





CubeSat Background

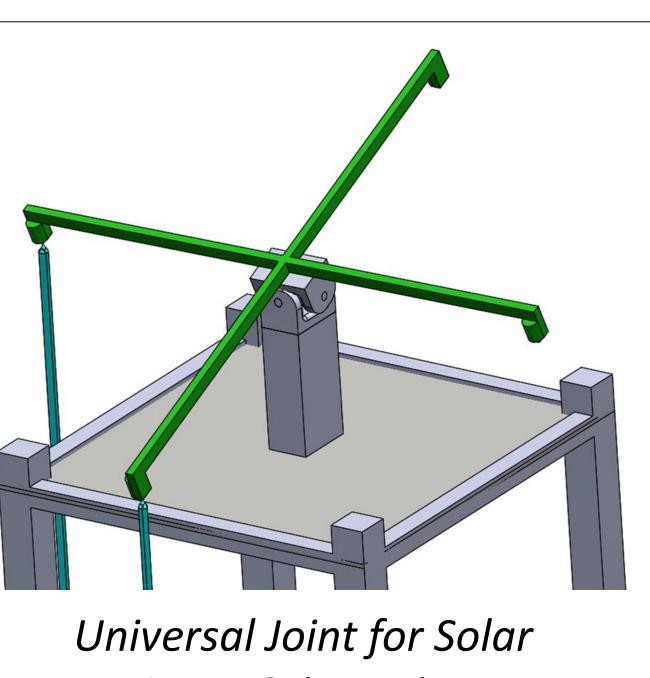
- Design originated in late 1990's
- Small satellites built up from individual units
- 1 unit = 10cm x 10cm x 10cm cube
- CubeSats make space more accessible to small institutions
- Causing a revolution in Space, Defense, and Conservation Research



Photograph of San Francisco, CA taken from a CubeSat and 3 Unit CubeSat (Courtesy of Planet Labs Inc., San Francisco, CA)

Design Philosophy

- Use a Universal Joint to allow for full range spatial pointing
- Use smart materials as prismatic actuators to control position



Array Orientation

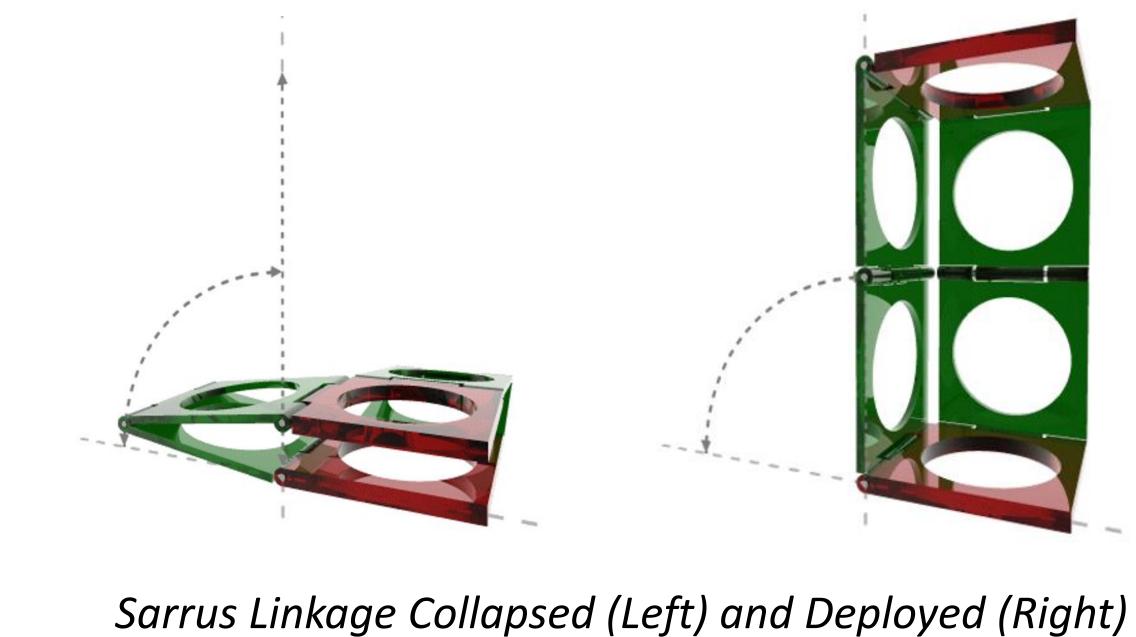
Development of a Self-Orienting Solar Array for CubeSats By: Eric McGill

