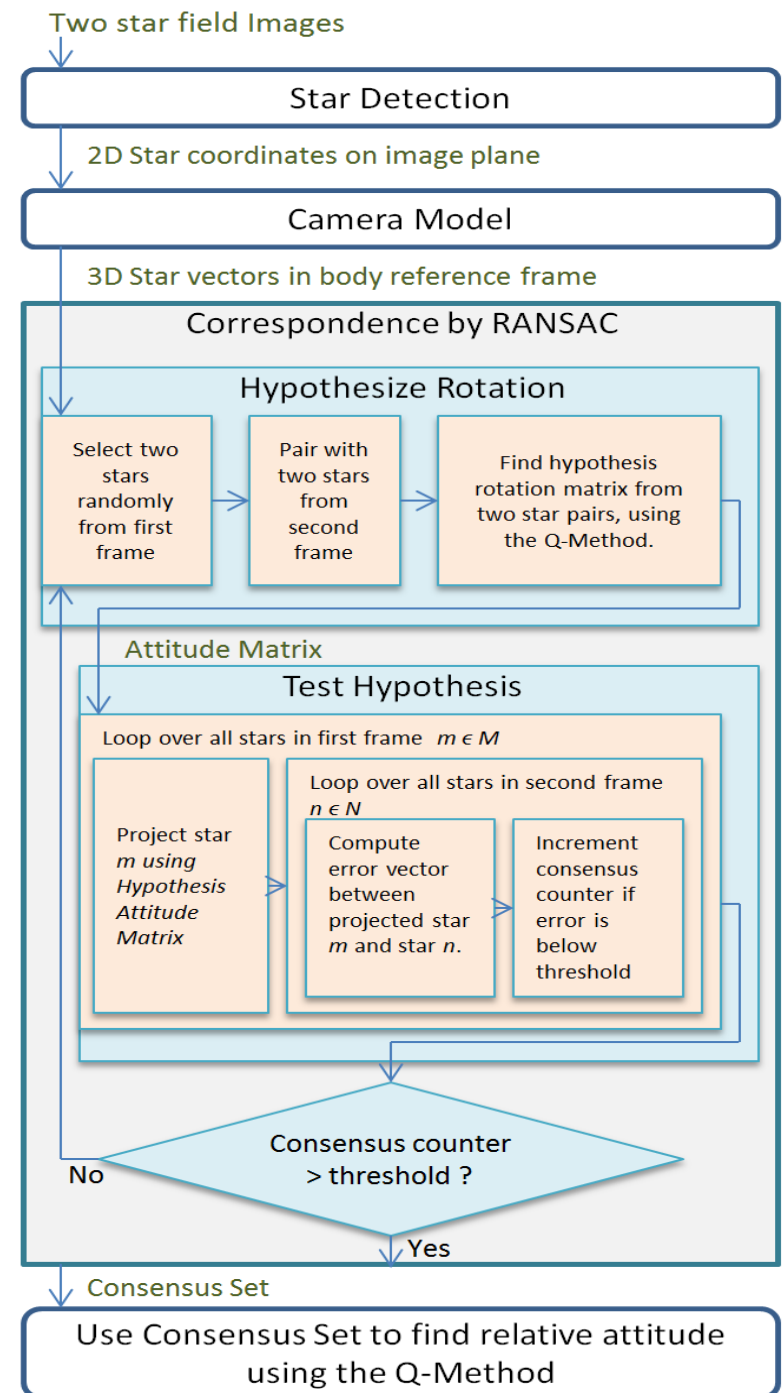
Detected Stars in first frame  
Detected Stars in second frame  
Paired stars using RANSAC



# Random Sample Consensus (RANSAC)

- ▶ **RANSAC:** iterative method to estimate parameters of a mathematical model from a set of observed data which is contaminated a large number of outliers that do not fit the model.
- ▶ The steps of RANSAC can be summarized as
  - ▶ **Hypothesize:** A hypothesis rotation is calculated using randomly selected star pairs across frames.
  - ▶ **Test:** The estimated rotation matrix is tested against all the stars in the two frames. Stars that show consensus are counted towards the Consensus Set (CS).
  - ▶ **Iterate:** RANSAC iterates between the above two steps until a random hypothesis finds “enough” consensus to some selected threshold.

