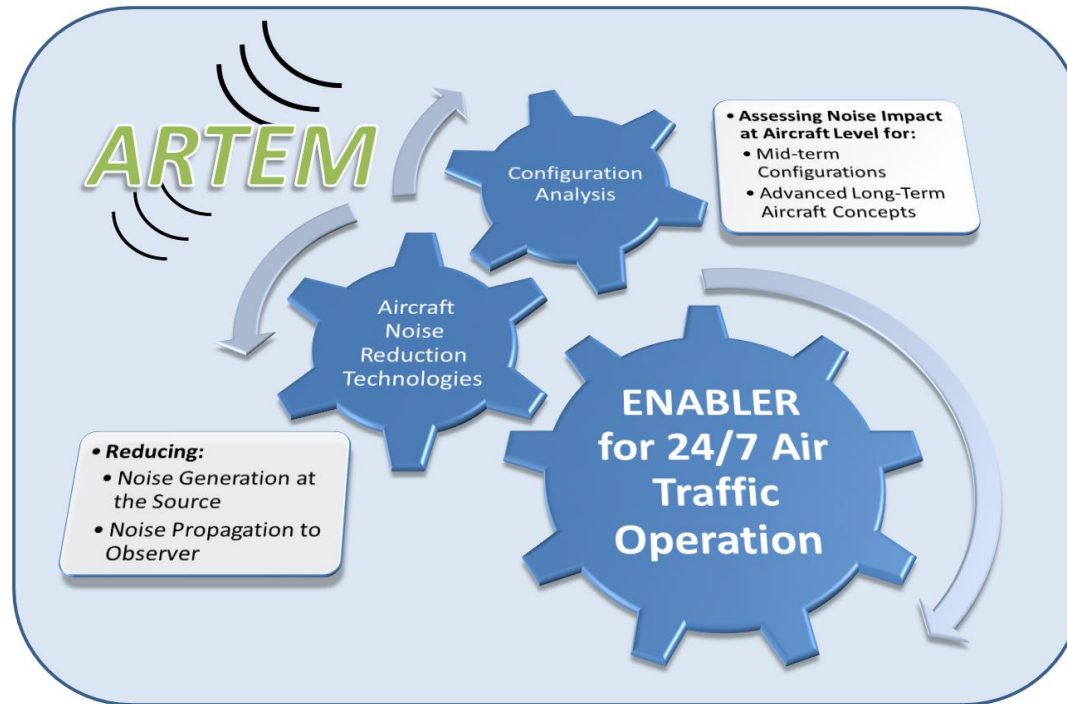


# Noise reduction pathways

**Noise Reduction Technologies for Future Aircraft Concepts -  
First results of the H2020 project "ARTEM"**

**Dr.-Ing. Karsten Knobloch  
German Aerospace Center (DLR)**

# ARTEM is an Enabler Project



***By assessing "Generation 3" Noise Reduction Technologies, ARTEM is the "Technology Project" among the projects of MG-1-2-2017 "Reducing aviation noise"***

