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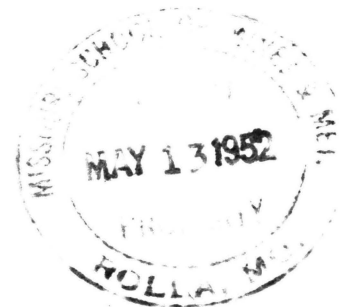
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**TRANSFER FUNCTION OF THE MODIFIED
KRAMER SYSTEM**

BY
DAVID L. HILLHOUSE



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A
THESIS
submitted to the faculty of the
SCHOOL OF MINES AND METALLURGY OF THE UNIVERSITY OF MISSOURI
in partial fulfillment of the work required for the
DEGREE OF
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John Laborsky
Professor of Electrical Engineering

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INTRODUCTION

The use of operational mathematics in the solution of engineering problems has grown tremendously in the last two decades, particularly during the last war. This has come about because of the facility with which problems which would be almost impossibly complicated or tedious to solve by classical methods can be solved using operational calculus.

The operational calculus, particularly the Laplace transform method, has become especially useful in the field of servomechanisms and controls. Here the constantly increasing complexity of systems used has resulted in very high order differential equations. The fact that the Laplace transform method makes it possible to inject the boundary conditions into the initial equations is often a great advantage in their solution. In addition, the fact that much can often be learned about the behavior of a system by a study of its operational equations, without having to resort to an actual solution, makes this approach even more valuable.

This paper proposes to develop the complete operational solution, or transfer function, of a specific control system, namely, the modified Kramer system for speed control of induction motors. This transfer function is derived as a basis for further study of the performance of the system, or of its components.

First, a brief qualitative discussion of the original Kramer system seems in order. This system has been in use for many years in steel mill power applications. It has been applied chiefly in situations where the amount of power wasted in the rotor circuit of an induction motor by use of resistances to control speed became large enough that it was economically feasible to recover this power and return it to the

line.

Figure 1 shows a block diagram of the Kramer system. In operation the power developed in the rotor of the main machine, A, is converted to DC by the synchronous converter, B, and fed into the DC motor, C which in turn drives machine D as an induction generator; returning to the system the otherwise wasted rotor power. This is of course minus the losses of machines B, C, and D.

Speed variation is accomplished in the following manner. It is well known from induction motor theory that the speed is a function of the resistance of the rotor circuit. This rotor resistance of course represents a voltage drop. Whether this drop is caused by a passive impedance, or by an active element in opposition to the induced rotor voltage makes no difference as far as the induction machine is concerned. In the Kramer system a synchronous converter is inserted in the rotor circuit of the induction machine. The voltage developed by the converter affects the main machine's speed just as would a rotor resistance. The DC machine represents a load supplied by the DC side of the converter. By varying the excitation of the DC machine, therefore, the voltage of the DC side of the converter can be caused to vary. The voltage of the AC side of the converter is in fixed ratio to that of the DC side. Thus a variable voltage is inserted in the rotor circuit of the induction machine by varying the excitation of the DC machine, thereby causing variation in the induction motor's speed. Since the frequency of the converter must always equal the slip frequency of the induction motor, i.e., the frequency of the supply for the converter stator, it is apparent that the speed variation of B will be equal and opposite to that of machine A. Automatic speed control can be accomplished by providing the proper feedback link from the speed of machine A to the field of machine

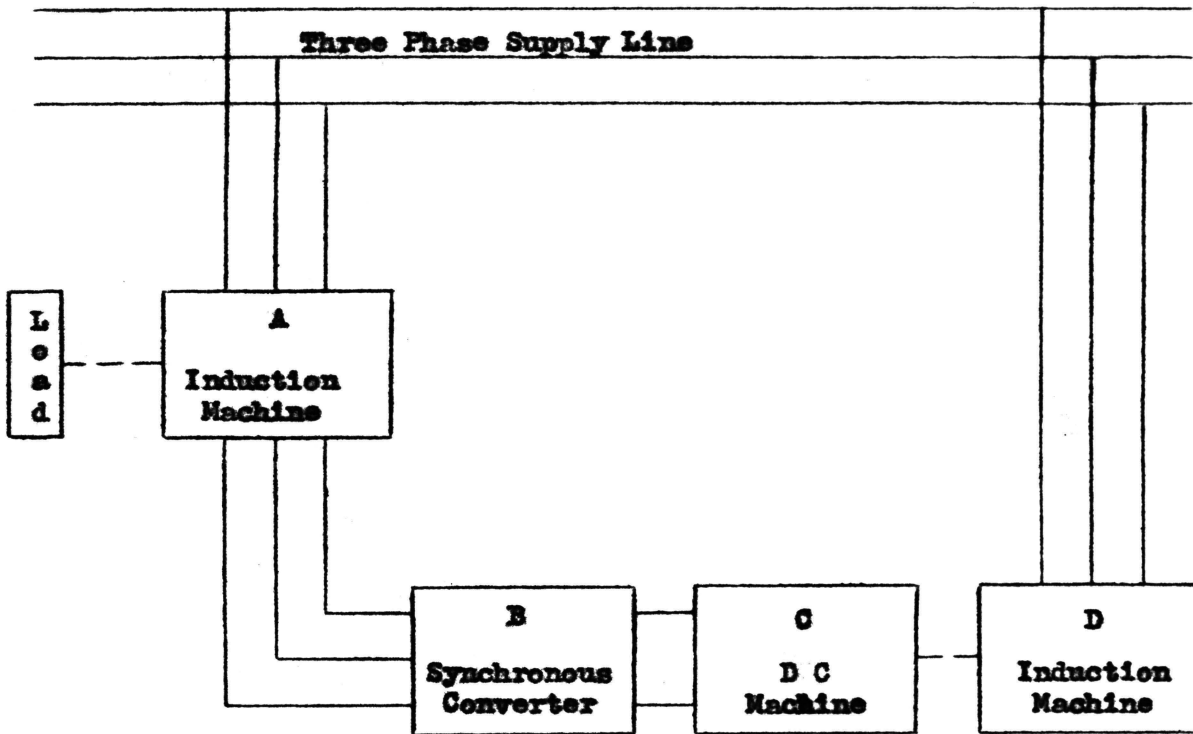


FIGURE 1 - Block Diagram of Kramer System

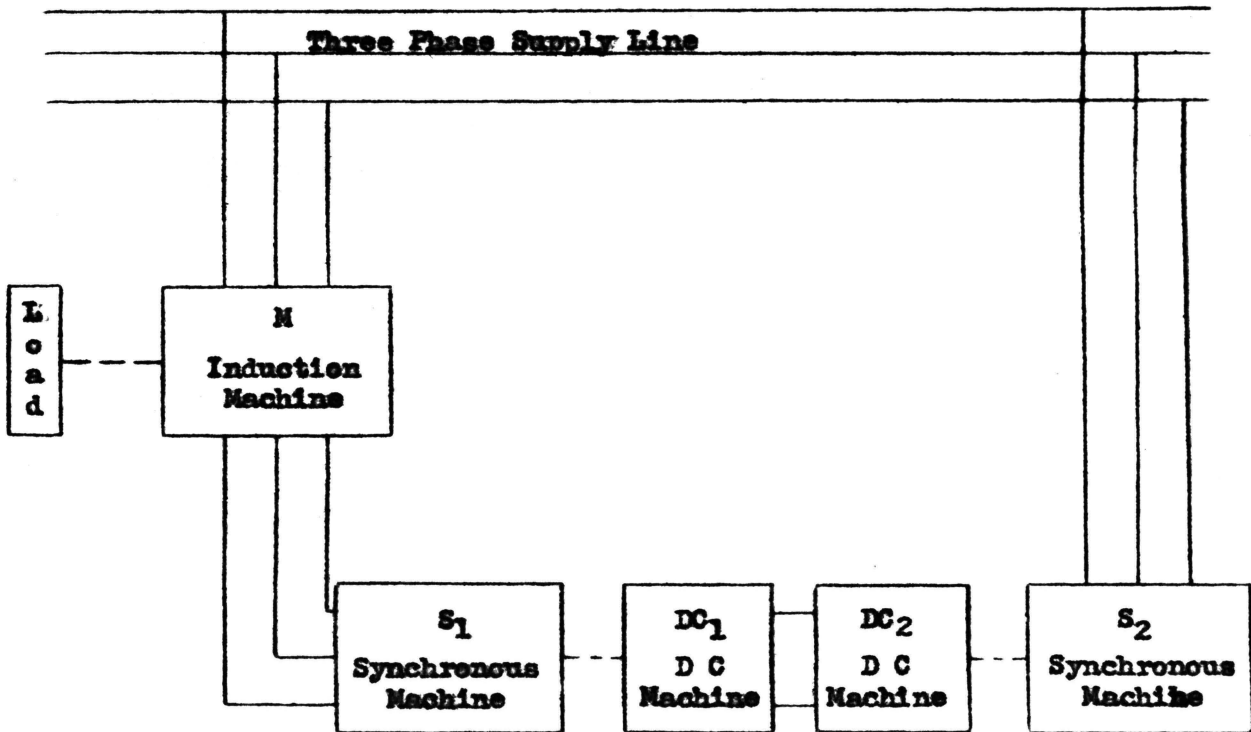


FIGURE 2 - Block Diagram of Modified Kramer System

c.

A block diagram of the modified Kramer system is shown in Figure 2. In operation it is very similar to the Kramer system. The modified system, however, utilizes a synchronous motor-generator set instead of the synchronous converter; and a second synchronous motor-generator set instead of the induction generator-DC motor set of the original Kramer system. The chief advantage of the modified system is that in general synchronous motor-generator sets are more readily available, cheaper, and require less auxiliary equipment than, for example, a converter. Also, the DC side of the MG set S_1 -DC₁ is no longer rigidly tied, voltage-wise, to the AC side, as in the converter; and set S_2 -DC₂ is of necessity a constant speed set.

Due to saturation, hysteresis, and non-uniform air gaps, electrical machinery is inherently non-linear in behavior. Therefore in order to obtain a mathematical analysis of its performance, it is conventional to make certain simplifying assumptions. These will be made in the development which follows, and will be tabulated at the very beginning of the discussion.

The development will be carried forward in three main sections, the last two of which will be further subdivided. In the first section the general assumptions will be discussed. In the second, relations for each of the individual machine groups will be developed. In the third, these relations will be matched at their boundaries to obtain the final solution.

REVIEW OF LITERATURE

There is very little in the literature concerning the specific subject of this paper. Very brief qualitative mention of the Kramer system is made in most of the standard college texts on electrical machinery. The modified Kramer system made its first appearance about 1941, in a Wright Air Force Base wind tunnel application. (1),(2) The

-
- (1) Dickey, A. D., Laffon, C. M., and Kilgore, L. A., Variable Speed Drive For USAAC Wind Tunnel, Wright Field, Dayton, Ohio, AIEE Transactions, Vol. 61, 1942, pp. 126-130.
 (2) Clymer, C. C., Large Adjustable-Speed Wind Tunnel Drive, AIEE Transactions, Vol. 61, 1942, pp. 156-158.
-

only publication dealing directly with the problem in this paper is an article by M. M. Liwschitz and L. A. Kilgore. (3) That article deals

-
- (3) Liwschitz, M. M., and Kilgore, L. A., A Study of the Modified Kramer or Asynchronous-Synchronous Cascade Variable-Speed Drive, AIEE Transactions, Vol. 61, 1942, pp. 256-260.
-

primarily with the steady-state stability of the system, and develops complete equations only for the double-fed machine. This paper approaches the problem from a somewhat different angle, and extends the analysis to obtain a transfer function for the complete system. The Liwschitz-Kilgore article works in the Heaviside calculus, whereas this paper uses the Laplace transforms.

Although as stated the literature dealing specifically with the subject of this paper is very limited in quantity, this review would be incomplete without mention of the classic article on the operational analysis of electrical machinery, namely, the "Park article". (4) Vir-

-
- (4) Park, R. H., Two-Reaction Theory of Synchronous Machines-Generalized Method of Analysis-Part I, AIEE Transactions, Vol. 48, 1929, pp. 716-727.

tually all the works listed in the Bibliography of this paper in some way stem from or are based upon the above paper by Park. Noteworthy among these succeeding papers are: the Waring and Crary (5) article,

- (5) Waring, M. L., and Crary, S. B., The Operational Impedances of a Synchronous Machine, G. E. Review, Vol. 35, 1932, pp. 578-582.
-

which presents Park's work on the basis of reciprocal mutual reactances; the Stanley article, (6) which solves the induction machine by a trans-

- (6) Stanley, H. C., An Analysis of the Induction Machine, AIEE Transactions, Vol. 57, 1938, pp. 751-755.
-

formation to a set of stationary α - β axes; and the papers by Rankin (7),(8) which further correlate the work of Park with Waring and Crary

- (7) Rankin, A. W., The Equations of the Idealized Synchronous Machine, G. E. Review, Vol. 47, June, 1944, pp. 31-36.
 (8) Rankin, A. W., Per-Unit Impedances of Synchronous Machines, AIEE Transactions, Vol. 64, 1945, part I, pp. 569-573.
-

and others. This paper draws on all the above listed publications, particularly the Stanley article.

DISCUSSION

1. Discussion of Complete Schematic Diagram and Tabulation of General Assumptions

Refer to Fig. 3. The symbols used in the figure are defined thereon. It should be noted that the masses and frictional effects of each complete motor-generator set have been lumped together as J_2, B_2, J_3, B_3 , respectively. All shafts have been assumed to be rigid. This is generally valid because of their relatively large diameters and short lengths in most electrical machines.

For simplicity in writing the equations, and because it is very probably so in an actual application, synchronous machine S_1 and S_2 as well as machines DC_1 and DC_2 , will be assumed to be identical.

Since it is desired to control the system by varying the excitation of one of the DC machines, three of the four excitation voltages shown on Fig. 3 can be taken as constants. E_2 has arbitrarily been chosen as the variable excitation. Therefore, E_{f1}, E_{f2} , and E_1 are constants. Furthermore, since the system is tied to a very large source of supply, the line voltage and frequency applied to motor M and synchronous machine S_2 are constants.

Finally, the presence of the two extra short circuited windings in the rotors of S_1 and S_2 is an approximate attempt to account for the effect of amortisseur windings. More will be said of this last assumption later on.

The general assumptions discussed above are listed below:

1. Masses and frictions of mechanically coupled elements are lumped
2. All shafts are rigid.
3. Machines S_1 and S_2 are identical, as are DC_1 and DC_2 .

4. E_{f1} , E_{f2} , and E_1 are constant.
5. Line voltage and frequency to M and S_2 are constant.
6. Amortisseur windings can be represented by two short - circuited rotor windings.

2. Transfer Functions of Individual Machine Groups

Induction Machine, M

The development in this section follows that of Stanley ⁽⁹⁾ very

(9) Stanley, H. G., op. cit., pp. 751-755.

closely in form and notation, and draws on an earlier paper by Levine ⁽¹⁰⁾

(10) Levine, S. J., An Analysis of the Induction Motor, AIEE Transactions, Vol. 54, 1935, pp. 526-529.

for the initial equations. However, there is an important difference from the Stanley solution. This is that a reference frame rotating at synchronous speed has been used rather than Stanley's fixed frame. As will be brought out later in the development, this change of reference is the key to the Laplace transformable solution obtained.

