

# 'Correlating and Predicting the Air Infiltration Through the Cracks of Suspended Timber Floors

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This paper shows that the visible gap length may be used as a physical parameter when correlating pressure difference to volume flow rate (or air velocity) through the cracks between floorboards and can be termed the *Equivalent Crack Length*. The crack widths, which previously have been determined separately from the edge effect coefficient and the laminar flow coefficient, were significantly different. When analysed graphically they can be shown to correlate by an empirical relationship. A generalised equation is proposed that, in conjunction with the empirical relationship, allows predictions to be made of the volume flow rate through the cracks between floorboards for a known pressure difference.

## Symbols

Q = volume flow rate,

$\Delta P$  = pressure difference

n = exponent

A & B =constants that relate to crack parameters

C<sub>2</sub> is a constant, found by regression analysis,

$\mu$ = dynamic viscosity

z = total distance through the crack

L= length of the crack as measured on the horizontal plane of the floor

d= crack width or thickness

$\rho$ = air density

C<sub>b</sub>= a constant that relates to the number of bends in the crack.

w<sub>lam</sub>= equivalent crack length calculated from the Laminar coefficient

w<sub>edge</sub> = equivalent crack length calculated from the Edge coefficient

## 1.0 Introduction

It is established that air infiltration from a basement or sub-floor void is a drain on the heat requirement of a building<sup>(1)</sup> as well as a potential medium for the transportation of ground contaminants<sup>(2,3)</sup>. In previous work<sup>(4)</sup> the authors presented the results of an investigation of three methods for correlating air infiltration through timber floors with pressure difference, namely the power relationship, the linear relationship and the quadratic relationship. It was shown that both the power relationship and the quadratic relationship offer an effective means of describing the flow characteristics associated with air infiltration through timber floorboards for the pressure difference range of 2 – 50 Pascals.

In the case of the quadratic relationship, knowledge of its coefficients allows the physical parameters of the crack geometry to be estimated. However, the determination of the coefficients is likely to be affected by (a) the nature and type of the experiment being undertaken; (b) the availability and spread of data; and (c) the

method of data analysis. Consequently, the parametric values thus deduced are strictly applicable only to the case under investigation. Furthermore, caution must be exercised when using measurements taken at low flow rates, i.e. for pressures differences less than 5 Pa because of the practical constraints. In extreme cases (where the pressure difference is less than 2 Pa), extrapolation of data offers the only means of analysis. Although the quadratic relationship is well established and its coefficients can be related to crack parameter, the following two items need addressing;

Can the quadratic relationship be made to be independent of case studies?

Can the uncertainty associated with measurements at flow rates be minimised?

The main emphasis of the present study is devoted to tackling these two issues and the purpose of this paper is therefore two-fold

to provide more experimental data to enable the search of a universal quadratic relationship;

to show the effectiveness of using an apparent floorboard crack length for parametric calculations.

## **2. The Experiment**

For this study a purpose-built floor has been constructed to investigate the flow characteristics associated with low flow rates through a suspended timber floor. This would enable the crack dimensions to be readily measured, so that comparison could be made with predicted values from mathematical models. The technique used to construct the floor was typical of that used in current U.K. housing stock.

## 2.1 Test rig

In order to exercise greater control over the air flow rate and pressure difference across the chamber floor, a rig of 1.8 m x 1.2 m x 0.60 m (Figure 1) was constructed with a suspended timber floor dividing the rig into two 300 mm deep chambers. The rig was constructed of clear 6 mm perspex held together by 25 mm aluminium and mild steel angles. The angle-section/perspex interface and corners of the rig were sealed with silicone sealant to achieve a near perfect air-tight seal. The chamber lid was made of 1.8 m x 1.2 m x 6 mm perspex that was screwed on to the angle sections and a seal achieved by using a silicone sealant bead as the gasket. The floor was built of 125 mm x 25 mm (nom) tongue and groove boards on 50 mm x 50 mm carcassing timber. Air was supplied to the chamber through 51 mm flanged copper tubes and the flow rate was regulated by a variable speed centrifugal fan (250 mm) and a damper connected to a 1000 mm (*l*) x 51 mm(*i.d.*) copper tube. The mean velocity pressure through the inlet pipe was determined by means of a micromanometer connected to a Pitot-static tube located 1 m downstream of the entry point and 20 mm from the exit point where it was anticipated that the air flow would be reasonably developed. Bulk pressure across the floor was measured by an inclined manometer that was connected to pressure tappings above and below the floor, which enabled static pressure differences as low as 0.5 Pascals to be recorded.

## 2.2 Procedure

The purpose of the investigation was to measure volume flow rates and corresponding pressure differences across the timber floor for a pressure range of 0-50 Pascals

Observations of the flow conditions at the measuring station revealed a flattened velocity profile similar to that pertaining to a turbulent flow. Therefore, it was concluded that the air flow at the measuring station would always be turbulent and fully developed. The volume flow rate was calculated from *the Area of the pipe x Mean Air Velocity* which was found from a single, centre-line velocity pressure reading. Since the air velocity near the pipe wall is lower than that of the mean velocity, the calculated flow rate would slightly differ from the true flow rate. From recorded measurements the difference is estimated to be of the order of 10%.

### 2.3 Background Theory

Previous work<sup>(5)</sup> identified that the flow rate through the floorboards could be calculated by subtracting the leakage ( $Q_{leak1}$ ) from the total flow into the chamber ( $Q_1$ ). Although infiltration through the floorboards is measured directly, the flow between the skirting board and the floorboards is deduced from the difference between the measured flow rate through the *floor* and the measured flow rate through the *floor and skirting*. Measurement of the infiltration through the floorboards is possible by sealing the gap between the skirting and the floor with plastic tape. Background leakage from the chamber was determined by closing and sealing the chamber lid and comparing the inflow with the outflow. The inflow/outflow velocity pressure was measured in a 22 mm pipe and the measured leakage was calculated as less than 1% of the inflow rate in preliminary investigation and was subsequently neglected in other tests. Four test floors were prepared, each with pre-determined visible gap sizes between the floorboards and were separately tested (tests 1-4). All the tests had a skirting board at the edge of the floor that was sealed to the chamber

wall by silicone sealant to prevent unmeasured leakage. This arrangement simulated the typical construction detail of a wooden floor over a basement or floor-void. The results are analysed as the variation of the volume flow rate (through the floor and from the opening between the floor and skirting board) with pressure difference between the chambers and are subjected to regression analysis using the following relationship:

### 2.3.1 The quadratic equation

$$\Delta P = AQ + BQ^2 + C_2 \quad (1)$$

where  $Q$  = volume flow rate,  $\Delta P$  = pressure difference,  $A$  &  $B$  are constants that relate to crack parameters that were defined by Etheridge<sup>(6)</sup> and subsequently extended by Baker<sup>(7)</sup> as Equations 2 and 3.  $C_2$  is a constant, found by regression analysis, that is indicative of instrument error at zero pressure difference. By sealing the inlet tube and setting of the instruments to zero before each set of readings, instrument error was assumed to be negligible and set at zero for the analysis.

$$A = 12\mu z/Ld^3. \quad (\text{Laminar flow coefficient}) \quad (2)$$

$$B = \rho C_b / 2d^2 L^2 \quad (\text{Edge effect coefficient}) \quad (3)$$

where  $\mu$  = dynamic viscosity,  $z$  = total distance through the crack,  $L$  = length of the crack as measured on the horizontal plane of the floor,  $d$  = crack width or thickness,  $\rho$  = air density and  $C_b$  = a constant (described by Baker<sup>(7)</sup>) that relates to the number of bends in the crack.

## 3.0 Results and Discussion

### 3.1 Floorboard Crack Size

The experimental data for this work are plotted (Figures 2 – 5) using the regression technique, based on the volume flow rate - pressure difference quadratic relationship as defined in Equation 1. The results confirm the authors' previous observation<sup>(5)</sup> that the greatest pressure loss within the joint/crack is attributable to edge effects .

Substitution of the regression curve coefficients into Equation 3 enables the calculation of an approximate value of an *equivalent crack width* of the floorboard joint (Table I). Although the term *equivalent crack length* is used to describe the total length of the visible gap between floorboards it does not represent the true crack length because the tongue of one board rests on the lip of the adjoining board.

Inspection of the calculated values (Table I) shows that an equivalent crack width determined by using the laminar flow coefficient can result in vastly different values. However, because the laminar flow predominates at lower pressure differences, it is not wise to disregard it completely. When the calculated equivalent crack widths based on the laminar flow coefficient were plotted against those determined from the edge effect coefficient (Figure 7), a relationship is identified which can be represented by the following expression

$$w_{\text{lam}} = 0.0711(w_{\text{edge}})^{0.2805} \quad (4) \quad \text{where}$$

$w_{\text{lam}}$  = is the equivalent crack length calculated from the laminar flow coefficient

$w_{\text{edge}}$  = is the equivalent crack length calculated from the edge effect coefficient. Combining Equations 3 and 4 and substituting into Equation 1 yields

$$\Delta P = (\rho C_b / 2(w_{\text{edge}})^2 L^2) * Q^2 + \{ 12 \mu z / L [0.0711(w_{\text{edge}})^{0.2805}] \}^3 * Q \quad (5)$$

Thus, the empirical relationship given by Equation 4 allows the infiltration flow rate Q to be predicted based on a knowledge of the equivalent crack width deduced from

the *edge effect coefficient*, which can be calculated from the regression analysis of experimental data. .