**Duration: 4 years, Start: 01.12.2017**

**7.5M€ funding from EC**

**Coordination:**



**24 Partners from**

**Industry**

**SME**

**Research Establishments**

**Universities**

**AIRBUS**

**AEDS**



**Bucharest, 27-30 May 2019**



## **ARTEM will advance low-TRL technologies**

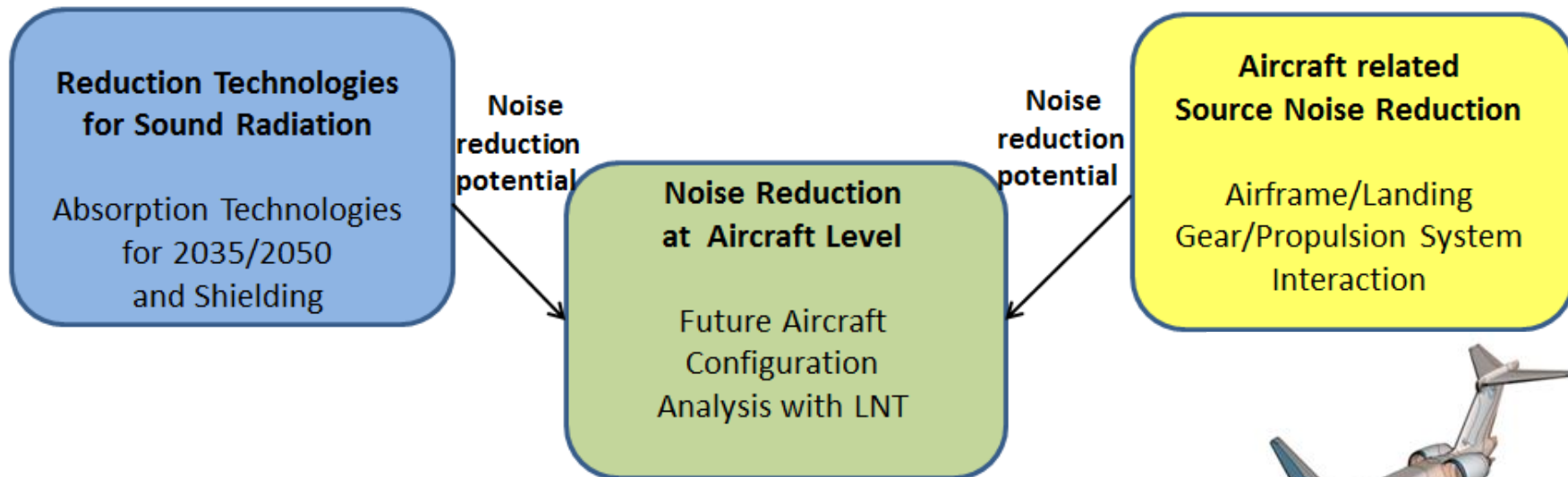
Novel and innovative ideas for liners and dissipative surfaces will be taken up and their potential assessed.

## **ARTEM considers the interaction noise sources**

The interaction of components and associated noise generation must be understood to reduce the overall aircraft noise signature.

**The assessment will be performed for future aircrafts concepts ranging from 2025-2050**

# ARTEM considers technologies and configurations for 2035/2050

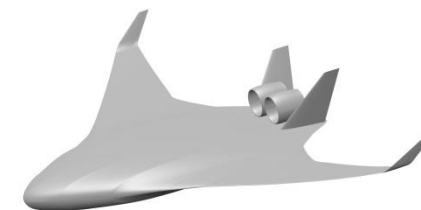


***Define novel aircraft configurations for 2035 and 2050***

***Ensure that these configurations will build on similar platforms considered in other European and national research programs***



BLI



# Configurations to be considered

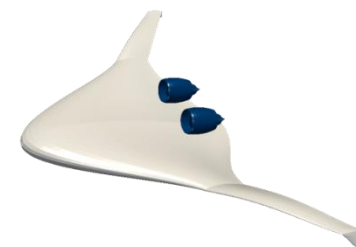
2025-2030: defined outside ARTEM,  
e.g. advanced tube-and-wing with UHBR engines

2035: NOVA concept of ONERA,  
tube and wing with rear-mounted engines  
with BLI



BLI

2050: **B**lended Wing Body  
Optimized with **LNT** (BOLT)



2050: **RE**gional **Bwb** **E**lectric-propelled (REBEL)

# WP1 „Reduction Technologies for Sound Radiation“

## Objectives

Investigation of **novel noise reduction** means to be **applied** in the mid (2035) and long term scenarios (2050) of **all types of future aircraft configurations**.

**Qualification of novel concepts from pure ideas and technology concepts (TRL1/2) to TRL3/4** (experimental proof of concept and validation in lab environment) for further development towards industrial applications.

Study the **capability to reduce noise signature** at the ground by making full use of the available design space of airframe and propulsion system placement and to **shield and reflect the generated noise** away from the ground.

