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# Fast Dynamic Model of a Moving-base 6-DOF Parallel Manipulator

António M. Lopes

*Unidade de Integração de Sistemas e Processos Automatizados,  
Faculdade de Engenharia, Universidade do Porto  
Portugal*

## 1. Introduction

Dynamic models play an important role in parallel manipulators simulation and control. Mainly in the later case, the efficiency of the involved computations is of paramount importance, because manipulator real-time control is usually necessary (Zhao & Gao, 2009). The dynamic model of a parallel manipulator operating in free space can be represented in Cartesian coordinates by a system of nonlinear differential equations, which may be written in matrix form as

$$\mathbf{I}(\mathbf{x}) \cdot \ddot{\mathbf{x}} + \mathbf{V}(\mathbf{x}, \dot{\mathbf{x}}) \cdot \dot{\mathbf{x}} + \mathbf{G}(\mathbf{x}) = \mathbf{f} \quad (1)$$

$\mathbf{I}(\mathbf{x})$  being the inertia matrix,  $\mathbf{V}(\mathbf{x}, \dot{\mathbf{x}})$  the Coriolis and centripetal terms matrix,  $\mathbf{G}(\mathbf{x})$  a vector of gravitational generalized forces,  $\mathbf{x}$  the generalized position of the moving platform (or end-effector) and  $\mathbf{f}$  the controlled generalized force applied on the end-effector. Thus,

$$\mathbf{f} = \mathbf{J}^T(\mathbf{x}) \cdot \boldsymbol{\tau} \quad (2)$$

where  $\boldsymbol{\tau}$  is the generalized force developed by the actuators and  $\mathbf{J}(\mathbf{x})$  is the inverse kinematics jacobian matrix (Merlet, 2006).

Dynamic modelling of parallel manipulators presents an inherent complexity, mainly due to system closed-loop structure and kinematic constraints. Several approaches have been applied to the dynamic analysis of parallel manipulators, the Newton-Euler and the Lagrange methods being the most popular ones (Do & Yang, 1988; Reboulet & Berthomieu, 1991; Ji, 1994; Dasgupta & Mruthyunjaya, 1998; Khalil & Ibrahim, 2007; Riebe & Ulbrich, 2003; Guo & Li, 2006; Nguyen & Pooran, 1989; Lebrete et al., 1993; Di Gregório & Parenti-Castelli, 2004; Caccavale et al., 2003; Dasgupta & Choudhury, 1999). These methods use classical mechanics principles, as is the case for all the approaches found in the literature, namely the ones based on the principle of virtual work (Staicu et al., 2007; Tsai, 2000; Wang & Gosselin, 1998), screw theory (Gallardo et al., 2003), recursive matrix method (Staicu & Zhang, 2008), Hamilton's principle (Miller, 2004), and Kane's equation (Liu et al., 2000).

Thus, all approaches are equivalent, leading to equivalent dynamic equations. Nevertheless, these equations can present different levels of complexity and associated computational loads (Zhao & Gao, 2009). Minimizing the number of operations involved in the computation of the manipulator dynamic model has been the main goal of recent proposed techniques (Zhao & Gao, 2009; Staicu & Zhang, 2008; Abdellatif & Heimann, 2009; Wang et al., 2007; Sokolov & Xirouchakis, 2007; Bhattacharya et al., 1997; Carricato & Gosselin, 2009; Lopes, 2009).

This book chapter presents the generalized momentum concept to model the dynamics of a Stewart platform manipulator having a non-stationary base platform. This is important, for example, in macro/micro robotic applications, where a small manipulator is attached in series to a big one. The later performs large amplitude movements, while the former is only responsible for the small motions. The book chapter is organized as follows. Section 2 presents a brief description of the parallel manipulator under study. In section 3 a complete dynamic model is developed. The generalized momentum approach is used and the motion of the manipulator base platform is considered. Conclusions are drawn in section 4.

## 2. Manipulator Kinematic Structure

A Stewart platform manipulator is being considered. It comprises a (usually) fixed platform (the base) and a moving platform (the payload platform), linked together by six independent, identical, open kinematic chains (Raghavan, 1993). In this book chapter a particular design will be considered as shown in Figure 1 (Fichter, 1986). In this case, each chain (leg) comprises a cylinder and a piston (or spindle) that are connected together by a prismatic joint,  $l_i$ . The upper end of each leg is connected to the moving platform by a spherical joint whereas the lower end is connected to the fixed base by a universal joint. Points  $B_i$  and  $P_i$  are the connecting points to the base and moving platforms, respectively (Figure 2). They are located at the vertices of two semi-regular hexagons inscribed in circumferences of radius  $r_B$  and  $r_P$ . The separation angles between points  $B_1$  and  $B_6$ ,  $B_2$  and  $B_3$ , and  $B_4$  and  $B_5$  are denoted by  $2\phi_B$ . In a similar way, the separation angles between points  $P_1$  and  $P_2$ ,  $P_3$  and  $P_4$ , and  $P_5$  and  $P_6$  are denoted by  $2\phi_P$  (Figure 2).

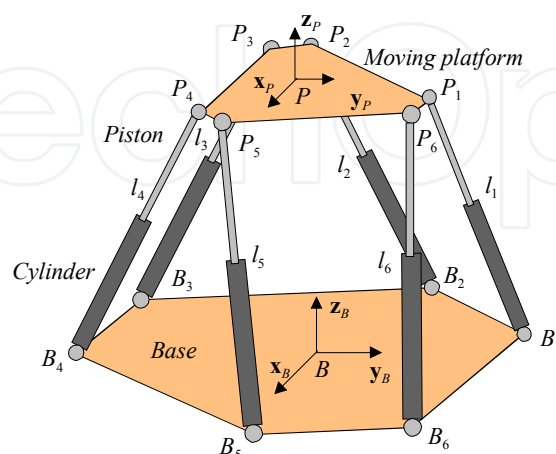


Fig. 1. Stewart platform kinematic structure

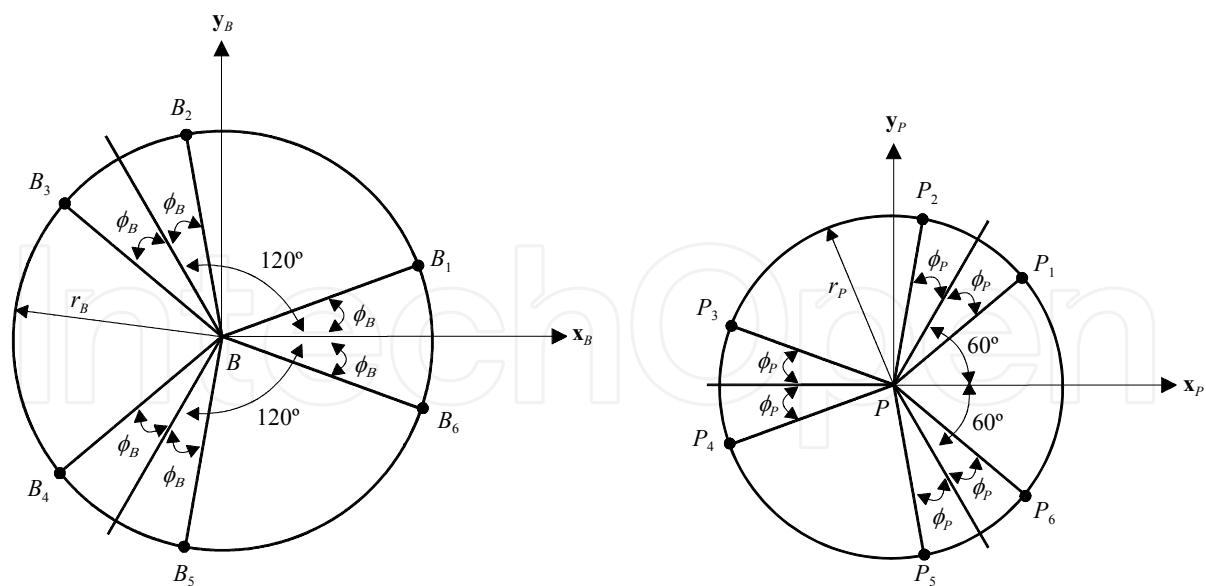


Fig. 2. Position of the connecting points to the base and payload platforms

For kinematic modelling purposes, two frames, {P} and {B}, are attached to the moving and base platforms, respectively. Its origins are the platforms centres of mass. The generalized position of frame {P} relative to frame {B} may be represented by the vector:

