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I am submitting herewith a thesis written by James L. Scott entitled "A Calorimetric Investigation of Zirconium, Titanium, and Zirconium Alloys from 60 to 960°C." I have examined the final electronic copy of this thesis for form and content and recommend that it be accepted in partial fulfillment of the requirements for the degree of Doctor of Philosophy, with a major in .

E.E. Stansbury, Major Professor

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ARRAY(0x7f6ffe240f68)

Accepted for the Council: Dixie L. Thompson

Vice Provost and Dean of the Graduate School

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May 24, 1957

To the Graduate Council:

I am submitting herewith a thesis written by James L. Scott entitled "A Calorimetric Investigation of Zirconium, Titanium, and Zirconium Alloys from 60 to 960°C." I recommend that it be accepted in partial fulfillment of the requirements for the degree of Doctor of Philosophy, with a major in Metallurgy.

Otanst

Major Professor

We have read this thesis and recommend its acceptance:

ilsen

Accepted for the Council:

Dean of the Graduate School

A CALORIMETRIC INVESTIGATION OF ZIRCONIUM, TITANIUM,

AND ZIRCONIUM ALLOYS FROM 60 to 960°C

A THESIS

Submitted to

The Graduate Council

of

The University of Tennessee

in

Partial Fulfillment of the Requirements

for the degree of Doctor of Philospphy

> by James L. Scott

> > **June** 1957

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CHAPTER I

SUMMARY

The favorable nuclear properties of zirconium make it a very attractive construction material for nuclear reactors. Full benefit of the material, however, can only be gained by the addition of alloys which increase the high_temperature strength. Consequently, a fundamental knowledge of the alloying behavior of zirconium is desirable. A better insight to this behavior may be secured by studying the similarities and differences between alloys of zirconium, titanium, and hafnium since these elements appear to be quite similar in the metallic state.

One approach to the study of the alloying behavior of a metal is the determination of the changes of the thermodynamic functions with the addition of a given solute. For such a determination, **a** knowledge of the specific heats and heats of transformation of the alloys of interest is required. Consequently, the measurement of the specific heats and heats of transformation of a series of alloys of zirconium as well as titanium was undertaken.

The pure metals and alloys studied were iodide zirconium, iodide titanium, and alloys of zirconium with 0.881 and 5.37 weight per cent silver, 7.77 weight per cent indium, 17.5 weight per cent niobium, 34.4 weight per cent titanium, and 300 parts per million of hydrogen. Measurements were made over the temperature range of 60 to 960°C through the use of a high-temperature-adiabatic calorimeter. The results are presented graphically in the text. The conclusions which were drawn from this investigation are:

1. The specific heat of beta zirconium is lower than that of alpha at the transformation temperature of $870 \pm 2^{\circ}$ C. Analogously the specific heat of beta titanium is lower than that of alpha at the transformation temperature of $883 \pm 2^{\circ}$ C. For a zirconium - 17.5 weight per cent niobium alloy, which should correspond quite well to pure zirconium, the specific heat of the beta phase has the constant value 0.069 \pm 0.002 calories/gram-°C from room temperature to 800° C. Within the limits of error the addition of 0.746 atomic per cent of silver or 6.27 atomic per cent of indium does not affect the specific heat of either the alpha or beta phase over their ranges of stability above room temperature. An alloy of higher silver content (4.58 atomic per cent) does have a slightly higher specific heat of the intermetallic compound Zr_0Ag .

2. The heat of transformation of zirconium is 993 ± 25 calories per gram atom at 870 °C, and the heat of transformation of titanium is 978 ± 25 calories per gram atom at 883 °C. Within the limits of experimental error the heats of transformation of zirconium and titanium would have the same value if the transformations occurred at the same temperature. Although the transformation from the alpha to the beta phase in a 0.746 atomic per cent silver alloy occurs over the temperature range 830 - 865 °C, the heat of transformation at an arbitrary intermediate temperature is the same as the value zirconium would have if it transformed at the same temperature within the limits of error. This is also true for an alloy of zirconium with 6.27 atomic per cent of indium,

which transforms over the range of 910 - 940°C. The temperature of the eutectoid transformation in the zirconium silver system is 825 ± 2 °C and the heat of transformation of the eutectoid (3.73 atomic per cent silver) is 1240 ± 25 calories per gram atom.

3. The changes in the thermodynamic functions ${\bigtriangleup}\ {\rm H}_{\underset{\mbox{\scriptsize o}}{}}$ and

$$\int_{O}^{T} dT \int_{O}^{T} \Delta C^{X} d \ln T$$

are quite small, so that the precision of the calorimeter used was not sufficient to permit their evaluation.

4. The heat due to the solution of ZrH_2 in alpha zirconium may be detected calorimetrically even when the hydrogen content is as low as 28.5 ppm. The heat of solution is approximately 11,000 calories per mole of hydrogen.

5. There exists a metastable transition in a zirconium - 17.5 weight per cent niobium alloy below 520°C which has decided effects on the specific heat curves and on the hardness but cannot be identified by metallographic or x-ray investigations. This transformation is tentatively called the omega transition. A second metastable phase which occurs at lower temperature can be identified metallographically and with x-rays, but has little effect on the hardness or the specific heat.

6. An alloy containing 50 atomic per cent titanium with zirconium shows the expected specific heat behavior in the alpha phase, but the specific heat of the beta phase is definitely abnormal. The beta specific heat decreases linearly from 650°C to 800°C. This effect may be interpreted as a disordering of the beta phase. The transformation occurred over the range $550 - 640^{\circ}$ C on heating and $558.5 - 560.5^{\circ}$ C during cooling. The heat of transformation of 740 ± 20 calories per gram atom at 600°C is abnormally low.

CHAPTER II

INTRODUCTION

Structural materials for nuclear reactors must meet three stringent conditions: they must have adequate mechanical properties at the operating temperature, they must be corrosion resistant to the heat transfer medium (air, water, or alkali metals), and they must not absorb neutrons excessively. This last requirement is a particularly stringent one, for it is met by only four accessible metals: aluminum, magnesium, beryllium, and zirconium, as Table I shows.

For reactors which operate at low temperatures, aluminum serves as an excellent constructional material, but it cannot be used for more efficient high-temperature reactors, because of its low elevated temperature strength associated with its low melting point of $660\,^{\circ}$ C. Although magnesium has a lower absorption cross section than aluminum, it has a lower melting point ($650\,^{\circ}$ C) and poorer corrosion resistance, thus offering no particular advantages over aluminum. The use of beryllium is impractical except for very special applications, because of its extreme brittleness and the relative scarcity of its ores. In contrast, zirconium ores are abundant; the metal is fairly easily produced; it has satisfactory mechanical properties and excellent corrosion resistance at temperatures above the practical operating range for aluminum; and techniques of fabrication are now well developed. Thus zirconium comes at once to the foreground as a structural material for the construction of nuclear reactors.

TABLE	I
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Element	Atomic Weight	Absorption (barns)	Scattering (barns)
Beryllium	9.01	0.009	7.0
Magnesium	24.32	0.060	3.6
Aluminum	26.98	0.21	1.4
Titanium	47.90	5.6	4.0
Chromium	52.01	2.9	3.0
Iron	55.85	2.4	11.0
Nickel	58.69	4.5	17.5
Copper	63.54	3.6	7.2
Zirconium	91.22	0.18	8.0
Niobium	92.91	1.1	5.0
Molybdenum	95.95	2.4	7.0
Silver	107.88	60.0	6.0
Hafnium	1 7 8.6	115.0	

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MICROSCOPIC THERMAL-NEUTRON CROSS SECTIONS OF WHOLE ELEMENTS (1)

No one element can ever meet the insatiable demands of the design engineer, who is ever striving for operation at higher and higher temperatures, and zirconium is no exception. With respect to mechanical properties zirconium behaves as a low melting point element instead of the high melting metal it is, and its strength drops rapidly with increasing temperature such that it is no longer useful above about 350°C. By the addition of alloying elements the operating temperature may be increased to 500°C or possibly higher; hence an understanding of the alloying behavior of zirconium is absolutely essential to its projected use in a reactor system.

Zirconium is one of the Group IV transition elements. Other members of this group are titanium, which has a high strength to weight ratio, a property much valued for aircraft construction; and hafnium, whose high absorption cross section for neutrons makes it a useful reactor control rod construction material. From its position in the Periodic Table thorium would also be included in this group. In reality, however, the properties of this element are more closely related to those of the other members of the actinide series than to those of the Group IV transition metals.

The chemical similarity of the Group IV transition metals is surprising. All three metals form tetravalent compounds. Their oxides $(TiO_2, ZrO_2, and HfO_2)$, for instance, are among the most refractory materials. This chemical similarity between zirconium and hafnium is one of the deterrents to the use of zirconium in reactors, since its ores generally contain from one to three per cent hafnium. If a good

neutron economy in a reactor is to be secured, the high cross section hafnium must be separated from the zirconium. Although this separation is difficult, low hafnium zirconium is now being produced. That this similarity persists in the metallic state is intriguing to theoretical metallurgists and presents an opportunity to study the similarities and differences between alloys of various solutes in zirconium, hafnium, and titanium. Such studies should lead to a better understanding of alloy formation in general and specifically the importance of the Hume-Rothery factors: atom size, valence electron-atom ratios, and electronegative valency effect on the alloying behavior of the transition elements.

The physical metallurgy of zirconium and titanium is now well established. Less is known about hafnium. All three metals transform from a closed-packed hexagonal structure which is stable at low temperatures to a body-centered-cubic structure which is stable at higher temperatures. The former structure, according to the usual terminology, is called alpha, whereas the latter structure is denoted as beta. The lattice parameter of these phases for zirconium, titanium, and hafnium together with the transition temperatures and other physical constants are shown in Table II.

The existence of an allotropic transition in a pure metal with rising temperature may be understood in terms of the free energies of the the allotropic forms, the criterion for stability being that at constant temperature and pressure the phase is stable which has the lowest free energy. For a given phase, say alpha, the free energy F at a temperature T may be expressed by the equation

Property	Titanium	Zirconium	Hafnium
Crystal Structure:			
Alpha phase, c.p.h., A (25°C)			
a o c	2.950 4.683	3.232 5.147	3.186 5.037
c _o /a _o	1.587	1.593	1.581
Beta phase (b.c.c.) A a _o (900°)	3.306	3.61	-
Transition temperature	882	870	~1310
Melting point, °C	1795	1845	1975
Specific resistivity (25°C) ohm-cm x 10 ⁶	45	40	32
Young's modulus lb/in. ² x 10 ⁻⁶	16	13.7	20

PROPERTIES OF HAFNIUM, ZIRCONIUM, AND TITANIUM (2)

TABLE II

$$F_{\alpha} = H_{\alpha}^{\circ} - \int_{0}^{T} dT \int_{0}^{T} \frac{Cp}{T} dT$$
(1)

where $H_{o\alpha}^{O}$ is the enthalpy of the alpha phase at absolute zero and Cp_{α} is the specific heat at constant pressure. In the same manner the free energy of the beta phase may be expressed as

$$F_{\beta} = H_{\beta\beta}^{O} - \int_{O}^{T} \int_{O}^{T} \frac{Cp_{\beta}}{T} dT.$$
 (2)

The fact that beta becomes stable at an elevated temperature means that

$$/\int_{O}^{T} dT \int_{O}^{T} \frac{Cp}{T} dT /$$

increases more rapidly as the temperature increases than

$$\int_{O}^{T} \int_{O}^{T} \frac{Cp_{\alpha}}{T} dT/.$$

This observation is shown more clearly by Figure 1, which gives a plot of the alpha and beta phases as a function of the absolute temperature.



Figure 1. Hypothetical plot of the free energy of the α and β phases of zirconium as a function of temperature.

Thus at absolute zero $H_{O\beta}^{O} > H_{O\alpha}^{O}$ makes alpha the stable phase, but as the temperature increases F_{β} decreases more rapidly than F_{α} until at T_{c} , $F_{\beta} = F_{\alpha}$. At temperatures greater than T_{c} , $F_{\beta} < F_{\alpha}$ makes beta the stable phase. This more rapid decrease of the free energy of beta with increasing temperature may be due to a higher heat capacity of the beta phase or a higher Debye temperature for the alpha phase. If the alpha to beta transitions in pure titanium, zirconium and hafnium are to be understood, then accurate thermodynamic data on both the alpha and beta phases must be secured.

The earliest determinations of the specific heats of both titanium and zirconium were made by Jaeger and his associates (3) (4) using a dropping calorimeter with materials purified by the decomposition of zirconium and titanium iodide, a process which was developed by van Arkel and de Boer. The samples were sealed in platinum capsules and dropped into a copper block after they had reached equilibrium in a furnace at a predetermined temperature. The measured-temperature rise in the copper block was compared to the rise produced by known quantities of heat. The heat effect of the platinum was subtracted to give the enthalpy of zirconium and titanium as a function of temperature. The temperature derivative of this function is the specific heat. More recently Skinner (5) and Coughlin and King (6) have determined the specific heat and the heat of transformation of zirconium using the same method. Redmond and Lones (7) have also measured the specific heat of zirconium using an ice calorimeter. In an ice calorimeter the enthalpy decrease of the sample corresponds to a certain volume of ice which is

melted by dropping the sample into the receiving tube which is compared to the volume change produced by known quantities of heat. The specific heat and heat of transformation of titanium has been measured by Skinner, Johnston, and Beckett (8) using the same apparatus Skinner used for zirconium. The specific heats for zirconium, which have been reported by the above investigators are shown in Figure 2 together with the results of the present investigation. The temperatures and heats of transformation for zirconium, which were reported, are compared in Table III.

In the same fashion values for the specific heat and the heat of transformation of titanium are shown in Figure 3 and Table IV.

The wide variations in these results are due partly to impurity effects of oxygen, nitrogen, and hydrogen, and partly to inherent difficulties in making high temperature calorimetric measurements. All investigations show that the beta specific heats of both titanium and zirconium are lower than those of the alpha phase at the transition temperature, suggesting that a lower Debye temperature of the beta phase is responsible for greater decrease of the free energy of beta as the temperature is increased. The measurements of the heats of transformation of zirconium and titanium were integral parts of the measurements of the specific heats except for the value for titanium reported by Schofield (9). This value was calculated from measurements of the change in the time for the transformation at different cooling rates. The method used by Backhurst (10) for the determination of the heat of transformation of titanium has not been published.



Figure 2. The reported values of the specific heat of zirconium.

TABLE III

Investigator	Heat of Transformation cal/gram atom	Temperature of Transformation °C
Skinner (5)	1040	870
Coughlin and King (6)	920	837 - 917
This investigation	993 ± 25	870 ± 2

REPORTED VALUES OF THE HEAT OF TRANSFORMATION OF ZIRCONIUM



Figure 3. The reported values of the specific heat of titanium.

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Investigator	Heat of Transformation cal/gram/atom	Temperature of Transformation °C
Jaeger, Rosenbohm, and Font ey ne (4)	950	882
Skinner, Johnson, and Beckett (8)	943	883
Schofield (9)	813 ± 80	885 ± 2.5
Backhurst (10)	880 ± 44	-
This investigation	978 ± 25	883 ± 2

REPORTED VALUES OF THE HEAT OF TRANSFORMATION OF TITANIUM

TABLE IV

The alpha to beta transition in zirconium and titanium is of considerable practical interest. For example, certain alloy additions tend to stabilize the low-temperature form to higher temperatures thereby avoiding stress and dimensional changes produced by the transformation during fabrication or use. In other instances solutes may be added which will stabilize the high-temperature form at lower temperatures. These additions may also permit metastable phases or precipitation hardening to occur which lead to more favorable physical properties, as in the heat treatment of steel or aluminum alloys.

