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ABSTRACT OF DISSERTATION

Shuhua Lai

The Graduate School University of Kentucky 2006 Subdivision Surface based One-Piece Representation

ABSTRACT OF DISSERTATION

A dissertation submitted in partial fulfillment of the requirements for the degree of Doctor of Philosophy in the College of Engineering at the University of Kentucky

By

Shuhua Lai

Lexington, Kentucky

Director: Dr. Fuhua (Frank) Cheng, Professor of Computer Science

Lexington, Kentucky

2006

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ABSTRACT OF DISSERTATION

Subdivision Surface based One-Piece Representation

Subdivision surfaces are capable of modeling and representing complex shapes of arbitrary topology. However, methods on how to build the control mesh of a complex surface are not studied much. Currently, most meshes of complicated objects come from triangulation and simplification of raster scanned data points, like the Stanford 3D Scanning Repository. This approach is costly and leads to very dense meshes.

Subdivision surface based one-piece representation means to represent the final object in a design process with only one subdivision surface, no matter how complicated the object's topology or shape. Hence the number of parts in the final representation is always one.

In this dissertation we present necessary mathematical theories and geometric algorithms to support subdivision surface based one-piece representation. First, an explicit parametrization method is presented for exact evaluation of Catmull-Clark subdivision surfaces. Based on it, two approaches are proposed for constructing the one-piece representation of a given object with arbitrary topology. One approach is to construct the one-piece representation by using the interpolation technique. Interpolation is a natural way to build models, but the fairness of the interpolating surface is a big concern in previous methods. With similarity based interpolation technique, we can obtain better modeling results with less undesired artifacts and undulations. Another approach is through performing Boolean operations. Up to this point, accurate Boolean operations over subdivision surfaces are not approached yet in the literature. We present a robust and error controllable Boolean operation method which results in a one-piece representation. Because one-piece representations resulting from the above two methods are usually dense, error controllable simplification of one-piece representations is needed. Two methods are presented for this purpose: adaptive tessellation and multiresolution analysis. Both methods can significantly reduce the complexity of a one-piece representation and while having accurate error estimation.

A system that performs subdivision surface based one-piece representation was implemented and a lot of examples have been tested. All the examples show that our approaches can obtain very good subdivision based one-piece representation results. Even though our methods are based on Catmull-Clark subdivision scheme, we believe they can be adapted to other subdivision schemes as well with small modifications.

KEYWORDS: Solid Modeling, Subdivision Surface, One-Piece Representation, Interpolation, Boolean Operation.

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8/15/2006

Subdivision Surface based One-Piece Representation

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8/15/2006

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To my parents, my wife Shuting and son Jianjian

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Chapter 1 Introduction

1.1 Motivation

Subdivision surfaces [1, 2, 3, 4, 7, 9, 11, 12, 17, 22] have become popular recently in graphical modeling and animation because of their capability in modeling and representing complex shapes of arbitrary topology, their relatively high visual quality, and their stability and efficiency in numerical computation. Subdivision surfaces can model and represent complex shapes of arbitrary topology because there is no limit on the shape and topology of the control mesh of a subdivision surface [1, 2, 3]. Actually, subdivision surfaces have already been used as primitives in several commercial systems such as Alias|Wavefronts's Maya, Pixar's Renderman, Nichimen's Mirai, and Newtek's Lightwave 3D.

Basically, subdivision is a method for generating smooth surfaces, which first appeared as an extension of splines to arbitrary topology control meshes [1, 2, 3]. Subdivision schemes can be considered as an algorithmic generalization of classical spline techniques enabling *control meshes* with arbitrary topology. They provide easy access to globally smooth surfaces of arbitrary shape by iteratively applying simple refinement rules to the given control mesh. A sequence of meshes generated by this process quickly converges to a smooth limit surface. For most practical applications, the refined meshes are already sufficiently close to the smooth limit after only a few refinement steps. Complex smooth surfaces can be derived in a reasonably predictable way from relatively simple meshes. There are many subdivision schemes that have been proposed in the past thirty years [1, 2, 3, 4, 13, 15, 23, 24, 25, 48]. The Catmull-Clark subdivision scheme [1], the Doo-Sabin subdivision scheme [2] and the Loop subdivision scheme [3] are the most wellknown, and are used in many high-end modeling and animation packages. There are quite a few other schemes. The Butterfly scheme [25] and a closely related scheme known as Modified Butterfly [24] also use triangle meshes. Others include the Kobbelt's Interpolation Quadrilateral scheme [23], the Local Surface Fitting scheme [48], and many variations on the above schemes. The rules given for the above schemes are usually only suitable for closed surfaces. Surfaces with boundaries need special case rules to handle the boundary without unpleasant artifacts.

With the parametrization technique for subdivision surfaces becoming available [8, 17, 18, 58] and with the fact that non-uniform B-spline and NURBS surfaces are special cases of subdivision surfaces becoming known [15], we now know that subdivision surfaces cover both *parametric forms* and *discrete forms*. Parametric forms are good for design and representation, and discrete forms are good for machining and tessellation (including *Finite Element* mesh generation). Hence, we have a representation scheme that is good for all graphics and CAD/CAM applications.

Research work for subdivision surfaces has been done in several important areas, such as surface texture mapping [60, 116], surface interpolation [11, 23, 24, 25, 26, 55], exact surface evaluation [8, 16, 17, 18, 20], surface trimming [12], Boolean operations [7, 64, 92], deformation [117], mesh editing [21, 110, 111], and computer animation [9, 19] etc.

Although subdivision surfaces are capable of modeling and representing complex shapes of arbitrary topology and are well studied in many applications, methods on how to build the control mesh of a complex surface are not studied much. Currently, most meshes of complicated objects come from triangulation and simplification of raster scanned data points, like the Stanford 3D Scanning Repository. This approach is costly and leads to very dense meshes.

The objective of this research work is to develop necessary mathematical theories and geometric algorithms to support subdivision surface based one-piece representation. Subdivision surface based one-piece representation means to represent the final object in a design process with only one subdivision surface (i.e. a sparse control mesh), no matter how complicated the object's topology or shape. No decomposition of the object into simpler components is necessary. Hence the number of parts in the final representation is always the minimum: one. Another goal of this work is to build a system that can represent any complicated 3D object and their Boolean operation results with only one sparse mesh structure. Once every given object can be represented with one simple mesh, it would be very convenient and efficient to render, manipulate, store and transmit any virtual environment.

1.2 Subdivision Surfaces

We consider primarily *stationary subdivision* schemes in this research work, which means that the choices of the refinement rules do not depend on the subdivision level. Once a mesh is refined, the old mesh will not be used in computing the next level of vertices. The positions of vertices for the next subdivision step only rely on the topology and position of the current mesh. This requirement makes the implementation highly efficient and also makes the analysis of subdivision surfaces much simpler.

Many different schemes exist for the actual subdivision process [1, 2, 3, 4, 13, 15, 23, 24, 25, 48]. The first two were developed in 1978 by two different pairs of people. The Doo-Sabin scheme [2] and the Catmull-Clark scheme [1] are the most well-known, and are used in many high-end modeling and animation packages. A third popular scheme developed relatively recently, the Loop scheme [3], works only on triangle meshes. There are also other subdivision schemes, like $\sqrt{3}$ scheme [4] etc. Most subdivision methods are approximating. But there are also some interpolating subdivision schemes whose limit surface interpolates the given initial control points. Such scheme includes the Butterfly scheme [25] and the Kobbelt interpolating scheme [23] etc.

Given an initial mesh, subdivision computes a sequence of refined meshes converging to a *limit surface*. The refined meshes are obtained by adding new vertices to the mesh and connecting them with old vertices. The positions of new vertices are computed as functions of positions of the old vertices. The positions of old vertices in the refined mesh can be modified as well. To specify a subdivision scheme, two rules need to be established: a topological rule for obtaining the graph of the refined mesh from the graph of the initial mesh and a rule for computing the positions of the new vertices and modifying positions of the old vertices. As an example, the Catmull-Clark subdivision scheme [1] is shown below.



Figure 1.1: Basic Concept of the Catmull-Clark Scheme.

Given a control mesh, a Catmull-Clark subdivision surface (CCSS) is generated by iteratively refining the control mesh [1]. The limit surface is called a subdivision surface because the mesh refining process is a generalization of the uniform B-spline surface subdivision technique. Therefore, CCSSs include uniform B-spline surfaces and piecewise Bézier surfaces as special cases. It is now known that CCSSs include non-uniform Bspline surfaces and NURBS surfaces as special cases as well [15]. The valence of a mesh vertex is the number of mesh edges adjacent to the vertex. A mesh vertex is called an extraordinary vertex if its valence is different from four. Vertex \mathbf{V} in Figure 1.1 is an extraordinary vertex of valence five. A mesh face with an extraordinary vertex is called an extraordinary face. The valance of an extraordinary face is the valence of its extraordinary vertex. In the following, for the sake of simplicity, a mesh face and the corresponding surface patch will be denoted by the same notation.

In the Catmull-Clark subdivision scheme[1], each mesh refining step involves the construction of three new types of points: *face points*, *edge points* and *vertex points*, see Figure 1.1. New points are connected to form a new control mesh. These control meshes

converge to a limit surface. A face point is created for each old polygon, defined as the average of every point in the polygon, i.e., the centroid of each face:

$$\mathbf{F}_i^{k+1} = \frac{\mathbf{V}^k + E_i^k + \mathbf{F}_i^k + \mathbf{E}_{i+1}^k}{4}$$

where k is the subdivision level. An edge point is created for each old edge, defined as the average of the midpoint of the original edge and the midpoint of the two new face points for the polygons that adjoin the original edge:

$$\mathbf{E}_{i}^{k+1} = \frac{\mathbf{V}^{k} + \mathbf{E}_{i}^{k} + \mathbf{F}_{i-1}^{k+1} + \mathbf{F}_{i}^{k+1}}{4}$$

And finally, new vertex points are defined as follows.

$$\mathbf{V}^{k+1} = \frac{n-2}{n}\mathbf{V}^k + \frac{1}{n^2}\sum_{i=1}^{n}\mathbf{E}_i^k + \frac{1}{n^2}\sum_{i=1}^{n}\mathbf{F}_i^k$$

where n is the valance of vertex V and k is the subdivision depth. In this dissertation, we consider general CCSSs. That is, the new *vertex point* V^{k+1} is computed as follows:

$$\mathbf{V}^{k+1} = \alpha_n \mathbf{V}^k + \frac{\beta_n}{n} \sum_{i=1}^n \mathbf{E}_i^k + \frac{\gamma_n}{n} \sum_{i=1}^n \mathbf{F}_i^k$$
(1.1)

where α_n , β_n and γ_n are positive numbers whose sum equals one.

The new points are then connected, see Figure 1.1. Each face point connects to an edge point, which connects to a new vertex point, which connects to the edge point of the adjoining edge, which returns to the face point. This is done for each quadruple, forming new quadrilaterals around the faces. The scheme only produces quadrilaterals, although they are not necessarily planar.

CCSSs can model/represent complex shapes of arbitrary topology because there is no limit on the shape and topology of the control mesh of a CCSS [1]. See Figure 1.2(d) for the representation of a ventilation control component with a single CCSS. The initial control mesh of the surface, the control mesh after one refinement and two refinements are shown in Figure 1.2(a), Figure 1.2(b) and Figure 1.2(c), respectively. The ventilation control component is a solid with seventeen holes (handles). It can not be represented by a single trimmed B-spline or NURBS surface.



Figure 1.2: An example of Catmull-Clark Subdivision Surfaces.

1.3 Subdivision Surface based One Piece Representation

Subdivision surfaces have an important impact on several areas of geometric modeling:

- **Representation:** Subdivision surfaces provide a more general surface representation scheme to the design community because subdivision surfaces include traditional surface representation schemes as special cases. For instance, a NURBS surface can be generated as a subdivision surface through knot insertion. Subdivision surfaces also provide a different way to generate traditional surfaces.
- Modeling Capability: Subdivision surfaces provide more flexibility in shape modeling than traditional surface representation schemes. It is possible to represent any complex shape with only one subdivision surface. It is even possible to represent the result of a Boolean operation of two surfaces by a single subdivision surface. This is due to the fact that the control mesh of a subdivision surface can be of any shape and of any topology.

- Numerical Stability: The construction process of a subdivision surface is numerically stable no matter how complicated the shape of the surface. This is because the mesh refining process of a subdivision surface is a local process. It shares the same kind of numerical stability as the *deCasteljau algorithm* and the *De Boor algorithm*. Note that these algorithms represent some mesh refining processes as well.
- Smoothness and Discretization: Subdivision surfaces can be represented both in parametric form and discrete form. Therefore subdivision surfaces enjoy advantages of both representation schemes. The polygonal mesh form of a subdivision surface is extremely suitable for machining and tessellation (including FE mesh generation). On the other hand, it is possible to generate smooth parametric subdivision surfaces of any shape and any topology for any design purpose. These surfaces can be C^1 , G^1 , C^2 or G^2 continuous everywhere except at a few extraordinary points where smoothness of the surface is only one order lower than that at other points. Hence we have a representation scheme that is good for all CAD/CAM applications.

However, geometric algorithms and modeling technologies required in subdivision surface based modeling operations have not yet been studied thoroughly [118]. For instance, even though it is known that one can use a subdivision surface to model/represent complex shape of arbitrary topology, a methodology on how to build the control mesh of such a surface has never been presented. The construction is basically a trial-and-error process.

Hence we need approaches to construct one-piece represented control meshes for complex shape with arbitrary topology. Subdivision surface based one-piece representation means to represent the final object in a design process with only one subdivision surface (i.e. a sparse control mesh), no matter how complicated the object's topology or shape. No decomposition of the object into simpler components is necessary. Hence the number of parts in the final representation is always the minimum: one.



Figure 1.3: Subdivision surface based one-piece and multi-piece Representations.

In this dissertation research, we study two possible approaches to build a one-piece represented control mesh for a given model. One possibility is to use the subdivision surface interpolation technique to approximate the surface of the given model. But this approach would be difficult to build features such as cusps, creases and darts into the resultant surface in such a process. Another approach is to construct a mesh through Boolean operations and multiresolution analysis. Both approaches can achieve good results and can have explicit error control.

A comparison of subdivision surface based one-piece and multi-piece representation is given in Figure 1.3. Figure 1.3(a) is the control mesh of the one-piece representation surface shown in Figure 1.3(b) and Figure 1.3(c) is the mesh of the multi-piece representation surface shown in Figure 1.3(d), where different colors denote different parts. We can see from Figure 1.3 that with subdivision surface based one-piece representation, the number of components in the representation is only one.

1.4 Structure of the Subdivision based One Piece Representation System

The objective of the subdivision based one piece representation system is to represent the final object in a design process with only one subdivision surface (i.e. a sparse control mesh), no matter how complicated the object's topology or shape. No decomposition of the object into simpler components is necessary. Hence the output of our system is always a sparse control mesh whose Catmull-Clark subdivision surface approximates the target model. The system provides two possible ways to construct a sparse control mesh: interpolation and Boolean operations. Our system supports CSG (Constructive Solid Geometry) operations as well as long as the CSG primitives are represented in subdivision surfaces. The overall framework of the system is shown in Figure 1.4. The main steps of the framework will be discussed in the later chapters.

1.5 Contributions

Our research work for the dissertation is focused on developing necessary mathematical theories and geometric algorithms to support subdivision surface based one-piece representation. In this section we summarize the contributions of the dissertation research work as follows.

• **Parametrization** of CCSSs [58]:

Subdivision methods for evaluating surfaces rely on performing repeated subdivisions until the control structure approximates the limit surface within some tolerance. It is then possible to push the control points to their limit positions and bilinearly interpolate values across an inexact surface patch. One of the main problems that may hinder the usage of subdivision surfaces in shape design is the exponential growth rate of the number of vertices in the refined mesh with respect to the subdivision depth. This would make both the real time rendering and the necessary accuracy difficult to achieve for complicated objects. Hence an explicit



Figure 1.4: Structure of the Subdivision surface based One Piece Representation System

and exact evaluation and parametrization method for subdivision surfaces is indispensable.

However, powerful evaluation and analysis techniques for subdivision surfaces have not been fully developed yet. Parametrization methods that have been developed so far are suitable for evaluation purposes only, not for analysis purposes, because these methods either do not have an explicit expression, or are too complicated for each part to be explicit. For instance, in [17], eigen functions are pre-computed numerically and stored in a file. Therefore, they can be used for evaluation purpose only.



Figure 1.5: Direct and exact evaluation of Catmull-Clark subdivision surfaces.

We proposed a new parametrization technique for general Catmull-Clark subdivision surfaces. The new technique extends J. Stam's work [17] by redefining all the eigen basis functions in the parametric representation for general Catmull-Clark subdivision surfaces and giving each of them an explicit form. The entire eigen structure of the subdivision matrix and its inverse are computed exactly and explicitly with no need to precompute anything. Therefore, the new representation can be used not only for evaluation purposes, but for analysis purposes as well [58]. Our new approach is based on an Ω -partition of the parameter space and a detoured subdivision path. This results in a block diagonal matrix with constant size diagonal blocks (7 × 7) for the corresponding subdivision process. Consequently, eigen decomposition of the matrix is always possible and is simpler and more efficient. Furthermore, since the number of eigen basis functions required in the new approach is only one half of the previous approach [17], our new parametrization is also more efficient for evaluation purposes. The camel model shown in Figure 1.5 is rendered using our parametrization techniques with all the positions and normals computed exactly, not approximated. Hence, the quality of the image is better than those generated through the subdivision process.

• Interpolation of meshes of arbitrary topology [55]:

Interpolation is a direct approach for building a sparse mesh structure of a given model in our subdivision based one-piece representation system. Although there are some interpolation methods using subdivision surfaces [11, 25, 26, 52], most of them cannot handle open meshes and the resultant surface exhibits undesired artifacts and undulations. We proposed a new method for constructing a sparse mesh whose smooth Catmull-Clark subdivision surface (CCSS) interpolates the vertices of a mesh with arbitrary topology. The new method handles both open and closed meshes. Normals or derivatives specified at any vertices of the mesh (which can actually be anywhere) can also be interpolated. The construction process is based on the assumption that, in addition to interpolating the vertices of the given mesh, the interpolating surface is also similar to the limit surface of the given mesh. Therefore, construction of the interpolating surface can use information from the given mesh as well as its limit surface. This approach, called *similarity based interpolation* [55], gives us more control of the smoothness of the interpolating



Figure 1.6: Subdivision surface based one-piece representation using interpolation techniques.

surface. Consequently, it avoids the need of shape fairing in the construction of the interpolating surface. The computation of the interpolating surface's control mesh follows a new approach, which does not require the resulting global linear system to be solvable. An approximate solution provided by any fast iterative linear system solver is sufficient. Nevertheless, interpolation of the given mesh is guaranteed. This is an important improvement over previous methods because with these features, the new method can handle meshes with a large number of vertices efficiently. Although the new method is presented for CCSSs, the concept of similarity based interpolation can be used for other subdivision surfaces as well.

Figure 1.6 gives us an example of subdivision surface based one-piece representation using interpolation techniques. Figure 1.6(a) is a given model for which we need to construct a spars control mesh. Figure 1.6(b) is a mesh whose control points are directly sampled from the model shown in Figure 1.6(a). By using our interpolation technique, we can construct a sparse control mesh shown in Figure 1.6(d), whose limit surface interpolates the sampled mesh shown in Figure 1.6(b) and would also be almost the same as the given model shown in Figure 1.6(c). Figure 1.6(c) is the limit surface of the mesh shown in Figure 1.6(b), which can be used as a reference surface in the process of constructing the mesh shown in Figure 1.6(d).



Figure 1.7: Voxelization of free-form solids.

 128×128

 512×512

The mesh shown in Figure 1.6(d) is called the subdivision surface based one-piece representation of the given model shown in Figure 1.6(a).

• Voxelization of free-form solids [57]:

A voxelization technique and its applications for objects with arbitrary topology are presented. With parametrization techniques for subdivision surfaces becoming available, it is now possible to model and represent any continuous but topologically complex object with an analytical representation. We proposed a method to convert a free-form object from its continuous geometric representation into a set of voxels that best approximates the geometry of the object. Unlike traditional 3D scan-conversion based methods, our voxelization method is performed by recursively subdividing the 2D parameter space and sampling 3D points from selected 2D parameter space points. Because we can calculate every 3D point position explicitly and accurately, uniform sampling on surfaces with arbitrary topology is not a problem any more.

Moreover, our discretization of 3D closed objects is guaranteed to be leak-free when a 3D flooding operation is performed. This is ensured by proving that our voxelization results satisfy the properties of separability, accuracy and minimality. In addition, a 3D volume flooding algorithm using dynamic programming techniques is proposed which significantly speeds up the volume flooding process. Hence our method is suitable for visualization of complex scenes, measuring object volume, mass, surface area, determining intersection curves of multiple surfaces and performing accurate Boolean/CSG operations. These capabilities are demonstrated by test examples shown in the dissertation. For example, Figure 1.7 gives two results of voxelization of the rocker arm model shown in Figure 1.7(b), whose mesh is shown in Figure 1.7(a). The result shown in Figure 1.7(c) is obtained by voxelizing the rocker arm model using resolution $128 \times 128 \times 128$, while Figure 1.7(d) is obtained using resolution $512 \times 512 \times 512$. It is easy to see, when resolution is high enough, the voxelization result would be close enough to the original model.

• Boolean operations on free-form solids [64]:

A method for performing robust and error controllable Boolean operations on freeform solids represented by Catmull-Clark subdivision surfaces (CCSSs) was developed. The given objects are voxelized to make Boolean operations more efficient. However, different from previous voxelization based approaches, the final result of the Boolean operations in our method is represented with a continuous geometric representation (i.e. a polygonal mesh). This is achieved by doing the Boolean operations in the parameter spaces of the solids, instead of the object space. The 2D parameter space is recursively subdivided until a keep-or-discard decision can be made for each resulting subpatch using results of the voxelization process. This approach allows us to easily compute a continuous approximation of the intersection curve and, consequently, build a continuous geometric representation for the Boolean operation result. To make the Boolean operation result more accurate, a secondary local voxelization can be performed for intersecting subpatches. Because the voxelization process itself is very fast and robust, the overall process is fast and robust too. Most importantly, error of Boolean operation results can be estimated, so error control is possible. In addition, our method can handle any cases of Boolean operations as long as the given solids are represented by CCSSs. Therefore there are no special or degenerated cases to take care of. Although the new method is presented for CCSSs, the concept actually works for any subdivision scheme whose limit surfaces can be parameterized. See Figure 1.8 for an example of performing Boolean operations between two solids represented by Catmull-Clark subdivision surfaces. Figure 1.8(a) is the union of the cylinder and the bunny model, while Figure 1.8(b) is the subtraction of one from the other.



(a) Union

(b) Difference

Figure 1.8: Boolean operations on free-form solids.

• Adaptive tessellation of CCSSs [56, 59, 61]:

The Catmull-Clark subdivision scheme provides a powerful method for building smooth and complex surfaces. But the number of faces in the uniformly refined meshes increases exponentially with respect to subdivision depth. Adaptive tessellation reduces the number of faces needed to yield a smooth approximation to the limit surface and, consequently, makes the rendering process more efficient. We have developed a new adaptive tessellation method for general Catmull-Clark subdivision surfaces. Different from previous control mesh refinement based approaches, which generate approximate meshes that usually do not interpolate the limit surfaces, the new method is based on direct evaluation of the limit surface to generate an inscribed polyhedron of the limit surface. With explicit evaluation of general Catmull-Clark subdivision surfaces becoming available, the new adaptive tessellation method can precisely measure error for every point of the limit surface. Therefore, it has complete control of the accuracy of the tessellation result. Cracks



(a) Given model



(d) Adaptive tessellation



(b) Uniform tessellation



(e) Adaptive tessellation



(c) Adaptive tessellation





Figure 1.9: Adaptive tessellation on surfaces with arbitrary topology.

are avoided by using a recursive color marking process to ensure that adjacent patches or subpatches use the same limit surface points in the construction of the shared boundary. The new method performs limit surface evaluation only at points that are needed for the final rendering process. Therefore it is very fast and memory efficient. The new method is presented for the general Catmull-Clark subdivision scheme, but it can be used for any subdivision scheme that has an explicit evaluation method for its limit surface. An example of adaptive tessellation is shown in
Figure 1.9. Figure 1.9(a) is the given gargoyle model, which is a subdivision surface. Figure 1.9(b) is a uniform tessellation of Figure 1.9(a), which is very dense. Figures 1.9(c), 1.9(d) and 1.9(e) are three adaptive tessellations of the given model with different error tolerances. From these figures we can see that sparse polygonal approximation can be achieved through adaptive tessellation. Figure 1.9(f) is the triangulated tessellation of Figure 1.9(e), which consists of only triangles in the polygonal approximation.



Figure 1.10: A snapshot of the subdivision surface based one-piece representation system.

• A system that supports subdivision surface based one-piece representation has been implemented and a lot of examples have been tested. All the examples show that our approaches can obtain very good subdivision based one-piece representation results. Figure 1.10 is a snapshot of our subdivision surface based one-piece representation system.

• Other results:

We also achieved some other good results during the dissertation research. Although they are not directly related to our subdivision surface based one-piece representation system, they all are good applications of subdivision surfaces and may have some impacts in future research work.

- Constrained Scaling of CCSSs [62]:

A method to scale a Catmull-Clark subdivision surface while holding the shape and size of specific features (sub-structures) unchanged is presented. The basic idea of the method, *fix-and-stretch*, is similar to a previous approach for trimmed NURBS surfaces, i.e., the new surface is formed by fixing selected regions of the given subdivision surface that contain the features, scaling and stretching the remaining part; the goal is to ensure that the resulting surface reflects the shape and curvature distribution of the unconstrained scaled version of the given surface. However, the stretching process, which is the core of the entire process, is more complicated because of the complexity of a subdivision surface's topology. The major contributions of the new constrained scaling technique include new strain energy computation techniques and energy optimization techniques for regions around extraordinary points. The new method is more powerful than the previous method in that it can handle more complicated shapes and, consequently, can be used for more challenging applications. Test results on the rocker arm model that can not be represented by trimmed NURBS surfaces are shown in Figure 1.11. The left-hand model in Figures 1.11(a) and 1.11(b) is the given model. The blue parts in these figures are the specified feature which will not be changed in the constrained scaling process. The right-hand model of Figure 1.11(a) is constrained scaling with scale factors $S_x = 1.3$, $S_y = 1.2$ and $S_z = 1.1$. The right-hand model of Figure 1.11(b) is obtained in the constrained scaling process with scale factor $S_x = 0.8, S_y = 0.95$ and $S_z = 0.9$.



Figure 1.11: Constrained scaling of Catmull-Clark subdivision surfaces.

- Texture mapping on surfaces of arbitrary topology [60]:

A very simple and yet highly efficient, high quality texture mapping method for surfaces of arbitrary topology has been presented. The new method projects the given surface from the 3D object space into the 2D texture space to identify the 2D texture structure that will be used to texture the surface. The object space to texture space projection is optimized to ensure minimum distortion of the texture mapping process. The optimization is achieved through a commonly used norm preserving minimization process on edges of the surface. The main difference here is, that by using an initial value approach, the optimization problem can be set up as a quadratic programming problem and, consequently, solved by a linear least squares method. Three methods to choose a good initial value are presented. Test cases show that the new method works well on surfaces of arbitrary topology, with the exception of surfaces with exceptionally abnormal curvature distribution. Other advantages of the new method include uniformity and seamlessness of the texture mapping process. The new method is suitable for applications that do not require precise texture mapping results but demand highly efficient mapping



(a) Uniform texture Mapping



(b) Uniform texture Mapping



(c) Patch based texture Mapping



(d) Patch based texture Mapping

Figure 1.12: Texture Mapping on surfaces of arbitrary topology.

process such as computer animation or video games. Examples of texture mapping on surfaces of arbitrary topology are shown in Figure 1.12. Figures 1.12(a) and 1.12(b) are obtained with a global norm preserving based optimization, while Figures 1.12(c) and 1.12(d) are obtained through patch based parametrization. It is easy to see that the global method can obtain uniform texture mapping results, which are more realistic than non-uniform ones.

Rendering live scenes by using view dependent textured splatting techniques [63]:

We presented a novel approach for rendering low resolution point clouds with

multiple high resolution textures, the type of data commonly generated by real-time vision systems. The low precision, noisy, and sometimes incomplete nature of such data sets is not suitable for existing point-based rendering techniques that are designed to work with high precision and high density point clouds. Our new algorithm, View-dependent Textured Splatting (VDTS), combines traditional splatting with a view-dependent texturing strategy to increase rendering quality of low resolution data sets with high resolution images. VDTS requires no pre-processing, addresses texture visibility and antialiasing on the fly, and can be efficiently accelerated by commodity graphics hardware. Therefore it is particularly well-suited for rendering dynamic scenes in real time and online.

1.6 Notions

The following notational conventions are adopted in this dissertation. Space objects such as points, lines and parametric functions are denoted by bold upper case Roman characters, e.g., **V**. Linearly transformed items or Fourier points are denoted by bold lower case Roman characters, e.g., **v**. All vectors are assumed to be columns. Vectors of ordinary items (respectively linearly transformed items or Fourier points) are denoted by upper (respectively lower) case italicized characters, e.g., V (respectively g). Matrices are denoted by uppercase Roman characters, e.g., M. The transpose of a vector V (respectively matrix M) is denoted by V^T (respectively M^T).

1.7 Overview

The organization of this dissertation is as follows:

• First, an explicit parametrization method is presented for exact evaluation of Catmull-Clark subdivision surfaces in **Chapter 2**. With an explicit parametrization, subdivision is no longer a must in order to obtain the limit surface of a given mesh, because direct and exact evaluation can now be directly applied.

- In Chapter 3, an interpolation method for meshes of arbitrary topology is presented. Using interpolation, a subdivision surface based one-piece representation for any model with arbitrary topology can be achieved.
- A voxelization technique and its applications for objects with arbitrary topology are presented in **Chapter 4**. The new technique converts a free-form object from its continuous geometric representation into a set of voxels that best approximates the geometry of the object. The voxelization results can further be used for performing robust and error controllable Boolean operations in our subdivision surface based one-piece representation system.
- A technique for performing robust and error controllable Boolean operations on freeform solids represented by Catmull-Clark subdivision surfaces (CCSSs) is presented in **Chapter 5**. After the Boolean operations, a one-piece representation can be achieved in this stage, although the resulting meshes could be dense.
- Because subdivision surface based one-piece representations obtained from interpolation or Boolean operations are usually dense meshes, good mesh simplification or reduction methods are needed. One method is adaptive tessellation, which reduces the number of faces needed to yield a smooth approximation to the limit surface and, therefore, makes the rendering process more efficient. In **Chapter 6**, we present a new adaptive tessellation method for general Catmull-Clark subdivision surfaces which can significantly reduce the number of polygons for representing a CCSS with accurate error control.
- Multiresolution analysis is another good method for simplifying dense meshes with arbitrary topology. In **Chapter 7** multiresolution representation for subdivision surface based one-piece representations is presented, which results in much sparser control meshes and has explicit error estimation.

• We conclude the dissertation in **Chapter 8** and point out some directions for future work.

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Chapter 2

Parametrization and Evaluation of General Catmull-Clark Subdivision Surfaces

In this chapter, a new parametrization technique and its applications for general Catmull-Clark subdivision surfaces are presented. Our new technique [58] extends J. Stam's work [17] by redefining all the eigen basis functions in the parametric representation for general Catmull-Clark subdivision surfaces and giving each of them an explicit form. The entire eigen structure of the subdivision matrix and its inverse are computed exactly and explicitly with no need to precompute anything. Therefore, the new representation can be used not only for evaluation purpose, but for analysis purpose as well. The new approach is based on an Ω -partition [17] of the parameter space and a detoured subdivision path. This results in a block diagonal matrix with constant size diagonal blocks (7 × 7) for the corresponding subdivision process. Consequently, eigen decomposition of the matrix is always possible and is simpler and more efficient. Furthermore, since the number of eigen basis functions required in the new approach is only one half of the previous approach [17], the new parametrization is also more efficient for evaluation purpose. This is demonstrated by several applications of the new techniques in texture mapping, special feature generation, surface trimming, boolean operations and adaptive rendering.

The organization of this chapter is: A brief introduction is given in Section 1. Section 2 gives a brief review of the Catmull-Clark subdivision scheme and previous evaluation techniques. Section 3 shows an intuitive but expensive approach in parameterizing an extraordinary Catmull-Clark patch. Section 4 shows a more efficient approach in parameterizing a Catmull-Clark patch using an extended subdivision path. Section 5 shows how to compute the eigen structure of the subdivision matrix of the extended subdivision path. Section 6 shows the evaluation process of the new parametric representation at an arbitrary point of a Catmull-Clark patch. Section 7 gives some examples of analysis with our explicit representation around an extraordinary vertex. Section 8 shows application examples of the new scheme in texture mapping, special feature generation, surface trimming, adaptive rendering, mesh interpolation and boolean operations. The concluding remarks are given in Section 9.

2.1 Introduction

Subdivision surfaces have become popular recently in graphical modelling and animation because of their capability in modeling/representing complex shape of arbitrary topology [9], their relatively high visual quality, and their stability and efficiency in numerical computation. Subdivision surfaces can model/represent complex shape of arbitrary topology because there is no limit on the shape and topology of the control mesh of a subdivision surface.

Subdivision methods for evaluating surfaces rely on performing repeated subdivisions until the control structure approximates the limit surface within some tolerance. It is then possible to push the control points to their limit positions and bilinearly interpolate values across an inexact surface patch. But in some applications, the exact evaluation is critical. Hence a good parametrization for subdivision surface is indispensable.

However, powerful evaluation and analysis techniques for subdivision surfaces have not been fully developed yet. Parametrization methods that have been developed so far are suitable for evaluation purpose only, not for analysis purpose, because these methods either do not have an explicit expression, or are too complicated for each part to be explicit. For instance, in [17], eigen functions are pre-computed numerically and stored in a file. So they can be used for evaluation purpose only. Note that exact evaluation at a point of a subdivision surface is possible only if there is an explicit parametrization of the surface. Hence, an explicit parametrization is not only critical for analysis purpose, but for evaluation and rendering purpose as well.

In this chapter we will present an Ω -partition based approach to solve several important problems of subdivision surfaces: (1) computation of new control vertices at a specified subdivision level, (2) explicit parametrization of an extraordinary patch, and (3) surface evaluation at arbitrary parameter space point with eigen functions computed on the fly.

The new approach is based on the observation that the subdivision process on the control vertices can be broken into subdivision processes on smaller, same frequency groups after a few linear transformations. Using a different ordering of the vertices and the idea of enlarging the subdivision matrix, the subdivision matrix can be transformed into a block matrix with each block being circulant [5, 22]. Hence it is natural to use the Fourier matrices to transform them into diagonal matrix. Each such subdivision process on points of the same frequency is independent of the *valence* of the extraordinary vertex. The dimension of the corresponding subdivision matrix for each frequency group is 7×7 . Therefore, the process of using a large subdivision matrix to perform the subdivision process on the control vertices can be replaced with a set of 7×7 matrices on the subdivision matrices always possible, but also simpler and more efficient. Inverses of the eigenvector matrices can also be explicitly computed.

2.2 Previous Work

2.2.1 Catmull-Clark Subdivision Surfaces

Given a control mesh, a *Catmull-Clark subdivision surface* (CCSS) is generated by iteratively refining the control mesh [1]. The limit surface is called a *subdivision surface* because the mesh refining process is a generalization of the uniform B-spline surface *subdivision technique*. The *valence* of a mesh vertex is the number of mesh edges adjacent to the vertex. A mesh vertex is called an *extraordinary vertex* if its valence is different



Figure 2.1: (a) Control vertices that influence an extraordinary patch. (b) New control vertices (solid dots) generated after a Catmull-Clark subdivision.

from four. Vertex \mathbf{V} in Figure 2.1(a) is an extraordinary vertex of valence five. A mesh face with an extraordinary vertex is called an *extraordinary face*. The *valance* of an extraordinary face is the valence of its extraordinary vertex. In the following, for the sake of simplicity, a mesh face and the corresponding surface patch will be treated the same and denoted by the same notation.

Given an extraordinary face $\mathbf{S} = \mathbf{S}_{0,0}$. If the valence of its extraordinary vertex is n, then the surface patch corresponding to this extraordinary face is influenced by 2n + 8 control vertices [1, 17]. The control vertices shown in Figure 2.1(a) are the ones that influence the patch marked with an " $\mathbf{S} = \mathbf{S}_{m-1,0}$ ". In general, if $\mathbf{S}_{m-1,0}$ is the extraordinary subpatch generated after m - 1 subdivision steps, then by performing a Catmull-Clark subdivision step on the control vertices of $\mathbf{S}_{m-1,0}$, one gets 2n + 17 new control vertices. See Figure 2.1(b) for the new control vertices generated for the patch $\mathbf{S}_{m-1,0}$ shown in (a). These 2n + 17 new control vertices define four subpatches: $\mathbf{S}_{m,b}$, b = 0, 1, 2, 3 (Figure 2.1(b)). $\mathbf{S}_{m,0}$ is again an extraordinary patch but $\mathbf{S}_{m,1}$, $\mathbf{S}_{m,2}$, and $\mathbf{S}_{m,3}$ are regular uniform bicubic B-spline patches. Iteratively repeat this process, one gets a sequence of regular bicubic B-spline patches ($\mathbf{S}_{m,b}$), $m \ge 1$, b = 1, 2, 3, a sequence of extraordinary patches ($\mathbf{S}_{m,0}$), $m \ge 0$, and a sequence of extraordinary vertices. The extraordinary patches converge to the limit point of the extraordinary vertices [11]. The regular bicubic B-spline patches ($\mathbf{S}_{m,b}$), $m \ge 1$, b = 1, 2, 3, and the limit point of the

2.2.2 Previous Parametrization and Evaluation Methods

An algorithm for the evaluation of a subdivision surface at an arbitrary point was first proposed by J. Stam in 1998 for Catmull-Clark subdivision surfaces [17] and then in 1999 for Loop subdivision surfaces [18]. Stam's approach shows that an extraordinary surface patch and its derivatives can be represented as a linear combination of the control points with weights defined by a set of 2n + 8 eigenbasis functions where n is the valence of the extraordinary patch. The representation satisfies simple scaling relations and can be easily evaluated in constant time. However, even though analytical expressions for the eigenbasis functions have been derived, some of them are too complicated to be reported in the paper [17]. Besides, some of the eigenbasis functions are redundant. We will show in this chapter that only n + 6 eigenbasis functions are actually needed and, consequently, the evaluation process can be made more efficient. J. Stam's approach [17] is mainly developed for evaluation purpose. As we shall present, our parametrization results [58] can be used not only for evaluation, but for analysis purpose as well.

Warrent and Weimer presented a method in [22] for computing all eigenvalues and eigenvectors of the subdivision matrix by writing the subdivision matrix for the 2-ring in block circulant form. Ball and Storry [5] also used the similar approach to compute the eigen structure of the subdivision matrix. However, as far as we know, the inverse of the matrix of the eigenvectors has never been computed explicitly, and the overall explicit eigen structure has never been integrated into the parametrization formula. In this paper, based on the eigen analysis results of [5], an explicit and exact evaluation formula is derived.

Zorin extended the work of J. Stam by considering subdivision rules for piecewise smooth surfaces with parameter-controlled boundaries [20]. The main contribution of their work is the usage of a different set of basis vectors for the evaluation process which, unlike eigenvectors, depend continuously on the coefficients of the subdivision rules. The advantage of this algorithm is that it is possible to define evaluation for parametric families of rules without considering excessive number of special cases, while improving numerical stability of calculation.

In addition to Stam's approach, two different parameterizations of Catmull-Clark subdivision surfaces have been proposed by Boier-Martin and Zorin [8]. The motivation of their work is to provide parametrization techniques that are differentiable everywhere. Although all the natural parameterizations of subdivision surfaces are not C^1 around extraordinary vertices of valence higher than four[8], the resulting surfaces are still C^2 almost everywhere. Moreover, despite of the fact that the partial derivatives diverge around an extraordinary vertex, in this paper, we will show that an standardized normal vector can be calculated explicitly everywhere. As we know, precisely calculated normal vector is indispensable for surface shading purposes.