$${}^B\mathbf{x}_{P|_{B/E}} = [x_P \quad y_P \quad z_P \quad \psi_P \quad \theta_P \quad \varphi_P]^T = [{}^B\mathbf{x}_{P(pos)}^T|_B \quad {}^B\mathbf{x}_{P(o)}^T|_E]^T \tag{3}$$

where  ${}^B\mathbf{x}_{P(pos)|_B} = [x_P \quad y_P \quad z_P]^T$  is the position of the origin of frame {P} relative to frame {B}, and  ${}^B\mathbf{x}_{P(o)|_E} = [\psi_P \quad \theta_P \quad \varphi_P]^T$  defines an Euler angles system representing orientation of frame {P} relative to {B}. The used Euler angles system corresponds to the basic rotations (Vukobratovic & Kircanski, 1986):  $\psi_P$  about  $\mathbf{z}_P$ ;  $\theta_P$  about the rotated axis  $\mathbf{y}_P$ ; and  $\varphi_P$  about the rotated axis  $\mathbf{x}_P$ . The rotation matrix is given by:

$${}^B\mathbf{R}_P = \begin{bmatrix} C\psi_P C\theta_P & C\psi_P S\theta_P S\varphi_P - S\psi_P C\varphi_P & C\psi_P S\theta_P C\varphi_P + S\psi_P S\varphi_P \\ S\psi_P C\theta_P & S\psi_P S\theta_P S\varphi_P + C\psi_P C\varphi_P & S\psi_P S\theta_P C\varphi_P - C\psi_P S\varphi_P \\ -S\theta_P & C\theta_P S\varphi_P & C\theta_P C\varphi_P \end{bmatrix} \tag{4}$$

$S(\cdot)$  and  $C(\cdot)$  correspond to the sine and cosine functions, respectively. The manipulator position and velocity kinematic models are well known, being obtainable from the geometrical analysis of the kinematics chains. The velocity kinematics is represented by the Euler angles jacobian matrix,  $\mathbf{J}_E$ , or the kinematic jacobian,  $\mathbf{J}_C$  (Merlet, 2006). These jacobians relate the velocities of the active joints (the actuators) to the generalized velocity of the moving platform:

$$\dot{\mathbf{i}} = \mathbf{J}_E \cdot {}^B \dot{\mathbf{x}}_{P|B/E} = \mathbf{J}_E \cdot \begin{bmatrix} {}^B \dot{\mathbf{x}}_{P(pos)|B} \\ {}^B \dot{\mathbf{x}}_{P(o)|E} \end{bmatrix} \quad (5)$$

$$\dot{\mathbf{i}} = \mathbf{J}_C \cdot {}^B \dot{\mathbf{x}}_{P|B} = \mathbf{J}_C \cdot \begin{bmatrix} {}^B \dot{\mathbf{x}}_{P(pos)|B} \\ {}^B \boldsymbol{\omega}_{P|B} \end{bmatrix} \quad (6)$$

with

$$\dot{\mathbf{i}} = \begin{bmatrix} \dot{i}_1 & \dot{i}_2 & \dots & \dot{i}_6 \end{bmatrix}^T \quad (7)$$

$${}^B \boldsymbol{\omega}_{P|B} = \mathbf{J}_A \cdot {}^B \dot{\mathbf{x}}_{P(o)|E} \quad (8)$$

and (Vukobratovic & Kircanski, 1986)

$$\mathbf{J}_A = \begin{bmatrix} 0 & -S\psi_p & C\theta_p C\psi_p \\ 0 & C\psi_p & C\theta_p S\psi_p \\ 1 & 0 & -S\theta_p \end{bmatrix} \quad (9)$$

Vectors  ${}^B \dot{\mathbf{x}}_{P(pos)|B} \equiv {}^B \mathbf{v}_{P|B}$  and  ${}^B \boldsymbol{\omega}_{P|B}$  represent the linear and angular velocity of the moving platform relative to {B}, and  ${}^B \dot{\mathbf{x}}_{P(o)|E}$  represents the Euler angles time derivative.

### 3. Complete Dynamic Modelling Using the Generalized Momentum Approach

It is well known the generalized momentum of a rigid body,  $\mathbf{q}_c$ , may be computed from the following general expression:

$$\mathbf{q}_c = \mathbf{I}_c \cdot \mathbf{u}_c \quad (10)$$

Vector  $\mathbf{u}_c$  represents the generalized velocity (linear and angular) of the body and  $\mathbf{I}_c$  is its inertia matrix. Vectors  $\mathbf{q}_c$  and  $\mathbf{u}_c$  and inertia matrix  $\mathbf{I}_c$  must be expressed in the same frame of reference.

Equation (10) may also be written as

$$\mathbf{q}_c = \begin{bmatrix} \mathbf{Q}_c \\ \mathbf{H}_c \end{bmatrix} = \begin{bmatrix} \mathbf{I}_{c(tr)} & \mathbf{0} \\ \mathbf{0} & \mathbf{I}_{c(rot)} \end{bmatrix} \cdot \begin{bmatrix} \mathbf{v}_c \\ \boldsymbol{\omega}_c \end{bmatrix} \quad (11)$$

where  $\mathbf{Q}_c$  is the linear momentum vector due to rigid body translation and  $\mathbf{H}_c$  is the angular momentum vector due to body rotation.  $\mathbf{I}_{c(tr)}$  is the translational inertia matrix and  $\mathbf{I}_{c(rot)}$  is the rotational inertia matrix.  $\mathbf{v}_c$  and  $\boldsymbol{\omega}_c$  are the body linear and angular velocities.

The inertial component of the generalized force (force and moment) acting on the body can be obtained from the time derivative of equation (10):

$$\mathbf{f}_{c(in)} = \dot{\mathbf{q}}_c = \dot{\mathbf{I}}_c \cdot \mathbf{u}_c + \mathbf{I}_c \cdot \dot{\mathbf{u}}_c \quad (12)$$

with force and moment expressed in the same frame and  $\mathbf{f}_{c(in)} = \begin{bmatrix} \mathbf{F}_{c(in)}^T & \mathbf{M}_{c(in)}^T \end{bmatrix}^T$ .

Equivalently, force and moment vectors are:

$$\mathbf{F}_{c(in)} = \dot{\mathbf{Q}}_c = \mathbf{I}_{c(tr)} \cdot \dot{\mathbf{v}}_c + \dot{\mathbf{I}}_{c(tr)} \cdot \mathbf{v}_c \quad (13)$$

$$\mathbf{M}_{c(in)} = \dot{\mathbf{H}}_c = \mathbf{I}_{c(rot)} \cdot \dot{\boldsymbol{\omega}}_c + \dot{\mathbf{I}}_{c(rot)} \cdot \boldsymbol{\omega}_c \quad (14)$$

### 3.1 Payload Platform Modelling

Considering the Stewart platform manipulator base motion, i.e., the motion of frame {B} relative to a fixed world frame {W}= $\{\mathbf{x}_W, \mathbf{y}_W, \mathbf{z}_W\}$ , the position of the payload platform, {P}, relative to {W} and expressed in {W}, may be given by

$${}^W \mathbf{p}_{P|W} = {}^W \mathbf{p}_{B|W} + {}^B \mathbf{p}_{P|B} \quad (15)$$

where  ${}^W \mathbf{p}_{B|W}$  is the position of frame {B}, and  ${}^B \mathbf{p}_{P|B}$  represents the position of frame {P} relative to {B} and expressed in {W}.