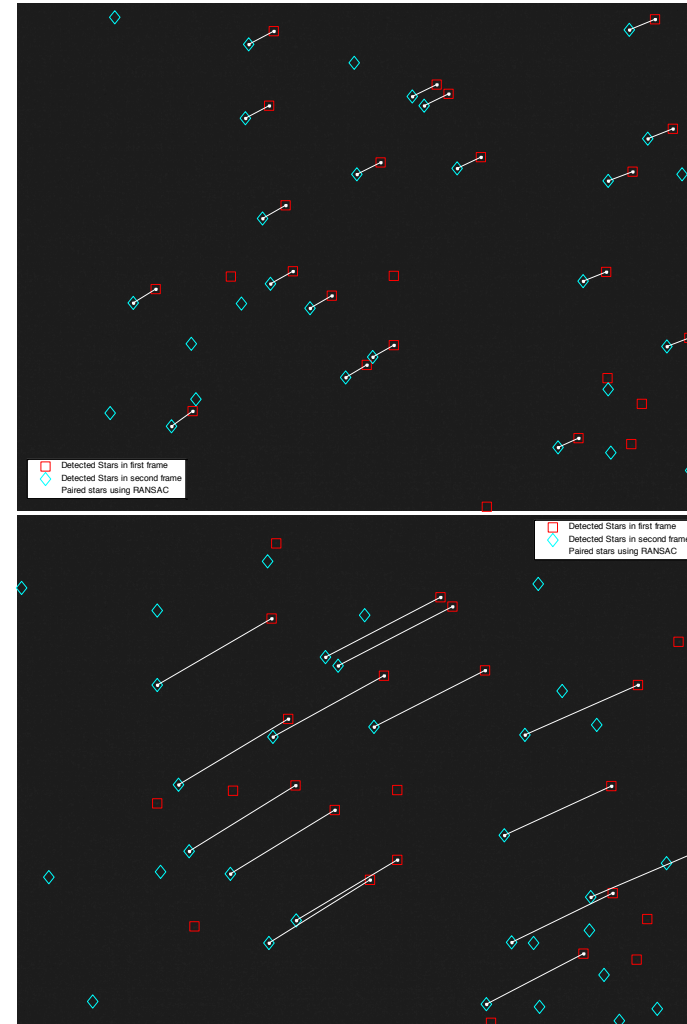
*S. A. Rawashdeh, J. E. Lumpp, “Image-Based Attitude Propagation for Small Satellites using RANSAC”, IEEE Transactions on Aerospace and Electronic Systems, vol. 50, no. 3 pp 1864-1875, 2014.*





# False Star Rejection using RANSAC

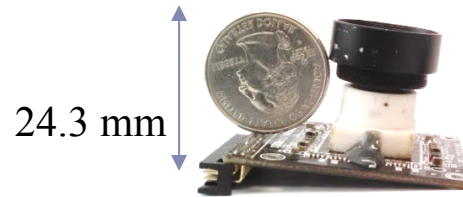
- ▶ Random Sample Consensus (RANSAC) approach is **effective at rejecting stars that do not show agreement with underlying motion.**
- ▶ RANSAC can **tolerate up to 50% un-pairable “stars”** (noise, stars leaving or entering, shot noise, etc).
- ▶ Hypothesis: RANSAC as a filtering step before star database search



# Imager Design



2014, XIMEA corp.



