

REAL TIME OPTIMAL IMPLEMENTATION OF STABILIZING CONTROLLER FOR INVERTED PENDULUM

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ABSTRACT

This paper present real time control of an inverted pendulum. The system is inherently unstable and multivariable. It is mostly used in laboratories to study, verify and validate new control ideas. The dynamic model of the system was derived based on Lagrange approach and it was linearized. Linear Quadratic Regulator (LQR) controller was designed to stabilize the system in an upright position. The robustness of the control algorithm was tested based on disturbance rejection. Simulation and experimental results showed a good performance was achieved and the controller is robust to external disturbances.

KEYWORDS: *Nonlinear; LQR; inverted pendulum; disturbance rejection; Lagrange*

1.0 INTRODUCTION

An inverted pendulum is a multivariable, an under-actuated, nonlinear, and unstable system, (Mus & Tovornik, 2006), (Riachy et al, 2007), (Jerome et al, 2013). Due to these dynamics, it is mostly used by researchers to investigate the control algorithms. Modelling and control of the nonlinear inverted pendulum system is one of the major areas of research with lots of potentials in the field of robotics and automation. Various researchers have proposed different control algorithms and techniques for swing up, tracking and stabilization of an inverted pendulum.

Balancing an inverted pendulum mobile robot using LQG and LQR has been proposed by Hauser and Saccon (2005), and their performances were compared. Modelling and a predictive controller based on the nonlinear model of the system have been presented in Chalupa (2008). Similarly, Prasad (2014), Singh and Yadav (2012) and Gupta (2014) compared the stabilization and swing up control performances of LQR and PID controllers for an inverted pendulum. The stabilization control of an inverted pendulum using Pole Placement and LQR has been presented in Kumar et al (2012), a comparison of the control algorithms has also been presented.

Furthermore, an MODE-based optimized LQR developed in Tijani (2013) shows a superior performance. Singh et al (2014) used a modified PSO based PID sliding mode

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for swing up and stabilization control of an inverted pendulum. In addition, Chakraborty et al (2013) investigated the optimization of PID controller using Genetic Algorithm. A dynamic modelling and optimal control of wheel inverted pendulum were proposed in Shamsudin et al (2013) using an optimally tuned partial-state PID. Swing up and stabilization of double inverted pendulum using LQR and LQR based fuzzy was proposed in Bhangal (2013), and their performances were analyzed and compared. An intelligent control algorithm has also been proposed in A-hadithi (2012) and implemented for swing up and stabilization of a double inverted pendulum system.

The combinations of intelligent and conventional control have shown good performances and robustness of the control algorithms. An Adaptive Neural Network for motion control of a wheeled, inverted pendulum has also been presented in Yang et al (2014). In Chalupa and Bobal (2008), Model Predictive Controller has been proposed. A novel PSO based Sliding Mode Control for stabilization of an inverted pendulum was presented by Singh et al (2014). In Brisilla and Sankaranarayanan (2015), a stabilization of an inverted pendulum using a nonlinear control algorithm has been presented.

However, to implement a simple control algorithm and obtained stability under external disturbances is a big challenging task. Thus, using LQR the stability and control of all system poles are guaranteed. This paper proposed LQR for real time stabilization of an inverted pendulum under wind disturbances of magnitude of 0.2 N. Time response specification and level of disturbances rejections were used as the performance index of the control algorithms.

2.0 SYSTEM DESCRIPTION

In this part, the inverted pendulum dynamics is presented.

2.1 An Inverted Pendulum System

In this work, the laboratory scale feedback digital pendulum system version 33-000-V73 was used. The system consists of the following parts; the cart, two pendulums, a rail, and a D.C motor. The pendulum is hinged at the center of the cart in a manner that they can rotate freely for 360°. The cart can move freely horizontally on the rail with the aid of the D.C motor. The mechanical system is as shown in Figure1.

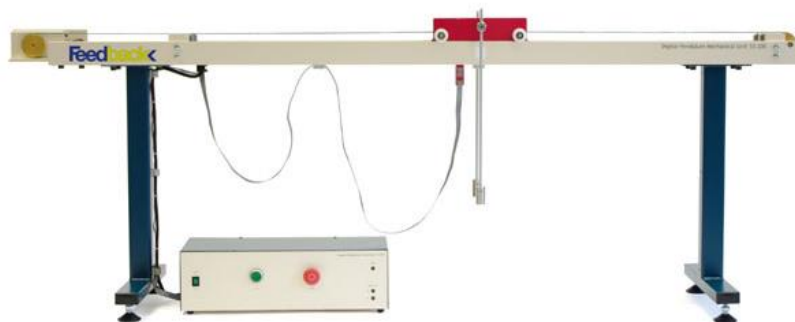


Figure 1. The laboratory-based Inverted Pendulum

When the pendulum is in the vertical position, the system is unstable and when the pendulum is in a downward position, the system is completely stable. The pendulum is completely unstable for small deviations from the equilibrium position. Figure 2 shows the activity zone of the inverted pendulum within which control can be achieved.