T1.1 Absorption technologies for 2035

T1.2 Absorption technologies for 2050

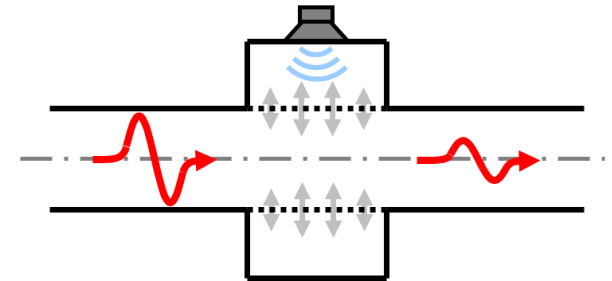
T1.3 Shielding

# WP1 - Technologies

## **Absorption Technologies 2035:**

*...aiming at low-frequency and broadband reduction capabilities*

- *3-D printing of complex liner geometries*
- *Zero-Mass-Flow-concept*
- *Assessment of surface roughness effects*



## **Absorption Technologies 2050:**

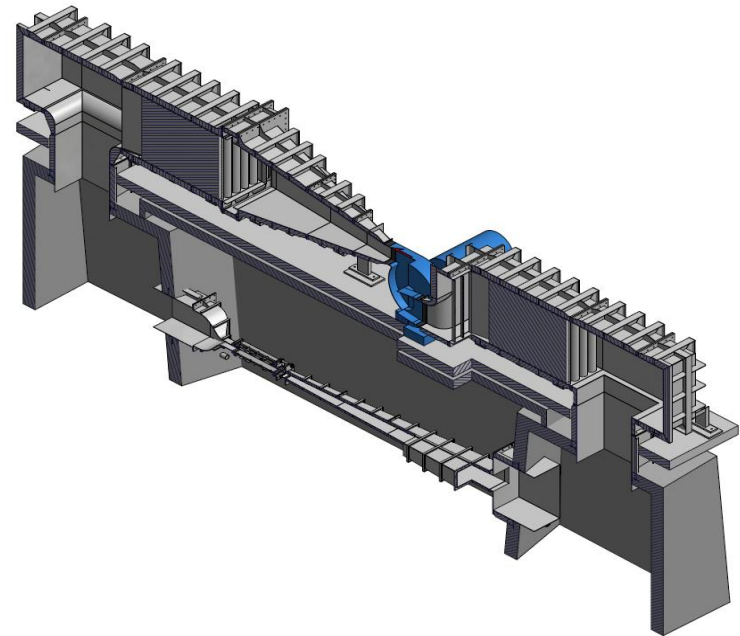
*...qualify innovative ideas and assess potential of*

- *Friction-powder liners*
- *Semi-active membrane liners*
- *Plasma actuators*
- *Numerical modelling of innovative absorption technologies*
- *Metamaterial-based innovative devices*



# Absorption technologies results assessment

- *Use of common facilities for comparative testing and benchmarking of technologies after initial development*
- *Transfer results to WP3 for full scale engine/aircraft assessment*



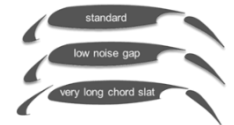
# WP2 „Aircraft related Source Noise Reduction “



## Objectives

- *Assess installation noise sources on conventional and novel aircraft configurations (aerodynamic and aero-acoustic installation effects)*
- *Assess potential noise reduction solutions (geometrical and technological,*
- *Investigate numerically and experimentally to improve the understanding of physical phenomena*

T2.1 High lift dev. noise



T2.2 Jet installation noise



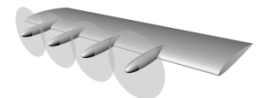
T2.3 Boundary layer ingestion effects



T2.4 Landing gear installation effects

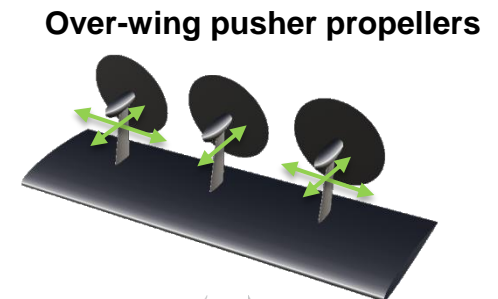
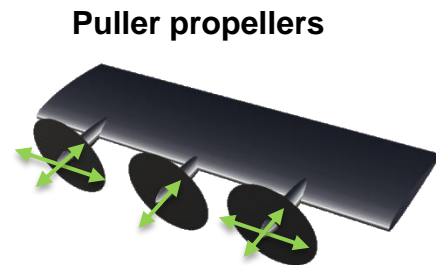


