Robust Control of Uncertain Nonlinear Systems: A Nonlinear DOBC Approach

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ABSTRACT

This paper advocates disturbance observer based control (DOBC) for uncertain nonlinear systems. Within this framework, a nonlinear controller is designed based on the nominal model in the absence of disturbance and uncertainty where the main design specifications are to stabilize the system and achieve good tracking performance. Then a nonlinear disturbance observer is designed to not only estimate external disturbance but also system uncertainty/unmodelled dynamics. With described uncertainty, rigorous stability analysis of the closed-loop system under the composite controller is established in this paper. Finally, the robust control problems of a missile roll stabilization and a mass spring system are addressed to illustrative the distinct features of the nonlinear DOBC approach.

1 Introduction

Robust control of uncertain nonlinear systems is deemed as one of the most crucial topics in modern control theory [1–3]. High-gain feedback domination technique is a major design tool utilized for robust suppression of uncertainties in nonlinear control theory [3]. However, it is generally claimed that most of existing robust control approaches are criticized as worst case based design, that is, the objective of the controller design is to achieve best performance in the presence of the

worst uncertainties [4, 5]. Such a design philosophy involving in most of the existing robust control design methodologies would result in the following two major obstacles for its practical applications [6]. On the one hand, excessive control energy is usually required for the robust control to suppress the worst case uncertainties [5]. This would cause the waste of control energy as well as unreasonable selection of the actuator since powerful actuators are required to perform unnecessarily excessive control action demanded by the robust controllers. To this end, the conventional robust controller in practical engineering usually leads to input saturation problems [7–9]. On the other hand, the robustness performance of those control performance in the nominal case is usually not considered in the objective function of robust control [4, 6]. However, most practical systems are generally working around its nominal operation point and have a limited chance to operate far away from their nominal operation point. Taking a flight control system as an example, the flight envelop of the vehicle is generally around its nominal flight conditions. The extremely severe flight environment will but rarely appear in the whole flight process. The worst case based robust control design results in over conservative design. As such, a promising and practical robust controller should consider not only the robustness against the worst case of uncertainties but also the control performance around the nominal operation points.

Disturbance observer based control (DOBC) provides an alternative approach for robust control of nonlinear systems [10–13]. In the framework of nonlinear DOBC, the total uncertainties including external disturbances and internal model uncertainties of nonlinear systems are estimated online and then compensated within the closed-loop systems. As a practical robust control approach, nonlinear DOBC has been utilized for uncertainty compensation in engineering practices, for example, see altitude control of missile [14], speed regulation of motor drives [15, 16], tracking control of helicopter [18], DC-bus voltage control of microgrid [19], and control of haptic feedback system [17]. A remarkable superiority of DOBC against most of the existing robust control lies in that it is not a worst case based design approach. In the context of DOBC, a disturbance observer acting as a patch on the baseline controller is employed to estimate the uncertainties. In the absence of uncertainties, the disturbance observer will not be activated and a baseline feedback controller maintains the nominal control performances. In the presence of uncertainties, the disturbance observer is activated for uncertainty estimation and compensation, which achieves the robustness against uncertainties without involving excessive control energy. In general, the DOBC framework facilitates a good trade off between the nominal performance and robustness.

Stability of nonlinear DOBC for nonlinear systems subject to various disturbances (e.g. unknown constant [10], generated by an exogenous system [12] and with bounded variation rate) has been well established. However, it is not applicable when DOBC is used as a robust control method to deal with uncertainties. In this case, different from disturbance, the influence of the uncertainties is actually a function of system states. As such, in spite of promising properties as demonstrated in a number of applications and growing interests in DOBC, the rigorous qualitative robustness analysis on theoretical side lags far behind. It is not reasonable to ignore the coupling between the system dynamics and the observer dynamics when investigating its stability. Recently, in the field of active disturbance rejection control known as another kind of disturbance estimator based control, the robustness stability of the closed-loop systems under uncertainties are established but in the context of linear nominal systems [5] and lower-triangular nonlinear systems [20].

The major contribution of this paper is to develop a general framework for establishing rigorous robust stability for

uncertain nonlinear systems under the nonlinear DOBC approach. The nominal dynamics of the uncertain nonlinear systems under consideration are nonlinear, which is different from [5, 20]. The Lyapunov stability theory is employed for stability derivation and the results show that the exponentially stability of the uncertain nonlinear system under nonlinear DOBC is guaranteed under certain condition. Simulation studies on a missile roll stabilization and a mass spring system are conducted to show the effectiveness of the nonlinear DOBC approach. The relationship between the disturbance observer gains and quantity of uncertainties are further explored via simulation studies.

