

CONVECTION HEAT AND MASS TRANSFER IN
A HYDROMAGNETIC FLOW OF A MICROPOLAR FLUID
OVER A POROUS MEDIUM

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Convection heat and mass transfer in a hydromagnetic flow of a micropolar fluid over a porous medium

B.I.Olajuwon* J.I.Oahimire† M.A.Waheed‡

Abstract

This study presents a mathematical analysis of a hydromagnetic boundary layer flow, heat and mass transfer characteristics on steady two-dimensional flow of a micropolar fluid over a stretching sheet embedded in a non-Darcian porous medium with uniform magnetic field in the presence of thermal radiation. The governing system of partial differential equations is first transformed into a system of non-linear ordinary differential equation using the usual similarity transformation. The resulting coupled non-linear ordinary differential equations are then solved using perturbation technique. With the help of graphs, the effects of the various important parameters entering into the problem on the velocity, temperature and concentration fields within the boundary layer are separately discussed. The effects of the pertinent parameters on the wall temperature, wall solutal concentration, skin friction coefficient and the rate of heat and mass transfer are presented numerically in tabular form. The results obtained showed that these parameters have significant influence on the flow.

Keywords: mass transfer, hydromagnetic flow, porous medium, radiation, perturbation technique, heat transfer, stretching sheet

1 Introduction

The study of heat and mass transfer has attracted the interest of numerous researchers due to its application in sciences and engineering problems. Such

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applications include nuclear reactor, MHD generators, geothermal energy extractions, analyzing the behaviour of exotic lubricant, the flow of colloidal suspension or polymeric fluid and the boundary layer controlling the field of aerodynamics. In nature and industrial applications many transport processes exist where the transfer of heat and mass takes place simultaneously as a result of combined buoyancy effects of thermal diffusion and diffusion of chemical species. In chemical process industries such as food processing and polymer production, the phenomenon of heat and mass transfer is also encountered. Rebhi (2007) studied unsteady natural convection heat and mass transfer of micropolar fluid over a vertical surface with constant heat flux. The governing equations were solved numerically using McCormack's technique and effects of various parameters were investigated on the flow. Vian et al (2003) studied the unsteady boundary layer flow of a micropolar fluid, which started impulsively in motion with a constant velocity from rest near the rear stagnation point of an infinite plane wall. The transformed non-similar boundary-layer equations were solved numerically using a very efficient finite-difference method known as Keller-box method. Eldabe and Ouaf (2006) solved the problem of heat and mass transfer in a hydromagnetic flow of a micropolar fluid past a stretching surface with ohmic heating and viscous dissipation using the Chebyshev finite difference method. Keelson and Desseaux (2001) studied the effect of surface conditions on the flow of a micropolar fluid driven by a porous stretching surface. The governing equations were solved numerically. Sunil et al (2006) studied the effect of rotation on a layer of micropolar ferromagnetic fluid heated from below saturating a porous medium. The resulted non-linear coupled differential equations from the transformation were solved using finite-difference method. Rahman and Sultan (2008) studied the thermal radiation interaction of the boundary layer flow of micropolar fluid past a heated vertical porous plate embedded in a porous medium with variable suction as well as heat flux at the plate. The governing equations were solved numerically by an efficient, iterative, finite – difference method. Mahmoud (2007) investigated thermal radiation effect on magneto hydrodynamic flow of a micropolar fluid over a stretching surface with variable thermal conductivity. The solution was obtained numerically by iterative, Runge-Kuta order-four method. Magdy (2004) studied unsteady free convection flow of an incompressible electrically conducting micropolar fluid, bounded by an infinite vertical plane surface of constant temperature with thermal relaxation including heat sources. The governing equations were solved using Laplace transformation. The inversion of the Laplace transforms was carried out with a numerical method. Chaudhary (2008) investigated the

effects of chemical reactions on magneto hydrodynamic micropolar fluid flow past a vertical plate in slip-flow regime. Heat and mass transfer effects on the unsteady flow of a micropolar fluid through a porous medium bounded by a semi-infinite vertical plate in a slip flow regime was studied taking into account a homogeneous chemical reaction of the first order. Finite difference method was used to obtain solution for the governing equations. Mohammed and Abo-Dahah (2009) investigated the effects of chemical reaction and thermal radiation on heat and mass transfer in magneto hydrodynamic micropolar flow over a vertical moving porous plate in a porous medium with heat generation. The solution was obtained numerically by finite-difference method. Bayomi et al (2009) consider magneto hydrodynamic flow of a micropolar fluid along a vertical semi-infinite permeable plate in the presence of wall suction or injection effects and heat generation or absorption. The obtained self-similar equation were solved numerically by an efficient implicit, iterative, infinite-difference method. Reena and Rana (2009) investigated double-diffusive convection in a micropolar fluid layer heated and soluted from below saturating a porous medium. A linear stability analysis theory and normal mode analysis method was used. Mohammed et al (2010) studied magneto hydrodynamic convection with thermal radiation and mass transfer of micropolar fluid through a porous medium occupying a semi-infinite region of the space bounded by an infinite vertical porous plate with constant suction velocity in the presence of chemical reaction, internal heat source, viscous and Darcy's dissipation. The highly non-linear coupled differential equations governing the boundary layer flow, heat and mass transfer were solved using finite difference method.

Heat and mass transfer in a hydromagnetic flow have many applications in science and engineering. This present model have applications in biomedical and engineering. For instance in the dialysis of blood in artificial kidney, flow in oxygenation, e.t.c. Engineering applications include the porous pipe design, design of filter, e.t.c. Motivated by the above previous works and possibly applications, the present paper studies heat and mass transfer in a hydromagnetic flow of a micropolar fluid over a porous medium using Boussineq model in the presence of uniform magnetic field. The transformed non-linear boundary layer equations together with the boundary conditions are solved analytically using perturbation technique.

2 Mathematical formulation

We consider a steady, two-dimensional mixed convection flow of an incompressible, electrically conducting micropolar fluid over a stretching sheet. The fluid flows towards a surface coinciding with the plane $y = 0$, the flow region $y > 0$. The origin is fixed as shown in Fig. 1.

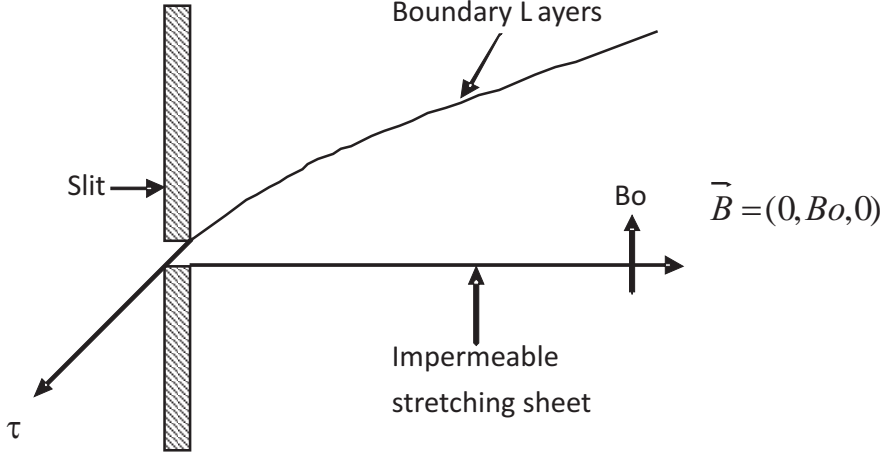


Figure 1: Physical Model

The x-axis is taken in the direction along the sheet and y-axis is taken perpendicular to it. The flow is generated by the action of two equal and opposite forces along the x-axis and the sheet is stretch in such a way that the velocity at any instant is proportional to the distance from the origin ($x=0$). The flow field is exposed to the influence of an external transverse magnetic field of strength $\vec{B} = (0, B_0, 0)$.

