

## Active Control of a Non-Linear Ship model with External and Parametric Excitation

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### ABSTRACT

The response of a ship model with non-linearly coupled pitch and roll modes under modulated external and parametric solved and studied. The active control is applied to reduce the vibration of the system . The method of multiple scale perturbation technique is applied to obtain the periodic response equation near the primary resonance in the presence of internal resonance of the system. The objective of this research is focused on the stability of this periodic solution, dynamical properties and chaotic response. The stability of the obtained numerical solution is studied using both frequency response equation and phase-plane methods. The effects of some parameters on the vibrating system are investigated and reported in this paper.

**Keywords:** Pitch-roll ship ; Vibration; Nonlinear system; Periodic solution; Stability; Chaos; Phase-Plane.



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## 1- INTRODUCTION

Nayfeh and others [1- 4] studied the non-stationary responses of a non-linearly two-degree-of-freedom system under non-stationary excitation. The basic two-degree-of freedom ship model under a sinusoidal excitation with either a slowly varying amplitude or frequency . Nayfeh, Mook and Marshall's study [5-7]. They used first order multiple time scale method to analyze this system of ship model and a saturation phenomenon of the second mode has been reported. The authors [8, 9] have studied the same saturation phenomenon theoretically and experimentally. Many investigators have studied chaos in non-linear multi degree-of-freedom systems with internal resonance. For instance, quadratically coupled oscillators with 1:2 internal resonance [10,11], a harmonically excited mechanical system with one to one internal resonance [12], a parametrically and externally excited dynamical system with 2:1 internal resonance [13,14] have been studied. Davis and Pan studied a system of a ship model using both first order multiple time scale and averaging method, and a more accurate analytical solution and bifurcation diagrams are obtained and reported [15-17]. They also studied the same system of ship model, when it has both a 2:1 internal resonance. Furthermore, the case of periodic response and chaotic response of the same system under modulated excitation has been studied. Also, the stability of the two modes of a ship motions near primary resonance in the presence of internal resonance has been studied and reported [10, 14, 17]. The stability of the numerical solution is investigated using both phase plane methods and frequency response equations. The stability of the proposed solution is determined applying Lyapunov's first method and the stability of the obtained numerical solution of the considered system is studied applying Runge-Kutta method [18,19]. Eissa and El-Bassiouny [20] construct a second-order uniform expansion of the non-linear rolling response of a ship in regular beam seas by method of multiple time scales. The analysis took into consideration linear, quadratic, cubic, quintus, and seven terms in the polynomial expansion of the relative roll angle. Eissa, El-Ganaini and Hamed [21] Saturation phenomena may occur in non-linear vibrating systems. This phenomena is very useful in suppressing the undesired vibrations and saturation is investigated in a non-linear oscillating system subject to multi-parametric excitation. Kamel [22] studied the response of a two-degree-of-freedom system with quadratic coupling under a modulated amplitude sinusoidal excitation is studied and solved.. EL-Sayed, Kamel and Eissa [23] studied an application of passive vibration control to a non-linear spring pendulum system simulating a ship's roll motion. This leads to a four-degree-of-freedom (4-DOF) system subjected to multi external and parametric excitations. Sayed and Hamed [24] studied deals with the response of a two- degree-of-freedom(2DOF) system with quadratic coupling under parametric and harmonic excitations. The method of multiple scale perturbation technique is applied to solve the nonlinear differential equations and obtain approximate solutions up to and including the second order approximations.

Sayed and Kamel [25, 26] studied the effect of different controllers on the vibrating system and the saturation control of a linear absorber to reduce vibrations due to rotor blade flapping motion. Kamel, El-Ganaini and Hamed[27] the coupling of two non-linear oscillators of the main system and absorber representing ultrasonic cutting process is investigated. This leads to a two-degree-of-freedom system subjected to multi-external excitation force. Dostal, Kreuzer and Navaratnam[28] studied Multi-degree-of-freedom ship motion and ship stability in random seas are of major interest for the development of new advanced intact stability criteria and improve the safety of new ship designs, but the results are relevant also for other engineering systems involving multiple scales.

We focus on roll-pitch and roll-heave motion in random seas. The random wave excitation is modeled by a non-white stationary process. This process is derived from a spectral description of the random seaway using traveling effective wave. The aim of this work is to control the main system behavior at simultaneous primary and internal resonance condition, where the system damage is probable. Multiple scale perturbation method is applied to obtain the solution up to the second order approximations and Some of resonance cases are investigated. The effects of natural and excitation frequencies on the response of the system are investigated and discussed.

## 2- MATHEMATICAL MODELING

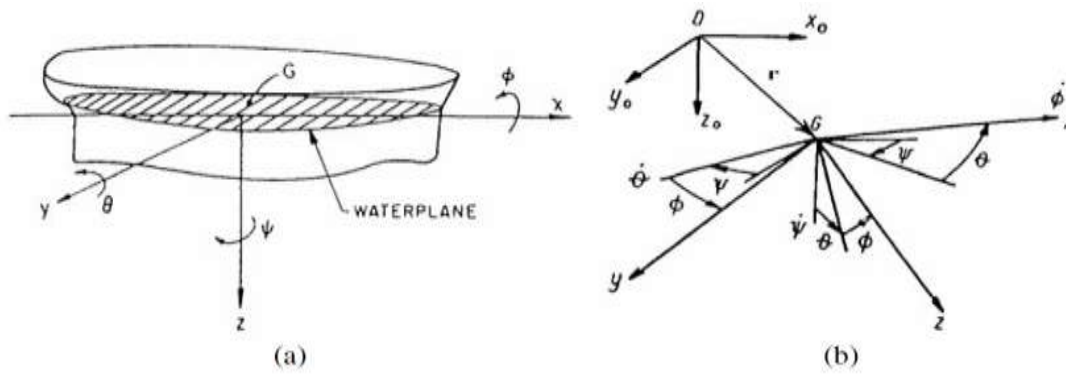


Fig.1a The body - fixed coordinates system  $Ox_o y_o z_o$ . b coordinates system  $Gxyz$  and  $Ox_o y_o z_o$

In this paper, we consider a ship model with non linear coupled pitch and roll modes subjected to a sinusoidal harmonic excitation and parametric excitations [24]. The nonlinear system can be written as:

$$\ddot{X} + 2\tilde{\mu}_1 \dot{X} + \omega_1^2 X + \tilde{\alpha}_1 \dot{X} \dot{Y} = \tilde{G}_o \cos \Omega_1 t + X \tilde{F}_1 \sin \Omega_2 t + T_1 \quad (1a)$$

$$\ddot{Y} + 2\tilde{\mu}_2 \dot{Y} + \omega_2^2 Y + \tilde{\alpha}_2 \dot{X}^2 = \tilde{F}_o \cos \Omega_1 t + Y \tilde{F}_2 \sin \Omega_2 t + T_2 \quad (1b)$$

where  $X$  and  $Y$  are the roll and pitch mode amplitudes,  $\tilde{\mu}_1$  and  $\tilde{\mu}_2$  the modal damping coefficients,  $\omega_1$  and  $\omega_2$  the natural angular frequencies of the roll and pitch modes, and  $\Omega_1, \Omega_2$  the excitations or wave frequencies.  $\tilde{G}_o, \tilde{F}_1$  and  $\tilde{F}_o, \tilde{F}_2$  are the excitation force amplitudes of the roll and pitch modes,  $\tilde{\alpha}_1$  and  $\tilde{\alpha}_2$  non linear coefficients.  $T_1 = -\varepsilon G_1 \dot{X}^3$ , and  $T_2 = -\varepsilon G_2 \dot{Y}^3$  are the absorbers of the system. All the coefficients in the above equations are complicated functions of the various ship moments of inertia, fluid parameters, boat speed, etc. The linear viscous damping forces, exciting forces and controller are assumed to be

$$\tilde{\mu}_n = \varepsilon \mu_n \quad \tilde{F}_n = \varepsilon F_n \quad \tilde{\alpha}_n = \varepsilon \alpha_n \quad \tilde{G}_o = \varepsilon G_o \quad \tilde{F}_o = \varepsilon F_o \quad (n = 1, 2)$$

where  $\varepsilon$  is a small perturbation parameter and  $0 < \varepsilon < 1$ .  $G_1$  and  $G_2$  are gains of the absorbers.