*Current Design*

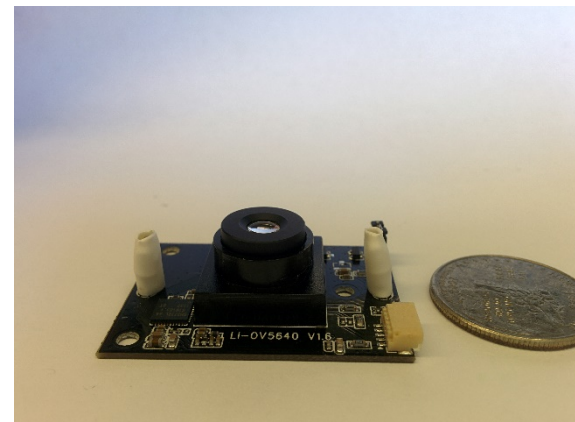
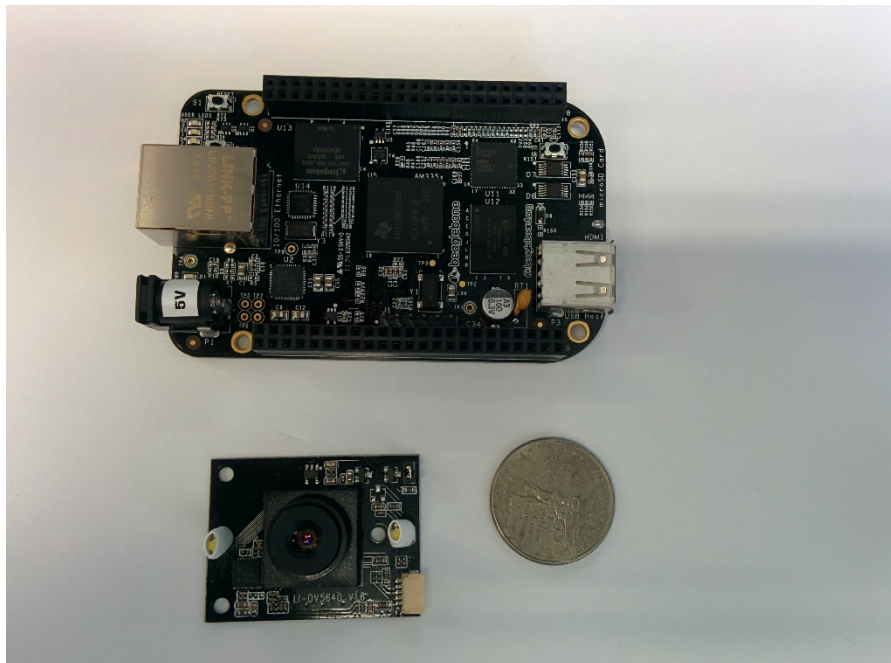


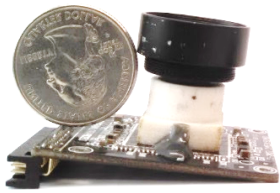
*Proposed Design*

<i>Sensor</i>	Aptina MT9P031 (5MP, 1/2.5")	OmniVision OV5640 (5MP, 1/4")
<i>Lens</i>	Hi Res 16mm, F/1.2, 1/3"	Hi Res 6.3mm, F/2.0, 1/2.5"
<i>Lens Height</i>	Image plane to lens top: 24.3 mm	Image plane to lens top: <b><u>11mm</u></b>
<i>Field of View</i>	15.2° x 20.2 °	24.5° x 32.5 °
<i>Dimmest Star Visible</i>	Magnitude 5.7	<b><u>To be tested.</u></b>
<i>Number of stars in view</i>	Avg 22.9, Min 8 stars (threshold at magnitude 5.7, as found in tests)	Avg 15.3, Min 5 (with conservative threshold at magnitude 4.5)
<i>Connectivity</i>	Parallel data bus (1 camera, 1 processor board)	USB, smaller centralized processor, <b><u>multiple camera nodes</u></b>

# New Topology

- ▶ A centralized image processing system, and multiple miniature cameras (facing various directions)
- ▶ Software developed under Linux; deployable to your favorite Linux system (rad-hard, smaller, multi-purpose, etc)