2 Problem Formulation

In the presence of uncertainty and disturbance, a class of uncertain nonlinear systems may be represented as

$$\dot{x} = f(x) + \Delta f(x) + [g(x) + \Delta g(x)]u + g_1(x)w \tag{1}$$

where $x \in R^n$, f(x), g(x) are smooth nonlinear functions, $w \in R^m$ is the external disturbance, $\triangle f(x)$ and $\triangle g(x)$ are the system uncertainties, and $u \in R^m$ is the control input.

It is assumed that the influence of the uncertainty and disturbance can be equivalent to the input channel of the nonlinear system

$$\dot{x} = f(x) + g(x)[\triangle(x, u, w) + u] \tag{2}$$

where $g(x) \triangle (x, u, w) = \triangle f(x) + \triangle g(x)u + g_1(x)w$. That is, the disturbance and uncertainties satisfy the so called matching condition.

In the context of DOBC approach, the influence of the uncertainty and the disturbance are lumped together and a nonlinear disturbance observer is designed to estimate them. Letting $d = \triangle(x, u, w)$, the system (2) becomes

$$\dot{x} = f(x) + g(x)(d+u) \tag{3}$$

In the presence of the matched disturbance, the control *u* in the DOBC is proposed in the following form

$$u = \alpha(x) - \hat{d} \tag{4}$$

where $\alpha(x)$ is designed by any nonlinear control method based on the nominal plant, i.e. f(x). Normally, it is designed to achieve stability and tracking/regulation specifications. The estimate \hat{d} of the uncertainties is utilized to compensate the influence of lumped uncertainties d. In this paper, the nonlinear disturbance observer proposed in [10, 12] is adopted to estimate the influence of uncertainties

$$\begin{cases} \hat{d} = \xi + p(x), \\ \dot{\xi} = -\gamma l(x)g(x)\xi - \gamma l(x)[f(x) + g(x)u + g(x)p(x)], \end{cases}$$
(5)

where p(x) and l(x) satisfy the relationship

$$\frac{\partial p(x)}{\partial x} = \gamma l(x) \tag{6}$$

and γ is a positive scalar to be designed. The block diagram of the nonlinear DOBC (4) for uncertain nonlinear system (1) is shown by Fig. 1.



Fig. 1. Block diagram of nonlinear DOBC.

As discussed above, in most cases, DOBC is regarded as a robust control method due to its promising robustness. As an alternative methodology to robust control, DOBC can alleviate some shortcomings (most notably, worst case based design) in robust control. This can be explained by using the basic diagram of DOBC as in Fig. 1 under the assumption of no external disturbance. The inner disturbance observer loop acts as an adaption mechanism in the presence of uncertainty. Rather than directly estimating the uncertain parameters as in many adaptive control algorithms [21], as shown in Eq. (2), it estimates the total difference between the nominal model and the physical system, which may include both structured or unstructured uncertainty. So it is a "crude" adaptive control mechanism. Although it might not achieve the same control performance as other adaptive control algorithms when the uncertainty is structured (e.g. represented by linear functions of unknown parameters), it is more *robust* than most of adaptive control algorithms. On the other side, as the inner loop is not activated when there is no difference between the nominal model and the physical system, the nominal tracking and regulation performance is maintained. In the presence of uncertainty, the inner disturbance observer loop is activated to estimate and attenuate the influence of the uncertainty. Compared with robust control methods based on the worst case design, the robustness of DOBC may be not as good as them as these methods are designed to achieve best possible performance in the presence of the worst uncertainty. So in the sense, DOBC can be regarded as a "refined" robust control method. However, it provides a promise approach for trading off between the nominal performance and robustness. In summary, DOBC provides an alternative approach to widely used robust control and adaptive control methods for dealing with uncertain (linear or nonlinear) systems.

Stability and performance of the DOBC under various external disturbances have been investigated in [10, 12]. However, so far there is no rigorous stability and performance analysis when the DOBC is applied to deal with uncertainties. This paper focuses on the stability analysis of DOBC under uncertainties. To this end, in the following of this paper, only the influence of the uncertainties is considered, i.e. $d = \Delta(x)$. The key challenge in investigating the stability of DOBC is, different from

the external disturbance which is independent of the system, the influence of the uncertainty depends on the state of the nonlinear system.

