# Root Locus Plotter for a Dual Tank System Under Feedback Control 

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# ROOT LOCUS PLOTTER FOR A DUAL TANK SYSTEM UNDER FEEDBACK CONTROL 

BY
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## RESEARCH REPORT

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## ABSTRACT

A root locus graphics routine was written in Turbo Pascal for the analysis and design of a linearized dual tank control system. The routine is a subprogram to be incorporated with an editor written by L. Fadden. This editor allows for the saving and changing of parameters to the system.

The dual tank system is a good example for classical feedback control analysis. A brief description of the process and system is presented. The system may be described by linearized differential and algebraic equations. From these, a characteristic equation is derived, which gives rise to the root locus. The root locus is a plot of the poles of the closed loop system. Poles or roots of the characteristic equation are found using the Lin-Bairstow algorithm. This method may be used to solve for the zeros of an $n^{\text {th }}$ degree polynomial.

The root locus plotter was exercised by attempting to optimally tune the system's controller. Corroboration of the results was provided by step response plots from the TUTSIM simulation program.

Minor modifications allow the root locus plotter to run without the editor. Graphics subroutines are provided by
the Turbo Graphix Toolbox. When run under the editor, the plotter is one interactive design module of the dual tank system analysis and design program. The subprogram was designed principally for user ease, error checking, and effective graphics.

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## INTRODUCTION

This computer program was developed as a module to be run with a simulation program by Leon Fadden (1986). It is a design routine which draws from the linearized model of a dual tank fluid system. With minor modification the module can run alone. Both programs are in support of a manuscript on system analysis being written by Dr. Harold Klee (University of Central Florida).

The root locus design tool presented herein allows for prediction of system stability, response characteristics, and aids in optimum tuning of a proportional-integralderivative (P-I-D) controller. The root solving method used is based upon the Lin-Bairstow algorithm, which is good for any order polynomial. However, the two tank system under P-I-D control gives rise to a maximum third order characteristic polynomial equation. This algorithm was chosen because it is well known in numerical methods, converges rapidly, and is readily understood. If the system is modified for greater accuracy, or a more complex controller is incorporated, the characteristic equation may become higher order. This program would, then, still be useful. The reader can apply the root solving part of the program for other problems which contain higher order polynomials.

The program was written in Turbo Pascal because of the language's power, readability, and current popularity. High resolution monochrome graphics were realized with the aid of the Turbo Graphix Toolbox. Minor modifications were made to the Toolbox in order to obtain enhanced results for this particular application. Some necessary system time response graphs were obtained using the TUTSIM simulation program (Applied i 1985).

## CHAPTER 1 A DUAL TANK SYSTEM

The system under investigation consists of a process which has two fluid holding tanks interconnected by a pipe (Klee 1986). Such a process might be part of a chemical batch production unit, or a flow regulating unit for the coolant of a power plant. The principal components to be analyzed in a simplified model of the system (Figure 1) are described as follows.

A constant displacement motor driven pump impels the inlet fluid into the first tank. Both tanks are unpressurized, i.e., open to the atmosphere. The intertank flow is a function of the pres-sure head of both tanks. It is assumed that the bottoms of the tanks as well as the inter-tank pipe are at ground reference level. This pipe has an adjustable valve which may be considered a load variable $\left(\theta_{1}\right)$ or disturbance input.

For design and analysis purposes we may let the two tank areas vary between simulation runs. At the outlet of the second tank is a discharge pipe with another hand actuated valve $\left(\theta_{2}\right)$ at some height above the reference. The discharge flow is a function of the tank 2 fluid level as well as the outlet valve's opening position. We also include some direct disturbance flow ( $F_{L}$ ) to tank 2. This


Figure 1. Schematic Diagram of Dual Tank System
will have some impact upon the system equations, however, the three load variables $\theta_{1}, \theta_{2}$, and $F_{L}$ will not influence the root locus of the system.

The components described so far represent an uncontrolled process, which is rarely useful in engineering operations. We can include certain other components which allow an operator to control some aspect of the process, e.g., the height of tank 2. A sensing device or transmitter is required to detect the height of the tank. The transmitter converts the height of fluid in the tank into an electrical signal (voltage), and provides an amplified signal to a controller unit (figures 2 and 3 ).

The controller considered herein is of the proportional-integral-derivative (P-I-D) type such as is commonly found in practice. It operates upon an error signal, i.e., the difference between some reference height and the height returned by the transmitter. Heights are first converted to analog electrical signals which can be recognized by the controller. Depending upon parameters set to adjust controller action, a voltage signal will be sent to the pump's motor. The inlet flow through the pump is considered to be a manipulated or controlled variable. Common configurations of controller parameters and their characteristics are as follows. For proportional control only, the response for this particular system is second


Figure 2. Block Diagram of Dual Tank System


Figure 3. Structure Blocks of Dual Tank System for TUTSIM
order. That is, the Laplace transformed transfer function has a second order characteristic polynomial in the denominator. This implies that the system may be overdamped or underdamped. An increase in gain speeds up the response and pushes the system closer towards oscillation. The major drawback to proportional control is that the steady state step response is offset from the changed set point (servo action). Using the final value theorem upon the tranfer function, the step response is

$$
\begin{equation*}
r=\frac{k_{c}}{1+k_{c}} \cdot c \tag{1}
\end{equation*}
$$

where $c$ is the magnitude of the step change in input and $K_{C}$ is the controller gain. Offset is obvious from Eq. [1] above.

The addition of integral action eliminates steady state offset, however response speed is reduced (Weber 1973). The open loop transfer function acquires an additional pole and the system becomes third order. Setting controller parameters such that the proportional gain is low and integral action low (long reset time), the system behaves like a second order system. As integral action increases the system becomes more sluggish. Increasing the gain adds oscillations, and the system tends towards instability.

If derivative action is added the system remains third order, however response speed is improved. The system
becomes more stable allowing for a higher maximum gain and shorter reset time.

Changes to controller modes, e.g., proportional to proportional-integral, are not additive. This makes tuning the controller for optimum response a complex task. Certain methods have been developed for achieving this, including the Ziegler-Nichols reaction curve method and the continuous cycling method (Weber 1973). Each method has its drawbacks. The most common method of tuning a controller in practice is by operator trial and error. The root locus gives some insight as to what the response will be for various controller settings. This alleviates the problem of experimenting on a real process, which may be time consuming, expensive, and potentially disastrous.

The system model used with this design tool must be linear. The real system undoubtedly incorporates many nonlinearities. By "linear" it is meant that the response to the sum of two signals is the same as the sum of the responses to each signal input. A linearized model is usually valid for relatively small changes about a set of design conditions. A model type (linear or non-linear) may be identified by the nature of its describing differential and algebraic equations.

Design conditions are found by setting all external inputs to desired values (Klee 1986). Then the internal,
dependent variables may be found by taking the system differential equations and rewriting them for steady state behavior. The resulting values determine the quiescent operating point about which new equations for the linearized model may be developed.

## CHAPTER 2 THE ROOT LOCUS

The root locus is a graphical path drawn on the complex s-plane. Each point along the path indicates a pole of the closed loop (Laplace domain) transfer function for a system with fixed controller parameters. As a controller parameter is perturbed (typically the controller gain) the poles of the system transfer function change. A pole is a value of $s$ on the complex plane which causes the denominator of the transfer function to go to zero, hence it causes the transfer function to go to infinity.

When set equal to zero, the denominator is called the characteristic equation of the system. The following discussion shows how the characteristic equation arises from time domain, linearized model equations. Whether or not load disturbance variables are introduced the characteristic equation does not change. For simplicity, load variables are assumed to be fixed at system design conditions, hence they do not appear in the following equations which use deviation variables.