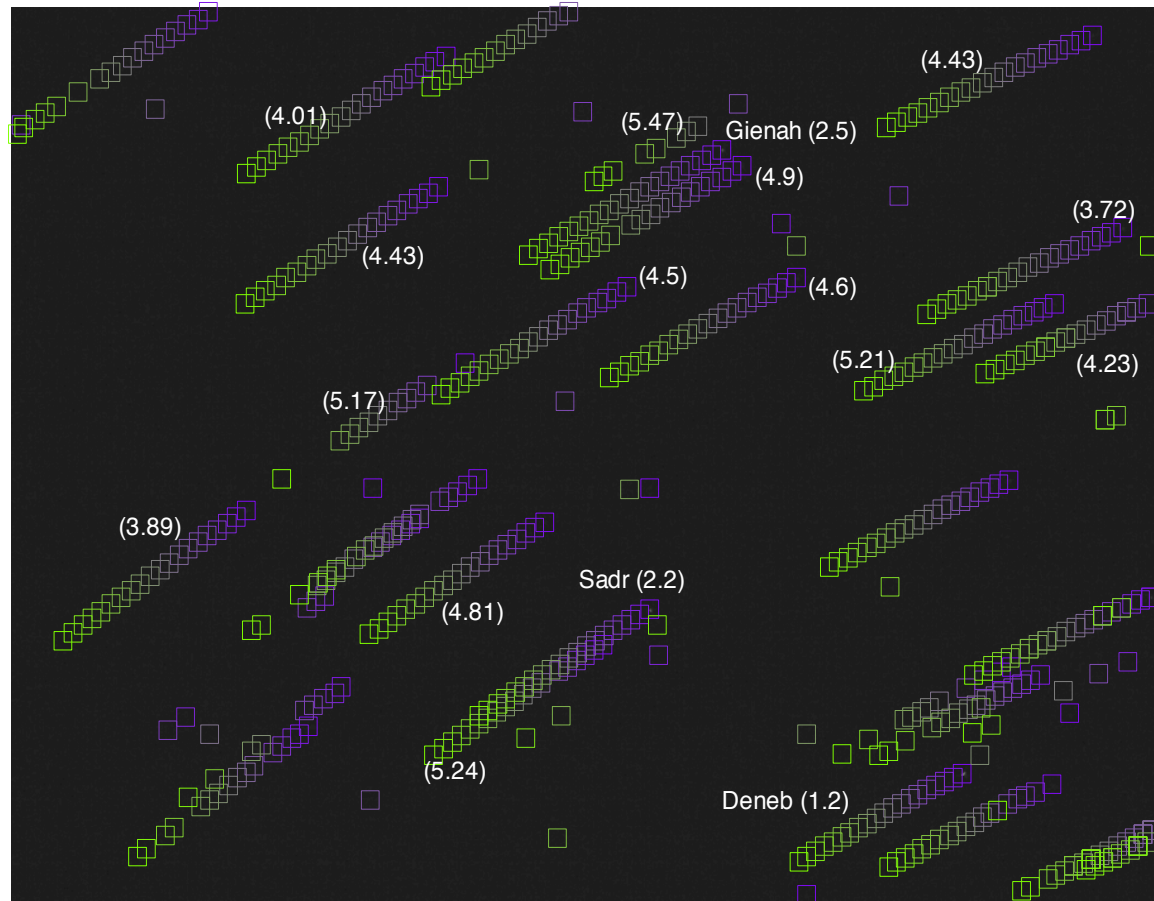


# Night-Sky Tests - Sensitivity

Illustration of detected stars in first photo set of the Cygnus constellation. A photo was taken every minute as Earth rotated in inertial space, every color represents star detections in a single photo and star apparent magnitudes are marked.

*Reliably detected stars of magnitude ~5.2*

*Expected on orbit: magnitude ~5.7*



*S. A. Rawashdeh, J. E. Lumpp, "Image-Based Attitude Propagation for Small Satellites using RANSAC", IEEE Transactions on Aerospace and Electronic Systems, vol. 50, no. 3 pp 1864-1875, 2014.*