With these assumptions, the continuity equation, momentum equation, angular momentum equation, energy equation and mass diffusion equation governing the flow are:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0, \quad (1)$$

$$\begin{aligned} \frac{u\partial u}{\partial x} + \frac{v\partial u}{\partial y} = & \left(v + \frac{k_1^*}{\rho} \right) \frac{\partial^2 u}{\partial y^2} + \frac{k_1^*}{\rho} \frac{\partial N}{\partial y} - \frac{v\varphi u}{k} - \frac{C_b}{\sqrt{k}} \varphi u^2 \\ & - \frac{\sigma}{\rho} B_0^2 u + g\beta_t (T - T_\infty) + g\beta_c (c - c_\infty), \end{aligned} \quad (2)$$

$$\rho j \left(u \frac{\partial N}{\partial x} + v \frac{\partial N}{\partial y} \right) = \gamma \frac{\partial^2 N}{\partial y^2} - k_1^* \left(2N + \frac{\partial u}{\partial y} \right), \quad (3)$$

$$\begin{aligned} \frac{u \partial T}{\partial x} + \frac{v \partial T}{\partial y} &= \frac{1}{\rho C_p} \frac{\partial}{\partial y} \left(\kappa \frac{\partial T}{\partial y} \right) - \frac{1}{\rho C_p} \frac{\partial q_r}{\partial y} \\ &+ \frac{\sigma B_0^2}{\rho C_p} u^2 + \frac{q^{111}}{\rho C_p} + \frac{u}{\rho C_p} \left(\frac{\partial u}{\partial y} \right)^2, \end{aligned} \quad (4)$$

$$u \frac{\partial c}{\partial x} + v \frac{\partial c}{\partial y} = \frac{D \partial^2 c}{\partial y^2}, \quad (5)$$

where u and v are the velocity components along x and y directions, ρ is the density, T is the temperature of the fluid, C_b is the form of drag coefficient which is independent of viscosity and other properties of the fluid but is dependent on geometry of the medium, k is permeability of the porous medium, C_p is the specific heat at constant pressure, ν is the kinematic viscosity σ is the electrical conductivity of the fluid, N is the components of micro rotation or angular velocity whose rotation is in the direction of the x - y plane and j , γ and k_1^* are the microinertia per unit mass, spin gradient viscosity and vortex viscosity respectively. The spin gradient viscosity γ , which defines the relationship between the coefficient of viscosity and microinertia are as follows:

$$\gamma = \mu (1 + K/2) j \quad (6)$$

in which $K = k_1^*/\mu (> 0)$ is the material parameter. Here all the material constants, γ , μ , k , j are non-negative and we take $j = v/b$ is taken as a reference length. The appropriate physical boundary conditions for the problem under study are given by

$$u = u_w = bx, \quad v = 0, \quad N = -n \frac{\partial u}{\partial y} \quad \text{at } y = 0, \quad (7)$$

$$-k \frac{\partial T}{\partial y} = q_w = D_o \left(\frac{x}{l} \right)^2, \quad -D \frac{\partial c}{\partial y} = m_w = D_1 \left(\frac{x}{l} \right)^2 \quad \text{at } y = 0, \quad (8)$$

$$u \rightarrow 0, \quad N \rightarrow 0 \quad \text{as } y \rightarrow \infty, \quad (9)$$

$$T \rightarrow T_\infty, \quad C \rightarrow C_\infty \quad \text{as } y \rightarrow \infty, \quad (10)$$

where l is the characteristic length, T_w is the wall temperature of the fluid and T_∞ is the temperature of the fluid far away from the sheet, C_w is the wall

concentration of the solute and C_∞ is the concentration of the solute far away from the sheet, D_o, D_1 are constant and $k = k_\infty (1 + \varepsilon\theta(\eta))$ [Chiam(1998)]. n is a constant such that $0 \leq n \leq 1$. The case when $n = 0$, is called strong concentration, which indicates $N = 0$ near the wall represents concentrated particle flows in which the micro-elements close to the wall surface are unable to rotate [Mathur(1981)]. The case when $n = 1/2$ indicates the vanishing of anti-symmetric part of the stress tensor and denotes weak concentrations where as $n = 1$ is used for modelling of turbulent boundary layer flows [Ahamed (1976)]. It is worth mentioning that $k = 0$ describes the classical Navier-Stokes equation for a viscous and incompressible fluid. The non-uniform heat source/sink q^{111} is given by

$$q''' = \frac{kU_w}{xv} [A^*(T_w - T_\infty) f' + B^*(T - T_\infty)], \quad (11)$$

where A^* and B^* are the coefficients of space and temperature - dependent heat source/sink, respectively. The case $A^* > 0$ and $B^* > 0$ corresponds to internal heat generation while $A^* < 0$ and $B^* < 0$ corresponds to internal heat absorption.

We use the following similarity variables and dimensionless steam function to transform equation (2) and (3)

$$U = bxf(\eta), \quad v = \sqrt{b\nu}f(\eta), \quad \eta = \sqrt{\frac{b}{\nu}}y, \quad N = bx \left(\frac{b}{\nu}\right)^{1/2} g(\eta). \quad (12)$$

Substituting (12) into equation (2) and (3), we have

$$f'^2 - ff'' = (1 + K)f''' - Da^{-1}f' - af'^2 + Kg' - Ha^2f' + G_t\theta + G_c\phi, \quad (13)$$

$$f'g - fg' = \left(1 + \frac{K}{2}\right)g'' - K(2g + f''), \quad (14)$$

where $\alpha = \frac{C_b\phi x}{\sqrt{k}}$ is local inertia coefficient parameter [Mustafa(2006)], $Da^{-1} = \frac{\phi v}{kb}$ is inverse Darcy number, $Ha = \sqrt{\frac{\sigma}{\rho b}}B_0$ is the Hartmann number, $G_t = \frac{g\beta_t(T - T_\infty)}{b^2}$ is local temperature Grashof number, $G_c = \frac{g\beta_c(C - C_\infty)}{b^2}$ is local concentration Grashof number and $K = \frac{k_1^*}{\mu}$ is material parameter.

The appropriate boundary conditions (7) and (9) now become

$$\begin{aligned} f(\eta) = 0, \quad f'(\eta) = 1, \quad g(\eta) = -nf''(\eta) \quad \text{at } \eta = 0, \\ f'(\eta) \rightarrow 0, \quad g(\infty) \rightarrow 0 \quad \text{as } \eta \rightarrow \infty, \end{aligned} \quad (15)$$

where $n = 1/2$. If we take $g(\eta) = -\frac{1}{2}f''$, then combining (13) and (14), these two will reduce to a single non-linear ordinary differential equation as

$$f'^2 - ff'' = \left(1 + \frac{K}{2}\right)f''' - Da^{-1}f' - af'^2 - Ha^2f' + G_t\theta + G_c\phi, \quad (16)$$

subject to the appropriate boundary conditions

$$f(\eta) = 0, \quad f'(\eta) = 1 \text{ at } \eta = 0, \quad f'(\eta) \rightarrow 0 \text{ as } \eta \rightarrow \infty. \quad (17)$$

Following Rosseland approximation [Brewster(1972)] radiative heat flux q_r is modeled as

$$q_r = -\frac{4\sigma^*}{3k^*} \frac{\partial T^4}{\partial y}, \quad (18)$$

where σ^* is the Stefan – Boltzman constant and k^* is the mean absorption coefficient. Assuming that the difference in temperature within the flow are such that T^4 can be expressed as a linear combination of the temperature. We expand T^4 in Taylor series about T_∞ as follows

$$T^4 = T_\infty^4 + 4T_\infty^3(T - T_\infty) + 6T_\infty^2(T - T_\infty)^2 + \dots \quad (19)$$

