

1 **THEORETICAL AND EXPERIMENTAL ANALYSIS**
2 **OF THE EXACT RECEPTANCE FUNCTION**
3 **OF A CLAMPED-CLAMPED BEAM WITH**
4 **CONCENTRATED MASSES**

5 **Nguyen Viet Khoa***, **Dao Thi Bich Thao**
6 *Institute of Mechanics, VAST, 18 Hoang Quoc Viet, Hanoi, Vietnam*
7 E-mail: nvkhoe@imech.vast.vn

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9 **Abstract.** This paper establishes the exact receptance function of a clamped-clamped beam
10 carrying concentrated masses. In this paper, the derivation of exact receptance and the
11 numerical simulations are provided. The proposed receptance function can be used as a
12 convenient tool for predicting the dynamic response at arbitrary point of the beam acted
13 by a harmonic force applied at arbitrary point. The influence of the concentrated masses
14 on the receptance is investigated. The numerical simulations show that peak in the recep-
15 tance will decrease when there is a mass located close to that peak position. The numerical
16 results have been compared to the experimental results to justify the theory.

17 *Keywords:* receptance, frequency response function, concentrated mass.

18 **1. INTRODUCTION**

19 The receptance function is very important in vibration problems such as control de-
20 sign, system identification or damage detection since it interrelates the harmonic excita-
21 tion and the response of a structure in the frequency domain. The receptance method
22 was first introduced by Bishop and Johnson [1]. This method has been developed and
23 applied widely in mechanical systems and structural dynamics. Milne [2] proposed a
24 general solution of the receptance function of uniform beams which can be can be applied
25 for all combinations of beam end conditions. Yang [3] derived the exact receptances of
26 non-proportionally damped dynamic systems. In this work an iteration procedure is de-
27 veloped based on a decomposition of the damping matrix, which does not require matrix
28 inversion and eliminate the error caused by the undamped model data. Lin and Lim [4]
29 proposed the receptance sensitivity with respect to mass modification and stiffness mod-
30 ification from the limited vibration test data. Mottershead [5] investigate the measured
31 zeros from frequency response functions and its application to model assessment and
32 updating. Gurgoze [6] presented the receptance matrices of viscously damped systems

33 subject to several constraint equations. In this paper, the frequency response matrix of the
 34 unconstrained system and the coefficient vectors of the constraint equations was used to
 35 obtain the frequency response matrix of the constrained system. Gürgöze and Erol [7]
 36 established the frequency response function of a damped cantilever simply supported
 37 beam carrying a tip mass. In this paper, the frequency response function was derived
 38 by using a formula established for the receptance matrix of discrete linear systems sub-
 39 jected to linear constraint equations, in which the simple support was considered as a
 40 linear constraint imposed on generalized co-ordinates. Burlon et al. [8] derived an ex-
 41 act frequency response function of axially loaded beams with viscoelastic dampers. The
 42 method relies on the theory of generalized functions to handle the discontinuities of the
 43 response variables, within a standard 1D formulation of the equation of motion. In an-
 44 other work, Burlon et al. [9] presented an exact frequency response of two-node coupled
 45 bending-torsional beam element with attachments. Karakas and Gürgöze [10] extended
 46 the work in [3] in which the receptance matrix was obtained directly without using the it-
 47 erations as presented in [3] to form the receptance matrix of non-proportionally damped
 48 dynamic systems. Muscolino and Santoro [11] developed the explicit frequency response
 49 functions of discretized structures with uncertain parameters. Recently, the authors of
 50 this paper [12] presented the exact formula of the receptance function of a cracked beam
 51 and its application for crack detection. However, the exact form of frequency response
 52 function of a beam with concentrated masses has not been established yet.

53 The aim of the present paper is to present an exact receptance function of a clamped-
 54 clamped beam carrying an arbitrary number of concentrated masses. The proposed for-
 55 mula of receptance function is simple and can be applied easily to investigate the dy-
 56 namic response of beam at an arbitrary point under a harmonic force applied at any
 57 point along the beam. The influence of concentrated masses on the receptance of the
 58 clamped-clamped beam is investigated. The comparison between numerical simulations
 59 and experimental results have been carried out to justify the proposed method.

60 2. THEORETICAL BACKGROUND

61 Considering the Euler-Bernoulli beam carrying concentrated masses subjected to a
 62 force as shown in Fig. 1, the governing bending motion equation of the beam can be
 63 extended from [13] as follows

$$EIy'''' + \left[m + \sum_{k=1}^n m_k \delta(x - x_k) \right] \ddot{y} = \delta(x - x_f) f(t) \quad (1)$$

64 where E is the Young's modulus, I is the moment of inertia of the cross sectional area of
 65 the beam, μ is the mass density per unit length, m_k is the k^{th} concentrated mass located
 66 at x_k , $y(x, t)$ is the bending deflection of the beam at location x and time t , $f(t)$ is the force
 67 acting at position x_f , $\delta(x - x_f)$ is the Dirac delta function.