Exact evaluation of piecewise smooth Catmull-Clark surfaces near sharp and semisharp features is considered in [16]. Constant-time performance is achieved by employing Jordan decomposition of the subdivision matrix. In this paper we will show that special features can be generated using ordinary Catmull-Clark rules with constant-time evaluation performance as well.

2.3 Parametrization of a Patch

The regular bicubic B-spline patches $\{\mathbf{S}_{m,b}\}, m \ge 1, b = 1, 2, 3$, induce a partition on the unit square $[0, 1] \times [0, 1]$. The partition is defined by : $\{\mathbf{\Omega}_{m,b}\}, m \ge 1, b = 1, 2, 3$, with

$$\begin{split} \mathbf{\Omega}_{m,1} &= \left[\frac{1}{2^m}, \frac{1}{2^{m-1}}\right] \times \left[0, \frac{1}{2^m}\right], \\ \mathbf{\Omega}_{m,2} &= \left[\frac{1}{2^m}, \frac{1}{2^{m-1}}\right] \times \left[\frac{1}{2^m}, \frac{1}{2^{m-1}}\right], \\ \mathbf{\Omega}_{m,3} &= \left[0, \frac{1}{2^m}\right] \times \left[\frac{1}{2^m}, \frac{1}{2^{m-1}}\right] \end{split}$$

(see Figure 2.2 for an illustration of the partition [17]). For any $(u, v) \in [0, 1] \times [0, 1]$ but $(u, v) \neq (0, 0)$, there is an $\Omega_{m,b}$ that contains (u, v). To find the value of **S** at (u, v), first map $\Omega_{m,b}$ to the unit square. If (u, v) is mapped to (\bar{u}, \bar{v}) by this mapping, then compute the value of $\mathbf{S}_{m,b}$ at (\bar{u}, \bar{v}) . The value of **S** at (0, 0) is the limit of the



Figure 2.2: Ω -partition of the unit square [17].

extraordinary vertices. For convenience of subsequent reference, the above partition will be called an Ω -partition of the unit square.

In the above process, m and b can be computed as follows:

$$\begin{split} m(u,v) &= \min\{\lceil \log_{\frac{1}{2}}u\rceil, \lceil \log_{\frac{1}{2}}v\rceil\} \ ,\\ b(u,v) &= \begin{cases} 1, & if \ 2^{m}u \geq 1 \ and \ 2^{m}v < 1 \\ 2, & if \ 2^{m}u \geq 1 \ and \ 2^{m}v \geq 1 \\ 3, & if \ 2^{m}u < 1 \ and \ 2^{m}v \geq 1 \ . \end{cases} \end{split}$$

The mapping from $\Omega_{m,b}$ to the unit square is defined as:

$$(u,v) \rightarrow (\bar{u},\bar{v}) = (\phi(u),\phi(v)),$$

where

$$\phi(t) = \begin{cases} 2^m t, & \text{if } 2^m t \le 1\\ 2^m t - 1, & \text{if } 2^m t > 1 \end{cases}.$$
(2.1)

Since each $\mathbf{S}_{m,b}$ is a standard B-spline surface, it can be expressed as

$$\mathbf{S}(u,v) = W^T(\bar{u},\bar{v})\mathbf{M}G_{m,b}$$

where $G_{m,b}$ is the control point vector of $\mathbf{S}_{m,b}$, W(u, v) is a vector containing the 16 power basis functions:

$$W^{T}(u,v) = [1, u, v, u^{2}, uv, v^{2}, u^{3}, u^{2}v, uv^{2}, v^{3}, u^{3}v, u^{2}v^{2}, uv^{3}, u^{3}v^{2}, u^{2}v^{3}, u^{3}v^{3}],$$

and M is the B-spline coefficient matrix. An important observation is, $W^T(\bar{u}, \bar{v})$ can be expressed as the product of $W^T(u, v)$ and two matrices:

$$W^T(\bar{u}, \bar{v}) = W^T(u, v) \mathbf{K}^m \mathbf{D}_{\mathbf{b}},$$

where K is a diagonal matrix

$$K = Diag(1, 2, 2, 4, 4, 4, 8, 8, 8, 8, 16, 16, 16, 32, 32, 64)$$

and D_b is an upper triangular matrix depending on b only. D_b can be obtained by replacing \bar{u} , \bar{v} in $W(\bar{u}, \bar{v})$ with $\phi(u)$, $\phi(v)$ defined in Eq. (2.1). Therefore, we have

$$\mathbf{S}(u, v) = W^T(u, v) \mathbf{K}^{\mathrm{m}} \mathbf{D}_{\mathrm{b}} \mathbf{M} G_{m, b}$$

The computation of the control vertices of $\mathbf{S}_{m,b}$ involves two matrices, A and \overline{A} [17]. \overline{A} is a $(2n + 17) \times (2n + 8)$ matrix, representing the subdivision process shown in Figure 2.1(b). A is a $(2n + 8) \times (2n + 8)$ submatrix of \overline{A} , representing the process of mapping the 2n + 8 control vertices of the given extraordinary patch to the 2n + 8 control vertices of its extraordinary subpatch. Let

$$G = [\mathbf{V}, \mathbf{E}_1, \cdots, \mathbf{E}_n, \mathbf{F}_1, \cdots, \mathbf{F}_n, \mathbf{I}_1, \cdots, \mathbf{I}_7]$$

then G (See Figure 2.1(a) for its labelling) is the column vector representing the control vertices of **S**. By applying A to G (m-1) times we get the 2n + 8 control vertices of the extraordinary subpatch $\mathbf{S}_{m-1,0}$. Now by applying \bar{A} to the control vertices of $\mathbf{S}_{m-1,0}$ (represented as G_{m-1}), we get 2n+17 new control points which include the 2n+8 control vertices of $\mathbf{S}_{m,0}$. Let \bar{G}_m be the column vector representation of these 2n + 17 vertices, we have

$$\bar{G}_m = \bar{A}G_{m-1} = \bar{A}A^{m-1}G$$

Then by multiplying \bar{G}_m with an appropriate "picking" matrix P_b , we get the control vertices of the subpatch $\mathbf{S}_{m,b}$:

$$G_{m,b} = P_b \bar{G}_m = P_b \bar{A} A^{m-1} G$$
.

Hence we have

$$\mathbf{S}(u,v) = W^T(u,v) \mathbf{K}^{\mathbf{m}} \mathbf{D}_{\mathbf{b}} \mathbf{M} \mathbf{P}_b \bar{\mathbf{A}} \mathbf{A}^{m-1} G.$$
(2.2)

This is a parametrization of an extraordinary patch. However, this is a costly process to use because it involves m - 1 multiplications of the $(2n + 8) \times (2n + 8)$ matrix A. In the next section, we will present an efficient approach to calculate $G_{m,b}$ for any b and m.



Figure 2.3: The extended subdivision diagram.

2.4 Calculate Control Vertices after *m* Subdivisions

The goal here is to show that instead of using the direct path from G to G_{m-1} to compute $G_{m-1} = A^{m-1}G$ in the above equation, one should use the indirect, longer path $(G \to g \to g_{m-1} \to G_{m-1})$ in Figure 2.3 to do the job. The reason for doing so is: the corresponding matrix T is a block diagonal matrix with each diagonal block of dimension 7×7 only. Therefore, the process of computing their eigen decompositions is not only always possible, but also much simpler and more efficient.

Details of this new approach and definitions of related mappings are given below. We consider a general CCSS here. That is, the new *vertex point* \mathbf{V}' after one subdivision is computed as defined in Eq. 1.1. New *face points* and *edge points* are computed the same way as in [1].

First, to prepare G for the major transformation, we extend G into a vector of seven equal-length components, called \hat{G} :

$$\hat{G} = (V^T, E^T, F^T, I_1^T, I_2^T, I_3^T, I_4^T)^T,$$

where

$$V = (\mathbf{V}, \mathbf{V}, \dots, \mathbf{V})^{T}, E = (\mathbf{E}_{1}, \mathbf{E}_{2}, \dots, \mathbf{E}_{n})^{T}, F = (\mathbf{F}_{1}, \mathbf{F}_{2}, \dots, \mathbf{F}_{n})^{T}, I_{k} = (\mathbf{I}_{k}, \mathbf{I}_{k+4}, \mathbf{0}, \dots, \mathbf{0})^{T}, \qquad k = 1, 2, 3 I_{4} = (\mathbf{I}_{4}, \mathbf{0}, \mathbf{0}, \dots, \mathbf{0})^{T}$$

with all of them having the same length of n. We can get \hat{G} from G by a simple extension matrix H_1 , i.e., $\hat{G} = H_1 G$. Note that the matrix inducing \hat{G}_i to \hat{G}_{i+1} , i.e., $H_1 A H_1^{-1}$, is a $7n \times 7n$ block matrix with each block $(n \times n)$ being circulant [5, 22]. Therefore, each of these blocks can be diagonalized exactly using the discrete Fourier transform. Let \hat{g} be the result of applying the discrete Fourier transform L to the components of \hat{G} :

Each component of \hat{g} has the same length n, but is indexed from 0 to n-1. We can get \hat{g} from \hat{G} by combining all L's into a single matrix H_2 , i.e., $\hat{g} = H_2\hat{G}$. It is easy to see that H_2 is a block diagonal matrix. If we re-arrange the elements of \hat{g} into a set of same frequency groups:

$$g = (h_0^T, h_1^T, \cdots, h_{n-1}^T)^T,$$

where $h_{\omega} = (v_{\omega}, e_{\omega}, f_{\omega}, i_{1\omega}, i_{2\omega}, i_{3\omega}, i_{4\omega})^T$, with $0 \leq \omega \leq n - 1$. We can get g from \hat{g} through a $7n \times 7n$ permutation matrix H_3 , i.e., $g = H_3\hat{g}$. The above relationships hold for g_j , G_j , \hat{g}_j and \hat{G}_j , $j \geq 1$, as well (See Figure 2.3). Since H_1 , H_2 and H_3 are invertible, we can easily calculate g_j and G_j from each other.

For each $j \ge 1$, the subdivision process performed on G_{j-1} to get G_j can be reflected on g_{j-1} and g_j through H_1 , H_2 and H_3 . The induced subdivision process [5] on g_{j-1} can be represented by a $7n \times 7n$ matrix T as:

$$g_j = \mathrm{T}g_{j-1} = \mathrm{T}^j g.$$

T is a block diagonal matrix with each diagonal block T_{ω} ($\omega = 0, 1, 2, \dots, n-1$), being a 7×7 matrix. The expression of each T_{ω} can be found in [5]. Therefore, for each $m \ge 1$, we have (See Figure 2.3):

$$\mathbf{A}^{m-1} = \mathbf{H}_1^{-1} \mathbf{H}_2^{-1} \mathbf{H}_3^{-1} \mathbf{T}^{m-1} \mathbf{H}_3 \mathbf{H}_2 \mathbf{H}_1$$

By combining the above expression with (2.2), we have

$$\mathbf{S}(u,v) = W^T \mathbf{K}^m \mathbf{D}_b \mathbf{M} \mathbf{P}_b \bar{\mathbf{A}} \mathbf{H}_1^{-1} \mathbf{H}_2^{-1} \mathbf{H}_3^{-1} \mathbf{T}^{m-1} \mathbf{H}_3 \mathbf{H}_2 \mathbf{H}_1 G$$
(2.3)

For a given (u, v), every matrix in (2.3) is known to us if valance n is known. Hence it can be used to exactly and explicitly evaluate the position of $\mathbf{S}(u, v)$.

2.5 Eigen analysis of T

Equation (2.3) provides a formal parametrization of an extraordinary patch. This parametrization, however, is still costly to evaluate because it involves m - 1 multiplications of the matrix T. The evaluation process can be considerably simplified if T is decomposed as $T = X^{-1}\Lambda X$, where Λ is a diagonal matrix of eigenvalues of T and X is an invertible matrix whose columns are the corresponding eigenvectors. Therefore, the evaluation of T^{m-1} becomes the evaluation of $X^{-1}\Lambda^{m-1}X$ only.

Note that T is a block diagonal matrix. To find the eigen decomposition of T, we first find the eigen decomposition of each diagonal block T_{ω} of T:

$$T_{\omega} = X_{\omega}^{-1} \Lambda_{\omega} X_{\omega}, \ (\omega = 0, 1, \cdots, n-1).$$

Since each diagonal block T_{ω} is of size 7×7 , its eigen decomposition can be calculated explicitly. X, Λ and X^{-1} are then formed as block diagonal matrices with diagonal blocks being X_{ω} , Λ_{ω} and X_{ω}^{-1} , respectively. Consequently, $\mathbf{S}(u, v)$ can be expressed as:

$$\mathbf{S}(u,v) = W^T \mathbf{K}^m \mathbf{Z}_b \Lambda^{m-1} \mathbf{Z} G \tag{2.4}$$

where $Z = XH_3H_2H_1$ and $Z_b = D_bMP_b\bar{A}Z^{-1}$. For any given *n*, these matrices are known explicitly.

There are totally n + 6 different eigenvalues in Λ . These different eigenvalues of T are:

$$\begin{array}{rcl} \lambda_{0} & = & (4\alpha_{n} - 1 + \sqrt{16\alpha_{n}^{2} - 8\alpha_{n} + 8\beta_{n} - 3} \)/8 \\ \lambda_{1} & = & (4\alpha_{n} - 1 - \sqrt{16\alpha_{n}^{2} - 8\alpha_{n} + 8\beta_{n} - 3} \)/8 \\ \lambda_{2\omega} & = & (c_{\omega} + 5 + \sqrt{c_{\omega}^{2} + 10c_{\omega} + 9} \)/16 \\ \lambda_{2\omega+1} & = & (c_{\omega} + 5 - \sqrt{c_{\omega}^{2} + 10c_{\omega} + 9} \)/16 \\ \lambda_{n+1} & = & 1 \\ \lambda_{n+2} & = & 1/8 \\ \lambda_{n+3} & = & 1/16 \\ \lambda_{n+4} & = & 1/32 \\ \lambda_{n+5} & = & 1/64 \end{array}$$

where $1 \leq \omega \leq n/2$, $c_{\omega} = \cos(2\pi\omega/n)$, and α_n and β_n are defined in (1.1). It is easy to check that $\lambda_0 > \lambda_1$ and $\lambda_2 > \lambda_i$ for $3 \leq i \leq n$.

2.6 Evaluation of a CCSS Patch

In this section we show how can Eq. (2.4) be used in the efficient evaluation of a CCSS patch at a given (u, v). Eq. (2.4) can be used for both extraordinary and regular patches because the derivation of Eq. (2.4) did not use the assumption that $n \neq 4$.

First note that $\mathbf{S}(u, v)$ defined in Eq. (2.4) can be written as a linear combination of these different eigenvalues in Λ to the (m-1)st power:

$$\mathbf{S}(u,v) = W^T \mathbf{K}^m \sum_{j=0}^{n+5} \lambda_j^{m-1} \ (\mathbf{Z}_b \Theta_j \mathbf{Z}) G_j$$

where Θ_j is a $7n \times 7n$ matrix with all the entries being zero except the ones corresponding to λ_j in matrix Λ . Those entries of Θ_j are 1. Let $M_{b,j} = Z_b \Theta_j Z$. We get

$$\mathbf{S}(u,v) = W^T \mathbf{K}^m \sum_{j=0}^{n+5} \lambda_j^{m-1} \mathbf{M}_{b,j} G.$$
 (2.5)

The exact expressions of $M_{b,j}$ are shown in the end of this chapter. Eq. (2.5) is the most important result of this dissertation [55, 56, 57, 58, 59, 60, 61, 62, 64]. This equation can be used to evaluate a CCSS patch at any point (including (0,0)), and it can also be used to compute the derivative of a CCSS patch at any point (including (0,0)) as well). The patch can be regular or extraordinary.

Note that for any $m \ge 0$, we have $W^T(u, v) \mathbf{K}^m = W^T(2^m u, 2^m v)$. Define

$$\Phi_{b,j}(u,v) = W^T(2^m u, 2^m v) \lambda_j^{m-1} \mathcal{M}_{b,j} \Phi_b(u,v) = \sum_{j=0}^{n+5} \Phi_{b,j}(u,v).$$

 $\Phi_{b,j}(u, v)$ are called the *j*th *eigen basis function* of CCSSs. There are totally n + 6 eigen basis functions and for any given (u, v), every eigen basis function can be exactly and explicitly represented. It is easy to check that all the eigen basis functions satisfy the so called *scaling relation* [17, 20]:

$$\Phi_{b,j}(u/2, v/2) = \lambda_j \Phi_{b,j}(u, v)$$

With the above definition, Eq. (2.5) can be represented as

$$\mathbf{S}(u,v) = \Phi_b(u,v) \ G,$$

which is used for fast rendering in our implementation.

One can compute the derivatives of $\mathbf{S}(u, v)$ to any order simply by differentiating W(u, v) in Eq. (2.5) accordingly. For example,

$$\frac{\partial}{\partial u} \mathbf{S}(u, v) = \left(\frac{\partial W}{\partial u}\right)^T \mathbf{K}^m \sum_{j=0}^{n+5} \lambda_j^{m-1} \mathbf{M}_{b,j} G.$$
(2.6)

2.7 Behavior Around an extraordinary Point

2.7.1 Limit Point of an extraordinary Vertex

Eq. (2.5) not only can be used for evaluation purpose, but analytic derivation as well. For example, one gets the limit point of an extraordinary vertex simply by setting u = v = 0and $m \to \infty$ in Eq. (2.5):

$$\mathbf{S}(0,0) = \begin{bmatrix} 1, 0, \cdots, 0 \end{bmatrix} \cdot M_{2,n+1} \cdot G$$

$$= \frac{5\mathbf{V} + (12\beta_n + 8\gamma_n)\bar{\mathbf{E}} + (2\beta_n + 8\gamma_n)\bar{\mathbf{F}}}{5 + 14\beta_n + 16\gamma_n}$$
(2.7)

where $\bar{\mathbf{E}} = (\sum_{i=1}^{n} \mathbf{E}_i)/n$ and $\bar{\mathbf{F}} = (\sum_{i=1}^{n} \mathbf{F}_i)/n$. This result generalizes Eq. (13) of [11].

2.7.2 Partial Derivatives Around an extraordinary Vertex

It is known the first partial derivatives of $\mathbf{S}(u, v)$ at (0, 0) diverge in a natural parametrization [8]. However, knowing the directions of them is sufficient in many applications. As pointed out by [5], when $\lambda_0 \geq \lambda_2$, a general Catmull-Clark subdivision surface is not C^1 continuous. Suppose $\lambda_0 < \lambda_2$, dividing both sides of Eq. (2.6) by $2^m \lambda_2^{m-1}$, and by setting u = v = 0 and $m \to \infty$, we get

$$D_u(0,0) = [0, 1, 0, 0, \cdots, 0] \cdot M_{2,2} \cdot G$$

$$D_v(0,0) = [0, 0, 1, 0, \cdots, 0] \cdot M_{2,2} \cdot G$$

where D_u and D_v are the direction vectors of $\frac{\partial \mathbf{S}(0,0)}{\partial u}$ and $\frac{\partial \mathbf{S}(0,0)}{\partial v}$, respectively. The normal vector at (0,0) is the cross product of them. Similarly, when $\lambda_0 < \lambda_2$, it is easy to calculate the second partial derivatives at (0,0). These derivatives are listed as follows.

$$D_{uu}(0,0) = [0,0,0,2,0,\cdots,0] \cdot M_{2,2} \cdot G$$

$$D_{uv}(0,0) = [0,0,0,0,1,0,\cdots,0] \cdot M_{2,2} \cdot G$$

$$D_{vv}(0,0) = [0,0,0,0,0,2,0,\cdots,0] \cdot M_{2,2} \cdot G$$

where D_{uu} , D_{uv} and D_{vv} are the direction vectors of $\frac{\partial^2 \mathbf{S}(0,0)}{\partial u^2}$, $\frac{\partial^2 \mathbf{S}(0,0)}{\partial u \partial v}$ and $\frac{\partial^2 \mathbf{S}(0,0)}{\partial v^2}$, respectively. Since $M_{2,2}$ is explicitly and exactly known, all these vectors can be calculated once G is given.

2.7.3 Proof of tangent plane continuity

With the explicit expressions of partial derivatives of $\mathbf{S}(u, v)$ at (0, 0), some properties of CCSS at an extraordinary point can be proved easily. For instance, one can prove that when $\lambda_0 < \lambda_2$, there exists a common tangent plane at an extraordinary point.

The tangent plane continuity property has been proven by many people with different approaches [5, 11, 13, 14]. Here a simple proof using our parametrization results is given below.

Expand D_u and D_v , we have

$$D_u = \sum_{i=1}^n \bar{e}_i \cdot \mathbf{E}_i + \sum_{i=1}^n \bar{f}_i \cdot \mathbf{F}_i$$
$$D_v = \sum_{i=1}^n \hat{e}_i \cdot \mathbf{E}_i + \sum_{i=1}^n \hat{f}_i \cdot \mathbf{F}_i$$

where

where $c_{\omega} = \cos(2\pi\omega/n)$. All scalars x_{ij} 's in the above definitions depend on valance n only and can be derived from $M_{b,2}$ explicitly. To prove the existence of a common

tangent plane at an extraordinary point, one needs to show that computation of the normal vector is independent of k (the ID of a face adjacent to an extraordinary point [5], which determines the order of the control points of a patch):

$$\left(\sum_{i=1}^{n} \bar{e}_i \mathbf{E}_{i+k} + \sum_{i=1}^{n} \bar{f}_i \mathbf{F}_{i+k}\right) \times \left(\sum_{i=1}^{n} \hat{e}_i \mathbf{E}_{i+k} + \sum_{i=1}^{n} \hat{f}_i \mathbf{F}_{i+k}\right).$$

To prove this, it is sufficient to show that $\sum \bar{e}_i \mathbf{E}_{i+k} \times \sum \hat{e}_i \mathbf{E}_{i+k}$ is independent of k. The other parts can be proved similarly. Note

$$\sum_{i=1}^{n} \bar{e}_{i-k} \mathbf{E}_i \times \sum_{j=1}^{n} \hat{e}_{j-k} \mathbf{E}_j = \sum_{i \le j} (\bar{e}_{i-k} \hat{e}_{j-k} - \bar{e}_{j-k} \hat{e}_{i-k}) \mathbf{E}_i \times \mathbf{E}_j$$

To prove the above expression is independent of k, we only need to prove $(\bar{e}_{i-k}\hat{e}_{j-k} - \bar{e}_{j-k}\hat{e}_{i-k})$ is independent of k:

$$\begin{split} \bar{e}_{i-k} \hat{e}_{j-k} &- \bar{e}_{j-k} \hat{e}_{i-k} \\ &= \sum_{1 \le s, t \le 5} x_{s1} x_{t2} (c_{i-k-s+2} c_{j-k-t+2} - c_{j-k-s+2} c_{i-k-t+2}) \\ &= \sum_{1 \le s, t \le 5} x_{s1} x_{t2} \ (c_{i-j-s+t} - c_{j-i-s+t}) / 2 \end{split}$$

which is independent of k. Hence all the patches sharing a common extraordinary point have the same normal vector at the extraordinary point. Therefore, there exists a common tangent plane at an extraordinary point.

When $\lambda_0 \geq \lambda_2$, it can be proved similarly that the resulting surface does not have a common tangent place [5]. In fact, Eq. (2.5) and Eq. (2.6) can be used for many other analytic purposes as well. For example, the curvature property at an extraordinary point can be explicitly analyzed using these two formulas [6].

Although most of these properties of CCSS around an extraordinary vertex are well known, an explicit parametrization of CCSS nevertheless makes the analyzing process much more simpler and intuitive. Moreover, our results possibly can be used for studying other unknown properties of CCSS as well. For instance, it is possible to investigate the integrability of a CCSS using the parametrization technique presented in this chapter.

2.8 Applications

2.8.1 Fast, Exact and Explicit Rendering

Eq. (2.5) not only gives us an explicit method to evaluate $\mathbf{S}(u, v)$, but also a faster and convenient way to render $\mathbf{S}(u, v)$. Note that $M_{b,j}$ depend on the valence of the extraordinary vertex only. They can be explicitly and analytically computed for every different valence. For a given valence, we only need to perform such calculation once, no matter how many patches in the mesh are with such a valence. Once the step sizes for u and vare given, we can calculate all $\Phi_b(u_i, v_k)$ beforehand and store them in a look-up table. Therefore, the evaluation of $\mathbf{S}(u, v)$ at each point (u_i, v_k) basically is just a multiplication of $\Phi_b(u_i, v_k)$ and G only. An algorithm of the fast rendering process is shown below:

CCSS-Rendering(Mesh, ustep, vstep, β_n, γ_n)

- 1. For each valance n involved in input Mesh
- 2. For u = 0: 1: ustep and For v = 0: 1: vstep
- 3. Calculate $\Phi_b(u, v)$
- 4. For each patch whose valance is n in input Mesh
- 5. Find its 2n + 8 control points G
- 6. For u = 0: 1: ustep and For v = 0: 1: vstep
- 7. calculate each $\mathbf{S}(u, v)$ and its normal using Eq. (2.5)
- 8. Display all these $\mathbf{S}(u, v)$'s

All the examples shown in this chapter are rendered using this algorithm. One can see that it is essentially the same as the rendering process of a regular patch. An important difference between this approach and the previous approach [17] is that nothing need to be precomputed when our method is used, while the the Stam method [17] need to precompute a huge number of eigen basis functions and stored them in a file. In addition, the previous approach [17] was developed for special α_n and β_n only. Therefore, it cannot handle general eigen basis functions while we can calculate all the eigen basis



Figure 2.4: Left: Control mesh of a horse model, Right: exactly evaluated Catmull-Clark subdivision surface.

functions explicitly with only a small overhead. The horse shown in Figure 2.4 (right) is rendered using this algorithm with all the positions and normals exactly computed, not approximated. Hence, the quality of the image is better than those generated through the subdivision process. Figure 2.4 (left) is the control mesh of the shape shown in Figure 2.4 (right).

2.8.2 Generating Special Features

Eq. (2.5) can be used to render subdivision surfaces with special features. As we know, special features can be generated by properly arranging the control mesh. For instance, tripling a line in the control mesh generates a ridge or edge-like feature; tripling a control point generates a dart-like feature. One can get subdivision surfaces with complicated features and, consequently, complicated shape through this process. However, no matter how complicated the topology of the control mesh, as long as it is a two-manifold (to satisfy the definition of a CCSS), Eq. (2.5) will always generate the correct result. An example of a CCSS with sharp edges, corners and several genera is shown in Figure 2.5. The control mesh of the surface is shown in Figure 2.5(a). Since the features are generated from parametrization of the control mesh directly, the result shown in Figure 2.5(b) is better than those generated by Boolean operations.



Figure 2.5: Generating special features using Catmull Clark subdivision surfaces

2.8.3 Texture Mapping

Precise texture mapping on a CCSS is possible only if a proper parametric representation is available for each extraordinary patch. Without a proper parametrization,texture mapping on object of any topology is almost impossible. Now with Eq. (2.5), texture mapping is doable on any object of any genus.



Figure 2.6: Regular division of the control mesh of a CCSS.

However, to implement texture mapping on a CCSS, one needs to divide the interior faces of the control mesh into regions such that each region is of a rectangular structure first. Such a division will be called a *regular division*. The division is not unique.



Figure 2.7: Texture mapping on Catmull-Clark subdivision surfaces

Figure 2.6 shows a division of the interior faces of a CCSS into seven rectangular regions. Once a regular division of the interior faces of the control mesh is available, one simply performs texture mapping on each of these regions using standard approach. Examples of texture mapping on three subdivision surface represented objects: a rocker arm, a space station and a leopard are shown in Figure 2.7(a), 2.7(b), and 2.7(c), respectively. The regular division usually is not unique. Different divisions of the interior faces of the control mesh would lead to different texture outputs.

2.8.4 Surface Trimming

Surface trimming is another important application used in computer graphics, CAD and CAM. The trimming loops are defined in the parameter space of the surface and isoparametric lines in the parameter space are clipped against the trimming loops to have the trimmed regions removed. Hence, a global or local parametrization is necessary for precise and efficient rendering of a trimmed CCSS. In Figure 2.8.4, trimmed CCSSs surface are shown. In Figure 2.8(a), the trimmed regions are defined by the logo of the 2006 International CAD Conference, and in Figure 2.8(b), the trimmed regions are defined by the boundaries of the word 'SIGGRAPH'.



Figure 2.8: Surface trimming on Catmull-Clark subdivision surfaces

The CCSS surface has four extraordinary vertices in the trimmed region, but partitioning of the control mesh is not required here because the surface is rendered on the basis of individual patches.

2.8.5 Adaptive Rendering



Figure 2.9: Adaptive tessellation of Catmull-Clark subdivision surfaces

Adaptive rendering is a technique for fast rendering of complicated objects. The rendering process of a patch depends on its flatness. A flat patch will not be tessellated as densely as other patches. Adaptive rendering is not a problem with (2.5) because Eq. (2.5) is capable of generating any point of the surface required in the tessellation process. One thing we must keep in mind is that, in order to avoid crack, we must generate the same number of points on the shared boundary of adjacent faces. But we can generate any number of points, even zero, inside a patch. An example of adaptive rendering is shown in Figure 2.8.5. Figure 2.9(c) is the given ventilation control component model which is represented by a single CCSS. Its control mesh is shown in Figure 2.9(a). The adaptive tessellation of the model is shown in Figure 2.9(b). The flatness of patches is determined by the maximum norm of the second order forward differences of its control points. More details about the adaptive tessellation technique is presented in **Chapter 6**.

2.8.6 Interpolation



Figure 2.10: Interpolation using Catmull-Clark subdivision surfaces

Performing exact interpolation on meshes with arbitrary topology has been done by many people [24, 25, 23, 11, 26]. Given an control mesh the goal is to produce a smooth and visually pleasing surface whose shape matches the original data points or given normals in the given mesh exactly. Usually many constrains on the interpolatory surface need to be considered when optimization is used. For example, in [11], some energy fairing constrains are taken into account in building a global system. Because there was not an available explicit parametrization, the fairing process appeared to be very complicated in [11]. However, with our explicit parametrization and evaluation, all kinds of constrains can be integrated into the global system. For example, Figure 2.10(b) is the interpolating result of the mesh given in Figure 2.10(a) using the first, second and third derivatives as the constrains. More details about the interpolating meshes of arbitrary topology are presented in **Chapter 3**.

2.8.7 Boolean Operations

Figure 2.11: Performing Boolean operations on Catmull-Clark subdivision surfaces

In solid modelling, an object is formed by performing Boolean operations on simpler objects or primitives. A CSG tree is used in recording the construction history of the object and is also used in the ray-casting process of the object. Surface-surface intersection (including the in-on-out test) and ray-surface intersection are the core operations in performing the Boolean operations and the ray-casting process. Each operation requires a parametrization of the surface to do the work. This is especially important for the in-on-out test. None of these is a problem with Eq. (2.5). Examples of performing Boolean operations on two and three cows are presented in Figure 2.11(a) and 2.11(b), respectively. A *difference* operation is first performed to remove some portions from each of these cows and a *union* operation is then performed to join them together. Performing Boolean operations on subdivision surfaces has been studied by Biermann, Kristjansson, and Zorin [7]. The emphasis of their work is different though - they focus on construction of the approximating multiresolution surface for the result, instead of precise computation of the surface-surface intersection curves. More details about performing Boolean operations on subfirary topology are presented in **Chapter 5**.

2.9 Summary

New parametrization and evaluation techniques for extraordinary patches of CCSSs are presented in this paper. The parametrization is obtained by performing subdivision on a group of same-frequency point sets after a few linear transformations, not on the control vertices themselves directly. This results in a block diagonal matrix with constant size diagonal blocks (7×7) for the corresponding subdivision process. Consequently, eigen decomposition of the subdivision matrix is always possible and is simpler and more efficient. Besides, the new approach works for the general CCSSs, not just a special case. The evaluation process using this parametrization works for both extraordinary and regular CCSS patches.

One thing has to be pointed out here. The exponent m in (2.5) can not be cancelled out. This is because when λ_j is not a multiple of 1/2, m - 1 in K^{m-1} and $\lambda_j^{m-1}M_{b,j}$ does not cancel out. Hence, when $n \neq 4$, there does not exist a matrix M such that $S(u, v) = W^T MG$.

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Chapter 3

Similarity based Interpolation using Catmull-Clark Subdivision Surfaces

As we discussed before, there are two possible approaches to build a one-piece represented control mesh for a given model. One is to use the subdivision surface interpolation technique to approximate the surface of the given model. Another approach is to construct a mesh structure through Boolean operations and multiresolution analysis. Both approaches can achieve a one piece represented control mesh whose Catmull-Clark subdivision surface results in the given model. In this chapter we discuss the interpolation based one-piece representation method [55], i.e., to construct a one piece represented control mesh, whose Catmull-Clark subdivision surface (CCSS) interpolates the vertices of a given mesh of arbitrary topology. The Boolean operation based one-piece representation method [64] will be discussed in **Chapter 5**.

Our new interpolation method [55] handles both open and closed meshes. Normals or derivatives specified at any vertices of the mesh (which can actually be anywhere) can also be interpolated. The construction process is based on the assumption that, in addition to interpolating the vertices of the given mesh, the interpolating surface is also similar to the limit surface of the given mesh. Therefore, construction of the interpolating surface can use information from the given mesh as well as its limit surface. This approach, called *similarity based interpolation*, gives us more control on the smoothness of the interpolating surface and, consequently, avoids the need of shape fairing in the construction of the interpolating surface. The computation of the interpolating surface's control mesh follows a new approach, which does not require the resulting global linear system to be solvable. An approximate solution provided by any fast iterative linear system solver is sufficient. Nevertheless, interpolation of the given mesh is guaranteed. This is an important improvement over previous methods [11] because with these features, the new method can handle meshes with large number of vertices efficiently. Although the new method is presented for CCSSs, the concept of similarity based interpolation can be used for other subdivision surfaces as well [55].

This remaining part of this chapter is organized as follows: The Section 1 gives a brief introduction to related and previous interpolation methods. Also a overview of our our interpolation method is given in this section. In Section 2, the similarity based interpolation technique for closed meshes is discussed detailedly. A technique that works for open meshes is presented in Section 3. Implementation issues and test results are presented in Section 4. A summary is given in Section 5.

3.1 Introduction

Given a 3D mesh, there exist infinitely many smooth surfaces that interpolate the mesh vertices. Any of them can be used as a solution to the interpolation problem. But, to a shape designer, usually only one of them is the surface he really wants. That surface, called the *designer's concept surface*, is a piece of important information for the interpolation process. If that information is available to the interpolation system, then by constructing an interpolating surface whose shape is most '*similar*' to the designer's concept surface, we get the best result one can get for the interpolation process. We call an interpolation process *similarity based interpolation* if the interpolation also depends on establishing '*similarity*' with a *reference surface*. In the above case, the reference surface is the designer's concept surface.

The result of a similarity based interpolation depends on the quality of the reference surface. The closer the shape of the reference surface to the designer's concept surface, the better the result. The designer's concept surface usually is not available to the interpolation system. But it is reasonable to assume that the given mesh carries a shape similar to the designer's concept surface. After all, these are the vertices the user extracted from his concept surface. Consequently, limit surface of the given mesh, when viewed as the control mesh of a Catmull-Clark subdivision surface [1], would be similar to the designer's concept surface. Therefore, using the limit surface as the reference surface in the interpolation process, i.e., constructing an interpolating surface of a given mesh that is also similar to the limit surface of the given mesh, we should get an interpolating surface that is relatively close to the designer's concept surface. This interpolation concept has not been studied with subdivision surfaces before, although interpolation using subdivision surfaces has already been studied for a while [23, 25, 26, 42, 48].

3.1.1 Previous Work: A Brief Review

There are two major ways to interpolate a given mesh with a subdivision surface: *interpolating subdivision* [23, 24, 25, 45, 48] or *global optimization* [26, 42, 55]. In the first case, a subdivision scheme that interpolates the control vertices, such as the Butterfly scheme[25], Zorin et al's improved version [24] or Kobbelt's scheme [23], is used to generate the interpolating surface. New vertices are defined as local affine combinations of nearby vertices. This approach is simple and easy to implement. It can handle meshes with large number of vertices. However, since no vertex is ever moved once it is computed, any distortion in the early stage of the subdivision will persist. This makes interpolating subdivision very sensitive to the irregularity in the given mesh. In addition, it is difficult for this approach to interpolate normals or derivatives.

The second approach, global optimization, usually needs to build a global linear system with some constraints [47]. The solution to the global linear system is an interpolating mesh whose limit surface interpolates the control vertices in the given mesh. This approach usually requires some fairness constraints in the interpolation process, such as the energy functions presented in [42], to avoid undesired undulations. Although this approach seems more complicated, it results in a traditional subdivision surface. For example, the method in [42] results in a Catmull-Clark subdivision surface (CCSS), which is C^2 continuous almost everywhere and whose properties are well studied and understood. The problem with this approach is that a global linear system needs to be built and solved. Hence it is difficult to handle meshes with large number of control vertices. There are also subdivision techniques that produce surfaces to interpolate given curves or surfaces that near- (or quasi-)interpolate given meshes [46]. But those techniques are either of different natures or of different concerns and, hence, will not be discussed here.

3.1.2 Overview

In this chapter, we address the one-piece representation problem by using similarity based interpolation technique developed for CCSSs. Given a 3D mesh P with arbitrary topology, our new method [55] calculates a control mesh Q whose CCSS interpolates the vertices of P. The CCSS of Q is constructed with the additional assumption that its shape is *similar* to a reference surface, the limit surface of P. A shape fairing process is not required in the construction process of the interpolating surface. The computation of the control mesh Q follows a new approach which does not require the resulting global linear system to be solvable. An approximate solution provided by any fast iterative linear system solver is sufficient. Hence, handling meshes with large number of vertices is not a problem. Nevertheless, interpolation of the given mesh is guaranteed. The new method can handle both closed and open meshes. The interpolating surface can interpolate not only vertices of a given mesh, but also derivatives and normals anywhere in the parameter space of the surface.

3.2 Similarity based Interpolation

3.2.1 Mathematical Setup

Given a 3D mesh with *n* vertices: $P = \{\mathbf{P}_1, \mathbf{P}_2, \cdots, \mathbf{P}_n\}$, the goal here is to construct a control mesh Q whose CCSS interpolates P (the vertices of P, for now). The construction of Q follows the following path. First, we perform one or more levels of Catmull-Clark subdivision on P to get a finer control mesh G. G satisfies the following property: each face of G is a quadrilateral and each face of G has at most one *extra-ordinary vertex*. The vertices of G are divided into two groups. A vertex of G is called a *Type I vertex* if it corresponds to a vertex of P. Otherwise it is called a *Type II vertex*. Q is then

defined as a control mesh with the same number of vertices and the same topology as G. We assume Q has m vertices $Q = \{\mathbf{Q}_1, \mathbf{Q}_2, \dots, \mathbf{Q}_m\}, m > n$, and the first n vertices correspond to the n Type I vertices of G (and, consequently, the n vertices of P). These n vertices of Q will also be called Type I vertices and the remaining m - n vertices Type II vertices. This way of setting up Q is to ensure the parametric form developed for a CCSS patch [17, 58] can be used for the limit surface of Q, denoted $\mathbf{S}(Q)$, and we have enough degree of freedom in our subsequent work. Note that m is usually much bigger than n. The remaining job then is to determine the position of each vertex of Q.

In previous methods [26, 42] the *n* Type I vertices of *Q* are set as independent variables, the m - n Type II vertices are represented as linear combinations of the Type I vertices. Since m - n is bigger than *n*, this setting leads to an over-determined system. Without any freedom in adjusting the solution of the system, one has no control on the shape of the resulting interpolating surface $\mathbf{S}(Q)$ even if it carries undesirable undulations. In our approach [55], instead, the m - n Type II vertices are set as independent variables and the *n* Type I vertices are represented as linear combinations of the Type II vertices. This approach provides us with enough degrees of freedom to adjust the solution of the resulting linear system and, consequently, more control on the shape of the interpolating surface $\mathbf{S}(Q)$.

