Adaptive Inverse Control of a vibrating Coupled Vessel-Riser System With Input Backlash

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Abstract—This article involves the adaptive inverse control of a coupled vessel-riser system with input backlash and system uncertainties. By introducing an adaptive inverse dynamics of backlash, the backlash control input is divided into a mismatch error and an expected control command, and then a novel adaptive inverse control strategy is established to elimnate vibration, tackle backlash, and compensate for system uncertainties. The bounded stability of the controlled system is analyzed and demonstrated by exploiting the Lyapunov's criterion. The simulation comparison experiments are finally presented to verify the feasibility and effectiveness of the control algorithm.

13 *Index Terms*—Adaptive inverse control, boundary control, 14 flexible risers, input backlash, vibration control.

I. INTRODUCTION

¹⁶ A DAPTIVE control as a common method for handling ¹⁷ Parametric uncertainty, provides techniques and algo-¹⁸ rithms for parameter estimation and is introduced in many ¹⁹ literatures [1]–[6]. In recent decades, significant advance-²⁰ ments in adaptive control for the nonlinear systems have ²¹ been documented. To list some, in [7] and [8], switched ²² nonlinear systems were stabilized by developing an adap-²³ tive neural tracking control and the semiglobal boundedness ²⁴ was ensured. In [9]–[11], an adaptive finite-time convergence

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control for uncertain nonlinear systems was investigated via 25 parameter estimation. Liu et al. [12], Zhang et al. [13], and Liu et al. [14] explored an adaptive neural control method-27 ologies for uncertain nonlinear systems subject to constraints. 28 In [15] and [16], an adaptive fuzzy sliding-mode control was 29 designed for nonlinear systems to compensate for unknown 30 upper bounds. However, the aforesaid results were just con-31 cerned on the adaptive control analysis of ordinary differential 32 equation systems and it cannot be applied in partial differential 33 equation systems. 34

The flexible marine riser is crucial in the exploitation of 35 ocean petroleum and natural gas resources, and receives more 36 and more attention in recent years [17]. Generally, vibra-37 tion and deformation appear in flexible risers due to the face 38 of harsh conditions, however, the undesired vibration may 39 shorten service life, lead to fatigue failure, and even cause 40 serious environmental pollution [18]. Hence, how to develop 41 the effective active control strategies [19], [20] for eliminat-42 ing the riser's vibration has attracted many scholars, and 43 they have presented many control approaches including model 44 reduction method [21]–[23] and boundary control [24]–[27]. 45 Boundary control, the implementation of which is generally 46 considered to be nonintrusive actuation and sensing [28]-[38], 47 is more realistic and effective for stabilizing flexible riser 48 systems due to the circumvention of control spillover result-49 ing from the reduced-order model method [39]-[41], and 50 the recent developments have been documented. To men-51 tion a few examples, in [18], a boundary adaptive control 52 framework was raised for the stabilization of an uncertain 53 flexible riser system. In [42], an anti-disturbance control 54 was put forward to damp the riser's oscillation and real-55 ize the extrinsic disturbance elimination. In [43], the riser 56 vibration decrease was achieved using the presented bound-57 ary robust output feedback control, which simultaneously 58 ensured the controlled system state's convergence. In [44], 59 three-dimensional (3-D) extensible risers were exponentially 60 stabilized under the designed boundary control scheme. 61 Meanwhile, the well-posedness and stability analysis were 62 also presented. In [45], boundary controllers were proposed to 63 address the large in-plane deflection reduction and the global 64 and exponential stabilization of unshearable and extensible 65 flexible risers subject to sea loads. In [46], 3-D longitudi-66 nal and transverse vibrations of flexible risers with bending couplings were suppressed via boundary simultaneous controllers. However, note that the above-mentioned approaches 69 were confined to suppress vibrations, which are invalid 70

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⁷¹ for flexible riser systems with nonsmooth input nonlinear ⁷² constraints.

In recent years, significant attention has focused on con-73 74 trol of nonlinear systems subject to input nonlinearities, such 75 as backlash, deadzone, saturation, and hysteresis [47]-[53], 76 which are common and tough issues in mechanical 77 connections, piezoelectric translators, and hydraulic servo 78 valves [54]–[56]. Recently, boundary control has achieved 79 rapid development on handling the input constraints in flex-80 ible riser systems [57]-[59]. In [60], an input-restricted riser 81 system was significantly stabilized by using anti-saturation 82 vibration control strategies. In [57], backstepping technique 83 was employed to construct an adaptive control for riser 84 systems to resolve the oscillation elimination, input saturation, 85 and output constraint. Further, anti-saturation control strate-⁸⁶ gies were presented to restrain the oscillation of flexible risers 87 with input constraint by introducing the Nussbaum function 88 in [58]. Note that the chattering phenomenon caused by the 89 discontinuous sign function in [57] was removed. In [59], 90 hybrid input deadzone and saturation constraint issue in the ⁹¹ riser system was addressed by exploiting the auxiliary function 92 to propose a boundary control law. However, in the aforemen-⁹³ tioned research, the design was confined to eliminate vibration, ⁹⁴ tackle input saturation, or eliminate mixed input deadzone and 95 saturation in the riser system.