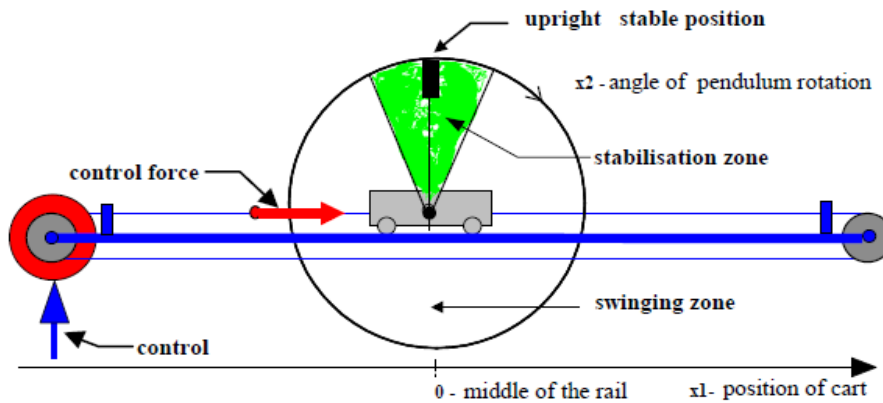


Figure 2. Activity zone of the control algorithm.

2.1.1 System Modelling

The system schematics diagram is shown in Figure 3, with θ , x , and $f(t)$ as pendulum angle, the cart displacement and applied force in (N) respectively. M is the mass of the cart (kg), l is the pendulum length (m), and I is the moment of inertia of the rod from the centre of mass (kgm^2). C and b are the translation and viscous damping of the cart (Ns/m) and that of the pendulum (Nm/rad) respectively (Zhang and Tu, 2006).

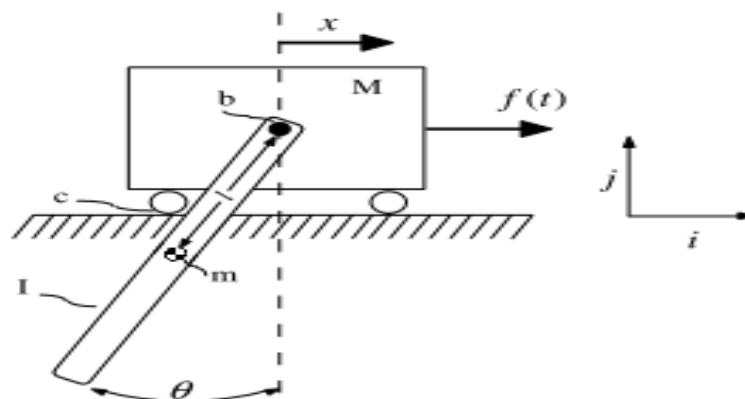


Figure 3. Schematic diagram of the pendulum-cart system

The energy in the system is simply its kinetic energy and potential energy. The total kinetic energy of the system can be expressed as;

$$T = T_M + T_m \tag{1}$$

where T is the total kinetic energy, T_M is the kinetic energy of the cart and T_m is the kinetic energy of the pendulum. Hence,

$$T_M = \frac{1}{2}M\dot{x}^2 \quad (2)$$

$$T_m = \frac{1}{2}mV^2 + \frac{1}{2}I\omega^2 \quad (3)$$

where V is the velocity of the pendulum's centre of mass and I is the moment of inertia of the pendulum around the centre of mass and ω is the angular velocity of the pendulum. Also, the position vector can be given as;

$$r = (x + l\sin\theta)\hat{i} + l\cos\theta\hat{j} \quad (4)$$

And the velocity vector can be express as;

$$V = \frac{dr}{dt} = (\dot{x} + l\cos\theta\dot{\theta})\hat{i} + l\sin\theta\dot{\theta}\hat{j} \quad (5)$$

Putting $\omega = \dot{\theta}$, the kinetic energy of the pendulum becomes;

$$T_m = \frac{1}{2}m(\dot{x}^2 + 2\dot{x}l\cos\theta\dot{\theta} + l^2\cos^2\theta\dot{\theta}^2 + l^2\sin^2\theta\dot{\theta}^2) + \frac{1}{2}I\dot{\theta}^2 \quad (6)$$

Hence, the total kinetic energy of the system is as;

$$T = T_M + T_m = \frac{1}{2}M\dot{x}^2 + \frac{1}{2}m(\dot{x}^2 + 2\dot{x}l\cos\theta\dot{\theta} + l^2\cos^2\theta\dot{\theta}^2 + l^2\sin^2\theta\dot{\theta}^2) + \frac{1}{2}I\dot{\theta}^2 \quad (7)$$

