

# Tetrahedral Mesh Improvement by Shell Transformation

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## ABSTRACT

Existing flips for tetrahedral meshes simply make a selection from a few possible configurations within a single shell (i.e., a polyhedron that can be filled up with a mesh composed of a set of elements that meet each other at one edge), and their effectiveness is usually confined. A new topological operation for tetrahedral meshes named *shell transformation* is proposed. Its recursive callings execute a sequence of shell transformations on neighboring shells, acting like *composite edge removal* transformations. Such topological transformations are able to perform on a much larger element set than that of a single flip, thereby leading the way towards a better local optimum solution. Hence, a new mesh improvement algorithm is developed by combining this recursive scheme with other schemes, including smoothing, point insertion and point suppression. Numerical experiments reveal that the proposed algorithm can well balance some stringent and yet sometimes even conflict requirements of mesh improvement, i.e., resulting in high-quality meshes and reducing computing time at the same time. Therefore, it can be used for mesh quality improvement tasks involving millions of elements, in which it is essential not only to generate high-quality meshes, but also to reduce total computational time for mesh improvement.

**KEY WORDS:** mesh improvement; mesh generation; shell transformation; mesh smoothing; topological transformation; tetrahedral meshes

## 1. INTRODUCTION

For numerical simulations with complex geometries, mesh generation typically represents a large portion of the overall computational time. Thus, the ability of performing computations on large-scale tetrahedral elements has always been regarded as an important issue. The fundamental reason is mainly because a theoretically valid tetrahedral mesh can always be automatically generated for a valid 3D domain [1-5], despite that this is not always the case for other specific types of volume elements. Despite of the validity, the quality of an initial tetrahedral mesh produced by a mesher may not be high enough for simulations. A follow-up mesh improvement step is thus indispensable to remove those poorly shaped elements contained in the initial meshes to prevent their adverse effects on the stability and accuracy of the simulations.

In general, a mesh improver executes the following types of local operations iteratively:

- (1) *Smoothing*, which repositions mesh points to improve the quality of adjacent elements.
- (2) *Local reconnection*, which replaces a local mesh with another mesh that fills up the same region. The new mesh will have the same point set as the old mesh but applying different point connections.

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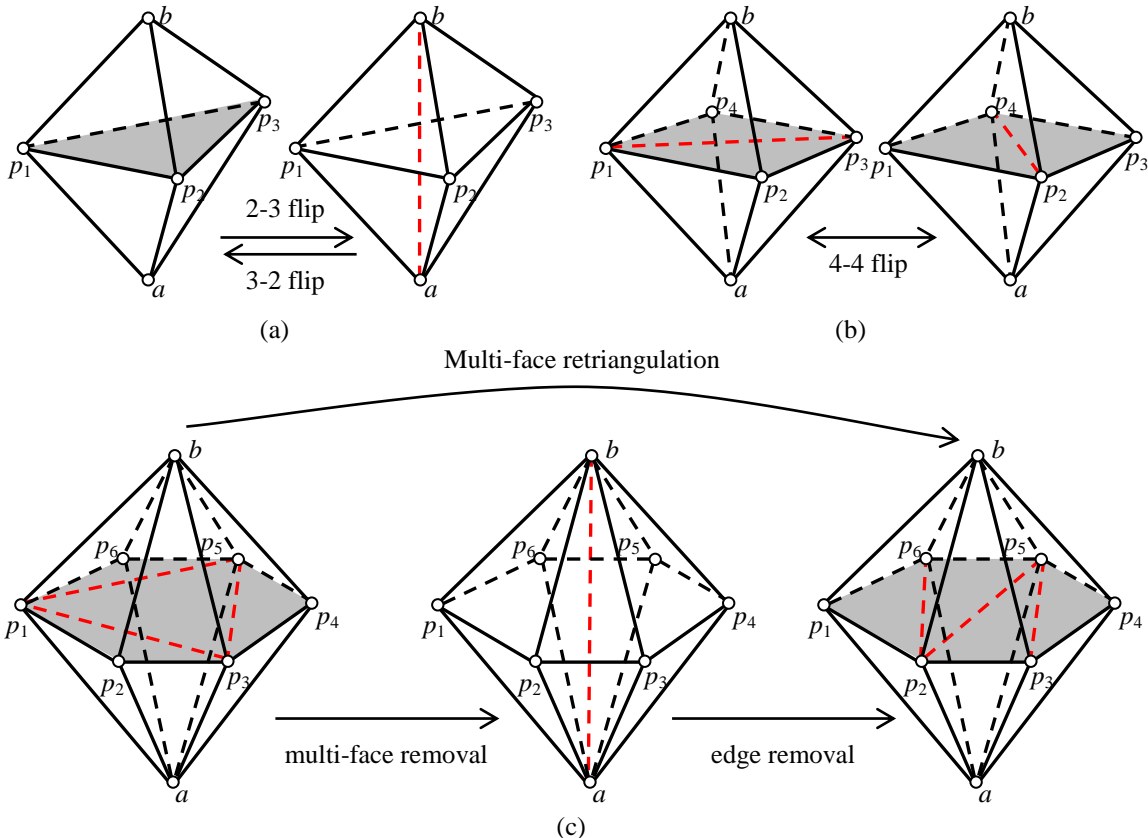
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41 (3) *Point insertion/suppression*, which improves a mesh by inserting new points into the  
 42 mesh or removing existing points from the mesh.

43 Our primary focus in this study is on local reconnection, although all the local operations  
 44 mentioned above will be combined in the developed mesh improver. If the point set is fixed,  
 45 the quality of mesh elements is apparently determined by how these points are connected. It is  
 46 unrealistic to search for a global optimal mesh topology by directly iterating a large number of  
 47 possible solutions to connect a point set because this number could expand exponentially with  
 48 the increase of the number of points. Thus, heuristics prevail in improving the quality of a  
 49 mesh by iteratively changing the local connections of points.

50 The most frequently used local reconnection technique for tetrahedral meshes is based on  
 51 *elementary flips* [6], including 2-3, 3-2 and 4-4 flips (note that the numbers in these names  
 52 denote the number of tetrahedra removed and created by the flips, respectively; see Figures 1a  
 53 and 1b). Because the elementary flips simply make a selection from several possible  
 54 configurations within a relatively small region, their effectiveness in mesh quality  
 55 improvement is usually confined. To overcome this limit, three advanced flips that involve  
 56 more elements were later suggested, i.e., *edge removal* [7], *multi-face removal* [8] and  
 57 *multi-face retriangulation* [9] (see Figure 1c). They enrich the possible configurations within  
 58 relatively larger regions and therefore behave more effectively in mesh quality improvement  
 59 than the elementary flips.

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64 **Figure 1.** Existing flips for a tetrahedral mesh: (a) 2-3 flip and 3-2 flip; (b) 4-4 flip; (c) multi-  
 65 face removal, edge removal and multi-face retriangulation.

66 In general, the effect of a local reconnection technique highly depends on the size of a local  
 67 mesh it treats. The more elements it treats during one operation; the more possibility it  
 68 improves the quality of a local mesh to a higher level. This motivates the development of  
 69 more aggressive local reconnection techniques for mesh quality improvement. For instance,  
 70 Joe [6] once proposed nine schemes to *combine the elementary flips* for mesh improvement.  
 71 Later, Shewchuk [10] pointed out that the composite flips proposed by Joe [6] could be

72 expressed as one or two edge removal operations. Thus, Shewchuk suggested that the study of  
73 composite edge removal transformations was a fruitful direction for mesh improvement  
74 research. Unfortunately, Shewchuk did not present details about what such composite  
75 transformations are and how to implement such transformations efficiently. As a result, it was  
76 observed that no composite transformations are actually incorporated into the open-source  
77 tetrahedral improver (namely Stellar thereafter) developed by Klinger and Shewchuk [11, 12].

78 The main contribution of this present study is the development of a new local reconnection  
79 technique that could act like composite edge removal transformations. This technique is based  
80 on the recursive callings of a new flip named *shell transformation*. The single calling of shell  
81 transformation could be considered as an enhanced version of edge removal transformation.  
82 However, an essential difference exists between two approaches, and enables shell  
83 transformation to be executed recursively (see Section 2.4 for details). Thus, edge removal  
84 only makes a selection from a few possible configurations within a single shell (i.e., a  
85 polyhedron that can be filled up with a mesh composed of a set of elements that meet each  
86 other at one common edge). However, the recursive callings of shell transformations can  
87 execute a sequence of shell transformations on neighboring shells. In other words, recursive  
88 shell transformations could be performed on a much larger element set than that of edge  
89 removal, thereby leading the way towards a better local optimum solution.

90 Another focus of this study is about the efficient implementation of the new local  
91 reconnection technique, because local reconnections need to be employed for a large number  
92 of times during the entire mesh improvement workflow. The dynamic programming algorithm  
93 suggested by Shewchuk for edge removal [10, 13] will be revisited at first and then further  
94 enhanced to implement the basic shell transformation routine. Besides, the computing  
95 efficiency of recursive callings of shell transformations is investigated carefully because the  
96 number of such callings may increase exponentially when the recursive level increases.  
97 Reasonable restrictions are provided to prevent inefficient recursive callings. Meanwhile,  
98 several strategies are suggested to improve the efficiency.

99 Finally, the ability of the proposed local reconnection technique will be demonstrated by  
100 performing various mesh improvement tasks, some of which involve millions of elements. In  
101 a pipeline of producing meshes of this magnitude, mesh generation itself may only consume a  
102 few seconds computing time, owing to recent advancement in the field of fast and parallel  
103 mesh generation techniques [14-17]. However, a mesh improver possibly consumes many  
104 minutes computing time or even longer in order to manage such a big mesh. Therefore, to  
105 ensure the applicability of this newly developed mesh improver for large-scale problems, an  
106 essential requirement we will take into account is the *cost-effectiveness* of the mesh improver,  
107 i.e., the ability to balance the conflict requirements of resulting in a high-quality mesh and  
108 saving computing time of mesh improvement. Following this concept, a set of existing  
109 smoothing, point insertion and point suppression schemes are selected. Combining these  
110 schemes with the proposed local reconnection technique, a cost-effective improver applicable  
111 to large-scale meshes is therefore developed and verified.

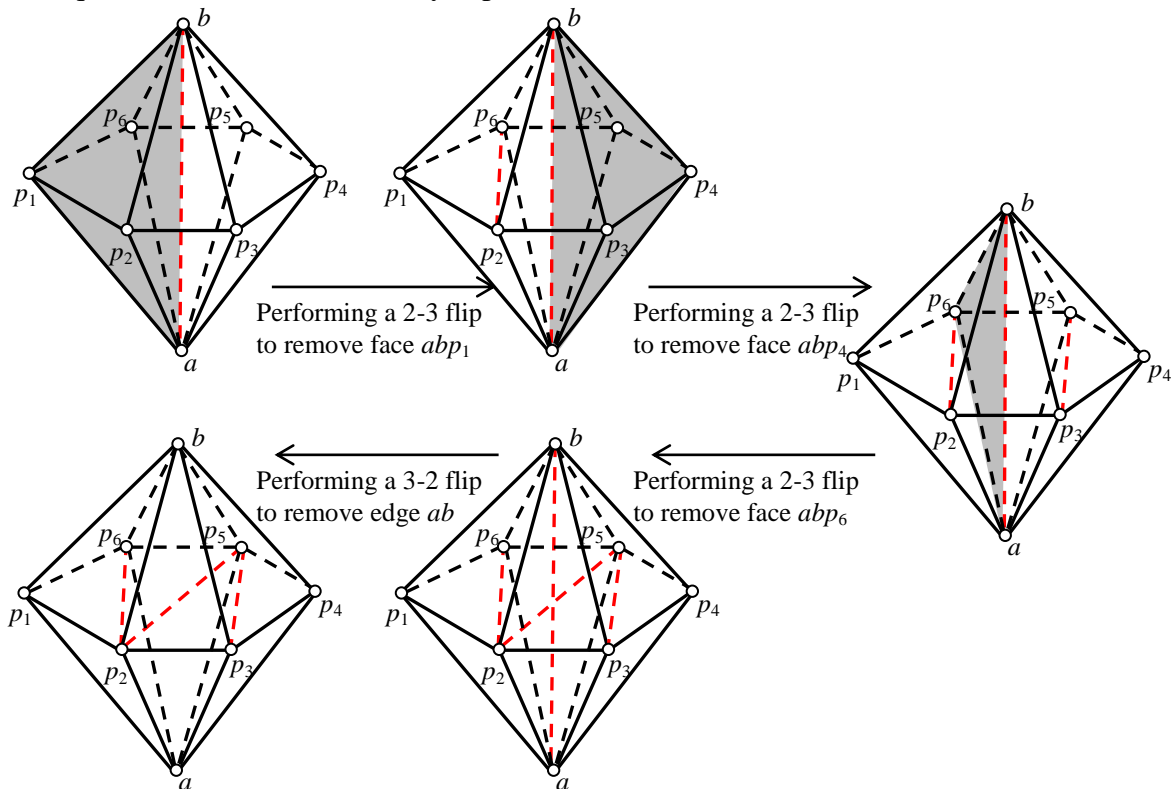
112 The remainder of this article will be organized as follows. In Section 2, related works are  
113 firstly reviewed, followed by the introduction of the new local reconnection technique in  
114 Section 3. Section 4 describes the basic implementation of shell transformation, while  
115 Section 5 presents the recursive scheme of shell transformation and the local reconnection  
116 scheme based on this recursive scheme. Section 6 introduces other local operations that are  
117 combined to form the developed new mesh improver. Section 7 provides various examples of  
118 numerical experiments demonstrating the effectiveness and efficiency of the proposed  
119 scheme. Section 8 concludes with outcomes of the study.