However, the effect of the input backlash nonlinearity char-96 97 acteristic was not considered in these mentioned literatures. 98 Backlash, which describes a dynamical input-output relation-99 ship, exists in various physical systems and devices, such 100 as electronic relay circuits, mechanical actuators, electro-¹⁰¹ magnetism, biology optics, and other areas [50]. The effects 102 of input backlash nonlinearity can seriously deteriorate system 103 performance, give rise to undesirable inaccuracy or oscilla-104 tions, and even result in closed-loop instability [50]. In [61], 105 an adaptive control with an adjustable update law were 106 established by discomposing and treating the backlash as "disturbance-like" items. To the best of our knowledge, despite 107 108 great advances in boundary control design for flexible riser 109 systems subjected to input nonlinearities have been made, the 110 framework on how to develop an adaptive inverse control for 111 tackling the simultaneous effects of the input backlash non-112 linearity and uncertainties in the riser system has not been 113 reported thus far in the literature. It is what to motivate this 114 research and, in this article, we consider and investigate a 115 vessel-riser system depicted in Fig. 1 simultaneously affected ¹¹⁶ by input backlash and system uncertainties.

The main contributions of this article are summarized as follows: 1) the input backlash is reformulated in a sum of a desired control signal and a mismatch error by introducing an adaptive inverse backlash dynamics, rather than resolving and visualizing the backlash as disturbance-like items and 2) a new adaptive inverse control strategy with online update laws as is developed to achieve the vibration attenuation, backlash elimination, and uncertainties compensation for the coupled vessel-riser system.

This article is laid out as follows: a dynamical model transformation of the system and preliminaries are arranged in Section II. Reserved to the stability analysis and the controller



design. Section IV makes and analyzes numerical simulations. ¹²⁹ Finally, Section V draws a conclusion. ¹³⁰

II. PROBLEM STATEMENT 131

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A. System Model

As displayed in Fig. 1, p(s, t) describes the vibrational ¹³³ deflection of the riser whose length is w. d(t) denotes the ¹³⁴ extraneous disturbance acting on the vessel whose mass and ¹³⁵ damping coefficient are m > 0 and $d_a > 0$. f(s, t) denotes the ¹³⁶ distributed disturbance which is acted on the riser, and $\tau(t)$ ¹³⁷ represents control input which is put on the vessel. In addition, ¹³⁸ some notations for simplification are presented: $(*) = \partial(*)/\partial t$, ¹³⁹ $(*)'' = \partial(*)/\partial s$, $(*)' = \partial^2(*)/\partial s \partial t$, $(*)''' = \partial^2(*)/\partial s^2$, ¹⁴⁰ $(*)''' = \partial^2(*)/\partial s^3$, $(*)'''' = \partial^2(*)/\partial s^4$, and $(*) = \partial^2(*)/\partial t^2$. ¹⁴¹

In this article, the goal is to propose an adaptive inverse control for damping the vibration deflection and simultaneously 143 handling the backlash nonlinearity and system uncertainties. 144 To realize this objective, we model the dynamics of the 145 considered vessel-riser system as [18] 146

$$\rho \ddot{p} + EIp'''' - Tp'' + c\dot{p} - f = 0, \ 0 < s < w$$
(1) 147

$$p(0,t) = p'(0,t) = p''(w,t) = 0$$
(2) 140

$$m\ddot{p}(w,t) + Tp'(w,t) - EIp'''(w,t)$$
 149

$$+ d_a \dot{p}(w, t) = \tau(t) + d(t)$$
 (3) 15

where EI > 0, c > 0, $\rho > 0$, and T > 0 express the bending stiffness, damping coefficient, mass per unit length, and the tension of the riser, respectively. (153)

B. Input Backlash Analysis

For the convenience of adaptive inverse control design, we 155 present the expression of the backlash nonlinearity [62] shown 156



Fig. 2. Backlash nonlinearity.

157 in Fig. 2 as follows:

$$\tau(t) = \mathcal{B}(\gamma)$$

$$= \begin{cases} \varrho(\gamma(t) - B), & \text{if } \dot{\gamma} > 0 \text{ and } \tau(t) = \varrho(\gamma(t) - B) \\ \varrho(\gamma(t) + B), & \text{if } \dot{\gamma} < 0 \text{ and } \tau(t) = \varrho(\gamma(t) + B) \\ \tau(t_{-}), & \text{otherwise} \end{cases}$$

$$(4)$$

¹⁶¹ where $\tau(t)$ denotes the control input, $\gamma(t)$ expresses the ¹⁶² expected control to be developed, ρ represents the slope, ¹⁶³ *B* denotes the "crossing," and $\tau(t_{-})$ shows no change in $\tau(t)$.

164 C. Preliminaries

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We provide the related assumptions, lemmas, and remarks for facilitating the subsequent analysis and design.