The linear velocity of the payload platform, {P}, relative to {W} and expressed in {W}, may be obtained taking the time derivative of the previous equation, that is,

$${}^W \mathbf{v}_{P|W} = {}^W \mathbf{v}_{B|W} + {}^B \mathbf{v}_{P|B} + {}^W \boldsymbol{\omega}_{B|W} \times {}^B \mathbf{p}_{P|B} \quad (16)$$

where  ${}^W \mathbf{v}_{B|W}$  is the linear velocity of frame {B},  ${}^B \mathbf{v}_{P|B}$  is the linear velocity of frame {P} as seen by an observer fixed in {B},  ${}^W \boldsymbol{\omega}_{B|W}$  represents the angular velocity of frame {B} relative to {W}, and  ${}^B \mathbf{p}_{P|B} \equiv {}^B \mathbf{x}_{P(pos)|B}$  represents the position of {P} relative to {B} and expressed in {W}.

In the following analysis, knowledge of the generalized position of frame {B} relative to {W},  ${}^W \mathbf{x}_{B|W|E} = [x_B \ y_B \ z_B \ \psi_B \ \theta_B \ \phi_B]^T = [{}^W \mathbf{x}_{B(pos)|W}^T \ {}^W \mathbf{x}_{B(o)|E}^T]^T$ , shall be assumed:

${}^W \mathbf{x}_{B(pos)|W}$  represents the position vector expressed in frame {W}, and  ${}^W \mathbf{x}_{B(o)|E}$  represents the orientation expressed in an Euler angles system. Knowledge of its first and second time derivatives shall also be supposed i.e.,  ${}^W \dot{\mathbf{x}}_{B|W|E}$  and  ${}^W \ddot{\mathbf{x}}_{B|W|E}$ , respectively. Therefore, the orientation matrix,  ${}^W \mathbf{R}_B$ , of frame {B} relative to {W} can be easily computed, and the jacobian,  $\mathbf{J}_G$ , relating the angular velocity of the base frame relative to {W},  ${}^W \boldsymbol{\omega}_{B|W}$ , to the first time derivative of the Euler angles,  ${}^W \dot{\mathbf{x}}_{B(o)|E}$ , is given by

$${}^W \boldsymbol{\omega}_{B|W} = \mathbf{J}_G \cdot {}^W \dot{\mathbf{x}}_{B(o)|E} \quad (17)$$

Considering equation (16), in frame {B}, the following equation can be written:

$${}^W\mathbf{v}_{P|B} = {}^W\mathbf{v}_{B|B} + {}^B\mathbf{v}_{P|B} + {}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{p}_{P|B} \quad (18)$$

Therefore, the total linear momentum of {P} expressed in frame {B} will be

$$\mathbf{Q}_{P(tot)|B} = m_P \cdot {}^W\mathbf{v}_{P|B} \quad (19)$$

$m_P$  being the payload platform mass.

Taking the time derivative of the previous equation results in

$$\dot{\mathbf{Q}}_{P(tot)|B} = m_P \cdot \dot{{}^W\mathbf{v}}_{P|B} \quad (20)$$

Knowing that,

$${}^W\boldsymbol{\omega}_{P|B} = {}^B\boldsymbol{\omega}_{P|B} + {}^W\boldsymbol{\omega}_{B|B} \quad (21)$$

$${}^W\dot{\boldsymbol{\omega}}_{P|B} = {}^B\dot{\boldsymbol{\omega}}_{P|B} + {}^W\boldsymbol{\omega}_{B|B} \times {}^B\boldsymbol{\omega}_{P|B} + {}^W\dot{\boldsymbol{\omega}}_{B|B} \quad (22)$$

results in

$${}^W\dot{\mathbf{v}}_{P|B} = {}^W\dot{\mathbf{v}}_{B|B} + {}^B\dot{\mathbf{v}}_{P|B} + 2 {}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{v}_{P|B} + {}^W\dot{\boldsymbol{\omega}}_{B|B} \times {}^B\mathbf{p}_{P|B} + {}^W\boldsymbol{\omega}_{B|B} \times ({}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{p}_{P|B}) \quad (23)$$

The inertial part of the total force applied in {P}, due to the payload platform translation, expressed in frame {B} will be

$${}^P\mathbf{F}_{P(ine)(tot)|B} = \dot{\mathbf{Q}}_{P(tot)|B} \quad (24)$$

That is,

$${}^P\mathbf{F}_{P(ine)(tot)|B} = {}^P\mathbf{F}_{P(ine)(fix)|B} + {}^P\mathbf{F}_{P(ine)(man)|B} \quad (25)$$

$${}^P\mathbf{F}_{P(ine)(fix)|B} = m_P \cdot {}^B\dot{\mathbf{v}}_{P|B} \quad (26)$$

$${}^P\mathbf{F}_{P(ine)(man)|B} = m_P \cdot ({}^W\dot{\mathbf{v}}_{B|B} + 2 {}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{v}_{P|B} + {}^W\dot{\boldsymbol{\omega}}_{B|B} \times {}^B\mathbf{p}_{P|B} + {}^W\boldsymbol{\omega}_{B|B} \times ({}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{p}_{P|B})) \quad (27)$$

where  ${}^P\mathbf{F}_{P(ine)(fix)|B}$  represents the inertial part of the force, considering the base platform is not moving, and  ${}^P\mathbf{F}_{P(ine)(man)|B}$  represents the inertial part of the force which results from the base motion.

On the other hand, the angular momentum of the moving platform, about its centre of mass and expressed in frame {B} will be

$$\mathbf{H}_{P(tot)|_B} = \mathbf{I}_{P(rot)|_B} \cdot^W \boldsymbol{\omega}_P|_B \quad (28)$$

$\mathbf{I}_{P(rot)|_B}$  represents the rotational inertia matrix of the moving platform, expressed in the base frame, {B}. This inertia matrix is given by:

$$\mathbf{I}_{P(rot)|_B} = {}^B \mathbf{R}_P \cdot \mathbf{I}_{P(rot)|_P} \cdot {}^B \mathbf{R}_P^T \quad (29)$$

where  $\mathbf{I}_{P(rot)|_P}$  is a constant matrix representing the rotational inertia matrix of the moving platform, expressed in frame {P}. Considering that  $I_{P_{xx}}$ ,  $I_{P_{yy}}$  and  $I_{P_{zz}}$  are the moments of inertia of the moving platform expressed in its own frame, this matrix may be written as:

$$\mathbf{I}_{P(rot)|_P} = \text{diag}([I_{P_{xx}} \quad I_{P_{yy}} \quad I_{P_{zz}}]) \quad (30)$$

Taking the time derivative of equation (28) results in

$$\dot{\mathbf{H}}_{P(tot)|_B} = \dot{\mathbf{I}}_{P(rot)|_B} \cdot^W \boldsymbol{\omega}_P|_B + \mathbf{I}_{P(rot)|_B} \cdot^W \dot{\boldsymbol{\omega}}_P|_B \quad (31)$$

The inertial part of the total moment applied in {P}, due to the payload platform rotation, expressed in frame {B} will be

$${}^P \mathbf{M}_{P(ine)(tot)|_B} = \dot{\mathbf{H}}_{P(tot)|_B} \quad (32)$$

that is,

$${}^P \mathbf{M}_{P(ine)(tot)|_B} = {}^P \mathbf{M}_{P(ine)(fix)|_B} + {}^P \mathbf{M}_{P(ine)(man)|_B} \quad (33)$$

where,

$${}^P \mathbf{M}_{P(ine)(fix)|_B} = \mathbf{I}_{P(rot)|_B} \cdot^B \dot{\boldsymbol{\omega}}_P|_B + \dot{\mathbf{I}}_{P(rot)|_B} \cdot^B \boldsymbol{\omega}_P|_B \quad (34)$$

$${}^P \mathbf{M}_{P(ine)(man)|_B} = \dot{\mathbf{I}}_{P(rot)|_B} \cdot^W \boldsymbol{\omega}_B|_B + \mathbf{I}_{P(rot)|_B} \cdot \left( {}^W \dot{\boldsymbol{\omega}}_B|_B + {}^W \boldsymbol{\omega}_B|_B \times {}^B \boldsymbol{\omega}_P|_B \right) \quad (35)$$

${}^P \mathbf{M}_{P(ine)(fix)|_B}$  represents the inertial part of the moment, considering the base platform is not moving, and  ${}^P \mathbf{M}_{P(ine)(man)|_B}$  represents the inertial part of the moment which results from the base motion.