T2.5 Distributed elect. propulsion



# Selection of Noise Reduction Technologies and Effects

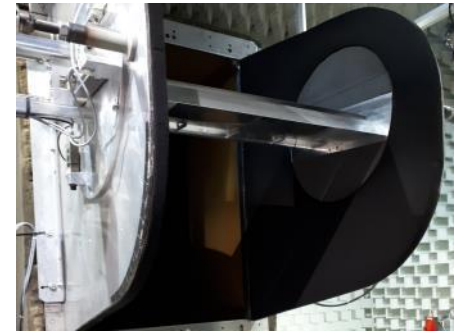
- *HLD:*  
*porous flaps, serrated trailing edges, micro-surface structuring, thrust gates, Chevrons, plasma actuators,...*
- *Assess the acoustic implications of BLI installation*
- *Assess installation options (and associated interaction effects) for landing gear*
- *Assess effects of distributed (electric) propulsion*



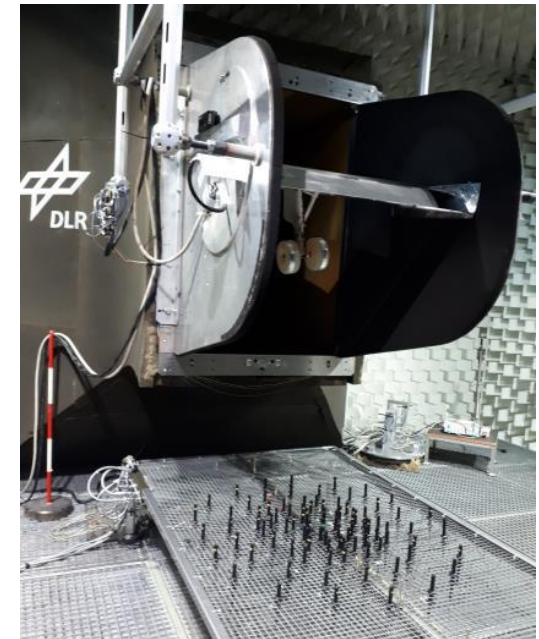
# Current Achievements -1

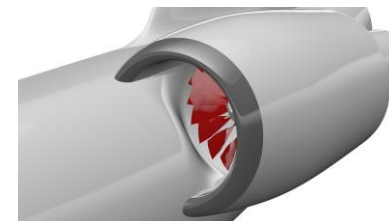


*First test campaign on Landing gear configurations in aero-acoustic wind tunnel AWB finished (May 2019)*



- Global aerodynamic characterisation of 3-element wing for 3 different slat/flap settings
- Local aerodynamics + far-field acoustic measurements of wing with and without LG (3 high lift settings)
- Local aerodynamics + far-field acoustic measurements of fuselage/belly fairing + wing with and without LG (2 high lift settings)





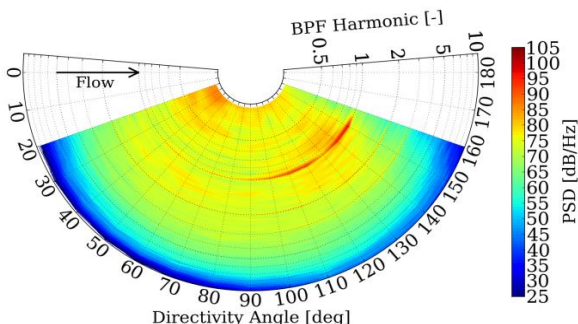
# Current Achievements -2

Full-scale assessment on BLI installation of a modified NASA SDT engine for NOVA configuration (for low-speed conditions), comparison with isolated engine