In addition, the total potential energy of the system is given as;

$$V = V_M + V_m \quad (8)$$

where V_M is the potential energy of the cart and V_m is the potential energy of the pendulum. But, $V_M = 0$, and the term V_m is given as

$$V_m = mgl\cos\theta \quad (9)$$

Therefore, the total potential energy can be written as;

$$V = V_M + V_m = mgl\cos\theta \quad (10)$$

The system dynamics equation of motion can be derive using Lagrange's equation as in Mishra and Chandra (2014).

$$\frac{d}{dt}\left(\frac{\partial L}{\partial \dot{x}}\right) - \frac{\partial L}{\partial x} = \Sigma_x \quad (11)$$

where, (L) is expressed in form of kinetic energy and potential energy of the system given as;

$$L = T - V \quad (12)$$

Substituting T and V in Equation (12), it is obtained as;

$$L = \frac{1}{2}M\dot{x}^2 + \frac{1}{2}m(\dot{x}^2 + 2\dot{x}l\cos\theta\dot{\theta} + l^2\cos^2\theta\dot{\theta}^2 + l^2\sin^2\theta\dot{\theta}^2) + \frac{1}{2}I\dot{\theta}^2 - mgl\cos\theta \quad (13)$$

Substituting Equation (13) into (11) and solving for the partial derivatives yields;

$$\frac{d}{dt}(M\dot{x} + m\dot{x} + ml\cos\theta\dot{\theta}) = f(t) - c\dot{x} \quad (14)$$

The above equation can be represented as;

$$(M + m)\ddot{x} + ml\cos\theta\ddot{\theta} + ml\sin\theta\dot{\theta}^2 + c\dot{x} = f(t) \quad (15)$$

Moreover, for θ is given as;

$$\frac{d}{dt}\left(\frac{\partial L}{\partial \dot{\theta}}\right) - \frac{\partial L}{\partial \theta} = \Xi_{\theta} \quad (16)$$

Substituting Equation 13 into 16 and solve for the partial derivatives yields;

$$\frac{d}{dt}(ml\dot{x}\cos\theta + ml^2\dot{\theta} + I\dot{\theta}) + ml\dot{x}\sin\theta\dot{\theta} + mgl\sin\theta = -b\dot{\theta} \quad (17)$$

Also, Equation 17 can be further simplified as;

$$(ml^2 + I)\ddot{\theta} + ml\ddot{x}\cos\theta - ml\dot{x}\sin\theta\dot{\theta} + ml\dot{x}\sin\theta\dot{\theta} + mgl\sin\theta + b\dot{\theta} = 0 \quad (18)$$

After re-arranging, the overall dynamic equations of the system were obtained as;

$$\begin{cases} (M + m)\ddot{x} + c\dot{x} + ml\cos\theta\ddot{\theta} + ml\sin\theta\dot{\theta}^2 = f(t) \\ (ml^2 + I)\ddot{\theta} + ml\ddot{x}\cos\theta + mgl\sin\theta + b\dot{\theta} = 0 \end{cases} \quad (19)$$

2.1.1.1 Model linearization

To linearize the nonlinear system, two points have to be considered for equilibrium points. The pendulum in the upright position (unstable, $\theta = 0$) and pendulum in the downward position (stable, $\theta = \pi$). This can be achieved by Taylor's series approximation, for a small angle deviation around an equilibrium point θ_o ;

$$\theta = \theta_o + \varepsilon \quad (20)$$

Taylor's series first order approximation is given as;

$$f(\theta) \approx f(\theta_o) + \varepsilon \frac{df}{d\theta} \quad (21)$$

where $\varepsilon \approx \theta$, a small angle deviation from equilibrium and $\varepsilon^2 \approx 0$, higher order neglected. Pendulum in upright position (unstable, $\theta = 0$). The following functions can be linearized as

$$\begin{aligned} \sin\theta &\cong \theta, \\ \cos\theta &\cong 1, \\ \dot{\theta}^2 &= 0 \end{aligned}$$

Thus, substituting these into Equation (19), the motion equations were obtained as;

$$\begin{cases} (M+m)\ddot{x} + c\dot{x} + ml\ddot{\theta} = f(t) \\ (ml^2 + I)\ddot{\theta} + ml\ddot{x} + mgl\theta + b\dot{\theta} = 0 \end{cases} \quad (22)$$

Pendulum downward position (stable, $\theta = \pi$). The following functions can also be linearized as;

$$\begin{aligned} \sin\theta &\cong -\theta, \\ \cos\theta &\cong -1, \\ \dot{\theta}^2 &= 0 \end{aligned}$$

Thus, the motion Equations were obtained as;

$$\begin{cases} (M+m)\ddot{x} + c\dot{x} - ml\ddot{\theta} = f(t) \\ (ml^2 + I)\ddot{\theta} - ml\ddot{x} - mgl\theta + b\dot{\theta} = 0 \end{cases} \quad (23)$$