For disturbances d(t) and f(s, t) that possess finite energy, that is, $d(t), f(s, t) \in \mathcal{L}_{\infty}$ [63]–[65], we make an assumption for these disturbances in the following.

Assumption 1: We assume that extraneous disturbances d(t)and f(s, t) acting on the vessel and riser are bounded and there trace exist $D, F \in \mathbb{R}^+$ satisfying $|d(t)| \leq D$, and $|f(s, t)| \leq F$, $\forall (s, t) \in [0, w] \times [0, +\infty)$.

174 Lemma 1 [66]: Let $\delta_1(s, t), \delta_2(s, t) \in \mathbb{R}, \phi > 0$ with 175 $(s, t) \in [0, w] \times [0, +\infty)$, then

176
$$\delta_1 \delta_2 \le \frac{1}{\phi} \delta_1^2 + \phi \delta_2^2. \tag{5}$$

Lemma 2 [67]: Let $\delta(s, t) \in \mathbb{R}$ be under the condition $\delta(0, t) = 0$, where $(s, t) \in [0, w] \times [0, +\infty)$, then

$$\delta^2 \le w \int_0^w {\delta'}^2 ds. \tag{6}$$

Lemma 3 [68]: The following inequality is provided to the derive our main results:

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$$0 \le |\varpi(t)| - \varpi(t) \tanh(\varpi(t)) \le 0.2785.$$
 (7)

III. CONTROL DESIGN

System parameters T, EI, d_a , and m, and upper bounds of the disturbance D are utilized in the control design. However, these parameters may be unavailable in real system, thus it will bring a challenge for the control design and even make control approaches unusable. In this section, we present a new adaptive inverse controller to handle system uncertainties. Moreover, this control scheme can stabilize the vessel-riser system and eliminate the backlash nonlinearity. Subsequently, ¹⁹¹ we analyze the closed-loop system's stability in theory. ¹⁹²

A. Boundary Adaptive Inverse Control With Input Backlash 193 Invoking [62], the inverse backlash is presented as follows: 194

$$\gamma^{*}(t) = \mathcal{BI}(\tau_{d}(t)) = \begin{cases} \frac{1}{\varrho}\tau_{d}(t) + B, & \text{if } \dot{\tau}_{d}(t) > 0\\ \frac{1}{\varrho}\tau_{d}(t) - B, & \text{if } \dot{\tau}_{d}(t) < 0\\ \gamma^{*}(t_{-}), & \text{otherwise} \end{cases}$$
(8) 195

where $\mathcal{BI}(\cdot)$ represents a backlash inverse function and $\tau_d(t)$ ¹⁹⁶ denotes the expected control command.

Then, according to the above analysis and [62], we propose 198 an adaptive inverse of backlash as 199

$$\dot{\alpha}(t) = \dot{\tau}_d(t) - \frac{\beta}{B_m} |\dot{\tau}_d(t)| \alpha(t)$$
 (9) 200

$$\gamma(t) = \widehat{\mathcal{BI}}(\tau_d(t)) = \frac{1}{\varrho} \tau_d(t) + \widehat{\lambda} \beta \alpha(t)$$
(10) 201

where β denotes a positive gain parameter, B_m denotes a nominal backlash value, $\widehat{BI}(\cdot)$ denotes an adaptive backlash inverse compensator, and $\widehat{\lambda}$ represents an adaption to adjust B_m so as to match the actual backlash spacing B with $B = \lambda B_m$, $\lambda \leq \varepsilon, \varepsilon \in \mathbb{R}^+$, and $\widetilde{\lambda} = \widehat{\lambda} - \lambda$.

Meanwhile, we bring a mismatch error as follows:

$$\tau(t) = \mathcal{B}\big(\mathcal{BI}(\tau_d(t))\big) = \tau_d(t) + \tau_e(t) \tag{11} 200$$

where we formulate the mismatch error $\tau_e(t)$ as

$$\tau_e(t) = \rho \lambda B_m sgn(\dot{\tau}_d(t)) = \rho \lambda \beta \alpha(t).$$
(12) 210

Invoking (9)–(12), we then rewrite (3) as

$$\ddot{p}(w,t) + Tp'(w,t) - EIp'''(w,t) + d_a \dot{p}(w,t) - d(t)$$

$$= \tau(t) = \mathcal{B}(\gamma(t))$$
(13) 213

Then, an adaptive inverse control is proposed as

$$\gamma(t) = \widehat{\mathcal{BI}}(\tau_d(t)) = \frac{1}{\varrho} \tau_d(t) + \widehat{\lambda} \beta \alpha(t)$$
(14) 215

where $\tau_d(t)$ is designed as

$$\tau_{d}(t) = -\kappa_{1}x(t) - \widehat{EI}p'''(w, t) + \widehat{T}p'(w, t) + \widehat{d}_{a}\dot{p}(w, t)$$

$$+ \widehat{m}(\kappa_{2}\dot{p}'''(w, t) - \kappa_{3}\dot{p}'(w, t)) - \tanh(x(t))\hat{D}$$
(15) 218