Neglecting higher order terms beyond the first degree in $(T - T_\infty)$, we have

$$T^4 \approx -3T_\infty^4 + 4T_\infty^3T. \quad (20)$$

Differentiating equation(18) with respect to y and using(20) we obtain

$$\frac{\partial q_r}{\partial y} = -\frac{16T_\infty^3\sigma^*}{3k^*} \frac{\partial^2 T}{\partial y^2}. \quad (21)$$

Substituting equation(21) into equation(4), we have

$$\begin{aligned} U \frac{\partial T}{\partial x} + V \frac{\partial T}{\partial y} &= \frac{1}{\rho C_p} \frac{\partial}{\partial y} \left[\left(K + 16 \frac{T_\infty^3 \sigma^*}{3k^*} \right) \frac{\partial T}{\partial y} \right] \\ &+ \frac{\sigma B_0^2 U^2}{\rho C_p} + \frac{q'''}{\rho C_p} + \frac{\mu}{\rho C_p} \left(\frac{\partial U}{\partial y} \right)^2. \end{aligned} \quad (22)$$

The thermal boundary conditions for solving (22) depend on the type of heating process considered. Now, the non-dimensional temperature $\theta(\eta)$ and concentration $\phi(\eta)$ are defined (in PHF case) as

$$\theta(\eta) = \frac{T - T_\infty}{T_w - T_\infty}, \quad \phi(\eta) = \frac{C - C_\infty}{C_w - C_\infty}, \quad (23)$$

where

$$T - T_\infty = \frac{D_0}{K_\infty} \left(\frac{x}{l}\right)^2 \sqrt{\frac{\nu}{b}} \theta(\eta) \quad \text{and} \quad T_w - T_\infty = \frac{D_0}{K_\infty} \left(\frac{x}{l}\right)^2 \sqrt{\frac{\nu}{b}}, \quad (24)$$

$$C - C_\infty = \frac{D_1}{D} \left(\frac{x}{l}\right)^2 \sqrt{\frac{\nu}{b}} \phi(\eta) \quad \text{and} \quad C_w - C_\infty = \frac{D_1}{D} \left(\frac{x}{l}\right)^2 \sqrt{\frac{\nu}{b}}. \quad (25)$$

Using equation (23) in equation (22), yields

$$(1 + N_r + \epsilon\theta)\theta'' + Pr(f\theta' - f'\theta) + \epsilon\theta^2 + PrHa^2 E_s f'^2 \\ (1 + \epsilon\theta)(Af' + B\theta) + PrE_s f''^2 \quad (26)$$

subject to the boundary conditions

$$\theta(\eta) = \frac{1}{1 + \epsilon} \quad \text{at} \quad \eta = 0, \quad \theta(\eta) \rightarrow 0 \quad \text{as} \quad \eta \rightarrow \infty. \quad (27)$$

Using equation (23) in equation (5), we have

$$\phi(\eta) = Sc(\phi''f - 2\phi f') = 0 \quad (28)$$

and the corresponding thermal boundary condition is

$$\phi(\eta) = -\frac{1}{1 + \epsilon}, \quad \phi(\eta) \rightarrow 0 \quad \text{as} \quad \eta \rightarrow \infty, \quad (29)$$

where $N_r = \frac{16T^3\sigma^*}{3K^*K_\infty}$ is thermal radiation parameter, $Pr = \frac{\mu C_p}{K_\infty}$ is Prandtl number, $E_s = E_C K_\infty \sqrt{\frac{b}{\nu}}$ is the scaled Eckert number, $E_C = \frac{b^2 l^2}{D_0 C_p}$ is the Eckert number and $S_c = \frac{\nu}{b}$ is the Schmidt number.

3 Method of solution

Equation (16), (26) and (28) are highly coupled non-linear ordinary differential equation and since $\epsilon \ll 1$, we assume a perturbation of the form

$$f = 1 + \epsilon f_1 + \epsilon^2 f_2, \quad (30)$$

$$\theta = -\frac{1}{1 + \epsilon} + \epsilon \theta_1 + \epsilon^2 \theta_2, \quad (31)$$

$$\phi = -\frac{1}{1 + \epsilon} + \epsilon \phi_1 + \epsilon^2 \phi_2, \quad (32)$$

where $f_0 = 1, \theta_0 = -\frac{1}{1+\varepsilon}, \phi_0 = \frac{1}{1+\varepsilon}$. Invoking equation (30)-(32) into equation (16), (26) and (28) and neglecting terms of $O(\varepsilon^3)$ and higher, we acquire the following sets of equations:

$$\left(1 + \frac{K}{2}\right) f_1''' + f_1'' - (Da^{-1} + Ha^2) f_1' = -G_t \theta_1 - G_c \phi_1, \quad (33)$$

$$(1 + N_r) \theta_1'' + Pr \theta_1' + B \theta_1 = -\frac{B}{(1+\varepsilon)^2}, \quad (34)$$

$$\phi_1'' + Sc \phi_1' = 0, \quad (35)$$

$$\begin{aligned} \left(1 + \frac{K}{2}\right) f_2''' + f_2'' - (Da^{-1} + Ha^2) f_2' = \\ f_1 f_1'' + f_1'^2 - \alpha f_1''^2 - G_t \theta_2 - G_c \phi_2, \end{aligned} \quad (36)$$

$$\begin{aligned} (1 + N_r) \theta_2'' + Pr \theta_2' + B \theta_2 = \frac{1}{(1+\varepsilon)} \theta_1'' - Pr f_1 \theta_1' - \frac{2}{(1+\varepsilon)} Pr f_1'^2 \\ - Pr Ha^2 E_s f_1'^2 - A f_1'^2 + \frac{2}{(1+\varepsilon)} \theta_1, \end{aligned} \quad (37)$$

$$\phi_2'' + Sc \phi_2' = -Sc \phi_1' f_1 - \frac{2}{(1+\varepsilon)} Sc f_1'^2. \quad (38)$$

Solving equations (33)-(38) with the boundary conditions lead to expressions for velocity, temperature and concentration distributions as follows:

$$\begin{aligned} f(\eta) = 1 + \varepsilon (A_3 + A_4 e^{-m_2 \eta} + A_6 e^{-m_1 \eta} + A_7 e^{-Sc \eta}) \\ + \varepsilon^2 (A_{25} + A_{26} e^{-m_2 \eta} + A_{27} \eta e^{-m_2 \eta} + A_{28} e^{-2m_2 \eta} \\ + A_{29} e^{-(m_1+m_2)\eta} + A_{30} e^{-(Sc+m_2)\eta} + A_{31} e^{-m_1 \eta} \\ + A_{32} e^{-2m_1 \eta} + A_{33} e^{-(Sc+m_1)\eta} + A_{34} e^{-Sc \eta} + A_{35} e^{-2Sc \eta}), \end{aligned} \quad (39)$$

$$\begin{aligned} \theta(\eta) = & -\frac{1}{(1+\varepsilon)} + \varepsilon \left(A_2 e^{-m_1 \eta} - \frac{1}{(1+\varepsilon)^2} \right) + \varepsilon^2 (A_{17} e^{-m_1 \eta} \\ & + A_{18} \eta e^{-m_1 \eta} + A_{19} e^{-(m_1+m_2)\eta} + A_{20} e^{-2m_1 \eta} + A_{21} e^{-(Sc+m_1)\eta} \quad (40) \\ & + A_{22} e^{-2m_2 \eta} + A_{23} e^{-(Sc+m_2)\eta} + A_{24} e^{-2Sc\eta}, \end{aligned}$$

