Hydroelasticity of three elastic coaxial shells interacting with viscous incompressible fluids between them under vibration

Dmitry V. Kondratov¹, Olga V. Elistratova², Lev I. Mogilevich³, Yulia N. Kondratova⁴

^{1, 2}Russian Presidential Academy of National Economy and Public Administration,

Saratov, Russian Federation

³Yuri Gagarin State Technical University of Saratov, Saratov, Russia

⁴National Research Saratov State University, Saratov, Russia

¹Corresponding author

E-mail: 1kondratovdv@yandex.ru, 2elistratovaov@yandex.ru, 3mogilevich@sgu.ru,

⁴kondratovaun@mail.ru

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Abstract. Elastic cylindrical shells interacting with a viscous incompressible fluid are widely used in various branches of science and technology, such as engineering and aviation engineering. They provide the possibility for solving a lot of problems, such as: reducing constructing weight and dimensions, equalizing dynamic influences and vibration level, as well as reducing friction and wearing, cooling. Mathematical model of the system, representing three coaxial cylindrical shells, freely supported at the ends, and interacting with viscous incompressible fluid between them under mechanical system harmonic vibration is constructed. This mathematical model represents a coupled system consisting of the Navier-Stokes, continuity for each fluid and equations and the ones of elastic coaxial cylindrical shells dynamics which are based on the Kirchhoff-Love hypotheses and the corresponding boundary conditions, namely: for the fluid non-flow and for free attaching for the shells. The constructed mathematical model allows to investigate the oscillations of a mechanical system consisting of coaxial elastic cylindrical shells interacting with viscous incompressible liquids in order to identify dangerous operating modes.

Keywords: hydroelasticity, oscillations, three coaxial shells, viscous fluid, vibration, mathematical modeling.

1. Introduction

Coaxial elastic thin-walled shells interacting with viscous incompressible fluid between them are of great significance for the development of modern high-tech products that are used in rocket, space and automobile industry, railway transport, agricultural machinery, fuel and energy complexes. They provide the possibility for solving a lot of problems, such as: reducing constructing weight and dimensions, equalizing dynamic influences and vibration level, as well as reducing friction and wearing, cooling. The investigation of thin-walled structures dynamics was made in references [1, 2]. Viscous incompressible fluid interaction with elastic plates is considered in references [3-7]. Geometrically nonlinear oscillations of circular cylindrical shells under internal fluid flow activity were investigated in references [8, 9]. In this case, the shell is modeled on the basis of Donnell's nonlinear theory of the shell, keeping geometric imperfections in the inertia plane. Experimental and theoretical parametric study is carried out to investigate transmural pressure effect of on the nonlinear dynamics and stability of circular cylindrical shells, there ends are fixed and loaded with internal fluid flow [10]. Cavitation effects in vibrations based on a model experimental setup for the study of fast flowing hydrodynamic processes with allowance for the interaction of liquid, gas, and deformable structure are investigated in [11]. A variant of a three-dimensional mathematical formulation and a numerical implementation algorithm is designed to determine the dynamic characteristics of arbitrary geometry shells. They are completely or partially filled with a compressible inviscous fluid and are considered in [12]. The analysis of circular cylindrical shells panel flutter with ideal compressible fluid and a streamlined supersonic gas flow is carried out [13]. Mathematical model of this system consisting of partial differential equations and describing viscous incompressible fluid dynamics and an elastic ribbed shell is presented in [14, 15]. References [16, 17] present the solution annular cross section hydroelasticity problem, consisting of viscous incompressible fluid dynamics equations and the ones of internal and external elastic cylindrical shells of finite length dynamics, the letter being based on the Kirchhoff-Love hypotheses with the corresponding boundary conditions under base vibration with a fixed the annular tube. No one's is investigating the problem hydroelasticity of three elastic coaxial shells interacting with viscous incompressible fluids between them under vibration. In this paper, we consider the construction of a mathematical model that will provide a joint consideration of the vibration of a mechanical system, the inertia of the motion of a viscous fluid, and the elasticity of three cylindrical shells of finite length freely supported at the ends of the mechanical system.