3.2.2 Interpolation Requirements

Recall that Type I vertices of Q are those vertices that correspond to vertices of P. Hence, each vertex of P is the limit point of a Type I vertex of Q. We assume the limit point of \mathbf{Q}_i is \mathbf{P}_i , $1 \le i \le n$. Then for each Type I vertex \mathbf{Q}_i $(1 \le i \le n)$, we have

$$\mathbf{Q}_i = C_i \cdot \tilde{Q} + c \mathbf{P}_i \tag{3.1}$$

where $\widetilde{Q} = {\mathbf{Q}_{n+1}, \mathbf{Q}_{n+2}, \cdots, \mathbf{Q}_m}$ is the vector of Type II vertices. Vector C_i and constant c depend on the topology of P and the degree of vertex \mathbf{P}_i . C_i and c can be easily obtained using the formula for calculating the limit point of a CCSS [17, 42, 58].

The conditions in Eq. (3.1) are called *interpolation requirements*, because they have to be exactly satisfied.

Note that the interpolation requirements in Eq. (3.1) form a system of linear equations. By solving this system of linear equations, we solve the interpolation problem [26]. But in this case one tends to get undesired undulations on the resulting interpolating surface [42].

3.2.3 Similarity Constraints

Two CCSSs are said to be *similar* if their control meshes have the same topology and they have similar *i*th derivatives $(1 \le i < \infty)$ everywhere. The first condition of this definition is a sufficient condition for the second condition to be true, because it ensures the considered CCSSs have the same parameter space. The CCSSs considered here, $\mathbf{S}(Q)$ and $\mathbf{S}(G)$, satisfy the first condition. Hence, we have the sufficient condition to make the assumption that $\mathbf{S}(Q)$ and $\mathbf{S}(G)$ are similar. In the following, we assume $\mathbf{S}(Q)$ and $\mathbf{S}(G)$ are similar in the sense of the above definition.

With explicit parametrization of a CCSS available [17], it is possible for us to consider derivatives of $\mathbf{S}(Q)$ and $\mathbf{S}(G)$ at any point of their parameter space. However, to avoid costly integration of derivative expressions, we will only consider derivatives sampled at the following parameter points [53]:

$$\{(k_1/2^i, k_2/2^j) \mid 0 \le i, j \le \infty, 0 \le k_1 \le 2^i, 0 \le k_2 \le 2^j\}$$
(3.2)

for each patch of $\mathbf{S}(Q)$ and $\mathbf{S}(G)$. In the above similarity definition, two derivatives are said to be *similar* if they have the same direction. In the following, we use the similarity condition to set up constraints in the construction process of $\mathbf{S}(Q)$.

Given two surfaces, let \mathbf{D}_u and \mathbf{D}_v be the u and v derivatives of the first surface and $\hat{\mathbf{D}}_u$ and $\hat{\mathbf{D}}_v$ the u and v derivatives of the second surface. These derivatives are *similar* if the following condition holds:

$$\mathbf{D}_u \times \hat{\mathbf{D}}_u = \mathbf{0} \quad \text{and} \quad \mathbf{D}_v \times \hat{\mathbf{D}}_v = \mathbf{0}$$
 (3.3)
A different condition, shown below, is used in [26, 42].

$$\mathbf{D}_{u} \cdot (\hat{\mathbf{D}}_{u} \times \hat{\mathbf{D}}_{v}) = 0 \quad \text{and} \quad \mathbf{D}_{v} \cdot (\hat{\mathbf{D}}_{u} \times \hat{\mathbf{D}}_{v}) = 0 \tag{3.4}$$

These two conditions are not necessarily equivalent. Our test cases show that Eq. (3.3) gives better interpolating surfaces. This is because Eq. (3.4) only requires the corresponding derivatives to lie in the same tangent plane, no restrictions on their directions. As a result, using Eq. (3.4) could result in unnecessary undulations. Note that Eq. (3.3) requires directions of \mathbf{D}_u and \mathbf{D}_v to be the same as that of $\hat{\mathbf{D}}_u$ and $\hat{\mathbf{D}}_v$, respectively.

Conditions of the type shown in Eq. (3.3) are called *similarity constraints*. These constraints do not have to be satisfied exactly, only to the extent possible. The interpolation method used in [26] considers interpolation requirements only. The method in [42] also includes fairness constraints to avoid undesired undulations and artifacts.

3.2.4 Global Linear System

If the derivatives of $\mathbf{S}(Q)$ and $\mathbf{S}(G)$ are sampled at a point in Eq. (3.2) then, according to Eq. (3.3) and the derivative of the parametric form of a CCSS patch [17, 53], we would have

$$(V^T \cdot Q) \times (V^T \cdot G) = \mathbf{0} \tag{3.5}$$

where V is a constant vector of scalars whose values depend on the type of the derivative and the point where the sampling is performed. This expression actually contains 3 equations, one for each component. Replace the Type I vertices $\mathbf{Q}_1, \mathbf{Q}_2, \cdots, \mathbf{Q}_n$ in the above expression with Eq. (3.1) and combine all the similarity constraints, we get a system of linear equations which can be represented in matrix form as follows:

$$\mathbf{D} \cdot X = C$$

where X is a vector of length 3(m-n), whose entries are the x, y and z components of \widetilde{Q} . D usually is not a square matrix. Hence we need to find an X such that $(D \cdot X - C)^T \cdot (D \cdot X - C)$ is minimized. This is a quadratic programming problem and can be solved using a linear least squares method. It is basically a process of finding a solution of the following linear system:

$$\mathbf{A} \cdot X = B \tag{3.6}$$

where $A = D^T D$ and $B = D^T C$. A is a symmetric matrix. Hence only half of its elements need to be calculated and stored. Once X is known, i.e., \tilde{Q} is known, we can find $\mathbf{Q}_1, \mathbf{Q}_2, \cdots, \mathbf{Q}_n$ using Eq. (3.1).

The matrix D could be very big if many sample points or constrains are used. Fortunately, we do not have to calculate and store the matrix D and the vector C. Note that A and B can be written as

$$\mathbf{A} = \sum D_i (D_i)^T \quad \text{and} \qquad B = \sum D_i c_i$$

where $(D_i)^T$ is the *i*th row of D and c_i is the *i*th entry of C. Note that the number of rows (constrains) of D can be as large as possible, but the number of its columns is fixed, 3(m-n). Suppose the *i*th constraint (See Eq. (3.5)), with $\mathbf{Q}_1, \mathbf{Q}_2, \cdots, \mathbf{Q}_n$ replaced, is written in vector form as $U^T \cdot X = u$. Then U^T is the *i*th row of matrix D and u is the *i*th entry of C. Hence rows of matrix D and entries of C can be calculated independently from Eq. (3.5) for each constraint of each sample point. Therefore, A and B can be accumulatively calculated, constraint by constraint. No matter how many sample points are used, and no matter how many constraints are considered for every sample point, only a fixed amount memory is required for the entire process and the size of matrix A is always the same, $3(m-n) \times 3(m-n)$.

Note that the solution of Eq. (3.6) only determines the positions of Type II vertices of Q. Type I vertices of Q are represented as linear combinations of Type II vertices in the interpolation requirements defined in Eq. (3.1). Since interpolation of the vertices of P is determined by the interpolation requirements (See Eq. (3.1)) only, this means as long as we can find a solution for Eq. (3.6), the task of constructing an interpolating surface that interpolates the vertices of P can always be fulfilled, even if the solution is not precise. Hence, an exact solution to the linear system Eq. (3.6) is not a must for our method. An approximate solution provided by a fast iterative linear system solver is sufficient. As a result, the new method can handle meshes with large number of vertices efficiently. This is an important improvement over previous methods.

With the similarity assumption, the surface interpolation problem is basically a process of using an iterative method to find an approximate solution for the global linear system Eq. (3.6). An initial guess for the iterative process can be obtained directly from G by scaling G properly, such that dimension of the scaled limit surface is the same as the interpolating surface. The required scaling factors s_x , s_y and s_z for such a task can be determined by the condition that the bounding box of the scaled limit surface is the same as the bounding box of the interpolating surface. This can easily be done by comparing the maxima and minima of the vertices of the given mesh in all three directions with the maxima and minima of their corresponding limit points. The scaled mesh called \hat{G} , is a good initial guess for the iterative process because \hat{G} is actually very close to the control mesh of the interpolating surface we want to obtain. In our implementation, the Gauss-Seidel method is used for the iterative process. The iterative process would converge to a good approximate solution very rapidly with this initial guess. However, it should be pointed out that there is no need to carry out the iterative process to a very precise level. According to our test cases, a residual tolerance of the size $\epsilon = 10^{-6}$ does not produce much noticeable improvement on the quality of the interpolating surface than a residual tolerance of the size $\epsilon = 10^{-3}$, while the former takes much more time than the latter. Therefore a relatively large residual tolerance can be supplied to the iterative linear system solver to prevent it from running too long on the iterative process, while not improving the quality of the interpolating surface much. This is especially important for processing meshes with large number of vertices.

3.2.5 Additional Interpolation Requirements

In addition to the *interpolation requirements* considered in Eq. (3.1), other *interpolation requirements* can be included in the global linear system as well. One can also modify or remove some of the *interpolation requirements* in Eq. (3.1). For example, if we wants

the first *u*-derivative of the interpolating surface at \mathbf{P}_i to be \mathbf{D}_u , we need to set up a condition similar to Eq. (3.5) as follows:

$$(V^T \cdot Q) \times \mathbf{D}_u = \mathbf{0}$$

where V is a constant vector. The difference here is, this is not a similarity constraint, but an interpolation requirement. However, if we want a particular normal to be interpolated, we should set up interpolation requirements for the u derivative and the v derivative whose cross product equals this normal, instead of setting up an interpolation requirement for the normal directly, to avoid the involvement of non-linear equations in the system. Then by combining all the new interpolation requirements with the original interpolation requirements in Eq. (3.1), we get all the expressions for vertices that are not considered independent variables in the linear system in Eq. (3.6). Note that including a new interpolation requirement in the interpolation requirement pool requires us to change a variable vertex in Eq. (3.6) to a non-variable vertex. Actually, interpolation requirements can be specified for any points of the interpolating surface, not just for vertices of P. This is possible because we have a parametric representation for each patch of a CCSS [17]. For example, if we want the position of a patch at parameter (1/2, 3/4) to be **T**, we can set up an interpolation requirement of the form: $V^T \cdot Q = \mathbf{T}$ where V is a constant vector whose values depend on (1/2, 3/4). Therefore the interpolating surface can interpolate positions, derivatives and normals anywhere in the parameter space.

3.2.6 Interpolation of Normal Vectors

The direction of normal vectors can be interpolated exactly by using additional interpolation requirements. The key idea is to change some similarity constrains to interpolation requirements, which means move some equations in Eq. (3.5) into the linear system in Eq. (3.1). Actually the direction of partial derivatives can also be interpolated by using such additional interpolation requirements. Additional interpolation requirements are conditions like Eq. (3.1) that are guaranteed to be satisfied and hence, are not involved in the solving of the global linear system in Eq. (3.6). However Eq. (3.5) is only good for exactly interpolating partial derivatives. For exactly interpolating normal vectors, we need to interpolate the derivatives in u- and vdirections respectively to avoid the involvement of non-linear systems. For example, for a given normal vector \mathbf{V} , whose direction is required to be interpolated at point \mathbf{P} in the interpolating surface. Assume the derivatives at point \mathbf{P} in the resulting interpolating surface in u- and v-directions are \mathbf{D}_1 and \mathbf{D}_2 , respectively. Then we need to integrate the following two equations into linear system Eq. (3.1):

$$\begin{cases} D_1 \times V = 0\\ D_2 \times V = 0 \end{cases}$$
(3.7)

Note that here $\mathbf{D_1}$ and $\mathbf{D_2}$ can be linearly represented using only the control points of the corresponding surface patch [17] and these control points are unknowns in Eq. (3.1) and Eq. (3.6). Because the above two equations in Eq. (3.7) now are in linear system Eq. (3.1), which is required to be satisfied exactly, the exact interpolation of the direction of normal vector \mathbf{V} is guaranteed. For example, Figure 3.1(f) is interpolated not only at vertex positions, but normal vectors at boundary vertices as well.

3.3 Handling Open Meshes



Figure 3.1: Interpolating an open mesh: (a) given mesh; (b) limit surface of (a); (c) extended version of (a); (d) limit surface of (c); (e) interpolating surface of (a) that uses (d) as a reference surface; (f) interpolating surface of (a) with additional requirements.

The interpolation process developed in the previous section can not be used for open meshes, such as the one shown in Figure 3.1(a), directly. This is because boundary

vertices of an open mesh have no corresponding limit points, nor derivatives, therefore, one can not set up interpolation requirements for these vertices, as required by the new interpolation process. One way to overcome this problem is to add an additional ring of vertices along the current boundary and connect the vertices of this ring with corresponding vertices of the current boundary to form an additional ring of faces, such as the example shown in Figure 3.1(c). The newly added vertices are called *dummy vertices*. We then apply the interpolation method to the extended open mesh as to a closed mesh except that there are no interpolation requirements for the dummy vertices. This technique of extending the boundary of a given mesh is similar to a technique proposed for uniform B-spline surface representation in [41].

Note that in this case, the interpolation process does not use the limit surface of the given mesh, but rather the limit surface of the extended mesh as a reference surface. Therefore, the shape of the interpolating surface depends on locations of the dummy vertices as well. Determining the location of a dummy vertex, however, is a tricky issue, and the user should not be burdened with such a tricky task. In our system, this is done by using locations of the current boundary vertices of the given mesh as the initial locations of the dummy vertices and then solving the global linear system in Eq. (3.6) to determine their final locations. This approach of generating dummy vertices works fine because dummy vertices only affect similarity constraints. Figure 3.1(e) is a surface that interpolates the mesh given in Figure 3.1(a) and uses 3.1(d) as a reference surface.

The above setting of the dummy vertices usually is not enough to create an interpolating surface with the desired boundary shape. Additional requirements (not constraints) are needed in the interpolation process. As explained in Section 3.2.5, a platform that allows us to define additional requirements can be created by treating the dummy vertices as non-variables in Eq. (3.6). We can then specify new derivative conditions or normal conditions to be satisfied at the original boundary vertices. With the additional interpolation requirements, a designer has more control on the shape of the interpolating surface in areas along the boundary and, consequently, can generate an interpolating surface with the desired boundary shape. For example, Figure 3.1(f) is an interpolating surface of the mesh given in Figure 3.1(a), but generated with additional interpolation requirements. The interpolating surface obviously looks more like a real glass now.

3.4 Test Results

The proposed approach has been implemented in C++ using OpenGL as the supporting graphics system on the Windows platform. Quite a few examples have been tested with the method described here. All the examples have extra-ordinary vertices. Some of the tested results are shown in Figures 3.1, 3.2 and 3.3. For all examples in Figures 3.1, 3.2 and 3.3, the limit surfaces of the given meshes and the interpolating surfaces are both shown so that one can tell if these surfaces are indeed similar to each other in the least squares sense. The one-piece meshes shown in Figures 3.2 and 3.3 are what we need to construct if we sample points from given models and need mesh structures whose limit surfaces approximate the given models. For example, if the model shown in Figure 3.2(c)is the given object, and we need to construct a one-piece represented mesh structure for it, the similarity based interpolation method could be used for this purpose. First we sample some representative points on the given model and connect these sampled points to form a mesh structure, which is shown in Figure 3.2(a). Once we have such a mesh structure, our similarity based interpolation technique can be applied. As a result, another mesh structure shown in Figure 3.2(d) is constructed, which can be regarded as a one-piece representation of the given model shown in Figure 3.2(c), because the limit surface of the resulting mesh structure (Figure 3.2(d)) is very close to the given model (Figure 3.2(c)).

In our implementation, only one subdivision is performed on the given mesh for each example and the first, second and third derivatives in u and v directions are used to construct interpolation constraints and build the global linear system. These derivatives are sampled at points with parameters $(\frac{k_1}{2^i}, \frac{k_2}{2^j})$, i, j = 0, 1 or ∞ , and $0 \le k_1 \le 2^i$, $0 \le k_2 \le 2^j$, for each patch. That is, 9 points are sampled for each patch, which is



Figure 3.2: Interpolating meshes with arbitrary topology.





(a) One-Piece Mesh



(d) Interpolating Surface



(e) One-Piece Mesh



(g) Limit Surface



(h) Interpolating Surface



(c) Limit Surface



(f) Given Mesh



(i) One-Piece Mesh

Figure 3.3: Interpolating meshes with arbitrary topology (Continued).

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good enough for most cases. For bigger patches one can use more sample points because patches do not have to be sampled uniformly.

The mesh shown in Figure 3.3(b) is an example of an open mesh with disconnected boundaries. Figure 3.3(d) is the interpolating surface without using additional interpolation requirements in the construction process.

As can be seen from Figure 3.2 and 3.3, all the resulting interpolating surface are very smooth and visually pleasing, except the interpolating surface shown in Figure 3.3(h). The surface has some undulations around the neck, but we do not think they are caused completely by our method. We believe this is more of a problem with the general interpolation concept. Note that the input mesh, Figure 3.3(f), has some abrupt changes of vertex positions and twists in the neck area. This is also reflected by some visible undulations in the neck area of the limit surface, Figure 3.3(g), even though they are not as clear as in the interpolating surface. An approximation curve/surface, like a spline curve, can be regarded as a low pass filter [24], which makes the given control polygon or mesh smoother. An interpolation curve/surface, on the other hand, can be regarded as a high pass filter, which magnifies undulations or twists in the input mesh. Since a limit surface is an approximation surface, it reduces the impact of abrupt vertex location changes and twists in the input mesh while the interpolating surface enhances it. This is why the undulations are more obvious in Figure 3.3(h) than in Figure 3.3(g).

The new interpolation method can handle meshes with large number of vertices in a matter of seconds on an ordinary PC (3.2GHz CPU, 512MB of RAM). For example, the meshes shown in Figures 3.3(f), 3.2(a) and 1.6(b) have 1022, 354 and 272 vertices, respectively. It takes 51, 14 and 3 seconds, respectively, to interpolate these meshes. For smaller meshes, like Figures 3.1(a), 3.2(i), 3.2(e) and 3.3(b), the interpolation process is done almost in real time. Hence our interpolation method is suitable for interactive shape design, where simple shapes with small or medium-sized control vertex sets are constructed using design or interpolation methods, and then combined using CSG trees to form complex objects.

3.5 Summary

A new interpolation method for meshes with arbitrary topology using general CCSSs is presented. This interpolation technique gives us a one-piece represented control mesh, whose limit surface approximates the target model. The development of the interpolation method is based on the assumption that the interpolating surface should be *similar* to the limit surface of the given mesh. Our test results show that this approach leads to good interpolation results even for complicated data sets.

The new method has several special properties. First, by using information from the vertices of the given mesh as well as its limit surface, one has more control on the smoothness of the interpolating surface. Hence, a *surface fairing* process is not needed in the new method. Second, there is no system solvability problem for the new method. The global linear system that the new method has to solve does not require an exact solution, an approximate solution is sufficient. The approximate solution can be provided by any fast iterative linear solver. Consequently the new method can process meshes with large number of vertices efficiently. Third, the new method can handle both open and closed meshes. It can interpolate not only vertices, but normals and derivatives as well. These normals and derivative can be anywhere, not just at the vertices of the given mesh. Therefore, the new interpolation method is general.

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Chapter 4

Voxelization of Free-form Solids Represented by Catmull-Clark Subdivision Surfaces

A voxelization technique [57] and its applications for objects with arbitrary topology are presented in this chapter. The voxelization technique will be used for performing accurate Boolean operations discussed in next chapter. By performing CSG or Boolean operations [64], we can obtain one-piece representations for objects of arbitrary topology.

With parametrization techniques for subdivision surfaces becoming available [17, 58], it is possible now to model and represent any continuous but topologically complex object with an analytical representation. In this chapter we present a method to convert a freeform object from its continuous geometric representation into a set of voxels that best approximates the geometry of the object. Unlike traditional 3D scan-conversion based methods [74, 75, 76, 91, 77], our voxelization method is performed by recursively subdividing the 2D parameter space and sampling 3D points from selected 2D parameter space points. Because we can calculate every 3D point position explicitly and accurately, uniform sampling on surfaces with arbitrary topology is not a problem any more. Moreover, our discretization of 3D closed objects is guaranteed to be leak-free when a 3D flooding operation is performed. This is ensured by proving that our voxelization results satisfy the properties of separability, accuracy and minimality. In addition, a 3D volume flooding algorithm using dynamic programming techniques is presented which significantly speeds up the volume flooding process. Hence our method is suitable for visualization of complex scenes, measuring object volume, mass, surface area, determining intersection curves of multiple surfaces and performing accurate Boolean/CSG operations. These capabilities are demonstrated by test examples shown in this chapter.

The structure of this chapter is as follows: A brief introduction is given in Section 1. Some background about 3D discrete space is introduced in Section 2. In Section 3, some related work is discussed. The voxelization method is presented in Section 4. The proof of the correctness of our voxelization method is given in Section 5. In Section 6, a dynamic programming method based volume flooding algorithm is presented. Some applications of the voxelization technique are discussed and some test examples are shown in Section 7. We draw some conclusions in Section 8.

4.1 Introduction

Volume graphics [69] represents a set of techniques aimed at modeling, manipulation and rendering of geometric objects, which have proven to be, in many aspects, superior to traditional computer graphics approaches. The main advantages of volume graphics are: (1) decoupling of voxelization from rendering, (2) uniformity of representation, and (3) support of Boolean, block and CSG operations. Two drawbacks of volume graphics techniques are their high memory and processing time demands. However, with the progress in both computers and specialized volume rendering hardware, these drawbacks are gradually losing their significance.

To be represented by the voxel raster, a geometric object has to go through a process called *voxelization*. This process is concerned with converting geometric objects from their continuous geometric representation into a set of voxels that best approximates the continuous object. Traditional voxelization methods (also referred to as 3D scanconversion) mimic the 2D scan-conversion process that pixelizes (rasterizes) 2D geometric objects. Hence traditional voxelization methods only work well for polygon based 3D objects. For surfaces with arbitrary topology, voxelization using 3D scan-conversion is not efficient, nor accurate.

Subdivision surfaces have become popular recently in graphical modeling, visualization and animation because of their capability in modeling/representing complex shape of arbitrary topology [1], their relatively high visual quality, and their stability and efficiency in numerical computation. Subdivision surfaces can model/represent complex shape of arbitrary topology because there is no limit on the shape and topology of the control mesh of a subdivision surface. With parametrization techniques for subdivision surfaces becoming available [17, 58] and with the fact that non-uniform B-spline and NURBS surfaces are special cases of subdivision surfaces becoming known [15], we now know that subdivision surfaces cover both *parametric forms* and *discrete forms*. Parametric forms are good for design and representation, discrete forms are good for machining and tessellation (including FE mesh generation). Hence, we have a representation scheme that is good for all graphics and CAD/CAM applications.

In this chapter we propose a voxelization method for free-form solids represented by Catmull-Clark subdivision surfaces. Instead of direct sampling of 3D points, the new method is based on recursive sampling of 2D parameter space points of a surface patch. Hence the new method is more efficient and less sensitive to numerical error.

Note that a voxelization process does not render the voxels but merely generates a database of the discrete digitization of the continuous object [90]. Some previous voxelization methods use quad-trees to store the voxelization result. This approach can save memory space but might sacrifice in time when used for applications such as Boolean operations or intersection curves determination. Nevertheless, with cheap and giga-byte memory chips becoming available, storage requirement is no longer a major issue in the design of an algorithm. People care more about the efficiency of the algorithm. The new method stores the voxelization result directly in a *Cubic Frame Buffer* for fast operation purpose.

4.2 Background: 3D Discrete Space

A 3D discrete space is a set of integral grid points in 3D Euclidean space defined by their Cartesian coordinates (x, y, z), with $x, y, z \in Z$. A voxel is a unit cube centered at the integral grid point. Usually a voxel is assigned a value of 0 or 1. The voxels assigned an '1', called the 'black' voxels, represent opaque objects. Those assigned a '0', called the 'white' voxels, represent the transparent background. Outside the scope of this paper is a non-binary approach where the voxel values are mapped onto the interval [0,1] representing either partial coverage, variable densities, or graded opacities. Due to its larger dynamic range of values, this approach can support higher quality rendering.

Two voxels are said to be 26-adjacent (See Figure 4.1(c)) if they share a vertex, an edge, or a face. Every given voxel has 26 such adjacent voxels: eight share a vertex (corner) with the given voxel, twelve share an edge, and six share a face. Accordingly, face-sharing voxels are said to be 6-adjacent (See Figure 4.1(a)), and edge-sharing and face-sharing voxels are said to be 18-adjacent (See Figure 4.1(b)).



Figure 4.1: *N*-adjacent, $N \in \{6, 18, 26\}$.

The prefix N is used to define the adjacency relation, with N = 6, 18, or 26. A sequence of voxels having the same value (e.g., 'black') is called an N-path if all consecutive pairs are N-adjacent. A set of voxels are said to be N-connected if there is an N-path between every pair of its voxels. It is easy to see that N-connectedness is an equivalent relation. Given three disjoint sets of voxels A, B and C, A is said to N-separate B and C if any N-path from a voxel of B to a voxel of C intersects A.

4.3 Previous Voxelization Techniques

Voxelization techniques can be classified into two major categories. The first category consists of methods that extend the standard 2D scan-line algorithm and employ numerical considerations to guarantee that no gaps appear in the resulting discretization. As we know polygons are fundamental primitives in 3D surface graphics in that they approximate arbitrary surfaces as a mesh of polygonal patches. Hence, early work on voxelization focused on voxelizing 3D polygon meshes [74, 75, 76, 91, 77] by using 3D scan-conversion algorithm. Although this type of methods can be extended to voxelize parametric curves, surfaces and volumes [78], it is difficult to deal with free-from surfaces of arbitrary topology.

The other widely used approach for voxelizing free-form solids is to use spatial enumeration algorithms which employ point or cell classification methods in either an exhaustive fashion or by recursive subdivision [84, 85, 86, 87]. However, 3D space subdivision techniques for models decomposed into cubic subspaces are computationally expensive and thus inappropriate for medium or high resolution grids. The voxelization technique that we will be presenting uses recursive subdivision. The difference is the new method performs recursive subdivision on 2D parameter space, not on the 3D object. Hence expensive distance computation between 3D points is avoided.

Like 2D pixelization, voxelization is a powerful technique for representing and modeling complex 3D objects. This is proved by many successful applications of volume graphics techniques in recently reported research work. For example, voxelization can be used for visualization of complex objects or scene [85]. It can also be used for measuring integral properties of solids, such as mass, volume and surface area [87]. Most importantly, it can be used for intersection curve calculation and performing accurate Boolean operations. For example, in [86, 88], a series of Boolean operations are performed on objects represented by a CSG tree. Voxelization is such an important technique that several hardware implementations of this technique have been reported recently [80, 81].

4.4 Voxelization based on Recursive 2D Parameter Space Subdivision

4.4.1 Basic Idea

Given a free-form object represented by a CCSS and a cubic frame buffer of resolution $M_1 \times M_2 \times M_3$, the goal is to convert the CCSS represented free-form object (i.e. continuous geometric representation) into a set of voxels that best approximates the geometry of the object. We assume each face of the control mesh is a quadrilateral and each face has at most one extra-ordinary vertex (a vertex with a *valence* different from 4). If this is not the case, simply perform Catmull-Clark subdivision on the control mesh of the CCSS twice.

With parametrization techniques for subdivision surfaces becoming available, it is possible now to model and represent any continuous but topologically complex object with an analytical representation [17, 18, 20, 58]. Consequently, any point in the surface can be explicitly calculated. On the other hand, for any given parameter space point (u, v), a surface point S(u, v) corresponding to this parameter space point can be exactly computed as well. Therefore, voxelization does not have to be performed in the 3D object space, as the previous recursive voxelization methods did, one can do voxelization in 2D space by performing recursive subdivision and testing on the 2D parameter space.

We first consider the voxelization process of a subpatch, which is a small portion of a patch. Given a subpatch of $\mathbf{S}(u, v)$ defined on $[u_1, u_2] \times [v_1, v_2]$, we voxelize it by assuming this given subpatch is small enough (hence, flat enough) so that all the voxels generated from it are the same as the voxels generated using its four corners:

$$\mathbf{V}_1 = \mathbf{S}(u_1, v_1), \quad \mathbf{V}_2 = \mathbf{S}(u_2, v_1), \quad \mathbf{V}_3 = \mathbf{S}(u_2, v_2), \quad \mathbf{V}_4 = \mathbf{S}(u_1, v_2).$$
(4.1)

Usually this assumption does not hold. Hence a test must be performed before the patch or subpatch is voxelized. It is easy to see that if the voxels generated using its four corners are not N-adjacent ($N \in \{6, 18, 26\}$) to each other, then there exist holes between them. In this case, the patch or subpatch is still not small enough. To make



Figure 4.2: Basic idea of parameter space based recursive voxelization.

it smaller, we perform a *midpoint subdivision* on the corresponding parameter space by setting

$$u_{12} = \frac{u_1 + u_2}{2}$$
 and $v_{12} = \frac{v_1 + v_2}{2}$

to get four smaller subpatches:

$$\begin{aligned} &\mathbf{S}([u_1, u_{12}] \times [v_1, v_{12}]), \quad \mathbf{S}([u_{12}, u_2] \times [v_1, v_{12}]), \\ &\mathbf{S}([u_{12}, u_2] \times [v_{12}, v_2]), \quad \mathbf{S}([u_1, u_{12}] \times [v_{12}, v_2]), \end{aligned}$$

and repeat the testing process on each of the subpatches. The process is recursively repeated until all the subpatches are small enough and can be voxelized using only their four corners.

The vertices of the resulting subpatches after the recursive parameter space subdivision are then used as vertices for voxelization that approximates the limit surface. For example, if the four rectangles in Figure 4.2(a) are the parameter spaces of four adjacent subpatches of $\mathbf{S}(u, v)$, and if the rectangles shown in Figure 4.2(b) are the parameter spaces of the resulting subpatches when the above recursive testing process stops, then 3D points will be *evaluated* at the 2D parameter space points marked with small solid circles to form voxels that approximate the limit surface.

To make things simple, we first normalize the input mesh to be of dimension $[0, M_1 - 1] \times [0, M_2 - 1] \times [0, M_3 - 1]$. Then for any 2D parameter space point (u, v) generated from the recursive testing process (See Figure 4.2), direct and exact evaluation is performed to

get its 3D surface position and normal vector at S(u, v). To get the voxelized coordinates (i, j, k) from S(u, v), simply set

$$i = \lfloor S(u, v) \cdot x + 0.5 \rfloor, \quad j = \lfloor S(u, v) \cdot y + 0.5 \rfloor, \quad k = \lfloor S(u, v) \cdot z + 0.5 \rfloor.$$
(4.2)

Once every single point marked in the recursive testing process is voxelized, the process for voxelizing the given patch is finished. The proof of the correctness of our voxelization results will be discussed in the next section.

Since the above process guarantees that shared boundary or vertex of patches or subpatches will be voxelized to the same voxel, we can perform voxelization of free-form objects represented by a CCSS patch by patch. One thing that should be pointed out is, to avoid stack overflow, only small subpatches should be fed to the recursive subdivision and testing process. This is especially true when a high resolution cubic frame buffer is given or some polygons are very big in the given control mesh. Generating small subpatches is not a problem for a CCSS once the parametrization techniques are available. For example, in our implementation, the size of subpatches (in the parameter space) fed to recursive testing is $\frac{1}{8} \times \frac{1}{8}$, i.e. each patch is divided into 8×8 subpatches before the voxelization process. In addition, feeding small size subpatches to the recursive testing process the assumption of our voxelization process to be satisfied, because the smaller the parameter size of a subpatch, the flatter the subpatch.

4.4.2 Voxelization Algorithms

The above voxelization method, based on recursive subdivision of the parameter space, is summarized into the following algorithms: *Voxelization* and *VoxelizeSubPatch*. The parameters to these algorithms are explained as follows. S: control mesh of a CCSS which represents the given object; N: an integer that specifies the N-adjacent relationship between adjacent voxels; M_1 , M_2 , and M_3 : resolution of the Cubic Frame Buffer; k: an integer that specifies the number of subpatches $(k \times k)$ that should be generated before fed to the recursive voxelization process. **Voxelization**(Mesh S, int N, int M_1 , int M_2 , int M_3 , int k)

- 1. Normalize S so that S is bounded $[0, M_1 1] \times [0, M_2 1] \times [0, M_3 1]$
- 2. for each patch pid in S

3. for
$$u = \frac{1}{k} : 1$$
, step size $\frac{1}{k}$

4. for
$$v = \frac{1}{k} : 1$$
, step size $\frac{1}{k}$

5. VoxelizeSubPatch $(N, pid, u - \frac{1}{k}, u, v - \frac{1}{k}, v);$

VoxelizeSubPatch(int N, int pid, float u_1 , float u_2 , float v_1 , float v_2)

1.
$$(i_1, j_1, k_1) = \text{Voxelize}(S(pid, u_1, v_1));$$

2. $(i_2, j_2, k_2) = \text{Voxelize}(S(pid, u_2, v_1));$

3.
$$(i_3, j_3, k_3) = \text{Voxelize}(S(pid, u_2, v_2));$$

- 4. $(i_4, j_4, k_4) = \text{Voxelize}(S(pid, u_1, v_2));$
- 5. $if(|u_2 u_1| < 1/\max\{M_1, M_2, M_3\})$ return;
- 6. $\Delta_i = \max\{|i_a i_b|\}, \text{ with } a \text{ and } b \in \{1, 2, 3, 4\};$
- 7. $\Delta_j = \max\{|j_a j_b|\}, \text{ with } a \text{ and } b \in \{1, 2, 3, 4\};$
- 8. $\Delta_k = \max\{|k_a k_b|\}$, with a and $b \in \{1, 2, 3, 4\}$;
- 9. if $(N = 6 \& \Delta_i + \Delta_j + \Delta_k \leq 1)$ return;
- 10. if $(N = 18 \& \Delta_i \le 1 \& \Delta_j \le 1 \& \Delta_k \le 1 \& \Delta_i + \Delta_j + \Delta_k \le 2)$ return;
- 11. if $(N = 26 \& \Delta_i \leq 1 \& \Delta_j \leq 1 \& \Delta_k \leq 1)$ return;
- 12. $u_{12} = (u_1 + u_2)/2;$ $v_{12} = (v_1 + v_2)/2;$
- 13. VoxelizeSubPatch $(N, pid, u_1, u_{12}, v_1, v_{12})$;
- 14. VoxelizeSubPatch $(N, pid, u_{12}, u_2, v_1, v_{12})$;
- 15. VoxelizeSubPatch $(N, pid, u_{12}, u_2, v_{12}, v_2)$;
- 16. VoxelizeSubPatch $(N, pid, u_1, u_{12}, v_{12}, v_2)$;

In algorithm 'VoxelizeSubPatch', corresponding surface points for the four corners are evaluated using Eq. (2.5), where *pid* tells us which patch we are currently working on. The routine 'Voxelize' voxelizes points by using Eq. (4.2). Lines 9, 10 and 11 are used to test if voxelizing the four corners of a subpatch is enough to generate a 6-, 18- and 26-adjacent voxelization, respectively. While Line 5 prevents the recursive process from non-stop dead loop in case Lines 9, 10 and 11 are always not satisfied.

4.5 Separability, Accuracy and Minimality

Let S be a C^1 continuous surface in R^3 . We denote by \overline{S} the discrete representation of S. \overline{S} is a set of black voxels generated by some digitalization method. There are three major requirements that \overline{S} should meet in the voxelization process. First, *separability* [90, 91], which requires to preserve the analogy between continuous and discrete space and to guarantee that \overline{S} is not penetrable since S is C^1 continuous. Second, *accuracy*. This requirement ensures that \overline{S} is the most accurate discrete representation of S according to some appropriate error metric. Third, *minimality* [90, 91], which requires the voxelization should not contain voxels that, if removed, make no difference in terms of separability and accuracy. The mathematical definitions for these requirements can be found in [91], which are based on [90].

First we can see that voxelization results generated using our recursive subdivision method satisfy the requirement of minimality. The reason is that voxels are sampled directly from the object surface. The termination condition of our recursive sampling process (i.e., Line 8, 9, 10 in algorithm 'VoxelizeSubPatch') and the coordinates transformation in Eq. (4.2) guarantee that every point in the surface has one and only one image in the resulting voxelization. In other words,

$$\forall P \in S, \exists Q \in \overline{S}, \text{ such that } P \in Q.$$
(4.3)

Note that here P is a 3D point and Q is a voxel, which is a unit cube. On the other hand, because all voxels are mapped directly from the object surface using Eq. (4.2), we have

$$\forall Q \in \overline{S}, \exists P \in S, \text{ such that } P \in Q.$$
(4.4)

Hence no voxel can be removed from the resulting voxelization, i.e., the property of minimality is satisfied. In addition, from Eq. (4.3) and Eq. (4.4) we can also conclude

that the resulting binary voxelization is the most accurate one with respect to the given resolution. Hence the property of accuracy is satisfied as well.

To prove that our voxelization results satisfy the separability property, we only need to show that there is no holes in the resulting voxelization. For simplicity, here we only consider 6-separability, i.e., there does not exist a ray from a voxel inside the freeform solid object to the outside of the free-form solid object in x, y or z direction that can penetrate our resulting voxelization without intersecting any of the black voxels. We prove the separability property by contradiction. As we know violating separability means there exists at least a hole (voxel) Q in the resulting voxelization that is not included int \overline{S} but is intersected by S and, there must also exist two 6-adjacent neighbors of Q that are not included in \overline{S} either and are on opposite sides of S. Because S intersects with Q, there exist at least one point P on the surface that intersects with Q. But the image of P after voxelization is not Q because Q is a hole. However, the image of P after voxelization must exist because of the termination condition of our recursive sampling process (i.e., Line 8, 9, 10 in algorithm 'VoxelizeSubPatch'). Moreover, according to our voxelization method, P can only be voxelized into voxel Q because of Eq. (4.2). Hence Q cannot be a hole, contradicting our assumption. Therefore, we conclude that \bar{S} is 6-separating.

4.6 Volume Flooding with Dynamic Programming4.6.1 Seed Selection

A seed must be designated before a flooding algorithm can be applied. In 2D flooding, a seed is usually given by the user interactively. However, in 3D flooding, for a closed 3D object, it is impossible for a user to designate a voxel as a seed by mouse-clicking because voxels inside a closed 3D object are invisible. Hence an automatic method is needed to select an inside voxel as a seed for volume flooding. Once we can correctly choose an inside voxel, the by applying a flooding operation, all inside voxels can be obtained. To select a voxel as a seed for volume flooding, we need to tell if a voxel is inside or outside

the 3D object. This is not a trivial problem. In the past In-Out test for voxels is not efficient and not accurate [87], especially for topologically complicated 3D objects.

With the availability of parametrization techniques for subdivision surfaces, we now can calculate derivatives and normals exactly and explicitly for each point located on the 3D object surface. Hence the normal for each voxel can also be exactly calculated in the voxelization process. Because the direction of a normal is perpendicular to the surface and points towards the outside of the surface, the closest voxel in its opposite direction must be located either inside or on the surface (Assume the voxelization resolution is high enough). For a given voxel (called *start voxel*), to choose the closest voxel in its normal's opposite direction, we just need to calculate the dot product of its normal and one of the axis vectors. These vectors are: $\{1, 0, 0\}, \{-1, 0, 0\}, \{0, 1, 0\}, \{0, -1, 0\}, \{0, 0, 1\}, \{0,$ $\{0, 0, -1\}$ corresponding to x, -x, y, -y, z and -z direction, respectively. The direction with biggest dot product is chosen for finding an inside voxel. If the closest voxel in this chosen direction is also a black voxel (i.e., located on the 3D object surface), another start voxel has to be selected and the above process is repeated until an inside voxel is found. The found inside voxel can be designated as a seed for inside volume flooding. Similarly, an outside voxel can also be found for outside volume flooding. In this case, the seed voxel should not be chosen from the normal's opposite direction, but along the normal's direction.