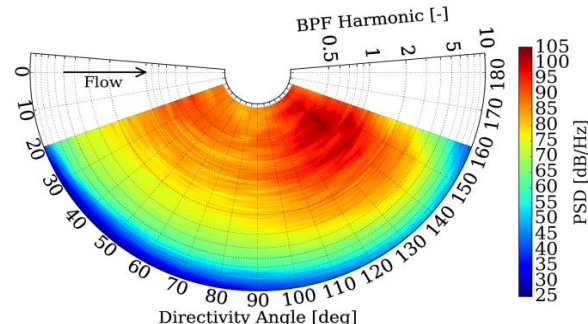
→ TU Delft presented at AIAA-Aeroacoustics Conference May 2019, Delft/The Netherlands, Paper AIAA2019-2429

Below: PSD of far-field noise directivity – 10 m arc at  $\phi=0^\circ$ :

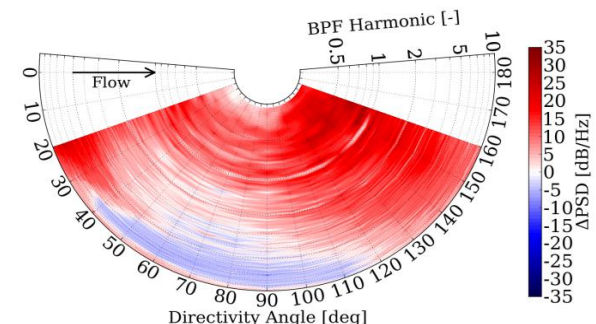
- Noise radiated most efficiently downstream for both engines
- High broadband levels for BLI engine
- 10-20 dB higher noise levels for BLI engine (except for  $40^\circ$ - $110^\circ$  sector above BPF-2)



