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MATRIX COMPUTER ANALYSIS  
OF CURVILINEAR GRID SYSTEMS

BY

DAVID LEON FENTON

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A

DISSERTATION

submitted to the faculty of

THE UNIVERSITY OF MISSOURI AT ROLLA

in partial fulfillment of the requirements for the

Degree of

DOCTOR OF PHILOSOPHY IN CIVIL ENGINEERING

Rolla, Missouri

1967

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## ABSTRACT

A general equilibrium-stiffness method of matrix structural analysis was adapted and applied to the solution of the member end forces and moments of each of the members in a curvilinear structural grid system. A structural system of this nature is commonly used as the supporting framework for a "steel-framed dome", in addition to being a basic structural component in many aerospace applications.

An integral part of the development of the analysis was the development of a computer program to perform the many complex operations required to obtain the solution. The engineer, by supplying the appropriate structural data and load data to the computer program, is able to obtain the forces and moments at the ends of each of the members in the structural system corresponding to the six possible degrees of freedom of each one of the joints, or nodal points as they are referred to.

David Leon Fenton

## ACKNOWLEDGEMENT

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## LIST OF SYMBOLS

$P'$	Member end-loads in system coordinates.
$\delta'$	Member end-displacements in system coordinates
$K'$	Appropriate member stiffness matrix in system coordinates
$F$	Flexibility matrix
$p_x, p_y, p_z$	Member end-forces and moments in member coordinates.
$m_x, m_y, m_z$	
$H$	Equilibrium matrix.
*	Rigid body end-displacements.
$K$	Appropriate member stiffness matrix in member coordinates.
$T$	Transformation matrix.
$K''$	System stiffness matrix
$S$	One half of the cord length of the member.
$R$	Radius of curvature of the member.
$e$	Perpendicular distance from radius focus to the end of the member.
$\emptyset$	One half of the total angle subtended by the member.
$\Theta$	Member end rotation
$\theta$	General reference angle
$N$	Normal force on any cross-section of a member.
$A$	Cross-sectional area of a member.

$M_x, M_y, M_z$	General moments at any point along a member with respect to member coordinate axes.
$M_T, M_{YN}$	Moments with respect to the major axis of the cross-section.
$I_N$	Equivalent polar moment of inertia.
$I_{yN}$	Moment of inertia about major "y" axis.
$I_z$	Moment of inertia about major "z" axis.
E	Modulus of elasticity
G	Shear modulus.
$\lambda$	Direction cosine.
$\alpha$	Rotation of major axis of a member about member x-axis.
$\beta$	Angle between system x-axis and the projection of the member x-axis on the system xz-plane
$\gamma$	Vertical angle between the member x-axis and its projection on the system xa-plane.

## Chapter 1

### INTRODUCTION

The appearance of the dome structure in today's modern space age society is as accepted and as appropriate as it was hundreds of years ago. Today's modern materials and advancements made in construction methods have made the dome structure one of the most economical space-structure systems in use today.

There is an ever growing trend to enclose larger and larger areas and in so doing man is endeavoring to control his environment in order to eliminate the influence of weather on his daily activities. By enclosing large areas such as shopping centers the loss of business due to unfavorable weather can be reduced. The same is true of athletic stadiums, civic centers and auditoriums, green houses, opera houses and a multitude of other applications. The present demands, in addition to demands of the future where domes spanning a mile or more may be desired, present today's engineer with a tremendous challenge. The analysis of these structural systems will require talented and creative engineers with more than average ability, and, in addition, there will be a continuing need for the development of more efficient and sophisticated methods of analysis.

Modern matrix analysis of large structural systems together with the use of the high speed computer is becoming as common to the structural engineer's world today as the slope-deflection and moment-distribution methods have been in the past. The analysis of space frames, as well as flat grids, consisting of straight members has been well documented.<sup>1,2,3</sup> On the other hand it is felt that there is a definite need for a clean straight forward approach to the analysis of space-structure systems composed of curved members.

Much of the work that has been done with curved members deals with two dimensional structure such as arch and multi-arch frames.<sup>5</sup> In the area of members curved in space the following approaches have been taken, a matrix procedure using a trial and error approach was suggested by, Baron,<sup>7</sup> in the paper by Eisemann, Woo, and Namyet<sup>1</sup>, it is suggested that curved members or members with variable cross-sections could be approximated by the introduction of additional rigid joints along the length of the member resolving the member into a series of straight segments. This approach, however, would greatly increase the number of equations required to solve and would be highly inefficient for large systems. Another approach to the solution of "curvilinear grid frames" has been suggested<sup>4</sup>, which involves the use of trigonometric series to approximate moments and deflections.

In a paper on "Metallic Dome-Structure Systems," by Shu-t'ien Li<sup>9</sup>, it is suggested that any surface of revolution supported on a horizontal thrust ring may, if symmetrically loaded by distributed loads, be regarded as supporting itself directly by tension or compression. This paper then suggests that by adapting the established theory of thin-shell spherical domes, the analysis of the lattice dome can be made quite simply. The compressive stresses toward the pole and around lines of latitude when multiplied by the rib spacing will give the rib stress. The shell approach, however, does not include the bending and torsion taking place in the lattice structure, which is necessary in order to describe more accurately the behavior of the structure.