Figure 4.3: A voxel with multiple pieces of object surface in it.

However, if the voxelization resolution is not high enough, the closest voxel in the normal's opposite direction might be an outside voxel. For example, in Figure 4.3, *ABCD*

denotes a voxel and part of the object surface passes through this voxel. Differently, there are two pieces of surface that are not connected but are all inside this voxel. If we choose P_1 as the start point in Figure 4.3 to find an inside voxel using the above seed selection method, an outside voxel will be wrongly chosen. Hence the above method is no longer applicable in this case. To resolve the problem in this situation, higher voxelization resolution could be used. However, no matter how high the voxelization resolution is, we still cannot guarantee cases like the one shown in Figure 4.3 will not occur. Hence other approach is needed.

Fortunately, voxels that have multiple pieces of surface passing through, like the one shown in Figure 4.3, can be easily identified in the voxelization process. To identify these voxels, we need to calculate normals for each voxel. For example, in Figure 4.3, if surface point P_1 is mapped to voxel ABCD, then the normal at P_1 which is N_1 , is also memorized as the normal of this voxel. Next time if another surface point, say P_2 , is also mapped to voxel ABCD, then the normal at P_2 which is N_2 , will be first compared with the memorized normal of voxel ABCD by calculating their dot product. If $N_1 \cdot N_2 > 0$, then nothing need to be done. Otherwise, say surface point P_3 , which is mapped to the same voxel and its normal is N_3 , if $N_1 \cdot N_3 \leq 0$, then this voxel is marked as a voxel that has multiple piece passing through. Once every voxel that has multiple pieces of surface passing through is marked, we can easily solve the problem simply by not choosing these marked voxels as the start voxels.

4.6.2 3D Flooding using Dynamic Programming

Here we only present flooding algorithms using 6-separability, but the idea can be applied to N-separability with N = 18 or 26, Although 6-separability is used in the flooding process, the voxelization itself can be N-adjacent with N = 6, 18 or 26, Once a seed is chosen, 3D flooding algorithms can be performed in order to fill all the voxels that are 6-connected with this seed voxel. The simplest flooding algorithm is *recursive flooding*, which recursively search adjacent voxels in 6 directions for 6-connected voxels. This method sounds ideally reasonable but does not work in real world because even for a very low resolution, it would still cause stack overflow.

Another method that can be used for flooding is called *linear flooding*, which searches adjacent voxels that are 6-connected with the given the seed voxel, linearly from the first voxel to the last voxel in the cubic frame buffer, and marks all the found voxels with gray. The search process is repeated until no more white ('0') voxels is found that are 6-connected with one of the gray voxels. Linear flooding is simple and does not require extra memory in the flooding process. However, it is very slow, especially when a high resolution is used in the voxelization process.

In many applications, 3D flooding operations are required to be fast with low extra memory consumption. To make a 3D flooding algorithm applicable and efficient, we can combine the recursive flooding and the linear flooding methods using the so called dynamic programming technique.

Dynamic programming usually breaks a problem into subproblems, and these subproblems are solved and the solutions are memorized, in case they need to be solved again. This is the essentiality of dynamic programming. To use dynamic programming in our 3D flooding algorithm, we use a sub-routine *FloodingXYZ* which marks inside voxels having the same x, y or z coordinates as the given seed voxel, and all marked voxels are memorized by pushing them into a stack called *GRAYSTACK*. Note here the stack has a limited space, whose length is specified by the user. When the stack reaches its maximal capacity, no gray voxels can be pushed into it. Hence it guarantees limited memory consumption. The 3D flooding algorithm with dynamic programming can improve the flooding speed significantly. For ordinary resolution, say, $512 \times 512 \times 512$, a flooding operation can be done almost in real time. The pseudo code for the 3D volume flooding algorithm is given as follows and the parameters (s_i, s_j, s_k) are the coordinates of the given seed voxel. **VolumeFlooding**(int s_i , int s_j , int s_k)

```
1.
     FloodingXYZ(s_i, s_j, s_k);
2.
     loop = 1;
3.
     while(loop)
4.
        while (GRAYSTACK is not empty)
5.
           (i, j, k) = \text{GRAYSTACK.Pop}();
6.
           FloodingXYZ(i, j, k)
7.
        loop = 0;
8.
        for(i = 0; i < M_1; i++)
           for(j = 0; j < M_2; j++)
9.
               for (k = 0; k < M_3; k++)
10.
11.
                   if (Voxel (i, j, k) is white and is 6-adjacent with a gray voxel)
12.
                      FloodingXYZ(i, j, k);
13.
                      loop = 1;
```

4.7 Applications

4.7.1 Visualization of Complex Scenes

Ray tracing is a commonly used method in the field of visualization of volume graphics. This is due to its ability to enhance spatial perception of the scene using techniques such as transparency, mirroring and shadow casting. However, there is a main disadvantage for ray tracing approach: large computational demands. Hence rending using this method is very slow. Recently, surface splatting technique for point based rendering has become popular [63, 89]. Surface splatting requires the position and normal of every point to be known, but not their connectivity. With explicit position and exact normal information for each voxel in our voxelization results, now it is much easier for us to render discrete voxels using surface splatting techniques. The rendering is fast and high quality results can be obtained. For example, Figure 4.4(f) is the given mesh, Figure 4.4(g) is the corresponding limit surface. After the voxelization process, Figure 4.4(h) is generated

only using basic point based rendering techniques with explicitly known normals to each voxel. While Figure 4.4(i) is rendered using splatting based techniques. The size of cubic frame buffer used for Figure 4.4(h) is $512 \times 512 \times 512$. The voxelization resolution used for Figure 4.4(i) is $256 \times 256 \times 256$. Although the resolution is much lower, we can tell from Figure 4.4, that the one using splatting techniques is smoother and closer to the corresponding object surface given in Figure 4.4(g).

4.7.2 Integral Properties Measurement

Another application of voxelization is that it can be used to measure integral properties of solid objects such as mass, volume and surface area. Without discretization, these integral properties are very difficult to measure, especially for free-form solids with arbitrary topology.

Volume can be measured simply by counting all the voxels inside or on the surface boundary because each voxel is a unit cube. With efficient flooding algorithm, voxels inside or on the boundary can be precisely counted. But the resulting measurement may not be accurate because boundary voxels do not occupy all the corresponding unit cubes. Hence for higher accuracy, higher voxelization resolution is needed. Once the volume is known, it is easy to measure the mass simply by multiplying the volume with density. Surface area can be measured similarly. But using this approach would lead to big error because we do not know how surfaces pass through their corresponding voxels. Fortunately, surface area can be measured much more precisely in the voxelization process. As we know, during the recursive voxelization process, if the recursive process stops, all the marked parameter points of a patch or subpatch (See Figure 4.2) are points used for final voxelization. Hence all these quadrilaterals corresponding to these marked parameter points can be used for measuring surface area after these marked parameter space points are mapped to 3D space. The flatness of these quadrilaterals is required to be tested if high accuracy is needed. The definition of patch flatness and the flatness testing method can be found in [59].



Figure 4.4: Applications of Voxelization

4.7.3 Performing Boolean and CSG Operations

The most important application of voxelization is to perform Boolean and CSG operations on free-form objects. In solid modeling, an object is formed by performing Boolean operations on simpler objects or primitives. A CSG tree is used in recording the construction history of the object and is also used in the ray-casting process of the object. Surface-surface intersection (including the in-on-out test) and ray-surface intersection are the core operations in performing the Boolean and CSG operations. With voxelization, all of these problems become much easier set operations. For instance, Figure 4.4(d) is generated by subtracting a cylinder from the Venus model. While Figure 4.4(k) and Figure 4.4(1) are the union and difference results of the cow model and the rocker arm model shown in Figure 4.4(g). Note that all these union and difference pairs are positioned the same way when Boolean operations are performed. Figure 4.4(j) is generated by subtracting the heart model shown in Figure 4.4(c), from rock arm model shown in Figure 4.4(g). And Figure 4.4(c) is generated by subtracting the rock arm model shown in Figure 4.4(g) from the heart model. A mechanical part is also generated in Figure 4.4(e) using CSG operations. Intersection curves can be similarly generated by searching for common voxels of objects. The black curve shown in Figure 4.4(b) and Figure 4.4(a) is the intersection curve generated from two different objects.

4.8 Summary

A method to convert a free-form object from its continuous geometric representation to a set of voxels that best approximates the geometry of the object is presented. The new voxelization method can be used furthermore in next chapter for our subdivision surface based one-piece representation system when Boolean operation is used to construct onepiece represented mesh structure. Unlike traditional 3D scan-conversion based methods, the new method does the voxelization process by recursively subdividing the 2D parameter space and sampling 3D surface points only at selected 2D parameter space positions. Because of the capability to calculate every 3D point position explicitly and accurately, uniform sampling on surfaces with arbitrary topology is not a problem for the approach at all. Moreover, the new method guarantees that discretization of 3D closed objects is leak-free when a 3D flooding operation is performed. This is ensured by proving that voxelization results of the new method satisfy the properties of separability, accuracy and minimality. In addition, a 3D volume flooding algorithm using dynamic programming techniques is presented which significantly speeds up the volume flooding process. Hence the new method is suitable for visualization of complex scenes, measuring object volume, mass, surface area, determining intersection curve of multiple surfaces and performing accurate Boolean/CSG operations.

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Chapter 5

Robust and Error Controllable Boolean Operations on Free-Form Solids Represented by Catmull-Clark Subdivision Surfaces

In this chapter, a method for performing robust and error controllable Boolean operations on free-form solids represented by Catmull-Clark subdivision surfaces (CCSSs) is presented [64]. The given objects are voxelized [57] using the voxelization method presented in **chapter 4** to make Boolean operations more efficient. However, different from previous voxelization-based approaches, the final results of the Boolean operations in our method are represented with a continuous geometric representation, that is, our results after Boolean operations are one-piece representations of solid objects. They are represented with topologically correct mesh structure [64]. This is achieved by doing the Boolean operations in the parameter spaces of the solids, instead of the object space. The 2D parameter space is recursively subdivided until a keep-or-discard decision can be made for each resulting subpatch using results of the voxelization process. This approach allows us to easily compute a parametric approximation of the intersection curve and, consequently, to build a continuous geometric representation for the Boolean operation result. To make the Boolean operation result more accurate, a secondary local voxelization can be performed for intersecting subpatches. Because the voxelization process itself is very fast and robust, the overall process is fast and robust too. Most importantly, the error of the Boolean operation results can be estimated, so error control is possible. In addition, our method can handle any cases of Boolean operations as long as the given solids are represented by CCSSs. Therefore there are no special or degenerated cases to

take care of. Although the new method is presented for CCSSs, the concept actually works for any subdivision scheme whose limit surfaces can be parameterized.

The remaining part of the chapter is arranged as follows. In section 1, a brief introduction is given. Some background and previous works related to this one are given in Section 2. The process of performing Boolean operations on solids represented by CCSSs is discussed in Section 3. The local voxelization technique is presented in Section 4. Error control is given in Section 5. Implementation issues and test cases are shown in Section 6. Concluding remarks are given in Section 7.

5.1 Introduction

Boolean operations are a natural way of constructing complex solid objects from simpler primitives. For example, the *Constructive Solid Geometry* (CSG) representation scheme allows users to define complex 3D solid objects by hierarchically combining simple geometric primitives using Boolean operations and affine transformations. However, for many applications CSG is not the most efficient approach. Another major representation scheme used in solid modeling is *boundary representation* (B-rep). But because higher order B-reps are needed for complicated objects, it is usually very difficult to find the intersecting curve analytically. In addition, care always has to be taken to handle special cases and degenerated cases [94]. Hence, accurate Boolean operations are usually not fast, nor robust, although excellent results have been achieved by some commercial solid modeling engines.

Voxelization of 3D objects has been studied and used for 3D object modeling and rendering for a while. With voxelization, it is actually very simple to get all the resulting voxels after Boolean operations because now Boolean operations become simple set operations. The difficult part is how to represent the resulting object properly and accurately when voxelization is used in the Boolean operation process. Traditionally, results of Boolean operations are represented as sets of voxels [104, 105] and special volumetric rendering algorithms are developed for visualizing Boolean operation results [89, 107]. The main disadvantage of this approach is that there is no continuous geometric representation for the resulting objects. Consequently, the results of Boolean operations cannot be scaled seamlessly or smoothly because of the nature of discretization.

In this chapter, we present a method for performing robust and error-controllable Boolean operations on free-form solids represented by Catmull-Clark subdivision surfaces (CCSSs). The given solids are voxelized so that Boolean operations can be performed more efficiently and robustly. However, the final results of Boolean operations in our method are still represented with a continuous geometric representation. This is achieved by performing Boolean operations subpatch by subpatch in 2D parameter space. Each subpatch is small enough to ensure the resulting voxels are either adjacent or overlapping. Consequently, connectivity of adjacent voxels can be easily constructed and the intersection curve can be easily identified. Because Boolean operations are performed subpatch by subpatch in 2D parameter space, our method can handle solids with arbitrary topology. There are no special cases or degenerated cases to take care of. Therefore our method is robust. Most importantly, error control is possible in our method. To make the Boolean results more accurate according to our error estimation formula, a secondary local voxelization can be performed for each pair of intersecting subpatches.

5.2 Background & Related Work

5.2.1 Subdivision Surfaces

Given a control mesh, a subdivision surface is generated by iteratively refining (subdividing) the control mesh to form new and finer control meshes. The refined control meshes converge to a limit surface called a subdivision surface. So a subdivision surface is determined by the given control mesh and the mesh refining (subdivision) process. Popular subdivision surfaces include Catmull-Clark subdivision surfaces (CCSSs) [1], Doo-Sabin subdivision surfaces [10] and Loop subdivision surfaces [3]. All these subdivision schemes can be considered as an algorithmic generalization of classical spline techniques enabling control meshes with arbitrary topology [1, 3, 10]. They provide easy access to globally smooth surfaces of arbitrary shape by iteratively applying simple refinement rules to the given control mesh. A sequence of meshes generated by this process quickly converges to a smooth limit surface. For most practical applications, the refined meshes are already sufficiently close to the smooth limit after only a few refinement steps.

Subdivision surfaces are by far the most general surface representation scheme. They include non-uniform B-spline and NURBS surfaces as special cases [15]. In this chapter we only consider performing Boolean operations on free-form solids represented by CCSSs. However, our approach can be used for any subdivision scheme whose parametrization is available.

5.2.2 Voxelization

Like 2D pixelization, voxelization of surfaces [90, 91] is a powerful technique for representing and modeling complex 3D objects. This is proved by many successful applications of volume graphics techniques in recently reported research work. For example, voxelization can be used for visualization of complex objects or scenes [57, 89, 107]. It can also be used for measuring integral properties of solids, such as mass, volume and surface area. Most importantly, it can be used for intersection curve calculation and, consequently, Boolean operations [57, 105]. For example, in [105], a series of Boolean operations are performed on objects represented by a CSG tree.

A good voxelization should meet three requirements in the voxelization process: *separability, accuracy,* and *minimality* [90, 91]. The first requirement demands similarity between the continuous space and the discrete space to be preserved and the resulting voxelization to be impenetratable since the given solid is closed and continuous. The second requirement ensures that the resulting voxelization is the most accurate discrete representation of the given solid according to some appropriate error metric. The third requirement requires that the voxelization does not contain voxels that, if removed, make no difference in terms of separability and accuracy. The mathematical definitions of these requirements can be found in [90, 91].

Note that a voxelization process does not render the voxels but merely generates a database of the discrete digitization of the continuous object [90]. Some previous voxelization methods use quad-trees to store the voxelization result [106]. This approach can save memory space but might take a long time when used for applications such as Boolean operations or intersection curve determination. Nevertheless, with cheap gigabyte memory chips becoming available, storage requirements are no longer a major issue in the design of a voxelization algorithm. People care more about the efficiency of the algorithm. Our new method stores the voxelization result directly in a *Cubic Frame Buffer* [90] for the purpose of fast operation.

5.2.3 Boolean Operations on Free-Form Solids

Performing Boolean operations is a classic problem in geometric modeling. Many approaches have been reported in the literature, such as [7, 64, 92, 98, 102, 104, 105, 106], to name a few. Currently, most solid modelers can support Boolean operations on solids composed of polyhedral models or quadric surfaces (like spheres, cylinders etc.). Over the last few years, modeling using free-form surfaces has become indispensable throughout the commercial CAD/CAM industry. However, the major bottleneck is in performing robust, efficient and accurate Boolean operations on free-form objects. The topology of a surface patch becomes quite complicated when a number of Boolean operations are performed and finding a convenient representation for these topologies has been a major challenge. As a result, some solid modelers [92] use polyhedral approximation to these surfaces and apply Boolean operations on these approximate polyhedral objects. Although these approaches seem simple, there are always some special cases or degenerated cases [94] that are difficult to take care of. Some modelers use point (or surfel) based approaches [106] to perform Boolean operations and quite good results are obtained. However, error control is difficult in such approaches. Zorin etc. proposed a method [7] to perform approximate Boolean operations on free-form solids represented by subdivision surfaces. The main contribution of their method is the algorithms that are able to generate a control mesh for a multiresolution surface approximating the Boolean results.

Most of the recent work in the literature on Boolean operations of curved models are focused on computing the surface intersection [93, 95, 97, 99, 101, 103]. However, the algebraic degree of the resulting curve can typically be very high (up to 324 for a pair of bicubic Bézier surfaces) [92] and the genus is also non-zero. Hence it is very difficult to represent the intersection curve analytically and the current methods are aimed at computing approximations to the intersection curve.

5.3 Performing Boolean Operations on 3D Free-Form Solids

Because we perform Boolean Operations on Free-Form Solids by voxelizing these solids, Boolean operations performed on three or more objects can be regarded as a series of Boolean operations performed on two objects. Therefore, here we only need to consider Boolean operations performed on two free-form solids A and B. As a result, only two cubic frame buffers are needed in the whole process, one for each object. The results of Boolean operations can share a cubic frame buffer with any of them. Once voxelization is done (See **Chapter 4**), a volume flooding (see **Chapter 4**) must be performed to mark the voxels located inside a given solid. After all these steps, there are three types of voxels in each cubic frame buffer: (1) *inside voxels*, (2) *boundary voxels* and (3) *outside voxels*.

Several possible Boolean operations may be specified by the users. However, the essential process is almost the same. Here we illustrate the process by assuming that the given Boolean operation is to find the intersection of two solid objects.

With voxelization, it is actually quite simple to get the resulting voxels for a Boolean operation. For example, the voxels left after an intersection operation are those located inside or on the boundary of both objects. The difficult part is how to represent the resulting part properly and accurately. Traditionally the results of Boolean operations are represented with just voxels. The main disadvantage of this method is that the results
cannot be scaled seamlessly because of the nature of discretization. In the following sections, we present an approach that represents the final result with a continuous geometric representation.

5.3.1 Boolean Operations based on Recursive 2D Parameter Space Subdivision and Voxelization

For a subpatch of $\mathbf{S}(u, v)$ of solid A defined on $[u_1, u_2] \times [v_1, v_2]$, we voxelize it one more time using the method discussed in **Chapter 4**. However, this time we do not write the voxels into A's cubic frame buffer, but look up the voxel values in both solid A and solid B's cubic frame buffers. Recall that we are performing an intersection operation of A and B. If all the voxel values of the whole subpatch in both cubic frame buffers are not *outside*, then this is a subpatch to keep. Subpatches of this type are called Ksubpatches (subpatches to be kept). If the voxel values of this subpatch are all outside in both A and B's cubic frame buffer, then this is a subpatch to discard. Subpatches of this type are called *D*-subpatches (subpatches to be discarded). Otherwise, i.e., if some of the voxel values are *inside*, *boundary* and some of the voxel values are *outside*, then this is a patch with some parts to keep and other parts to discard. Subpatches whose voxel values contain all of *inside*, *boundary* and *outside* are called *I-subpatches* (intersecting subpatches). For example, the rectangles shown in Figure 5.1 (a) are the parameter spaces of the resulting subpatches when the recursive voxelization process stops and the dashed polyline is part of the intersection curve of the two given solids in this patch's 2D parameter space. We can see that subpatch $A_1A_2A_4A_3$ in Figure 5.1 (a) is an I-subpatch. Note here all the marked (dark circles) adjacent points, when evaluated and voxelized, will be mapped to either the same voxel or adjacent voxels (see **Chapter 4**). For example, no voxel exists between voxels corresponding to parameter points A_1 and A_3 . Therefore, even though the intersection curve does not pass through A_1 or A_3 , the voxel corresponding to the intersection point I_1 will fall into the closest voxel corresponding to parameter point A_1 or A_3 . In this case, it falls into the voxel corresponding to A_1 .



Figure 5.1: Performing Boolean operations on 2D parameter space.

An intersecting voxel is a voxel whose voxel value is boundary in both cubic frame buffers. Hence it is very easy to find all the intersecting voxels, which compose the intersection curve (but at this moment we do not know how to connect these intersecting voxels yet. This will be explained shortly). For example, in Figure 5.1(a), parameter points A_1 and B_7 are intersecting voxels. Once all the intersecting voxels are identified, a continuous geometric representation for the Boolean operation result can be generated as follows.

K-subpatches and D-subpatches are easy to handle. They are either kept (for K-subpatches) or discarded (for D-subpatches) totally. For example, in Figure 5.1(b), $\mathbf{A_4 A_5 A_7 A_6}$ is a K-subpatch, hence $\mathbf{A_4 A_5 A_7 A_6}$ will be output wholly in the tessellation or rendering process. For an I-subpatch, one can determine which part of the subpatch to keep by traversing all the marked points attached to this subpatch. For example, for the subpatch $\mathbf{B_0 B_1 B_2 B_3 B_7}$ in Figure 5.1(a), after a traverse of the marked vertices, it is easy to see that the part to keep is the triangle $\mathbf{B_2 B_3 B_7}$. Hence $\mathbf{B_2 B_3 B_7}$ will be used in the tessellation and rendering process and another region of the subpatch $\mathbf{B_0 B_1 B_2 B_3 B_7}$ in Figure 5.1(a) will be discarded. Note here the intersection point $\mathbf{I_2}$, after voxelization, maps to the voxel $\mathbf{B_7}$. In Figure 5.1(b) the shaded part is the result after performing the Boolean operation in the 2D parameter space. Once we have the result of the Boolean operation in 2D parameter space, the 3D result can be easily obtained by directly evaluating and tessellating these shaded polygons. Note here we obtain not only the polygons,

but also their connectivity. Hence a mesh structure can be achieved in the above process. It is the mesh structure that we can consider as a one-piece representation of the results of Boolean operations. In this stage, we have a continuous geometric representation (the mesh) as well as a discrete voxel-based representation (the cubic frame buffer) for our resulting shape of Boolean operations. Because we now have both representations, a connected intersection curve can be easily constructed as well by picking intersecting voxels (from the discrete voxel-based representation) and traversing the mesh structure (information of the continuous geometric representation). For example, in Figure 5.1, the intersection curve (inside this patch) is $A_1A_4A_6B_2B_7B_8$.

The above voxelization process and Boolean operations guarantee that a shared boundary or vertex of patches or subpatches will be chopped, kept or discarded in exactly the same way, no matter which patch the operation is performed on. Therefore, in our approach, Boolean operations of free-form objects represented by CCSSs can be performed on the basis of individual patches.

5.3.2 Crack Prevention

Due to the fact that adjacent patches might be tessellated by quadrilaterals corresponding to subpatches from different levels of the midpoint subdivision process mentioned in the above section, cracks could occur between adjacent patches or subpatches. For instance, in Figure 6.3, the left patch $\mathbf{A}_1\mathbf{A}_2\mathbf{A}_5\mathbf{A}_6$ is approximated by one quadrilateral but the right patch is approximated by 7 quadrilaterals. Consider the boundary shared by the left patch and the right patch. On the left side, that boundary is a line segment defined by two vertices : \mathbf{A}_2 and \mathbf{A}_5 . But on the right side, the boundary is a polyline defined by four vertices : \mathbf{A}_2 , \mathbf{C}_4 , \mathbf{B}_4 , and \mathbf{A}_5 . They would not coincide unless \mathbf{C}_4 and \mathbf{B}_4 lie on the line segment defined by \mathbf{A}_2 and \mathbf{A}_5 . But that is usually not the case. Hence, cracks would appear between the left patch and the right patch.

Fortunately, cracks can be eliminated simply by replacing each boundary of a patch or subpatch with the one that contains all the evaluated points for that boundary. For example, in Figure 6.3, all the dotted lines should be replaced with the corresponding polylines. In particular, boundary A_2A_5 of patch $A_1A_2A_5A_6$ should be replaced with the polyline $A_2C_4B_4A_5$. As a result, polygon $A_1A_2A_5A_6$ is replaced with polygon $A_1A_2C_4B_4A_5A_6$ in the tessellation process. For rendering purposes this is fine because graphics systems like OpenGL can handle polygons with non-co-planar vertices and polygons with any number of sides. However, it should be pointed out that through a simple zigzag technique, triangulation of those polygons is actually a simple and very fast process. More details about the crack prevention problem are presented in **Chapter** 6.

Cracks could also occur if solids A and B are not connected properly in the intersecting area. For example, in Figure 5.1 (a), if intersection point \mathbf{I}_1 after evaluation and voxelization falls to the voxel corresponding to 2D parameter point \mathbf{A}_1 of solid A, and if \mathbf{I}_1 falls to the voxel corresponding to 2D parameter point $\mathbf{\bar{A}}_1$ of solid B, then after evaluation, $\mathbf{S}_A(\mathbf{A}_1)$ might not equal $\mathbf{S}_B(\mathbf{\bar{A}}_1)$ exactly. Therefore a crack occurs. To eliminate this kind of crack, we cannot use the exact 3D positions evaluated from 2D parameter points for intersection points. Instead we use the average of $\mathbf{S}_A(\mathbf{A}_1)$ and $\mathbf{S}_B(\mathbf{\bar{A}}_1)$ as the intersection point. In this way, solids A and B will have exactly the same intersection positions and intersection curve. As a result, solids A and B can be connected seamlessly. Note that for K-subpatches, their vertices will be evaluated directly from parameter points. Only intersection points of partially kept I-subpatches are approximated by the average of $\mathbf{S}_A(\mathbf{A}_1)$ and $\mathbf{S}_B(\mathbf{\bar{A}}_1)$.

5.4 Local Voxelization

The voxelization process presented in the above section is called a *global voxelization*, because it is performed for the entire object space. After all of the Boolean operations are performed, a fine scale voxelization, called a *local voxelization*, will also be performed. The goal of the local voxelization is to improve the accuracy of the I-subpatches. For example, in Figure 5.1(a), $\mathbf{A}_1\mathbf{A}_2\mathbf{A}_4$ is used to approximate the area of the I-subpatch $A_1A_2A_4A_3$ that should be kept. The accuracy of this approximation depends on the resolution of the global cubic frame buffer, which is always not high enough because of limited memory resources. However, we can do a secondary voxelization, which has lower resolution, but is only applied to a very small portion of the object space. As a result, high accuracy can still be achieved at the intersecting area.

The process and the approach used for a local voxelization are the same as for a global voxelization. The only difference is that they are applied to a different size of volumes in the object space. In order to perform local voxelization, information about which subpatches of solid A intersecting with which subpatches of solid B must be known first. This information is very difficult to obtain in previous voxelization-based methods. Fortunately, in our method, it can be readily obtained when performing the Boolean operations, as mentioned in Section 5.3.1. If we mark these intersecting subpatches of solids A and B during the keep-or-discard test process, we would know exactly which subpatches of solid A intersect with which subpatches of solid B. Once all intersecting subpatches are known, local voxelization can be directly performed for each pair of intersecting subpatches. For example, if subpatch p_1 of object A intersects with subpatches q_1 and q_2 of object B, then a local voxelization is performed on these 3 subpatches only. Their intersection curve is used to replace the intersection curve obtained using the global voxelization process. The local voxelization process is applied to every pair of intersecting subpatches of solids A and B. Consequently, a more accurate intersection curve can be computed. For instance, in Figure 5.1(a), the intersection curve A_4A_1 will be replaced with $\mathbf{V}_1 \mathbf{V}_2 \cdots \mathbf{V}_k$, k = 10, if \mathbf{V}_i , $i = 1 \cdots 10$ are the new intersecting voxels in the corresponding local cubic frame buffers and polygon $A_1A_2A_4V_1V_2\cdots V_k$ will be used in the tessellation and rendering process. Similar to global voxelization, only two local cubic frame buffers are needed for local voxelization. The local cubic frame buffers can be reused for each new pair of intersecting subpatches. Hence local voxelization does not require a lot of memory.

5.5 Error Control

Given an ϵ , the purpose of error control is to make sure the error of the resulting solid after performing Boolean operations using our method is less than ϵ to the one hundred percent accurate result. There are two kinds of error that might occur when our method is applied to perform Boolean operations among closed free-form solids represented by Catmull-Clark subdivision surfaces. They are discussed as follows.

The first inaccuracy possibly occurring using our method is the approximation of resulting solids with polygonal meshes. Because all obtained resulting solids are approximated with polygonal meshes, even although the approximating meshes are dense and are very close to the true surface, error inevitably occurs. However, the error caused by approximation of polygonal meshes can be accurately measured [56, 64]. Therefore, error control for this type of error is possible. The measurement of this kind of error is discussed in **Chapter 6**.

Another source that could introduce error in the result of the Boolean operations is the voxelization process. Both the global and the local voxelization can cause inaccuracy. The kind of error caused by voxelization is easy to estimate if the resolutions of cubic frame buffers are known. For example, if the cubic frame buffer resolution is $R_1 \times R_2 \times R_3$ and the object space is of size $X_1 \times X_2 \times X_3$, then we can see that each voxel is of size $\frac{X_1}{R_1} \times \frac{X_2}{R_2} \times \frac{X_2}{R_3}$. It is easy to see that the maximum error of voxelization is half the size of a voxel. If we perform local voxelization for every pair of intersecting subpatches, then global voxelization will not cause any error. Here we can also see why local voxelization can improve the accuracy dramatically. In local voxelization, because the size of the subpatches being voxelized is very small, even with a low resolution the voxel size is still very small.

Therefore, the overall error caused by polygonalization and voxelization is the sum of the errors caused by each of them. To make the error of the final Boolean operation results less than the given ϵ everywhere, the test condition in Eq. (6.5) has to be changed to the following form:

$$\begin{cases} \sqrt{d(\bar{u}, \bar{v})} + \sqrt{d(\hat{u}, \hat{v})} &\leq \epsilon/2\\ \text{size of each voxel} &\leq \epsilon \end{cases}$$
(5.1)

where (\hat{u}, \hat{v}) and (\bar{u}, \bar{v}) are defined the same way as in Eq. (6.5). The first equation in Eq. (5.1) ensures that the patch (or subpatch) and its approximating polygon are both located inside two quadrilaterals that are $\epsilon/2$ away. The second equation in Eq. (5.1) ensures that the error caused by voxelization is not bigger than $\epsilon/2$. Hence the total error in the whole process is guaranteed to be less than ϵ .

5.6 Test Results

The proposed approach has been implemented in C++ using OpenGL as the supporting graphics system on the Windows platform. Quite a few examples have been tested with the method described here. All the examples have extra-ordinary vertices. Some of the tested results are shown in Figure 5.2. The resolution of global voxelization is $512 \times 512 \times 512$ for all the test examples, and the error for all of them is set to 10^{-3} . The size of each example is normalized to [0, 1] before voxelization and Boolean operations are performed. Resolutions of the local voxelization process depend on error tolerance and the given meshes. Hence resolution of local voxelization is different for each of the examples shown in Figure 5.2. For example, resolution of local voxelization used for Figures 5.2(k) and 5.2(l) is $8 \times 8 \times 8$, while for Figures 5.2(g), 5.2(h), 5.2(i) and 5.2(j), the resolution are different, the overall error is the same in the final results. From Eq. (5.1) we can see that this difference is because intersecting subpatches in Figures 5.2(g), 5.2(h), 5.2(i) and 5.2(j) are bigger than Figures 5.2(k) and 5.2(l).

In Figure 5.2, all of the Difference and Intersection operations are performed on solids positioned exactly the same as in the Union operation so that we can easily tell if the results of the Boolean operations are correct within the given error tolerance. For example, Figures 5.2(j) and 5.2(g) are results of a Difference operation and a Union



Figure 5.2: Boolean Operations Performed on Solids Represented by CCSSs.

operation, respectively, on solids placed in the same positions. Similarly, Figure 5.2(i) corresponds to 5.2(h), 5.2(b) corresponds to 5.2(a), 5.2(d) corresponds to 5.2(c), 5.2(f) corresponds to 5.2(e) and 5.2(l) corresponds to 5.2(k).

5.7 Summary

A new method for performing robust and error-controllable Boolean operations on freeform solids represented with CCSSs is presented. The resulting solids after Boolean operations are represented with a continuous geometric representation, that is, our results after Boolean operations are one-piece representations of solid objects. They are represented with topologically correct mesh structure. Test results show that this approach leads to good results even for complicated solids with arbitrary topology.

The new method has several special properties. First, Boolean operations can be performed on 2D parameter spaces on the basis of individual patches. There is no need to take care of special cases or degenerated cases. Hence the method is robust. Second, although voxelization is performed to facilitate Boolean operations, the result of a Boolean operation in our method is still represented with a continuous geometric representation. Hence our Boolean operation results can be scaled seamlessly and smoothly. Third, error of Boolean operation results can be precisely estimated. According to the error estimation formula, a secondary local voxelization can be performed for intersecting subpatches only. Therefore, higher accuracy can be achieved. Finally, although the new method is presented for CCSSs, the concept actually works for any subdivision scheme whose limit surfaces can be parameterized.

Chapter 6

Adaptive Tessellation of General Catmull-Clark Subdivision Surfaces based on Inscribed Approximation

Catmull-Clark subdivision scheme provides a powerful method for building smooth and complex surfaces. But the number of faces in the uniformly refined meshes increases exponentially with respect to subdivision depth. Adaptive tessellation reduces the number of faces needed to yield a smooth approximation to the limit surface and, consequently, makes the rendering process more efficient.

In this chapter, we present a new adaptive tessellation method for general Catmull-Clark subdivision surfaces. The new adaptive tessellation method can be used to precisely measure error caused by polygonal approximation. For example the error control in our Boolean operation process presented in **Chapter 5** employs this method. The new adaptive tessellation method also can be used for significantly reducing face number of dense meshes with accurate error estimation. As a result our one-piece representation obtained from either interpolation (See **Chapter 2**) or performing Boolean operations (See **Chapter 5**), can be substantially simplified using the new adaptive tessellation method.

Different from previous control mesh refinement based approaches, which generate approximate meshes that usually do not interpolate the limit surface, the new method is based on direct evaluation of the limit surface to generate an inscribed polyhedron of the limit surface. With explicit evaluation of general Catmull-Clark subdivision surfaces becoming available, the new adaptive tessellation method can precisely measure error for every point of the limit surface. Hence, it has complete control of the accuracy of the tessellation result. Cracks are avoided by using a recursive color marking process to ensure that adjacent patches or subpatches use the same limit surface points in the construction of the shared boundary. The new method performs limit surface evaluation only at points that are needed for the final rendering process. Therefore it is very fast and memory efficient. The new method is presented for the general Catmull-Clark subdivision scheme. But it can be used for any subdivision scheme that has an explicit evaluation method for its limit surface.

The structure of this chapter is arranged as follows: We give a brief introduction in the Section 1. Some previous works related to this one is given in Section 2. A description of the basic idea of our adaptive tessellation technique is given in Section 3. The issue of crack elimination is discussed in Section 4. Two settings of patch flatness are discussed in Section 5. Algorithms of our technique are presented in Section 6. Test results are shown in Section 7. The concluding remarks are given in Section 8.

6.1 Introduction

Subdivision based evaluation process of a subdivision surface relies on performing repeated subdivision of the control mesh until the refined mesh is close enough to the limit surface (within some given tolerance). It is then possible to push the control points (mesh vertices) to their limit positions. But the number of faces in the uniformly refined meshes increases exponentially with the recursive steps of subdivision. See Figure 1.9(b) for an example where the control mesh of a Gargoyle is uniformly subdivided only twice and yet the resulting mesh is already quite dense. Hence, a good method for reducing the number of faces in the refined mesh while keeping the precision of the approximation is necessary. For instance, in Figure 1.9(c), 1.9(d), and 1.9(e), the same model is adaptively subdivided 4, 3 and 2 times, respectively. The resulting meshes have a higher or similar precision while the number of facets in the resulting meshes is much less than the uniform case. Such a method is important for both rendering and finite-element mesh generation. The criterion for rendering, however, is different from the criterion for finite-element mesh generation. In the first case, the number of sides of the mesh faces could be different while, in the second case, the mesh faces are either all triangles or all quadrilaterals. Figure 1.9(f) shows a triangulated result of Figure 1.9(e).

Research work for reducing the number of faces in a mesh has been done in several directions. *Mesh simplification* [29] is the most popular one over the past decade. It aims at removing some of the overly sampled vertices in a mesh and produces approximate meshes with various levels of detail. Another main method for reducing the number of faces in a mesh, called *adaptive tessellation*, is to apply adaptive or local refinement schemes to areas specified by a user or determined by an application. The resulting mesh should be crack-free and have the same limit surface as the uniformly refined mesh.

There are two possible approaches for adaptive tessellation of subdivision surfaces. One is a *mesh refinement based* approach. It approximates the limit surface by adaptively refining the control mesh of the surface. The resulting mesh usually does not interpolate points of the limit surface. The other one is a surface evaluation based approach. This approach approximates the limit surface by generating an inscribed polyhedron of the limit surface, with vertices of the polyhedron taken (evaluated) adaptively from the limit surface. The mesh refinement based approach needs a subdivision scheme, such as the Catmull-Clark method or the Doo-Sabin method, to refine the input mesh. Most methods proposed in the literature for adaptive tessellation of subdivision surfaces belong to this category. The second approach needs a parametrization/evaluation method for the limit surface. With the availability of direct evaluation methods of subdivision surfaces recently [17, 18, 20, 58], the second approach could be more appealing for adaptive tessellation of subdivision surface because of its simplicity in nature. Currently there is only one paper published in this category [32]. This paper works for parametrization that reproduces linear functions [39]. For more general parameterizations [17, 18, 20, 58], it does not work well.

In this chapter we will present a surface evaluation based approach for adaptive tessellation of subdivision surfaces. Our method is different from [32] in that our method works with any parametrization method and has a precise error estimate. The new approach is presented for the general Catmull-Clark subdivision surfaces [1], but it can be easily extended to work for any subdivision surface that has an exact evaluation method for its limit surface.

6.2 Previous Work

A number of adaptive tessellation methods for subdivision surfaces have been proposed [40, 30, 31, 32, 35, 36]. Most of them are mesh refinement based, i.e., approximating the limit surface by adaptively refining the control mesh. This approach requires the assignment of a subdivision depth to each region of the surface first. In [40], a subdivision depth is calculated for each patch of the given Catmull-Clark surface with respect to a given error tolerance ϵ . In [30], a subdivision depth is estimated for each vertex of the given Catmull-Clark surface by considering factors such as curvature, visibility, membership to the silhouette, and projected size of the patch. The approach used in [40] is error controllable. An error controllable approach for Loop surface is proposed in [32], which calculates a subdivision depth for each patch of a Loop surface by estimating the distance between two bounding linear functions for each component of the 3D representation.

Several other adaptive tessellation schemes have been presented as well [36, 35, 31]. In [31], two methods of adaptive tessellation for triangular meshes are proposed. The adaptive tessellation process for each patch is based on angles between its normal and normals of adjacent faces. A set of new error metrics tailored to the particular needs of surfaces with sharp creases is introduced in [35].