Because of the possible improvement of engineering properties of zirconium by the addition of alloys and its unique role in reactor design, an intensive research program has been carried out since 1950 under the auspices of the Atomic Energy Commission to determine the phase diagrams of various binary systems with zirconium. Compilations of this research have shown that zirconium-binary-alloy systems may be catalogued into five types (11):

1. Peritectic and Peritectoid (raising of the alpha-beta transition): Zr-Al, Zr-O, Zr-Sn, Zr-N, Zr-In.

2. Extended beta (depression of the alpha-beta transition): Zr-Nb, Zr-Ta, Zr-Th, Zr-U.

3. Eutectic and Eutectoid: Zr-Ag, Zr-Be, Zr-Co, Zr-Cr, Zr-Cu, Zr-Fe, Zr-Mn, Zr-Mo, Zr-Ni, Zr-V, Zr-W, and Zr-H.

4. Miscellaneous (including types 1, 2, and 3): Zr-C, Zr-Ge, Zr-Si.

5. Complete solid solubility: Zr-Hf, Zr-Ti.

These results indicate that in general the alpha-beta boundaries of zirconium alloys vary regularly with the position of the solute in the Periodic Table. Thus when zirconium is alloyed with transition element solutes to the right in the Periodic Table and including the elements of Group IB, the alpha-beta boundaries are depressed; whereas when the solute is a member of Groups IIIB, IVB, or VB, the alpha-beta boundaries are elevated.

Betterton and Frye (12) have expressed these systematic alloying effects in terms of the valency rule, and have shown that such a rule does apply for zirconium. The valency rule states that for those solutes which depress the alpha to beta transition the degree of lowering the transition temperature should be a linear function of the difference in valency between the solute and the solvent; and for those solutes which raise the transformation temperature the degree of raising of the transition temperature should also be a linear function of the difference in valency between the solute and the solvent. In addition positive valency differences should have an opposite effect to negative valency differences. Silver, for instance, which is usually considered to be monovalent, lowers the transformation temperature of zirconium, whereas indium, which is ordinarily given a valence of three, raises the transformation temperature. If it is assumed that these valences are correct, zirconium, according to the valency rule, must have a valence of two, and those elements with lower valency would produce eutectoids while those elements with a higher valence would form peritectoids. Dilute

solutions of divalent solutes would be expected to have little effect on the transition temperature unless size effects or electronegativity effects come into importance. The transition elements, chromium, iron, cobalt, and nickel would be assigned their usual valence of zero to agree with this scheme.

The sign of the valency difference is reversed, but the theory otherwise holds, if the valences for metals which have been derived by Pauling (13) are applied. Pauling states that the valence of silver is 5.56, indium is 3.56, and zirconium is 4. According to this valency theory those elements with a higher valence than zirconium form eutectoids and those elements which have a lower valence form peritectoids. The valence of six is assigned to the elements chromium, iron, cobalt, and nickel by Pauling. This valence gives agreement to the valency rule for these elements. Betterton and Frye have shown that with either set of valencies which one wishes to adopt, titanium alloys follow the same valency rule as zirconium if the difference in size between the zirconium and titanium atoms is taken into account, the titanium atom being roughly nine per cent smaller than that of zirconium.

It is possible to express the effects of solutes of different valences on the transformation temperature in terms of changes in the thermodynamic parameters of the alpha and beta phases. If one assumes that a given solute forms a regular solution in both the alpha and beta phases, he may write the free energies of the alpha and beta solutions, which have a solute mole fraction of X, as follows:

 $F_{\alpha} = H_{\alpha}^{O}(X) - \int_{O}^{T} dt \int_{O}^{T} Cp_{\alpha}(X) d \ln T + KT[X \ln X - (1-X)\ln(1-X)]$ (3)

and

$$F_{\beta} = H_{\beta}^{O}(X) - \int_{O}^{T} DT \int_{O}^{T} Cp_{\beta}(X) d \ln T + KT [X \ln X + (1-X) \ln (1-X)] .$$
(4)

A change in the mole fraction of a solute with a given valency would be expected to shift the enthalpies at absolute zero, $H^{O}_{\alpha}(X)$ and $H^{O}_{\beta}(X)$, and the integrated specific heat terms

$$\int_{0}^{T} dT \int_{0}^{T} Cp_{\alpha}(X) d \ln T$$

and

$$\int_{o}^{T} dT \int_{o}^{T} Cp_{\beta}(X) d \ln T.$$

A change in one or both of these parameters for the alpha phase produced by a change in X would be expected to be different from the change in . the beta phase. In addition higher valence solutes would be expected to shift these parameters in an opposite direction to those produced by lower valence solutes. Using the above expressions for the free energies of the alpha and beta phases, respectively, and the usual conditions of equilibrium, i.e.,

$$\frac{\partial F_{\alpha}}{\partial X_{\alpha}} = \frac{\partial F_{\beta}}{\partial X_{\beta}} = \frac{F_{\alpha} - F_{\beta}}{X_{\alpha} - X_{\beta}}$$
(5)

Betterton and Frye (12) derived the following expressions for the compositions of the phase boundaries, X_{α} and X_{β} , at a given temperature T:

$$X_{\alpha} = -\frac{e}{KT} + 1$$

$$-e^{-\frac{N}{KT}} + 1$$
(6)

21

and

$$x_{\beta} = \frac{e^{+\frac{L}{KT}} - 1}{e^{+\frac{N}{KT}} - 1}$$
(7)

$$L = \Sigma_{\alpha} - \Sigma_{\beta} - X_{\alpha}\dot{\Sigma}_{\alpha} + X_{\beta}\dot{\Sigma}_{\beta}$$
(8)

and

$$N = \dot{\Sigma}_{\beta} - \dot{\Sigma}_{\alpha} .$$
 (9)

In these expressions

$$\Sigma_{\alpha} \equiv H_{\alpha}^{O}(X_{\alpha}) - \int_{O}^{T} dT \int_{O}^{T} C_{\alpha}(X_{\alpha}) d \ln T, \qquad (10)$$

$$\Sigma_{\beta} \equiv H_{\beta}^{O}(X_{\beta}) - \int_{O}^{T} dT \int_{O}^{T} C_{\beta}(X_{\beta}) d \ln T, \qquad (11)$$

$$\Sigma_{\alpha} \Xi \frac{\partial \Sigma_{\alpha}}{\partial \Sigma_{\alpha}}$$
, (12)

and

$$\Sigma_{\beta} \equiv \frac{\partial \Sigma_{\beta}}{\partial x_{\beta}} , \qquad (13)$$

Because of a lack of thermodynamic data Betterton and Frye assumed that $L = \Delta F_{_{\rm O}} \tag{14}$

and

$$N = K_1 \bigtriangleup V - K_2 S. \tag{15}$$

In the above equations ΔF_0 is the free energy change for the pure solvent, ΔV is the difference in valence between the solute and the solvent, and S is the square of the difference between the solute and solvent atom diameters divided by the square of the diameter of the solvent atom. They then calculated values for N for different solutes from their phase diagrams and solved the constants for K_1 and K_2 by least squares. Using these values of K_1 and K_2 they recomputed the phase boundaries for the various solutes in zirconium and titanium and found a good correlation between the calculated and experimental $\alpha/\alpha + \beta$ and $\alpha + \beta/\beta$ boundaries. In addition the values of the constants K_1 and K_2 for titanium were found to be approximately the same as those for zirconium. Betterton and Frye suggested that a better correlation could be obtained if accurate thermodynamic data for the various alloys were available.

Because of this need for calorimetric data on zirconium and its alloys, and because of the wide scatter in the reported values of the thermodynamic functions of zirconium, the measurement of the specific heat and heats of transformation of zirconium and alloys of zirconium with silver, indium, hydrogen, niobium, and titanium was undertaken by the author at the University of Tennessee in January, 1956. As a corollary study, the measurement of the specific heat and heats of transformation of titanium was also undertaken. The zirconium-silver and zirconium-indium systems were selected for study because the phase diagrams have been very carefully worked out by Betterton et al (14) (15), and because they serve as typical representatives of lower and higher valence solutes in zirconium with corresponding reversed effects on the stability of the alpha and beta phases. Zirconium-hydrogen alloys were studied to explain anomalous peaks in the specific heat curve for zirconium which were observed during the course of the investigation. Alloys of zirconium with titanium and niobium were studied because in these systems the beta phase is suppressed to temperatures much lower

than 870°C and thus information can be obtained on the beta phase over a wider range of temperatures. All measurements were made in the University of Tennessee high-temperature adiabatic calorimeter developed by G. E. Elder (16), M. L. Picklesimer (17) and D. L. McElroy (18).

There are characteristic features of each alloy system studied, and the results of this investigation will be presented in eight sections. These are:

Section I.	General Information
Section II.	Zirconium
Section III.	Zirconium-Silver
Section IV.	Zirconium-Indium
Section V.	Zirconium-Hydrogen
Section VI.	Zirconium-Niobium
Section VII.	Titanium
Section VIII.	Zirconium-Titanium

Because the University of Tennessee calorimeter operates only above room temperature, these results will serve only as a partial thermodynamic analysis of the systems, and are not intended to be complete. For a complete analysis one needs calorimetric data down to absolute zero on both the basic components and the alloys, and data on the difference of free energies of alloys and their pure constituents such as those one obtains from vapor pressure or electromotive-forcecell measurements. These data should, however, be of much value to future studies on zirconium alloys and will illustrate the usefulness of calorimetric measurements on the study of alloy systems.

CHAPTER III

THE PREPARATION OF THE ALLOYS

It is difficult to prepare a homogeneous alloy of zirconium containing insufficient oxygen, nitrogen, or hydrogen to affect seriously its mechanical properties. For this reason the methods of preparing the alloys used in this research will be described in detail.

To prepare a high-purity alloy one must start with pure zirconium, which is itself a difficult material to produce since it is among the most reactive of metals at elevated temperatures, a property illustrated by its use as a getter in electronic tubes. Not only does zirconium react with oxygen, hydrogen, and nitrogen, but it also dissolves the reaction products rapidly above the following temperatures: oxygen -700°C, hydrogen - 300°C, and nitrogen - 1000°C (19). The resulting solutions may be quite brittle and fabricated only with difficulty. For this reason zirconium must always be melted and heat treated in extremely inert atmospheres.

Zirconium of sufficient purity for most processes is produced by the Kroll process (20). In this process zirconium oxide (ZrO_2) , which has been highly purified, is reacted with carbon to produce zirconium carbide. This carbide is chlorinated to produce zirconium tetrachloride $(ZrCl_4)$ which is then reacted with molten magnesium under an inert atmosphere to produce metallic zirconium and magnesium chloride. The chloride is removed by washing and any excess magnesium is recovered by a distillation process. The resulting zirconium is a porous-spongy

mass which is then remelted in suitable furnaces into a useful form. The current price of low hafnium sponge zirconium is about \$15 per pound.

For reactor applications the stipulation of low hafnium is made because the normal hafnium content in zirconium ores, if allowed to remain, would make the neutron absorption cross section of the final sponge prohibitive. Because the chemical properties of zirconium and hafnium are almost identical, the removal of the one to three per cent hafnium is quite difficult. The separation is permitted, however, by the slightly different solubilities of hafnium and zirconium salts such as $(NH_4)_2HfF_6$ and $(NH_4)_2ZrF_6$ in dilute hydrofluoric acid solutions. By a continuous fractional crystallization process with about 10 stages, a salt with a hafnium-zirconium ratio of less than 0.001 may be produced (21).

If a product of greater purity than sponge is required, as in fundamental research, zirconium (or hafnium or titanium) may be purified by the van Arkel and de Boer iodide decomposition process (22). In this process the zirconium sponge is reacted with iodine at 450°C to produce zirconium iodide (ZrI_4) vapor. This compound is then decomposed by and deposited on a filament of zirconium at a temperature of 1200°C. The resulting zirconium is in the form of a dense rod which is low in total impurities and is ready to fabricate.

The high reactivity of zirconium at high temperatures also makes its fabrication difficult. Melting and casting, especially, are difficult processes to carry out without contaminating the metal. There

is no crucible material which is inert to liquid zirčonium, for it readily reduces oxides, etc., and dissolves up to 0.4 per cent carbon from graphite. Carbon, like oxygen, embrittles the zirconium. Arc melting of zirconium in a water-cooled copper crucible has proved to be the most satisfactory method. The conductivity of copper is so much higher than that of zirconium that the first layer of zirconium to solidify on the copper surface acts as a layer of insulation to the metal adjacent to it, so that zirconium is actually melted in a zirconium crucible. In order to have molten zirconium (melting point 1845°C) in contact with solid copper a method of generating a large quantity of heat in a confined area is essential. This is done by utilizing an electric arc either from an inert tungsten electrode or a consumable electrode of zirconium itself. Alloys are made by melting the alloy and the zirconium in the arc at the same time.

The materials used in this research were iodide zirconium obtained from the Westinghouse Atomic Power Laboratory and iodide titanium purchased from the Foote Mineral Company. The compositions of the alloys produced from them are shown in Table V. The iodide zirconium had been deposited in the form of a one-inch diameter rod, and the calorimetric specimens of dimensions two inches long and seven-eighths inch diameter were machined directly from the deposited material. Since the as deposited titanium was only a half inch spiny rod, it required remelting to produce an ingot large enough for a calorimeter specimen. The melting of the titanium was effected by the author at the Oak Ridge National Laboratory in a consumable elèctrode arc furnace.
Property	Iodide	Zirconium	Zr + 0.881	w/o Ag	Zr + 5.	37 w/o Ag	Zr + 7.	77 w/o In	Iodide Tita	anium	Zr + 17.5 w/o Nb	Zr + 34.4 w/o Ti
DPH Hardness ^{1,2}	66.5		95.7		162		147		79.2		194	366
Solute Content	Wet	Activation	Ingot Top	Ingot Bottom	Ingot Top	Ingot Bottom	Ingot Top	Ingot Bottom	Wet	Spectro- graphic		
Al ppm	4	l								30		
Ag w/o			0.879	0.883	5.44	5.30						
Ca ppm	6									trace		
Co ppm	1	0.5								+		
Cr ppm	1	23	ho	60	20	10			10	trace		
Сарры Беррт	130	186	360	330	280	300	210		300	50		
Hf ppm	70	162	500	500	200	500	LIU		500)0		
In w/o	1-	200					7.77					
K ppm	40											
Mg ppm	5									100		
Mo ppm	3.5	1.9										
Na ppm	30	1.5										
Ni ppm	30	49								trace	ND - 17.5 W/O	
Pb ppm	18	6								500		÷
Si ppm	10	0)00		
יייס אומין דייט דיין דייש	2								Remainder			34.4 4/0
W ppm	18	2.4							,			5
C ppm 3	220		140	100	130	320	67		320			
H ppm 1	35		4.4	4.0	11	67	5.1		67			
N ppm	75		8.6	4.6	4.9	7.7	30		7.7			
O ppm ⁴	150		150	180	220	110	160		110			

TABLE V

1. 10 Kg load
 2. All specimens furnace cooled
 3. Direct combustion
 4. Vacuum fusion

To prepare the silver-zirconium and the indium-zirconium alloys, plates of iodide zirconium, again obtained from Westinghouse Atomic Power Division, were pressed and swaged into tubes. After a thorough cleaning with hydrofluoric acid, water, and alcohol, three short lengths of tube were welded together in a dry box to form a pipe of sufficient length for use in the consumable electrode arc furnace. Wires of 99.999 per cent indium or silver of the proper diameter to produce the desired composition were then inserted inside the tube along its full length. The tubes were then consumably melted to produce the desired alloy. Although the original intention was to remelt the resulting ingots, results of hardness measurements, metallographic and chemical analyses indicated that the alloys were homogeneous after a single pass. Finally the alloys were homogenized in a vacuum of 10⁻⁶ mm of mercury for two weeks at 1300°C. Since the hardness is a good measure of the quality of the zirconium alloy, the hardness values of the pure metals and the alloys are also shown in Table V.