where $\kappa_1, \kappa_2, \kappa_3 > 0, \widehat{D}, \widehat{T}, \widehat{EI}, \widehat{d}_a$, and \widehat{m} are the estimated ²¹⁹ values of *D*, *T*, *EI*, d_a , and *m*, and we define the auxiliary ²²⁰ variable x(t) as ²²¹

$$x(t) = \dot{p}(w, t) - \kappa_2 p'''(w, t) + \kappa_3 p'(w, t).$$
(16) 222

Now, we present the following inverse backlash dynamics: 223

$$\dot{\alpha}(t) = \dot{\tau}_d(t) - \frac{\beta}{B_m} |\dot{\tau}_d(t)| \alpha(t) \tag{17} \quad 224$$

and the estimation $\hat{\lambda}$ is obtained from the following:

$$\widehat{\lambda} = -\frac{\varrho}{v}\beta x(t)\alpha(t) - \widehat{\lambda}.$$
(18) 226

At this time, the adaptive laws are presented when system $_{227}$ parameters *T*, *EI*, d_a , and *m*, and upper bounds of the $_{228}$ disturbance *D* are not available $_{229}$

$$\widehat{D} = x(t) \tanh(x(t)) - \varsigma_1 \widehat{D}$$
(19) 230

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$$\widehat{T} = -x(t)p'(w,t) - \varsigma_2 \widehat{T}$$
(20)

$$\widehat{EI} = x(t)p'''(w,t) - \varsigma_3 \widehat{EI}$$
(21)

$$\widehat{d}_a = -x(t)\dot{p}(w,t) - \varsigma_4 \widehat{d}_a \tag{22}$$

$$\widehat{m} = x(t) \left(\kappa_3 \dot{p}'(w, t) - \kappa_2 \dot{p}'''(w, t) \right) - \varsigma_5 \widehat{m}$$
(23)

235 where $\zeta_i > 0, i = 1 \cdots 5$.

²³⁶ The estimation errors are defined as

237

$$\widetilde{D} = \widehat{D} - D, \quad \widetilde{T} = \widehat{T} - T,$$
238

$$\widetilde{EI} = \widehat{EI} - EI, \quad \widetilde{d}_a = \widehat{d}_a - d_a, \quad \widetilde{m} = \widehat{m} - m. \quad (24)$$

Remark 1: Note that available boundary signals p'''(w, t), p'(w, t), p(w, t), $\dot{p}'''(w, t)$, $\dot{p}'(w, t)$, and $\dot{p}(w, t)$ consist of the proposed control law (15), where p'''(w, t) is measured by shear force sensors, p'(w, t) is measured by inclinometers, and μ_{43} p(w, t) is measured by laser displacement sensors. Moreover, we can use the backward difference algorithm to achieve the one-order time derivative of some measurable signals $\dot{p}'''(w, t)$, μ_{46} $\dot{p}'(w, t)$, and $\dot{p}(w, t)$ in the designed controller.

Now, we will present the stability analysis for deriving our main results.

249 B. Stability Proof

250 Select the Lyapunov candidate function as

$$\Theta(t) = \Theta_1(t) + \Theta_2(t) + \Theta_3(t) + \Theta_4(t)$$

252 where

 $\Theta_2(t) = \frac{\sigma}{2}mx^2(t) \tag{27}$

$$\Theta_3(t) = \frac{\sigma}{2}\widetilde{D}^2 + \frac{\sigma}{2}\widetilde{T}^2 + \frac{\sigma}{2}\widetilde{E}I^2 + \frac{\sigma}{2}\widetilde{d}_a^2 + \frac{\sigma}{2$$

$$\Theta_4(t) = \psi \rho \int_0^w s \dot{p} p' ds \tag{29}$$

259 with ξ , σ , $\psi > 0$ being constants.

Lemma 4: The constructed function (25) is positive

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$$0 \leq \iota_1[\Theta_1(t) + \Theta_2(t) + \Theta_3(t)] \leq \Theta(t)$$
262

$$< \iota_2[\Theta_1(t) + \Theta_2(t) + \Theta_3(t)]$$
(30)

263 where $\iota_1, \iota_2 > 0$.

264 Proof: We invoke Lemma 1 and combine (29) to derive

$$|\Theta_4(t)| \le \frac{\psi \rho w}{2} \int_0^w \left(\dot{p}^2 + p'^2\right) ds \le \vartheta \Theta_1(t) \tag{31}$$

²⁶⁶ where $\vartheta = (\psi \rho w / \min(\xi \rho, \xi T, \xi EI)).$

267 We select ξ and ψ appropriately to satisfy the following:

$$\xi > \frac{\psi \rho w}{\min(\rho, T, EI)}.$$
(32)

Equation (32) indicates $0 < \vartheta < 1$. Then, rearranging (31) 270 and adding (26) gives

$$0 \le (1 - \vartheta)\Theta_1(t) \le \Theta_1(t) + \Theta_4(t) \le (1 + \vartheta)\Theta_1(t).$$
(33)

Invoking (25) and (33) yields

$$0 \le \iota_1[\Theta_1(t) + \Theta_2(t) + \Theta_3(t)] \le \Theta(t)$$
²⁷³

$$\leq \iota_2[\Theta_1(t) + \Theta_2(t) + \Theta_3(t)]$$
(34) 274

where $\iota_1 = \min(1 - \vartheta, 1) > 0$ and $\iota_2 = \max(1 + \vartheta, 1) > 1$. *Lemma 5:* The time derivative of (25) is upper bounded as ²⁷⁶

$$\dot{\Theta}(t) \le -\iota\Theta(t) + \chi \tag{35} \ _{277}$$

where $\iota, \chi > 0$.