The total inertial component of the generalized force applied to {P} and expressed in {B} will be



$${}^P \mathbf{f}_{P(ine)(tot) | B} = \begin{bmatrix} {}^P \mathbf{F}_{P(ine)(tot) | B}^T & {}^P \mathbf{M}_{P(ine)(tot) | B}^T \end{bmatrix}^T \quad (36)$$

The inertial components of the forces in the manipulator actuators (actuating forces) will be

$$\boldsymbol{\tau}_{P(ine)(fix)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{P(ine)(fix) | B} \quad (37)$$

$$\boldsymbol{\tau}_{P(ine)(man)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{P(ine)(man) | B} \quad (38)$$

On the other hand, regarding the gravitational part of the generalized force, if the base platform orientation changes, then the force applied to {P} and expressed in {B} results in

$${}^P \mathbf{f}_{P(gra) | B} = {}^W \mathcal{R}_B^T \cdot {}^P \mathbf{f}_{P(gra) | W} \quad (39)$$

where,

$${}^W \mathcal{R}_B = \begin{bmatrix} {}^W \mathbf{R}_B & \mathbf{0} \\ \mathbf{0} & {}^W \mathbf{R}_B \end{bmatrix} \quad (40)$$

and  ${}^P \mathbf{f}_{P(gra) | W}$  is the gravitational generalized force applied to {P} and expressed in {W}. This force can be computed using a simplified model that considers both a non-moving base platform, frame {B} parallel to {W}, and  $\mathbf{z}_B \equiv -\hat{\mathbf{g}}$ , i.e.,

$${}^P \mathbf{f}_{P(gra) | W} = \frac{\partial P_p \left( {}^B \mathbf{x}_P |_{B|E} \right)}{\partial {}^B \mathbf{x}_P |_{B|E}} \quad (41)$$

$P_p = m_p \cdot g \cdot z_p$  representing the mobile platform potential energy.

The gravitational component of the actuating forces due to the moving platform,  $\boldsymbol{\tau}_{P(gra)}$ , is given by equation (42), which can be added to equations (37) and (38).

$$\boldsymbol{\tau}_{P(gra)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{P(gra) | B} \quad (42)$$

### 3.2 Cylinder Modelling

Position of the cylinder  $i$ , relative to {W} and expressed in {W}, may be computed using the following equation:

$${}^W \mathbf{p}_{C_i | W} = {}^W \mathbf{p}_B | W + {}^B \mathbf{p}_{C_i | W} \quad (43)$$

The linear velocity of the cylinder, relative to {W} and expressed in {W}, may be obtained taking the time derivative of the previous equation, that is,

$${}^W \mathbf{v}_{C_i|W} = {}^W \mathbf{v}_{B|W} + {}^B \mathbf{v}_{C_i|B} + {}^W \boldsymbol{\omega}_{B|W} \times {}^B \mathbf{p}_{C_i|W} \quad (44)$$

Considering that frame  $\{C_i\}$  is attached to the cylinder  $i$  and positioned at its centre of mass, then  ${}^B \mathbf{v}_{C_i|B}$  is the linear velocity of frame  $\{C_i\}$  as seen by an observer fixed in  $\{B\}$ , and  ${}^B \mathbf{p}_{C_i|B}$  represents the position of  $\{C_i\}$  relative to  $\{B\}$  and expressed in  $\{W\}$ . In frame  $\{B\}$  the following equation can be written:

$${}^W \mathbf{v}_{C_i|B} = {}^W \mathbf{v}_{B|B} + {}^B \mathbf{v}_{C_i|B} + {}^W \boldsymbol{\omega}_{B|B} \times {}^B \mathbf{p}_{C_i|B} \quad (45)$$

Considering the centre of mass of each cylinder is located at a constant distance,  $b_C$ , from the cylinder to base platform connecting point,  $B_i$ , (Figure 3), then its position relative to frame  $\{B\}$  is

$${}^B \mathbf{p}_{C_i|B} = b_C \cdot \hat{\mathbf{l}}_i + \mathbf{b}_i \quad (46)$$

where,

$$\hat{\mathbf{l}}_i = \frac{\mathbf{l}_i}{\|\mathbf{l}_i\|} = \frac{\mathbf{l}_i}{l_i} \quad (47)$$

$$\mathbf{l}_i = {}^B \mathbf{x}_{P(pos)|B} + {}^P \mathbf{p}_{i|B} - \mathbf{b}_i \quad (48)$$

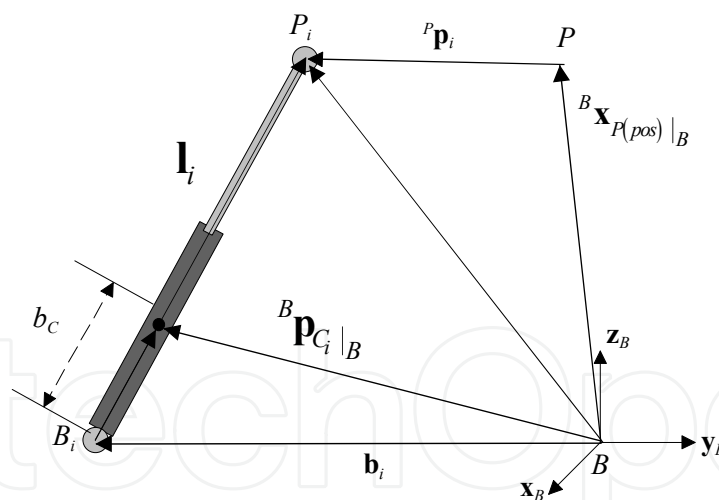


Fig. 3. Position of the centre of mass of the cylinder  $i$

The linear velocity of the cylinder centre of mass,  ${}^B \dot{\mathbf{p}}_{C_i|B}$ , relative to  $\{B\}$  and expressed in the same frame, may be computed as:

$${}^B \dot{\mathbf{p}}_{C_i|B} = {}^B \boldsymbol{\omega}_{l_i|B} \times b_C \cdot \hat{\mathbf{l}}_i \quad (49)$$

where  ${}^B \boldsymbol{\omega}_{l_i|B}$  represents the leg angular velocity, which can be found from:

$${}^B\boldsymbol{\omega}_{l_i|_B} \times \mathbf{l}_i = {}^B\mathbf{v}_{p|_B} + {}^B\boldsymbol{\omega}_{p|_B} \times {}^P\mathbf{p}_{i|_B} \quad (50)$$