Because of the fact that only a small quantity of 99.9 per cent niobium was available, the zirconium-niobium alloy used in this investigation was made in a slightly different way. Enough zirconium and niobium were weighed out quantitatively to produce 200 grams of a 17.5 weight per cent niobium alloy. These ingredients were then placed in a copper hearth, tungsten electrode arc furnace and melted by J. B. Flynn at Oak Ridge National Laboratory. The resulting ingot was removed from the furnace, cut into small pieces and remelted twice. This care was taken to improve the homogeneity of the final alloy. Due to the limited size of this "button" melting assembly, the final calorimeter specimen after being swaged to a round cross section was about a half inch in diameter. This, however, was of sufficient size to make an adequate calorimeter specimen. Since neither component of this alloy is volatile the final composition was assumed to be the same as the initial charge. An alloy of zirconium with 34.4 weight per cent titanium was produced in the same way from the iodide materials.

The other specimen used in this investigation, the zirconiumhydrogen alloy, was produced by heating a zirconium calorimeter specimen in a rarefied atmosphere of hydrogen. The actual pressure of the system was uncertain and unfortunately no chemical analyses were made so that the actual hydrogen content was unknown. There are indications which are discussed later that the final hydrogen content was about 300 ppm.

The chemical analyses of the alloys produced for this investigation were made by the Analytical Chemistry Division of Oak Ridge National Laboratory. These methods included vacuum fusion for hydrogen, oxygen and nitrogen, direct combustion for carbon, flame spectrophotometry for indium, and the chloride precipitation of silver. Other analyses on the original zirconium and titanium were supplied by Westinghouse Atomic Power Division and the Foote Mineral Company. A complete resume on the analytical methods used by Westinghouse Atomic Power Division has been given by Read (23).

CHAPTER IV

DESCRIPTION OF THE CALORIMETER

The calorimeter used in this investigation was a high-temperatureadiabatic unit designed and described in detail by D. L. McElroy (18). McElroy's design incorporated improvements over previous calorimeters built at the University of Tennessee by M. L. Picklesimer (17) and G. E. Elder (16) under the guidance of E. E. Stansbury.

The original objective of the calorimeter program at the University of Tennessee was to utilize modern high-precision controllers such as the Leeds and Northrup Speedomax G and DAT 50 unit as the control mechanisms for a high-temperature adiabatic calorimeter which would operate continuously from room temperature to 1000°C. The underlying principle for such a calorimeter had been proposed previously by Quinney and Taylor (24) and Sykes and Jones (25). This principle is to install a heater inside the specimen to be studied in such a way that all the power dissipated by the heater will be absorbed by the specimen. The \cdot specimen with its heater is then placed in a furnace of special design which serves as an adiabatic shield and whose heating circuit is fed by an independent power supply. The power input to the shield is controlled by a differential thermocpuple between the specimen and the shield so that no temperature difference between the two will exist. With this arrangement one measures the electrical energy dissipated in the specimen heater when the temperature of a specimen of known mass is raised a certain number of degrees. The heat capacity is then computed from the

following equation:

$$Cp = \frac{EI\Delta t'f}{m\Delta T} .$$
 (16)

In this expression Cp is the heat capacity, E is the voltage across the specimen heater, I is the heater current, Δt is the time required for the specimen to rise ΔT degrees, m is the mass of the specimen, and f is a factor which converts electrical units to heat units. The heat capacity thus computed is the average heat capacity over the temperature range included in ΔT .

The primary application of this type of calorimeter is to measure the specific heats of materials at elevated temperatures. It may, however, also be used to measure the heat of transformation of a substance from one allotropic form to another, the heat of fusion, the heat involved in an order-disorder reaction, or the heat associated with the demagnetization of a ferromagnetic material. By the addition of separate thermocouples to the specimen and to the adiabatic shield, the temperature of the shield may be held constant while one varies the temperature of the specimen. Such an arrangement permits the use of the instrument as an isothermal calorimeter, to measure the heat released during the isothermal transformation of an alloy below the critical temperature. Finally, by controlling the power to the shield in such a manner that a given deflection of the galvanometer is maintained, one can induce in a specimen any cooling rate which is desired below the maximum rate obtained by turning the power off and thus operate the calorimeter as a precise thermal analysis unit. Thus, an adiabatic

calorimeter appears to be a simple instrument with a variety of quite useful applications.

Unfortunately the design and operation of a high-precision, hightemperature adiabatic calorimeter entails a number of discouraging problems. These problems may be listed as follows:

1. The design of a truly adiabatic shield.

2. The response of the adiabatic shield.

3. The inaccuracies of thermocouples.

4. The instability of thermocouples.

5. The conduction of ceramics at elevated temperatures.

6. The vaporization of metals at elevated temperatures.

7. The emission of electrons from heaters at elevated temperatures.

8. The low thermal conductivity of high strength elevated temperature materials.

9. The corrosion of materials at elevated temperatures.

10. The thermal stresses.

Each of these difficulties places a separate requirement on the design and the simultaneous fulfillment of all the requirements changes the calorimeter from a simple device in principle to an intricate network of wires, pipes, pumps, and electronic apparatus.

The calorimeter which McElroy designed to optimize these diverse requirements is shown schematically in Figure 4. A smaller, but otherwise identical calorimeter designed to operate over a smaller temperature



Figure 4. Schematic representation of the calorimeter system.

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range utilizing the same control system as the McElroy calorimeter has been described in detail by Pawel (26). Since the wiring diagrams for the control and detecting circuitry have been thus described, only a general description to explain the schematic will be given here.

The material for the adiabatic shield consisted of a tube of monel with a one and one-half inch internal diameter which was thirtyeight inches long. Monel was selected because of its relatively high thermal conductivity and freedom from heat capacity anomalies. The long tube length was used because it was felt that a very long tube evenly wound with a resistance wire would result in a longer constant temperature section than would a shorter tube. To reduce heat transfer between the specimen and the wall a vacuum of approximately 0.01 micron of Hg was maintained on the inside of the tube. A helium atmosphere was maintained on the outside of the tube to prevent oxidation of the monel and to even out temperature gradients along the tube. On the outside of the helium chamber was a layer of insulation to reduce the power requirements for the system. The heating elements of the shield consisted of nichrome and were insulated with high purity Alundum beads. Semicircular-spiral grooves of the same diameter as the beads were machined into the outer diameter of the tube to hold the heating elements in place and to reduce the mass of the shield for more rapid response.

Four separate control circuits were utilized to reduce temperature gradients in the shield. These control units were the specimen cylinder control, the upper guard control G_1 , the lower guard control G_2 , and the inner guard control G_3 . The specimen cylinder control

maintained the surface of the inner wall of the tube adjacent to the specimen at the same temperature as that of the center of the outer surface of the specimen. G_1 and G_2 kept the temperature of the upper and lower ends, respectively, of the inner surface of the tube the same as that of the center of the inner surface of the tube. G_3 prevented the top surface of the specimen from radiating heat to the cover of the calorimeter through the end of the tube.

The control units for these systems were all null balance Speedomax G's with DAT 50 (pulse type proportioning control) except for the G₃ which was controlled by a Liston Becker unit. The sensitivity of the Speedomax units was increased through the use of a Leeds and Northrup D. C. Microvolt Amplifier on each circuit. The resulting sensitivity was such that one small division on the Speedomax G chart represented 0.1 microvolt. Except at high temperatures control deviations were no greater than one-tenth of one small division even when heating rates in the calorimeter were as high as 7 degrees a minute.

The thermocouples used for the control circuitry were platinum/ platinum-13 per cent rhodium which had been tested for thermals in a special temperature gradient furnace described by McElroy (18). These thermocouples, which were selected for their high stability at the expense of a high emf, were carefully installed to insure a minimum of cold working. Rubber gloves were worn by those who installed thermocouples to prevent contamination of the thermocouple wires or the highpurity Alundum insulation. The thermocouples were attached to the specimen and to the cylinder by spot welding the thermocouples to tabs

of platinum or nichrome for the cylinder and titanium or zirconium for the specimen and then spot welding the tabs to the respective components. The necessity for special care with thermocouples and the fabrication of the special tabs that gave good thermal contact and no electrical contact required for the control of the auxiliary heating circuits has been described by Pawel.

The specimens, which were suspended in the tube by the heater leads for a minimum contact between the specimen and the surroundings, were cylinders two inches long and seven-eighths inch in diameter which were machined to receive a special heater of nichrome insulated with high-purity Alundum. A standard zirconium specimen and heater are shown in Figure 5, together with the smaller zirconium-niobium specimen which was cast in the button-type arc furnace. Attached to the heater directly at the top of the specimen are leads for measuring the voltage drop across the heater. Also in the heater circuit is a standard one ohm resistance which, by measurement of the voltage drop across it, permits determination of the heater current.

Since a very uniform rate of temperature increase is required for energy input measurement and successful controller performance, the power source to the heater circuit must be very stable. This stability was obtained in the present calorimeter by using a heavy duty 12 volt automobile storage battery as the source of power. By varying the voltage applied to the specimen circuit by the two volt intervals which were available directly from the battery, different heating rates for a given specimen were obtainable.



Figure 5. Typical calorimeter specimens and a specimen heater. Note particularly the metal deposit on the ceramic of the specimen heater. The white areas were covered with powdered alumina. The surface roughness produced by several cycles through the alpha-beta transition may be seen on the iodide-zirconium specimen on the right. The roughness of the surface of the zirconium-niobium sample in the center is due primarily to the swaging operation which produced a circular cross section. The specimen heater circuit represents one of the critical problems limiting the precision of the data. Two contradictory requirements must be met by the specimen heater, good thermal contact to facilitate the heat transfer from the heater to the specimen, and poor electrical contact so that no current can flow from the heater circuit to ground through the differential thermocouple between the specimen and the shield. A secondary requirement is that the mass of the heater should be a minimum because a correction must be made for the specific heat of the heater. Difficulties arise due to decrease in the electrical resistance of ceramics at high temperatures and surface conduction due to the deposition of vaporized metal atoms onto the ceramics. This deposition was found to be particularly high with zirconium. For this reason, powdered Al_20_3 was pressed into the oavity between the heater and the specimen. High-purity aluminum oxide has been found to be the best available ceramic for high-temperature heater insulation.

Another problem associated with the specimen heater is the emission of electrons from the very hot heater filament. These electrons collect on the specimen and flow to ground through the differential thermocouple in the same way as leakage current mentioned above. Such a current flow produces an emf between the junctions of the differential thermocouple and affects the control system the same way as a temperature difference. Since the control system reacts to induce zero potential between the specimen and the shield, a temperature difference equal electrically and opposite to the leakage current results. If this

effect is not compensated for, the measured heat capacity progressively deviates from the true value as the temperature increases. Compensation for this effect, which generally first appears at about 600°C, is made by the thermal compensation circuit described by Pawel, which is essentially a system for placing a biasing voltage between heater circuit and ground. The error potential across the platinum-13 per cent rhodium leads from the specimen and the shield thermocouple was monitored continually during the course of a run and the bias was adjusted to maintain zero potential. This essential part of the specimen heater circuit was omitted from the schematic diagram for the sake of simplicity.

The operation of the calorimeter was carried out by bringing each component of the calorimeter into thermal equilibrium with the specimen and then applying a d.c. voltage to the specimen. The drive mechanism for each circuit was adjusted and the unit allowed again to attain dynamic equilibrium. Ordinarily the temperature of the specimen had reached 60 degrees Celsius by the time the dynamic equilibrium was established. A potential corresponding to a given temperature was set on a Rubicon B potentiometer used to read the specimen thermocouple. The instant the deflection of the galvanometer was zero, one of a pair of precision chronographs was started. The Rubicon B potentiometer was then set at an emf corresponding to a higher temperature. When the galvanometer was again zero, the first chronograph was stopped and the second was started. Immediately after starting a chronograph the voltage and current to the specimen was read with the same potentiometer. The measurements consisted of the time Δt required for the specimen of mass

m to rise AT degrees when the power applied to the heater was the EI product. The heat capacity was then computed from Equation 16. Between readings the four separate control systems were checked and the bias to the heater circuit was adjusted. The temperature intervals were selected so that one set of data was taken approximately every five minutes. The duration of a run from room temperature to 950°C required from four to ten hours. Temperatures were taken from the tables of emf versus temperature for platinum/platinum-13 per cent rhodium derived by Picklesimer (17), which are smoothed values of the table published by the Bureau of Standards (NBS Bulletin No. 508 or 561).

As one would expect, certain operational troubles are sure to arise during the operation of a device as complicated as this calorimeter, especially at high temperatures. A few of the more serious problems encountered are listed as follows:

1. Burn out of a shield heater circuit.

2. Power failures.

3. Breaking of thermocouples inside the calorimeter due to thermal stress.

4. Controller drifts or failure.

5. Failure of the specimen heater.

6. Loose connections.

7. Failure of variac drive mechanisms.

One or a combination of these troubles resulted in approximately half of the total number of the runs made being discarded. The first twelve runs were discarded because of the inability of an inexperienced operator to get good control at the higher temperatures where emf pickup occurred.

CHAPTER V

EXPERIMENTAL DATA AND RESULTS

1. General Information

In the course of the present investigation ten specimens were studied. These specimens consisted of two iodide-zirconium samples, two alloys of zirconium - 0.881 weight per cent silver, a sample of zirconium - 5.37 weight per cent silver, a sample of zirconium - 7.77 weight per cent fitanium, a sample of zirconium - 300 ppm hydrogen, and a sample of iodide titanium. Details on the weights of these specimens and their heaters together with information on the successful runs made with each specimen are given in Table VI.

The ratio of the heater weight to the specimen weight is a very important quantity in the determination of the specific heat of the material, for in addition to supplying the heat to raise the temperature of the specimen, the heater must also supply the energy required to heat itself and whatever insulation it is mounted on. If one defines the total specific heat of a specimen and its heater to be $C_{\rm T}$, the specific heat of the specimen alone to be $C_{\rm S}$ and the specific heat of the heater alone to be $C_{\rm H}$, and assumes that the temperature of the heater is the same as that of the specimen at all times, then a simple heat balance will yield the relation

$$C_{S} = C_{T} + \frac{W_{H}}{W_{G}} (C_{T} - C_{H})$$
(17)

where $W_{\rm H}/W_{\rm S}$ is the ratio of the heater weight to that of the specimen.

TABLE VI

					#	
Specimen	Atomic Per Cent Solute	Mean Atomic Weight	Specimen Weight - Grams	Heater Weight - Grams	Total Weight - Grams	Runs Used
Iodide Zirconium A	0.00	91.22	120.5731	2.4777	123.0508	17,18,24,26,28
Iodide Zirconium B	0.00	91.22	112.3564	3.2612	115.6176	60,62,63,64
Zirconium-0.881 w/o Silver A	0.746 Ag	91.34	122.0947	2.5221	124.6168	13,14,15
Zirconium-0.881 w/o Silver B	0.746 Ag	92.34	118.6492	2.6761	121.3253	50
Zirconium-5.37 w/o Silver	4.58 Ag	92.11	122.6848	2.4728	125.1576	32,33,36,37
Zirconium-7.77 w/o Indium	6.27 In	93.05	122.1519	3.4750	125.6269	54,55,56,58,59
Zirconium-17.5 w/o Niobium	17.24 Nb	91.50	81.6971	3.8345	85.5316	67,68,69,70,77,78
Zirconium-34.4 w/o Titanium	50.00 Ti	69.56	98.1422	3.2193	101.3615	43,45,46,47,48,49,79
Zirconium-0.03 w/o Hydrogen	2.75 H	88.74	109.5424	3.5232	113.0656	71,72,73,74,75
Iodide Titanium	0.00	47.90	82.6217	2.2572	84.8789	38,39,40,41,42

DESCRIPTION OF CALORIMETER SPECIMENS

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Since a larger fraction of the total weight of the heater is Alundum (Al_2O_3) , whose specific heat is approximately 0.27 cal/gram °C, and the specific heat of the metals being studied are in the range of 0.06 - 0.16 cal/gram °C, the difference $(C_S - C_H)$ is not a negligible quantity. Thus the smaller is the ratio W_H/W_S , the lower is the heater correction. Actually the temperature of the heater must be much higher than that of the specimen to produce the experimental heating rates so that equation (17) does not apply. By assuming that only radiant heat transfer from the heater to the specimen occurred, and that the heat transferred was independent of temperature, Pawel derived the expression:

$$C_{\rm g} = C_{\rm T} + \frac{W_{\rm H}}{W_{\rm g}} (C_{\rm T} - 0.55 C_{\rm H}) \cdot$$
 (18).