$$\begin{aligned} \varphi(\eta) = & -\frac{1}{(1+\varepsilon)} + \varepsilon A_1 e^{-Sc\eta} + \varepsilon^2 (A_9 e^{-Sc\eta} + A_{10} \eta e^{-Sc\eta} \\ & + A_{11} e^{-(Sc+m_2)\eta} + A_{12} e^{-(Sc+m_1)\eta} + A_{13} e^{-2Sc\eta} \quad (41) \\ & + A_{14} e^{-2m_2 \eta} + A_{15} e^{-(m_1+m_2)\eta} + A_{16} e^{-2m_1 \eta}, \end{aligned}$$

with

$$\begin{aligned} m_1 &= \frac{Pr + \sqrt{Pr^2 - 4(1+N_r)B^*}}{2(1+N_r)}, \\ m_2 &= \frac{1 + \sqrt{1 + 4\left(1 + \frac{K}{2}\right)(Da^{-1} + Ha^2)}}{2\left(1 + \frac{K}{2}\right)}, \\ A_1 &= -\frac{1}{1+\varepsilon}; \quad A_2 = -\frac{1}{1+\varepsilon} + \frac{1}{(1+\varepsilon)^2}, \\ A_6 &= \frac{-GtA_2}{-\left(1 + \frac{K}{2}\right)m_1^3 + m_1^2 + (Da^{-1} + Ha^2)m_1}, \\ A_7 &= \frac{-GcA_1}{-\left(1 + \frac{K}{2}\right)Sc^3 + Sc^2 + (Da^{-1} + Ha^2)Sc}, \\ A_4 &= \frac{-1 + ScA_7 + m_1A_6}{m_2}, \\ A_3 &= -(A_4 + A_6 + A_7), \\ A_{10} &= \frac{Sc^2 A_1 A_3}{3Sc}, \\ A_{11} &= \frac{(1+\varepsilon)Sc^2 A_1 A_4 - 4Sc^2 m_2 A_4 A_7}{(1+\varepsilon)\left((Sc+m_2)^2 - Sc(Sc+m_2)\right)}, \\ A_{12} &= \frac{(1+\varepsilon)Sc^2 A_1 A_6 - 4Sc^2 m_1 A_6 A_7}{(1+\varepsilon)\left((Sc+m_1)^2 - Sc(Sc+m_1)\right)}, \end{aligned}$$

$$\begin{aligned}
A_{13} &= \frac{(1 + \varepsilon) Sc^2 A_1 A_7 - 2Sc^3 A_7^2}{(1 + \varepsilon) 2Sc^2}, \\
A_{14} &= \frac{-2Scm_2^2 A_4^2}{(4m_1^2 - 2Scm_2)}, \\
A_{15} &= \frac{-4Scm_1 m_2 A_4 A_6}{(1 + \varepsilon) ((m_1 + m_2)^2 - Sc(m_1 + m_2))}, \\
A_{16} &= \frac{-2Scm_1^2 A_6^2}{(1 + \varepsilon) (4m_1^2 - Scm_1)}, \\
A_9 &= - \left(\frac{1}{1 + \varepsilon} + A_{11} + A_{12} + A_{13} + A_{14} + A_{15} + A_{16}, \right), \\
A_{18} &= \frac{A_2 m_1^2 + (1 + \varepsilon) Pr A_2 m_1 A_3}{1 + \varepsilon}, \\
A_{19} &= \frac{(1 + \varepsilon) Pr A_2 m_1 A_4 - 2Pr + (1 + \varepsilon) Pr Ha^2 Es - (1 + \varepsilon) A^*}{(1 + \varepsilon) \left((1 + Nr) (m_1 + m_2)^2 - Pr (m_1 + m_2) + B^* \right)}, \\
A_{20} &= \frac{(1 + \varepsilon) Pr A_2 m_1 A_6 - 2Pr m_1^2 A_6^2 + (1 + \varepsilon) Pr H_a^2 Es m_1^2 A_6^2 - (1 + \varepsilon) Am_1^2 A_6^2}{(1 + \varepsilon) \left((1 + Nr) 4m_1^2 - 2Pr m_1 + B^* \right)}, \\
A_{21} &= \frac{(1 + \varepsilon) Pr A_2 m_1 A_6 - 4Pr Scm_1 A_6 A_7 + (1 + \varepsilon) 2Pr Es H_a^2 m_1 A_6 A_7 - 2(1 + \varepsilon) Am_1 A_6 A_7}{(1 + \varepsilon) \left((1 + Nr) (Sc + m_1)^2 - Pr (Sc + m_1) + B^* \right)}, \\
A_{22} &= \frac{-2Pr m_1^2 A_4^2 - (1 + \varepsilon) Pr H_a^2 Es m_2^2 A_4^2 + (1 + \varepsilon) Am_2^2 A_4^2}{(1 + \varepsilon) \left((1 + Nr) 4m_2^2 - 2Pr m_2 + B^* \right)}, \\
A_{23} &= \frac{-2m_2 A_4 Sc A_7 (2Pr + (1 + \varepsilon) Pr H_a^2 Es - (1 + \varepsilon) A^*)}{(1 + \varepsilon) \left((1 + Nr) (m_2 + Sc)^2 - Pr (m_2 + Sc) + B^* \right)}, \\
A_{24} &= \frac{-Sc^2 A_7^2 (2Pr + (1 + \varepsilon) Pr H_a^2 Es - (1 + \varepsilon) A^*)}{(1 + \varepsilon) \left((1 + Nr) 4Sc^2 - Pr Sc + B^* \right)}, \\
A_{17} &= - \left(\frac{1}{1 + \varepsilon} + A_{19} + A_{20} + A_{21} + A_{22} + A_{23} + A_{24} \right), \\
A_{27} &= \frac{-A_3 A_4 m_2^2}{\left(1 + \frac{K}{2}\right) 3m_2^2 - 2m_2 - (Da^{-1} + H_a^2)}, \\
A_{28} &= \frac{A_4^2 m_2^2 + (\alpha - 1) (m_2^2 A_4^2 + Gt A_{22} + Gc A_{14})}{\left(1 + \frac{K}{2}\right) 8m_2^3 - 4m_2^2 - (Da^{-1} + Ha^2) 2m_2},
\end{aligned}$$

$$A_{29} = \frac{A_4 A_6 (m_2^2 + m_1^2 + 2(\alpha - 1)m_1 m_2) + GtA_{19} + GcA_{15}}{(1 + \frac{K}{2})(m_1 + m_2)^3 - (m_1 + m_2)^2 - (Da^{-1} + Ha^2)(m_1 + m_2)},$$

$$A_{30} = \frac{A_4 A_7 (m_2^2 - Sc^2 - 2(\alpha - 1)m_2 Sc) + GtA_{23} + GcA_{11}}{(1 + \frac{K}{2})(Sc + m_2)^3 - (Sc + m_2)^2 - (Da^{-1} + Ha^2)(Sc + m_2)},$$

$$A_{31} = \frac{A_6 A_3 m_1^2 - GtA_{17}}{(1 + \frac{K}{2})m_1^3 - m_1^2 - (Da^{-1} + Ha^2)m_1},$$

$$A_{32} = \frac{A_6^2 m_1^2 (1 + (\alpha - 1)) + GtA_{20} + GcA_{16}}{(1 + \frac{K}{2})8m_1^3 + 4m_1^2 + 2(Da^{-1} + Ha^2)m_1},$$