## 2.1-Perturbation analysis

The method of multiple time scale is applied to determine a first order uniform expansion for the solution of equations (1a) and (1b) as in the form:

$$X(t, \varepsilon) = x_o(T_o, T_1) + \varepsilon x_1(T_o, T_1) + \varepsilon^2 x_2(T_o, T_1) + \varepsilon^3 x_3(T_o, T_1) + O(\varepsilon^4) \quad (2a)$$

$$Y(t, \varepsilon) = y_o(T_o, T_1) + \varepsilon y_1(T_o, T_1) + \varepsilon^2 y_2(T_o, T_1) + \varepsilon^3 y_3(T_o, T_1) + O(\varepsilon^4) \quad (2b)$$

where  $\varepsilon$  is a small perturbation parameter,  $T_o = t, T_1 = \varepsilon t$  are fast and slow time scales respectively, and the time derivatives became

$$\frac{d}{dt} = D_o + \varepsilon D_1 + \varepsilon^2 D_2, \quad \frac{d^2}{dt^2} = D_o^2 + 2\varepsilon D_o D_1 + \varepsilon^2 (D_1^2 + 2D_o D_2) \quad (3)$$

Substituting equations (2a),(2b) and (3) in to equations (1a) and (1b) and equating the coefficients of the same power of  $\varepsilon$  in both sides, we obtain

$$(D_o^2 + \omega_1^2)x_o = 0 \quad (4a)$$

$$(D_o^2 + \omega_2^2)y_o = 0 \quad (4b)$$



$$(D_o^2 + \omega_1^2)x_1 = -2D_o D_1 x_o - 2\mu_1 D_o x_o - \alpha_1 (D_o x_o)(D_o y_o) + x_o F_1 \sin \Omega_2 t - G_1 (D_o x_o)^3 + G_0 \cos \Omega_1 t \quad (5a)$$

$$(D_o^2 + \omega_2^2)y_1 = -2D_o D_1 y_o - 2\mu_2 D_o y_o - \alpha_2 (D_o x_o)^2 + F_o \cos \Omega_1 t + y_o F_2 \sin \Omega_2 t - G_2 (D_o y_o)^3 \quad (5b)$$

$$(D_o^2 + \omega_1^2)x_2 = -D_1^2 x_o - 2D_o D_1 x_1 - 2\mu_1 (D_o x_1 + D_1 x_o) - \alpha_1 ((D_o x_o)(D_o y_1) - 3G_1 (D_o x_o)^2 (D_o x_1 + D_1 x_o) + (D_o x_o)(D_1 y_o) + (D_o y_o)(D_o x_1) + (D_o y_o)(D_1 x_o)) + x_1 F_1 \sin \Omega_2 t \quad (6a)$$

$$(D_o^2 + \omega_2^2)y_2 = -D_1^2 y_o - 2D_o D_1 y_1 - 2\mu_2 (D_o y_1 + D_1 y_o) - 2\alpha_2 (D_o x_o)(D_o x_1 + D_1 x_o) + y_1 F_2 \sin \Omega_2 t - 3G_2 (D_o y_o)^2 (D_o y_1 + D_1 y_o) \quad (6b)$$

$$(D_o^2 + \omega_1^2)x_3 = -D_1^2 x_1 - 2D_o D_1 x_2 - 2\mu_1 (D_o x_2 + D_1 x_1) - \alpha_1 ((D_o x_1 + D_1 x_o) \times (D_o y_1 + D_1 y_o) + (D_o x_o)(D_o y_2 + D_1 y_1) + (D_o y_o)(D_o x_2 + D_1 x_1)) + x_2 F_1 \sin \Omega_2 t - 3G_1 ((D_o x_o)(D_o x_1 + D_1 x_o)^2 + (D_o x_o)^2 (D_o x_2 + D_1 x_1)) \quad (7a)$$

$$(D_o^2 + \omega_2^2)y_3 = -D_1^2 y_1 - 2D_o D_1 y_2 - 2\mu_2 (D_o y_2 + D_1 y_1) - \alpha_2 ((D_o x_1 + D_1 x_o)^2 + 2(D_o x_o)(D_o x_2 + D_1 x_1)) + y_2 F_2 \sin \Omega_2 t - 3G_2 ((D_o y_o)(D_o y_1 + D_1 y_o)^2 + (D_o y_o)^2 (D_o y_2 + D_1 y_1)) \quad (7b)$$

The general solution of equations (4a) and (4b) are given by

$$x_o(T_o, T_1) = A_o(T_1)e^{i\omega_1 T_o} + \bar{A}_o(T_1)e^{-i\omega_1 T_o} \quad (8a)$$

$$y_o(T_o, T_1) = B_o(T_1)e^{i\omega_2 T_o} + \bar{B}_o(T_1)e^{-i\omega_2 T_o} \quad (8b)$$

where  $A_o, B_o$  are complex function in  $T_1$  and cc represents the complex conjugate of the preceding terms. Substituting equations (8a) and (8b) in to equations (5a) and (5b), and eliminating the secular terms, then the general solution obtained as

$$x_1(T_o, T_1) = C_1 e^{i\omega_1 T_o} + C_2 e^{3i\omega_1 T_o} + C_3 e^{i\Omega_1 T_o} + C_4 e^{i(\omega_1 + \omega_2) T_o} + C_5 e^{i(\omega_1 - \omega_2) T_o} + C_6 e^{i(\Omega_2 + \omega_1) T_o} + C_7 e^{i(\Omega_2 - \omega_1) T_o} + cc \quad (9a)$$

$$y_1(T_o, T_1) = C_1^* e^{i\omega_2 T_o} + C_2^* e^{3i\omega_2 T_o} + C_3^* e^{2i\omega_1 T_o} + C_4^* e^{i\Omega_1 T_o} + C_5^* e^{i(\Omega_2 + \omega_2) T_o} + C_6^* e^{i(\Omega_2 - \omega_2) T_o} + C_7^* e^{i(\Omega_2 - \omega_2) T_o} + C_7^* + cc \quad (9b)$$

where  $C_i$  and  $C_i^*$  ( $i = 1, 2, \dots, 7$ ) are complex function in  $T_1$ , cc are complex conjugate. Similarly, Substituting from equations (8a), (8b), (9a) and (9b) into equations (6a) and (6b) we get

$$x_2(T_o, T_1) = N_1 e^{i\omega_1 T_o} + N_2 e^{3i\omega_1 T_o} + N_3 e^{5i\omega_1 T_o} + N_4 e^{i\Omega_1 T_o} + N_5 e^{i(\omega_1 + \omega_2) T_o} + N_6 e^{i(\omega_1 - \omega_2) T_o} + N_7 e^{i(\omega_1 + 2\omega_2) T_o} + N_8 e^{i(\omega_1 - 2\omega_2) T_o} + N_9 e^{i(\omega_1 + 3\omega_2) T_o} + N_{10} e^{i(\omega_1 - 3\omega_2) T_o} + N_{11} e^{i(3\omega_1 + \omega_2) T_o} + N_{12} e^{i(3\omega_1 - \omega_2) T_o} + N_{13} e^{i(\Omega_1 + \Omega_2) T_o} + N_{14} e^{i(\Omega_1 - \Omega_2) T_o} + N_{15} e^{i(\Omega_1 + \omega_1) T_o} + N_{16} e^{i(\Omega_1 - \omega_1) T_o} + N_{17} e^{i(\Omega_1 + 2\omega_1) T_o} + N_{18} e^{i(\Omega_1 - 2\omega_1) T_o} + N_{19} e^{i(\Omega_1 + \omega_2) T_o} + N_{20} e^{i(\Omega_1 - \omega_2) T_o} + N_{21} e^{i(\Omega_2 + \omega_1) T_o} + N_{22} e^{i(\Omega_2 - \omega_1) T_o} + N_{23} e^{i(\Omega_2 + 3\omega_1) T_o} + N_{24} e^{i(\Omega_2 - 3\omega_1) T_o} + N_{25} e^{i(2\Omega_2 + \omega_1) T_o} + N_{26} e^{i(2\Omega_2 - \omega_1) T_o}$$



$$+ N_{27}e^{i(\Omega_2+(\omega_1+\omega_2))T_o} + N_{28}e^{i(\Omega_2-(\omega_1+\omega_2))T_o} + N_{29}e^{i(\Omega_2+(\omega_1-\omega_2))T_o} + N_{30}e^{i(\Omega_2-(\omega_1-\omega_2))T_o} + cc \quad (10a)$$

$$y_2(T_o, T_1) = M_1e^{i\omega_2T_o} + M_2e^{3i\omega_2T_o} + M_3e^{5i\omega_2T_o} + M_4e^{2i\omega_1T_o} + M_5e^{4i\omega_1T_o} + M_6e^{i(2\omega_1+\omega_2)T_o}$$

$$+ M_7e^{i(2\omega_1-\omega_2)T_o} + M_8e^{2i(\omega_1+\omega_2)T_o} + M_9e^{2i(\omega_1-\omega_2)T_o} + M_{10}e^{i\Omega_1T_o} + M_{11}e^{i\Omega_2T_o}$$

$$+ M_{12}e^{i(\Omega_1+\Omega_2)T_o} + M_{13}e^{i(\Omega_1-\Omega_2)T_o} + M_{14}e^{i(\Omega_1+\omega_1)T_o} + M_{15}e^{i(\Omega_1-\omega_1)T_o} + M_{16}e^{i(\Omega_1+2\omega_2)T_o}$$