The heater correction is reduced somewhat in the expression. In the present case the alumina, which was packed around the heater to prevent the deposition of zirconium on the surface of the heater ceramic, probably acted somewhat as an insulation, making the temperature of the heater even higher. Since, however, the alumina also inhibited radiant heat transfer to some extent, the heat transfer process is difficult to describe quantitatively and a simple analytic expression for the correction could not be derived. If the temperature of the heater were higher than when the heat transfer was by a radiant mechanism, the heater correction would probably be smaller than Pawel's estimate. For lack of a better value, Pawel's expression was assumed to apply, and although the relative weights of heater wire and Alundum varied from specimen to specimen, it was assumed that seventy-five per cent of the weight of the heater was Alundum. This assumption greatly simplified the heater correction calculations and introduced less that 0.25 per cent error in the final specific heat.

The fact that the temperature of the heater is higher than that of the specimen introduces another important consideration. This is that in an adiabatic calorimeter which utilizes an internally heated specimen, heat flows into the specimen, but not out of it. The heat flow can only occur if there is a temperature gradient so that there must be a temperature gradient in the specimen. That this temperature gradient exists makes any heat capacity measurement an average value rather than an absolute one. This is relatively unimportant in the present work since data were taken over twenty or forty degree intervals and an average value was computed. The temperature gradient, however, also causes the inside of the specimen to reach a transformation temperature before the outside. For this reason one would expect a transformation to appear to occur over a range of temperatures even when it would otherwise occur at a single point.

The temperature gradient may be estimated, if one assumes that the specimen consists of an infinite tube of outer radius a and inner radius b. If every point in the specimen is rising in temperature at the same rate, and this rate is constant, the solution to the Fourier conduction equation for adiabatic conditions has been shown by Awberry and Snow (27) to be

$$\theta_{\rm b} - \theta_{\rm a} = \frac{bQ}{K(a^2 - b^2)} \left[a^2 \ln \frac{a}{b} - \frac{a^2 - b^2}{2}\right]$$
 (19)

where Q is the heat input to the specimen per unit area of inner surface, K is the thermal conductivity of the specimen and θ is the temperature at a given radius. The rate of temperature rise n of any point in the specimen is

$$n = \frac{2h^2 bQ}{K (a^2 - b^2)}$$
(20)

where h = K/Cp is the thermal diffusivity of the specimen, Cp is the specific heat of the specimen, and ρ is the density. If one uses the data of Bing et al (28) for the thermal conductivity of zirconium (0.187 watts/cm deg C at 300°C), measures a (1.1 cm) and b(0.3 cm) and assumes 5 watts are dissipated over a length of 5.3 centimeters, the temperature difference across the sample is calculated to be approximately one degree. The corresponding rate of heating is 41 degrees in five minutes, a value that is close to the observed heating rate for this power input. Since most specimens were run with lower power, one degree represents a maximum temperature gradient. The lower thermal conductivity of the alloys would raise this temperature gradient perhaps to as high as three degrees. For the button-type specimens (Zr-Nb and Zr-Ti), however, the decreased value of the outer radius would result in a smaller temperature gradient.

In the above discussion it was assumed that the heat transfer at the outer surface of the specimen was zero. This would only be true if the shield were ideally adiabatic, a condition that is most difficult if not impossible to obtain except over a very limited temperature range. If there are any temperature gradients along the surface of the tube which serves as a shield, these will result in a heat loss

or gain to the specimen and introduce an error in the measurement of the specific heat. In order to determine how near to the ideal adiabatic condition he is operating at a given temperature one may turn off the specimen power and observe how the temperature of the specimen changes or "drifts" as a function of time. Another way one may determine the adiabaticity is to measure the specific heat of a material at a series of different heating rates and compare the resulting values. Only if the calorimeter is truly adiabatic will the curves superimpose. Otherwise the heat capacity will be higher the slower the heating rate if the specimen is losing heat and lower the slower the heating rate if the specimen is absorbing heat from the surroundings. Pawel (26) showed that two ways of determining the nonadiabaticity were comparable, so that only the easier of the two measurements, the drift rate, need be made to correct for nonadiabatic conditions.

When drift rates are measured in the McElroy calorimeter, a very odd effect is found. If the measurements are made on a given specimen at a given temperature after successive runs, they are quite reproducible.. If, on the other hand, the specimen is removed, and then returned to the calorimeter, one finds that he can no longer (duplicate the drift measurement. If the control thermocouples are moved and rewelded, again the drift rates are not reproducible with previously determined values. These observations indicate that the relative adiabaticity of a given specimen in the calorimeter is very strongly dependent on its exact position in the tube and the exact position of each controlling thermocouple. If one is very careful to

insure that the precise position is repeated each time a specimen is replaced in the calorimeter, this effect is lessened. When a drift rate is measured in the calorimeter, one has three choices. The first, a rather impractical one, is to disassemble the calorimeter and redesign it. The second is to offset one or more of the control units to the calorimeter so that the drift will be decreased. The third and usual choice is to allow the shield to be nonadiabatic and then correct the data by measuring the drift rates of the individual specimen at a series of temperatures. Unfortunately it takes an hour or so for the calorimeter to reach its equilibrium drift rate once the specimen power is turned off, so that the measurement of the drift rate is a time consuming task. In the present work the upper guard heater was offset to reduce the drift so that a smaller correction would be needed, but otherwise method three was utilized.

During the course of the investigation certain accidents such as breaking of thermocouple welds caused the adiabaticity in the calorimeter to change, so that different drift rates were found for different sets of runs. These drift rates even went from large positive values to large negative ones. The drift rates which were used to correct the data are shown in Figure 6.

The drift rate correction is found by a simple heat balance to be

$$Cp_{T} = \frac{Cp_{A}}{1 - \frac{\Delta t}{NT} \frac{R(T)}{3600}}$$
(21)

where CpA is the apparent total specific heat (EI Δ t'f/(m_s + m_H) Δ T),



Figure 6. Values of the measured drift rates.

 Cp_T is the total corrected specific heat (specimen plus heater), $\Delta t/\Delta T$ is the reciprocal of the heating rate sec/deg., and R(T) is the drift rate with no power input, deg./hr. The S and H refer to the specimen and heater, respectively. The actual correction factor is the dimensionless term $\frac{\Delta t}{\Delta T} \frac{R(T)}{3600}$ which will subsequently be called ϵ . This correction is smaller—the greater one makes the reciprocal heating rate $\Delta t/\Delta T$, so that for a minimal correction one would use a large power input to the specimen. Sample calculations illustrating the drift and heater corrections are given in Table VIII of the Appendix.

Such a large power input, while decreasing ϵ , has inherent disadvantages. The first is that the specimen is always further from equilibrium at high heating rates. For if a transformation occurs, it would tend to occur at a higher temperature at a higher heating rate. A second disadvantage is the increased difficulty in attaining precision control on the adiabatic shield. Since the temperature of the specimen is changing more rapidly at the high heating rates, it becomes more difficult for the control system to respond and maintain control to within 0.01°C. When the specimen temperature levels out due to a transformation, there is a much greater tendency for the control system to overshoot. This requires that at high temperatures the operator must anticipate a transformation and assist the control system by reducing the power to the shield windings when a transformation occurs. In an analogous fashion the operator must sense the end of a transformation and increase the power as the specimen begins a more rapid temperature increase. One must balance the opposing desires for a low

 ε and high precision control by selecting an optimum range of power the input levels for each specimen.

In the plot of the drift rate as a function of temperature it may be observed that the data are given in terms of a set of runs and not in terms of data on each specimen. It is possible to correlate drift data for different specimens, if they are carefully placed in exactly the same depth in the calorimeter and if it is assumed that the function MCpR(T) is a constant. For specimens with different outer diameters the function which is constant is $\frac{M Cp R(T)}{A}$, where A is the external surface area of the specimen.

The use of the drift rate correction is very restricted in the temperature range where a transformation occurs, because the drift is immeasurably small during a transformation. In order to correct the measured enthalpy data taken during a transformation, one must rely on some type of extrapolation from the drift in the all-alpha or allbeta regions. In this work the corrections on the transformation data were made by assuming that ϵ varied linearly over the range of transformation temperatures. The value of the enthalpy over a given temperature range in a region of transformation is computed from the equation

$$\Delta H = \frac{\Delta H \text{ measured}}{1 - \overline{\epsilon}}$$
(22)

where $\overline{\epsilon}$ is the average value of ϵ over the temperature range associated with ΔH . The heats of transformation are then determined graphically by measuring the height of the discontinuous jump in ΔH . The use of the assumption of linearity of ϵ causes no greater scatter in the heat of transformation data than one finds in the specific heat data where R(T) is known.

The fact that there is scatter in the specific heat data amounting to two per cent at low temperatures and three to five per cent at high temperatures indicates that the heat loss is a function of the heating rate as well as the time interval. A method of measuring the effect of the heating rate on the nonadiabaticity has not yet been developed. The scatter that is produced in the data is much higher than one would expect from errors in reading $\triangle t$, $\triangle T$, E or I as discussed by Pawel. Thus the scatter appears to be due to loss of precise control at high temperatures and spurious emfs resulting from contaminated thermocouple protection tubes. As a specimen was rerun more and more, the ability to maintain precise control became progressively more difficult. Another effect of the progressive contamination was the loss of precision at low temperatures. Thus if the calorimeter had never been operated above 600°C a precision of about 0.5 per cent could be expected from the results of Pawel. After a few runs to 900 degrees with zirconium alloys the precision at low temperatures was decreased to about 2 per cent. Since the characteristics of the calorimeter change completely when the thermocouples are replaced, and the calibration runs would serve to contaminate the system, one is forced to accept the lowered precision until some type of specimen shielding to prevent its seeing the control thermocouples can be devised. Although the absolute errors are difficult to estimate, it is believed that the absolute accuracy is within the 2-5 per cent spread of the calculated values.

2. Iodide Zirconium

The corrected values of the specific heat of iodide zirconium for nine successful runs are shown in Figure 7. The data are also given in tabular form in Table IX of the Appendix. Included in the data are runs with high and low heating rates for runs with both high positive drift rates (Runs 17, 18, 24, 26, 28) and high negative ones (60, 62, 63, 64); thus any consistent errors in the first set would be expected to be opposite to those of the latter. Although even in the alpha region the spread in the data is high, several runs agreed within one per cent in the alpha region. The curve which is believed to be the best value was the set of mean points between these closely agreeing curves.

In the beta region the spread of the data is such that it is difficult to assign a best value with any confidence. For this reason an approximate mean value was adopted. Although the specific heat seems to increase with increasing temperature in the beta region, a constant value was assumed to apply for the specific heat over the range studied. That this value should be constant is indicated by the results on the zirconium - 17.5 weight per cent niobium alloy discussed in Section V. In any case, the specific heat of the beta phase is considerably lower than that of the alpha at the transformation temperature. This result agrees with the findings of Coughlin and King (6) and Skinner (5) as shown in Figure 2.

Although the specific heat values for alpha zirconium are somewhat lower than those reported by the previous investigations, they do



Figure 7. The measured values of the specific heat of zirconium.

indicate an abnormal rise in the specific heat with increasing temperature as previously reported.

The peak in the alpha specific heat curve in the neighborhood of 240°C will be shown to be due to the solution of zirconium hydride in the alpha zirconium in Section VII.

The plot of the measured enthalpies of zirconium above 800° C is shown in Figure 8. These data are also given in Table XVII of the Appendix. For comparison the enthalpy values of zirconium-silver alloys, a zirconium-indium alloy, and titanium are included in the same figure. The zirconium data indicate that the transformation of iodide zirconium is $870 \pm 2^{\circ}$ C and that the heat of transformation of zirconium is $993 \pm$ 25 calories per gram atom. This value is compared to previously reported results in Table III.

It should be noted that there is an abnormal rise in the enthalpy curve for zirconium above 850°C. This rise is believed to be due to the iron contamination of approximately 150 parts per million. Betterton (13) showed that the purest obtainable iodide zirconium has two phases in this range of temperatures.

3. Zirconium-Silver Alloys

The zirconium-silver phase diagram has been thoroughly studied by Betterton and co-workers (14). The resulting diagram, which is shown in Figure 9, shows that silver lowers the temperature at which the beta phase is stable to 821°C, the eutectoid composition being 3.76 atomic



Figure 8. The measured values of the transformation enthalpies of zirconium, titanium, and alloys of zirconium with silver and indium.



Figure 9. The silver-zirconium phase diagram.

per cent silver. The maximum solubility of silver in the alpha phase is 1.10 atomic per cent. Two intermetallic compounds are formed in this system, Zr_0Ag and ZrAg.

In order to determine the thermodynamic characteristics of the eutectoid decomposition, two alloys of silver and zirconium were studied, an alloy in the alpha-solid solution range containing 0.748 atomic per cent silver and a hypereutectoid alloy containing 4.58 atomic per cent silver. More difficulty was experienced in maintaining control with the silver alloys than with zirconium, because of the high volatility of silver even in the dilute solutions.

The specific heat data for the zirconium - 0.746 atomic per cent silver alloy is shown in Figure 10. The data are also tabulated in Table X of the Appendix. The data indicate that the specific heat of the alpha-solid solution is about two per cent higher than the specific heat of the alpha phase in pure zirconium, the difference increasing at the higher temperatures. The data also indicate that there is only a small solution heat effect from the solution of Zr_2Ag in alpha zirconium. The scatter in the data is such that it is impossible to tell if the slightly high specific heats in run 14 and 15 in the neighborhood of 700°C are due to impurity effects or experimental errors. That the measured heat of solution should be small is in agreement with the data of Betterton which show that the change in the terminal solubility of silver in alpha with decreasing temperature is small. No explanation can be given for the higher specific heat at low temperatures found in Run 50, but the effect is probably due to hydrogen. The peak at 160°C



Figure 10. The measured specific heat values for a zirconium - 0.746 atomic per cent silver alloy.

is definitely due to hydrogen. The sharp drop in the specific heat values in the first few measurements of Run 50 is due to the fact that a higher rate of heating was used in Run 50 than in Runs 13, 14, and 15. This drop in the first few specific heat values for the higher heating rates was observed in several of the runs.

The specific heat data for the 4.58 atomic per cent silver alloy are shown in Figure 11 and are given in Table XI. In the alpha range these data are about four per cent higher than the specific heat values for the 0.746 atomic per cent silver alloy. Such a behavior would be expected since the specific heat of the intermetallic compound Zr_2Ag would be expected to be high. Due to the uncertainties of the measured curves, it was not believed to be worthwhile to calculate the specific heat of the compound on the basis of this information.

There is a rather rapid rise in the specific heat of the zirconium - 4.58 atomic per cent silver alloy above 600° C. This rise is believed to be due to an increased solubility of zirconium in the compound Zr_2Ag with increasing temperature, since the change in the alpha solubility was found to be small. The total heat of solution was found to be approximately 100 calories per mole by extrapolating the specific heat data of the alloy to 825° C, subtracting the specific heat of pure zirconium, and integrating from 600 to 825° C. Such a calculation is only an approximation since it is not known how much heat due to this reaction is included in the heat of transformation. Within the limits of error the specific heat of the beta phase was unchanged by the addition of silver.



Figure 11. The measured specific heat values for a zirconium - 4.58 atomic per cent silver alloy.

The heats of transformation and temperatures of transformation for the 0.746 atomic per cent silver alloy were determined from the enthalpytemperature plot shown previously (Figure 8). These enthalpy data are also tabulated in Table XVIII of the Appendix. The temperature of the start of the alpha-beta transition in the zirconium - 0.746 atomic per cent silver was found to be about 830°C in agreement with the work of Betterton. The completion of the transformation, disregarding hysteresis effects, would be 870° for the alloy studied. This is about five degrees higher than the temperature reported by Betterton. This increase in the transformation temperature would be expected if the oxygen content were high. Since the alloy was melted in a consumable-electrode arc furnace whereas the pure zirconium was not remelted, more oxygen would be expected to be found in the alloy. The chemical analyses did not show any significant pickup of oxygen on melting, but such an analysis is not only difficult, but also relatively inaccurate.