$$A_{33} = \frac{A_6 A_7 (m_1^2 + Sc^2 + 2m_1(\alpha - 1) + GtA_{21} + GcA_{12}}{(1 + \frac{K}{2})(Sc + m_1)^3 - (Sc + m_1)^2 + (Da^{-1} - Ha^2)(Sc + m_1)},$$

$$A_{34} = \frac{A_7 A_3 Sc^2 + GcA_9}{(1 + \frac{K}{2})Sc^3 - Sc^2 + (Da^{-1} - Ha^2)Sc},$$

$$A_{35} = \frac{A_7^2 Sc^2 (1 + (\alpha - 1) + GtA_{24} + GcA_{13}}{(1 + \frac{K}{2})8SC^3 + 4Sc^2 + (Da^{-1} + Ha^2)2Sc},$$

$$A_{26} = \frac{1 + 2m_2 A_{28} + (m_1 + m_2)A_{29} + (Sc + m_2)A_{30} + m_1 A_{31}}{m_2}$$

$$+ \frac{2m_1 A_{32} + (Sc + m_1)A_{33} + ScA_{34} + 2ScA_{35}}{m_2},$$

$$A_{25} = -(A_{26} + A_{28} + A_{29} + A_{30} + A_{31} + A_{32} + A_{33} + A_{34} + A_{35}).$$

The physical quantities of most interest in science and engineering are the skin-friction coefficient C_f , Nusselt number N_u and Sherwood number S_h which are defined by the following relations:

$$C_f = \frac{\tau_w}{\rho U_w^2 / 2}, \quad (42)$$

whereas the skin-friction on the plate τ_w is given by

$$\tau_w = \{(\mu + k_1^*) \frac{\partial u}{\partial y}\}_{y=0}. \quad (43)$$

Substituting equation (12) in (42) and using (43), we have

$$C_f Re_x^{1/2} = (1 + K) f'', \tag{44}$$

where $Re_x = U_w x / \nu$ is the local Reynolds number.

The local Nusselt number is given by

$$Nu = \frac{\left\{ \frac{\partial T}{\partial y} \right\}_{y=0}}{(T_w - T_\infty) \sqrt{\frac{b}{\nu}}} = -\theta'(0) \tag{45}$$

and the local Sherwood number reads

$$Sh = \frac{\left\{ \frac{\partial C}{\partial y} \right\}_{y=0}}{(C_w - C_\infty) \sqrt{\frac{b}{\nu}}} = -\varphi'(0). \tag{46}$$

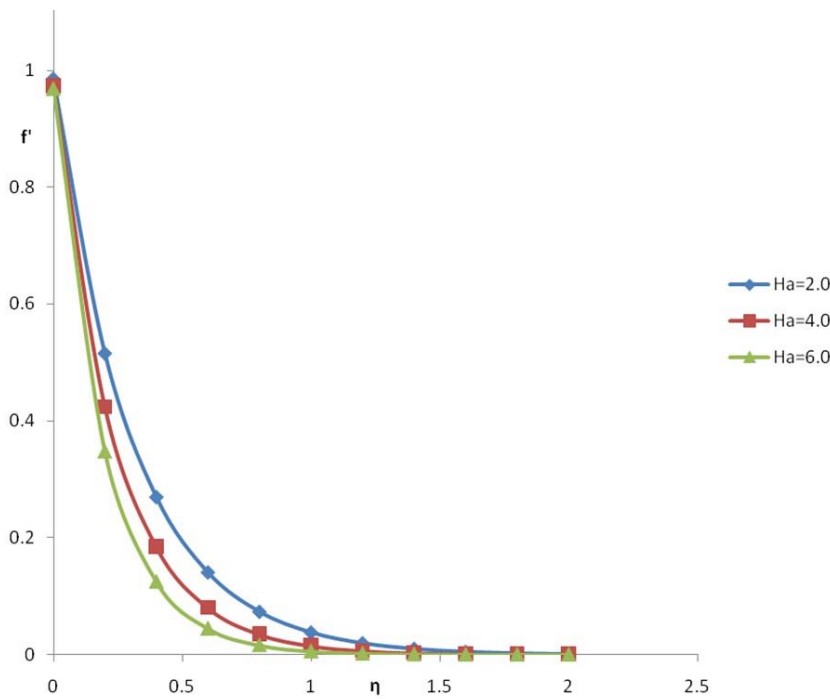


Figure 2: Variation of f' for different values of Ha when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Pr = 5.0, Da^{-1} = 0.5, Nr = 3.5, Gt = 6.0, Gc = 5.0, \varepsilon = 0.01, \alpha = 0.1$.

4 Results and discussion

The formulation of the hydromagnetic boundary layer flow, heat and mass transfer characteristics on steady two-dimensional flow with uniform magnetic field in the presence of thermal radiation of a micropolar fluid over a porous medium has been performed in the preceding sections. In order to understand the physical situation of the problem and hence the manifestation of the various parameters entering the problem, we have carried out the numerical calculations for distribution of velocity, temperature and concentration across the boundary layer for different values of the parameters. In this present study we have chosen $A^* = 0.01, B^* = 0.01, K = 0.2, Es = 0.05, \varepsilon = 0.01$ while $Ha, Gt, Gc, Da^{-1}, \alpha, Nr, Pr$ and Sc are varied over a range being listed in figures legends.

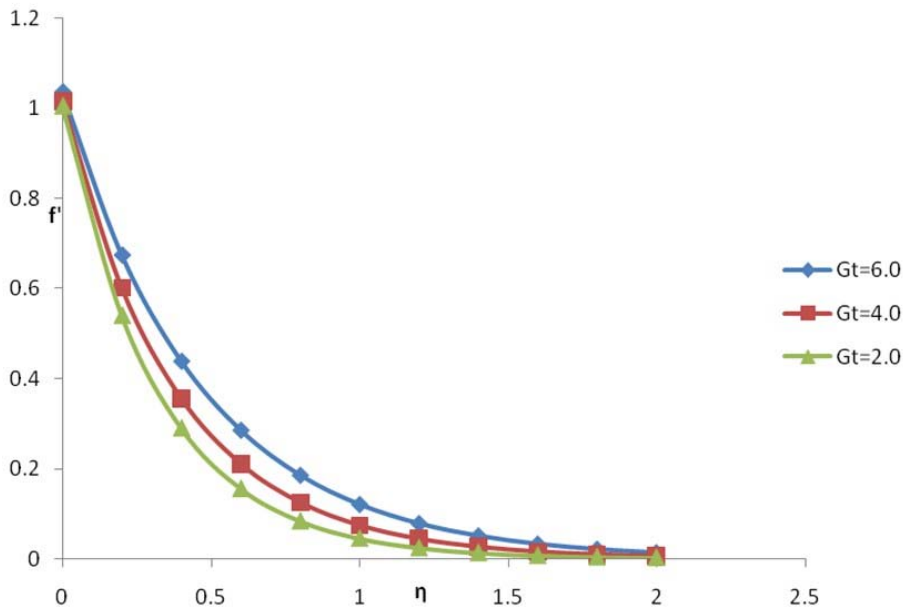


Figure 3: Variation of f' for different values of Gt when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Pr = 5.0, Da - 1 = 2.0, Nr = 3.5, Ha = 0.1, Gc = 0.1, \varepsilon = 0.01, \alpha = 0.1$.