68 Eq. (1) can be rewritten in the form

$$EIy'''' + m\ddot{y} = \delta(x - x_f) f(t) - \sum_{k=1}^n m_k \delta(x - x_k) \ddot{y} \quad (2)$$

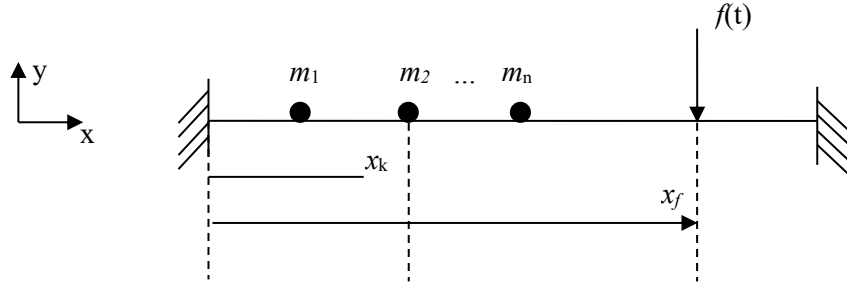


Fig. 1. A clamped-clamped beam with concentrated masses

69 Eq. (2) can be considered as the equation of forced vibration of a beam without con-
 70 centrated masses which is acted by the inertia forces of n concentrated masses and the
 71 external force $f(t)$. The solution of Eq. (2) can be expressed in the form

$$y(x, t) = \sum_{i=1}^{\infty} \phi_i(x) q_i(t), \quad (3)$$

72 where ϕ_i is the i^{th} mode shape of the beam without concentrated masses and q_i is the i^{th}
 73 generalized coordinate.

74 Substituting (3) into (2), yields

$$EI \sum_{i=1}^{\infty} \phi_i''''(x) q_i(t) + m \sum_{i=1}^{\infty} \phi_i(x) \ddot{q}_i(t) = - \sum_{k=1}^n \delta(x - x_k) m_k \sum_{i=1}^{\infty} \phi_i(x) \ddot{q}_i(t) + \delta(x - x_f) f(t). \quad (4)$$

75 Multiplying Eq. (4) by $\phi_j(x)$ and integrating from 0 to L and considering the defini-
 76 tion of the Dirac delta function, one obtains

$$\begin{aligned} & \int_0^L EI \sum_{i=1}^{\infty} \phi_i''''(x) \phi_j(x) q_i(t) dx + \int_0^L m \sum_{i=1}^{\infty} \phi_i(x) \phi_j(x) \ddot{q}_i(t) dx \\ & = - \sum_{k=1}^n m_k \phi_i(x_k) \phi_j(x_k) \ddot{q}_i(t) + \phi_j(x_f) f(t). \end{aligned} \quad (5)$$

The orthogonality of the normal mode shapes of the beam without concentrated masses can be addressed here

$$\int_0^L \phi_i(x) EI \phi_i''''(x) dx = 0 \quad \text{if } i \neq j \quad (6)$$

$$\int_0^L \phi_i(x) m \phi_j(x) dx = 0 \quad \text{if } i \neq j \quad (7)$$

77 Integrating the first equation in Eq. (6) twice by parts, yields

$$\phi_i(x) EI\phi_j'''(x)|_0^L - \phi_i'(x) EI\phi_j''(x)|_0^L + \int_0^L \phi_i''(x) EI\phi_j''(x) dx = 0 \quad \text{if } i \neq j. \quad (8)$$

78 For general boundary conditions the first two terms in Eq. (8) vanish. Thus, from
79 Eq. (8) we have

$$\int_0^L \phi_i''(x) EI\phi_j''(x) dx = \begin{cases} 0 & \text{if } i \neq j \\ \int_0^L \phi_i''^2(x) EI dx & \text{if } i = j \end{cases} \quad (9)$$

80 Applying Eqs. (6)–(9), Eq. (5) can be rewritten as

$$\left[m \int_0^L \phi_i^2(x) dx + \sum_{k=1}^n m_k \phi_i^2(x_k) \right] \ddot{q}_i(t) + \left[EI \int_0^L \phi_i''^2(x) dx \right] q_i(t) = \phi_j(x_f) f(t). \quad (10)$$

By introducing notations

$$\mathbf{M} = m \begin{bmatrix} \int_0^L \phi_1^2(x) dx + \sum_{k=1}^n \bar{m}_k \phi_1^2(x_k) & \sum_{k=1}^n \bar{m}_k \phi_1(x_k) \phi_2(x_k) & \dots & \sum_{k=1}^n \bar{m}_k \phi_1(x_k) \phi_N(x_k) \\ \sum_{k=1}^n \bar{m}_k \phi_2(x_k) \phi_1(x_k) & \int_0^L \phi_2^2(x) dx + \sum_{k=1}^n \bar{m}_k \phi_2^2(x_k) & \dots & \sum_{k=1}^n \bar{m}_k \phi_2(x_k) \phi_N(x_k) \\ \dots & \dots & \dots & \dots \\ \sum_{k=1}^n \bar{m}_k \phi_N(x_k) \phi_1(x_k) & \sum_{k=1}^n \bar{m}_k \phi_N(x_k) \phi_2(x_k) & \dots & \int_0^L \phi_N^2(x) dx + \sum_{k=1}^n \bar{m}_k \phi_N^2(x_k) \end{bmatrix},$$