The enthalpy data for the zirconium - 4.58 atomic per cent silver alloy which are shown in Figure 8 and tabulated in Table XIX of the Appendix indicate that the eutectoid temperature is 825° C, and that both the heat of solution of some phase in the three phase region and the heat of solution of Zr_2Ag in beta significantly affect the enthalpy of the alloy and broaden the apparent transformation range.

By extrapolating the alpha and beta specific heats and measuring graphically the enthalpy change during the transformation at an arbitrary intermediate temperature, the heat of transformation of the zirconium - 0.746 atomic per cent silver alloy was found to be
995 ± 25 calories per gram atom at 865° C. This heat of transformation is the same as that for zirconium at the same temperature. The heat of transformation of the zirconium - 4.58 atomic per cent silver alloy was found to be 1360 ± 25 calories per gram atom at 825°C. This is 310 calories per mole higher than the heat of transformation of zirconium at the same temperature. The heat of transformation of the eutectoid (3.73 atomic per cent silver) may be estimated by subtracting out the contribution of the reaction; excess Zr_2Ag above 3.73 atomic per cent silver goes into solution in the beta. Betterton (14) has reported the heat of solution of Zr_2Ag in beta to be 14,400 calories per gram atom of silver. The excess silver for the alloy studied is 0.0085 gram atoms per gram atom of alloy. The heat of solution of the amount of the eutectoid, then, is 1240 ± 25 calories per gram atom.

Zirconium-Indium Alloys

Indium has an opposite effect on the alpha-beta transformation to that of silver. This has been shown by Betterton (15) who developed the phase diagram presented in Figure 12. This diagram shows that indium raises the maximum temperature of alpha stability to a temperature of 995°C, the peritectoid composition being 10 atomic per cent indium. The gamma phase is an intermetallic compound with the approximate composition Zr_3In .



Figure 12. The indium-zirconium phase diagram to 14 atomic per cent indium.

The specific heat curves of an alloy containing 6.27 atomic per cent indium are shown in Figure 13. The data, which are also given in Table XII, 'show that within the limits of experimental accuracy the specific heat of this indium alloy is the same as that of pure zirconium. Two chemical effects due to the solution of the gamma phase were found. This alloy is so near to the saturation limit of indium in the alpha phase, that no effect from the solution of Zr_3In would be expected.

The data on the enthalpy of transformation are shown in Figure 8 and are tabulated in Table XX of the Appendix. These data indicate that the alpha to beta transition starts at 915° and is complete at about 960°C. This is a wider spread than Betterton has shown, and also is at a somewhat lower temperature for the given composition. The increased spread is probably due to oxygen absorbed during the casting operation. The iron content may also be affecting a shift in the reaction temperature.

The heat of transformation of the zirconium - 6.27 indium alloy was found to be 930 ± 25 calories per gram atom at 930°C by the same method used previously for silver alloys. This is the same value that the heat of transformation of pure zirconium would have if it transformed at the same temperature within experimental error.

The fact that the heats of transformation of solid-solution alloys of silver and indium in zirconium are the same as the values zirconium would have if it transformed at the same temperatures places certain restrictions on the thermodynamic properties. This may be shown by equating the free energy change, $\Delta F_{T_c}^{O}$, for the transformation of pure



Figure 13. The measured specific heat values for a zirconium-6.27 atomic per cent indium alloy.

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alpha zirconium to pure beta at the equilibrium transformation temperature, T_c , with the free energy change, $\triangle F_T^X$, for the hypothetical reaction alpha of composition x to beta of the same composition for a binary alloy of zirconium at a temperature T in the alpha plus beta region. If x is the composition of an alloy for which the enthalpy change has been measured, T will be taken to be the temperature at which $F_{\alpha} = F_{\beta}$ for this composition. It will also be assumed that

$$\Delta H_{o} = \Delta H_{o}^{O} + \Delta H_{o}^{X}$$
(23)

and

$$\Delta C = \Delta C^{\circ} + \Delta C^{X}.$$
(24)

In these equations $\triangle H_{O}^{O}$ and $\triangle C^{O}$ refer to the enthalpy difference at absolute zero, $H_{O\beta}^{O} - H_{O\alpha}^{O}$, and the difference in the specific heats, $C_{\beta}^{O} - C_{\alpha}^{O}$, for pure zirconium. $\triangle H_{O}^{X}$ and $\triangle C^{X}$ are the enthalpy differences due to alloying at absolute zero and the specific heat differences due to alloying in the sense beta minus alpha for the alloy of composition X. Hence, $\triangle H_{O}$ and $\triangle C$ are the total enthalpy at absolute zero and specific heat differences between the beta and alpha phases of the alloy. From equations (1) and (2),

$$\Delta F_{\mathrm{T}}^{\mathrm{X}} = \Delta H_{\mathrm{o}}^{\mathrm{o}} + \Delta H_{\mathrm{o}}^{\mathrm{X}} - \int_{\mathrm{o}}^{\mathrm{T}} \mathrm{d} \mathrm{T} \int_{\mathrm{o}} (\overset{\mathrm{T}}{\Delta} \mathrm{C}^{\mathrm{o}} + \Delta \mathrm{C}^{\mathrm{X}}) \mathrm{d} \ln \mathrm{T} = 0.$$
(25)

Likewise for the pure zirconium at the transformation temperature,

$$\Delta F_{T_c}^{o} = \Delta H_{o}^{o} - \int_{o}^{T_c} dT \int_{o}^{T} c^{o} d \ln T = 0.$$
 (26)

If equation 26 is subtracted from equation 25, the result is

$$\Delta H_{o}^{X} + \int_{T_{c}}^{T} \Delta C^{o} dT - (T-T_{o}) \int_{o}^{T_{c}} \Delta C^{o} dlnT - T \int_{T_{c}}^{T} \Delta C^{o} dlnT + \int_{o}^{T} \Delta C^{X} dT - T \int_{o}^{T} \Delta C^{X} dlnT = 0 \quad (27)$$

Since the enthalpy for the transformation of the pure zirconium, λ_c , is equal to $\Delta \operatorname{S}^{O}T_c$, where $\Delta \operatorname{S}^{O}$ is the entropy change associated with the transformation and

$$\triangle S^{\circ} = \int_{\circ}^{T_{co}} \Delta C^{\circ} d \ln T,$$

certain simplifications of equation 27 may be made. If it is further observed that $\triangle C^{\circ} \stackrel{\sim}{=} -1$ for zirconium in the neighborhood of T_{c} and $\ln \frac{T}{T_{c}} \stackrel{\sim}{=} \frac{T}{T_{c}} - 1$, equation 27 may be written as follows: $\triangle H_{\circ}^{X} - (T-T_{c}) - \frac{(T-T_{c})}{T_{c}}\lambda_{c} + \frac{(T-T_{c})}{T_{c}}T + \int_{\circ}^{T}\Delta C^{X}dT - T\int_{\circ}^{T}\Delta C^{X}d \ln T \stackrel{\sim}{=} 0$ (28).

Since $\lambda_c \stackrel{\sim}{=} T$ for zirconium in the neighborhood of T_c ,

$$\Delta H_{O}^{X} + \int_{O}^{T} \Delta C^{X} dT - T \int_{O}^{T} \Delta C^{X} d \ln T \stackrel{\sim}{=} T - T_{C} \qquad (29)$$

Next consider the enthalpy difference $\triangle H_T^x$ at temperature T and composition x, which can be measured. This quantity may be written

$$\Delta H_{\rm T}^{\rm X} = \Delta H_{\rm o}^{\rm O} + \int_{\rm o}^{\rm T_{\rm C}} dT + \int_{\rm T}^{\rm T} \Delta C^{\rm o} dT + \Delta H_{\rm o}^{\rm X} + \int_{\rm o}^{\rm T} \Delta C^{\rm X} dT , \qquad (30)$$

or

$$\Delta H_{\rm T}^{\rm X} = \lambda_{\rm c} + \int_{\rm T} \int_{\rm c}^{\rm T} \Delta C^{\rm o} dT + \Delta H_{\rm o}^{\rm X} + \int_{\rm o}^{\rm T} \Delta C^{\rm X} dT \quad . \tag{31}$$

The experimental data indicate

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$$\Delta H_{\rm T}^{\rm X} = \lambda_{\rm c} + \int_{\rm T} \frac{T}{\Delta C^{\rm o} dT} \pm 25 \, {\rm cal/g \, atom} \,, \qquad (32)$$

so that

$$\Delta H_{O}^{X} + \int_{O}^{T} \Delta C^{X} dT = \pm 25 \text{ cal/g atom} .$$
 (33)

Substitution of equation 33 into equation 29 yields the result

$$\int_{O}^{T} \Delta C^{X} d \ln T = -\frac{T - T_{C}}{T} \pm 25 \text{ cal/g atom.}$$
(34)

If in equation 32,

$$\Delta H_{T} = \lambda_{c} + \int_{T_{c}}^{T} \Delta C^{0} dT,$$

equation 34 shows that $\triangle C^X$ could not be zero at all temperatures, but the scatter of the data is such that one cannot separate the effects of $\triangle C^X$ and $\triangle H^X_O$. In any event equation 34 indicates that $\triangle C^X$ is quite small. If this is the case equation 29 indicates that $\triangle H^X_O$ is also quite small. It is doubtful that an adiabatic calorimeter with the precision to separate these terms could ever be developed except for systems where very anomalous results are found.

5. Zirconium-Hydrogen Alloys

In the study of the specific heat of pure zirconium an anomalous, but reproducible, peak always appeared in the first runs made on a given specimen. As the specimen was recycled to high temperatures in the high vacuum maintained within the calorimeter, the magnitude of this peak decreased. Because of this decrease in effect after vacuum annealing, it was suspected that the peak was due to hydrogen. To prove the validity of this supposition a specimen of iodide zirconium was purposely heated in an atmosphere of rarified hydrogen. The pressure of hydrogen was adjusted so that the equilibrium content of hydrogen in the zirconium would be 250 ppm at the temperature of the absorption. The equilibrium-pressure temperature data of Gulbransen and Andrew (29) was used to select the proper pressure. The resulting specific heat curve of the hydrogenated zirconium, together with low temperature data on the iodide zirconium runs are presented in Figure 14 and in Table XIII of the Appendix. The data show that a decided peak in the specific heat curve of zirconium is produced by the addition of hydrogen.

The peak is undoubtedly due to the solution of the hydride ZrH₂ in alpha phase as the temperature increased. When all the hydrogen from the hydride had gone into solution, the specific heat of the zirconiumhydrogen alloy was identical to that of pure zirconium. A study of the specific heat curve under different heating rates showed that within the available range of heating rates the curve is independent of the heating rate. This would imply that the rate of reaction is quite high.

The phase diagram for the hydrogen-zirconium system has been studied by Edwards, Levesque, and Cubicciotti (30), Vaughan and Bridge (31), Schwartz and Mallett (33), Gulbransen and Andrew (29) and Ells and McQuillan (33). The constitutional diagram determined by these investigators is shown in Figure 15. A better description of the alphaterminal solubility is given in Figure 16 which shows the logarithm of the terminal solubility in atomic per cent as a function of the reciprocal of the absolute temperature.

If one assumes that equilibrium was essentially maintained in the hydrogen-zirconium system under the conditions of operation of the calorimeter, a correlation between the terminal-solubility curve and



Figure 14. The measured specific heat values for alloys of zirconium with 0.26 and 2.75 atomic per cent hydrogen.

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Figure 15. The hydrogen-zirconium phase diagram.





the specific heat curve can be made. For if equilibrium were attained, the peak in the specific heat curve would indicate the temperature at which one crossed the terminal-solubility boundary. This temperature then corresponds to a certain hydrogen content. For the iodide zirconium the peak, which occurred at 242°C, corresponds to 28.5 parts per million of hydrogen in good agreement with the value of 35 parts per million obtained by vacuum fusion analysis. The hydrogenatedzirconium peak occurred at 450°C which corresponds to 300 parts per million of hydrogen. Unfortunately a direct analysis of the specimen was not made before the run and the hydrogen was removed by the vacuum anneal associated with the specific heat measurements. The heat associated with the solution process in both the iodide specimen and the hydrogenated specimen were measured by integrating the area under each peak numerically. The results are shown in Table VII.

The agreement is surprisingly good considering the long extrapolation in the solubility curve. The heat of solution may also be computed by the approximate equation:

$$S = S_{o}e^{-\frac{\Delta R}{RT}}$$

From the slope of the curve of log S versus 1/T the heat of solution was calculated to be 8,616 calories per gram atom of hydrogen using the data of Gulbransen. This is also in fair agreement with the measured heats of solution. Thus the calorimetric data substantiate the reported terminal solubility which had been previously reported. The

TABLE VII

COMPUTED HEAT OF SOLUTION OF ${\rm ZrH}_2$ IN ALPHA ZIRCONIUM

Specimen			Gran	Gram Atoms of Hydr			gen Heat of Solution Calories				Heat of Solu- tion Calories	
				Gran	Atoms	of	Specimen	Gram	Atom	of	Spec.	Gram Atom of Hydrogen
												· ·
Zr	+	28.5	ppm	H	0.0	0026	5	C	.2999	9		10,500
Zr	+	300	ppm	Н	0.0	0275	5		3.4678	3		11,500

data also show that an adiabatic calorimeter would be an excellent instrument for studying the zirconium-hydrogen phase diagram and transformation kinetics.

6. Zirconium-Niobium Alloys

The small range of temperatures between the transformation temperature of zirconium (870°C) and the maximum operating temperature of the calorimeter (960°C) made the determination of the specific heat of the beta phase and the effect of temperature on this quantity very difficult. For this reason an alloy was prepared in which the beta stability was lowered to 610 degrees Celsius and in which the solute, niobium, was known to have the same structure and almost the same mass as the beta phase of zirconium. A similarity of these properties is desirable, because the specific heat would be less affected by the addition of such a solute. The composition of the zirconium-niobium alloy which was run in the adiabatic calorimeter was 17.5 weight per cent niobium.

The phase diagram for the zirconium-niobium system, which was developed by Rodgers and Atkins (34), is shown in Figure 17. This diagram shows the beta phase to be stable as low as 610°C at a composition of 17.5 weight per cent niobium. Rodgers and Atkins reported that the decomposition of the beta phase is quite sluggish so that the beta phase may be retained to room temperature if the niobium content is greater than 5 weight per cent. It was hoped that such a retention of the beta phase would permit the determination of its specific heat over the range of 100-900°C.



Figure 17. The niobium-zirconium phase diagram.

The measured specific heat curves for the zirconium-17.5 weight per cent niobium alloy are shown in Figure 18 and are given in Table XIV of the Appendix. The first two runs (67 and 68) were made only up to a temperature of three hundred degrees to prevent any transformation from obscuring the specific heat of the beta phase in the vicinity of room temperature. On the third run (No. 69) which was carried to higher temperatures, a slight peak was observed at 320°C. Then there was a large dip at approximately 400°, and finally a decided peak was again observed at 480°. From 520° to 800° the specific heat of the alloy was constant within experimental error. The heating rate of this run was seven degrees a minute at 200 degrees. Thus the reaction consisting of a precipitation and resolution of a new phase, which undoubtedly occurred, was exceedingly rapid.

After the specimen had cooled in the calorimeter to room temperature the specific heat curve was redetermined. The same result was found except that the dip at 400°C was more severe. This indicated that even with the relatively slow cooling rates associated with the cooling of the specimen in a vacuum, some nuclei for the precipitation might have formed, but the reaction by no means had gone to completion. When the temperature reached 460° the power to the specimen was turned off, and the specimen was allowed to drift for two hours to determine if further precipitation would occur isothermally. The higher peak on resolution (Run 70) indicated that there was indeed additional precipitation. Again above 520° specific heat was essentially constant which indicated that the structure was all beta.