$$+ M_{17}e^{i(\Omega_1-2\omega_2)T_o} + M_{18}e^{i(\Omega_2+2\omega_1)T_o} + M_{19}e^{i(\Omega_2-2\omega_1)T_o} + M_{20}e^{i(\Omega_2+\omega_2)T_o} + M_{21}e^{i(\Omega_2-\omega_2)T_o}$$

$$+ M_{22}e^{i(\Omega_2+3\omega_2)T_o} + M_{23}e^{i(\Omega_2-3\omega_2)T_o} + M_{24}e^{i(2\Omega_2+\omega_2)T_o} + M_{25}e^{i(2\Omega_2-\omega_2)T_o} + M_{26} + cc \quad (10b)$$

where  $N_m$  ( $m = 1, 2, \dots, 30$ ) and  $M_i$  ( $i = 1, 2, \dots, 26$ ) are complex functions in  $T_1$ , cc are complex conjugates. Similarly, substituting from equations (8a), (8b), (9a), (9b), (10a) and (10b) in to equations (7a) and (7b) we get,

$$x_3(T_o, T_1) = w_1e^{i\omega_1T_o} + w_2e^{3i\omega_1T_o} + w_3e^{5i\omega_1T_o} + w_4e^{7i\omega_1T_o} + w_5e^{i\Omega_1T_o} + w_6e^{2i\Omega_1T_o}$$

$$+ w_7e^{i(\omega_1+\omega_2)T_o} + w_8e^{i(\omega_1-\omega_2)T_o} + w_9e^{i(\omega_1+2\omega_2)T_o} + w_{10}e^{i(\omega_1-2\omega_2)T_o}$$

$$+ w_{11}e^{i(\omega_1+3\omega_2)T_o} + w_{12}e^{i(\omega_1-3\omega_2)T_o} + w_{13}e^{i(\omega_1+4\omega_2)T_o} + w_{14}e^{i(\omega_1-4\omega_2)T_o}$$

$$+ w_{15}e^{i(\omega_1+5\omega_2)T_o} + w_{16}e^{i(\omega_1-5\omega_2)T_o} + w_{17}e^{i(\omega_1+6\omega_2)T_o} + w_{18}e^{i(\omega_1-6\omega_2)T_o}$$

$$+ w_{19}e^{i(3\omega_1+\omega_2)T_o} + w_{20}e^{i(3\omega_1-\omega_2)T_o} + w_{21}e^{i(3\omega_1+2\omega_2)T_o} + w_{22}e^{i(3\omega_1-2\omega_2)T_o}$$

$$+ w_{23}e^{3i(\omega_1+\omega_2)T_o} + w_{24}e^{3i(\omega_1-\omega_2)T_o} + w_{25}e^{i(5\omega_1+\omega_2)T_o} + w_{26}e^{i(5\omega_1-\omega_2)T_o}$$

$$+ w_{27}e^{i(\Omega_1+\Omega_2)T_o} + w_{28}e^{i(\Omega_1-\Omega_2)T_o} + w_{29}e^{i(\Omega_1+2\Omega_2)T_o} + w_{30}e^{i(\Omega_1-2\Omega_2)T_o}$$

$$+ w_{31}e^{i(\Omega_1+\omega_1)T_o} + w_{32}e^{i(\Omega_1-\omega_1)T_o} + w_{33}e^{i(\Omega_1+2\omega_1)T_o} + w_{34}e^{i(\Omega_1-2\omega_1)T_o}$$

$$+ w_{35}e^{i(\Omega_1+3\omega_1)T_o} + w_{36}e^{i(\Omega_1-3\omega_1)T_o} + w_{37}e^{i(\Omega_1+4\omega_1)T_o} + w_{38}e^{i(\Omega_1-4\omega_1)T_o}$$

$$+ w_{39}e^{i(\Omega_1+\omega_2)T_o} + w_{40}e^{i(\Omega_1-\omega_2)T_o} + w_{41}e^{i(\Omega_1+2\omega_2)T_o} + w_{42}e^{i(\Omega_1-2\omega_2)T_o}$$

$$+ w_{43}e^{i(\Omega_1+3\omega_2)T_o} + w_{44}e^{i(\Omega_1-3\omega_2)T_o} + w_{45}e^{i(2\Omega_1+\omega_1)T_o} + w_{46}e^{i(2\Omega_1-\omega_1)T_o}$$

$$+ w_{47}e^{i(\Omega_2+\omega_1)T_o} + w_{48}e^{i(\Omega_2-\omega_1)T_o} + w_{49}e^{i(\Omega_2+3\omega_1)T_o} + w_{50}e^{i(\Omega_2-\omega_1)T_o}$$

$$+ w_{51}e^{i(\Omega_2+5\omega_1)T_o} + w_{52}e^{i(\Omega_2-5\omega_1)T_o} + w_{53}e^{i(2\Omega_2+\omega_1)T_o} + w_{54}e^{i(2\Omega_2-\omega_1)T_o}$$

$$+ w_{55}e^{i(2\Omega_2+3\omega_1)T_o} + w_{56}e^{i(2\Omega_2-3\omega_1)T_o} + w_{57}e^{i(3\Omega_2+\omega_1)T_o} + w_{58}e^{i(3\Omega_2-\omega_1)T_o}$$

$$+ w_{59}e^{i(\Omega_1+(\omega_1+\omega_2))T_o} + w_{60}e^{i(\Omega_1-(\omega_1+\omega_2))T_o} + w_{61}e^{i(\Omega_1+(\omega_1-\omega_2))T_o} + w_{62}e^{i(\Omega_1-(\omega_1-\omega_2))T_o}$$

$$+ w_{63}e^{i(\Omega_1+(\omega_1+2\omega_2))T_o} + w_{64}e^{i(\Omega_1-(\omega_1+2\omega_2))T_o} + w_{65}e^{i(\Omega_1+(\omega_1-2\omega_2))T_o} + w_{66}e^{i(\Omega_1-(\omega_1-2\omega_2))T_o}$$

$$+ w_{67}e^{i(\Omega_1+(2\omega_1+\omega_2))T_o} + w_{68}e^{i(\Omega_1-(2\omega_1+\omega_2))T_o} + w_{69}e^{i(\Omega_1+(2\omega_1-\omega_2))T_o} + w_{70}e^{i(\Omega_1-(2\omega_1-\omega_2))T_o}$$



$$\begin{aligned}
& +W_{71}e^{i(\Omega_2+(\omega_1+\omega_2))T_o} + W_{72}e^{i(\Omega_2-(\omega_1+\omega_2))T_o} + W_{73}e^{i(\Omega_2+(\omega_1-\omega_2))T_o} + W_{74}e^{i(\Omega_2-(\omega_1-\omega_2))T_o} \\
& +W_{75}e^{i(\Omega_2+(\omega_1+2\omega_2))T_o} + W_{76}e^{i(\Omega_2-(\omega_1+2\omega_2))T_o} + W_{77}e^{i(\Omega_2+(\omega_1-2\omega_2))T_o} + W_{78}e^{i(\Omega_2-(\omega_1-2\omega_2))T_o} \\
& +W_{79}e^{i(\Omega_2+(\omega_1+3\omega_2))T_o} + W_{80}e^{i(\Omega_2-(\omega_1+3\omega_2))T_o} + W_{81}e^{i(\Omega_2+(\omega_1-3\omega_2))T_o} + W_{82}e^{i(\Omega_2-(\omega_1-3\omega_2))T_o} \\
& +W_{83}e^{i(\Omega_2+(3\omega_1+\omega_2))T_o} + W_{84}e^{i(\Omega_2-(3\omega_1+\omega_2))T_o} + W_{85}e^{i(\Omega_2+(3\omega_1-\omega_2))T_o} + W_{86}e^{i(\Omega_2-(3\omega_1-\omega_2))T_o} \\
& +W_{87}e^{i(2\Omega_2+(\omega_1+\omega_2))T_o} + W_{88}e^{i(2\Omega_2-(\omega_1+\omega_2))T_o} + W_{89}e^{i(2\Omega_2+(\omega_1-\omega_2))T_o} + W_{90}e^{i(2\Omega_2-(\omega_1-\omega_2))T_o} \\
& +W_{91}e^{i((\Omega_1+\Omega_2)+\omega_1)T_o} + W_{92}e^{i((\Omega_1+\Omega_2)-\omega_1)T_o} + W_{93}e^{i((\Omega_1-\Omega_2)+\omega_1)T_o} + W_{94}e^{i((\Omega_1-\Omega_2)-\omega_1)T_o} \\
& +W_{95}e^{i((\Omega_1+\Omega_2)+2\omega_1)T_o} + W_{96}e^{i((\Omega_1+\Omega_2)-2\omega_1)T_o} + W_{97}e^{i((\Omega_1-\Omega_2)+2\omega_1)T_o} + W_{98}e^{i((\Omega_1-\Omega_2)-2\omega_1)T_o} \\
& +W_{99}e^{i((\Omega_1+\Omega_2)+\omega_2)T_o} + W_{100}e^{i((\Omega_1+\Omega_2)-\omega_2)T_o} + W_{101}e^{i((\Omega_1-\Omega_2)+\omega_2)T_o} + W_{102}e^{i((\Omega_1-\Omega_2)-\omega_2)T_o} + cc \tag{11a}
\end{aligned}$$