Hence, the system equations are represented in state space form as;

$$\dot{x} = Ax + Bu \quad (24)$$

$$y = Cx \quad (25)$$

The states vector of the system can be assigned as;

$$z = \begin{bmatrix} z_1 \\ z_2 \\ z_3 \\ z_4 \end{bmatrix} = \begin{bmatrix} \theta \\ \dot{\theta} \\ x \\ \dot{x} \end{bmatrix} \quad (26)$$

where, θ , $\dot{\theta}$, x and \dot{x} are the pendulum angle, angular velocity, Cart displacement and velocity of the Cart respectively (Kizir et al, 2010). Hence the dynamics equations can be represented in state space form as;

$$\begin{bmatrix} \dot{z}_1 \\ \dot{z}_2 \\ \dot{z}_3 \\ \dot{z}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ \frac{(M+m)gml}{q} & \frac{(M+m)b}{q} & 0 & \frac{-cml}{q} \\ 0 & 0 & 0 & 1 \\ \frac{(ml)^2g}{q} & \frac{bml}{q} & 0 & \frac{-c(I+ml^2)}{q} \end{bmatrix} \begin{bmatrix} z_1 \\ z_2 \\ z_3 \\ z_4 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{mlg}{q} \\ 0 \\ \frac{(I+ml^2)}{q} \end{bmatrix} f \quad (27)$$

$$y = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} z_1 \\ z_2 \\ z_3 \\ z_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \end{bmatrix} f \quad (28)$$

3.0 FULL STATE FEEDBACK CONTROLLER DESIGN

In this section, LQR controller was designed and implemented on real time system. Controllability, observability and stability test was carried out before applying the controller.

3.1.1 Controllability Test

The system is said to be controllable if an input to the plant can take all the states from a desired initial state to a desired final state in a final time interval, otherwise is uncontrollable. Table 1 show the system parameters. The controllability matrix is given as;

$$M_c = [B \quad AB \quad A^2B \quad A^3B] \quad (29)$$

Table 1. System Parameters

Parameter	Value
Mass of cart (M)	2.4 kg
Mass of pole (m)	0.23 kg
Length of pole (l)	0.38 m
Moment of inertia of the pole (I)	0.099 kg/m ²
Coefficient of friction of cart (b)	0.05 Ns/m
Damping coefficient of pendulum (d)	0.005 Nms/rad
Gravity (g)	9.8 m/s ²

For controllability, the rank (M_c) of the matrix must be equal to the number of states of the system. Using the MATLAB command, the M_c was obtained to be 4 hence the system is controllable.

3.1.2 Observability Test

If all states of a system can be determined from an observation of $y(t)$ over a finite time interval, the system is completely observable otherwise is unobservable (Ogata, 1997). The observability matrix is given as;

$$M_o = [C^T \quad A^T C^T \quad (A^T)^2 C^T \quad (A^T)^3 C^T] \quad (30)$$

For observability, the rank (M_o) of the matrix must be equal to the number of the outputs of the system. Using the MATLAB command, the M_o was obtained as 4 hence the system is completely observable.

3.1.3 Stability Test

Inverted pendulum system is an unstable system, thus before applying any control algorithms there is need to test the stability of the system before and after control action so that the performance of the controller can be observed. Using the pole zero map

shown in Figure 4, it can be observed that some poles of the system is in the right hand plant hence the system is unstable.

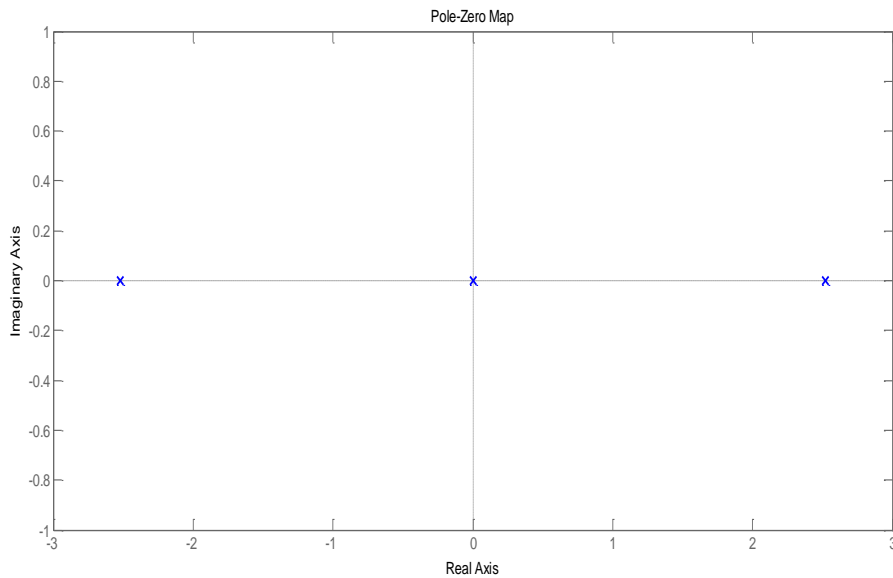


Figure 4. Poles and zero map of the open loop system

3.2 LQR Control Design

The LQR is a full state feedback controller that is usually used in industries for mechanical system control. The control of inverted pendulum using classical PID is mostly difficult because the system has higher state variables than the controller. Hence, as shown in Figure 5, a full state feedback controller is most suitable (Hauser and Saccon, 2005) and (Tahir et al, 2017).