Starting with the process equations (Klee 1986), by conservation of mass for tank 1 and tank 2:

$$
\begin{aligned}
& A_{1} \dot{H}_{1}(t)+F_{2}(t)=F_{1}(t) \\
& A_{2} H_{2}(t)+F_{0}(t)=F_{2}(t)
\end{aligned}
$$

Symbol definitions are found in Table I. Succeeding variables are functions of time unless otherwise noted. From Bernoulli's equation the inter-tank flow is

$$
\begin{equation*}
F_{2}=c_{1}\left(\bar{H}_{1}-\bar{H}_{2}\right)^{1 / 2} \tag{2}
\end{equation*}
$$

and the discharge flow is

$$
\begin{equation*}
F_{0}=c_{2}\left(\bar{H}_{2}-\bar{H}_{3}\right)^{1 / 2} \tag{3}
\end{equation*}
$$

Using deviation variables $(\Delta X$ is a relatively small deviation from design point $\bar{X}$ ) it follows from eqs.[2] and [3] that
and

$$
\begin{align*}
& \mathrm{A}_{1} \Delta \dot{\mathrm{H}}_{1}+\Delta \mathrm{F}_{2}=\Delta \mathrm{F}_{1}  \tag{4}\\
& \mathrm{~A}_{2} \Delta \dot{\mathrm{H}}_{2}+\Delta \mathrm{F}_{0}=\Delta \mathrm{F}_{2} \tag{5}
\end{align*}
$$

Since $\mathrm{F}_{2}$ is a function of $\mathrm{H}_{1}$ and $\mathrm{H}_{2}$,

$$
\Delta \mathrm{F}_{2}=\frac{\partial \mathrm{F}_{2} \Delta \mathrm{H}_{1}}{{\overline{\partial H_{1}}}_{1}}+\frac{\partial \mathrm{F}_{2} \Delta \mathrm{H}_{2}}{\overline{\partial H}_{2}}
$$

It is necessary to substitute functions of $H_{1}$ and $H_{2}$ for $F_{0}$ and $F_{2}$ into eqs. [4] and [5]. Taking the partials of $F_{2}$ and evaluating at design conditions yields

$$
\begin{aligned}
& \partial F_{2}=\bar{F}_{2}
\end{aligned}
$$

a linear approximation which we define as $1 / R 1$ where $R_{1}$ is called the fluid resistance. Similarly

$$
\begin{array}{lll}
\partial \mathrm{F}_{2}= & -\overline{\mathrm{F}}_{2} & =-1 \\
\overline{\partial \mathrm{H} 2} 2(\overline{\mathrm{H} 1}-\overline{\mathrm{H} 2}) & \overline{\mathrm{R} 1} .
\end{array}
$$

TABLE I
PROCESS AND SYSTEM SYMBOLS

CONSTANT
DESCRIPTION

| $A_{1}$ | area of tank 1 |
| :---: | :---: |
| $A_{2}$ | area of tank 2 |
| $R_{1}, R_{2}$ | linearized fluid resistances |
| $C_{1}, C_{2}$ | valve constants |
| $K_{C}$ | controller gain |
| $K_{p}$ | process gain |
| $K_{t}$ | controller integral (reset) time |
| $T_{i}$ | controller derivative time |
| $T_{d}$ | process time constants |

TIME VARYING
DESCRIPTION

| $\mathrm{H}_{1}$ | tank 1 fluid level |
| :--- | :---: |
| $\mathrm{H}_{2}$ | tank 2 fluid level |
| $\mathrm{F}_{1}$ | input flow to tank 1 |
| $\mathrm{~F}_{2}$ | inter-tank flow |
| $\mathrm{F}_{0}$ | discharge flow from tank 2 |

Eq. [5] can be rewritten as

$$
\begin{equation*}
\Delta^{F_{2}}=\frac{\Delta^{H_{1}}-\Delta^{H_{2}}}{\mathrm{R}_{1}} \tag{6}
\end{equation*}
$$

By a parallel argument, for $\mathrm{F}_{0}$ a function of $\mathrm{H}_{2}$ and $\mathrm{H}_{3}\left(\mathrm{H}_{3}\right.$ constant) it follows that

$$
\begin{gather*}
\Delta F_{0}=\frac{\partial F_{0}}{\partial H_{2}} \Delta_{2} \\
\frac{\partial F_{0}}{\partial H_{2}}=\frac{\bar{F}_{0}}{2\left(\bar{H}_{2}-\bar{H}_{3}\right)}=\frac{1}{R_{2}} \\
\Delta F_{0}=\frac{\Delta H_{2}}{\overline{R_{2}}}
\end{gather*}
$$

Substituting Eq. [6] into [4]

$$
\begin{equation*}
\mathrm{A}_{1} \Delta \dot{\mathrm{H}}_{1}+\frac{\Delta \mathrm{H}_{1}-\Delta^{H_{2}}}{\mathrm{R}_{1}}=\Delta \mathrm{F}_{1} \tag{8}
\end{equation*}
$$

Substituting eqs.[6] and [7] into [5]

$$
\begin{equation*}
\mathrm{A}_{2} \Delta \dot{\mathrm{H}}_{2}+\frac{\Delta \mathrm{H}_{2}}{\frac{\mathrm{R}_{2}}{}}=\frac{\Delta \mathrm{H}_{1}-\Delta \mathrm{H}_{2}}{\mathrm{R}_{1}} \tag{9}
\end{equation*}
$$

Rearranging eqs.[8] and [9] into standard form where the dependent variables are on the left-hand side

$$
\begin{equation*}
\mathrm{A}_{1} \mathrm{R}_{1} \Delta^{\mathrm{H}_{1}}+\Delta \mathrm{H}_{1}-\Delta_{2} \mathrm{H}_{2}=\mathrm{R}_{1} \Delta \mathrm{~F}_{1} \tag{10}
\end{equation*}
$$

and

$$
\begin{equation*}
\mathrm{A}_{2} \frac{\mathrm{R}_{1} \mathrm{R}_{2}}{\mathrm{R}_{1}+\mathrm{R}_{2}} \Delta_{H_{2}}+\Delta \mathrm{H}_{2}-\frac{\mathrm{R}_{2}}{\mathrm{R}_{1}+\mathrm{R}_{2}} \Delta^{H_{1}}=0 \tag{11}
\end{equation*}
$$

which are a pair of coupled first order linear differential equations (Eq. [ll] also is homogeneous).

If the tanks are initially at design conditions (zero initial conditions), the Laplace transforms of eqs.[10] and [ll] are
and

$$
\left(A_{1} R_{1} s+1\right)_{\Delta} H_{1}(s)-\Delta_{2}(s)=R_{1} \Delta F_{1}(s)
$$

$$
-R_{2} \Delta_{1}(s)+\left[A_{2} R_{1} R_{2} s+\left(R_{1}+R_{2}\right)\right]_{\Delta} H_{2}(s)=0
$$

Note that we have moved from the time domain to the complex s domain.

Solving for $\Delta_{2}(s)$, the controlled variable by Cramer's rule is

$$
\begin{aligned}
\Delta H_{2}(s) & =\left|\begin{array}{lc}
A_{1} R_{1} s+1 & R_{1} \Delta F_{1} \\
-R_{2} & 0 \\
\frac{A_{1} R_{1} s+1}{-R_{2}} & -1 \\
A_{2} R_{1} R_{2} s+R_{1}+R_{2}
\end{array}\right| \\
& =\frac{R_{2} \Delta F_{1}(s)}{A_{1} A_{2} R_{1} R_{2} s_{2}+\left[A_{1}\left(R_{1}+R_{2}\right)+A_{2} R_{2}\right] s+1}
\end{aligned}
$$

Dividing both sides by $\triangle F l(s)$ gives the transfer function of the process, $G_{p}(s)$. The denominator of the above equation is the characteristic polynomial of the process. Since it is second order it can be rewritten for convenience as

$$
\begin{aligned}
\left(T_{A} s+1\right)\left(T_{B} s+1\right)= & A_{1} A_{2} R_{1} R_{2} s^{2} \\
& +\left[A_{1}\left(R_{1}+R_{2}\right)+A_{2} R_{2}\right] s+1
\end{aligned}
$$

If we let $K_{p}=R_{2}$ (the process gain), then

$$
G_{p}(s)=\frac{K_{p}}{\left(T_{A} s+1\right)\left(T_{B} s+1\right)}
$$

Looking at the open loop system transfer function $G(s)$, we have

$$
\begin{equation*}
G(s)=K_{t} G_{C}(s) K_{m} G_{p}(s) \tag{12}
\end{equation*}
$$

$G_{C}(s)$ and $G_{p}(s)$ are, respectively, the controller and process transfer functions. Other terms are defined in Table I.

The closed loop system transfer function is

$$
\begin{align*}
& \Delta_{2}^{H_{2}}(s)=\frac{G(s)}{1+G(s)}  \tag{13}\\
& \Delta^{H} H_{2 s}(s)
\end{align*}
$$

assuming no change of load variables from design conditions. The transfer function for a P-I-D controller is

$$
\begin{equation*}
G_{c}(s)=K_{c}\left[1+\frac{1}{T_{i} s}+T_{d} s\right] \tag{14}
\end{equation*}
$$

Therefore substituting eqs.[12] and [14] into Eq. [13]
$\frac{\Delta^{H_{2}}(s)}{\Delta^{H}{ }_{2 s}(s)} \frac{K_{0}\left(T_{d} s+1\right) T_{i} s+1}{\left(T_{A} s+1\right)\left(T_{B} s+1\right) T_{i} s+K_{0}\left[\left(T_{d} s+1\right) T_{i} s+1\right]}$
where $K_{0}=K_{t} K_{c} K_{m} K_{p}$ is the loop gain.
The characteristic equation of the system, which comes from the denominator of Eq. [15] can be rewritten as

$$
s^{3}+\frac{\left(T_{A}+T_{B}+K_{0} T_{d}\right) s^{2}}{T_{A} T_{B}}+\frac{\left(K_{0}+1\right) s}{T_{A} T_{B}}+\frac{K_{0}}{T_{A} T_{B} T_{i}}=0
$$

The roots of this equation yield points which may be plotted on the complex $s$ plane. As $K_{c}$ varies the roots change and a locus of points may be drawn.