# New Sensor Preliminary Results: Night-Sky Tests - Sensitivity

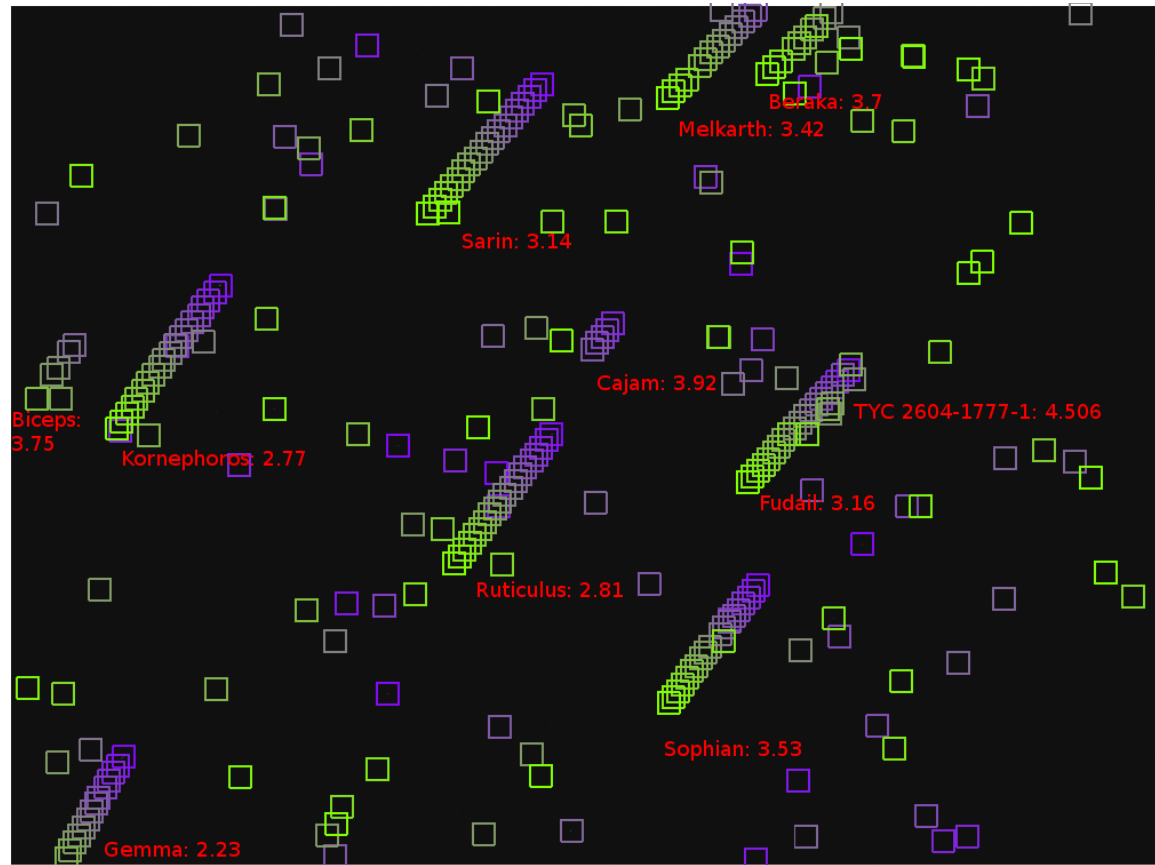
Illustration of detected stars in first photo set of the Hercules constellation. A photo was taken every minute as Earth rotated in inertial space, every color represents star detections in a single photo and star apparent magnitudes are marked.

*Reliably detected stars of magnitude ~3.5*

*Expected on orbit: magnitude ~4.5*

Weather Conditions in test:

- Moon Phase: Waxing Gibbous
- Humidity: 55% to 60%



# Star Database Simulations

- ▶ For star detection threshold of magnitude 4.5 and FOV of  $24.5^\circ \times 32.5^\circ$
- ▶ Sweeping the sky using SKY2000 Star Catalog:
  - ▶ On average 15.3 stars are in view
  - ▶ At least 5 stars in view in the darkest parts of the sky