As the leg (both the cylinder and piston) cannot rotate along its own axis, the angular velocity along  $\hat{\mathbf{l}}_i$  is always zero, and vectors  $\mathbf{l}_i$  and  ${}^B\boldsymbol{\omega}_{l_i|_B}$  are always perpendicular. This enables equation (50) to be rewritten as:

$${}^B\boldsymbol{\omega}_{l_i|_B} = \frac{1}{\mathbf{l}_i^T \cdot \mathbf{l}_i} \cdot \left[ \mathbf{l}_i \times \left( {}^B\mathbf{v}_{p|_B} + {}^B\boldsymbol{\omega}_{p|_B} \times {}^P\mathbf{p}_{i|_B} \right) \right] \quad (51)$$

or,

$${}^B\boldsymbol{\omega}_{l_i|_B} = \mathbf{J}_{D_i} \cdot \begin{bmatrix} {}^B\mathbf{v}_{p|_B} \\ {}^B\boldsymbol{\omega}_{p|_B} \end{bmatrix} \quad (52)$$

where jacobian  $\mathbf{J}_{D_i}$  is given by:

$$\mathbf{J}_{D_i} = \begin{bmatrix} \tilde{\mathbf{l}}_i & \tilde{\mathbf{l}}_i^P \tilde{\mathbf{p}}_{i|_B}^T \end{bmatrix} \quad (53)$$

$$\tilde{\mathbf{l}}_i = \frac{\mathbf{l}_i}{\mathbf{l}_i^T \cdot \mathbf{l}_i} \quad (54)$$

and, for a given vector  $\mathbf{a} = [a_x \ a_y \ a_z]^T$ ,

$$\tilde{\mathbf{a}} = \begin{bmatrix} 0 & -a_z & a_y \\ a_z & 0 & -a_x \\ -a_y & a_x & 0 \end{bmatrix} \quad (55)$$

On the other hand, equation (49) can be rewritten as:

$${}^B\dot{\mathbf{p}}_{C_i|_B} = \mathbf{J}_{B_i} \cdot \begin{bmatrix} {}^B\mathbf{v}_{p|_B} \\ {}^B\boldsymbol{\omega}_{p|_B} \end{bmatrix} \quad (56)$$

where the jacobian  $\mathbf{J}_{B_i}$  is given by:

$$\mathbf{J}_{B_i} = \begin{bmatrix} b_c \cdot \tilde{\mathbf{l}}_i^T \cdot \tilde{\mathbf{l}}_i & b_c \cdot \tilde{\mathbf{l}}_i^T \cdot \tilde{\mathbf{l}}_i^P \tilde{\mathbf{p}}_{i|_B}^T \end{bmatrix} \quad (57)$$

The total linear momentum of the cylinder  $i$ , expressed in frame {B} will be

$$\mathbf{Q}_{C_i(tot)}|_B = m_C \cdot {}^W \dot{\mathbf{v}}_{C_i}|_B \quad (58)$$

$m_C$  being the cylinder mass.

Taking the time derivative of the previous equation results in

$$\dot{\mathbf{Q}}_{C_i(tot)}|_B = m_C \cdot {}^W \ddot{\mathbf{v}}_{C_i}|_B \quad (59)$$

where,

$${}^W \dot{\mathbf{v}}_{C_i}|_B = {}^W \dot{\mathbf{v}}_B|_B + {}^B \dot{\mathbf{v}}_{C_i}|_B + 2 {}^W \boldsymbol{\omega}_B|_B \times {}^B \mathbf{v}_{C_i}|_B + {}^W \dot{\boldsymbol{\omega}}_B|_B \times {}^B \mathbf{p}_{C_i}|_B + {}^W \boldsymbol{\omega}_B|_B \times ({}^W \boldsymbol{\omega}_B|_B \times {}^B \mathbf{p}_{C_i}|_B) \quad (60)$$

The inertial part of the total force applied in  $\{C_i\}$ , due to the cylinder translation, expressed in frame  $\{B\}$  will be

$${}^{C_i} \mathbf{F}_{C_i(ine)(tot)}|_B = \dot{\mathbf{Q}}_{C_i(tot)}|_B \quad (61)$$

That is,

$${}^{C_i} \mathbf{F}_{C_i(ine)(tot)}|_B = {}^{C_i} \mathbf{F}_{C_i(ine)(fix)}|_B + {}^{C_i} \mathbf{F}_{C_i(ine)(man)}|_B \quad (62)$$

$${}^{C_i} \mathbf{F}_{C_i(ine)(fix)}|_B = m_C \cdot {}^B \dot{\mathbf{v}}_{C_i}|_B = m_C \cdot \mathbf{J}_{B_i}^T \cdot {}^B \dot{\mathbf{x}}_P|_B + m_C \cdot \mathbf{J}_{B_i}^T \cdot {}^B \ddot{\mathbf{x}}_P|_B \quad (63)$$

$${}^{C_i} \mathbf{F}_{C_i(ine)(man)}|_B = m_C \cdot ({}^W \dot{\mathbf{v}}_B|_B + 2 {}^W \boldsymbol{\omega}_B|_B \times {}^B \mathbf{v}_{C_i}|_B + {}^W \dot{\boldsymbol{\omega}}_B|_B \times {}^B \mathbf{p}_{C_i}|_B + {}^W \boldsymbol{\omega}_B|_B \times ({}^W \boldsymbol{\omega}_B|_B \times {}^B \mathbf{p}_{C_i}|_B)) \quad (64)$$

where  ${}^{C_i} \mathbf{F}_{C_i(ine)(fix)}|_B$  represents the inertial part of the force, considering the base platform is not moving, and  ${}^{C_i} \mathbf{F}_{C_i(ine)(man)}|_B$  represents the inertial part of the force which results from the base motion.

When equation (61) is pre-multiplied by  $\mathbf{J}_{B_i}^T$ , the inertial component of the generalized force applied to  $\{P\}$ , due to each cylinder translation is obtained in frame  $\{B\}$ :

$${}^P \mathbf{f}_{C_i(ine)(tot)(tra)}|_B = \mathbf{J}_{B_i}^T \cdot {}^{C_i} \mathbf{F}_{C_i(ine)(tot)}|_B \quad (65)$$

The inertial components of the actuating forces will be

$$\boldsymbol{\tau}_{C_i(ine)(fix)(tra)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{C_i(ine)(fix)(tra)}|_B \quad (66)$$

$$\boldsymbol{\tau}_{C_i(ine)(man)(tra)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{C_i(ine)(man)(tra)}|_B \quad (67)$$

On the other hand, the total angular momentum of the cylinder about its centre of mass and expressed in frame  $\{B\}$  will be:

$$\mathbf{H}_{C_i(rot)|_B} = \mathbf{I}_{C_i(rot)|_B} \cdot {}^W\boldsymbol{\omega}_{C_i|_B} \quad (68)$$

Taking the time derivative of the previous equation results in

$$\dot{\mathbf{H}}_{C_i(rot)|_B} = \dot{\mathbf{I}}_{C_i(rot)|_B} \cdot {}^W\boldsymbol{\omega}_{C_i|_B} + \mathbf{I}_{C_i(rot)|_B} \cdot {}^W\dot{\boldsymbol{\omega}}_{C_i|_B} \quad (69)$$

where,

$${}^W\boldsymbol{\omega}_{C_i|_B} = {}^B\boldsymbol{\omega}_{C_i|_B} + {}^W\boldsymbol{\omega}_{B|_B} \quad (70)$$

$${}^W\dot{\boldsymbol{\omega}}_{C_i|_B} = {}^B\dot{\boldsymbol{\omega}}_{C_i|_B} + {}^W\boldsymbol{\omega}_{B|_B} \times {}^B\boldsymbol{\omega}_{C_i|_B} + {}^W\dot{\boldsymbol{\omega}}_{B|_B} \quad (71)$$

Considering that  $\mathbf{I}_{C_i(rot)|_{C_i}}$  is the inertia constant matrix of the rotating cylinder  $i$ , expressed in the frame fixed to the cylinder itself,  $\{C_i\} \equiv \{\mathbf{x}_{C_i}, \mathbf{y}_{C_i}, \mathbf{z}_{C_i}\}$ , then

$$\mathbf{I}_{C_i(rot)|_B} = {}^B\mathbf{R}_{C_i} \cdot \mathbf{I}_{C_i(rot)|_{C_i}} \cdot {}^B\mathbf{R}_{C_i}^T \quad (72)$$

where  ${}^B\mathbf{R}_{C_i}$  is the orientation matrix of each cylinder frame,  $\{C_i\}$ , relative to the base frame,  $\{B\}$ .

