Advances in Aircraft Structures

Editor

Jaffar Syed Mohamed Ali

Erwin Sulaema



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Chapter 1

Design of IIUM Aircraft Fuselage Using Composite Material

Y. Aminanda, Mir Amirul Shah Bin Esa, Said Hamadi Said Mohamed

Abstract

The main objective of this chapter is to structurally design a minimum weight fuselage which has enough strength to overcome loads while flying or on ground. Structure is related not only to the safety but also to the cost and to the performance of an airplane. Due to these reasons, a proper steps need to be taken during the whole process. In this chapter a complete structural design calculation of the IIUM made aircraft's fuselage will be discussed. The design process discussed in this chapter is a standard practice applied in aircraft manufacturing industries. Step-by-step procedures are included whenever possible. The stress calculation and sizing of the structures is carried out using hand calculations throughout this chapter..

Keywords: Composite, fuselage, stress, design, optimum sizing

1. Introduction

Fuselage is one of the important components of an airplane. Generally it is use to protect airplane payloads. Payloads here may include human, cargoes and aircraft system components. Since the applications of the fuselage are huge and significant, appropriate tasks are needed to build a fuselage. The requirement of this project is to structurally design a fuselage which has minimum weight. It