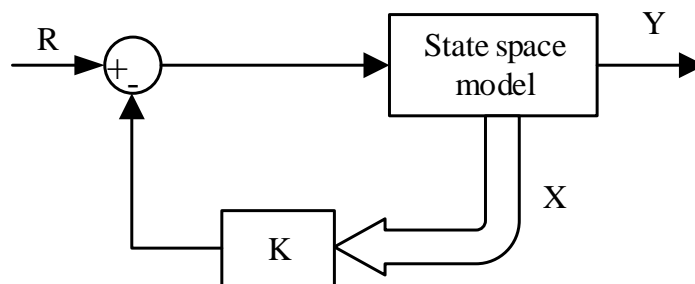


Figure 5. Typical LQR control system

In this control technique, a control law is selected $u = \psi(x)$ to regulate the state x and minimize the performance index:

$$J = \int_0^{\infty} x^T(t)Qx(t) + u^T(t)Ru(t) \tag{31}$$

where J is the performance index function, $Q \geq 0$ and $R > 0$ are weight matrices for the state variable $x(t)$ and control variable $u(t)$ respectively. Q and R are the semi-positive definite matrix and positive definite matrix respectively (Ogata, 2010) and

(Tahir et al, 2016). Thus, the gain vector K can be obtained to satisfy the feedback control law given as in (Tahir et al, 2017).

$$u = -Kx = -R^{-1}B^T Px \quad (32)$$

where P is the solution of the Riccati equation given as;

$$A^T P + PA + Q - PBR^{-1}B^T P = 0 \quad (33)$$

In which;

$$K = R^{-1}B^T P \quad (34)$$

Bryson's rule states that a first choice of matrices Q and R is to select the matrices diagonals giving by;

$$Q_{ii} = \frac{1}{\text{maximum acceptable value of } X_i^2} \quad i \in \{1,2,3, \dots, l\} \quad (35)$$

$$R_{jj} = \frac{1}{\text{maximum acceptable value of } u_j^2} \quad j \in \{1,2,3, \dots, l\} \quad (36)$$

where X_i and u_i are the maximum expected value of the state and that of control signal respectively (Lingyan et al, 2009). Therefore, Q and R parameters were used in the following forms;

$$Q = \begin{bmatrix} Q_{11} & 0 & 0 & 0 \\ 0 & Q_{22} & 0 & 0 \\ 0 & 0 & Q_{33} & 0 \\ 0 & 0 & 0 & Q_{44} \end{bmatrix} \quad (37)$$

$$R = R_1 \quad (38)$$

The cart is constrained to lie between $-0.5 \leq x(t) \leq 0.5(m)$ and the input to the motor is constrained to lie between $-2.5 \leq u(t) \leq 2.5 (v)$. Thus, $Q_{11} = \frac{1}{0.2^2} = 25$, is the weight due to angle, assuming a 0.2 rad maximum deviation expected from the upright position. $Q_{33} = \frac{1}{0.2^2} = 25$, is the weight due to the position of the cart and was assume that the cart should not exceed 0.2m from the centre of the rail. $R_1 = \frac{1}{2.5^2} = 0.16$, is the weight due to the control voltage. The close loop controller gain for both simulation and experiment were obtained as $K = [58.2411 \quad 22.8208 \quad -12.5000 \quad -12.2993]$ and $K = [44.72 \quad 200.8 \quad -49.77 \quad -27.38]$ respectably.

4.0 RESULTS AND DISCUSSION

The real time optimal control implementation of an inverted pendulum was presented. LQR was designed and implemented for swing up and stabilization control of an inverted pendulum. A Gaussian distributed noise and wind disturbance of magnitude 0.2 N was used to test the robustness of the control algorithms in both simulations and experiments respectively..

4.1 Simulation Results

The system is completely stable after applying the controller. Figure 6 shows the poles zeros map of the system, with details of the poles, damping, frequency, and overshoot of the system.