In addition to those already mentioned, the following assumptions will be made:

1. Hysteresis, saturation, and eddy currents will be neglected.

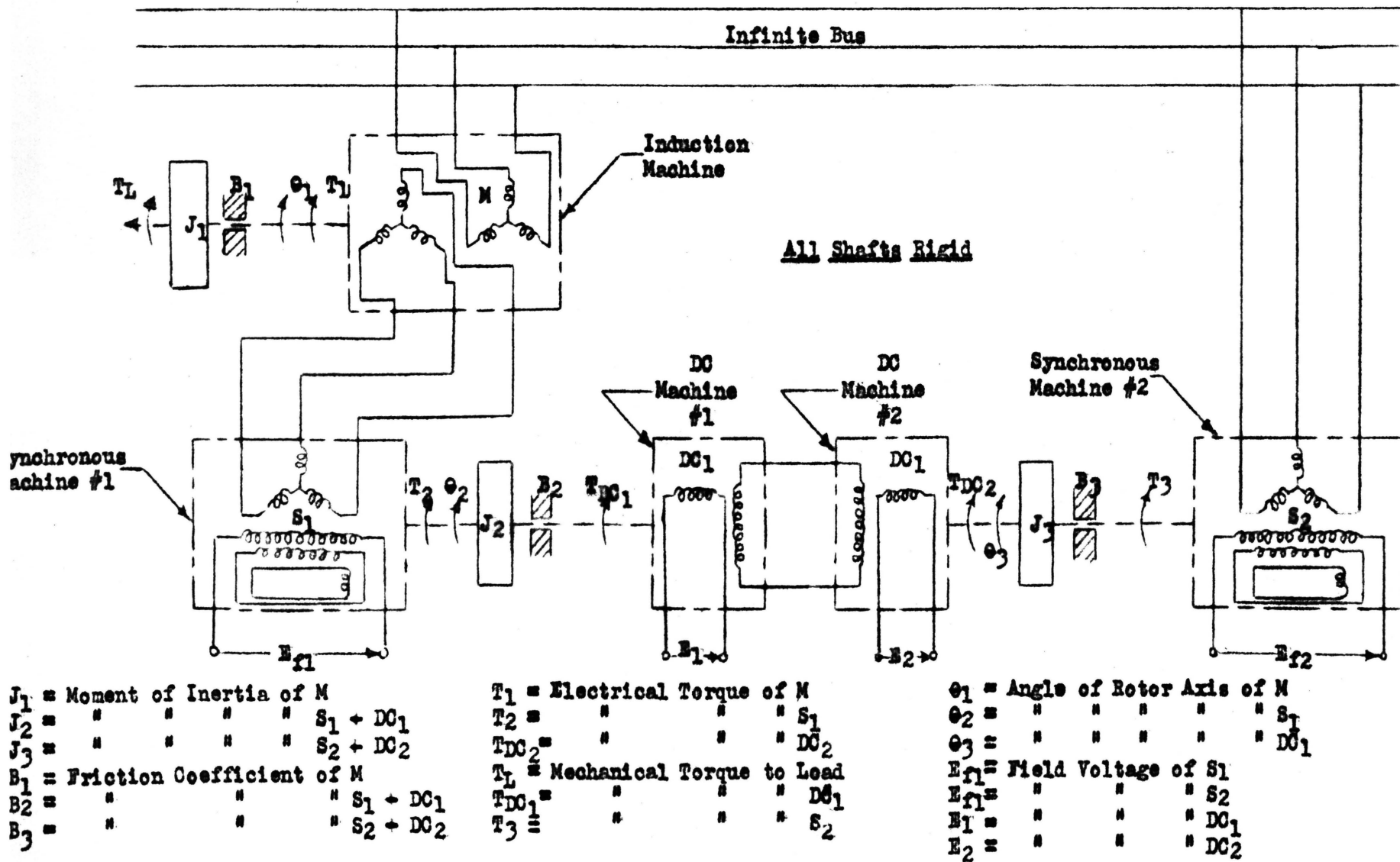


FIGURE 3 - Schematic Diagram of Modified Kramer System

2. Rotor and stator windings are balanced.
3. Rotor is smooth, i.e., self inductances of all windings are independent of rotor position.
4. Coefficient of mutual inductance between rotor and stator windings is a cosine function of the angle between the axes of the two windings.

It can be stated as a general relation for any electric circuit that:

$$e = ir + \frac{d\psi}{dt}$$

where

e = instantaneous applied voltage

i = instantaneous positive current

r = resistance

ψ = total instantaneous flux linkages with the circuit, including those with coupled circuits, such linkages being positive when they produce positive drop in the circuit.

By the fundamental definition for induced voltages

$$\frac{d\psi}{dt} = L \frac{di}{dt} + \sum M_n \frac{di_n}{dt} = \frac{d\psi_{self}}{dt} + \frac{d\psi_{mut}}{dt}$$

where

L = circuit self-inductance

M_n = mutual inductance with the nth circuit

i_n instantaneous current in the nth circuit

Then

$$(1) \quad e = ir + L \frac{di}{dt} + \sum_0^n M_n \frac{di_n}{dt}$$

and

$$\Psi_n = M_n i_n$$

$$\Psi_{\text{self}} = Li$$

Under the assumptions made, the three-phase induction motor can be represented by six circuits, each having self and mutual inductances as shown in Fig. 4, but with the rotor circuit free to rotate with reference to the stator. On the basis of assumptions (2) and (3) the self inductances of each phase and the mutual inductances between phases on the stator and rotor are equal, i.e.,

$$L_{sa} = L_{sb} = L_{sc}$$

$$M_{sab} = M_{sbc} = M_{sac}$$

$$L_{ra} = L_{rb} = L_{rc}$$

$$M_{rab} = M_{rbc} = M_{rac}$$

The following is actually an explanation of assumption (4). The $M_{3\text{ph}}$ represents the maximum value of mutual inductance between one rotor phase and one stator phase, i.e., that inductance which occurs when the magnetic axes of the given phases are opposite each other. Referring to phases a and 1 , Fig. 5, it can be seen that the mutual coupling between any stator phase and a given rotor phase is a function of the cosine of the angle between them. With this in mind, and referring

to Fig. 4 again, it may be written that

$$\begin{aligned}\Psi_a &= L_1 i_a + M_1 (i_b + i_c) + M_{3ph} [i_1 \cos \theta + i_2 \cos (\theta + 120) + i_3 \cos (\theta - 120)] \\ \Psi_b &= L_1 i_b + M_1 (i_a + i_c) + M_{3ph} [i_1 \cos (\theta - 120) + i_2 \cos \theta + i_3 \cos (\theta + 120)] \\ (2) \quad \Psi_c &= L_1 i_c + M_1 (i_a + i_b) + M_{3ph} [i_1 \cos (\theta + 120) + i_2 \cos (\theta - 120) + i_3 \cos \theta] \\ \Psi_1 &= L_2 i_1 + M_2 (i_2 + i_3) + M_{3ph} [i_a \cos \theta + i_b \cos (\theta - 120) + i_c \cos (\theta + 120)] \\ \Psi_2 &= L_2 i_2 + M_2 (i_1 + i_3) + M_{3ph} [i_a \cos (\theta + 120) + i_b \cos \theta + i_c \cos (\theta - 120)] \\ \Psi_3 &= L_2 i_3 + M_2 (i_1 + i_2) + M_{3ph} [i_a \cos (\theta - 120) + i_b \cos (\theta + 120) + i_c \cos \theta]\end{aligned}$$

But it is also true that

$$\begin{aligned}i_a + i_b + i_c &= 0 \\ (3) \quad i_1 + i_2 + i_3 &= 0\end{aligned}$$

from which, referring to equations (2) above

$$\begin{aligned}\Psi_a + \Psi_b + \Psi_c &= 0 \\ (4) \quad \Psi_1 + \Psi_2 + \Psi_3 &= 0\end{aligned}$$

The above equations quickly become unwieldy because of the involved trigonometric relations they contain. Therefore two successive simplifying transformations will be made. First, replace the three phase machine of Fig. 4 with its equivalent two phase machine. To make this equivalence it is necessary only to adjust the inductance values so that the same flux linkage relations are maintained. In the equivalent machine of Fig. 6, let the following current relations be assumed: (11)

(11) Kimbark, E. W., Two Phase Coordinates of a Three-Phase Circuit, AIEE Transactions, Vol. 58, 1939, p. 894.

$$(5) \quad i_x = i_a = \frac{2i_a - i_b - i_c}{3}$$

$$(6) \quad i_y = j i_a = \frac{i_b - i_c}{\sqrt{3}}$$

$$(7) \quad i_d = i_1 = \frac{2i_1 - i_2 - i_3}{3}$$

$$(8) \quad i_g = j i_1 = \frac{i_2 - i_3}{\sqrt{3}}$$

Then

$$(9) \quad \Psi_x = L_s i_x + M(i_d \cos \theta - i_g \sin \theta)$$

$$\Psi_y = L_s i_y + M(i_d \sin \theta + i_g \cos \theta)$$

$$(10) \quad \Psi_d = L_R i_d + M(i_x \cos \theta + i_y \sin \theta)$$

$$\Psi_g = L_R i_g + M(i_x \sin \theta + i_y \cos \theta)$$

From the symmetry of equations (3) and (4) it may also be stated, analogous to equations (5) through (8) that

$$(11) \quad \Psi_x = \Psi_a$$

$$(12) \quad \Psi_y = \frac{\Psi_b - \Psi_c}{\sqrt{3}}$$

$$(13) \quad \Psi_d = \Psi_1$$

$$(14) \Psi_q = \frac{\Psi_2 - \Psi_3}{\sqrt{3}}$$

Now, the values of L_S , L_R , and M can be determined by substituting in (11) and (13) from (2), (9), and (10):

$$(15) \quad L_S i_x + M [i_d \cos \theta - i_q \sin \theta] = L_1 i_a + M_1 (i_b + i_c) \\ + M_{3ph} [i_1 \cos \theta + i_2 \cos(\theta + 120) + i_3 \cos(\theta - 120)]$$

$$(16) \quad L_R i_d + M [i_x \cos \theta + i_y \sin \theta] = L_2 i_1 + M_2 (i_2 + i_3) \\ + M_{3ph} [i_a \cos \theta + i_b \cos(\theta - 120) + i_c \cos(\theta + 120)]$$

Using equations (3) and certain trigonometric identities, the right hand sides of (15) and (16) simplify, so that

$$L_S i_x + M [i_d \cos \theta - i_q \sin \theta] = (L_1 - M_1) i_a \\ + \frac{3}{2} M_{3ph} [i_1 \cos \theta - \frac{1}{\sqrt{3}} (i_2 - i_3) \sin \theta] \\ L_R i_d + M [i_x \cos \theta + i_y \sin \theta] = (L_2 - M_2) i_1 \\ + \frac{3}{2} M_{3ph} [i_a \cos \theta + \frac{1}{\sqrt{3}} (i_b - i_c) \sin \theta]$$

which by the definition of the two phase quantities becomes

$$L_S i_x + M [i_d \cos \theta - i_q \sin \theta] = (L_1 - M_1) i_x \\ + \frac{3}{2} M_{3ph} [i_d \cos \theta - i_q \sin \theta] \\ L_R i_d + M [i_x \cos \theta + i_y \sin \theta] = (L_2 - M_2) i_d \\ + \frac{3}{2} M_{3ph} [i_x \cos \theta + i_y \sin \theta]$$

Inspection of these last two equations shows that the equivalent two phase inductances must be

$$L_s = L_1 - M_1$$

$$(17) \quad L_R = L_2 - M_2$$

$$M = \frac{3}{2} M_{3ph}$$

A complete family of equations can now be written for the two phase machine. These will include equations (9) and (10) and the following:

$$e_x = i_x r_s + \frac{d\psi_x}{dt}$$

$$(18) \quad e_y = i_y r_s + \frac{d\psi_y}{dt}$$

$$e_d = i_d R_R + \frac{d\psi_d}{dt}$$

$$e_q = i_q R_R + \frac{d\psi_q}{dt}$$

A further simplification, consisting of a transformation to a set of right angle axes, α and β , rotating at synchronous speed, will next be made. Fig. 7-a shows the actual two phase machine at a given instant. In the steady state the flux, current, and voltage in each phase of the stator pulsate at synchronous frequency, and are in space and time quadrature with the corresponding quantities in the other phase. The quantities in phase x lead those in phase y. It can readily be shown that the resultant of each pair of pulsating quantities, e.g., i_x and i_y , referred to a fixed reference is a vector of constant magnitude rotating clockwise at synchronous speed, ω_f . The rotor turns clockwise at a speed ω_r , so that the stator vectors sweep past it at

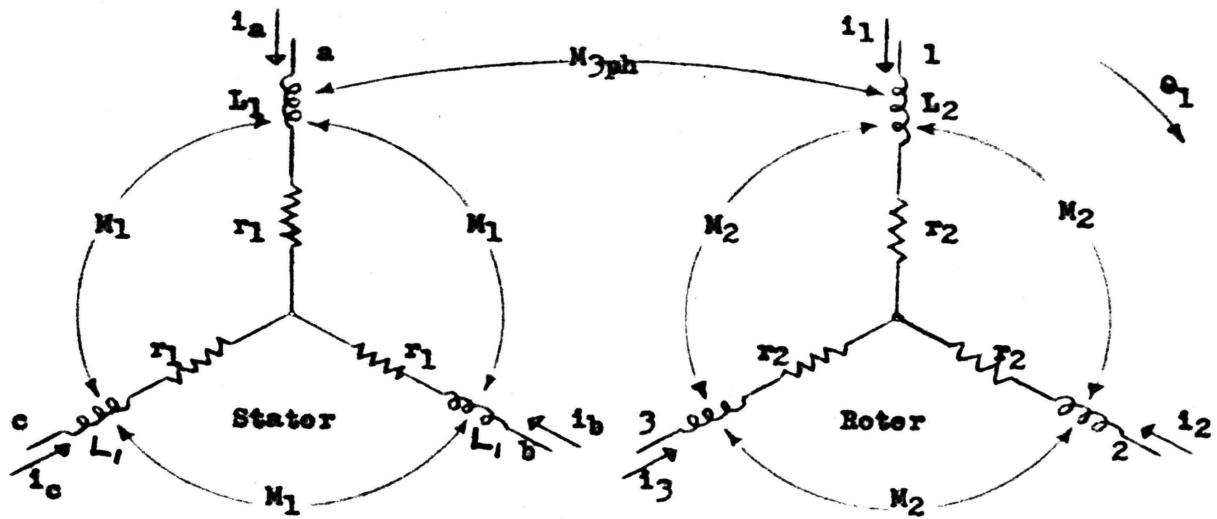


FIGURE 4 - Induction Motor Circuits.

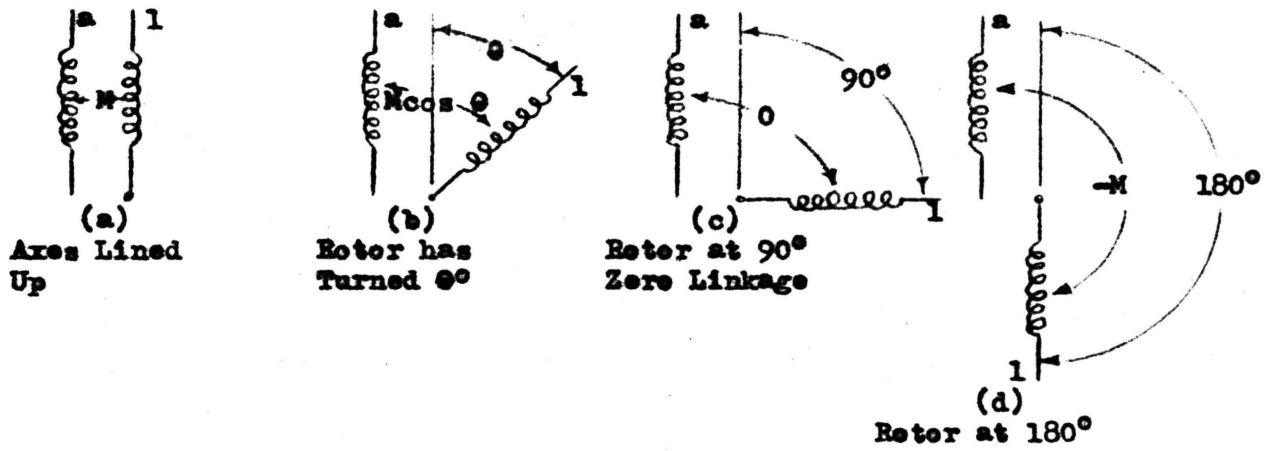


FIGURE 5 - Variation of N with Rotor Angle.