Figure 2. shows the behaviour of velocity profile for different values of Hartmann number Ha . It is well known that the Hartmann number represents the importance of magnetic field on the flow. As depicted in figure 2., when

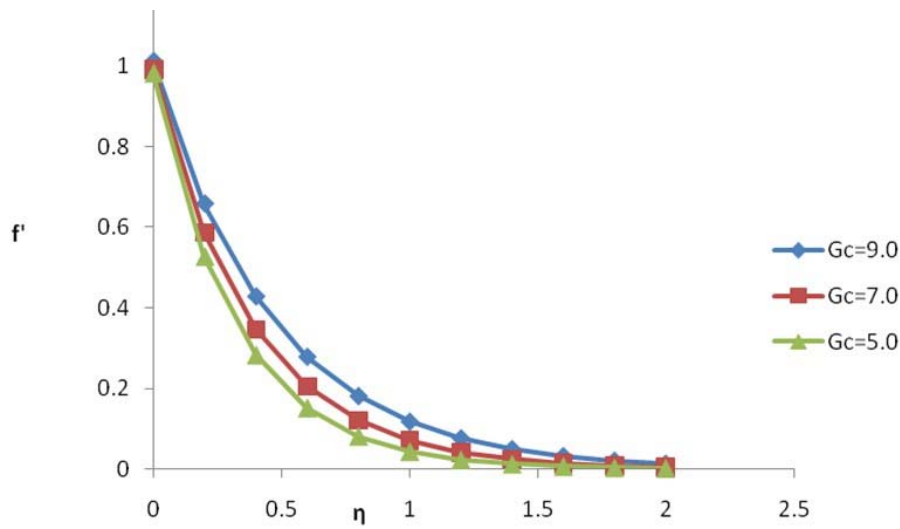


Figure 4: Variation of f' for different values of Gc when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Pr = 5.0, Da^{-1} = 2.0, Nr = 3.5, Ha = 0.1, \varepsilon = 0.01, Gt = 0.2, \alpha = 0.1$.

the Hartmann number increases, the velocity profile decreases. This is due to the fact that the introduction of transverse magnetic field normal to the flow direction has a tendency to create a drag due to Lorentz force and hence results in retarding the velocity profile. Thus when the Hartmann number increases, the Lorentz force also increases due to decrease of velocity profile. From figure 3. and figure 4., it is observed that the effect of increasing the value of the thermal Grashof number Gr and concentration Grashof number Gc is to increase the velocity profile.

Figure 5. illustrates the variation of velocity profile with η for various values of inverse Darcy number. The plot shows that velocity profile decreases with increase in the inverse Darcy number which shows the effect of increasing inverse Darcy number with decrease of the velocity profile. Similar effects are seen in case of increasing inertia coefficient parameter α as shown in figure 6. Figure 7. represents the temperature profiles for various values of thermal radiation parameter Nr in the boundary layer. This figure shows that the effect of thermal radiations to enhance heat transfer because of the fact that thermal boundary layer thickness increases with increase in the thermal radiation. Thus it is pointed out that the radiation should be minimized to have the cooling process at a faster rate. Figure 8. illustrates the variation of

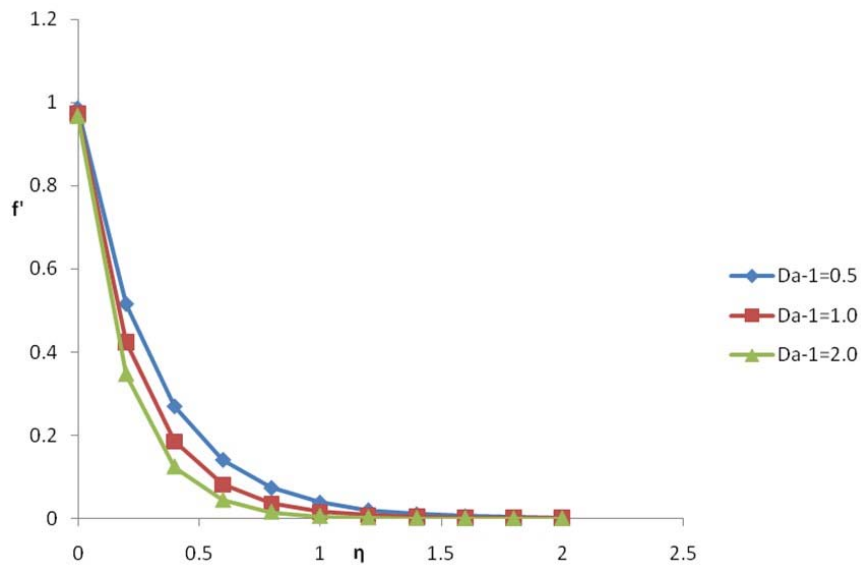


Figure 5: Plot of f' for different values of Da^{-1} when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Nr = 3.5, \alpha = 0.1, \varepsilon = 0.01, Gt = 0.2, Gc = 0.1$.

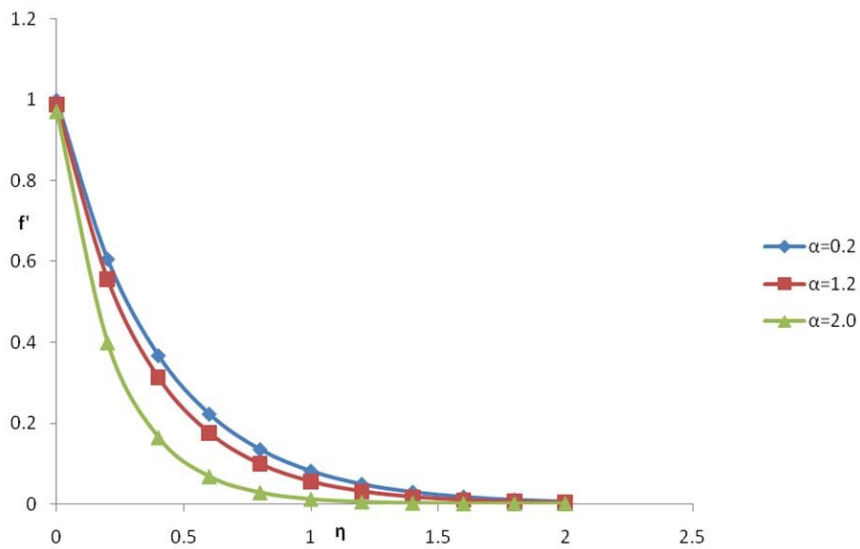


Figure 6: Variation of f' for different values of α when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Pr = 5.0, Ha = 0.1, Nr = 3.5, Gt = 6.0, Gc = 5.0, Da^{-1} = 2.0$.

temperature profile for various values of Prandtl number. It is seen that the temperature decreases with increasing the values of Prandtl number in the boundary layer. From this it is evident that temperature in the boundary layer falls very quickly for large value of the Prandtl number because of the fact that thickness of the boundary layer decreases with increase in the value of the Prandtl number. Figure 9. shows the variation of temperature profile with η for various values of inverse Darcy number Da^{-1} . It depicts that temperature increases with increase in the value of inverse Darcy number which is due to the fact that obstruction on the fluid motion is produced by the presence of porous medium which generates heat and thereby temperature increases in the thermal boundary layer.

Table 1: Values of $f''(0)$, $\theta(0)$, $\phi(0)$ for various different values of $K, Ha, Es, Gt, Gc, Pr, Nr$ and Da^{-1} when $A = 0.01, B = 0.01$ and $\varepsilon = 0.0$.