The validity of this approach can be quantified as follows. When the pressure difference across the chambers was maintained at 2 Pa, the calculated flow rate ( $Q$ ) was found to be within 3.5% of the values predicted using the regression analysis coefficients for tests 2- 4 This figure increases to 11 % for a full size test floor. It should be noted that the difference between Equation 5 predictions and those from the regression analysis can vary significantly at very low pressure. For example, the difference in predictions for test No.1 (figure 2) can be as high as 54% at 0.4 Pa and decreases logarithmically to 10 % at 50 Pa. The reason for the large discrepancy is unknown and this may be attributable to practical difficulties in measuring data at low pressures.

### 3.2 Normalised results

It is generally accepted that the upper pressure difference for testing air flow through building components should be limited to 50 Pa. . Because of the similarity between each curve, the data can be conveniently normalised with respect to a maximum pressure difference of 50 Pascals and the corresponding volume flow rate as follows; -

$$y = \Delta P_i / \Delta P_{50} \quad (6)$$

$$x = Q_i / Q_{50} \quad (7)$$

Where the subscript ‘i’ denotes a quantity measured at a pressure difference less than 50 Pa; and subscript ‘50’ denotes a quantity measured at 50 Pa.

The normalised results of this work are presented in Figure 8, along with the results of British Gas's work on crack flow<sup>(6)</sup> and the authors' previous work<sup>(5)</sup> for comparison. Presenting the data in a dimensionless form allows any operational bias from various experimental set-up to be minimised. With most of the data following a definite pattern, a generalised quadratic relationship (Equation 9) can be identified which has achieved a coefficient of Determination of 0 .99.

$$y = 0.7635x^2 + 0.2621x \quad (8)$$

or

$$0.7635x^2 + 0.2621x - y = 0$$

By solving the generalised quadratic relationship, theoretical infiltration flow rates can be calculated for a given pressure ratio. The validity of Equation 8 has been assessed by comparing the predicted flow rates with measured flow rates. The comparison of all the results over the full range (0 – 50 Pa) appears encouraging although the error in the 0-3 Pascal pressure difference is higher than expected due to practical difficulties when measuring very low pressure differences. However, analysis of the 3 – 10 Pa range showed that out of thirty-five case studies taken from seventeen tests, thirty-one predictions (88%) are within 15% of the measured flow rates (Figure 9), while twenty predictions are within 10%. Thus, given the floorboard dimensions and an equivalent crack width, this approach allows acceptable estimations to be made of the air infiltration through the gaps between floorboards.

#### **4.0 Conclusions**

The above discussion examines the results of the present investigation , the authors' previous study and the work of Etheridge's. Some interesting points have been highlighted that enables the following conclusions to be made.

A new parameter known as the *equivalent crack length* has been proposed to describe the total length of the visible gap between floorboards..

An empirical relationship has been proposed for determining floorboard equivalent crack width based on edge effect coefficient, that can be used with Equation 5 to calculate air infiltration across a traditional wooden floor.

The correlation between pressure difference and volume flow rate can be normalised onto a single curve which can be used to predict infiltration, within a range of 3-10 Pa, through a traditional wooden floor with better than 85% agreement.

## References

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**TABLE**

Table I Comparison of Calculated Crack widths for Differing Floorboard Configurations					
Gap width (mm)	Edge Coefficient A	Crack width (mm) $w_A$	Laminar Coef.* B	Crack width (mm) $w_B$	Widths Ratio $w_B/w_A$
0	$1.81 \times 10^8$	0.01	6750	1.4	140
1.25	246466	0.21	1605	2.8	13.3
3.5	600324	0.14	1585	2.9	20.7
Random <sup>1</sup>	21819	0.71	397	5.8	8.2
1.75 <sup>2</sup>	4507.1	0.25	109.3	4.2	16.8

<sup>1</sup> The floor was laid with large non-uniform, random size cracks between the boards and 2.5mm is a reasonable approximation of the average gap size

<sup>2</sup> The tests were performed on a full-size test floor that had settled with non-uniform gap widths. An estimation of the average size of the gap width is approximately 1.75mm

\* This is included for comparison with the edge effect coefficient

## LIST OF FIGURES

Fig.1 Diagram of the Test Rig

Fig.2 Pressure Difference versus Flow Rate (Test 1)

Fig.3) Pressure Difference versus Flow Rate (Test 2)

Fig.4 Pressure Difference versus Flow Rate (Test 3)

Fig.5 Pressure Difference versus Flow Rate (Test 4)

Fig.6 Pressure Difference versus Flow Rate (<sup>Ref.5</sup>)

Fig.7 Laminar Coefficient Width versus Edge Effect Coefficient Width

Fig.8 Normalised Pressure Difference versus Normalised Flow Rate

Fig. 9 Magnitude of Error With Pressure Difference

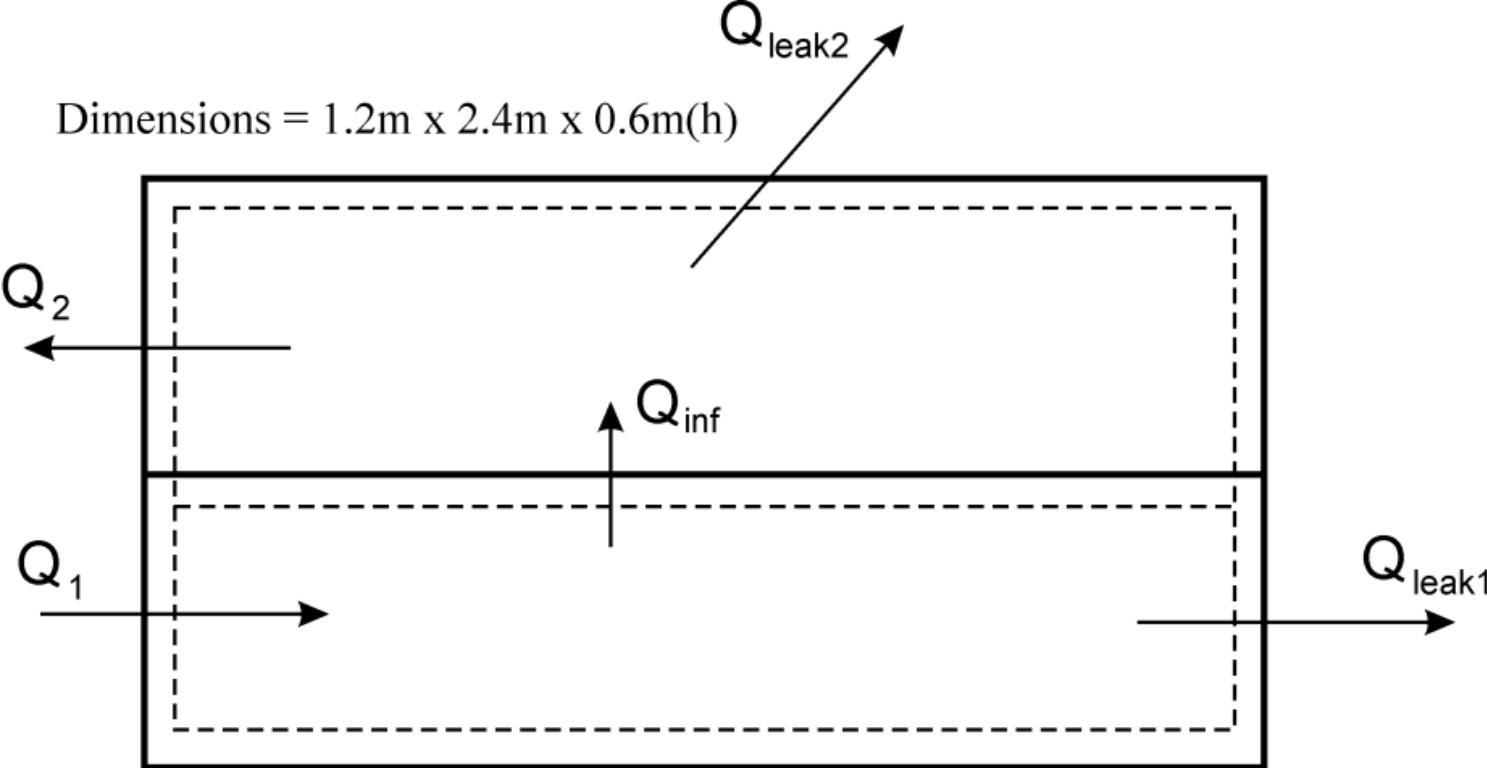
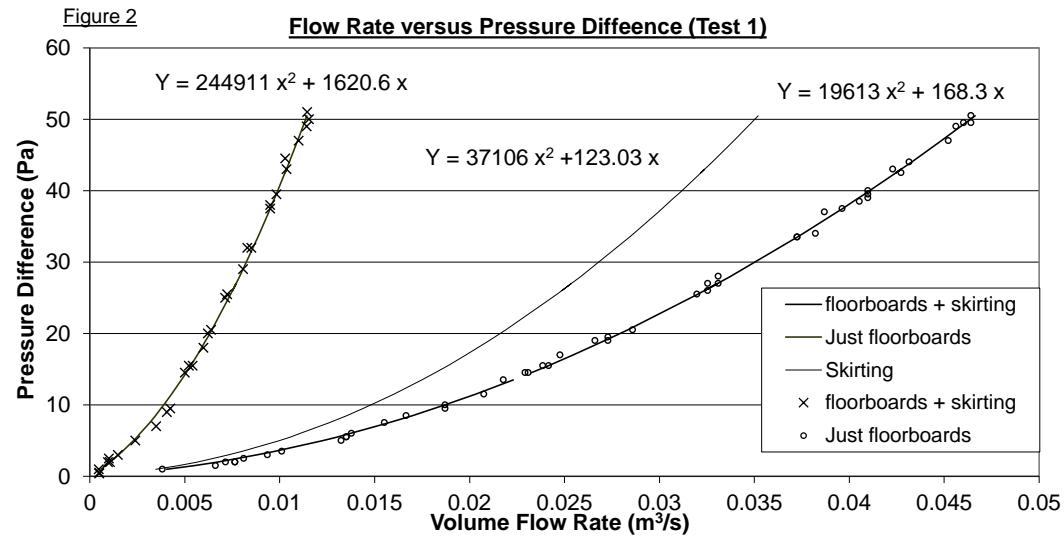