$$\mathbf{K} = EI \begin{bmatrix} \int_0^L \phi_1''^2(x) dx & 0 & \dots & 0 \\ 0 & \int_0^L \phi_2''^2(x) dx & \dots & 0 \\ \dots & \dots & \dots & \dots \\ 0 & 0 & \dots & \int_0^L \phi_N''^2(x) dx \end{bmatrix},$$

$$\Phi(x) = [\phi_1(x), \dots, \phi_N(x)]^T, \quad \ddot{\mathbf{q}}(t) = [\ddot{q}_1(t), \ddot{q}_2(t), \dots, \ddot{q}_N(t)]^T,$$

$$\mathbf{q}(t) = [q_1(t), q_2(t), \dots, q_N(t)]^T, \quad \bar{m}_k = \frac{m_k}{m}.$$

81 Eq. (10) can be expressed in matrix form as follows

$$\mathbf{M}\ddot{\mathbf{q}}(t) + \mathbf{K}\mathbf{q}(t) = \Phi(x_f) f(t). \quad (11)$$

82 The natural frequency Ω of beam carrying concentrated masses can be obtained by
83 solving the eigenvalue problem associated with Eq. (11), that is

$$\det [\mathbf{K} - \Omega^2 \mathbf{M}] = 0. \quad (12)$$

84 If the force is harmonic $f(t) = \bar{f}e^{i\omega t}$ then the solution of Eq. (11) can be found in
85 the form

$$\mathbf{q}(t) = \bar{\mathbf{q}}e^{i\omega t} \quad (13)$$

86 Substituting Eq. (13) into Eq. (11) yields

$$(\mathbf{K} - \omega^2 \mathbf{M}) \bar{\mathbf{q}} = \Phi(x_f) \bar{f}. \quad (14)$$

87 The receptance function is defined as the frequency response function in which the
88 response is the displacement. This means that in the frequency domain: receptance =
89 displacement/force. Thus, left multiplying Eq. (14) with $\frac{\Phi^T(x)}{\bar{f}} [(\mathbf{K} - \omega^2 \mathbf{M})]^{-1}$ the re-
90 ceptance at x due to the force at x_f is obtained

$$\alpha(x, x_f, \omega) = \frac{\Phi^T(x) \bar{\mathbf{q}}}{\bar{f}} = \Phi^T(x) (\mathbf{K} - \omega^2 \mathbf{M})^{-1} \Phi(x_f). \quad (15)$$

91 It is note that when the infinite modes are applied, i.e. $N \rightarrow \infty$, Eq. (15) becomes the
92 exact formula of the receptance function.

93 For the clamped-clamped beam, following relations can be derived:

$$\begin{aligned} \phi_i(x) &= \frac{\sin \alpha_i L + \sinh \alpha_i L}{\cos \alpha_i L - \cosh \alpha_i L} (\sin \alpha_i x - \sinh \alpha_i x) + \cos \alpha_i x - \cosh \alpha_i x, \\ \phi''_i(x) &= -\alpha_i^2 \left[\frac{\sin \alpha_i L + \sinh \alpha_i L}{\cos \alpha_i L - \cosh \alpha_i L} (\sin \alpha_i x + \sinh \alpha_i x) + \cos \alpha_i x + \cosh \alpha_i x \right], \\ \int_0^L \phi_i^2(x) dx &= L, \\ \int_0^L [\phi''_i(x)]^2 dx &= L \alpha_i^4, \end{aligned} \quad (16)$$

94 where α_i is the solution of the frequency equation $\cos \alpha L \cosh \alpha L - 1 = 0$.

95 From Eq. (16) the matrices \mathbf{M} and \mathbf{K} are derived

$$\mathbf{M} = m \begin{bmatrix} L + \beta_{11} & \beta_{12} & \dots & \beta_{1N} \\ \beta_{21} & L + \beta_{22} & \dots & \beta_{2N} \\ \dots & \dots & \dots & \dots \\ \beta_{N1} & \dots & \dots & L + \beta_{NN} \end{bmatrix}, \quad \mathbf{K} = EIL \begin{bmatrix} \alpha_1^4 & 0 & \dots & 0 \\ 0 & \alpha_2^4 & \dots & 0 \\ \dots & \dots & \dots & \dots \\ 0 & \dots & \dots & \alpha_N^4 \end{bmatrix} \quad (17)$$

96 where

$$\begin{aligned} \beta_{ij} &= \sum_{k=1}^n \left\{ \bar{m}_k \left[\frac{\sin \alpha_i L + \sinh \alpha_i L}{\cos \alpha_i L - \cosh \alpha_i L} (\sin \alpha_i x_k - \sinh \alpha_i x_k) + \cos \alpha_i x_k - \cosh \alpha_i x_k \right] \right. \\ &\quad \times \left. \left[\frac{\sin \alpha_j L + \sinh \alpha_j L}{\cos \alpha_j L - \cosh \alpha_j L} (\sin \alpha_j x_k - \sinh \alpha_j x_k) + \cos \alpha_j x_k - \cosh \alpha_j x_k \right] \right\}. \end{aligned} \quad (18)$$

The exact formula of the receptance of the clamped-clamped beam carrying concentrated masses will be derived from Eqs. (16)–(18).

