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Stabilized finite element method for the transient Navier–Stokes equations based on a pressure gradient projection

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Abstract

In this paper we present a stabilized finite element formulation for the transient incompressible Navier–Stokes equations. The main idea is to introduce as a new unknown of the problem the projection of the pressure gradient onto the velocity space and to add to the incompressibility equation the difference between the Laplacian of the pressure and the divergence of this new vector field. This leads to a pressure stabilization effect that allows the use of equal interpolation for both velocities and pressures. In the case of the transient equations, we consider the possibility of treating the pressure gradient projection either implicitly or explicitly. In the first case, the number of unknowns of the problem is substantially increased with respect to the standard Galerkin formulation. Nevertheless, iterative techniques may be used in order to uncouple the calculation of the pressure gradient projection from the rest of unknowns (velocity and pressure). When this vector field is treated explicitly, the increment of computational cost of the stabilized formulation with respect to the Galerkin method is very low. We provide a stability estimate for the case of the simple backward Euler time integration scheme for both the implicit and the explicit treatment of the pressure gradient projection. © 2000 Elsevier Science S.A. All rights reserved.

1. Introduction

The purpose of this paper is to present a stabilized finite element method for the transient Navier–Stokes equations that allows to use equal velocity–pressure interpolation. This is possible due to a stabilization technique based on the introduction as unknown of the discrete problem of the projection of the pressure gradient onto the finite element space of continuous vector fields. The divergence of the difference between these two vectors (pressure gradient and its projection) is introduced in the continuity equation.

The idea of this stabilization method was originally developed for the stationary Stokes problem in [1] and extended to the non-linear Navier–Stokes problem in [2]. The motivation for the design of this formulation was to inherit the stability properties of some fractional step time integration schemes that use a Poisson equation to compute the pressure (see [3] for further discussion). From the computational point of view, the main drawback of this approach is the introduction of the projection of the pressure gradient as a new unknown of the problem, thus increasing substantially the number of nodal unknowns of the final discrete system. However, iterative strategies may be devised to make the method efficient, with a computational cost similar to that of other stabilization methods.

When the transient Navier–Stokes equations are discretized in time using a finite difference scheme, the projection of the pressure gradient can be treated explicitly, that is to say, evaluated at the previous time step. In this case, the increase of cost of the formulation with respect to the standard Galerkin method is

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very low. First, a stabilization matrix must be built up and at the end of each time step the pressure gradient must be projected. This leads to a linear system of equations with a Gramm system matrix which can be solved by a simple Jacobi iteration or approximated by a diagonal system. The number of unknowns is not increased.

In this paper it is shown that the formulation just described is stable. A stability estimate is obtained for both velocities and pressures, without the need to satisfy the standard inf–sup condition on the velocity and pressure finite element spaces encountered when the Galerkin method is used. The stabilization technique is applied to both a monolithic velocity–pressure formulation of the problem using the trapezoidal rule and to an incremental fractional step method.

It has to be remarked that the only purpose of the stabilization technique presented here is to stabilize the pressure, in a similar way to the methods discussed for example in [4] and [5], even though we consider the most general case of transient Navier–Stokes equations. The instabilities due to the convective term when the viscosity is very small are not considered in our formulation. This is reflected by the fact that the stability estimates that we obtain depend on the viscosity. To overcome this, further modifications of the formulation presented herein are required (see for example Refs. [6,7], where formulations accounting both the convective and the pressure instabilities are analyzed).

2. Problem statement

Let us consider the transient Navier–Stokes equations for an incompressible fluid. Let Ω be an open, bounded and polyhedral domain of \mathbb{R}^d , where $d = 2$ or 3 is the number of space dimensions, $\Gamma = \partial\Omega$ its boundary and $[0, T]$ the time interval of analysis. The Navier–Stokes problem consists in finding a velocity \mathbf{u} and a pressure p such that

$$\frac{\partial \mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla) \mathbf{u} - \nu \Delta \mathbf{u} + \nabla p = \mathbf{f} \quad \text{in } \Omega, \quad t \in (0, T), \quad (1)$$

$$\nabla \cdot \mathbf{u} = 0 \quad \text{in } \Omega, \quad t \in (0, T), \quad (2)$$

$$\mathbf{u} = \mathbf{0} \quad \text{on } \Gamma, \quad t \in (0, T), \quad (3)$$

$$\mathbf{u} = \mathbf{u}^0 \quad \text{in } \Omega, \quad t = 0, \quad (4)$$

where ν is the kinematic viscosity, \mathbf{f} is the force vector and \mathbf{u}^0 is the velocity initial condition. We have considered the homogeneous Dirichlet boundary condition (3) for simplicity.

To write the weak form of problem (1)–(4) we need to introduce some notation. As usual, we denote by $H^m(\omega)$ the Sobolev space of m th order in a set ω , consisting of functions whose distributional derivatives of order up to m belong to $L^2(\omega)$ and by $H_0^1(\omega)$ the subspace of $H^1(\omega)$ of functions with zero trace on Γ . A bold character is used for the vector counterpart of these spaces. The L^2 scalar product is denoted by $(\cdot, \cdot)_\omega$, and the H^m norm by $\|\cdot\|_{m,\omega}$. The subscript m is omitted when $m = 0$ and so is ω when it is Ω .

Let us now consider the spaces $\mathbf{V}_{\text{st}} = \mathbf{H}_0^1(\Omega)^d$ and $\mathcal{Q}_{\text{st}} = \{q \in L^2(\Omega) \mid \int_\Omega q = 0\}$ and the forms

$$\begin{aligned} a(\mathbf{u}, \mathbf{v}) &:= \nu(\nabla \mathbf{u}, \nabla \mathbf{v}), \\ b(q, \mathbf{v}) &:= (q, \nabla \cdot \mathbf{v}), \\ c(\mathbf{u}, \mathbf{v}, \mathbf{w}) &:= ((\mathbf{u} \cdot \nabla) \mathbf{v}, \mathbf{w}) + \frac{1}{2}((\nabla \cdot \mathbf{u}) \mathbf{v}, \mathbf{w}), \end{aligned} \quad (5)$$

with $\mathbf{u}, \mathbf{v}, \mathbf{w} \in \mathbf{V}_{\text{st}}$ and $q \in \mathcal{Q}_{\text{st}}$. All these forms are continuous and c is skew-symmetric in its last two arguments. If $\nabla \cdot \mathbf{u} = 0$ it coincides with the standard trilinear form coming from the convective term of (1). However, for the finite element formulation that we shall consider the discrete velocity field will not be weakly divergence free and the use of the trilinear form given by (5) simplifies the analysis.

The norms of a and c are denoted by N_a and N_c , respectively, the former being proportional to the viscosity ν . Moreover, a is coercive as a consequence of the Poincaré–Friedrichs inequality, and b satisfies the inf–sup or Babuška–Brezzi condition for the spaces \mathbf{V}_{st} and \mathcal{Q}_{st} .

If $\langle \cdot, \cdot \rangle$ denotes the duality pairing between V_{st} and its topological dual space V'_{st} , the weak form of problem (1)–(4) consists in finding $(\mathbf{u}, p) \in V \times Q$, with $V = L^2(0, T; V_{st})$ and $Q = L^2(0, T; Q_{st})$, such that

$$\left(\frac{\partial \mathbf{u}}{\partial t}, \mathbf{v} \right) + c(\mathbf{u}, \mathbf{u}, \mathbf{v}) + a(\mathbf{u}, \mathbf{v}) - b(p, \mathbf{v}) = \langle \mathbf{f}, \mathbf{v} \rangle \quad \forall \mathbf{v} \in V_{st}, \tag{6}$$

$$b(q, \mathbf{u}) = 0 \quad \forall q \in Q_{st}, \tag{7}$$

and satisfying the initial condition in a weak sense. These equations hold weakly in $L^2(0, T)$.

If instead of having $\mathbf{f} \in L^2(0, T; \mathbf{H}^{-1}(\Omega))$ we require $\mathbf{f} \in L^2(0, T; L^2(\Omega))$ and Γ is sufficiently smooth, it is known that the solution of problem (6) and (7) verifies $\mathbf{u} \in L^2(0, T; V_{st} \cap \mathbf{H}^2(\Omega))$ and $p \in L^2(0, T; Q_{st} \cap H^1(\Omega))$, that is, the regularity of the solution increases (see, e.g., [8]). Also, the duality $\langle \mathbf{f}, \mathbf{v} \rangle$ in (7) can be replaced by (\mathbf{f}, \mathbf{v}) . The finite element approximation proposed in this paper will be based on the fact that $p \in L^2(0, T; H^1(\Omega))$. We say that problem (1)–(4) is *p-regular* if $p \in L^2(0, T; Q_{st} \cap H^1(\Omega))$ whenever $\mathbf{f} \in L^2(0, T; L^2(\Omega))$.

3. Stabilization of the stationary problem

Let us consider now the stationary version of problem (1)–(3) and (5) that is, dropping the time dependence in both the velocity and the pressure. In this section we present the stabilized method we have developed for it and collect the most important results we have obtained. In Section 4 we present the extension of this stabilization method to the transient case.

Let \mathcal{T}_h denote a finite element partition of the domain Ω of diameter h , from which we construct the finite element spaces Q_h, V_h and $V_{h,0}$, approximations to $Q_{st}, \mathbf{H}^1(\Omega)^d$ and V_{st} , respectively. The former is made up with continuous functions of degree k_q and the other two with continuous vector functions of degree k_v , the latter verifying the homogeneous Dirichlet boundary conditions.

The standard Galerkin finite element counterpart of the stationary version of problem (6) and (7) can now be written as follows: find $(\mathbf{u}_h, p_h) \in V_{h,0} \times Q_h$ such that

$$c(\mathbf{u}_h, \mathbf{u}_h, \mathbf{v}_h) + a(\mathbf{u}_h, \mathbf{v}_h) - b(p_h, \mathbf{v}_h) = (\mathbf{f}, \mathbf{v}_h) \quad \forall \mathbf{v}_h \in V_{h,0}, \tag{8}$$

$$b(q_h, \mathbf{u}_h) = 0 \quad \forall q_h \in Q_h. \tag{9}$$

It is well known that if this formulation is used the velocity and pressure finite element spaces have to satisfy the discrete analogue of the inf–sup condition. In our case this happens if $k_v = k_q + 1$ in the case $d = 2$, i.e., for 2D Taylor–Hood type elements [9,10]. This condition is not necessary using the method described next.

Let $\alpha_K, K \in \mathcal{T}_h$, be a family of mesh parameters depending on the element sizes h_K . The modification of problem (8) and (9) that we propose is: find $(\mathbf{u}_h, p_h, \xi_h) \in V_{h,0} \times Q_h \times V_h$ such that

$$c(\mathbf{u}_h, \mathbf{u}_h, \mathbf{v}_h) + a(\mathbf{u}_h, \mathbf{v}_h) - b(p_h, \mathbf{v}_h) = (\mathbf{f}, \mathbf{v}_h), \tag{10}$$

$$\sum_K \alpha_K^2 (\nabla p_h, \nabla q_h)_K - \sum_K \alpha_K (\xi_h, \nabla q_h)_K + b(q_h, \mathbf{u}_h) = 0, \tag{11}$$

$$- \sum_K \alpha_K (\nabla p_h, \boldsymbol{\eta}_h)_K + (\xi_h, \boldsymbol{\eta}_h) = 0 \tag{12}$$

for all $(\mathbf{v}_h, q_h, \boldsymbol{\eta}_h) \in V_{h,0} \times Q_h \times V_h$.