Figure 1 Diagrammatic of Test Chamber

**Fig 2:**



just fbds	1620.6	244911
fbds+skts	168.34	19613
flow rate Pressure Difference		
0.000503	0.5	0.000295
0.001006	2.5	0.001291
0.002394	5	0.002292
0.00406	9	0.003598
0.005218	15.5	0.005307
0.006221	20	0.006315
0.007133	25	0.007323
0.008508	32	0.008591
0.009503	37.5	0.0095
0.010368	43	0.010349
0.011545	50	0.011358
0.000503	0.375	0.000224
0.000954	2	0.001063
0.00148	3	0.001508
0.00349	7	0.002979
0.004233	9.5	0.003744
0.005421	15.5	0.005307
0.006391	20.5	0.00642
		0.002335

	fbds+ sktng	jst fbds	jst sktng
1	0.00403936	0.000568	0.003471
1.5	0.005449979	0.000823	0.004627
2	0.00668072	0.001063	0.005617
2	0.00668072	0.001063	0.005617
2.5	0.007786695	0.001291	0.006496
3	0.008799566	0.001508	0.007292
3.5	0.00973951	0.001715	0.008024
5	0.012241767	0.002292	0.00995
5.5	0.012995551	0.002471	0.010525
5.5	0.012995551	0.002471	0.010525
6	0.013717814	0.002645	0.011073
7.5	0.015728867	0.003139	0.01259
8.5	0.016964125	0.003448	0.013516
9.5	0.018131437	0.003744	0.014388
10	0.018692871	0.003887	0.014806
11.5	0.020300391	0.004301	0.016
13.5	0.022292971	0.00482	0.017473

Pressure D	Flow Rate	floorboards	Just floorbc	Skirting	floorboards	Just floorboards
1	0.004039	1				
1.5	0.00545	1.5				
2	0.006681	2				
2	0.006681	2				
2	0.006681	2				
2.5	0.007787	2.5				
3	0.0088	3				
3.5	0.00974	3.5				
5	0.012242	5				
5.5	0.012996	5.5				
5.5	0.012996	5.5				
6	0.013718	6				
7.5	0.015729	7.5				
8.5	0.016964	8.5				
9.5	0.018131	9.5				
10	0.018693	10				
11.5	0.020203	11.5				
13.5	0.022293	13.5				

0.008076	29	0.008065	0.008076	14.5	0.023235225	0.005067	0.018168	14.5	0.023235	14.5
0.009503	38	0.00958	0.009503	14.5	0.023235225	0.005067	0.018168	14.5	0.023235	14.5
0.011005	47	0.010934	0.011005	15.5	0.024146276	0.005307	0.018839	15.5	0.024146	15.5
0.011421	49	0.011218	0.011421	15.5	0.024146276	0.005307	0.018839	15.5	0.024146	15.5
0.000468	1	0.000568	0.000468	17	0.025460591	0.005656	0.019805	17	0.025461	17
0.001063	2	0.001063	0.001063	19	0.027127602	0.0061	0.021027	19	0.027128	19
0.005008	14.5	0.005067	0.005008	19	0.027127602	0.0061	0.021027	19	0.027128	19
0.005986	18	0.005881	0.005986	19.5	0.027530713	0.006208	0.021323	19.5	0.027531	19.5
0.007233	25.5	0.007418	0.007233	20.5	0.02832199	0.00642	0.021902	20.5	0.028322	20.5
0.008294	32	0.008591	0.008294	25.5	0.032020652	0.007418	0.024602	25.5	0.032021	25.5
0.009836	39.5	0.009815	0.009836	27	0.033058897	0.0077	0.025359	27	0.033059	27
0.010299	44.5	0.010571	0.010299	26	0.032370001	0.007513	0.024857	26	0.03237	26
0.011452	51	0.011496	0.011452	27	0.033058897	0.0077	0.025359	27	0.033059	27
				28	0.033735316	0.007884	0.025851	28	0.033735	28
				33.5	0.037259248	0.008846	0.028413	33.5	0.037259	33.5
				33.5	0.037259248	0.008846	0.028413	33.5	0.037259	33.5
				34	0.037564897	0.00893	0.028635	34	0.037565	34
				37	0.039353861	0.00942	0.029934	37	0.039354	37
				37.5	0.039644941	0.0095	0.030145	37.5	0.039645	37.5
				38.5	0.040221139	0.009659	0.030563	38.5	0.040221	38.5
				39	0.040506833	0.009737	0.03077	39	0.040507	39
				39.5	0.040790469	0.009815	0.030975	39.5	0.04079	39.5
				40	0.041072331	0.009893	0.03118	40	0.041072	40
				43	0.042728037	0.010349	0.032379	43	0.042728	43
				42.5	0.042456158	0.010274	0.032182	42.5	0.042456	42.5
				44	0.043267131	0.010497	0.03277	44	0.043267	44
				47	0.044848942	0.010934	0.033915	47	0.044849	47
				49	0.045875781	0.011218	0.034658	49	0.045876	49
				49.5	0.046129224	0.011288	0.034841	49.5	0.046129	49.5
				49.5	0.046129224	0.011288	0.034841	49.5	0.046129	49.5
				50.5	0.046632325	0.011427	0.035205	50.5	0.046632	50.5
					0.000568				1	
					0.000823				1.5	
					0.001063				2	
					0.001063				2	
					0.001063				2	
					0.001291				2.5	
					0.001508				3	
					0.001715				3.5	
					0.002292				5	
					0.002471				5.5	
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					0.002645				6	
					0.003139				7.5	
					0.003448				8.5	
					0.003744				9.5	
					0.003887				10	
					0.004301				11.5	
					0.00482				13.5	
					0.005067				14.5	
					0.005067				14.5	

0.005307	15.5
0.005307	15.5
0.005656	17
0.0061	19
0.0061	19
0.006208	19.5
0.00642	20.5
0.007418	25.5
0.0077	27
0.007513	26
0.0077	27
0.007884	28
0.008846	33.5
0.008846	33.5
0.00893	34
0.00942	37
0.0095	37.5
0.009659	38.5
0.009737	39
0.009815	39.5
0.009893	40
0.010349	43
0.010274	42.5
0.010497	44
0.010934	47
0.011218	49
0.011288	49.5
0.011288	49.5
0.011427	50.5
0.003471	1
0.004627	1.5
0.005617	2
0.005617	2
0.005617	2
0.006496	2.5
0.007292	3
0.008024	3.5
0.00995	5
0.010525	5.5
0.010525	5.5
0.011073	6
0.01259	7.5
0.013516	8.5
0.014388	9.5
0.014806	10
0.016	11.5
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0.018168	14.5
0.018839	15.5
0.018839	15.5
0.019805	17