$$\begin{aligned}
y_3(T_o, T_1) = & Z_1e^{i\omega_2T_o} + Z_2e^{2i\omega_2T_o} + Z_3e^{3i\omega_2T_o} + Z_4e^{5i\omega_2T_o} + Z_5e^{6i\omega_2T_o} + Z_6e^{7i\omega_2T_o} \\
& +Z_7e^{2i\omega_1T_o} + Z_8e^{4i\omega_1T_o} + Z_9e^{6i\omega_1T_o} + Z_{10}e^{i\Omega_1T_o} + Z_{11}e^{2i\Omega_1T_o} + Z_{12}e^{i\Omega_2T_o} \\
& +Z_{13}e^{2i\Omega_2T_o} + Z_{14}e^{i(\omega_1+2\omega_2)T_o} + Z_{15}e^{i(\omega_1-2\omega_2)T_o} + Z_{16}e^{i(\omega_1+3\omega_2)T_o} \\
& +Z_{17}e^{i(\omega_1-3\omega_2)T_o} + Z_{18}e^{i(\omega_1+4\omega_2)T_o} + Z_{19}e^{i(\omega_1-4\omega_2)T_o} + Z_{20}e^{i(2\omega_1+\omega_2)T_o} \\
& +Z_{21}e^{i(2\omega_1-\omega_2)T_o} + Z_{22}e^{2i(\omega_1+\omega_2)T_o} + Z_{23}e^{2i(\omega_1-\omega_2)T_o} + Z_{24}e^{i(2\omega_1+3\omega_2)T_o} \\
& +Z_{25}e^{i(2\omega_1-3\omega_2)T_o} + Z_{26}e^{i(2\omega_1+4\omega_2)T_o} + Z_{27}e^{i(2\omega_1-4\omega_2)T_o} + Z_{28}e^{i(4\omega_1+\omega_2)T_o} \\
& +Z_{29}e^{i(4\omega_1-\omega_2)T_o} + Z_{30}e^{2i(2\omega_1+\omega_2)T_o} + Z_{31}e^{2i(2\omega_1-\omega_2)T_o} + Z_{32}e^{i(\Omega_1+\omega_1)T_o} \\
& +Z_{33}e^{i(\Omega_1-\omega_1)T_o} + Z_{34}e^{i(\Omega_1+2\omega_1)T_o} + Z_{35}e^{i(\Omega_1-2\omega_1)T_o} + Z_{36}e^{i(\Omega_1+3\omega_1)T_o} \\
& +Z_{37}e^{i(\Omega_1-3\omega_1)T_o} + Z_{38}e^{i(\Omega_1+2\omega_2)T_o} + Z_{39}e^{i(\Omega_1-2\omega_2)T_o} + Z_{40}e^{i(\Omega_1+3\omega_2)T_o} \\
& +Z_{41}e^{i(\Omega_1-3\omega_2)T_o} + Z_{42}e^{i(\Omega_1+4\omega_2)T_o} + Z_{43}e^{i(\Omega_1-4\omega_2)T_o} + Z_{44}e^{i(2\Omega_1+\omega_2)T_o} \\
& +Z_{45}e^{i(2\Omega_1-\omega_2)T_o} + Z_{46}e^{i(\Omega_2+2\omega_1)T_o} + Z_{47}e^{i(\Omega_2-2\omega_1)T_o} + Z_{48}e^{i(\Omega_2+4\omega_1)T_o} \\
& +Z_{49}e^{i(\Omega_2-4\omega_1)T_o} + Z_{50}e^{i(\Omega_2+\omega_2)T_o} + Z_{51}e^{i(\Omega_2-\omega_2)T_o} + Z_{52}e^{i(\Omega_2+2\omega_2)T_o} \\
& +Z_{53}e^{i(\Omega_2-2\omega_2)T_o} + Z_{54}e^{i(\Omega_2+3\omega_2)T_o} + Z_{55}e^{i(\Omega_2-3\omega_2)T_o} + Z_{56}e^{i(\Omega_2+5\omega_2)T_o} \\
& +Z_{57}e^{i(\Omega_2-5\omega_2)T_o} + Z_{58}e^{2i(\Omega_2+\omega_1)T_o} + Z_{59}e^{2i(\Omega_2-\omega_1)T_o} + Z_{60}e^{i(2\Omega_2+\omega_2)T_o} \\
& +Z_{61}e^{i(2\Omega_2-\omega_2)T_o} + Z_{62}e^{i(2\Omega_2+3\omega_2)T_o} + Z_{63}e^{i(2\Omega_2-3\omega_2)T_o} + Z_{64}e^{i(3\Omega_2+\omega_2)T_o} \\
& +Z_{65}e^{i(3\Omega_2-\omega_2)T_o} + Z_{66}e^{i(\Omega_1+(\omega_1+\omega_2))T_o} + Z_{67}e^{i(\Omega_1-(\omega_1+\omega_2))T_o} + Z_{68}e^{i(\Omega_1+(\omega_1-\omega_2))T_o} \\
& +Z_{69}e^{i(\Omega_1-(\omega_1-\omega_2))T_o} + Z_{70}e^{i(\Omega_1+(\omega_1+2\omega_2))T_o} + Z_{71}e^{i(\Omega_1-(\omega_1+2\omega_2))T_o} + Z_{72}e^{i(\Omega_1+(\omega_1-2\omega_2))T_o}
\end{aligned}$$



$$\begin{aligned}
 &+ Z_{73}e^{i(\Omega_1-(\omega_1-2\omega_2))T_o} + Z_{74}e^{i(\Omega_1+(2\omega_1+\omega_2))T_o} + Z_{75}e^{i(\Omega_1-(2\omega_1+\omega_2))T_o} + Z_{76}e^{i(\Omega_1+(2\omega_1-\omega_2))T_o} \\
 &+ Z_{77}e^{i(\Omega_1-(2\omega_1-\omega_2))T_o} + Z_{78}e^{i(\Omega_2+(\omega_1+3\omega_2))T_o} + Z_{79}e^{i(\Omega_2-(\omega_1+3\omega_2))T_o} + Z_{80}e^{i(\Omega_2+(\omega_1-3\omega_2))T_o} \\
 &+ Z_{82}e^{i(\Omega_2+(2\omega_1+\omega_2))T_o} + Z_{83}e^{i(\Omega_2-(2\omega_1+\omega_2))T_o} + Z_{84}e^{i(\Omega_2+(2\omega_1-\omega_2))T_o} + Z_{85}e^{i(\Omega_2-(2\omega_1-\omega_2))T_o} \\
 &+ Z_{86}e^{i(\Omega_2+(2\omega_1+\omega_2))T_o} + Z_{87}e^{i(\Omega_2-2(\omega_1+\omega_2))T_o} + Z_{88}e^{i(\Omega_2+2(\omega_1-\omega_2))T_o} + Z_{89}e^{i(\Omega_2-2(\omega_1-\omega_2))T_o} \\
 &+ Z_{90}e^{i(\Omega_1+\Omega_2)T_o} + Z_{91}e^{i(\Omega_2-\Omega_2)T_o} + Z_{92}e^{i(\Omega_1+2\Omega_2)T_o} + Z_{93}e^{i(\Omega_1-2\Omega_2)T_o} \\
 &+ Z_{94}e^{i((\Omega_1+\Omega_2)+\omega_1)T_o} + Z_{95}e^{i((\Omega_1+\Omega_2)-\omega_1)T_o} + Z_{96}e^{i((\Omega_1-\Omega_2)+\omega_1)T_o} + Z_{97}e^{i((\Omega_1-\Omega_2)-\omega_1)T_o} \\
 &+ Z_{98}e^{i((\Omega_1+\Omega_2)+2\omega_2)T_o} + Z_{99}e^{i((\Omega_1+\Omega_2)-2\omega_2)T_o} + Z_{100}e^{i((\Omega_1-\Omega_2)+2\omega_2)T_o} + Z_{101}e^{i((\Omega_1-\Omega_2)-2\omega_2)T_o} \\
 &+ Z_{102} + cc
 \end{aligned} \tag{11b}$$

where  $w_m (m = 1, 2, \dots, 103)$  and  $Z_i (i = 1, 2, \dots, 102)$  are complex functions in  $T_1$ ,  $cc$  are complex conjugates.