3 Main Results

Define the estimation error

$$e_d = d - \hat{d} \tag{7}$$

Substituting the nonlinear controller (4) and the nonlinear disturbance observer (5) into the uncertain system (3) gives the closed-loop system

$$\begin{cases} \dot{x} = f(x) + g(x)\alpha(x) + g(x)e_d, \\ \dot{e}_d = -\gamma l(x)g(x)e_d + \dot{d} \end{cases}$$
(8)

Let

$$\dot{x} = f_c(x) = f(x) + g(x)\alpha(x) \tag{9}$$

be the nominal closed-loop dynamics. Based on the definition of d, the dynamics of d can be represented by

$$\dot{d} = \frac{\partial \triangle}{\partial x} \dot{x}$$

$$= \frac{\partial \triangle}{\partial x} (f_c(x) + g(x)e_d) = L_{f_c} \triangle (x) + L_g \triangle (x)e_d,$$
(10)

Inserting Eq. (10) into Eq. (8) obtains

$$\begin{cases} \dot{x} = f_c(x) + g(x)e_d, \\ \dot{e}_d = \left[-\gamma l(x)g(x) + L_g \bigtriangleup(x)\right]e_d + L_{f_c}\bigtriangleup(x), \end{cases}$$
(11)

We will provide the stability proof for the proposed approach. To this end, Lemma 1 will be firstly introduced.

Lemma 1 [22] (see p. 368) Consider the nonlinear system $\dot{x} = \psi(x, \alpha)$, where $\psi(x, \alpha)$ is continuously differentiable and the Jocobian matrices satisfying $||\frac{\partial \psi(x,\alpha)}{\partial x}|| \le L_1$ and $||\frac{\partial \psi(x,\alpha)}{\partial \alpha}|| \le L_2||x||$ for all $(x, \alpha) \in D \times \Gamma, D \stackrel{\triangle}{=} \{x \in R^n \mid ||x|| < r\}$. Let k, λ, r_0 be positive constant with $r_0 < r/k$ and define $D_0 = \{x \in R^n \mid ||x|| < r_0\}$. Assume that the trajectory of the system satisfies $||x(t)|| \le k||x(0)||e^{-\lambda t}$, $\forall x(0) \in D_0, \alpha \in \Gamma, t \ge 0$. Then there is a function $V : D_0 \times \Gamma \to R$ such that the following conditions hold

$$\begin{split} b_1 ||x||^2 &\leq V(x, \alpha) \leq b_2 ||x||^2, \\ \frac{\partial V}{\partial x} \psi(x, \alpha) &\leq -b_3 ||x||^2, \\ |\frac{\partial V}{\partial x} || &\leq b_4 ||x||, ||\frac{\partial V}{\partial \alpha} || \leq b_5 ||x||^2. \end{split}$$

where b_i ($i = 1, \dots, 5$) is positive constant. Moreover, if all the assumptions hold globally (in *x*), then $V(x, \alpha)$ is defined and satisfies the above conditions on $\mathbb{R}^n \times \Gamma$.

In this paper, the following assumptions are imposed to the nonlinear system (1) and the nominal controller in (4):

- (A1) $\triangle(0) = 0.$
- (A2) $f, g, \alpha, \triangle, \frac{\partial \triangle}{\partial x}$ are all continuously differentiable.
- (A3) $f_c(0) = 0, \dot{x} = f_c(x)$ is exponentially stable.

Assumption A1 is for the purpose of convenience and not necessary. As a matter of fact, if $\triangle(0) \neq 0$, the uncertainties $\triangle(x)$ can be separated into two parts as $\triangle(x) = \overline{\triangle}(x) + \triangle(0)$ with $\overline{\triangle}(x) = \triangle(x) - \triangle(0)$ and $\overline{\triangle}(0) = 0$. In this case, only the term $\overline{\triangle}(x)$ has to be considered and the term $\triangle(0)$ can be simply merged into the external disturbance term. Assumption A2 is a necessary condition for robustness analysis of DOBC when tackling uncertainties, which has also been employed in related disturbance estimation based control for uncertainties compensation, for example see [5, 20]. Assumption A3 is a basic condition that shall be fulfilled by all successful nonlinear control design.

Theorem 1. Suppose that Assumptions A1-A3 are satisfied. The closed-loop system (11) under the proposed controller (4) and (5) is exponentially stable if the observer gain l(x) is selected such that $\dot{e}_d = -\gamma l(x)g(x)e_d$ is exponentially stable for all $x \in B_r \stackrel{\triangle}{=} \{x \mid ||x|| < r\}$, where $\gamma > 0$ is parameter that can be chosen sufficiently large.

