

Direct numerical simulation of gas transfer across the air-water interface driven by buoyant convection

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A series of direct numerical simulations of mass transfer across the air-water interface driven by buoyancy-induced convection has been carried out to elucidate the physical mechanisms that play a role in the transfer of heat and atmospheric gases. The buoyant instability is caused by the presence of a thin layer of cold water situated on top of a body of warm water. In time, heat and atmospheric gases diffuse into the uppermost part of the thermal boundary layer and are subsequently transported down into the bulk by falling sheets and plumes of cold water. Using a specifically-designed numerical code for the discretization of scalar convection and diffusion, it was possible to accurately resolve this buoyant instability induced transport of atmospheric gases into the bulk at a realistic Prandtl number ($Pr = 6$) and Schmidt numbers ranging from $Sc = 20$ to $Sc = 500$. The simulations presented here provided a detailed insight into instantaneous gas transfer processes. The falling plumes with highly gas-saturated fluid in their core were found to penetrate deep inside the bulk. With an initial temperature difference between the water surface and the bulk of slightly above 2 K peaks in the instantaneous heat flux in excess of 1600 W/m^2 were observed, proving the potential effectiveness of buoyant convective heat and gas transfer. Furthermore, the validity of the scaling law for the ratio of gas and heat transfer velocities $K_L/H_L \propto (Pr/Sc)^{0.5}$ for the entire range of Schmidt numbers considered was confirmed. A good time-accurate approximation of K_L was found using surface information such as velocity fluctuations and convection cell size or surface divergence. A reasonable time-accuracy for the K_L estimation was obtained using the horizontal integral length scale and the root-mean-square of the horizontal velocity fluctuations in the upper part of the bulk.

Key words:

1. Introduction

During the last decade gas transfer across the air-water interface received increasing interest because of its importance to the global green house gas budget. At the air-water interface, heat and gas fluxes are dominated by molecular diffusion, which can be enhanced significantly by convective mixing. In nature three sources of near-surface

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convective mixing can be identified. The first and most studied one is mixing by wind-shear induced turbulence at the water surface. The second is turbulence generated at the bottom of a stream or by ocean currents that subsequently diffuses upwards to the surface where it enhances turbulent mixing. The third and least studied mechanism is due to a buoyant-convective instability which is responsible for mixing during low wind-speed conditions. In the ocean, free convection due to buoyant instabilities is often caused by vertical temperature and/or salinity gradients. This free convection induces near-surface turbulence which enhances the exchange of heat and green-house gases (in particular carbon dioxide) between the atmosphere and the oceans.

Wind-shear induced mixing has attracted significant attention because turbulence introduced immediately at the water surface is a very effective mechanism for promoting gas transfer. Especially larger wind-speeds can induce significant wave-growth and wave breaking which further aid the mixing process (MacIntyre *et al.* 1999). Hence, in most cases the predictive estimation of heat and green-house gas fluxes only takes wind-shear into account, which often results in an over- or underprediction of the actual gas flux, especially in low wind conditions (e.g. Crusius & Wanninkhof 2003; Nightingale *et al.* 2000; Smith *et al.* 2011).

Sheltered inland water bodies and equatorial ocean regions are examples of regions with low wind-speed conditions that typically have large air-water CO₂ concentration gradients. These regions contribute significantly to the global CO₂ budget (Feely *et al.* 2002), both as a major source of climatically-important gases (in lakes and reservoirs : Cole *et al.* 2007; MacIntyre *et al.* 2010) as well as a major sink for CO₂ (in the equatorial ocean : Ishii *et al.* 2014). MacIntyre *et al.* (2010) showed that for wind speeds lower than 5 m/s the gas transfer velocity K_L is independent of wind-speed and the buoyancy-driven flux dominates. In the ocean (Soloviev & Klinger 2001; Rutgersson *et al.* 2011) have reported that in low wind-speed conditions buoyant convection due to surface cooling contributes significantly to the surface gas exchange.

Another important example of buoyant convective mixing in nature is the absorption of oxygen into lakes or reservoirs driven by surface cooling. At night, the gas-saturated surface layer is cooled and becomes slightly heavier than the water underneath. Small fluctuations may trigger an instability resulting in cold oxygen-saturated surface water plunging down and being replaced by warmer unsaturated fluid from below. This process results in the reaeration of sheltered bodies of water, which is needed for bio-degradation of organic wastes.

In order to improve current models used for the prediction of gas fluxes there is a need for a detailed understanding of buoyant-convective mixing processes. To explore the effect of these processes requires an in-depth investigation of the interaction between the near-surface flow field and the interfacial mass flux which - in contrast to the relatively abundant empirical quantification of the gas transfer velocity - is currently still lacking.

The gas flux across the air-water interface j_z is defined by

$$j_z = K_L(C_s - C_b), \quad (1.1)$$

where K_L is the transfer velocity, C_s is the saturated gas concentration at the interface and C_b is the gas concentration in the bulk. The transfer velocity K_L is governed by the complex interaction of molecular diffusion and turbulent transport on either side of the surface and also depends on surface conditions as well as the chemical reactivity of the dissolved gas. At a constant temperature Henry's law states that at equilibrium the solubility of an atmospheric gas in water increases linearly with its partial pressure in air, p_a , so that $C_s = p_a/H$, where H denotes Henry's constant. (Note that the evaluation of C_s is important in order to accurately determine j_z .) The partition coefficient of

a gas between the atmosphere and water is defined by $\mathcal{K} = \frac{C_g}{C_s}$, where C_g is the gas concentration in the air. (For an ideal gas, \mathcal{K} is related to H by $\mathcal{K} = \frac{H}{RT}$, where R is the gas constant and T is the temperature.) As shown by Liss (1973), the total transfer velocity K_L can be related to the individual gas transfer coefficients (k_L and k_g for the liquid and gas phase, respectively) by

$$\frac{1}{K_L} = \frac{1}{k_L} + \frac{1}{\mathcal{K}k_g}. \quad (1.2)$$

This "resistance-in-series model" shows that the ratio $k_L/\mathcal{K}k_g$ determines whether the concentration boundary layer on the air or on the water side controls the gas transfer process (offers more resistance). With increasing solubility (which may be caused by a decreasing temperature) the control of the atmospheric gas transfer process will gradually shift to the air side. Environmentally important gases typically have a low (O_2 , CO , NO) to moderate (CO_2) solubility. Because of this and their significantly lower diffusivities in water than in air (hence $k_L \ll k_g$), for such gases $k_L/\mathcal{K}k_g$ is small and their transfer process at the air-water interface is concentrated in an extremely thin boundary layer ($10 - 1000\mu m$) on the liquid side (see Liss 1973; Jähne & Haussecker 1998). It should be noted that CO_2 is a chemical reactive gas in water and can be hydrated at the water surface which chemically enhances the transfer velocity (see Wanninkhof & Knox 1996).

The small size of the concentration boundary layer on the liquid side makes it extremely difficult to carry out direct measurements of the diffusive ($-D\frac{\partial C}{\partial z}$) and turbulent ($w''C''$) vertical mass fluxes, where C is the concentration of the solute, w is the velocity in the vertical (z) direction and D is the molecular diffusion coefficient (double primes denoting fluctuations with respect to statistical averages). Together, the diffusive and turbulent mass fluxes determine the total mass flux j_z in the vertical direction. The mass flux averaged over horizontal planes is a function of depth (z) and time (t), as given by

$$\langle j_z \rangle (z, t) = - \left[D \frac{\partial \langle C \rangle}{\partial z} - \langle w'' C'' \rangle \right]. \quad (1.3)$$

Because it is difficult to perform measurements inside the tiny layer of gas-saturated fluid at the water surface, previous measurements focused mainly on measuring the overall transfer velocity K_L . Equations (1.1) and (1.3) imply that to gain a fundamental understanding of the overall transfer velocity K_L it is needed to resolve both the diffusive and turbulent mass transfer terms.

Numerous studies proposed empirical models which relate the transfer velocity K_L to global parameters such as the bulk velocity of a stream, the bottom slope and friction (e.g. Tsivoglou & Wallace 1972; Chu & Jirka 2003). As mentioned above, for the case where buoyancy is likely to dominate, K_L has been mostly related to wind-shear alone (e.g. Liss & Slater 1974; Nightingale *et al.* 2000; Crusius & Wanninkhof 2003). This has led to considerable inaccuracies in the estimation of K_L , especially for wind speeds below 5 m/s.

Furthermore, previous studies on buoyant-convective induced flow were mainly concerned with the heat flux and flow patterns in a fluid layer confined by a heated plate at the bottom and a cooled plate at the top (Amati *et al.* 2005; Verzicco & Sreenivasan 2008; Schumacher 2009), where no-slip boundary conditions apply. In many applications in nature such conditions do not apply, and as the current work is motivated by the interfacial mass transfer at a gas-liquid interface, a flat surface assumption with a free-slip boundary condition is far more appropriate to model the water surface.

While Rayleigh-Bénard convection between two solid boundaries has been studied quite extensively, buoyant convection induced by a cooled water surface has received much less

attention. Spangenberg & Rowland (1961) were among the first who visualized the surface and vertical temperature patterns during evaporative cooling using the schlieren imaging technique. They observed that the cool surface water forms a net-like pattern when it collects along lines as it plunges down forming vertical sheets. Similar observations were made by Katsaros *et al.* (1977). The infrared images of Volino & Smith (1999), however, did not reveal any net-like pattern at the surface. Instead, they only observed regions with vortical structures, which could be explained by their relatively high Rayleigh number and/or the presence of surface contamination. Flack *et al.* (2001), for instance, observed that surface contamination can significantly affect the surface temperature pattern. A detailed investigation on how the surface pattern in buoyancy-driven flow is affected by surface contamination, Rayleigh number, or Schmidt number is still lacking.

Only a limited number of quantitative investigations of this problem were performed. Deadorff *et al.* (1969) measured the mean temperature profile of the bulk water and horizontal length scales, while Katsaros *et al.* (1977) conducted temperature measurements within the boundary layer. The latter found that the boundary layer profile was highly nonlinear due to unsteadiness. Bukhari & Siddiqui (2006) used PIV to measure the turbulent structure below the surface, while Volino & Smith (1999) performed simultaneous temperature and PIV measurements at several horizontal and vertical cross-sections of the fluid. They observed that the surface temperature did not always correlate well with the velocity field.

While all buoyant-convective flow investigations studied the evolution of flow structures, only a limited number related buoyant convection to gas exchange across the air-water interface. For instance, Eugster *et al.* (2003) and MacIntyre *et al.* (2010) measured the CO₂ exchange rate between the atmosphere and a lake. Their quantification of the global K_L confirmed the importance of buoyancy-driven convective mixing in relation to interfacial gas-transfer. Schladow *et al.* (2002) performed gas transfer measurements in the laboratory using LIF, where the buoyant-convective instability was triggered by surface cooling. The synoptic 2D-PIV-LIF measurements of Jirka *et al.* (2010) showed that the oxygen-rich fluid which is transported by cold-plumes from the surface into the bulk does initially form mushroom-like structures, which eventually deform into an anchor-like shape. To further elucidate the actual dynamics of the falling plumes, three-dimensional visualizations are needed. Also, the accuracy of the laboratory measurements was insufficient to resolve the minute fluctuations in C'' and w'' which is needed to quantify directly the turbulent mass flux and to relate the near surface turbulence to the mixing properties.

Direct numerical simulations (DNS) related to heat/mass transfer across the free surface of an open channel flow have been performed by various research groups (Nagaosa 1999; Yamamoto *et al.* 2001). These simulations showed a correlation between the vortices ejected from the bottom region and the near-surface concentration field and proved the usefulness of DNS in obtaining detailed statistics of the scalar fields. Other researchers, such as Kunugi & Satake (2002), simulated the gas transfer promoted by wind-shear-generated turbulence. They were able to establish a relationship between low-speed streaks and the transfer of CO₂ from the interface into the bulk region. Magnaudet & Calmet (2006) used large-eddy simulations (LES) to analyse the concentration and flow field in the near-surface region of an open channel flow at a high Reynolds number. They reported large-scale turbulent upwelling and downwelling motions. Because of the high Reynolds number and the relatively large size of the computational domain, the numerical resolution of the very thin diffusive sublayers was only marginal. Kermani & Shen (2009) used DNS to study characteristics of interfacial mass transfer driven by free-surface turbulence. They focused on the quantification of the surface age related to

the surface renewal models pioneered by Higbie’s penetration theory (Higbie 1935) and further elaborated by Danckwert’s random surface renewal model Danckwerts (1951). These models assume that the turbulence in the bulk region of the fluid transports the low-gas-saturated fluid portion up to the surface where gas transfer takes place. Hasegawa & Kasagi (2009) performed a hybrid DNS/LES numerical simulation of a coupled air-water turbulent flow and associated mass transfer to investigate the relation between the interfacial mass transfer and the surface divergence. Khakpour *et al.* (2011) performed DNS with turbulent shear flow to study the effect of a surfactant contaminated surface on the gas transfer process. It should be noted that due to limitations in computing capacity, the above DNS studies were mostly conducted at low Schmidt numbers (less than 10). Recently, Herlina & Wissink (2014) carried out a series of DNS to study the effect of bottom-shear-induced turbulence on interfacial gas transfer at realistic Schmidt numbers. To resolve the scalar flow field for Schmidt numbers up to $Sc = 500$ a dual mesh strategy was implemented together with a fifth-order accurate WENO scheme for scalar convection, see Kubrak *et al.* (2013).

This paper reports on the results obtained in a series of DNS calculations of interfacial gas transfer driven by buoyant convection. The heat and mass transfer at a Prandtl number of $Pr = 6$ and Schmidt numbers up to $Sc = 500$, which are relevant for the transport of atmospheric gases in water, were calculated using a slightly adapted version of the code by Kubrak *et al.* (2013). The complex mass transfer process is elucidated by fully resolving the thin concentration boundary layer at the air-water interface as well as the thin filaments present in the deeper bulk region. This allowed us to carry out an in-depth investigation into the dynamics of the falling sheets and plumes that are responsible for the heat and mass transfer from the surface into the bulk. Also, the growth rate of the convection cells in time was investigated as well as the relation between the flow field, the temperature and the gas concentration. Finally, the dependence of the transfer velocity K_L on the Schmidt number, Rayleigh number, typical surface information (e.g. the average size of the convection cells and the surface divergence), and typical bulk information (e.g. bulk integral length scale and rms of the velocity) was assessed.