For a function $q \in H^1(\Omega)$ let us define $\nabla^h q$ by

$$\nabla^h q|_K = \alpha_K \nabla q|_K, \quad K \in \mathcal{T}_h. \tag{13}$$

From (12) it is seen that ξ_h is precisely the projection of $\nabla^h p_h$ onto V_h . We shall call it simply the pressure gradient projection. Our stability and convergence analysis is strongly based on a decomposition of the pressure gradient that we describe next. Let $\nabla^h Q_h$ denote the space of vector functions in $L^2(\Omega)$ which are of the form $\nabla^h q_h$, with $q_h \in Q_h$ and consider the vector space

$$E_h := V_h + \nabla^h Q_h = E_{h,1} \oplus E_{h,2} \oplus E_{h,3},$$

where $\mathbf{E}_{h,i}$, $i = 1, 2, 3$, are three mutually L^2 -orthogonal subspaces defined as

$$\mathbf{E}_{h,1} := \mathbf{V}_{h,0}, \quad \mathbf{E}_{h,2} := \mathbf{V}_{h,0}^\perp \cap \mathbf{V}_h, \quad \mathbf{E}_{h,3} := \mathbf{V}_h^\perp \cap \mathbf{E}_h.$$

Let us denote by $P_{h,i}$ the orthogonal projection from \mathbf{E}_h to $\mathbf{E}_{h,i}$, and $P_{h,ij} := P_{h,i} + P_{h,j}$, $i, j = 1, 2, 3$. Also, we denote $\mathbf{E}_{h,ij} := \mathbf{E}_{h,i} \oplus \mathbf{E}_{h,j}$. In order to prove that the pressure gradient in the discrete problem is stable, we shall bound independently the three terms in the decomposition

$$\nabla^h p_h = P_{h,1}(\nabla^h p_h) + P_{h,2}(\nabla^h p_h) + P_{h,3}(\nabla^h p_h). \tag{14}$$

The bound for the first component of the pressure gradient decomposition in (14) can be obtained from the momentum Eq. (10), whereas the third component can be bounded only assuming that α_K verifies condition (16) stated below. Thus, the stability provided by the method in comparison with the standard problem (8) and (9) is precisely in the control over the term $P_{h,3}(\nabla^h p_h)$, that is, the component of the pressure gradient orthogonal to the space of continuous vector fields \mathbf{V}_h .

The second component in (14) deserves special attention. It depends on the properties of the finite element spaces and not on the problem actually solved. We assume that there is a positive constant K'_2 such that

$$\|\nabla^h q_h\| \leq K'_2 \|P_{h,13}(\nabla^h q_h)\| \quad \forall q_h \in Q_h,$$

which means that $\|P_{h,2}(\nabla^h q_h)\|$ can be bounded by $\|P_{h,13}(\nabla^h q_h)\|$. This condition is equivalent to the existence of a constant $K_2 = 1/K'_2 > 0$ such that

$$\inf_{q_h \in Q_h} \sup_{\mathbf{v}_h \in \mathbf{E}_{h,13}} \frac{(\nabla^h q_h, \mathbf{v}_h)}{\|\nabla^h q_h\| \|\mathbf{v}_h\|} \geq K_2, \tag{15}$$

which is similar to the inf–sup condition of the standard problem, although much weaker and, in particular, verified when equal interpolation is used. This was proved in [1] for P_k interpolations (and a very similar proof can also be applied to the Q_1 case). The analysis in this reference is based on a generalization of the macroelement technique presented in [11].

Let us collect now the stability and convergence results obtained for the formulation that we have proposed here. The proof of all these results can be found in [2]. We assume that the viscosity is large enough so as to ensure uniqueness of solution of the stationary Navier–Stokes equations. This condition can be written as

$$\frac{N_c \|\mathbf{f}\|_{-1}}{K_a^2} < 1,$$

where K_a is the coercivity constant of the bilinear form a .

In order to prove stability we need to introduce a further assumption, now on the behavior of the numerical parameters α_K . There exist constants α_0 and α_1 , independent of h_K , such that

$$\alpha_0 h_K \leq \alpha_K \leq \alpha_1 h_K \quad \forall K \in \mathcal{T}_h. \tag{16}$$

Under this assumption we have:

Theorem 1. *Suppose that the uniqueness condition holds and that the family of finite element partitions $\{\mathcal{T}_h\}_{h>0}$ is non-degenerate and such that (15) holds. If α_K satisfies (16) then, for h small enough, there exists a unique solution to the discrete finite element problem (10) and (12) that verifies the stability estimate*

$$\|\mathbf{u}_h\|_1 + \|\nabla^h p_h\| + \|\boldsymbol{\xi}_h\| \leq C \|\mathbf{f}\|,$$

for a constant C independent of h .

Let us give now a convergence result for the solution of the discrete problem to the solution of the continuous Navier–Stokes problem in the norm in which stability has been proven, that is, in the \mathbf{H}^1 norm for the velocity and the L^2 norm of the mesh dependent pressure gradient defined in (13) and its projection

onto V_h . This is what we call ‘natural’ norm of the finite element method presented. Next, L^2 error estimates can be obtained using classical duality arguments, similar to those using in [5] for the GLS method.

To prove convergence we need to explicitly assume that the problem is p -regular. Furthermore, we also need a technical condition on the family of finite element partitions discussed in [2]. Under all these assumptions we can prove [2].

Theorem 2. *Suppose that the Navier–Stokes problem is p -regular, the uniqueness condition holds and that the family of finite element partitions $\{\mathcal{T}_h\}_{h>0}$ is non-degenerate and such that (15) holds. Assume also that the parameters α_K satisfy (16). Then, for h small enough, the solution of the discrete finite element problem satisfies the error estimate*

$$\|\mathbf{u} - \mathbf{u}_h\|_1 + \|\nabla^h p - \nabla^h p_h\| + \|\nabla^h p - \xi_h\| \leq CE(h), \tag{17}$$

where

$$E(h) := \inf_{\mathbf{v}_h \in V_{h,0}} \|\mathbf{u} - \mathbf{v}_h\|_1 + \inf_{\mathbf{v}_h \in V_{h,0}} \sum_K \frac{1}{h_K} \|\mathbf{u} - \mathbf{v}_h\|_{0,K} + \inf_{q_h \in Q_h} \|p - q_h\| + \inf_{q_h \in Q_h} \|\nabla^h p - \nabla^h q_h\| + \inf_{\boldsymbol{\eta}_h \in V_h} \|\nabla^h p - \boldsymbol{\eta}_h\| \tag{18}$$

and \mathbf{u} and p are the solution of the continuous problem.

Clearly, estimate (17) is optimal. From the standard approximation properties of the finite element spaces it follows that if $\mathbf{u} \in \mathbf{H}^r(\Omega) \cap \mathbf{V}$, $r \geq 1$, and $p \in H^s(\Omega) \cap Q$, $s \geq 1$, then the error function $E(h)$ in (18) behaves like h^k , with $k = \min\{r - 1, s, k_v, k_q + 1\}$.

4. Stabilization of the transient problem

In this section we present the extension of the stabilization technique described for the steady problem in the previous section to the transient case. We shall concentrate on two particular time integration schemes, namely, the monolithic trapezoidal rule (solving for the velocity and the pressure at the same time) and a particular fractional step method that allows to compute the different variables by solving uncoupled equations for each. The two particular cases considered are representative of monolithic and fractional step methods and the extension of the stabilization method to other time integration schemes of these types is straightforward.

4.1. Monolithic schemes

Let us consider first the trapezoidal rule. The discretization of (6) and (7) in this case consists in solving the following problem: from known \mathbf{u}^n and p^n , find $\mathbf{u}^{n+1} \in V_{st}$ and $p^{n+1} \in Q_{st}$ such that

$$\left(\frac{\delta \mathbf{u}^n}{\delta t}, \mathbf{v} \right) + c(\mathbf{u}^{n+\theta}, \mathbf{u}^{n+\theta}, \mathbf{v}) + a(\mathbf{u}^{n+\theta}, \mathbf{v}) - b(p^{n+1}, \mathbf{v}) = \langle \mathbf{f}^{n+1}, \mathbf{v} \rangle \quad \forall \mathbf{v} \in V_{st}, \tag{19}$$

$$b(q, \mathbf{u}^{n+1}) = 0 \quad \forall q \in Q_{st}, \tag{20}$$

where δt is the time step size, superscripts refer to the time step level, $\theta \in [0, 1]$ and we use the notation

$$\mathbf{u}^{n+\theta} := \theta \mathbf{u}^{n+1} + (1 - \theta) \mathbf{u}^n \text{ and } \delta \mathbf{u}^n := \mathbf{u}^{n+1} - \mathbf{u}^n.$$

The force term in (19) and below has to be understood as the time average of the force in the interval $[t^n, t^{n+1}]$, even though we use a superscript $n + 1$ to characterize it. The pressure value computed here has been identified as the pressure evaluated at t^{n+1} , although this is irrelevant for the velocity approximation. The problem can be written in terms of $\mathbf{u}^{n+\theta}$ by expressing \mathbf{u}^{n+1} in terms of it. Also, instead of using $c(\mathbf{u}^{n+\theta}, \mathbf{u}^{n+\theta}, \mathbf{v})$ for the convective term it is also possible to use a linearized form of it with the same order of approximation.

The Galerkin finite element approximation of problem (19) and (20) is standard. As for the stationary problem, the finite element spaces used to interpolate the velocity and the pressure need to satisfy the discrete inf–sup condition (see [12] for a complete discussion and analysis of this problem when $\theta = 1/2$). On the other hand, it is possible to avoid the need to satisfy this condition by extending the stabilization technique of the previous section to the transient case. This extension is straightforward. Using the same notation as before, the transient analogue of problem (10) and (12) is to find finite element approximations $(\mathbf{u}_h^{n+\theta}, p_h^{n+1})$ to $(\mathbf{u}^{n+\theta}, p^{n+1})$ and also ξ_h^{n+1} such that

$$\left(\frac{\delta \mathbf{u}_h^n}{\delta t}, \mathbf{v}_h \right) + c(\mathbf{u}_h^{n+\theta}, \mathbf{u}_h^{n+\theta}, \mathbf{v}_h) + a(\mathbf{u}_h^{n+\theta}, \mathbf{v}_h) - b(p_h^{n+1}, \mathbf{v}_h) = (\mathbf{f}^{n+1}, \mathbf{v}_h), \tag{21}$$

$$\sum_K \alpha_K^2 (\nabla p_h^{n+1}, \nabla q_h)_K - \sum_K \alpha_K (\xi_h^{n+\beta}, \nabla q_h)_K + b(q_h, \mathbf{u}_h^{n+1}) = 0, \tag{22}$$

$$- \sum_K \alpha_K (\nabla p_h^{n+1}, \boldsymbol{\eta}_h)_K + (\xi_h^{n+1}, \boldsymbol{\eta}_h) = 0 \tag{23}$$

for all $(\mathbf{v}_h, q_h, \boldsymbol{\eta}_h) \in V_{h,0} \times Q_h \times V_h$.

In this equations we have introduced the parameter β , whose values of interest are $\beta = 0$ and $\beta = 1$. In the first case, the pressure gradient projection is treated explicitly, whereas in the second it is treated implicitly. We shall see in the next section that this is possible, in the sense that if the parameters α_K are properly chosen the resulting time integration scheme is stable.

The choice of β is especially relevant from the computational point of view. To discuss this, it is convenient to introduce the matrix form of the algebraic system resulting from problem (21)–(23), which is

$$\mathbf{M}_0 \frac{\delta \mathbf{U}^n}{\delta t} + \mathbf{K}(\mathbf{U}^{n+\theta}) \mathbf{U}^{n+\theta} + \mathbf{G} \mathbf{P}^{n+1} = \mathbf{F}^{n+1}, \tag{24}$$

$$-\mathbf{L}_\alpha \mathbf{P}^{n+1} - \mathbf{G}_\alpha^t \boldsymbol{\Xi}^{n+\beta} + \mathbf{D} \mathbf{U}^{n+1} = \mathbf{0}, \tag{25}$$

$$-\mathbf{G}_\alpha \mathbf{P}^{n+1} + \mathbf{M} \boldsymbol{\Xi}^{n+1} = \mathbf{0}. \tag{26}$$

The notation employed here is as follows. First, we have used capital letters to denote the vectors of nodal unknowns of the corresponding lower case variables. Vector \mathbf{F}^{n+1} is the resulting force vector, matrix \mathbf{M}_0 is the Gram matrix for the velocity space $V_{h,0}$ and \mathbf{M} for the whole space V_h . Matrix \mathbf{K} comes from the discretization of both the viscous and convective terms and matrices \mathbf{G} , \mathbf{D} and \mathbf{L} come from the approximation of the gradient, the divergence and the Laplacian operators, respectively. When these matrices are modified by the appearance of the α_K parameters in the integrals we have introduced a subscript α .