Proof: Define a Lyapunov function candidate

$$V_0(e_d) = e_d^2/2$$

for the observer error dynamics in (11). Taking the derivative of $V_0(e_d)$ along the dynamics (11) gives

$$\dot{V}_0(e_d) = \left[-\gamma l(x)g(x) + L_g \bigtriangleup (x)\right]e_d^2 + L_{f_c} \bigtriangleup (x)e_d,$$

$$\leq -\varepsilon e_d^2 + |e_d| \cdot |L_{f_c} \bigtriangleup (x)|,$$
(12)

where $\varepsilon > 0$ and γ are chosen such that

$$\gamma \ge \left(\max_{x \in B_r} |L_g \bigtriangleup(x)| + \varepsilon\right) / \min_{x \in B_r} l(x)g(x).$$
(13)

It further follows from (12) that

$$\dot{V}_0(e_d) \le -\varepsilon(1-\theta)e_d^2, \,\forall \, |e_d| \ge \frac{|L_{f_c} \triangle(x)|}{\varepsilon \theta},\tag{14}$$

where $0 < \theta < 1$, which shows that the disturbance estimation error dynamics are input-to-state stable by taking $L_{f_c} \triangle(x)$ as an input and e_d as a state. As such, the response of the disturbance estimation error at an initial state $e_d(0) = e_d^0$ satisfies $e_d(t) \in B_{\rho}$ and

$$\rho = \max\{\lambda_1 | e_d^0 |, \lambda_2 \max_{x \in B_r} | L_{f_c} \bigtriangleup(x) | \}$$

where λ_1 and λ_2 are positive constants.

For the reduced-order nominal system $\dot{x} = f_c(x)$, it follows from Assumption A3 that there exists a Lyapunov function V(x) such that [22]

$$c_{1}||x||^{2} \leq V(x) \leq c_{2}||x||^{2},$$

$$\frac{\partial V}{\partial x}f_{c}(x) \leq -c_{3}||x||^{2},$$

$$||\frac{\partial V}{\partial x}|| \leq c_{4}||x||.$$
(15)

for all $x \in B_{r_0}$, $r_0 \leq r$.

The exponential stability of the nonlinear disturbance observer regardless of *x* (that is, $\dot{e}_d = -\gamma l(x)g(x)e_d$ is exponentially stable) implies that there exists a k > 0 and $\lambda > 0$ such that the trajectory of the observer error dynamic system satisfies $||e_d(t)|| \le k||e_d(0)||e^{-\lambda t}, \forall e_d(0) \in D_0, x \in \Gamma, t \ge 0$ where $D_0 = \{e_d \in R^n | ||e_d|| < \rho_0\}$ with $\rho_0 < \rho/k$. It follows from Lemma 1 that there exists a function $W(e_d, x) : D_0 \times \Gamma \to R$ such that

$$a_{1}||e_{d}||^{2} \leq W(e_{d}, x) \leq a_{2}||e_{d}||^{2},$$

$$\frac{\partial W}{\partial e_{d}}(-l(x)g(x)e_{d}) \leq -a_{3}||e_{d}||^{2},$$

$$||\frac{\partial W}{\partial e_{d}}|| \leq a_{4}||e_{d}||, ||\frac{\partial W}{\partial x}|| \leq a_{5}||e_{d}||^{2}.$$
(16)

for all $e_d \in D_0$.

Define a Lyapunov function candidate $\Phi(x, e_d)$ for closed-loop system (11) as

$$\Phi(x, e_d) = V(x) + W(e_d, x).$$

Taking the derivative of $\Phi(x, e_d)$ along the trajectory of the closed-loop system (11), one can obtain

$$\begin{split} \dot{\Phi} &= \frac{\partial V}{\partial x} \dot{x} + \frac{\partial W}{\partial x} \dot{x} + \frac{\partial W}{\partial e_d} \dot{e}_d \\ &= \frac{\partial V}{\partial x} (f_c(x) + g(x)e_d) + \frac{\partial W}{\partial x} (f_c(x) + g(x)e_d) \\ &+ \frac{\partial W}{\partial e_d} ([-\gamma l(x)g(x) + L_g \bigtriangleup (x)]e_d + L_{f_c} \bigtriangleup (x)) \\ &\leq -c_3 ||x||^2 + (\frac{\partial V}{\partial x} + \frac{\partial W}{\partial x})g(x)e_d + \frac{\partial W}{\partial x}f_c(x) - \gamma a_3 ||e_d||^2 \\ &+ \frac{\partial W}{\partial e_d} L_g \bigtriangleup (x)e_d + \frac{\partial W}{\partial e_d} L_{f_c} \bigtriangleup (x). \end{split}$$
(17)

In addition, based on Eqs. (15) and (16), we can obtain the following inequalities.