2. Statement of the problem

We consider the following mechanical model (Fig. 1), consisting of outer shell 1, middle shell 2 and inner shell 3, which are elastic coaxial cylindrical ones. The inner radius of the outer shell is denoted as R_1 , and the outer radius of the inner shell is R_3 , and the outer and inner radius of the middle shell are R_{21} and R_{22} , respectively. Fastening of the inner, middle and outer shells ensures their free support. The viscous incompressible fluid 1 and fluid 2 completely fills the gap between the walls of the three shells. We assume that the radial clearance of a cylindrical gap in the cylinder $1 \delta_1$ is much smaller than the outer radius of the middle shell 2, that is, $\delta_1 = R_1 - R_{21} << R_{21}$. We also assume that the radial clearance of a cylindrical gap in the cylinder $2 \delta_2$ is much smaller than the inner radius of the middle shell 2, that is, $\delta_2 = R_{22} - R_3 << R_3$. The mechanical system is influenced by the time-harmonic inertial force of moving space. The shifts of the inner and middle shells relative to the outer one at the ends is absent. Also, there is no inner and middle shells shifts relative to each other. The mechanical system is considered to be thermally stabilized.

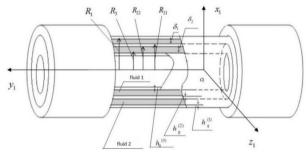


Fig. 1. Mechanical model

3. The theory and solution

To construct a mathematical model of the mechanical system under consideration, we introduce the coordinate system $O_1x_1y_1z_1$ associated with the base to which the mechanical system is attached, that is, the center O_1 is located in the geometric center of the coaxial shells in the unperturbed state. We assume that the component of vibration acceleration along the axis O_1y_1 is absent. We denote the vibration acceleration of the base by \ddot{x}_0 , \ddot{z}_0 . In addition, we introduce a cylindrical coordinate system r, θ , y (\bar{n}_r , \bar{n}_θ , \bar{j} – the unit of the cylindrical system), which poles and the axes directions Oy, O_1y_1 coincide the cylindrical and Cartesian coordinate systems. (Fig. 2)

The mathematical model of the presented mechanical system has the form. Equation of Navier-Stocks [18]:

$$\overline{W}_{11} = -\frac{1}{\rho_1} \nabla p_1 + \nu_1 \Delta \overline{V}_1, \ \nabla \cdot \overline{V}_1 = 0, \ \overline{W}_{22} = -\frac{1}{\rho_2} \nabla p_2 + \nu_2 \Delta \overline{V}_2, \ \nabla \cdot \overline{V}_2 = 0,$$

and boundary conditions for cylindrical axes on impenetrable surface of shells in a cylindrical clearance are:

$$\begin{split} V_{1r} &= \frac{\partial u_{3}^{(i)}}{\partial t}, \ V_{1\theta} &= \frac{\partial u_{2}^{(i)}}{\partial t}, \ V_{1y} = -\frac{\partial u_{1}^{(i)}}{\partial t}, \ r = R_{3} + \delta_{1} + h_{0}^{(2)} + \delta_{2} + u_{3}^{(1)}, \\ V_{1r} &= \frac{\partial u_{3}^{(i)}}{\partial t}, \ V_{1\theta} &= \frac{\partial u_{2}^{(i)}}{\partial t}, \ V_{1y} = -\frac{\partial u_{1}^{(i)}}{\partial t}, \ r = R_{3} + h_{0}^{(2)} + \delta_{2} + u_{3}^{(2)}, \\ V_{2r} &= \frac{\partial u_{3}^{(i)}}{\partial t}, \ V_{2\theta} &= \frac{\partial u_{2}^{(i)}}{\partial t}, \ V_{2y} = -\frac{\partial u_{1}^{(i)}}{\partial t}, \ r = R_{3} + u_{3}^{(3)}, \\ V_{2r} &= \frac{\partial u_{3}^{(i)}}{\partial t}, \ V_{2\theta} &= \frac{\partial u_{2}^{(i)}}{\partial t}, \ V_{2y} = -\frac{\partial u_{1}^{(i)}}{\partial t}, \ r = R_{3} + \delta_{2} + u_{3}^{(2)}, \end{split}$$