From above-proposed solution, the reported resonance cases are :

- (a) Trivial resonance:  $\Omega_1 \cong \Omega_2 \cong \omega_1 \cong \omega_2 = 0$
- (b) Primary resonance:  $\Omega_1 \cong \omega_1, \Omega_2 \cong \omega_2$
- (c) Internal resonance: (1)  $\omega_1 \cong n\omega_2$  (2)  $\omega_2 \cong n\omega_1, n=1, 2, 3, 4, 5$  (3)  $n\omega_1 \cong m\omega_2, n=2, 3, m=3, 5$
- (d) Sub-harmonic resonance: (1)  $\Omega_1 \cong n\omega_1$  (2)  $\Omega_1 \cong n\omega_2, n=2, 3, 4, 5$  (3)  $\Omega_2 \cong m\omega_1$   
 (4)  $\Omega_2 \cong m\omega_2, m=2, 4, 6$
- (e) Combined resonance: (1)  $\Omega_1 \cong \omega_1 \pm \omega_2$  (2)  $\Omega_1 \cong 2(\omega_1 \pm \omega_2)$  (3)  $\Omega_1 \cong (2\omega_1 \pm 3\omega_2)$   
 (4)  $\Omega_1 \cong (\omega_1 \pm 2\omega_2)$  (5)  $\Omega_1 \cong (2\omega_1 \pm \omega_2)$  (6)  $2\Omega_2 \cong (2\omega_1 \pm \omega_2)$  (7)  $\Omega_2 \cong 2(\omega_1 \pm \omega_2)$   
 (8)  $\Omega_2 \cong (4\omega_1 \pm \omega_2)$  (9)  $\Omega_2 \cong (2\omega_1 \pm \omega_2)$
- (f) Simultaneous resonance:

Any combination of the above resonance cases is considered as a simultaneous resonance one.

### 2.2-Stability Analysis

Stability analysis is limited to the first order approximation. Then all the solution coefficients are functions in  $T_1$  only.

Applying an absorber to the main system at internal resonance  $\omega_2 \cong 2\omega_1$ . Introducing the detuning parameters  $\sigma_1$  and  $\sigma_2$  in the primary and internal resonance to convert the small-divisor terms into the secular terms according to

$$\Omega_1 \cong \omega_1 + \varepsilon\sigma_1 \text{ and } \omega_2 \cong 2\omega_1 + \varepsilon\sigma_2 \tag{12}$$

Eliminating the secular terms of both  $x_1$  and  $y_1$  of equations (5a) and (5b), leads to the solvability conditions, and noting that  $A_o$  and  $B_o$  are functions in  $T_1$  only, we get

$$[-2i\omega_1(D_1A_o + \mu_1A_o) - 3iG_1\omega_1^3A_o^2\bar{A}_o]e^{i\omega_1T_o} + \frac{G_o}{2}e^{i\Omega_1T_o} - \alpha_1\omega_1\omega_2\bar{A}_oB_o e^{i(\omega_1-\omega_2)T_o} = 0 \tag{13a}$$



$$[-2i\omega_2(D_1B_o + \mu_2B_o) - 3iG_2\omega_2^3B_o^2\bar{B}_o]e^{i\omega_2T_o} + \alpha_2\omega_1^2A_o^2e^{2i\omega_1T_o} = 0 \quad (13b)$$

Putting the polar form

$$A_o = \frac{1}{2}a_1(T_1)e^{i\gamma_1(T_1)}, \quad B_o = \frac{1}{2}a_2(T_1)e^{i\gamma_2(T_1)} \quad (14)$$

where  $a_1, a_2, \gamma_1$  and  $\gamma_2$  are real. Substituting Equations (14) into equations (13a), (13b), and separating real and imaginary parts we get the following

$$a_1' = -\mu_1a_1 - \frac{3}{8}G_1\omega_1^2a_1^3 + \frac{G_o}{2\omega_1}\sin\theta_1 - \frac{\alpha_1}{4}\omega_2a_1a_2\sin\theta_2 \quad (15a)$$

$$a_1\gamma_1' = \frac{\alpha_1}{4}\omega_2a_1a_2\cos\theta_2 - \frac{G_o}{2\omega_1}\cos\theta_1 \quad (15b)$$

$$a_2' = -\mu_2a_2 - \frac{3}{8}G_2\omega_2^2a_2^3 - \frac{\alpha_2\omega_1^2a_1^2}{4\omega_2}\sin\theta_2 \quad (15c)$$

$$a_2\gamma_2' = -\frac{\alpha_2\omega_1^2a_1^2}{4\omega_2}\cos\theta_2 \quad (15d)$$

where  $\theta_1 = (\sigma_2T_1 + \gamma_2 - 2\gamma_1)$  and  $\theta_2 = \sigma_1T_1 - \gamma_1$ . For steady state solutions,  $a_1' = a_2' = \theta_1' = \theta_2' = 0$ , and equations (15a), (15b), (15c) and (15d) becomes

$$0 = -\mu_1a_1 - \frac{3}{8}G_1\omega_1^2a_1^3 + \frac{G_o}{2\omega_1}\sin\theta_1 - \frac{\alpha_1}{4}\omega_2a_1a_2\sin\theta_2 \quad (16a)$$

$$a_1\sigma_1 = \frac{\alpha_1}{4}\omega_2a_1a_2\cos\theta_2 - \frac{G_o}{2\omega_1}\cos\theta_1 \quad (16b)$$

$$0 = -\mu_2a_2 - \frac{3}{8}G_2\omega_2^2a_2^3 - \frac{\alpha_2\omega_1^2a_1^2}{4\omega_2}\sin\theta_2 \quad (16c)$$

$$(2\sigma_1 - \sigma_2)a_2 = -\frac{\alpha_2\omega_1^2a_1^2}{4\omega_2}\cos\theta_2 \quad (16d)$$

Squaring equations (16a), (16b) and adding the result, we get the corresponding frequency response equations (FRE) are

$$a_1^2\sigma_1^2 + (\mu_1a_1 + \frac{3}{8}\omega_1^2G_1a_1^3)^2 - \frac{G_o^2}{4\omega_1^2} - \frac{1}{16}\alpha_1^2\omega_2^2a_1^2a_2^2 + \frac{G_o\alpha_1\omega_2}{\omega_1}a_1a_2 = 0 \quad (17)$$

Similarly, from equations (16c) and (16d), we get

$$a_2^2(2\sigma_1 - \sigma_2)^2 + (\mu_2a_2 + \frac{3}{8}G_2\omega_2^2a_2^3)^2 - (\frac{\alpha_2\omega_1^2a_1^2}{4\omega_2})^2 = 0 \quad (18)$$

From equations (17) and (18) we have the following cases:

(i) Case 1  $a_1 \neq 0, a_2 = 0$





$$a_1^2 \sigma_1^2 + (\mu_1 a_1 + \frac{3}{8} \omega_1^2 G_1 a_1^3)^2 - \frac{G_o^2}{4\omega_1^2} = 0 \quad (19)$$

(ii) Case 2  $a_1 = 0, a_2 \neq 0$

$$a_2^2 (2\sigma_1 - \sigma_2)^2 + (\mu_2 a_2 + \frac{3}{8} G_2 \omega_2^2 a_2^3)^2 = 0 \quad (20)$$

(iii) Case 3  $a_1 \neq 0, a_2 \neq 0$  represented by equations (17) and (18). The steady state solution of the obtained fixed points will be determined as follows:

Let  $A_o$  and  $B_o$  Expressed in Cartesian form as following:

$$A_o(T_1) = \frac{1}{2}(p_1 - iq_1)e^{iv_1 T_1}, B_o(T_1) = \frac{1}{2}(p_2 - iq_2)e^{iv_2 T_1} \quad (21)$$

where  $p_n$  and  $q_n$ , ( $n=1,2$ ) are real values and. Inserting equations (19) in to the linear form of equations (13a) ,(13b) and separating real and imaginary parts, the following system of equations is obtained as:

$$p_1' + v_1 p_1 - \eta_1 q_1 = 0 \quad (22a)$$

$$q_1' + v_1 q_1 + \eta_1 p_1 = 0 \quad (22b)$$

$$p_2' + v_2 p_2 - \eta_2 q_2 = 0 \quad (22c)$$

$$q_2' + v_2 q_2 + \eta_2 p_2 = 0 \quad (22d)$$

where  $\eta_1 = \frac{\mu_1}{\omega_1}, \eta_2 = \frac{\mu_2}{\omega_2}, v_1 = \frac{\sigma_1}{\omega_1}, v_2 = \frac{\sigma_2}{\omega_2}$

The stability of linear solution is investigated from the zero characteristics matrix

$$\begin{vmatrix} \lambda + v_1 & -\eta_1 & 0 & 0 \\ \eta_1 & \lambda + v_1 & 0 & 0 \\ 0 & 0 & \lambda + v_2 & -\eta_2 \\ 0 & 0 & \eta_2 & \lambda + v_2 \end{vmatrix} = 0$$

The eigen values are given by

$$\lambda^4 + r_1 \lambda^3 + r_2 \lambda^2 + r_3 \lambda + r_4 = 0$$

where,  $r_1 = 2(\eta_1 + \eta_2), r_2 = \eta_1^2 + \eta_2^2 + 4\eta_1 \eta_2 + v_1^2 + v_2^2$

$$r_3 = 2\eta_1 \eta_2 (\eta_1 + \eta_2) - 2\eta_2 v_1^2 + 2\eta_1 v_2^2, r_4 = (\eta_1^2 + v_1^2)(v_2^2 + \eta_2^2)$$

According to the Routh-Hurwitz criterion, the linear solution is stable if the following are satisfied

$$r_1 > 0, \quad r_1 r_2 - r_3 > 0, \quad r_3(r_1 r_2 - r_3) - r_1^2 r_4 > 0, \quad r_4 > 0$$

#### 4- RESULTS AND DISCUSSION

A non –linear control law is proposed to suppress the vibration of the pitch-roll ship under modulated external and parametric excitation. The system is modeled by two second order non-linear ordinary differential equations and the

control is based on cubic velocity feedback. The numerical solutions are given by Runge–kutta 4<sup>th</sup> order method at non resonance case as shown in Fig. 2, which consider as basic case. From this figure it can be seen that the amplitudes of roll(x)and pitch (y)are about 0.02 and 0.012 respectively.