Figure 18. The measured specific heat values for a zirconium-17.5 weight per cent niobium alloy.

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The results of these initial runs had little correlation with the phase diagram which had been reported. It was suspected that this lack of correlation was due to the rapid precipitation and resolution of a nonequilibrium structure, omega, which will be discussed later, and the sluggishness of the equilibrium reaction. To check this hypothesis the specimen was removed from the calorimeter, sealed in a quartz tube with rarefied argon, and annealed for two weeks at 600°C, ten degrees below the reported equilibrium transformation temperature. It was hoped that a two-week anneal at this temperature would produce the equilibrium structure in the alloy. After the heat treatment, the specimen was returned to the calorimeter and Run 77 was made. The pronounced peaks and dips observed in Runs 69 and 70 were now almost completely removed, but traces were still evident. Above 600° the specific heat rose sharply indicative of a transformation. The fact that the specific heat remained abnormally high up to 800° showed that this transformation was quite sluggish. An integration of the specific heat-temperature curve gave a heat of transformation of only 332 calories per gram atom. At the peak temperature of 700° the heat of transformation of pure zirconium would be approximately 1270 calories per gram atom.

After the specimen cooled to room temperature, a final run (Run 78) was made. This run showed essentially the same characteristics as the run made on the beta-quenched specimen (Run 69).

Additional information on the reactions occurring during the calorimetric runs was obtained by sectioning samples from the original specimen and giving them heat treatments which corresponded approximately

to the thermal history received in the calorimeter. These samples were then analyzed metallographically, with hardness measurements, and with x-rays. The x-ray technique used was the analysis of the patterns obtained from filings in a standard Debye-Sherrer camera. The x-ray study was done by R. M. Steele of the Metallurgy Division of the Oak Ridge National Laboratory.

The sample which had been held for two weeks at 600°C will be discussed first. This sample consisted of two phases which were identified as alpha and beta (a = 3.526A). No niobium-rich beta prime (β' , a \approx 3.35) was detected. A visual examination of the relative intensities of the x-ray lines indicated that the structure was approximately 30 per cent alpha. A lineal analysis of the microstructure indicated that the amount of alpha was 17.5 per cent; however, due to the difficulties in etching zirconium-niobium alloys, microscopic interpretations are only qualitative. If the latter figure is assumed to be correct, the heat of transformation of the alpha to beta in this alloy at 700° is approximately 1840 calories per mole. That this is higher than the heat of transformation of zirconium at the same temperature (1270 cal/mole) would indicate that there is also beta prime in the structure which is not resolvable metallographically or with x-rays. Additional heat would be released if this were the case corresponding to the heat of resolution of beta prime in beta. If, on the other hand, no beta prime is present, the alpha would indicate that there was metastable equilibrium between the alpha and beta. The amount would correspond to the calculated value obtained by applying the lever

rule to the extrapolated alpha-alpha plus beta, and beta-alpha plus beta boundary if the eutectoid temperature is raised to 620°C. At 610° this calculation would predict seven per cent of alpha. Thus the nature of the structure obtained by annealing a 17.5 weight per cent niobium alloy for two weeks at 600° is uncertain other than that it is definite that equilibrium was not attained. The hardness of this specimen was 173 DPH.

A second sample was annealed for two hours at 800°C and then water quenched. The hardness of this specimen was 179 DPH and the metallographic study showed that there was a needlelike phase comprising two per cent of the structure. This phase was not detected by x-rays.

A phase of similar appearance was observed in a sample cooled from 800° in the calorimeter. In this instance the matrix adjoining the phase was depleted in one of the constituents so that the specimen looked like a cored casting. The hardness of the calorimeter sample was 194 DPH, fifteen points harder than the quenched sample. When this specimen was reheated to 450° for two hours the hardness rose to 251 DPH. A decrease in the amount of the metallographically observable second phase indicated that this phase was going back into solution. X-ray data indicated the phase in the calorimeter cooled specimen had a face-centered cubic structure with a lattice constant 5.61A. After the two-hour anneal at 450°, insufficient second phase for x-ray detection remained. Thus x-ray and metallographic examination indicated a low-temperature phase but failed to show a structure which could account for the high hardness after a two-hour anneal at 450° and the

anomalous dips and peaks in the specific heat curves below 520°. Little hardening is associated with the low-temperature phase.

Although it cannot be detected except by hardness and calorimetric measurements, the transition which causes the pronounced dip and rise in the specific heat is believed to be the beta-omega transition which has been reported previously in the zirconium-niobium system (35). This transition, which was first reported in titanium alloys with beta stabilizers, is not yet fully understood, although the increased hardness associated with it is of considerable practical significance. Omega is known to be a metastable phase, which ultimately decomposes to equilibrium constituents after prolonged ageing treatments. The metastable phase forms because of its ability to form much more rapidly than the equilibrium constituents, particularly during a quench and reheat treatment (36). Although x-ray data have been reported for the omega structure in zirconium-molybdenum (35) and in titanium alloys (37), it has not been observed metallographically at the present time.

The calorimetric data indicate the omega phase retrogresses to beta above a temperature of 520°C, about one hundred degrees below the equilibrium transformation. The data also indicate that the amount of omega increases with time at a temperature of 450°C. Prolonged heating at 600° greatly decreases the tendency for beta to form omega, possibly because of the formation of alpha in favorable nucleation sites. Although further work is needed to rationalize the omega transition in the zirconium system, these data indicate the usefulness of the adiabatic calorimeter in the study of the kinetics of phase transformations.

If one ignores the beta-omega transition, he will find that the specific heat of the beta phase in this alloy is essentially constant from room temperature to 800°C. It is for this reason that the specific heat of beta zirconium was assumed to be constant from 870° to 950°C in Figure 18.

7. Iodide Titanium

The corrected specific heat data for iodide titanium are given in Figure 19 and Table XV of the Appendix. The data are compared with values determined by previous investigators in Figure 3. The values given in the present determination are lower than those of Jaeger, Rosenbohm, and Fonteyne (4) and Skinner, Johnston, and Beckett (8). It will be remembered that the data given for the specific heat of zirconium were also lower than the results previously reported. As reported by previous investigators, there is an abnormal increase in the specific heat of the alpha phase in comparison to other metals and the specific heat of the beta phase is lower than that of the alpha phase at the transformation temperature.

The enthalpy of titanium above 800° C is given in Figure 8, and in Table XXI of the Appendix. These data indicate that the transformation temperature of titanium is $883 \pm 2^{\circ}$ C and the heat of transformation of titanium is the same as the value zirconium would have if it transformed at the same temperature. In addition the gram-atomic



Figure 19. The measured specific heat values for titanium.

specific heat of titanium and zirconium are the same within experimental error. These observations emphasize the similarity between zirconium and titanium.

7. Zirconium-Titanium Alloys

The phase diagram for the zirconium-titanium system, which was determined by Fast (36) and Hayes et al (37) is shown in Figure 20. This diagram is surprising in that the beta phase is stable down to a temperature of approximately 550°C at a composition of 34.4 weight per cent titanium (50 atomic per cent). Smoluchowski (38) has rationalized this decrease in the transformation temperature by the argument that preferential ordering in the beta phase, because of the greater importance of the size effect in body-centered cubic alloys, leads to a lowering of the free energy of the beta phase so that it is stable at lower temperatures. It was felt by the author that such an effect would be indicated by calorimetric data if it were indeed true. Thus an alloy of 50 atomic per cent titanium and zirconium was studied.

The results of the calorimetric studies are given in Figure 21 and Table XVI of the Appendix. These data indicate that the specific heat of the alpha phase on a gram-atomic basis is the same as that of zirconium up to 550° C. From 550 to 640° a sluggish transformation from alpha to the beta phase occurs with a heat of transformation of 740 ± 20 calories per gram atom of the alloy at 600° C. On slow cooling the transformation begins at 558.5° C and the temperature rises to a maximum temperature



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Figure 20. The titanium-zirconium phase diagram.



Figure 21. The measured specific heat values for a zirconium-50 atomic per cent titanium alloy.

of 560.5°C. Above 640 degrees the specific heat of the beta phase decreases linearly up to a temperature of 900°C, the highest temperature studied. That this decrease is reversible is shown by the fact that when the specimen was cooled to 650° and rerun to a temperature of 900° the linear decrease was again observed. In an attempt to obtain data on the beta phase at lower temperatures, the calorimeter specimen was homogenized at 800°C for two hours and water quenched. After the sample was returned to the calorimeter Run 79 was obtained. This run showed a dip in the alpha specific heat between 320° and 480°C which can be interpreted by assuming that 4.7 per cent of the beta was retained on quenching. Thus data for the beta phase at temperatures below 640° were not obtainable.

The linear decrease in the specific heat of the beta phase can be interpreted as being due to disordering of an ordered beta structure in agreement with the hypothesis of Smoluchowski. The fact that at 900 degrees these data fall to the value 3R, the classical value of the specific heat, cannot be explained.

The peaks at 190°C in Run 43 and 150°C in Run 45 are undoubtedly due to hydrogen. The effect of hydrogen on the specific heat dissappeared, because the hydrogen was removed by the vacuum anneal associated with these runs.

CHAPTER VI

CONCLUSIONS

The following conclusions may be drawn from the results of this investigation:

1. The specific heat of beta zirconium is lower than that of alpha at the transformation temperature of $870 \pm 2^{\circ}$ C. Analogously the specific heat of beta titanium is lower than that of alpha at the transformation temperature of $883 \pm 2^{\circ}$ C. For a zirconium-17.5 weight per cent niobium alloy, which should correspond quite well to pure zirconium, the specific heat of the beta phase has the constant value 0.069 ± 0.002 calories/gram-°C from room temperature to 800° C. Within the limits of error the addition of 0.746 atomic per cent of silver or 6.27 atomic per cent of indium does not affect the specific heat of either the alpha or beta phase over their ranges of stability above room temperature. An alloy of higher silver content (4.58 atomic per cent) does have a slightly higher specific heat of the intermetallic compound Zr_oAg .

2. The heat of transformation of zirconium is 993 ± 25 calories per gram atom at 870° C, and the heat of transformation of titanium is 978 ± 25 calories per gram atom at 883° C. Within the limits of experimental error the heats of transformation of zirconium and titanium would have the same value if the transformations occurred at the same temperature. Although the transformation from the alpha to the beta

phase in a 0.746 atomic per cent silver alloy occurs over the temperature range 830 - 854°C, the heat of transformation at an arbitrary intermediate temperature is the same as the value zirconium would have if it transformed at the same temperature within the limits of error. This is also true for an alloy of zirconium with 6.27 atomic per cent of indium, which transforms over the range of 910 - 940°C. The temperature of the eutectoid transformation in the zirconium-silver system is $825 \pm 2^{\circ}$ C and the heat of transformation of the eutectoid (3.73 atomic per cent silver) is 1240 ± 25 calories per gram atom.

3. The changes in the thermodynamic functions $\triangle H_{O}^{x}$ and $\int_{O}^{T} dT \int_{O}^{T} \triangle C^{x} d \ln T$

are quite small, so that the precision of the calorimeter used was not sufficient to permit their evaluation.

4. The heat due to the solution of ZrH_2 in alpha zirconium may be detected calorimetrically even when the hydrogen content is as low as 28.5 ppm. The heat of solution is approximately 11,000 calories per mole of hydrogen.

5. There exists a metastable transition in a zirconium-17.5 weight per cent niobium alloy below 520°C which has decided effects on the specific heat curves and on the hardness but cannot be identified by metallographic or x-ray investigations. This transformation is tentatively called the omega transition. A second metastable phase which occurs at lower temperature can be identified metallographically and with x-rays, but has little effect on the hardness or the specific heat. 6. An alloy containing 50 atomic per cent titanium with zirconium shows the expected specific heat behavior in the alpha phase, but the specific heat of the beta phase is definitely abnormal. The beta specific heat decreases linearly from 650°C to 800°C. This effect may be interpreted as a disordering of the beta phase. The transformation occurred over the range 550 - 640°C on heating and 558.5 - 560.5°C during cooling. The heat of transformation of 740 \pm 20 calories per gram atom at 600°C is abnormally low.

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APPENDIX
TABLE VIII

SAMPLE CALCULATIONS

							Zirconi	um - 0.03	Weight Pe	er Cent Hy	drogen, Run 74					
т (°с)	∆t sec.	Τ (μv)	E (volts)	E (volts)	I (amps)	Ī (amps)	ĒĪ∆t (watt- sec.)	Cp _A cal/g °C	∆t ∆T (sec,^C)	R(T) (°C/hr)	$1 - \frac{\Delta t}{\Delta T} \frac{R(T)}{3600}$ (dimensionless)	^{Cp} T cal/g °C	Cp _T 55C _H cal/g °C	$\frac{WH}{W_{S}}(Cp_{T}55C_{H})$ cal/g °C	C _s cal/g°C	፹ (°C)
100 140 180 220 260	0 347.1 349.2 350.8 348.4	645.5 957.0 1251.0 1643.7 2012.6	5 • 3935 5 • 3928 5 • 3894 5 • 3860 5 • 3826	5.3931 5.3911 5.3877 5.3843	0.74418 0.74392 0.74321 0.74276 0.74206	0.74495 0.74356 0.74298 0.74241	1392.82 1399.81 1404.24 1396.68	0.07359 0.073 9 9 0.07419 0.07379	8.68 8.73 8.77 8.74	0.90 1.67 2.80 4.16	1.0022 1.0041 1.0068 1.0101	0.07343 0.07369 0.07369 0.07305	0,0607 0.0604 0.0604 0.0610	0.0019 5 0.00194 0.00194 0.00196	0.07148 0.07175 0.07175 0.07109	120 160 200 240
300	354.4	2395.6	5 • 3794	5.3810	0.74132	0.74169	1414.42	0.07473	8.86	5.53	1.0136	0.07372	0.0604	0.00194	0.07178	280
340	359.5	2750.4	5 • 3760	5.3777	0.74060	0.74096	1432.49	0.07569	8.99	6.91	1.0173	0.07440	0.0597	0.00192	0.07248	320
380	362.0	3195.1	5 • 3735	5.3747	0.73995	0.74027	1440.30	0.07610	9.05	8.20	1.0206	0.07456	0.0595	0.00191	0.07265	360
420	365.5	3508.4	5 • 3706	5.3720	0.73923	0.73959	1452.16	0.07672	9.14	9.61	1.0244	0.07489	0.0592	0.00190	0.07299	400
460	370.7	4029.9	5 • 3679	5.3692	0.73851	0.73887	1470.62	0.07770	9.27	16.9	1.0281	0.07558	0.0585	0.00188	0.07370	440
500	374.6	4459.4	5.3691	5.3685	0.73830	0.73840	1484.95	0.07845	9.37	12.81	1.0333	0.07593	0.0582	0.00187	0.07406	480
540	378.9	4896.6	5.3752	5.3711	0.73876	0.73853	1502.99	0.07941	9.47	13.49	1.0355	0.07669	0.0574	0.00185	0.07484	520
580	383.7	5341.5	5.3801	5.3776	0.73881	0.73878	1524.39	0.08054	9.59	14.50	1.0386	0.07755	0.0565	0.00182	0.07573	560
620	390.9	5794.1	5.3843	5.3822	0.73855	0.73868	155 ¹ 4.11	0.08211	9.77	15.58	1.0423	0.07878	0.0553	0.00178	0.07700	600
660	398.1	6254.4	5.3829	5.3836	0.73801	0.73828	1582.29	0.08360	9.95	16.50	1.0456	0.07995	0.0541	0.00174	0.07821	640
700	405.8	6722.3	5.3919	5.3874	0.73736	0.73768	1604.77	0.08479	10.15	17.40	1.0491	0.08082	0.0533	0.00171	0.07911	680
740	408.5	7197.9	5.3903	5.3911	0.73586	0.73661	1622.21	0.08571	10.21	18.28	1.0518	0.08149	0.0526	0.00169	0.07980	720
780	416.1	7681.0	5.3940	5.3921	0.73506	0.73546	1650.12	0.08718	10.40	19.10	1.0552	0.08262	0.0515	0.00165	0.08096	760