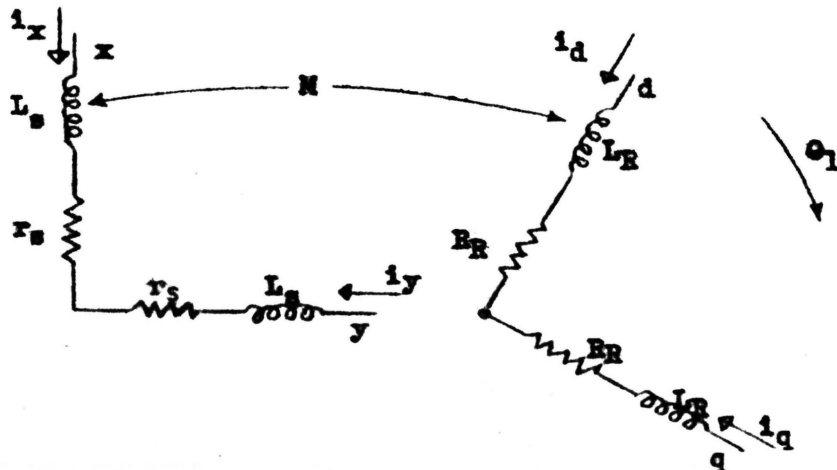
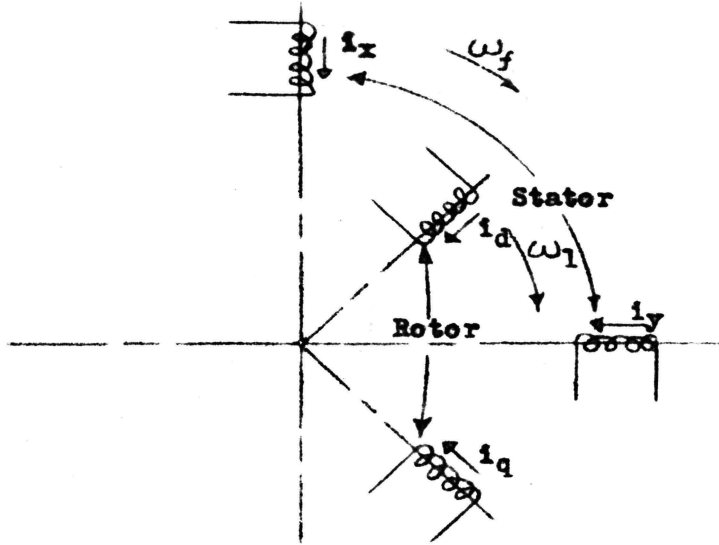


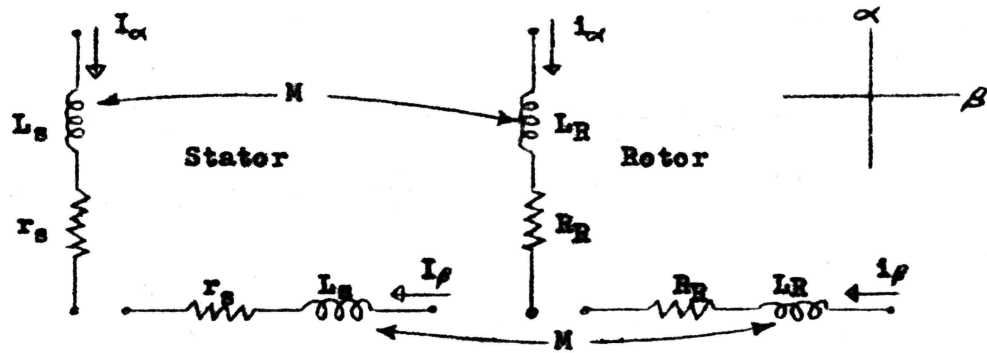
FIGURE 6 - Equivalent Two Phase Machine.

$\omega_f - \omega_1 = \omega_s$ or slip speed. By the converse of the relation which produces the rotating stator vectors, the rotating vectors set up slip frequency pulsating fluxes, currents and voltages in the rotor phases. These quantities in turn may be resolved into constant amplitude vectors, rotating clockwise at slip speed with reference to the rotor, or at synchronous speed relative to a fixed reference frame. Thus it is seen that both the stator and rotor quantities resolve into constant magnitude synchronously rotating vectors. Then if the system is viewed from a reference frame which is also rotating clockwise at synchronous speed, it reduces to a set of stationary constant magnitude vectors. If a set of right angle coordinate axes, α and β is set up on this synchronously rotating reference frame, each vector can be broken up into components along these axes. There will then be in the α axis, for instance, constant components of the stator vectors, I_α , Ψ_α , and E_α , respectively. A little consideration will show that three identical quantities would be present in a hypothetical coil oriented in the α axis with voltage E_α applied, current I_α flowing, and flux linkages Ψ_α present due to $I_\alpha L_s$ and to currents in any mutually coupled coil. Identical reasoning can be used for stator components in the β axis, I_β , Ψ_β , and E_β ; and for corresponding α and β axis rotor components, i_α , ψ_α , e_α , and i_β , ψ_β , e_β . Thus the actual two phase machine can be replaced, using the synchronously rotating α - β axes, by four stationary coils as shown in Fig. 7-b. In the steady state these coils will carry direct current and be linked by steady fluxes. The linkage relations are immediately very much simplified, in each case reducing to a self and mutual direct current component. (See equations 21). As will be seen when the specific transformation relations are introduced and the solution carried out, all unwieldy trigonometric relations are ultimately



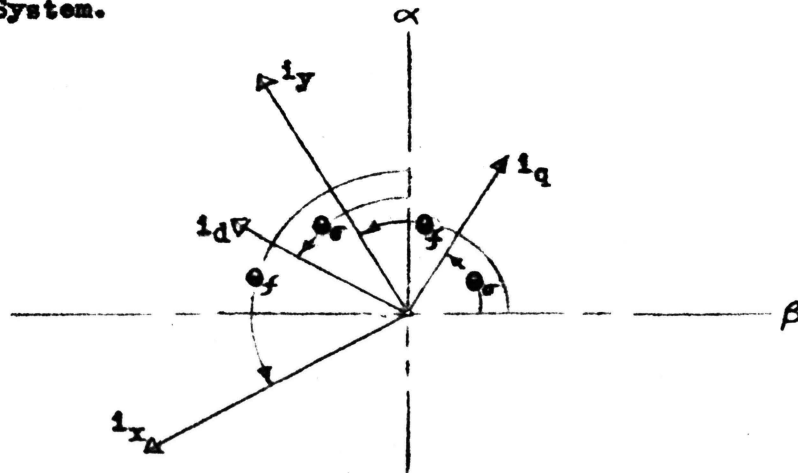
(a)

Actual Two Phase Machine, Showing Rotor and Stator Coils, and Rotation of Rotor and of Stator Flux.



(b)

Equivalent Circuit of Induction Machine in the Rotating Coordinate System.



(c)

Instantaneous Diagram, Showing Rotor and Stator Vectors and their Rotational Angles Relative to the α - β Axes

eliminated.

The preceding has been a more or less quantitative discussion of the α - β axis transformation. It will now be necessary to develop specific relations between the two phase quantities and the α - β quantities. These can be brought out by referring to Fig. 7-c, which is an instantaneous picture of the stator and rotor current vectors in reference to the synchronously rotating α - β axes. Each current vector is actually aligned with the physical axis of the coil indicated by its subscript. Thus vectors i_x and i_y rotate counterclockwise at synchronous speed ω_f ; vectors i_d and i_q rotate counterclockwise at slip speed ω_s . θ_f and θ_s are defined respectively as the instantaneous angles between i_x and the α axis and i_d and the α axis. The stator current I_α is then the projection of i_x and i_y on the α axis; I_β is the projection of i_x and i_y on the β axis, so that

$$(19-a) \quad I_\alpha = i_x \cos \theta_f + i_y \sin \theta_f$$

$$I_\beta = i_x \sin \theta_f + i_y \cos \theta_f$$

Also, by exact analogy

$$E_\alpha = e_x \cos \theta_f + e_y \sin \theta_f$$

$$(19-b) \quad E_\beta = -e_x \sin \theta_f + e_y \cos \theta_f$$

$$\Psi_\alpha = \Psi_x \cos \theta_f + \Psi_y \sin \theta_f$$

$$\Psi_\beta = -\Psi_x \sin \theta_f + \Psi_y \cos \theta_f$$

Similarly, the rotor current i_α is the projection of i_d and i_q on the α axis and i_β the projection of i_d and i_q on the β axis.

Then

$$i_{\alpha} = i_d \cos \theta_s + i_q \sin \theta_s$$

(20-a)

$$i_{\beta} = -i_d \sin \theta_s + i_q \cos \theta_s$$

and, as for the stator,

$$e_{\alpha} = e_d \cos \theta_s + e_q \sin \theta_s$$

$$e_{\beta} = -e_d \sin \theta_s + e_q \cos \theta_s$$

(20-b)

$$\Psi_{\alpha} = \Psi_d \cos \theta_s + \Psi_q \sin \theta_s$$

$$\Psi_{\beta} = -\Psi_d \sin \theta_s + \Psi_q \cos \theta_s$$

As was stated earlier, very simple flux linkage relations exist in this reference system. Referring to Fig. 7-b, and remembering that there can be no linkage between coils oriented 90° apart it can be written that

$$\Psi_{\alpha} = I_{\alpha} L_s + M i_{\alpha}$$

$$\Psi_{\beta} = I_{\beta} L_s + M i_{\beta}$$

(21)

$$\Psi_{\alpha} = i_{\alpha} L_R + M I_{\alpha}$$

$$\Psi_{\beta} = i_{\beta} L_R + M I_{\beta}$$

As the first step in solving relations (19), (20) and (21), take the time derivative of Ψ_{α} :

$$\begin{aligned} \frac{d\Psi_{\alpha}}{dt} &= -\Psi_d \sin \theta_s \frac{d\theta_s}{dt} + \frac{d\Psi_d}{dt} \cos \theta_s + \Psi_q \cos \theta_s \frac{d\theta_s}{dt} + \frac{d\Psi_q}{dt} \sin \theta_s \\ &= \frac{d\theta_s}{dt} (-\Psi_d \sin \theta_s + \Psi_q \cos \theta_s) + \frac{d\Psi_d}{dt} \cos \theta_s + \frac{d\Psi_q}{dt} \sin \theta_s \end{aligned}$$

But from equation (18)

$$(18) \quad \begin{aligned} e_d &= \frac{d\psi_d}{dt} + R_R i_d \\ e_g &= \frac{d\psi_g}{dt} + R_R i_g \end{aligned}$$

so that, since $\psi_\beta = -\psi_d \sin\theta_\sigma + \psi_g \cos\theta_\sigma$

$$\begin{aligned} \frac{d\psi_\alpha}{dt} &= \psi_\beta \frac{d\theta_\sigma}{dt} + (e_d - R_R i_d) \cos\theta_\sigma + (e_g - R_R i_g) \sin\theta_\sigma \\ &= \psi_\beta \frac{d\theta_\sigma}{dt} + (e_d \cos\theta_\sigma + e_g \sin\theta_\sigma) - R_R (i_d \cos\theta_\sigma + i_g \sin\theta_\sigma) \\ &= \psi_\beta \frac{d\theta_\sigma}{dt} + e_\alpha - R_R i_\alpha \end{aligned}$$

or, solving for

$$(22-a) \quad e_\alpha = \frac{d\psi_\alpha}{dt} - \psi_\beta \frac{d\theta_\sigma}{dt} + R_R i_\alpha$$

By an identical process, it can be shown that

$$\begin{aligned} e_\beta &= \frac{d\psi_\beta}{dt} + \psi_\alpha \frac{d\theta_\sigma}{dt} + R_R i_\beta \\ (22-b) \quad E_\alpha &= \frac{d\psi_\alpha}{dt} - \psi_\beta \frac{d\theta_\sigma}{dt} + r_s I_\alpha \\ E_\beta &= \frac{d\psi_\beta}{dt} + \psi_\alpha \frac{d\theta_\sigma}{dt} + r_s I_\beta \end{aligned}$$

To get equations (22) into terms of currents and voltages only, substitute equations (21) into them:

$$(23) \quad \begin{aligned} e_\alpha &= \frac{d}{dt} (L_R i_\alpha + M I_\alpha) - (L_R i_\beta + M I_\beta) \frac{d\theta_\sigma}{dt} + R_R i_\alpha \\ e_\beta &= \frac{d}{dt} (L_R i_\beta + M I_\beta) + (L_R i_\alpha + M I_\alpha) \frac{d\theta_\sigma}{dt} + R_R i_\beta \end{aligned}$$

$$E_{\alpha} = \frac{d}{dt} (L_s I_{\alpha} + M i_{\alpha}) - (L_s I_{\beta} + M i_{\beta}) \frac{d\theta}{dt} + r_s I_{\alpha}$$

$$E_{\beta} = \frac{d}{dt} (L_s I_{\beta} + M i_{\beta}) + (L_s I_{\alpha} + M i_{\alpha}) \frac{d\theta}{dt} + r_s I_{\beta}$$

One more equation can be written relating to the performance of the machine, namely, the electrical torque equation. Maxwell's equation for mechanical force between electric circuits can be stated for two circuits as

$$(24) \quad T = \frac{1}{2} I_1^2 \frac{dL_1}{d\theta} + \frac{1}{2} I_2^2 \frac{dL_2}{d\theta} + I_1 I_2 \frac{dM_{12}}{d\theta}$$

Since the circuits are rigid, and since the rotor and stator have been assumed to be smooth, the self inductances, L_1 and L_2 , are not functions of the angular position. Therefore $\frac{dL_1}{d\theta} = \frac{dL_2}{d\theta} = 0$ and

$$(25) \quad T = I_1 I_2 \frac{dM_{12}}{d\theta}$$

or, expressed in terms of the actual machine quantities

$$(26) \quad T_i = \sum I_R I_S \frac{dM}{d\theta}$$

It has been shown (see Fig. 5) that M is a cosine function of θ .

Therefore $\frac{dM}{d\theta}$ is a sine function of θ , or

$$(27) \quad T_i = \sum I_R I_S M (-\sin \theta) \times \frac{I_R}{I_S}$$

Then from Fig. 6, and equation (27)

$$T_i = M \left\{ \begin{array}{l} -i_x i_d \sin \theta_1 - i_x i_q \sin(\theta_1 + 90^\circ) \\ -i_y i_d \sin(\theta_1 - 90^\circ) - i_y i_q \sin \theta_1 \end{array} \right\}$$

$$= M \left\{ -i_x i_d \sin \theta_1 - i_x i_q \cos \theta_1 + i_y i_d \cos \theta_1 - i_y i_q \sin \theta_1 \right\}$$

$$\begin{aligned}
&= M \left\{ \begin{aligned} &-i_x i_d \sin(\theta_f - \theta_\sigma) - i_x i_q \cos(\theta_f - \theta_\sigma) \\ &+ i_y i_d \cos(\theta_f - \theta_\sigma) - i_y i_q \sin(\theta_f - \theta_\sigma) \end{aligned} \right\} \\
&= \left\{ \begin{aligned} &M \cos(\theta_f - \theta_\sigma) [-i_x i_q + i_y i_d] \\ &+ M \sin(\theta_f - \theta_\sigma) [-i_x i_d - i_y i_q] \end{aligned} \right\} \\
&= \left\{ \begin{aligned} &M(\cos \theta_f - \cos \theta_\sigma + \sin \theta_f \sin \theta_\sigma) [-i_x i_q + i_y i_d] \\ &+ M(\sin \theta_f \cos \theta_\sigma - \cos \theta_f \sin \theta_\sigma) [-i_x i_d - i_y i_q] \end{aligned} \right\} \\
&= M \left\{ \begin{aligned} &-i_x \sin \theta_f (i_d \cos \theta_\sigma + i_q \sin \theta_\sigma) \\ &-i_x \cos \theta_f (-i_d \sin \theta_\sigma + i_q \cos \theta_\sigma) \end{aligned} \right\} \\
&+ M \left\{ \begin{aligned} &-i_y \sin \theta_f (i_d \sin \theta_\sigma + i_q \cos \theta_\sigma) \\ &+ i_y \cos \theta_f (i_d \cos \theta_\sigma + i_q \sin \theta_\sigma) \end{aligned} \right\} \\
&= M \left\{ \begin{aligned} &(i_d \cos \theta_\sigma + i_q \sin \theta_\sigma) (-i_x \sin \theta_f + i_y \cos \theta_f) \\ &-(-i_d \sin \theta_\sigma + i_q \cos \theta_\sigma) (i_x \cos \theta_f + i_y \sin \theta_f) \end{aligned} \right\}
\end{aligned}$$

and

$$(28) \quad T_1 = M (I_\sigma i_\alpha - I_\alpha i_\sigma)$$

from equations (19-a) and (20-a). This relation could have been written directly, using equation (27) and referring to Fig. 7-(b).