and also, the dynamic equations of elastic shell based on Kirchhoff-Love hypotheses are:

$$\begin{split} &\frac{\partial^2 u_1^{(i)}}{\partial y^2} + \frac{1 - \mu_0^{(i)}}{2} \frac{1}{(R^{(i)})^2} \frac{\partial^2 u_1^{(i)}}{\partial \theta^2} - \frac{1 + \mu_0^{(i)}}{2} \frac{1}{R^{(i)}} \frac{\partial^2 u_2^{(i)}}{\partial y \partial \theta} - \frac{\mu_0^{(i)}}{R^{(i)}} \frac{\partial u_3^{(i)}}{\partial y} = \frac{1 - \left(\mu_0^{(i)}\right)^2}{E^{(i)} h_0^{(i)}} \\ &\cdot \left[\rho_0^{(i)} h_0^{(i)} W_{0s}^{(i)} - G_1^{(i)} \right], \\ &- \frac{1 + \mu_0^{(i)}}{2} \frac{1}{R^{(i)}} \frac{\partial^2 u_1^{(i)}}{\partial y \partial \theta} + \frac{1 - \mu_0^{(i)}}{2} \frac{\partial^2 u_2^{(i)}}{\partial y^2} + \frac{1}{(R^{(i)})^2} \frac{\partial^2 u_2^{(i)}}{\partial \theta^2} \\ &+ \left(a_0^{(i)} \right)^2 \left[2 \left(1 - \mu_0^{(i)} \right) \frac{\partial^2 u_2^{(i)}}{\partial y^2} + \frac{1}{(R^{(i)})^2} \frac{\partial^2 u_2^{(i)}}{\partial \theta^2} \right] + \frac{1}{(R^{(i)})^2} \frac{\partial u_2^{(i)}}{\partial \theta} \\ &- \left(a_0^{(i)} \right)^2 \left[\left(2 - \mu_0^{(i)} \right) \frac{\partial^3 u_3^{(i)}}{\partial y^2 \partial \theta} + \frac{\partial^3 u_3^{(i)}}{\partial \theta^3} \right] = \frac{1 - \left(\mu_0^{(i)} \right)^2}{E^{(i)} h_0^{(i)}} \left[\rho_0^{(i)} h_0^{(i)} W_{0\theta}^{(i)} - G_2^{(i)} \right], \\ &- \frac{\mu_0^{(i)}}{R^{(i)}} \frac{\partial u_1^{(i)}}{\partial y} + \frac{1}{(R^{(i)})^2} \frac{\partial u_2^{(i)}}{\partial \theta} - \left(a_0^{(i)} \right)^2 \left[\left(2 - \mu_0^{(i)} \right) \frac{\partial^3 u_2^{(i)}}{\partial y^2 \partial \theta} + \frac{1}{(R^{(i)})^2} \frac{\partial^3 u_2^{(i)}}{\partial \theta^3} \right] + \frac{u_3^{(i)}}{R^{(i)}} \\ &+ \left(a_0^{(i)} \right)^2 \left[\left(R^{(i)} \right)^2 \frac{\partial^4 u_3^{(i)}}{\partial y^4} + 2 \frac{\partial^4 u_3^{(i)}}{\partial y^2 \partial \theta^2} + \frac{1}{(R^{(i)})^2} \frac{\partial^4 u_3^{(i)}}{\partial \theta^4} \right] = \frac{1 - \left(\mu_0^{(i)} \right)^2}{E^{(i)} h_0^{(i)}} \\ &\cdot \left[- \rho_0^{(i)} h_0^{(i)} W_{0n}^{(i)} + G_3^{(i)} \right], \end{split}$$

and boundary conditions of shells free attaching are:

$$\frac{\partial u_1^{(i)}}{\partial y} = 0$$
, $u_2^{(i)} = 0$, $u_3^{(i)} = 0$, $\frac{\partial^2 u_3^{(i)}}{\partial y^2} = 0$, $y = \pm \frac{l}{2}$, $i = 1,2,3$,