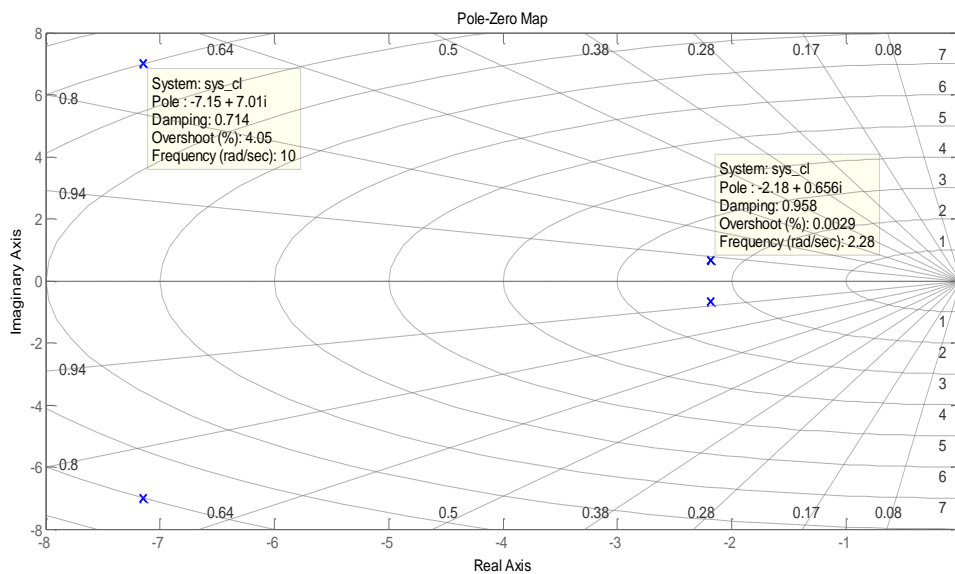


Figure 6. Closed loop poles-zeros map

The cart position, swing angle and control signal of the system were as shown in Figure 7, simulated with an initial condition of 0.1rad. The systems stabilized at 2 sec with 0.262 m undershoot of cart position and 0.08 m of the swing angle.

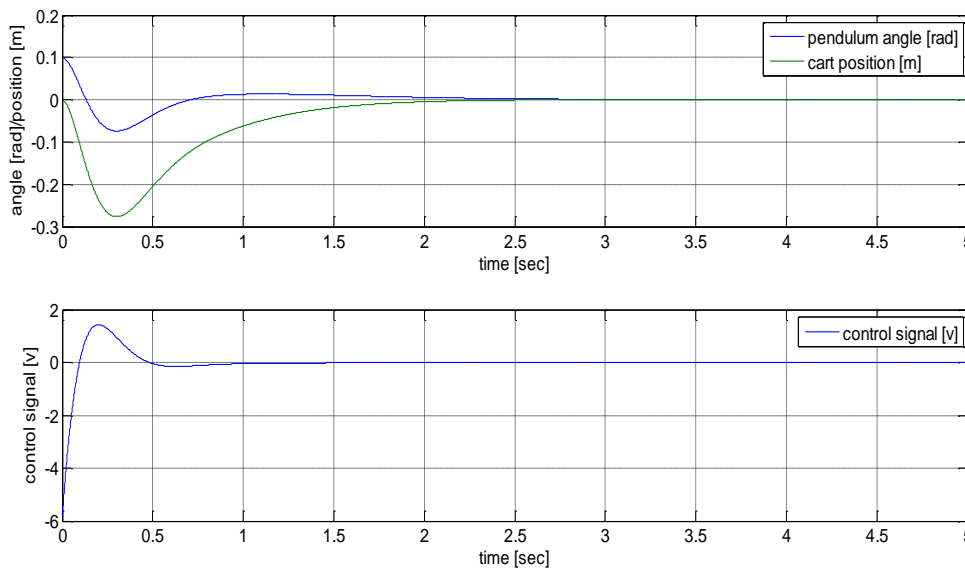


Figure 7. LQR control action of the system

In simulation, a Gaussian distributed noise of mean = 0.0001 and variance = 0.00001 was injected into the control system, at a sampling time of 0.5sec. As shown in Figure 8, the performance of the controller remained the same however, some negligible amplitude and frequency variations can be observed in the system.