Inspection of equations (23) and (28) shows that each equation contains terms involving the product of two time functions, i.e.,

$i_\beta(t) \theta_\sigma(t)$, $I_\sigma(t) i_\alpha(t)$, etc. In general it is not practical to attempt to take the Laplace transform of such products. Unless both time functions are known and their product is directly transformable,

the transform of a product is a complex convolution integral. This is an implicit transform, and the integration may be difficult or impossible to carry out. However, the input and output disturbances encountered in control systems, as opposed to those in servomechanisms, are usually small. Therefore it will be permissible to carry out the analysis on the basis of incremental variations of currents, voltages, and speeds. Then each of the variables in equations (23) and (28) can be written as a steady state plus an incremental term. Accordingly

$$\begin{aligned}
 e_{\alpha 0} + \Delta e_{\alpha} &= \frac{d}{dt} [L_R (i_{\alpha 0} + \Delta i_{\alpha}) + M (I_{\alpha 0} + \Delta I_{\alpha})] \\
 &\quad - [L_R (i_{\beta 0} + \Delta i_{\beta}) + M (I_{\beta 0} + \Delta I_{\beta})] \frac{d}{dt} (\theta_{s 0} + \Delta \theta_s) \\
 &\quad + R_R (i_{\alpha 0} + \Delta i_{\alpha}) \\
 &= \left[\frac{d}{dt} (L_R i_{\alpha 0} + M I_{\alpha 0}) - (L_R i_{\beta 0} + M I_{\beta 0}) \frac{d\theta_{s 0}}{dt} + R_R i_{\alpha 0} \right] \\
 &\quad + \left[\frac{d}{dt} (L_R \Delta i_{\alpha} + M \Delta I_{\alpha}) - \frac{d\theta_{s 0}}{dt} (L_R \Delta i_{\beta} + M \Delta I_{\beta}) - \frac{d\Delta \theta_s}{dt} (L_R i_{\beta 0} + M I_{\beta 0}) \right. \\
 &\quad \left. + R_R \Delta i_{\alpha} \right] \\
 &\quad + \left[\frac{d\Delta \theta_s}{dt} (L_R \Delta i_{\beta} + M \Delta I_{\beta}) \right]
 \end{aligned}$$

The steady state terms can be cancelled directly. Then, since increments are very small changes, the products of increments become negligibly small, and may be dropped. The first term of the above equation is the steady state portion, and the third is the product of two incremental quantities. Therefore

$$\begin{aligned}
 (29-a) \quad \Delta e_{\alpha} &= \frac{d}{dt} (L_R \Delta i_{\alpha} + M \Delta I_{\alpha}) - \frac{d\theta_{s 0}}{dt} (L_R \Delta i_{\beta} + M \Delta I_{\beta}) \\
 &\quad - \frac{d\Delta \theta_s}{dt} (L_R i_{\beta 0} + M I_{\beta 0}) + R_R \Delta i_{\alpha}
 \end{aligned}$$

By the same process

$$\Delta E_{\beta} = \frac{d}{dt} (L_R \Delta + M \Delta I_{\beta}) + \frac{d\theta_{s0}}{dt} (L_R \Delta i_{\alpha} + M \Delta I_{\alpha})$$

$$+ \frac{d\Delta\theta_s}{dt} (L_R i_{\alpha 0} + M I_{\alpha 0}) + R_R \Delta i_{\beta}$$

$$\Delta E_{\alpha} = \frac{d}{dt} (L_S \Delta I_{\alpha} + M \Delta i_{\beta}) - \frac{d\theta_{r0}}{dt} (L_S \Delta I_{\beta} + M \Delta i_{\beta})$$

(29-b)

$$- \frac{d\Delta\theta_r}{dt} (L_S I_{\beta 0} + M i_{\beta 0}) + r_S \Delta I_{\alpha}$$

$$\Delta E_{\beta} = \frac{d}{dt} (L_S \Delta I_{\beta} + M \Delta i_{\alpha}) + \frac{d\theta_{r0}}{dt} (L_S \Delta I_{\alpha} + M \Delta i_{\alpha})$$

$$- \frac{d\Delta\theta_r}{dt} (L_S I_{\alpha 0} + M i_{\alpha 0}) + r_S \Delta I_{\beta}$$

But $\frac{d\theta_{s0}}{dt} = \omega_{s0}$, and $\frac{d\Delta\theta_s}{dt} = \Delta\omega_s$. Furthermore, since the line voltage and frequency are constant, $\frac{d\theta_{r0}}{dt} = \omega_f$, and $\Delta E_{\alpha} = \Delta E_{\beta}$
 $= \frac{d\Delta\theta_r}{dt} = 0$. Also

$$\omega_s = \omega_f - \omega_1$$

(30)

$$\omega_{s0} + \Delta\omega_s = \omega_f - \omega_{10} - \Delta\omega_1$$

and

$$\Delta\omega_s = -\Delta\omega_1$$

which is to say, the variation of slip speed is equal in magnitude and opposite in sign to the variation of rotational speed. Also, from equation (21)

$$(31) \quad L_R i_{\alpha 0} + M I_{\alpha 0} = \Psi_{\alpha 0}$$

$$L_R i_{\beta 0} + M I_{\beta 0} = \Psi_{\beta 0}$$

Then equations (29) reduce to

$$\Delta e_x = \frac{d}{dt} (L_R \Delta i_\alpha + M \Delta I_\alpha) - \omega_{s0} (L_R \Delta i_\beta + M \Delta I_\beta)$$

$$+ \Psi_{\beta 0} \Delta \omega_1 + R_R \Delta i_\alpha$$

$$\Delta e_\beta = \frac{d}{dt} (L_R \Delta i_\beta + M \Delta I_\beta) + \omega_{s0} (L_R \Delta i_\alpha + M \Delta I_\alpha)$$

(32)

$$- \Psi_{\alpha 0} \Delta \omega_1 + R_R \Delta i_\beta$$

$$0 = \frac{d}{dt} (L_S \Delta I_\alpha + M \Delta i_\alpha) - \omega_f (L_S \Delta I_\beta + M \Delta i_\beta) + r_s \Delta I_\alpha$$

$$0 = \frac{d}{dt} (L_S \Delta I_\beta + M \Delta i_\beta) + \omega_f (L_S \Delta I_\alpha + M \Delta i_\alpha) + r_s \Delta I_\beta$$

Inspection of equations (29) and (32) will now show that time function products are no longer present, since as was shown in the discussion of transformation to the α - β axis system, steady state currents, voltages, and fluxes are constants in this system. These equations may now be Laplace transformed very simply. The notation (S) to indicate that the variables are now functions of S will be omitted for convenience in writing. Then

$$(33) \quad \Delta e_\alpha = (L_R S + R_R) \Delta i_\alpha + M S \Delta I_\alpha - \omega_{s0} (L_R \Delta i_\beta + M \Delta I_\beta) + \Psi_{\beta 0} \Delta \omega_1$$

$$(34) \quad \Delta e_\beta = (L_R S + R_R) \Delta i_\beta + M S \Delta I_\beta + \omega_{s0} (L_R \Delta i_\alpha + M \Delta I_\alpha) + \Psi_{\alpha 0} \Delta \omega_1$$

$$(35) \quad 0 = (L_s s + r_s) \Delta I_\alpha + M s \Delta i_\alpha - \omega_f (L_s \Delta I_\beta + M \Delta i_\beta)$$

$$(36) \quad 0 = (L_s s + r_s) \Delta I_\beta + M s \Delta i_\beta + \omega_f (L_s \Delta I_\alpha + M \Delta i_\alpha)$$

The torque equation (28) can be handled in the same manner. Thus

$$\begin{aligned} T_{10} + \Delta T_1 &= M \left[(I_{\beta 0} + \Delta I_\beta)(i_{\alpha 0} + \Delta i_\alpha) - (I_{\alpha 0} + \Delta I_\alpha)(i_{\beta 0} + \Delta i_\beta) \right] \\ &= M (I_{\beta 0} i_{\alpha 0} - I_{\alpha 0} i_{\beta 0}) + M (I_{\beta 0} \Delta i_\alpha + i_{\alpha 0} \Delta I_\beta) \\ &\quad + M (I_{\alpha 0} \Delta i_\beta - i_{\beta 0} \Delta I_\alpha) + M (\Delta I_\beta \Delta i_\alpha - \Delta I_\alpha \Delta i_\beta) \end{aligned}$$

from which

$$(37) \quad \Delta T_1 = M (I_{\beta 0} \Delta i_\alpha + i_{\alpha 0} \Delta i_\beta - I_{\alpha 0} \Delta i_\beta - i_{\beta 0} \Delta I_\alpha)$$

The transformed equation has the same form as (37), therefore it will not be necessary to rewrite it.

Since the rotor of the induction machine feeds the stator of the synchronous machine S_1 , the equations of the two machines must be matched through the common parameters, e_α , e_β , i_α , i_β . Therefore ΔI_α and ΔI_β are of no interest, and may be eliminated from equations (33) through (37). Accordingly, regroup equations (35) and (36)

$$(35) \quad (L_s s + r_s) \Delta I_\alpha - \omega_f L_s \Delta I_\beta = \omega_f M \Delta i_\beta - M s \Delta i_\alpha$$

$$(36) \quad \omega_f L_s \Delta I_\alpha + (L_s s + r_s) \Delta I_\beta = \omega_f M \Delta i_\alpha - M s \Delta i_\beta$$

Then, by determinants

$$\begin{aligned}
 (38) \quad \Delta I_{\alpha} &= \frac{\begin{vmatrix} M(\omega_f \Delta i_{\beta} - s \Delta i_{\alpha}) & -\omega_f L_s \\ -M(\omega_f \Delta i_{\alpha} + s \Delta i_{\beta}) & (L_s s + r_s) \end{vmatrix}}{\begin{vmatrix} (L_s s + r_s) & -\omega_f L_s \\ \omega_f L_s & (L_s s + r_s) \end{vmatrix}} \\
 &= \frac{M(\omega_f \Delta i_{\beta} - s \Delta i_{\alpha})(L_s s + r_s) - \omega_f L_s M(\omega_f \Delta i_{\beta} - s \Delta i_{\alpha})}{(L_s s + r_s)^2 + \omega_f^2 L_s^2} \\
 &= \frac{(X_{Mf} Z_s - X_{sf} M s) \Delta i_{\beta} - (X_{sf} X_{Mf} + M Z_s s) \Delta i_{\alpha}}{\delta_1}
 \end{aligned}$$

where

$$Z_s \triangleq L_s s + r_s$$

$$X_{sf} \triangleq \omega_f L_s = 60\text{-cycle stator self reactance.}$$

$$X_{Mf} \triangleq \omega_f M = 60\text{-cycle rotor-stator mutual reactance.}$$

$$\delta_1 \triangleq Z_s^2 + X_{sf}^2$$

Similarly

$$(39) \quad \Delta I_{\beta} = \frac{-(X_{sf} X_{Mf} + M Z_s s) \Delta i_{\beta} - (X_{Mf} Z_s - X_{sf} M s) \Delta i_{\alpha}}{\delta_1}$$

The above values for ΔI_{α} and ΔI_{β} may be substituted in equations (33), (34) and (37), resulting in three equations involving rotor currents, voltages, speed, and electrical torque. After some regrouping, these equations are given below.

$$\Delta e_{\alpha} = \left\{ Z_R + \frac{X_{Mf} (X_{Mf} Z_s - X_{sf} M s) - M s (X_{sf} X_{Mf} + M Z_s s)}{\delta_1} \right\} \Delta i_{\alpha} +$$

$$(40) \quad + \left\{ \frac{MS(X_{Mf}Z_s - X_{sf}MS) + X_{M\sigma 0}(X_{sf}X_{Mf} + MZ_s S)}{\delta_1} - X_{R\sigma 0} \right\} \Delta i_{\beta} + \psi_{\beta 0}$$

$$\Delta e_{\beta} = - \left\{ \frac{MS(X_{Mf}Z_s - X_{sf}MS) + X_{M\sigma 0}(X_{sf}X_{Mf} + MZ_s S)}{\delta_1} - X_{R\sigma 0} \right\} \Delta i_{\alpha}$$

$$(41) \quad + \left\{ Z_R + \frac{X_{M\sigma 0}(X_{Mf}Z_s - X_{sf}MS) - MS(X_{sf}X_{Mf} + MZ_s S)}{\delta_1} \right\} \Delta i_{\beta} - \psi_{\alpha 0} \Delta \omega_1$$

where

$X_{R\sigma 0} \triangleq \omega_{s0} L_R = \text{steady-state slip frequency rotor self reactance.}$

$X_{M\sigma 0} \triangleq \omega_{s0} M = \text{steady-state slip frequency rotor-stator mutual reactance.}$

$$(42) \quad \Delta T_1 = M \left\{ I_{\beta 0} + \frac{i_{\beta 0}(X_{sf}X_{Mf} + MZ_s S) - i_{\alpha 0}(X_{Mf}Z_s - X_{sf}MS)}{\delta_1} \right\} \Delta i_{\alpha}$$

$$-M \left\{ I_{\alpha 0} + \frac{i_{\alpha 0} (X_{sf} X_{mf} + M Z_s s) + i_{\beta 0} (X_{mf} Z_s - X_{sf} M s)}{\delta_1} \right\} \Delta i_{\beta}$$

In order to simplify subsequent manipulations, equations (40), (41), and (42) are now rewritten as follows:

$$(43) \quad \Delta e_{\alpha} = g_1(s) \Delta i_{\alpha} + g_2(s) \Delta i_{\beta} + \psi_{\beta 0} \Delta \omega_1$$

$$(44) \quad \Delta e_{\beta} = -g_2(s) \Delta i_{\alpha} + g_1(s) \Delta i_{\beta} - \psi_{\alpha 0} \Delta \omega_1$$

$$(45) \quad \Delta T_1 = g_3(s) \Delta i_{\alpha} - g_4(s) \Delta i_{\beta}$$

where $g_1(s)$, $g_2(s)$, $g_3(s)$, $g_4(s)$ are defined by the brackets in equations (40), (41), and (42).

By reference to Fig. 3, it is seen that T_1 is the electrical torque of machine M. However, since disturbances must originate at the load, T_L is the variable of chief interest. It can then be written that

$$(46) \quad T_1 = B_1 \frac{d\theta_1}{dt} + J_1 \frac{d^2\theta_1}{dt^2} + T_L = B_1 \omega_1 + J_1 \frac{d\omega_1}{dt} + T_L$$

Again taking increments

$$(47) \quad \Delta T_1 = B_1 \Delta \omega_1 + J_1 \frac{d\Delta \omega_1}{dt} + \Delta T_L$$

Transformed, equation (47) becomes

$$(48) \quad \begin{aligned} \Delta T_1 &= B_1 \Delta \omega_1 + J_1 s \Delta \omega_1 + \Delta T_L = (J_1 s + B_1) \Delta \omega_1 + \Delta T_L \\ &= Z_1 \Delta \omega_1 + \Delta T_L \end{aligned}$$

where

$$Z_1 \triangleq J_1 s + B_1 = \text{mechanical impedance of machine M.}$$

Substituting equation (45) into (48) and rearranging, there results

$$(49) \quad \Delta T_L = g_3(s) \Delta i_{\alpha} - g_4(s) \Delta i_{\beta} - Z_1 \Delta \omega_1$$

Equations (43), (44), and (49) completely describe the performance of the main machine, M , in terms of load torque and rotor quantities. These equations are perfectly general for small variations of input and output, and can be applied to an ordinary induction motor tied to a large source simply by making Δe_{α} and Δe_{β} equal to zero in equations (43) and (44). If the source cannot be considered infinite, ΔE_{α} and ΔE_{β} cannot be considered equal to zero and slightly different equations will result.

Synchronous Machine, S_1

The assumptions made for the induction machine will again be made. In addition, it will be assumed that all rotor circuits in addition to the main field winding can be adequately represented by two short-circuited windings, one in the direct, or main-field axis, the other in the quadrature axis. The ordinary synchronous machine may have, in addition to the main field winding, an amortisseur winding, a field collar, and circuits through the spider iron. In general, the amortisseur winding is a continuous ladder-like circuit completely encircling the rotor. Each bar forms a loop with every other bar, and all loops are mutually linked with each other and with all other rotor and stator circuits. The result is an extremely complex group of short circuited windings, each of which under transient conditions contributes a decaying exponential term to the disturbance effect. Exact analysis of such a circuit would be almost impossibly complicated. A notion of the complexity of the circuits encountered can be gotten by reference to a paper by Linville ⁽¹²⁾, in

(12) Linville, T. M., Starting Performance of Salient-Pole Synchronous Motors, AIEE Transactions, Vol. 49, 1930, pp. 531-547.

which some relatively exact rotor equivalent circuits are developed. Fig. 8 shows the decrement curve of a simple RL circuit superimposed on a typical decrement curve of an actual machine. The two curves actually coincide at only three times, 0, T, and ∞ . However, by proper selection of R and L the approximation can be made fair enough to make the mathematical simplification thus achieved more than worthwhile.

Making the assumption just discussed, the three-phase equivalent circuit of the synchronous machine can be represented as in Fig. 9. Impedances, currents, etc., used in writing the initial equations will be as indicated in the figure. M_{sf} , M_{sd} , and M_{sq} are the maximum values of mutual inductance between the stator and the main field, the short-circuited direct axis circuit, and the short-circuited quadrature axis circuit, respectively. Rotation is clockwise and at a speed ω_2 which in the steady state is equal to the ω_{s0} of induction machine M since the synchronous machine is fed by the induction machine rotor.

