



Autonomous Pose Estimation for In-Orbit Self-Assembly of the Intelligent Self-powered Modules

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Content Intelligent ISM Self-powered ISM Docking **Pose Estimation** Modules (ISM) Electromagnetic Motivation Flat Docking Relative Position System Applications Relative Attitude • Torque/Force ISM Mission Model • Results Scenario



Motivation

- Small, light and cost- efficient satellites
- Large space structures
- Interchangeability, upgrading and maintenance in orbit

In-orbit assembly of modular spacecraft

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Applications

- Multiple Spacecraft Docking Missions:
 - magnetic shielding
 - large telescopic mirrors
 - large solar panels.







Intelligent Self-powered Modules (ISM)

- Cubic Core

 --> Electronics
- Electromagnetic
 Flat Docking System
- Solar cells





ISM Mission Scenario

other –satellite

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Docking Systems

Docking Systems

Cone / Probe

- •Requires fine alignment
- Smooth contact
- Plume impingement effects.

Propellant-less

- capture using EM forces
- Controlled contact
- No extra propellant
- No plume impingement effects.



Propellant-less Docking

Electro-magnetic Forces

Electro-static Forces





Propellant-less Docking

Electro-magnetic Forces

Electro-static Forces

The ISM mission concept uses the (2003) electromagnetic forces as a means for completing multiple docking attempts without damaging the satellites' sensitive parties the space of the space

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Electromagnetic Docking



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ISM Pose Estimation



ISM body structure

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ISM axial facet



Relative Position



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Partial Occlusion

False estimates ^ | | | Partial occlusion





Relative Attitude









Facet detection







12th August 2009, 8:45AM MDT, Utah State University, Logan, Utah



Results





Results



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Conclusion

- The pose of a neighbouring ISM structure was determined
 - Relative position - > passive
 - Relative attitude - > active
- Algorithm tested on a 3-by-3 structure
- Accuracy depends on the camera specifications

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THANK YOU!