CHAPTER 3 THE LIN-BAIRSTOW ROOT SOLVING ALGORITHM

## Advantages of the Method

The roots of any order polynomial may be solved for by using this algorithm (McCalla 1967) even though the highest order characteristic equation generated by the model is third order. However, a more complex controller could be incorporated. The model might be expanded. Each of these changes would probably induce a higher order characteristic equation. This root solving method would, then, still be useful. Furthermore, most linear systems could employ the algorithm.

In addition to its general utility, the instructional value of Lin-Bairstow's method contributed to its selection. It is very efficient since it converges to each root quadratically. A root locus usually involves the calculation of complex roots. The Lin-Bairstow method has an advantage of not requiring any complex arithmetic. It requires only real arithmetic to calculate a complex zero and, simultaneously, its conjugate.

The Algorithm
A polynomial of any order greater than two can be factored into products of quadratic factors and perhaps one linear factor, all of which involve only real
coefficients (McCalla 1967). The roots of the quadratic may, of course, be complex. The main idea of the algorithm is to provide an efficient way of extracting the quadratic factors iteratively, and if necessary, the linear factor. Flow chart Figure 4 provides an overview of the method.

Suppose that some polynomial $P_{n}(x)$ of order $n$ is divided by an arbitrary quadratic factor $x^{2}+r x+s$. We obtain a polynomial $P_{n-2}(x)$ two orders lower and a remainder term $R x+S$. If the remainder term were zero then our trial factor would be an exact factor of $P_{n}(x)$. Therefore, operative constraints are that
and

$$
\begin{align*}
& R(r, s)=0  \tag{16}\\
& S(r, s)=0 . \tag{17}
\end{align*}
$$

The remainder coefficients are written as functions of $r$ and s since variations in these produce different remainders. Equations [16] and [17] are two non linear equations in two unknowns. Newton's method may now be applied. For sufficiently close initial estimates the method will converge (Dorn 1972).

Using a first order Taylor series approximation about an initial $r$ and $s$, and in terms of differentials

$$
\begin{align*}
d R & =R(r+d r, s+d s)-R(r, s) \\
& =R_{r} d r+R_{s} d s+\ldots \cdot \tag{18}
\end{align*}
$$

and

$$
\begin{align*}
d s & =s(r+d r, s+d s)-s(r, s) \\
& =s_{r} d r+s_{s} d s+\ldots \cdot \tag{19}
\end{align*}
$$



Figure 4 . Flow Chart of Root Solving Method
where " . . . " indicates higher order terms which may be dropped. Now if $r_{0}$ and $s_{0}$ are estimates for the factor $x^{2}+r_{0} x+s_{0}$ such that
and

$$
\begin{aligned}
& R\left(r_{0}, s_{0}\right) \neq 0 \\
& s\left(r_{0}, s_{0}\right) \neq 0,
\end{aligned}
$$

dr and as must be found such that these constraints are true:

$$
\begin{aligned}
& R\left(r_{0}, s_{0}\right)+d R=0 \\
& s\left(r_{0}, s_{0}\right)+d s=0,
\end{aligned}
$$

and
or

$$
\begin{align*}
& d R=-R\left(r_{0}, s_{0}\right)  \tag{20}\\
& d s=-s\left(r_{0}, s_{0}\right) \tag{21}
\end{align*}
$$

Therefore from Eqs. [18], [19], [20], and [21]
and

$$
\begin{align*}
& R_{r} d r+R_{s} d s=-R\left(r_{0}, s_{0}\right)  \tag{22}\\
& s_{r} d r+s_{s} d s=-s\left(r_{0}, s_{0}\right) \tag{23}
\end{align*}
$$

The last two equations are called "differential-correction" equations.

By solving for $d r$ and as we can satisfy our original constraints
and

$$
\begin{align*}
& R\left(r_{0}+d r, s_{0}+d s\right)=0  \tag{24}\\
& s\left(r_{0}+d r, s_{0}+d s\right)=0
\end{align*}
$$

Since $r_{0}+d r=r_{1}$ is a first order approximation to the zero of $R$, we can refine our solution using the above technique iteratively, starting from $r_{1}$. A similar procedure is needed for finding the zero of $S$. If the initial guess $r_{0}$ and $s_{0}$ is sufficiently close we can
converge towards the roots of eqs. [22] and [23] to within some arbitrarily small number epsilon.

## Calculations from Polynomial Coefficients

In order to solve for $d r$ and is in eqs. [24] and
it is necessary to obtain six numbers from the original polynomial. Suppose it is given that

$$
\begin{aligned}
P_{n}(x)= & x^{n} \\
= & a_{1} x_{n-1}+a_{2} x_{n-2}+\ldots+a_{n-1} x+a_{n} \\
= & \left(x^{2}+r x+s\right)\left(x_{n-2}+b_{1} x_{n-3}+b_{2} x_{n-4}+\ldots\right. \\
& \left.+b_{n-3} x+b_{n-2}\right)+R x+s .
\end{aligned}
$$

The a's are coefficients of the original polynomial, and b's are coefficients of the reduced polynomial. It is shown that (McCall 1967) after quadratic factor division:

$$
\begin{align*}
R & =b_{n-1}=a_{n-1}-r b_{n-2}-s b_{n-3}  \tag{26}\\
b_{n} & =a_{n}-r b_{n-1}-s b_{n-2} \\
s & =b_{n}+r b_{n-1} . \tag{27}
\end{align*}
$$

Furthermore, using the notation
and

$$
\begin{aligned}
& \mathrm{p}_{\mathrm{k}}=\frac{\partial \mathrm{b}_{\mathrm{k}}}{\partial \mathrm{r}} \\
& \mathrm{q}_{\mathrm{k}}=\frac{\partial \mathrm{b}_{\mathrm{k}}}{\partial \mathrm{~s}}
\end{aligned}
$$

it is also shown that (McCalla 1967):

$$
\begin{align*}
p_{k} & =-b_{k-1}-r p_{k-1}-s p_{k-1}-s p_{k-2} \\
q_{k} & =-b_{k-2}-r q_{k-1}-s q_{k-2} \\
R_{r}= & \partial b_{n-1}=p_{n-1}  \tag{28}\\
&
\end{align*}
$$

$$
\begin{align*}
& R_{s}=\frac{\partial b_{n-1}}{\partial s}=q_{n-1} \\
& s_{r}=p_{n}+r p_{n-1}+b_{n-1}  \tag{29}\\
& s_{s}=q_{n}+r q_{n-1} . \tag{30}
\end{align*}
$$

Eqs. [26], [27], and [28] through [31] are the six numbers obtained from recursion formulas for solving the differential correction equations.

## CHAPTER 4 PROGRAM DESCRIPTION

The root locus plotter can be divided into two major sections. The first is concerned with obtaining variable controller parameters from the editor, calculating a system characteristic polynomial, and iteratively finding the roots of the characteristic as the controller gain is varied. The roots are then stored in real number arrays for plotting.

The second section makes use of the Turbo Graphix Toolbox for IBM monochrome high resolution graphing of the root locus. Procedures needed from the utility package are accessed via "include files," which are compiled integrally with the main program. The procedures used feature windowing; axis drawing; drawing of points, lines, and numbers; an automatic world coordinate system; and a virtual memory screen. A few of the Turbo Graphix routines were modified slightly for improved results.

Figures 5 and 6 are macroscopic flow charts which provide an overview of program control flow. For a closer look at how the program is structured, and for details regarding subroutines the reader may refer to the commented program source code in Appendix II.

Certain problems unique to this application were encountered. When a fixed step size is specified for the


Figure 5. Flow Chart for Root Locus Program Numerical Routines


Figure 6. Flow Chart for Root Locus Program Graphics Routines
controller gain $K_{c}$, the spacing of points on the root locus varies. A moderate step size should be selected to produce a quick but smooth curve. However when the roots break away from or reenter the real axis, point spacing suddenly widens. Therefore it was necessary to incorporate a variable step size option.

This requires two passes to find the roots. The first pass uses a fixed step size and stores the calculated points (roots) in a two dimensional plot array. (The imaginary part of the root corresponds to the root locus' vertical axis.) A second pass examines the spacing between any two points found in the first pass. A linear interpolation is made to find the desired step size:

$$
\frac{\text { fixed step size }}{\text { desired step size }}=\frac{\text { first pass point spacing }}{\text { desired point spacing }}
$$

The procedure which handles this calculation is "getstepsize." The effect of this linear interpolation is to moderate the step size as needed for most points, but near the break away, spacing change is too large for this method alone. Therefore the step size is additionally decreased by a factor of one-fifth. A counter ensures that this reduced step size is used five times. A similar procedure is used for re-entry. Since either fixed or variable step size options may be selected by the user, they may be compared for best results.

