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Backside delamination detection in composites through local defect resonance induced nonlinear source behavior

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CRediT author statement

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Backside delamination detection in composites through local 1 defect resonance induced nonlinear source behaviour 2 3 Joost Segers^{1#}, Saeid Hedayatrasa^{1,2}, Gaétan Poelman¹, Wim Van Paepegem¹ and Mathias 4 5 Kersemans¹ 6 ¹Mechanics of Materials and Structures (UGent-MMS), Department of Materials, Textiles and 7 Chemical Engineering (MaTCh), Ghent University, Technologiepark-Zwijnaarde 46, 9052 Zwijnaarde, 8 Belgium 9 [#]Corresponding author, Joost.Segers@ugent.be 10 ² SIM Program M3 DETECT-IV, Technologiepark-Zwijnaarde 48, B-9052 Zwijnaarde, Belgium 11 12 Abstract 13 Shallow damages can be detected by searching for local defect resonance behavior in the 14 fundamental vibrational response. However, this is not possible for defects which are deeper than 15 half the thickness of the coupon due to the limited bending stiffness difference between the defect 16 and the sound material. 17 In this study, it is shown that delaminations under harmonic excitation at the local defect resonance 18 frequency behave as sources of higher harmonic vibration components. These higher harmonic 19 components are detectable when examining the delaminated component from the side where the 20 delamination is close to the surface, but more importantly, also from the other side where the 21 delamination is at the backside. As such, monitoring of these harmonics allows for the detection of 22 backside delaminations which cannot be found using linear LDR techniques. First, a finite element 23 simulation with nonlinear contact conditions is performed as a proof of concept. Experimental 24 validation is performed for a delaminated CFRP coupon using low-power piezoelectric actuation and 25 scanning laser Doppler vibrometer response measurements. 26 27 Keywords 28 Composites, NDT, Nonlinearity, Local defect resonance, Higher harmonics, Laser Doppler vibrometry 29 30

1. Introduction

31 The non-destructive testing (NDT) method developed in this study builds on the concept of local 32 defect resonance (LDR) [1, 2]. As the name indicates, an LDR is a localized amplification of the 33 vibrational response (i.e. a resonance) at the defect. When the specimen is excited at an LDR 34 frequency, the defect can be pinpointed, and to a certain extent evaluated, by measuring the 35 structure's operational deflection shape (ODS). In general, a scanning laser Doppler vibrometer 36 (SLDV) is used to measure the full-field vibrational surface response. As the existence of LDR is linked 37 to the difference in stiffness between the defect and the surrounding sound material, it is clear that 38 the detection of defects, which are located deep inside the coupon, with LDR is problematic. A recent 39 study of the current authors indicated that the classical LDR can detect defects up to a depth of 50% 40 (relative to full thickness of the sample) [3]. Enhanced defect detection is possible using nonlinear 41 defect imaging, exploiting the relatively high vibrational activity of the defect under LDR frequency 42 excitation [4-8]. The high vibrational activity at the defect results in a nonlinear response to 43 vibrations caused by multiple mechanisms: contact, friction, viscoelastic damping, etc. [8-11]. Which 44 of these mechanisms is dominant depends on the properties of the defect as well as the 45 characteristics of the vibration field (e.g. dominant in-plane versus out-of-plane) [11, 12]. However, 46 until now deep defect detection in monolithic composite components has not been demonstrated47 under LDR condition.

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In this study, the nonlinear response of delamination in carbon fiber reinforce polymer components is further investigated. It is demonstrated that delaminations behave as sources of higher harmonics (HH) when the excitation frequency matches the LDR frequency. Moreover, it is numerically and experimentally proven that this nonlinear source behavior is visible from both sides of the component. This allows for the detection of shallow delaminations, and more importantly, it allows for the detection of deep, backside, delaminations.