2. Computational aspects

2.1. Numerical Method

The problem under consideration is the gas transfer across the air-water interface driven by buoyant convection caused by temperature differences between the water surface and the bulk. The initial temperature difference is assumed to be small, in our case $T_{b,0} - T_s$ was 2.2K, where $T_{b,0}$ and T_s are the temperatures at the bulk and the surface, respectively. In this region the density is approximately a linear function of the temperature and the Boussinesq approximation is used to describe the effect of the small variations in density on the fluid flow.

The fluid motion is governed by the non-dimensional Navier-Stokes equations. As in Balachandar (1992) the flow is driven by an unstable temperature gradient and the non-dimensionalization is carried out using a length scale L and a velocity scale $U = \kappa/L$, where $\kappa = 0.146 \times 10^{-6} \text{ m}^2/\text{s}$ is the thermal diffusivity of water at 298.15K.

Note that this study only concerns the initial development of the Rayleigh instability, hence the length scale is chosen to be fixed at a value of $L = 0.00925 \text{ m}$ that is independent of the actual depth of the computational domain. The resulting continuity equation reads

$$\frac{\partial u_i}{\partial x_i} = 0, \quad (2.1)$$

and the momentum equations are given by

$$\frac{\partial u_i}{\partial t} + \frac{\partial u_i u_j}{\partial x_j} = -\frac{\partial p}{\partial x_i} + Pr \frac{\partial^2 u_i}{\partial x_j \partial x_j} - Ra_L|_{t=0} Pr (1 - T^*) \delta_{i3} \quad i = 1, 2, 3 \quad (2.2)$$

where $x_1 = x$, $x_2 = y$ are the horizontal directions and $x_3 = z$ is the vertical direction, $u_1 = u$, $u_2 = v$ and $u_3 = w$ are the velocities in the x , y , and z directions, respectively, p is the pressure and t denotes time. The last term on the right hand side represents the buoyancy force in the z -direction where δ_{i3} is the Kronecker delta,

$$T^* = \frac{T - T_s}{T_{b,0} - T_s} \quad (2.3)$$

is the non-dimensional temperature, $Pr = \frac{\nu}{\kappa} = 6$ is the Prandtl number corresponding to the ratio of the momentum and the thermal diffusivities of water at 298.15K and

$$Ra_L = \frac{\alpha \Delta T g L^3}{\kappa \nu} \quad (2.4)$$

is the macroscale Rayleigh number in which α is the thermal expansion factor in K^{-1} , $g = 9.81 \text{ m/s}^2$ is the gravitational acceleration and $\Delta T = (T_b - T_s)$ is the temperature difference between the bulk and the surface. Note that because of the adiabatic boundary conditions at the bottom, T_b depends on time. In (2.2), $Ra_L|_{t=0}$ is calculated using $T_b = T_{b,0}$.

The non-dimensional transport equation of the temperature T^* is given by

$$\frac{\partial T^*}{\partial t} + \frac{\partial u_j T^*}{\partial x_j} = \frac{\partial^2 T^*}{\partial x_j \partial x_j}. \quad (2.5)$$

Similarly, the passive scalar transport is governed by the dimensionless three-dimensional convection-diffusion equation of the non-dimensional scalar concentration C^*

$$\frac{\partial C^*}{\partial t} + \frac{\partial u_j C^*}{\partial x_j} = \frac{Pr}{Sc} \frac{\partial^2 C^*}{\partial x_j \partial x_j}, \quad (2.6)$$

where

$$C^* = \frac{C - C_{b,0}}{C_s - C_{b,0}}, \quad (2.7)$$

with $C_{b,0}$ and C_s defined analogously to $T_{b,0}$ and T_s , respectively. In the remainder of the paper T , C rather than T^* , C^* will be used to refer to the non-dimensional temperature and scalar concentration, respectively. To improve clarity, time will be displayed in seconds. Unless mentioned otherwise, T_b and C_b will be evaluated at $\frac{1}{2}d$, where d is the domain depth.

For the simulations, the full set of governing equations mentioned above was solved using the code described in Kubrak *et al.* (2013). The solver was developed specifically with the aim to resolve the details of low-diffusivity scalar transport, which is governed by a very thin boundary layer at the air-water interface. For the scalar transport a fifth-order accurate WENO scheme (Liu *et al.* 1994) for the convection was combined with a fourth-order accurate central method for the diffusion in order to accurately capture high concentration gradients. For the time-integration of the scalar a three-stage Runge-Kutta scheme was employed. To solve the incompressible 3D Navier-Stokes equations for the fluid flow, a central finite-difference approach with a fourth-order accurate discretization of the diffusion and a fourth-order-accurate kinetic energy conserving discretization of the convection (Wissink 2004) was used. The spatial discretization was performed on a non-uniform mesh using a staggered variable arrangement, where the scalar concentrations,

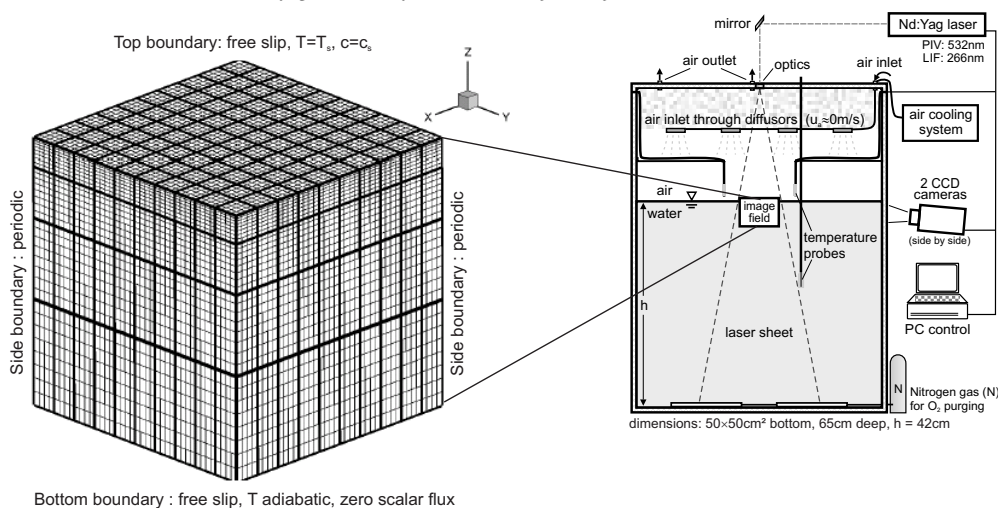


FIGURE 1. Schematic of the computational domain (left). Only part of the experimental domain (right) near the surface is modeled using a reduced depth.

temperature and pressure were all defined in the middle of the grid cells. The time-integration was performed using the second-order-accurate Adams-Bashforth method. The Poisson equation for the pressure, which is obtained after substituting the discretized momentum equation into the continuity equation, was solved using a conjugate gradient solver with a simple diagonal preconditioning.

As the scalar diffusivity is much smaller than the momentum diffusivity, significantly more grid points are needed to accurately resolve the evolution of the scalar. To deal with this in a computationally efficient manner a dual-meshing strategy was employed in which the scalar concentration field was solved on a finer mesh than the velocity field.

2.2. Overview of simulations

The setup of the numerical simulations was inspired by earlier experiments performed at KIT (Jirka *et al.* 2010). In the experiments, the water temperature was held fixed at 293.15K and oxygen was stripped out. The body of water was subsequently exposed to a cold gas at its surface leading to a temperature difference between bulk and surface. As a result, a thin thermal boundary layer of cool water formed adjacent to the air-water interface, with an even thinner layer of gas-saturated water at the top. As the cool water is slightly heavier than the water in the bulk region below, small disturbances may trigger a Rayleigh instability by which the cold water (partially saturated with gas) from above will penetrate the bulk region below thereby interchanging unsaturated water with saturated water. While the experiments were performed in a $50 \times 50 \times 65 \text{ cm}^3$ tank with a water depth of 42 cm, the computational domain in the simulations only covered part of the physical domain, as illustrated in Fig.1. The domain and grid sizes used in the various simulations are given in table 1. The mesh is stretched in the z -direction in order to obtain a finer resolution near the interface. The node distribution is given by

$$z(k) = \left[1 - \frac{\tanh(z_\phi)}{\tanh(z_1)} \right] z(0) + \left[\frac{\tanh(z_\phi)}{\tanh(z_1)} \right] z(n_z) \quad (2.8)$$

for $k = 1, \dots, n_z - 1$, with

$$z_1 = \sigma/2,$$

Run	Sc	Domain	Base mesh size	f_{RS}	$\alpha\Delta T$	$Ra_L _{t=0}$
VR1	–	$5L \times 5L \times 5L$	$280 \times 280 \times 184$	1	0.000572	34000
VR2	–	$5L \times 5L \times 5L$	$400 \times 400 \times 256$	1	0.000572	34000
VR3	–	$5L \times 5L \times 5L$	$500 \times 500 \times 320$	1	0.000572	34000
SR1	500	$5L \times 5L \times 5L$	$400 \times 400 \times 256$	1	0.000572	34000
SR2	500	$5L \times 5L \times 5L$	$400 \times 400 \times 256$	2	0.000572	34000
SR3	500	$5L \times 5L \times 5L$	$400 \times 400 \times 256$	3	0.000572	34000
BC1	20; 500	$5L \times 5L \times 5L$	$400 \times 400 \times 256$	1; 3	0.000572	34000
BC2	500	$5L \times 5L \times 10L$	$400 \times 400 \times 400$	3	0.000572	34000
BC3	20; 500	$10L \times 10L \times 10L$	$800 \times 800 \times 512$	1; 3	0.000572	34000
BC4	20; 500	$10L \times 10L \times 5L$	$800 \times 800 \times 352$	1; 3	0.000572	34000
BC5	20; 500	$5L \times 5L \times 5L$	$400 \times 400 \times 256$	1; 3	0.000381	23000

TABLE 1. Overview of the simulations. In all simulations the Prandtl number was set to $Pr = 6$ (typical for water at 298.15K), f_{RS} is the refinement factor for the scalar mesh, $\alpha\Delta T$ is the expansion factor and $Ra_L|_{t=0}$ is the initial macroscale Rayleigh number. Note that the conclusions presented in this paper are based on results mostly from BC3.

$$z_\phi = \frac{k z_1}{n_z}$$

where n_z is the number of nodes in the z -direction. The stretching is controlled by the parameter σ , which is set to $\sigma = 3$ in all simulations.

As mentioned above, a dual-meshing strategy is employed in which the gas-concentration field is solved on a finer mesh (dual grid) than the velocity and temperature fields. The boundary conditions used in the simulations are illustrated in figure 1 (left). In the horizontal directions periodic boundary conditions were employed for all variables. For the velocity field, free-slip boundary conditions were used at the top and bottom of the computational domain. Hence, fluid is blocked from crossing the bottom boundary. The scalars (T and C) at the top were fixed at $T = 0$ and $C = 1$, respectively, while at the bottom a zero scalar flux was assumed. Expansion factors $\alpha\Delta T = 0.000381$ and 0.000572 were employed. For water at 298.15K this corresponds to temperature differences between the water surface and the bulk of $\Delta T = 1.48\text{K}$ and 2.22K , respectively.

Initially at $t = 0$ the velocity field was set to zero. Below the top boundary, the concentration field was set to $C = 0$ and the temperature field to $T = 1$. After allowing the thermal and concentration boundary layers to develop for $t = 9.6\text{s}$, small random disturbances (uniformly distributed between $T/\Delta T = 0$ and 0.020) were added to the temperature field to trigger the instability.

2.3. Grid refinement study

According to the Grötzbach criterion (Grötzbach 1983), a scalar is sufficiently resolved if the geometric mean of the grid size, $\bar{\Delta} = \sqrt[3]{\Delta x \times \Delta y \times \Delta z}$, is smaller than πL_B , where L_B is the Batchelor scale defined by

$$L_B = (\nu^3/\epsilon)^{0.25} Pr^{-0.5} \text{ and } L_B = (\nu^3/\epsilon)^{0.25} Sc^{-0.5}, \quad (2.9)$$

for the temperature and mass transport, respectively. It can be seen in table 2 that in all simulations the geometric mean of the grid size is less than πL_B and near the air-water interface the vertical size of the grid cells is significantly smaller than the thermal and concentration boundary layer thicknesses so that all employed meshes fulfill the Grötzbach criterion.

To further ensure an accurate resolution of all near-surface details of the buoyant

Run	Δz (L)	Δz_R (L)	$\overline{\Delta}/\pi$ (L)	$\overline{\Delta_R}/\pi$ (L)	Pr_6	L_B (L) Sc_{20}	Sc_{500}	Pr_6	$\delta_{cf,min}$ (L) Sc_{20}	Sc_{500}
BC1	0.0059	0.0020	0.00310	0.0010	0.0353	0.0193	0.0039	0.0847	0.0478	0.0097
BC2	0.0075	0.0025	0.0034	0.0011	0.0313	—	0.0034	0.0839	—	0.0098
BC3	0.0059	0.0020	0.00310	0.0010	0.0353	0.0193	0.0039	0.0836	0.0467	0.0096
BC4	0.0059	0.0020	0.00310	0.0010	0.0353	0.0193	0.0039	0.0868	0.0480	0.0097
BC5	0.0059	0.0020	0.00310	0.0010	0.0406	0.0222	0.0044	0.100	0.0541	0.0110

TABLE 2. Comparison of the mean-width and near-surface vertical grid spacings with the Batchelor scale and boundary layer thicknesses, where Δz and Δz_R are the vertical sizes at the surface of the base-grid and the refined grid, respectively.

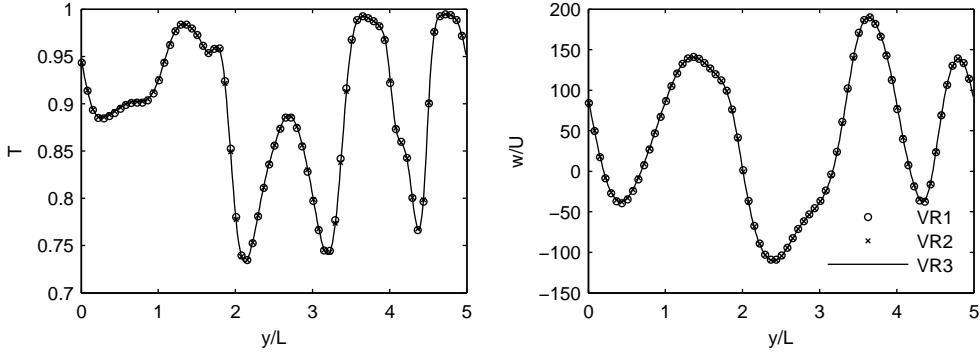


FIGURE 2. Extracted temperature (a) and w -velocity component (b) profiles after 48.1s at $z/L = 3$ and $x/L = 2.5$ obtained using different mesh-sizes. Only every fourth data point is shown.

convective instability down to the Batchelor scale a rigorous grid refinement study for the base mesh was performed. To facilitate this study, in a separate simulation, similar to VR2 (see table 1), random disturbances were added to the temperature field at $t = 9.6$ s. This simulation was subsequently run for a further 1.9s after which the resulting perturbed temperature field was saved and used as the initial condition (starting the simulations at $t = 11.5$ s) for the temperature in VR1, VR2 and VR3. This procedure allows a direct comparison of instantaneous flow and temperature fields calculated on different meshes. VR1 has the coarsest mesh with $280 \times 280 \times 184$ points, the typical mesh used in VR2 is approximately 1.4 times finer in each direction while the mesh used in VR3 is 1.8 times finer in each direction. Typical results obtained after $t = 48.1$ s are shown in figure 2, It can be seen that the profiles of both the temperature and the w -velocity component at $z/L = 3$ and $x/L = 2.5$ are in excellent agreement. Similar observations were made for profiles extracted at other locations as well as for other velocity components. Hence, it can be concluded that the mesh used in VR2 is sufficiently fine to resolve the buoyancy induced instability.