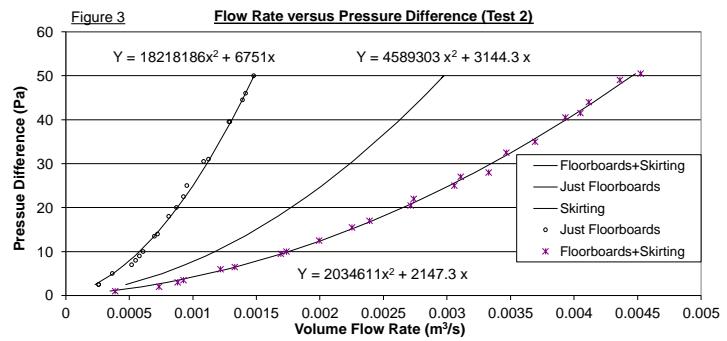
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0.028413	33.5
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0.030145	37.5
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0.030975	39.5
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0.032379	43
0.032182	42.5
0.03277	44
0.033915	47
0.034658	49
0.034841	49.5
0.034841	49.5
0.035205	50.5
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0.001006	2.5
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0.00406	9
0.005218	15.5
0.006221	20
0.007133	25
0.008508	32
0.009503	37.5
0.010368	43
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0.000503	0.375
0.000954	2
0.001448	3
0.00349	7
0.004233	9.5
0.005421	15.5
0.006391	20.5
0.008076	29
0.009503	38
0.011005	47
0.011421	49
0.000468	1
0.001063	2
0.005008	14.5
0.005986	18

0.007233	25.5
0.008294	32
0.009836	39.5
0.010299	44.5
0.011452	51
0.003822	1
0.006621	1.5
0.007151	2
0.007645	2
0.007645	2
0.008108	2.5
0.009363	3
0.010113	3.5
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0.013514	5.5
0.013782	6
0.015527	7.5
0.016661	8.5
0.018726	9.5
0.018726	10
0.020761	11.5
0.021791	13.5
0.022934	14.5
0.023093	14.5
0.023871	15.5
0.024175	15.5
0.024772	17
0.02662	19
0.027297	19
0.027297	19.5
0.028604	20.5
0.03198	25.5
0.032546	27
0.032546	26
0.033103	27
0.033103	28
0.037256	33.5
0.037256	33.5
0.038224	34
0.038699	37
0.039631	37.5
0.040542	38.5
0.04099	39
0.04099	39.5
0.04099	40
0.042306	43
0.042735	42.5
0.043161	44
0.045227	47
0.045629	49
0.046028	49.5

0.046423  
0.046423

49.5  
50.5

**fig 3**



just fbds	6751	18218186
fbds+skts	2147.3	2034611
<b>Flow Rate Pressure C Floorboards Floorboard Skirting</b>		
0.00039	1	0.000113
0.000734	2	0.000194
0.000881	3	0.000261
0.000927	3.5	0.000291
0.001218	6	0.000418
0.00133	6.5	0.00044
0.001693	9.5	0.00056
0.001737	10	0.000578
0.001995	12.5	0.000664
0.002253	15.5	0.000756
0.002391	17	0.000798
0.002714	20.5	0.000892
0.002739	22	0.000929
0.003059	25	0.001001
0.003108	27	0.00104
0.003328	28	0.001068
0.003467	32.5	0.001163
0.003693	35	0.001213
0.003932	40.5	0.001317
0.004045	41.5	0.001335
0.004116	44	0.00138
0.004361	49	0.001465
0.004524	50.5	0.00149
0.00026	2.5	0.000229
0.00026	2.5	0.000229
0.000367	5	0.000307
0.000519	7	0.000462
0.000551	8	0.000503
0.000581	9	0.000542
0.000609	10	0.000578
0.000699	13.5	0.000695
0.000723	14	0.000711
0.000811	18	0.000826
0.000871	20	0.000879
0.000927	22.5	0.000941
0.000954	25	0.001001
0.001086	30.5	0.001122
0.001124	31	0.001132
0.001285	39.5	0.001299
0.001292	39.5	0.001299
0.001392	44.5	0.001389
0.001416	46	0.001414
0.001418	50	0.001482
0.001509	49.5	0.001473

$(-\$V\$1+(POWER((\$V\$1*\$V\$1)+(4*\$W\$1*Z10)),0.5))/2*\$W\$1)$

Flow Rate Floorboard Just Floort Skirting Just Floot Floorboards+Skirting

0.00035 1

0.000595 2

0.000796 3

0.000886 3.5

0.001269 6

0.001336 6.5

0.001697 9.5

0.001751 10

0.002007 12.5

0.002282 15.5

0.002411 17

0.00269 20.5

0.002803 22

0.003017 25

0.003153 27

0.003454 28

0.003504 32.5

0.003653 35

0.003965 40.5

0.004019 41.5

0.004153 44

0.004408 49

0.004482 50.5

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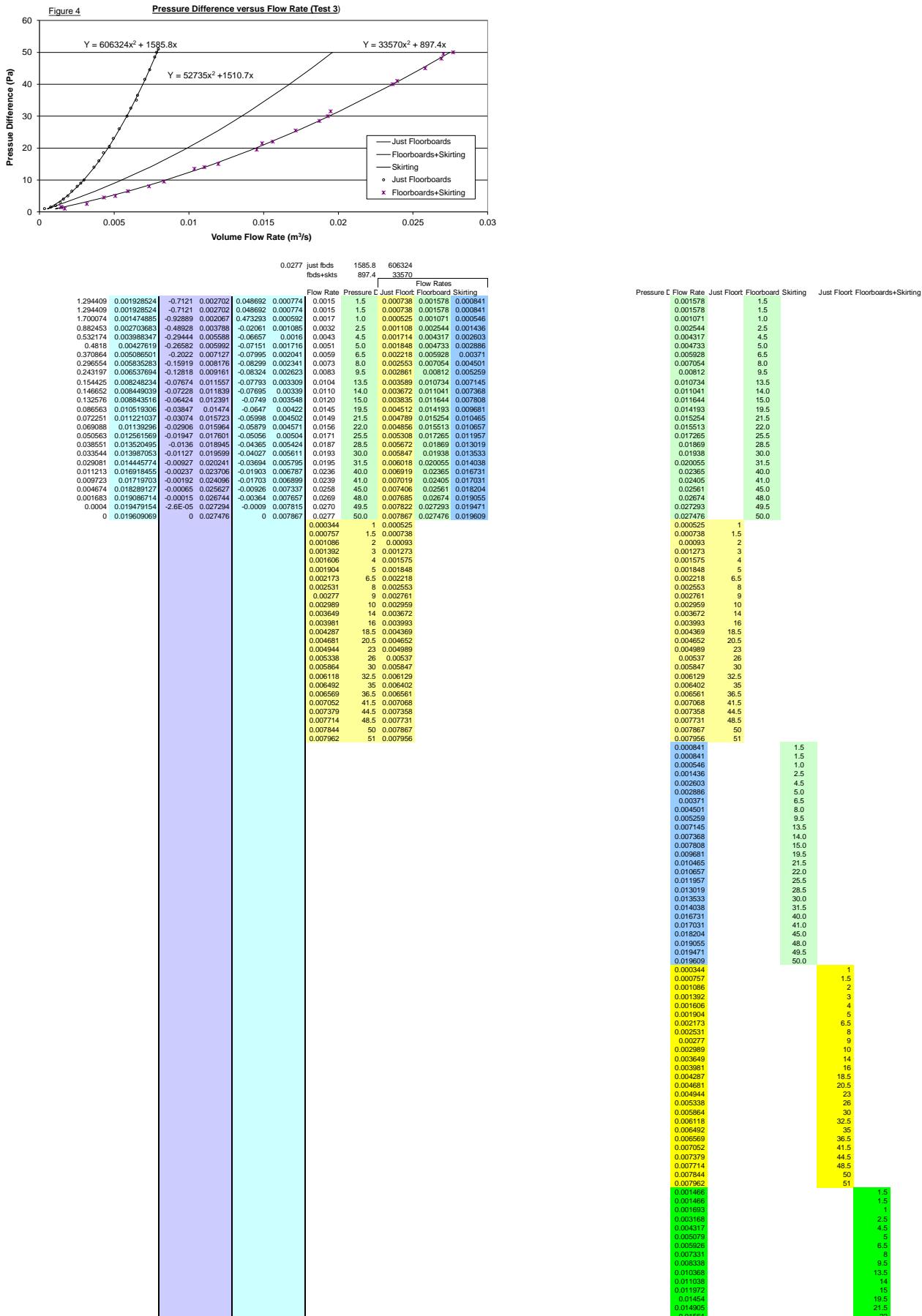
0.004482 50.5