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2. Simulation: proof of concept

57 As a proof of concept, an explicit finite element simulation (ABAQUS) is performed for a 180x165x2 58 mm³ CFRP coupon, with layup [(0/90)₂]_s, using 115736 linear hexahedral elements. Orthotropic 59 material properties are used in accordance with the material properties of the experimentally 60 investigated coupon (see Table 3 in Ref. [13]). A delamination defect of 20x20 mm² is modeled as a 2 61 µm gap between the first and second ply, and both normal (hard contact) and tangential (friction) 62 interactions are prescribed. This defect model allows for contact acoustic nonlinearity at the gap 63 closures when the defect is resonated. A harmonic force excitation of 50 kPa amplitude and duration 64 2 ms is applied at the lower left corner over an circular area (diameter 20 mm). The excitation 65 frequency is set to 28 kHz which corresponds to an LDR frequency of the modelled delamination. The 66 vibrational surface response of both sides of the sample is analyzed in the frequency domain. At the 67 fundamental frequency of 28 kHz, the LDR behavior of the defect is clearly visible from the Shallow 68 Side (see Figure 1(a)). From the Deep Side however, no indication of the defect is observed due to 69 the limited stiffness mismatch between sound material and the thick side of the delamination (Figure 70 1(b). At the second and third higher harmonic frequencies (f_{HH2} = 2 x 28 kHz and f_{HH3} = 3 x 28 kHz 71 respectively), the delamination shows a relatively high amplitude of vibration. While this high 72 amplitude of vibration is especially visible from the Shallow Side (Figure 1(c,e)), it is clear that the HH 73 components are also clearly visible from the Deep Side (Figure 1(d,f)). 74

> **Shallow Side Deep Side** 6 (b) (a) f_{fund} = 28 kHz LDR @ Defect mm/s mm/s Excitation $f_{HH2} = 2 \times 28 \text{ kHz}$ (c) (d) 33 1.11.11.25 mm/s mm/s f_{HH3} = 3 x 28 kHz (f) (e) 111111 24 mm/s mm/s

Figure 1: Operational deflection shapes at fundamental excitation frequency f_{fund} and at second and third higher harmonic frequency (f_{HH2} , f_{HH3}) for sine excitation at LDR frequency – <u>Simulation results</u>.

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Furthermore, experimental verification is performed using an eight ply 290x140x2.2 mm³ CFRP 79 coupon with layup $[(0/90)_2]_s$ (see Figure 2(a)). Similar to the FE model, a 20x20 mm² delamination is 80 81 present between the first and second ply. The delamination is made by inserting two layers of 25 µm 82 thick brass foil between the plies. Vibrations are introduced using two low power piezoelectric 83 actuators (EPZ-20MS64W Ekulit) bonded to the Deep Side surface using phenyl salicylate. In all 84 experiments, both actuators are supplied with an identical signal. A Falco systems wma-300 voltage 85 amplifier is used to increase the voltage supplied to the actuators. The surface of the coupon is 86 covered with reflective tape (3M[™]Scotchlite[™]) to increase the signal-to-noise ratio of the acquired 87 signals. The full-field surface vibrations are obtained using a 3D infrared SLDV (Polytec PSV-500-3D-88 Xtra). In this study, only the out-of-plane velocity component V_z is used. The vibrations are measured 89 from both the Shallow Side as well as the Deep Side.

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3. Experiment: validation

91 As a first step, the LDR behavior of the delamination is investigated. To this end, the actuators are 92 supplied with a 50 Vpp sweep signal from 1 to 100 kHz and the measured velocity response is 93 analyzed for both sides of the coupon. At each frequency, the defect-to-background ratio *DBR* is

94 calculated as:

$$DBR(f) = \frac{\Omega_{\text{healthy}}}{\Omega_{\text{defect}}} \frac{\sum_{i=1}^{n_{\text{defect}}} V_Z(x_i, y_{i,i}, f)}{\sum_{i=1}^{n_{\text{healthy}}} V_Z(x_i, y_{i,i}, f)}$$
(1)