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TABLE 1	IX
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and the second state of the second							
Run 17		Run 18		Run 18	Run 18		
EI = 2	.410 watts	EI = 2	.546 watts	EI = 2	EI = 2.546 watts		
Т°С	Cp cal gram °C	Τ°C	Cp cal gram °C	Τ°C	Cp <u>cal</u> gram_°C		
90 110	0.06838 0.06903	(°.90 130	0.06849 0.07009	845	0.09938		
130	0.06926	170 210	0.07106	TRANSFO	ORMATION		
170	0.07041	250	0.07276	892.5 910	0.07361 0.07179		
190 210 230	0.07079 0.07160	290 330	0.07124 0.07195	930 950	0.07102 0.07224		
250 250 270	0.07201 0.07042	430 450	0.07325 0.07366	Run 24			
	0.0700			EI = 1	.402 watts		
290 310 330	0.07091 0.07126	470 490 510	0.07448 0.07512 0.07563	T°C	Cp <u>cal</u> gram °C		
360 390	0.07173 0.07257	530 550	0.07646 0.07668	270 290 310	0.07243 0.07161 0.07149		
410 430 450	0.07256 0.07317 0.07359	570 590 610	0.07699 0.07783 0.07857	330 350	0.07209 0.07257		
470 490	0.07410 0.07445	630 650	0.07914 0.07991	370 390 210	0.07349 0.07371 0.07/139		
510 530 550	0.07506 0.07560 0.075 90	670 700 730	0.08094 0.08223 0.08543	430 450	0.07454 0.07450		
570 590	0.07696 0.07779	750 770	0.08660	470 490 510	0.07584 0.07646 0.07736		
610	0.07826	790 805 815	0.09086 0.09212 0.09235	530 550	0.07814 0.07873		
		825 835	0.09296 0.09464	570 590	0.07919 0.07973		

TRUE SPECIFIC HEAT VALUES OF IODIDE ZIRCONIUM

TABLE IX(Continued)

Run 26		Run 28		Run 60	- <u> </u>	
EI = 2	.454 watts	EI = 3	.780 watts	EI = 1	EI = 1.441 watts	
Τ°C	Cp <u>cal</u> gram °C	Τ°C	Cp cal gram °C	Т°С	Cp cal gram °C	
60 80 100 120 140	0.07012 0.07004 0.07095 0.07122 0.07145	280 320 360 400 440	0.07196 0.07232 0.07266 0.07289 0.07362	260 280 300 320 340	0.07099 0.07051 0.07092 0.07146 0.07143	
160 180 200 220 240	0.07180 0.07202 0.07146 0.07122 0.07130	480 520 560 600 640	0.07448 0.07 57 8 0.07718 0.07801 0.07937	360 380 400 Run 62	0.07163 0.07191 0.07216	
260	0.07136	680	0.08035	EI = 3	.928 watts	
280 300 320	0.07170 0.07178 0.07187	720 760 800	0.08113 0.08201 0.08483	T°C	Cp gram°C	
350 390 430	0.07257 0.07335 0.07441	2257 Run 60 335 441 EI = 1.441 watts		320 360 400 440 480	0.07231 0.07268 0.07301 0.07339 0.07341	
4 /U Dum 08	0.07543	T°C	Cp cal oc	520	0:07263	
EI = 3.	780	60 80 100	0.06816 0.06809 0.06840	560 600 640	0.07477 0.07600 0.07726	
T°C	Cp gram °C	120 140	0.07007 0.07100	680	0.07814	
80 120 160 200	0,07045 0.07081 0.07152 0.07135	160 180 200 220	0.07138 0.07166 0.07183 0.07233	720 760 800 840	0.07970 0.08107 0.08493 0.10069	
240	0.07118	240	0.07260	TRANSFO	DRMATION	
				900 920 940 960 980	0.08030 0.07504 0.07418 0.07462 0.07497	

TRUE SPECIFIC HEAT VALUES OF IODIDE ZIRCONIUM

TABLE IX (Continued)

Run 64		Run 74		Run 75 cont.		
EI = 2.54	0 watts	EI = 3.96	8 watts	EI = 3.986 watts		
Т°С	Cp cal cram °C	Τ°C	Cp cal grams °C	Τ°C	$Cp \frac{cal}{gram} \circ C$	
60 100 140 180 220	0.07008 0.06891 0.07026 0.07127 0.07112	120 160 200 240 280	0.07148 0.07175 0.07175 0.07109 0.07178	340 380 420 480 540	0.07269 0.07291 0.07342 0.07465 0.07553	
260 300 340 380 4 <i>2</i> 0	0.07110 0.07 16 9 0.07213 0.07258 0.07310	320 360 400 440 480	0.07248 0.07265 0.07299 0.07370 0.07406	580 620 660 700	0.07657 0.07768 0.07834 0.08016	
460 500 540 580 620	0.07403 0.07506 0.07604 0.07726 0.07877	520 560 600 640 680	0.07484 0.07573 0.07700 0.07821 0.07911			
660 700 740 780 820	0.07996 0.08065 0.08131 0.08309 0.08472	720 760 Run 75	0.07980 0.08096			
850	0.10497	EI = 3.900	cal			
TRANSFORMATION			Cp gram °C			
900 920 940	0.07553 0.06918 0.06901	140 180 220 260 390	0.07104 0.07161 0.07122 0.07157 0.07210			

TRUE SPECIFIC HEAT VALUES OF IODIDE ZIRCONIUM

TABLE X

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 0.881 WEIGHT PER CENT SILVER ALLOY

Run 13	(Run 14		Run 15		
EI = 2	.511 watts	EI = 2	.590 watts	EI = 2	EI = 2.500 watts	
Τ°C	Cp <u>cal</u> gram °C	Τ°C	Cp cal c	Τ°C	Cp <u>cal</u> gram °C	
60 90 100 120 140	0.06814 0.06764 0.06847 0.06932 0.06985	100 140 180 220 260	0.06894 0.07000 0.07038 0.07005 0.07014	70 90 110 130 150	0.06802 0.06795 0.06918 0.06911 0.07068	
160 180 210 240 260	0.06989 0.07025 0.06960 0.06996 0.06994	300 340 380 420 460	0.07102 0.07161 0.07218 0.07295 0.07422	170 190 210 230 250	0.07043 0:07028 0.06972 0.07079 0.06819	
280 300 320 340 360	0.07031 0. 0 7039 0.07082 0.07131 0.07153	500 540 580 620 660	0.07558 0.07638 0.07764 0.07885 0.08044	270 290 310 330 350	0.06996 0.07015 0.07072 0.07121 0.07146	
380 400 420 440 460	0.07186 0.07232 0.07302 0.07351 0.07420	690 710 730 750 770	0.08201 0.08436 0.08521 0.08590 0.08632	370 390 410 430 450	0.07197 0.07242 0.07242 0.07332 0.07353	
480 500 520 540:: 560	0.07467 0.07527 0.07612 0.07664 0.07719	790 810 830	0.08782 0.09477 0.10544	470 490 510 530 550	0.07554 0.07488 0.07541 0.07609 0.07666	
580 600	0.07766 0.07843			570 590 610 630 650	0.07710 0.07779 0.07885 0.07875 0.08008	

TABLE X(Continued)

Run 15 cont.			Run 50		
EI = 2.500 watts			EI = 3.821 watts		
т°С	Cp cal cram °C	Τ°C	Cp cal gram °C		
670 690 710 730 750	0.08196 0.08357 0.08555 0.08588 0.08649	80 120 160 200 240	0.07405 0.07237 0.07212 0.07165 0.07192		
770 790 810 TRANSFOR	0.08891 0.09153 0.09160 MATION	280 320 360 40 0 440	0.07254 0.07299 0.07349 0.07390 0.07427		
905 915 925 935 945	0.07397 0.06833 0.06772 0.06819 0.06857	480 520 560 600 640	0.07488 0.07555 0.07677 0.07828 0.07966		
		680 720 760 800	0.08093 0.08238 0.08403 0.08811		
		TRANSFO	ORMATION		
		920 960	0.07290 0.07030		

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 0.881 WEIGHT PER CENT SILVER ALLOY

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Run 32		Run 32	cont	Run 33	Run 33 cont.	
EI = 2	.361 watts	EI = 2	.361 watts	EI = 3	EI = 3.649 watts	
Τ°C	Cp cal c	Τ°C	$Cp \frac{cal}{gram} \circ_C$	Τ°C	$C_p \frac{cal}{gram} \circ_C$	
70 90 110 130 150	0.07111 0.07010 0.07088 0.07135 0.07173	670 690 710 730 750	0.08343 0.08560 0.08650 0.08975 0.093 5 9	500 540 580 620 660	0.07743 0.07791 0.07863 0.07907 0.08057	
170 190 210 230 250	0.07140 0.07148 0.07183 0.07226 0.07223	770 790 805 850 870	0.09552 0.09802 0.10276 0.10586 0.10045	700 740 780 805 Run 36	0.08171 0.08554 0.09061 0.09386	
270 290 310 330 350	0.07286 0.07338 0.07337 0.07390 0.07434	890 910 930 950	0.08728 0.07588 0.07436 0.07561	EI = 3 T°C	.649 watts Cp <u>cal</u>	
370 390	0.07461 0.07523	Run 33 EI = 3	.649 watts	360 400 440	0.07490 0.07566 0.07638	
410 430 450	0.07585 0.07653 0.07701	Τ°C	Cp <u>cal</u> gram. °C	480 520	0.07689 0.07750	
470 490 510 530	0.07 7 3 0.07808 0.07888 0.07946 0.07982	110 140 180 220 260	0.07163 0.07244 0.07261 0.07295 0.07346	560 600 640 680 720	0.07796 0.07894 0.07981 0.08216 0.08711	
570 590 610 630 650	0.08048 0.08092 0.08137 0.08174 0.08224	300 340 380 420 460	0.07440 0.07453 0.07526 0.07587 0.07661	760	0.09049	

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 5.37 WEIGHT PER CENT SILVER ALLOY

TABLE XI (Continued)

Ru	n 37	andonena gijaj provinskom
EI = 3.8	B10 watts	
T°C	Cp cal gram °C	
100 140 180 220 260	0.07414 0.07251 0.07234 0.07229 0.07291	
300 340 380 420 460	0.07364 0.07451 0.07527 0.07595 0.07692	
500 540 580 620 660	0.07756 0.07855 0.07910 0.08006 0.08110	
700 740 780 805 815	0.08373 0.08887 0.09432 0.09862 0.16941	
TRANSFOR	MATION	
850 880	.0.14361 0.10803	

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 5.37 WEIGHT PER CENT SILVER ALLOY

TABLE XII

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 7.77 WEIGHT PER CENT INDIUM ALLOY

Run 54		Run 55		Run 56	cont.	
EI = 2.4	85 watts	E I = 3	.866 watts	EI = 3	EI = 3.884 watts	
Т°С	Cp <u>cal</u> °C	Τ°C	Cp <u>cal</u> gram °C	Τ°C	$C_p \frac{cal}{gram} C$	
80 120 160 200 240	0.07082 0.06970 0.06980 0.06991 0.07032	80 120 160 200 240	0.06924 0.06960 0.06969 0.06991 0.07065	270 310 350 390 430	0.07109 0.07168 0.07220 0.07244 0.07360	
280 320 360 400 440	0.07052 0.07109 0.07175 0.07184 0.07215	280 320 360 400 440	0.07153 0.07185 0.07227 0.07247 0.07296	470 510 550 590 630	0.07381 0.07503 0.07611 0.07692 0.07748	
480 520 560 600 640	0.07310 0.07438 0.07626 0.07705 0.07772	480 520 560 600 640	0.07397 0.07529 0.07582 0.07642 0.07653	670 710 750 790 830	0.07802 0.07938 0.08024 0.08175 0.08403	
680 720 760 800 830	0.07863 0.07845 0.08053 0.08209 0.08211	680 720 760 800 840	0.07668 0.07880 0.08039 0.08024 0.08113	870 900 Run 58	0.08465 0.11912	
850 870 885	0.08071 0.08465 0.08982	880 Bun 56	0.08643	EI = 2. T°C	.492 watts Cp <u>cal</u> °C	
895 905	0.09436 0.10867	EI = 3	.884 watts	60 100 140	0.06999 0.06903 0.06964	
		70 110 150 190 230	Cp gram °C 0.07034 0.06965 0.06976 0.06980 0.07041	180 220	0.06986 0.07044	

TABLE XII (Continued)

Run 58	cont.	Run 59		
EI = 2	.492 watts	EI = 3.914 watts		
Т°С	Cp cal gram °C	Т°С	Cp $\frac{\text{cal}}{\text{gram}}$ °C	
260 300 340 380 420 460 500 540 580 620	0.07103 0.07142 0.07197 0.07261 0.07303 0.07409 0.07540 0.07669 0.07758 0.07837	60 100 140 180 220 260 300 340 380 420	0.07017 0.06916 0.06964 0.06959 0.07015 0.07015 0.07116 0.07169 0.07199 0.07226	
660 710 760 800 840	0.07886 0.08011 0.08111 0.08074 0.08028	460 500 580 620 660 700 740 780 820 860	0.07285 0.07384 0.07444 0.07501 0.07538 0.07591 0.07659 0.07712 0.07773 0.07922 0.07992	

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 7.77 WEIGHT PER CENT INDIUM ALLÓY

TABLE XIII

Run 71		Run 72	ж	Run 73	
EI = 2.5	56 watts	EI = 3.9	27 watts	EI = 3.91	L8 watts
Т°С	Cp cal gram °C	Т°С	Cp <u>cal</u> gram <u>°c</u>	Τ°C	Cp cal gram °C
50 70 90 110 130 150 170 190 210 230	0.07175 0.07316 0.07215 0.07205 0.07208 0.07242 0.07267 0.07267 0.07327 0.07387 0.07487	60 100 140 180 220 260 300 340 380 420	0.07306 0.07201 0.07154 0.07209 0.07378 0.07673 0.07892 0.08219 0.08219 0.08637 0.09077	80 120 160 200 240 280 320 360 400 440	0.07118 0.07117 0.07173 0.07271 0.07463 0.07749 0.08039 0.08405 0.08425 0.08825 0.09178
250 270 290 310 330	0.07645 0.07740 0.07823 0.07995 0.08144	460 500 540 580	0.09077 0.07661 0.07638 0.0774	480 520 560 600 640	0.08355 0.07546 0.07717 0.07806 0.07917
350 370 390	0.08322 0.08526 0. 0 8735			680 710 730 750 770	0.07962 0.08185 0.08706 0.09765 0.09795
				790	0.11249

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 300 PPM HYDROGEN ALLOY

TABEE XIV

Run 67		Run 69		Run 70	cont.	
EI = 1.	380 watts	EI = 3.	EI = 3.850 watts		EI = 2.455 watts	
Т°С	Cp <u>cal</u> gram °C	Τ°C	Cp cal gram°C	Τ°C	Cp cal °C	
60 80 100 120 140	0.06948 0.06713 0.06671 0.06714 0.06813	70 110 150 190 230	0.06987 0.06943 0.06896 0.06888 0.07008	300 340 380 420 Specime	0.07073 0.06304 0.04733 0.05463 en held 2 hr	
160 180 200 220 240	0.06932 0.06990 0.07029 0.07136 0.07228	270 310 350 390 430	0.07253 0.07672 0.07180 0.05491 0.06347	450 470 490 510 530	0.10552 0.13819 0.16627 0.12829 0.07760	
260 280 Run 68	0.07308 0.07369	470 510 550 590 630	0.10815 0.09580 0.06921 0.06854 0.06829	550 570 590 620 660	0.07134 0.07038 0.07001 0.06949 0.06894	
EI = 1.	402 watts	0 90	0.0002)	000	010000/1	
T°C	Cp cal gram °C	670 710 750	0.06847 0. 06 897 0.06800	Run 77 EI = 2.	396 watts	
60 80 100	0.06940 0.06940 0.06830	790 Run 70	0.06729	T°C	Cp <u>cal</u> gram °C	
120 140	0.07058 0.07047	EI = 2.	455 watts	60 100	0.06960 0.06956	
160 180	0.07052 0.07069	T°C	Cp cal gram °C	140 180 220	0.06979 0.06869 0.06890	
200 220 240	0.07075 0.07136 0.07160	100 140 180	0.06998 0.06991 0.06991	260 300 340	0.07039 0.07156 0.07110	
260 280	0.07264 0.07363	220 260	0.07113 0.07269	380 420	0.06968 0.06938	