$$(\frac{\partial V}{\partial x} + \frac{\partial W}{\partial x})g(x)e_d \leq (||\frac{\partial V}{\partial x}|| + ||\frac{\partial W}{\partial x}||)||g(x)|| \cdot ||e_d||$$

$$\leq (c_4||x|| + a_5||e_d||^2)||g(x)|| \cdot ||e_d||$$

$$\leq c_4||x||G||e_d|| + a_5||e_d||^3G$$

$$(18)$$

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where $G \stackrel{\triangle}{=} \max_{x \in B_r} ||g(x)||$. The third term of the right hand side of (17) is estimated as

$$\frac{\partial W}{\partial x} f_c(x) \le ||\frac{\partial W}{\partial x}|| \cdot ||f_c(x)||$$

$$\le a_5 ||e_d||^2 \cdot L_3 ||x||$$

$$= d_1 ||e_d||^2 ||x||$$
(19)

where $L_3 = \max_{x \in B_r} ||\frac{\partial f_c(x)}{\partial x}||$ and $d_1 = a_5 L_3$. Based on Eq. (16), it is obtained that

$$\frac{\partial W}{\partial e_d} L_g \bigtriangleup (x) e_d \le || \frac{\partial W}{\partial e_d} || \cdot || L_g \bigtriangleup (x) || \cdot || e_d ||$$

$$\le a_4 || e_d || \cdot L_1 \cdot || e_d ||$$

$$\le a_6 || e_d ||^2$$
(20)

where $L_1 = \max_{x \in B_r} ||L_g \triangle(x)||$ and $a_6 = a_4 L_1$. Based on similarly reason, we can also obtain

$$\frac{\partial W}{\partial e_d} L_{f_c} \Delta(x) \leq ||\frac{\partial W}{\partial e_d}|| \cdot ||\frac{\partial \Delta(x)}{\partial x}|| \cdot ||f_c(x)|| \\
\leq a_4 ||e_d|| \cdot L_2 \cdot L_3 ||x|| \\
\leq a_7 ||e_d|| \cdot ||x||$$
(21)

where $L_2 = \max_{x \in B_r} ||\frac{\partial \triangle(x)}{\partial x}||$ and $a_7 = a_4 L_1 L_3$. Inserting Eqs. (18)-(21) into (17) obtains

$$\dot{\Phi}(x,e_d) \le -c_3 ||x||^2 + c_4 G ||x|| \cdot ||e_d|| + a_5 G ||e_d||^3 + d_1 ||e_d||^2 \cdot ||x|| - \gamma a_3 ||e_d||^2 + a_6 ||e_d||^2 + a_7 ||e_d|| \cdot ||x||$$
(22)

For all $e_d \in B_{\rho_0}$, $d_1 ||e_d||^2 ||x|| \le \tilde{d}_1 ||e_d|| ||x||$, $a_5 G ||e_d||^3 \le \tilde{a}_5 G ||e_d||^2$, where $\tilde{d}_1 = d_1 \rho_0$ and $\tilde{a}_5 = a_5 \rho_0$. Then Eq. (22) satisfies

$$\begin{split} \dot{\Phi}(x,e_d) &\leq -c_3 ||x||^2 + (c_4 G + a_7 + \tilde{d}_1) ||x|| \cdot ||e_d|| \\ &+ (-\gamma a_3 + \tilde{a}_5 G + a_6) ||e_d||^2 \\ &\leq -c_3 ||x||^2 + \frac{c_3}{2} ||x||^2 + c_8 ||e_d||^2 + (-\gamma a_3 + \tilde{a}_5 G + a_6) ||e_d||^2 \\ &= -\frac{c_3}{2} ||x||^2 - (\gamma a_3 - \tilde{a}_5 G - a_6 - c_8) ||e_d||^2 \end{split}$$

$$(23)$$

where $c_8 = (c_4G + a_7 + \tilde{d}_1)^2/(2c_3)$. This implies that the exponential stability of the closed-loop system under the proposed controller is guaranteed by choosing a sufficiently large γ such that

$$\gamma > (\tilde{a}_5 G + a_6 + c_8)/a_3.$$

A general design procedure of nonlinear DOBC for robust control of nonlinear system is summarized as follows:

Step 1) Design a baseline nonlinear controller $u_b = \alpha(x)$ for the nominal system (1) in the absence of disturbances and uncertainties to obtain satisfied stability and other performance specifications.

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- Step 2) Design a nonlinear disturbance observer to estimate the disturbances and uncertainties online. Select the observer gain l(x) such that the observer error dynamics is exponentially stable. Tune the observer scalar γ to achieve satisfactory estimation accuracy.
- Step 3) Integrate the baseline nonlinear control with the disturbance compensation raised by nonlinear disturbance observer to formulate the composite control law.