(25)

(28)

Proof: We differentiate (25) to derive

$$\dot{\Theta}(t) = \dot{\Theta}_1(t) + \dot{\Theta}_2(t) + \dot{\Theta}_3(t) + \dot{\Theta}_4(t).$$
 (36) 280

Invoking (1) and applying Lemma 1, $\dot{\Theta}_1(t)$ is obtained as 281

$$\dot{\Theta}_{1}(t) \leq \frac{\xi EI}{2\kappa_{2}} x^{2}(t) - \frac{\xi EI\kappa_{2}}{2} p'''^{2}(w, t) - \frac{\xi EI}{2\kappa_{2}} \dot{p}^{2}(w, t)$$

$$\xi EI\kappa_{2}^{2} \qquad \xi EI\kappa_{2}$$

$$-\frac{\xi EI \kappa_3}{2\kappa_2} p'^2(w,t) + (\xi T - \frac{\xi EI \kappa_3}{\kappa_2}) p'(w,t) \dot{p}(w,t)$$
 283

$$+\xi EI\kappa_{3}p'''(w,t)p'(w,t) - (c-v_{1})\xi \int_{0} \dot{p}^{2}ds \qquad 284$$

$$\frac{\xi}{\nu_1} \int_0^u f^2 ds,$$
 (37) 280

where $v_1 > 0$. Combining (11)–(16), we derive $\dot{\Theta}_2(t)$ as 286

$$\times \left[Ip(w,t) - EIp''(w,t) \right]$$
286

$$+ a_{a}p(w, t) + m(\kappa_{2}p^{-}(w, t) - \kappa_{3}p^{-}(w, t))] = 286$$

$$\sigma x(t) \tanh(x(t))\hat{D} + \sigma x(t)d(t) - \sigma \kappa_{1}x^{2}(t). = 296$$

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Invoking (17)–(24), we obtain $\dot{\Theta}_3(t)$ as

$$\dot{\Theta}_{3}(t) \leq \frac{\sigma \varsigma_{5}}{2}m^{2} - \frac{\sigma \varsigma_{1}}{2}\widetilde{D}^{2} + \frac{\sigma \varsigma_{1}}{2}D^{2} + \sigma x(t) \tanh(x(t))\widetilde{D}(t) \qquad ^{293} \\ + \sigma x(t) [\widetilde{T}p'(w,t) - \widetilde{E}Ip'''(w,t) + \widetilde{d}_{a}\dot{p}(w,t) \qquad ^{294}$$

$$+ \widetilde{m} \left(\kappa_2 \dot{p}^{\prime\prime\prime}(w,t) - \kappa_3 \dot{p}^{\prime}(w,t) \right) \left] - \frac{\sigma \zeta_2}{2} \widetilde{T}^2 - \frac{\sigma \zeta_3}{2} \widetilde{EI}^2 \quad _{295}$$

$$-\frac{\sigma \xi_4}{2}\widetilde{d}_a^2 - \frac{\sigma \xi_5}{2}\widetilde{m}^2 + \frac{\sigma \xi_2}{2}T^2 + \frac{\sigma \xi_3}{2}EI^2 + \frac{\sigma \xi_4}{2}d_a^2 \qquad _{296}$$
$$-\sigma \alpha \widetilde{\lambda}\beta\alpha(t)\mathbf{r}(t) - \frac{\sigma \upsilon}{2}\widetilde{\lambda}^2 + \frac{\sigma \upsilon}{2}\lambda^2 \qquad (39)$$

$$-\sigma \varrho \widetilde{\lambda} \beta \alpha(t) x(t) - \frac{\sigma \nu}{2} \widetilde{\lambda}^2 + \frac{\sigma \nu}{2} \lambda^2.$$
(39) 297

We invoke (1) and apply Lemma 1 to derive $\dot{\Theta}_4(t)$ as

$$\dot{\Theta}_4(t) \le -w\psi EIp'''(w,t)p'(w,t) - \frac{3\psi EI}{2}\int_0^w p''^2 ds$$
 29

$$+ \frac{w\psi c}{v_2} \int_0^w \dot{p}^2 ds + \frac{\psi \rho w}{2} \dot{p}^2(w, t) + \frac{w\psi}{v_3} \int_0^w f^2 ds \quad {}_{300}$$

$$+ \frac{\psi T w}{2} p^{\prime 2}(w,t) - \frac{\psi \rho}{2} \int_0^w \dot{p}^2 ds \qquad (40) \ _{302}$$

where $v_2, v_3 > 0$.