Clearly, when $\beta = 0$ Eqs. (24) and (25) can be solved independently of (26), which can be solved for $\boldsymbol{\Xi}^{n+1}$ once the pressure \mathbf{P}^{n+1} is known. This makes the stabilized method only marginally more expensive than the standard Galerkin approach. The stability when $\beta = 0$ will be discussed in the next section.

From (26) it is found that the nodal unknowns of the pressure gradient projection are

$$\boldsymbol{\Xi}^{n+\beta} = \mathbf{M}^{-1} \mathbf{G}_\alpha \mathbf{P}^{n+\beta}.$$

The inversion of \mathbf{M} is trivial if it approximated by the standard lumped mass matrix. Otherwise, a linear system needs to be solved, although this can be done efficiently by a sort of iterative techniques.

Inserting the expression for $\boldsymbol{\Xi}^{n+\beta}$ in (25) yields

$$-\mathbf{L}_\alpha \mathbf{P}^{n+1} - \mathbf{G}_\alpha^t \mathbf{M}^{-1} \mathbf{G}_\alpha \mathbf{P}^{n+\beta} + \mathbf{D} \mathbf{U}^{n+1} = \mathbf{0}.$$

This is the modified discrete continuity equation solved by the stabilized method instead of the equation $\mathbf{D} \mathbf{U}^{n+1} = \mathbf{0}$. For $\beta = 1$ it can be written as

$$\mathbf{B}_\alpha \mathbf{P}^{n+1} + \mathbf{D} \mathbf{U}^{n+1} = \mathbf{0},$$

where we have introduced the matrix

$$\mathbf{B}_\alpha := -\mathbf{L}_\alpha - \mathbf{G}_\alpha^t \mathbf{M}^{-1} \mathbf{G}_\alpha.$$

It can be proved that this matrix is positive-semidefinite [3], thus explaining from a purely algebraic point of view why the stability of the pressure is increased by the present stabilization technique.

4.2. Fractional step schemes

4.2.1. Matrix form

Let us discuss now the application of this stabilization method when a fractional step scheme is used for the time integration. For that, we consider a purely algebraic presentation of these types of methods. Let us split (24) into the following two equations:

$$M_0 \frac{1}{\delta t} (\hat{U}^{n+1} - U^n) + K(U^{n+\theta})U^{n+\theta} + \gamma GP^n = f^{n+1}, \tag{27}$$

$$M_0 \frac{1}{\delta t} (U^{n+1} - \hat{U}^{n+1}) + G(P^{n+1} - \gamma P^n) = \mathbf{0}, \tag{28}$$

where \hat{U}^{n+1} is an auxiliary vector of unknowns and $\gamma = 0$ or 1 . If we write $\hat{U}^{n+\theta} = \theta \hat{U}^{n+1} + (1 - \theta)U^n$, Eq. (27) can be solved independently of (28) if the approximation

$$K(\hat{U}^{n+\theta})\hat{U}^{n+\theta} \approx K(U^{n+\theta})U^{n+\theta} \tag{29}$$

is used. This is the source of splitting error. Since from (28) it follows that

$$U^{n+1} = \hat{U}^{n+1} - \delta t M_0^{-1} G(P^{n+1} - \gamma P^n), \tag{30}$$

it is clear that the approximation (29) is formally of order $O(\delta t)$ if $\gamma = 0$ and of order $O(\delta t^2)$ if $\gamma = 1$.

Using the expression for U^{n+1} obtained in (30) and inserting it into (25), the resulting discrete continuity equation is

$$\begin{aligned} \mathbf{0} &= -L_\alpha P^{n+1} - G_\alpha^t \Xi^{n+\beta} + D U^{n+1} \\ &= -L_\alpha P^{n+1} - G_\alpha^t \Xi^{n+\beta} + D \hat{U}^{n+1} - \delta t D M_0^{-1} G(P^{n+1} - \gamma P^n). \end{aligned} \tag{31}$$

A further approximation is now required in order to make the final discrete problem easier to solve. It consists of approximating the discrete Laplacian matrix appearing in the last term of (31) computed as $D M_0^{-1} G$ by the standard discrete Laplacian L , that is, to take

$$D M_0^{-1} G(P^{n+1} - \gamma P^n) \approx L(P^{n+1} - \gamma P^n). \tag{32}$$

Using this, together with the splitting (27) and (28) of (24) and the approximation (29), the final algebraic system of equations to be solved at each time step is

$$M_0 \frac{1}{\delta t} (\hat{U}^{n+1} - U^n) + K(\hat{U}^{n+\theta})\hat{U}^{n+\theta} + \gamma GP^n = F^{n+1}, \tag{33}$$

$$\delta t L(P^{n+1} - \gamma P^n) = D \hat{U}^{n+1} - L_\alpha P^{n+1} - G_\alpha^t \Xi^{n+\beta}, \tag{34}$$

$$-G_\alpha P^{n+1} + M \Xi^{n+1} = \mathbf{0}, \tag{35}$$

$$M_0 \frac{1}{\delta t} (U^{n+1} - \hat{U}^{n+1}) + G(P^{n+1} - \gamma P^n) = \mathbf{0}. \tag{36}$$

This equations are very attractive from the implementation point of view because they are uncoupled. Eq. (33) can be solved for \hat{U}^{n+1} , and then the pressure can be computed directly from (34) if $\beta = 0$ or coupling this equation together with (35) if $\beta = 1$. Finally, (36) can be solved for U^{n+1} .

4.2.2. Motivation from a continuous problem

To motivate the method given by (33)–(36) let us assume for a moment that the parameters α_K are constant and equal to a value α . If we take the trapezoidal rule as given by (21)–(23) as the starting point for

the design of the fractional step method, it is seen that the continuous equations that are being approximated are

$$\frac{\partial \mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla) \mathbf{u} - \nu \Delta \mathbf{u} + \nabla p = \mathbf{f}, \tag{37}$$

$$-\alpha^2 \Delta p + \alpha \nabla \cdot \boldsymbol{\xi} + \nabla \cdot \mathbf{u} = 0, \tag{38}$$

$$-\alpha \nabla p + \boldsymbol{\xi} = \mathbf{0}, \tag{39}$$

with the appropriate boundary conditions. Let us introduce the abbreviation

$$N(\mathbf{u}) := (\mathbf{u} \cdot \nabla) \mathbf{u} - \nu \Delta \mathbf{u}.$$

To discretize problem (37)–(39) let us consider the simplest fractional step scheme [13,14], which consists of solving the problem

$$\frac{1}{\delta t} (\hat{\mathbf{u}}^{n+1} - \mathbf{u}^n) + N(\hat{\mathbf{u}}^{n+1}) + \gamma \nabla p^n = \mathbf{f}^{n+1}, \tag{40}$$

$$\frac{1}{\delta t} (\mathbf{u}^{n+1} - \hat{\mathbf{u}}^{n+1}) + \nabla (p^{n+1} - \gamma p^n) = \mathbf{0}, \tag{41}$$

$$-\alpha^2 \Delta p^{n+1} + \alpha \nabla \cdot \boldsymbol{\xi}^{n+\beta} + \nabla \cdot \mathbf{u}^{n+1} = 0, \tag{42}$$

$$-\alpha \nabla p^{n+1} + \boldsymbol{\xi}^{n+1} = \mathbf{0}. \tag{43}$$

In this equations, $\hat{\mathbf{u}}^{n+1}$ is an intermediate velocity which is computed from a pressure guess and for which the continuity equation is not enforced. If Eqs. (40) and (41) are added up, we recover the standard discretization of the momentum equation using the trapezoidal rule for $\theta = 1$, except for the fact that N is evaluated with $\hat{\mathbf{u}}^{n+1}$ instead of being evaluated with \mathbf{u}^{n+1} . Here is where the splitting error comes in. This error is of order $O(\delta t^2)$ if $\gamma = 1$ (that is, the pressure guess is the pressure obtained in the previous time step) and of order $O(\delta t)$ if $\gamma = 0$ (the pressure guess is zero). Of course, the splitting error is in addition to the error associated to the time integration scheme being used, which in the case considered is also of order $O(\delta t)$.

In principle, the way to solve (40)–(43) is to start with (40), which is independent of the other equations, and then solve (41)–(43), with the possibility of uncoupling the last equation if $\beta = 0$. A further computational advantage can be achieved if we take the divergence of (41) and use the modified continuity Eq. (42). This leads to

$$\left(1 + \frac{\alpha^2}{\delta t}\right) \Delta p^{n+1} = \frac{1}{\delta t} \nabla \cdot \hat{\mathbf{u}}^{n+1} + \gamma \Delta p^n + \frac{\alpha}{\delta t} \nabla \cdot \boldsymbol{\xi}^{n+\beta}. \tag{44}$$

For $\beta = 0$, this allows to solve, successively, (40) for $\hat{\mathbf{u}}^{n+1}$, (44) for p^{n+1} , (43) for $\boldsymbol{\xi}^{n+1}$ and finally (41) for \mathbf{u}^{n+1} . This fact is extremely attractive from the computational standpoint and is probably the main reason for using fractional step or splitting techniques (see [15] for a review of such methods). However, there is a point that we have deliberately omitted, which is the treatment of the boundary conditions. This is irrelevant for our purpose of showing how to apply the stabilization method to this type of schemes.

Once the finite element approximation of the problem has been performed, the final algebraic system of equations to be solved is (33)–(36), which is obtained from the discretization of Eqs. (40), (44), (43) and (41). Now the parameters α_K can be considered again as defined element by element and take the resulting formulation as the proposed stabilized fractional step finite element method.

4.2.3. Discrete variational form

The matrix system given by (33)–(36) corresponds to the following discrete variational equations:

$$\frac{1}{\delta t} (\hat{\mathbf{u}}_h^{n+1} - \mathbf{u}_h^n, \mathbf{v}_h) + c(\hat{\mathbf{u}}_h^{n+\theta}, \hat{\mathbf{u}}_h^{n+\theta}, \mathbf{v}_h) + a(\hat{\mathbf{u}}_h^{n+\theta}, \mathbf{v}_h) - \gamma b(p_h^n, \mathbf{v}_h) = (\mathbf{f}^{n+1}, \mathbf{v}_h), \tag{45}$$

$$\sum_K (\alpha_K^2 \nabla p_h^{n+1} - \alpha_K \xi_h^{n+\beta}, \nabla q_h)_K + \sum_K \delta t (\nabla p_h^{n+1} - \gamma \nabla p_h^n, \nabla q_h)_K + b(q_h, \hat{\mathbf{u}}_h^{n+1}) = 0, \tag{46}$$

$$- \sum_K \alpha_K (\nabla p_h^{n+1}, \boldsymbol{\eta}_h)_K + (\xi_h^{n+1}, \boldsymbol{\eta}_h) = 0, \tag{47}$$

$$\frac{1}{\delta t} (\mathbf{u}_h^{n+1} - \hat{\mathbf{u}}_h^{n+1}, \mathbf{v}_h) - b(p_h^{n+1} - \gamma p_h^n, \mathbf{v}_h) = 0, \tag{48}$$

which must be satisfied for all $(\mathbf{v}_h, q_h, \boldsymbol{\eta}_h) \in \mathbf{V}_{h,0} \times Q_h \times \mathbf{V}_h$. This variational form of the problem allows us to draw some conclusions.