# Camera Array

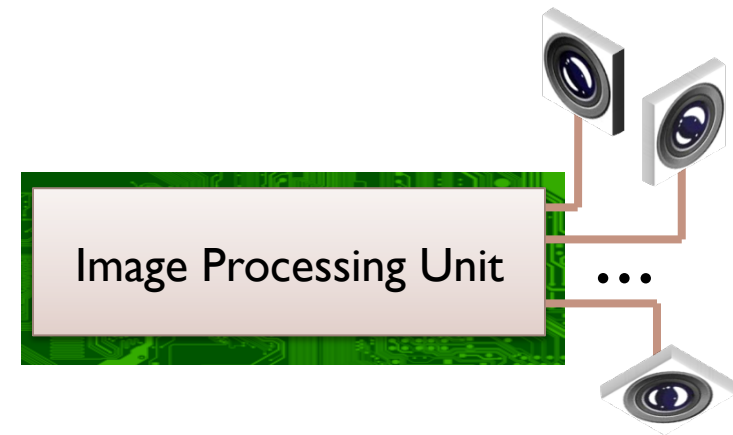
- ▶ Star Tracker FOV typically  $8 \sim 15^\circ$
- ▶ Miniature Camera FOV =  $24.5^\circ \times 23.5^\circ$

## Advantages:

Wide view increases chances of bright stars in view; i.e. camera does not have to be as sensitive to dim stars as narrower cameras.

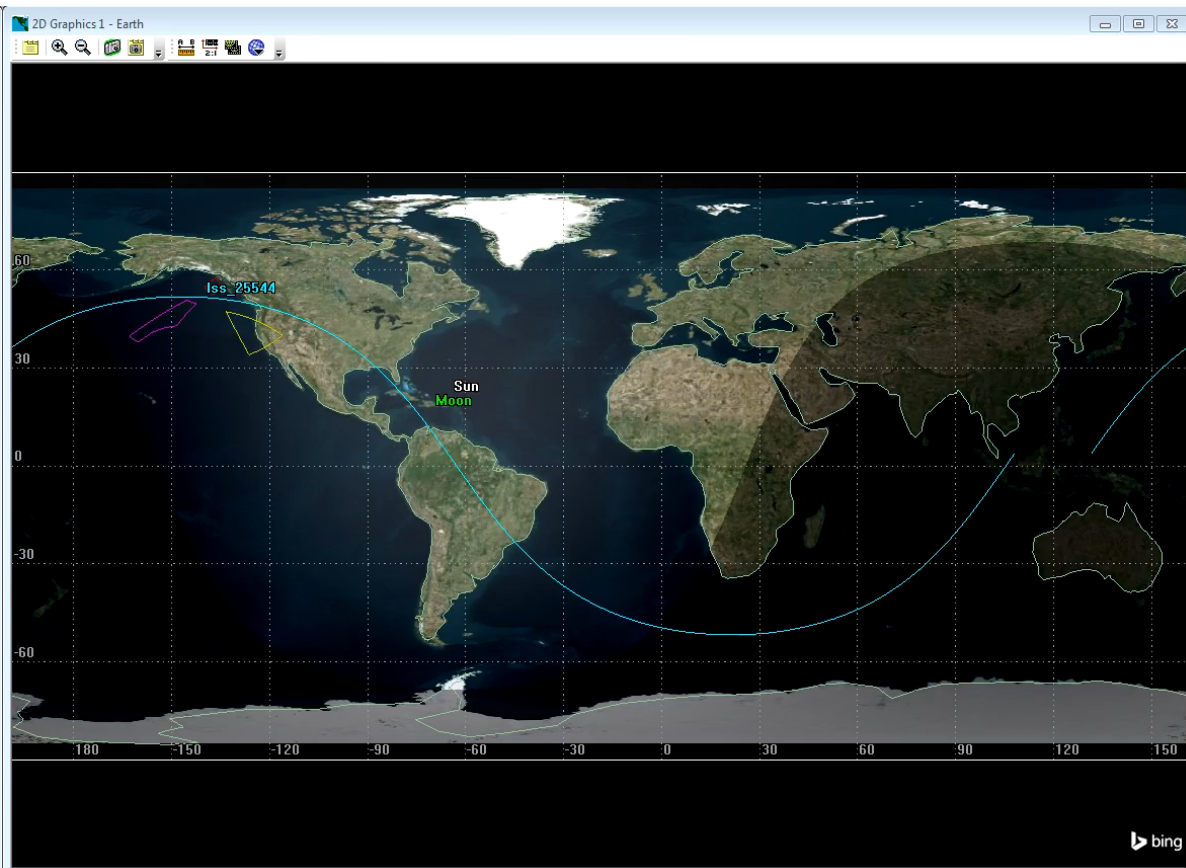
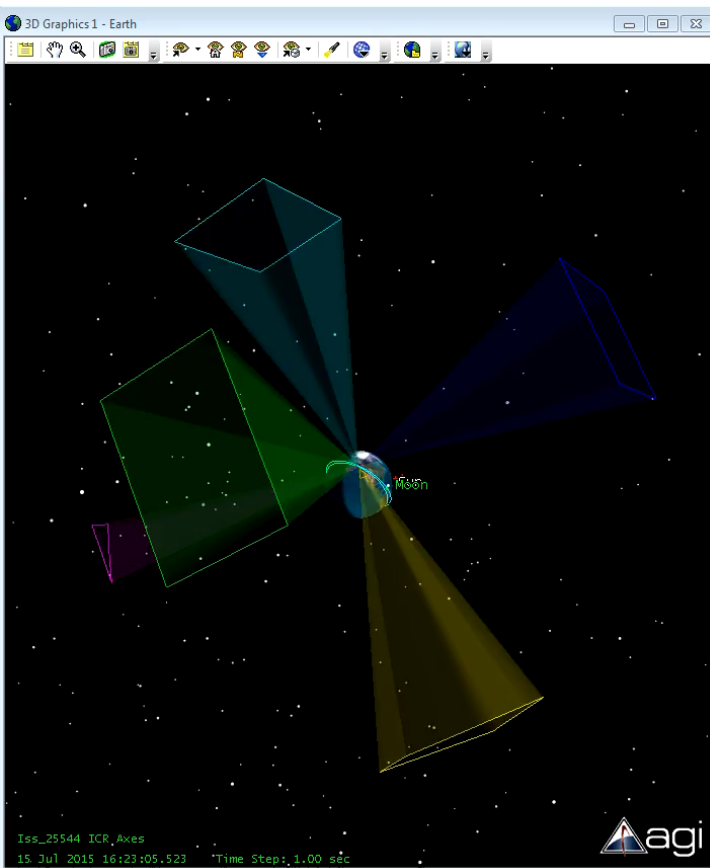
## Disadvantages:

Wide view increases chances of obstruction in view (Earth, Moon, etc.)



# Camera Array View of Sky

- ▶ When one camera is obscured, another camera may provide a view of the stars.
- ▶ A level of fault tolerance comes with multiple imagers in the system.



# Conclusion

- ▶ Modern Linux single-board computers and USB camera modules enable a camera array topology with centralized processing
- ▶ Miniature cameras produce noisy images, primarily because of the small lens aperture.
- ▶ Using “Random Sample Consensus” (RANSAC) by taking two photos and detecting the underlying rotation produces a “Consensus Set” in the presence of up to 50% noise (unpairable stars).
- ▶ This form of RANSAC could be used as a filtering step or as a search approach to identify stars (star tracker).
- ▶ Where the field of view using a small lens may be considered too wide (obstructed too often), a camera array can be used, enabled by the small size and the USB bus advantage.

# Thank You

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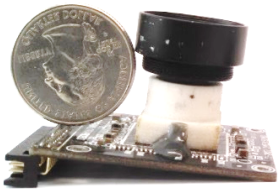
E-mail: [rawashdeh@umich.edu](mailto:rawashdeh@umich.edu)

Web: [sar-lab.net](http://sar-lab.net)



## Acknowledgement:

This project is supported by a seed grant from the NASA Michigan Space Grant Consortium (MSGC)



# Stellar Gyroscope – Simulated Image Set

Essentially: For two images of a star field, produce rotation estimates (three degree-of-freedom / quaternion).