Then, in a manner exactly similar to that for the induction machine, the following equations may be written:

$$\begin{aligned} \Psi_{a1} = & L_{a3} i_1 + M_{a3} (i_2 + i_3) + I_f M_{af3} \cos \theta_2 \\ & + I_d M_{ad3} \cos \theta_2 - I_q M_{aq3} \sin \theta_2 \end{aligned}$$

$$\begin{aligned} \Psi_{a2} = & L_{a3} i_2 + M_{a3} (i_1 + i_3) + I_f M_{af3} \cos(\theta_2 - 120) \\ & + I_d M_{ad3} \cos(\theta_2 - 120) - I_q M_{aq3} \sin(\theta_2 - 120) \end{aligned}$$

$$\begin{aligned} (50) \quad \Psi_{a3} = & L_{a3} i_3 + M_{a3} (i_1 + i_2) + I_f M_{af3} \cos(\theta_2 + 120) \\ & + I_d M_{ad3} \cos(\theta_2 + 120) - I_q M_{aq3} \sin(\theta_2 + 120) \end{aligned}$$

$$\Psi_f = L_f I_f + M_{fd} I_d + i_1 M_{af3} \cos \theta_2 \\ + i_2 M_{af3} \cos(\theta_2 - 120) + i_3 M_{af3} \cos(\theta_2 + 120)$$

$$\Psi_d = L_d I_d + M_{fd} I_f + i_1 M_{ad3} \cos \theta_2 \\ + i_2 M_{ad3} \cos(\theta_2 - 120) + i_3 M_{ad3} \cos(\theta_2 + 120)$$

$$\Psi_g = L_g I_g - i_1 M_{ag3} \sin \theta_2 - i_2 M_{ag3} \sin(\theta_2 - 120) \\ - i_3 M_{ag3} \sin(\theta_2 + 120)$$

where θ_2 is the angle between phase a and the rotor direct axis. The same two simplifying transformations will again be made. Accordingly the stator is first replaced by its equivalent two phase stator, resulting in the circuit of Fig. 10, in which

$$i_d = \frac{2i_1 - i_2 - i_3}{3} = i_1 \quad (51)$$

$$i_g = \frac{i_2 - i_3}{\sqrt{3}} = -j i_1$$

$$\Psi_{ad} = \frac{2\Psi_{a1} - \Psi_{a2} - \Psi_{a3}}{3} = \Psi_{a1}$$

$$\Psi_{ag} = \frac{\Psi_{a2} - \Psi_{a3}}{\sqrt{3}} = -j \Psi_{a1} \quad (52)$$

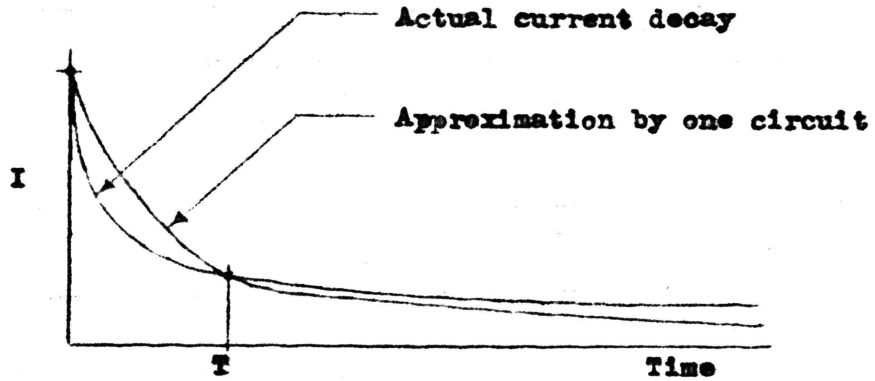


FIGURE 8 - Comparison of Transient Current Curve for One Rotor Circuit with the Curve of a Possible Actual Circuit.

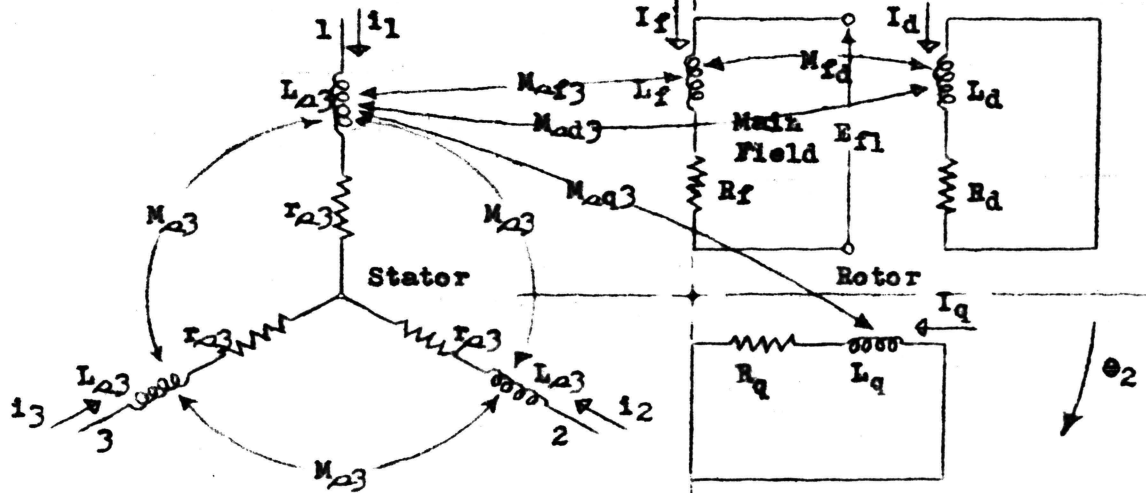


FIGURE 9 - Synchronous Machine Circuits.

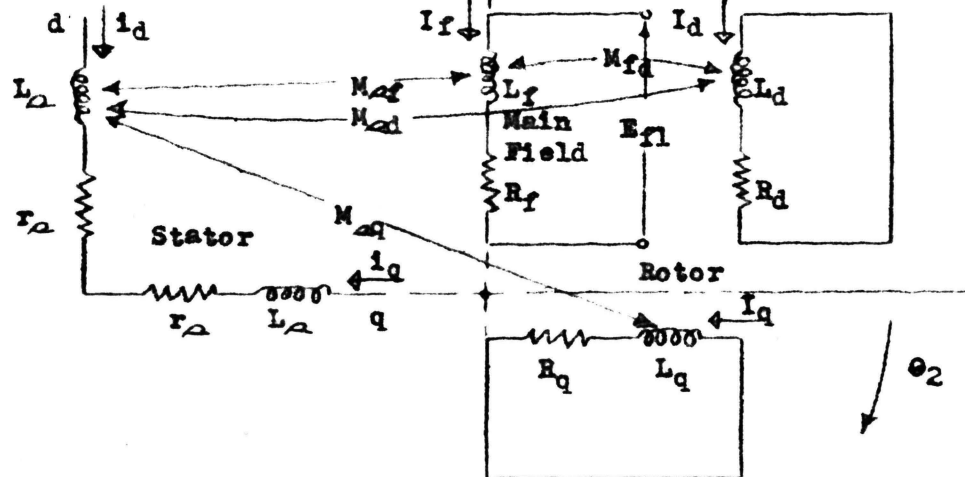


FIGURE 10 - Equivalent Two Phase Machine.

Then the linkage equations become

$$\Psi_{ad} = L_{ad} i_d + I_f M_{af} \cos \theta_2 + I_d M_{ad} \cos \theta_2 - I_g M_{ag} \sin \theta_2$$

$$\Psi_{ag} = L_{ag} i_g + I_f M_{af} \sin \theta_2 + I_d M_{ad} \sin \theta_2 + I_g M_{ag} \cos \theta_2$$

$$(53) \quad \Psi_f = L_f I_f + I_d M_{fd} + i_d M_{af} \cos \theta_2 + i_g M_{af} \sin \theta_2$$

$$\Psi_d = L_d I_d + I_f M_{fd} + i_d M_{ad} \cos \theta_2 + i_g M_{ad} \sin \theta_2$$

$$\Psi_g = L_g I_g - i_d M_{ag} \sin \theta_2 + i_g M_{ag} \cos \theta_2$$

As was the case for the induction machine, the values of the inductances must be altered in order to make the two phase machine exactly equivalent. From equations (52), $\Psi_{ad} = \Psi_{a1}$. Therefore substituting the proper values from equations (50) and (53) for Ψ_{ad} and Ψ_{a1} , there is, after cancellations

$$L_{ad} i_d = L_{a3} i_1 + M_{a3} (i_2 + i_3)$$

But from equations (51)

$$i_1 = i_d$$

$$i_2 + i_3 = -i_d$$

then

$$L_{ad} i_d = L_{a3} i_d - M_{a3} i_d$$

and

$$(54) \quad L_{ad} = L_{a3} - M_{a3}$$

Since the field circuits are unchanged, it follows that Ψ_f , Ψ_d , and Ψ_g must remain unchanged. Therefore again equating the appropriate terms from equations (50) and (53), there results

$$\begin{aligned}
M_{af}(i_d \cos \theta_2 + i_q \sin \theta_2) &= M_{af3} [i_1 \cos \theta_2 + i_2 \cos(\theta_2 - 120) + i_3 \cos(\theta_2 + 120)] \\
&= \frac{3}{2} M_{af3} \left[i_1 \cos \theta_2 + \left(\frac{i_2 - i_3}{\sqrt{3}} \right) \sin \theta_2 \right] \\
&= \frac{3}{2} M_{af3} (i_d \cos \theta_2 + i_q \sin \theta_2)
\end{aligned}$$

from equations (51). Thus

$$(55) \quad M_{af} = \frac{3}{2} M_{af3}$$

and since the Ψ_d equations are identical in form

$$(56) \quad M_{ad} = \frac{3}{2} M_{ad3}$$

In a similar manner

$$\begin{aligned}
-i_d M_{aq} \sin \theta_2 + i_q M_{aq} \cos \theta_2 &= -M_{aq3} [i_1 \sin \theta_2 + i_2 \sin(\theta_2 - 120) + i_3 \sin(\theta_2 + 120)]
\end{aligned}$$

The right hand side of the above equation can be shown to be equal to

$$-M_{aq3} \left\{ \left[i_1 - \frac{(i_2 + i_3)}{2} \right] \sin \theta_2 - \left[\frac{\sqrt{3}}{2} (i_2 - i_3) \right] \cos \theta_2 \right\}$$

which, again from equations (51) reduces so that

$$\text{and} \quad -M_{aq} (i_d \sin \theta_2 - i_q \cos \theta_2) = -\frac{3}{2} M_{aq3} (i_d \sin \theta_2 - i_q \cos \theta_2)$$

$$(57) \quad M_{aq} = \frac{3}{2} M_{aq3}$$

Equations (54) through (57) completely prescribe the relations which must hold between the three phase and the equivalent two phase machine.

Referring everything to a rotating reference frame, this time rotating clockwise at rotor speed ω_2 , the synchronous machine reduces to a static network in the same manner as did the induction machine. This network is shown in Fig. 11. Since in this case the rotor is turning at reference frame speed, the direct and quadrature axes of the rotor may be taken directly as references, and the rotor quantities remain unchanged.

The stator of the original machine is rotating counterclockwise at a speed ω_2 in the new reference system. Then the new stator equations are

$$\begin{aligned}
 i_\alpha &= i_d \cos \theta_2 + i_q \sin \theta_2 \\
 i_\beta &= -i_d \sin \theta_2 + i_q \cos \theta_2 \\
 e_\alpha &= e_d \cos \theta_2 + e_q \sin \theta_2 \\
 (58) \quad e_\beta &= -e_d \sin \theta_2 + e_q \cos \theta_2 \\
 \psi_{\alpha d} &= \psi_{sd} \cos \theta_2 + \psi_{sq} \sin \theta_2 \\
 \psi_{\alpha q} &= -\psi_{sd} \sin \theta_2 + \psi_{sq} \cos \theta_2
 \end{aligned}$$

The rotor relations are unchanged, but will be repeated here for easy reference:

$$\begin{aligned}
 I_{f\alpha} &= I_f \\
 E_{f\alpha} &= E_f \\
 \Psi_{f\alpha} &= \Psi_d \\
 I_{d\alpha} &= I_d \\
 (59) \quad \Psi_{d\alpha} &= \Psi_d \\
 I_{g\beta} &= I_g \\
 \Psi_{g\beta} &= \Psi_g \\
 I_{f\beta} = E_{f\beta} = \Psi_{f\beta} = I_{d\beta} = E_{d\alpha} = E_{d\beta} = \Psi_{d\beta} = I_{g\alpha} = E_{g\alpha} = E_{g\beta} = \Psi_{g\alpha} &= 0
 \end{aligned}$$

By a procedure identical to that for the induction machine, it can be shown that

$$(60) \quad \begin{aligned} e_{\alpha} &= \frac{d\psi_{s\alpha}}{dt} - \psi_{s\beta} \frac{d\theta_2}{dt} + r_{s\alpha} i_{\alpha} \\ e_{\beta} &= \frac{d\psi_{s\beta}}{dt} + \psi_{s\alpha} \frac{d\theta_2}{dt} + r_{s\beta} i_{\beta} \end{aligned}$$

Next the loop equation for each rotor circuit can be written,

$$(61) \quad \begin{aligned} E_f &= \frac{d\psi_f}{dt} + R_f I_f \\ 0 &= \frac{d\psi_d}{dt} + R_d I_d \\ 0 &= \frac{d\psi_q}{dt} + R_q I_q \end{aligned}$$

and referring to Fig. 11, the following simple flux relations are seen to hold:

$$(62) \quad \begin{aligned} \psi_{s\alpha} &= L_{s\alpha} i_{\alpha} + M_{sf} I_f + M_{sd} I_d \\ \psi_{s\beta} &= L_{s\beta} i_{\beta} + M_{sq} I_q \\ \psi_f &= L_f I_f + M_{fd} I_d + M_{f\alpha} i_{\alpha} \\ \psi_d &= L_d I_d + M_{fd} I_f + M_{d\alpha} i_{\alpha} \\ \psi_q &= L_q I_q + M_{sq} i_{\beta} \end{aligned}$$

The flux terms can now be eliminated from equations (60) and (61) by substitution of equations (62) into them:

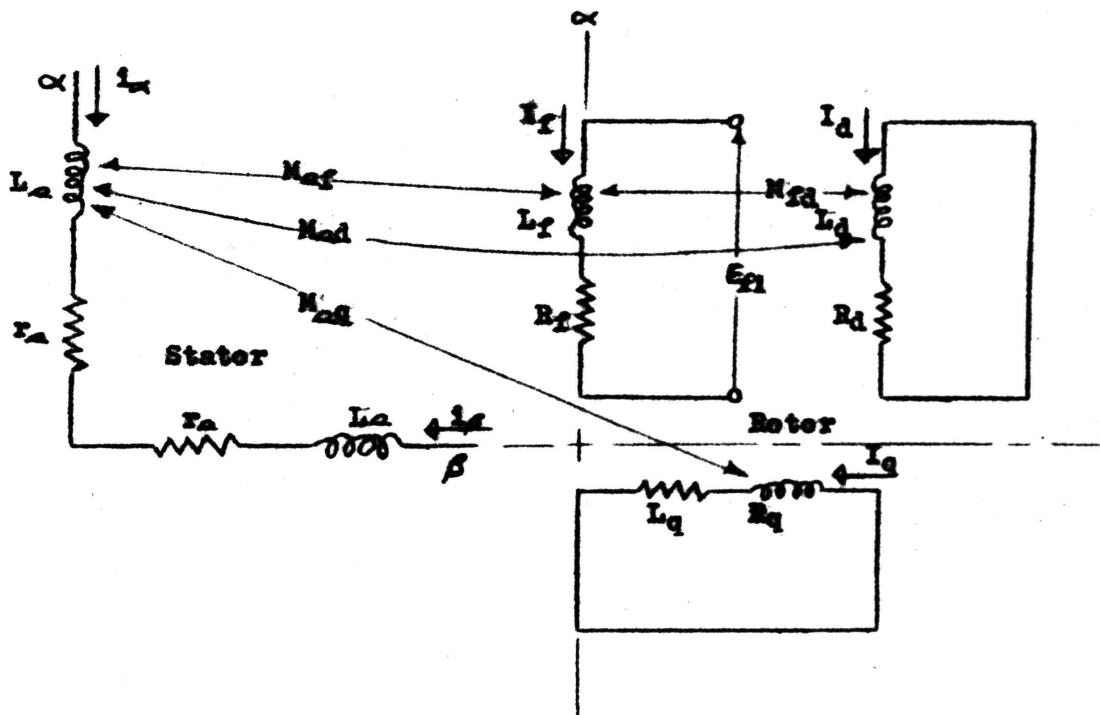


FIGURE 11 -- Equivalent Machine Reduced to Static Network Fixed to Rotating α - β Reference Frame.

$$e_{\alpha} = \frac{d}{dt} (L_{\alpha} i_{\alpha} + M_{\alpha f} I_f + M_{\alpha d} I_d) - (L_{\alpha} i_{\beta} + M_{\alpha g} I_g) \frac{d\theta_2}{dt} + r_{\alpha} i_{\alpha}$$

$$e_{\beta} = \frac{d}{dt} (L_{\alpha} i_{\beta} + M_{\alpha g} I_g) + (L_{\alpha} i_{\alpha} + M_{\alpha f} I_f + M_{\alpha d} I_d) \frac{d\theta_2}{dt} + r_{\alpha} i_{\beta}$$

$$(63) \quad E_f = \frac{d}{dt} (L_f I_f + M_{fd} I_d + M_{\alpha f} i_{\alpha}) + R_f I_f$$

$$0 = \frac{d}{dt} (L_d I_d + M_{\alpha d} I_f + M_{\alpha d} i_{\alpha}) + R_d I_d$$

$$0 = \frac{d}{dt} (L_g I_g + M_{\alpha g} i_{\beta}) + R_g I_g$$

Using equation (27) and Fig. 11, the torque equation is next obtained:

$$(64) \quad T_2 = i_{\beta} (M_{\alpha f} I_f + M_{\alpha d} I_d) - M_{\alpha g} i_{\alpha} I_g$$

Torques tending to turn the stator circuit in the direction of the actual stator's rotation with respect to the reference frame have been taken as positive, and vice versa.