Cylinder frames were chosen in the following way: axis  $\mathbf{x}_{C_i}$  coincides with the leg axis and points towards  $P_i$ ; axis  $\mathbf{y}_{C_i}$  is perpendicular to  $\mathbf{x}_{C_i}$  and always parallel to the base plane, this condition being possible given the existence of a universal joint at point  $B_i$ , that negates any rotation along its own axis; axis  $\mathbf{z}_{C_i}$  completes the referential following the right hand rule, and its projection along axis  $\mathbf{z}_B$  is always positive. Thus, matrix  ${}^B\mathbf{R}_{C_i}$  becomes:

$${}^B\mathbf{R}_{C_i} = [\mathbf{x}_{C_i} \quad \mathbf{y}_{C_i} \quad \mathbf{z}_{C_i}] \quad (73)$$

where

$$\mathbf{x}_{C_i} = \hat{\mathbf{l}}_i \quad (74)$$

$$\mathbf{y}_{C_i} = \left[ -\frac{l_{iy}}{\sqrt{l_{ix}^2 + l_{iy}^2}} \quad \frac{l_{ix}}{\sqrt{l_{ix}^2 + l_{iy}^2}} \quad 0 \right]^T \quad (75)$$

$$\mathbf{z}_{C_i} = \mathbf{x}_{C_i} \times \mathbf{y}_{C_i} \quad (76)$$

So, the inertia matrices of the cylinders can be written as:

$$\mathbf{I}_{C_i(rot)|_{C_i}} = \text{diag}([I_{C_{xx}} \quad I_{C_{yy}} \quad I_{C_{zz}}]) \quad (77)$$

where  $I_{C_{xx}}$ ,  $I_{C_{yy}}$  and  $I_{C_{zz}}$  are the cylinder moments of inertia expressed in its own frame.

The inertial part of the total moment applied in  $\{C_i\}$ , due to the cylinder rotation, expressed in frame  $\{B\}$  will be

$${}^{C_i}\mathbf{M}_{C_i(ine)(tot)}|_B = \dot{\mathbf{H}}_{C_i(tot)}|_B \quad (78)$$

that is,

$${}^{C_i}\mathbf{M}_{C_i(ine)(tot)}|_B = {}^{C_i}\mathbf{M}_{C_i(ine)(fix)}|_B + {}^{C_i}\mathbf{M}_{C_i(ine)(man)}|_B \quad (79)$$

where,

$${}^{C_i}\mathbf{M}_{C_i(ine)(fix)}|_B = \frac{d}{dt} \left( \mathbf{I}_{C_i(rot)}|_B \cdot \mathbf{J}_{D_i}^B \dot{\mathbf{x}}_P|_B \right) + \mathbf{I}_{C_i(rot)}|_B \cdot \mathbf{J}_{D_i}^B \ddot{\mathbf{x}}_P|_B \quad (80)$$

$${}^{C_i}\mathbf{M}_{C_i(ine)(man)}|_B = \dot{\mathbf{I}}_{C_i(rot)}|_B \cdot {}^W\boldsymbol{\omega}_B|_B + \mathbf{I}_{C_i(rot)}|_B \cdot \left( {}^W\dot{\boldsymbol{\omega}}_B|_B + {}^W\boldsymbol{\omega}_B|_B \times {}^B\boldsymbol{\omega}_{C_i}|_B \right) \quad (81)$$

${}^{C_i}\mathbf{M}_{C_i(ine)(fix)}|_B$  represents the inertial part of the moment, considering the base platform is not moving, and  ${}^{C_i}\mathbf{M}_{C_i(ine)(man)}|_B$  represents the inertial part of the moment which results from the base motion.

When equation (78) is pre-multiplied by  $\mathbf{J}_{D_i}^T$ , the inertial component of the generalized force applied to  $\{P\}$ , due to each cylinder rotation is obtained in frame  $\{B\}$ :

$${}^P\mathbf{f}_{C_i(ine)(tot)(rot)}|_B = \mathbf{J}_{D_i}^T \cdot {}^{C_i}\mathbf{M}_{C_i(ine)(tot)}|_B \quad (82)$$

The inertial components of the actuating forces will be

$$\boldsymbol{\tau}_{C_i(ine)(fix)(rot)} = \mathbf{J}_C^{-T} \cdot {}^P\mathbf{f}_{C_i(ine)(fix)(rot)}|_B \quad (83)$$

$$\boldsymbol{\tau}_{C_i(ine)(man)(rot)} = \mathbf{J}_C^{-T} \cdot {}^P\mathbf{f}_{C_i(ine)(man)(rot)}|_B \quad (84)$$

Now, with reference to the gravitational part of the generalized force, if the base platform orientation changes, then the force applied in  $\{P\}$  and expressed in  $\{B\}$  results in

$${}^P\mathbf{f}_{C_i(gra)}|_B = {}^W\mathfrak{R}_B^T \cdot {}^P\mathbf{f}_{C_i(gra)}|_W \quad (85)$$

where  ${}^P\mathbf{f}_{C_i(gra)}|_W$  is the gravitational generalized force applied in  $\{P\}$  and expressed in  $\{W\}$ .

This force can be computed using the model that considers both a non-moving base platform, frame  $\{B\}$  parallel to  $\{W\}$ , and  $\mathbf{z}_B \equiv -\hat{\mathbf{g}}$ , i.e.,

$${}^P\mathbf{f}_{C_i(gra)}|_W = \frac{\partial_{C_i} \left( {}^B\mathbf{x}_P|_{B|E} \right)}{\partial {}^B\mathbf{x}_P|_{B|E}} \quad (86)$$

$P_{C_i}$  being the cylinder potential energy. Using equation (46) it can be computed by:

$$P_{C_i} = m_C \cdot g \cdot \frac{b_C}{l_i} \left( z_P + {}^P p_{i|Bz} \right) \quad (87)$$

The gravitational component of the actuating forces due to each cylinder,  $\tau_{C_i(gra)}$ , is

$$\tau_{C_i(gra)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{C_i(gra)|_B} \quad (88)$$

### 3.3. Piston Modelling

Position of the piston  $i$ , relative to  $\{W\}$  and expressed in  $\{W\}$ , may be computed using the following equation:

$${}^W \mathbf{p}_{S_i|_W} = {}^W \mathbf{p}_{B|_W} + {}^B \mathbf{p}_{S_i|_W} \quad (89)$$

The linear velocity of the piston, relative to  $\{W\}$  and expressed in  $\{W\}$ , may be obtained taking the time derivative of the previous equation, that is,

$${}^W \mathbf{v}_{S_i|_W} = {}^W \mathbf{v}_{B|_W} + {}^B \mathbf{v}_{S_i|_W} + {}^W \boldsymbol{\omega}_{B|_W} \times {}^B \mathbf{p}_{S_i|_W} \quad (90)$$

Considering that frame  $\{S_i\}$  is attached to the piston  $i$  and positioned at its centre of mass, then  ${}^B \mathbf{v}_{S_i|_W}$  is the linear velocity of frame  $\{S_i\}$  as seen by an observer fixed in  $\{B\}$ , and  ${}^B \mathbf{p}_{S_i|_W}$  represents the position of  $\{S_i\}$  relative to  $\{B\}$  and expressed in  $\{W\}$ . In frame  $\{B\}$  the following equation can be written:

$${}^W \mathbf{v}_{S_i|_B} = {}^W \mathbf{v}_{B|_B} + {}^B \mathbf{v}_{S_i|_B} + {}^W \boldsymbol{\omega}_{B|_B} \times {}^B \mathbf{p}_{S_i|_B} \quad (91)$$

If the centre of mass of each piston is located at a constant distance,  $b_s$ , from the piston to moving platform connecting point,  $P_i$ , (Figure 4), then its position relative to frame  $\{B\}$  is:

$${}^B \mathbf{p}_{S_i|_B} = -b_s \cdot \hat{\mathbf{l}}_i + {}^B \mathbf{p}_{i|_B} + {}^B \mathbf{x}_{P(pos)|_B} \quad (92)$$