Another problem encountered was that the Graphix Toolbox was not able to produce round, even numbers for axi numbering in spite of a variable axis density scheme provided. It was necessary to incorporate a "world" finding routine into the plotter program. This routine, "findXlimYlim," determines the order of magnitude and size limits of the roots to be plotted for a simulation. The values obtained are rounded off. Thus the Toolbox routine "Findworld" was bypassed. Furthermore, the axis drawing routine was modified in order to yield nice, even numbers on the axes for most cases.

The user may elect to temporarily pause plotting. A memory based virtual screen is used to save the display, and auxiliary windows pop up which contain certain variables and parameters.

Occasionally due to certain odd controller parameters entered by the user, the root solving algorithm does not converge to within user specifications. A diagnostic warning is displayed, and the user may elect to abort further calculations, or continue with unpredictable plotter results. Error checking is incorporated for all user input to ensure that entries are reasonable and within range. Figure 7 shows two typical input screens.

The plotter program, including Turbo Graphix "include" files, yields an instruction code segment of more than forty

| CONTROLLER |
| :---: |
| Change a parameter. Select one. |
| 1. Kc |
| $2 . \mathrm{Td}$ |
| $3 . \mathrm{Ti}$ |
| -1 no change |


| OPTIONS |
| :---: |
|  |
| Vary step size? |
| Yes No |

kilobytes. When coupled with the calling editor and other subprograms the code is larger than the Turbo Pascal compiler can handle easily. A Turbo Extender shell program (TurboPower 1986) was used to assist in compilation of the entire program. Note again that the plotter subprogram can run alone with minor modifications, and will compile using only Turbo Pascal.

Heap and stack memory management was necessary, but this was easily implemented with standard Turbo Pascal functions. Since the compiler window procedure does not work with the Toolbox window routines the Toolbox was, again, modified.

## CHAPTER 5 DISCUSSION OF THE RESULTS

The following illustrates how the root locus plotter may be used to analyze and design for the dual tank system. Sample graphs of the plotter are provided. The TUTSIM simulation program is also used to validate or extend plotter results.

In order to find some unique combination of controller parameters it is necessary to specify some goal to be achieved with respect to the system. One such goal is to tune the controller optimally by the quarter decay ratio method, as first published by Ziegler and Nichols (Weber 1973).

By quarter decay it is meant that the step response should exhibit damped oscillations (underdamped response) such that the second peak is one-quarter of the height of the first peak (overshoot). While there is no unique combination of P-I-D controller parameters which yield such behavior, a logical set of values may be found by first examining pure proportional control.

Suppose the system under feedback control is brought to design conditions. The controller is switched to manual mode, which breaks open the feedback loop. A unit step voltage directly to the pump motor will produce a typical
second order response at the transmitter output since the process under consideration is second order. An optional controller chart recorder can graph the output (Figure 8). Extend a tangent line of maximum slope down to the time axis. The time intercepted is, in effect, a delay. The open loop system can be approximated by a first order transfer function in series with a pure delay element. The slope line and delay time found as detailed above characterize the approximate model. Such a simplified model can mathematically be shown to have the quarter decay response characteristic desired (Smith 1985) when the controller is tuned as follows (thus closing the loop):

$$
K c=\frac{1}{S L}
$$

where $S$ is the slope of the tangent line and $L$ is the delay time described above. For our system, values of $\mathrm{S}=$ $0.013 / \mathrm{min}, \mathrm{L}=3.57 \mathrm{~min}$, and $\mathrm{Kc}=21.55$ were found.

Having found some particular value to set the proportional control, the root locus plotter may be employed for further analysis. Figure 9 shows the root locus stopped at this value. The locus appears to be heading straight up as the gain increases. This observation is confirmed by Figure 17. Using a simulation with $\mathrm{Kc}=21.55$, a closed loop step response was graphed (Figure 10). A decay ratio smaller than 0.25 is evident, therefore the recommended setting is too conservative. It turns out that


Figure 8. Reaction Curve Method for Tuning the Controller


Figure 9. Optimally Tuned Controller Using Proportional Control Only


Figure 10. Step Response for P - I Controller and Proportional Controller Tuned Optimally by Reaction Curve Method
a small increase in gain yields the desired response. It should be noted that the goal of quarter decay ratio is more useful in controlling load changes than set point changes. For set point changes this method produces too much overshoot, but it prevents load changes from deviating too far from design without being too oscillatory (Smith 1985). Another method which has the same quarter decay goal and may employ the root locus principally is the continuous cycling or ultimate gain method. The method dictates that the closed loop system be placed under proportional control only. Increase the controller gain until the system step response oscillates continuously. This point corresponds to the vertical axis crossing of the root locus. At that point read the ultimate gain $K_{c u}$ and the natural frequency $W_{n u}$. Controller parameters for optimum control may be calculated as follows (Weber 1971). P

$$
\begin{aligned}
\mathrm{K}_{\mathrm{c}} & =0.5 \mathrm{~K}_{\mathrm{cu}} \\
\mathrm{~K}_{\mathrm{c}} & =0.45 \mathrm{~K}_{\mathrm{cu}} \\
\mathrm{~T}_{\mathrm{i}} & =\frac{2 I I}{1.2 \mathrm{w}_{\mathrm{nu}}} \\
\mathrm{~K}_{\mathrm{c}} & =0.6 \mathrm{~K}_{\mathrm{cu}} \\
\mathrm{~T}_{\mathrm{i}} & =\frac{2 I I}{2.0 \mathrm{w}_{\mathrm{nu}}} \\
\mathrm{~T}_{\mathrm{d}} & =\frac{2 I I}{8.0 \mathrm{w}_{\mathrm{nu}}}
\end{aligned}
$$

Unfortunately, for the dual tank system this method does not work. As already mentioned, increasing the gain for the
system under proportional control causes a vertical locus. The system never becomes unstable, which is characteristic of a second order process under proportional control. Most real world processes are of higher order, and some value of gain will cause instability.

One inportant item yielded by the root locus plotter is the damping ratio. Standard second order response curves are available for specified damping factors, which allow the designer to predict what the response will be, i.e., how fast the oscillations will die out. Many higher order systems are characterized by two poles which dominate the response, and can be approximated by a second order system. Another item is the natural frequency, $w_{n}$, which indirectly gives the period of oscillation. Figure 10 also indicates the step response for a P-I controller tuned at the Ziegler-Nichols optimum. It is evident from the corresponding root locus (Figure 11) that the system is far from unstable, but that increasing the gain will eventually cause more oscillations since the path is turning back towards the Y axis. A natural frequency of $\mathrm{w}_{\mathrm{n}}=0.2343$ predicts that the period of oscillation is

$$
\begin{aligned}
T & =\frac{2 I I}{\left(1-z^{2}\right) w_{n}} \\
& =\frac{6.28}{0.968(0.234)}=27.7 \mathrm{~min}
\end{aligned}
$$



Figure 11. Optimally Tuned Controller Using Proportional - Integral Control
which agrees well with the time response (Figure 10 ). ( $z$ is the damping ratio.) Three peaks are found over a 75-minute interval. Notice that the addition of integral control adds a third pole which moves horizontally along the horizontal axis. The contribution to the response is a negative exponential which dies out more quickly as the gain increases. The addition of derivative action to P-I control reveals that the system has 3 poles and remains third order (Figure 12). The locus reveals a greater degree of stability as the path continues to move away from the jw axis. Compared to P-I control an increased value of gain is permitted for optimum tuning $(\mathrm{Kc}=25.85)$. Figure 13 shows the corresponding step response which has less initial overshoot, and a faster settling out to steady state. Note that the step input has been arbitrarily delayed for 10 minutes for better graphics. The faster settling time could be predicted from the root locus plot, by noting a higher damping factor (0.32). If we increase the gain of the optimally tuned $P-I$ controller to that of the $P-I-D$ controller, i.e., from $K c=19.3$ to $K c=25.8$, the response becomes too oscillatory, as shown by Figure 14.

The system model was further exercised and certain unique conditions were observed. A controller setting was quickly determined by using the root locus to find a point of marginal stability (Figure 15). The corresponding step


Figure 12. Optimal.1y Tuned Controller Using Proportional - Integral Derivative Control


Figure 13. Step Response for P - I - D Controller Tuned Optimally by Reaction Curve Method


Figure 14. Step Response for P - I Controller Gain Set Higher Than Recommended

Root Locus Plotter


Figure 15. Controller Set for System at Marginal Stability
response agreed nicely, producing relatively constant oscillations (Figure 16). At the point of breakaway the response is critically damped. Using the root locus, this condition was easily found (Figures 18 and 19). For certain systems any overshoot might be unacceptable. The root locus could then be used to find controller settings for the fastest response with no overshoot, as done above.