## 2. RELATED WORKS

121 Firstly, related works on local reconnection techniques are reviewed in details (see Section  
 122 2.1). After that, a brief review on other types of local operations (see Sections 2.2 and 2.3,  
 123 respectively) is presented to justify our choices of these types of operations in the developed  
 124 mesh improver.

### 125 2.1 Related work on local reconnection techniques

126 Local reconnection techniques are frequently used in various circumstances of mesh  
 127 generation, such as Delaunay refinement [18], mesh adaptation [19], boundary recovery [1-5],  
 128 and mesh quality improvement [6-12, 20, 21]. The first type of local reconnection techniques  
 129 is based on the flips presented in Figure 1. The 3-2, 2-3 and 4-4 flips are defined as  
 130 *elementary flips*, not only because they are special cases of the advanced flips, but also  
 131 because they could be combined to form the advanced flips. For instance, the edge removal  
 132 transformation could be implemented as a sequence of 2-3 flips followed by a single 3-2 flip  
 133 (see Figure 2) [10]. This possibly explains why Shewchuk [10] judged that the composite  
 134 transformations of the elementary flips proposed earlier by Joe [6] could be expressed as one  
 135 or two edge removal transformations. However, this does not mean that a local reconnection  
 136 technique based on the elementary flips could achieve the same effect as the techniques based  
 137 on the advanced flips. In general, the advanced flips provide elaborate ‘*patterns*’ for how to  
 138 combine the elementary flips such that those involved elementary flips could work together in  
 139 a much larger element set. As a result, the local reconnection techniques based on the  
 140 advanced flips could usually improve the quality of a mesh to a higher level than that by the  
 141 techniques based on the elementary flips.



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144 **Figure 2.** Implementing edge removal as a sequence of 2-3 flips followed by a 3-2 flip.

145 Differently, Liu *et al.* proposed to improve mesh quality by a local remeshing technique  
 146 named *small polyhedron reconnection* (SPR) [21]. The SPR algorithm initializes an empty

147 polyhedron in the neighbourhood of a poorly shaped element, and then performs an  
148 exhaustive search to find the *optimal* tetrahedralization of this polyhedron. Typically, the  
149 advanced flips treat only a local mesh composed of a few tetrahedral elements, while the SPR  
150 algorithm can search for the optimal tetrahedralization of a polyhedron composed of 20-40  
151 tetrahedral elements. Consequently, it was reported that the SPR algorithm could achieve a  
152 better mesh improvement result than that of the flip-based algorithm [21]. However, the main  
153 issue of the SPR algorithm is its computing complexity, since the problem of meshing an  
154 empty polyhedron is NP hard. Although some strategies have been proposed in the past to  
155 improve the timing-related performance of the SPR algorithm [4, 21], our experience shows  
156 that, if the local reconnection scheme of a mesh improver is completely dependent on the SPR  
157 algorithm. In addition, the runtime of a mesh improver may be beyond the user's expectation,  
158 in particular when the treated mesh is composed of one million or more elements.

## 159 2.2 Related work on mesh smoothing

160 The mesh smoothing methods can be classified into two categories in general, i.e., the  
161 Laplacian-type method [22] and the optimization-based method [20, 23-28]. The Laplacian-  
162 type method moves each mesh vertex towards the *center* of its neighboring vertices. It  
163 provides no guarantee for the improvement of mesh quality, therefore leading to either low  
164 quality or even possibly invalid elements. To overcome this drawback, various optimization-  
165 based approaches were proposed. These approaches can be divided into two types in  
166 accordance with the choice of objective functions. The *local* method maximizes a quality  
167 function for the elements surrounding each mesh vertex by relocating mesh vertices  
168 iteratively [20, 23, 24]. The *global* method maximizes a quality function for all elements by  
169 relocating all mesh vertices simultaneously [25-28]. While the local method may improve an  
170 element at a cost of degrading its neighboring elements, the global method relieves this  
171 conflict to some extents by considering the quality of the entire mesh as a whole.  
172 Nevertheless, because the global method needs to solve a large-scale optimization model, the  
173 choice of solution methods for this optimization model will be the key step to achieve  
174 acceptable computing time and performance [28]. Likewise, in the case of surface mesh  
175 smoothing, various *local* and/or *global* shape-preserving approaches have been developed, in  
176 which the global approach based on geometric flows is now still prevailing because of its  
177 powerful ability to preserve geometric features and thus reduce volume shrinkage [25].

178 Since the cost-effectiveness of a new mesh improver is our primary goal to achieve, we  
179 select a local approach instead of a more time-consuming global approach for mesh  
180 smoothing. More specifically, we combine an optimization-based algorithm [20] with the  
181 Laplacian smoothing [22]. See Section 5.4 later for more details.

## 182 2.3 Related work on point insertion and point suppression

183 It is intuitive to eliminate low-quality elements by inserting vertices into existing meshes.  
184 Typically, a new vertex can be located at the circumcenter [5, 29] or centroid of a poor  
185 element [1, 4], or the midpoint of the longest edge of this element [2, 30]. One typical strategy  
186 to insert this kind of vertex into a mesh is by Delaunay refinement [1-5, 29, 31], or more  
187 simply, by splitting the element(s) containing the new vertex (which may result in temporary  
188 low-quality elements) firstly and then improving the mesh by combining smoothing and local  
189 reconnection [19]. Besides, Klingner and Shewchuk suggested an effective but very time-  
190 consuming point insertion scheme that combines a Delaunay-type algorithm with smoothing  
191 operations [11]. It is possible that hundreds of elements are involved in just one single  
192 operation. Nevertheless, to meet our goal of developing a cost-effective mesh improver for  
193 large-scale problem inputs, we adopt an *edge-splitting* based point insertion scheme.

194 As a reverse operation of point insertion, it is not surprising that point suppression can also  
 195 improve local mesh quality by remeshing an empty polyhedron composed of all elements  
 196 surrounding the vertex to be removed. The challenge mainly comes from those polyhedra that  
 197 cannot be tetrahedralized if no Steiner points are allowed [31]. Meanwhile, even if a  
 198 polyhedron can be tetrahedralized without inserting Steiner points (although the prediction of  
 199 this is NP hard [31]), it is not easy to find an optimal mesh to fill in that polyhedron (it is also  
 200 NP hard [4, 21]). Theoretically, the SPR algorithm mentioned in Section 2.1 [4, 21] could be  
 201 a good candidate for remeshing an empty polyhedron because it can provide an optimal  
 202 solution when the polyhedron is meshable. Nevertheless, as we pointed earlier, the main issue  
 203 of the SPR algorithm is its relatively poor timing performance. Therefore, we adopt an *edge-*  
 204 *contraction* based point suppression routine that is available from Stellar [12, 13].

205 See Section 5.5 later for the developed point insertion and suppression schemes.

### 206 3. SHELL TRANSFORMATION AND ITS RECURSIVE CALLINGS: 207 THE MAIN IDEA

208 For the completeness, we have briefly reviewed different types of local operations in order to  
 209 justify our choices. However, it must be emphasized that the main contribution of this study is  
 210 the development of a new local reconnection technique. With respect to other local  
 211 operations, we only select a suitable operation among various existing approaches to meet our  
 212 goal of developing a cost-effective mesh improver.

213 Before introducing the new local reconnection technique, Figure 3 illustrates some  
 214 terminologies in relation with a shell structure because all of the flips discussed in this study,  
 215 including those presented earlier in Figure 1 and the new proposed shell transformation, will  
 216 involve this structure.

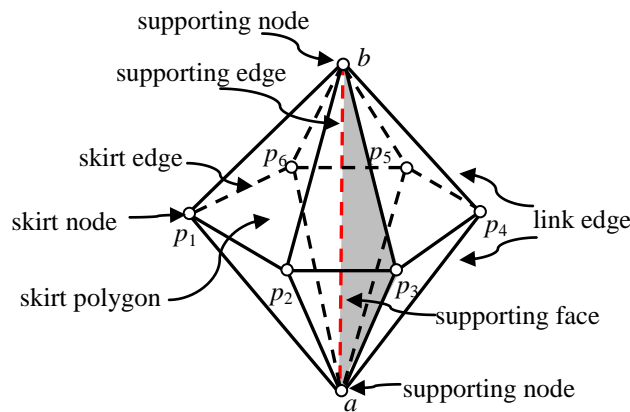
217 It was reported that edge removal might be the most effective flip for mesh quality  
 218 improvement [12], although other flips could provide marginal improvements additionally. To  
 219 remove a low-quality element (for instance, the element  $abp_5p_6$  in Figure 4), we could pick up  
 220 an edge of this element (denoted by  $e$ ), and perform the edge removal transformation in the  
 221 shell of  $e$  (i.e.,  $e$  is the supporting edge of the shell). For instance,  $e$  refers to  $ab$  in Figure 4. If  
 222 the output covering mesh of the transformation is better than the old one, edge removal  
 223 succeeds and the low-quality element is removed (since  $e$  is removed). In such a case, the  
 224 skirt polygon of the shell is *completely triangulated* (see Figure 4). However, it is not  
 225 uncommon that the edge removal could not provide a better mesh than the old one and thus  
 226 fails to remove  $e$  and the low-quality element bounded by  $e$ . Instead, a covering mesh of the  
 227 shell that is better than the old one might be the one shown in the bottom of Figure 4, where a  
 228 *core* referring to the unmeshed part of the skirt polygon exists in the resulting mesh. In this  
 229 case, we say the shell of  $e$  can only be partially reduced<sup>†</sup>, and the remaining supporting faces  
 230 must be removed to reduce the shell further. Obviously, if one of the link edges that bound a  
 231 supporting face  $f$  is removed,  $f$  will be removed accordingly. For the case shown in the bottom  
 232 of Figure 4,  $f$  could be the face  $abp_5$ , and the link edge could be  $ap_5$  or  $bp_5$ . Assuming that  $ap_5$   
 233 is picked up for removal, the above transformation is called again to reduce the shell of  $ap_5$ . If  
 234 the reduced shell of  $ap_5$  does not contain the face  $abp_5$  and any new supporting faces sharing

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<sup>†</sup>Here, the *degree* of a covering mesh refers to the number of elements that share the supporting edge in this mesh. A shell is *reduced* if the degree of the new covering mesh is becoming smaller than that of the old mesh. In particular, if the degree of the new mesh becomes zero, the shell is *completely reduced*, and the new mesh is a *completely reduced mesh*; otherwise, the shell is *partially reduced*, and the new mesh is a *partially reduced mesh*.