4 Simulation Studies

4.1 Roll Stabilization for Precision Guided Munitions Class Missiles

The roll stabilization for PGM (precision guided munitions) class missiles is investigated to illustrate the proposed method in this paper. The dynamics of a PGM clsass missile in roll plane, ignoring the flexible mode, is described as follows [23]

$$\ddot{\phi} = C_{\alpha} \sin 4\phi - w_{RR} \dot{\phi} + K_{\delta} \delta_a + \delta_f C_{\alpha} \sin 4\phi$$
(24)

where ϕ and $\dot{\phi}$ is roll angle and roll rate of PGM missiles, δ_a is fin deflection; C_{α} , w_{RR} , and K_{δ} is the nominal disturbance coefficient, roll rate bandwith and fin effectiveness; δ_f is parameter perturbation coefficient of nominal disturbance. The nominal parameters are chosen as $C_{\alpha} = 70$, $w_{RR} = 5$ and $K_{\delta} = 1450$. Letting $x_1 = \phi$, $x_2 = \dot{\phi}$ and $u = \delta_a$, the dynamic model (24) can be written in state-space form as

$$\dot{x}_{1} = x_{2}$$

$$\dot{x}_{2} = C_{\alpha} \sin 4x_{1} - w_{RR}x_{2} + K_{\delta}u + \delta_{f}C_{\alpha} \sin 4x_{1}$$
(25)

The nonlinear disturbance observer for system (25) is designed as:

$$\hat{d} = \xi + \gamma x_2$$

$$\dot{\xi} = -\gamma \xi - \gamma (C_{\alpha} \sin 4x_1 - w_{RR} x_2 + K_{\delta} u + \gamma x_2)$$
(26)

The baseline nonlinear controller for system (25) is designed as

$$\alpha(x) = -\frac{1}{K_{\delta}} (C_{\alpha} \sin 4x_1 - w_{RR} x_2 + K_P x_1 + K_D x_2)$$
(27)

Parameters K_P , K_D are the controller gains and designed as $K_P = 225$, and $K_D = 15$. The nonlinear DOBC for system (25) is finally designed as

$$u = \alpha(x) - \frac{1}{K_{\delta}}\hat{d}$$
(28)

The following simulation scenarios as listed in Table 1 are implemented to illustrate the main results of this paper.

Case I	$\gamma = 10$	$\delta_{f1} = 0.5, \ \delta_{f2} = 1, \ \delta_{f3} = 1.8$
Case II	$\delta_f = 0.9$	$\gamma_1 = 1, \ \gamma_2 = 10, \ \gamma_3 = 100$

Table 1. Simulation Scenarios Setting of Roll Stabilization

4.1.1 Robustness of DOBC with Determined Observer Parameters

Firstly, we will investigate the robustness of the DOBC with fixed controller and observer parameters. This simulation scenario setting is based on the consideration that the DOBC with fixed control parameters in general have limited robustness performances, that is, the closed-loop system with fix control parameters possibly becomes unstable in the presence of some extremely severe uncertainties. The robustness of the DOBC with observer scalar $\gamma = 10$ is investigated under different parameter perturbation coefficients: $\delta_{f1} = 0.5$, $\delta_{f2} = 1$, $\delta_{f3} = 1.8$ in this subsection. The response curves of roll angle, roll rate, and fin deflection under DOBC with $\gamma = 10$ in the presence of uncertainties of *Case I* are shown by Figs. 2.

As shown by Fig. 2, the DOBC with given control parameters could effectively compensate the plant uncertainties of $\delta_{f_1} = 0.5$ and $\delta_{f_2} = 1$. However, the closed-loop system becomes unstable once the uncertainty is up to $\delta_{f_3} = 1.8$. As a result, it is concluded from the above analysis that the DOBC with determined parameters have limited robustness performances. In the next subsection, we will investigate how to tune the parameters of DOBC so that the closed-loop system could cover a larger stability region.

4.1.2 Robustness of DOBC with Different Observer Scalars

In this subsection, the simulation studies are conducted for the roll stabilization for PGM class missiles with fixed uncertainties of *Case II*: $\delta_f = 0.9$ under DOBC with different observer scalars. The purpose of those simulation studies is to show how the observer scalar γ can be tuned to achieve better robustness and performances in the presence of uncertainties. The variable responses of the roll stabilization system under DOBC in the presence of uncertainties described by Case II are shown by Figs. 3.

In the presence of uncertainty $\delta_f = 0.9$, it can be observed from Fig. 3 that the closed-loop system is unstable with an observer scalar $\gamma = 1$. By increasing the observer scalar to $\gamma = 10$, it is shown by Fig. 3 that the output of closed-loop system converges to its desired reference asymptotically. As shown by Fig. 3, the variable response curves of the roll stabilization system under DOBC with an observer scalar $\gamma = 100$ is much closer to the nominal control system, indicating a much better transient control performance, which is referred to as transient performance recovery in [2]. In conclusion, the simulation results in this subsection reveal that a larger observer scalar generally indicates a larger stability region, which confirms the theoretic results presented in Section 3.