The six equations (63) and (64) describe the performance of the synchronous machine. However it will again be necessary to take increments in order to make them transformable, i.e., to consider only small deviations in input and output. Thus

$$\Delta e_{\alpha} = \frac{d}{dt} (L_{\alpha} \Delta i_{\alpha} + M_{\alpha f} \Delta I_f + M_{\alpha d} \Delta I_d) - \frac{d\theta_{20}}{dt} (L_{\alpha} \Delta i_{\beta} + M_{\alpha g} \Delta I_g) - \frac{d\Delta\theta_2}{dt} (L_{\alpha} i_{\beta 0} + M_{\alpha g} I_{g0}) + r_{\alpha} \Delta i_{\alpha}$$

$$(65) \quad \Delta e_{\beta} = \frac{d}{dt} (L_{\alpha} \Delta i_{\beta} + M_{\alpha g} \Delta I_g) + \frac{d\theta_{20}}{dt} (L_{\alpha} \Delta i_{\alpha} + M_{\alpha f} \Delta I_f + M_{\alpha d} \Delta I_d) + \frac{d\Delta\theta_2}{dt} (L_{\alpha} i_{\alpha 0} + M_{\alpha f} I_{f0} + M_{\alpha d} I_{d0}) + r_{\alpha} \Delta i_{\beta}$$

$$\Delta E_f = \frac{d}{dt} (L_f \Delta I_f + M_{fd} \Delta I_d + M_{af} \Delta i_\alpha) + R_f \Delta I_f$$

$$0 = \frac{d}{dt} (L_d \Delta I_d + M_{fd} \Delta I_f + M_{ad} \Delta i_\alpha) + R_d \Delta I_d$$

$$0 = \frac{d}{dt} (L_g \Delta I_g + M_{ag} \Delta i_\beta) + R_g \Delta I_g$$

But $\frac{d\theta_{20}}{dt} = \omega_{20} = \omega_{s0}$; $\frac{d\Delta\theta_2}{dt} = \Delta\omega_2$ and from equations (62)

$$(66) \quad L_a i_{\beta 0} + M_{ag} I_{g0} = \Psi_{a\beta 0}$$

$$L_a i_{\alpha 0} + M_{af} I_{f0} + M_{ad} I_{d0} = \Psi_{a\alpha 0}$$

Also, it was originally assumed that \mathbb{E}_f was to be constant. Therefore,

$\Delta E_f = 0$, so that equations (65) can be rewritten as

$$\Delta e_\alpha = \frac{d}{dt} (L_a \Delta i_\alpha + M_{af} \Delta I_f + M_{ad} \Delta I_d) - \omega_{s0} (L_a \Delta i_\beta + M_{ag} \Delta I_g) - \Psi_{a\beta 0} \Delta\omega_2 + r_a \Delta i_\alpha$$

$$\Delta e_\beta = \frac{d}{dt} (L_a \Delta i_\beta + M_{ag} \Delta I_g)$$

$$(67) \quad + \omega_{s0} (L_a \Delta i_\alpha + M_{af} \Delta I_f + M_{ad} \Delta I_d) + \Psi_{a\alpha 0} \Delta\omega_2 + r_a \Delta i_\beta$$

$$0 = \frac{d}{dt} (L_f \Delta I_f + M_{fd} \Delta I_d + M_{af} \Delta i_\alpha) + R_f \Delta I_f$$

$$0 = \frac{d}{dt} (L_d \Delta I_d + M_{fd} \Delta I_f + M_{ad} \Delta i_\alpha) + R_d \Delta I_d$$

$$0 = \frac{d}{dt} (L_g \Delta I_g + M_{ag} \Delta i_\beta) + R_g \Delta I_g$$

Taking increments in equation (64) there is obtained

$$(68) \quad \Delta T_2 = i_{\beta 0} (M_{af} \Delta I_f + M_{ad} \Delta I_d) + (M_{af} I_{f0} + M_{ad} I_{d0}) \Delta i_\beta - M_{ag} (i_{\alpha 0} \Delta I_g + I_{g0} \Delta i_\alpha)$$

Equations (67) and (68) may now be transformed:

$$\Delta e_\alpha = Z_a \Delta i_\alpha + s(M_{af} \Delta I_f + M_{ad} \Delta I_d) - \omega_{s0}(L_a \Delta I_\beta + M_{ag} \Delta I_g) - \Psi_{a\beta 0} \Delta \omega_2$$

$$\Delta e_\beta = Z_a \Delta i_\beta + sM_{ag} \Delta I_g + \omega_{s0}(L_a \Delta i_\alpha + M_{af} \Delta I_f + M_{ad} \Delta I_d) - \Psi_{a\alpha 0} \Delta \omega_2$$

$$(69) \quad 0 = Z_f \Delta I_f + s(M_{fd} \Delta I_d + M_{af} \Delta i_\alpha)$$

$$0 = Z_d \Delta I_d + s(M_{fd} \Delta I_f + M_{ad} \Delta i_\alpha)$$

$$0 = Z_g \Delta I_g + sM_{ag} \Delta i_\beta$$

$$\Delta T_2 = i_{\beta 0} (M_{af} \Delta I_f + M_{ad} \Delta I_d) + (M_{af} I_{f0} + M_{ad} I_{d0}) \Delta i_\beta - M_{ag} (i_{\alpha 0} \Delta I_g + I_{g0} \Delta i_\alpha)$$

where

$$Z_a \triangleq L_a s + R_a$$

$$Z_f \triangleq L_f s + R_f$$

$$Z_d \triangleq L_d s + R_d$$

$$Z_g \triangleq L_g s + R_g$$

Again the rotor currents are of no interest. Accordingly they may be eliminated from equations (69). From the quadrature axis equation

$$(70) \quad \Delta I_g = - \frac{sM_{ag} \Delta i_\beta}{Z_g}$$

and from the main field and direct axis damper equations solved simultaneously:

$$(71) \quad \Delta I_f = \frac{s(M_{fd} M_{ad} - M_{af} Z_d)}{\delta_2}$$

$$(72) \quad \Delta I_d = \frac{s(M_{fd} M_{af} - M_{ad} Z_f)}{\delta_2}$$

where

$$\delta_2 = Z_f Z_d - s^2 M_{fd}^2$$

Substituting equations (70) through (72) into equations (69) there is obtained

$$\begin{aligned} \Delta e_\alpha &= \frac{s^2 [M_{af}(M_{fd} M_{ad} - M_{af} Z_d) + M_{ad}(M_{fd} M_{af} - M_{ad} Z_f)]}{\delta_2} \Delta i_\alpha \\ &\quad + Z_a \Delta i_\alpha - \omega_{s0} \left(L_a - \frac{s M_{ag}^2}{Z_g} \right) \Delta i_\beta - \psi_{s\beta 0} \Delta \omega_2 \\ (73) \quad \Delta e_\beta &= \omega_{s0} \left\{ L_a + \frac{s [M_{af}(M_{fd} M_{ad} - M_{af} Z_d) + M_{ad}(M_{fd} M_{af} - M_{ad} Z_f)]}{\delta_2} \right\} \Delta i_\alpha \\ &\quad + \left(Z_a - \frac{s^2 M_{ag}^2}{Z_g} \right) \Delta i_\beta + \psi_{s\alpha 0} \Delta \omega_2 \\ \Delta T_2 &= i_{\beta 0} s \left[\frac{M_{af}(M_{fd} M_{ad} - M_{af} Z_d) + M_{ad}(M_{fd} M_{af} - M_{ad} Z_f)}{\delta_2} \right] \Delta i_\alpha \\ &\quad - M_{ag} I_{g0} \Delta i_\alpha + \left(M_{af} I_{f0} + M_{ad} I_{d0} + \frac{i_{\alpha 0} M_{ag}^2}{Z_g} \right) \Delta i_\beta \end{aligned}$$

After some regrouping, equations (73) may be written as

$$(74) \Delta e_{\alpha} = \left\{ Z_a + \frac{s^2 [M_{af}(SM_{fd}M_{ad} - M_{af}Z_d) + M_{ad}(SM_{fd}M_{af} - M_{ad}Z_f)]}{\delta_2} \right\} \Delta i_{\alpha} \\ - \left(X_{a60} - \frac{SM_{ag60}M_{ag}}{Z_g} \right) \Delta i_{\beta} - \psi_{a\beta 0} \Delta \omega_2$$

$$(75) \Delta e_{\beta} = \left\{ X_{a60} + \frac{s [X_{M_{af}60}(SM_{fd}M_{ad} - M_{af}Z_d) + X_{M_{ad}60}(SM_{fd}M_{af} - M_{ad}Z_f)]}{\delta_2} \right\} \Delta i_{\alpha} \\ - \left(Z_a - \frac{s^2 M_{ag}}{Z_g} \right) \Delta i_{\beta} + \psi_{a\alpha 0} \Delta \omega_2$$

$$(76) \Delta T_2 = \left\{ \frac{i_{\beta 0} s [M_{af}(SM_{fd}M_{ad} - M_{af}Z_d) + M_{ad}(SM_{fd}M_{af} - M_{ad}Z_f)]}{\delta_2} - M_{ag} I_{g0} \right\} \Delta i_{\alpha} \\ + \left(M_{af} I_{f0} + M_{ad} I_{d0} + \frac{SM_{ag}^2 i_{\alpha 0}}{Z_g} \right) \Delta i_{\beta}$$

Where

$X_{a60} = \omega_{60} L_a =$ Slip frequency stator self reactance

$X_{M_{ag}60} = \omega_{60} M_{ag} =$ Slip frequency stator-quadrature axis mutual reactance

$X_{M_{af}60} = \omega_{60} M_{af} =$ Slip frequency stator-main field mutual reactance

$X_{M_{ad}60} = \omega_{60} M_{ad} =$ Slip frequency stator-direct axis mutual reactance

Then for simplification the above equations are rewritten as follows:

$$(77) \Delta e_{\alpha} = h_1(s) \Delta i_{\alpha} - h_2(s) \Delta i_{\beta} - \psi_{a\beta 0} \Delta \omega_2$$

$$(78) \quad \Delta e_{\beta} = h_3(s) \Delta i_{\alpha} + h_4(s) \Delta i_{\beta} + \Psi_{a\alpha 0} \Delta \omega_2$$

$$(79) \quad \Delta T_2 = h_5(s) \Delta i_{\alpha} + h_6(s) \Delta i_{\beta}$$

Where $h_1(s)$, etc., are defined by the appropriate brackets in equations (74) through (76). ΔT_2 is the electrical torque of the synchronous machine. As a final step, the effects of inertia and friction of this machine plus machine DC, will be included. Thus

$$(80) \quad \Delta T_2 = B_2 \Delta \omega_2 + J_2 s \Delta \omega_2 + \Delta T_{DC1} = Z_2 \Delta \omega_2 + T_{DC1}$$

where

$$Z_2 = J_2 s + B_2 = \text{mechanical impedance of machines } S_1 \text{ and } DC_1.$$

Finally, from equations (79) and (80)

$$(81) \quad \Delta T_{DC1} = h_5(s) \Delta i_{\alpha} + h_6(s) \Delta i_{\beta} - Z_2 \Delta \omega_2$$

Equations (77), (78), and (81) completely describe the performance of synchronous machine S_1 working into a mechanical load DC_1 . These equations are general for small variations of input and output. They have been derived on the assumption that the machine is being supplied from a non-infinite source, i.e., variations in load may change the input voltage. To apply them to a machine supplied from a large source and having variable excitation, Δe_{α} and Δe_{β} would be equated to zero, whereas ΔE_f would no longer be zero.

Solution for the Double-Fed Machine

Since the rotor of machine M and the stator of machine S_1 are tied together, obviously the quantities Δe_{α} , Δe_{β} , Δi_{α} , and Δi_{β} are common in the two machines. It is now desired to obtain a relation bridging the two machines, i.e., between ΔT_1 and ΔT_{DC1} . This can

be done by equating Δe_α and Δe_β from equations (43) and (44) to the corresponding quantities from equations (78) and (79). This eliminates Δe_α and Δe_β . Since the speed control variations are to come through ΔT_{Dc} , it is also apparent that Δi_α and Δi_β are no longer of interest. Accordingly from equations (43) and (77)

$$(82) \quad g_1(s)\Delta i_\alpha + g_2(s)\Delta i_\beta + \psi_{\beta 0}\Delta\omega_1 = h_1(s)\Delta i_\alpha - h_2(s)\Delta i_\beta - \psi_{\alpha\beta 0}\Delta\omega_2$$

and from equations (44) and (78)

$$(83) \quad -g_2(s)\Delta i_\alpha + g_1(s)\Delta i_\beta - \psi_{\alpha 0}\Delta\omega_1 = h_3(s)\Delta i_\alpha + h_4(s)\Delta i_\beta + \psi_{\alpha\alpha 0}\Delta\omega_2$$

Equations (82) and (83) may be rearranged as follows:

$$(84) \quad [h_1(s) - g_1(s)]\Delta i_\alpha - [h_2(s) + g_2(s)]\Delta i_\beta = \psi_{\beta 0}\Delta\omega_1 + \psi_{\alpha\beta 0}\Delta\omega_2$$

$$(85) \quad -[g_2(s) + h_3(s)]\Delta i_\alpha + [g_1(s) - h_4(s)]\Delta i_\beta = \psi_{\alpha 0}\Delta\omega_1 + \psi_{\alpha\alpha 0}\Delta\omega_2$$

Solving (84) and (85) for Δi_α and Δi_β , there is obtained, after regrouping

$$(86) \quad \Delta i_\alpha = \frac{\left[\begin{aligned} &\{ \psi_{\beta 0} [g_1(s) - h_4(s)] + \psi_{\alpha 0} [g_2(s) + h_2(s)] \} \Delta\omega_1 \\ &+ \{ \psi_{\alpha\beta 0} [g_1(s) - h_4(s)] + \psi_{\alpha\alpha 0} [g_2(s) + h_2(s)] \} \Delta\omega_2 \end{aligned} \right]}{\delta_3}$$

$$(87) \quad \Delta i_\beta = \frac{\left[\begin{aligned} &\{ \psi_{\beta 0} [g_2(s) + h_3(s)] - \psi_{\alpha 0} [g_1(s) - h_1(s)] \} \Delta\omega_1 \\ &+ \{ \psi_{\alpha\beta 0} [g_2(s) + h_3(s)] - \psi_{\alpha\alpha 0} [g_1(s) - h_1(s)] \} \Delta\omega_2 \end{aligned} \right]}{\delta_3}$$

where

$$\delta_3 = [h_1(s) - g_1(s)][g_1(s) - h_4(s)] + [g_2(s) + h_3(s)][g_2(s) + h_2(s)]$$

For convenience in writing define

$$f_1(s) = \Psi_{\beta 0} [g_1(s) - h_4(s)] + \Psi_{\alpha 0} [g_2(s) + h_2(s)]$$

$$f_2(s) = \Psi_{\alpha \beta 0} [g_1(s) - h_4(s)] + \Psi_{\alpha \alpha 0} [g_2(s) + h_2(s)]$$

$$f_3(s) = \Psi_{\beta 0} [g_2(s) + h_3(s)] - \Psi_{\alpha 0} [g_1(s) - h_1(s)]$$

$$f_4(s) = \Psi_{\alpha \beta 0} [g_2(s) + h_3(s)] - \Psi_{\alpha \alpha 0} [g_1(s) - h_1(s)]$$

Then

$$(88) \quad \Delta i_{\alpha} = \frac{f_1(s) \Delta \omega_1 + f_2(s) \Delta \omega_2}{\delta_2}$$

$$(89) \quad \Delta i_{\beta} = \frac{f_3(s) \Delta \omega_1 + f_4(s) \Delta \omega_2}{\delta_2}$$

Using equations (88) and (89), Δi_{α} and Δi_{β} may now be eliminated from equations (49) and (81):

$$(90) \quad \Delta T_L = g_3(s) \left[\frac{f_1(s) \Delta \omega_1 + f_2(s) \Delta \omega_2}{\delta_3} \right] - g_4(s) \left[\frac{f_3(s) \Delta \omega_1 + f_4(s) \Delta \omega_2}{\delta_3} \right] - Z_1 \Delta \omega_1$$

$$(91) \quad \Delta T_{DC1} = h_5(s) \left[\frac{f_1(s) \Delta \omega_1 + f_2(s) \Delta \omega_2}{\delta_3} \right] + h_6(s) \left[\frac{f_3(s) \Delta \omega_1 + f_4(s) \Delta \omega_2}{\delta_3} \right] - Z_1 \Delta \omega_2$$

or, rearranged

$$(90) \quad \Delta T_L = \left[\frac{g_3(s) f_1(s) - g_4(s) f_3(s)}{\delta_3} - Z_1 \right] \Delta \omega_1 + \left[\frac{g_3(s) f_2(s) - g_4(s) f_4(s)}{\delta_3} \right] \Delta \omega_2$$

$$(91) \quad \Delta T_{DC_1} = \left[\frac{h_5(s)f_1(s) + h_6(s)f_3(s)}{\delta_3} \right] \Delta \omega_1 + \left[\frac{h_5(s)f_2(s) + h_6(s)f_4(s)}{\delta_3} - Z_2 \right] \Delta \omega_2$$

or, in more abbreviated form

$$(92) \quad \Delta T_L = G_1(s) \Delta \omega_1 + G_2(s) \Delta \omega_2$$

$$(93) \quad \Delta T_{DC_1} = G_3(s) \Delta \omega_1 + G_4(s) \Delta \omega_2$$

Eliminating $\Delta \omega_2$ from the above equations, the final equation is obtained.

$$(94) \quad \Delta T_{DC_1} = \frac{G_4(s)}{G_2(s)} \Delta T_L + \left[G_3(s) - \frac{G_1(s)G_4(s)}{G_2(s)} \right] \Delta \omega_1$$

Equation (94) is the complete solution of the double-fed machine in terms of load torques and induction motor speed. This solution can be tied through ΔT_{DC_1} to the solution for the direct current machine group.