where p_1 – pressure fluid 1; p_2 – pressure fluid 2; ρ_1 – fluid 1 density; ρ_2 – fluid 2 density; ν_1 – kinematic viscosity of the fluid 1; ν_2 – kinematic viscosity of the fluid 2; Δ – Laplace operator; ∇ – Hamiltonian operator; $\overline{W}_{11} = \overline{W}_1 + \frac{\partial \overline{V}_1}{\partial t} + (\overline{V}_1 \cdot \nabla) \overline{V}_1$ – absolute fluid 1 volume unit acceleration; $\overline{W}_{22} = \overline{W}_1 + \frac{\partial \overline{V}_2}{\partial t} + (\overline{V}_2 \cdot \nabla) \overline{V}_2$ – absolute fluid 2 volume unit acceleration;

 $\overline{W}_1 = W_{1x1}\overline{i} + W_{1z1}\overline{k}$ - base drag acceleration; $\overline{r} = r\,\overline{n}_r + y\overline{j}$ - the radius vector of the liquid particle mass center with respect to the pole O_1 of the cylindrical coordinate system r, θ , y; $\overline{V}_1 = V_{1r}\overline{n}_r + V_{1\theta}\overline{n}_\theta + V_{1y}\overline{j}$ - fluid 1 velocity of the relative to the cylindrical slot in the projections on the axis r, θ , y; $\overline{V}_2 = V_{2r}\overline{n}_r + V_{2\theta}\overline{n}_\theta + V_{2y}\overline{j}$ - fluid 2 velocity of the relative to the cylindrical slot in the projections on the axis r, θ , y; \overline{n}_r , \overline{n}_θ , \overline{j} - cylindrical coordinate system r, θ , y base unit; $u_1^{(i)} = u_1^{(i)}(y,\theta,t)$ - elastic longitudinal shell shift which is positive in the direction \overline{n}_s , opposite in the direction \overline{j} ; $u_2^{(i)} = u_2^{(i)}(y,\theta,t)$ - circumferential elastic movement in the direction \overline{n}_θ , $u_3^{(i)} = u_3^{(i)}(y,\theta,t)$ - shell sagging which is positive in the direction \overline{n} , coinciding with \overline{n}_r and opposite in a direction to a curvature center; $\overline{u}^{(i)} = u_1^{(i)}\overline{n}_s + u_2^{(i)}\overline{n}_\theta + u_3^{(i)}\overline{n}$ - vector of elastic movements of shells; $E^{(i)}$ - Young's modulus, $\mu_0^{(i)}$ - Poisson's ratio, $\rho_0^{(i)}$ - material density, $R^{(i)}$ - middle surface radius, $h_0^{(i)}$ - shell thickness, $W_{0s}^{(i)}$, $W_{0\theta}^{(i)}$, $W_{0n}^{(i)}$ - projection of shells middle surface unit area the absolute acceleration; $\overline{W}_0^{(i)} = \overline{W}_1 + \frac{\partial^2 \overline{u}^{(i)}}{\partial t^2}$, i = 1, 2, 3 at axes \overline{n}_s , \overline{n}_θ , \overline{n}_s i = 1 - outer shell, i = 2 - middle shell; i = 3 - inner shell:

$$\begin{split} G_{1}^{(1)} &= q_{s}^{(1)}, \quad G_{1}^{(2)} &= q_{sz}^{(2)} + q_{st}^{(2)}, \quad G_{1}^{(3)} &= q_{3}^{(3)}, \quad G_{2}^{(1)} &= q_{\theta}^{(1)}, \quad G_{2}^{(2)} &= q_{\theta2}^{(2)} + q_{\theta1}^{(2)}, \\ G_{2}^{(3)} &= q_{\theta}^{(3)}, \quad G_{3}^{(1)} &= -q_{n}^{(1)}, \quad G_{3}^{(2)} &= q_{n2}^{(2)} - q_{n1}^{(2)}, \quad G_{3}^{(3)} &= q_{n}^{(3)}, \\ F^{(1)} &= R_{3} + \delta_{1} + h_{0}^{(2)} + \delta_{2} + u_{3}^{(1)}, \quad r^{(21)} &= R_{3} + \delta_{2} + h_{0}^{(2)} + u_{3}^{(2)}, \\ F^{(22)} &= R_{3} + \delta_{2} + u_{3}^{(2)}, \quad r^{(3)} &= R_{3} + u_{3}^{(3)}, \quad W_{0s}^{(i)} &= \frac{\partial^{2} u_{1}^{(i)}}{\partial t^{2}}, \\ W_{0\theta}^{(i)} &= W_{1x1} \cos \theta - W_{1z1} \sin \theta + \frac{\partial^{2} u_{2}^{(i)}}{\partial t^{2}}, \quad W_{0n}^{(i)} &= W_{1x1} \sin \theta + W_{1z1} \cos \theta + \frac{\partial^{2} u_{3}^{(i)}}{\partial t^{2}}, \\ i &= 1,2,3, \\ g_{s}^{(1)} &= & - \left[P_{ry}^{(1)} \cos^{(1)}(\bar{n},\bar{n}_{r}) + P_{\theta y}^{(1)} \cos^{(1)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(1)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(1)}}, \\ q_{0}^{(1)} &= & \left[P_{r\theta}^{(1)} \cos^{(1)}(\bar{n},\bar{n}_{r}) + P_{\theta y}^{(1)} \cos^{(1)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(1)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(1)}}, \\ q_{1}^{(2)} &= & \left[P_{ry}^{(1)} \cos^{(1)}(\bar{n},\bar{n}_{r}) + P_{\theta y}^{(1)} \cos^{(1)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(1)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(21)}}, \\ q_{21}^{(2)} &= & - \left[P_{ry}^{(1)} \cos^{(21)}(\bar{n},\bar{n}_{r}) + P_{\theta y}^{(1)} \cos^{(21)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(21)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(21)}}, \\ q_{22}^{(2)} &= & \left[P_{r\theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{r}) + P_{\theta \theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(21)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(21)}}, \\ q_{22}^{(2)} &= & \left[P_{r\theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{r}) + P_{\theta \theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(21)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(21)}}, \\ q_{23}^{(2)} &= & \left[P_{r\theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{r}) + P_{\theta \theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(1)} \cos^{(21)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(21)}}, \\ q_{3}^{(3)} &= & \left[P_{r\theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{r}) + P_{\theta \theta}^{(2)} \cos^{(22)}(\bar{n},\bar{n}_{\theta}) + P_{yy}^{(2)} \cos^{(22)}(\bar{n},\bar{j}) \right] \right|_{r=r^{(3)}}, \\ q_{3}^{(3)} &= & \left[P_{r\theta}^{(2)} \cos^{(3)}(\bar{n},\bar{n}_{r}) + P_{\theta \theta}^{(2)} \cos^{(3)}($$