Let us consider first the case $\alpha_K = 0$, that is, the classical fractional step scheme. In the case $\gamma = 0$, it is seen from (46) that a pressure Laplacian multiplied by δt is introduced in the incompressibility equation. This is the reason why the resulting scheme allows to use equal velocity–pressure interpolation, an observation found for example in [16–19]. In [20], the scheme is shown to be stable by considering it as a (perturbed) problem for the intermediate velocity $\hat{\mathbf{u}}_h^{n+1}$ instead of a problem for the end-of-step velocity \mathbf{u}_h^{n+1} . However, the fact that the stability (and therefore the convergence) of the scheme relies on the time step size δt has severe drawbacks. If δt is too large, the incompressibility constraint will be poorly approximated, whereas if it is too small there will be a lack of stability. From the analysis presented for the stationary case it follows that δt must be of order $O(h^2)$ to have both good stability and accuracy. It is known that good results are obtained if δt is computed as the critical time step of the explicit approximation to the equations (that is, $\theta = 0$).

The case $\gamma = 1$ leads to a scheme in which the splitting error is of second order, and therefore with a better approximation to the incompressibility of the flow. However, in this case it is observed from (46) that stability may deteriorate as the steady state is reached, that is, as $p_h^{n+1} - p_h^n \rightarrow 0$ (in a certain norm). In order to prove convergence of the fully discrete scheme to the solution of the continuous problem the discrete velocity–pressure spaces must satisfy the inf–sup condition (see [21]).

The use of α_K satisfying condition (16) allows to circumvent this misbehavior, since the stability does not rely any more on the time step size δt . Now $\gamma = 1$ can be used, yielding a scheme with second order splitting error and keeping the most interesting properties of fractional schemes, namely, the uncoupling of the equations and the possibility of using equal velocity–pressure interpolations.

5. A stability estimate

The purpose of this section is to prove a stability estimate for a simplified case of the fully discrete problem (21)–(23). The simplifications that will be assumed are that we shall take α_K equal for all the elements to a parameter α and that we shall take $\theta = 1$. These assumptions are not restrictive if, as in our case, the purpose of the analysis is to get insight on the stability nature of the scheme analyzed.

The problem that we consider is therefore

$$\left(\frac{\delta \mathbf{u}_h^n}{\delta t}, \mathbf{v}_h \right) + c(\mathbf{u}_h^{n+1}, \mathbf{u}_h^{n+1}, \mathbf{v}_h) + a(\mathbf{u}_h^{n+1}, \mathbf{v}_h) - b(p_h^{n+1}, \mathbf{v}_h) = (\mathbf{f}^{n+1}, \mathbf{v}_h), \tag{49}$$

$$\alpha^2 (\nabla p_h^{n+1}, \nabla q_h) - \alpha (\xi_h^{n+\beta}, \nabla q_h) + b(q_h, \mathbf{u}_h^{n+1}) = 0, \tag{50}$$

$$-\alpha (\nabla p_h^{n+1}, \boldsymbol{\eta}_h) + (\xi_h^{n+1}, \boldsymbol{\eta}_h) = 0, \tag{51}$$

for all $(\mathbf{v}_h, q_h, \boldsymbol{\eta}_h) \in \mathbf{V}_{h,0} \times Q_h \times \mathbf{V}_h$. As it is seen from (50), we still consider the possibility of treating the pressure gradient projection either implicitly or explicitly.

We shall also make an assumption on the finite element mesh, namely, that it is quasi-uniform and of diameter h . In this case, the inverse estimates

$$\|\nabla \mathbf{v}_h\| \leq \frac{C_{\text{inv}}}{h} \|\mathbf{v}_h\|, \quad \|\mathbf{v}_h\|_{L^\infty(\Omega)} \leq \frac{C_{\text{inv}}}{h^{d/2}} \|\mathbf{v}_h\| \quad \forall \mathbf{v}_h \in \mathbf{V}_h, \tag{52}$$

hold true. Again, the same results as for the steady-state case can be proved only assuming that the mesh is non-degenerate, but this does not introduce any further insight on the method.

From now on we use C , possibly with subscripts, to denote positive constants independent of h and δt , not necessarily the same at different occurrences.

The idea is to obtain a stability estimate for the sequence of velocities \mathbf{u}_h^n as well as for the sequence $\nabla^h p_h^n = h \nabla p_h^n$. In order to write down these estimates, let us introduce the following definition. If $\{f^n\}_{n=0}^N$ is a sequence of functions in a Banach space X with norm $\|\cdot\|_X$, we say that

$$\{f^n\} \in \ell^p(X) \quad \text{iff} \quad \sum_{n=0}^N \|f^n\|_X^p \delta t \leq C < \infty, \tag{53}$$

for $1 \leq p < \infty$, and

$$\{f^n\} \in \ell^\infty(X) \quad \text{iff} \quad \max_{n=0, \dots, N} \|f^n\|_X \leq C < \infty. \tag{54}$$

Our main stability result for problem (49)–(51) the following:

Theorem 3. *Suppose that the family of finite element partitions $\{\mathcal{T}_h\}_{h>0}$ is quasi-uniform and such that (15) holds. If α satisfies (16), then the solution to problem (49)–(51) verifies*

$$\{\mathbf{u}_h^n\} \in \ell^\infty(\mathbf{L}^2(\Omega)) \cap \ell^2(\mathbf{H}^1(\Omega)), \tag{55}$$

$$\{\nabla^h p_h^n\} \in \ell^1(\mathbf{L}^2(\Omega)), \tag{56}$$

provided $\{f^n\} \in \ell^2(\mathbf{L}^2(\Omega))$. In particular, there exist constants C_1 and C_2 , independent of h and δt , such that

$$\|\mathbf{u}_h^{N+1}\|^2 + \sum_{n=0}^N \delta t \nu \|\nabla \mathbf{u}_h^{n+1}\|^2 + \sum_{n=0}^N \delta t h \|\nabla p_h^{n+1}\| \leq C_1 \sum_{n=0}^N \delta t \|f^{n+1}\|^2 + C_2.$$

Proof. To simplify the notation in what follows, let us write

$$\mathbf{g}_i^n := P_{h,i}(\alpha \nabla p_h^n), \quad i = 1, 2, 3.$$

From (51) we have that

$$\boldsymbol{\xi}_h^{n+1} = \mathbf{g}_1^{n+1} + \mathbf{g}_2^{n+1},$$

and therefore the continuity equation can be written as

$$\alpha(\mathbf{g}_3^{n+1}, \nabla q_h) + \bar{\beta} \alpha(\boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n, \nabla q_h) + b(q_h, \mathbf{u}_h^{n+1}) = 0, \tag{57}$$

where $\bar{\beta} := 1 - \beta$. We will be interested only in the cases $\bar{\beta} = 0$ and $\bar{\beta} = 1$. Adding up (57) with the momentum equation (49) taking $\mathbf{v}_h = \mathbf{u}_h^{n+1}$ and $q_h = p_h^{n+1}$ it is found that

$$\left(\mathbf{u}_h^{n+1}, \frac{\delta \mathbf{u}_h^n}{\delta t} \right) + \nu \|\nabla \mathbf{u}_h^{n+1}\|^2 + \|\mathbf{g}_3^{n+1}\|^2 + \bar{\beta} (\boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n, \alpha \nabla p_h^{n+1}) = (f^{n+1}, \mathbf{u}_h^{n+1}). \tag{58}$$

Taking into account that

$$\begin{aligned} (\boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n, \alpha \nabla p_h^{n+1}) &= (\boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n, \boldsymbol{\xi}_h^{n+1}) = \frac{1}{2} \|\boldsymbol{\xi}_h^{n+1}\|^2 - \frac{1}{2} \|\boldsymbol{\xi}_h^n\|^2 + \frac{1}{2} \|\boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n\|^2, \\ (f^{n+1}, \mathbf{u}_h^{n+1}) &\leq C \|f^{n+1}\|^2 + \frac{\nu}{2} \|\nabla \mathbf{u}_h^{n+1}\|^2, \end{aligned}$$

it follows from (58) that

$$\begin{aligned} & \| \mathbf{u}_h^{n+1} \|^2 - \| \mathbf{u}_h^n \|^2 + \| \mathbf{u}_h^{n+1} - \mathbf{u}_h^n \|^2 + \delta t \nu \| \nabla \mathbf{u}_h^{n+1} \|^2 + \delta t \| \mathbf{g}_3^{n+1} \|^2 \\ & + \delta t \bar{\beta} \left(\| \boldsymbol{\xi}_h^{n+1} \|^2 - \| \boldsymbol{\xi}_h^n \|^2 + \| \boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n \|^2 \right) \leq C \delta t \| \mathbf{f}^{n+1} \|^2, \end{aligned}$$

and, adding these equations up from $n = 0$ to the last time interval N it follows that

$$\begin{aligned} & \| \mathbf{u}_h^{N+1} \|^2 + \sum_{n=0}^N \| \mathbf{u}_h^{n+1} - \mathbf{u}_h^n \|^2 + \sum_{n=0}^N \delta t \nu \| \nabla \mathbf{u}_h^{n+1} \|^2 + \sum_{n=0}^N \delta t \| \mathbf{g}_3^{n+1} \|^2 \\ & + \delta t \bar{\beta} \left(\| \boldsymbol{\xi}_h^{N+1} \|^2 + \sum_{n=0}^N \| \boldsymbol{\xi}_h^{n+1} - \boldsymbol{\xi}_h^n \|^2 \right) \leq C_1 \sum_{n=0}^N \delta t \| \mathbf{f}^{n+1} \|^2 + C_2. \end{aligned} \tag{59}$$

Of course this is also true for any $N_0 \leq N$ and proves (55). To prove the stability estimate for the pressure, observe from (59) that \mathbf{g}_3^n is already bounded in $\ell^2(\mathbf{L}^2(\Omega))$, and therefore it is also bounded in $\ell^1(\mathbf{L}^2(\Omega))$. Assuming that condition (15) holds, it only remains to obtain a bound for \mathbf{g}_1^n . For that, let us take $\mathbf{v}_h = P_{h,1}(\nabla P_h^{n+1})$ in the momentum equation. Using the inverse estimate (52) and Schwarz’s inequality we obtain

$$\begin{aligned} \| P_{h,1}(\nabla P_h^{n+1}) \|^2 &= - \left(P_{h,1}(\nabla P_h^{n+1}), \frac{\delta \mathbf{u}_h^n}{\delta t} \right) + (\mathbf{f}^{n+1}, P_{h,1}(\nabla P_h^{n+1})) \\ &\quad - a(\mathbf{u}_h^{n+1}, P_{h,1}(\nabla P_h^{n+1})) - c(\mathbf{u}_h^{n+1}, \mathbf{u}_h^{n+1}, P_{h,1}(\nabla P_h^{n+1})) \\ &\leq \| P_{h,1}(\nabla P_h^{n+1}) \| \left\| \frac{\delta \mathbf{u}_h^n}{\delta t} \right\| + \| \mathbf{f}^{n+1} \| \| P_{h,1}(\nabla P_h^{n+1}) \| \\ &\quad + \left(N_a \| \mathbf{u}_h^{n+1} \| + N_c \| \mathbf{u}_h^{n+1} \|^2 \right) \frac{C_{\text{inv}}}{h} \| P_{h,1}(\nabla P_h^{n+1}) \|. \end{aligned}$$

Dividing both sides by $P_{h,1}(\nabla P_h^{n+1})$ (assuming it is $\neq 0$), multiplying by $h \delta t$ and adding the resulting equations for $n = 0, 1, \dots, N$ it follows that

$$\sum_{n=0}^N \delta t h \| P_{h,1}(\nabla P_h^{n+1}) \| \leq \sum_{n=0}^N \| \mathbf{u}_h^{n+1} - \mathbf{u}_h^n \| + \sum_{n=0}^N \delta t \| \mathbf{f}^{n+1} \| + \sum_{n=0}^N \delta t N_a \| \mathbf{u}_h^{n+1} \| + \sum_{n=0}^N \delta t N_c \| \mathbf{u}_h^{n+1} \|^2.$$

The bound for \mathbf{g}_1^n follows from the behavior assumed for α and from estimate (59). This proves (56). \square

The stability estimate for the pressure gradient projection given by (56) can be improved in some cases, although not in the general case of the 3D Navier–Stokes equations. This is what the following result states.