As the scalar diffusivity is much smaller than the momentum and thermal diffusivities, a refined mesh is used to be able to accurately resolve the scalar concentration for Schmidt numbers up to $Sc = 500$. To test which refinement factor is needed, a series of simulations (SR1, SR2, SR3) with the same $400 \times 400 \times 256$ base-mesh and refinement factors of $f_{RS} = 1, 2, 3$ were performed. In figure 3 a snapshot at $t = 52.9$ s of the scalar concentration

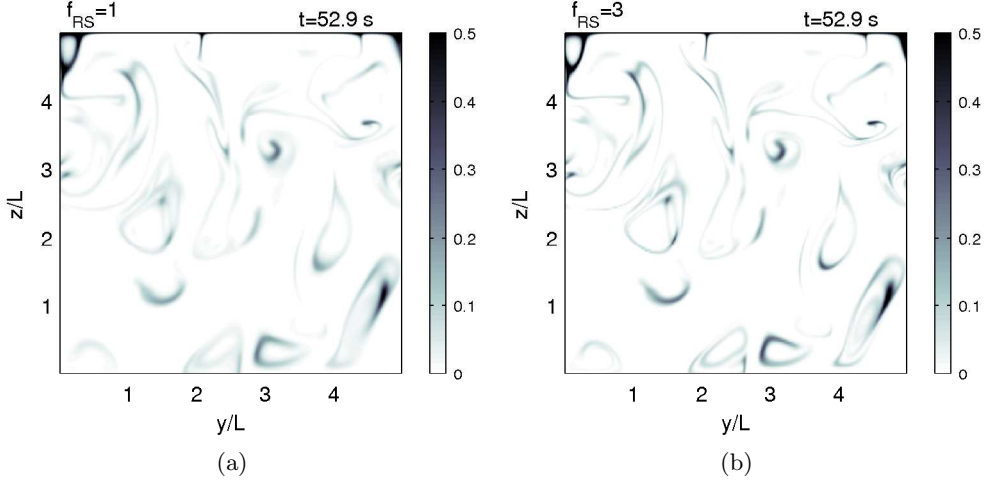


FIGURE 3. Snapshot of concentration field ($Sc = 500$) at $t = 52.9$ s and $x/L = 2.5$ resolved using scalar-meshes with refinement factor a) $f_{RS} = 1$ and b) $f_{RS} = 3$. For clarity the color coding is truncated at $C=0.5$.

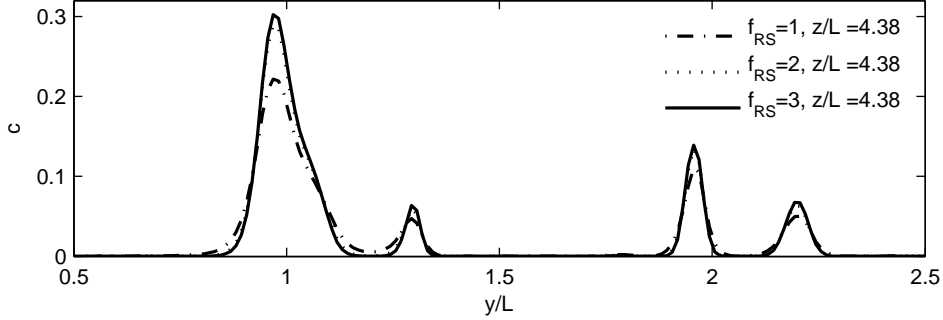


FIGURE 4. Extracted concentration profile ($Sc = 500$) at $t = 52.9$ s, $x/L = 2.5$, $0.5 \leq y/L \leq 2.5$ and $z/L = 4.38$ from simulations with different scalar-mesh refinement factors (SR1, SR2 and SR3).

($Sc = 500$), obtained in SR1 and SR3 at $x/L = 2.5$ is presented. The snapshot shows the ability of the employed code to resolve regions where steep gradients occur without introducing spurious oscillations, including in the deeper bulk region where the mesh-size becomes relatively coarse (see figure 1). In figure 4 the cross-sectional profiles at $x/L = 2.5$, $z/L = 4.38$ obtained in SR1, SR2 and SR3 are compared. The resolution is significantly improved when increasing the scalar-mesh refinement factor from 1 to 2, while an increment from 2 to 3 yields only a slight further improvement. In the near surface region, which is our actual area of interest, the results using $f_{RS} = 2$ and $f_{RS} = 3$ are nearly identical. Thus, a refinement factor $f_{RS} \geq 2$ is deemed to be sufficient to accurately resolve the scalar transport.

3. Restriction on simulation time due to domain size

The simulations aim to elucidate the physics of gas transfer across the air-water interface driven by buoyant convection in deep calm bodies of water. To fully describe this

phenomenon, the size of the computational domain would have to be large enough to allow all cold water plumes to move downward until they naturally lose their negative buoyancy. Unfortunately, the current computational resources are insufficient to fully simulate this problem. Hence, it was decided to concentrate on the initial development of the Rayleigh instability and the early stages of transition, including the formation of convection cells and their influence on interfacial gas transfer. The maximum simulation time, during which the physics does not depend on the size of the computational domain, was determined by performing simulations using computational domains with a variety of sizes.

To determine the maximum simulation time during which bottom-effects are negligible, two simulations were performed with different depth-sizes (BC1 and BC2) using exactly the same disturbance to the initial temperature field at $t = 9.6$ s. As illustrated by the snapshots in figures 5a and b, the surface temperature contours of simulations BC1 and BC2 up to $t = 57.7$ s were found to be identical even after the first plumes reached the bottom of the domain of simulation BC1 at $t = 48.1$ s. When comparing figures 5c and d, a very slight difference can be seen close to the bottom of BC1 in the snapshots at $t = 57.7$ s showing temperature contours in the plane $x = 2.5L$ from simulations BC1 and BC2. Higher up in the domain, however, the contours were found to be in very good agreement. Figure 6a shows that also the w'_{peak} versus time profiles are identical up to $t = 57.7$ s, where w'_{peak} is the maximum of the vertical $\sqrt{\langle w'^2 \rangle}$ profile, with $\langle \cdot \rangle$ denoting horizontal averaging. Hence, in the free-fall regime the vertical velocity in BC1 and BC2 does not depend on depth. This regime starts at $t \approx 29.3$ s and ends when the bottom effects begin to influence the flow dynamics. As illustrated in figure 6b, the effect of the bottom on the total heat flux is felt first at $t = 63.5$ s which is about 15.4 s after the plumes reached the bottom of the shallower domain (BC1). The results also suggest that for deep-calm water bodies the near surface characteristics including the heat flux and the topology of the convection cells at the surface only depend on the buoyancy flux, not on depth.

The results from the simulations with a larger horizontal domain (BC3 and BC4) are also included in figure 6. Because the initial disturbances added to the temperature field were not identical (though of the same level), these results do not show the same degree of agreement as in BC1 and BC2. By extrapolating these results, it is to be expected that after the plumes reached the bottom and the fall velocity becomes constant, the convection cell size would continue to increase in size provided there are no horizontal restrictions.

As the focus in this paper is on gas transfer induced by a buoyant-convective instability in deep waters the remainder of the paper will mainly focus on the near surface fluid flow and the scalar transport process before bottom effects become noticeable. Unless mentioned differently, the results presented below are all taken from the largest simulation BC3.

4. Results

4.1. Visualization of thermal structures

Three-dimensional structures

The sequence of snapshots in figure 7 shows isosurfaces of the temperature field from BC3. It illustrates the 3D transport process driven by a buoyant-convective instability that occurs due to surface cooling. At first heat transfer is dominated entirely by molecular diffusion leading to a thickening of the thermal (cooler) boundary layer (not shown

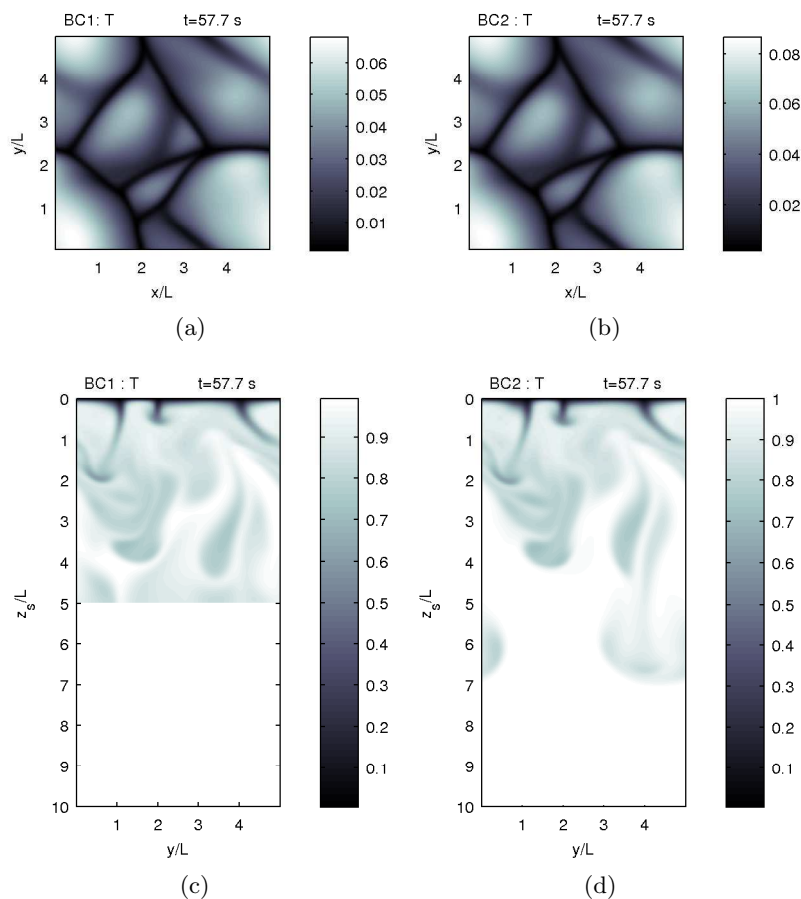


FIGURE 5. Evaluation of the effect of the domain size. Near the surface identical snapshots of the temperature fields in BC1 and BC2 are observed up to $t = 57.7$ s.

here). In this unstably stratified configuration small fluctuations added to the temperature field at $t = 9.6$ s trigger an instability that is characterized by the formation of thin falling "sheets" of saturated, cold water. To replenish the falling cold water, warmer water from the bulk starts to flow upward resulting in the formation of convection cells. In the interior of the cells one or more sources of upwelling warm water exist. Once the warm water reaches the surface, it moves radially outward approximately parallel to the surface. While flowing along the surface the warm water is cooled and slowly becomes increasingly heavy until it starts falling down usually in the form of vertical sheets (see also Spangenberg & Rowland 1961) along the edges of the cells at a relatively high velocity as can be seen by the coloring of the isosurface. At intersections of three or more convection cells the added quantity of cold water increases the local sink speed, which results in the formation of mushroom-like plumes that are clearly visible in the snapshots at $t = 38.9$ s and $t = 42.8$ s. As the plumes sink, they tend to rotate about their axis thereby distorting the mushroom-like shape and changing it into something resembling a blade. The rotational movement appears to enhance the stretching and downward movement of the blade-shaped plumes. The latter is reflected in figure 6a where w'_{peak} (after a transient period of $t \approx 43.3$ s) does appear to increase somewhat in time as long as the

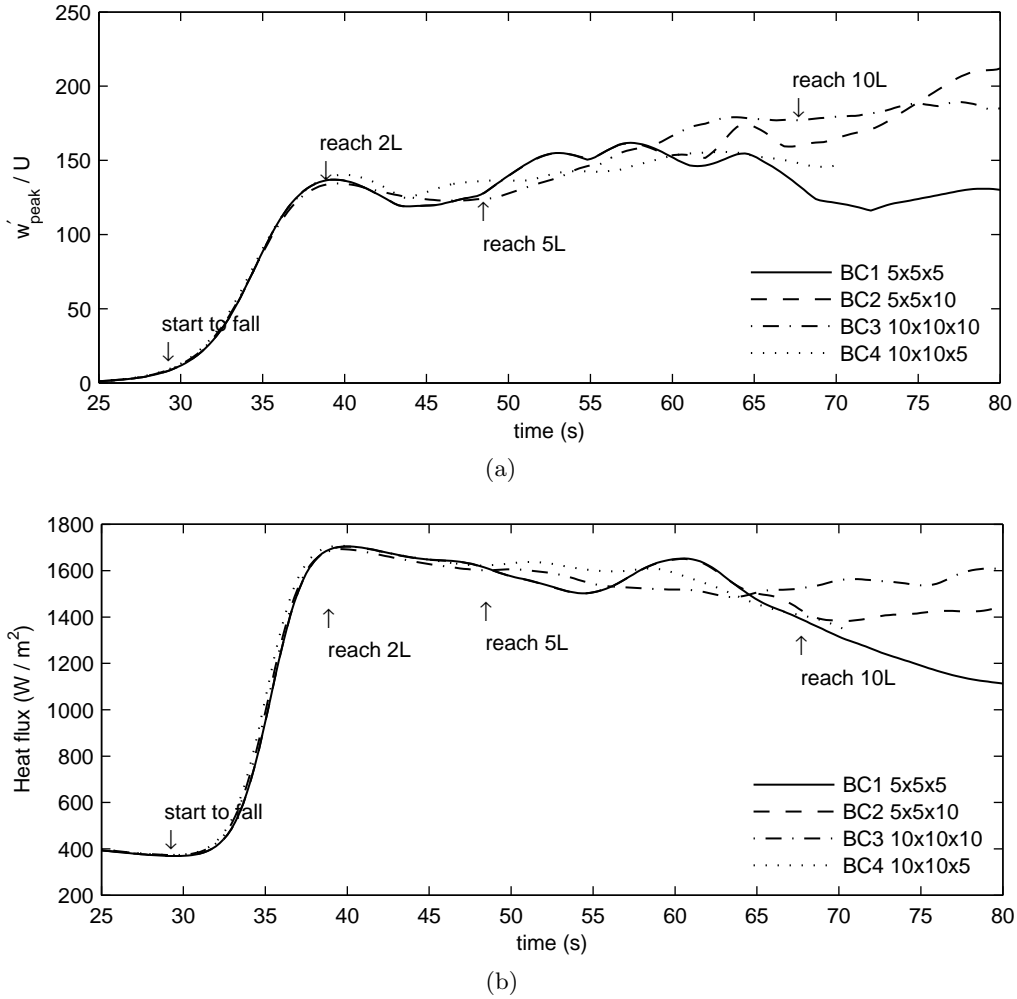


FIGURE 6. Evaluation of the effect of the domain size. BC1 and BC2 were started using exactly the same disturbance to the initial temperature field. Also shown are simulations BC3 and BC4 which have a larger horizontal domain and different initial disturbances to the temperature field. a) variation of the maximum vertical velocity fluctuation w'_{peak} with time. b) Variation of the dimensional heat flux with time. The heat flux was computed based on the temperature gradient near the surface, see (4.5).

bottom is not reached. As the convection cells grow, separate plumes merge into larger plumes and separate sheets into larger sheets.