In a recent paper by J. Michalos<sup>10</sup>, a general approach to the analysis of "space networks" is described. The method of analysis described, employs the use of both the displacement and force methods of matrix analysis, however, the method of formulation still uses an iterative approach. The resisting forces and moments in each branch of the network are first computed by the force method, while the rotations and displacements of the ends of a branch are prevented. Correction moments and forces resulting from rotations and displacements of the nodes are then computed from the displacement method and finally the correction moments and forces are added to the resisting moments and forces previously determined.

The number of operations and steps required by this analysis would still encourage the development of a more straightforward and simplified approach.

In view of the preceding discussion, a need for a more direct method of analysis for the curvilinear space structure still exists. The fact that a given matrix method of formulation has been developed does not eliminate the possibility of the existence of a more efficient and more direct method of matrix formulation.

The majority of the structures built are not designed with unusual assortments of member shapes and cross-sectional patterns, for the obvious reasons of economy. In most cases the members are either all straight or all curved and in some cases there may be a combination of straight and curved members. In structures where curved members are used, they are usually of the same type, (i.e. all segments of circles or parabolas, etc.).

It is for these reasons that this author favors deriving the equations for the member flexibility coefficients directly from energy considerations, in terms of the geometric and elastic constants of a typical member and then substitute the appropriate geometric and elastic constants of each individual member into these equations to obtain the member flexibility matrices. The member stiffness matrices are then obtained simply by inverting the member flexibility matrices.

It was with this philosophy that the following method of analysis was developed. The method used to formulate the analysis was a general equilibrium-stiffness formulation in which the load-displacement equations for the individual members are used to generate directly the system stiffness matrix for the entire structure, from which the nodal displacements are obtained. These displacements are then substituted into the member load-displacement equations to obtain the final member-end loads.

Finally, a computer program was developed to effectively carry out the necessary computations, and an experimental model was build to correlate the results.



## Chapter 2

### GENERAL FORMULATION

#### 2.1 Method of Formulation

The equilibrium (or displacement) method has been chosen as the most suitable approach to the formulation of the curvilinear space grid. The load-displacement equations are first derived for a single segmental circular element, (or in general any other element for that matter) in terms of the most convenient element coordinate system, which in most cases is different from the coordinate system chosen to represent the entire structure. Similar equations for all the elements of the structure are then combined to form a set of load-displacement equations for the structure, the details, of which, will be described later.

The elements of a structure may be oriented in any manner relative to the structural system. It is, therefore, necessary to express the member equations in terms of the system coordinates before conditions of compatibility and equilibrium can be applied. The following equations (2-1), for example, are illustrative of the equations which contain a complete description of the load-displacement characteristics of a single element in system coordinates.

$$\begin{aligned}
 P'_1 &= K'_{11} \delta'_1 + K'_{12} \delta'_2 \\
 P'_2 &= K'_{21} \delta'_1 + K'_{22} \delta'_2
 \end{aligned}
 \tag{2-1}$$

Where  $P'$  and  $\delta'$  represent end loads and displacements respectively,  $K'$  represents the appropriate stiffness matrix and the primes indicate that these quantities are expressed in system coordinates.

One important advantage of this method is that equations (2-1) can be derived for a single element without regard to how this element will be oriented in the structure. The formulation of the problem can be separated into two separate parts, (1) developing the equations (2-1) for a single element using the elastic properties and geometry of the element, (2) applying the conditions of compatibility and equilibrium which are concerned with the topology of the structure. Each of these parts are entirely independent of one another which is a tremendous advantage in the formulation of complex structural systems. It is assumed in the formulation which follows that the theory of small deflections holds true.

## 2.2 Flexibility and Stiffness Matrices for a Single Element

It has been found in the case of a curved member, that it is much easier to derive the flexibility matrix and invert it to obtain the stiffness matrix than to derive the stiffness matrix directly as would be the case

for a straight member. The inversion can be done quite efficiently and rapidly on the computer since the maximum size of an element flexibility matrix would be 6 X 6.

It will be necessary to consider a member as having direction. As shown in Figure 2.1 the direction of a member is from its 1-end to its 2-end.

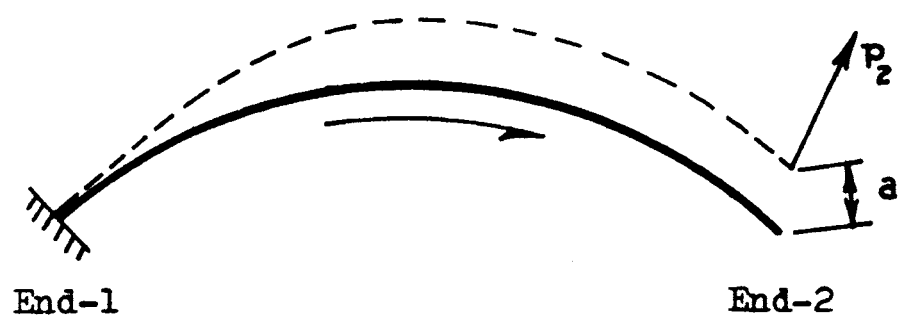


Figure 2.1 Fundamental member load-displacement.