Theorem 4. *Under the same conditions as in Theorem 3, suppose also that either $c = 0$ (Stokes problem) or $d = 2$ (2D flows). In the case $\beta = 0$ suppose also that $\delta t \geq Ch^2$. Then the solution to problem (49)–(51) verifies the stability estimate*

$$\{ \nabla^h p_h^n \} \in \ell^2(\mathbf{L}^2(\Omega)). \tag{60}$$

Proof. Let us consider now the continuity Eq. (57) for time step n and $n + 1$. Subtracting both it is found that

$$\frac{\alpha}{\delta t} (\mathbf{g}_3^{n+1} - \mathbf{g}_3^n, \nabla q_h) + \bar{\beta} \frac{\alpha}{\delta t} (\boldsymbol{\xi}_h^{n+1} - 2\boldsymbol{\xi}_h^n + \boldsymbol{\xi}_h^{n-1}, \nabla q_h) - \left(\nabla q_h, \frac{\delta \mathbf{u}_h^n}{\delta t} \right) = 0. \tag{61}$$

Taking now $q_h = p_h^{n+1}$ we get

$$\begin{aligned} \left(P_{h,1}(\nabla p_h^{n+1}), \frac{\delta \mathbf{u}_h^n}{\delta t} \right) &= \frac{1}{\delta t} (\mathbf{g}_3^{n+1} - \mathbf{g}_3^n, \mathbf{g}_3^{n+1}) + \bar{\beta} \frac{1}{\delta t} (\boldsymbol{\xi}_h^{n+1} - 2\boldsymbol{\xi}_h^n + \boldsymbol{\xi}_h^{n-1}, \boldsymbol{\xi}_h^{n+1}) \\ &= \frac{1}{2\delta t} \left(\|\mathbf{g}_3^{n+1}\|^2 - \|\mathbf{g}_3^n\|^2 + \|\mathbf{g}_3^{n+1} - \mathbf{g}_3^n\|^2 \right) \\ &\quad + \bar{\beta} \frac{1}{2\delta t} \left(\|\boldsymbol{\xi}_h^{n+1}\|^2 + \|\boldsymbol{\xi}_h^{n-1}\|^2 - \|2\boldsymbol{\xi}_h^n\|^2 - \|2\boldsymbol{\xi}_h^n - \boldsymbol{\xi}_h^{n-1}\|^2 \right) \\ &\quad + \|\boldsymbol{\xi}_h^{n+1} - 2\boldsymbol{\xi}_h^n + \boldsymbol{\xi}_h^{n-1}\|^2. \end{aligned} \tag{62}$$

Taking now $\mathbf{v}_h = P_{h,1}(\nabla p_h^{n+1})$ in the momentum equation, using the continuity of a in $\mathbf{H}^1(\Omega)^2$ and the continuity of c in $\mathbf{L}^2(\Omega) \times \mathbf{H}^1(\Omega) \times \mathbf{L}^\infty(\Omega)$ and the inverse estimates in (52) it follows that

$$\begin{aligned} \left(P_{h,1}(\nabla p_h^{n+1}), \frac{\delta \mathbf{u}_h^n}{\delta t} \right) + \|P_{h,1}(\nabla p_h^{n+1})\|^2 &= (\mathbf{f}^{n+1}, P_{h,1}(\nabla p_h^{n+1})) \\ &\quad - a(\mathbf{u}_h^{n+1}, P_{h,1}(\nabla p_h^{n+1})) - c(\mathbf{u}_h^{n+1}, \mathbf{u}_h^{n+1}, P_{h,1}(\nabla p_h^{n+1})) \\ &\leq \|\mathbf{f}^{n+1}\| \|P_{h,1}(\nabla p_h^{n+1})\| + N_a \|\mathbf{u}_h^{n+1}\| \|P_{h,1}(\nabla p_h^{n+1})\| \\ &\quad + N_c \|\mathbf{u}_h^{n+1}\| \|\mathbf{u}_h^{n+1}\| \|P_{h,1}(\nabla p_h^{n+1})\|_{L^\infty(\Omega)} \\ &\leq \|\mathbf{f}^{n+1}\| \|P_{h,1}(\nabla p_h^{n+1})\| \\ &\quad + C \|\mathbf{u}_h^{n+1}\| \left(\frac{N_a}{h} + \frac{N_c}{h^{d/2}} \|\mathbf{u}_h^{n+1}\| \right) \|P_{h,1}(\nabla p_h^{n+1})\|. \end{aligned}$$

Since the velocities are in $\ell^\infty(\mathbf{L}^2(\Omega))$, the norm inside the parenthesis is bounded. Also, either if $N_c = 0$ (Stokes problem) or if $d = 2$ this parenthesis behaves as $1/h$. Therefore, from Young’s and Poincaré’s inequalities we have

$$\left(P_{h,1}(\nabla p_h^{n+1}), \frac{\delta \mathbf{u}_h^n}{\delta t} \right) + \|P_{h,1}(\nabla p_h^{n+1})\|^2 \leq C \|\mathbf{f}^{n+1}\|^2 + C \frac{\nu}{h^2} \|\nabla \mathbf{u}_h^{n+1}\|^2 + \frac{1}{2} \|P_{h,1}(\nabla p_h^{n+1})\|^2,$$

and, using (62) and multiplying by $h^2 \delta t$,

$$\begin{aligned} \delta t h^2 \|P_{h,1}(\nabla p_h^{n+1})\|^2 + h^2 \left(\|\mathbf{g}_3^{n+1}\|^2 - \|\mathbf{g}_3^n\|^2 + \|\mathbf{g}_3^{n+1} - \mathbf{g}_3^n\|^2 \right) \\ + \bar{\beta} h^2 \left(\|\boldsymbol{\xi}_h^{n+1}\|^2 + \|\boldsymbol{\xi}_h^{n-1}\|^2 - \|2\boldsymbol{\xi}_h^n\|^2 - \|2\boldsymbol{\xi}_h^n - \boldsymbol{\xi}_h^{n-1}\|^2 + \|\boldsymbol{\xi}_h^{n+1} - 2\boldsymbol{\xi}_h^n + \boldsymbol{\xi}_h^{n-1}\|^2 \right) \\ \leq C_1 \delta t \|\mathbf{f}^{n+1}\|^2 + C_2 \delta t \|\nu \nabla \mathbf{u}_h^{n+1}\|^2. \end{aligned} \tag{63}$$

Adding up from $n = 0$ up to N (and neglecting some positive terms) we get

$$\begin{aligned} \sum_{n=0}^N \delta t h^2 \|P_{h,1}(\nabla p_h^{n+1})\|^2 + \bar{\beta} h^2 \|\boldsymbol{\xi}_h^{N+1}\|^2 \\ \leq C_1 \sum_{n=0}^N \delta t \|\mathbf{f}^{n+1}\|^2 + C_2 \sum_{n=0}^N \delta t \|\nu \nabla \mathbf{u}_h^{n+1}\|^2 + \bar{\beta} h^2 \left(\|\boldsymbol{\xi}_h^N\|^2 + 2 \sum_{n=0}^N \|\boldsymbol{\xi}_h^n - \boldsymbol{\xi}_h^{n-1}\|^2 \right) \end{aligned} \tag{64}$$

Finally, we can use condition (16) on α in (59) to obtain an estimate for the norm of $hP_{13}(\nabla p_h^{n+1})$, which allows to obtain an estimate for the norm of the whole pressure gradient if the weak inf–sup condition (15) holds. The combination of (59) and (64) yields therefore

$$\sum_{n=0}^N \delta t h^2 \|\nabla p_h^{n+1}\|^2 \leq \left(C_1 + C_2 \bar{\beta} \frac{h^2}{\delta t} \right) \left(\sum_{n=0}^N \delta t \|\mathbf{f}^{n+1}\|^2 + C_3 \right).$$

This completes the proof of the theorem. \square

It is observed from Theorems 3 and 4 that each of the members of the sequence of velocities are L^2 -stable, whereas the L^2 spatial norm of the velocity gradient and the pressure gradient (multiplied by h) are stable in the sense that the discrete approximation to their L^2 norm (or L^1 norm) in time is bounded. This is frequently expressed by saying that the velocities are strongly stable in L^2 and weakly in H^1 , whereas the pressure gradient (multiplied by h) is weakly stable in L^2 .

A further remark refers to the need for the condition $\delta t \geq Ch^2$ in Theorem 4 when the projection of the pressure gradient is treated explicitly. This expresses the fact that the term added to the incompressibility constraint must be small compared to δt . However, it is observed from (59) that without this assumption we already have control over the norm of \mathbf{g}_3^{n+1} and only the norm of $P_{h,1}(\nabla p_h^{n+1})$ needs to be controlled (assuming that the weak inf–sup condition holds true). On the other hand, very few components of ∇p_h belong only to $V_{h,0}$ (none for linear and multilinear elements!) and thus a bound for $P_{h,3}(\alpha \nabla p_h^{n+1})$ implies stability for most of the pressure components. In practice, we have never observed the need for having δt large enough and we have always obtained perfectly stable solutions when the projection of the pressure gradient is treated explicitly. Moreover, in Theorem 3 we have seen that no conditions on δt are needed to obtain stability in $\ell^1(L^2(\Omega))$ for the pressure gradient.

6. Numerical tests

In this section we show the numerical results obtained with the stabilized, monolithic scheme (21)–(23) in two different problems, a test case with an analytical solution and the standard problem of the flow past a circular cylinder.

6.1. A case with an analytical solution

We consider the following analytical solution of the unsteady, incompressible Navier–Stokes equations in two dimensions, called the Taylor vortex problem (see [22,23]):

$$\begin{aligned} u(x, y, t) &= -\cos(\pi x) \sin(\pi y) e^{-2v\pi^2 t} \\ v(x, y, t) &= \sin(\pi x) \cos(\pi y) e^{-2v\pi^2 t} \\ p(x, y, t) &= (-1/4) (\cos(2\pi x) + \cos(2\pi y)) e^{-4v\pi^2 t} \end{aligned} \tag{65}$$

where $v > 0$ is the fluid’s kinematic viscosity (which we took equal to 0.1). We employ this example to check the accuracy properties of the numerical scheme (21)–(23) with respect to the time discretization for different values of the parameter θ . The case $\theta = 1$ is studied analytically in [24], where we prove first order error estimates in the time step size and optimal order estimates in the mesh size with the given finite element interpolation for the fully discrete solution.

We took a computational domain consisting of the unit square $\Omega = [0, 1] \times [0, 1]$, and imposed time-dependent, Dirichlet-type boundary conditions given by the value of the analytical solution (65) on the boundary of Ω . We used a uniform 31×31 noded mesh to discretize the domain, and the same mesh points were used to define both a Q_1 and a Q_2 quadrilateral element interpolation, for comparison purposes.

We integrated the Navier–Stokes Eqs. (1)–(4) with the scheme (21)–(23) for $\theta = 1$ and $\theta = 1/2$ with different time step sizes, starting from (65) at $t = 0$ as a velocity initial condition, to a final time $T = 1$. We then calculated the L^2 -norm of the velocity, the velocity gradient and the pressure gradient errors with respect to the analytical solution (65) at $t = 1$, these are the variables the stability of which is established in Theorem 3 and for which error estimates are derived in [24]. For each value of θ , we considered the cases $\beta = 0, 1$, corresponding to an explicit/implicit approximation of the pressure gradient. Also, when $\theta = 1$ we considered the possibility of treating explicitly the advection velocity of the non-linear convective term within each time step, that is to say, to take this convective term as $c(\mathbf{u}_h^n, \mathbf{u}_h^{n+1}, \mathbf{v}_h)$. This also leads to a first order time integration scheme.