Note that a sheet typically spans the joint edge of two convection cells while a blade originates from a plume that forms at the intersection of three or more convection cells and typically has a much smaller width and larger penetration depth than a sheet.

Top structures : netlike pattern, size of convection cells

Snapshots of the non-dimensionalized temperature distribution just beneath the top boundary, at a distance of $z_s = 0.0029L$ from the surface, are shown in figure 8. The convection cells discussed above are responsible for the formation of this typical net-like pattern at the surface. The snapshots show that the convection cells tend to grow in

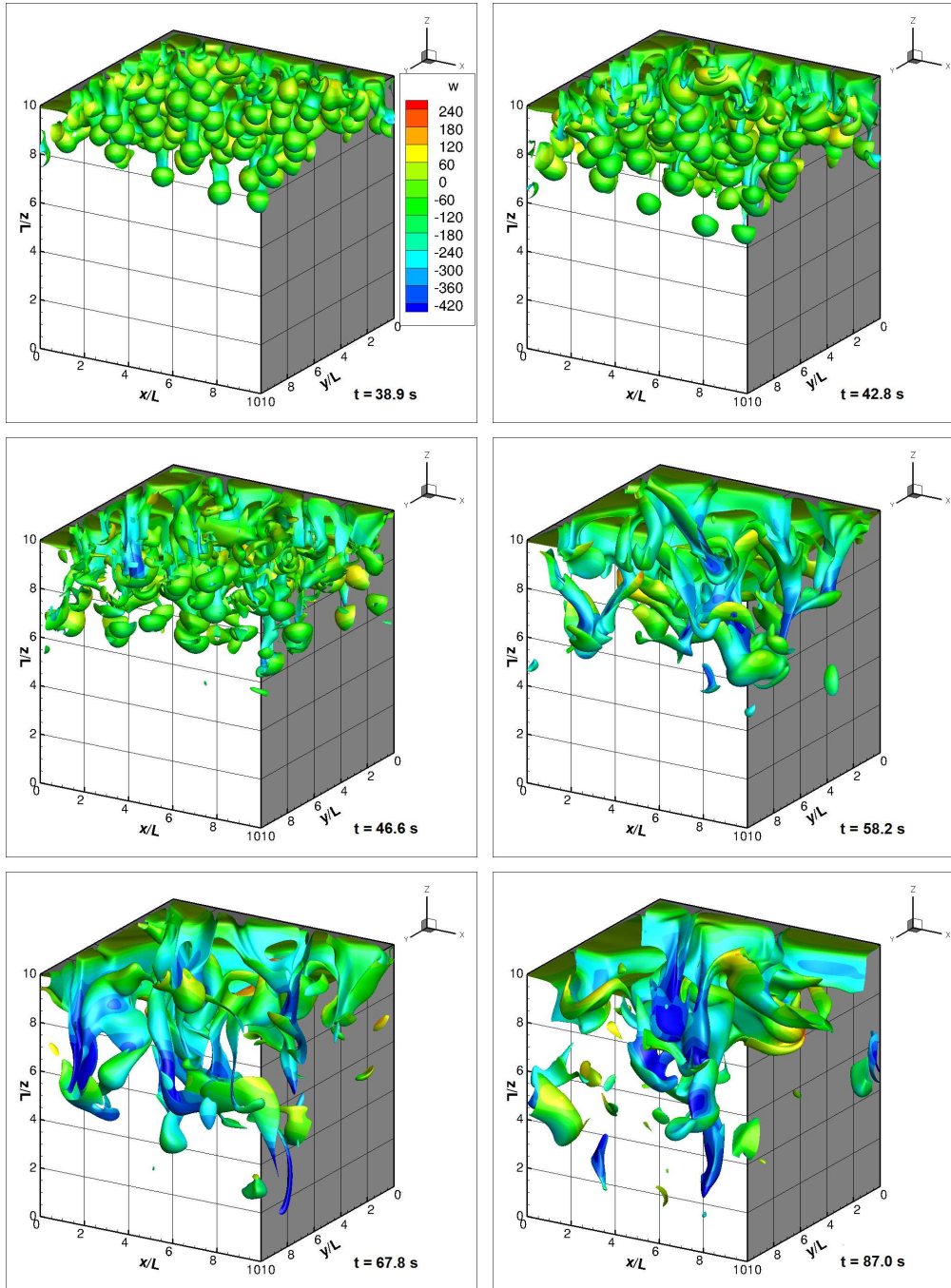


FIGURE 7. Isosurface of temperature ($Pr = 6$) field from simulation BC3 at $T = 0.75$ (with T non-dimensionalised such that $T = 0$ is the coldest water at the surface and $T = 1$ is the initial temperature of the warmer bulk water) coloured by the vertical velocity.

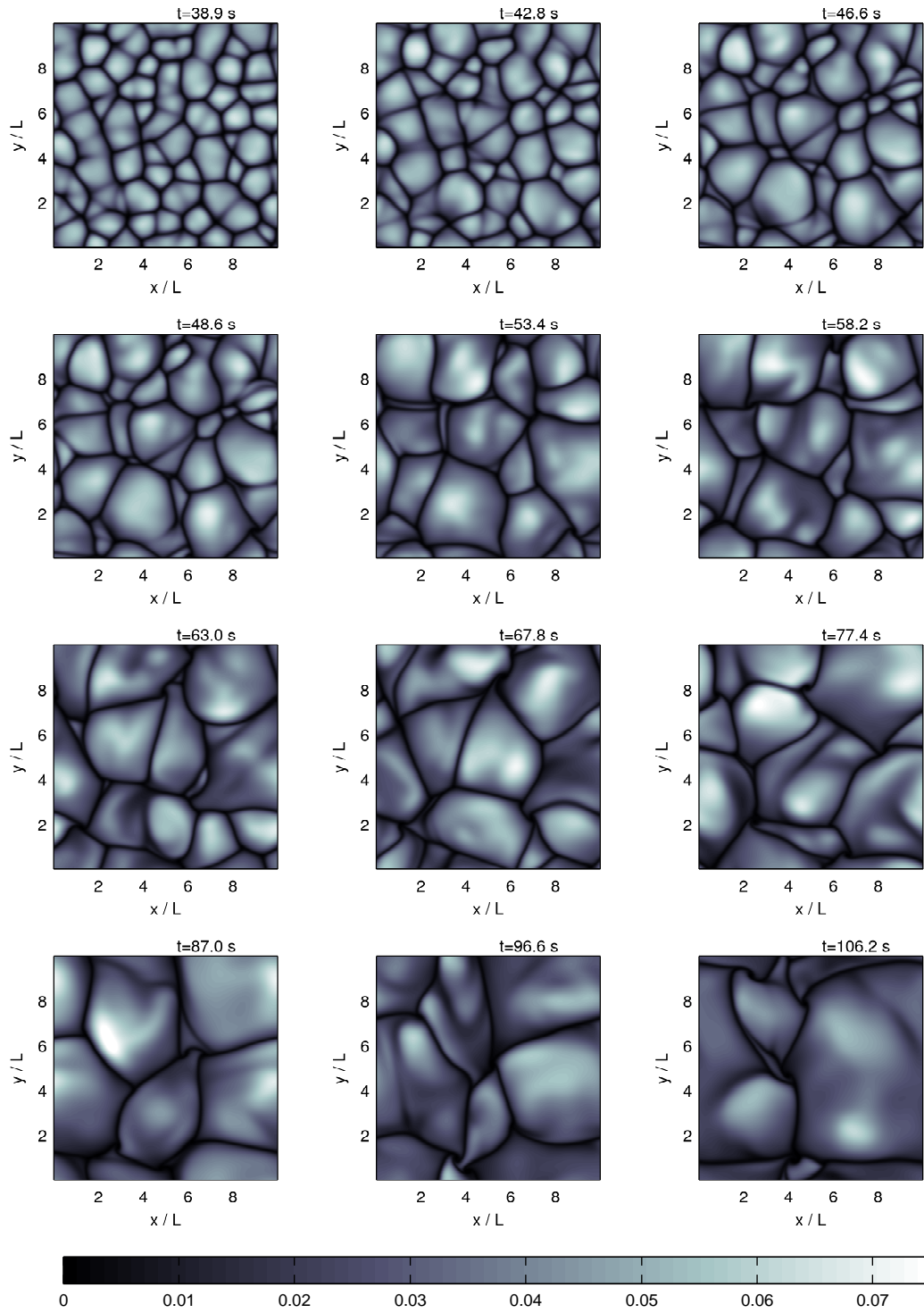


FIGURE 8. Top grid plane temperature distribution from simulation BC3, shown in a normalized form such that $T = 0$ is the coldest water at the surface and $T = 1$ is the initial temperature of the warmer bulk water.

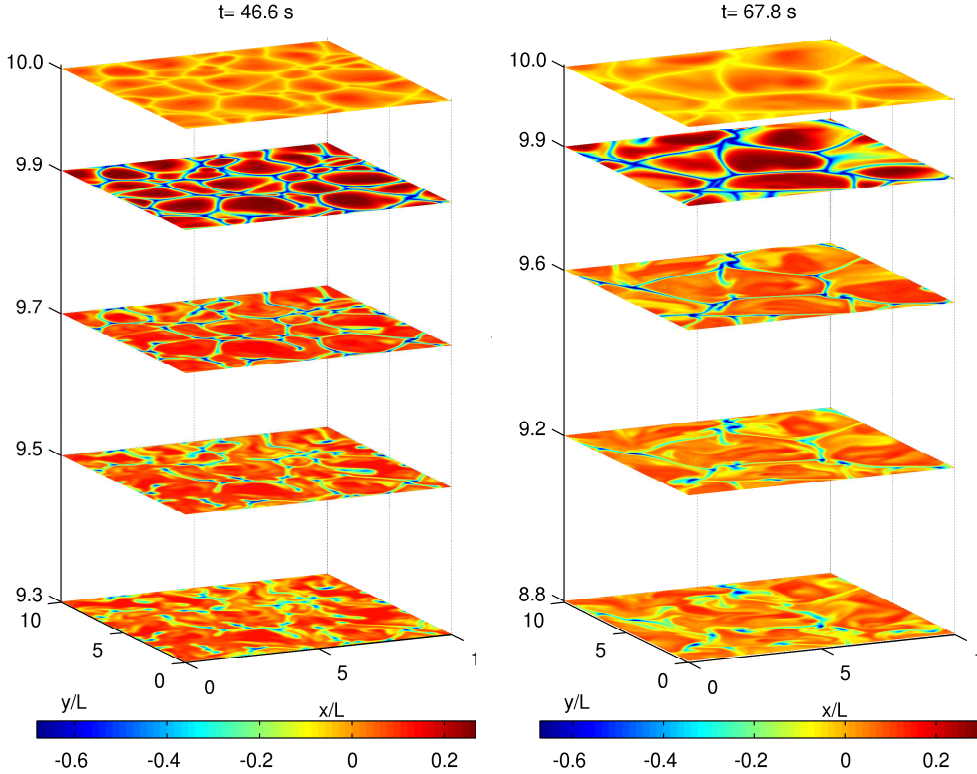


FIGURE 9. Contours of the fluctuating temperature $T - \langle T \rangle$ in various z -planes from simulation BC3.

time, as illustrated by the fact that the average cell size at $t = 46.6$ s is significantly smaller than at $t = 87$ s. The increased area of high-surface temperature that can be seen inside many individual cells at $t = 87$ s, indicates that the upwelling is stronger than at $t = 46.6$ s. This is indirectly confirmed by the isosurfaces shown in figure 7, where the downward falling plumes are much more pronounced at $t = 87$ s than at $t = 46.6$ s. As a result, a stronger upwelling motion would be needed to replenish the water at the surface.

Snapshots at $t = 46.6$ and 67.8 s of the fluctuating temperature $T - \langle T \rangle$ in various z -planes are shown in figure 9. It can be seen that the same net-like pattern that is observed at the surface is also identifiable in the upper bulk up to depths of $z_s \approx 0.5L$ and $z_s \approx 0.8L$ at $t = 46.6$ and 67.8 s, respectively. For larger depths, the convection cells lose their coherence because of the limited penetration depth of the sheets of falling cold water (for the smaller structures) and the interaction between the larger structures as they fall deeper into the bulk (cf. figure 8). The figure also shows the significant change in the temperature fluctuation intensity with depth. The intensity changes dramatically within a distance of $0.1L$ from the surface. Further down the intensity decreases again as the bulk becomes more mixed by the interacting falling plumes.