Another item of interest concerns the plotter's variable step size feature. Although proportionalderivative control is seldom used in practice because of the resulting offset, the root locus plot is interesting as it forms a complete loop (Figure 20). Note how much improved the plotter draws Figure 21 which incorporates a variable step size. At breakaway and reentry, point spacing suddenly increases. A fixed $K_{C}$ is especially undesirable for this plot. Standard root locus plotters, such as found in the Matlab Control System Toolbox (Figure 22), do not incorporate automatic variable spacing (Moler 1985). Several of the plots made with the root locus were run on Matlab for validation, and there were no discrepancies found between the two plotter routines.


Figure 16. Step Response for Marginally Stable System

A subprogram was written for the IBM PC/AT to be used as one design tool which is called by an editor program. The editor allows for making, changing, and saving various configurations of the dual tank system. This design tool is the first in a series of several subprograms to be developed by other students. The entire software is written in support of a manuscript on system analysis to be later published as a textbook.

In writing the subprogram my intent was to make the numerical routine modular in the event that the reader wishes to use a Turbo Pascal root finder for some other application. The graphics routines which use the Turbo Graphix Toolbox are set apart, and not essential to the first part of the subprogram. The reader may wish to incorporate his own graphics routines to plot points residing in arrays, since the Toolbox carries a large overhead in subroutines (about 2500 total source lines of code) not used for this application. With minor changes the subprogram can run alone without a calling editor.

Although the subprogram source code (excluding Graphix Toolbox subroutines) is about 1200 lines, its real-time execution is very fast. It is even necessary to incorporate delays in order to allow the user to pause the graphing. The goal in writing the program was to allow for user ease, error checking, and informative graphics. Code minimization was of minor concern.

Rudimentary differential equations for the system led to a derivation of the characteristic equation. Although the system under consideration gave rise to a maximum of third degree polynomials, it was instructive to obtain a higher capability root solving method and make it available in Turbo Pascal. Since the roots of the characteristic equation were almost always small numbers the initial guess of ( $r_{0}, s_{0}=0,0$ ) guaranteed convergence.

The root locus plotter was exercised by attempting to optimally tune the system's controller according to the quarter decay ratio method. It was demonstrated how the root locus could be used to obtain quick information about the time response of a system. Principal items obtained included the relative stability, damping ratio, and the frequency of oscillations of the system.

## APPENDIX I ADDITIONAL FIGURES



Figure 17. System Under Proportional Control for Very High Gain


Figure 18. System with Gain Slightly Above Breakaway


Figure 19. Step Response for System at Breakaway. Controller Using Proportional Gain Only


Figure 20. System Under P - D Control. Plot Using Fixed Step Size


Figure 21. System Under P - D Control. Plot Using Variable Step Size



Figure 22. Sample Matlab Input and Output

## APPENDIX II PROGRAM SOURCE CODE LISTING

```
{ a comment generally refers to the procedure
    directly following that comment}
```

```
procedure RootLoci (Kc, Ti, Td, areal, area2: real);
```

```
procedure RootLoci (Kc, Ti, Td, areal, area2: real);
```

```
type
    stringy = string[ 78];
    Onedim = Array[ 1..l00] of real;
    dummypointer = ^dummy;
    dummy = record { used to release heap pointer}
        end;
var heapaddress: dummypointer;
    apoints, bpoints, cpoints: PlotArray;
    II, JJ, Imax : integer;
    xlim, ylim, Xlo, Ylo: real;
    eps, ro, s0, R1, R2, Kp, Km, Kt,
    Kcsaved, Kcfinalsaved, stepsizesaved, Tisaved,
    Tdsaved: real;
    alpha, alphasaved: real;
    Kstore, Kstore2: Onedim;
    linerequested, abort, firstRun, modify, exit,
    thirdorder: boolean;
    message: stringy;
```

\{ check that the response to an option message
displayed is yes or no only\}
procedure check_response( message: stringy;
requestnum: integer);
var inp: char;
ValidChar: boolean;
begin
GotoXY ( 35, 2);
Textcolor( Lightred);
write( ' OPTIONS ');
Textcolor( Yellow);
Window ( 1, 1, 80, 25) ;
GotoXY ( 34,13 );
writeln( message);
GotoXY( 37,23 );
Textcolor( Lightgreen) ;
write( 'Y');
Textcolor( Cyan);
write( 'es');
Textcolor( Lightgreen);
write( ' N');

```
    Textcolor( Cyan);
    write( 'o');
    repeat
        read (kbd, inp);
        inp := upcase( inp);
        case inp of
        'N': exit := true;
        'Y': case requestnum of
            l: modify := true;
            2: linerequested := true;
            3: abort := true;
            4: ;
            end {case}
        else
            begin
                Sound (880);
                Delay (200);
                Nosound
            end {else}
        end; {case}
        validchar := inp in ['N','Y']
        until validchar;
        Textcolor( Yellow);
        Window(4, 23, 78, 24);
        ClrScr;
        Window( 1, 1, 80, 25)
    end; {check response}
{ display a warning or error, such as invalid
    input}
procedure display_error( message: stringy);
begin
    Sound (880);
    Delay (200);
    Nosound;
    Window(3, 4, 78, 21);
    ClrScr;
    Window(1, 1, 80, 25);
    GotoXY( 3,5);
    TextBackground( red);
    TextColor( white);
    write( message);
    TextBackground( black);
    TextColor( Yellow);
    Delay( 3000);
    Window( 3, 4, 78, 21);
    ClrScr;
```

```
    Window( 1, 1, 80, 25)
end; {display error}
```

\{ large supervisory routine, which does numerical calculations, and stores required root locus points in plot arrays. All succeeding procedures are directly or indirectly controlled by this routine until graphics procedures are reached\}
procedure getpoints;

```
type TwoDimArray = array[ 1..1200, l..2] of real;
```

            OneDimArray \(=\) array [1..3] of real;
    const TwoDimMax $=1200$; OneDimMax $=3$;
var infile: text;
delta_r, delta_s, $r, s$, capR, capS, Rr, Rs, Sr, Ss: real; Kcfinal, stepsize, rholim: real;
$n, 1$, capN, JJMax: integer;
a, b,imagroot, realroot: OneDimArray;
fwpoints: TwoDimArray;
$\mathrm{ap}, \mathrm{bp}, \mathrm{cp}:$ PlotArray;
in_range, breakaway: boolean;
\{ The next five routines are within getpoints. All are involved with initialization of arrays which if not performed can yield undesirable side effects $\}$
procedure zeroOut( var thisarray: OneDimArray); var I: integer;
begin
for $I:=1$ to oneDimMax do thisArray[I] := 0.0
end;
procedure initializelDArrays;

```
begin
    zeroOut (a);
    zeroOut (b);
    zeroOut (imagroot);
    zeroOut (realroot)
end;
```

```
procedure zero( var thisarray: TwoDimArray);
    var I,J: integer;
    begin
        for J := l to 2 do
        for I := l to TwoDimMax do
            thisArray[I,J] := 0.0
    end;
```

```
procedure NullOut( var thisarray: plotArray);
    var I,J: integer;
    begin
        for \(J:=1\) to 2 do
        for \(I:=1\) to MaxPlotGlb do
            thisArray[I,J] := 0.0
    end;
```

procedure initializePlotArrays;

```
begin
    NullOut (apoints);
    NullOut (bpoints);
    NullOut (cpoints);
    Nullout (ap);
    NullOut (bp);
    NullOut (cp);
end;
```

\{ get auxiliary system parameters from a disk file if root locus procedure is called from editors
procedure get_parameters;
begin
assign (infile,'RLINPUT.DTA');
reset (infile);
read (infile, eps, r0, so);
read (infile, areal, area2, R1, R2, Kp, Km, Kt);
read (infile, Kc, Kcfinal, stepsize, Td, Ti);
\{initialize " saved" variables\}
Kcsaved := Kc;
Kcfinalsaved := Kcfinal;
stepsizesaved := stepsize;
Tdsaved := Td;