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Substituting (37)–(40) into (36) and using Lemmas 1–3 and (13), $\dot{\Theta}(t)$ gives

$$\begin{array}{ll} 306 \qquad \dot{\Theta}(t) \leq -\frac{3\psi EI}{2} \int_{0}^{w} p^{\prime 2} ds \\ 307 \qquad -\left(\frac{\xi EI\kappa_{3}^{2}}{2\kappa_{2}} - \frac{\xi |T - EI\kappa_{3}/\kappa_{2}|}{2\nu_{4}}\right) \end{array}$$

$$-\frac{EI|\xi\kappa_3 - w\psi|v_5}{2} - \frac{\psi Tw}{2} p^{\prime 2}(w,t) - \frac{\sigma \varsigma_1}{2} \widetilde{D}^2$$

$$- \left(\frac{\xi EI}{2\kappa_2} - \frac{\xi |I| - EI\kappa_3/\kappa_2/\nu_4}{2} - \frac{\psi p w}{2}\right)\dot{p}^2(w, t) - \left(\frac{\xi EI\kappa_2}{2} - \frac{|\xi \kappa_3 - w\psi|}{2}\right)p^{\prime\prime\prime 2}(w, t) + \frac{\sigma \xi_1}{2}D^2$$

$$= \frac{1}{2} \left(\frac{2}{\sqrt{2}} - \frac{2}{\sqrt{2}} \frac{1}{\sqrt{2}} - \frac{\sqrt{2}}{\sqrt{2}} \frac{1}{\sqrt{2}} - \frac{\sqrt{2}}{\sqrt{2}} \frac{1}{\sqrt{2}} \frac{1}{\sqrt{2}} - \frac{\sqrt{2}}{\sqrt{2}} \frac{1}{\sqrt{2}} \frac{1}{\sqrt{2}}$$

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$$-\left(\sigma\kappa_{1} - \frac{\xi EI}{2\kappa_{2}}\right)x^{2}(t) - \frac{\sigma\varsigma_{2}}{2}\widetilde{T}^{2} - \frac{\sigma\varsigma_{3}}{2}\widetilde{EI}^{2} + 0.2785\sigma D$$

$$-\frac{\sigma}{2}\frac{\varsigma_4}{2}\widetilde{d}_a^2 - \frac{\sigma}{2}\frac{\varsigma_5}{2}\widetilde{m}^2 + \frac{\sigma}{2}\frac{\varsigma_2}{2}T^2 + \frac{\sigma}{2}\frac{\varsigma_3}{2}EI^2 + \frac{\sigma}{2}\frac{\varsigma_4}{2}d_a^2$$

$$+ \left(\frac{\xi}{\nu_1} + \frac{w\psi}{\nu_3}\right)\int_0^w f^2 ds - \frac{\sigma}{2}\widetilde{\lambda}^2 + \frac{\sigma}{2}\lambda^2$$

$$(41)$$

³¹⁶ where $\nu_4, \nu_5 > 0$ and we choose $\psi, \sigma, \xi, \kappa_i, i = 1 \cdots 3, \nu_j$, ³¹⁷ for $j = 1 \cdots 5$ to satisfy

$$\frac{\xi E I \kappa_3^2}{2\kappa_2} - \frac{\xi |T - E I \kappa_3 / \kappa_2|}{2\nu_4} - \frac{E I |\xi \kappa_3 - w\psi| \nu_5}{2} - \frac{\psi T w}{2} \ge 0$$
(42)

$$_{320} \qquad \frac{\xi EI}{2\kappa_2} - \frac{\xi |T - EI\kappa_3/\kappa_2|\nu_4}{2} - \frac{\psi \rho w}{2} \ge 0 \tag{43}$$

$$\frac{\xi E I \kappa_2}{2} - \frac{|\xi \kappa_3 - w\psi|}{2\nu_5} \ge 0$$
(44)

322
$$\omega_1 = \xi c - \xi v_1 - \frac{w\psi c}{v_2} + \frac{\psi \rho}{2} > 0$$
(45)

$$\omega_2 = \frac{\psi I}{2} - \psi v_2 cw - \psi v_3 w > 0 \tag{46}$$

$$\omega_3 = \sigma \kappa_1 - \frac{\xi EI}{2\kappa_2} > 0 \tag{47}$$

325
$$\chi = \left(\frac{\xi}{\nu_1} + \frac{w\psi}{\nu_3}\right) wF^2 + \frac{\sigma\varsigma_1}{2}D^2 + \frac{\sigma\varsigma_2}{2}T^2 + \frac{\sigma\varsigma_3}{2}EI^2 + \frac{\sigma\varsigma_4}{2}d_a^2 + \frac{\sigma\varsigma_5}{2}m^2 + \frac{\sigma\upsilon}{2}\varepsilon^2 + 0.2785\sigma D < +\infty.$$