The existence of net-like patterns at the surface has been known for a long time, see e.g. the qualitative description in Spangenberg & Rowland (1961). The well resolved time-accurate data produced in the present DNS allows a quantitative estimation of the average convection cell-size at the surface (L_C). Here L_C is estimated by twice the

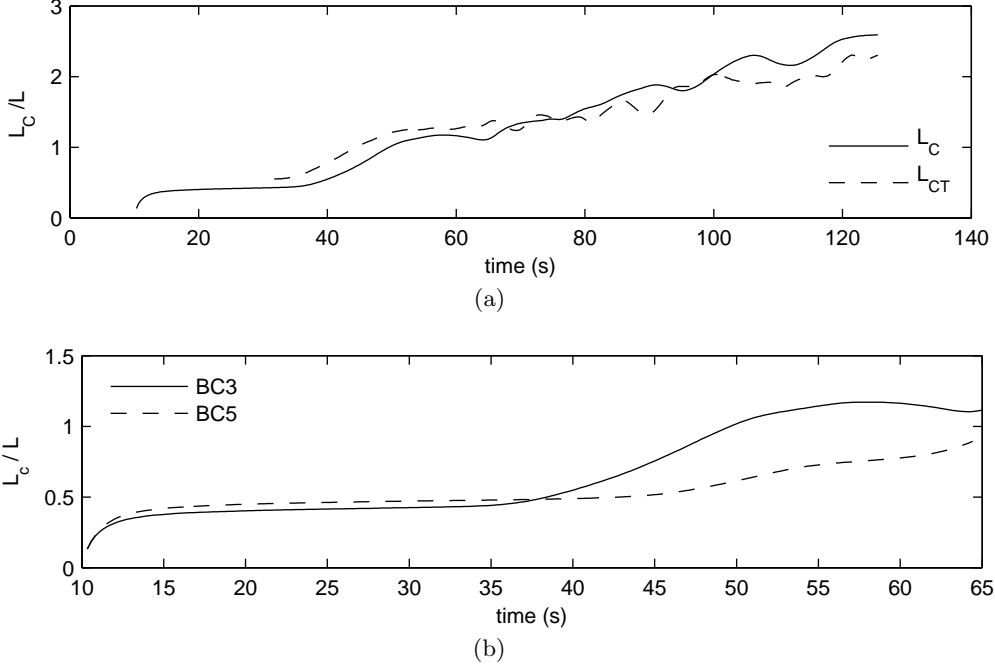


FIGURE 10. The growth of the average convection cell size of the top structures estimated using (4.1). a) Simulation BC3. b) Detailed comparison of BC3 ($Ra_L|_{t=0} = 34000$) and BC5 ($Ra_L|_{t=0} = 23000$).

average integral length scale at the surface

$$L_C = 2\sqrt{L_x L_y}, \quad (4.1)$$

where $\sqrt{L_x L_y}$ is the geometric mean of the integral length scales in the x and y -directions (L_x and L_y) based on the u and v velocities, respectively. Figure 10a depicts the average convection cell size L_C from BC3 estimated using (4.1). Also shown in the plot is the alternative estimation of the cell size L_{CT} which is calculated similarly to L_C , after replacing the interfacial u and v velocities by the temperature in the grid-plane immediately below the surface. It should be noted that the structures found in the interfacial $\partial w/\partial z$ contours and in the temperature contours seen in figure 8 are almost identical, indicating a strong correlation between the temperature and the near surface vertical velocity (and, consequently, the horizontal velocities). Based on this, it is expected that L_C and L_{CT} are of comparable size. This is confirmed by the fact that the estimation of both L_C and L_{CT} using (4.1) is found to correspond well with the mean size of the convection cells displayed in figure 8.

Figure 10a shows that the horizontal integral length scales L_C and L_{CT} tend to grow in time, which is in agreement with observations made in figures 8 and 10, where sequences of clearly identifiable structures show a significant growth of the convection cell size in time. A comparison of the growth of the cell-size obtained in BC3 and BC5 with Rayleigh numbers $Ra_L|_{t=0} = 34000$ and $Ra_L|_{t=0} = 23000$, respectively, is given in figure 10b. In both cases, the cell-size begins to increase rapidly about 7s after the first plumes start to break away from the thermal boundary layer, which happens at $t \approx 29.3$ s for BC3 and at $t \approx 36.1$ s for BC5. Furthermore, figure 10a shows that the integral length scale in BC3 keeps increasing, indicating a continued growth of the convection cells until the

end of the simulation. It is expected that this growth will only end when one convection cell spans the entire computational domain.

4.2. Correlation of T and c

Figure 11 contrasts the diffusivities of the temperature and the scalar $C_{Sc=500}$ by combining contours of T with isolines of $C_{Sc=500}$. A good correlation in the middle of the downward falling sheets is obtained between the cold water from the surface and high scalar concentrations which is maintained even in the head of the mushroom. As the difference in diffusivities of T and C , becomes less, areas of low T and high C will tend to coincide more and more. This is illustrated in figure 12 by the almost perfect negative correlation between T and $C_{Sc=20}$ of $\rho(T, C_{Sc=20}) \approx -0.92$ compared to values of $\rho(T, C_{Sc=500})$ between -0.455 and -0.495 .

It is interesting to study the evolution of the scalar distribution of $C_{Sc=500}$ in more detail. In an idealized situation, at the location where two equally strong convection cells meet, we would expect the formation of a symmetric 2D sheet of falling, saturated water to form in the middle of the planar jet (corresponding to the coldest water). Because of the free-slip boundary conditions, the near-surface flow will be almost uniform. Hence, it is likely that at the locations where convection cells meet the relative scaling of the thermal and concentration boundary layer thicknesses (see figure 19) would be at least partially conserved in the falling sheets of cold water. Indeed, a relative scaling of $\delta_C/\delta_T \approx (Sc/Pr)^{-0.5}$ was retrieved in the center of the falling sheets, where δ_T and δ_C are the thicknesses of the thermal and concentration layers inside the sheets. This almost perfect conservation of the relative boundary layer thicknesses in the falling sheets imply that the almost uniform flow extends further down than just the near-surface region.

4.3. Thermal boundary layer, Nusselt and Rayleigh numbers

Before the instability sets in, the colder temperature at the surface drives molecular diffusion and the thermal boundary layer thickness increases with time according to $\delta = (\pi\kappa t)^{1/2}$, where κ is the thermal diffusivity (see e.g. Turner 1979). Figure 13a shows the temporal variation of the thermal boundary layer thicknesses δ_{cf} and δ_q . The former is identified by the distance from the surface where the temperature fluctuation is maximum, while the latter is defined by

$$\delta_q = \frac{\Delta T}{(\partial T/\partial z)|_i}, \quad (4.2)$$

where subscript i denotes the interface.

The small disturbances added to the temperature field at $t = 9.6$ s were found to result in an almost instantaneous growth of the Rayleigh instability. This is proven by the fact that immediately after the addition of the disturbances, organised fluid movements (although very small) at the top surface started to appear. Figures 13a and b illustrate that at least up to $t = 27$ s diffusion dominates as the growth of the boundary layer thickness adheres to $\delta = (\pi\kappa t)^{1/2}$. At approximately $t = 29.3$ s the horizontally averaged boundary layer thickness reaches its maximum before starting to decrease abruptly. This instance matches the time when the cold fluid collected along the border of small cells first starts to plunge down in the form of thin vertical sheets and plumes. As in Howard (1964), the thermal boundary layer Rayleigh number Ra_δ is defined by

$$Ra_\delta = \frac{\alpha\Delta T g \delta_q^3}{\kappa\nu}. \quad (4.3)$$

Figure 13b shows a detailed plot of the thermal boundary layer evolution without and

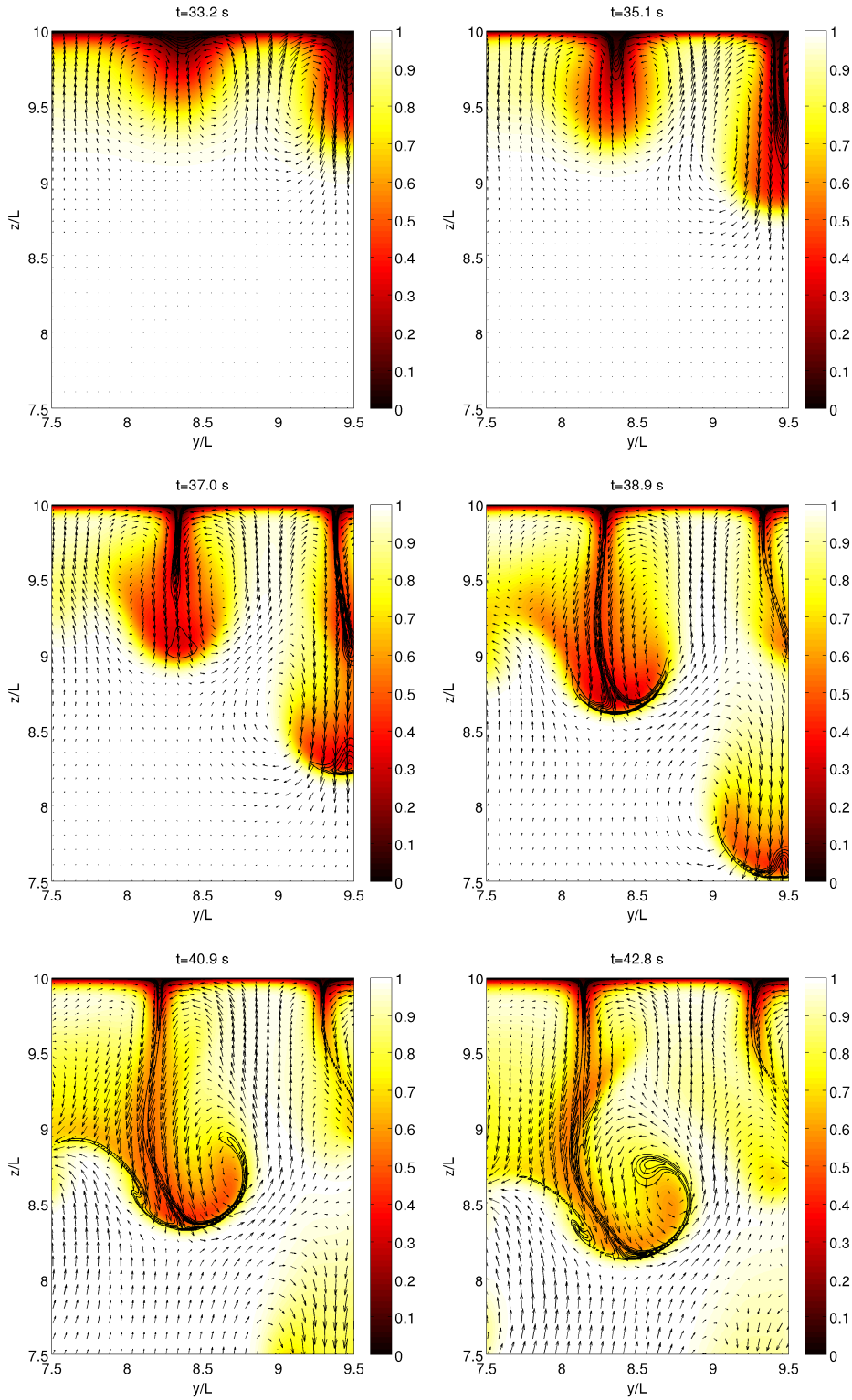


FIGURE 11. Simulation BC3 : correlation between temperature field and the $Sc = 500$ scalar, the arrows illustrate the (v, w) velocity field in the planes extracted at $x/L = 5$. The isolines of $C_{Sc=500}$ range from $C = 0.1$ to 0.6 .

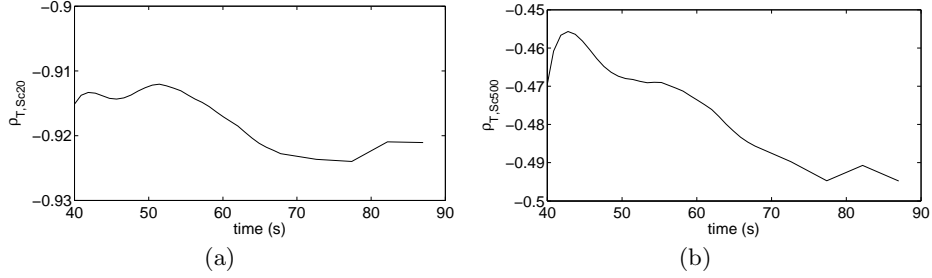


FIGURE 12. Simulation BC3 : variation of the correlation of temperature and scalar concentration in time. a) $Sc = 20$. b) $Sc = 500$.

with disturbances for two different expansion factors $\alpha\Delta T = 0.000572$ and 0.000381 , as used in BC3 and BC5, respectively. Based on the boundary layer thickness reached at the time when fluid elements first break away from the boundary layer in BC3 and BC5, critical Ra_δ of approximately 2000 and 1800 were found, respectively. The initial evolution of the thermal boundary layer thickness in all simulations with identical $\alpha\Delta T$ and initial disturbance level was found to be the same. It should be noted that in our case ΔT is the temperature difference between the water surface and the bulk, while in Howard (1964) ΔT is the temperature difference between the lower hot and upper cold surface. For the case of two-rigid boundaries, Howard (1964) estimated a critical Ra_δ of the order of 1000.

In figures 13a and b it can also be seen that for BC3 immediately after $t = 29.3$ s, both the mean boundary layer thickness and Ra_δ rapidly decrease and finally reach plateaus at approximately $0.083L$ and 18, respectively.