0.004482 50.5

0.004482 50.5

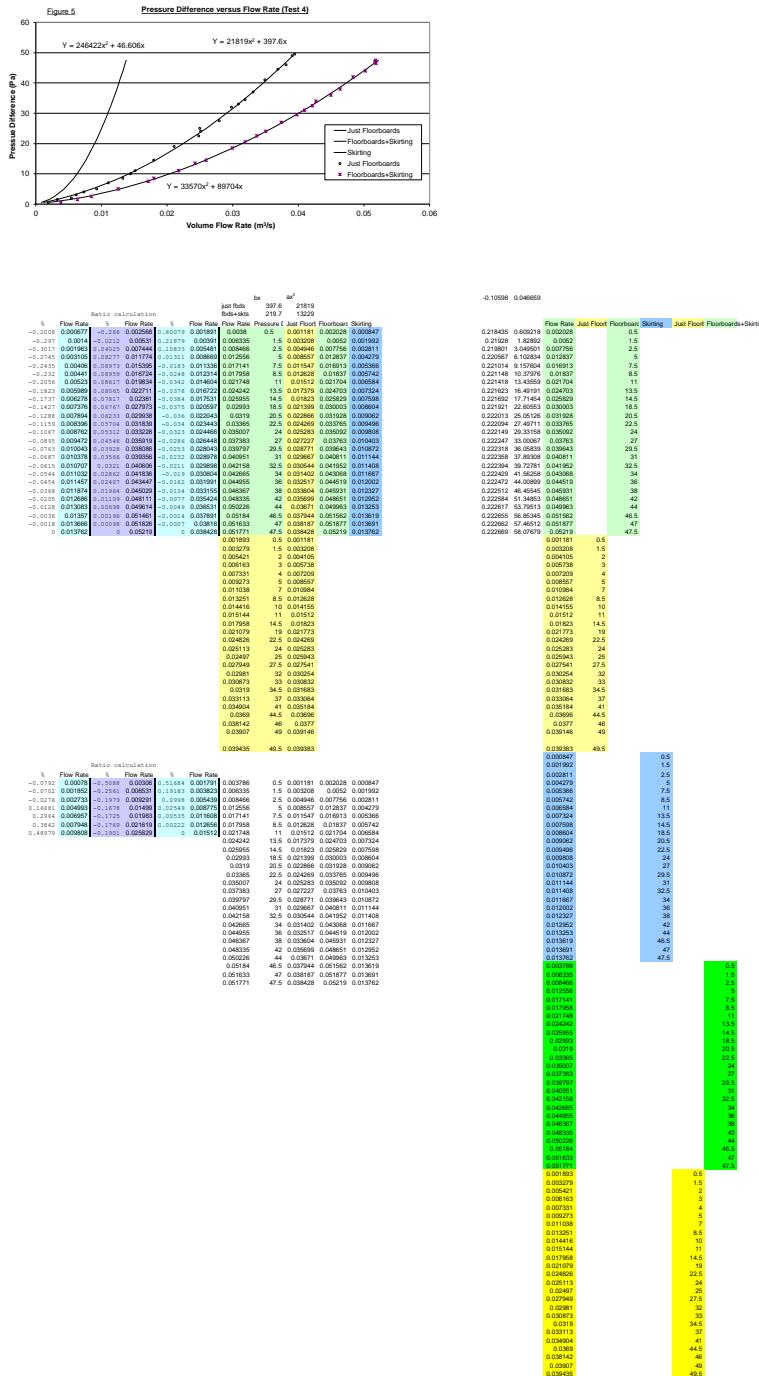
0.00148	50
0.00039	1
0.00734	2
0.00881	3
0.00927	3.5
0.001218	6
0.00133	6.5
0.001693	9.5
0.001737	10
0.001995	12.5
0.002253	15.5
0.002391	17
0.002714	20.5
0.002739	22
0.003059	25
0.003108	27
0.003328	28
0.003467	32.5
0.003693	35
0.003932	40.5
0.00405	41.5
0.004116	44
0.004361	49
0.004524	50.5

**Fig 4**





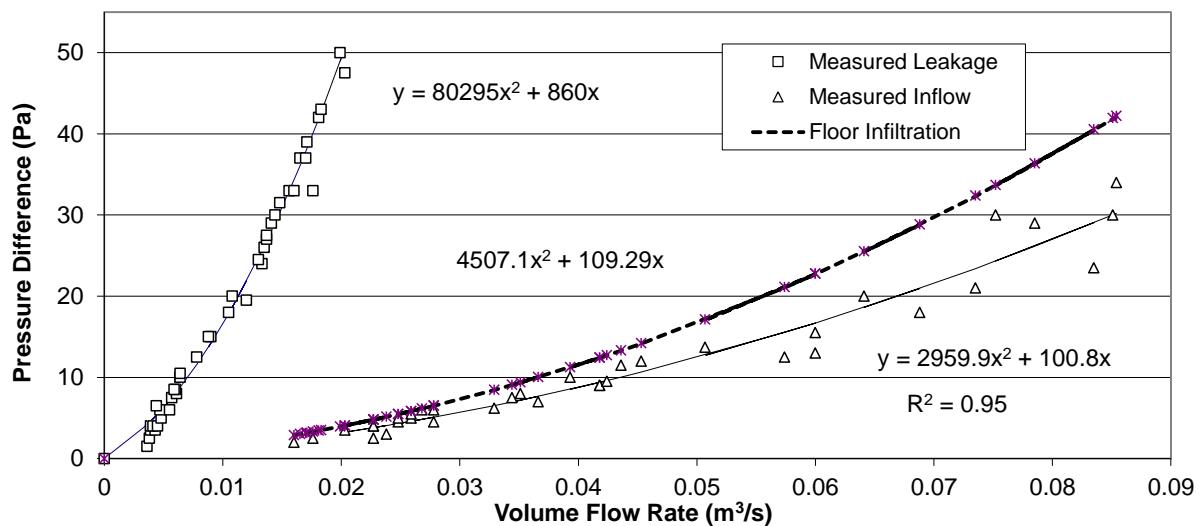
Fia 5



**Fig 6**

Figure 6

**Flow Rate versus Pressure Difference<sup>(ref.5)</sup>**



Marta data

Rawdata below on lines 101

### Marta data

Table of results for the Quadratic curves (fig.4)						Floor Infiltration	
Flow Rate	leakage		inflow		Predict	Predict	Floor Infiltration
	Measured	Quadratic	Press dif	PressDiff			
0	0	0	0	0	0	0	0
0.0036	1.5	4.1387832					
0.0038	2.5	4.4297398					
0.0039	3.5	4.57762695					
0.0043	3.5	5.18523455					
0.0039	4	4.57762695					
0.0041	4	4.87821895					
0.0045	4	5.49867375					
0.0048	5	5.9808768					
0.0047	6	5.81853655					
0.0055	6	7.16222375					
0.0044	6.5	5.3411512					
0.0057	7.5	7.51420455					
0.0061	8	8.23743695					
0.006	8.5	8.05422					
0.0059	8.5	7.87260895					
0.0064	10	8.7967232					
0.0064	10	8.7967232					
0.0064	10.5	8.7967232					
0.0078	12.5	11.5978278					
0.009	15	14.249295					
0.0088	15	13.7913248					
0.0105	18	17.88882375					
0.012	19.5	21.88968					
0.0108	20	18.6600888					
0.0133	24	25.64936255					
0.013	24.5	24.757655					

0.0135	26	26.25186375				
0.0137	27	26.86078855				
0.0137	27.5	26.86078855				
0.0141	29	28.09790895				
0.0144	30	29.0426112				
0.0148	31.5	30.3246968				
0.0156	33	32.9659512				
0.016	33	34.32512	2	2.371494	2.902406	2.89426
0.0165	37	36.06021375			3.030289	3.021789
0.017	37	37.835455			3.160426	3.151566
0.0171	39	38.19532095			3.186724	3.177791
0.0176	33	40.0187392	2.5	2.691995	3.319565	3.310264
0.0181	42	41.88230495			3.45466	3.444985
0.0183	43	42.63897255			3.509329	3.499503
0.0203	47.5	50.55894655			4.075849	4.064459
0.0199	50	48.92356295			3.95966	3.948591
0.0227			2.5	3.814729	4.803267	4.789872
0.0238			3	4.077074	5.154019	5.139659
0.0203			3.5	3.267203	4.075849	4.064459
0.0227			4	3.814729	4.803267	4.789872
0.0227			4	3.814729	4.803267	4.789872
0.0248			4.5	4.321785	5.48235	5.467087
0.0278			4.5	5.091437	6.521426	6.503309
0.0248			5	4.321785	5.48235	5.467087
0.0259			5	4.597805	5.853924	5.837641
0.0259			5.5	4.597805	5.853924	5.837641
0.0268			6	4.828967	6.166053	6.148912
0.0278			6	5.091437	6.521426	6.503309
0.0329			6.2	6.522119	8.474041	8.450575
0.0366			7	7.65644	10.03739	10.00965
0.0344			7.5	6.972211	9.09296	9.067801
0.0351			8	7.186812	9.38873	9.362762
0.0418			9	9.387604	12.44313	12.40882
0.0418			9	9.387604	12.44313	12.40882
0.0424			9.5	9.597654	12.7364	12.70129
0.0393			10	8.535334	11.2561	11.22503
0.0436			11.5	10.02415	13.33267	13.29594
0.0453			12	10.64294	14.19961	14.16051
0.0574			12.5	15.54152	21.12277	21.06485
0.06			13	16.70724	22.78265	22.72022
0.0507			13.7	12.722	17.12622	17.07916
0.06			15.5	16.70724	22.78265	22.72022
0.0688			18	20.94968	28.85286	28.77396
0.0641			20	18.62679	25.52397	25.4541
0.0735			21	23.40333	32.38088	32.29241
0.0835			23.5	29.05897	40.54983	40.43925
0.0785			29	26.15715	36.35268	36.25346
0.0752			30	24.32302	33.70601	33.61395
0.0851			30	30.01881	41.94052	41.82617
0.0854			34	30.20047	42.20384	42.08878

