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Aerodynamics investigation of delta wing at low reynold's number

(Article)

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Abstract

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A numerical simulation has been carried out to investigate the aerodynamics of an oscillating delta wing and to evaluate the flow structure over the leading edge at different low Reynolds numbers. The numerical results of coefficient of lift (CL) vs angle of attack is validated with the experimental results from the previous study. Effects of Reynolds number and angle of attack is investigated for the wing lift coefficient, and the aerodynamic efficiency. Pressure contour and turbulence kinetic energy are also observed for each case. Vortex formation from each case is noted from pressure contour and lift coefficient. The stall condition is also observed. Considering the three Reynolds number at max CL, which is at angle of attack of 40 degrees, the CL only increased by 0.01, which is not a very significant increase. But L/D ratio has in-creased significantly for Reynolds number in the range from 8x104 to 1.5x105 and from Re 1.5x105 to 2.68x105. Whereas for each Reynolds number, the lift coefficient CL seems to attain highest value of 0.4 at angle of incidence between 10 degrees and -10 degrees to-20 degrees. © 2019 PENERBIT AKADEMIA BARU-All rights reserved.

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