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 17.5 WEIGHT PER CENT NIOBIUM ALLOY

TABLE XIV (Continued)

	and the second	a and the second statement of the second statements and the second statements and the second statements and the	the second s			
Run 77	cont.	Run 79	Run 79			
EI = 2.	396 watts	EI = 2.3	394 watts			
Т°С	Cp cal gram °C	Τ°C	Cp <u>cal</u> gram °C			
460 500 540 570 590	0.07213 0.07041 0.07036 0.07070 0.07259	100 140 180 220 260	0.06913 0.06892 0.06905 0.07020 0.07073			
610 630 650 670 690	0.07513 0.07929 0.08820 0.09802 0.10367	300 345 390 430 470	0.06868 0.06973 0.06589 0.06340 0.09606			
710 730 750 770 790	0.09923 0.090 <u>1</u> 5 0.08110 0.07255 0.07041	510 550 590 630 670 710	0.09220 0.06806 0.06620 0.07337 0.07235 0.06855			

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 17.5 WEIGHT PER CENT NIOBIUM ALLOY

TABLE	XV
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-	an a		ann an tao an ann an tao ann an tao ann an tao a T		an a
Run 38 Run 38 cont.		Run 39	cont.		
EI = 2	.338 watts	EI = 2	.338 watts	EI = 3.	705 watts
Т°С	$Cp \frac{cal}{gram} \circ C$	Τ°C	$Cp \frac{cal}{gram} °C$	Τ°C	$Cp \frac{cal}{gram} \circ C$
50 70 90 110 130	0.1397 0.1387 0.1400 0.1404 0.1359	650 670 690 710 730	0.1514 0.1521 0.1526 0.1536 0.1545	560 600 640 680 720	0.1508 0.1527 0.1532 0.1539 0.1552
150 170 190 210 230	0.1357 0.1363 0.1369 0.1379 0.1388	750 770 790 810 830	0.1558 0.1588 0.1623 0.1652 0.1665	750 770 790 810 830	0.1563 0.1588 0.1610 0.1632 0.1641
250 270 290 310	0.1394 0.1404 0.1411 0.1415	TRANSF 900 920	ORMATION 0.1841 0.1636	850 865 875	0.1644 0.1656 0.1830
350 370 390 410	0.1431 0.1440 0.1447 0.1451	940 960 Run 39 EI = 3	0.1502 0.1502	910 930 950	0.1709 0.1608 0.1538
430 450	0.1456 0.1463	T°C	Cp cal gram °C	Run 40 EI = 3.	674 watts
490 510 530	0.1489 0.1471 0.1483 0.1488	160 200 240 280	0.1370 0.1371 0.1394 0.1413	T°C 90 130	Cp <u>cal</u> <u>gram</u> °C 0.1336 0.1355
550 570 590 610 630	0.1490 0.1497 0.1502 0.1507 0.1513	320 360 400 440 480 520	0.1428 0.1444 0.1454 0.1465 0.1479 0.1494	170 210 250	0.1368 0.1380 0.1397

TRUE SPECIFIC HEAT VALUES OF IODIDE TITANIUM

TABLE XV (Continued)

Run 40 cont.		Run 42		Run 42 cont.	
EI = 3.67	4 watts	EI = 2.30	7 watts	EI = 2.307	7 watts
Т°С	Cp cal gram °C	Τ°C	Cp cal gram °C	Т [®] С	Cp <u>cal</u> gram °C
290 330 370 410 450	0.1416 0.1434 0.1447 0.1459 0.1473	60 80 100 120 140	0.1312 0.1321 0.1328 0.1336 0.1351	660 680 700 720 740	0.1521 0.1556 0.1561 0.1557 0.1557
490 530 570 610 650	0.1485 0.1500 0.1516 0.1532 0.1547	160 180 200 220 240	0.1357 0.1363 0.1370 0.1380 0.1386	760	0.1583
690 730 770 Run 41	0.1562 0.1570 0.1 60 1	260 280 300 320 340	0.1402 0.1407 0.1415 0.1427 0.1437		al.
EI = 5.14	7 watts	360 380	0.1443 0.1452		
T°C	Cp <u>cal</u> gram °C	400 420	0.1457 0.1465		
360 400 440 480	0.1436 0.1448 0.1459 0.1471	440 460 480 500	0.1472 0.1480 0.1486 0.1495		
520 560 600	0.1480 0.1491 0.1511	540 560	0.1501		
640 680	0.1521 0.1533	580 600 620	0.1517 0.1521 0.1522		
720 760 800 840	0.1573 0.1613 0.1640	640	0.1222		

TRUE SPECIFIC HEAT VALUES OF IODIDE TITANIUM

TABLE XVI

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 34.4 WEIGHT PER CENT TITANIUM ALLOY

 Run 43		Run 43	Run 43 cont.		cont.
EI = 3.	120 watts	EI = 3.	120 watts	EI = 3.	110 watts
T°C	Cp cal gram °C	Τ°C	Cp $\frac{ ext{cal}}{ ext{gram}}$ °C	Τ°C	Cp <u>cal</u> gram °C
50 70 90 110 130	0.09685 0.09787 0.09741 0.09826 0.09952	592.5 597.5 602.5 607.5 612.5	0.21651 0.30203 0.44915 0.63227 0.41325	270 310 350 390 430	0.09498 0.09810 0.09929 0.10169 0.10304
150 170 190 210 230	0.10099 0.10186 0.10243 0.10182 0.10084	617.5 635 660 680 700	0.19066 0.21250 0.10063 0.09974 0.09885	470 510 550 580 595	0.10374 0.10420 0.10737 0.12580 0.13778
250 270 290 310 330	0.09966 0.09938 0.09893 0.09889 0.09972	720 740 760 780 Run 45	0.09789 0.09671 0.09554 0.09557	tranf of 640 660 680 700	RMATION 0.12261 0.09948 0.09816 0.09680
350 370 390 410 430	0.09877 0.09882 0.09942 0.10088 0.10230	EI = 3. T°C	llO watts Cp <u>cal</u> gram °C	720 750 790 830	0.09634 0.09541 0.09427 0.09274
450 470 490 510 530	0.10314 0.10369 0.10475 0.10620 0.10957	60 80 100 120 140	0.09502 0.09434 0.09532 0.09669 0.09762	870 900 920 Rerun d	0.08932 0.08640 0.08695 of the ß region
545 555 565 575 585	0.11257 0.11741 0.12112 0.13263 0.16226	160 1 8 0 200 220 240	0.09754 0.09523 0.09471 0.09516 0.09597	680 700 720	0.09682 0.09573 0.09340

TABLE XVI (Continued)

Run 45	cont.	Run 46	cont.	Run 48		
EI = 3	.110 watts	EI = 3	.094 watts	EI = 4.	EI = 4.398 watts	
Т°С	Cp <u>cal</u> gram °C	Т°С	Cp $\frac{\text{cal}}{\text{gram}}$ °C	Т°С	$Cp \frac{cal}{gram} C$	
750 790 830 870 910	0.09236 0.09067 0.08961 0.08605 0.08464	830 870 Rerun 680	0.09011 0.08844 of β region 0.09841	220 260 300 340 380	0.09812 0.09769 0.09 8 39 0.09940 0.10131	
Run 46		720 760 800	0.09501 0.09128 0.08679	420 460	0.10270	
EI = 3	.094 watts	000	0.00079	500	0.10329	
Т°С	Cp <u>cal</u> gram °C	840 880	0.08456 0.08338	540 580	0.10442 0.11632	
1.30	0.09399	Run 47		TRANSFO	RMATION	
170 210 250 290	0.09443 0.09493 0.09644	EI = 4 T°C	.415 watts Cp <u>gram</u> °C	660 700 740 780	0.10646 0.09817 0.09697 0.09548	
330 370	0.09905 0.10052	100 140 180	0.09456 0.09317 0.09354	820 860	0.09325	
410 450 490	0.10313 0.10396 0.10 459	220 260	0.09469 0.09634	Run 49 EI = 4.	398 watts	
530	0.10628	300 340	0.09820 0.09941	Т°С	Cp <u>cal</u> gram °C	
560 590 595	0.10728 0.12245 0.12838	380 420 460	0.10121 0.10251 0.10313	80 120 160	0.09547 0.09380 0.09319	
TRANSFO	DRMATION	500	0.10351	240	0.09305	
640 670 710 750 790	0.15559 0.10334 0.09983 0.09803 0.09560	580	0.11581	280 320 360 400 440	0.09639 0.09748 0.09872 0.10099 0.10206	

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 34.4 WEIGHT PER CENT TITANIUM ALLOY

TABLE XVI (Continued)

 	and and a second s	Sector and the sector of the	
Run 49 cont.		Run 79	
EI = 4.37	2 watts	EI = 3.49	95 watts
T°C	Cp cal gram °C	Т°С	Cp <u>cal</u> gram °č
480 520 560 TRANSFORM	0.10273 0.10314 0.10659 MATION	60 100 140 180 220	0.09515 0.09362 0.09401 0.09416 0.09479
660 700 740 780 820 860 Rerun of	0.10528 0.09720 0.09610 0.09568 0.09400 0.09248 β region	260 300 340 380 420 460 500	0.09615 0.08764 0.09807 0.09635 0.09635 0.10049 0.10398
690 730 770 810 850 890	0.09421 0.09064 0.08922 0.08675 0.08488 0.08253	540 580 620 660 700 740 780	0.10728 0.12285 0.31808 0.10640 0.09950 0.09756 0.09613

TRUE SPECIFIC HEAT VALUES OF A ZIRCONIUM - 34.4 WEIGHT PER CENT TITANIUM ALLOY

TABLE XVII

Run 18		Run 63		Run 65	
Τ°C	$H-H_{800} \frac{cal}{g \text{ atom}}$	Τ°C	$H-H_{800} \frac{cal}{g \text{ atom}}$	Д°С	H-H ₈₀₀ <u>cal</u>
800 810 820 830 840 850 855 860 865 870 875	<pre> 0.00 86.71 173.88 261.93 351.81 446.15 498.73 565.57 724.44 1504.86 1621.68 </pre>	800 820 860 870 880 900 920 940 950	0.00 161.49 536.44 1238.23 1569.72 1745.53 1884.11 2017.35 2085.28	800 810 850 860 870 880 900 920 940 950	0.00 83.49 436.42 558.40 1130.42 1607.66 1788.59 1926.97 2058.82 2127.02
885 900 920	1701 .89 1811.99 1954.28	run 64 T°C	H-H ₈₀₀ cal g atom		
940 950	2093.17 2161.52	800	0.00		
Run 62		840 860 870	327.16 533.94		
Т°С	$H-H_{800} \frac{cal}{g \text{ atom}}$	880 890	1529.39		
800 820 860 970 890 910 930 950	0.00 160.29 552.93 1444.61 1698.52 1847.60 1989.04 2130.79	910 930 950	1760.80 1891.98 2021.98		

TRANSFORMATION ENTHALPY DATA FOR IODIDE ZIRCONIUM

TABLE XVIII

Run 15		Run 50		Run 52	
Τ°C	$H-H_{800} \frac{\text{cal}}{\text{g atom}}$	₽°C	$H-H_{800} \frac{cal}{g atom}$	Т°С	$H-H_{800} \frac{cal}{g \text{ atom}}$
800 820 830 840 845 850 855 860 865 870 885 880 885 880 900 910 920 930 940 950	0.00 190.23 289.26 390.17 445.55 507.61 587.65 720.97 1035.80 1285.44 1443.67 1541.93 1609.79 1666.83 1750.80 1822.66 1889.07 1954.35 2019.31 2082.54	800 820 840 860 880 900 940 950	0,00 173.79 382.23 759.05 1568.12 1730.60 2009.57 2078.28	800 820 830 840 850 860 870 880 900 920 940 950	0.00 184.50 289.20 399.40 540.34 858.21 1385.76 1582.97 1749.13 1885.47 2016.00 2082.26

TRANSFORMATION ENTHALPY DATA FOR A ZIRCONIUM - 0.881 WEIGHT PER CENT SILVER ALLÓY

TABLE XIX

Run 3	2 ·	Run 37	
Τ°C	H-H800 cal gratom	Τ°C	$H-H_{800} \frac{cal}{g \text{ atom}}$
800 810 825 820 825 830 840 860 880 900	0.00 98.36 164.17 295.12 824.63 1225.71 1488.64 1700.44 1889.99 2055.05	800 810 820 825 830 840 860 900	0.00 93.81 234.06 525.41 989.63 1361.70 1643.28 2051.28

TRANSFORMATION ENTHALPY DATA FOR A ZIRCONIUM - 5.37 WEIGHT PER CENT SILVER ALLOY

TABLE XX

	-7.77	WEIGHT	PER CENT INDIUM	ALLOY	
Run 5 T°C	H-H800 cal g atom	Run 55 T°C	H-H ₈₀₀ cal g atom	Run 56 T°C	H-H ₈₀₀ cal g atom
800 820 840 860 880 900 910 920 930 935 940 950 960	0.00 152.74 305.53 455.71 613.24 700.36 792.20 899.37 1071.94 1536.34 1733.23 1858.87 1998.68 2095.95	800 820 860 880 900 920 940 960	0.00 149.32 451.29 612.13 780.05 1056.83 1801.99 2058.44	800 810 850 880 890 910 930 950 960	0.00 76.07 388.83 625.13 707.41 942.32 1520.73 1975.99 2073.75

TRANSFORMATION ENTHALPY DATA FOR A ZIRCONIUM -7.77 WEIGHT PER CENT INDIUM ALLOY

Run . 38		Run 39	
Т°С	H-H ₈₀₀ $\frac{\text{cal}}{\text{g atom}}$	Τ°C	$H-H_{800} \frac{\text{cal}}{\text{g atom}}$
800 820 840 860 870 880 885 890 910 930	0.00 162.70 323.50 483.95 564.60 724.71 1398.99 1647.04 1808.36 1945.90	800 820 840 860 870 880 885 890 900 920 920 930	0.00 160.69 320.31 480.22 560.78 6 ^{1,} 9.78 1074.35 1540.48 1703.29 1855.64 1927.21

TRANSFORMATION ENTHALPY DATA FOR IODIDE TITANIUM

TABLE XXI

TABLE XXII

			5-10-11-15,		
Run 4	3	Run 45		Run 46	
Τ°C	$H-H_{500} \frac{\text{cal}}{\text{g atom}}$	Τ°C	$H-H_{500} \frac{cal}{g \text{ atom}}$	Τ°C	$H-H_{500} \frac{cal}{g \text{ atom}}$
500 520 540 550 560 570 580 590 595 600 605 610 615 620 650 670 690 700	0.00 154.11 312.87 394.32 479.13 566.50 661.78 778.34 856.11 964.59 1125.92 1353.03 1501.46 1569.94 1798.92 1945.35 2090.53 2162.50	500 530 570 590 600 610 620 630 650 670 690 700	0.00 227.02 538.47 719.68 818.66 955.63 1326.10 1641.73 1818.52 1963.34 2106.34 2176.90	500 510 550 590 600 610 620 630 650 690 700	0.00 75.92 383.36 541.02 718.78 811.16 918.24 1105.66 1621.67 1846.74 2147.06 2219.71

TRANSFORMATION ENTHALPY DATA FOR A ZIRCONIUM - 34.4 WEIGHT PER CENT TITANIUM ALLOY