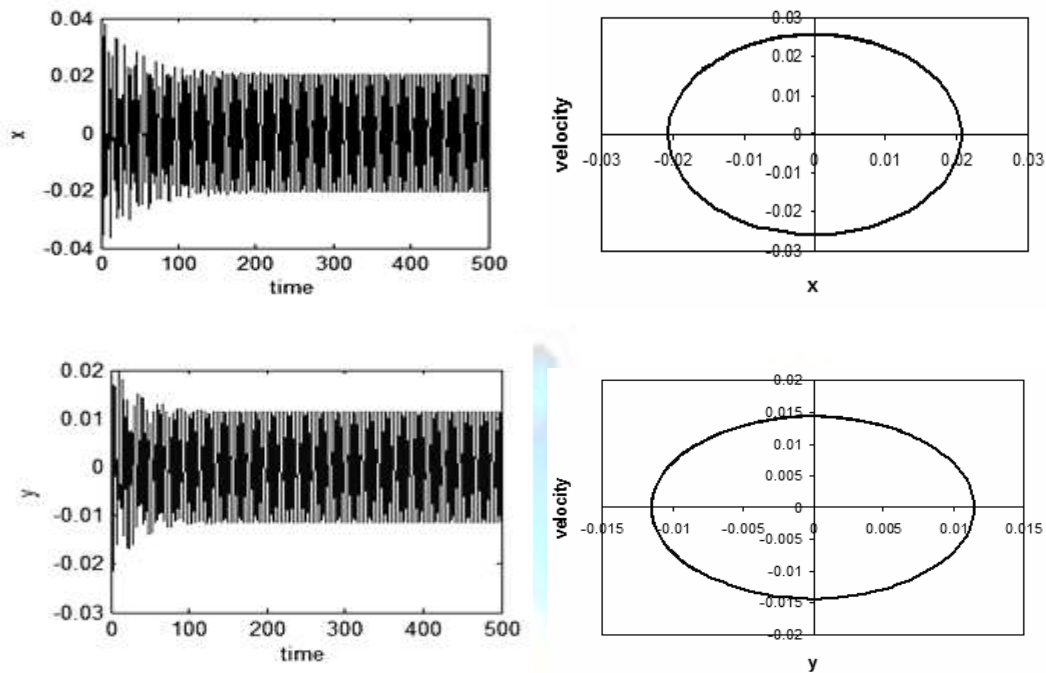


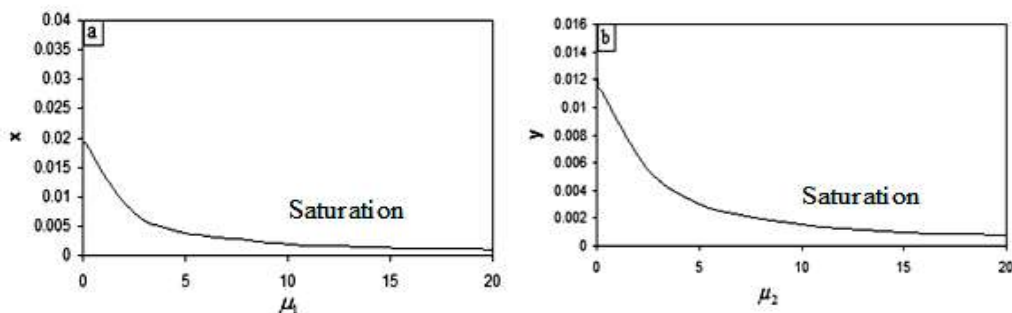
Fig.2. The Plot of basic case

#### 4.1- Effect of parameters

- i) For the positive damping coefficients the amplitudes of roll and pitch are monotonic decreasing functions on  $\mu_1$  and  $\mu_2$  respectively as shown in Figs. 3a, 3b, and more increasing values of  $\mu_1$  and  $\mu_2$  leads to saturation phenomena.
- ii) From Figs. 3c, 3d, we can see that the steady state amplitude of the roll and pitch are monotonic decreasing functions of the nonlinear parameters  $\alpha_1$  and  $\alpha_2$ .
- iii) The steady state amplitude of the roll and pitch are monotonic increasing functions of the excitation force amplitude  $G_o$ ,  $F_o$ ,  $F_1$  and  $F_2$ , which leads to the system is becomes un stable. as shown in Figs. 3e, 3f, 3g, 3h, and 3i respectively.

#### 4.2- Effect of the control

From Figs. 3j and 3k, the steady state amplitude of the roll and pitch are monotonic decreasing functions of the gains of control  $G_1$  and  $G_2$  respectively. but more increasing of values  $G_1$  and  $G_2$  leads to saturation phenomena.



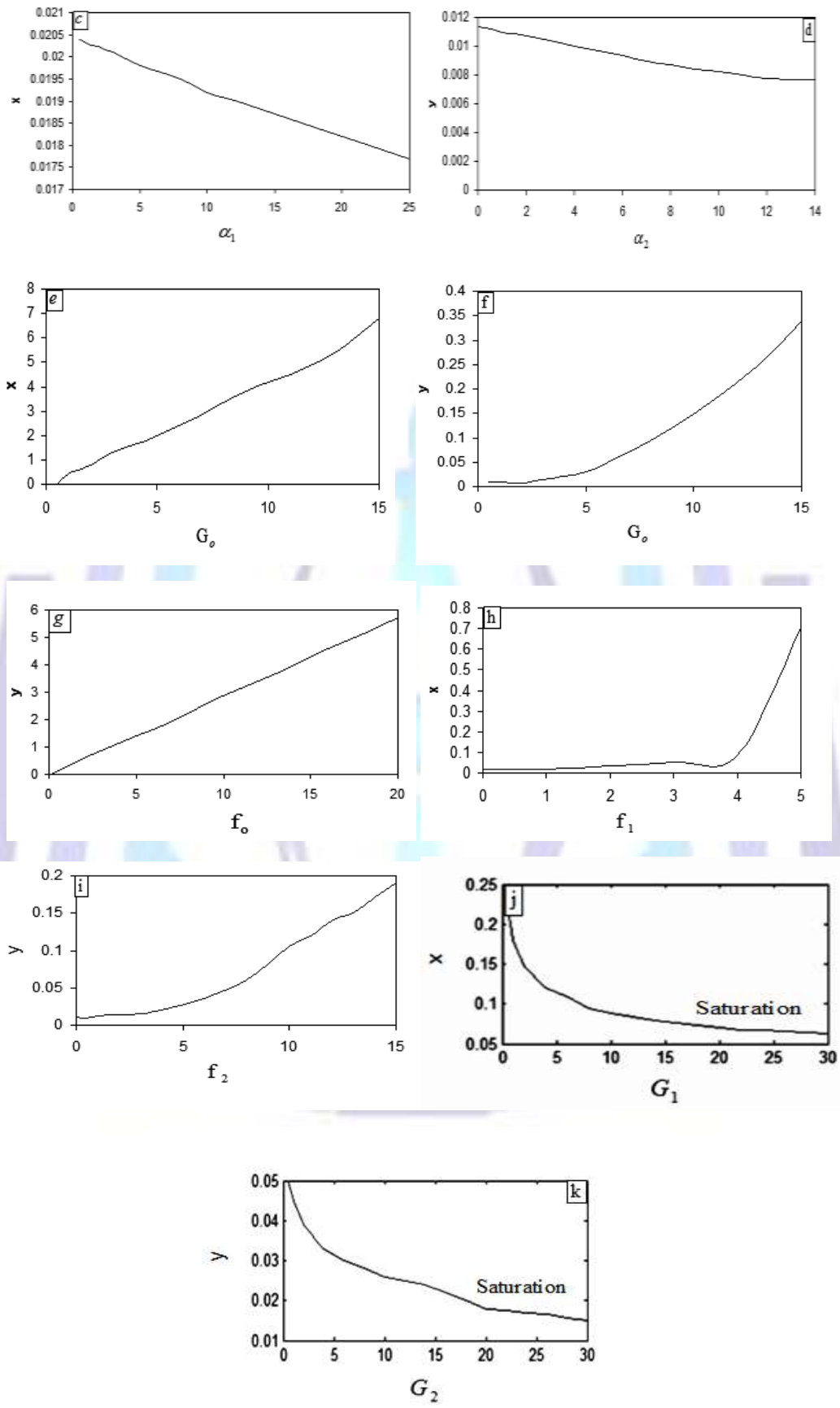


Fig.3. Effects of parameters

### 4.3- Resonance cases

All resonance cases of the system which are studied numerically obtain the worst case. From table 1, we find that the worst resonance case is simultaneous primary resonance  $\Omega_1 \cong \omega_1, \omega_2 \cong 2\omega_1$ , so we take this case to study the effect of control on the system Fig.4, shows the steady state amplitudes of the system at  $\Omega_1 \cong \omega_1, \omega_2 \cong 2\omega_1$ , without control ( $G_1 = G_2 = 0$ ), from this figure we see that the amplitude of the roll is increased to about 0.67 (34 times) and the amplitude of the pitch is increased to about 0.064(6times) respectively, of the basic case which shown in Fig.2.