Tisaved := Ti
end; \{get_parameters\}
\{ if repeating the root locus procedure, then reset the controller parameters to what they were when first called from editor\}
procedure reset_parameters;
begin
Kc := Kcsaved;
Kcfinal := Kcfinalsaved;
stepsize := stepsizesaved;
Ti := Tisaved;
Td := Tdsaved
end; \{reset parameters\}
\{ display the menu which allows user to change
various controller parameters\}
procedure submenu;
var selecting, validnum: boolean;
selection: char;
\{within submenu\}
procedure display_old( parameter: real);
begin
GoToXY ( 20, 13) ;
TextBackground( blue);
TextColor( lightgray);
write( parameter:9:3);
TextBackground( black);
TextColor ( yellow);
end;
procedure display_new( parameter: real);
begin
GotoXY( 20, 13) ;
TextBackground( blue);
TextColor( yellow);
write( parameter:9:3);
TextBackground( black);
end;

```
{ error checking to ensure that all inputs for
controller parameters are within a valid range}
procedure check_real( var number: real);
    begin
    Textcolor( lightgreen);
    GotoXY( 38, 23);
    write( '? ');
    Textcolor( yellow);
    {$I-}
    readln( number);
    {$I+}
    if( (IOresult <> 0) or
        (number < 0 ) or (number > IE5) or
                (Ti = 0.0)) then
            display_error
        ('Invalid number. Please re-enter.')
                else
            validnum := true;
    GotoXY( 38, 23);
    write(
    end; {check real}
{gets user parameter changes from the keyboard}
procedure obtain( var param, paramsaved: real;
                                    message: stringy; Y: integer);
var tempparam: real;
begin
    validnum := false;
    tempparam := param;
    repeat
        GoToXY( 25, Y);
        display_old( tempparam);
        write( message);
        check_real( param)
    until validnum;
    paramsaved := param;
    display_new( paramsaved);
    Delay( I000);
end; {obtain}
```

```
{display prompt for user}
procedure writehere( message: stringy; X, Y: integer);
    begin
            GoToXY( X, Y);
            write( message);
    end;
```

```
begin \{submenu\}
selecting := true;
while selecting do begin
    Window (3, 4, 78, 21);
    ClrScr;
    Window ( 1, 1, 80, 25) ;
    GotoXY ( 30,2 );
    Textcolor( lightred);
    writehere ( 'CONTROLLER', 35, 2 );
    Textcolor ( yellow) ;
    writehere
    ('Change a parameter. Select one.', 25, 5);
    writehere( 'l. Kc', 38, 12);
    writehere( '2. Td', 38, 13);
    writehere( '3. Ti', 38, 14);
    writehere( ', l no change', 34, 21);
    Textcolor( lightgreen);
    writehere( '? ', 38, 23);
    Textcolor ( yellow);
    read (kbd, selection);
    Window (3, 4, 78, 21) ;
    ClrScr;
    Window ( 1, 1, 80, 25) ;
    writehere( \(1,38,23\) );
    case selection of
    'l': begin
                obtain( Kc, Kcsaved,
            ' initial Kc (Return: no change)', 13);
            obtain( Kcfinal, Kcfinalsaved,
                                    final Kc? \(\quad\), 14);
            obtain( stepsize, stepsizesaved,
                step size? (, 15);
        end;
    '2': obtain( Td, Tdsaved, ' Td ', 13);
    '3': obtain( Ti, Tisaved, ' Ti ', 13);
    \#13: selecting := false;
        else
            sound (660);
            delay (250);
            nosound
```

end \{case\}
end; \{while\}
writehere ( $1,35,2$ ) end; \{submenu\}
\{from the various controller parameters and other system parameters construct the characteristic polynomial for the closed loop system\}
procedure construct_polynomial;
const Tlarge: integer $=100$;
var tauAtauB, tauAplustauB, Ko: real;
begin
tauAtauB := areal * area2 * R1 * R2; tauAplustauB $:=$ areal * (R1 + R2) + area2 * R2;
Ko : = Kp * Km * Kc * Kt;
$a[l]:=($ tauAplustauB $+($ Ko * Td)) / tauAtauB;
$a[2]:=(K 0+1) /$ tauAtauB;
$\mathrm{a}[3]:=\mathrm{KO} /$ (tauAtauB * Ti) ;
if (Ti >= Tlarge) then begin
capN := 2;
thirdorder := false end
else begin
capN := 3;
thirdorder := true end; $\{$ else $\}$
end; \{construct_polynomial\}
\{ this routine and its subroutines implement the LinBairstow algorithm, as described in the text Ch. 3\}
procedure roots_driver;
const lmax: integer $=20$;
var $k, m, j:$ integer;
stop, continue: boolean;
st, steps: string[ 6];
procedure init_next_factor;
begin

$$
\mathrm{n}:=\operatorname{capN}-(2 * m) ;
$$

```
    1 := 0;
    r := r0;
    s := s0;
end; {init next factor}
function testdisc (capR, capS: real): integer;
var disc: real;
begin
    disc := capR * capR - 4.0 * capS;
    if disc < O then
        testdisc := -1
    else
        if disc = 0 then
            testdisc := 0
        else
            testdisc := l
end; {testdisc}
procedure quadroots (R, S:real);
var rad: real;
begin
    case testdisc(R, S) of
    -l: begin
                            rad := sqrt (4.0 * S - R * R);
                    realroot [J] := -R / 2.0;
                realroot [J+1] := -R / 2.0;
                imagroot [J] := rad /2.0;
                imagroot [J+1] := -rad / 2.0
            end;
            0: begin
            realroot [J] := -R /2.0;
            realroot [J+1] := -R / 2.0;
            imagroot [J] := 0.0;
            imagroot [J+l] := 0.0
            end;
            1: begin
                rad := sqrt (R * R - 4.0 * S);
                realroot [J] := (-R + rad) / 2.0;
                realroot [J+1] := (-R - rad) / 2.0;
                imagroot [J] := 0;
                imagroot [J+1] := 0
            end
    end {case}
end; {quadroots}
```

```
procedure reduce_polynomial;
begin
    \(b[1]:=a[l]-r ;\)
    \(\mathrm{b}[2]:=\mathrm{a}[2]-(r * \mathrm{~b}[1])-\mathrm{s} ;\)
    for \(k:=3\) to \(n\) do
        \(\mathrm{b}[\mathrm{k}]:=\mathrm{a}[\mathrm{k}]-(\mathrm{r} * \mathrm{~b}[\mathrm{k}-1])-\mathrm{s} * \mathrm{~b}[\mathrm{k}-2]\);
    capR \(:=b[n-1]\);
    capS \(:=b[n]+(r\) * \(b[n-l])\);
end; \{reduce polynomial\}
```

procedure partials;

```
var p, q: array [l..6] of real;
begin
    p[1]:= -1.0;
    p[2] := r - b[l];
    for k := 3 to n do
        p[k] := -b[k-l] - r * p[k-l] - s * p[k-2];
    Rr := p[n-l];
    Sr := p[n]+ (r * p[n-l]) + b[n-l];
    q[l] := 0.0;
    q[2] := -1.0;
    for k := 3 to n do
        q[k] := -b[k-2] - r * q[k-l] - s * q[k-2];
    Rs := q[n-l];
    Ss :=q[n] +r * q[n-l]
end; {partials}
```

procedure differential_corrections;

```
var denom: real;
begin
    denom := Rr * Ss - Rs * Sr;
    delta_r := (-capR * Ss + capS * Rs)/ denom;
    delta_s := (-Rr * capS + Sr * capR)/ denom;
    r:= r + delta_r;
    s := s+ delta_s;
end; {diff'l corrections}
```

procedure replace_polynomial;
var newN: integer;

```
    begin
    m := m+1;
    j:= j+2;
    newN := capN - (2* m);
    for k:= l to newN do
    a[k] := b[k];
end; {replace polynomial}
begin {roots driver}
    abort := false;
    k := 0; m := 0; j := l; stop := false;
    continue := false;
    while not stop do
        begin
        init_next_factor;
        if n- < 2 Ehen
            begin
            realroot [J] := -a[l];
            stop := true
            end
        else
            if n}=2\mathrm{ then
            begin
                    quadroots(a[l],a[2]);
                    stop := true
            end
        else
            begin
                    repeat
                            reduce polynomial;
                            partials;
                            differential corrections;
                        l:= l+1;
                        stop := (l > lmax);
                        continue := not stop and
                    ((abs(delta_r) > eps)
                    or (abs(del\overline{ta_s) > eps));}
                until not continue;
                if not stop then
                    begin
                    quadroots(r,s);
                    replace_polynomial;
                    end {if not stop}
                    else if (l > lmax) then {stop true }
                    begin
                        str( 1:3,st);
                        str(eps:6,steps);
```

```
    message := 'roots have not'+
    ' converged after ' + st +
    ' iterations to within '+ steps;
    display_error( message);
    check_response( ' Abort? ', 3);
    quadroots( r, s);
    replace_polynomial
end {else}
            end {else begin repeat}
        end {while}
    end; {roots_driver}
```