just fbds		109.2	4507.1							
just fbds		137	4507.1							
<b>A=137</b>		<b>A=109</b>		% Flow Rate				pred Q		
	Flow Rate	Flow Rate		<b>A=137</b>	<b>A=109</b>			from ratio		
								Equation	% Diff	
0.016	2.89426	0.014351	0.015973	10.15809	10.31	0.2	<b>0.0142</b>		11.53	2.89426
0.0165	3.021789	0.014826	0.016473	9.99768	10.15	0.2	<b>0.0146</b>		11.34	3.021789
0.017	3.151566	0.015301	0.016972	9.842138	9.99	0.2	<b>0.0151</b>		11.16	3.151566
0.0171	3.177791	0.015397	0.017072	9.811594	9.96	0.2	<b>0.0152</b>		11.13	3.177791
0.0176	3.310264	0.015873	0.017571	9.66161	9.81	0.2	<b>0.0157</b>		10.95	3.310264
0.0181	3.444985	0.016351	0.018070	9.516039	9.66	0.2	<b>0.0161</b>		10.79	3.444985
0.0183	3.499503	0.016542	0.018270	9.459004	9.61	0.2	<b>0.0163</b>		10.72	3.499503
0.0203	4.064459	0.018459	0.020267	8.923418	9.07	0.2	<b>0.0182</b>		10.10	4.064459
0.0199	3.948591	0.018074	0.019868	9.025728	9.17	0.2	<b>0.0179</b>		10.22	3.948591
0.0227	4.789872	0.020770	0.022664	8.354367	8.50	0.2	<b>0.0206</b>		9.44	4.789872
0.0238	5.139659	0.021833	0.023762	8.116731	8.26	0.2	<b>0.0216</b>		9.17	5.139659
0.0203	4.064459	0.018459	0.020267	8.923418	9.07	0.2	<b>0.0182</b>		10.10	4.064459
0.0227	4.789872	0.020770	0.022664	8.354367	8.50	0.2	<b>0.0206</b>		9.44	4.789872
0.0227	4.789872	0.020770	0.022664	8.354367	8.50	0.2	<b>0.0206</b>		9.44	4.789872
0.0248	5.467087	0.022802	0.024761	7.911956	8.06	0.2	<b>0.0226</b>		8.93	5.467087

0.0278	6.503309	0.025715	0.027756	7.354492	7.50	0.2	<b>0.0255</b>	8.28	6.503309
0.0248	5.467087	0.022802	0.024761	7.911956	8.06	0.2	<b>0.0226</b>	8.93	5.467087
0.0259	5.837641	0.023868	0.025859	7.698145	7.84	0.2	<b>0.0237</b>	8.68	5.837641
0.0259	5.837641	0.023868	0.025859	7.698145	7.84	0.2	<b>0.0237</b>	8.68	5.837641
0.0268	6.148912	0.024742	0.026758	7.5315	7.68	0.2	<b>0.0245</b>	8.49	6.148912
0.0278	6.503309	0.025715	0.027756	7.354492	7.50	0.2	<b>0.0255</b>	8.28	6.503309
0.0329	8.450575	0.030692	0.032849	6.566095	6.71	0.2	<b>0.0305</b>	7.37	8.450575
0.0366	10.00965	0.034318	0.036544	6.091394	6.24	0.2	<b>0.0341</b>	6.81	10.00965
0.0344	9.067801	0.032161	0.034347	6.36509	6.51	0.2	<b>0.0319</b>	7.13	
0.0351	9.362762	0.032847	0.035046	6.2754	6.42	0.2	<b>0.0326</b>	7.03	
0.0418	12.40882	0.039429	0.041737	5.528784	5.67	0.2	<b>0.0392</b>	6.16	
0.0418	12.40882	0.039429	0.041737	5.528784	5.67	0.2	<b>0.0392</b>	6.16	
0.0424	12.70129	0.040020	0.042336	5.470437	5.61	0.2	<b>0.0398</b>	6.09	
0.0393	11.22503	0.036970	0.039240	5.785805	5.93	0.2	<b>0.0368</b>	6.46	
0.0436	13.29594	0.041202	0.043534	5.357336	5.50	0.2	<b>0.0410</b>	5.95	
0.0453	14.16051	0.042878	0.045232	5.204836	5.35	0.2	<b>0.0427</b>	5.78	
0.0574	21.06485	0.054835	0.057315	4.327025	4.47	0.1	<b>0.0547</b>	4.75	
0.06	22.72022	0.057410	0.059912	4.175546	4.32	0.1	<b>0.0573</b>	4.57	
0.0507	17.07916	0.048208	0.050624	4.772952	4.91	0.1	<b>0.0480</b>	5.27	
0.06	22.72022	0.057410	0.059912	4.175546	4.32	0.1	<b>0.0573</b>	4.57	
0.0688	28.77396	0.066135	0.068700	3.732974	3.87	0.1	<b>0.0660</b>	4.05	
0.0641	25.45411	0.061473	0.064006	3.95702	4.10	0.1	<b>0.0613</b>	4.31	
0.0735	32.29241	0.070800	0.073393	3.532869	3.67	0.1	<b>0.0707</b>	3.81	
0.0835	40.43925	0.080736	0.083380	3.171037	3.31	0.1	<b>0.0807</b>	3.39	
0.0785	36.25346	0.075767	0.078387	3.342214	3.48	0.1	<b>0.0757</b>	3.59	
0.0752	33.61395	0.072489	0.075091	3.465659	3.61	0.1	<b>0.0724</b>	3.73	
0.0851	41.82617	0.082326	0.084978	3.119895	3.26	0.1	<b>0.0823</b>	3.33	
0.0854	42.08878	0.082625	0.085277	3.110489	3.25	0.1	<b>0.0826</b>	3.32	

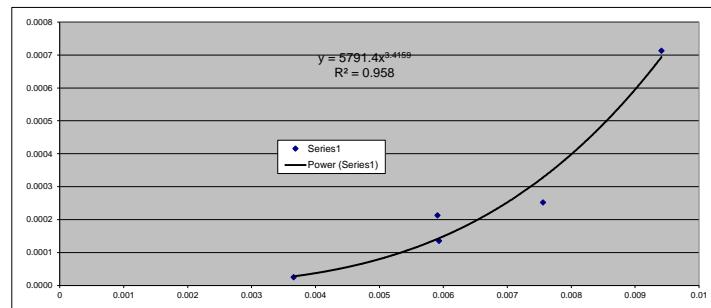
Equation = 
$$\begin{matrix} 0.7423 & 0.2577 & 0.05 \\ 0.7423 & 0.2577 & 0.05 \end{matrix}$$

pressure Difference 50	Crack Width 0.000251877 0.006958998	Edge Coef 4507	Laminar coef 137.22
Lam width =			
<b>Flow Rate at 50 Pa</b>	<b>0.0912</b>		
Predicted Flow Rate=	<b>0.0126</b>		

x= -0.4858  
0.1387

Lam.Coeff	Edge.Coeff	Lam.Coeff
0.005921	0.000247	0.00590088
0.005909	0.0021	0.005909333
0.005934	0.00014	0.005934084
0.009414	0.00071	0.009413836
0.007561	0.00025	0.00756062

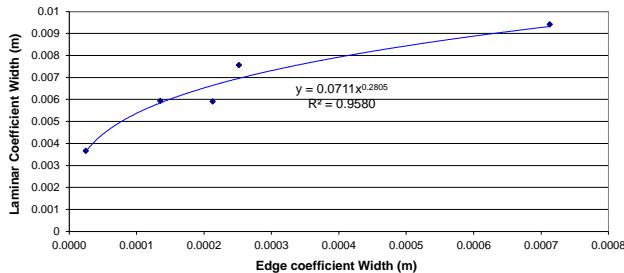
Edgeef(Lam)	% diff
2.05441E-08	-3.515337171
0.000140784	33.84110348
0.000142697	5.659363511
0.000740471	-3.86221847
0.000338908	-34.5530933



Gap size	Edge effect calculation	crack width	Laminar coef	crack width	Lam.Coeff
0 (test 2)	18218186	6.08738E-10	0.0000246726	6750	4.90667E-08
1.25(test 1)	244911	4.52822E-09	0.0002127961	1605	2.06355E-07
(2 fits) 3.5	606324	1.82907E-08	0.0001352432	1585	2.08959E-07
random	21819	5.08278E-07	0.0007129358	397	8.34257E-07
1.75(Marta)	4507	6.34418E-08	0.0002518765	107	4.32188E-07