The backward Euler case $\theta = 1$ displayed first order accuracy in all three variables and all four methods. The results obtained in this case with a Q_1 element interpolation can be seen in Figs. 1–3, where we plot the

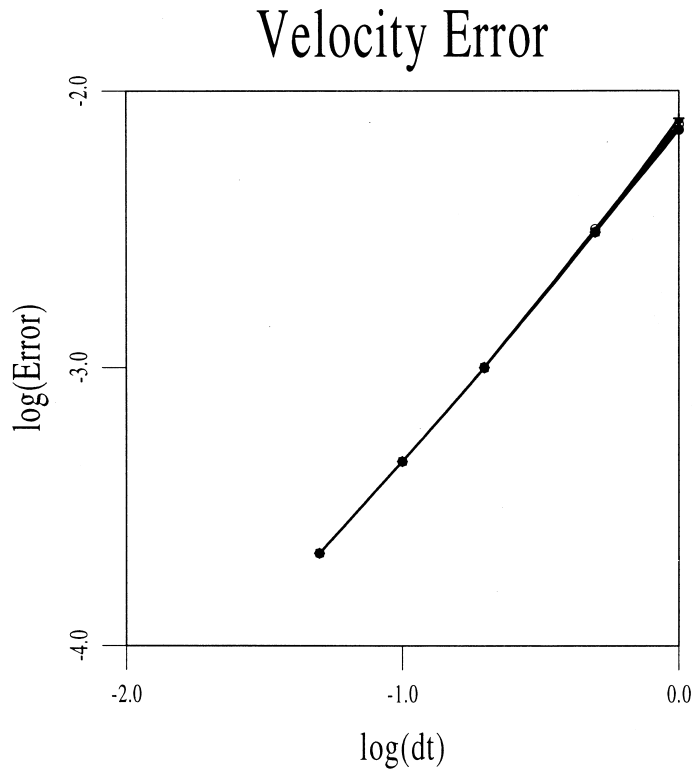


Fig. 1. Taylor flow, $\theta = 1$, Q_1 element, velocity errors: \circ $\beta = 0$, linearized scheme; \bullet $\beta = 1$, linearized scheme; $+$ $\beta = 0$, non-linear scheme; \times $\beta = 1$, non-linear scheme.

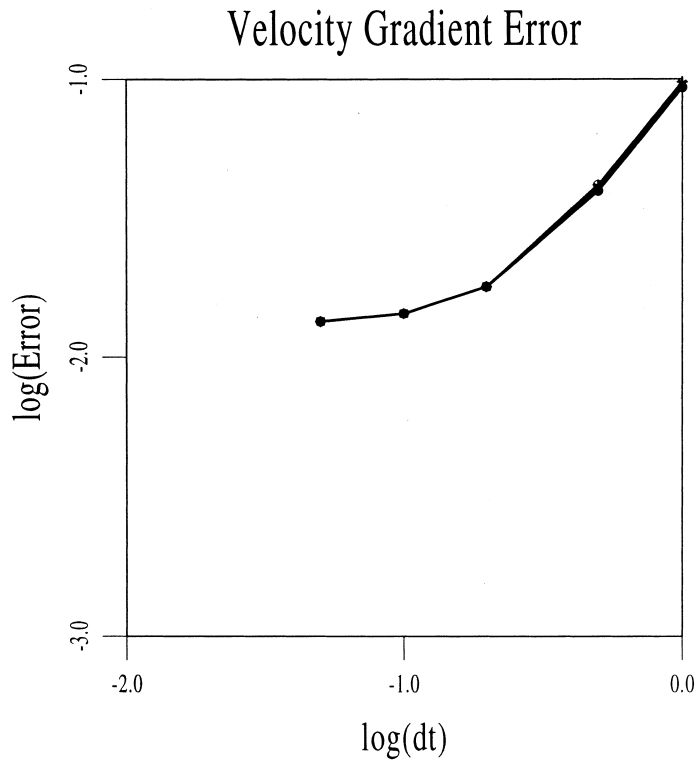


Fig. 2. Taylor flow, $\theta = 1$, Q_1 element, velocity gradient errors: \circ $\beta = 0$, linearized scheme; \bullet $\beta = 1$, linearized scheme; $+$ $\beta = 0$, non-linear scheme; \times $\beta = 1$, non-linear scheme.

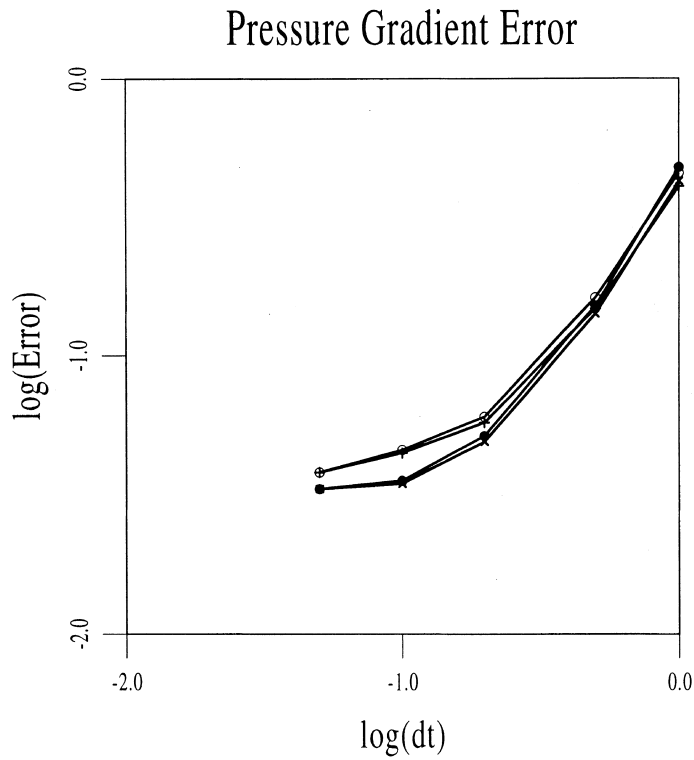


Fig. 3. Taylor flow, $\theta = 1$, Q_1 element, pressure gradient errors: \circ $\beta = 0$, linearized scheme; \bullet $\beta = 1$, linearized scheme; $+$ $\beta = 0$, non-linear scheme; \times $\beta = 1$, non-linear scheme.

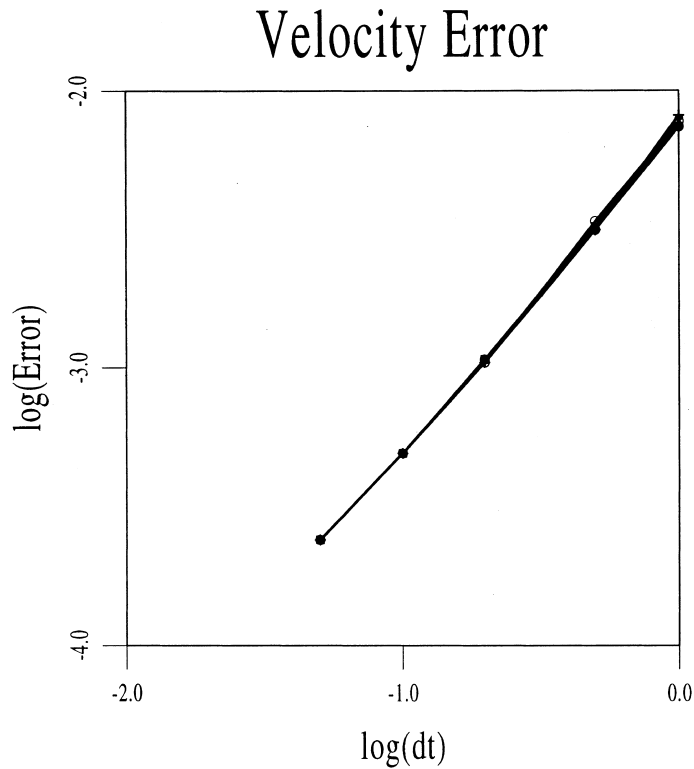


Fig. 4. Taylor flow, $\theta = 1$, Q_2 element, velocity errors: \circ $\beta = 0$, linearized scheme; \bullet $\beta = 1$, linearized scheme; $+$ $\beta = 0$, non-linear scheme; \times $\beta = 1$, non-linear scheme.

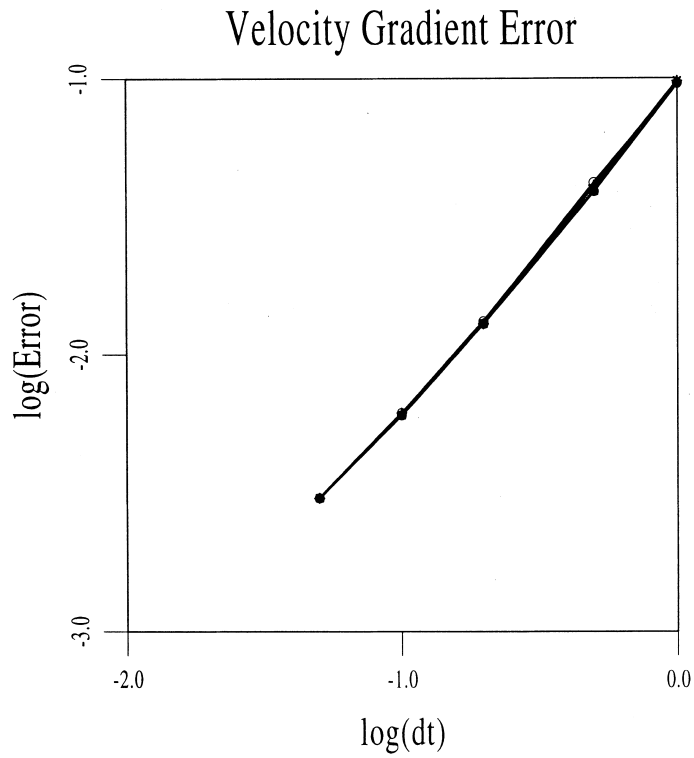


Fig. 5. Taylor flow, $\theta = 1$, Q_2 element, velocity gradient errors: \circ $\beta = 0$, linearized scheme; \bullet $\beta = 1$, linearized scheme; $+$ $\beta = 0$, non-linear scheme; \times $\beta = 1$, non-linear scheme.

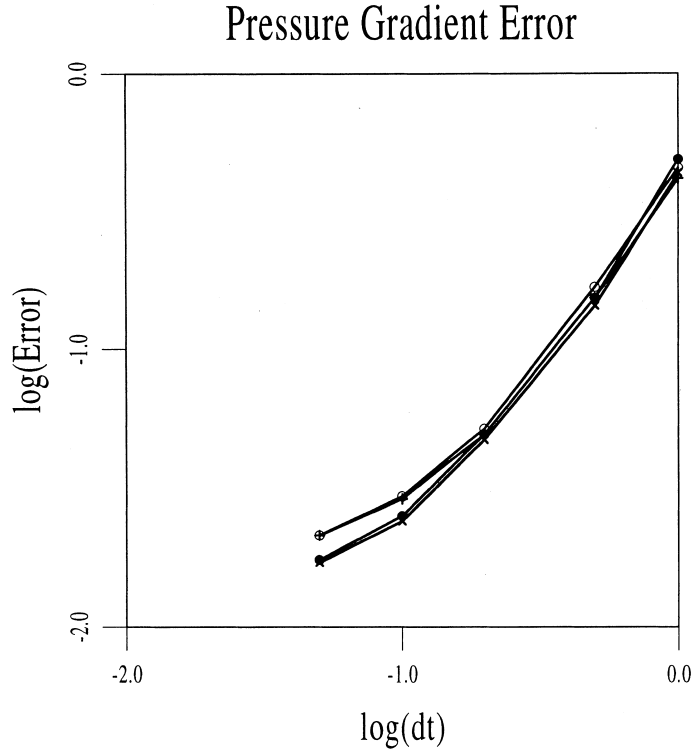


Fig. 6. Taylor flow, $\theta = 1$, Q_1 element, pressure gradient errors: \circ $\beta = 0$, linearized scheme; \bullet $\beta = 1$, linearized scheme; $+$ $\beta = 0$, non-linear scheme; \times $\beta = 1$, non-linear scheme.

different errors versus the time step size in logarithmic scale. However, first order accuracy is lost in the low δt range for the velocity gradient and the pressure gradient, since the space error is reached then, the velocity gradient is shown in [24] to have $O(h)$ spatial accuracy for the Q_1 element case while the pressure gradient is only bounded. For the mesh considered, we have $h \simeq 1/30$. The velocity spatial error, which is $O(h^2)$, is not reached.