Fig. 2. Response curves of the roll stabilization system (25) under DOBC (28) with $\gamma = 10$ in the presence of various cases of uncertainties δ_f (Case I): (a) roll angle, (b) roll rate, (c) fin deflection.



Fig. 3. Response curves of the roll stabilization system (25) under DOBC (28) with various observer scalar γ in the presence of uncertainties $\delta_f = 0.9$ (Case II): (a) roll angle, (b) roll rate, (c) fin deflection.

4.2 Displacement Tracking for Nonlinear Spring Mechanical System

A mechanical system with a spring of nonlinear dynamics as shown in Fig. 4 is introduced to illustrate the proposed method in this paper. The dynamical equation of the system is described as follows [24, 25]

$$m\ddot{x} + c\dot{x} + f_1 x + f_3 x^3 = k_t u \tag{29}$$

where x, u, m, c and k_t are the displacement, controller force, mass, damping and the torque constant, respectively. f_1 and f_3 are the parameters to describe the characteristics of the nonlinear spring. The dynamic model (29) can be written in



Fig. 4. Mass spring system

state-space form as

$$\dot{x}_1 = x_2 \dot{x}_2 = -\frac{c}{m}x_2 - \frac{f_1}{m}x_1 - \frac{f_3}{m}x_1^3 + \frac{k_t}{m}u$$
(30)

where x_1 , x_2 and u are the displacement, velocity and controller force, respectively. There always exists some uncertainties in the parameters of the nonlinear spring. Let $\delta_{f_1} = \frac{f_1 - f_{10}}{f_1}$ and $\delta_{f_3} = \frac{f_3 - f_{30}}{f_3}$ represent the uncertain parts of f_1 and f_3 , where f_{10} and f_{30} denote the nominal values of f_1 and f_3 , respectively. The nominal parameters are referred to [24, 25] and listed in Table 2.

Parameter	т	С	f_{10}	f_{30}	<i>k</i> _t
Unit	kg	Ns/m	N/m	N/m^3	1
Value	1	5	100	500 000	1

Table 2. Nominal Parameters of the Mass Spring System

The nonlinear disturbance observer for system (30) is designed as:

$$\hat{d} = \xi + \frac{\gamma m}{k_t} x_1
\dot{\xi} = -\gamma \xi - \gamma (u + \frac{\gamma m}{k_t} x_1) + \frac{\gamma}{k_t} (c x_2 + f_{10} x_1 + f_{20} x_1^3)$$
(31)

The baseline nonlinear controller for system (30) is designed as

$$\alpha(x) = \frac{1}{k_t}(cx_2 + f_1x_{10} + f_{30}x_1^3) + \frac{m}{k_t} \left[-K_P(x_1 - x_{1d}) - K_Dx_2 - K_I \int_0^t (x_1 - x_{1d})dt \right]$$
(32)

where x_{1d} is the reference signal of the displacement x_1 . Parameters K_P , K_I and K_D are the controller gains to be determined, which are designed as $K_P = 48$, $K_I = 64$ and $K_D = 12$, respectively. The desired reference of displacement is set as $x_{1d} = 0.02$. The nonlinear DOBC for system (30) is finally designed as

$$u = \alpha(x) - \frac{m}{k_t}\hat{d} \tag{33}$$

4.2.1 Robustness of DOBC Against Uncertainties

 Table 3. Simulation Scenarios Setting of Robustness Test

Scenarios	Observer Scalar	Uncertainties
Case I	$\gamma = 6$	$\delta_{f_3} = 0, \ \delta_{f_1} = -40\%, \ -84\%, \ 84\%$
Case II	$\gamma = 6$	$\delta_{f_1} = 0, \ \delta_{f_3} = -8\%, \ -13.5\%, \ 13.5\%$

First, we will investigate the robustness of the DOBC against various uncertainties with fixed controller and observer parameters. This simulation scenario setting is based on the consideration that the DOBC with fixed control parameters in general have limited robustness performances, that is, the closed-loop system with fixed control parameters possibly becomes unstable in the presence of some extremely severe uncertainties. The robustness of the DOBC with observer scalar $\gamma = 6$ is investigated under two cases of uncertainties (*Case I*: $\delta_{f_1} = -40\%$, -84%, 84% and *Case II*: $\delta_{f_3} = -8\%$, -13.5%, 13.5%, as shown by Table 2) in this subsection. The response curves of displacement, velocity and control force under DOBC with $\gamma = 6$ in the presence of uncertainties of *Case I* and *Case II* are shown by Figs. 5 and 6, respectively.