$$\begin{split} &P_{\theta\theta}(^{2)} = -p_2 + 2\rho_2 v_2 \left(\frac{1}{r} \frac{\partial V_{2\theta}}{\partial \theta} + \frac{V_{2r}}{r}\right), \ P_{ry}(^{2)} = \rho_2 v_2 \left(\frac{\partial V_{2y}}{\partial r} + \frac{\partial V_{2r}}{\partial y}\right), \\ &P_{yy}(^{2)} = -p_2 + 2\rho_2 v_2 \frac{\partial V_{2y}}{\partial y}, \ P_{\theta y}(^{2)} = \rho_2 v_2 \left(\frac{\partial V_{2\theta}}{\partial y} + \frac{1}{r} \frac{\partial V_{2y}}{\partial \theta}\right), \cos^{(1)}(\bar{n}, \bar{n}_r) = \frac{r}{|\bar{N}|^{(1)}}, \\ &\cos^{(1)}(\bar{n}, \bar{n}_{\theta}) = -\frac{1}{|\bar{N}|^{(1)}} \frac{\partial u_3^{(1)}}{\partial \theta}, \cos^{(1)}(\bar{n}, \bar{j}) = -\frac{r}{|\bar{N}|^{(1)}} \frac{\partial u_3^{(2)}}{\partial y} \bigg|_{r=r^{(1)}}, \\ &\cos^{(21)}(\bar{n}, \bar{n}_r) = \frac{r}{|\bar{N}|^{(21)}}, \cos^{(21)}(\bar{n}, \bar{n}_{\theta}) = -\frac{1}{|\bar{N}|^{(21)}} \frac{\partial u_3^{(2)}}{\partial \theta}, \\ &\cos^{(1)}(\bar{n}, \bar{j}) = -\frac{r}{|\bar{N}|^{(21)}} \frac{\partial u_3^{(2)}}{\partial y} \bigg|_{r=r^{(21)}}, \cos^{(22)}(\bar{n}, \bar{n}_r) = \frac{r}{|\bar{N}|^{(22)}} \frac{\partial u_3^{(2)}}{\partial y} \bigg|_{r=r^{(22)}}, \\ &\cos^{(22)}(\bar{n}, \bar{n}_{\theta}) = -\frac{1}{|\bar{N}|^{(3)}} \frac{\partial u_3^{(2)}}{\partial \theta}, \cos^{(2)}(\bar{n}, \bar{j}) = -\frac{r}{|\bar{N}|^{(22)}} \frac{\partial u_3^{(2)}}{\partial y} \bigg|_{r=r^{(22)}}, \\ &\cos^{(3)}(\bar{n}, \bar{n}_r) = \frac{r}{|\bar{N}|^{(3)}}, \cos^{(3)}(\bar{n}, \bar{n}_{\theta}) = -\frac{1}{|\bar{N}|^{(3)}} \frac{\partial u_3^{(3)}}{\partial \theta}, \\ &\cos^{(3)}(\bar{n}, \bar{j}) = -\frac{r}{|\bar{N}|^{(3)}} \frac{\partial u_3^{(3)}}{\partial y} \bigg|_{r=r^{(3)}}, (a_0^{(1)})^2 = \frac{(h_0^{(1)})^2}{12(R^{(1)})^2}, \\ &|\bar{N}|^{(1)} = \left\{ (r^{(21)})^2 \left[1 + \frac{\partial u_3^{(2)}}{\partial y} \right]^2 + \left(\frac{\partial u_3^{(1)}}{\partial \theta}\right)^2 \right\}^{\frac{1}{2}}, \\ &|\bar{N}|^{(22)} = \left\{ (r^{(22)})^2 \left[1 + \frac{\partial u_3^{(2)}}{\partial y} \right]^2 + \left(\frac{\partial u_3^{(3)}}{\partial \theta}\right)^2 \right\}^{\frac{1}{2}}, \\ &|\bar{N}|^{(3)} = \left\{ (r^{(3)})^2 \left[1 + \frac{\partial u_3^{(2)}}{\partial y} \right]^2 + \left(\frac{\partial u_3^{(3)}}{\partial \theta}\right)^2 \right\}^{\frac{1}{2}}. \end{aligned}$$

4. Conclusions

Thus, we constructed a mathematical model of a mechanical system consisting of three coaxial elastic cylindrical shells interacting through viscous incompressible fluid layers under of in the presence of external vibration. The mathematical model of the system under consideration: shell-liquid-shell-liquid-shell is a connected system of equations, including non-linear partial Navier-Stokes equations and the continuity equation for describing the fluids dynamics of between elastic cylindrical shells, and partial differential ones for describing elastic cylindrical shells dynamics, based on Kirchhoff-Love hypotheses and the corresponding boundary conditions. The constructed mathematical model allows to investigate the oscillations of a mechanical system consisting of coaxial elastic cylindrical shells interacting with viscous incompressible liquids in order to identify dangerous operating modes.

Calculations have shown (Table 1) that the liquid pressure $d(\omega)$ can really drop below 0.2 atmospheres at the resonance frequencies of the deflection of one of the shells, which corresponds

to the conditions of cavitation.

Table 1. Liquid pressure $d(\omega)$) at the resonance fre	equencies of the middle shell
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- *****						
ω, rad/s	6669	22225	64491	79909	110630	114836
$d(\omega)$, Pa	9,90E+03	1,20E+09	1,49E+04	1,09E+05	4,95E+05	1,66E+05

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