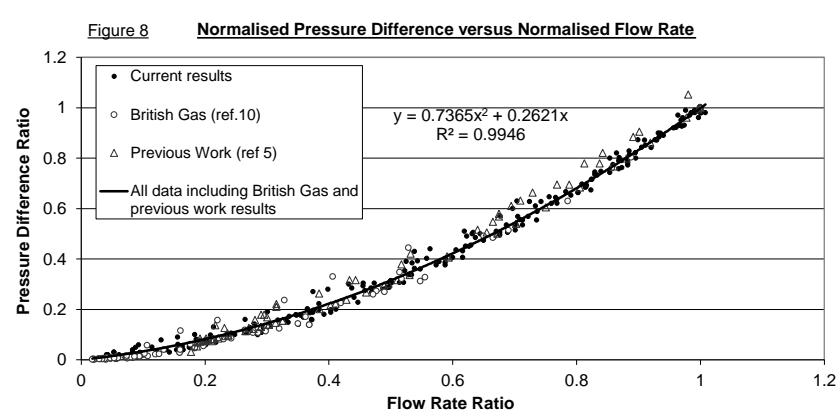
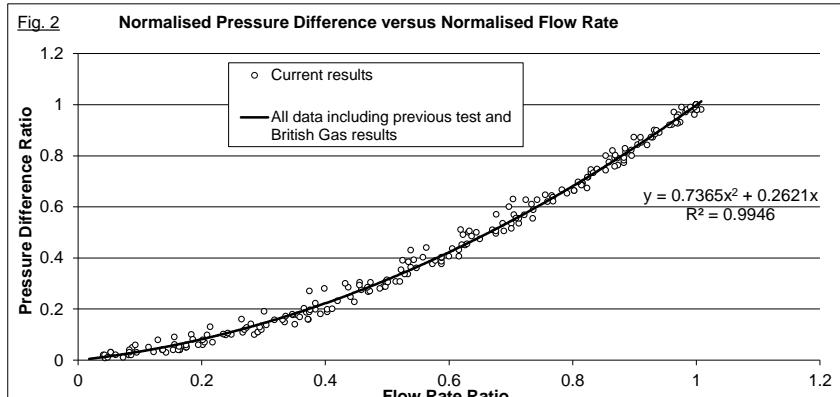
Lam.Coeff	Edge.Coeff	Lam.Coeff	Edge.Coeff	Lam.Coeff	Edge.Coeff
0.003660965	0.000247	2.76E-05	2.75594E-05	-11.70	33.53
0.005909333	0.00021	0.000141	0.000141443	3.69	-6.09
0.00664	0.009413836	0.00071	0.000694	2.65	0.00694021
0.005847	0.00756062	0.00025	0.000328	-30.30	0.000328207
0.00932					0.006961

Figure 1 Laminar Coefficient Width versus Edge Effect Coefficient Width



**Fig 8**

Ratio flow rate	Current res	British Gas	Previous Work (ref 5)
0.009804	0.043911	0.009804	
0.04902	0.087823	0.04902	
0.098039	0.209083	0.098039	
0.176471	0.354518	0.176471	
0.303922	0.455687	0.303922	
0.392157	0.543214	0.392157	
0.490196	0.622879	0.490196	
0.627451	0.742908	0.627451	
0.735294	0.829774	0.735294	
0.843137	0.905357	0.843137	
0.980392	1.008163	0.980392	
0.007353	0.043911	0.007353	
0.039216	0.083316	0.039216	
0.078431	0.129272	0.078431	
0.107843	0.266653	0.107843	
0.137255	0.304789	0.137255	
0.186275	0.369611	0.186275	
0.303922	0.473333	0.303922	
0.401961	0.5581	0.401961	
0.568627	0.705172	0.568627	
0.745098	0.829774	0.745098	
0.921569	0.960988	0.921569	
0.960784	0.997264	0.960784	
0.019608	0.040879	0.019608	
0.058824	0.092805	0.058824	
0.284314	0.437329	0.284314	
0.352941	0.522708	0.352941	
0.5	0.631591	0.5	
0.627451	0.724286	0.627451	
0.77451	0.858898	0.77451	
0.872549	0.899301	0.872549	
1	1	1	
0.019802	0.082339	0.019802	
0.029703	0.142615	0.029703	
0.039604	0.154042	0.039604	
0.039604	0.164677	0.039604	
0.039604	0.164677	0.039604	
0.049505	0.174667	0.049505	
0.059406	0.201688	0.059406	
0.069307	0.217849	0.069307	
0.09901	0.28523	0.09901	
0.108911	0.291111	0.108911	
0.108911	0.291111	0.108911	
0.118812	0.296876	0.118812	
0.148515	0.334461	0.148515	
0.168317	0.358906	0.168317	
0.158416	0.372804	0.158416	
0.188119	0.403376	0.188119	
0.19802	0.403376	0.19802	
0.227723	0.447214	0.227723	
0.267327	0.469403	0.267327	
0.287129	0.494032	0.287129	
0.287129	0.497451	0.287129	
0.306931	0.514203	0.306931	
0.306931	0.520756	0.306931	
0.336634	0.533616	0.336634	
0.376238	0.573423	0.376238	
0.376238	0.588016	0.376238	
0.386139	0.588016	0.386139	
0.405941	0.616166	0.405941	
0.50495	0.688895	0.50495	
0.534653	0.701089	0.534653	
0.514851	0.701089	0.514851	
0.534653	0.713074	0.534653	
0.554455	0.713074	0.554455	
0.663366	0.802538	0.663366	
0.663366	0.802538	0.663366	
0.673267	0.823387	0.673267	
0.732673	0.833616	0.732673	
0.742574	0.853706	0.742574	
0.762376	0.873334	0.762376	
0.772277	0.882984	0.772277	
0.782178	0.882984	0.782178	
0.792079	0.882984	0.792079	
0.851485	0.911322	0.851485	
0.841584	0.920575	0.841584	
0.871287	0.929735	0.871287	
0.930693	0.974245	0.930693	
0.970297	0.982905	0.970297	
0.980198	0.991489	0.980198	
0.980198	1	0.980198	
1	1	1	
0.05	0.175412	0.05	
0.05	0.175412	0.05	
0.1	0.248069	0.1	
0.14	0.350823	0.14	
0.16	0.372104	0.16	
0.18	0.392232	0.18	
0.2	0.411377	0.2	
0.27	0.47231	0.27	
0.28	0.488325	0.28	
0.36	0.547723	0.36	
0.4	0.588348	0.4	
0.45	0.626345	0.45	
0.5	0.644503	0.5	
0.61	0.733799	0.61	
0.62	0.759555	0.62	
0.79	0.868243	0.79	
0.79	0.872662	0.79	



0.89	0.94054	0.89
0.92	0.956757	0.92
1	1	1
0.019802	0.086102	0.019802
0.039604	0.162355	0.039604
0.059406	0.194657	0.059406
0.069307	0.204963	0.069307
0.118812	0.269235	0.118812
0.128713	0.294093	0.128713
0.188119	0.37421	0.188119
0.19802	0.383988	0.19802
0.247525	0.440907	0.247525
0.306931	0.497934	0.306931
0.336634	0.528434	0.336634
0.405941	0.599973	0.405941
0.435644	0.605433	0.435644
0.49505	0.676141	0.49505
0.534653	0.687018	0.534653
0.554455	0.735654	0.554455
0.643564	0.766365	0.643564
0.693069	0.816326	0.693069
0.80198	0.869111	0.80198
0.821782	0.895256	0.821782
0.871287	0.909855	0.871287
0.970297	0.963926	0.970297
1	1	1
0.03	0.05295	0.03
0.03	0.05295	0.03
0.02	0.061142	0.02
0.05	0.114386	0.05
0.09	0.155882	0.09
0.1	0.183425	0.1
0.13	0.213996	0.13
0.16	0.264752	0.16
0.19	0.301088	0.19
0.27	0.374415	0.27
0.28	0.398596	0.28
0.3	0.432338	0.3
0.39	0.525072	0.39
0.43	0.538256	0.43
0.44	0.563699	0.44
0.51	0.619013	0.51
0.57	0.676716	0.57
0.6	0.697124	0.6
0.63	0.703795	0.63
0.8	0.853799	0.8
0.82	0.864675	0.82
0.9	0.932287	0.9
0.96	0.971558	0.96
0.99	0.976356	0.99
1	1	1
0.019608	0.043147	0.019608
0.029412	0.095092	0.029412
0.039216	0.136444	0.039216
0.058824	0.174886	0.058824
0.078431	0.201721	0.078431
0.098039	0.239125	0.098039
0.127451	0.272888	0.127451
0.156863	0.317905	0.156863
0.176471	0.347866	0.176471
0.196078	0.375443	0.196078
0.27451	0.458374	0.27451
0.313725	0.5	0.313725
0.362745	0.538418	0.362745
0.401961	0.588001	0.401961
0.45098	0.620998	0.45098
0.509804	0.670424	0.509804
0.588235	0.736582	0.588235
0.637255	0.768392	0.637255
0.686275	0.81541	0.686275
0.715686	0.825137	0.715686
0.813725	0.885763	0.813725
0.872549	0.926845	0.872549
0.95098	0.968932	0.95098
0.980392	0.985264	0.980392
1	1	1
0.010526	0.073127	0.010526
0.031579	0.122365	0.031579
0.052632	0.163517	0.052632
0.105263	0.242536	0.105263
0.157895	0.331098	0.157895
0.178947	0.346873	0.178947
0.231579	0.420084	0.231579
0.284211	0.468243	0.284211
0.305263	0.501338	0.305263
0.389474	0.578122	0.389474
0.431579	0.616181	0.431579
0.473684	0.649969	0.473684
0.505263	0.676118	0.505263
0.568421	0.722074	0.568421
0.621053	0.768700	0.621053
0.652632	0.790992	0.652632
0.684211	0.814311	0.684211
0.715789	0.824102	0.715789
0.757895	0.868336	0.757895
0.8	0.895622	0.8
0.884211	0.933626	0.884211
0.926316	0.970143	0.926316
0.978947	1.001336	0.978947
0.989474	0.997323	0.989474
1	1	1
0.010101	0.048002	0.010101
0.030303	0.083141	0.030303
0.040404	0.137455	0.040404
0.060606	0.156282	0.060606