Fig.5 illustrate the response of the system with control, which shows that the steady state of the roll is decreased to about 0.14 and steady state of the pitch is decreased to about 0.005, which means that the control are effective and  $E_a$  is 4.7 and 12.8 respectively.

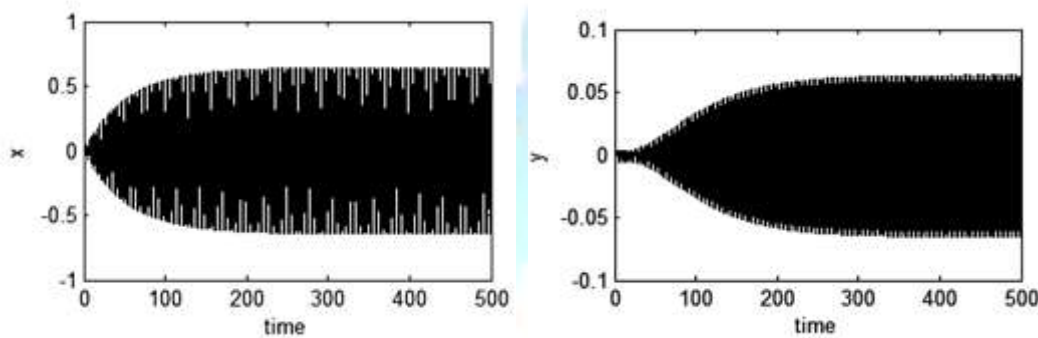


Fig. 4. Shows that the steady state amplitude without control at simultaneous primary resonance  $\Omega_1 \cong \omega_1, \omega_2 \cong 2\omega_1$

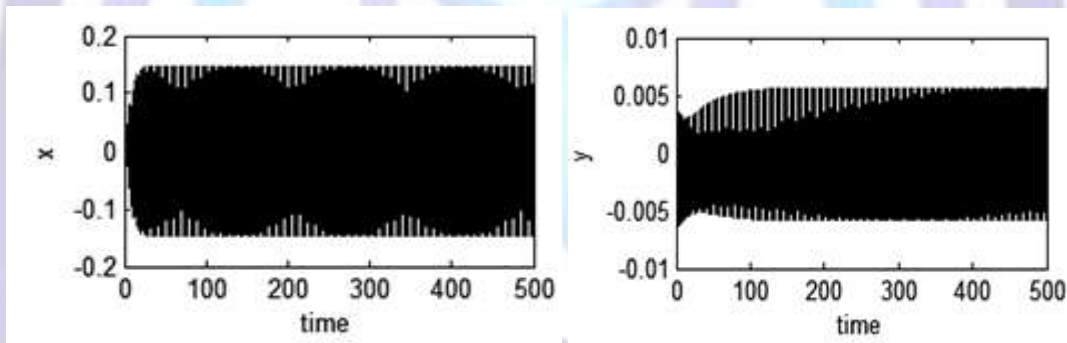


Fig. 5. The response for the system with absorber at the simultaneous primary resonance  $\Omega_1 \cong \omega_1, \omega_2 \cong 2\omega_1$



To summaries the worst resonance cases with and without absorber

**Table 1. Summary of worst resonance case**

Cases	Without control		With control		$E_a$	$E_a$	Remarks
	x	y	x	y	x	y	
$\Omega_1 \cong \omega_1$	0.63	0.037	0.14	0.0368	4.5	1	Limit cycle
$\Omega_1 \cong \omega_2$	0.046	0.3	0.0458	0.10	1	3	Limit cycle
$\omega_1 \cong (\Omega_2 + \omega_2) / 4$	0.99	0.020	0.22	0.012	4.5	1.7	Limit cycle
$\Omega_1 \cong \omega_1, \Omega_2 \cong \omega_1$	0.63	0.0358	0.14	0.0352	4.5	1	Limit cycle
$\Omega_1 \cong \omega_1, \omega_2 \cong \omega_1$	0.62	0.33	0.147	0.116	4.2	2.8	Limit cycle
$\Omega_1 \cong \omega_1, \omega_2 \cong 2\omega_1$	0.67	0.064	0.14	0.005	4.7	12.8	Multi Limit cycle
$\Omega_1 \cong \omega_1, \omega_1 \cong 3\omega_2 / 4$	0.65	0.018	0.14	0.012	4.6	1.5	Limit cycle
$\Omega_1 \cong \omega_1, \omega_2 \cong 3\Omega_2 / 2$	0.63	0.014	0.147	0.002	4.3	7	Limit cycle
$\Omega_2 \cong \omega_1, \omega_2 \cong 2\omega_1$	0.048	0.037	0.014	0.0112	3.4	3.3	Limit cycle
$\Omega_1 \cong \omega_1, \Omega_2 \cong \omega_1, \omega_2 \cong \omega_1$	0.63	0.33	0.136	0.12	4.6	2.75	Limit cycle

#### 4.4- Frequency response curves

The frequency response equations 17-20 are nonlinear equations of the amplitudes of the roll ( $a_1$ ) and the pitch ( $a_2$ ) against the detuning parameters  $\sigma_1, \sigma_2$ , which solved numerically as shown in Figs. (6-8) respectively. From these figures we find the amplitude of the roll is monotonic decreasing functions of the damping coefficient  $\mu_1$  and natural frequency  $\omega_1$  and the nonlinear parameter  $\alpha_1$  and the gain  $G_1$  as shown in Figs. 6a, b, c and Figs. 8a, b, c. But the steady state amplitude of the roll ( $a_1$ ) is monotonic increasing function in the excitation amplitude  $G_o$  as shown in Figs. 6d, 8d. Similarly, the steady state of the pitch ( $a_2$ ) is monotonic decreasing function of the damping coefficient  $\mu_2$  and natural frequency  $\omega_2$  and the gain  $G_2$  and the nonlinear parameter  $\alpha_2$  as shown in Figs. 7a, b, c and Figs. 8e, f, g, h which are a good agreement of the effect of parameter shown in Fig.2.

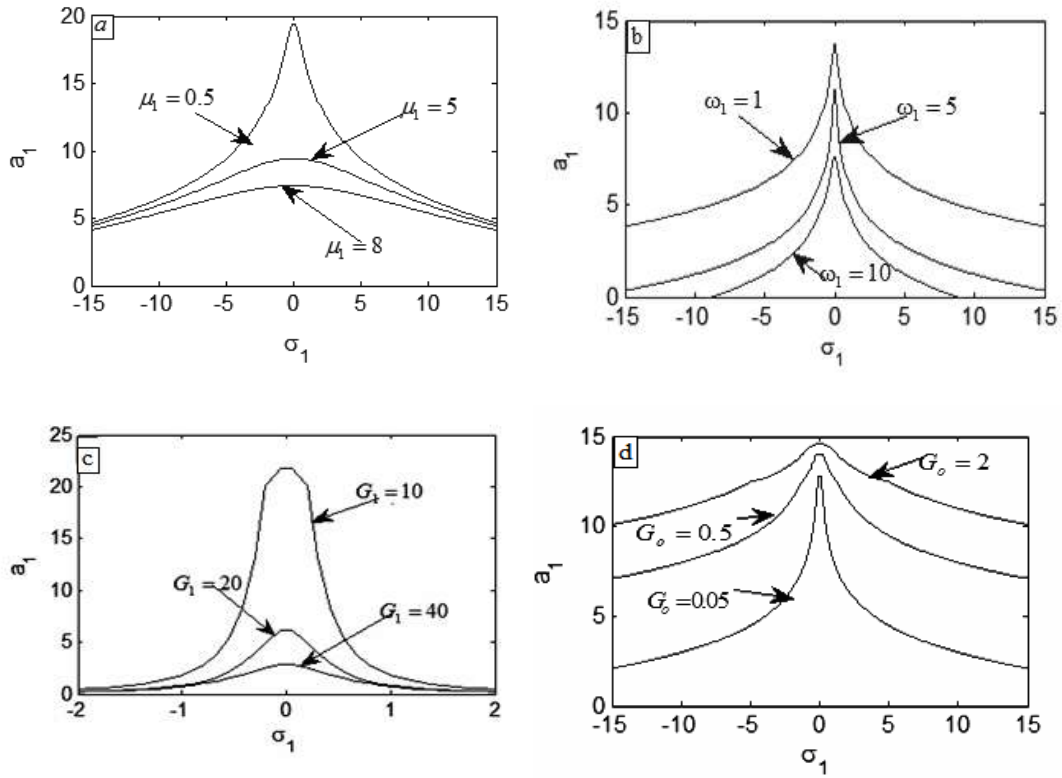


Fig.6 Frequency response curves of the first case ( $a_1 \neq 0, a_2 = 0$ )