Direct Current Machines, DC and DC₂

The DC machine relations used in this development are well known, and may be found in any standard text on DC Machinery, e.g., Dawes.⁽¹³⁾

(13) Dawes, C. L., *Electrical Engineering, Vol. I., Direct Currents*, McGraw-Hill Book Co., N. Y., 1936, pp. 397-495.

The usual assumptions will be made neglecting hysteresis and eddy currents. The effect of armature slots will also be neglected.

Since the armature circuits of the two machines form one simple loop, they may conveniently be considered together. The combined electric circuit is as shown in Fig. 12. The parameter which ties the machines together is of course I_A .

In any DC machine the torque is proportional to flux and armature current. Therefore the following equations can be written:

$$(95) T_{DC_1} = K_1 \Phi_1 I_A$$

$$(96) T_{DC_2} = K_1 \Phi_2 I_A \quad (14)$$

where

$$K_1 = \frac{PN}{20\pi P'} \times \frac{1}{4.45 \times 10^5 \times 2.5 \times 12}$$

(14) Dawes, C. L., op. cit., pages 401 and 484.

and

P = number of poles

N = number of conductors

P' = number of parallel armature paths.

The field flux linking the armature at open circuit is

$$\Phi_f = K_f I_{f1}$$

Where K_f is a coefficient of such value that $(L_{f1} - K_f) I_{f1}$ represents the field leakage flux. When the armature is carrying current, however, the flux is decreased because of armature reaction, and as the current increases, because of saturation. When current flows in the armature, some of the flux it sets up acts in direct opposition to the field flux, i.e., is a demagnetizing flux. This demagnetizing flux is a function of current, brush position, and reluctance of the flux paths. In addition, the coefficient K_f is a function of reluctance. The reluctance of any circuit containing iron is in turn a function of saturation, so that since K_f has been defined at no load, it cannot be said to hold strictly when the armature carries current. No simple relation can be written to express the overall change in flux due to these factors. However, an ap-

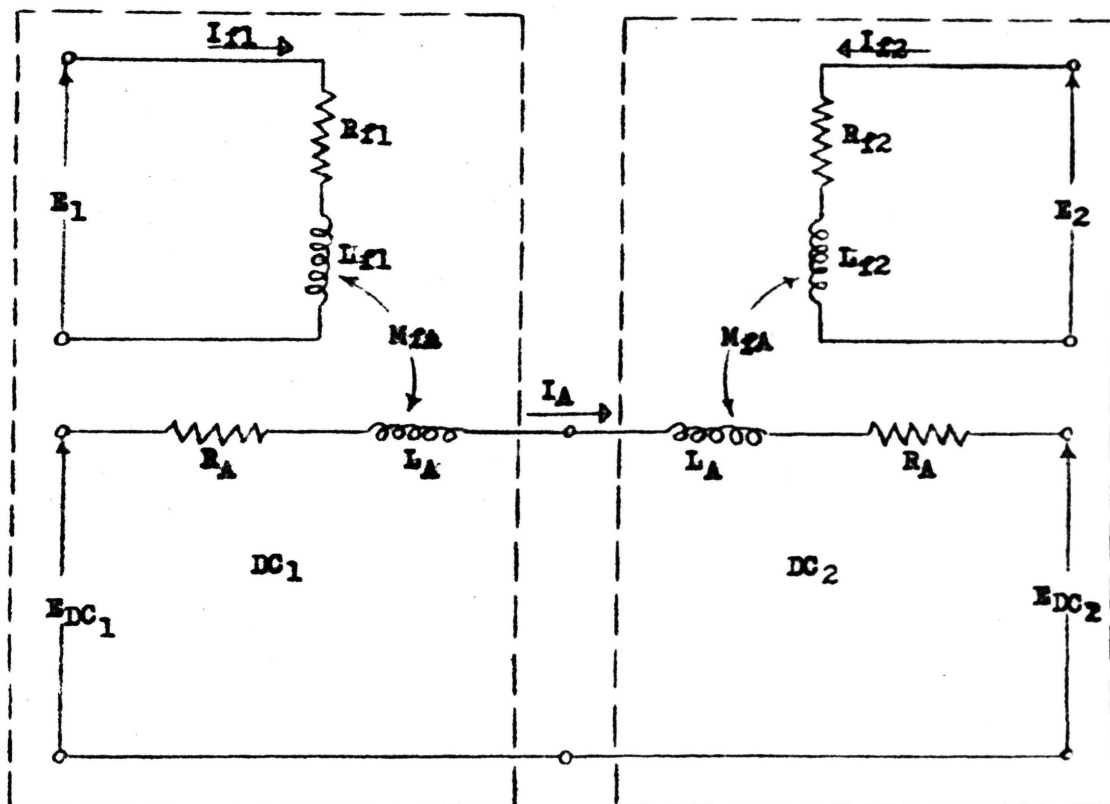


FIGURE 12 - Electric Circuit of DC Machines

proximation can be made by assuming the variation in flux is directly proportional to armature current. This is a good approximation over a considerable range because of the air gaps in the magnetic circuit.

Then for machine DC₁ operating as a generator

$$(97) \quad \Phi_1 = K_f I_{f1} - A_1 I_A$$

and for DC₂, operating as a motor

$$(98) \quad \Phi_2 = K_f I_{f2} - A_2 I_A$$

Also, referring to the field circuits of DC₁ and DC₂, Fig. 12,

$$(99) \quad E_1 = L_{f2} \frac{dI_{f1}}{dt} + I_{f1} R_{f2} + M_{fA} \frac{dI_A}{dt}$$

$$(100) \quad E_2 = L_{f2} \frac{dI_{f2}}{dt} + I_{f2} R_{f2} + M_{fA} \frac{dI_A}{dt}$$

since $L_{f1} = L_{f2}$ and $R_{f1} = R_{f2}$. M_{fA} is an ordinary mutual inductance coefficient, defining the voltage induced in the field by changes in armature current. Very strictly speaking, M_{fA} is an approximation because it too is a function of saturation, hence of current. However, it will be considered constant for this solution.

From equations (95) through (100) a solution in terms of T_{DC1} , T_{DC2} , and E_2 can be obtained. Taking increments

$$(101) \quad \Delta T_{DC1} = K_1 (\Phi_{10} \Delta I_A + I_{A0} \Delta \Phi_1)$$

and similarly

$$(102) \quad \Delta T_{DC2} = K_1 (\Phi_{20} \Delta I_A + I_{A0} \Delta \Phi_2)$$

$$(103) \quad \Delta \Phi_1 = K_f \Delta I_{f1} - A_1 \Delta I_A$$

Also

$$(104) \quad \Delta \Phi_2 = K_f \Delta I_{f2} - A_2 \Delta I_A$$

$$E_{20} + \Delta E_2 = L_{f2} \frac{d}{dt} (I_{f20} + \Delta I_{f2}) + R_{f2} (I_{f20} + \Delta I_{f2}) \\ + M_{fA} \frac{d}{dt} (I_{A0} + \Delta I_A)$$

$$(105) \quad \Delta E_2 = L_{f2} \frac{d \Delta I_{f1}}{dt} + R_{f2} \Delta I_{f2} + M_{fA} \frac{d \Delta I_A}{dt}$$

and since \mathbb{E}_1 is held constant

$$(106) \quad 0 = L_{f2} \frac{d \Delta I_{f1}}{dt} + R_{f2} \Delta I_{f1} + M_{fA} \frac{d \Delta I_A}{dt}$$

Taking the Laplace transforms of equations (101) through (106) there is

$$(107) \quad \Delta T_{DC1} = K_1 (\Phi_{10} \Delta I_A + I_{A0} \Delta \Phi_1)$$

$$(108) \quad \Delta T_{DC2} = K_1 (\Phi_{20} \Delta I_A + I_{A0} \Delta \Phi_2)$$

$$(109) \quad \Delta \Phi_1 = K_f \Delta I_{f1} - A_1 \Delta I_A$$

$$(110) \quad \Delta \Phi_2 = K_f \Delta I_{f2} - A_2 \Delta I_A$$

$$(111) \quad \Delta E_2 = (L_{f2} S + R_{f2}) \Delta I_{f2} + M_{fA} S \Delta I_A = Z_{f2} \Delta I_{f2} + M_{fA} S \Delta I_A$$

$$(112) \quad 0 = (L_{f2} S + R_{f2}) \Delta I_{f1} + M_{fA} S \Delta I_A = Z_{f2} \Delta I_{f1} + M_{fA} S \Delta I_A$$

where

$$Z_{f2} = L_{f2} S + R_{f2}$$

Eliminating ΔI_{f1} from equation (109) with equation (112), then substituting $\Delta \Phi_1$ and $\Delta \Phi_2$ from equations (109) and (110) into equations

(107) and (108) and regrouping:

$$(113) \quad \Delta T_{DC1} = K_1 \left[(\Phi_{10} - A_1 I_{A0}) - \frac{K_f M_{fA} I_{A0}}{Z_{f2}} \right] \Delta I_A$$

$$(114) \quad \Delta T_{DC2} = K_1 \left[(\Phi_{20} - A_2 I_{A0}) \Delta I_A + K_f I_{A0} \Delta I_{f2} \right]$$

Solving equation (113) for ΔI_A , and substituting into equations (111) and (114), there is obtained

$$(115) \quad \Delta E_2 = Z_{f2} \Delta I_{f2} + \frac{M_{fA} S}{K_1 \left[(\Phi_{10} - A_1 I_{A0} - \frac{K_f M_{fA} I_{A0} S}{Z_{f2}}) \right]} \Delta T_{DC1}$$

$$(115-a) \quad \Delta T_{DC2} = \left\{ \frac{(\Phi_{20} - A_2 I_{A0})}{(\Phi_{10} - A_1 I_{A0} - \frac{K_f M_{fA} I_{A0} S}{Z_{f2}})} \right\} \Delta T_{DC1} + K_1 K_f I_{A0} \Delta I_{f2}$$

Then, eliminating ΔI_{f2} from equations (115) and (115-a) and rearranging

$$(115-b) \quad \Delta T_{DC1} = D_1(s) \Delta T_{DC2} - D_2(s) \Delta E_2$$

where

$$D_1(s) = \frac{Z_{f2} (\Phi_{10} - A_1 I_{A0}) - K_f M_{fA} I_{A0} S}{Z_{f2} (\Phi_{20} - A_2 I_{A0}) - K_f M_{fA} I_{A0} S}$$

$$D_2(s) = \frac{K_1 K_f I_{A0}}{Z_{f2}} D_1(s)$$

Now, from Fig. 3

$$(115-c) \quad \Delta T_{DC2} = J_3 S \Delta \omega_3 + B_3 \Delta \omega_3 + \Delta T_3 = Z_3 \Delta \omega_3 + \Delta T_3$$

where

$$Z_3 = J_3 S + B_3 = \text{mechanical impedance of machines DC}_2 \text{ and}$$

S₂.

Substituting equation (115-c) into (115-b) the final transfer function for the DC machines is obtained.

$$(116) \quad \Delta T_{DC1} = D_1(s) [Z_3 \Delta \omega_3 + \Delta T_3] - D_2(s) \Delta E_2$$

As expected, this solution states that the input torque of the generator is a function of the excitation, speed, and load torque of the motor.

The load torque T_3 is of course the input to the synchronous machine S₂, feeding power back to the system. A relation for this torque will be ob-

tained in the following section.

Synchronous Machine S₂

The solution for synchronous machine S₂ will differ from that developed for S₁ only in that S₂ is tied to an infinite bus. Hence and $E_{\beta 3}$ will be constants, as will be the frequency, which now is synchronous frequency. Refer then to equations (69). Using the subscript (3) to distinguish terms in the S₂ equations which differ from corresponding terms in the S₁ equations, and remembering that $\Delta e_{\alpha 3}$ and $\Delta e_{\beta 3}$ equal zero and that ω_{s0} is replaced by ω_f , there can be written

$$0 = Z_a \Delta I_{\alpha 3} + S(M_{af} \Delta I_{f3} + M_{ad} \Delta I_{d3}) \\ - \omega_f (L_a \Delta I_{\beta 3} + M_{ag} \Delta I_{g3}) - \psi_{\alpha\beta 30} \Delta \omega_3$$

$$0 = Z_a \Delta i_{\beta 3} + S M_{ag} \Delta I_{g3} \\ + \omega_f (L_a \Delta i_{\alpha 3} + M_{af} \Delta I_{f3} + M_{ad} \Delta I_{d3}) + \psi_{\alpha\alpha 30} \Delta \omega_3$$

$$(117) \quad 0 = Z_f \Delta I_{f3} + S(M_{fd} \Delta I_{d3} + M_{af} \Delta i_{\alpha 3})$$

$$0 = Z_d \Delta I_{d3} + S(M_{fd} \Delta I_{f3} + M_{ad} \Delta i_{\alpha 3})$$

$$0 = Z_g \Delta I_{g3} + S M_{ag} \Delta i_{\beta 3}$$

$$\Delta T_3 = i_{\beta 30} (M_{af} \Delta I_{f3} + M_{ad} \Delta I_{d3}) + (M_{af} I_{f30} + M_{ad} I_{d30}) \Delta i_{\beta 3}$$

$$- M_{ag} (L_{\alpha 30} \Delta I_{g3} + I_{g30} \Delta i_{\alpha 3})$$

After eliminating ΔI_{f3} , ΔI_{d3} , and ΔI_{g3} and regrouping, there may be written, analogous to equations (74), (75) and (76)

$$0 = \left\{ Z_a + \frac{s^2 \left[M_{af} (SM_{fd} M_{ad} - M_{af} Z_d) + M_{ad} (SM_{fd} M_{af} - M_{ad} Z_f) \right]}{\delta_2} \right\} \Delta i_{\alpha 3}$$

$$- \left(X_{af} - \frac{SM_{af} M_{ag}}{Z_g} \right) \Delta i_{\beta 3} - \Psi_{a\beta 30} \Delta \omega_3$$

$$0 = \left\{ X_{af} + \frac{s \left[X_{M_{af}} (SM_{fd} M_{ad} - M_{af} Z_d) + X_{M_{fd}} (SM_{fd} M_{af} - M_{ad} Z_f) \right]}{\delta_2} \right\} \Delta i_{\alpha 3}$$

$$+ \left(Z_a - \frac{s^2 M_{ag}^2}{Z_g} \right) \Delta i_{\beta 3} + \Psi_{a\alpha 30} \Delta \omega_3$$

$$\Delta T_3 = \left\{ \frac{i_{\beta 30} s \left[M_{af} (SM_{fd} M_{ad} - M_{af} Z_d) + M_{ad} (SM_{fd} M_{af} - M_{ad} Z_f) \right]}{\delta_2} - M_{ag} I_{g30} \right\} \Delta i_{\alpha 3}$$

$$+ \left(M_{af} I_{f30} + M_{ad} I_{d30} + \frac{SM_{ag}^2 i_{\alpha 30}}{Z_g} \right) \Delta i_{\beta 3}$$

where

$$X_{af} = 60\text{-cycle stator self reactance} = \omega_s L_a$$

$$X_{M_{af}} = \omega_s M_{af} = 60\text{-cycle stator-main field mutual reactance}$$

$X_{M_{\alpha d f}} = \omega_f M_{\alpha d} = 60\text{-cycle stator-direct axis mutual reactance}$

$X_{M_{\beta q f}} = \omega_f M_{\beta q} = 60\text{-cycle stator-quadrature axis mutual reactance}$
or, in abbreviated form

$$(121) \quad 0 = h_{13}(s) \Delta i_{\alpha 3} - h_{23}(s) \Delta i_{\beta 3} - \psi_{\alpha \beta 30} \Delta \omega_3$$

$$(122) \quad 0 = h_{33}(s) \Delta i_{\alpha 3} + h_{43}(s) \Delta i_{\beta 3} + \psi_{\alpha \alpha 30} \Delta \omega_3$$

$$(123) \quad \Delta T_3 = h_{53}(s) \Delta i_{\alpha 3} + h_{63}(s) \Delta i_{\beta 3}$$

where $h_{13}(s)$, etc., are defined by the appropriate brackets above.