In addition to various adaptive tessellation schemes, there are also applications of these techniques. D. Rose et al. used adaptive tessellation method to render terrain [38] and K. Müller et al. combined ray tracing with adaptive subdivision surfaces to generate some realistic scenes [34]. Adaptive tessellation is such an important technique that an API has been designed for its general usage [37]. Actually hardware implementation of this technique has been reported recently as well [33]. A problem with the mesh-refinement-based, adaptive tessellation techniques is the so called gap-prevention requirement. Because the number of new vertices generated on each boundary of the control mesh depends on the subdivision depth, gaps (or, cracks) could occur between the control meshes of adjacent patches if these patches are assigned different subdivision depths. Hence, each mesh-refinement-based adaptive tessellation method needs some special mechanism to eliminate gaps. This is usually done by performing additional subdivision or splitting steps on the patch with lower subdivision depth. As a result, many unnecessary polygons are generated in the tessellation process. In this paper, we will adaptively tessellate a subdivision surface by taking points from the limit surface to form an inscribed polyhedron of the limit surface, instead of refining the control mesh. Our method simplifies the process of gap detecting and elimination. It does not need to perform extra or unnecessary evaluations either.

6.3 Basic Idea



Figure 6.1: Inscribed and Circumscribed Approximation.

6.3.1 Inscribed Approximation

One way to approximate a curve (surface) is to use its control polygon (mesh) as the approximating polyline (polyhedron). For instance, in Figure 6.1(a), at the top are a cubic Bézier curve and its control polygon. For a better approximation, we can refine the control polygon using midpoint subdivision. The solid polyline at the bottom of Figure 6.1(a) is the approximating control polygon after one refinement. This method relies on

performing iterative refinement of the control polygon or control mesh to approximate the limit curve or surface. Because this method approximates the limit shape from control polygon or control mesh "outside" the limit shape, we call this method *circumscribed approximation*.

Another possible method is *inscribed approximation*. Instead of approximating the limit curve (surface) by performing subdivision on its control polygon (mesh), one can approximate the limit curve (surface) by inscribed polygons (polyhedra) whose vertices are taken from the limit curve (surface) directly. The easiest approach to get vertices of the inscribed polygons (polyhedra) is to perform uniform midpoint subdivision on the parameter space and use the evaluated vertices of the resulting subsegments (subpatches) as vertices of the inscribed polylines (polyhedra). For instance, in Figure 6.1(b), at the top are a cubic Bézier curve and its approximating polygon with vertices evaluated at parameter points 0, 1/2 and 1. Similarly, the solid polygon at the bottom of Figure 6.1(b) is an approximating polygon with vertices evaluated at five parameter points.

Because inscribed approximation uses points directly located on the limit curve or surface, in most cases, it has faster convergent rate than the circumscribed approximation. As one can see clearly from Figure 6.1 that the inscribed polygon at the bottom of Figure 6.1(b) is closer to the limit curve than the circumscribed polygon shown at the bottom of Figure 6.1(a) even though the inscribed polygon actually has less segments than the circumscribed polygon.

Inscribed approximation requires explicit evaluation of a CCSS Patch. Several approaches [17, 18, 20, 58] have been presented for exact evaluation of an extraordinary patch at any parameter point (u, v). In our implementation, we follow the parametrization technique presented in [58], because this method is numerically stable, employs less eigen basis functions, and can be used for the evaluation of 3D position and normal vector of any point in the limit surface exactly and explicitly. Some most related results of [58] are presented in **Chapter 2**.

However, the problem with both Inscribed or circumscribed approximation approaches is that, with uniform subdivision, no matter it is performed on the control mesh or the parameter space, one would get unnecessarily small and dense polygons for surface patches that are already flat enough and, consequently, slow down the rendering process. To speed up the rendering process, a flat surface patch should not be tessellated as densely as a surface patch with big curvature. The adaptive tessellation process of a surface patch should be performed based on the flatness of the patch. This leads to our adaptive inscribed approximation.

6.3.2 Adaptive Inscribed Approximation

For a patch of $\mathbf{S}(u, v)$ defined on $u_1 \leq u \leq u_2$ and $v_1 \leq v \leq v_2$, we try to approximate it with the quadrilateral formed by its four vertices $\mathbf{V}_1 = \mathbf{S}(u_1, v_1)$, $\mathbf{V}_2 = \mathbf{S}(u_2, v_1)$, $\mathbf{V}_3 = \mathbf{S}(u_2, v_2)$ and $\mathbf{V}_4 = \mathbf{S}(u_1, v_2)$. If the distance (to be defined below) between the patch and its corresponding quadrilateral is small enough (to be defined below), then the patch is considered *flat* enough and will be (for now) replaced with the corresponding quadrilateral in the tessellation process. Otherwise, we perform a midpoint subdivision on the parameter space by setting

$$u_{12} = \frac{u_1 + u_2}{2}$$
 and $v_{12} = \frac{v_1 + v_2}{2}$

to get four subpatches: $[u_1, u_{12}] \times [v_1, v_{12}]$, $[u_{12}, u_2] \times [v_1, v_{12}]$, $[u_{12}, u_2] \times [v_{12}, v_2]$, $[u_1, u_{12}] \times [v_{12}, v_2]$, and repeat the *flatness* testing process on each of the subpatches. The process is recursively repeated until the distance between all the subpatches and their corresponding quadrilaterals are small enough. The vertices of the resulting subpatches are then used as vertices of the inscribed polyhedron of the limit surface. For instance, if the four rectangles in Figure 6.2(a) are the parameter spaces of four adjacent patches of $\mathbf{S}(u, v)$, and if the rectangles shown in Figure 6.2(b) are the parameter spaces of the resulting subpatches when the above flatness testing process stops, then the limit surface



Figure 6.2: Basic idea of the construction of an inscribed polyhedron.

will be evaluated at the points marked with small solid circles to form vertices of the inscribed polyhedron of the limit surface.

In the above flatness testing process, to measure the difference between a patch (or subpatch) and its corresponding quadrilateral, we need to parameterize the quadrilateral as well. The quadrilateral can be parameterized as follows:

$$\mathbf{Q}(u,v) = \frac{v_2 - v}{v_2 - v_1} \left(\frac{u_2 - u}{u_2 - u_1} \mathbf{V}_1 + \frac{u - u_1}{u_2 - u_1} \mathbf{V}_2\right) + \frac{v - v_1}{v_2 - v_1} \left(\frac{u_2 - u}{u_2 - u_1} \mathbf{V}_4 + \frac{u - u_1}{u_2 - u_1} \mathbf{V}_3\right)$$
(6.1)

where $u_1 \leq u \leq u_2$, $v_1 \leq v \leq v_2$. The *difference* between the patch (or subpatch) and the corresponding quadrilateral at (u, v) is defined as

$$d(u,v) = \| \mathbf{Q}(u,v) - \mathbf{S}(u,v) \|^2 = (\mathbf{Q}(u,v) - \mathbf{S}(u,v)) \cdot (\mathbf{Q}(u,v) - \mathbf{S}(u,v))^T \quad (6.2)$$

where $\|\cdot\|$ is the second norm and \mathbf{A}^T is the transpose of \mathbf{A} . The *distance* between the patch (or subpatch) and the corresponding quadrilateral is the maximum of all the differences:

$$D = \max\{ \sqrt{d(u,v)} \mid (u,v) \in [u_1,u_2] \times [v_1,v_2] \}$$

To measure the distance between a patch (or subpatch) and the corresponding quadrilateral, we only need to measure the norms of all local minima and maxima of d(u, v). Note that $\mathbf{Q}(u, v)$ and $\mathbf{S}(u, v)$ are both C^1 -continuous, and $d(\mathbf{V}_1)$, $d(\mathbf{V}_2)$, $d(\mathbf{V}_3)$ and $d(\mathbf{V}_4)$ are equal to 0. Therefore, by Mean Value Theorem, the local minima and maxima must lie either inside $[u_1, u_2] \times [v_1, v_2]$ or on the four boundary curves. In other words, they must satisfy at least one of the following three conditions:

$$\begin{cases} \frac{\partial d(u,v)}{\partial u} = 0\\ v = v_1 \text{ or } v = v_2\\ u_1 \le u \le u_2 \end{cases} \qquad \begin{cases} \frac{\partial d(u,v)}{\partial v} = 0\\ u = u_1 \text{ or } u = u_2\\ v_1 \le v \le v_2 \end{cases} \qquad \begin{cases} \frac{\partial d(u,v)}{\partial u} = 0\\ \frac{\partial d(u,v)}{\partial v} = 0\\ (u,v) \in (u_1,u_2) \times (v_1,v_2) \end{cases}$$
(6.3)

For a patch (or subpatch) that is not adjacent to an extraordinary point (i.e., $(u_1, v_1) \neq (0, 0)$), m is fixed and known $(m(u, v) = min\{\lceil log_{\frac{1}{2}}u \rceil, \lceil log_{\frac{1}{2}}v \rceil\})$. Hence Eq. (6.3) can be solved explicitly. With the valid solutions, we can find the difference for each of them using Eq. (6.2). Suppose the one with the biggest difference is (\hat{u}, \hat{v}) . Then (\hat{u}, \hat{v}) is also the point with the biggest distance between the patch (or subpatch) and its corresponding quadrilateral. We consider the patch (or subpatch) to be *flat* enough if

$$D = \sqrt{d(\hat{u}, \hat{v})} \leq \epsilon$$
(6.4)

where ϵ is a given error tolerance. In such a case, the patch (or subpatch) is replaced with the corresponding quadrilateral in the tessellation process. If a patch (or subpatch) is not flat enough yet, i.e., if Eq. (6.4) does not hold, we perform a midpoint subdivision on the patch (or subpatch) to get four new subpatches and repeat the flatness testing process for each of the new subpatches. This process is recursively repeated until all the subpatches satisfy Eq. (6.4).

For a patch (or subpatch) that is adjacent to an extraordinary point (i.e. $(u_1, v_1) = (0, 0)$ in Eq. (6.3)), m is not fixed and m tends to ∞ (see Figure 2.2). As a result, Eq. (6.3) can not be solved explicitly. One way to resolve this problem is to use nonlinear numerical method to solve these equations. But numerical approach cannot guarantee the error is less than ϵ everywhere. For precise error control, a better choice is needed. In the following, an alternative method is given for that purpose.

Eq. (2.7) shows that $\mathbf{S}(u, v)$ and $\mathbf{Q}(u, v)$ both converge to $\mathbf{S}(0, 0)$ when $(u, v) \rightarrow (0, 0)$. Hence, for any given error tolerance ϵ , there exists an integer m_{ϵ} such that if $m \geq m_{\epsilon}$, then the distance between $\mathbf{S}(u, v)$ and $\mathbf{S}(0, 0)$ is smaller than $\epsilon/2$ for any $(u, v) \in$

 $[0, 1/2^m] \times [0, 1/2^m]$, and so is the distance between $\mathbf{Q}(u, v)$ and $\mathbf{S}(0, 0)$. Consequently, when $(u, v) \in [0, 1/2^m] \times [0, 1/2^m]$, the distance between $\mathbf{S}(u, v)$ and $\mathbf{Q}(u, v)$ is smaller than ϵ . The value of m_{ϵ} , in most of the cases, is a relatively small number and can be explicitly calculated. In next subsection, we will show how to calculate m_{ϵ} .

For other regions of the unit square with $\lceil \log_{\frac{1}{2}} u_2 \rceil \leq m < m_{\epsilon}$ (see Figure 2.2), Eq. (6.3) can be used directly to find the difference between $\mathbf{S}(u, v)$ and $\mathbf{Q}(u, v)$ for any fixed $m \in (\lceil \log_{\frac{1}{2}} u_2 \rceil, m_{\epsilon})$. Therefore, by combining all these differences, we have the distance between the given extra-ordinary patch (or subpatch) and the corresponding quadrilateral. If this distance is smaller than ϵ , we consider the given extra-ordinary patch (or subpatch) to be flat, and use the distance quadrilateral to replace the extra-ordinary patch (or subpatch) in the tessellation process. Otherwise, repeatedly subdivide the patch (or subpatch) and perform flatness testing on the resulting subpatches until all the subpatches satisfy Eq. (6.4).

6.3.3 Calculating m_{ϵ}

For a given $\epsilon > 0$, an integer k_{ϵ} will first be computed so that if k is bigger than k_{ϵ} , then the subpatch of $\mathbf{S}(u, v)$ with $0 \le u, v \le 1/2^k$ is contained in a sphere with center $\mathbf{S}(0, 0)$ and diameter ϵ (called an ϵ -sphere). A subpatch is contained in an ϵ -sphere if all points of the subpatch are $\epsilon/2$ away from $\mathbf{S}(0, 0)$.

To find such k_{ϵ} , we need a few properties from [58]. Recall that an extra-ordinary patch $\mathbf{S}(u, v)$ can be expressed as

$$\mathbf{S}(u,v) = \sum_{j=0}^{n+5} \Phi_{b,j}(u,v) \cdot G$$

where $\Phi_{b,j}$ are eigen basis functions defined in [58] and G is the vector of control points of **S**. The eigen basis functions satisfy the scaling relation [17, 58], i.e.,

$$\Phi_{b,j}(u/2^k, v/2^k) = \lambda_j^k \Phi_{b,j}(u, v)$$

for any positive integer k, where λ_j are eigen values of the Catmull-Clark subdivision

matrix [58]. The eigen values are indexed so that

$$1 = \lambda_{n+1} > \lambda_2 \ge \lambda_i > 0$$

where $0 \le i \le n+5$ and $i \ne n+1$. Also recall that $\Phi_{b,j}(0,0) = 0$ when $j \ne n+1$, and $\Phi_{b,n+1}(u,v)$ is a constant vector, its value is independent of (u,v) [58]. Hence,

$$(\Phi_{b,n+1}(u,v) - \Phi_{b,n+1}(u',v')) \cdot G_r = 0$$

for any (u, v) and (u', v') where $r \in \{x, y, z\}$ and G_r is the x-, y- or z-component of G.

Hence for any $1/2 \le u \le 1$ or $1/2 \le v \le 1$, and for any k we have

$$\begin{aligned} |\mathbf{S}_{r}(u/2^{k}, v/2^{k}) - \mathbf{S}_{r}(0, 0)| &= |\sum_{j=0}^{n+5} (\lambda_{j}^{k} \Phi_{b,j}(u, v) - \Phi_{b,j}(0, 0)) \cdot G_{r}| \\ &\leq \sum_{j \neq n+1} \lambda_{j}^{k} |(\Phi_{b,j}(u, v) \cdot G_{r}| < \lambda_{2}^{k} \sum_{j \neq n+1} |(\Phi_{b,j}(u, v) \cdot G_{r}| \end{aligned}$$

Similarly, the three conditions in Eq. (6.3) can be used to find the maxima of $|(\Phi_{b,j}(u, v) \cdot G_r|$ for any j. Note that because here $(u, v) \notin [0, 1/2] \times [0, 1/2]$, the corresponding m is equal to 1 (See figure 2.2). Hence we can easily find the maximum in its domain $\{(u, v)|1/2 \leq u \leq 1 \text{ or } 1/2 \leq v \leq 1\}$. Let the maximum of $|(\Phi_{b,j}(u, v) \cdot G_r)|$ be F_{rj} and $F_r = \sum_{j \neq n+1} F_{rj}$. Then, for any k > 0 we have

$$|\mathbf{S}_r(u/2^k, v/2^k) - \mathbf{S}_r(0, 0)| \le \lambda_2^k F_r.$$

Therefore if $(\lambda_2^k F_x)^2 + (\lambda_2^k F_y)^2 + (\lambda_2^k F_z)^2 \le (\epsilon/2)^2$, we have

$$\| \mathbf{S}(u/2^k, v/2^k) - \mathbf{S}(0, 0) \| \le \epsilon/2.$$

If we define k_{ϵ} as follows

$$k_{\epsilon} = \left\lceil \log_{\lambda_2} \frac{\epsilon}{2\sqrt{F_x^2 + F_y^2 + F_z^2}} \right\rceil$$

then it is easy to see that when $k \ge k_{\epsilon}$, the subpatch $\mathbf{S}(u, v)$ with $(u, v) \in [0, 1/2^k] \times [0, 1/2^k]$ is inside an ϵ -sphere whose center is $\mathbf{S}(0, 0)$.

In addition, $\mathbf{S}(0,0)$ is a fixed point and has an explicit expression for any patch (see Eq. 2.7), and $\mathbf{Q}(u, v)$ also has an explicit parametrization (See Eq. (6.1)). Hence, similarly, by using the method of Eq. (6.3), it is easy to find an integer \tilde{k}_{ϵ} , such that for any given

 $\epsilon > 0$, when $k \ge \widetilde{k}_{\epsilon}$, we have $\| \mathbf{Q}(u, v) - \mathbf{S}(0, 0) \| \le \epsilon/2$, where $(u, v) \in [0, 1/2^k] \times [0, 1/2^k]$. Once we have k_{ϵ} and \widetilde{k}_{ϵ} , simply set m_{ϵ} as the maximum of k_{ϵ} and \widetilde{k}_{ϵ} .

$$m_{\epsilon} = \max\{k_{\epsilon}, k_{\epsilon}\}$$

With this m_{ϵ} , it is easy to see that when $m \ge m_{\epsilon}$, we have $|| \mathbf{S}(u, v) - \mathbf{Q}(u, v) || \le \epsilon$, where $(u, v) \in [0, 1/2^m] \times [0, 1/2^m]$.

6.4 Crack Elimination



Figure 6.3: Crack elimination.

Due to the fact that adjacent patches might be approximated by quadrilaterals corresponding to subpatches from different levels of the midpoint subdivision process, cracks could occur between adjacent patches. For instance, in Figure 6.3, the left patch $A_1A_2A_5A_6$ is approximated by one quadrilateral but the right patch is approximated by 7 quadrilaterals. Consider the boundary shared by the left patch and the right patch. On the left side, that boundary is a line segment defined by two vertices : A_2 and A_5 . But on the right side, the boundary is a polyline defined by four vertices : A_2 , C_4 , B_4 , and A_5 . They would not coincide unless C_4 and B_4 lie on the line segment defined by A_2 and A_5 . But that usually is not the case. Hence, cracks would appear between the left patch and the right patch.

Fortunately Cracks can be eliminated simply by replacing each boundary of a patch or subpatch with the one that contains all the evaluated points for that boundary. For example, in Figure 6.3, all the dashed lines should be replaced with the corresponding polylines. In particular, boundary A_2A_5 of patch $A_1A_2A_5A_6$ should be replaced with the polyline $A_2C_4B_4A_5$. As a result, polygon $A_1A_2A_5A_6$ is replaced with polygon $A_1A_2C_4B_4A_5A_6$ in the tessellation process. For rendering purpose this is fine because graphics systems like OpenGL can handle polygons with non-co-planar vertices and polygons with any number of sides. However, it should be pointed out that through a simple zigzag technique, triangulation of those polygons is actually a simple and very fast process.

A potential problem with this process is the new polygons generated by the crack elimination algorithm might not satisfy the flatness requirement. To ensure the flatness requirement is satisfied everywhere when the above crack elimination method is used, we need to change the test condition in Eq. (6.4) to the following one:

$$\sqrt{d(\bar{u},\bar{v})} + \sqrt{d(\hat{u},\hat{v})} \le \epsilon \tag{6.5}$$

where (\hat{u}, \hat{v}) and (\bar{u}, \bar{v}) are solutions of Eq. (6.3) and they satisfy the following conditions:

- Among all the solutions of Eq. (6.3) that are located on one side of Q(u, v), i.e. solutions that satisfy (Q − S) · ((V₁ − V₃) × (V₂ − V₄)) ≥ 0, d(û, î) is the biggest. If there does not exist any solution such that this condition holds, then d(û, î) is set to 0;
- Among all the solutions of Eq. (6.3) that are located on the other side of Q(u, v),
 i.e. solutions that satisfy (Q − S) · ((V₁ − V₃) × (V₂ − V₄)) < 0, d(ū, v̄) is the biggest. If there does not exist any solution such this condition holds, then d(ū, v̄) is set to 0.

From the definition of (\hat{u}, \hat{v}) and (\bar{u}, \bar{v}) , we can see that satisfying Eq. (6.5) means that the patch being tested is located between two quadrilaterals that are ϵ away.

Note that all the evaluated points lie on the limit surface. Hence, for instance, in Figure 6.3, points A_2, C_4, B_4 and A_5 of patch $A_2A_3A_4A_5$ are also points of patch

 $A_1A_2A_5A_6$. With the new test condition in Eq. (6.5), we know that a patch or subpatch is flat enough if it is located between two quadrilaterals that are ϵ away. Because boundary points A_2 , C_4 , B_4 and A_5 are on the limit surface, they must be located between two quadrilaterals that are ϵ away. So is the polygon $A_1A_2C_4B_4A_5A_6$. Now the patch (or subpatch) and its approximating polygon are both located inside two quadrilaterals that are ϵ away. Hence the overall error between the patch (or subpatch) and its approximating polygon is guaranteed to be smaller than ϵ .

In previous methods for adaptive tessellation of subdivision surfaces [40, 30, 31, 35], the most difficult part is crack prevention. Yet in our method, this part becomes the simplest part to handle and implement. The resulting surface is error controllable and guaranteed to be crack free.

6.5 Degree of Flatness

Just like numerical errors have two different settings, the flatness of a patch, which can be viewed as a numerical error from the approximation point of view, has two different aspects as well, depending on if the flatness is considered in the absolute sense or relative sense. The flatness of a patch is called the *absolute flatness* (AF) if the patch is not transformed in any way. In that case, the value of ϵ in Eq. (6.4) and (6.5) is set to whatever precision the flatness of the patch is supposed to meet. AF should be considered for operations that work on physical size of an object such as machining or prototyping.

For operations that do not work on the physical size of an object, such as the rendering process, we need a flatness that does not depends on the physical size of a patch. Such a flatness must be Affine transformation invariant to be a constant for any transformed version of the patch. Such a flatness is called the *relative flatness* of the patch. More specifically, if \mathbf{Q} is the corresponding quadrilateral of patch \mathbf{S} , the relative flatness (RF) of \mathbf{S} with respect to \mathbf{Q} is defined as follows:

$$\mathrm{RF} = \frac{d}{\max\{D_1, D_2\}}$$

where d is the maximal distance from S to Q, and D_1 , D_2 are lengths of the diagonal

lines of \mathbf{Q} . It is easy to see that RF defined this way is Affine transformation invariant. Note that when D_1 and D_2 are fixed, smaller RE means smaller d. Hence, RE indeed measures the flatness of a patch. The difference between RF and AF is that RF measures the flatness of a patch in a global sense while AF measures flatness of a patch in a local sense. Therefore, RF is more suitable for operations that have data sets of various sizes but with a constant size display area such as the rendering process. Using RF is also good for adaptive tessellation process because it has the advantage of keeping the number of polygons low in the tessellation process.

6.6 Algorithms of Adaptive Tessellation

In this section, we discuss the important steps of the adaptive tessellation process and present the corresponding algorithms.

6.6.1 Global Index ID

Currently, all the subdivision surface parametrization and evaluation techniques are patch based [17, 20, 58]. Hence, no matter which method is used in the adaptive tessellation process, a patch cannot see vertices evaluated by other patches from its own (local) structure even though the vertices are on its own boundary. For example, in Figure 6.3, vertices C_4 and B_4 are on the shared boundary of patches $A_1A_2A_5A_6$ and $A_2A_3A_4A_5$. But patch $A_1A_2A_5A_6$ can not see these vertices from its own structure because these vertices are not evaluated by this patch. To make adjacent patches visible to each other and to make subsequent crack elimination work easier, one should assign a *global index* ID to each evaluated vertex so that

- all the evaluated vertices with the same 3D position have the same index *ID*;
- the index ID's are sorted in v and then in u, i.e., if $(u_i, v_i) \ge (u_j, v_j)$, then $ID_i \ge ID_j$, unless ID_i or ID_j has been used in previous patch evaluation.

With a global index ID, it is easy to do crack prevention even with a patch based approach. Actually, subsequent processing can all be done with a patch based approach

and still performed efficiently. For example, in Figure 6.3, patch $A_1A_2A_5A_6$ can see both C_4 and B_4 even though they are not evaluated by this patch. In the subsequent rendering process, the patch simply output all the marked vertices (to be defined below) on its boundary that it can see to form a polygon for the rendering purpose, i.e., $A_1A_2C_4B_4A_5A_6$.

6.6.2 Adaptive Marking

The purpose of *adaptive marking* is to mark those points in uv space where the limit surface should be evaluated. With the help of the global index ID, this step can be done on an individual patch basis. Initially, all (u, v) points are marked white. If surface evaluation should be performed at a point and the resulting vertex is needed in the tessellation process, then that point is marked in black. This process can be easily implemented as a recursive function. A pseudo code for this step is given below.

AdaptiveMarking($\mathbf{P}, u_1, u_2, v_1, v_2$)

- 1. Evaluate ($\mathbf{P}, u_1, u_2, v_1, v_2$),
- 2. AssignGlobalID($\mathbf{P}, u_1, u_2, v_1, v_2$),
- 3. if (FlatEnough($\mathbf{P}, u_1, u_2, v_1, v_2$))
- 4. MarkBlack($\mathbf{P}, u_1, u_2, v_1, v_2$)
- 5. else
- 6. $u_{12} = (u_1 + u_2)/2$
- 7. $v_{12} = (v_1 + v_2)/2$
- 8. AdaptiveMarking($\mathbf{P}, u_1, u_{12}, v_1, v_{12}$)
- 9. AdaptiveMarking($\mathbf{P}, u_{12}, u_2, v_1, v_{12}$)
- 10. AdaptiveMarking($\mathbf{P}, u_{12}, u_2, v_{12}, v_2$)
- 11. AdaptiveMarking($\mathbf{P}, u_1, u_{12}, v_{12}, v_2$)

This routine adaptively marks points in the parameter space of patch \mathbf{P} . Function 'Evaluate' evaluates limit surface at the four corners of patch or subpatch \mathbf{P} defined on $[u_1, u_2] \times [v_1, v_2]$. Function 'FlatEnough' uses the method given in section 6.3 and Eq. (6.4) to tell if a patch or subpatch is flat enough. Function 'MarkBlack' marks the four corners of patch or subpatch **P** defined on $[u_1, u_2] \times [v_1, v_2]$ in black. All the marked corner points will be used in the tessellation process.

6.6.3 Adaptive Rendering a Single Patch

The purpose of this step is to render the limit surface with as few polygons as possible, while preventing the occurrence of any cracks. Note that the limit surface will be evaluated only at the points marked in black, and the resulting vertices are the only vertices that will be used in the rendering process. To avoid cracks, each marked points must be rendered properly. Hence special care must be taken on adjacent patches or subpatches. With the help of *adaptive marking*, this process can easily be implemented as a recursive function as well. A pseudo code for this step is given below.

AdaptiveRendering($\mathbf{P}, u_1, u_2, v_1, v_2$)

- 1. if (NoMarkedPointInside($\mathbf{P}, u_1, u_2, v_1, v_2$))
- 2. RenderPolygon($\mathbf{P}, u_1, u_2, v_1, v_2$)
- 3. else
- 4. $u_{12} = (u_1 + u_2)/2$
- 5. $v_{12} = (v_1 + v_2)/2$
- 6. AdaptiveRendering($\mathbf{P}, u_1, u_{12}, v_1, v_{12}$)
- 7. AdaptiveRendering($\mathbf{P}, u_{12}, u_2, v_1, v_{12}$)
- 8. AdaptiveRendering($\mathbf{P}, u_{12}, u_2, v_{12}, v_2$)
- 9. AdaptiveRendering($\mathbf{P}, u_1, u_{12}, v_{12}, v_2$)

This routine adaptively renders marked points in patch or subpatch **P**. Function 'NoMarkedPointInside' tests if none of the points inside $[u_1, u_2] \times [v_1, v_2]$, excluding the boundary points, are marked. If all the interior points are in white (i.e. not marked), it returns TRUE. Function 'RenderPolygon' is defined as follows.

RenderPolygon($\mathbf{P}, u_1, u_2, v_1, v_2$)

- 1. glBegin(RenderModel)
- 2. Output all the marked points between
- 3. $(u_1, v_1) \to (u_2, v_1)$
- 4. $(u_2, v_1) \to (u_2, v_2)$
- 5. $(u_2, v_2) \to (u_1, v_2)$
- 6. $(u_1, v_2) \to (u_1, v_1)$
- 7. glEnd()

6.6.4 Adaptive Rendering a CCSS

The overall algorithm for rendering a general CCSS is given below. The algorithm takes the control mesh of the surface as input.

CCSSAdaptiveRendering(Mesh M)

- 1. for each face \mathbf{P} in \mathbf{M}
- 2. AdaptiveMarking(\mathbf{P} ,0,1,0,1)
- 3. for each face \mathbf{P} in \mathbf{M}
- 4. AdaptiveRendering($\mathbf{P}, 0, 1, 0, 1$)

6.7 Implementation and Test Results

The proposed approach has been implemented in C++ using OpenGL as the supporting graphics system on the Windows platform. Quite a few examples have been tested with the method described here. Some of the tested results are shown in Figures 1.9, 6.4 and 6.5. We also summarize those tested results in Table 6.1. The column underneath

Figure	Object	AUT	polygons	A/U Ratio	Depth	Error	RF
Figure 1.9(b)	Gargoyle	U	16384	100.00%	2	0.0055	12%
Figure 1.9(c)	Gargoyle	А	14311	5.46%	4	0.0030	6%
Figure 1.9(d)	Gargoyle	А	5224	7.97%	3	0.0045	9%
Figure 1.9(e)	Gargoyle	А	2500	15.26%	2	0.0055	12%
Figure 1.9(f)	Gargoyle	Т	6139	37.47%	2	0.0055	12%
Figure 6.4(a)	Bunny	U	65536	100.00%	3	0.0008	3%
Figure 6.4(b)	Bunny	А	32894	12.55%	4	0.0001	1%
Figure $6.4(c)$	Bunny	А	9181	14.01%	3	0.0008	3%
Figure $6.4(d)$	Bunny	А	3412	20.82%	2	0.0010	5%
Figure 6.4(e)	Bunny	Т	7697	46.98%	2	0.0010	5%
Figure 6.4(f)	Venus	U	65536	100.00%	2	0.00095	8%
Figure $6.4(g)$	Venus	А	29830	2.84%	4	0.00015	3%
Figure 6.4(h)	Venus	А	21841	2.08%	4	0.00035	4%
Figure 6.4(i)	Venus	А	9763	3.72%	3	0.00060	6%
Figure 6.4(j)	Venus	А	6178	9.43%	2	0.00095	8%
Figure 6.5(a)	Rockerarm	U	90624	100.00%	4	1.2	3%
Figure $6.5(b)$	Rockerarm	А	36045	9.94%	5	0.85	1%
Figure $6.5(c)$	Rockerarm	А	10950	3.02%	5	1.0	2%
Figure $6.5(d)$	Rockerarm	А	5787	6.39%	4	1.2	3%
Figure 6.5(e)	Rockerarm	А	2901	12.80%	3	1.5	5%
Figure $6.5(f)$	Rockerarm	Т	6917	30.53%	3	1.5	5%
Figure $6.5(g)$	Beethoven	U	65536	100.00%	2	0.041	10%
Figure 6.5(h)	Beethoven	А	20893	1.99%	4	0.006	4%
Figure 6.5(i)	Beethoven	А	15622	1.48%	4	0.026	6%
Figure 6.5(j)	Beethoven	А	7741	2.95%	3	0.035	8%
Figure 6.5(k)	Beethoven	А	5230	7.99%	2	0.041	10%

Table 6.1: Experiment data of Figures 1.9, 6.4 and 6.5

A|U|T in Table 6.1 indicates the type of tessellation technique (Adaptive, Uniform or Triangulated after adaptive tessellation) used in the rendering process. For instance, Figure 1.9(b) is generated using uniform subdivision, while Figures 1.9(c), 1.9(d), 1.9(e) are tessellated with the adaptive technique presented in this paper, and Figure 1.9(f) is the triangulated result of Figure 1.9(e). Also Figure 6.4(e) and Figure 6.5(f) are the triangulated results of Figure 6.4(d) and Figure 6.5(e), respectively. The term A/U ratio means the ratio of number of polygons in an adaptively tessellated CCSS to its counter part in a uniformly tessellated CCSS with the same accuracy. The term *Depth* means the number of iterative uniform subdivisions that have to be performed on the control

mesh of a CCSS to satisfy the error requirement. From Table 6.1 we can see that all the adaptively tessellated CCSS's have relatively low A/U ratios. This shows the proposed method can indeed significantly reduce the number of faces in the resulting mesh while satisfying the given error requirement.

The 'Error' column in Table 6.1 represents absolute error. We can easily see that, for the same model, the smaller the error, the lower the A/U ratio. For example, Figure 6.4(b) has lower A/U ratio than Figure 6.4(c) and Figure 6.4(d) because the former has smaller error tolerance than the last two. However, for the same model, if the difference of two error tolerances is not big enough, the resulting adaptive tessellation would have the same subdivision depth (see information on Figures 6.4(g) and 6.4(h) or Figures 6.5(b)and 6.5(c) in Table 6.1). As a result, the one with smaller error tolerance would have higher A/U ratio, because the corresponding uniformly subdivided meshes are the same. Another interesting fact is that Figure 6.5(a) uses much more polygons than Figure 6.5(b)does, while the former is less accurate than the latter. This shows the presented adaptive tessellation method is capable of providing a higher accuracy with less polygons.

From Table 6.1 we can easily see that for different models the absolute errors differ very much. Therefore, for different models, comparing their absolute errors might not make any practical sense because absolute error is not affine transformation invariant. In the mean while, from Table 6.1, we can see that RF is a much better and more understandable measurement for users to specify the error requirement in the adaptive tessellation process.

From Table 6.1, we can also see that triangulated tessellations usually have higher A/U ratio, because triangulation increases the number of polygons by at lease 2 times. Hence triangulation will slow down the rendering process while it does not improve accuracy. From the view point of rendering, triangulation is not really necessary. But for some special applications, such as Finite Element Analysis, triangulation is indispensable. As mentioned above, performing triangulation on the resulting mesh of our adaptive tessellation process is straightforward and fast.



Figure 6.4: Adaptive rendering on surfaces with arbitrary topology.



Figure 6.5: Adaptive rendering on surfaces with arbitrary topology (Continued).

The proposed adaptive tessellation method is good for models that have large flat or nearly flat regions in its limit surface and would save significant amount of time in the final rendering process, but may not have low A/U ratios when it is applied to surfaces with extraordinary curvature distribution or surfaces with very dense control meshes. One main disadvantage of all the current adaptive tessellation methods (including the method proposed here) is that they only eliminate polygons inside a patch. They do not take the whole surface into consideration. For instance, all the flat sides of the rocker arm model in Figure 6.5 are already flat enough, yet a lot of polygons are still generated there.

6.8 Summary

A surface-evaluation-based adaptive tessellation method for general Catmull-Clark subdivision surfaces is presented. The new method only evaluates those limit surface points that are needed in the final rendering process. On the other hand, while previous methods use a significant amount of effort to prevent the occurrence of cracks between adjacent patches, it takes almost no effort for the new method to eliminate cracks in the resulting inscribed polyhedron of the limit surface. Hence the new method is both computation efficient and memory efficient.

The new inscribed approximation based adaptive tessellation method can be used to measure error caused by polygonal approximation. It also can be used for substantially reducing face number of dense meshes with precise error estimation.

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Chapter 7 Multiresolution Representation

In this chapter we will use a multiresolution analysis method [110, 111] to simplify onepiece representation based mesh structures obtained from either interpolation (See Chapter 3) or performing Boolean operations (See Chapter 5). The resulting meshes after multiresolution analysis are still topologically correct control meshes, whose Catmull-Clark subdivision surfaces are approximations of the limit surfaces of the original dense meshes obtained from either interpolation or performing Boolean operations. The new subdivision surfaces obtained from simplified one-piece representation based mesh structures are very close to the original limit surfaces and the error between them can be explicitly calculated. More importantly, we can reconstruct the original control mesh from a sparse one-piece representation based mesh structure without losing any detail. Hence multiresolution-based one-piece representations are good for mesh editing, levelof-detail control and other applications.

The remaining part of the chapter is arranged as follows. In section 1, a brief introduction is given. Some background about wavelet transform and multiresolution analysis is given in Section 2. In section 3, we introduce how to use Lounsbery et al's multiresolution analysis method [110] and Eck et al's mesh conversion method [111] to simplify our one-piece representation based mesh structures. Some concluding remarks are given in Section 4.

7.1 Introduction

Multiresolution analysis and *wavelets* have received considerable attention in recent years, fueled largely by the diverse collection of problems that benefit from their use [19, 21, 110, 111, 112, 113, 114]. The basic idea behind multiresolution analysis is to decompose

a complicated function into a "simpler" low resolution part, together with a collection of perturbations, called *wavelet coefficients*, which are necessary to recover the original function. For many of the functions encountered in practice, a large percentage of the wavelet coefficients are small, meaning that good approximations can be obtained by using only a few of the largest coefficients. Excellent compression rates for images have been achieved using this type of approximation [115].

Multiresolution techniques and the use of hierarchy have a long history in computer graphics. Most recently these approaches have received a significant boost and increased interest through the introduction of the mathematical framework of wavelets. With their roots in signal processing and harmonic analysis, wavelets have lead to a number of efficient and easy-to-implement algorithms. Wavelets have already had a major impact in several areas of computer graphics [19]:

- Image Compression and Processing: some of the most powerful compression techniques for still and moving images are based on wavelet transforms;
- Global Illumination: wavelet radiosity and radiance algorithms are asymptotically faster than other finite element techniques;
- **Hierarchical Modeling**: using multiresolution representations for curves and surfaces accelerates and simplifies many common editing tasks [113];
- Animation: the large constrained optimization tasks which arise in physically based modeling and animation that are subject to goal constraints can be solved faster and more robustly with wavelets;
- Volume Rendering and Processing: wavelets can greatly facilitate dealing with huge data sets since they can be used for compression as well as feature detection and enhancement;
- **Multiresolution Painting**: By using multiresolution analysis, one can build high enough resolution paint systems;

• **Image Query**: using a small number of the largest wavelet coefficients of an image results in a perceptually useful signature for fast search and retrieval.

In computer graphics and geometric modeling, shapes are often represented by polygonal meshes. With the advent of laser scanning systems, meshes of extreme complexity are rapidly becoming commonplace. Such meshes are notoriously expensive to store, transmit, render, and are awkward to edit. Multiresolution analysis offers a simple, unified, and theoretically sound approach to dealing with these problems. In addition, geometric modeling is traditionally based on piecewise polynomial surface representations. However, while special polynomial basis functions are well suited for describing and modifying smooth triangular or quadrilateral patches, it turns out to be rather difficult to smoothly join several pieces of a composite surface along common (possibly trimmed) boundary curves. As flexible patch layout is crucial for the construction of non-trivial geometric shapes, spline-based modeling tools do spend much effort to maintain the global smoothness of a surface.

In this chapter we show that by using techniques from subdivision surfaces, multiresolution analysis can be extended to functions defined on domains of arbitrary topological type [110, 111]. This generalization, which is termed as *subdivision wavelets* [114], dramatically extends the class of applications to which multiresolution analysis can be applied. Multiresolution mesh representations are particularly convenient for a number of geometric modeling applications, including [110, 111]:

- **Compression/simplification**: A multiresolution mesh can be compressed by removing small wavelet coefficients. Moreover, the threshold for removal can be chosen such that the resulting approximation is guaranteed to be within a specified error tolerance of the original mesh.
- **Progressive display** and **transmission**: An attractive method for displaying a complex object is to begin with a low resolution version that can be quickly rendered, and then progressively improve the display as more detail is obtained

from disk or over a network. Using a multiresolution representation, this is simply achieved by first displaying the base mesh, and then progressively adding the contributions of wavelet coefficients in order of decreasing magnitude.