We then repeated the calculation with a Q_2 element interpolation; we plot the results in Figs. 4–6. First order time accuracy is also attained in this case. Moreover, the velocity gradient and pressure gradient errors confirmed their respective $O(h^2)$ and $O(h)$ theoretical estimates (see [24]).

The second order Crank–Nicholson case $\theta = 1/2$ proved very sensitive to the values of the stabilization parameters α_K . We finally took them as:

$$\alpha_K = \left(\frac{4v}{\alpha_0 h_K^2} + \frac{2v_K}{h_K} \right)^{-1/2}, \tag{66}$$

where h_K is the size of element K , v_K is a representative velocity of that element at the current time step and α_0 is a given constant (which we took equal to $1/3$ for bilinear elements). These are the optimal values of the stabilization parameters of the GLS method. For large values of the viscosity and small element sizes, (66) behaves as condition (16) dictates. In general cases, we have found good results using (66).

Although not as sharp as in the previous case, it displayed second order accuracy for the velocity solution both in the L^2 -norm and in the H^1 -norm, and both for the Q_1 and Q_2 finite element spatial interpolations (in the first case, the spatial error also degraded the accuracy for low δt values). The errors found in this case can be seen in Figs. 7 and 8, for the Q_1 element case, and Figs. 9 and 10 for the Q_2 element.

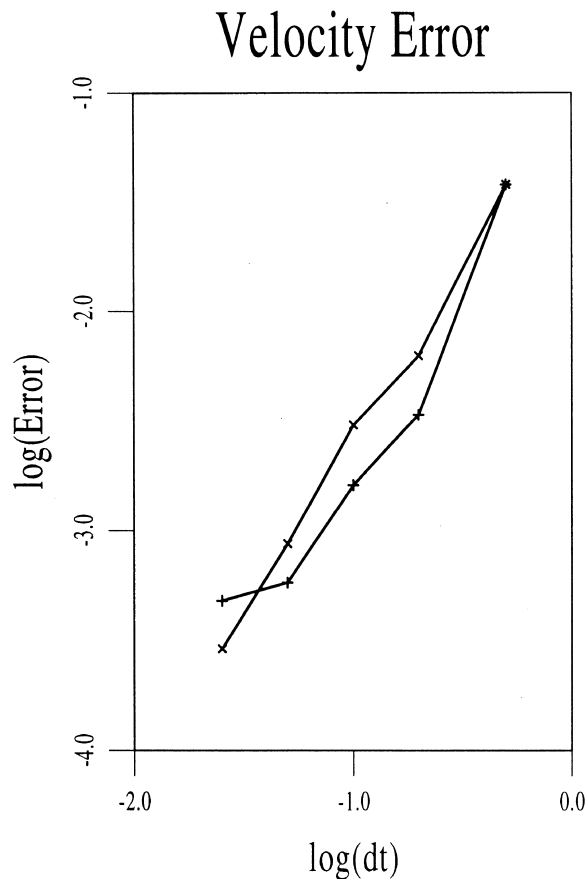


Fig. 7. Taylor flow, $\theta = 1/2$, Q_1 element, velocity errors: + $\beta = 0$; $\times \beta = 1$.

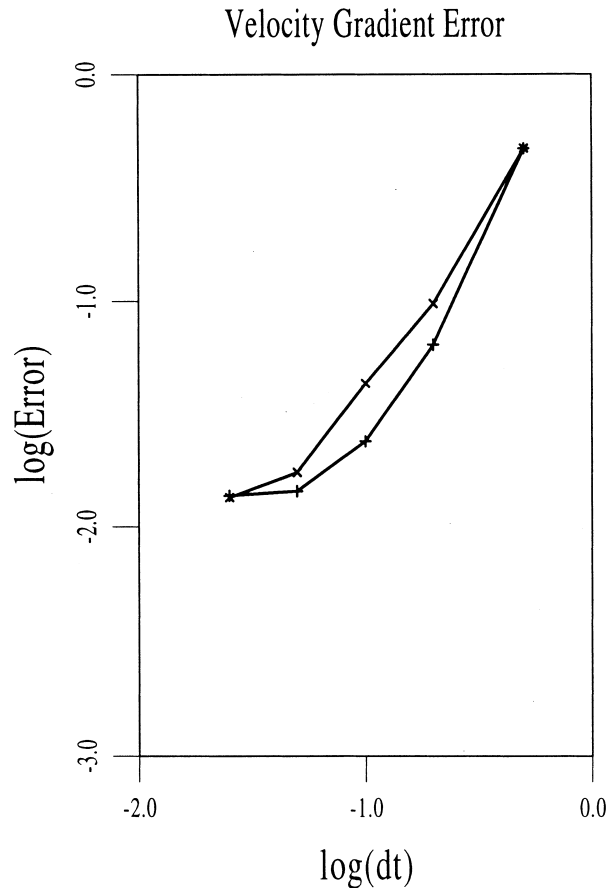


Fig. 8. Taylor flow, $\theta = 1/2$, Q_1 element, velocity gradient errors: + $\beta = 0$; x $\beta = 1$.

Finally, we plot the streamlines and pressure contours obtained at the end of one of the runs in Figs. 11 and 12, respectively.

6.2. Flow past a circular cylinder

We then considered the problem of the flow past a circular cylinder, which has become a compulsory benchmark test for transient algorithms for the Navier–Stokes equations.

It is well known that for low values of the Reynolds number, the solution is steady and symmetric about a line parallel to the free-stream flow through a cylinder diameter; a pair of symmetrical *eddies* develop downstream of the cylinder. But beyond a critical value of Re (which is larger than 40), the steady solution becomes unstable and a periodic solution develops, so that vortex shedding sets in.

The shedding period of the solution is one of the most studied quantities to compare quantitatively the results obtained in this problem. Equivalently, one can look at the Strouhal number or adimensional frequency, defined as $St = D/u_0\tau$, D being the cylinder diameter, u_0 the inflow velocity and τ the shedding period of the solution. It is now known that the shedding characteristics (and the Strouhal number, in particular) are significantly affected by the dissipation introduced by the time-stepping scheme, as well as by the spatial interpolation used (see [25], for instance). Moreover, the computed unsteady flow is also affected by the introduction of finite boundaries in the domain. While the location of the upstream (or inflow) and the downstream (or outflow) boundaries does not seem to affect the solution much (as long as the latter is further than 6 cylinder diameters, see [26]), the shedding solution is far more sensitive to the location of the lateral boundaries, as was recently found in [27]. In particular, the shedding period can become too small

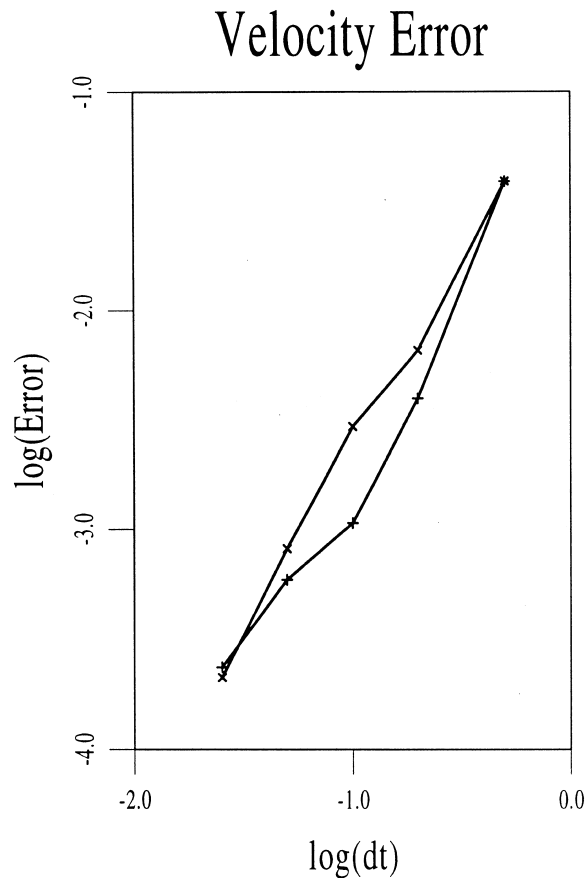


Fig. 9. Taylor flow, $\theta = 1/2$, Q_2 element, velocity errors: + $\beta = 0$; x $\beta = 1$.

when the lateral boundaries are placed too close to the cylinder. We placed these boundaries 12 cylinder diameters away from the cylinder, which according to [27] should be enough not to affect the solution.

Thus, we considered a cylinder of unit diameter and took a computational domain consisting of the rectangle $[0, 50] \times [0, 24]$, the center of the cylinder being situated at the point (12., 12.). The boundary conditions taken are the ones usually imposed in this problem, a unit free-stream horizontal velocity at the inflow, a solid wall condition on the cylinder surface, zero normal velocity and zero tangential stress on the lateral boundaries and a traction-free condition at the outflow. The mesh used in this case can be seen in Fig. 13; it consists of 3000 nodes and 2880 of the Q_1 elements.

We used the explicit pressure gradient, non-linear convection scheme (that is, (21)–(23) with $\beta = 0$), with local values of the coefficients α given again by (66). At each time step, the non-linearity of the problem was solved by a Picard iteration with an error tolerance of 10^{-3} (for which 2 or 3 iterations were enough); the velocity–pressure systems of linear equations were solved by the iterative GMRES method, with an error tolerance of 10^{-5} , while the pressure gradient system was solved by the conjugate gradient algorithm with a consistent mass matrix and the same error tolerance.

We first solved the problem for a Reynolds number of 40, which is based upon the free-stream velocity and the cylinder diameter, starting from the fluid at rest but for the prescribed boundary conditions. We performed 10 time steps of size $\delta t = 0.5$ of the fully implicit scheme $\theta = 1$ (so as to reach the steady state fastest), which yielded the steady solution very accurately.