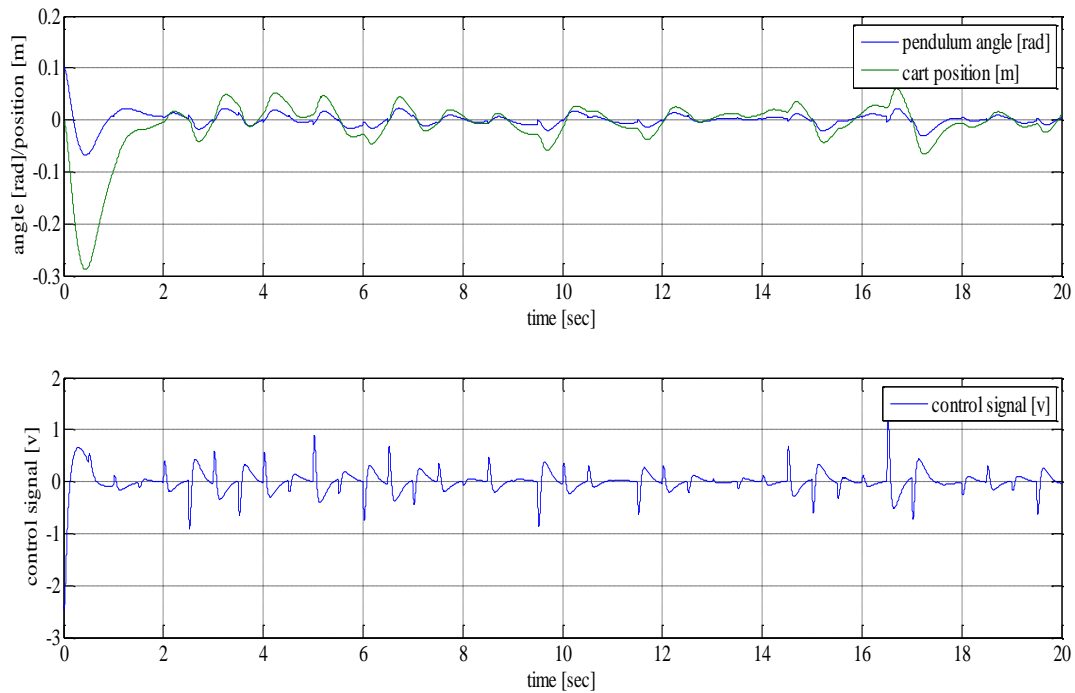


Figure 8. Disturbance injection response of the system with LQR

4.1.1 Experimental Results

The experiment was conducted at simulation and control laboratory Glasgow Caledonian University UK. Figures 9 and 10 showed the photos of the real time experiment with swing up and stabilizing controller respectively, using the feedback instruments laboratory scale inverted pendulum system.

The algorithm was implemented in real time, the control signal, swing angle, and cart position is as shown in Figure 11, in which the pendula swing up and stabilized within the constraints voltage. It was observed that the controller shows a very good performance as it stabilized the system within the stabilization zone of 0.2 m. A good stabilization and swing up control was also achieved as in Figure 12, with the control action only starting at 13 sec. This is also within the stabilization zone of the system.

In addition, the experiment was conducted under a wind disturbance magnitude of 0.2 N and a very good performance was also achieved as shown in Figure 13 but from the control signal it can be observed that under disturbance, more voltage was consumed.