- Level-of-detail control: High performance rendering systems often use a levelof-detail hierarchy, that is, a sequence of approximations at various levels-of-detail. The crudest approximations are used when the viewer is far from the object, while higher detail versions are substituted as the viewer approaches. Multiresolution representations naturally support this type of display by adding successively smaller wavelet coefficients as the viewer approaches the object, and by removing them as the viewer recedes. Moreover, the coefficients can be added smoothly, thereby avoiding the visual discontinuities encountered when switching between approximations of different resolution.
- Multiresolution editing: Editing at various scales can proceed along the lines developed by Finkelstein and Salesin [112] by ordering coefficients according to their support, that is, by the spatial extent of their influence. Multiresolution editing has been done by several people with different methods [21, 110, 112, 114].

7.2 Background

7.2.1 Wavelet Transform

The wavelet transform is a tool for carving up functions, operators, or data into components of different frequency, allowing one to study each component separately. Wavelet transform is capable of providing the time and frequency information simultaneously, hence giving a time-frequency representation of a signal. The term *wavelet* was itself created in 1982, according to Ingrid Daubechies. *Wavelet analysis* may be considered as a generalization of analysis by Hilbert transform or short-time Fourier transform.

The continuous wavelet transform of a function x(t) is defined as follows [109].

$$\Psi_x^{\psi}(\tau,s) = \frac{1}{\sqrt{(|s|)}} \int x(t)\psi^*(\frac{t-\tau}{s})dt$$
where τ represents translation, s represents scale and $\psi(t)$ is the mother wavelet. $\psi^*(t)$ is the complex conjugate of $\psi(t)$. Here are two mother wavelets commonly used in wavelet analysis. The Mexican Hat wavelet is defined as the second derivative of the Gaussian function:

$$\psi(t) = \frac{1}{\sqrt{2\pi}\sigma^3} \left(e^{\frac{-t^2}{2\sigma^2}} \cdot \left(\frac{t^2}{\sigma^2} - 1\right) \right)$$

And the *Morlet wavelet*, named after Jean Morlet, originally formulated by Goupillaud, Grossmann and Morlet in 1984, is defined as

$$\psi(t) = c_{\sigma} \pi^{-\frac{1}{4}} e^{-\frac{t^2}{2}} (e^{i\sigma t} - k_{\sigma})$$

where σ is the scaling parameter that affects the width of the window, $c_{\sigma} = (1 + e^{-\sigma^2} - 2e^{-\frac{3\sigma^2}{4}})^{-1/2}$ is a modulation parameter, and $k_{\sigma} = e^{-\frac{\sigma^2}{2}}$ is defined by the admissibility criterion.

The term *wavelet* means a small wave. The smallness refers to the condition that this (window) function is of finite length (compactly supported). The wave refers to the condition that this function is oscillatory. The term *mother* implies that the functions with different regions of support that are used in the transformation process are derived from the main mother wavelet. In other words, the mother wavelet is a prototype for generating the other window functions. The translation parameter τ is used in the same sense as it was used in the Short Time Fourier Transform (STFT); it is related to the location of the window, as the window is shifted through the signal. This term, obviously, corresponds to time information in the transform domain. However, there is not a frequency parameter. Instead, a scale parameter is defined as 1/frequency. The parameter scale s in the wavelet analysis is similar to the scale used in maps. As in the case of maps, high scales correspond to a non-detailed global view (of the signal), and low scales correspond to a detailed view. Similarly, in terms of frequency, low frequencies (high scales) correspond to a global information of a signal (that usually spans the entire signal), whereas high frequencies (low scales) correspond to a detailed information of a hidden pattern in the signal (that usually lasts a relatively short time). Fortunately in practical applications, low scales (high frequencies) do not last for the entire duration of the signal, but they usually appear from time to time as short bursts, or spikes. High scales (low frequencies) usually last for the entire duration of the signal. Scaling, as a mathematical operation, either dilates or compresses a signal. Larger scales correspond to dilated (or stretched out) signals and small scales correspond to compressed signals.

7.2.2 Wavelet Synthesis

The continuous wavelet transform is a reversible transform [109], provided that Equation (7.1) is satisfied. Fortunately, this is a very non-restrictive requirement. The continuous wavelet transform is reversible if Equation (7.1) is satisfied, even though the basis functions in general may not be orthonormal. The reconstruction is possible by using the following reconstruction formula:

$$x(t) = \frac{1}{c_{\psi}^2} \int_s \int_t \Psi_x^{\psi}(\tau, s) \frac{1}{s^2} \psi(\frac{t-\tau}{s}) d\tau ds$$

where c_{ψ} is a constant that depends on the wavelet used. The success of the reconstruction depends on this constant, called the *admissibility constant*, to satisfy the following *admissibility condition*:

$$c_{\psi} = \sqrt{2\pi} \int_{-\infty}^{\infty} \frac{|\hat{\psi}(\xi)|^2}{|\xi|} d\xi < \infty$$
(7.1)

where $\hat{\psi}(\xi)$ is the Fourier transform of $\psi(t)$. The above equation implies that $\hat{\psi}(0) = 0$, which is $\int \psi(t) dt = 0$.

As mentioned above, the above equation Eq. (7.1) is not a very restrictive requirement since many mother wavelet functions can be found whose integral is zero. For the above equation to be satisfied, the mother wavelet must be oscillatory.

7.2.3 Multiresolution Analysis

A multiresolution analysis [109] of the space $L^2(\mathbb{R})$ consists of a sequence of nested subspaces

$$\cdots \subset V_0 \subset V_1 \subset \cdots \vee V_n \subset V_{n+1} \subset \cdots \subset L^2(\mathbb{R})$$

that satisfies certain self-similarity relations in time/space and scale/frequency, as well as completeness and regularity relations: • Self-similarity in time demands that each subspace V_k is invariant under shifts by integer multiples of 2^{-k} . That is, for each $f \in V_k$, $m \in \mathbb{Z}$, there is a $g \in V_k$ with

$$\forall x \in \mathbb{R} : \quad f(x) = g(x + m2^{-k}).$$

Self-similarity in scale demands that all subspaces V_k ⊂ V_l, k < l, are time-scaled versions of each other, with scaling dilation factor 2^{l-k}, respectively. I.e., for each f ∈ V_k there is a g ∈ V_l with

$$\forall x \in \mathbb{R}: \quad g(x) = f(2^{l-k}x).$$

If f has limited support, then as the support of g gets smaller, the resolution of the l-th subspace is higher than the resolution of the k-th subspace.

- Regularity demands that the model subspace V_0 be generated as the linear hull (algebraically or even topologically closed) of the integer shifts of one or a finite number of generating functions ψ . Those integer shifts should at least form a frame for the subspace $V_0 \subset L^2(\mathbb{R})$, which imposes certain conditions on the decay at infinity. The generating functions are also known as *scaling functions* or *father wavelets*. In most cases those functions need to be piecewise continuous with compact support.
- Completeness demands that those nested subspaces fill the whole space, i.e., their union should be dense in $L^2(\mathbb{R})$, and that they are not too redundant, i.e., their intersection should only contain the zero element.

Because of $V_0 \subset V_1$, a finite sequence of coefficients $a_k = 2\langle \psi(x), \psi(2x-k) \rangle$, for $|k| \leq N$ and $a_k = 0$ for |k| > N, such that

$$\psi(x) = \sum_{k=-N}^{N} a_k \psi(2x - k).$$

Defining another function, known as mother wavelet or just *wavelet*

$$\psi(x) := \sum_{k=-N}^{N} (-1)^k a_{1-k} \psi(2x-k),$$

one can see that the space $W_0 \subset V_1$, which is defined as the linear hull of its integer shifts, is the orthogonal complement to W_0 inside V_1 . Or put differently, V_1 is the orthogonal sum of W_0 and V_0 . By self-similarity, there are scaled versions W_k of W_0 and by completeness one has

$$L^2(\mathbb{R}) = \bigcup_{k \in \mathbb{Z}} W_k,$$

thus the set

$$\{\psi_{k,n}(x) = 2^{k/2}\psi(2^k x - n) : k, n \in \mathbb{Z}\}\$$

is a countable complete orthonormal wavelet basis in $L^2(\mathbb{R})$.

7.3 Multiresolution Analysis for 3D Surfaces with One-Piece Representation

7.3.1 Multiresolution Mesh Representation

As mentioned in the above section, a main idea behind multiresolution analysis is the decomposition of a function (in this case a polyhedron expressed as a parametric function on the sphere) into a low resolution part and a "detail" part [110]. Figure 7.1(middle) is a low resolution version of the polyhedron shown in Figure 7.1(left). The vertices in Figure 7.1(middle) are obtained by applying linear transforms on the vertices in Figure 7.1(left). These linear transforms essentially implement a low pass filter denoted as A. The detail part consists of a set of abstract coefficients, called wavelet coefficients, that are computed as weighted differences of the vertices in Figure 7.1(left). These differencing weights form a high-pass filter B. The decomposition process, technically called *analysis* [110], can be used to further transform Figure 7.1(middle) into an even lower resolution version and corresponding wavelet coefficients. This cascade of analysis processes, referred to as a *filter bank algorithm* [110], repeats with an even much coarser level representation shown in Figure 7.1(right), together with wavelet coefficients at each level.

The analysis filters A and B are constructed so that the original polyhedron can be recovered exactly from the low-resolution version and the wavelet coefficients. Recovery, technically called *synthesis* [110], can be used to reconstruct Figure 7.1(left) from Figure



Figure 7.1: Illustration of Multiresolution Analysis.

7.1(middle) together with the finest-level wavelet coefficients. Synthesis refines each patch of Figure 7.1(middle) into more subpatches by introducing new vertices around the patch, and then by perturbing the resulting set of vertices according to the wavelet coefficients. The refining and perturbing steps can be described by two different filters P (the *refining filter*) and Q (the *perturbing filter*), collectively called *synthesis filters* [110]. The task is to develop the four analysis and synthesis filters so that [110]:

- the low-resolution versions are good approximations of the original object (in a least-squares sense);
- the magnitude of a wavelet coefficient reflects a coefficient's importance by measuring the error introduced when the coefficient is set to zero; and
- analysis and synthesis filter banks should have time complexity that is linear in the number of vertices.

7.3.2 Lounsbery et al's Multiresolution Analysis Method

Multiresolution analysis was first formalized by Mallat [108] for functions defined on \mathbb{R}^n . Lounsbery et al. [110] have recently extended the notion of multiresolution analysis to functions defined on base complexes of arbitrary topological type, so that multiresolution mesh representation can be calculated robustly and efficiently. However, their results can only be used to construct multiresolution representations of meshes with subdivision connectivity. The purpose of this subsection is to summarize their basic results and algorithms at a high level.

As we know, the two basic ingredients of multiresolution analysis are a sequence of nested linear function spaces and an inner product. Lounsbery et al. use a sequence of spaces V_0, V_1, \cdots associated with meshes of arbitrary topology. To describe meshes, the approximation spaces V_j consist of piecewise linear functions that can also be regarded as the space of continuous piecewise linear functions over a partition K_j of K_0 , created by performing j times of recursive subdivision to the faces of K_0 . As j increases, the mesh K_j becomes more and more dense. As a result, the functions in V_j can better represent arbitrary continuous functions on K_0 . The inner product used by Lounsbery et al. is the standard inner product defined as

$$\langle f,g \rangle := \int_{x \in K_0} f(x)g(x)dx$$

where dx is the differential area of K_0 embedded in \mathbb{R}^m , so that all faces have unit area.

The above inner product can be used to define the following orthogonal complement spaces that is also called *wavelet spaces*,

$$W_j := \{ f \in V_{j+1} | < f, g \ge 0, \forall g \in V_j \}$$

Basically, W_j describes the detail that is discarded when a function in V_{j+1} is approximated by a function in V_j .

Basis functions for V_j are also called *scaling functions*. It is easy to see that a *wavelet* $\psi_i^k(x)$ is a basis function for one of the wavelet spaces W_k . Lounsbery et al. [110] give constructions for wavelet bases on arbitrary mesh K_0 . Lounsbery et al. shows that a wavelet basis for V_j consists of a basis for V_0 together with bases for the wavelet spaces W_0, \dots, W_{j-1} .

In [110], the parametrization $\rho^J \in V_J$ for V_j is expanded in the hat function basis as

$$\rho^J(x) = \sum_i v_i^J \psi_i^J, \qquad x \in K_0$$

where v_i^J denotes the vertex positions of M^J . A multiresolution representation of ρ^J is the expansion in a wavelet basis:

$$\rho^{J}(x) = \sum_{i} v_{0} \psi_{i}^{0}(x) + \sum_{j=0}^{J-1} \sum_{i} w_{i}^{j} \psi_{i}^{j}(x), \qquad x \in K_{0},$$

where w_i^j denotes the wavelet coefficients. The full detail model, described by $\rho^J(x)$ can be successively decomposed into lower resolution approximations together with collections of coefficients. The results are coarser meshes together with wavelet coefficients at various levels of detail. The analysis process itself involves operations of sparse matrices which initially were calculated and distributed by Lounsbery et al. [110].

7.3.3 Subdivision Connectivity Requirement

An important assumption in Lounsbery et al's multiresolution analysis method is that the geometric connectivity of the input mesh must be of the form of a mesh M^{j} that results from subdividing a simple mesh M^0 j times. This requirement is called *subdi*vision connectivity [111]. Therefore Lounsbery et al's multiresolution analysis method [110] can only be applied to a restricted class of meshes with subdivision connectivity. Unfortunately, meshes used in practice usually do not meet this requirement. For example, the one-piece representation meshes we obtain by using the Boolean operation method presented in Chapter 5 do not satisfy this requirement. The one-piece representation meshes obtained by the interpolation technique presented in Chapter 3 do not satisfy this requirement either. Hence, in order to represent these complex meshes which do not satisfy the subdivision connectivity requirement in multiresolution meshes, we need to first convert them into meshes that satisfy the subdivision connectivity requirement. In this section we introduce a method presented by Eck et al [111] to overcome the subdivision connectivity restriction. In other words, with Eck et al's method, completely arbitrary meshes can be converted to multiresolution representation. The method is based on the approximation of an arbitrarily given mesh M by a mesh M^{j} that has subdivision connectivity and is guaranteed to be very close to M (within a given error tolerance ϵ).

The key part [111] of the conversion procedure is the construction of a parametrization of M over a base complex K_0 possessing a small number of faces. The parametrization is then sampled to produce the remesh. Considerable care is taken to create a parametrization and a sampling pattern so that the resulting remesh can be well approximated with relatively few wavelet coefficients.

The basic idea of remeshing [111] is to construct a parametrization of M over a suitably determined domain mesh K_0 . This parametrization is then resampled to produce a mesh M_J that has subdivision connectivity and is of the same topological type as M. The remeshing algorithm presented by Eck et al. [111] consists of three steps:

- Partitioning: Partition M into a number of triangular regions T₁,..., T_r. We want the number r of regions to be small, because the lowest complexity approximation we can construct has r faces. Basic tools used in partitioning are harmonic maps, which are maps that preserve as much of the metric structure (lengths, angles, etc.) of M as possible. Harmonic maps are described in [111]. A detailed description of the partitioning algorithm is given in [111]. Let m be the number of vertices or nodes of the triangulation T₁,..., T_r. This triangulation determines the structure of a simple mesh K₀, called the base complex [111], with a face corresponding to each of the r triangular regions. This mesh serves as the domain of the parametrization constructed in the next step.
- **Parametrization**: For each region T_i of M construct a (local) parametrization $\rho_i : F_i \to T_i$ over the corresponding face F_i of the base complex K_0 . The local parameterizations are made to fit together continuously, meaning that collectively they define a globally continuous parametrization $\rho : K_0 \to M$. The coordinate functions of the parametrization are required to vary as little as possible since such functions have multiresolution approximations with few significant wavelet coefficients, leading to high compression ratios. Harmonic maps in a sense minimize distortion and therefore are particularly well suited for this purpose. The description of the parametrization step is presented in [111].

• Resampling: Perform J recursive 4-to-1 splits on each of the faces of K_0 . This results in a triangulation K_J of K_0 with subdivision connectivity. The remesh M_J is obtained by mapping the vertices of K_J into \mathbb{R}^3 using the parametrization ρ , and constructing an interpolating mesh in the obvious way; M_J therefore has vertices lying on M, and has subdivision connectivity. The resampling step is described more fully in [111], and it is shown that J can be determined so that M_J and Mdiffer by no more than a specified remeshing tolerance ϵ .

7.3.4 One-Piece Representation using Multiresolution

With the availability of converting meshes of arbitrary topology to meshes satisfying the subdivision connectivity requirement [111], now multiresolution analysis of an arbitrary mesh M thus proceeds in two steps:

- First use the method of Eck et al. [111] to convert M to a mesh M_J which satisfies the subdivision connectivity requirement, and
- Then use the method of Lounsbery et al. [110] to convert M_J to a multiresolution mesh representation.

Because multiresolution analysis can now be applied to meshes with arbitrary topology and arbitrary connectivity, we certainly can use multiresolution analysis to simplify our one-piece results obtained from Boolean operations described in Chapter 5 or onepiece represented meshes obtained from the interpolation method presented in Chapter 3. As depicted in Chapter 5, we have a dense (depends on accuracy requirement) one piece represented polygonal surface obtained through performing one or more Boolean operations. The polygons of the one piece surface can be regarded as the control mesh of the surface because the mesh itself is very dense. The limit surface of the dense mesh would be almost the same as the target surface. Hence by applying multiresolution analysis to the dense mesh, we can obtain a sparser mesh together with some small wavelet coefficients (see Figure 7.1). The limit surface of the resulting sparser mesh is somewhat different from the original surface, because some details are lost in the wavelet coefficients, but their limit surface would be very similar because the lost information is small. Furthermore, the error between the limit surface from the sparser mesh and the limit surface of the original mesh can be exactly measured [110]. Hence it gives users more flexibility in modeling. For example, in the design process, a relatively low resolution mesh (i.e. sparser mesh) is desired for fast modeling/rendering purposes, because at a low resolution level, it is easier to control (for example, directly edit) the mesh. Also in the final stage of the design process, we can use the inverted multiresolution analysis process, i.e. filterbank synthesis process to synthesize a sparser mesh and wavelet coefficients back to a finer and denser mesh, whose limit surface is closer to the final modeling result.



Figure 7.2: Multiresolution of surfaces after Boolean operations: (a) two given objects; (b) one-piece representation after a difference operation; (c) mesh of (b); (d) low resolution representation of (c); (e) lower resolution representation of (c)

The above Figure 7.2 is an example of multiresolution mesh representation of one piece represented objects after performing Boolean operations. Figure 7.2(a) is two objects positioned for Boolean operations. Figure 7.2(b) is the resulting object after a *difference* operation between the two given objects. Figure 7.2(c) is the mesh structure of the one-piece represented object shown in Figure 7.2(b). This mesh is obtained using the method presented in Chapter 5. Figure 7.2(d) and Figure 7.2(e) are two lower resolution

representations of the mesh shown in Figure 7.2(c). As we can see, the meshes are much sparser than the original one, but lose only some small details in the resulting objects. Together with wavelet coefficients, the mesh shown in Figure 7.2(e) can be synthesized back to the mesh shown in Figure 7.2(d) exactly, and furthermore, the mesh shown in Figure 7.2(d) can be synthesized back to the mesh shown in Figure 7.2(c) exactly too. Hence, in a design process, users can conveniently choose a resolution level for the representation. Even if the error tolerance is changed in the future, the modeling results do not need to be redesigned because a different resolution level of the one-piece multiresolution representation can be obtained from each other as long as the list of wavelet coefficients exists as well.

7.4 Summary

The multiresolution analysis method is introduced in this chapter for the purpose of simplifying dense one-piece representation based mesh structures. The Multiresolution analysis process is based on the method of Lounsbery et al's multiresolution analysis method [110] and Eck et al's mesh conversion method [111]. There are a list of versions of resulting meshes after multiresolution analysis is applied. All of them have different resolutions with different details. However, each of them can be regarded as a control mesh, whose limit surface is an approximation of the original limit surface. The error between them is explicitly known. Hence for a given error tolerance ϵ , we can find the sparsest one-piece representation based mesh structure that satisfies the error control condition. Furthermore, even though error requirements are changed later, we do not need to remodel the resulting solids because we can reconstruct the original control mesh from a sparse one-piece representation based mesh structure, without losing any detail.

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Chapter 8

Conclusion and Future Work

8.1 Conclusion

This dissertation presents our research work in subdivision surface based one-piece representation. Our research work related to the dissertation is summarized as follows.

• In Chapter 2 we develop a foundation for subdivision surface based one-piece representation. Our new parametrization technique for general Catmull-Clark subdivision surfaces extends J. Stam's work [17] by redefining all the eigen basis functions in the parametric representation for general Catmull-Clark subdivision surfaces and giving each of them an explicit form. The entire eigen structure of the subdivision matrix and its inverse are computed exactly and explicitly with no need to precompute anything. Therefore, the new representation can be used not only for evaluation purposes, but for analysis purposes as well [58]. Furthermore, since the number of eigen basis functions required in the new approach is only one half of the previous approach [17], our new parametrization is also more efficient for evaluation purposes.

With our parametrization technique available, we could then develop necessary mathematical theories and geometric algorithms in the other chapters to support subdivision surface based one-piece representation.

• We propose two ways for constructing subdivision surface based one-piece representation in our dissertation research. **Chapter 3** introduces one of them. We establish explicit conditions for building a sparse mesh structure of a given model using an interpolation technique. Although some interpolation methods using subdivision surfaces [11, 25, 26, 52] exist, most of them cannot handle open meshes and the resultant surface exhibits undesired artifacts and undulations. We develop a new method for constructing a sparse mesh whose smooth Catmull-Clark subdivision surface (CCSS) interpolates the vertices of a mesh with arbitrary topology. The new method handles both open and closed meshes. Normals or derivatives specified at any vertices of the mesh (which can actually be anywhere) can also be interpolated. The construction of the interpolating surface uses information from the given mesh as well as its limit surface, hence it gives us more control on the smoothness of the interpolating surface and, consequently, avoids the need of the shape fairing process in the construction of the interpolating surface. Most importantly, the computation of the interpolating surface's control mesh follows a new approach, which does not require the resulting global linear system to be solvable. An approximate solution provided by any fast iterative linear system solver is sufficient. Nevertheless, interpolation of the given mesh is guaranteed. This is an important improvement over previous methods because with these features, the new method can handle meshes with a large number of vertices efficiently.

• In Chapter 4 we establish necessary and sufficient conditions for accurately converting a subdivision surface into a set of voxels. Our discretization of 3D closed objects is guaranteed to be leak-free when a 3D flooding operation is performed. This is ensured by proving that our voxelization results satisfy the properties of separability, accuracy and minimality.

Based on the voxelization technique, a method for performing robust and error controllable Boolean operations on free-form solids represented by Catmull-Clark subdivision surfaces is developed in **Chapter 5**. Different from previous voxelizationbased approaches, the final result of the Boolean operations in our method is represented with a continuous geometric representation (i.e. a polygonal mesh). This is achieved by doing the Boolean operations in the parameter spaces of the solids, instead of the object space. This approach allows us to easily compute a continuous approximation of the intersection curve and, consequently, to build a one-piece representation for the whole Boolean operation result.

- The one-piece representation resulting from either performing Boolean operations or interpolation could be dense. In **Chapter 6** we develop a method called adaptive tessellation, for simplifying the control meshes. All the simplified meshes have accurate error measurement. Hence error control is no problem. Adaptive tessellation reduces the number of faces needed to yield a smooth approximation to the limit surface and, consequently, makes the rendering process more efficient. The main advantages of our method are that it handles crack problems easily and has explicit error estimation.
- Another mesh reduction approach for simplifying one-piece represented control meshes is introduced in **Chapter 7**. By using the multiresolution analysis method [110, 111] we can significantly simplify one-piece representation based mesh structures obtained from either interpolation or performing Boolean operations. The resulting meshes after multiresolution analysis are still topologically correct control meshes, and are good approximations to the original mesh with explicit error calculation. More importantly, from a sparse one-piece representation based mesh structure obtained from a multiresolution analysis process, we can reconstruct the original control mesh without losing any detail. Hence multiresolution based one-piece representations are very good for simplification of our subdivision surface based one-piece representation.
- We have built a system that supports subdivision surface based one-piece representation. The system can perform all the functions necessary for constructing a subdivision surface based one-piece representation. A lot of examples have been tested within the system. All the examples shown in this dissertation are generated by this system.

8.2 Future Work

We have done some fundamental research related to representations of topologically complex 3D objects and obtained some good results. More work needs to be done to make the subdivision surface based one-piece representation system a mature and useful tool for designing and modeling. The following are a number of problems inspired by my dissertation research, and are theoretically significant and practically valuable. They will constitute my main research directions in the near future.

• More Convenient and Accurate Representations of Topologically Complex 3D Objects:

Convenient representations should be easily understood by users, easily constructed by modelers and easily manipulated by programmers. One-piece representation is one type of convenient representation. But there are still a lot of things that need to be improved. For example, for solid objects obtained from a sequence of Boolean operations, their subdivision surface based one-piece representation would be irregular and redundant because we did not take the regularity of the original solids into account while performing Boolean operations. Hence such one-piece representation can still be improved in future work.

Accurate representations of complex 3D objects are also of great importance in many disciplines ranging from engineering and manufacturing to medicine and biology. With the wide availability of powerful computational resources and ever better acquisition technologies such as 3D laser scanning and volumetric MRI or CAT imaging, the geometries used in applications are becoming increasingly complex. Traditional representations of geometry are inefficient and even ineffective in representing such complex topologies. Increasing complexity of geometric data calls for new efficient and accurate representations. The goal of my future research is to study more flexible, effective and accurate representations of complex 3D objects. The representations to be studied in my future research would result in more efficient ways of manipulating and processing topologically complex 3D objects.

• Algorithms for Construction, Manipulation, Processing, Transmission, Rendering and Storage of Topologically Complex 3D Objects:

Topologically complex 3D objects have become increasingly common in applications arising in diverse areas. Plastic surgeons use computer models to plan and perform operations. Mechanical engineers virtually assemble and test cars and planes. Animators create complex characters and effects indistinguishable from reality. Now it is even feasible to create 3D computer models of sculptures capturing the most subtle details using geometric design. Increasing complexity of geometric data requires more efficient algorithms for construction, rendering, manipulation, processing, transmission, compression and storage of such a large amount of complex data.

Therefore part of our immediate future work is to design efficient algorithms for 3D modeling to improve the quality of the resulting objects without sacrificing much performance. Both theoretical properties of the object representations (regularity, convergence rates, fairness, stability) and their practical applications need to be studied. The algorithms to be developed will have immediate applications in many fields, such as computer-aided design, medical visualization and computer animation.

• Interdisciplinary Research and Applications:

My research area is inherently interdisciplinary, assimilating ideas from computer graphics, computer vision, image processing, geometric modeling, applied mathematics, bio-informatics and computer engineering to develop 3D models for design and visualization. I will continue my efforts of exploring the theoretical foundation and practical algorithms of accurate representation and efficient manipulation of 3D objects and pursue their applications in interdisciplinary areas. For example, our research results could be applied to virtual surgery because human bodies, blood vessels, organs or even complex brains can be accurately represented. As a matter of fact, modeling of macro- and microscopic biological structures is becoming increasingly important for medical research, training, and treatment support. However, because such body structures often have extremely complex shapes and topology (e.g., the blood vessel system, facial muscles, folded protein molecules), it is still a challenging research work. Hence putting our research results into such applications is another immediate part of future work for us.

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Appendix A

Eigen Basis Matrices $M_{b,j}$

We introduce two notations first:

$$M_{b,j}[k]$$
 and $M_{b,j} = [Y_1, Y_2, \cdots, Y_{2n+8}].$

The first notation, $M_{b,j}[k]$, represents the kth row of the matrix $M_{b,j}$. The second notation means that $M_{b,j}$ is a matrix of 2n + 8 columns and the *i*th column of $M_{b,j}$ is Y_i . Recall that $c_{\omega} = \cos(2\pi\omega/n)$. Also, we define

$$f(k,\omega, x_1, x_2, x_3, x_4, x_5) = \sum_{i=1}^{5} x_i c_{(k-i+2)\omega}$$

The matrices $M_{b,j}$, b = 1, 2, 3, j = 0, 1, ..., n + 5, are shown below in eight groups.

1. $M_{b,0}$ and $M_{b,1}$ (corresponding to eigenvalues λ_0 and λ_1): let

$$\begin{split} h &= (r-1)(8r-1)(32r-1)(64r-1)(r-s); \\ Y_b^T &= \frac{1}{h} \mathbf{Q}_b \cdot [1, r, r^2, r^3, r^4, r^5, r^6]; \\ \hat{Y}_b &= -\frac{8s}{n} Y_b; \\ \bar{Y}_b &= \frac{8s-1}{n} Y_b \end{split}$$

where $r = \lambda_0$, $s = \lambda_1$, and Q_b (see **Appendix B**) is a constant matrix of dimension 16×7 . Then

$$\mathbf{M}_{b,0} = [Y_b, \hat{Y}_b, \cdots, \hat{Y}_b, \bar{Y}_b, \cdots, \bar{Y}_b, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}]$$

where **0** is a zero vector of dimension 16. $M_{b,1}$ is obtained by switching r and s in the above items.

2. $M_{b,2\omega}$ and $M_{b,2\omega+1}$ (corresponding to eigenvalues $\lambda_{2\omega}$ and $\lambda_{2\omega+1}$), $1 \le \omega < n/2$: let

$$\begin{split} h &= (64r-1)(32r-1)(16r-1)(r-s)/2; \\ V_k^T &= [c_{(k+1)\omega}, c_{k\omega}, c_{(k-1)\omega}, c_{(k-2)\omega}, c_{(k-3)\omega}]; \\ B_k^T &= [V_k^T, rV_k^T, r^2V_k^T, r^3V_k^T, r^4V_k^T, r^5V_k^T]; \\ \hat{Y}_k &= \frac{1}{nh} Q_{b1} B_k; \\ \bar{Y}_k &= \frac{4s-1}{nh(1+c_\omega)} Q_{b2} B_k \end{split}$$

where $1 \leq k \leq n$, $r = \lambda_{2\omega}$, $s = \lambda_{2\omega+1}$ and Q_{b1} and Q_{b2} (see **Appendix B**) are constant matrices of dimension 16×30 . Then

$$M_{b,2\omega} = [\mathbf{0}, \hat{Y}_1, \cdots, \hat{Y}_n, \bar{Y}_1, \cdots, \bar{Y}_n, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}].$$

 $M_{b,2\omega+1}$ is obtained by switching r and s in the above items. Note that if n is odd, the last two matrices in this group are $M_{b,n-1}$ and $M_{b,n}$. Otherwise, the last two matrices are $M_{b,n-2}$ and $M_{b,n-1}$.

3. $M_{b,n}$ when n is even (corresponding to eigenvalue λ_n): set

$$\hat{Y}^T = [1, -1, 1, -1, \cdots, (-1)^{n+1}]/(4n),
\bar{Y}^T = [0, 0, 0, 0, \cdots, 0]$$

where \hat{Y} and \bar{Y} are of length n. Then for b = 1, 2 and 3, we have

$$\begin{aligned} \mathbf{M}_{b,n}[4] &= [0, \ \hat{Y}^T, \ \bar{Y}^T, 0, 0, 0, 0, 0, 0, 0]; \\ \mathbf{M}_{b,n}[5] &= [0, \ \bar{Y}^T, \ \hat{Y}^T, 0, 0, 0, 0, 0, 0, 0]; \\ \mathbf{M}_{b,n}[6] &= [0, \ -\hat{Y}^T, \ \bar{Y}^T, 0, 0, 0, 0, 0, 0, 0]; \end{aligned}$$

and all the other rows are zero.

4. $M_{b,n+1}$ (corresponding to eigenvalue= 1): set

$$t = 5 + 14\beta_n + 16\gamma_n;$$

$$\hat{h}_k = 4(3\beta_n + 2\gamma_n)/(nt);$$

$$\bar{h}_k = 2(\beta_n + 4\gamma_n)/(nt),$$

 $1 \leq k \leq n$. Then for b = 1, 2 and 3, we have

$$\mathbf{M}_{b,n+1}[1] = [5/t, \hat{h}_1, \dots, \hat{h}_n, \bar{h}_1, \dots, \bar{h}_n, 0, 0, 0, 0, 0, 0, 0]$$

and all the other rows are zero.

5. $M_{b,n+2}$ (corresponding to eigenvalue= 1/8): set

$$\hat{y}_{k} = \begin{cases} \frac{-11(n-1)}{144n}, & k = 1\\ \frac{1}{144n}, & 1 < k \le n \end{cases}$$

$$\bar{y}_{k} = \begin{cases} \frac{5(n-2)}{288n}, & k = 1 \text{ or } k = n\\ \frac{-5}{144n}, & 1 < k < n \end{cases}$$

$$t = 48(8\gamma_{n} - 1);$$

$$\hat{h} = (11 - 24\beta_{n} + 8\gamma_{n})/(3tn);$$

$$\bar{h} = (-5 + 24\beta_{n} - 32\gamma_{n})/(3tn).$$

Then for b = 1, 2 and 3 we have

$$\begin{split} \mathbf{M}_{b,n+2}[7] &= \begin{bmatrix} -1/t, \hat{h} + \hat{y}_1, \hat{h} + \hat{y}_2, \cdots, \hat{h} + \hat{y}_n, \bar{h} + \bar{y}_1, \\ \bar{h} + \bar{y}_2, \cdots, \bar{h} + \bar{y}_n, \frac{1}{288}, \frac{1}{72}, \frac{1}{288}, 0, 0, 0, 0]; \\ \mathbf{M}_{b,n+2}[10] &= \begin{bmatrix} -1/t, \hat{h} + \hat{y}_n, \hat{h} + \hat{y}_1, \cdots, \hat{h} + \hat{y}_{n-1}, \bar{h} + \bar{y}_n, \\ \bar{h} + \bar{y}_1, \cdots, \bar{h} + \bar{y}_{n-1}, 0, 0, 0, 0, \frac{1}{288}, \frac{1}{72}, \frac{1}{288} \end{bmatrix} \end{split}$$

and all the other rows are zero.

6. $M_{b,n+3}$ (corresponding to eigenvalue= 1/16): set

$$\hat{y}_k = \sum_{\omega=0}^n \frac{7 \cdot f(k,\omega,0,1,0,-1,0)}{64n(2c_\omega - 7)}; \bar{y}_k = \sum_{\omega=0}^n \frac{f(k,\omega,-5,-23,23,5,0)}{192n(2c_\omega - 7)},$$

where $1 \le k \le n$. Then for b = 1, 2 and 3 we have

and all the other rows are zero.

7. $M_{b,n+4}$ (corresponding to eigenvalue= 1/32): set

$$\hat{y}_{k} = \sum_{\omega=0}^{n} \frac{-f(k,\omega,0,41,180,41,0)}{384n(4c_{\omega}-45)}, \\
\bar{y}_{k} = \sum_{\omega=0}^{n} \frac{f(k,\omega,10,100,100,10,0)}{384n(4c_{\omega}-45)}, \\
t = 32(-41+96\beta_{n}-32\gamma_{n}); \\
\hat{h} = (-131+192\beta_{n}+160\gamma_{n})/(6tn); \\
\bar{h} = (55-96\beta_{n}-80\gamma_{n})/(3tn),$$

 $1 \leq k \leq n$. Then for b = 1, 2 and 3, we have

and all the other rows are zero.

8. $M_{b,n+5}$ (corresponding to eigenvalue= 1/64): set

$$\hat{y}_{k} = \sum_{\omega=0}^{n} \frac{f(k,\omega,0,27,461,461,27)}{768n(8c_{\omega}-217)}; \\ \bar{y}_{k} = \sum_{\omega=0}^{n} \frac{-f(k,\omega,10,229,888,229,10)}{1152n(8c_{\omega}-217)}; \\ t = -209 + 448\beta_{n} - 64\gamma_{n}; \\ \hat{h} = (61 - 92\beta_{n} - 64\gamma_{n})/(48tn); \\ \bar{h} = (-683 + 1216\beta_{n} + 752\gamma_{n})/(576tn),$$

 $1 \leq k \leq n.$ Then for b=1,2 and 3, we have

$$\mathbf{M}_{b,n+5}[16] = \begin{bmatrix} \frac{-45}{256t}, \hat{h} + \hat{y}_1, \cdots, \hat{h} + \hat{y}_n, \bar{h} + \bar{y}_1, \cdots, \\ h + \bar{y}_n, \frac{-1}{2304}, \frac{1}{768}, \frac{-1}{768}, \frac{1}{2304}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{-1}{768}, \frac{1}{768}, \frac{-1}{768}, \frac{-1}{$$

and all the other rows are zero.

Appendix B

Matrices Q_b , Q_{b1} and Q_{b2}

The matrices Q_b , Q_{b1} and Q_{b2} , (b = 1, 2, 3) are listed as follows. Matrices Q_b are of size 16×7 and Matrices Q_{b1} and Q_{b2} are of size 16×30 . In order to put each of them into a page, all matrices Q_{b1} and Q_{b2} are divided into two parts: $Q_{b1}[1:15]$, $Q_{b1}[16:30]$, $Q_{b2}[1:15]$ and $Q_{b2}[16:30]$. $Q_{b1}[1:15]$ are the submatrices of Q_{b1} by taking the first 15 columns of Q_{b1} and $Q_{b1}[16:30]$ are the submatrices of Q_{b1} by taking the last 15 columns of Q_{b1} . Similarly $Q_{b2}[1:15]$ and $Q_{b2}[16:30]$ are submatrices of Q_{b2} .