\{ determine a world coordinate system. Bypass the Turbo
Graphix procedure in order to condition the limits
obtained. Scan the array of points for max \& min
values\}
procedure FindXlimYlim( A:TwoDimArray; NPoints:integer);
var j:integer;
begin
NPoints $:=$ abs ( NPoints) ;
if NPoints $>=2$ then
begin
$\mathrm{Xlim}:=A[1,1]$;
$\operatorname{Ylim}:=A[1,2]$;
Xlo: $=$ Slim;
Yo := Ylim;
for $j:=2$ to NPoints do
begin
if $A[j, 1]>X l i m$ then
$\mathrm{Xlim}:=A[j, 1]$
else
if $A[j, 1]$ < Xlo then
Plo $:=A[j, 1] ;$
if $A[j, 2]>$ Ylim then
Ylim $:=A[j, 2]$
else
if $A[j, 2]<Y l o ~ t h e n ~$
Ylo $:=A[j, 2]$
end; \{for\}
end \{if NPoints $\}$
else error (7,4);
end; \{FindXlimylim\}
\{after calculating the roots of a polynomial for some Kc, store that point in an array\}
procedure fillarrays
( var Aarray, Barray, Carray: plotarray; var Karray: Onedim) ;

```
begin
    II := II + I;
    Aarray [II,l] := realroot [l];
    Aarray [II,2] := imagroot [l];
    JJ := JJ + 1;
    fwpoints[JJ,1] := Aarray[II,l];
    fwpoints[JJ,2] := Aarray[II,2];
    Barray [II,l] := realroot [2];
    Barray [II,2] := imagroot [2];
    JJ := JJ + 1;
    fwpoints[JJ,l] := Barray [II,I];
    fwpoints[JJ,2] := Barray [II,2];
    if thirdorder then
        begin
            Carray [II,l] := realroot [3];
            Carray [II,2] := imagroot [3];
            JJ := JJ + 1;
            fwpoints [JJ,I] := Carray [II,I];
            fwpoints [JJ,2] := Carray [II,2];
            end; {if}
    Karray[ II] := Kc
end; {fillarrays}
{ensure that user requirements are not too large
    or too small for the arrays}
procedure test_range;
    var st: string[8];
    begin
        if ( II > MaxPlotGlb) or ( II < l) then
            begin
            II := MaxPlotGlb + l;
            str( II:8, st);
            in_range := false;
            message := ' # of points to be plotted, '+ st
            + #13#10 + 'is too large.'+
            ' Please change requirements';
            display_error( message);
            Delay(3000)
            end {if}
        end; {testrange}
```

```
{if a fixed increment is used for Kc, then pass l
obtains the required array of points. Otherwise
these points are necessary for comparison when
calculating a variable step size}
```

procedure passl;
begin
Window ( 3, 4, 78, 21);
ClrScr;
Window(1, 1, 80, 25) ;
GotoXY ( 35, 2) ;
Textcolor( lightred);
write ('CALCULATING');
GoToXY $(29,12)$;
TextColor ( yellow) ;
TextBackground( black);
write ( $\#$ of points to be plotted:' );
TextColor ( Brown) ;
repeat
construct_polynomial;
roots_driver;
if (nōt abort) then
begin
fillarrays( ap, bp, cp, Kstore) ;
GoToXY ( 39,13) ;
write( II:3);
test_range;
Kc $: \equiv \mathrm{Kc}+$ stepsize
end \{if not abort\}
until ( Kc > Kcfinal) or abort or ( not in_range);
TexTColor( Yellow);
end; \{passl\}
\{if a variable step size is required, a scan is made of the distance between points obtained in pass 1. for each pair of adjacent points a linear interpolation is made to obtain the desired step size. New points are calculated and stored in separate arrays
procedure pass2;
var count, count2, imin: integer; maskbreakout, maskbreakin: boolean;
\{within pass2. Does linear interpolation as per text Ch. 4. Upon break away or reentry at $x$ axis, the step size is additionally reduced for a few

## steps $\}$

procedure getstepsize;
const small = $1 \mathrm{E}-6$;
var aNum, deltaAl, deltaA2, deltaA: real; i: integer;
stopcondition, breakout, breakin: boolean;
begin
i $:=$ imin;
repeat
if (i < Imax) then
stopcondition $:=$ ( Kstore[ i-l] <= Kc ) and ( Kc < Kstore[ i])
else
begin
stopcondition $:=($ Kstore $[i-1]<=\mathrm{Kc})$; end;
i $:=1+1 ;$
until stopcondition;
i:=i-1;
imin := i;
deltaAl $:=a p[i, l]-a p[i-1,1] ;$
deltaA2 $:=a p[i, 2]-a p[i-1,2] ;$
deltaA $:=\operatorname{sqrt}(\operatorname{sqr}(\operatorname{deltaAl})+\operatorname{sqr}(\operatorname{deltaA} 2)) ;$ stepsize := stepsizesaved
( rholim / deltaA) * ( $1.0 / 60.0$ );
breakout $:=(\operatorname{abs}(a p[i, 2])>$ small) and
breakin $:=(\operatorname{abs}(\operatorname{ap}[i-1,2])<$ small) i ( abs(ap[i-1, 2]) > small);
if not maskbreakout then
if breakout then
count :=5;
if count $>0$ then
begin
count $:=$ count -1 ; maskbreakout $:=$ true;
stepsize := stepsize/ 5
end;
if not maskbreakin then
if breakin then
count2 : = 10;
if count2 $>0$ then begin

```
                    count2 := count2 - 1;
                    maskbreakin := true;
                    stepsize := stepsize/ 3
            end;
    end; { get step size}
    begin {pass 2}
    maskbreakout := false;
    maskbreakin := false;
    count := 0;
    count2 := 0;
    II := 1;
    imin := 2;
    Kc := Kcsaved;
    rholim := sqrt( sqr( Xlim-Xlo) + sqr( Ylim-Ylo) );
    TextColor( Brown);
    repeat
        getstepsize;
        Kc := Kc + stepsize;
        construct_polynomial;
        roots_driver;
        if no\overline{t}}\mathrm{ abort then
            begin
                fillarrays( apoints, bpoints,
                                cpoints, Kstore2);
                GoToXY( 39,13);
                write( II: 3);
                test_range
            end {if not abort}
        until ( Kc > Kcfinal) or abort or ( not in_range);
    TextColor( Yellow)
end; {pass2}
begin {getpoints}
    initializelDArrays;
    initializePlotArrays;
    zero( fwpoints);
    repeat
    in_range := true;
    if firstRun then
        begin
            get_parameters;
            firstRun := false
        end
    else
        reset_parameters;
    submenu;
```

```
    modify := false;
    Window (3, 4, 78, 21);
    ClrScr;
    Window (1, 1, 80, 25);
    check_response( 'Vary step size? ', 1);
    linerequested := false;
    check_response( 'Lines instead of points? 1, 2);
    II :=-0; \{index associated with plotarrays\}
    JJ := 0; \{index associated with fwpoints array only\}
    passl;
    if (not abort) and ( in_range) then
        begin
            Imax := II;
            JJmax := JJ; \{was set in fillarrays,
                                    now used to find Xlim, Ylims
                    findXlimYlim( fwpoints, JJmax);
                if modify then
                pass2
                else
                    begin
                            apoints \(:=\) ap;
                            bpoints \(:=\mathrm{bp}\);
                            cpoints \(:=c p ;\)
                    Kstore2 := Kstore
                end \{else\}
            end \{if not abort\}
    until in_range
end; \{getpōints\}
(*** graphics routines ***)
\{initializes the root locus graph by using initial-
    ization procedures from the Turbo Graphix Toolbox.
    Incorporates a round off procedure to make the
    graph limits round numbers\}
procedure initializeGraph;
    var
        XMinAdj, YMinAdj, XMaxAdj, YMaxAdj: integer;
        function round_off( number:real): real;
        var firstDigit, exponent, stringnum: string[8];
            num, code, digit: integer;
            temp: real;
        begin
```

```
            str( number:8, firstDigit);
            str( number:8, exponent);
            if number >= 0 then begin
    delete( firstDigit, 2, 7);
    val( firstDigit, digit, code);
    num := l + digit end
    else begin
    delete( firstDigit, 3, 6);
    val( firstDigit, digit, code);
    num := -l + digit end;
    delete( exponent, 1, 4);
    str( num, stringnum);
    val( stringnum + exponent, temp, code);
    round_off := temp;
end;
begin
    DefineWindow (l, 0, 0, XMaxGlb, YMaxGlb);
    DefineHeader (1, 'Root Locus Plotter');
    XMinAdj := 4;
    YMinAdj := 16;
    XMaxAdj := XMaxGlb-2;
    YMaxAdj := YMaxGlb-14;
    DefineWindow (2, XMinAdj, YMinAdj, XMaxAdj, YmaxAdj);
    DefineWindow (3, XMinAdj, YMinAdj + 4,
                                XMinAdj + 10, YMinAdj + 26);
    DefineHeader (3, 'Kc');
    DefineWindow (4, trunc( XMaxAdj/2.0) - 3, YMinAdj +4,
                                trunc( XMaxAdj/2.0) + 5,
                                YMinAdj + 26 );
    DefineHeader (4, 'damping');
    DefineWindow (5, XMinAdj, YMinAdj + 28, XMinAdj + 10,
        YMinAdj + 50);
    DefineHeader (5, 'Wn');
    DefineWindow (6, XMinAdj, YMinAdj + 52, XMinAdj + 10,
                                YMinAdj + 74);
    DefineHeader (6, 'Ti');
    DefineWindow (7, XMinAdj, YMinAdj +100, XMinAdj + 10,
        YMinAdj + 122);
    DefineHeader (7, 'Td');
    SelectWindow (1);
    SetHeaderOn;
    DrawBorder;
    IMax := II;
    {final value II was set by fill arrays,
    used for repeat control below}
if linerequested then
```

```
    begin
        II := 2;
            {reset plot array index to beginning,
            #2 required for DrawLine. Now first
            two points equal to remove side effects.}
            apoints[ 1, l] := apoints[ 2, 1];
            apoints[ 1, 2] := apoints[ 2, 2];
            bpoints[ 1, l] := bpoints[ 2, l];
            bpoints[ 1, 2] := bpoints[ 2, 2];
            cpoints[ 1, l] := cpoints[ 2, 1];
            cpoints[ 1, 2] := cpoints[ 2, 2]
    end
else
    II := l;
    if ( abs( Ylo) < eps) and ( abs( Ylim) < eps) then
    begin
            Ylo := -1.0;
            Ylim := 1.0
    end;
Xlo := round_off( Xlo);
Ylo := round off( YlO);
Ylim := roun\overline{d}off( Ylim);
DefineWorld( I, Xlo, Ylo, Xlim, Ylim);
SelectWindow (3);
SetHeaderToBottom;
SetHeaderOn;
DrawBorder;
SelectWindow (4);
SetHeaderToBottom;
SetHeaderOn;
DrawBorder;
SelectWindow (5);
SetHeaderToBottom;
SetHeaderOn;
DrawBorder;
end; {initializeGraph}
{next two subroutines are from the Toolbox, and are
used to draw point positions which pop up when the
plotting is temporarily stopped by the user}
function StringNumber(XI:real;
                                    MaxExponent:integer):wrkstring;
var y:wrkstring;
```

```
begin
    str(Xl*exp(-MaxExponent*ln(10.0)):5:2,y);
    StringNumber:=y;
end;
```

```
function GetExponent(Xl:real):integer;
begin
    GetExponent:=0;
    if Xl<>0.0 then
            if abs(Xl)>=1.0 then GetExponent:=
            trunc(ln(abs(Xl))/ln(10.0))
            else GetExponent:=
            -trunc(abs(ln(abs(X1)))/ln(10.0)+1.0);
end;
```