**Isolated engine**



**BLI engine**



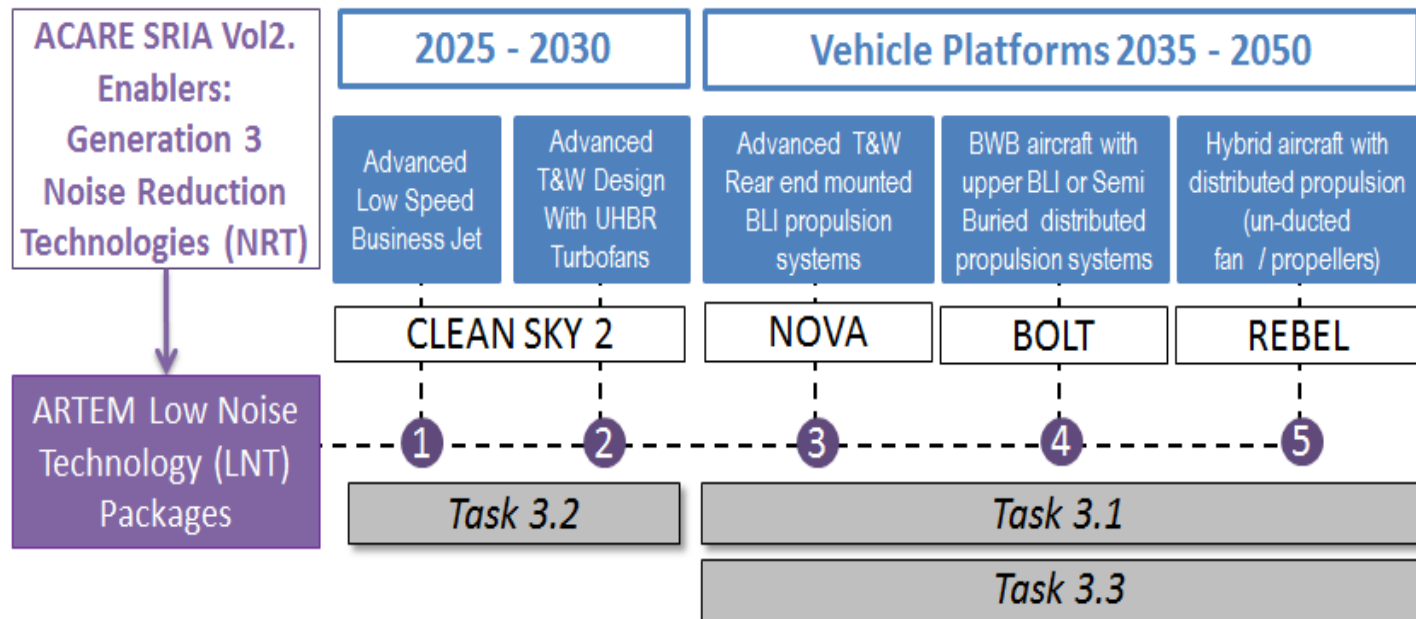
**BLI - isolated**

→ Mitigation means: BLI intake and engine internal design optimization, acoustic liners,...

# WP3 “Noise Reduction at Aircraft Level”



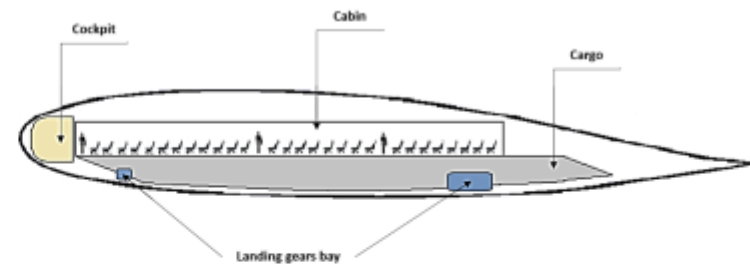
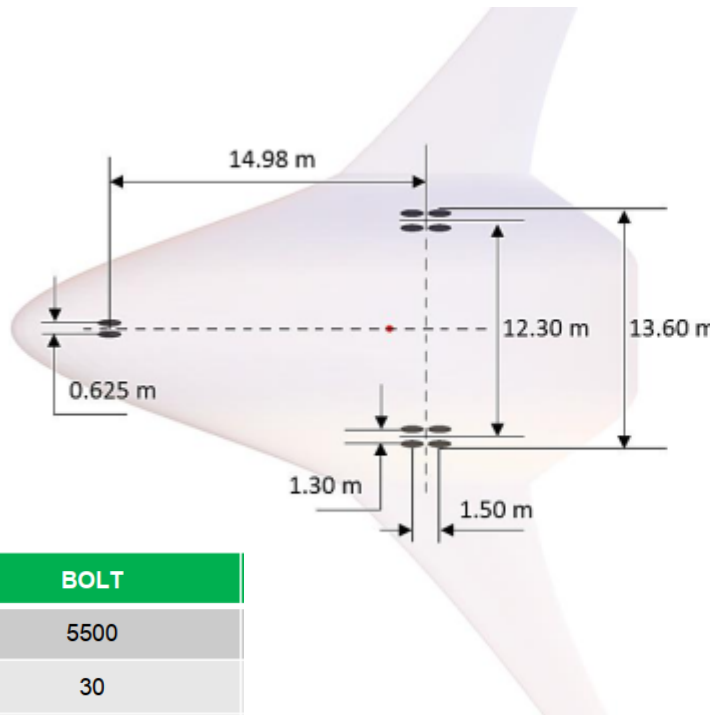
- Configuration analysis with and without Low-Noise Technology
- Noise predictions for fly-over
- Auralization



# Current Achievements



*Configuration definition analysis for BOLT and REBEL using multi-dimensional optimization tool FRIDA (U Roma3)*



Landing gear bays volume		$m^3$	12,6
Main landing gear	No.	#	2
	Wheels per struct	#	4
	Tyre diameter	$m$	1,35
	Struct. length	$m$	3,11
Nose landing gear	No.	#	1
	Wheels per struct	#	2
	Tyre diameter	$m$	1,01
	Struct. length	$m$	2,33