3. NUMERICAL SIMULATION

3.1. Reliability of the theory

In order to check the reliability of the proposed receptance, frequency parameters $\alpha_i L$ of a clamped-clamped beam carrying two masses are calculated from Eq. (12) and compared to Ref. [14]. Five lowest frequency parameters of the clamped-clamped beam with two concentrated masses $\bar{m}_1 = \bar{m}_2 = 0.5$ attached at $0.25L$ and $0.75L$ obtained by two methods are listed in Tab. 1. As can be seen from this table, the first five frequency parameters of the present work are in excellent agreement with Ref. [14]. This result justifies the reliability of the proposed receptance function.

Table 1. Frequency parameters of the clamped-clamped beam

Frequency parameters	Ref. [14]	Present paper	Error (%)
$\alpha_1 L$	4.0973	4.0976	0.00007
$\alpha_2 L$	5.8984	5.8995	0.00019
$\alpha_3 L$	9.1453	9.1534	0.00089
$\alpha_4 L$	13.7527	13.7567	0.00029
$\alpha_5 L$	16.9258	16.9399	0.00083

3.2. Influence of location of the concentrated masses on the receptance

In this paper, the numerical simulations of a clamped-clamped beam with two masses are presented. Parameters of the beam are: Mass density $\rho = 7800 \text{ kg/m}^3$; modulus of elasticity $E = 2.0 \times 10^{11} \text{ N/m}^2$; $L = 1 \text{ m}$; $b = 0.02 \text{ m}$; $h = 0.01 \text{ m}$. Two equal concentrated masses of 0.6 kg are attached on the beam in different scenarios. The first five mode shapes are used to calculate the receptance. The receptance matrices are calculated at 50 points spaced equally on the beam while the force moves along these points.

The receptance of the clamped-clamped beam without masses is calculated first. Fig. 2 presents the receptance matrices when the forcing frequencies equal to the first, second and third natural frequencies of the beam-mass system, respectively. As can be seen from Fig. 2(a) when the forcing frequency is equal to the first natural frequency, the receptance is maximum at the middle of the beam which corresponds to the position where the amplitude of the first mode is maximum. As can be observed from Fig. 2(b) that when the forcing frequency is equal to the second natural frequency, the receptance is maximum at position of about $0.3L$ and $0.7L$ from the left end of the beam which are the positions where the amplitude of the second mode shape is maximum. Meanwhile, the receptance is smallest at the middle of beam which corresponds to the position where the amplitude of the second mode shape is minimum. Fig. 2(c) presents the receptance matrix of the beam when the frequency of the force is equal to the third natural frequency. The receptance of the beam is maximum at the positions of about $0.2L, 0.5L$ and

128 $0.8L$ where the amplitude of the third mode shape is maximum. The receptance is minimum
 129 at positions of about $0.35L, 0.65L$ where the amplitude of the third mode shape is
 130 minimum. It can be concluded that, when the excitation frequency is equal to a natural
 131 frequency the positions of maxima and minima in the receptance are the same with the
 132 positions of maxima and minima in the corresponding mode shape. Therefore, similar
 133 to the mode shape, we call the maxima in the receptance “peaks of receptance” and the
 134 minima in the receptance “nodes of receptance”.

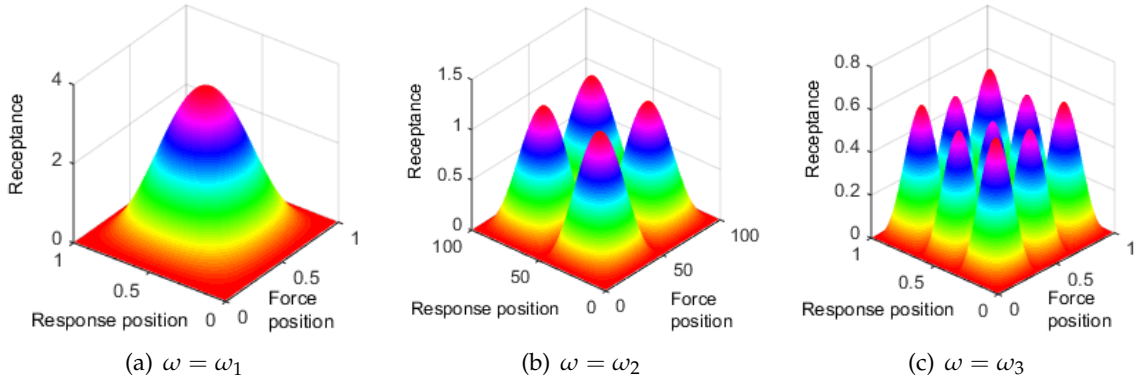


Fig. 2. Receptance of beam without a masse

135 When there is a concentrated mass, the receptance matrix of the beam is changed.
 136 Fig. 3 presents the receptance matrices of the beam when the forcing frequency is equal
 137 to the first natural frequency of the beam-mass system. As can be seen from this figure,
 138 when the mass is located $0.25L$ the position of the peak of receptance “moves” to the
 139 left end of beam. However, when the mass is located at the middle of the beam, the
 140 shape of the receptance is unchanged. The change of position of the peak of receptance
 141 is depicted clearer in Fig. 4 when the force is fixed at position $0.5L$. As can be observed
 142 from this figure, the peak of receptance moves to the position of $0.4L$ when the mass is
 143 located at $0.25L$. The receptance seems to be “pulled’ toward the mass position.