	1	•		1				
	$\begin{bmatrix} \frac{1}{144} \end{bmatrix}$	$-\frac{7}{8}$	$\frac{217}{6}$	-620	$\frac{13888}{3}$	-14336	$\frac{131072}{9}$ -	1
	-1/48	$\frac{125}{48}$	$-\frac{1271}{12}$	$\frac{5270}{3}$	$-\frac{36704}{3}$	$\frac{97280}{3}$	$-\frac{65536}{3}$	
	0	0	0	0	0	0	0	
	1/48	$-\frac{41}{16}$	$\frac{807}{8}$	-1565	9488	-18944	$\tfrac{32768}{3}$	
	0	0	0	0	0	0	0	
	$-\frac{1}{96}$	$\frac{43}{32}$	-58	1070	-8864	29696	$-\frac{65536}{3}$	
	$-\frac{1}{144}$	$\tfrac{119}{144}$	$-rac{2191}{72}$	$\frac{7429}{18}$	$-rac{17528}{9}$	$\tfrac{30464}{9}$	$-\tfrac{16384}{9}$	
0	0	0	0	0	0	0	0	
$Q_1 =$	1/32	$-\frac{125}{32}$	$\frac{1271}{8}$	-2635	18352	-48640	32768	
	$\frac{1}{192}$	$-\frac{133}{192}$	$\frac{1505}{48}$	-612	$\frac{15568}{3}$	$-\frac{46592}{3}$	$\tfrac{32768}{3}$	
	0	0	0	0	0	0	0	
	-1/32	$\frac{117}{32}$	$-\frac{1053}{8}$	1800	-10632	25344	-16384	
	$-\frac{1}{64}$	$\frac{125}{64}$	$-\tfrac{1271}{16}$	$\frac{2635}{2}$	-9176	24320	-16384	
	$\frac{1}{96}$	$-\frac{101}{96}$	$\frac{713}{24}$	$-\frac{1054}{3}$	$\frac{5704}{3}$	$-\frac{12928}{3}$	$\frac{8192}{3}$	
	$\frac{1}{64}$	$-\frac{109}{64}$	$\frac{899}{16}$	$-\frac{1581}{2}$	4960	-12416	8192	
	$-\frac{1}{192}$	$\frac{77}{192}$	$-\frac{539}{48}$	$\frac{865}{6}$	$-\frac{2576}{3}$	$\frac{6272}{3}$	$-\frac{4096}{3}$	
Г	1/48	$-\frac{47}{18}$	$\frac{961}{9}$	$-\frac{16120}{9}$	$\frac{115072}{9}$	$-\frac{323584}{9}$	$\frac{262144}{9}$	
	-1/24	$\frac{125}{24}$	$-\frac{1271}{6}$	$\frac{10540}{3}$	$-\frac{73408}{3}$	$\tfrac{194560}{3}$	$-\frac{131072}{3}$	
	-1/24	$\frac{125}{24}$	$-\frac{1271}{6}$	$\tfrac{10540}{3}$	$-\frac{73408}{3}$	$\tfrac{194560}{3}$	$-\frac{131072}{3}$	
	1/32	$-\frac{371}{96}$	$\frac{923}{6}$	$-\frac{7330}{3}$	$\tfrac{46816}{3}$	$-\tfrac{105472}{3}$	$\frac{65536}{3}$	
	1/16	$-\frac{125}{16}$	$\frac{1271}{4}$	-5270	36704	-97280	65536	
	1/32	$-\frac{371}{96}$	$\frac{923}{6}$	$-\frac{7330}{3}$	$\tfrac{46816}{3}$	$-\tfrac{105472}{3}$	$\frac{65536}{3}$	
	$-\frac{5}{576}$	$\tfrac{601}{576}$	$-\frac{5653}{144}$	$\frac{5032}{9}$	$-\frac{26704}{9}$	$\tfrac{54784}{9}$	$-\frac{32768}{9}$	
	-1/32	$\frac{125}{32}$	$-\frac{1271}{8}$	2635	-18352	48640	-32768	
$Q_2 =$	-1/32	$\frac{125}{32}$	$-\frac{1271}{8}$	2635	-18352	48640	-32768	
	$-\frac{5}{576}$	$\tfrac{601}{576}$	$-\frac{5653}{144}$	$\frac{5032}{9}$	$-\frac{26704}{9}$	$\frac{54784}{9}$	$-\frac{32768}{9}$	
	$\frac{1}{192}$	$-\frac{125}{192}$	$\frac{1271}{48}$	$-\frac{2635}{6}$	$\tfrac{9176}{3}$	$-\frac{24320}{3}$	$\tfrac{16384}{3}$	
	0	-1/4	$\frac{109}{4}$	-835	7720	-23296	16384	
	$\frac{1}{192}$	$-\frac{125}{192}$	$\frac{1271}{48}$	$-\frac{2635}{6}$	$\frac{9176}{3}$	$-\frac{24320}{3}$	$\frac{16384}{3}$	
	$\frac{1}{192}$	$-\frac{77}{192}$	$\frac{155}{48}$	$\frac{527}{6}$	$-\frac{3472}{3}$	$\frac{11392}{3}$	$-\frac{8192}{3}$	
	$\frac{1}{192}$	$-\frac{77}{192}$	$\frac{155}{48}$	$\frac{527}{6}$	$-\frac{3472}{3}$	$\frac{11392}{3}$	$-\frac{8192}{3}$	
l	$-\frac{1}{288}$	$\frac{53}{288}$	$-\frac{173}{72}$	$-\frac{20}{9}$	$\frac{1448}{9}$	$-\frac{5504}{9}$	$\frac{4096}{9}$	

[$-\frac{1}{144}$	$-\frac{7}{8}$	$\frac{217}{6}$	-620	$\frac{13888}{3}$	-14336	$\frac{131072}{9}$
	0	0	0	0	0	0	0
	-1/48	$\frac{125}{48}$	$-\tfrac{1271}{12}$	$\tfrac{5270}{3}$	$-rac{36704}{3}$	$\frac{97280}{3}$	$-\frac{65536}{3}$
	$-\frac{1}{96}$	$\frac{43}{32}$	-58	1070	-8864	29696	$-rac{65536}{3}$
	0	0	0	0	0	0	0
	1/48	$-\frac{41}{16}$	$\frac{807}{8}$	-1565	9488	-18944	$\frac{32768}{3}$
	$\frac{1}{192}$	$-\frac{133}{192}$	$\tfrac{1505}{48}$	-612	$\frac{15568}{3}$	$-\frac{46592}{3}$	$\frac{32768}{3}$
0	1/32	$-\frac{125}{32}$	$\frac{1271}{8}$	-2635	18352	-48640	32768
~~3 —	0	0	0	0	0	0	0
	1						
	$-\frac{1}{144}$	$\frac{119}{144}$	$-\frac{2191}{72}$	$\frac{7429}{18}$	$-\frac{17528}{9}$	$\frac{30464}{9}$	$-\frac{16384}{9}$
	$-\frac{1}{144}$ $-\frac{1}{64}$	$\frac{119}{144}$ $\frac{125}{64}$	$-\frac{2191}{72}$ $-\frac{1271}{16}$	$\frac{\frac{7429}{18}}{\frac{2635}{2}}$	$-\frac{17528}{9}$ -9176	$\frac{30464}{9}$ 24320	$-\frac{16384}{9}$ -16384
	$-\frac{1}{144}$ $-\frac{1}{64}$ -1/32	$ \frac{119}{144} $ $ \frac{125}{64} $ $ \frac{117}{32} $	$-\frac{2191}{72} \\ -\frac{1271}{16} \\ -\frac{1053}{8}$	$\frac{7429}{18}$ $\frac{2635}{2}$ 1800	$-\frac{17528}{9}$ -9176 -10632	$\frac{30464}{9}$ 24320 25344	$-\frac{16384}{9}$ -16384 -16384
	$-\frac{1}{144}$ $-\frac{1}{64}$ $-\frac{1}{32}$ 0	$ \frac{119}{144} \frac{125}{64} \frac{117}{32} 0 $	$-\frac{2191}{72} \\ -\frac{1271}{16} \\ -\frac{1053}{8} \\ 0$	$\frac{7429}{18}$ $\frac{2635}{2}$ 1800 0	$-\frac{17528}{9}$ -9176 -10632 0	$\frac{30464}{9}$ 24320 25344 0	$-\frac{16384}{9}$ -16384 -16384 0
	$-\frac{1}{144}$ $-\frac{1}{64}$ $-\frac{1}{7}$ $-\frac{1}{64}$	$\frac{119}{144}$ $\frac{125}{64}$ $\frac{117}{32}$ 0 $-\frac{109}{64}$	$-\frac{2191}{72} - \frac{1271}{16} - \frac{1053}{8} 0 \frac{899}{16}$	$\frac{\frac{7429}{18}}{\frac{2635}{2}}$ 1800 0 $-\frac{1581}{2}$	$-\frac{17528}{9}$ -9176 -10632 0 4960	$\frac{30464}{9}$ 24320 25344 0 -12416	$-\frac{16384}{9}$ -16384 -16384 0 8192
	$-\frac{1}{144}$ $-\frac{1}{64}$ $-1/32$ 0 $\frac{1}{64}$ $\frac{1}{96}$	$\frac{119}{144}$ $\frac{125}{64}$ $\frac{117}{32}$ 0 $-\frac{109}{64}$ $-\frac{101}{96}$	$-\frac{2191}{72} - \frac{1271}{16} - \frac{1053}{8} 0 \frac{899}{16} \frac{713}{24}$	$\frac{7429}{18}$ $\frac{2635}{2}$ 1800 0 $-\frac{1581}{2}$ $-\frac{1054}{3}$	$-\frac{17528}{9} \\ -9176 \\ -10632 \\ 0 \\ 4960 \\ \frac{5704}{3}$	$\frac{30464}{9}$ 24320 25344 0 -12416 - $\frac{12928}{3}$	$-\frac{16384}{9} - 16384 - 16384 0 8192 \frac{8192}{3}$

 $Q_{11}[1:15] =$

0	$\frac{-1}{72}$	$\frac{25}{144}$	$\frac{-1}{72}$	0	0	$\frac{215}{144}$	$\frac{-1357}{72}$	$\frac{215}{144}$	0	0	$\frac{-136}{3}$	$\frac{5149}{9}$	$\frac{-136}{3}$	0
0	1/24	$\frac{-25}{48}$	1/24	0	0	$\frac{-71}{16}$	$\frac{453}{8}$	$\frac{-71}{16}$	0	0	131	$\frac{-5179}{3}$	131	0
0	1/48	0	-1/48	0	0	$\frac{-31}{12}$	0	$\frac{31}{12}$	0	0	$\frac{310}{3}$	0	$\frac{-310}{3}$	0
0	-1/24	$\frac{25}{48}$	-1/24	0	0	$\frac{209}{48}$	$\frac{-1357}{24}$	$\frac{209}{48}$	0	0	$\frac{-243}{2}$	$\frac{5149}{3}$	$\frac{-243}{2}$	0
0	-1/16	0	1/16	0	0	$\frac{61}{8}$	0	$\frac{-61}{8}$	0	0	-295	0	295	0
0	$\frac{-1}{96}$	$\frac{-25}{96}$	$\frac{-1}{96}$	0	0	$\frac{157}{96}$	$\tfrac{1357}{48}$	$\frac{157}{96}$	0	0	-87	$\frac{-5149}{6}$	-87	0
0	$\frac{1}{72}$	$\frac{-25}{144}$	$\frac{1}{72}$	0	0	$\frac{-67}{48}$	$\frac{443}{24}$	$\frac{-67}{48}$	0	0	$\frac{209}{6}$	$\tfrac{-4792}{9}$	$\frac{209}{6}$	0
0	1/16	0	-1/16	0	0	$\frac{-59}{8}$	0	$\frac{59}{8}$	0	0	$\frac{533}{2}$	0	$\frac{-533}{2}$	0
0	1/32	$\frac{25}{32}$	1/32	0	0	$\frac{-153}{32}$	$\frac{-1365}{16}$	$\frac{-153}{32}$	0	0	246	$\frac{5263}{2}$	246	0
0	$\frac{1}{576}$	$\frac{25}{192}$	$\frac{29}{576}$	0	0	$\frac{-223}{576}$	$\frac{-1357}{96}$	$\frac{-3527}{576}$	0	0	$\frac{473}{18}$	$\tfrac{5149}{12}$	$\frac{2102}{9}$	0
0	-1/48	0	1/48	0	0	$\frac{55}{24}$	0	$\frac{-55}{24}$	0	0	$\frac{-431}{6}$	0	$\frac{431}{6}$	0
0	-1/32	$\frac{-25}{32}$	-1/32	0	0	$\frac{145}{32}$	$\frac{1357}{16}$	$\frac{145}{32}$	0	0	$\frac{-435}{2}$	$\frac{-5149}{2}$	$\frac{-435}{2}$	0
0	$\frac{-1}{192}$	$\frac{-25}{64}$	$\frac{-29}{192}$	0	0	$\frac{215}{192}$	$\frac{1373}{32}$	$\frac{3295}{192}$	0	0	$\frac{-443}{6}$	$\frac{-5361}{4}$	$\frac{-1730}{3}$	0
0	$\frac{1}{96}$	$\frac{25}{96}$	$\frac{1}{96}$	0	0	$\frac{-43}{32}$	$\frac{-415}{16}$	$\frac{-43}{32}$	0	0	$\frac{111}{2}$	$\frac{4057}{6}$	$\frac{111}{2}$	0
0	$\frac{1}{192}$	$\frac{25}{64}$	$\frac{29}{192}$	0	0	$\frac{-199}{192}$	$\frac{-1357}{32}$	$\tfrac{-2831}{192}$	0	0	$\frac{193}{3}$	$\frac{5149}{4}$	$\frac{2311}{6}$	0
0	$\frac{-1}{576}$	$\frac{-25}{192}$	$\frac{-29}{576}$	0	0	$\frac{167}{576}$	$\tfrac{1133}{96}$	$\frac{1903}{576}$	0	0	$\frac{-142}{9}$	$\frac{-1287}{4}$	$\frac{-1369}{18}$	0

 $Q_{11}[16:30] =$

0	480	$\tfrac{-46496}{9}$	480	0	0	$\tfrac{-31744}{9}$	$\tfrac{139264}{9}$	$\tfrac{-31744}{9}$	0	0	$\frac{65536}{9}$	$\tfrac{-131072}{9}$	$\frac{65536}{9}$	0 .
0	$\frac{-3760}{3}$	15872	$\frac{-3760}{3}$	0	0	$\tfrac{24064}{3}$	$\frac{-154624}{3}$	$\tfrac{24064}{3}$	0	0	$\frac{-32768}{3}$	65536	$\frac{-32768}{3}$	0
0	$\frac{-4960}{3}$	0	$\frac{4960}{3}$	0	0	$\tfrac{31744}{3}$	0	$\tfrac{-31744}{3}$	0	0	$\tfrac{-65536}{3}$	0	$\frac{65536}{3}$	0
0	936	$\tfrac{-46496}{3}$	936	0	0	$\frac{-14080}{3}$	$\tfrac{139264}{3}$	$\frac{-14080}{3}$	0	0	$\frac{16384}{3}$	$\tfrac{-131072}{3}$	$\frac{16384}{3}$	0
0	4400	0	-4400	0	0	-24064	0	24064	0	0	32768	0	-32768	0
0	1760	$\frac{23248}{3}$	1760	0	0	$\frac{-31744}{3}$	$\frac{-69632}{3}$	$\frac{-31744}{3}$	0	0	$\frac{65536}{3}$	$\frac{65536}{3}$	$\frac{65536}{3}$	0
0	$\frac{-1492}{9}$	$\frac{12428}{3}$	$\frac{-1492}{9}$	0	0	$\frac{7552}{9}$	$\frac{-93568}{9}$	$\frac{7552}{9}$	0	0	$\frac{-8192}{9}$	8192	$\frac{-8192}{9}$	0
0	-3448	0	3448	0	0	14080	0	-14080	0	0	-16384	0	16384	0
0	-4720	-25152	-4720	0	0	24064	89600	24064	0	0	-32768	-98304	-32768	0
0	$\frac{-5440}{9}$	$\frac{-11624}{3}$	$\frac{-29576}{9}$	0	0	$\tfrac{31744}{9}$	$\tfrac{34816}{3}$	$\tfrac{130304}{9}$	0	0	$\frac{-65536}{9}$	$\frac{-32768}{3}$	$\tfrac{-180224}{9}$	0
0	$\frac{2132}{3}$	0	$\frac{-2132}{3}$	0	0	$\frac{-7552}{3}$	0	$\frac{7552}{3}$	0	0	$\frac{8192}{3}$	0	$\frac{-8192}{3}$	0
0	3768	23248	3768	0	0	-14080	-69632	-14080	0	0	16384	65536	16384	0
0	$\frac{4880}{3}$	13136	$\frac{19804}{3}$	0	0	$\frac{-24064}{3}$	-45824	$\frac{-76928}{3}$	0	0	$\frac{32768}{3}$	49152	$\frac{90112}{3}$	0
0	$\frac{-2452}{3}$	-5528	$\frac{-2452}{3}$	0	0	$\frac{7552}{3}$	$\frac{44096}{3}$	$\frac{7552}{3}$	0	0	$\frac{-8192}{3}$	-12288	$\frac{-8192}{3}$	0
0	$\frac{-3928}{3}$	-11624	$\frac{-12188}{3}$	0	0	$\tfrac{14080}{3}$	34816	$\frac{41408}{3}$	0	0	$\frac{-16384}{3}$	-32768	$\frac{-45056}{3}$	0
0	$\frac{2612}{9}$	$\frac{8096}{3}$	$\frac{6952}{9}$	0	0	$\frac{-7552}{9}$	$\frac{-21824}{3}$	$\tfrac{-21440}{9}$	0	0	$\frac{8192}{9}$	6144	$\frac{22528}{9}$	0

 $Q_{12}[1:15] =$

$-\frac{1}{144}$	$\frac{-23}{288}$	$\frac{-23}{288}$	$\frac{1}{144}$	0	$\frac{-215}{288}$	$\frac{833}{96}$	$\frac{833}{96}$	$\frac{-215}{288}$	0	$\frac{68}{3}$	$\frac{-4741}{18}$	$\frac{-4741}{18}$	$\frac{68}{3}$	0]
-1/48	$\frac{23}{96}$	$\frac{23}{96}$	-1/48	0	$\frac{71}{32}$	$\frac{-835}{32}$	$\frac{-835}{32}$	$\frac{71}{32}$	0	$\frac{-131}{2}$	$\frac{2393}{3}$	$\frac{2393}{3}$	$\frac{-131}{2}$	0
$\frac{-1}{96}$	$\frac{-1}{96}$	$\frac{1}{96}$	$\frac{1}{96}$	0	$\frac{31}{24}$	$\frac{31}{24}$	$\frac{-31}{24}$	$\frac{-31}{24}$	0	$\frac{-155}{3}$	$\frac{-155}{3}$	$\frac{155}{3}$	$\frac{155}{3}$	0
1/48	$\frac{-23}{96}$	$\frac{-23}{96}$	1/48	0	$\frac{-209}{96}$	$\frac{835}{32}$	$\frac{835}{32}$	$\frac{-209}{96}$	0	$\frac{243}{4}$	$\tfrac{-9569}{12}$	$\tfrac{-9569}{12}$	$\frac{243}{4}$	0
1/32	1/32	-1/32	-1/32	0	$\frac{-61}{16}$	$\frac{-61}{16}$	$\frac{61}{16}$	$\frac{61}{16}$	0	$\frac{295}{2}$	$\frac{295}{2}$	$\frac{-295}{2}$	$\frac{-295}{2}$	0
$\frac{1}{192}$	$\frac{13}{96}$	$\frac{13}{96}$	$\frac{1}{192}$	0	$\tfrac{-157}{192}$	$\frac{-957}{64}$	$\frac{-957}{64}$	$\frac{-157}{192}$	0	$\frac{87}{2}$	$\tfrac{5671}{12}$	$\frac{5671}{12}$	$\frac{87}{2}$	0
$\frac{-1}{144}$	$\frac{23}{288}$	$\frac{23}{288}$	$\frac{-1}{144}$	0	$\frac{67}{96}$	$\frac{-273}{32}$	$\frac{-273}{32}$	$\frac{67}{96}$	0	$\frac{-209}{12}$	$\tfrac{8957}{36}$	$\tfrac{8957}{36}$	$\frac{-209}{12}$	0
-1/32	-1/32	1/32	1/32	0	$\frac{59}{16}$	$\frac{59}{16}$	$\frac{-59}{16}$	$\frac{-59}{16}$	0	$\frac{-533}{4}$	$\frac{-533}{4}$	$\frac{533}{4}$	$\frac{533}{4}$	0
$\frac{-1}{64}$	$\frac{-13}{32}$	$\frac{-13}{32}$	$\frac{-1}{64}$	0	$\frac{153}{64}$	$\frac{2883}{64}$	$\frac{2883}{64}$	$\frac{153}{64}$	0	-123	$\frac{-5755}{4}$	$\frac{-5755}{4}$	-123	0
$\tfrac{-1}{1152}$	$\frac{-19}{288}$	$\frac{-13}{144}$	$\frac{-29}{1152}$	0	$\frac{223}{1152}$	$\frac{8365}{1152}$	$\frac{11669}{1152}$	$\frac{3527}{1152}$	0	$\frac{-473}{36}$	$\frac{-16393}{72}$	$\frac{-23855}{72}$	$\frac{-1051}{9}$	0
$\frac{1}{96}$	$\frac{1}{96}$	$\frac{-1}{96}$	$\frac{-1}{96}$	0	$\frac{-55}{48}$	$\frac{-55}{48}$	$\frac{55}{48}$	$\frac{55}{48}$	0	$\frac{431}{12}$	$\frac{431}{12}$	$\frac{-431}{12}$	$\frac{-431}{12}$	0
$\frac{1}{64}$	$\frac{13}{32}$	$\frac{13}{32}$	$\frac{1}{64}$	0	$\frac{-145}{64}$	$\frac{-2859}{64}$	$\frac{-2859}{64}$	$\frac{-145}{64}$	0	$\frac{435}{4}$	1396	1396	$\frac{435}{4}$	0
$\frac{1}{384}$	$\frac{19}{96}$	$\frac{13}{48}$	$\frac{29}{384}$	0	$\frac{-215}{384}$	$\tfrac{-8453}{384}$	$\frac{-11533}{384}$	$\tfrac{-3295}{384}$	0	$\tfrac{443}{12}$	$\tfrac{16969}{24}$	$\tfrac{23003}{24}$	$\frac{865}{3}$	0
$\frac{-1}{192}$	$\frac{-13}{96}$	$\frac{-13}{96}$	$\frac{-1}{192}$	0	$\frac{43}{64}$	$\tfrac{873}{64}$	$\frac{873}{64}$	$\frac{43}{64}$	0	$\frac{-111}{4}$	$\frac{-2195}{6}$	$\frac{-2195}{6}$	$\frac{-111}{4}$	0
$\frac{-1}{384}$	$\frac{-19}{96}$	$\frac{-13}{48}$	$\frac{-29}{384}$	0	$\frac{199}{384}$	$\tfrac{8341}{384}$	$\frac{10973}{384}$	$\frac{2831}{384}$	0	$\frac{-193}{6}$	$\tfrac{-16219}{24}$	$\frac{-20069}{24}$	$\tfrac{-2311}{12}$	0
$-\frac{1}{1152}$	$\frac{19}{288}$	$\tfrac{13}{144}$	$\frac{29}{1152}$	0	$\tfrac{-167}{1152}$	$\frac{-6965}{1152}$	$\frac{-8701}{1152}$	$\frac{-1903}{1152}$	0	$\frac{71}{9}$	$\frac{12151}{72}$	$\frac{14321}{72}$	$\tfrac{1369}{36}$	0

150

<u>)</u> @ 9	<u>5</u>	$\frac{13450}{-9}$	$\frac{15620}{-9}$	<u>34</u>	$\frac{76}{9}$ 0	<u>37</u>	<u>76</u>	$\tfrac{36512}{9}$	$\tfrac{43456}{9}$	$\tfrac{10720}{9}$	0	$\frac{4096}{-9}$	$\tfrac{31744}{-9}$	$\frac{38912}{-9}$	$\frac{11264}{-9}$
[1:	15] =													
[0	$\frac{-1}{72}$	1/3	1/3	$\frac{-1}{72}$	0	$\frac{91}{72}$	$\frac{-2623}{72}$	$\frac{-2623}{72}$	$\frac{91}{72}$	0	$\frac{-196}{9}$	$\frac{10102}{9}$	$\frac{10102}{9}$	$\frac{-196}{9}$
	0	1/24	$\frac{-17}{16}$	$\frac{-23}{48}$	1/48	0	$\frac{-91}{24}$	$\frac{5585}{48}$	$\frac{211}{4}$	$\frac{-91}{48}$	0	$\frac{196}{3}$	$\frac{-10820}{3}$	-1651	$\frac{98}{3}$
	0	1/48	$\frac{-23}{48}$	$\frac{-17}{16}$	1/24	0	$\frac{-91}{48}$	$\frac{211}{4}$	$\frac{5585}{48}$	$\frac{-91}{24}$	0	$\frac{98}{3}$	-1651	$\tfrac{-10820}{3}$	$\frac{196}{3}$
	0	-1/24	$\frac{33}{32}$	$\frac{7}{32}$	$\frac{-1}{96}$	0	$\frac{91}{24}$	$\tfrac{-10765}{96}$	$\frac{-1175}{48}$	$\frac{91}{96}$	0	$\frac{-196}{3}$	$\frac{10249}{3}$	$\frac{4757}{6}$	$\frac{-49}{3}$
	0	-1/16	$\frac{13}{8}$	$\frac{13}{8}$	-1/16	0	$\frac{91}{16}$	$\frac{-2867}{16}$	$\frac{-2867}{16}$	$\frac{91}{16}$	0	-98	5641	5641	-98
	0	$\frac{-1}{96}$	$\frac{7}{32}$	$\frac{33}{32}$	-1/24	0	$\frac{91}{96}$	$\tfrac{-1175}{48}$	$\frac{-10765}{96}$	$\frac{91}{24}$	0	$\frac{-49}{3}$	$\frac{4757}{6}$	$\frac{10249}{3}$	$\frac{-196}{3}$
	0	$\frac{1}{72}$	$\frac{-19}{64}$	$\tfrac{-17}{576}$	$\frac{1}{576}$	0	$\frac{-91}{72}$	$\frac{18317}{576}$	$\frac{331}{96}$	$\frac{-91}{576}$	0	$\frac{196}{9}$	$\frac{-8408}{9}$	$\frac{-485}{4}$	$\frac{49}{18}$
	0	1/16	$\frac{-49}{32}$	$\frac{-27}{32}$	1/32	0	$\frac{-91}{16}$	$\frac{5337}{32}$	$\frac{751}{8}$	$\frac{-91}{32}$	0	98	-5100	$\frac{-6019}{2}$	49
	0	1/32	$\frac{-27}{32}$	$\frac{-49}{32}$	1/16	0	$\frac{-91}{32}$	$\frac{751}{8}$	$\tfrac{5337}{32}$	$\frac{-91}{16}$	0	49	$\frac{-6019}{2}$	-5100	98
	0	$\frac{1}{576}$	$\frac{-17}{576}$	$\frac{-19}{64}$	$\frac{1}{72}$	0	$\frac{-91}{576}$	$\frac{331}{96}$	$\frac{18317}{576}$	$\frac{-91}{72}$	0	$\frac{49}{18}$	$\frac{-485}{4}$	$\frac{-8408}{9}$	$\frac{196}{9}$
	0	-1/48	$\frac{71}{192}$	$\frac{29}{192}$	$\frac{-1}{192}$	0	$\frac{91}{48}$	$\frac{-7685}{192}$	$\tfrac{-1615}{96}$	$\frac{91}{192}$	0	$\frac{-98}{3}$	$\frac{3616}{3}$	$\tfrac{6481}{12}$	$\frac{-49}{6}$
	0	-1/32	3/4	3/4	-1/32	0	$\frac{91}{32}$	$\tfrac{-2623}{32}$	$\tfrac{-2623}{32}$	$\frac{91}{32}$	0	-49	$\frac{5051}{2}$	$\frac{5051}{2}$	-49
	0	$\frac{-1}{192}$	$\frac{29}{192}$	$\frac{71}{192}$	-1/48	0	$\tfrac{91}{192}$	$\tfrac{-1615}{96}$	$\frac{-7685}{192}$	$\frac{91}{48}$	0	$\frac{-49}{6}$	$\tfrac{6481}{12}$	$\frac{3616}{3}$	$\frac{-98}{3}$
	0	$\frac{1}{96}$	$\frac{-7}{64}$	$\frac{-23}{192}$	$\frac{1}{192}$	0	$\frac{-91}{96}$	$\frac{2705}{192}$	$\frac{211}{16}$	$\frac{-91}{192}$	0	$\frac{49}{3}$	$\frac{-3175}{6}$	$\frac{-1651}{4}$	$\frac{49}{6}$
	0	$\frac{1}{192}$	$\frac{-23}{192}$	$\frac{-7}{64}$	$\frac{1}{96}$	0	$\tfrac{-91}{192}$	$\frac{211}{16}$	$\frac{2705}{192}$	$\frac{-91}{96}$	0	$\frac{49}{6}$	$\frac{-1651}{4}$	$\frac{-3175}{6}$	$\frac{49}{3}$
	0	$\frac{-1}{576}$	$\frac{-1}{144}$	$\frac{-1}{144}$	$\frac{-1}{576}$	0	$\frac{91}{576}$	$\frac{-887}{576}$	$\frac{-887}{576}$	$\frac{91}{576}$	0	$\frac{-49}{18}$	$\frac{2881}{36}$	$\frac{2881}{36}$	$\frac{-49}{18}$

 Q_{21}

-240	$\frac{21088}{9}$	$\frac{21088}{9}$	-240	0	$\frac{15872}{9}$	$\frac{17920}{-3}$	$\frac{17920}{-3}$	$\frac{15872}{9}$	0	$\frac{32768}{-9}$	$\frac{32768}{9}$	$\frac{32768}{9}$	$\frac{32768}{-9}$	0]
$\frac{1880}{3}$	$\frac{21928}{-3}$	$\frac{21928}{-3}$	$\frac{1880}{3}$	0	$\frac{12032}{-3}$	21760	21760	$\frac{12032}{-3}$	0	$\frac{16384}{3}$	$\tfrac{81920}{-3}$	$\tfrac{81920}{-3}$	$\tfrac{16384}{3}$	0
$\frac{2480}{3}$	$\frac{2480}{3}$	$\frac{2480}{-3}$	$\frac{2480}{-3}$	0	$\frac{15872}{-3}$	$\frac{15872}{-3}$	$\frac{15872}{3}$	$\frac{15872}{3}$	0	$\frac{32768}{3}$	$\tfrac{32768}{3}$	$\frac{32768}{-3}$	$\frac{32768}{-3}$	0
-468	$\tfrac{21844}{3}$	$\frac{21844}{3}$	-468	0	$\frac{7040}{3}$	-20864	-20864	$\frac{7040}{3}$	0	$\frac{8192}{-3}$	$\tfrac{57344}{3}$	$\tfrac{57344}{3}$	$\frac{8192}{-3}$	0
-2200	-2200	2200	2200	0	12032	12032	-12032	-12032	0	-16384	-16384	16384	16384	0
-880	$\frac{14264}{-3}$	$\frac{14264}{-3}$	-880	0	$\frac{15872}{3}$	16896	16896	$\frac{15872}{3}$	0	$\frac{32768}{-3}$	$\frac{65536}{-3}$	$\tfrac{65536}{-3}$	$\frac{32768}{-3}$	0
$\frac{746}{9}$	$\frac{17896}{-9}$	$\frac{17896}{-9}$	$\frac{746}{9}$	0	$\frac{3776}{-9}$	$\tfrac{14336}{3}$	$\tfrac{14336}{3}$	$\frac{3776}{-9}$	0	$\frac{4096}{9}$	$\frac{32768}{-9}$	$\frac{32768}{-9}$	$\frac{4096}{9}$	0
1724	1724	-1724	-1724	0	-7040	-7040	7040	7040	0	8192	8192	-8192	-8192	0
2360	14936	14936	2360	0	-12032	-56832	-56832	-12032	0	16384	65536	65536	16384	0
$\frac{2720}{9}$	$\frac{20156}{9}$	$\tfrac{32224}{9}$	$\frac{14788}{9}$	0	$\frac{15872}{-9}$	$\frac{68096}{-9}$	$\frac{117376}{-9}$	$\frac{65152}{-9}$	0	$\frac{32768}{9}$	$\tfrac{81920}{9}$	$\tfrac{139264}{9}$	$\tfrac{90112}{9}$	0
$\frac{1066}{-3}$	$\frac{1066}{-3}$	$\frac{1066}{3}$	$\frac{1066}{3}$	0	$\frac{3776}{3}$	$\tfrac{3776}{3}$	$\frac{3776}{-3}$	$\frac{3776}{-3}$	0	$\frac{4096}{-3}$	$\frac{4096}{-3}$	$\frac{4096}{3}$	$\frac{4096}{3}$	0
-1884	-13508	-13508	-1884	0	7040	41856	41856	7040	0	-8192	-40960	-40960	-8192	0
$\frac{2440}{-3}$	$\frac{22144}{-3}$	$\frac{29606}{-3}$	$\frac{9902}{-3}$	0	$\frac{12032}{3}$	$\tfrac{80768}{3}$	$\tfrac{107200}{3}$	$\tfrac{38464}{3}$	0	$\frac{16384}{-3}$	$\frac{90112}{-3}$	$\frac{118784}{-3}$	$\frac{45056}{-3}$	0
$\frac{1226}{3}$	$\frac{9518}{3}$	$\frac{9518}{3}$	$\frac{1226}{3}$	0	$\frac{3776}{-3}$	-8608	-8608	$\frac{3776}{-3}$	0	$\frac{4096}{3}$	$\frac{22528}{3}$	$\frac{22528}{3}$	$\frac{4096}{3}$	0
$\frac{1964}{3}$	$\tfrac{19400}{3}$	$\tfrac{23530}{3}$	$\tfrac{6094}{3}$	0	$\frac{7040}{-3}$	$\frac{59264}{-3}$	$\frac{72928}{-3}$	$\frac{20704}{-3}$	0	$\tfrac{8192}{3}$	$\tfrac{57344}{3}$	$\tfrac{71680}{3}$	$\tfrac{22528}{3}$	0
$\frac{1306}{-9}$	$\frac{13450}{-9}$	$\frac{15620}{-9}$	$\frac{3476}{-9}$	0	$\frac{3776}{9}$	$\frac{36512}{9}$	$\tfrac{43456}{9}$	$\frac{10720}{9}$	0	$\frac{4096}{-9}$	$\tfrac{31744}{-9}$	$\frac{38912}{-9}$	$\frac{11264}{-9}$	0

 $Q_{12}[16:30] =$

 $Q_{21}[16:30] =$

0	$\frac{-1280}{9}$	$\frac{-31424}{3}$	$\frac{-31424}{3}$	$\frac{-1280}{9}$	0	0	$\frac{278528}{9}$	$\frac{278528}{9}$	0	0	0	$\tfrac{-262144}{9}$	$\tfrac{-262144}{9}$	0
0	$\frac{1280}{3}$	$\tfrac{103552}{3}$	$\tfrac{47776}{3}$	$\frac{640}{3}$	0	0	$\tfrac{-342016}{3}$	$\tfrac{-139264}{3}$	0	0	0	131072	$\frac{131072}{3}$	0
0	$\frac{640}{3}$	$\frac{47776}{3}$	$\frac{103552}{3}$	$\frac{1280}{3}$	0	0	$\tfrac{-139264}{3}$	$\tfrac{-342016}{3}$	0	0	0	$\frac{131072}{3}$	131072	0
0	$\frac{-1280}{3}$	-31104	-8176	$\frac{-320}{3}$	0	0	$\frac{278528}{3}$	$\frac{69632}{3}$	0	0	0	$\frac{-262144}{3}$	$\frac{-65536}{3}$	0
0	-640	-55936	-55936	-640	0	0	187392	187392	0	0	0	-196608	-196608	0
0	$\frac{-320}{3}$	-8176	-31104	$\frac{-1280}{3}$	0	0	$\tfrac{69632}{3}$	$\frac{278528}{3}$	0	0	0	$\frac{-65536}{3}$	$\frac{-262144}{3}$	0
0	$\frac{1280}{9}$	$\tfrac{69016}{9}$	$\frac{12904}{9}$	$\frac{160}{9}$	0	0	$\frac{-179968}{9}$	$\frac{-34816}{9}$	0	0	0	16384	$\frac{32768}{9}$	0
0	640	46816	30784	320	0	0	-139264	-97792	0	0	0	131072	98304	0
0	320	30784	46816	640	0	0	-97792	-139264	0	0	0	98304	131072	0
0	$\frac{160}{9}$	$\frac{12904}{9}$	$\frac{69016}{9}$	$\frac{1280}{9}$	0	0	$\frac{-34816}{9}$	$\frac{-179968}{9}$	0	0	0	$\frac{32768}{9}$	16384	0
0	$\frac{-640}{3}$	$\frac{-31732}{3}$	$\frac{-16528}{3}$	$\frac{-160}{3}$	0	0	28800	16640	0	0	0	-24576	-16384	0
0	-320	-23568	-23568	-320	0	0	69632	69632	0	0	0	-65536	-65536	0
0	$\frac{-160}{3}$	$\frac{-16528}{3}$	$\frac{-31732}{3}$	$\frac{-640}{3}$	0	0	16640	28800	0	0	0	-16384	-24576	0
0	$\frac{320}{3}$	$\frac{15148}{3}$	$\tfrac{11944}{3}$	$\frac{160}{3}$	0	0	$\frac{-42304}{3}$	$\frac{-34816}{3}$	0	0	0	12288	$\frac{32768}{3}$	0
0	$\frac{160}{3}$	$\frac{11944}{3}$	$\frac{15148}{3}$	$\frac{320}{3}$	0	0	$\frac{-34816}{3}$	$\frac{-42304}{3}$	0	0	0	$\frac{32768}{3}$	12288	0
0	$\frac{-160}{9}$	$\frac{-7444}{9}$	$\frac{-7444}{9}$	$\frac{-160}{9}$	0	0	$\frac{6976}{3}$	$\frac{6976}{3}$	0	0	0	-2048	-2048	0