K	Ha	Es	Gt	Gc	Nr	Pr	Da ⁻¹	f''(0)	θ(0)	φ(0)
1.0	0.1	0.05	5.0	4.0	3.5	3.0	0.5	1.60434	0.70583	1.31521
2.0								1.32150	0.71848	1.35369
0.2	1.0							1.61046	0.74224	1.36108
	2.0							0.79668	0.87770	1.53207
		0.05						1.97707	0.71402	1.31493
		0.25						1.97922	0.71551	1.31356
			6.0					2.12904	0.70855	1.30228
			8.0					2.42051	0.69794	1.28087
				5.0				2.30513	0.69968	1.27993
				7.0				2.89612	0.65795	1.20833
					5.0			2.18918	0.83756	1.26950
					6.0			2.28906	0.89762	1.25801
						4.0		1.82115	0.60911	1.31143
						6.0		1.65949	0.50298	1.32867
							0.6	1.91600	0.69905	1.30365
							2.0	1.39126	0.73491	1.40513

Figure 10. shows that temperature increases with increase in the inertia coefficient parameter. In Figure 11. there is the plot of concentration distribution for various values of Schmidt number in the boundary layer. It is illustrated from the figure that the concentration decreases with increase in the value of Schmidt number. This is due to the fact that increase in Schmidt

number causes thinning of the solutal boundary layer thickness. It should be noted that the present results are in excellent agreement with the results reported by Abel et al(2008) as well as Dulal and Sewli (2009).

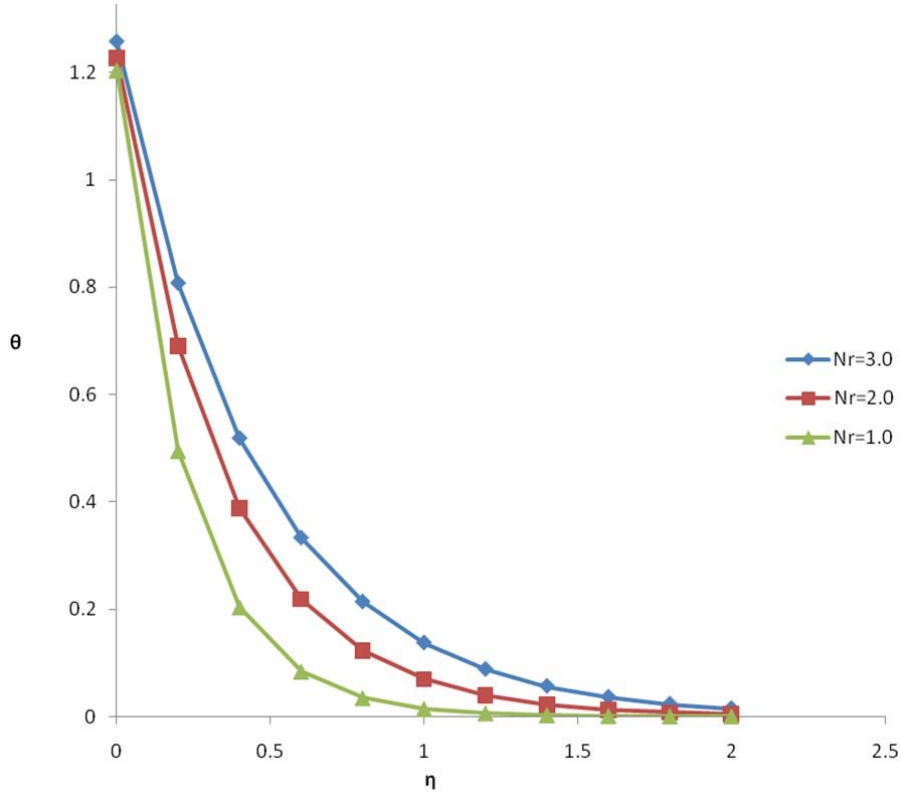


Figure 7: Variation of θ for different values of Nr when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Da - 1 = 0.5, Gt = 6.0, Ha = 0.2, \varepsilon = 0.01, \alpha = 0.1, Gc = 5.0$

Numerical values of the skin-friction coefficient $f''(0)$ the wall temperature $\theta(0)$ and the wall solutal concentration $\phi(0)$ are tabulated in table 1. for different values of material parameter K , Hartmann number Ha , Prandtl number Pr and inverse Darcy number Da^{-1} . The tabular data show that magnetic field, Prandtl number, inverse Darcy number and the material parameter reduce the skin-friction coefficient, whereas reverse trend is seen by increasing the values of Es, Gt, Gc and Nr . It is further observed that wall temperature increases with increase in K, Ha, Es, Nr and Da^{-1} whereas op-

posite effect is seen with increasing the value of $Gt, Gc,$ and Pr . The effect of increasing the values of K, Ha, Pr, Da^{-1} has the tendency to increase wall solutal concentration but the other parameters like Es, Gt, Gc and Nr have the effect of decreasing $\phi(0)$.

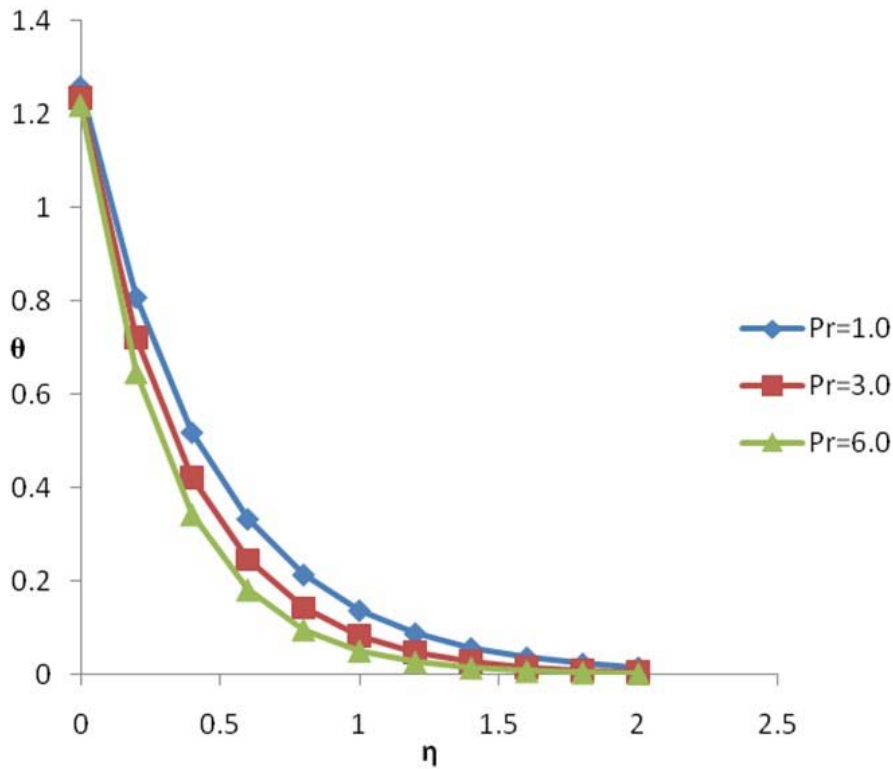


Figure 8: Variation of θ for different values of Pr when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Nr = 3.5, Da - 1 = 0.5, Gt = 6.0, Gc = 5.0, \epsilon = 0.01, \alpha = 0.1$.

Table 2. depicts the numerical values of coefficient of skin friction C_f , Nusselt number Nu and Sherwood number Sh for different values of Ha, Nr and Pr . The tabular data shows that Nusselt number and Sherwood number decrease with increasing in Hartmann number. On the other hand, Nusselt number increases as radiation parameter increases while the effect of increasing the value of Prandtl number is to decrease the skin friction coefficient and mass transfer rate but increase heat transfer rate which is in excellent agreement with Dulal and Sewli (2010) and Srinivasachanya and Ramreddy(2011).