328 Combining (42)–(48), (41) is derived as

$$\dot{\Theta}(t) \le \chi - \frac{\sigma \varsigma_1}{2} \widetilde{D}^2 - \omega_1 \int_0^w \dot{p}^2 ds - \omega_2 \int_0^w p'^2 ds - \frac{3\psi EI}{2} \int_0^w p''^2 ds - \omega_3 x^2(t) - \frac{\sigma \upsilon}{2} \widetilde{\lambda}^2$$

$$\begin{array}{rcl} & & & & & & \\ & & & & \\ {}^{331} & & & & -\frac{\sigma}{2}\frac{\sigma}{2}\widetilde{T}^2 - \frac{\sigma}{2}\frac{\sigma}{2}\widetilde{EI}^2 - \frac{\sigma}{2}\frac{\sigma}{2}\widetilde{d}_a^2 - \frac{\sigma}{2}\frac{\sigma}{2}\widetilde{m}^2 \\ & & \\ {}^{332} & & \leq -\iota_3[\Theta_1(t) + \Theta_2(t) + \Theta_3(t)] + \chi \end{array}$$

where $\iota_3 = \min(\frac{2\omega_1}{\xi\rho}, \frac{2\omega_2}{\xi T}, \frac{3\psi}{\xi}, \frac{2\omega_3}{\sigma m}, \varsigma_1, \varsigma_2, \varsigma_3, \varsigma_4, \varsigma_5, 1).$ We then invoke (30) and (49) to obtain

$$\dot{\Theta}(t) \le -\iota\Theta(t) + \chi \tag{50}$$

336 where $\iota = (\iota_3 / \iota_2)$.



Fig. 3. 3-D offset of the riser under no control.

Theorem 1: For the riser system with input backlash (4), ³³⁷ under the presented adaptive backlash inverse control (15), ³³⁸ online updating laws (18)–(23), and bounded initial conditions, ³³⁹ with the choice of design parameters ψ , σ , ξ , κ_i , $i = 1 \cdots 3$, ν_j , ³⁴⁰ for $j = 1 \cdots 5$ satisfying constraints (42)–(48), we arrive ³⁴¹ at a conclusion that the controlled system's state p(s, t) is ³⁴² uniformly ultimately bounded. ³⁴³

Proof: We multiply (35) by e^{tt} and then integrate the ³⁴⁴ consequence to derive ³⁴⁵

$$\Theta(t) \le \Theta(0)e^{-\iota t} + \frac{\chi}{\iota} \left(1 - e^{-\iota t}\right) \le \Theta(0)e^{-\iota t} + \frac{\chi}{\iota}.$$
 (51) 346

Invoking $\Theta_1(t)$, (30), and Lemma 2, we get

$$\frac{\xi T}{2w}p^2(s,t) \le \frac{\xi T}{2} \int_0^w p'^2(s,t) ds \le \Theta_1(t) \le \frac{1}{\iota_1} \Theta(t).$$
(52) 348

We substitute (51) into (52) to derive

$$p(s,t) \mid \leq \sqrt{\frac{2w}{\xi\iota_1 T}} \Big[\Theta(0)e^{-\iota t} + \frac{\chi}{\iota}\Big], \forall (s,t) \in [0,w] \times [0,+\infty).$$

$$(53) \quad 350$$

Combining (53) further gives

I

(48)

$$\lim_{t \to \infty} |p(s,t)| \le \sqrt{\frac{2w\chi}{\xi T \iota_1 \iota}}, \forall s \in [0,w].$$
(54) 353

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Remark 2: This article presents a framework of adaptive inverse control of uncertain vessel-riser systems subject ³⁵⁶ to input backlash, system uncertainties, and external disturbances, which is invalid for the system with input hysteresis. ³⁵⁸ To address this issue, the approaches in [50] will be employed ³⁵⁹ in the next step. Moreover, an issue about communication limit ³⁶⁰ in actuator is ignored in this article, and we will cope with it ³⁶¹ with recourse to the technique in [69]–[71]. ³⁶²

IV. NUMERICAL SIMULATION 363

On the basis of the vessel-riser system dynamical model $_{364}$ with input backlash (1), (2), (4), and (13), we exploit $_{365}$ the finite difference method [72] with time and space $_{366}$



Fig. 4. 3-D offset of the riser under proposed control.



Fig. 5. 3-D offset of the riser under previous control in [18].



Fig. 6. 3-D offset of the riser under previous control in [60].



Fig. 7. 2-D offset of the vessel under proposed control.







Fig. 9. 2-D offset of the vessel under previous control in [60].

³⁶⁷ steps as 2×10^{-5} and 0.05 to illustrate the dynam-³⁶⁸ ics of this nonlinear coupled system by setting system ³⁶⁹ parameters as $EI = 1.5 \times 10^7$ Nm², $\rho = 500$ kg/m, $T = 3.0 \times 10^8$ N, c = 1.0 Ns/m², w = 1000 m, $d_a = 370$ 1.5×10^5 Ns/m, and $m = 9.6 \times 10^6$ kg. System initial condi-371tions are presented as $p(s, 0) = 12 \sin(\frac{s}{w})$ and $\dot{p}(s, 0) = 0$. 372



Fig. 10. Previous control input in [18].