Figure 9. Pendulum in swing-up mode by swinging controller



Figure 10. Pendulum in an upright position controlled by stabilizing controller

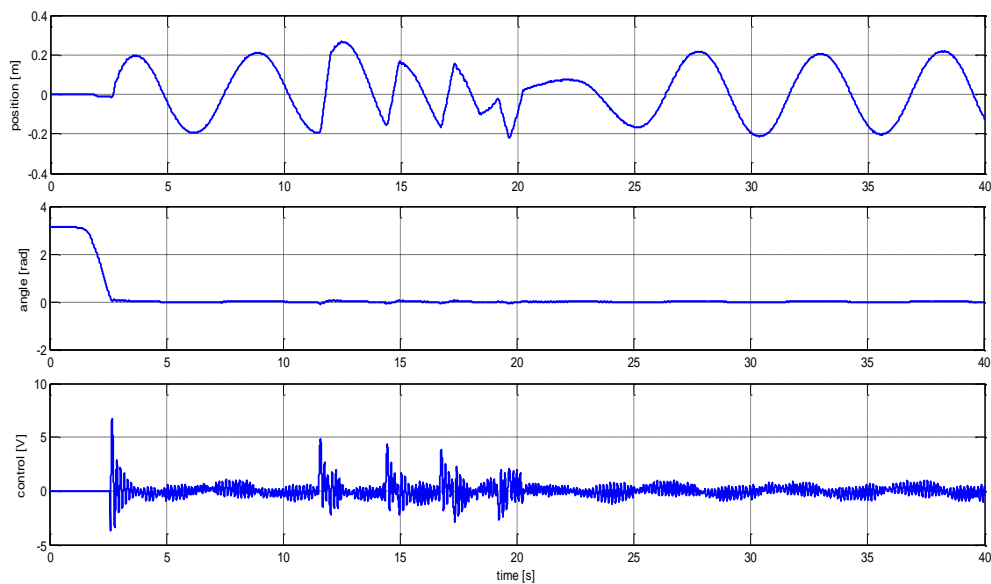


Figure 11. LQR stabilizing action experimental results

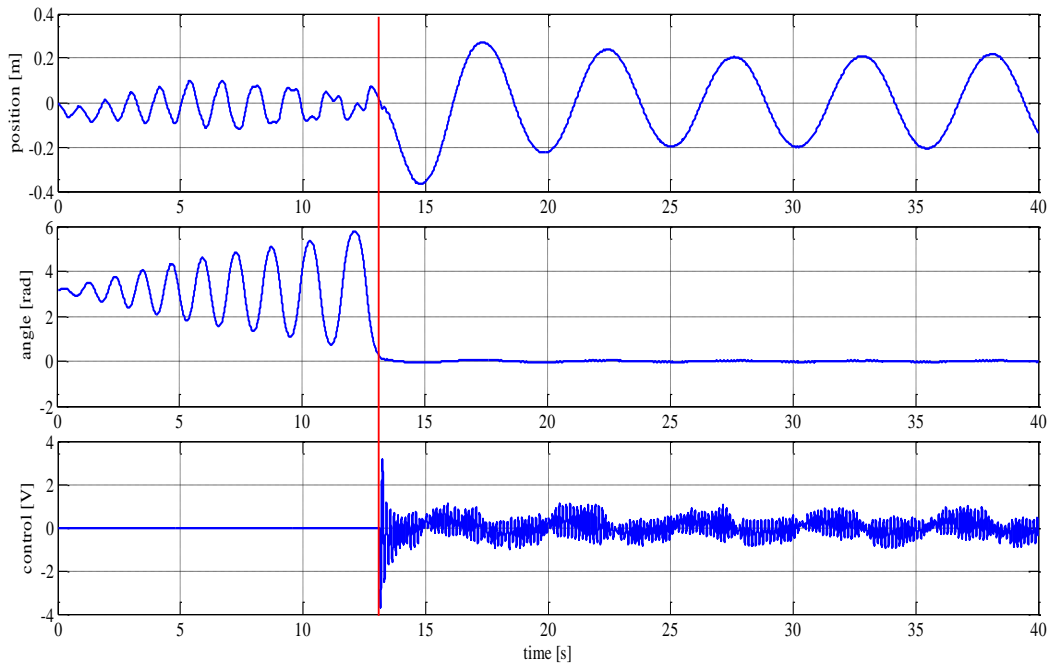


Figure 12. LQR Swing-up/stabilizing action experimental results

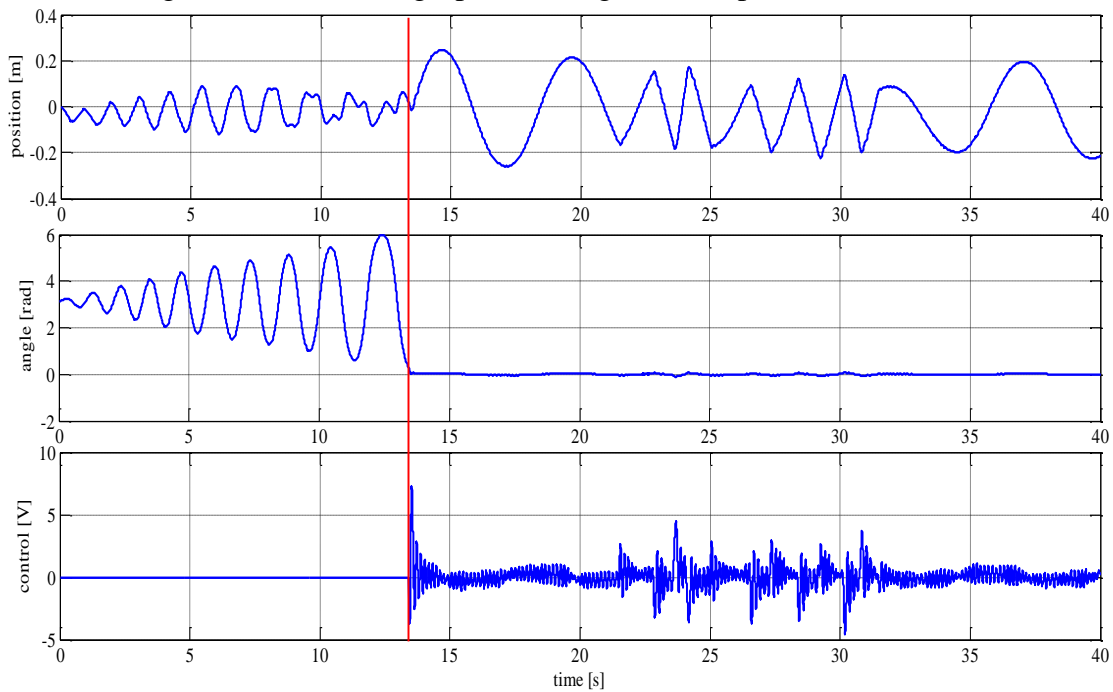


Figure 13. LQR Swing-up/stabilizing action experimental results under wind disturbance

5.0 CONCLUSION

In this paper, nonlinear inverted pendulum was linearized using Taylor series approach and LQR controller was designed to swing up and stabilize the pendulum. It was also implemented on real time system under wind disturbance. Time response specification and level of disturbances rejection were used as the performance index. Based on

Simulation and experimental results, a very good performance was achieved and the controller is robust to external disturbances. To further reduce the oscillations amplitude and frequencies, enhance and improve the modelling and control for real time accuracies, frictional coefficients should be taking into consideration.

ACKNOWLEDGEMENT

The authors are grateful to Glasgow Caledonian University, Scotland UK and Abubakar Tafawa Balewa University (ATBU) Bauchi, Nigeria for providing financial support and research resources.

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