Table 2: Values of C_f , Nu and Sh with different values of Ha , Nr and Pr when $A = 0.01$, $B = 0.01$, $K = 1.0$, $Es = 0.05$, $Gt = 5.0$, $Gc = 4.0$, $Da^{-1} = 0.5$, $\varepsilon = 0.01$ and $\varepsilon = 0.1$

Ha	Nr	Pr	C_f	Nu	Sh
0.0	3.5	3.0	1.97683	0.34286	0.47826
1.0	3.5	3.0	1.62968	0.24105	0.46217
2.0	3.5	3.0	0.80966	0.15098	0.41151
0.1	5.0	3.0	2.21928	0.12037	
0.1	6.0	3.0	2.32961	0.18692	
0.1	7.0	3.0	2.42161	0.24324	
0.1	3.5	4.0	1.84265	0.31832	0.20160
0.1	3.5	6.0	1.67848	0.41641	0.11923
0.1	3.5	8.0	1.58619	0.71502	0.01902

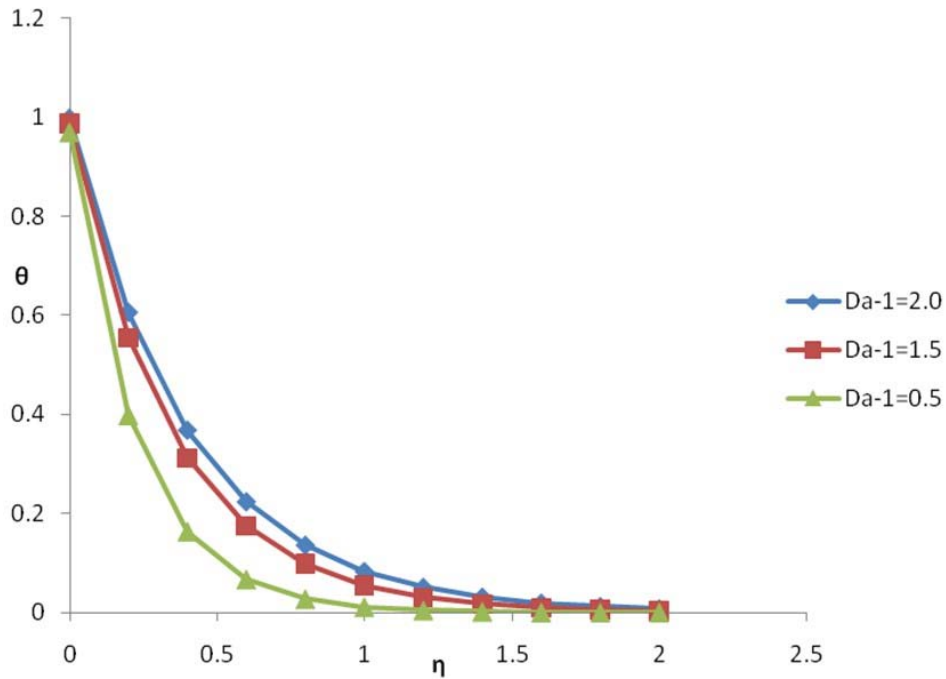


Figure 9: Variation of θ for different values of Da^{-1} when $K = 0.2$, $A = 0.01$, $B = 0.01$, $Es = 0.05$, $Pr = 5.0$, $Ha = 0.1$, $Nr = 3.5$, $Gt = 0.2$, $Gc = 0.1$, $\varepsilon = 0.01$.

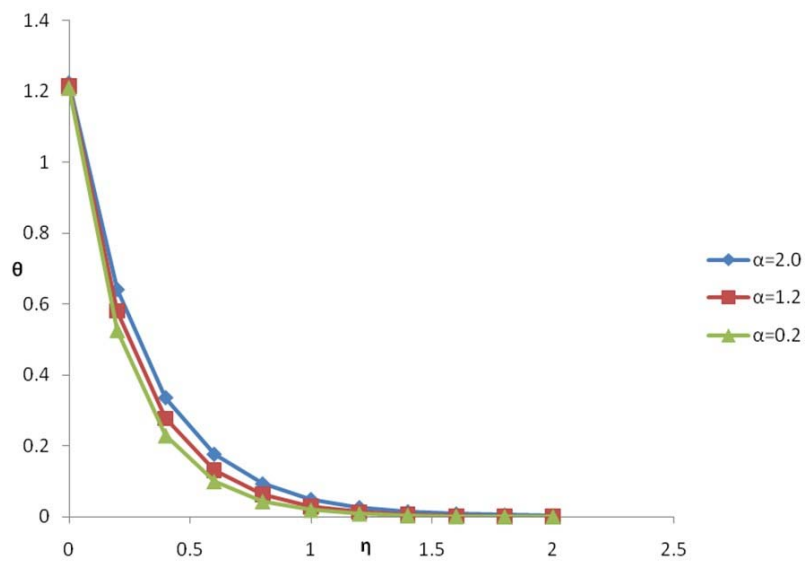


Figure 10: Plot of θ for different values of α when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Pr = 5.0, Ha = 0.1, Nr = 3.5, Gt = 6.0, Gc = 5.0, Da^{-1} = 2.0$.

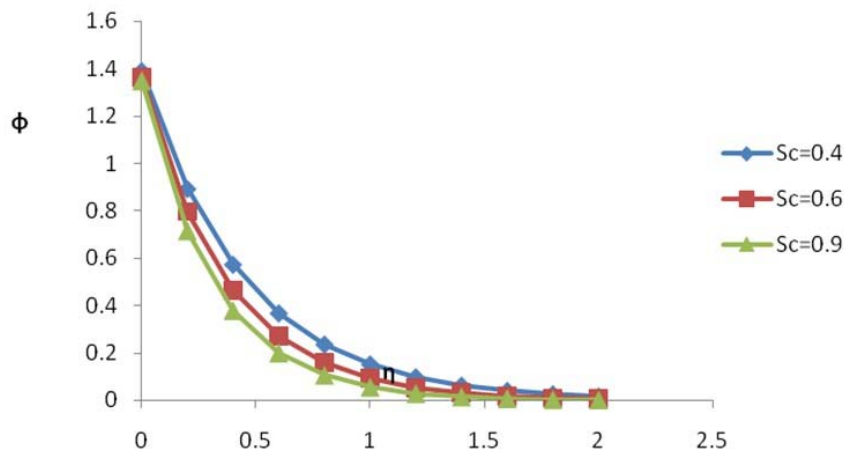


Figure 11: Variation of ϕ for different values of Sc when $K = 0.2, A = 0.01, B = 0.01, Es = 0.05, Pr = 5.0, Ha = 0.1, Nr = 3.5, Gt = 6.0, Gc = 5.0, Da^{-1} = 2.0$.

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Submitted in August 2013

Prenos toplote i mase strujanjem u hidromagnetskom tečenju mikropolarnog fluida preko porozne sredine

Ova studija predstavlja matematičku analizu tečenja u hidromagnetskom graničnom sloju, karakteristika prenosa toplote i mase na ravnomerna 2D strujanja mikropolarne tečnosti preko deformisane površi potopljene u ne-Darci poroznu sredinu sa ravnomernim magnetnim poljem u prisustvu toplotnog zračenja. Sistem parcijalnih diferencijalnih jednačina upravljanja je prvo preveden u sistem nelinearnih običnih diferencijalnih jednačina koristećenjem uobičajene transformacije sličnosti. Rezultujuće spregnute nelinearne obične diferencijalne jednačine se zatim rešavaju sredstvima perturbacione tehnike. Pomoću grafova, uticaji raznih vanih parametra na brzine, temperature i koncentracije polja u okviru graničnog sloja se posebno raspravljaju. Uticaji značajnih parametara na temperaturu zida, koncentracije na zidu, površinski koeficijent trenja, i brzinu promene prenosa toplote i mase su predstavljeni numerički u tabelarnom obliku. Rezultati su pokazali da ovi parametri imaju značajan uticaj na tečenje.