Figure 13c shows the temporal variation of $\Delta T = \langle T_b \rangle - T_s$, where $\langle \cdot \rangle$ denotes horizontal averaging, and the macro scale Rayleigh number Ra_L . The initial development of the instability and its affect on gas transfer across the air-water interface gives rise to time-dependent Rayleigh numbers and a time-dependent Nusselt number given by

$$Nu = \frac{H_L L}{\kappa}, \quad (4.4)$$

where $H_L = q/\Delta T$ is the heat transfer coefficient and the dimensional heat flux q is defined by

$$q = \kappa_c \left. \frac{\partial \langle T \rangle}{\partial z} \right|_i \times \frac{(T_s - T_{b,o})}{L}, \quad (4.5)$$

where κ_c is the thermal conductivity. From the definitions above it follows that

$$Nu = (Ra_L/Ra_\delta)^{1/3} \quad (4.6)$$

(e.g. Howard 1964). Figure 14 shows the relation between Nu and Ra_δ . The data points are calculated using instantaneous horizontally-averaged data. As the boundary layer thickness is correlated directly to the interfacial heat flux, the relation $Nu \propto Ra_\delta^{-1/3}$ holds for all times, including the diffusion dominated regime at the beginning of the simulation and the regime where the effects of the limited depth become important at the end of the simulation. In the free-fall regime, after $t \approx 41.3$ s, Ra_δ becomes virtually constant so that

$$Nu = c_{Ra} Ra_L^{1/3}, \quad (4.7)$$

with a value of c_{Ra} in the region between 0.35 and 0.4 (see figure 15a). Some time after

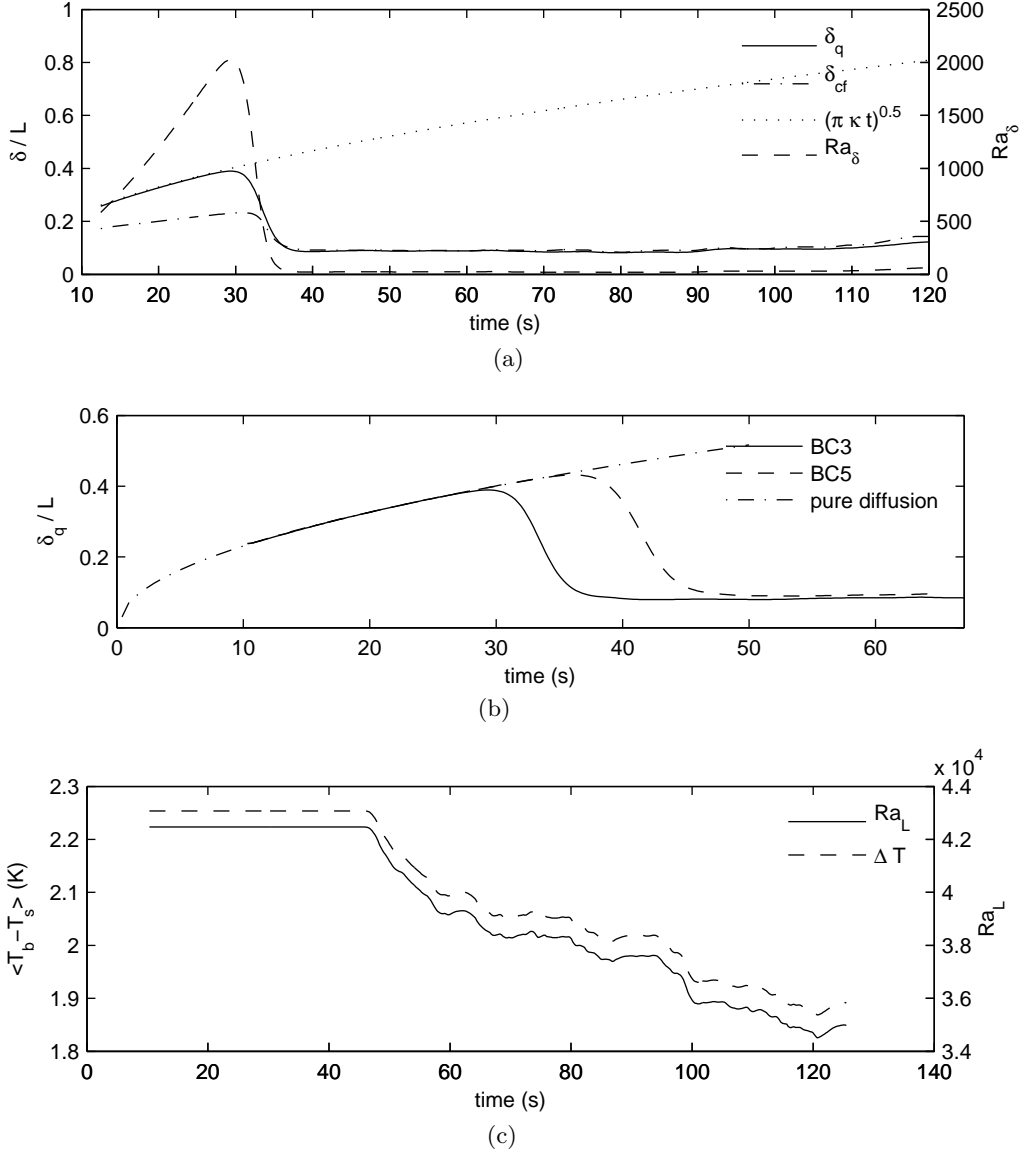


FIGURE 13. a) Time history of thermal boundary layer thicknesses (δ_{cf} and δ_q) and the resulting Ra_δ from BC3. b) A detailed plot of the δ_q evolution for BC3 and BC5. c) Decrease of the temperature difference between the bulk and the surface and the corresponding Ra_L with time.

the free-fall regime additional advective motions begin to affect the surface heat flux resulting in a reduction in c_{Ra} .

Because the process is unsteady, the constant of proportionality c_{Ra} in (4.7) was found to be sensitive to the distance from the surface, z_b , where T_b was evaluated (figure 15a and b for the case BC3). In the simulations with a domain depth $d = 10 L$ (BC2 and BC3), for instance, c_{Ra} slightly increased by ≈ 0.02 when z_b was decreased from $5 L$ to $2.5 L$. To allow a consistent comparison between simulations with different depths (figure 15c), for this case it was decided to fix z_b to a value of $2.5 L$, resulting in a constant of

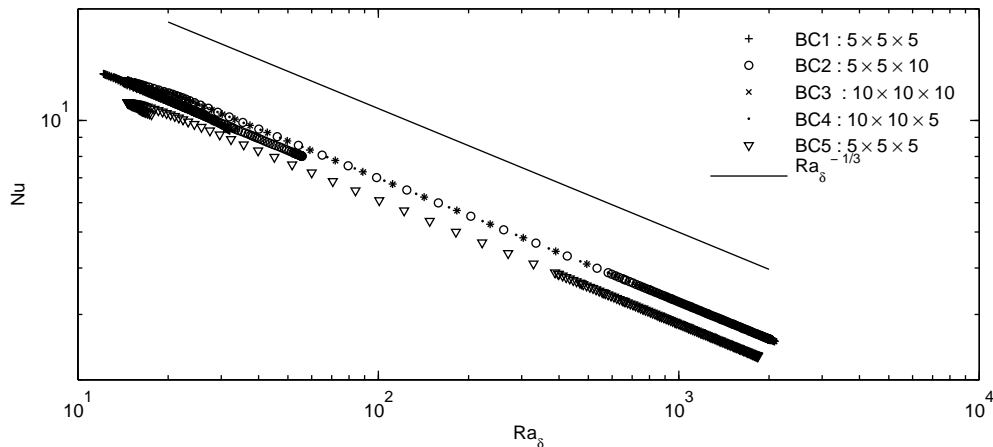


FIGURE 14. Nusselt vs boundary layer Rayleigh number (Ra_δ).

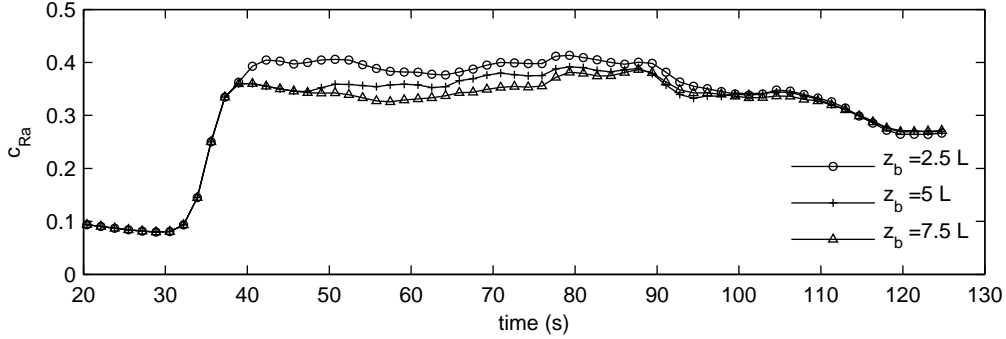
proportionality of $c_{Ra} = 0.39$. In the present simulations, temperature was only fixed at the top. To compare the present c_{Ra} to the free-slip results of Herring (1963) who used fixed temperatures at top and bottom, the different definitions of ΔT need to be taken into account. As Herring's ΔT is twice as large, his c_{Ra} needs to be pre-multiplied by $2^{1/3}$ giving a value of ≈ 0.39 which is identical to the value obtained here.

4.4. Vertical profiles

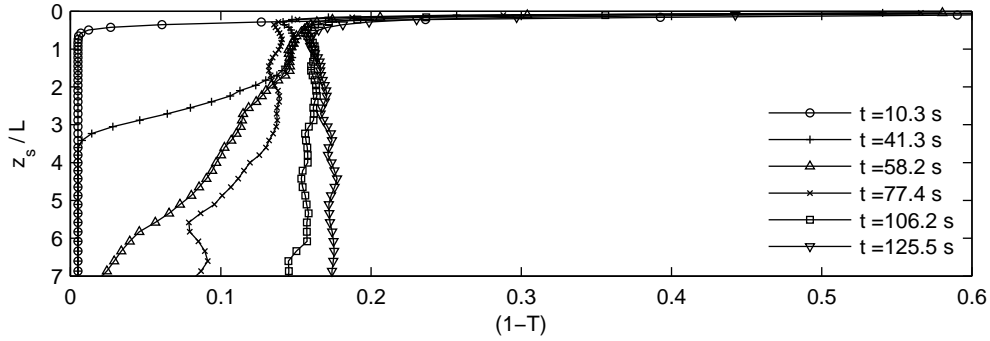
Figure 16 shows horizontally and time-averaged near-surface profiles of (a) the temperature T and (b) the temperature fluctuation T' for various 9.6 s time-intervals between $t = 41.3$ s and 87 s. Note that the purpose of the time-averaging over these short intervals is to slightly improve the statistics. When plotted against z_s/δ_{cf} , both the normalised mean temperature $\langle T_b - T \rangle / \langle T_b - T_s \rangle$ and the normalised rms temperature $T' / \langle T_b - T_s \rangle$ profiles can be seen to collapse reasonably well. This illustrates that in the quasi-steady state part of the free-fall regime, that is reached after a transient period (further detailed in Section 4.5), the instantaneous horizontally-averaged scalar profiles near the surface are approximately time-invariant (quasi-steady state). Furthermore, figure 16a shows that the horizontally-averaged temperature eventually becomes quasi-depth independent in the upper part of the bulk region.

Figure 17 shows the mean profiles of (a) the velocity fluctuations and (b) the scalar quantities T and C averaged over the entire free-fall period. The horizontal velocity fluctuations u' and v' reach their maximum of 82% of w'_{peak} at the surface where they affect the gas transfer to the unsaturated water flowing up from the bulk. For $z_s > 0.7 L$ the horizontal fluctuations become almost constant. The vertical fluctuations reach their peak value at $z_s \approx 1.9 L$.

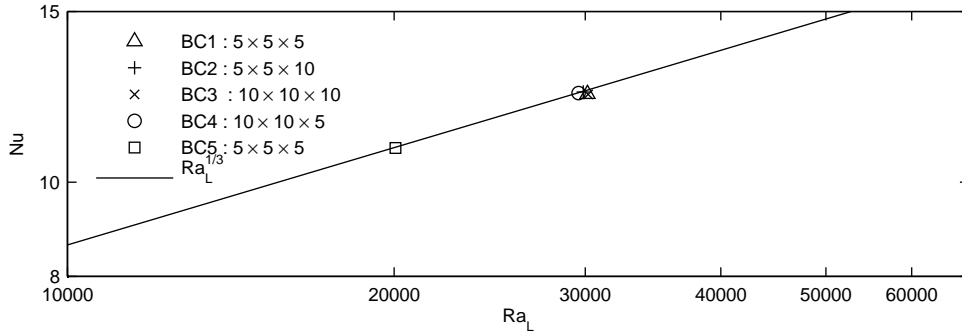
In figure 18a the normalised rms profiles of the temperature ($Pr = 6$) and concentrations ($Sc = 20$ and 500) can be seen to agree reasonably well with a peak value of approximately 0.25. The normalised diffusive and turbulent vertical mass fluxes shown in figure 18b were also found to collapse nicely. As expected, the turbulent vertical mass fluxes were observed to quickly increase with depth. Simultaneously, the diffusive mass flux reduced from its maximum value at the surface so that the sum of the two mass fluxes (in the region with quasi-steady horizontally averaged statistics) remained constant for all z_s in the upper bulk. At $z_s \approx 0.7\delta_{cf}$ the turbulent and diffusive mass fluxes



(a)



(b)



(c)

FIGURE 15. a) Sensitivity of c_{Ra} in $Nu = c_{Ra} Ra_L^{1/3}$ upon the selection of z_b , which is the distance from the surface where T_b is evaluated. b) Evolution of the horizontally averaged temperature profile. c) Nu vs Ra_L (time-averaged over $t = 41.3$ s to $t = 87$ s).

were found to be equal, while for larger z_s the turbulent vertical mass flux dominated the total mass flux. Finally, for $z_s \geq 2\delta_{cf}$ the diffusive mass flux was found to be zero while the turbulent mass flux reached its maximum value of $(D\partial\langle C\rangle/\partial z|_i)$ and fully dominated the total vertical mass flux. This result supports the usage of the eddy-correlation measurement technique (at the liquid side) in the bulk to determine the gas flux at the interface.

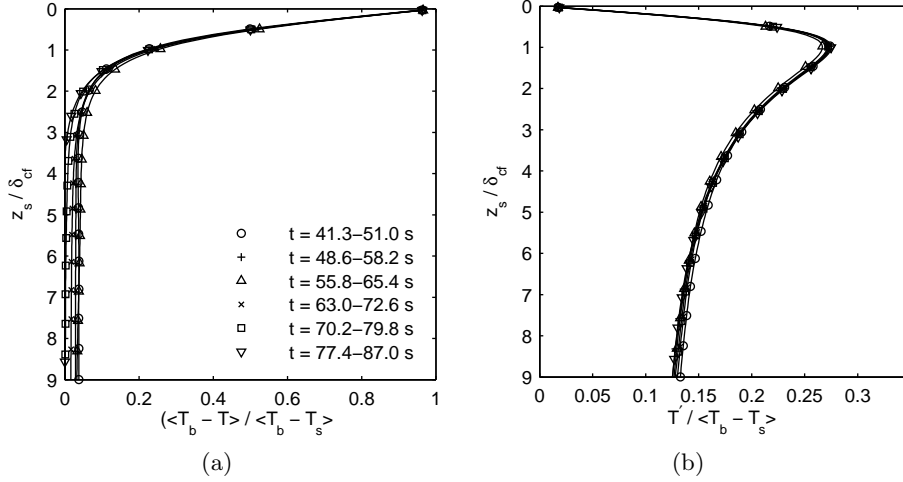


FIGURE 16. Simulation BC3: normalized mean vertical profiles (a) temperature and (b) temperature fluctuations.