\{when user pauses graphing, various numbers and windows display from a virtual memory screen. The user may resume graphing, whereupon these items vanish, or he may elect to terminate the graph prematurely (esc) \}
procedure pop_up;
var stal, sta2: string[ 5];
sta: string[ 13];
cposy, aposX, bposX, bposY: real; maxexponentX, maxexponentY: integer;
begin
CopyScreen;
SelectWindow ( 6);
SetHeaderOn;
DrawBorder;
GoToXY (7, 10);
write (Ti:5:2);
SelectWindow ( 7) ;
SetHeaderOn;
DrawBorder;
GoToXY $(7,16)$;
write (Td:5:2);
SelectWindow ( 2) ;
maxexponentX $:=$ GetExponent (Xlo);
maxexponenty $:=$ GetExponent (Ylo);
stal := StringNumber( apoints[ II, 1], maxexponentX);
sta2 := StringNumber( apoints[ II, 2], maxexponentY);
sta $:=$ stal $+1,1+$ sta2;

```
    aposX := apoints[II, l] - 0.25 *
    exp( maxexponentX * ln(10.0));
    drawtextw( aposX, -apoints[II, 2], 1, sta);
    stal := StringNumber( bpoints[ II, l], maxexponentX);
    sta2 := StringNumber( bpoints[ II, 2], maxexponentY);
    sta := stal + ',' + sta2;
    bposX := bpoints[II, 1] - 0.25 *
        exp( maxexponentX * ln(10.0));
    bposY := bpoints[II, 2] - 0.25 *
        exp( maxexponenty * ln(10.0));
    drawtextW( bposX, -bposY, l, sta);
    stal := StringNumber( cpoints[ II, l], maxexponentX);
    sta2 := StringNumber( -cpoints[ II, 2], maxexponentY);
    sta := stal + ',' + sta2;
    {reverses maxexponent to base lo number};
    cposY := -cpoints[II, 2] - 0.25 *
        exp( maxexponenty * ln(10.0));
    drawtextW( cpoints[ II, l], -cposY, l, sta);
    GotoXY( 39,23);
    write( 'Esc')
end;
```

\{actually plots the points which were stored in arrays. Also writes design information into auxiliary windows\}
procedure graphit;
var theta, zeta, Wn, Temp: real; maxexponentX: integer; ch: char;
\{controlled by graphit. Checks the keyboard at each plot iteration to see if the user wishes to pause and view auxiliary information, or abort\}
procedure check_keyboard_buffer;
var stopped: boolean;
begin
if KeyPressed then begin
read ( kbd, ch) ; if ( $\mathrm{ch}=\# 32$ ) then begin
stopped := true;

```
    Sound (880);
    Delay (200);
    NoSound;
    pop_up;
    while stopped do
    begin
        read (kbd, ch);
        if (ch = #32) and (II < Imax) then
        stopped := false
        else
            if (ch = #27) then
                begin
                    II := IMax + I;
                    stopped := false
                    end
    end; {while}
if ch = #32 then
    SwapScreen
    end {if ch}
    end {if KeyPressed}
    end; {check keyboard buffer}
{write root locus variables while plotting}
procedure write_to_subwindows;
begin
    GOTOXY ( 7, 4);
    Write (Kstore2 [II]:5:3);
    if ( apoints[ II, l] < IE-6) and
        ( apoints[ II, 2] < IE-6) then
        zeta := 1.0
    else
        begin
            theta := ArcTan
            ( apoints[ II,2] / apoints[ II,I]);
            zeta := cos (theta)
        end;
    Wn := sqrt( sqr( apoints[ II,2]) +
            sqr( apoints[ II,l]));
    GoToXY ( 37,4);
    if zeta < l then
        begin
            write (zeta:5:2);
            GoTOXY( 7,7);
            write( Wn:7:4)
        end
```

```
        else
    begin
        write( 'ovrdp');
        GotoXY( 7,7);
        write( 'ovrdp')
    end
    end; {write to sub windows}
begin {graphit}
    initializeGraph;
    with World[ l] do {invert world coord
    to Cartesian type coord}
begin
    Temp := Yl;
        Yl := Y2;
        Y2 := Temp
    end; {with}
SelectWorld (l);
SelectWindow (1);
DrawAxis( 8,7,0,0,0,0, -1, 0, false);
GotoXY ( 7,23);
writeln
    ('stop/start: space-bar ');
SelectWindow( 2);
maxexponentX := GetExponent( XLO);
Temp := 0.1 *
    exp( maxexponentX * ln(10.0));
Drawline( Temp, 0, -Temp, 0); {cross hair, origin}
repeat
    if linerequested then
        begin
            DrawLine (apoints[ II-1, 1], -apoints[ II-1, 2],
                apoints[ II, 1], -apoints[ II, 2]);
            DrawLine (bpoints[ II-1, 1], -bpoints[ II-1, 2],
                        bpoints[ II, 1], -bpoints[ II, 2]);
            if thirdorder then
                    DrawLine (cpoints[ II-1, 1],
                        -cpoints[ II-1, 2],
                                cpoints[ II, 1], -cpoints[ II, 2])
        end
    else
        begin
            DrawPoint (apoints[II,1], -apoints[II,2]);
            DrawPoint (bpoints[II,I], -bpoints[II,2]);
            if thirdorder then
                DrawPoint (cpoints[II,1], -cpoints[II,2])
        end; {else}
    write_to_subwindows;
    Delay (1000);
```

```
            check_keyboard_buffer;
            II :=- II + I;
            until II > IMax;
            if ch <> #27 then
            repeat until KeyPressed;
    end; {graphit}
begin {root loci}
    firstRun := true;
    putframe;
    repeat
            mark( heapaddress);
            getpoints;
            if not abort then
            begin
            InitGraphic;
                        ClearScreen;
                    HiResColor(Yellow);
                    graphit;
                    LeaveGraphic
            end;
            exit := false;
            ClrScr;
            putframe;
            check_response( 'Continue root locus? 1,4);
            release( heapaddress)
    until exit
end; {root loci}
```


## APPENDIX III "INCLUDE" FILES - NOTES

I. Since nesting is not allowed, Include Files used in this subprogram were compiled as part of the editor program (Fadden 1986). Include Files required are:

Turbo Graphix Toolbox files (Borland 1985)

```
TYPEDEF.SYS
    KERNEL.SYS
GRAPHIX.SYS
WINDOWS.SYS
AXIS.HGH
    4X6.FON
    8X8.FON
```

Editor files (Fadden 1986)
TEXTBOX.INC
PUTFRAME.INC
II. The following changes were made to Turbo Graphix Toolbox version 1.05:
A. file TYPEDEF.SYS
change MaxPlotGlb $=400$.
B. files TYPEDEF.SYS, KERNEL.SYS, GRAPHIX.SYS, WINDOWS.SYS, AXIS.HGH
change Window to WWindow
C. file AXIS.HGH

1. change line 193 to NPoints := Delta div 29
2. change line 199 to for $i:=1$ to NPoints -1 do
3. Change line 201 to xs $:=x s+29+$ Balance

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