Fig. 13. Designed control input.

³⁷³ Meanwhile, d(t) is given as $d(t) = [3 + 0.8\sin(0.7t) + 374 \ 0.2\sin(0.5t) + 0.2\sin(0.9t)] \times 10^5$.

Without any control, namely, $\tau(t) = 0$, 3-D and twodimensional (2-D) responses of this coupled system are portrayed in Figs. 3 and 7. From Fig. 3, it is seen that the marine flexible riser is vibrating with an equal amplitude under the external ocean disturbance. Fig. 7 illustrates the displacement of the vessel in the ocean surface. The persistent large deformation of this marine riser will lead to the produce fatigue problems, and it is crucial to reduce the vibration by implementing the effective control strategy.

With presented control law (15) by choosing control gains $\kappa_1 = 3 \times 10^8$, $\kappa_2 = 1$, $\kappa_3 = 20$, and control parameters $\kappa_1 = \zeta_2 = \zeta_3 = \zeta_4 = \zeta_5 = 0.001$, $\rho = 1$, $\beta = 1.2$, $\nu = 3$, $B = 3 \times 10^6$, and $B_m = 1 \times 10^7$, the spatio-temporal response and end point offset are depicted in Figs. 4 and 7. Son From Figs. 4 and 7, it is seen that the effects of the considered external ocean disturbance and the input backlash nonlinearity ³⁹¹ are eliminated under the proposed control (15). Moreover, it ³⁹² has a positive effect on the vibration attenuation of the marine ³⁹³ flexible riser, and the displacement of the vessel reduces to a ³⁹⁴ small neighborhood around the original position. Meanwhile, ³⁹⁵ Figs. 13 and 14 display 2-D responses of presented control ³⁹⁶ input and backlash input. ³⁹⁷

For the comparison with the proposed control law, we consider two control strategies presented in previous works [18] ³⁹⁹ and [60]. When exerting the previous control proposed in [18] ⁴⁰⁰ on the riser system with the given control parameters k = 401 3×10^8 , $k_1 = 1$, and $k_2 = 10$, Figs. 5, 8, and 10 display the responses of the marine flexible riser, vessel, and ⁴⁰³ the control law, respectively. Note that this previous research ⁴⁰⁴ does not consider the effect of the input constraint and ⁴⁰⁵ the control law presented in [18] requires longer convergent ⁴⁰⁶ time and larger convergent neighborhood than the proposed ⁴⁰⁷ control (15) for the marine vessel-riser system with input ⁴⁰⁸



Fig. 14. Backlash control input.

409 backlash nonlinearity. Under the action of previous anti-⁴¹⁰ saturation control developed in [60] on the riser system, when 411 the control design parameters are selected as $k_1 = k_3 = \beta_1 =$ ⁴¹² $\beta_2 = \beta_3 = \beta_4 = 1$, $k_2 = 20$, $k_4 = 0.01$, $k_5 = 8 \times 10^7$, ⁴¹³ $\sigma_1 = \sigma_2 = \sigma_3 = \sigma_4 = 1 \times 10^{-4}$, $\sigma = 1 \times 10^{-5}$, $\tau_{\text{max}} = 3 \times 10^5$, 414 and $\tau_{\rm min} = -2 \times 10^5$, the time and spatial responses are 415 described in Figs. 6, 9, 11, and 12. Note that the control 416 law presented in [60] also has the positive effect on reduc-417 ing the deformation of the coupled system, however it has 418 the large overshoot and convergent neighborhood than the 419 proposed control (15).

We observe from Figs. 3–14 that the vibration in the cou-420 421 pled vessel-riser system is observably suppressed under the 422 proposed adaptive inverse control, which achieves a better con-⁴²³ trol performance than the previous control; the end point offset $_{424} p(w, t)$ is stabilized at a small region around zero, and the 425 backlash nonlinearity in the control input is fairly obvious. In 426 other words, this approach leads to a good performance on 427 the vibration decrease, uncertainties compensation, and input 428 backlash elimination.

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V. CONCLUSION

The framework of the adaptive inverse control of uncer-430 431 tain vessel-riser systems possessing input backlash has been 432 presented in this article. The adaptive inverse of backlash 433 was used to formulate the nonlinear input backlash as a 434 desired control signal with a mismatch error. An adaptive 435 inverse control and relevant adaptive laws were presented for 436 stabilizing the riser's offset, eliminating the backlash, and 437 compensating for system uncertainties. Exploiting the rigor-438 ous analysis without recourse to model reducing technique, 439 the derived control ensured and realized the uniform stabil-440 ity of the controlled system. In conclusion, the simulation 441 comparison studies validated the control performance. Future 442 interesting topics include exploiting the intelligent control 443 techniques [73]–[79] to regulate the transient performance of 444 the controlled vessel-riser systems.

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