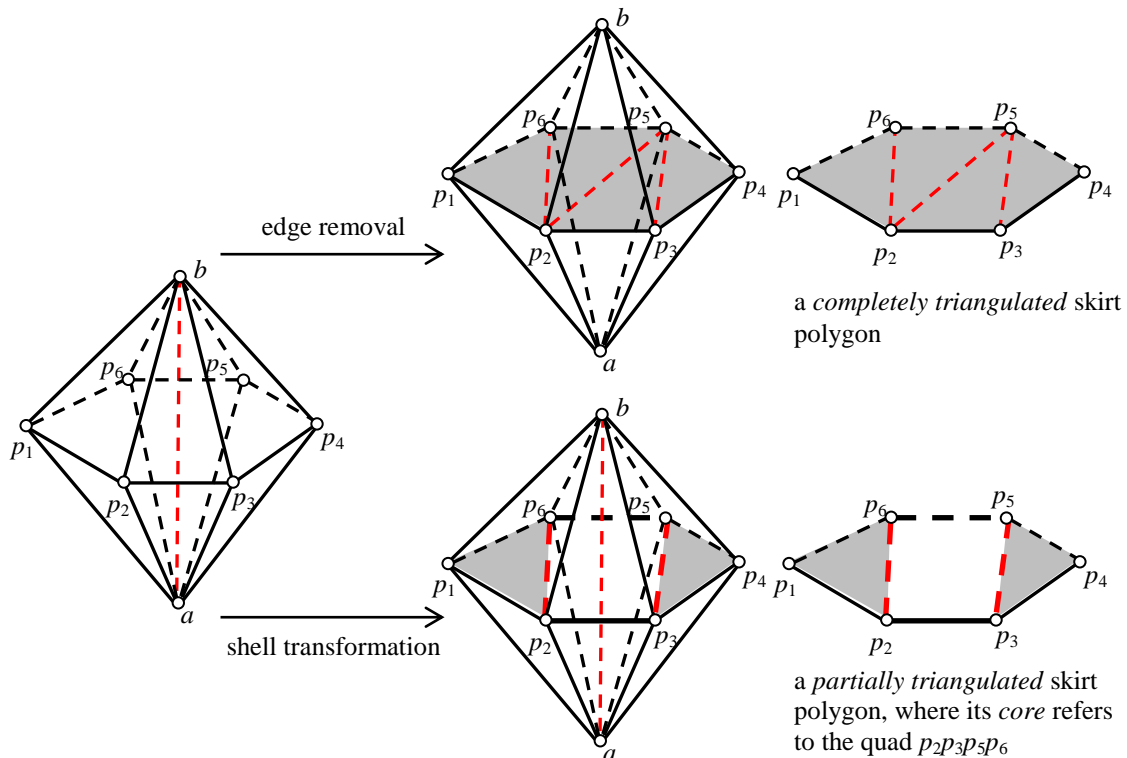
235  $ab$ , the shell of  $ab$  is reduced as well; otherwise, a process that attempts to remove the  
 236 supporting faces around  $ap_5$  is repeated.

237 To better understand the above recursive scheme, Figure 5 illustrates how this scheme  
 238 works on a local mesh composed of two shells (see Figure 5a), aimed at removing the edge  $ab$   
 239 from the mesh. Firstly, a transformation is called on the shell of  $ab$ . Since the shell cannot be  
 240 completely reduced, the edge  $ab$  still exists in the output mesh (see Figure 5b). Nevertheless,  
 241 the degree of the shell is reduced from 5 to 4. To reduce the shell further, a link edge  $bh$  is  
 242 picked up and a transformation is called on the shell of  $bh$  and reduces this shell completely.  
 243 Besides, the degree of the shell of  $ab$  is reduced from 4 to 3 after this step (see Figure 5c).  
 244 Finally, a transformation is called to update the shell of  $ab$  to remove  $ab$  by a single 3-2 flip  
 245 (see Figure 5d for the final output).



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**Figure 3.** The terms defined for the mesh entities of a shell.

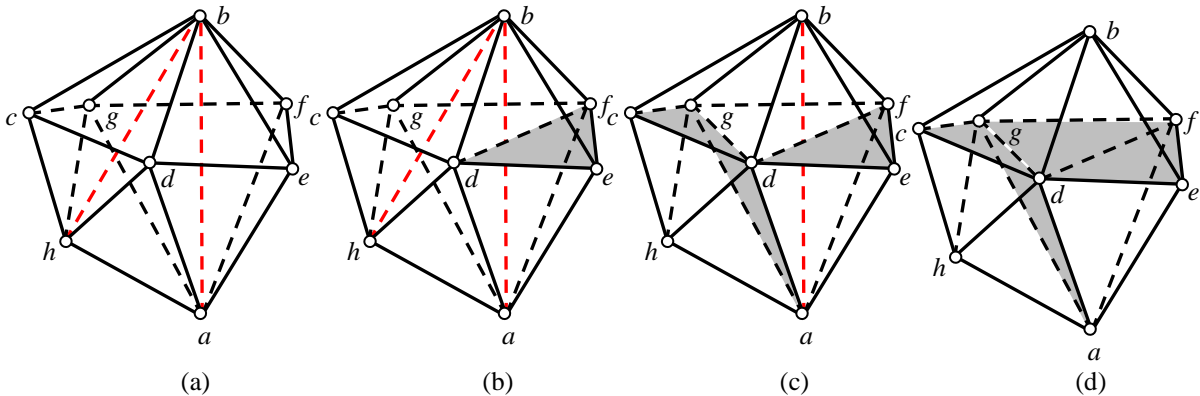


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**Figure 4.** The difference between edge removal and a single calling of shell transformation. This difference enables shell transformation to be called recursively while edge removal cannot. This recursive ability is the main advantage of shell transformation technique.

252 Here, given the covering mesh of a shell, the transformation that attempts to reduce the

253 shell is denoted as *shell transformation*. The main *difference* between shell transformation  
 254 and edge removal is that shell transformation allows the output mesh containing a *partially*  
 255 *triangulated* skirt polygon while edge removal does not. If the output covering mesh of shell  
 256 transformation does not contain a *core*, shell transformation is the same operation as edge  
 257 removal. In this respect, edge removal could be considered a special case of shell  
 258 transformation. In other words, shell transformation could be considered as an enhanced  
 259 version of edge removal because it considers more possibility to mesh a shell. Moreover, *the*  
 260 *main advantage of shell transformations is rooted in its recursive ability*. As Joe have  
 261 demonstrated [6] that the composite transformations of elementary flips could improve the  
 262 quality of a mesh to a much higher level than elementary flips, and the recursive callings of  
 263 shell transformations, acting like composite edge removal transformations, could perform  
 264 much better than edge removal in most mesh improvement tasks, as we will demonstrate in  
 265 Section 6.



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 267  
 268 **Figure 5.** Illustration for the recursive callings of shell transformations. (a) The input mesh.  
 269 (b) The output after the first shell transformation calling on the shell of  $ab$ . (c) The output  
 270 after the second shell transformation calling on the shell of  $bh$ . (d) The final output after the  
 271 third shell transformation calling on the shell of  $ab$ .

## 272 4. A SINGLE CALLING OF SHELL TRANSFORMATION

273 To implement a shell transformation procedure, a key step is to develop an algorithm that can  
 274 triangulate a skirt polygon *partially*. Meanwhile, among all of the valid triangulation schemes,  
 275 an algorithm needs to find the triangulation corresponding to an *optimal* covering mesh. In  
 276 this section, we will first reproduce a dynamic programming algorithm suggested by  
 277 Shewchuk [10], which can be used to triangulate the skirt polygon *completely* and optimally.  
 278 Next, the proposed shell transformation algorithm is described in details, which enhances the  
 279 Shewchuk's algorithm in order to triangulate the skirt polygon *partially* and optimally.

### 280 4.1 The algorithm proposed by Shewchuk [10]

281 Given a shell covered by a set of tetrahedra  $p_i p_{i+1} ab (i=1, 2, \dots, m)$ , each triangulation  
 282  $T = \{t_1, t_2, \dots, t_{m-2}\}$  of the skirt polygon induces a tetrahedralization of the shell:

$$283 \quad K = \{\text{conv}(t_i, a) \cup \text{conv}(t_i, b) \mid i=1, 2, \dots, m-2\}.$$

284 Here,  $\text{conv}(\cdot)$  refers to a tetrahedron formed by a face and a node. We define the quality of  
 285  $T$  to be the quality of the worst tetrahedron within the tetrahedralization  $K$ .

286 For each node  $p_i (1 \leq i \leq m)$ ,  $p_{i+m}$  is its alias.  $R_{i,j}$  defines a ring of edges whose ending  
 287 nodes are



$$P_{i,j} = \begin{cases} \{p_i, p_{i+1}, \dots, p_j\} & 1 \leq i < j \leq m \\ \{p_j, p_{j+1}, \dots, p_{i+m}\} & 1 \leq j < i \leq m \end{cases}.$$

Selecting a node  $p_k \in P_{i,j}$  other than  $p_i$  and  $p_j$ , the triangulation of  $R_{i,j}$  includes three parts, as shown in Figure 6:

$$T_{i,j} = T_{i,k} \cup T_{k,j} \cup \Delta p_i p_k p_j.$$

We can define a matrix  $M_q$  to record the quality of the optimal triangulation of  $R_{i,j}$  as:

$$M_q(i, j) = \max_{p_k \in P_{i,j}, p_k \neq p_i, p_j} \min\{M_q(i, k), M_q(k, j), q(a, p_i, p_j, p_k), q(p_i, p_j, p_k, b)\}, \quad (1)$$

where  $q(\cdot)$  is the quality function for tetrahedral elements.  $M_q(i, j) = \infty$  when  $j = i + 1$ . Since  $1 \leq i, j \leq m$  ( $m$  refers to the size of the skirt polygon), we know  $M_q$  is a  $m \times m$  matrix.

Based on Equation 1, Algorithm 1 fills in the upper triangular part of  $M_q$  by an order of decreasing  $i$  and increasing  $j$  so that  $M_q(i, k)$  and  $M_q(k, j)$  are computed before  $M_q(i, j)$ . Meanwhile, Algorithm 1 fills in another matrix  $M_k$  that records the values of  $k$  that maximize Equation 1 in order to reconstruct the optimal triangulation  $T^{\text{opt}}$ :

$$\begin{cases} T^{\text{opt}} = T_{1,m}^{\text{opt}} \\ T_{i,j}^{\text{opt}} = T_{i,k}^{\text{opt}} \cup T_{k,j}^{\text{opt}} \cup \Delta p_i p_k p_j \quad (k = M_k(i, j)) \end{cases}.$$

**Algorithm 1.** Filling in the upper triangular part of  $M_q$  and  $M_k$

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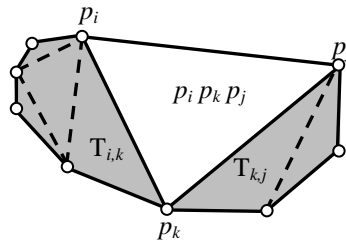
**fillInMatrices\_UpRight**( $a, b, P, M_q, M_k$ )

**Inputs:**

$a$  and  $b$ : the support nodes of a shell,

$P = \{p_1, p_2, \dots, p_m\}$ : the skirt polygon of a shell

1. **for**  $i = m - 2$  **downto** 1
  2.     **for**  $j = i + 2$  **to**  $m$
  3.         **for**  $k = i + 1$  **to**  $j - 1$
  4.              $q = \min\{q(a, p_i, p_k, p_j), q(p_i, p_k, p_j, b)\}$
  5.             **if**  $k < j - 1$
  6.                  $q = \min\{q, M_q(k, j)\}$
  7.             **if**  $k > i + 1$
  8.                  $q = \min\{q, M_q(i, k)\}$
  9.             **if**  $k = i + 1$  **or**  $q > M_q(i, j)$
  10.                  $M_q(i, j) = q$
  11.                  $M_k(i, j) = k$
- 



$$T_{i,j} = T_{i,k} \cup T_{k,j} \cup \Delta p_i p_k p_j$$

**Figure 6.** Decomposing a triangulation optimization problem into sub-problems.

## 306 4.2 The proposed shell transformation algorithm

307 Firstly, we introduce the concept of *triangulation graph*. It is a directed graph defined on a  
 308 polygon that is bounded by a set of nodes  $P = \{p_1, p_2, \dots, p_m\}$ . A *graph node* corresponds to a  
 309 polygon node, and a graph edge  $\langle p_i, p_j \rangle$  exists if there are valid triangulations for the ring  
 310  $R_{i,j}$ . Note that  $\mathbf{M}_q$  provides a representation of the triangulation graph:  $\mathbf{M}_q(i, j) > 0$   
 311 means that there is a valid triangulation for the ring  $R_{i,j}$ ; thus,  $\langle p_i, p_j \rangle$  is a graph edge.