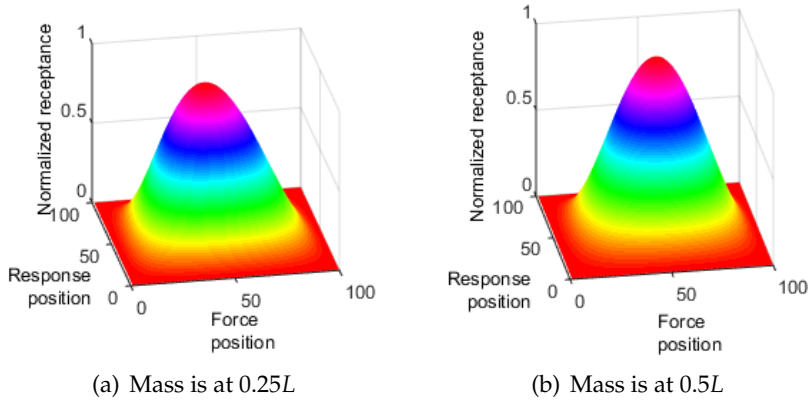


Fig. 3. Receptance matrices at $\omega = \omega_1$

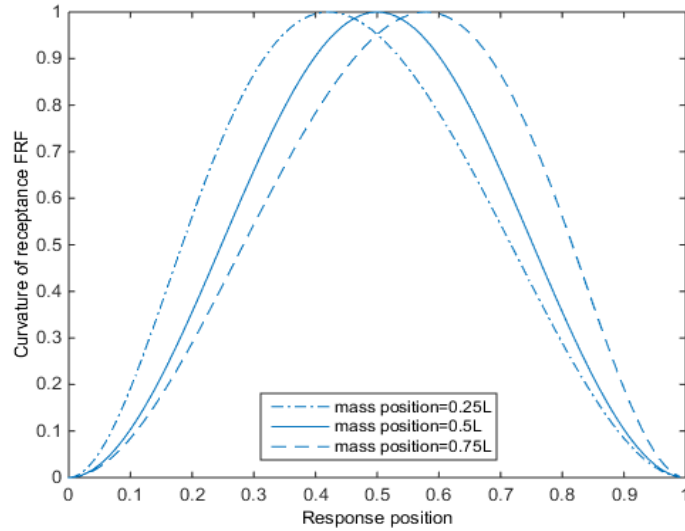


Fig. 4. Receptance of beam with force position is at $L/2, \omega = \omega_1$

144 Fig. 5 presents the receptance of beam carrying a concentrated mass at different positions
 145 when the forcing frequency is equal to the second natural frequency. As shown in
 146 Fig. 5(a), when the mass is located at $0.3L$, the peaks corresponding to either the response
 147 position of $0.3L$ or the force position of $0.3L$ decrease significantly. When the mass is located
 148 the middle of the beam, the receptance shape is unchanged as shown in Fig. 5(b).
 149 These results show that when the mass is attached at a peak of the receptance matrix, the
 150 peaks corresponding to either the response position or force position which is close to
 151 the mass position will decrease. Meanwhile, the shape of receptance is unchanged when
 152 the mass is attached at the nodes of the receptance. The change in receptance can be observed
 153 in more detail as presented in Fig. 6 when the force is fixed at position $0.25L$. As
 154 can be seen from this figure, the peak of receptance decreases. In addition, the peak of
 155 receptance moves slightly toward the mass position.