The linear velocity of the piston centre of mass,  ${}^B \dot{\mathbf{p}}_{S_i|_B}$ , relative to  $\{B\}$  and expressed in the same frame, may be computed as:

$${}^B \dot{\mathbf{p}}_{S_i|_B} = \dot{\mathbf{l}}_i + {}^B \boldsymbol{\omega}_{l_i|_B} \times (-b_s \cdot \hat{\mathbf{l}}_i) \quad (93)$$

$${}^B\dot{\mathbf{p}}_{S_i|B} = \mathbf{J}_{G_i} \cdot \begin{bmatrix} {}^B\mathbf{v}_P|B \\ {}^B\boldsymbol{\omega}_P|B \end{bmatrix} \tag{94}$$

where the jacobian  $\mathbf{J}_{G_i}$  is given by:

$$\mathbf{J}_{G_i} = \begin{bmatrix} \mathbf{I} - b_s \cdot \tilde{\mathbf{l}}^T \cdot \tilde{\mathbf{l}} & \left( \mathbf{I} - b_s \cdot \tilde{\mathbf{l}}^T \cdot \tilde{\mathbf{l}} \right) \cdot {}^P\tilde{\mathbf{p}}_i^T|B \end{bmatrix} \tag{95}$$

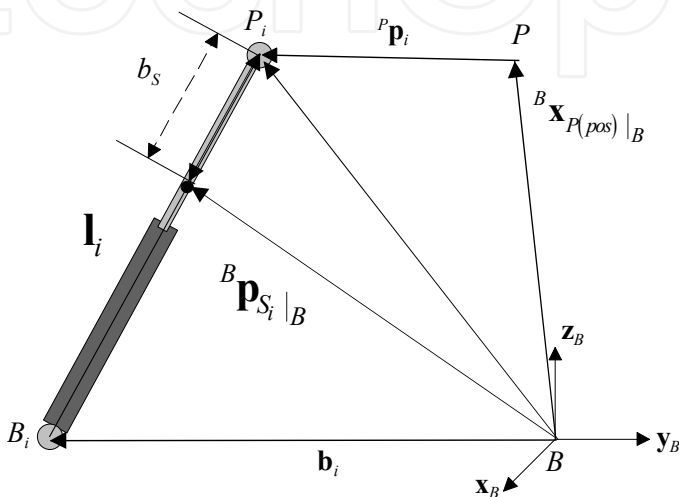


Fig. 4. Position of the centre of mass of the piston  $i$

The total linear momentum of the piston  $i$ , expressed in frame {B} will be

$$\mathbf{Q}_{S_i(tot)|B} = m_s \cdot {}^W\mathbf{v}_{S_i|B} \tag{96}$$

$m_s$  being the piston mass.  
Taking the time derivative of the previous equation results in

$$\dot{\mathbf{Q}}_{S_i(tot)|B} = m_s \cdot {}^W\dot{\mathbf{v}}_{S_i|B} \tag{97}$$

where,

$${}^W\dot{\mathbf{v}}_{S_i|B} = {}^W\dot{\mathbf{v}}_{B|B} + {}^B\dot{\mathbf{v}}_{S_i|B} + 2 {}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{v}_{S_i|B} + {}^W\dot{\boldsymbol{\omega}}_{B|B} \times {}^B\mathbf{p}_{S_i|B} + {}^W\boldsymbol{\omega}_{B|B} \times \left( {}^W\boldsymbol{\omega}_{B|B} \times {}^B\mathbf{p}_{S_i|B} \right) \tag{98}$$

The inertial part of the total force applied in  $\{S_i\}$ , due to the piston translation, expressed in frame {B} will be

$${}^{S_i}\mathbf{F}_{S_i(ine)(tot)|B} = \dot{\mathbf{Q}}_{S_i(tot)|B} \tag{99}$$

That is,

$${}^{S_i}\mathbf{F}_{S_i(ine)(tot)|B} = {}^{S_i}\mathbf{F}_{S_i(ine)(fix)|B} + {}^{S_i}\mathbf{F}_{S_i(ine)(man)|B} \tag{100}$$



$${}^{S_i} \mathbf{F}_{S_i(ine)(fix) | B} = m_S \cdot {}^B \dot{\mathbf{v}}_{S_i | B} = m_S \cdot \mathbf{J}_{G_i} \cdot {}^B \dot{\mathbf{x}}_P | B + m_S \cdot \mathbf{J}_{G_i} \cdot {}^B \ddot{\mathbf{x}}_P | B \quad (101)$$

$${}^{S_i} \mathbf{F}_{S_i(ine)(man) | B} = m_S \cdot \left( {}^W \dot{\mathbf{v}}_B | B + 2 {}^W \boldsymbol{\omega}_B | B \times {}^B \mathbf{v}_{S_i | B} + {}^W \dot{\boldsymbol{\omega}}_B | B \times {}^B \mathbf{p}_{S_i | B} + {}^W \boldsymbol{\omega}_B | B \times \left( {}^W \boldsymbol{\omega}_B | B \times {}^B \mathbf{p}_{S_i | B} \right) \right) \quad (102)$$

where  ${}^{S_i} \mathbf{F}_{S_i(ine)(fix) | B}$  represents the inertial part of the force, considering the base platform is not moving, and  ${}^{S_i} \mathbf{F}_{S_i(ine)(man) | B}$  represents the inertial part of the force which results from the base motion.

When equation (99) is pre-multiplied by  $\mathbf{J}_{G_i}^T$ , the inertial component of the generalized force applied to {P}, due to each piston translation is obtained in frame {B}:

$${}^P \mathbf{f}_{S_i(ine)(tot)(tra) | B} = \mathbf{J}_{G_i}^T \cdot {}^{S_i} \mathbf{F}_{S_i(ine)(tra) | B} \quad (103)$$

The inertial components of the actuating forces will be

$$\boldsymbol{\tau}_{S_i(ine)(fix)(tra)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{S_i(ine)(fix)(tra) | B} \quad (104)$$

$$\boldsymbol{\tau}_{S_i(ine)(man)(tra)} = \mathbf{J}_C^{-T} \cdot {}^P \mathbf{f}_{S_i(ine)(man)(tra) | B} \quad (105)$$

On the other hand, the total angular momentum of the piston about its centre of mass, expressed in frame {B}, will be:

$$\mathbf{H}_{S_i(rot) | B} = \mathbf{I}_{S_i(rot) | B} \cdot {}^W \boldsymbol{\omega}_{S_i | B} \quad (106)$$

Taking the time derivative of the previous equation results in

$$\dot{\mathbf{H}}_{S_i(rot) | B} = \dot{\mathbf{I}}_{S_i(rot) | B} \cdot {}^W \boldsymbol{\omega}_{S_i | B} + \mathbf{I}_{S_i(rot) | B} \cdot {}^W \dot{\boldsymbol{\omega}}_{S_i | B} \quad (107)$$

where,

$${}^W \boldsymbol{\omega}_{S_i | B} = {}^B \boldsymbol{\omega}_{S_i | B} + {}^W \boldsymbol{\omega}_B | B \quad (108)$$

$${}^W \dot{\boldsymbol{\omega}}_{S_i | B} = {}^B \dot{\boldsymbol{\omega}}_{S_i | B} + {}^W \boldsymbol{\omega}_B | B \times {}^B \boldsymbol{\omega}_{S_i | B} + {}^W \dot{\boldsymbol{\omega}}_B | B \quad (109)$$

Considering that  $\mathbf{I}_{S_i(rot) | S_i}$  is the inertia constant matrix of the rotating piston  $i$ , expressed in the frame fixed to the piston itself,  $\{S_i\}$ , then

$$\mathbf{I}_{S_i(rot) | B} = {}^B \mathbf{R}_{S_i} \cdot \mathbf{I}_{S_i(rot) | S_i} \cdot {}^B \mathbf{R}_{S_i}^T \quad (110)$$

where  ${}^B \mathbf{R}_{S_i}$  is the orientation matrix of each piston frame,  $\{S_i\}$ , relative to the base frame, {B}.