312 Algorithm 1 only fills in one half of  $\mathbf{M}_q$  and  $\mathbf{M}_k$ . To get a complete representation of  
 313 the triangulation graph, the lower left elements of  $\mathbf{M}_q$  and  $\mathbf{M}_k$  must be computed.  
 314 Algorithm 2 present a routine that fills in all of useful elements of  $\mathbf{M}_q$  and  $\mathbf{M}_k$ , which  
 315 could be located in either side of the main diagonal of the two matrices. In the new algorithm,  
 316 one diagonal of  $\mathbf{M}_q$  and  $\mathbf{M}_k$  is computed at a time in the increasing order of the size of  
 317  $R_{i,j}$  (i.e., number of vertices). Note that after calling Algorithm 2,  $\mathbf{M}_q(1, m)$ , and  
 318  $\mathbf{M}_q(2, 1)$ ,  $\dots$ , and  $\mathbf{M}_q(m, m-1)$  all records the quality of the optimal triangulation of the  
 319 complete skirt polygon, but with different start and end vertices. Based on Algorithm 2, we  
 320 could then implement a single calling of shell transformation (see Algorithm 3), where a key  
 321 step is to define the *core* of the shell. As shown in Figure 7, the core introduced in shell  
 322 transformation corresponds to a simple cycle of the triangulation graph of the skirt polygon.  
 323 Therefore, once the graph is set up by calling Algorithms 1 and 2, all of the simple cycles are  
 324 visited, and the optimal one is picked up to reconstruct the triangulation of the skirt polygon  
 325 (i.e., Line 5 of Algorithm 3). Here, a simple cycle is identified as optimal when it corresponds  
 326 to an optimal covering mesh (K). Note that the definition on the optimality of a mesh could be  
 327 application specific, see Section 4.3 for details.

### 328 **Algorithm 2.** Filling in $\mathbf{M}_q$ and $\mathbf{M}_k$

---

**fillInMatrices**( $a, b, P, \mathbf{M}_q, \mathbf{M}_k$ )

**Inputs:**

$a$  and  $b$ : the support nodes of a shell

$P = \{p_1, p_2, \dots, p_m\}$ : the skirt polygon of a shell

1. **for**  $d = 2$  **to**  $m - 1$
  2.     **for**  $i = 1$  **to**  $m$
  3.          $j' = i + d$
  4.         **for**  $k' = i + 1$  **to**  $j' - 1$
  5.              $j = j' > m ? j' - m : j'$
  6.              $k = k' > m ? k' - m : k'$
  7.              $q = \min\{q(a, p_i, p_k, p_j), q(p_i, p_k, p_j, b)\}$
  8.             **if**  $k' < j' - 1$
  9.                  $q = \min\{q, \mathbf{M}_q(k, j)\}$
  10.             **if**  $k' > i + 1$
  11.                  $q = \min\{q, \mathbf{M}_q(i, k)\}$
  12.             **if**  $k' = i + 1$  **or**  $q > \mathbf{M}_q(i, j)$
  13.                  $\mathbf{M}_q(i, j) = q$
  14.                  $\mathbf{M}_k(i, j) = k$
- 

329 Assuming that the node set of the core is  $P_c = \{p_{c_1}, \dots, p_{c_n}, p_{c_{n+1}} = p_{c_1}\}$ , the reconstructed  
 330 triangulation is (see Figure 7):  
 331

$$332 \quad T^{\text{opt}} = \{T_{c_j, c_{j+1}}^{\text{opt}} \mid j = 1, 2, \dots, n\}$$

333 The new covering mesh of the shell is:

$$K = K_1 \cup K_2$$

$$334 \quad K_1 = \{\text{conv}(t_i, a) \cup \text{conv}(t_i, b) \mid t_i \in T^{\text{opt}}\}, \quad (2)$$

$$K_2 = \{\text{tetr}(p_{c_j}, p_{c_{j+1}}, a, b) \mid j = 1, 2, \dots, n\}$$

335 where  $\text{tetr}(\cdot)$  refers to the tetrahedral element formed by four specified nodes.

336 **Algorithm 3.** A general routine of shell transformation

---

**shellTransformation**( $a, b, P, K_{\text{old}}$ )

**Inputs:**

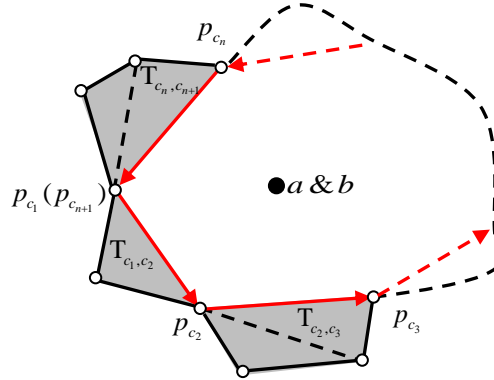
$a$  and  $b$ : the support nodes of the shell

$P = \{p_1, p_2, \dots, p_m\}$ : the skirt polygon of the shell

$K_{\text{old}}$ : the old covering mesh of the shell

1. **fillInMatrices\_UpRight**( $a, b, P, M_q, M_k$ )
  2. **fillInMatrices\_LowLeft**( $a, b, P, M_q, M_k$ )
  3.  $G$ : the triangulation graph with  $M_q$  as its matrix representation
  4.  $P_c$ : an *optimal* simple cycle of  $G$
  5.  $T^{\text{opt}}$ : the reconstructed triangulation from  $P_c$ , see Figure 6
  6.  $K$ : the new covering mesh of the shell, see Equation 2
  7. **if**  $K \neq \emptyset$  **and**  $K \neq K_{\text{old}}$
  8. Remesh the shell by replacing  $K_{\text{old}}$  with  $K$
- 

337



338

339 **Figure 7.** Illustration for the triangulation graph of a polygon, where those lines with arrows  
 340 are a group of graph edges, forming a simple cycle and thus defining a scheme that  
 341 triangulates the polygon *partially*.

## 342 5. RECURSIVE CALLINGS OF SHELL TRANSFORMATIONS

### 343 5.1 The basic routine

344 A single shell transformation calling only involves a small number of elements. It may not be  
 345 able to reduce a shell completely because of the constraints on the shell boundaries. Hence,  
 346 we develop a routine that calls shell transformations recursively to remove these constraints.

347 Algorithm 4 details the routine of recursive shell transformations. Given an edge  $e$ , the  
 348 calling **recursiveST**( $e, \emptyset, 0, l_{\text{max}}$ ) attempts to remove the edge  $e$ , where  $l_{\text{max}}$  limits the

349 maximally allowed recursive level. Given a face  $f$  and one of its boundary edges  $e$ , the calling  
 350 **recursiveST**( $e, f, 0, l_{\max}$ ) attempts to remove the face  $f$ .

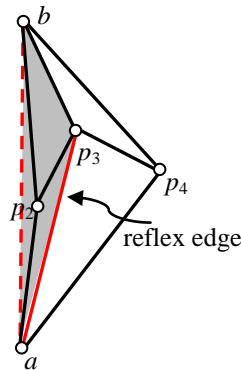
351 Note that Algorithm 4 expands an *edge tree*, where the input edge  $e$  is the *root* of this tree,  
 352 and those link edges ( $e'$ ) inputted for further recursions are *children* of the edge  $e$ . In this  
 353 manner, the tree can be expanded recursively. If a tree node  $v_1$  is an *ancestor* of another tree  
 354 node  $v_2$ , we say the edge corresponding to  $v_1$  is an *ancestor edge* of the edge corresponding to  
 355  $v_2$ .

### 356 5.2 Termination and efficiency

357 Two routines called by Lines 10 and 3 account for the timing-related performance of  
 358 Algorithm 4, namely **pickRecursiveLinkEdge** and **shellTransformation**. They determine  
 359 how many shell transformations are executed and how fast a single shell transformation can  
 360 run, respectively.

361 Given a supporting face  $f$  in the shell of  $e$ , the routine **pickRecursiveLinkEdge** checks  
 362 whether or not a further recursion is necessary. If yes, the routine returns a link edge between  
 363 two possible candidates. To filter inefficient recursions, the implementation of this routine can  
 364 be further improved through the following guidelines:

- 365 (1) Do not return a boundary edge.
- 366 (2) If a tetrahedra sharing  $f$  overlaps with the shells of an ancestor edge of  $e$ , return  
 367 nothing.
- 368 (3) Return a *reflex edge* only. In Figure 8, two faces  $ap_2p_3$  and  $ap_3p_4$  form a *reflex angle* if  
 369 viewed from a point  $b$ ; correspondingly,  $ap_3$  is called a *reflex edge* of the face  $abp_3$ .



370  
 371 **Figure 8.** Illustrative case of a *reflex edge*.

372 The termination of Algorithm 4 remains an issue because there is no guarantee that a mesh  
 373 edge could be removed by flips under the requirement that the quality of the mesh could not  
 374 be decreased by these flips. Although the second guideline mentioned above prevents an  
 375 infinite execution of the recursive callings, Algorithm 4 could still be very time-consuming  
 376 because the number of its shell transformation callings may increase exponentially when the  
 377 recursive level increases. Therefore, a user parameter  $l_{\max}$  is input to Algorithm 4 to limit the  
 378 maximal recursive level. Based on an analysis of many trial and error experimental results, we  
 379 choose  $l_{\max}$  to be 5 in this study to meet our goal of developing a cost-effective mesh  
 380 improver.

381 The routine **shellTransformation** (see Algorithm 3) includes two main steps. Firstly, it  
 382 employs Algorithms 1 and 2 to obtain the triangulation graph of the skirt polygon. Next, it  
 383 searches for a simple cycle in that graph to reconstruct an optimal covering mesh according to  
 384 Equation 2. The time consumptions of both steps are at an order of  $O(m^3)$ , where  $m$  is the  
 385 number of skirt nodes. A single calling of this routine may consume little time because  $m$  is  
 386 very small. However, the number of callings could be very large because of the recursive  
 387 nature of Algorithm 4. In our implementation, this routine has been speeded up through three  
 388 treatments as follows:

- 389 (1) Introduce the validity conditions (see Section 4.3) in the first step to simplify the  
390 triangulation graph;  
391 (2) Search for the *optimal* simple cycle first to prevent those unnecessary searches for low-  
392 quality cycles;  
393 (3) Record the results of time-consuming mesh validity and quality computations after  
394 they are executed for the first time so that simple queries can replace the repeated  
395 callings of these computations.

396 **Algorithm 4.** The routine of recursive shell transformations

---

**recursiveST**( $e, f, l, l_{\max}$ )

**Inputs:**

- the supporting edge, denoted  $e$
- a face containing  $e$  that the routine attempts to remove, denoted  $f$
- the recursive level with an initial value of zero, denoted  $l$
- the maximally allowed recursive level, denoted  $l_{\max}$

**Variables:**

- the ending nodes of an edge, denoted  $a(\cdot)$  and  $b(\cdot)$
- the skirt polygon of the shell of an edge, denoted  $P(\cdot)$
- the set of elements containing an edge, denoted  $S(\cdot)$
- a set of link faces contained in  $S(\cdot)$ , denoted  $F(\cdot) = \{f_1', f_2', \dots, f_m'\}$ , where  $m = |F(\cdot)|$

1. **if**  $|S(e)| \leq 0$  **or** ( $f \neq \emptyset$  **and**  $f \notin F(S(e))$ )
  2.     **return** success
  3.     **shellTransformation**( $a(e), b(e), P(e), S(e)$ )
  4. **if**  $|S(e)| \leq 0$  **or** ( $f \neq \emptyset$  **and**  $f \notin F(S(e))$ )
  5.     **return** success
  6. **if**  $l \geq l_{\max}$  /\* the recursive level is limited under  $l_{\max}$ . \*/
  7.     **return** fail
  8.      $m = |S(e)|$  /\* record the size of the shell  $S(e)$  \*/
  9.     **for**  $i = 1$  **to**  $m$
  10.      $e' = \text{pickRecursiveLinkEdge}(f_i')$  /\* filters are set to avoid inefficient recursions \*/
  11.     **if**  $e' \neq \emptyset$
  12.         **recursiveST**( $e', f_i', l+1, l_{\max}$ ) /\* recursive calling \*/
  13.     **if**  $|S(e)| < m$  /\*  $S(e)$  is reduced as well \*/
  14.         retrun **recursiveST**( $e, f, l, l_{\max}$ ) /\* recursive calling \*/
  15. **return** fail
- 

397 Another factor that affects the efficiency of Algorithm 4 is the routine that identifies a shell  
398 in the input mesh (referring to the *shell-find* routine thereafter), which is employed by Lines 4  
399 and 13 of Algorithm 4. The efficiency of this routine depends on the data structure adopted to  
400 represent a tetrahedral mesh. In our scheme, four incident vertices are stored for each  
401 tetrahedron, plus four neighboring elements of this tetrahedron. Meanwhile, for each mesh  
402 vertex, one element incident to this vertex is stored. This data structure requires a small  
403 amount of memories, and the *shell-find* routine based on it only needs to traverse the elements  
404 locally. Given two ending vertices of an edge (denoted by  $v_1$  and  $v_2$ , respectively), the *shell-*  
405 *find* routine is separated into two phases. In the first phase, one element that contains the input  
406 edge is searched by the following steps:  
407

- 408 (1) Get the stored element incident to  $v_1$  and push it into a stack;
- 409 (2) If the stack is empty, exit the routine and return NULL; otherwise, remove the top

- 410 element from the stack and go to Step 3;  
411 (3) If the top element contains  $v_2$ , return the top element and exit the routine;  
412 (4) Flag the top element as *visited*, and then visit its neighboring elements. For any  
413 unvisited neighbor, if it contains  $v_1$  as well, push it into the stack;  
414 (5) Go back to Step 2.