 $Q_{22}[1:15] =$

$\frac{1}{144}$	$\frac{-23}{144}$	-1/3	$\frac{-23}{144}$	$\frac{1}{144}$	$\frac{-91}{144}$	$\frac{211}{12}$	$\frac{2623}{72}$	$\frac{211}{12}$	$\frac{-91}{144}$	$\frac{98}{9}$	$\frac{-1651}{3}$	$\frac{-10102}{9}$	$\frac{-1651}{3}$	$\frac{98}{9}$
-1/48	$\frac{49}{96}$	$\frac{37}{48}$	$\frac{11}{48}$	$\frac{-1}{96}$	$\frac{91}{48}$	$\frac{-1801}{32}$	$\tfrac{-8117}{96}$	$\tfrac{-2441}{96}$	$\frac{91}{96}$	$\frac{-98}{3}$	$\frac{5312}{3}$	$\frac{15773}{6}$	$\frac{4855}{6}$	$\frac{-49}{3}$
$\frac{-1}{96}$	$\frac{11}{48}$	$\frac{37}{48}$	$\frac{49}{96}$	-1/48	$\frac{91}{96}$	$\tfrac{-2441}{96}$	$\tfrac{-8117}{96}$	$\frac{-1801}{32}$	$\frac{91}{48}$	$\frac{-49}{3}$	$\frac{4855}{6}$	$\frac{15773}{6}$	$\frac{5312}{3}$	$\frac{-98}{3}$
1/48	$\frac{-95}{192}$	-5/8	$\frac{-5}{48}$	$\frac{1}{192}$	$\frac{-91}{48}$	$\frac{3467}{64}$	$\frac{13115}{192}$	$\frac{753}{64}$	$\frac{-91}{192}$	$\frac{98}{3}$	$\frac{-3351}{2}$	$\tfrac{-25255}{12}$	$\frac{-1553}{4}$	$\frac{49}{6}$
1/32	$\frac{-25}{32}$	$\frac{-13}{8}$	$\frac{-25}{32}$	1/32	$\frac{-91}{32}$	$\frac{347}{4}$	$\frac{2867}{16}$	$\frac{347}{4}$	$\frac{-91}{32}$	49	$\frac{-5543}{2}$	-5641	$\frac{-5543}{2}$	49
$\frac{1}{192}$	$\frac{-5}{48}$	-5/8	$\frac{-95}{192}$	1/48	$\frac{-91}{192}$	$\frac{753}{64}$	$\frac{13115}{192}$	$\frac{3467}{64}$	$\frac{-91}{48}$	$\frac{49}{6}$	$\frac{-1553}{4}$	$\tfrac{-25255}{12}$	$\frac{-3351}{2}$	$\frac{98}{3}$
$\frac{-1}{144}$	$\tfrac{163}{1152}$	$\frac{47}{288}$	$\frac{1}{72}$	$\frac{-1}{1152}$	$\frac{91}{144}$	$\frac{-5863}{384}$	$\tfrac{-20303}{1152}$	$\frac{-1895}{1152}$	$\tfrac{91}{1152}$	$\frac{-98}{9}$	$\frac{4106}{9}$	$\frac{37997}{72}$	$\frac{4267}{72}$	$\frac{-49}{36}$
-1/32	$\frac{47}{64}$	$\frac{19}{16}$	$\frac{13}{32}$	$\frac{-1}{64}$	$\frac{91}{32}$	$\frac{-5155}{64}$	$\tfrac{-8341}{64}$	$\frac{-2913}{64}$	$\frac{91}{64}$	-49	2501	$\tfrac{16219}{4}$	$\frac{5921}{4}$	$\frac{-49}{2}$
$\frac{-1}{64}$	13	19	$\frac{47}{100}$	-1/32	91	-2913	-8341	-5155	91	-49	5921	16219	2501	40
04	32	16	64	1/01	64	64	64	64	32	2	4	4	2301	-49
$\frac{-1}{1152}$	$\frac{1}{72}$	$\frac{16}{288}$	$\frac{163}{1152}$	$\frac{-1}{144}$	$\frac{64}{1152}$	$\frac{64}{-1895}$ $\frac{-1895}{1152}$	$\frac{-20303}{1152}$	64 -5863 384	$\frac{32}{91}$ $\frac{91}{144}$	$\frac{-49}{36}$	$\frac{4267}{72}$	$\frac{4}{37997}$	$\frac{4106}{9}$	$\frac{-98}{9}$
$\frac{-1}{1152}$ $\frac{1}{96}$	$\frac{1}{72}$ $\frac{-67}{384}$	$ \frac{47}{288} \underline{-25}{96} $	$ \frac{163}{1152} \frac{-7}{96} $	$\frac{-1}{144}$ $\frac{1}{384}$	$ \frac{91}{1152} \frac{-91}{96} $				$\frac{32}{144}$ $\frac{-91}{384}$	$\frac{-49}{36}$ $\frac{49}{3}$	$ \frac{4267}{72} \frac{-1759}{3} $	$ \frac{4}{37997} \frac{-20945}{24} $	$\frac{4106}{9}$ $\frac{-6383}{24}$	$\frac{-98}{9}$ $\frac{49}{12}$
$ \frac{-1}{1152} $ $ \frac{1}{96} $ $ \frac{1}{64} $	32 $\frac{1}{72}$ $\frac{-67}{384}$ $\frac{-23}{64}$	$ \frac{47}{288} \frac{-25}{96} -3/4 $	$ \begin{array}{r} 64 \\ \underline{163} \\ \overline{1152} \\ \underline{-7} \\ 96 \\ \underline{-23} \\ \overline{64} \end{array} $	$\frac{-1}{144}$ $\frac{1}{384}$ $\frac{1}{64}$	$ \begin{array}{r} 64 \\ \underline{91} \\ 1152 \\ \underline{-91} \\ 96 \\ \underline{-91} \\ 64 \\ \end{array} $	$ \begin{array}{r} 64 \\ -1895 \\ 1152 \\ \overline{1321} \\ 384 \\ \underline{633} \\ 16 \\ \hline $		$ \begin{array}{r} \overline{64} \\ \underline{-5863} \\ \overline{384} \\ \underline{3139} \\ \overline{384} \\ \underline{633} \\ \overline{16} \\ \end{array} $	32 91 144 -91 384 -91 64	$ \frac{-49}{36} $ $ \frac{49}{3} $ $ \frac{49}{2} $	$ \frac{4267}{72} \\ \frac{-1759}{3} \\ \frac{-4953}{4} $	$\frac{\frac{37997}{72}}{\frac{-20945}{24}}$ $\frac{-5051}{2}$	$\frac{4106}{9} \\ \frac{-6383}{24} \\ \frac{-4953}{4}$	$-49 \\ -98 \\ 9 \\ 12 \\ 49 \\ 12 \\ 49 \\ 2 \\ 12 \\ 12 \\ 12 \\ 12 \\ 12 \\ 12 \\ 12$
$\frac{-1}{1152}$ $\frac{1}{96}$ $\frac{1}{64}$ $\frac{1}{384}$	$ \frac{1}{72} \\ -67 \\ 384 \\ -23 \\ 64 \\ -7 \\ 96 $	$ \begin{array}{r} 16 \\ \frac{47}{288} \\ -25 \\ \overline{96} \\ -3/4 \\ \underline{-25} \\ \overline{96} \\ \end{array} $	$ \begin{array}{r} 64 \\ \underline{163} \\ 1152 \\ \underline{-7} \\ 96 \\ \underline{-23} \\ 64 \\ \underline{-67} \\ 384 \\ \end{array} $	$\frac{-1}{144}$ $\frac{1}{384}$ $\frac{1}{64}$ $\frac{1}{96}$	$ \begin{array}{r} 64 \\ \underline{91} \\ 1152 \\ \underline{-91} \\ \overline{96} \\ \underline{-91} \\ 64 \\ \underline{-91} \\ \overline{384} \\ \end{array} $	$ \begin{array}{r} 64 \\ -1895 \\ 1152 \\ \overline{)} \\ 7321 \\ 384 \\ \underline{633} \\ 16 \\ \underline{3139} \\ 384 \\ \end{array} $	$ \begin{array}{r} \overline{64} \\ \underline{-20303} \\ \overline{1152} \\ \underline{10915} \\ \overline{384} \\ \underline{2623} \\ \overline{32} \\ \underline{10915} \\ \overline{384} \\ \end{array} $	$ \begin{array}{r} \overline{64} \\ \underline{-5863} \\ \overline{384} \\ \underline{3139} \\ \overline{384} \\ \underline{633} \\ \overline{16} \\ \underline{7321} \\ \overline{384} \end{array} $	$\frac{32}{91} \\ \frac{91}{144} \\ \frac{-91}{384} \\ \frac{-91}{64} \\ \frac{-91}{96} \\ $	$ \frac{-49}{36} \\ \frac{49}{3} \\ \frac{49}{2} \\ \frac{49}{12} $	$ \frac{4267}{72} \\ \frac{-1759}{3} \\ \frac{-4953}{4} \\ \frac{-6383}{24} $	$ \frac{\frac{37997}{72}}{-20945} \frac{-5051}{2} \frac{-20945}{24} $	$\frac{4106}{9} \\ \frac{-6383}{24} \\ \frac{-4953}{4} \\ \frac{-1759}{3}$	$ \frac{-98}{9} \\ \frac{49}{12} \\ \frac{49}{2} \\ \frac{49}{3} $
$ \frac{-1}{1152} \\ \frac{1}{96} \\ \frac{1}{64} \\ \frac{1}{384} \\ \frac{-1}{192} $	$ \frac{1}{72} \\ -67 \\ 384 \\ -23 \\ 64 \\ -7 \\ 96 \\ 19 \\ 384 $	$ \begin{array}{r} 16 \\ \underline{47} \\ 288 \\ \underline{-25} \\ 96 \\ -3/4 \\ \underline{-25} \\ 96 \\ \underline{11} \\ 96 \\ \end{array} $	$ \begin{array}{r} 64 \\ \underline{163} \\ 1152 \\ \underline{-7} \\ 96 \\ \underline{-23} \\ 64 \\ \underline{-67} \\ 384 \\ \underline{11} \\ 192 \\ \end{array} $	$\frac{-1}{144}$ $\frac{1}{384}$ $\frac{1}{64}$ $\frac{1}{96}$ $\frac{-1}{384}$	$ \begin{array}{r} 64 \\ \underline{91} \\ \underline{1152} \\ \underline{-91} \\ \underline{96} \\ \underline{-91} \\ \underline{64} \\ \underline{-91} \\ \underline{384} \\ \underline{91} \\ \underline{192} \\ \end{array} $	$\begin{array}{r} 64\\ -\underline{1895}\\ 1152\\ \hline 7321\\ 384\\ \hline 633\\ \hline 16\\ 3139\\ \overline{384}\\ -\underline{841}\\ 128\\ \end{array}$	$\frac{-20303}{1152}$ $\frac{10915}{384}$ $\frac{2623}{32}$ $\frac{10915}{384}$ $\frac{-5237}{384}$	$\frac{-5863}{384}$ $\frac{3139}{384}$ $\frac{633}{16}$ $\frac{7321}{384}$ $\frac{-2441}{384}$	$\frac{32}{91} \\ \frac{91}{144} \\ \frac{-91}{384} \\ \frac{-91}{64} \\ \frac{-91}{96} \\ \frac{91}{384}$	$ \frac{2}{-49} \\ \frac{49}{36} \\ \frac{49}{2} \\ \frac{49}{12} \\ \frac{49}{12} \\ \frac{-49}{6} \\ \hline $	$ \frac{4267}{72} \\ -1759 \\ -4953 \\ -4953 \\ -6383 \\ 24 \\ 3077 \\ 12 $	$\begin{array}{r} \underline{37997} \\ \underline{-20945} \\ \underline{-20945} \\ \underline{24} \\ \underline{-5051} \\ \underline{2} \\ \underline{-20945} \\ \underline{24} \\ \underline{11303} \\ \underline{24} \end{array}$	$\frac{4106}{9}$ $\frac{-6383}{24}$ $\frac{-4953}{4}$ $\frac{-1759}{3}$ $\frac{4855}{24}$	$ \begin{array}{r} -49 \\ -98 \\ 9 \\ \hline 49 \\ 12 \\ \hline 49 \\ 2 \\ \hline 49 \\ 3 \\ \hline -49 \\ 12 \\ \end{array} $
$ \begin{array}{r} -1 \\ -1 \\ 1152 \\ \hline 196 \\ \hline 164 \\ \hline 1384 \\ \hline -1 \\ 192 \\ \hline -1 \\ 384 \\ \end{array} $	$\begin{array}{c} 32\\ \frac{1}{72}\\ -67\\ 384\\ -23\\ 64\\ -7\\ 96\\ 19\\ 384\\ \frac{19}{384}\\ \frac{11}{192} \end{array}$	$ \begin{array}{r} 16 \\ \underline{47} \\ 288 \\ -25 \\ 96 \\ -3/4 \\ \underline{-25} \\ 96 \\ \underline{11} \\ 96 \\ \underline{11} \\ 96 \\ \end{array} $	$ \begin{array}{r} 64\\ \underline{163}\\ 1152\\ \underline{-7}\\ 96\\ \underline{-23}\\ 64\\ \underline{-67}\\ 384\\ \underline{11}\\ 192\\ \underline{19}\\ 384\\ \end{array} $	$\frac{-1}{144}$ $\frac{1}{384}$ $\frac{1}{64}$ $\frac{1}{96}$ $\frac{-1}{3844}$ $\frac{-1}{192}$	$ \begin{array}{r} 64\\ \underline{91}\\ \underline{-91}\\ 96\\ \underline{-91}\\ 64\\ \underline{-91}\\ \overline{64}\\ \underline{-91}\\ \overline{384}\\ \underline{91}\\ \underline{192}\\ \underline{91}\\ \underline{384}\\ \end{array} $	$\begin{array}{r} 64\\ -\underline{1895}\\ 1152\\ \hline 1152\\ 384\\ \hline 633\\ 16\\ \hline 3139\\ 384\\ -\underline{841}\\ 128\\ -\underline{2441}\\ 384\\ \end{array}$	$\begin{array}{r} -\frac{64}{64} \\ -20303 \\ 1152 \\ \hline 10915 \\ 384 \\ \hline 2623 \\ 32 \\ \hline 10915 \\ 384 \\ -5237 \\ \hline 384 \\ -5237 \\ \hline 384 \\ \hline -5237 \\ \hline 384 \\ \end{array}$	$\begin{array}{r} \hline 64 \\ \hline -5863 \\ \hline 384 \\ \hline 3384 \\ \hline 633 \\ \hline 16 \\ \hline 7321 \\ \hline 384 \\ \hline -2441 \\ \hline 384 \\ \hline -2441 \\ \hline -841 \\ \hline 128 \\ \end{array}$	$\frac{32}{91}$ $\frac{91}{144}$ $\frac{-91}{384}$ $\frac{-91}{64}$ $\frac{-91}{96}$ $\frac{91}{384}$ $\frac{91}{192}$	$ \frac{-49}{36} \\ \frac{49}{3} \\ \frac{49}{2} \\ \frac{49}{12} \\ \frac{-49}{6} \\ \frac{-49}{12} $	$ \frac{4}{4267} \frac{-1759}{3} \frac{-4953}{4} \frac{-6383}{24} \frac{3077}{12} \frac{4855}{24} $	$\begin{array}{r} \hline 4 \\ \hline 37997 \\ \hline -20945 \\ \hline 24 \\ \hline -5051 \\ \hline 2 \\ \hline -20945 \\ \hline 24 \\ \hline 11303 \\ \hline 24 \\ \hline 11303 \\ \hline 24 \\ \hline 11303 \\ \hline 24 \\ \hline \end{array}$	$\frac{4106}{9} \\ -\frac{6383}{24} \\ -\frac{4953}{4} \\ -\frac{1759}{3} \\ \frac{4855}{24} \\ \frac{3077}{12} \\ -\frac{12}{3} \\ -1$	$ \begin{array}{r} -49 \\ -98 \\ 9 \\ 12 \\ 49 \\ 12 \\ 49 \\ 2 \\ -49 \\ 12 \\ -49 \\ 12 \\ -49 \\ 6 \end{array} $
$ \begin{array}{r} -1 \\ -1 \\ 1152 \\ 1 \\ 96 \\ 1 \\ 64 \\ -1 \\ 192 \\ -1 \\ 384 \\ -1 \\ 192 \\ -1 \\ 384 \\ -1 \\ 1152 \\ \end{array} $	$\begin{array}{c} 32\\ \frac{1}{72}\\ -67\\ 384\\ -23\\ 64\\ -7\\ 96\\ 19\\ 384\\ \frac{19}{384}\\ \frac{11}{192}\\ 5\\ \overline{1152} \end{array}$	$ \begin{array}{r} 16\\ \frac{47}{288}\\ -25\\ 96\\ -3/4\\ -25\\ 96\\ \frac{11}{96}\\ \frac{11}{96}\\ \frac{1}{144}\\ \end{array} $	$\begin{array}{r} 64\\ \underline{163}\\ 1152\\ -\underline{7}\\ \overline{96}\\ -\underline{23}\\ 64\\ \underline{-67}\\ \overline{384}\\ \underline{11}\\ 192\\ \underline{19}\\ 384\\ \underline{5}\\ 1152\\ \end{array}$	$\frac{-1}{144}$ $\frac{1}{384}$ $\frac{1}{64}$ $\frac{1}{96}$ $\frac{-1}{384}$ $\frac{-1}{192}$ $\frac{1}{1152}$	$\begin{array}{r} 64\\ \underline{91}\\ \underline{-91}\\ \underline{-91}\\ \overline{96}\\ \underline{-91}\\ \overline{64}\\ \underline{-91}\\ \overline{384}\\ \underline{91}\\ \underline{192}\\ \underline{91}\\ \overline{384}\\ \underline{-91}\\ \underline{-91}\\ \underline{1152}\\ \end{array}$	$\begin{array}{r} 64\\ -\underline{1895}\\ 1152\\ 1\overline{384}\\ \underline{7321}\\ 384\\ \underline{633}\\ 16\\ \underline{3139}\\ 384\\ \underline{-841}\\ 128\\ \underline{-2441}\\ 384\\ \underline{199}\\ 288\\ \end{array}$	$\begin{array}{r} -\frac{64}{64} \\ -\frac{20303}{1152} \\ \hline 10915 \\ \hline 384 \\ 2623 \\ \hline 322 \\ \hline 323 \\ \hline 10915 \\ \hline 384 \\ -\frac{5237}{384} \\ \hline -5237 \\ \hline 384 \\ \hline 887 \\ \hline 576 \end{array}$	$\begin{array}{r} -64 \\ -5863 \\ 384 \\ 3139 \\ 384 \\ \hline 633 \\ 16 \\ \hline 7321 \\ 384 \\ \hline -2441 \\ 384 \\ \hline -2441 \\ 384 \\ \hline -841 \\ \hline 128 \\ \hline 199 \\ 288 \end{array}$	$\begin{array}{r} 3\overline{2} \\ 91 \\ 144 \\ -91 \\ \overline{384} \\ -91 \\ \overline{64} \\ -91 \\ \overline{96} \\ 91 \\ \overline{384} \\ 91 \\ \overline{192} \\ -91 \\ 1152 \end{array}$	$\begin{array}{r} \hline 2 \\ \hline -49 \\ \hline 36 \\ \hline 49 \\ \hline 2 \\ \hline 49 \\ \hline 2 \\ \hline -49 \\ \hline 6 \\ \hline -49 \\ \hline 12 \\ \hline 49 \\ \hline 36 \\ \hline \end{array}$	$\begin{array}{r} \hline 4 \\ \hline 4 \\ \hline 267 \\ \hline 77 \\ \hline 77 \\ \hline 3 \\ \hline -1759 \\ \hline 3 \\ \hline -4953 \\ \hline 4 \\ \hline -6383 \\ \hline 4 \\ \hline -6383 \\ \hline 4 \\ \hline 4 \\ \hline 5 \\ \hline 2 \\ \hline 4 \\ \hline 5 \\ \hline 2 \\ \hline 4 \\ \hline 5 \\ \hline 2 \\ \hline 4 \\ \hline 5 \\ \hline 2 \\ \hline 7 \\ 7 \\$	$\begin{array}{r} -\frac{4}{37997} \\ -20945 \\ 24 \\ -5051 \\ -20945 \\ 24 \\ -20945 \\ 24 \\ 11303 \\ 24 \\ 11303 \\ 24 \\ -2881 \\ 36 \end{array}$	$\begin{array}{r} \frac{4106}{9}\\ -\frac{6383}{24}\\ -\frac{4953}{4}\\ -\frac{1759}{3}\\ \frac{4855}{24}\\ \frac{3077}{12}\\ -\frac{2783}{72}\end{array}$	$ \begin{array}{r} -49 \\ -98 \\ 9 \\ \hline 12 \\ 49 \\ 2 \\ \hline 49 \\ 3 \\ \hline -49 \\ 12 \\ \hline -49 \\ 12 \\ \hline -49 \\ 6 \\ \hline 49 \\ 36 \end{array} $

0	0	$\frac{-1}{72}$	$\tfrac{25}{144}$	$\frac{-1}{72}$	0	0	$\frac{215}{144}$	$\frac{-1357}{72}$	$\frac{215}{144}$	0	0	$\frac{-136}{3}$	$\frac{5149}{9}$	$\frac{-136}{3}$
0	0	-1/48	0	1/48	0	0	$\frac{31}{12}$	0	$\frac{-31}{12}$	0	0	$\frac{-310}{3}$	0	$\frac{310}{3}$
0	0	1/24	$\frac{-25}{48}$	1/24	0	0	$\frac{-71}{16}$	$\frac{453}{8}$	$\frac{-71}{16}$	0	0	131	$\frac{-5179}{3}$	131
0	0	$\frac{-1}{96}$	$\frac{-25}{96}$	$\frac{-1}{96}$	0	0	$\frac{157}{96}$	$\frac{1357}{48}$	$\frac{157}{96}$	0	0	-87	$\frac{-5149}{6}$	-87
0	0	1/16	0	-1/16	0	0	$\frac{-61}{8}$	0	$\frac{61}{8}$	0	0	295	0	-295
0	0	-1/24	$\frac{25}{48}$	-1/24	0	0	$\frac{209}{48}$	$\frac{-1357}{24}$	$\frac{209}{48}$	0	0	$\frac{-243}{2}$	$\frac{5149}{3}$	$\frac{-243}{2}$
0	0	$\frac{29}{576}$	$\frac{25}{192}$	$\frac{1}{576}$	0	0	$\frac{-3527}{576}$	$\frac{-1357}{96}$	$\frac{-223}{576}$	0	0	$\frac{2102}{9}$	$\frac{5149}{12}$	$\tfrac{473}{18}$
0	0	1/32	$\frac{25}{32}$	1/32	0	0	$\frac{-153}{32}$	$\frac{-1365}{16}$	$\frac{-153}{32}$	0	0	246	$\frac{5263}{2}$	246
0	0	-1/16	0	1/16	0	0	$\frac{59}{8}$	0	$\frac{-59}{8}$	0	0	$\frac{-533}{2}$	0	$\frac{533}{2}$
0	0	$\frac{1}{72}$	$\frac{-25}{144}$	$\frac{1}{72}$	0	0	$\frac{-67}{48}$	$\frac{443}{24}$	$\frac{-67}{48}$	0	0	$\frac{209}{6}$	$\frac{-4792}{9}$	$\frac{209}{6}$
0	0	$\frac{-29}{192}$	$\frac{-25}{64}$	$\frac{-1}{192}$	0	0	$\frac{3295}{192}$	$\frac{1373}{32}$	$\frac{215}{192}$	0	0	$\frac{-1730}{3}$	$\frac{-5361}{4}$	$\frac{-443}{6}$
0	0	-1/32	$\frac{-25}{32}$	-1/32	0	0	$\frac{145}{32}$	$\frac{1357}{16}$	$\frac{145}{32}$	0	0	$\frac{-435}{2}$	$\frac{-5149}{2}$	$\frac{-435}{2}$
0	0	1/48	0	-1/48	0	0	$\frac{-55}{24}$	0	$\frac{55}{24}$	0	0	$\frac{431}{6}$	0	$\frac{-431}{6}$
0	0	$\frac{29}{192}$	$\frac{25}{64}$	$\frac{1}{192}$	0	0	$\tfrac{-2831}{192}$	$\frac{-1357}{32}$	$\tfrac{-199}{192}$	0	0	$\frac{2311}{6}$	$\frac{5149}{4}$	$\frac{193}{3}$
0	0	$\frac{1}{96}$	$\frac{25}{96}$	$\frac{1}{96}$	0	0	$\frac{-43}{32}$	$\frac{-415}{16}$	$\frac{-43}{32}$	0	0	$\frac{111}{2}$	$\frac{4057}{6}$	$\frac{111}{2}$
0	0	$\frac{-29}{576}$	$\frac{-25}{192}$	$\frac{-1}{576}$	0	0	$\tfrac{1903}{576}$	$\frac{1133}{96}$	$\frac{167}{576}$	0	0	$\frac{-1369}{18}$	$\frac{-1287}{4}$	$\frac{-142}{9}$

 $Q_{31}[1:15] =$

$\frac{640}{9}$	$\tfrac{47776}{9}$	$\tfrac{31424}{3}$	$\tfrac{47776}{9}$	$\frac{640}{9}$	0	$\tfrac{-139264}{9}$	$\tfrac{-278528}{9}$	$\tfrac{-139264}{9}$	0	0	$\tfrac{131072}{9}$	$\tfrac{262144}{9}$	$\frac{131072}{9}$	0]
$\frac{-640}{3}$	-17472	$\tfrac{-75664}{3}$	$\tfrac{-24208}{3}$	$\frac{-320}{3}$	0	$\frac{171008}{3}$	$\tfrac{240640}{3}$	$\tfrac{69632}{3}$	0	0	-65536	$\tfrac{-262144}{3}$	$\tfrac{-65536}{3}$	0
$\frac{-320}{3}$	$\tfrac{-24208}{3}$	$\tfrac{-75664}{3}$	-17472	$\frac{-640}{3}$	0	$\frac{69632}{3}$	$\tfrac{240640}{3}$	$\frac{171008}{3}$	0	0	$\tfrac{-65536}{3}$	$\tfrac{-262144}{3}$	-65536	0
$\frac{640}{3}$	$\tfrac{47296}{3}$	19640	$\frac{12424}{3}$	$\frac{160}{3}$	0	$\tfrac{-139264}{3}$	$\tfrac{-174080}{3}$	$\frac{-34816}{3}$	0	0	$\frac{131072}{3}$	$\tfrac{163840}{3}$	$\frac{32768}{3}$	0
320	28288	55936	28288	320	0	-93696	-187392	-93696	0	0	98304	196608	98304	0
$\frac{160}{3}$	$\tfrac{12424}{3}$	19640	$\tfrac{47296}{3}$	$\frac{640}{3}$	0	$\frac{-34816}{3}$	$\frac{-174080}{3}$	$\frac{-139264}{3}$	0	0	$\frac{32768}{3}$	$\tfrac{163840}{3}$	$\frac{131072}{3}$	0
$\frac{-640}{9}$	$\frac{-11716}{3}$	$\frac{-40960}{9}$	$\frac{-6532}{9}$	$\frac{-80}{9}$	0	$\frac{89984}{9}$	$\tfrac{107392}{9}$	$\frac{17408}{9}$	0	0	-8192	$\frac{-90112}{9}$	$\frac{-16384}{9}$	0
-320	-23728	-38800	-15552	-160	0	69632	118528	48896	0	0	-65536	-114688	-49152	0
-160	-15552	-38800	-23728	-320	0	48896	118528	69632	0	0	-49152	-114688	-65536	0
$\frac{-80}{9}$	$\frac{-6532}{9}$	$\frac{-40960}{9}$	$\frac{-11716}{3}$	$\frac{-640}{9}$	0	$\frac{17408}{9}$	$\tfrac{107392}{9}$	$\frac{89984}{9}$	0	0	$\frac{-16384}{9}$	$\frac{-90112}{9}$	-8192	0
$\frac{320}{3}$	$\tfrac{16186}{3}$	$\tfrac{24130}{3}$	$\frac{8344}{3}$	$\frac{80}{3}$	0	-14400	-22720	-8320	0	0	12288	20480	8192	0
160	11944	23568	11944	160	0	-34816	-69632	-34816	0	0	32768	65536	32768	0
$\frac{80}{3}$	$\frac{8344}{3}$	$\tfrac{24130}{3}$	$\tfrac{16186}{3}$	$\frac{320}{3}$	0	-8320	-22720	-14400	0	0	8192	20480	12288	0
$\frac{-160}{3}$	-2578	$\tfrac{-13546}{3}$	$\frac{-6052}{3}$	$\frac{-80}{3}$	0	$\frac{21152}{3}$	$\tfrac{38560}{3}$	$\frac{17408}{3}$	0	0	-6144	$\frac{-34816}{3}$	$\frac{-16384}{3}$	0
$\frac{-80}{3}$	$\frac{-6052}{3}$	$\frac{-13546}{3}$	-2578	$\frac{-160}{3}$	0	$\frac{17408}{3}$	$\frac{38560}{3}$	$\frac{21152}{3}$	0	0	$\frac{-16384}{3}$	$\frac{-34816}{3}$	-6144	0
80	3802	$\frac{7444}{9}$	3802	<u>80</u>	0	$\frac{-3488}{3}$	$\frac{-6976}{3}$	$\frac{-3488}{3}$	0	0	1024	2048	1024	0

 $Q_{22}[16:30] =$

 $Q_{31}[16:30] =$

0	0	480	$\tfrac{-46496}{9}$	480	0	0	$\tfrac{-31744}{9}$	$\tfrac{139264}{9}$	$\tfrac{-31744}{9}$	0	0	$\frac{65536}{9}$	$\tfrac{-131072}{9}$	$\frac{65536}{9}$
0	0	$\frac{4960}{3}$	0	$\frac{-4960}{3}$	0	0	$\tfrac{-31744}{3}$	0	$\tfrac{31744}{3}$	0	0	$\frac{65536}{3}$	0	$\tfrac{-65536}{3}$
0	0	$\frac{-3760}{3}$	15872	$\frac{-3760}{3}$	0	0	$\frac{24064}{3}$	$\tfrac{-154624}{3}$	$\frac{24064}{3}$	0	0	$\tfrac{-32768}{3}$	65536	$\tfrac{-32768}{3}$
0	0	1760	$\frac{23248}{3}$	1760	0	0	$\tfrac{-31744}{3}$	$\frac{-69632}{3}$	$\tfrac{-31744}{3}$	0	0	$\frac{65536}{3}$	$\frac{65536}{3}$	$\frac{65536}{3}$
0	0	-4400	0	4400	0	0	24064	0	-24064	0	0	-32768	0	32768
0	0	936	$\frac{-46496}{3}$	936	0	0	$\frac{-14080}{3}$	$\tfrac{139264}{3}$	$\frac{-14080}{3}$	0	0	$\frac{16384}{3}$	$\frac{-131072}{3}$	$\frac{16384}{3}$
0	0	$\tfrac{-29576}{9}$	$\frac{-11624}{3}$	$\frac{-5440}{9}$	0	0	$\tfrac{130304}{9}$	$\tfrac{34816}{3}$	$\tfrac{31744}{9}$	0	0	$\tfrac{-180224}{9}$	$\frac{-32768}{3}$	$\frac{-65536}{9}$
0	0	-4720	-25152	-4720	0	0	24064	89600	24064	0	0	-32768	-98304	-32768
0	0	3448	0	-3448	0	0	-14080	0	14080	0	0	16384	0	-16384
0	0	$\frac{-1492}{9}$	$\frac{12428}{3}$	$\frac{-1492}{9}$	0	0	$\frac{7552}{9}$	$\frac{-93568}{9}$	$\frac{7552}{9}$	0	0	$\frac{-8192}{9}$	8192	$\frac{-8192}{9}$
0	0	$\frac{19804}{3}$	13136	$\frac{4880}{3}$	0	0	$\frac{-76928}{3}$	-45824	$\frac{-24064}{3}$	0	0	$\tfrac{90112}{3}$	49152	$\frac{32768}{3}$
0	0	3768	23248	3768	0	0	-14080	-69632	-14080	0	0	16384	65536	16384
0	0	$\frac{-2132}{3}$	0	$\frac{2132}{3}$	0	0	$\frac{7552}{3}$	0	$\frac{-7552}{3}$	0	0	$\frac{-8192}{3}$	0	$\frac{8192}{3}$
0	0	$\frac{-12188}{3}$	-11624	$\frac{-3928}{3}$	0	0	$\frac{41408}{3}$	34816	$\frac{14080}{3}$	0	0	$\frac{-45056}{3}$	-32768	$\frac{-16384}{3}$
0	0	$\frac{-2452}{3}$	-5528	$\frac{-2452}{3}$	0	0	$\frac{7552}{3}$	$\frac{44096}{3}$	$\frac{7552}{3}$	0	0	$\frac{-8192}{3}$	-12288	$\frac{-8192}{3}$
0	0	$\frac{6952}{9}$	$\frac{8096}{3}$	$\frac{2612}{9}$	0	0	$\tfrac{-21440}{9}$	$\frac{-21824}{3}$	$\frac{-7552}{9}$	0	0	$\frac{22528}{9}$	6144	$\frac{8192}{9}$

 $Q_{32}[1:15] =$

0	$\frac{1}{144}$	$\frac{-23}{288}$	$\frac{-23}{288}$	$\frac{1}{144}$	0	$\frac{-215}{288}$	$\frac{833}{96}$	$\frac{833}{96}$	$\frac{-215}{288}$	0	$\frac{68}{3}$	$\frac{-4741}{18}$	$\frac{-4741}{18}$	$\frac{68}{3}$
0	$\frac{1}{96}$	$\frac{1}{96}$	$\frac{-1}{96}$	$\frac{-1}{96}$	0	$\frac{-31}{24}$	$\frac{-31}{24}$	$\frac{31}{24}$	$\frac{31}{24}$	0	$\frac{155}{3}$	$\frac{155}{3}$	$\frac{-155}{3}$	$\frac{-155}{3}$
0	-1/48	$\frac{23}{96}$	$\frac{23}{96}$	-1/48	0	$\frac{71}{32}$	$\frac{-835}{32}$	$\frac{-835}{32}$	$\frac{71}{32}$	0	$\frac{-131}{2}$	$\frac{2393}{3}$	$\frac{2393}{3}$	$\frac{-131}{2}$
0	$\frac{1}{192}$	$\frac{13}{96}$	$\frac{13}{96}$	$\frac{1}{192}$	0	$\frac{-157}{192}$	$\frac{-957}{64}$	$\frac{-957}{64}$	$\frac{-157}{192}$	0	$\frac{87}{2}$	$\frac{5671}{12}$	$\frac{5671}{12}$	$\frac{87}{2}$
0	-1/32	-1/32	1/32	1/32	0	$\frac{61}{16}$	$\frac{61}{16}$	$\frac{-61}{16}$	$\frac{-61}{16}$	0	$\frac{-295}{2}$	$\frac{-295}{2}$	$\frac{295}{2}$	$\frac{295}{2}$
0	1/48	$\frac{-23}{96}$	$\frac{-23}{96}$	1/48	0	$\frac{-209}{96}$	$\frac{835}{32}$	$\frac{835}{32}$	$\frac{-209}{96}$	0	$\frac{243}{4}$	$\frac{-9569}{12}$	$\frac{-9569}{12}$	$\frac{243}{4}$
0	$\frac{-29}{1152}$	$\frac{-13}{144}$	$\frac{-19}{288}$	$\frac{-1}{1152}$	0	$\frac{3527}{1152}$	$\frac{11669}{1152}$	$\frac{8365}{1152}$	$\frac{223}{1152}$	0	$\frac{-1051}{9}$	$\frac{-23855}{72}$	$\frac{-16393}{72}$	$\frac{-473}{36}$
0	$\frac{-1}{64}$	$\frac{-13}{32}$	$\frac{-13}{32}$	$\frac{-1}{64}$	0	$\frac{153}{64}$	$\frac{2883}{64}$	$\frac{2883}{64}$	$\frac{153}{64}$	0	-123	$\frac{-5755}{4}$	$\frac{-5755}{4}$	-123
0	1/32	1/32	-1/32	-1/32	0	$\frac{-59}{16}$	$\frac{-59}{16}$	$\frac{59}{16}$	$\frac{59}{16}$	0	$\frac{533}{4}$	$\frac{533}{4}$	$\frac{-533}{4}$	$\frac{-533}{4}$
0	$\frac{-1}{144}$	$\frac{23}{288}$	$\frac{23}{288}$	$\frac{-1}{144}$	0	$\frac{67}{96}$	$\frac{-273}{32}$	$\frac{-273}{32}$	$\frac{67}{96}$	0	$\frac{-209}{12}$	$\tfrac{8957}{36}$	$\frac{8957}{36}$	$\frac{-209}{12}$
0	$\frac{29}{384}$	$\frac{13}{48}$	$\frac{19}{96}$	$\frac{1}{384}$	0	$\frac{-3295}{384}$	$\frac{-11533}{384}$	$\frac{-8453}{384}$	$\frac{-215}{384}$	0	$\frac{865}{3}$	$\frac{23003}{24}$	$\tfrac{16969}{24}$	$\frac{443}{12}$
0	$\frac{1}{64}$	$\frac{13}{32}$	$\frac{13}{32}$	$\frac{1}{64}$	0	$\frac{-145}{64}$	$\frac{-2859}{64}$	$\tfrac{-2859}{64}$	$\frac{-145}{64}$	0	$\frac{435}{4}$	1396	1396	$\frac{435}{4}$
0	$\frac{-1}{96}$	$\frac{-1}{96}$	$\frac{1}{96}$	$\frac{1}{96}$	0	$\frac{55}{48}$	$\frac{55}{48}$	$\frac{-55}{48}$	$\frac{-55}{48}$	0	$\frac{-431}{12}$	$\frac{-431}{12}$	$\frac{431}{12}$	$\frac{431}{12}$
0	$\frac{-29}{384}$	$\frac{-13}{48}$	$\frac{-19}{96}$	$\frac{-1}{384}$	0	$\frac{2831}{384}$	$\frac{10973}{384}$	$\tfrac{8341}{384}$	$\frac{199}{384}$	0	$\frac{-2311}{12}$	$\tfrac{-20069}{24}$	$\tfrac{-16219}{24}$	$\frac{-193}{6}$
0	$\frac{-1}{192}$	$\frac{-13}{96}$	$\frac{-13}{96}$	$\frac{-1}{192}$	0	$\frac{43}{64}$	$\frac{873}{64}$	$\frac{873}{64}$	$\frac{43}{64}$	0	$\frac{-111}{4}$	$\frac{-2195}{6}$	$\frac{-2195}{6}$	$\frac{-111}{4}$
L o	$\frac{29}{1152}$	$\tfrac{13}{144}$	$\frac{19}{288}$	$\frac{1}{1152}$	0	$\frac{-1903}{1152}$	$\frac{-8701}{1152}$	$\frac{-6965}{1152}$	$\tfrac{-167}{1152}$	0	$\tfrac{1369}{36}$	$\frac{14321}{72}$	$\frac{12151}{72}$	$\frac{71}{9}$

 $Q_{32}[16:30] =$

0	-240	$\frac{21088}{9}$	$\frac{21088}{9}$	-240	0	$\frac{15872}{9}$	$\frac{17920}{-3}$	$\frac{17920}{-3}$	$\frac{15872}{9}$	0	$\frac{32768}{-9}$	$\frac{32768}{9}$	$\frac{32768}{9}$	$\frac{32768}{-9}$
0	$\frac{2480}{-3}$	$\frac{2480}{-3}$	$\frac{2480}{3}$	$\frac{2480}{3}$	0	$\frac{15872}{3}$	$\frac{15872}{3}$	$\frac{15872}{-3}$	$\frac{15872}{-3}$	0	$\frac{32768}{-3}$	$\frac{32768}{-3}$	$\frac{32768}{3}$	$\frac{32768}{3}$
0	$\frac{1880}{3}$	$\frac{21928}{-3}$	$\frac{21928}{-3}$	$\frac{1880}{3}$	0	$\frac{12032}{-3}$	21760	21760	$\frac{12032}{-3}$	0	$\tfrac{16384}{3}$	$\frac{81920}{-3}$	$\frac{81920}{-3}$	$\frac{16384}{3}$
0	-880	$\frac{14264}{-3}$	$\frac{14264}{-3}$	-880	0	$\frac{15872}{3}$	16896	16896	$\frac{15872}{3}$	0	$\frac{32768}{-3}$	$\frac{65536}{-3}$	$\frac{65536}{-3}$	$\frac{32768}{-3}$
0	2200	2200	-2200	-2200	0	-12032	-12032	12032	12032	0	16384	16384	-16384	-16384
0	-468	$\tfrac{21844}{3}$	$\frac{21844}{3}$	-468	0	$\frac{7040}{3}$	-20864	-20864	$\frac{7040}{3}$	0	$\frac{8192}{-3}$	$\tfrac{57344}{3}$	$\tfrac{57344}{3}$	$\frac{8192}{-3}$
0	$\frac{14788}{9}$	$\frac{32224}{9}$	$\tfrac{20156}{9}$	$\frac{2720}{9}$	0	$\frac{65152}{-9}$	$\frac{117376}{-9}$	$\frac{68096}{-9}$	$\frac{15872}{-9}$	0	$\tfrac{90112}{9}$	$\tfrac{139264}{9}$	$\frac{81920}{9}$	$\frac{32768}{9}$
0	2360	14936	14936	2360	0	-12032	-56832	-56832	-12032	0	16384	65536	65536	16384
0	-1724	-1724	1724	1724	0	7040	7040	-7040	-7040	0	-8192	-8192	8192	8192
0	$\frac{746}{9}$	$\frac{17896}{-9}$	$\frac{17896}{-9}$	$\frac{746}{9}$	0	$\frac{3776}{-9}$	$\frac{14336}{3}$	$\tfrac{14336}{3}$	$\frac{3776}{-9}$	0	$\frac{4096}{9}$	$\frac{32768}{-9}$	$\frac{32768}{-9}$	$\frac{4096}{9}$
0	$\frac{9902}{-3}$	$\tfrac{29606}{-3}$	$\tfrac{22144}{-3}$	$\frac{2440}{-3}$	0	$\tfrac{38464}{3}$	$\tfrac{107200}{3}$	$\tfrac{80768}{3}$	$\tfrac{12032}{3}$	0	$\tfrac{45056}{-3}$	$\tfrac{118784}{-3}$	$\frac{90112}{-3}$	$\frac{16384}{-3}$
0	-1884	-13508	-13508	-1884	0	7040	41856	41856	7040	0	-8192	-40960	-40960	-8192
0	$\frac{1066}{3}$	$\frac{1066}{3}$	$\frac{1066}{-3}$	$\frac{1066}{-3}$	0	$\frac{3776}{-3}$	$\frac{3776}{-3}$	$\frac{3776}{3}$	$\frac{3776}{3}$	0	$\frac{4096}{3}$	$\frac{4096}{3}$	$\frac{4096}{-3}$	$\frac{4096}{-3}$
0	$\frac{6094}{3}$	$\frac{23530}{3}$	$\frac{19400}{3}$	$\frac{1964}{3}$	0	$\frac{20704}{-3}$	$\frac{72928}{-3}$	$\frac{59264}{-3}$	$\frac{7040}{-3}$	0	$\tfrac{22528}{3}$	$\frac{71680}{3}$	$\tfrac{57344}{3}$	$\frac{8192}{3}$
0	$\frac{1226}{3}$	$\frac{9518}{3}$	$\frac{9518}{3}$	$\frac{1226}{3}$	0	$\frac{3776}{-3}$	-8608	-8608	$\tfrac{3776}{-3}$	0	$\frac{4096}{3}$	$\frac{22528}{3}$	$\frac{22528}{3}$	$\frac{4096}{3}$
L o	$\tfrac{3476}{-9}$	$\frac{15620}{-9}$	$\tfrac{13450}{-9}$	$\frac{1306}{-9}$	0	$\tfrac{10720}{9}$	$\tfrac{43456}{9}$	$\tfrac{36512}{9}$	$\tfrac{3776}{9}$	0	$\tfrac{11264}{-9}$	$\frac{38912}{-9}$	$\frac{31744}{-9}$	$\frac{4096}{-9}$.

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