Where Ω_{defect} is the defected area which contains n_{defect} measurement points and $\Omega_{healthy}$ is the 95 96 surrounding healthy area with n_{healthy} measurement points. $V_Z(x_i, y_i, f)$ is the magnitude of out-of-97 plane velocity of the scan point at location (x_i, y_i) for the ODS corresponding to frequency f. As a 98 result, a local maximum in the DBR curve is related to an LDR. The DBR curve, together with the average velocity amplitude at the location of the delamination (Vel_z), is shown in Figure 2(b-c) for the 99 100 Shallow and Deep Side of the delamination. From the Shallow Side, a clear LDR behavior is observed 101 at 27 kHz as shown by the ODS (Figure 2(a)). However, from the Deep Side of the sample, no LDR 102 behavior is present due to the limited stiffness mismatch between the sound material (8 plies) and 103 the thick part of the delamination (7 plies). Thus, the exploration of the nonlinear defect response is 104 necessary in order to detect the deep delamination. 105



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Figure 2: (a) Test specimen as seen from the side where the delamination is close to the surface (i.e. Shallow Side) with operational deflection shape at the LDR frequency of 27 kHz. Defect-to-background ratio *DBR* and defect velocity response *Vel_z* at delamination as measured from (b) Shallow Side and (c) Deep Side.

110 In the next step, the sample is harmonically excited at the above identified LDR frequency of 27 kHz

111 with a 150 V_{pp} signal. The resulting ODSs corresponding to the fundamental frequency (f_{fund} i.e. the

excitation frequency) and the higher harmonics (HH_i with $f_{HHi} = i * f_{fund}$) are shown in Figure 3. The

- 113 results are available in animated format in the online version of this article.
- 114



115 116 117 Figure 3: Operational deflection shapes at fundamental excitation frequency f_{fund} and at second and third higher harmonic 117 frequency (f_{HH2}, f_{HH3}) for sine excitation at LDR frequency – Experimental observations.

118 This experimental observation is highly similar to the numerical observation (see Figure 1). The ODS

119 at f_{fund} displays LDR behavior at the Shallow Side of the delamination, while there is no indication of

- the delamination from the Deep Side. The ODSs corresponding to the HH frequencies on the other hand show a local high amplitude at the defect and, more importantly, the defect behaves as a source, radiating these harmonics to the surrounding sound material (see also the animated figure). This source behavior is most pronounced at the Shallow Side but is also clearly visible from the Deep Side.
- 125
- 126 To further prove that the HH components are present due to the defect nonlinearity, and not as a
- 127 result of potential source nonlinearity, the specimen is excited using a sine frequency at f = 54 kHz.
- 128 The obtained fundamental ODSs (Figure 4(a,b)) are compared with the ODSs of the HH₂ component
- (sine excitation at f = 27 kHz) found earlier (Figure 4(c,d)). The large mismatch in ODSs indicates that
- 130 the HH_2 vibrations are indeed dominantly excited by the defect and are not related to source
- 131 nonlinearity.132



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134Figure 4: Operational deflection shapes at 54 kHz when excited using (a,b) 54 kHz sine, versus, (c,d) 27 kHz sine (see Fig
3(c,d)).

136 **4. Conclusion**

In this study, the nonlinear response of a vibrating delamination defect in a CFRP plate under
 harmonic excitation at the local defect resonance frequency is investigated experimentally as well as
 using numerical simulation.

140 It is shown that the defect under LDR behavior behaves as a radiating source of higher harmonic 141 components, visible from both sides of the component. This observation is of great importance for 142 the field of vibrometric non-destructive testing (NDT) as well as structural health monitoring (SHM). 143 It evidences that backside delaminations can be detected and localized by evaluating the local 144 amplitude of the HHs over the test specimen surface using SLDV measurements. Furthermore, the 145 observed source nature of the induced defect nonlinearity allows the use of HH source localization

- 146 algorithms in case of sparse array sensor networks.
- 147

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Declaration of interests

 \boxtimes The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

The authors declare the following financial interests/personal relationships which may be considered as potential competing interests:

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