Advised by Dr. Andrew Murray and Dr. David Myszka **Department of Mechanical and Aerospace Engineering**

Objective: Develop a solar array design for the CubeSat family of space craft. The design adheres to the following requirements: 1) Must have the ability to orient Solar Panels towards the Sun to maximize solar power generation; 2) Must within the space of a 3U CubeSat for launch; 3) The movement must use no power or computation internal to the CubeSat

Creating Clearance

- Universal Joint movement requires separation from the CubeSat Chassis Solution: Sarrus Linkage Straight-line Mechanism developed in the
- 1850's
- Spatial 6-Revolute Joint Mechanism

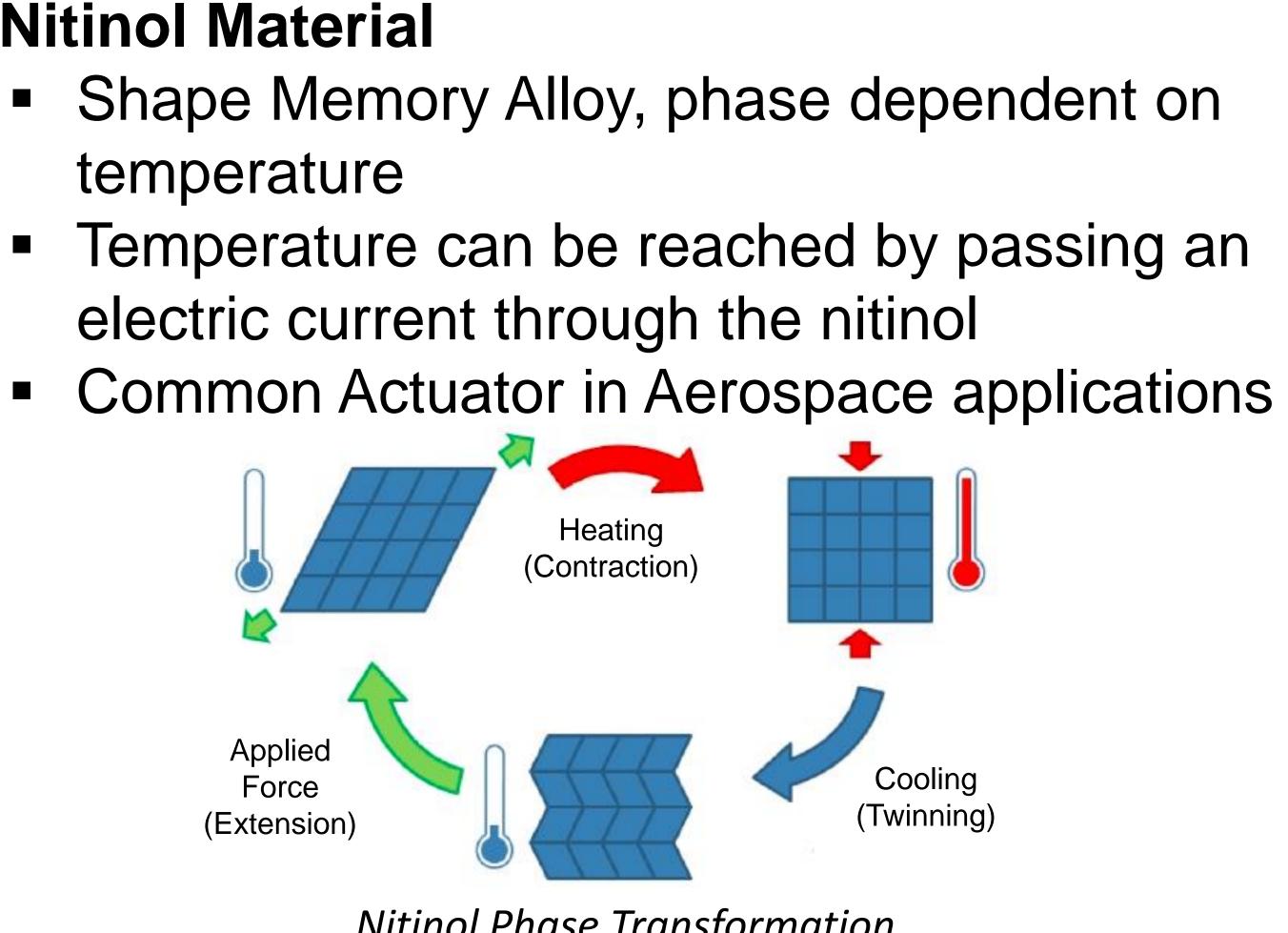


Nitinol Material

- Shape Memory Alloy, phase dependent on temperature
- Temperature can be reached by passing an electric current through the nitinol
 - Heating (Contraction) Applied Force (Extension) Nitinol Phase Transformation