We then raised the value of the Reynolds number to 100, we started the computation from the steady solution obtained for $Re = 40$ and performed 10,000 steps of size $\delta t = 0.1$ of the Crank–Nicholson method $\theta = 1/2$. The solution started oscillating freely from the start, the final periodicity was reached by $t = 750$. In this case, no artificial trick was needed to start up the periodic solution. In Fig. 14 we show the history of

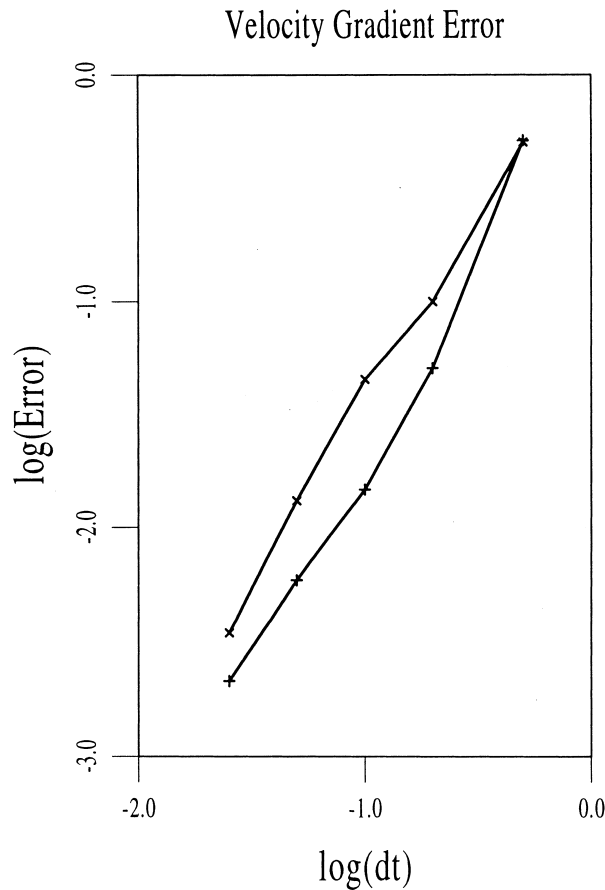


Fig. 10. Taylor flow, $\theta = 1/2$, Q_2 element, velocity gradient errors: + $\beta = 0$; $\times \beta = 1$.

the horizontal velocity at a node situated at the point (24.0, 14.), that is, downstream of the cylinder and slightly higher. In order to improve the accuracy of the solution, we then continued the calculation with a time step size $\delta t = 0.05$, with which we performed 500 steps. We finally found a Strouhal number of $St = 0.161$, or equivalently, a period of 6.2. To this order of precision, this is the same period that was found in [27] with a stabilized, space-time velocity–pressure formulation using a 7641 noded mesh (that is, more than two and a half times the number of nodes we used) in the present domain, a formulation which requires of many more stabilization terms than ours. (A period of 6.0 was found in this reference with a stabilized velocity–pressure–stress formulation also using a Crank–Nicholson time integration scheme, with the same 7641 noded mesh and 3 extra coupled unknowns in 2 dimensions). It has to be said that most researchers who have solved this problem numerically up to now have considered a smaller domain (that is, one with lateral boundaries closer to the cylinder surface), this is probably the reason why smaller periods (between 5.8 and 6.0) have usually been found.

The streamlines obtained at the end of the computation are shown in Fig. 15. In Fig. 16 we plot the stationary streamlines; the wakes behind the cylinder can be clearly seen there. Finally, we show the pressure contours in Fig. 17. All these results compare very well with other published solutions (see [28,29], for instance).

7. Conclusions

In this paper we have presented the extension of a stabilization technique originally developed for steady problems to the transient case. The goal is to be able to use equal velocity–pressure interpolations,

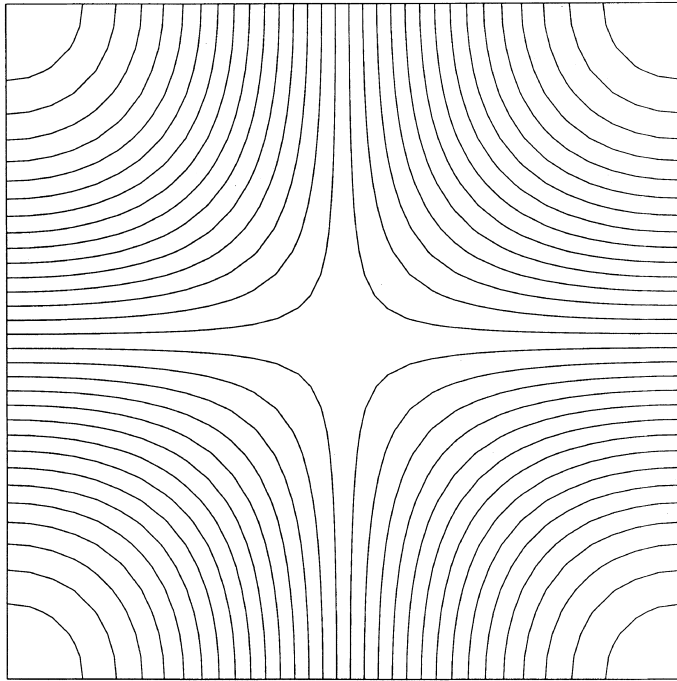


Fig. 11. Taylor flow, streamlines.

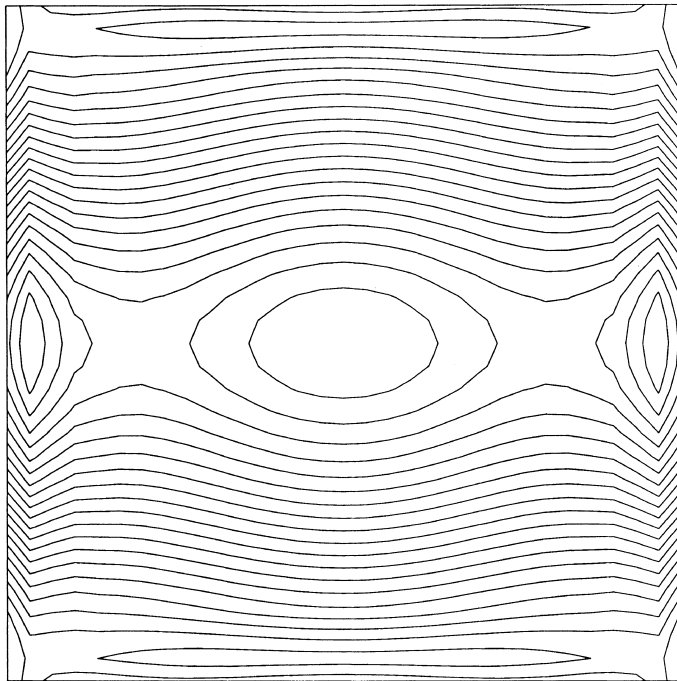


Fig. 12. Taylor flow, pressure contours.

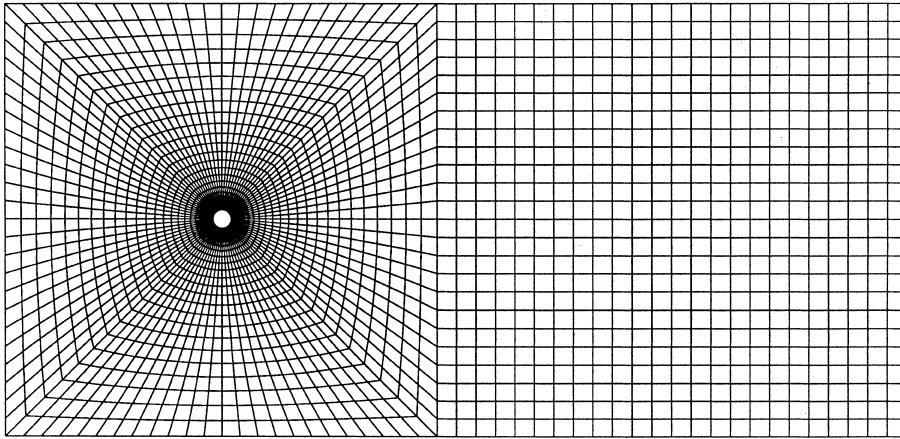


Fig. 13. Flow past a cylinder: mesh.

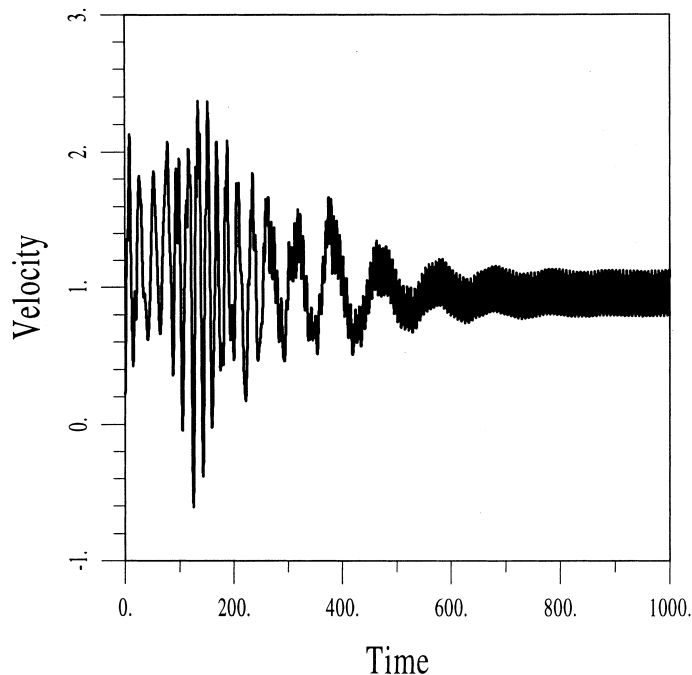


Fig. 14. Flow past a cylinder: nodal velocity history.

and thus other instabilities, such as that arising in convection dominated flows, have to be stabilized by other means.

Theoretical stability estimates and numerical examples demonstrate that the goal described is in fact achieved. The former can be obtained only with some weak assumptions on the finite element meshes employed and the velocity–pressure spaces. Numerical results show that the method is not only stable, but also that the formal accuracy of the time integration scheme is preserved by the use of the stabilization method.

From the computational point of view, the explicit treatment of the pressure gradient projection allows to adapt codes based on mixed div-stable formulations to equal order interpolations only with minor modifications. The continuity equation has to be modified by introducing a Laplacian-like term for the

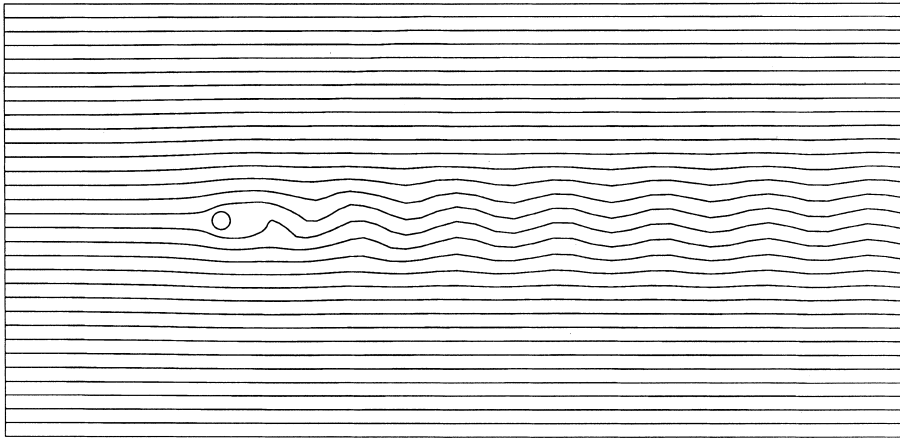


Fig. 15. Flow past a cylinder: streamlines.

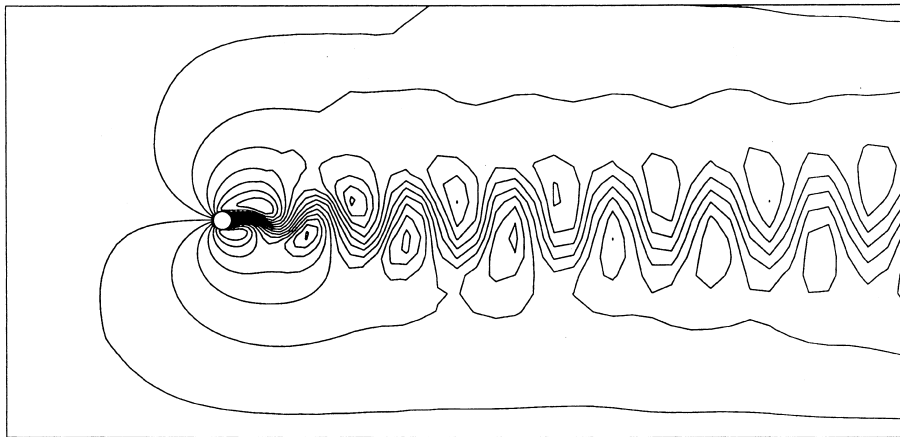


Fig. 16. Flow past a cylinder: stationary streamlines.



Fig. 17. Flow past a cylinder: pressure contours.

pressure and the divergence of the pressure gradient projection in the RHS. We have shown how to apply the stabilization technique to both monolithic and fractional step time integration methods.

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