415 If no valid element is returned by the above procedure, the shell is empty; otherwise,  
416 starting from the returned element, the entire shell can be visited by using the neighboring  
417 indices of elements. In the worst scenarios, the first phase visits all elements surrounding  $v_1$ ,  
418 and the number of such elements is close to 30 on average for real mesh examples. The  
419 second phase visits all elements surrounding the edge, and the number of such elements is  
420 about 5-7 on average<sup>‡</sup>. Therefore, the first phase dominates a general calling of the routine in  
421 terms of computing time. However, in many circumstances, one element surrounding an edge  
422 is stored somewhere before calling the shell-find routine. By using this element as an extra  
423 input, the timing-related performance of the shell-find routine can be improved remarkably by  
424 skipping over the first phase.

### 425 5.3 Validity and optimality conditions.

426 The shell transformation routine presented in Algorithm 3 needs to output an *optimal*  
427 covering mesh of a shell among all *valid* ones. Here, the definitions of validity and optimality  
428 depend on specific application purposes. For mesh quality improvement, three types of  
429 validation conditions are set for covering meshes as:

- 430 (1) *The basic condition*, which requires all elements have positive volumes.  
431 (2) *The recursive condition*, which requires each shell transformation should create no  
432 supporting faces around any *ancestor edge* of the current supporting edge.  
433 (3) *The application specific condition*, in the context of mesh improvement, which requires  
434 the quality of the output covering mesh of the shell should be higher than the quality of  
435 the input covering mesh.

436 It is possible that more than one *valid* covering mesh exists for a shell. The final output of a  
437 shell transformation calling is the covering mesh with the highest possible quality. See  
438 Section 5.1 for the definition of *mesh quality*.

### 439 5.4 The shell transformation based local reconnection scheme

440 If one edge or face of a low-quality element is removed, the element will be removed  
441 accordingly. Based on this concept, Algorithm 5 presents a local reconnection scheme that  
442 attempts to remove low-quality elements by removing the edges or faces of these elements.

443 All of low-quality elements are stored in a heap in an ascending order of the element  
444 quality. Firstly, Algorithm 4 is called on an edge of the first element of the heap. If the  
445 element is removed by Algorithm 4, Algorithm 4 succeeds; otherwise, Algorithm 4 is  
446 repeated on another edge of the element until all edges of the element are attempted. To  
447 protect the mesh boundary, the edges attempted for removal must be interior edges of the  
448 mesh. Next, if Algorithm 4 fails to remove the element, we attempt to remove the faces of this  
449 element individually. To protect the mesh boundary, the faces attempted for removal must be  
450 interior faces of the mesh. The 2-3 flip shown in Figure 1a is the simplest local scheme for  
451 face removal. However, a more effective alternative is *multi-face removal* (see Figure 1c).

---

<sup>‡</sup> It is worth noting that both numbers could vary case by case, depending on mesh topologies. Nevertheless, the meshes considered in this study are inputs for numerical simulations, where only a small percentage of elements are badly shaped. For different meshes of this type, it is observed that both numbers usually remain within the ranges we mentioned in the text. For instance, for the unimproved F16 and Bridge meshes to be presented in Section 7 for tests, the average numbers of elements surrounding interior mesh nodes are both 5.50. After mesh improvement, these two numbers are reduced to 5.21 and 5.20 respectively. The numbers of elements surrounding interior mesh edges are 26.05 and 26.21, respectively. After mesh improvement, these two numbers are reduced to 23.82 and 23.80 respectively.

452 Shewchuk [10] suggested an implementation of multi-face removal. In this study, we present  
453 an alternative solution based on the proposed shell transformation routine (i.e., Algorithm 3).

454 Given an interior face  $f$  for removal, the proposed algorithm takes the following steps as:

- 455 (1) Find two tetrahedra sharing face  $f$ , and denote the apexes of the tetrahedra opposite to  $f$   
456 as  $a$  and  $b$ , respectively.
- 457 (2) Find all of the faces opposite to  $a$  and  $b$ . As shown in Figure 9a, these faces may form  
458 several connected components.
- 459 (3) Select a component that includes the face (or faces) intersected by  $ab$ , as shown in  
460 Figure 9b.
- 461 (4) Define the boundary of the selected component as a polygon. If a point (such as  $p_7$  in  
462 Figure 9c) is contained in the interior of the polygon, remove one face (such as the face  
463  $p_1p_2p_7$  in Figure 9c) incident on this point from the component.
- 464 (5) Now we get a local mesh like the one shown in the left of Figure 1c. With this mesh as  
465 the input, *the shell transformation routine* (i.e., Algorithm 3) is called to search for a  
466 better covering mesh to fill in the shell region.

467 To avoid an infinite execution of the loop defined in Lines 2-16 of Algorithm 5, no matter  
468 the element for removal is removed or not, this element must be removed from the heap  
469 before the next iteration.

470 **Algorithm 5.** The combinational edge removal based on recursive shell transformation

---

**localReconnection**( $M, l_{\max}$ )

**Inputs:**

the mesh to be improved, denoted  $M$

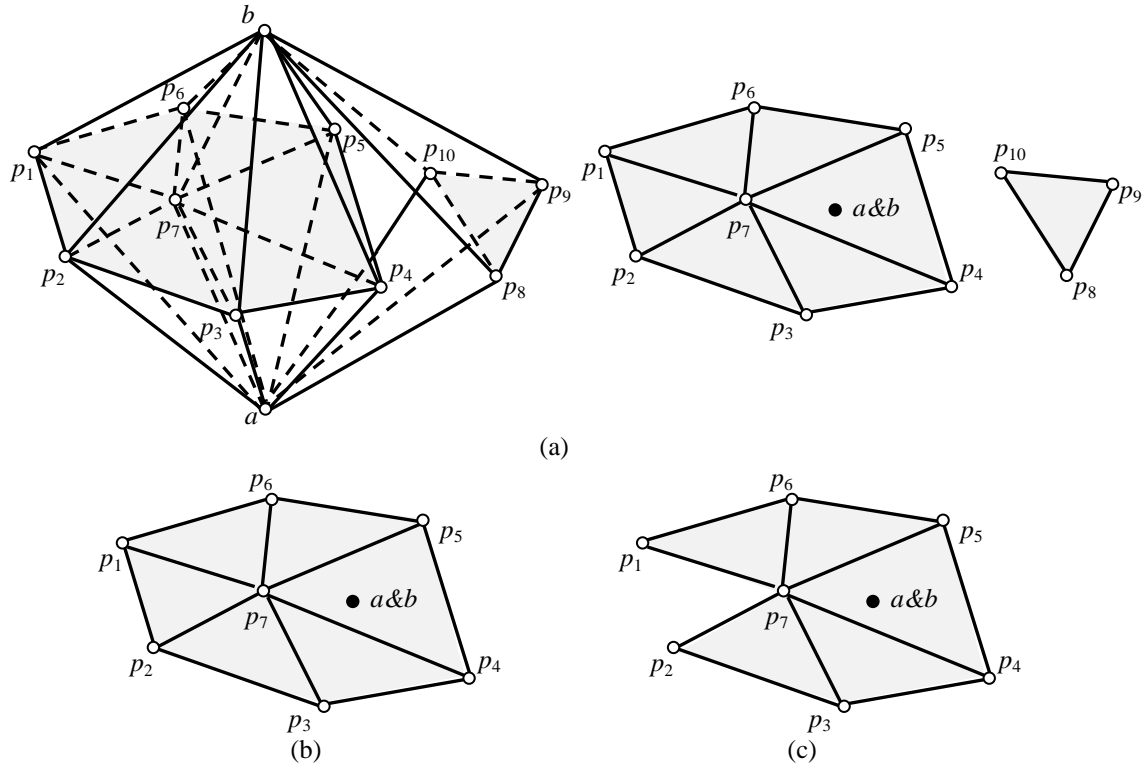
the maximally allowed recursive level, denoted  $l_{\max}$  and the default value is 5

**Variables:**

the heap that stores all of low-quality elements,  $T_{\text{bad}}$

1. Insert all of low-quality elements into  $T_{\text{bad}}$  in the ascending order of the element quality
  2. **while**  $T_{\text{bad}}$  is not empty
  3.      $t$ : the first element of  $T_{\text{bad}}$
  4.     **If**  $t$  has been removed from  $M$
  5.         **goto** line 16
  6.      $E = \{e_1, e_2, \dots, e_n\}$ : the set of edges of  $t$  qualified for removal ( $n \leq 6$ )
  7.     **for**  $j = 1$  **to**  $n$
  8.         **recursiveST**( $e_j, \emptyset, 0, l_{\max}$ )
  9.         **if**  $t$  is removed
  10.             **goto** line 16
  11.      $F = \{f_1, f_2, \dots, f_m\}$ : the set of faces of  $t$  qualified for removal ( $m \leq 4$ )
  12.     **for**  $j = 1$  **to**  $m$
  13.         Remove  $f_j$  by a shell transformation based multi-face removal routine
  14.         **if**  $t$  is removed
  15.             **break**
  16.     Remove  $t$  from  $T_{\text{bad}}$
- 

471  
472



473  
474

475  
476  
477

**Figure 9.** The procedure that prepares the inputs for the multi-face removal operation.

478

## 6. THE OVERALL MESH IMPROVEMENT ALGORITHM

479 The application goal of this study is to develop a cost-effective mesh improver. In this  
480 section, we first present some basic considerations that guide this development procedure.  
481 Then, we introduce the set of smoothing, point insertion and point suppression schemes  
482 incorporated in our mesh improver. Finally, the overall mesh improvement scheme is detailed.

### 483 6.1. The basic considerations

484 In this study, the *minimum sine* of dihedral angles is used as a default quality measure. The  
485 quality measure of a mesh is evaluated by a vector listing the quality of each tetrahedron  
486 contained by the mesh, in an order from the worst to the best. Since the worst tetrahedron in a  
487 mesh has far more influence than those average tetrahedra, the quality vectors of two meshes  
488 are compared *lexicographically* so that, for instance, an improvement in the second-worst  
489 tetrahedron improves the overall mesh quality even if the worst tetrahedron has not changed.

490 To ensure the heuristic algorithm never worsens the quality of a mesh, a *hill-climbing*  
491 method is adopted in all of the developed local schemes, which considers applying a local  
492 operation only if the quality of the changed mesh will be better than that of the original mesh.  
493 Local operations that do not improve the mesh quality are not applied. The method stops  
494 when no operation can achieve further improvement (i.e., the mesh is already locally optimal),  
495 or when a further optimization promises too little gain.

496 Meanwhile, since the focus is usually on the worst tetrahedron, only *bad elements* are  
497 treated to save the computing time, which refer to those elements whose minimum sine values  
498 are less than 0.5 in the following discussions, i.e., at least one dihedral angle of the element is  
499 either below  $30^\circ$  or above  $150^\circ$ .

500 The surface boundary of a volume mesh influences the mesh quality considerably. In the  
501 applications where the boundary can be changed to some extents, it is beneficial to extend the  
502 local schemes for mesh quality improvement from interior mesh entities to boundary entities.



503 However, in many applications, the improved mesh need to be consistent with a CAD model  
 504 or matched face-to-face with another mesh. To limit the discussions, the presented algorithm  
 505 regards the boundary configuration of the mesh as *untouchable*, i.e., vertices on the boundary  
 506 cannot be smoothed, and the connectivity between them cannot be changed.

### 507 6.2. Smoothing

508 To achieve the cost-effectiveness, we combine an optimization-based algorithm [20] with the  
 509 Laplacian smoothing to reposition each interior mesh point that is included by at least one bad  
 510 element (referred to as a *bad point* hereafter):

511 (1) Perform Laplacian smoothing. If the improved *ball* (referring to all of elements incident  
 512 on the point) contains no bad elements, the smoothing succeeds; otherwise, continue.

513 (2) Perform the optimization-based smoothing.

514 To save the smoothing time, a mesh point is flagged as *smoothed* after a successful  
 515 smoothing, and this flag is flushed only if the ball of the point is changed. In each smoothing  
 516 cycle, all of *non-smoothed bad* points are treated only once.