		BOLT
Range	nmi	5500
Loiter	min	30
Payload	#pax	400
Cruise altitude	ft	43000
Cruise Mach number	-	0,84

# Potential gaps and challenges

- *Support of industry is needed for assessment of technology development*
- *Overall schedule is challenging with assessment and exploitation depending on project results availability*
  - *data exchange is tested in advance with preliminary data*
- *Open access publications are requested, while the field of potential journals is evolving and changing quickly*
  - *need to judge between quality, outreach, cost,...*



# Cross-cutting issues

*Project results of **AERIALIST** regarding meta-materials will feed into ARTEM development work*

*→ direct link via the participation of U Roma3*

*ARTEM technologies will be assessed by **ANIMA** global coordination action in order to advise for future developments*

*→ links via direct involvement of partners (ONERA, UCP, DLR) in both project, exchange between project leads*

***CEAS-Workshop** planned for dissemination and discussion of project results*

# Impacts

- *Development (up to TRL 3/4) and assessment of potential of novel noise reduction technologies*  
→ *basis for future high TRL work*
- *Compilation of data bases for various effects (shielding, installation effects, distributed propulsion)*  
→ *enables tool validation*
- *Advancement of simulation capabilities and analytical/semi-empirical tools*
- *Direct assessment of technologies also for near-term configurations via CS-configuration analysis*

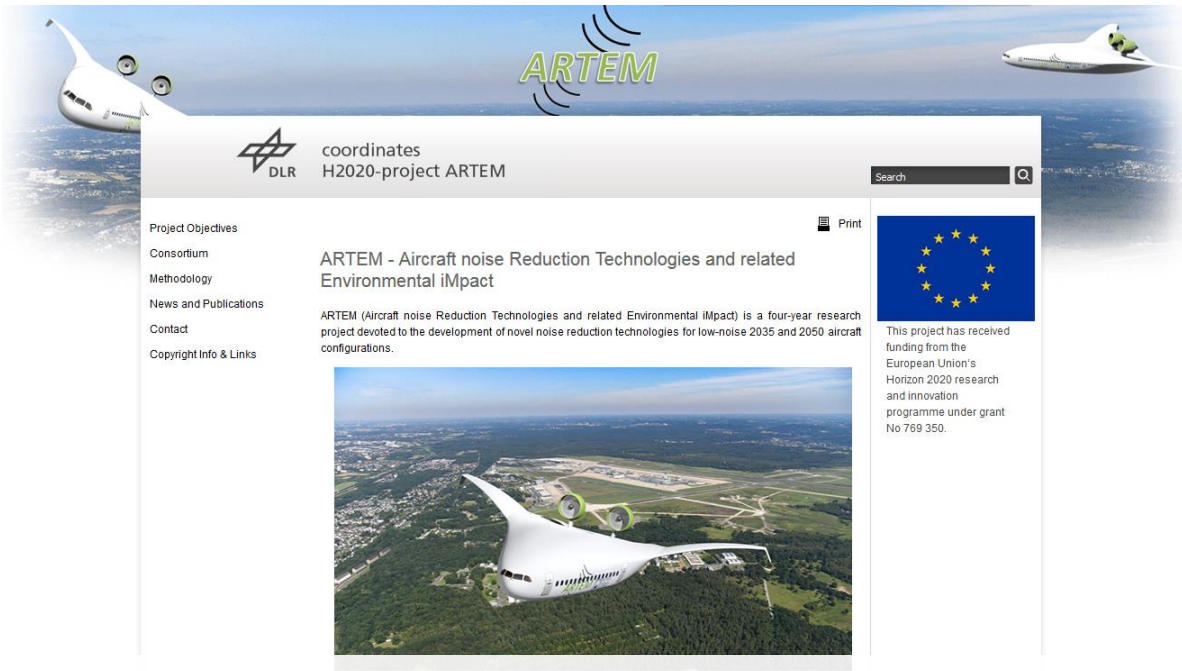
# Acknowledgements



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**Bucharest, 27-30 May 2019**