Equations (121) and (122) may now be solved simultaneously for $\Delta i_{\alpha 3}$ and $\Delta i_{\beta 3}$, and the results substituted in equation (123). By determinants

$$\Delta i_{\alpha 3} = \left[\frac{\psi_{\alpha \beta 30} h_{43}(s) - \psi_{\alpha \alpha 30} h_{23}(s)}{\delta_4} \right] \Delta \omega_3$$

$$\Delta i_{\beta 3} = - \left[\frac{\psi_{\alpha \beta 30} h_{33}(s) + \psi_{\alpha \alpha 30} h_{13}(s)}{\delta_4} \right] \Delta \omega_3$$

where

$$\delta_4 = h_{13}(s) h_{43}(s) + h_{33}(s) h_{23}(s)$$

from which

$$(124) \quad \Delta T_3 = \left\{ \frac{\psi_{\alpha \beta 30} [h_{53}(s) h_{43}(s) - h_{63}(s) h_{33}(s)] - \psi_{\alpha \alpha 30} [h_{53}(s) h_{23}(s) + h_{63}(s) h_{13}(s)]}{\delta_4} \right\} \Delta \omega_3$$

or, more simply

$$(125) \quad \Delta T_3 = H(s) \Delta \omega_3$$

Equation (124) expresses the behavior of the synchronous machine tied to an infinite bus, with constant field excitation.

3. Overall Open-Loop Transfer Function of the Modified Kramer System

Having already derived the transfer functions for the three principle machine groups, i.e., the double-fed machine M; direct current machines DC₁ and DC₂; and synchronous machine S₂; it remains necessary only to match them through their common terms. Those three solutions, equation (94), for the double-fed machine; equation (116), for the direct current machines; and equation (125), for synchronous machine S₂; are repeated below for convenient reference.

$$(94) \quad \Delta T_{DC_1} = \frac{G_4(s)}{G_2(s)} \Delta T_L + \left[G_3(s) - \frac{G_1(s) G_4(s)}{G_2(s)} \Delta \omega_1 \right]$$

$$(116) \quad \Delta T_{DC_1} = D_1(s) [Z_3 \Delta \omega_3 + \Delta T_3] - D_2(s) \Delta E_2$$

$$(125) \quad \Delta T_3 = H(s) \Delta \omega_3$$

As would be expected, the common terms, ΔT_{DC_1} and ΔT_3 , arise at the actual physical boundaries of the group, the shafts which transmit the torque from one group to another. It is desired to obtain finally an expression for changes in induction machine speed. Accordingly, equating (94) and (116), eliminating ΔT_3 with (125), and rearranging, the final overall solution is obtained:

$$(126) \quad \Delta \omega_1 = \left[\frac{G_4(s)}{G_1(s) G_4(s) - G_2(s) G_3(s)} \right] \Delta T_L - \left\{ \frac{D_1(s) G_2(s) [Z_3 + H(s)]}{G_1(s) G_2(s) - G_2(s) G_3(s)} \right\} \Delta \omega_3 \\ + \left[\frac{D_2(s) G_2(s)}{G_1(s) G_4(s) - G_2(s) G_3(s)} \right]$$

or, in abbreviated form

$$(127) \quad \Delta \omega_1 = F_1(s) \Delta T_L - F_2(s) \Delta \omega_3 + F_3(s) \Delta E_2$$

where $F_1 (S)$, etc., are defined by the proper brackets in equation (126). The speed of the induction motor is seen to be a function of its load, of the speed and excitation of the controlling DC machine, and the various machine parameters involved in the S functions. All the S functions defined are tabulated in the Appendix, in the order of their appearance in the paper.

SUMMARY

The object of the study made in this thesis was to obtain a solution, accurate within the limits of the idealizing assumptions necessary, for the open-loop transfer function of the modified Kramer speed control system. The intention was to establish relations which would be usable as the basis for further work leading to synthesis of a speed controller to close the loop from the speed of the induction machine to the excitation of the controlling direct current machine. The steps taken to obtain such a solution, and some of the difficulties encountered, are summarized briefly in the following discussion.

The work proceeded literally from left to right in the block diagram of Fig. 2. A solution was first obtained for induction machine M, then for synchronous machine S_1 . These were then combined to give the solution for the double-fed machine. To accomplish this, two major simplifications had to be made. First, the unwieldy trigonometric relations contained in the three phase equations, had to be eliminated, and, second, the resulting equations had to be Laplace transformable. Considerable trigonometric simplification was gained by the first step, replacing the three phase machines by equivalent two phase machines. The next, and most important, step was transformation to synchronously rotating axes. This entirely eliminated trigonometric quantities from the solution, and as was shown in the development, reduced the steady state currents, voltages, and flux linkages to steady direct current values. It should be mentioned in passing that the use of a synchronously rotating reference frame was not new in the solution for the synchronous machine, having been applied by Park (15) in the original article on the operation-

(15) Park, R. H., op. cit., pp. 716-730.

al solution of synchronous machinery. However, insofar as the author was able to ascertain in a survey of the literature, this reference frame has not been used before in the solution of the induction machine.

The third step was the treatment of the problem on an incremental basis. The incremental equations were found to be made up of products of steady state and incremental terms only. Because the steady state values in the rotating reference frame were constants, time function products were no longer present. Hence the equations were readily transformed, and solutions for the induction and the synchronous machines were obtained simply by algebraic manipulation. These are equations (43), (44), and (48) for the induction machine and equations (77), (78) and (81) for the synchronous machine. These six equations were combined through the voltages at the junction of the machines, and a solution for the double-fed machine obtained in terms of load torque, synchronous machine torque, and induction machine speed, as given in equation (94).

Solution of the direct current machines presented no particular difficulty. It was found convenient to lump the two together and solve them as one electric circuit, as in Fig. 12. The solution was obtained in terms of the torque input to DC_1 from synchronous machine S_1 , the torque output of DC_2 to synchronous machine S_2 , and the excitation and speed of DC_2 , as given in equation (116).

The solution for the second synchronous machine was a repetition of that for the first except that S_2 operated in the steady state at synchronous speed ω_f and with constant stator voltage, whereas, S_1 operated in the steady state at induction machine slip speed ω_{σ_0} and with variable

stator voltage. Accordingly the solution, equation (125), was considerably simpler in form involving only the torque and speed of S_2 .

Finally the group solutions were matched through their common torque terms to obtain the open-loop transfer function of the system. This is equation (127). This transfer function was not obtained as an explicit equation in descending powers of S because of the complexity of such a solution.

CONCLUSIONS

As has been stated, the relations developed in this paper constitute an accurate solution for both the modified Kramer system and the individual machines contained in it. It would now theoretically be possible through the use of Nyquist diagrams and other implements of servomechanism theory to determine the system performance for variations in E_2 or T_L , and from this determination proceed toward synthesis of a controller which would maintain desirable performance or correct improper performance. However, this would involve plotting each of the S functions contained in the Appendix; then combining these functions, by properly adding, subtracting, multiplying and dividing graphically, to obtain the final plot for the system. This would seem to involve a prohibitive amount of time and tedious labor.

A more logical approach to extension of the study would be an attempt to simplify the relations and reduce the order of the equations by various approximations. This would necessitate a detailed study of all the parameters of the several machines. Some of these, for instance the reactances and resistances of the induction and synchronous machines, are pretty well limited in range in any realizable or economical design at a given rating. Such parameters could be calculated or estimated numerically and substituted into the S functions. Such substitutions might lead to the possibility of neglecting some terms, cancelling others, etc. Having reduced the complexity as much as possible by the above procedure, further simplification and a more approximate solution might be achieved by deliberately neglecting some small but not completely negligible quantities. Finally the simplified transfer function could be plotted, and steps taken to obtain desirable performance of the system through syn

thesis of a controller and through adjustment of variable open-loop parameters such as the synchronous machine excitation voltage.

It should also be noted that the transfer functions of the components of the modified Kramer system could themselves be used in a completely independent study of the individual machines, or could serve as building blocks in setting up a solution for some altogether different combination of machines. The fact that synchronous and induction machines are by far the most common of all rotating electrical equipment renders their transfer functions doubly useful.

APPENDIX I**Tabulation of S Functions Used as Abbreviations in the Transfer Function Development****1. Induction Machine:**

$$g_1(s) = Z_R + \frac{X_{M60} (X_{Mf} Z_s - X_{sf} M_s) - M_s (X_{sf} X_{Mf} + M Z_s s)}{Z_s^2 + X_{sf}^2}$$

$$g_2(s) = \frac{M_s (X_{Mf} Z_s - X_{sf} M_s) + X_{M60} (X_{sf} X_{Mf} + M Z_s s)}{Z_s^2 + X_{sf}^2} - X_{R60}$$

$$g_3(s) = M \left[I_{\beta 0} + \frac{i_{\beta 0} (X_{sf} X_{Mf} + M Z_s s) - i_{\alpha 0} (X_{Mf} Z_s - X_{sf} M_s)}{Z_s^2 + X_{sf}^2} \right]$$

$$g_4(s) = M \left[I_{\alpha 0} + \frac{i_{\alpha 0} (X_{sf} X_{Mf} + M Z_s s) + i_{\beta 0} (X_{Mf} Z_s - X_{sf} M_s)}{Z_s^2 + X_{sf}^2} \right]$$

2. Synchronous Machine S₁:

$$h_1(s) = Z_a + \frac{s^2 [M_{af} (S M_{fd} M_{ad} - M_{af} Z_d) + M_{ad} (S M_{fd} M_{af} - M_{ad} Z_f)]}{Z_f Z_d - s^2 M_{fd}^2}$$

$$h_2(s) = X_{260} - \frac{S X_{M_{ag} 60} M_{ag}}{Z_g}$$

$$h_3(s) = X_{260} + \frac{s \left[X_{M_{af} 60} (S M_{fd} M_{ad} - M_{af} Z_d) + X_{M_{ad} 60} (S M_{fd} M_{af} - M_{ad} Z_f) \right]}{Z_f Z_d - s^2 M_{fd}^2}$$

$$h_4(s) = Z_a - \frac{s^2 M_{ag}^2}{Z_g}$$

$$h_5(s) = \frac{i_{\beta 0} s \left[M_{af} (s M_{fd} M_{ad} - M_{af} Z_d) + M_{ad} (s M_{fd} M_{af} - M_{ad} Z_f) \right]}{Z_f Z_d - s^2 M_{fd}^2} - M_{ag} I_{g0}$$

$$h_6(s) = M_{af} I_{f0} + M_{ad} I_{d0} + \frac{s M_{ag}^2 i_{\alpha 0}}{Z_g}$$

3. Double-Fed Machine:

$$f_1(s) = \Psi_{\beta 0} \left[g_1(s) - h_4(s) \right] + \Psi_{\alpha 0} \left[g_2(s) + h_2(s) \right]$$

$$f_2(s) = \Psi_{\alpha 0} \left[g_1(s) - h_4(s) \right] + \Psi_{\beta 0} \left[g_2(s) + h_2(s) \right]$$

$$f_3(s) = \Psi_{\beta 0} \left[g_2(s) + h_3(s) \right] - \Psi_{\alpha 0} \left[g_1(s) - h_1(s) \right]$$

$$f_4(s) = \Psi_{\alpha 0} \left[g_2(s) + h_3(s) \right] - \Psi_{\beta 0} \left[g_1(s) - h_1(s) \right]$$

$$G_1(s) = \frac{f_1(s) g_3(s) - f_3(s) g_4(s)}{\delta_3} - Z_1$$

$$G_2(s) = \frac{f_2(s) g_3(s) - f_4(s) g_4(s)}{\delta_3}$$

$$G_3(s) = \frac{f_1(s) h_5(s) + f_3(s) h_6(s)}{\delta_3}$$

$$G_4(s) = \frac{f_2(s) h_5(s) + f_4(s) h_6(s)}{\delta_3}$$

Where

$$\delta_3 = [h_1(s) - g_1(s)][g_1(s) - h_4(s)] + [g_2(s) + h_3(s)][g_2(s) + h_2(s)]$$

4. Direct Current Machines:

$$D_1(s) = \frac{Z_{f2} (\Phi_{10} - A_1 I_{A0}) - K_f M_{fA} I_{A0} s}{Z_{f2} (\Phi_{20} - A_2 I_{A0}) - K_f M_{fA} I_{A0} s}$$

$$D_2(s) = \frac{K_1 K_f I_{A0}}{Z_{f2}} D_1(s)$$

5. Synchronous Machine S2:

$$h_{13}(s) = Z_a + \frac{s_2 \left[M_{af} (s M_{fd} M_{ad} - M_{af} Z_d) + M_{ad} (s M_{fd} M_{af} - M_{ad} Z_f) \right]}{Z_f Z_d - s^2 M_{fd}^2}$$

$$h_{23}(s) = X_{af} - \frac{s M_{af} M_{ag}}{Z_g}$$

$$h_{33}(s) = X_{af} + \frac{s \left[X_{M_{Rf}} (s M_{fd} M_{ad} - M_{af} Z_d) + X_{M_{ad}} (s M_{fd} M_{af} - M_{ad} Z_f) \right]}{Z_f Z_d - s^2 M_{fd}^2}$$

$$h_{43}(s) = Z_a - \frac{s^2 M_{ag}^2}{Z_g}$$

$$h_{53}(s) = \frac{i_{\alpha 30} s \left[\begin{array}{c} M_{af} (s M_{fd} M_{ad} - M_{af} Z_d) \\ + M_{ad} (s M_{fd} M_{af} - M_{ad} Z_f) \end{array} \right]}{Z_f Z_d - s^2 M_{fd}^2} - M_{ag} I_{g30}$$

$$h_{63}(s) = M_{af} I_{f30} + M_{ad} I_{d30} + \frac{s M_{ag}^2 i_{\alpha 30}}{Z_g}$$

$$H(s) = \left\{ \frac{\begin{array}{c} \Psi_{\alpha\beta 30} [h_{53}(s) h_{43}(s) - h_{63}(s) h_{33}(s)] \\ - \Psi_{\alpha\alpha 30} [h_{53}(s) h_{23}(s) + h_{63}(s) h_{13}(s)] \end{array}}{\delta_3} \right\}$$

6. Final Solution:

$$F_1(s) = \frac{G_4(s)}{G_1(s) G_4(s) - G_2(s) G_3(s)}$$

$$F_2(s) = \frac{D_1(s) G_2(s) [z_3 + H(s)]}{G_1(s) G_4(s) - G_2(s) G_3(s)}$$

$$F_3(s) = \frac{D_2(s) G_2(s)}{G_1(s) G_4(s) - G_2(s) G_3(s)}$$

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VITA

The author was born in Clayton, New Mexico, on March 13, 1924. In 1927 his parents returned to the old family home near Branson, Missouri. He attended grade school in a rural district in Taney County, Missouri, and high school in Branson, Missouri, graduating from the latter in 1941.

He was drafted into the U. S. Army in February, 1943, and served three years. Included in this were ten months in the ASTP at the Universities of Kentucky and Tennessee; and one year of overseas duty with the Signal Corps in New Guinea, the Philippines, and Japan.

He entered the Missouri School of Mines in September, 1946, and received his B. S. in Electrical Engineering from there in June, 1950. In September, 1950 he re-entered the School of Mines as a graduate student in Electrical Engineering. At that time he accepted a position as graduate assistant in the Electrical Engineering Department, in which capacity he still serves.