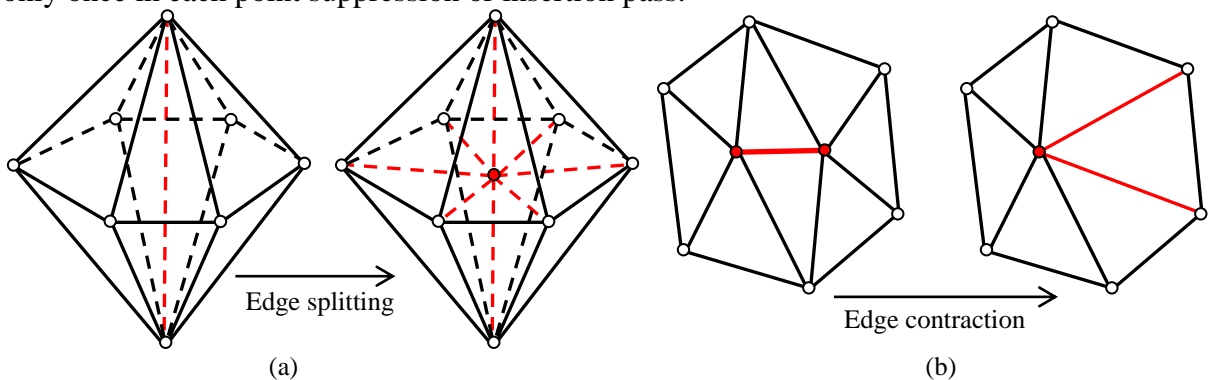
517 In each smoothing pass, the smoothing cycle is repeated until three indicators of the mesh  
 518 quality are not improved further: (1) the quality of the worst tetrahedral ( $q_{\text{worst}}$ ); (2) the  
 519 number of bad elements ( $n_{\text{bad}}$ ); and (3) the average quality of bad elements ( $q_{\text{aver}}$ ).

### 520 6.3. Point insertion and point suppression

521 We adopt an *edge-splitting* based point insertion scheme. It attempts to insert a point at the  
 522 middle of an interior edge and then to split those elements that meet at this edge; see Figure  
 523 10a. Finally, the new point is smoothed, and if the resulting mesh is better than the old one,  
 524 the mesh will be changed; otherwise, the old mesh is restored.

525 Besides, our mesh improver relies on an *edge contraction* operation to remove *bad* points  
 526 of the mesh. Figure 10b illustrates this operation using a 2D example. Each edge ended with  
 527 the point to be suppressed is contracted to the other endpoint, and the resulting mesh  
 528 configuration with the best quality is selected for further smoothing. To save computing time,  
 529 only the point that replaces the contracted edge is smoothed. The point suppression operation  
 530 fails if edge contraction (plus point smoothing) cannot produce a better mesh than the old one.

531 Since only bad elements are targeted, those edges included by bad elements are attempted  
 532 only once in each point suppression or insertion pass.



**Figure 10.** Illustration for the (a) edge splitting and (b) edge contraction operations.

### 536 6.4. The mesh improvement schedule

537 Algorithm 6 presents the proposed mesh improvement schedule, which combines different  
 538 local schemes to improve the mesh quality. This schedule begins with a smoothing pass, and  
 539 then executes the main loop of mesh improvement. In the main loop, a smoothing pass is  
 540 followed after the pass of each type of topological transformations to improve the mesh  
 541 quality further. The main loop is ended when three subsequent combinational passes fail to  
 542 make sufficient progress or the number of iteration steps exceeds a predefined threshold (in

543 the present study, the default value of this threshold is 30). We gauge progress using three  
 544 quality indicators mentioned in Section 5.4, i.e.,  $q_{\text{worst}}$ ,  $n_{\text{bad}}$  and  $q_{\text{aver}}$ .

545 **Algorithm 6.** The proposed mesh improvement schedule

---

**improveAMesh( $M$ )**

**Input:**

$M$ , the mesh to be improved

**Variables:**

$q_{\text{worst}}$ ,  $q'_{\text{worst}}$ , the quality of the worst tetrahedral

$n_{\text{bad}}$ ,  $n'_{\text{bad}}$ , the number of bad elements

$q_{\text{aver}}$ ,  $q'_{\text{aver}}$ , the average quality of bad elements

1.  $failed = 0$ ;  $itcount = 0$
  2. Smooth  $M$
  3. Query the mesh quality and store the indicators in  $q_{\text{worst}}$ ,  $n_{\text{bad}}$  and  $q_{\text{aver}}$ , respectively
  4. **while**  $failed < 3$  &&  $++itcount \leq 30$
  5.   **localReconnection**( $M$ , 5)
  6.   Smooth  $M$
  7.   Improve  $M$  by the point suppression scheme
  8.   Smooth  $M$
  9.   Improve  $M$  by the point insertion scheme
  10.   Smooth  $M$
  11.   Query the mesh quality and store the indicators in  $q'_{\text{worst}}$ ,  $n'_{\text{bad}}$  and  $q'_{\text{aver}}$ , respectively
  12.   **if** ( $q'_{\text{worst}} < q_{\text{worst}}$  ||  $n'_{\text{bad}} < n_{\text{bad}}$  ||  $q'_{\text{aver}} < q_{\text{aver}}$ )  $failed = failed + 1$
  13.   **else**  $failed = 0$
  14.    $q_{\text{worst}} = q'_{\text{worst}}$ ;  $n_{\text{bad}} = n'_{\text{bad}}$ ;  $q_{\text{aver}} = q'_{\text{aver}}$
- 

546

547

## 7. RESULTS

548 The numerical tests are conducted on a PC workstation (CPU: 3.5GHz, Memory: 24GB).  
 549 Results obtained from the developed mesh improver are compared with those obtained by  
 550 Grumpp (Version 0.3.4) and Stellar (Version 1.0), respectively. To our knowledge, Grumpp  
 551 [20, 32] and Stellar [12, 13] are among the best open-source improvers for tetrahedral meshes.  
 552 Although many common features exist between two codes, their differences are also evident  
 553 due to different start points of their development. The goal of Grumpp is to improve the  
 554 worst tetrahedra cost-effectively, while the goal of Stellar is to improve the worst tetrahedra  
 555 aggressively with speed as a secondary consideration.

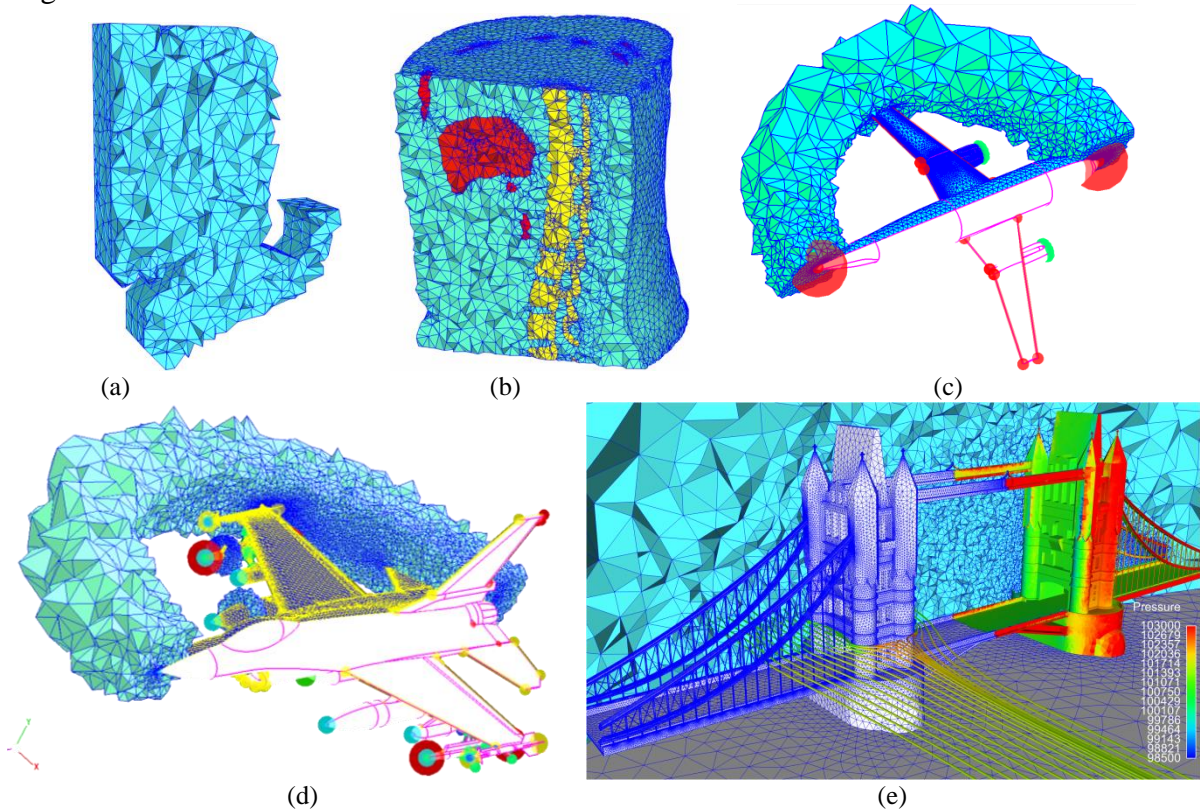
556 In default, Grumpp code takes four steps to improve a given mesh:

- 557 (1) Perform three passes of local reconnections for all elements.
- 558 (2) Perform two smoothing passes for elements containing angles below  $\theta$  degrees or above  
 559  $180 - \theta$  degrees. Here,  $\theta$  is a threshold initially set as  $25^\circ$  and then adaptively reduced  
 560 after each pass of smoothing operations.
- 561 (3) Repair a small fraction of the worst tetrahedra by a full range of swapping techniques.
- 562 (4) Repeat Step 2.

563 The default schedule of Stellar code begins with one smoothing pass, one local  
 564 reconnection pass and one edge contraction pass for all elements, and then combines these  
 565 local schemes and a point insertion scheme in a loop to improve the quality of a mesh  
 566 iteratively. Inside this loop, the smoothing and local reconnection routines target at all

567 elements, but the most passes of edge contraction and point insertion routines target at  
 568 elements with angles below  $40^\circ$  or above  $140^\circ$  except for a so-called *desperation pass*, which  
 569 targets at the worst 3.5% of tetrahedra.

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574 **Figure 11.** The selected meshes: (a) Tire; (b) Patient-organs02; (c) The F6 aircraft; (d) The  
 575 F16 aircraft; (e) The London Tower bridge. Surface meshes and grid sources for mesh sizing  
 576 control are displayed in both graph (c) and graph (d), and the surface mesh and inviscid flow  
 577 simulation results induced by a crosswind are displayed in graph (e), respectively.

578 As shown in Figure 11, five meshes are selected. The first two meshes are accessible from  
 579 the internet: the initial mesh of *tire* was ever analysed in literatures [20] and [32], and  
 580 included in the package of Grumpp [32] and Stellar [13], and the initial mesh of *patient-*  
 581 *organs02* is obtained from the AIM@SHAPE repository [33]. The last three meshes are  
 582 generated by our in-house codes following the same schedule as:

- 583 (1) Input a geometry model.
- 584 (2) Triangulate the surface by an advancing front technique.
- 585 (3) Tetrahedralize the volume by a Delaunay mesher [4].

586 The original geometry models of the last three examples are all accessible from the internet.  
 587 The London Tower Bridge model (referred to as *bridge* hereafter) and the DLR-F6 wing-  
 588 body-nacelle-pylon aircraft model (referred to as *F6* hereafter) are the selected test case  
 589 geometries for the meshing contest session of the 23rd International Meshing Roundtable  
 590 (IMR) and the 2nd AIAA CFD Drag Prediction Workshop, respectively. The F16 aircraft  
 591 model (referred to as *F16* hereafter) is obtained from GrabCAD [34].

592 Table 1 lists the initial mesh size statistics and the mesh quality data of those selected  
 593 examples, where  $\theta_{\min}$  and  $\theta_{\max}$  refer to the minimum and maximum dihedral angles, and  $\lambda$   
 594 refers to the percentage of bad dihedral angles, i.e., angles within the range of  $[0, 30^\circ]$  or  
 595  $[150^\circ, 180^\circ]$ , respectively. Meanwhile,  $\lambda_i$  ( $i=1-5$ ) is used to evaluate the distributions of bad  
 596 angles, which refers to the percentage of dihedral angles within the range of  $[6(i-1), 6i]$  or  
 597  $[180-6i, 180-6(i-1)]$  degrees. For instance,  $\lambda_2$  refers to the percentage of dihedral angles within  
 598 the range of  $6^\circ$  to  $12^\circ$  or  $168^\circ$  to  $174^\circ$ .