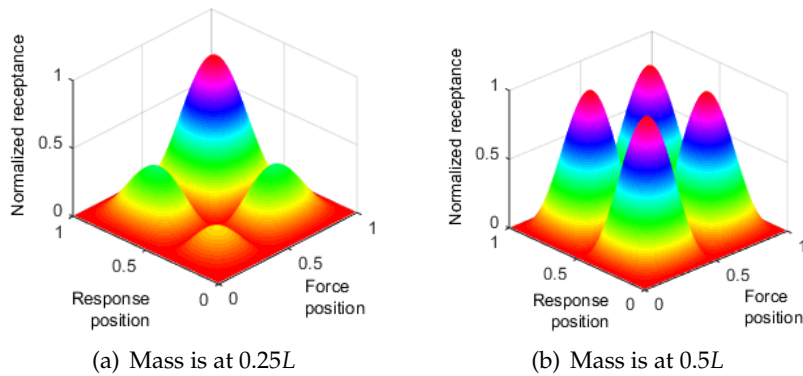


Fig. 5. Receptance of beam at $\omega = \omega_2$

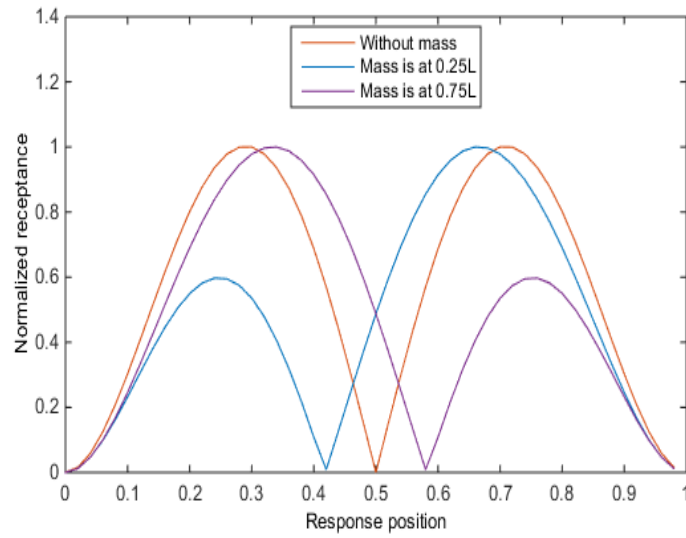


Fig. 6. Measured receptance with the force acting at $0.25L, \omega = \omega_2$

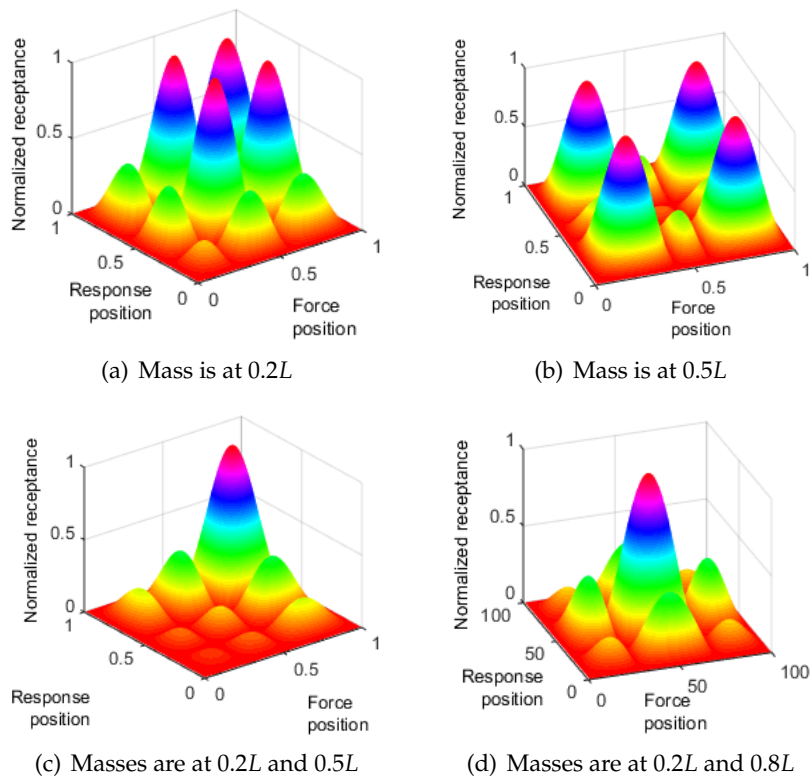


Fig. 7. Normalized receptance at $\omega = \omega_3$

156 Fig. 7 presents the receptance of beam when the forcing frequency is equal to the
 157 third natural frequency. As shown in Fig. 7(a), when one mass is located at $0.2L$ or $0.8L$,
 158 the peaks corresponding to either the response position of $0.2L$ or the force positions of
 159 $0.2L$ decrease significantly. When one mass is located at $0.5L$, the peaks corresponding
 160 to either the response position of $0.5L$ or the force position of $0.5L$ decrease significantly
 161 as shown in Fig. 7(b). When two masses are located at $0.2L$ and $0.5L$, the peaks corre-
 162 sponding to either the response positions of $0.2L, 0.5L$ or the force positions of $0.2L, 0.5L$
 163 decrease significantly as depicted in Fig. 7(c). When two masses are located at $0.2L$ and
 164 $0.8L$, the peaks corresponding to either the response positions of $0.2L, 0.8L$ or the force
 165 positions of $0.2L, 0.8L$ decrease significantly as shown in Fig. 7(d). The change in recep-
 166 tance can be seen in more detail when the force is fixed at position $0.2L$ as depicted in
 167 Fig. 8. Similar conclusion can be drawn from this figure that when the masses attached
 168 at peaks, these peaks will decrease significantly. When there is one mass the peaks of
 169 receptance move toward the mass position. When there are two masses attached sym-
 170 metrically at $0.2L$ and $0.8L$ the peak at $0.2L$ moves to the left end, while the peak at $0.8L$
 171 moves to the right end. When there are two masses attached at $0.2L$ and $0.5L$, the recep-
 172 tance is “pulled” to the left end. In this case, the receptance tends to “move” toward the
 173 heavier side of the beam.