As shown by Fig. 2, the DOBC with given control parameters could effectively compensate the plant uncertainties of $\delta_{f_1} = -40\%$ and $\delta_{f_1} = 84\%$. However, the closed-loop system becomes unstable once the uncertainty is up to $\delta_{f_1} = -84\%$. It is observed from Fig. 3 that the DOBC approach with above parameters has achieved appropriate robustness against the plant uncertainties of $\delta_{f_3} = -8\%$ and $\delta_{f_3} = 13.5\%$. When the uncertainty is up to $\delta_{f_3} = -13.5\%$, it can be seen from Fig. 3 that the response curves of the mass spring system under DOBC becomes oscillating. As a result, it is concluded from the above analysis that the DOBC with determined parameters have limited robustness performances. In the next subsection, we will investigate how to tune the parameters of DOBC so that the closed-loop system could cover a larger stability region.



Fig. 5. Response curves of the mass spring system (29) under DOBC (33) with $\gamma = 6$ in the presence of various cases of uncertainties δ_{f_1} (Case I): (a) displacement, (b) velocity, (c) control force.



Fig. 6. Response curves of the mass spring system (29) under DOBC (33) with $\gamma = 6$ in the presence of various cases of uncertainties δ_{f_3} (Case II): (a) displacement, (b) velocity, (c) control force.

4.2.2 Design Parameter Influence on Robustness of DOBC

Scenarios	Uncertainties	Observer Scalar
Case III	$\delta_{f_1} = -50\%, \ \delta_{f_3} = 0$	$\gamma_1 = 1.8, \gamma_2 = 5, \gamma_3 = 20$
Case IV	$\delta_{f_1} = 0, \ \delta_{f_3} = -30\%$	$\gamma_1 = 16.5, \ \gamma_2 = 22, \ \gamma_3 = 50$

Table 3. Simulation Scenarios Setting of Different DOBC parameters

In this subsection, the simulation studies are conducted for the mass spring system with two cases of fixed uncertainties (*Case III*: $\delta_{f_1} = -50\%$ and *Case IV*: $\delta_{f_3} = -30\%$, as shown by Table 3) under DOBC with different observer scalars. The purpose of those simulation studies is to show how the observer scalar γ can be tuned to achieve better robustness and performances in the presence of uncertainties. The variable responses of the mass spring system under DOBC in the presence of uncertainties described by Case III and Case IV are shown by Figs. 7 and 8, respectively.

In the presence of uncertainty $\delta_{f_1} = -50\%$, it can be observed from Fig. 7 that the closed-loop system is unstable with an observer scalar $\gamma = 1.8$. By increasing the observer scalar to $\gamma = 5$, it is shown by Fig. 7 that the output of closed-loop system converges to its desired reference asymptotically. As shown by Fig. 8, further increasing the observer scalar to $\gamma = 20$ brings a much better transient tracking performance including smaller tracking errors and shorter settling time. Similarly, for the case of uncertainty $\delta_{f_3} = -30\%$, it is observed from Fig. 8 that the closed-loop system is unstable with a small observer scalar of $\gamma = 16.5$. As the scalar increased to $\gamma = 22$, it is shown by Fig. 8 that the uncertainty can be effectively compensated from the closed-loop system. As shown by Fig. 8, the variable response curves of the mass spring system under DOBC with an observer scalar $\gamma = 50$ is much closer to the nominal control system, indicating a much better transient control performance, which is referred to as transient performance recovery in [2]. In conclusion, the simulation results in this subsection reveal that a larger observer scalar generally indicates a larger stability region, which confirms the theoretic results presented in Section 3.

5 Conclusions

Robust control of uncertain nonlinear systems has been investigated in this paper via a nonlinear DOBC approach. Rigorous robust stability analysis of uncertainties in nonlinear systems under nonlinear DOBC is established for matched disturbances and uncertainties. Design guidance of nonlinear DOBC has been suggested with the help of the stability analysis results. It has been shown that for certain given plant uncertainties, the nonlinear DOBC approach could achieve asymptotical stability with certain large observer gain γ . Simulation examples of roll stabilization for PGM classical missiles and a nonlinear mass spring mechanical system have been investigated to show the effectiveness of the nonlinear DOBC approach and the interaction between the uncertain level and the observer gain in the nonlinear disturbance observer. The simulation results have confirmed the theoretic results presented in the paper.



Fig. 7. Response curves of the mass spring system (29) under DOBC (33) with various observer scalar γ in the presence of uncertainties $\delta_{f_1} = -50\%$ (Case III): (a) displacement, (b) velocity, (c) control force.



Fig. 8. Response curves of the mass spring system (29) under DOBC (33) with various observer scalar γ in the presence of uncertainties $\delta_{f_3} = -30\%$ (Case IV): (a) displacement, (b) velocity, (c) control force.

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