As the relative motion between cylinder and piston is a pure translation,  $\{S_i\}$  can be chosen parallel to  $\{C_i\}$ . Therefore,  ${}^B\mathbf{R}_{S_i} = {}^B\mathbf{R}_{C_i}$ .

So, the inertia matrices of the pistons can be written as:

$$\mathbf{I}_{S_i(\text{rot})|_{S_i}} = \text{diag}([I_{S_{xx}} \quad I_{S_{yy}} \quad I_{S_{zz}}]) \quad (111)$$

where  $I_{S_{xx}}$ ,  $I_{S_{yy}}$  and  $I_{S_{zz}}$  are the piston moments of inertia expressed in its own frame.

The inertial part of the total moment applied in  $\{S_i\}$ , due to the piston rotation, expressed in frame  $\{B\}$  will be

$${}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{tot})|_B} = \dot{\mathbf{H}}_{S_i(\text{tot})|_B} \quad (112)$$

that is,

$${}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{tot})|_B} = {}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{fix})|_B} + {}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{man})|_B} \quad (113)$$

where,

$${}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{fix})|_B} = \frac{d}{dt} \left( \mathbf{I}_{S_i(\text{rot})|_B} \cdot \mathbf{J}_{D_i} \right)^B \dot{\mathbf{x}}_P|_B + \mathbf{I}_{S_i(\text{rot})|_B} \cdot \mathbf{J}_{D_i} \cdot {}^B\ddot{\mathbf{x}}_P|_B \quad (114)$$

$${}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{man})|_B} = \dot{\mathbf{I}}_{S_i(\text{rot})|_B} \cdot {}^W\boldsymbol{\omega}_B|_B + \mathbf{I}_{S_i(\text{rot})|_B} \cdot \left( {}^W\dot{\boldsymbol{\omega}}_B|_B + {}^W\boldsymbol{\omega}_B|_B \times {}^B\boldsymbol{\omega}_{S_i}|_B \right) \quad (115)$$

${}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{fix})|_B}$  represents the inertial part of the moment, considering the base platform is not moving, and  ${}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{man})|_B}$  represents the inertial part of the moment which results from the base motion.

When equation (112) is pre-multiplied by  $\mathbf{J}_{D_i}^T$ , the inertial component of the generalized force applied to  $\{P\}$ , due to each piston rotation is obtained in frame  $\{B\}$ :

$${}^P\mathbf{f}_{S_i(\text{ine})(\text{tot})(\text{rot})|_B} = \mathbf{J}_{D_i}^T \cdot {}^{S_i}\mathbf{M}_{S_i(\text{ine})(\text{tot})|_B} \quad (116)$$

The inertial components of the actuating forces will be

$$\boldsymbol{\tau}_{S_i(\text{ine})(\text{fix})(\text{rot})} = \mathbf{J}_C^{-T} \cdot {}^P\mathbf{f}_{S_i(\text{ine})(\text{fix})(\text{rot})|_B} \quad (117)$$

$$\boldsymbol{\tau}_{S_i(\text{ine})(\text{man})(\text{rot})} = \mathbf{J}_C^{-T} \cdot {}^P\mathbf{f}_{S_i(\text{ine})(\text{man})(\text{rot})|_B} \quad (118)$$

Now, with reference to the gravitational part of the generalized force, if the base platform orientation changes, then the force applied in  $\{P\}$  and expressed in  $\{B\}$  results in

$${}^P\mathbf{f}_{S_i(\text{gra})|_B} = {}^W\mathfrak{R}_B^T \cdot {}^P\mathbf{f}_{S_i(\text{gra})|_W} \quad (119)$$

where  ${}^P\mathbf{f}_{S_i(gra)|_W}$  is the gravitational generalized force applied in {P} and expressed in {W}. This force can be computed using the model that considers both a non-moving base platform, frame {B} parallel to {W}, and  $\mathbf{z}_B \equiv -\hat{\mathbf{g}}$ , i.e.,

$${}^P\mathbf{f}_{S_i(gra)|_W} = \frac{\partial P_{S_i}({}^B\mathbf{x}_P|_{B|E})}{\partial {}^B\mathbf{x}_P|_{B|E}} \quad (120)$$

$P_{S_i}$  being the piston potential energy. Using equation (92) it can be computed by:

$$P_{S_i} = m_s \cdot g \cdot \left(1 - \frac{b_s}{l_i}\right) \cdot (z_P + {}^Pp_i|_{Bz}) \quad (121)$$

The gravitational component of the actuating forces due to each piston,  $\boldsymbol{\tau}_{S_i(gra)}$ , is

$$\boldsymbol{\tau}_{S_i(gra)} = \mathbf{J}_C^T \cdot {}^P\mathbf{f}_{S_i(gra)|_B} \quad (122)$$

It should be noted that the base platform motion originates new inertial contributions to the parallel manipulator dynamic model, expressed by equations (38), (67), (84), (105) and (118). These contributions should be added to the corresponding ones resulting from the model that considers a fixed-base: equations (37), (66), (83), (104) and (117). Regarding the gravitational part of the dynamic model, the base platform motion modifies the gravitational force components, resulting in the equations (42), (88) and (122), which can also be added to the previous ones.

Computational efficiency of the proposed model has been evaluated by counting the number of scalar operations needed in the calculations (sums, multiplications and divisions). For this purpose, the Maple® software package was used. The results were then compared with the ones obtained by using the Lagrange formulation. The generalized momentum approach resulted in a much more efficient dynamic model. Regarding the total number of sums and multiplications involved in the two models, the ratio is five, approximately. This might be a great advantage if real-time simulation and control is needed.

## 4. Conclusion

A parallel manipulator is a complex multi-body dynamic system having several closed loops. Typically, it is composed of a (usually) fixed base platform and a moving payload platform, connected by at least two independent open kinematic chains. Dynamic modelling of parallel manipulators presents an inherent difficulty. Despite the intensive study in this topic of robotics, mostly conducted in the last two decades, additional research still has to be done in this area.

In this book chapter an approach based on the manipulator generalized momentum was explored and applied to the dynamic modelling of a Stewart platform manipulator. The system dynamic equations were obtained for the general case of a non-fixed base platform. It was shown the base platform motion originates new inertial force contributions to the parallel manipulator dynamic model. Compact analytical expressions for these contributions were presented and it was shown they can be easily added to the corresponding ones resulting from the model that considers a fixed base. On the other hand, regarding the gravitational part of the dynamic model, the base platform motion modifies the gravitational force components derived when considering a fixed base platform. Analytical expressions for these components were also presented. Computational efficiency of the dynamic model was evaluated by counting the number of scalar operations that are needed. The results were then compared with the ones obtained by using the Lagrange formulation. The generalized momentum approach results in a much more efficient dynamic model.

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The purpose of this volume is to encourage and inspire the continual invention of robot manipulators for science and the good of humanity. The concepts of artificial intelligence combined with the engineering and technology of feedback control, have great potential for new, useful and exciting machines. The concept of eclecticism for the design, development, simulation and implementation of a real time controller for an intelligent, vision guided robots is now being explored. The dream of an eclectic perceptual, creative controller that can select its own tasks and perform autonomous operations with reliability and dependability is starting to evolve. We have not yet reached this stage but a careful study of the contents will start one on the exciting journey that could lead to many inventions and successful solutions.

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51000 Rijeka, Croatia  
Phone: +385 (51) 770 447  
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### **InTech China**

Unit 405, Office Block, Hotel Equatorial Shanghai  
No.65, Yan An Road (West), Shanghai, 200040, China  
中国上海市延安西路65号上海国际贵都大饭店办公楼405单元  
Phone: +86-21-62489820  
Fax: +86-21-62489821

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