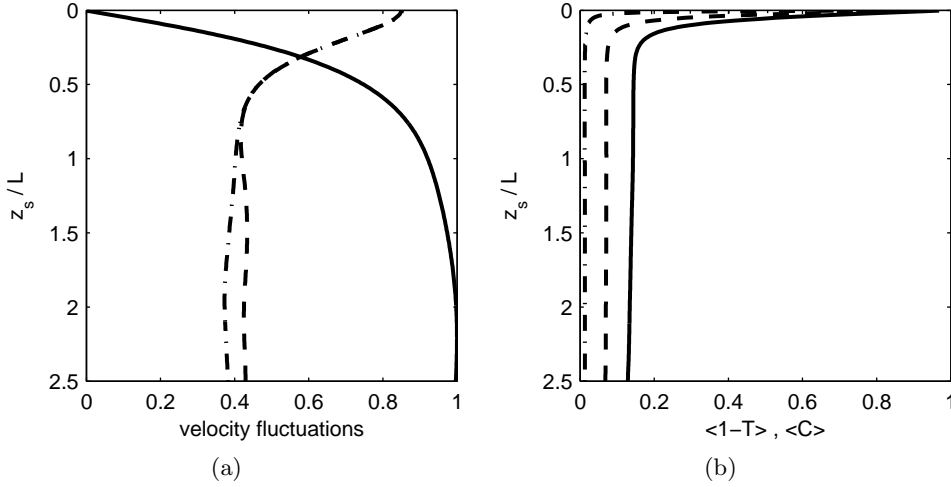


FIGURE 17. Simulation BC3: mean vertical profiles averaged over the free-fall regime (a) velocity fluctuations : - · - u'/w'_{peak} , - - - v'/w'_{peak} , — w'/w'_{peak} , (b) temperature and scalars : — temperature, - - - $C_{Sc=20}$, - · - $C_{Sc=500}$, ·

4.5. Estimation of H_L and K_L

In figure 6b the evolution of the temporal interfacial (dimensional) heat flux from BC3 was shown. With an initial temperature difference between the water surface and the bulk ($T_{b,0} - T_s$) of 2.2K, instantaneous peaks in the heat flux in excess of 1600 W/m² were observed. However, in time the heat flux was found to slowly decrease (while the convection cells size increased). In BC5, where $T_{b,0} - T_s$ was reduced to 1.48K, the maximum instantaneous heat flux decreased to about 1000 W/m². The reduction in the initial temperature difference from BC5 resulted in a thicker thermal boundary layer and hence a lower heat flux. The basic physical mechanism, however, is the same. Mainly because of the large initial temperature difference between the bulk and the water surface, in this study the heat fluxes were larger (though of similar order of magnitude) than most

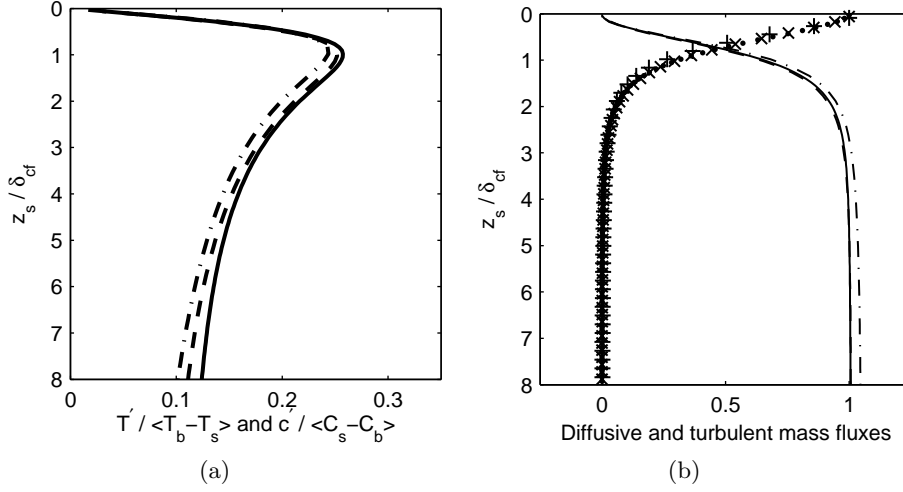


FIGURE 18. Simulation BC3: mean profiles of (a) temperature and concentration fluctuations — : T , - - - : $Sc = 20$, - · - : $Sc = 500$, (b) diffusive fluxes · : T , × : $Sc = 20$, + : $Sc = 500$ and turbulent mass fluxes (lines) normalized with mass flux at the surface $(D\partial\langle C \rangle / \partial z|_i)$ — : T , - - - : $Sc = 20$, - · - : $Sc = 500$.

reported values observed in the oceans or lakes (e.g. Shay & Gregg 1984, 1986; MacIntyre *et al.* 2002). It is clear that a one to one comparison between the heat fluxes obtained in the present simulations with the ones observed in nature is difficult. Apart from the additional uncertainties in nature, such as the presence of surface contamination that is known to reduce the interfacial mass transfer significantly, an accurate determination of the surface temperature in field and laboratory measurements is extremely difficult to achieve. In most experiments T_s is assumed to be identical to the air temperature some distance above the water surface. In still air, however, the formation of a very thick thermal boundary layer on the air side will result in a significant reduction of the temperature difference between the water surface and the bulk in a matter of seconds. On the other hand, evaporation effects (not considered here as the effects are confined to the surface where we assume a constant temperature) may result in a surface temperature that is lower than the air temperature some distance above the interface. In the present simulations we were able to fully resolve the initial (transient) development of the Rayleigh instability with relatively large heat fluxes and steep gradients that occur in nature when the water surface is suddenly cooled.

Figure 19 shows the horizontally and time-averaged gas transfer velocity and boundary layer thickness as a function of the Schmidt number. It can be seen that both K_L and δ_{cf} perfectly scale with $Sc^{-0.5}$. Time-averaging was performed from $t = 41.3$ s to $t = 87$ s. For $t \geq 87$ s bottom effects became non-negligible, though the scaling of both K_L and δ_{cf} with $Sc^{-0.5}$ was found to persist. This is in agreement with the K_L and δ scaling found in our previous DNS of grid-stirred turbulence driven gas transfer (Herlina & Wissink 2014) and is typical for cases with free-slip boundary conditions at the interface (perfectly clean water surface). Note that for contaminated surfaces the free-slip condition does not hold at the surface (see e.g. Khakpour *et al.* 2011).

The Sherwood number, defined as

$$Sh = \frac{K_L L}{D}, \quad (4.8)$$

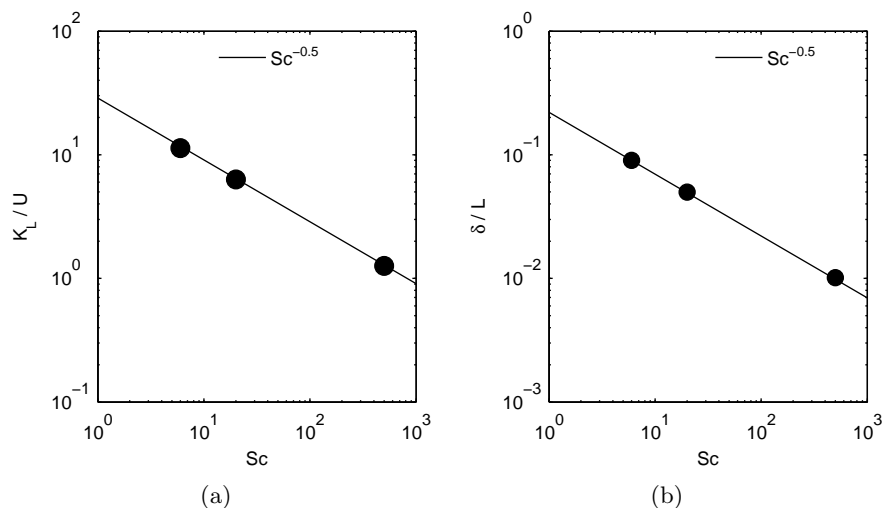


FIGURE 19. (a): K_L versus Sc (Pr) and (b): δ_{cf} versus Sc (Pr). K_L and δ_{cf} time-averaged from $t = 41.3$ to 87 s .

is the non-dimensional transfer velocity and is analogous to the Nusselt number (4.4) used in heat transfer problems. Since $H_{L,Pr=6} = K_{L,Sc=6}$ and $K_L \propto Sc^{-0.5}$ it immediately follows that

$$Sh = Nu(Sc/Pr)^{0.5}. \quad (4.9)$$

Based on the time history shown in figures 6, 13a, b and 20, for BC3 it is now possible to identify three different regimes :

(a) Diffusive regime : $0 < t < 29.3$ s. This regime is dominated by diffusive growth of the thermal and concentration boundary layers (figures 13a and b). It ends when the first thermal sheets and/or plumes separate from the boundary layer. Because the developing Rayleigh instability is still very weak, the surface fluctuations are very small (figure 20a).

(b) Free-fall regime : $29.3 < t < 87$ s. This regime follows the diffusive regime and ends when the near-surface fluid flow as well as the scalar transport become affected by the limited depth of the domain (figure 6). The free-fall regime can be further divided into : i) a transient part ($29.3 < t < 41.3$ s) and ii) a quasi-steady part ($41.3 < t < 87$ s). In the transient period, after the first plumes and/or sheets begin to fall both u'_{surf} and K_L rapidly increase. K_L is largely independent of the convection cell size, which remains approximately constant (see figure 10). After $t \approx 41.3$ s the normalised horizontally-averaged scalar profiles near the surface become approximately time-invariant indicating a quasi-steady state (figure 16). Here, both u'_{surf} and the cell size increase while K_L remains approximately constant.

(c) Bottom-influenced regime : $87 < t < 115$ s. This regime begins when the secondary flow -induced by the plumes reaching the bottom of the domain- first affects the interfacial mass transfer. Here u'_{surf} is found to become virtually constant while K_L decreases with increasing cell size.

An important aim of this study is to correlate surface information to the gas transfer velocity K_L . As can be seen in figures 7 and 8, the near surface flow in the present buoyancy driven gas transfer simulations is fully dominated by large structures. Based on this observation, it can be concluded that the large-eddy model of Fortescue & Pearson (1967) should be able to provide an accurate estimation of K_L . According to this model the surface renewal rate r can be estimated using velocity (U_r) and length (L_r) scales that

are typical for the larger structures in the flow so that $r = U_r/L_r$. Buoyancy driven flows exhibit a net-like structure of convection cells at the surface (see figure 8). The average diameter of these convection cells, L_C , (see figure 10) estimated by (4.1), appears to be the appropriate choice for L_r . Near the interface the flow in the convection cells is almost horizontal so that U_r can be based on the velocity fluctuations u'_{surf} at the surface, defined by $u'_{surf} = \sqrt{u'v'}$ (see figure 20a). Hence, a renewal rate of $r = u'_{surf}/L_C$ is obtained and K_L and H_L can be predicted using

$$K_L = c_K \sqrt{Du'_{surf}/L_C} \quad \text{and} \quad H_L = c_H \sqrt{\kappa u'_{surf}/L_C}, \quad (4.10)$$

where D and κ are the scalar and thermal diffusivities, respectively.

In figure 20b the predictions given in (4.10) are plotted together with K_L (for $Sc = 20$ and 500) and H_L . For $t \geq 38.5$ s a very good agreement between the aforementioned equations and both K_L and H_L is obtained using a value of 1.1 for both c_K and c_H . This good agreement is not only observed in the quasi-steady state part of the free-fall regime but also persists well after the plumes have reached the bottom of the domain and the unsteadiness introduced by the falling plumes starts to accumulate inside the computational domain. Only during the initial phase of the simulation, where diffusive mass transfer dominates the scaling of K_L and H_L , (4.10) is not applicable. Note that when replacing L_C by L_{CT} in (4.10) an approximation of K_L is obtained that is of equal quality for $t \geq 50$ s and slightly worse for $t < 50$ s.

As proposed by McCready *et al.* (1986), when very detailed surface information is available so that the surface divergence can be calculated, K_L and H_L can be estimated using

$$K_L \propto \sqrt{\beta' D} \quad \text{and similarly} \quad H_L \propto \sqrt{\beta' \kappa} \quad (4.11)$$

where β' is a measure of the surface divergence strength (see Turney & Banerjee 2013) given by

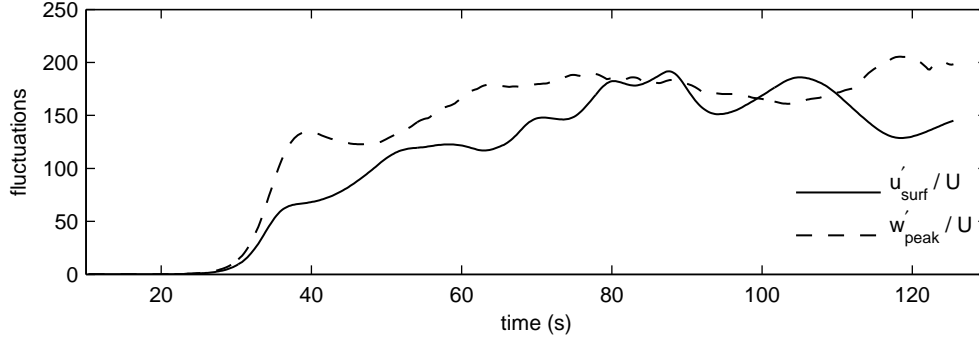
$$\beta' = \sqrt{\left\langle \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right)^2 \right\rangle} \quad (4.12)$$

In figure 20c a constant of proportionality of $c_\beta = 0.59$ in (4.11) is shown to give very good approximations. Note that Turney & Banerjee (2013) recently found that surface divergence theory is not applicable when a portion of the near surface motions holds small time scales. The time-scales in the buoyancy-driven flow presented here, however, are sufficiently long for surface divergence theory to be applicable.

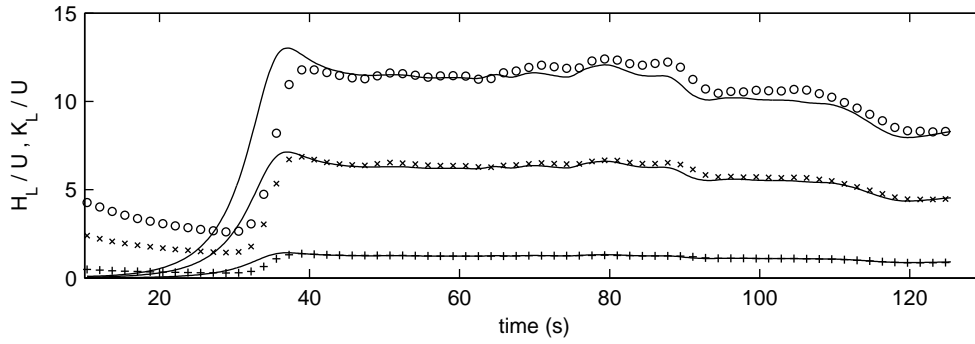
Figure 20a shows a comparison of w'_{peak} and u'_{surf} against time. It can be seen that initially the qualitative behaviour of w'_{peak} is very similar to that of K_L shown in figures 20b and c. In the free-fall regime for $t \geq 48.1$ s, w'_{peak} begins to grow while K_L reached a plateau. The reason for the very good initial agreement is that w'_{peak} is located near the surface so that $w'_{peak} \approx w'$ which is strongly related to the rms of the surface divergence.