**Table 1.** The initial mesh size statistics and mesh quality data.

Examples	#tetra.	#points	$\theta_{\min}$ ( $^{\circ}$ )	$\theta_{\max}$ ( $^{\circ}$ )	% of bad angles ( $\lambda$ )	Distribution of bad angles (%)				
						$\lambda_1$	$\lambda_2$	$\lambda_3$	$\lambda_4$	$\lambda_5$
Tire	11,098	2,570	0.66	178.88	4.58	0.12	0.46	0.77	1.28	1.96
Patient-organs02	280,911	51,124	2.89	175.23	7.65	0.0072	0.21	1.20	2.45	3.78
F6	1,023,532	172,664	2.6e-13	$\approx 180$	6.65	0.23	0.68	1.19	1.82	2.73
F16	18,065,336	2,906,056	2.6e-14	$\approx 180$	6.63	0.23	0.67	1.18	1.82	2.74
Bridge	37,772,656	6,205,571	2.1e-13	$\approx 180$	6.72	0.23	0.68	1.19	1.85	2.77

600 In the first test, the performance data of different local reconnection schemes are compared  
601 with each other. Stellar code executes edge removal and multi-face removal routines  
602 repeatedly to improve mesh topology, while Grummp code executes 2-3 flips and edge  
603 removal routines repeatedly. We improve the five initial meshes by performing one pass of  
604 the three local reconnection schemes, respectively. Instead of improving all elements, only  
605 *bad* elements are treated in this test. Table 2 presents the mesh quality and the computing time  
606 data comparison.

607 For *patient-organs02*, *F6* and *F16* cases, our scheme not only achieves the lowest  
608 percentage of bad angles ( $\lambda$ ), but also narrows the ranges of dihedral angles to the largest  
609 extent. For *bridge* case, our scheme also achieves the lowest  $\lambda$ ; however, all of the three local  
610 reconnection schemes fail to improve both the smallest and the largest angles to an acceptable  
611 level, although the values achieved by our scheme are slightly better. For *tire* case, Grummp  
612 code improves  $\theta_{\min}$  and  $\theta_{\max}$  at the same level as our scheme. Meanwhile, Grummp code  
613 achieves a slightly better value of  $\lambda$  than our scheme, while our scheme achieves smaller  
614 values of  $\lambda_1$  and  $\lambda_2$ . We believe that, for this mesh, more angles between  $12^{\circ}$  and  $30^{\circ}$  (or  
615 between  $150^{\circ}$  and  $168^{\circ}$ ) are generated when our scheme attempts to remove small angles  
616 between  $0^{\circ}$  and  $12^{\circ}$  or large angles between  $168^{\circ}$  and  $180^{\circ}$ , because the cost of improving the  
617 worst angle is possibly increased considerably due to the generation of more undesirable  
618 small/large angles.

619 In this test, Stellar code achieves the best performance with respect to the computational  
620 time, while Grummp code performs rather well for small meshes but very poor for big  
621 meshes. For our scheme, one pass of the proposed local reconnection scheme consumes more  
622 time than its counterpart in Stellar code, because of its recursive nature. Nevertheless, since  
623 the proposed scheme produces a much better mesh, this marginally more time consumption is  
624 acceptable.

**Table 2.** The mesh quality and computational time data for different local reconnection schemes.

Examples	#tetra.	$\theta_{\min}$ ( $^{\circ}$ )	$\theta_{\max}$ ( $^{\circ}$ )	$\lambda$	Distribution of bad angles (%)					Time (s)	
					$\lambda_1$	$\lambda_2$	$\lambda_3$	$\lambda_4$	$\lambda_5$		
Tire	Grummp	11,019	3.36	172.38	4.21	0.069	0.30	0.57	1.18	2.09	0.09
	Stellar	10,936	3.00	172.38	4.23	0.056	0.31	0.59	1.23	2.05	0.07
	Present	10,906	3.36	172.38	4.42	0.047	0.27	0.61	1.30	2.19	0.21
Patient-organs02	Grummp	265,086	5.68	165.38	3.46	6.3e-5	3.3e-3	0.12	0.96	2.37	2.6
	Stellar	261,485	6.55	167.70	2.98	0	2.4e-3	0.089	0.79	2.09	2.6
	Present	259,959	11.21	162.59	2.81	0	6.4e-5	0.034	0.67	2.10	5.9
F6	Grummp	955,584	0.78	178.4	0.75	3.0e-4	2.0e-3	0.016	0.10	0.63	9.8
	Stellar	951,623	0.73	178.7	0.58	3.7e-4	4.3e-3	0.022	0.094	0.46	4.6
	Present	946,534	3.69	174.85	0.31	7.0e-5	4.8e-4	2.5e-3	0.018	0.29	5.7
F16	Grummp	16,854,123	2.4e-4	$\approx 180$	0.74	3.9e-4	2.3e-3	0.018	0.11	0.61	258.0
	Stellar	16,791,528	1.3e-4	$\approx 180$	0.57	9.5e-4	4.7e-3	0.024	0.097	0.45	72.6
	Present	16,682,773	3.80	174.20	0.31	1.9e-5	4.1e-4	4.1e-3	0.025	0.28	85.3
Bridge	Grummp	35,233,789	3.0e-5	$\approx 180$	0.81	2.5e-4	2.0e-3	0.017	0.11	0.67	542
	Stellar	35,081,252	9.2e-5	$\approx 180$	0.62	4.0e-4	3.8e-4	0.022	0.10	0.50	160.8
	Present	34,853,521	8.6e-4	$\approx 180$	0.36	1.5e-5	1.6e-4	1.9e-3	0.025	0.33	192.2

627 In the second test, we compare the default schedules of Grummp code, Stellar code and the

628 proposed algorithm (i.e., Algorithm 6). In this test, the option that prohibits the change on the  
629 mesh surface is enabled for both Grummp and Stellar codes. In addition, because the first pass  
630 of edge contraction in Stellar code can coarsen the input mesh dramatically, this pass is thus  
631 disabled in this test.

632 Table 3 presents the mesh quality and computational time data from the second test. In all  
633 of the cases, Grummp code outputs the worst quality meshes. For *F16* and *bridge* cases, the  
634 meshes output by Grummp code contain extremely small and/or large angles, while Stellar  
635 code and our algorithm can improve them to an acceptable level for further numerical  
636 simulations. We believe that the following facts might account for the relatively poor  
637 performance of Grummp code. Firstly, Grummp code does not incorporate any point  
638 suppression and point insertion schemes. In practice, these two schemes are useful for  
639 eliminating extreme small and/or large angles of a mesh. Secondly, Grummp code only  
640 executes a fixed number of passes of swapping and smoothing operations. In both Stellar code  
641 and our algorithm, the adopted scheduling strategies that combine local mesh improvement  
642 schemes are far more aggressive.

643 **Table 3.** The mesh quality and computational time data of the default schedules of Grummp,  
644 Stellar and our improved method.

Examples		#tetra.	#points	$\theta_{\min}$ ( $^{\circ}$ )	$\theta_{\max}$ ( $^{\circ}$ )	$\lambda$	Distribution of bad angles (%)					Time (s)
							$\lambda_1$	$\lambda_2$	$\lambda_3$	$\lambda_4$	$\lambda_5$	
Tire	Grummp	11,039	2,570	13.67	158.55	1.7	0	0	0.030	0.21	1.47	0.5
	Stellar	10,973	2654	23.4	148.1	0.15	0	0	0	0.017	0.13	106
	Present	11,840	2,751	20.67	157.45	0.26	0	0	0	0.018	0.24	1.0
Patient -organs02	Grummp	264,954	51,124	8.93	160.40	2.91	0	1.3e-4	1.6e-3	0.06	2.85	21
	Stellar	227,775	46,237	31.7	141.58	0	0	0	0	0	0	3,220
	Present	266,631	52,392	20.52	149.93	8.3e-3	0	0	0	3.8e-4	7.9e-3	16
F6	Grummp	955,512	172,664	0.78	178.4	0.65	1.7e-4	3.0e-4	1.1e-3	3.5e-3	0.65	43
	Stellar	918,434	171,912	18.2	158.7	5.9e-3	0	0	0	4.7e-4	5.4e-3	1,193
	Present	935,608	172,167	10.64	159.79	0.017	0	5.3e-5	3.2e-4	1.4e-3	0.015	26
F16	Grummp	16,854,105	2,906,056	2.4e-4	179.99	0.63	1.5e-4	3.2e-4	2.6e-3	0.010	0.62	884
	Stellar	16,360,678	2,910,076	3.8	174.2	0.027	6.1e-6	3.1e-6	1.1e-4	9.1e-4	0.026	9,882
	Present	16,666,517	2,923,840	3.8	174.2	0.013	6.0e-6	1.3e-5	7.7e-5	1.2e-3	0.012	522
Bridge	Grummp	35,233,695	6,205,571	3.0e-5	179.99	0.71	1.1e-4	2.0e-4	8.6e-4	6.8e-3	0.70	1,521
	Stellar	34,188,981	6,211,120	7.5	171.5	0.13	0	1.0e-4	4.2e-4	9.5e-4	0.13	21,779
	Present	33,698,234	6,091,180	5.79	172.55	0.13	9.9e-7	7.0e-5	4.0e-4	1.9e-3	0.13	1,412

645 In the second test, Stellar code outputs the best quality meshes in most occasions, and it not  
646 only narrows the range of dihedral angles at most, but also reduces the percentage of bad  
647 angles as well. For instance, for *patient-organs02*, Stellar code improves the smallest and the  
648 largest angles to be  $31.7^{\circ}$  and  $141.6^{\circ}$ , respectively. In other words, the improved mesh does  
649 not contain any bad angles. The mesh improved by our algorithm is slightly worse, which  
650 contains 132 angles below  $30^{\circ}$  (of which 6 angles below  $24^{\circ}$ ), but no angles above  $150^{\circ}$ . For  
651 *F16* and *bridge* cases, the quality levels of the improved meshes by Stellar code and our  
652 algorithm are very close.

653 With respect to the computational time performance, the proposed algorithm performs the  
654 best in most cases, apart from the improvement of the mesh *tire*. For this smallest mesh,  
655 Grummp code performs the best in terms of the timing performance. Here, we define a  
656 *velocity* index to evaluate the timing performance as:

657 
$$v = \frac{\text{the number of elements contained in the input mesh}}{\text{the total computing time.}}$$

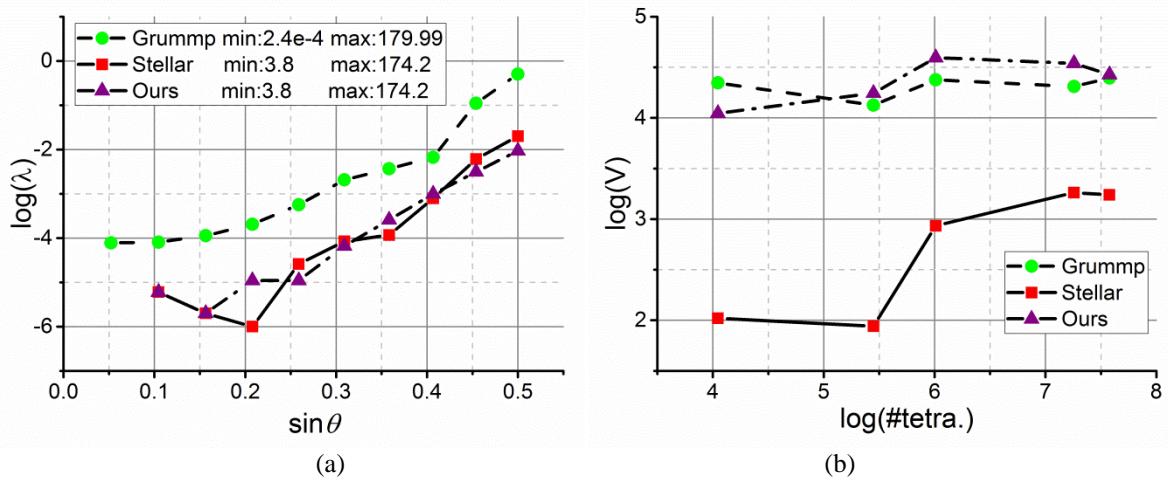
658 For those three inputs produced by the Delaunay mesher, i.e., *F6*, *F16* and *bridge*, Stellar  
659 code runs at a speed of 45.9, 18.9 and 15.4 times slower than that of the proposed algorithm,  
660 respectively. It is worth noting that the adopted Delaunay mesher runs very fast. For instance,  
661 the generation of the initial mesh of *bridge* consumes only 174 seconds, while the proposed  
662 algorithm can improve it to an acceptable mesh quality level for simulations in about 1,412  
663 seconds. However, if replacing the proposed algorithm by Stellar code, it will take about 6

664 hours to achieve only marginal improvement (compared with our results) in terms of the mesh  
 665 quality. In this respect, the proposed algorithm is undoubtedly a more *cost-effective* choice  
 666 than the current default schedule of Stellar code.