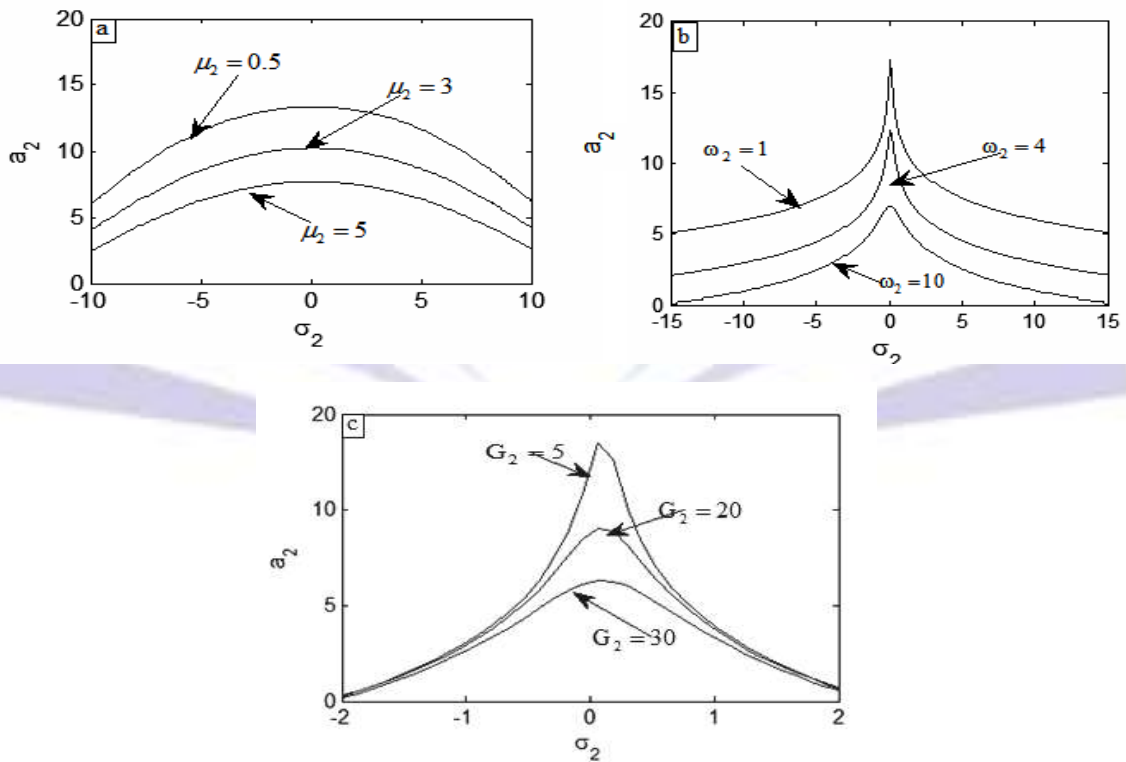


Fig. 7 Frequency response curves of the first case ( $a_1 = 0, a_2 \neq 0$ )

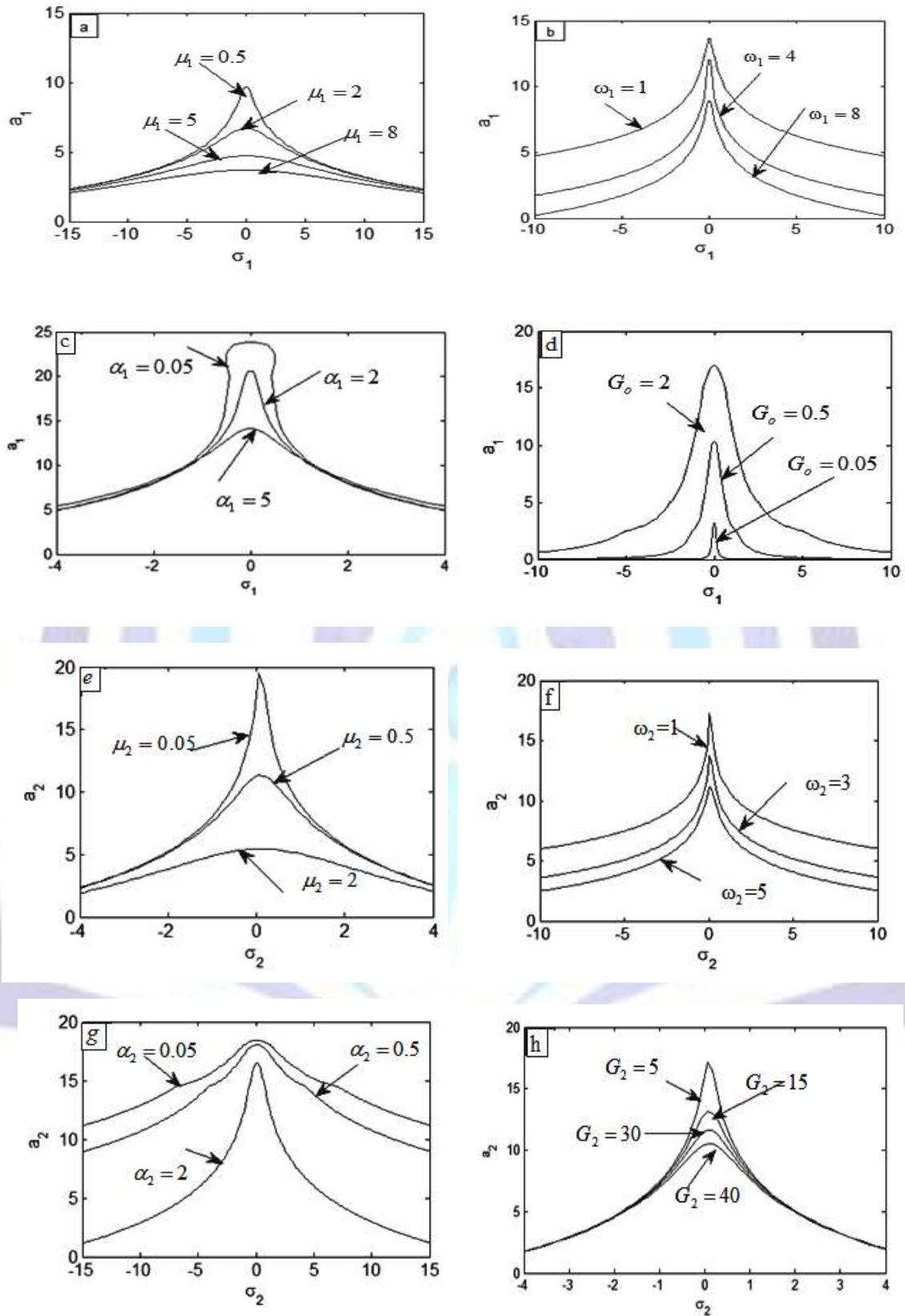


Fig.8. Frequency response curves of the first case ( $a_1 \neq 0, a_2 \neq 0$ )



## 5- CONCLUSIONS

The vibration of second order coupled system simulating the vibration of the roll-ship motion subjected to excitation forces is studied at the simultaneous primary resonance case. A control law based on cubic velocity feedback is proposed. The method of multiple scales is used to obtain the approximate solution of the system. The stability and effect of parameters are studied numerically. From the above study the following may be concluded

- 1- For the positive damping coefficients the amplitudes of roll and pitch are monotonic decreasing functions on  $\mu_1$  and  $\mu_2$ .
- 2- The steady state amplitude of the roll and pitch are monotonic decreasing functions of the nonlinear parameters  $\alpha_1$  and  $\alpha_2$ .
- 3- The steady state amplitude of the roll and pitch are monotonic increasing functions of the excitation force amplitude  $G_o$ ,  $F_o$ ,  $F_1$  and  $F_2$ , which leads to the system is unstable.
- 4- The worst resonance case is simultaneous primary resonance  $\Omega_1 \cong \omega_1, \omega_2 \cong 2\omega_1$ , which the amplitudes of the roll and the pitch are increased to about 0.67 (34 times) and 0.064 (6times) respectively, of the basic case.
- 5- The steady state amplitude of the roll ( $a_1$ ) is monotonic decreasing functions of the gain  $G_1$ , the steady state of the pitch ( $a_2$ ) is monotonic decreasing function of the gain  $G_2$ .

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