By comparing figures 20b and 20c it can be seen that the quality of the estimation of H_L and K_L using (4.10) or (4.11) is similar for $t \leq 87$ s. For $t > 87$ s, when bottom effects start to become important, a slightly better estimation is obtained when using the surface divergence model (4.11).

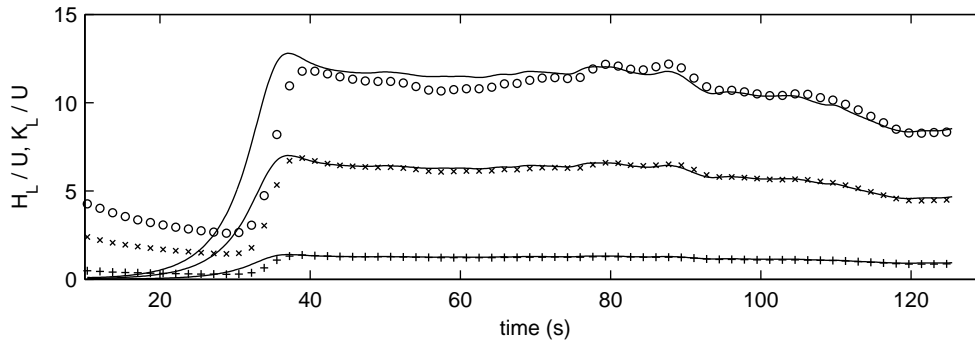
Figures 21a and b show estimations of the transfer velocities in various simulations using (4.10) and (4.11), respectively. The plotted data are time averaged over the quasi-steady part of the free-fall regime. Also displayed in figure 21b are results from experiments (Herlina & Jirka 2008; McKenna & McGillis 2004) and from the grid-stirred turbulence case of Herlina & Wissink (2014) where a c_β of 0.525 was found. In compari-



(a)



(b)



(c)

FIGURE 20. Simulation BC3 : a) Horizontal surface velocity fluctuations. b) Comparison of simulated K_L , H_L and the predictions from (4.10) with $c_{HL} = c_{KL} = 1.1$ (solid lines), \circ : $Pr = 6$, \times : $Sc = 20$, $+$: $Sc = 500$. c) Comparison of simulated K_L , H_L with $0.59\sqrt{D\beta'}$ (solid lines).

son, Magnaudet & Calmet (2006) obtained a c_β of 0.60 for turbulent open-channel flow, while Turney *et al.* (2005) reported a c_β of 0.5 in wind-driven gas transfer experiments (low and moderate wind speeds).

Above we have shown that a very good time-accurate approximation of K_L can be obtained using surface information. When no such information is available it might be possible to use sub-surface information instead, such as the vertical turbulent mass flux (see figure 18) or the velocity fluctuations combined with the integral length scale as

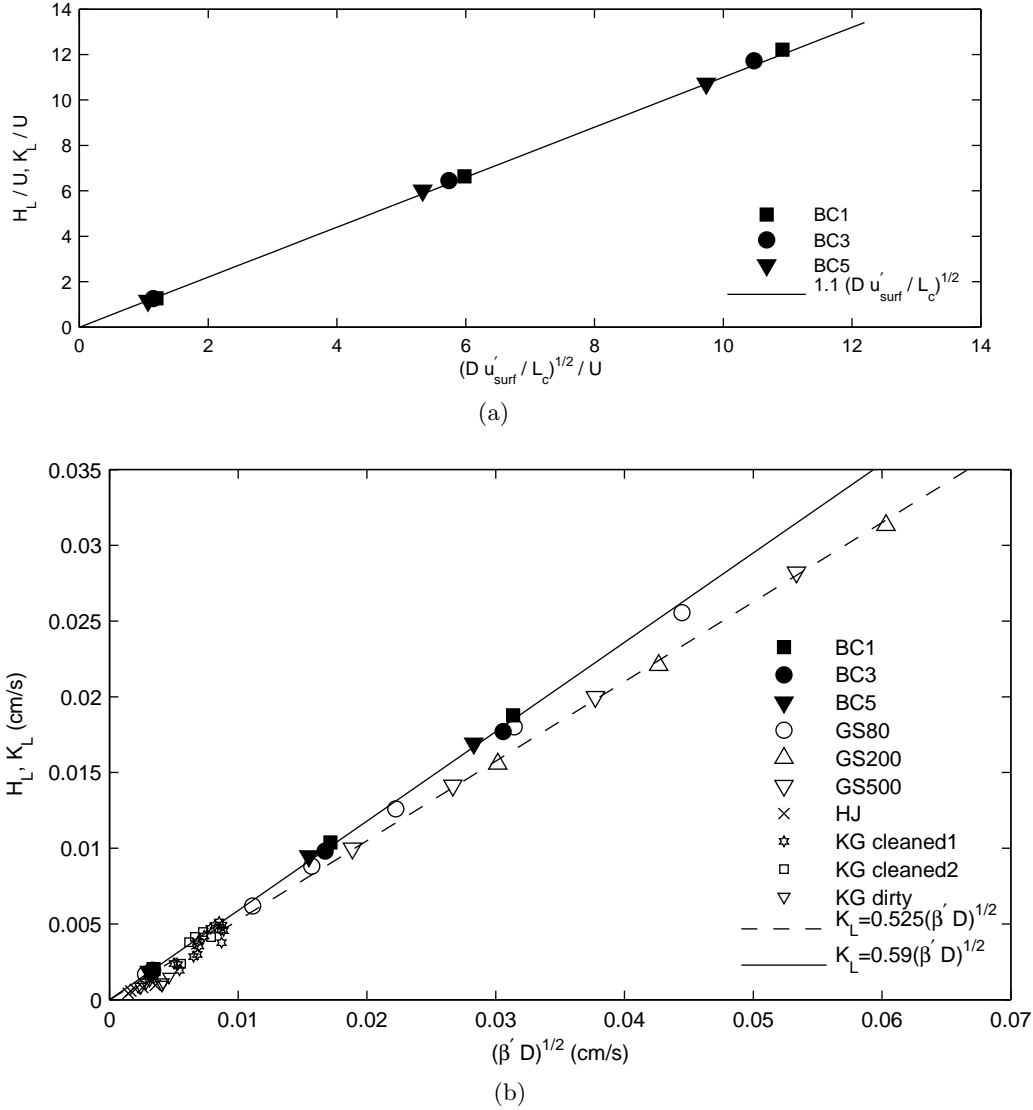


FIGURE 21. H_L and K_L estimations using data that was time-averaged over the quasi-steady part of the free-fall regime. (a): using (4.10) (b): using (4.11). The GS data points are from the numerical simulations by Herlina & Wissink (2014), the HJ and KG results are from experiments performed by Herlina & Jirka (2008) and McKenna & McGillis (2004), respectively.

detailed below. Analogously to the grid-stirred case of Herlina & Wissink (2014),

$$K_L = c_{RT} u' Sc^{-0.5} R_T^{-0.5} \quad (4.13)$$

can be employed to estimate the transfer velocity, where the bulk turbulent Reynolds number is given by

$$R_T = 2L_\infty u' / \nu, \quad (4.14)$$

L_∞ is the integral length scale and u' is the rms of the horizontal velocity fluctuations. Note that (4.13) is a variant of the large eddy model of Fortescue & Pearson (1967). To carry out a proper calculation of K_L using (4.13) it is needed that the bulk measurements

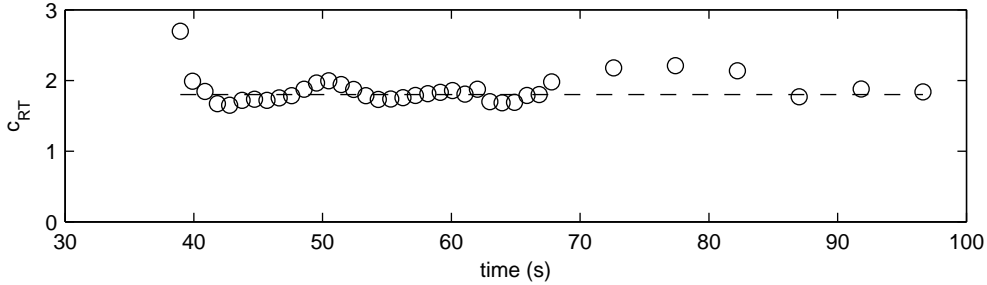


FIGURE 22. Coefficient of proportionality c_{RT} in (4.13). Bulk flow properties evaluated at $z_s = 2.5 L$. The dashed line corresponds to $c_{RT} = 1.8$.

are performed within the convective mixing layer. In the present simulations a very low $R_T \leq 50$ was obtained. As illustrated in figure 22, a reasonable time accuracy was achieved using a coefficient of proportionality of $c_{RT} = 1.8$, which is comparable to the value of 1.6 found in the grid-stirred case.

5. Conclusions

Large-scale direct numerical simulations (DNS) of mass transfer across a flat clean-surface driven by buoyant convection have been carried out. The simulations are motivated by previous experiments performed at KIT (Jirka *et al.* 2010). The aim of the DNS is to produce and analyse highly resolved data leading to a more in-depth and detailed understanding of the physical mechanisms that play a role in buoyancy driven gas transfer across the air-water interface. DNS calculations allow a full control of all boundary conditions, including the temperature and possible contaminations of the water surface. In the experiments, however, such a detailed control or knowledge of surface boundary conditions is never achieved. For instance, in the experiments by Jirka *et al.* (2010), the exact surface temperature could not be measured. Furthermore, in still air the large difference in thermal diffusivities of air and water will result in the formation of a very thick thermal boundary layer on the air side of the interface. As a result, the initial temperature difference between the water surface and the bulk will significantly reduce in a matter of seconds. Evaporation, on the contrary, could result in an interfacial temperature that is lower than the air temperature above. Another problem faced by experimentalists is unwanted surface contamination that tends to inhibit the free movement of water particles along the water surface. The latter also reduces the vertical fluxes that drive the gas transfer across the air-water interface.

The simulations were performed using a code especially developed to perform highly-accurate simulations of mass transfer at realistic (high) Schmidt and Prandtl numbers on a computationally feasible mesh while avoiding spurious oscillations of the scalar quantities (Kubrak *et al.* 2013). The purpose of the simulations was to investigate buoyancy-driven gas transfer across the air-water interface. The presented results are all transient and range from the onset of the Rayleigh instability until the limited depth of the domain begins to affect the gas flux at the surface. The Rayleigh instability was triggered by adding random disturbances to the temperature field at $t = 9.6$ s. The overall effect of the developing instability was found to be largely the same. In those cases where exactly the same disturbance field was employed (BC1 and BC2), it was possible to perform a one-to-one comparison of instantaneous flow, temperature and concentration fields. These fields were found to be identical until the plumes in the shallower simu-

lation, BC1, reached the bottom of the computational domain at $t \approx 48.1$ s, while it took a further time-lag of $\Delta t = 15.4$ s before the surface heat flux was affected. Hence, to describe this evolving Rayleigh instability depth-independent Rayleigh numbers were used, based on either the thermal boundary layer thickness or a fixed length-scale L . For larger temperature differences between the surface and the bulk (corresponding to larger Rayleigh numbers) a faster growing instability was obtained. Initially the thermal boundary layer (BL) thickness was fully determined by thermal diffusion. Eventually, the growing Rayleigh instability resulted in the vertical convective flux becoming strong enough to stop a further increase of the thermal BL thickness. The critical boundary layer Rayleigh number, Ra_δ , at maximum BL thickness was found to be $Ra_\delta \approx 2000$ for the simulations with an expansion factor of $\alpha\Delta T = 0.572 \times 10^{-3}$ and $Ra_\delta \approx 1800$ for $\alpha\Delta T = 0.381 \times 10^{-3}$.

The Rayleigh instability is seen to give rise to the formation of relatively small convection cells that tend to grow in time. At the water-surface these cells are visible as a net-like structure. In the interior of the cells one or more upwellings of low-saturated warm fluid from the bulk can be found. This fluid subsequently flows radially outward while it is cooled by the cold air and gets increasingly saturated with atmospheric gases. At the boundaries of the cells the cooled, now heavier fluid subsequently falls down forming thin vertical sheets of cold fluid. In the center plane of these sheets, where the fluid is coldest, the highest concentration of atmospheric gases was found. The scaling of the thermal BL thickness $\delta_T = \sqrt{\frac{Sc}{Pr}}\delta_C$ with the concentration boundary layer thickness, δ_C , was found to be conserved in the sheets. The mushroom-like structures that are often reported in literature were found to be formed at the intersection of three or more convection cells, leading to a local increase in the amount of cold fluid plunging down deep in the bulk at a relatively high velocity. Momentarily heat fluxes were obtained in excess of 1600 W/m^2 . The above illustrates the effectiveness of buoyant convective gas and heat transfer.

The ratio of the gas transfer and the heat transfer velocities, K_L/H_L was found to be equal to $(Pr/Sc)^{0.5}$. Hence, even when only information on H_L is available, estimates of K_L for Schmidt numbers between $Sc = 20$ up to (at least) $Sc = 500$ can be obtained.

To model the renewal rate r (Danckwerts 1951) it is possible to use the rms of the surface divergence β' as initially proposed by McCready *et al.* (1986). The best fit for $K_L \propto \sqrt{D\beta'}$ was obtained using a constant of proportionality of 0.59. Another accurate estimation of r was found using the geometric average of the rms of the velocities at the surface, $u'_{surf} = \sqrt{u'v'}$, and the average size of the convection cells L_C , giving $K_L = 1.1\sqrt{Du'_{surf}/L_C}$ as a best fit. The average cell size L_C was approximated by $2\sqrt{L_x L_y}$, where L_x and L_y are the integral length scales based on u in the x -direction and v in the y -direction, respectively.

Above, the transfer velocity was estimated using information that is available at the water surface. The two estimations were found to have a very good time-accuracy in and beyond the quasi-steady state part of the free-fall regime. When only sub-surface information is available the results showed that it is still possible to estimate K_L with reasonable time-accuracy. Such an estimation can be based, for instance, on the turbulent mass flux $\langle w''C'' \rangle$ or on u' and the integral length scale L_∞ , where all quantities need to be evaluated somewhere inside the convective mixing region. The estimations will improve in time as the convective mixing layer becomes well developed.

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