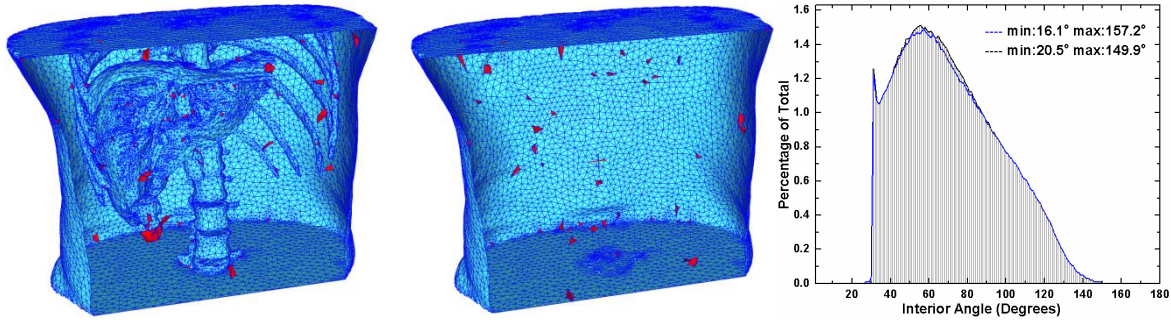
667 To demonstrate the performance difference between three mesh improvers more clearly,  
 668 Figure 12a compares the distributions of bad angles of three F16 meshes produced by  
 669 Grumpp code, Stellar code and our mesh improver. By default, each curve contains 10 data  
 670 points, and the  $\lambda$  value of the  $i$ th point refers to the percentage of bad dihedral angles within  
 671 the range of  $[3(i-1), 3i)$  or  $(180-3i, 180-3(i-1)]$  degrees ( $i=1-10$ ). Nevertheless, because the  
 672 meshes produced by Stellar code and our mesh improver contains no angles below  $3^\circ$  or  
 673 above  $177^\circ$ , their corresponding curves contains no data points referring to angles within this  
 674 range. Besides, Figure 12b compares the timing performance of three mesh improvers for five  
 675 test meshes of various sizes, evaluated by the velocity indices of these improvers mentioned  
 676 previously.

677 The  $\lambda$  value of each data point of the curve for Grumpp code is found larger than its  
 678 counterparts from Stellar code and our mesh improver by nearly one or two orders of  
 679 magnitude, while the  $\lambda$  values of data points of the curves for Stellar code and our mesh  
 680 improver are comparable in general. However, the velocity indices of Stellar code are lower  
 681 than their counterparts of Grumpp code and our mesh improver by one or two orders of  
 682 magnitude, while the velocity indices of Grumpp code and our improver is at the same order.  
 683 From the above analysis, we can conclude that our mesh improver presented in this study can  
 684 achieve an overall better balanced performance between the mesh quality and computational  
 685 time than other two state-of-the-art algorithms, and it is therefore more suitable for the quality  
 686 improvement tasks involving large-scale meshes.

687 It needs to be pointed out that the initial mesh of *patient-organs02* actually contains interior  
 688 constraints. However, because the current versions of Grumpp and Stellar codes provide no  
 689 options to input a mesh with interior constraints, all of the tests presented above choose not to  
 690 respect interior constraints. In fact, the proposed algorithm can respect interior constraints  
 691 very well. To demonstrate this, Figure 13 compares the meshes improved by the proposed  
 692 algorithm with or without interior constraints. Not surprisingly, the quality of the improved  
 693 mesh that respects interior constraints is slightly worse.



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 696 **Figure 12.** A comparison of Grumpp, Stellar and our improver in terms of mesh quality and  
 697 timing performance. (a) The distributions of bad angles of the F16 meshes produced by three  
 698 improvers. (b) The velocity indices of three improvers for five test meshes.



699

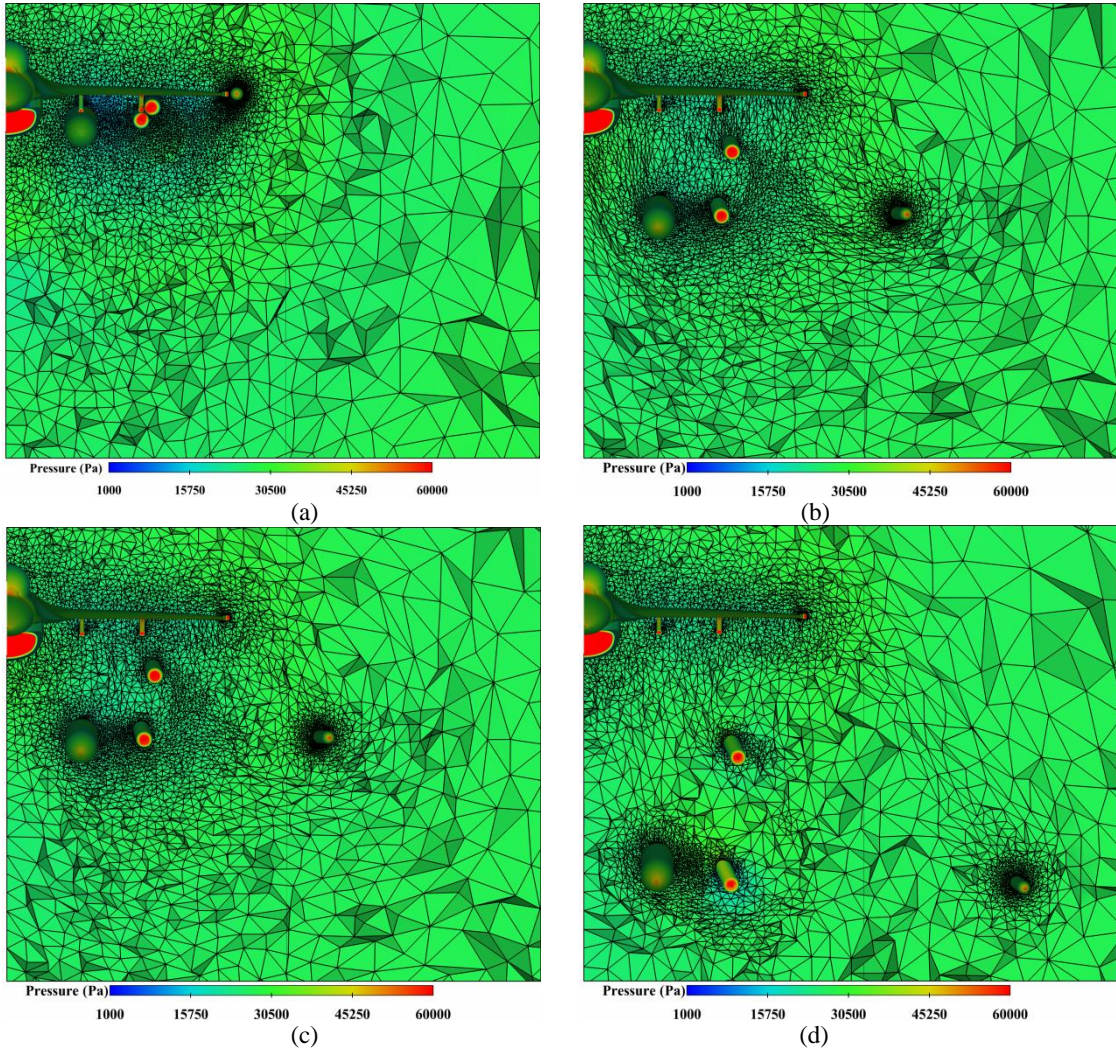
700 **Figure 13.** A comparison of the improved meshes of *patient-organs02* by the proposed  
 701 algorithm with interior constraints respected or not. In graphs (a) and (b), the blue triangles  
 702 are boundary triangles, and the red tetrahedra are elements containing dihedral angles below  
 703  $30^\circ$  or above  $150^\circ$ . Graph (c) compares the distributions of dihedral angles of both meshes.

704 Finally, the applicability of the developed mesh improver for real aerodynamics  
 705 simulations is demonstrated by a store separation simulation of a fully-loaded F16 aircraft. In  
 706 this test, four stores are separated from the aircraft to verify the robustness of our in-house  
 707 CFD system for complex flow simulations [35]. The main loop of this simulation includes  
 708 four main steps:

- 709 (1) Compute the unsteady flow by a finite volume solver.
- 710 (2) Compute aerodynamic forces and moments based on flow simulation results, with  
 711 which as inputs, the positions of moving bodies in the next time step are determined  
 712 using the six degrees-of-freedom equations of motion.
- 713 (3) Move the mesh points to adapt the movement of mesh boundaries.
- 714 (4) If mesh movement yields elements with unacceptable quality, the holes are formed by  
 715 deleting these elements. Next, a new mesh is formed by merging undeleted elements  
 716 and new elements filled in the holes. Finally, the solution is reconstructed by  
 717 interpolation.

718 The initial volume mesh is composed of about 3.69 million tetrahedral elements (see Figure  
 719 14a). Figures 14b and 14c compare the meshes before and after local remeshing at  $t_s = 0.147s$   
 720 ( $t_s$  refers to a physical time of separation). Figures 14d presents the mesh at  $t_s = 0.3s$ , instantly  
 721 after another local remeshing step is accomplished. Because the simulation involves very  
 722 complicated boundary movements, the proposed remeshing algorithm is employed very  
 723 frequently. Considering the simulation process until  $t_s = 0.5s$ , the remeshing algorithm is  
 724 employed for a total of 44 times. On average, each local remeshing step needs to generate and  
 725 improve a local mesh size composed of about 350K elements. Stellar code is obviously  
 726 inappropriate for this kind of application because of its huge time consumption. Before this  
 727 study, Grumpp code was ever employed for a local mesh improvement. It was observed that  
 728 Grumpp code occasionally failed to provide a qualified mesh for simulations. One possible  
 729 reason could be that mesh faces are largely stretched during the mesh deformation process  
 730 and some of them may even appear on the boundaries of the holes to be remeshed. Grumpp  
 731 code sometimes failed to remove those low-quality elements attaching to these stretched  
 732 faces. After replacing Grumpp code with the present mesh improver, no failing case has been  
 733 reported as far as this simulation is concerned.

734 Note that only the steps of the CFD solution and mesh deformation were executed in  
 735 parallel on 32 computer cores in this test, while other steps, including local remeshing, are  
 736 executed sequentially. Not surprisingly, the CFD solution step is most time-consuming, which  
 737 used 91.9% of the total computing time. The mesh deformation step only used 3.9% of the  
 738 total computing time because a simple spring-analogy approach was adopted [35]. Although  
 739 local remeshing calling is executed sequentially, its total time cost is very low (using only  
 740 2.9% of the total computing time).



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745 **Figure 14** Cut views of the volume meshes for the store separation problem of a full-loaded  
746 F16 aircraft at different physical time of separation ( $t_s$ ): (a)  $t_s=0s$ ; (b)  $t_s=0.147s$ , before local  
747 remeshing; (c)  $t_s=0.147s$ , after local remeshing; (d)  $t_s=0.3s$ , after local remeshing.

748

## 8. CONCLUSIONS

749 A new flip named shell transformation is proposed for mesh quality improvement. Its single  
750 calling could be considered as an enhanced version of the edge removal transformation, while  
751 its recursive scheme acts like “composite edge removal transformations”. In practice, this  
752 recursive scheme provides an elaborate pattern to combine multiple flips and perform these  
753 flips on hundreds of elements for a single goal, for instance, removing a low-quality element  
754 by removing one of its boundary edges. Accordingly, the possibility to achieve such a goal by  
755 the proposed recursive scheme is much higher than those based on performing single flips  
756 individually.

757 Furthermore, a new mesh improvement algorithm is developed by combining the proposed  
758 local reconnection scheme with smoothing and other topological transformation schemes.  
759 Numerical experiments have revealed that the proposed algorithm is capable of balancing the  
760 requirements for a high-quality mesh and a low computational time costs spent on the mesh  
761 improvement for large-scale engineering flow problems.

762

763 **Acknowledgements:** The authors appreciate the joint support for this project by the National Natural Science



764 Foundation of China (Grant No. 11172267, 11432013, 10872182) and Zhejiang Provincial Natural Science  
765 Foundation (Grant No. LR16F020002 and Y1110038). The first author acknowledges the joint support from  
766 Zhejiang University and China Scholarship Council and the host of Professor Oubay Hassan and Professor  
767 Kenneth Morgan for his recent research visit to Swansea University, UK.

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