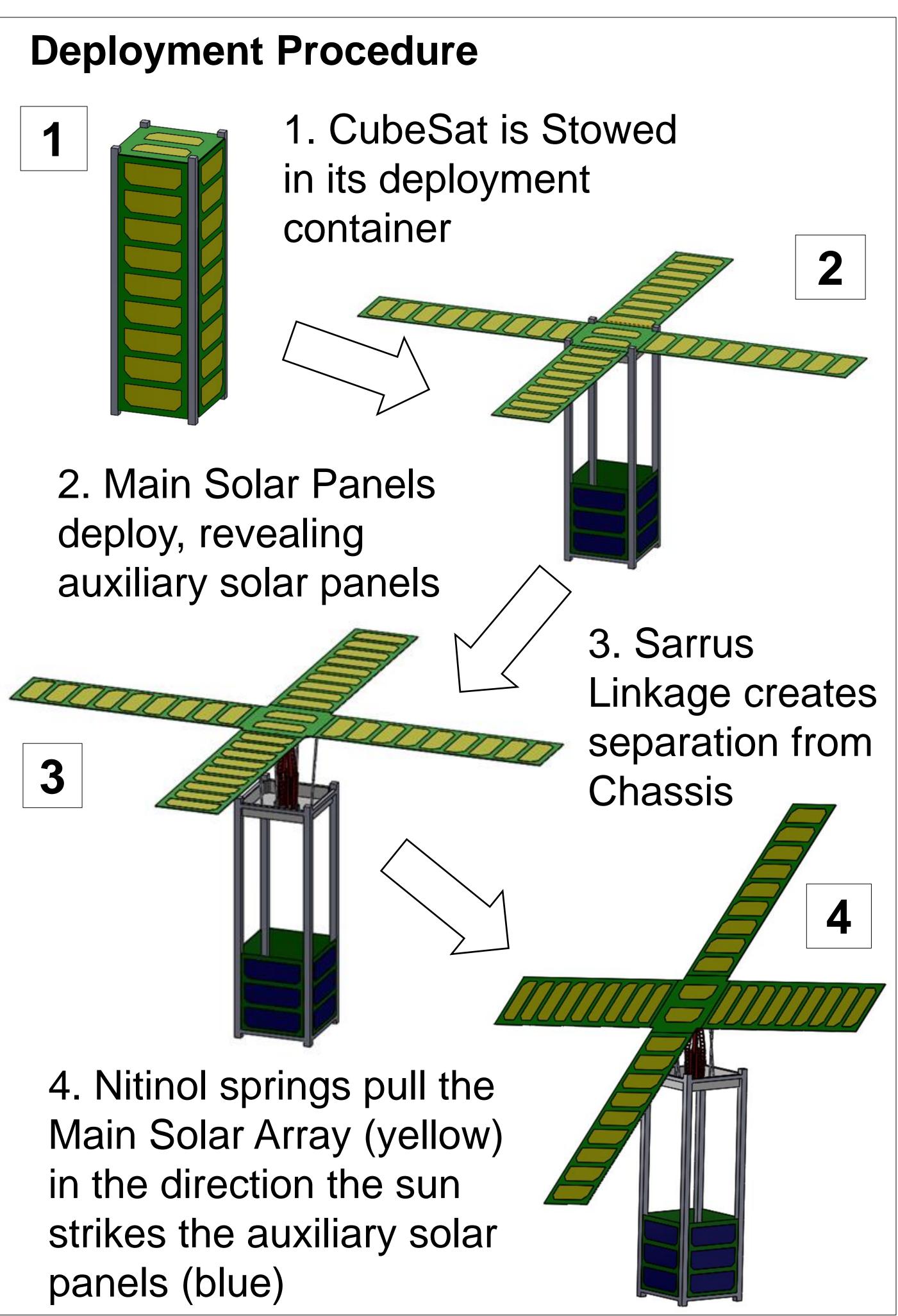
mage Source: J. P. Swensen and A. M. Dollar, "Optimization of Parallel Spring Antagonists

ctuators," in IEEE International Conference on Robotics & Automation (ICRA), Hong Kong, China, 2014.



Actuation Design







Pass electric current through Nitinol Springs Electric current from auxiliary solar panels • 4 Nitinol Springs, acting as antagonistic pair