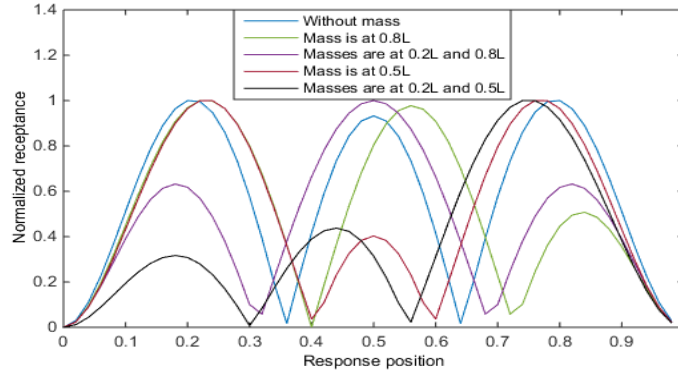


Fig. 8. Normalized receptance when the force is fixed at $0.2L, \omega = \omega_3$

174

4. EXPERIMENT RESULTS

175 The experimental setup is illustrated in Fig. 9. The clamped-clamped beam with
 176 the same parameters presented in Section 3.1 has been tested. The beam is excited by
 177 the Vibration Exciter Bruel & Kjaer 4808 and the response is measured by the instrument
 178 Polytec Laser Vibrometer PVD-100. Two equal concentrated masses of 0.6 kg are attached
 179 on the beam in different scenarios. The receptance is measured along the beam when the
 180 forcing frequency is set to the first three natural frequencies of the beam-mass system.
 181 The receptance matrix is obtained at 50 points spaced equally on the beam.

182 According to the simulation results, when the forcing frequency is equal to the first
 183 natural frequency, the change in receptance is simple that it has only one peak at the mid-
 184 dle of the beam and it moves toward the position of the attached mass. Meanwhile, when

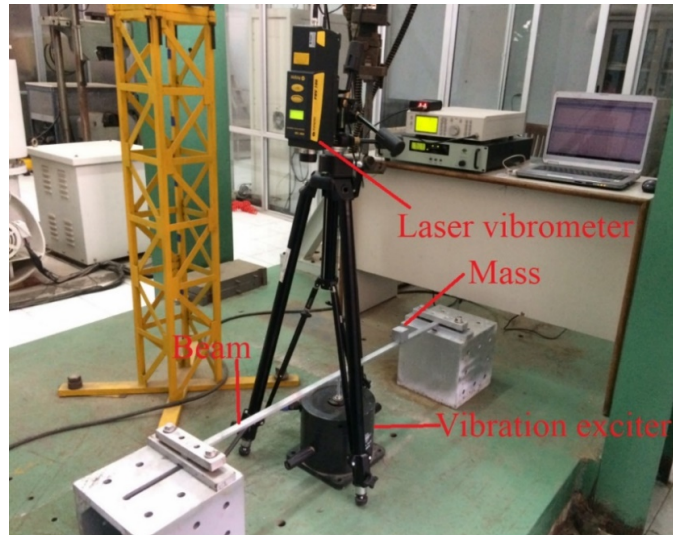


Fig. 9. Experimental setup

185 the forcing frequency is high the change in receptance becomes more complicated with
 186 different configurations of the attached masses. Therefore, when the forcing frequency
 187 is equal to the first natural frequency only the receptance curve extracted with the force
 188 fixed at one position is measured, while the whole receptance matrices are measured at
 189 the second and third natural frequencies.

190 When the mass is attached at the position of $L/4$, the force is fixed at $L/2$ and the
 191 forcing frequency is equal to the first natural frequency, the measured receptance moves
 192 to the left end as presented in Fig. 10. Comparing Figs. 4 and 10 it is concluded that the
 193 measured receptance and the simulation results are in very good agreement in both cases
 194 without and with an attached mass.

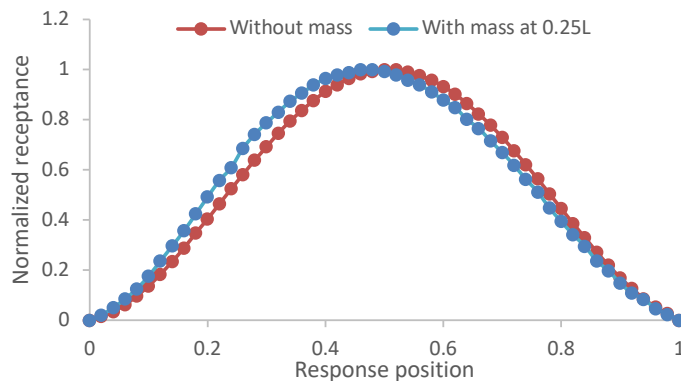


Fig. 10. Measured receptance curves of beam, force position = $L/4$, $\omega = \omega_1$