0.080808	0.185909	0.080808
0.10101	0.235159	0.10101
0.141414	0.279895	0.141414
0.171717	0.336011	0.171717
0.20202	0.365569	0.20202
0.222222	0.384012	0.222222
0.292929	0.455383	0.292929
0.383838	0.534522	0.383838
0.454545	0.629534	0.454545
0.484848	0.636812	0.484848
0.505051	0.633184	0.505051
0.555556	0.708734	0.555556
0.646465	0.755926	0.646465
0.666667	0.782881	0.666667
0.69697	0.808936	0.69697
0.747475	0.839684	0.747475
0.828283	0.885105	0.828283
0.89899	0.935722	0.89899
0.929293	0.967204	0.929293
0.989899	0.990741	0.989899

1	1	1
0.001302	0.019326	0.001302
0.003128	0.030644	0.003128

Test 4

Etheridge

0.004878	0.041651	0.004878
0.007713	0.056842	0.007713
0.011223	0.077099	0.011223
0.020646	0.101361	0.020646
0.082098	0.230073	0.082098
0.170892	0.362365	0.170892
0.321654	0.517664	0.321654
0.514004	0.690219	0.514004

1	1	1
0.057406	0.106557	0.057406
0.115313	0.160494	0.115313
0.156938	0.220339	0.156938
0.236843	0.328283	0.236843
0.330078	0.40625	0.330078
0.444876	0.528455	0.444876

1	1	1
0.003409	0.019461	0.003409
0.004954	0.028509	0.004954
0.007898	0.041139	0.007898
0.011513	0.056034	0.011513
0.017953	0.079268	0.017953
0.022625	0.095588	0.022625
0.06861	0.193452	0.06861
0.118355	0.264228	0.118355
0.225266	0.419355	0.225266
0.490905	0.65	0.490905

1	1	1
0.00934	0.04138	0.00934
0.01455	0.07059	0.01455
0.02389	0.10000	0.02389
0.03973	0.13333	0.03973
0.06145	0.17647	0.06145
0.08617	0.22222	0.08617
0.12926	0.30000	0.12926
0.28143	0.48000	0.28143
0.48919	0.66667	0.48919

1.00000	1.00000	1.00000
0.001654	0.039773	0.001654
0.00338	0.056452	0.00338
0.006386	0.074468	0.006386
0.008628	0.083333	0.008628
0.02857	0.159091	0.02857
0.138819	0.368421	0.138819

1	1	1
0.001654	0.039773	0.001654
0.00338	0.056452	0.00338
0.006386	0.074468	0.006386
0.008628	0.083333	0.008628
0.02857	0.159091	0.02857
0.138819	0.368421	0.138819

1	1	1
0.001383	0.022	0.001383
0.0031	0.035484	0.0031
0.004512	0.047667	0.004512
0.010511	0.080337	0.010511
0.015414	0.107519	0.015414
0.045935	0.183333	0.045935
0.11369	0.297917	0.11369
0.272653	0.476667	0.272653
0.484511	0.665116	0.484511

1	1	1
0.002381	0.020502	0.002381
0.003925	0.032111	0.003925
0.006155	0.047172	0.006155
0.011138	0.073019	0.011138
0.015714	0.090346	0.015714
0.054854	0.197422	0.054854
0.108317	0.28658	0.108317
0.327472	0.555249	0.327472
0.803397	0.888399	0.803397

1	1	1
0.00603	0.03467	0.00603
0.01138	0.05627	0.01138
0.01636	0.07367	0.01636
0.02471	0.10250	0.02471
0.06582	0.18860	0.06582
0.10736	0.27736	0.10736
0.25973	0.47151	0.25973
0.62946	0.78585	0.62946
1.00000	1.00000	1.00000
0.003866	0.038055	0.003866
0.006692	0.059406	0.006692

0.01329	0.094737	0.01329
0.048009	0.183673	0.048009
0.107378	0.285714	0.107378
0.347906	0.514286	0.347906
0.760126	0.857143	0.760126
1	1	1
0.001383	0.01791	0.001383
0.002882	0.033803	0.002882
0.007904	0.059259	0.007904
0.013946	0.094118	0.013946
0.021705	0.12	0.021705
0.040303	0.16	0.040303
0.124492	0.32	0.124492
0.269276	0.489796	0.269276
1	1	1
0.003087	0.029302	0.003087
0.006926	0.05339	0.006926
0.013076	0.086301	0.013076
0.023056	0.126	0.023056
0.085266	0.242308	0.085266
0.14349	0.35	0.14349
0.311916	0.547826	0.311916
1	1	1
0.068771	0.187354	0.068771 Etheridge
0.071801	0.193208	Marta floor
0.074885	0.199063	0.071801
0.075508	0.200234	0.074885
0.078656	0.206089	0.075508
0.081857	0.211944	0.078656
0.083152	0.214286	0.081857
0.093822	0.233021	0.083152
0.096575	0.237705	0.093822
0.096575	0.237705	0.096575
0.113811	0.265808	0.096575
0.113811	0.265808	0.113811
0.113811	0.265808	0.113811
0.122122	0.278689	0.113811
0.129902	0.290398	0.122122
0.129902	0.290398	0.129902
0.138706	0.303279	0.129902
0.138706	0.303279	0.138706
0.146102	0.313817	0.138706
0.154522	0.325527	0.146102
0.154522	0.325527	0.154522
0.200788	0.385246	0.154522
0.215453	0.40281	0.200788
0.222462	0.411007	0.215453
0.237831	0.428571	0.222462
0.266708	0.460187	0.237831
0.294834	0.489461	0.266708
0.301783	0.496487	0.294834
0.315911	0.510539	0.301783
0.336453	0.530445	0.315911
0.405798	0.593677	0.336453
0.500494	0.672131	0.405798
0.539824	0.702576	0.500494
0.539824	0.702576	0.539824
0.604778	0.750585	0.539824
0.683655	0.805621	0.604778
0.76725	0.860656	0.683655
0.798648	0.880562	0.76725
0.86136	0.919204	0.798648
0.960809	0.977752	0.86136
0.993761	0.996487	0.960809
1	1	0.993761
0.031579	0.17734	1 Marta floor
0.031579	0.17734	0.031579 Marta Total
0.052632	0.187192	0.052632
0.073684	0.192118	0.073684
0.084211	0.192118	0.084211
0.084211	0.20197	0.084211
0.073684	0.211823	0.073684
0.136842	0.216749	0.136842
0.084211	0.221675	0.084211
0.126316	0.231527	0.126316
0.105263	0.236453	0.105263
0.126316	0.270936	0.126316
0.157895	0.280788	0.157895
0.178947	0.29064	0.178947
0.178947	0.295567	0.178947
0.168421	0.300493	0.168421
0.210526	0.315271	0.210526
0.210526	0.315271	0.210526
0.221053	0.315271	0.221053
0.263158	0.384236	0.263158
0.315789	0.433498	0.315789
0.315789	0.44335	0.315789
0.378947	0.517241	0.378947
0.421053	0.53202	0.421053
0.410526	0.591133	0.410526
0.515789	0.640394	0.515789
0.505263	0.655172	0.505263
0.547368	0.665025	0.547368
0.568421	0.674877	0.568421
0.578947	0.674877	0.578947
0.610526	0.694581	0.610526
0.631579	0.709396	0.631579
0.663158	0.729064	0.663158
0.694737	0.768473	0.694737
0.694737	0.788177	0.694737
0.778947	0.812808	0.694737
0.778947	0.837438	0.778947
0.821053	0.842365	0.778947
0.884211	0.891626	0.821053
0.905263	0.901478	0.884211
1.052632	0.980296	0.905263
		1.052632

1 1

1 Marta

Fig 9