195 When the excitation frequency is equal to the second natural frequency and the mas
 196 is attached at the position of $L/4$, the measured receptance matrix presented in Fig. 5(a)
 197 and the simulation receptance matrix shown in Fig. 11(a) are in very good agreement. As
 198 can be seen from Fig. 11(a), three peaks corresponding to the position of $L/4$ in the re-
 199 ceptance matrix decrease significantly. When the mass is attached at $L/2$ and the forcing
 200 frequency is equal to the third natural frequency, five peaks of the receptance matrix cor-
 201 responding to the position of $L/2$ decrease significantly as can be observed in Fig. 11(b).
 202 This agrees with the simulation result depicted in Fig. 7(b). Fig. 12 presents the experi-
 203 mental receptance curves of beam without and with an attached mass at $L/4$ which was
 204 measured when the forcing frequency is equal to the second natural frequency. When
 205 there is no mass attached, these receptance has two peaks at $L/4$ and $3L/4$. These exper-
 206 imental results justify the correctness of the simulation results presented in Fig. 6. When
 207 there is a mass attached at the position of $L/4$, the peak at the mass position decreases
 208 clearly. Fig. 13 presents the receptance measured when the force frequency is equal to the
 209 third natural frequency. As can be seen from this figure, when there is no mass attached,
 210 the receptance has three peaks at $L/6, L/2$ and $5L/6$. When there are masses attached,

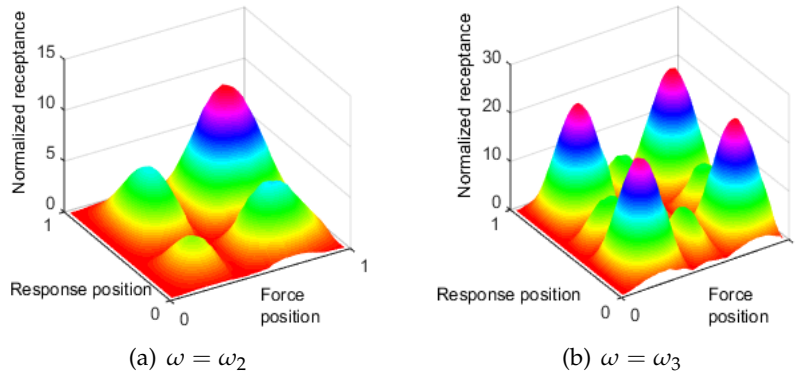


Fig. 11. Measured receptance matrices of beam

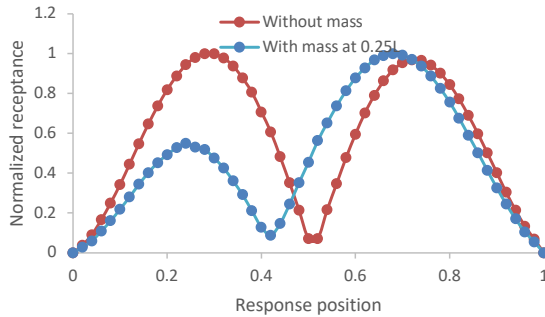


Fig. 12. Measured receptance curves of beam, force position = $L/4$, $\omega = \omega_2$

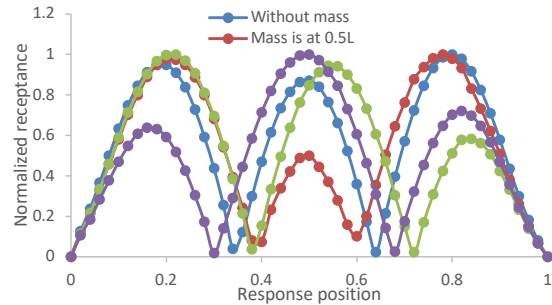


Fig. 13. Measured receptance curves of beam, force position = $L/6$, $\omega = \omega_3$

211 the receptance peaks decrease significantly at the mass positions. The experimental re-
 212 sults presented in Fig. 13 are in very good agreement with the simulation results depicted
 213 in Fig. 8.

214

5. CONCLUSIONS

215 In this paper, the exact receptance function of clamped-clamped beam carrying con-
 216 centrated masses is derived. The proposed receptance function can be applied easily
 217 for predicting the response of the beam under a harmonic excitation. The influence of
 218 the concentrated masses on the receptance of beam is also investigated. When the ex-
 219 citation frequency is equal to a natural frequency, the peaks and nodes positions of the
 220 receptance are the same with the maximum and minimum positions of the correspond-
 221 ing mode shape. When there are concentrated masses the shape of receptance is changed.
 222 When the mass positions are close to peaks of receptance, these peaks will decrease sig-
 223 nificantly. When the masses are located at the nodes of receptance, the receptance is not
 224 influenced. The influence of masses on the receptance matrices can be used to control the
 225 vibration amplitudes at some specific positions at given forcing frequencies.

226 The experiment has been carried out when the forcing frequency is set to the first
 227 three natural frequencies of the beam carrying concentrated masses. The experimental
 228 and simulation results are in very good agreement which justifies the proposed method.

229

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