The flexibility matrix will be derived by fixing the 1-end of the member and computing the general deformation vector "a" resulting from the general applied load vector "p<sub>2</sub>". Each of these vectors will contain six elements, the deformation vector having three translations and three rotations and the load vector having three forces and three moments. The relationship between the deformation and the load vector can be expressed in terms of the flexibility matrix, F, as follows,

$$a = Fp_2 \tag{2-2}$$

Since the 1-end of the member is completely fixed against any deformation, the deformations at the 2-end can be expressed uniquely in terms of  $p_2$  thereby establishing the validity for the existence of the inverse of  $F$ .  $p_2$  can then be written in terms of "a" and  $F^{-1}$  which equals  $K$ , the stiffness matrix, as

$$p_2 = F^{-1}a = Ka, \quad (2-3)$$

where both  $K$  and  $F$  are symmetric matrices.

If  $p_2$  can be expressed uniquely as in (2-3) it is a simple matter to obtain  $p_1$  from equilibrium considerations. Figure 2.2 shows a segmental circular member with forces and moments at each end directed in a positive sense, followed by the equilibrium equations for the member.

### End Loads

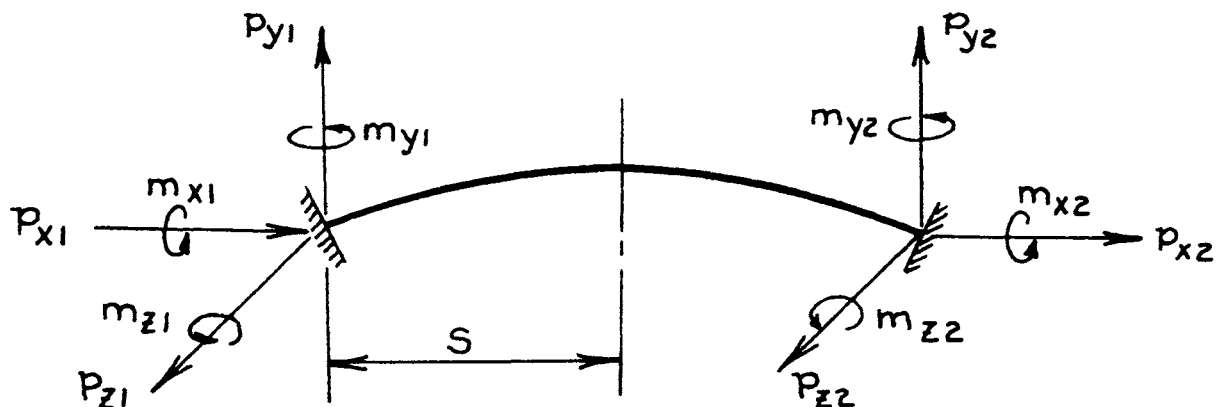


Figure 2.2 Member end load coordinates

$$\begin{aligned}
p_{x1} + p_{x2} &= 0 & m_{x1} + m_{x2} &= 0 \\
p_{y1} + p_{y2} &= 0 & m_{y1} + m_{y2} - 2Sp_{z2} &= 0 \\
p_{z1} + p_{z2} &= 0 & m_{z1} + m_{z2} + 2Sp_{y2} &= 0
\end{aligned} \tag{2-4}$$

The equilibrium equations can be written in matrix form as follows,

$$\begin{bmatrix} p_{x1} \\ p_{y1} \\ p_{z1} \\ m_{x1} \\ m_{y1} \\ m_{z1} \end{bmatrix} + \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & -2S & 0 & 1 & 0 \\ 0 & 2S & 0 & 0 & 0 & 1 \end{bmatrix} \times \begin{bmatrix} p_{x2} \\ p_{y2} \\ p_{z2} \\ m_{x2} \\ m_{y2} \\ m_{z2} \end{bmatrix} = 0 \tag{2-5}$$

or,

$$p_1 + Hp_2 = 0. \tag{2-5a}$$

In equation (2-5a) the matrix H is introduced and will be referred to as the "equilibrium matrix". H is a function which depends entirely on the geometry of the member, therefore, if the force vector at one end of the member is known, the force vector at the other end can be computed from simple matrix multiplication of H or  $H^{-1}$  by the appropriate force vector.

$$\text{i.e.} \quad p_1 = -Hp_2 \quad (2-5b)$$

$$\text{or,} \quad p_2 = -H^{-1}p_1.$$

The inverse of H can be obtained directly by matrix inversion, or it can be obtained more easily from physical considerations. This is accomplished by interchanging the 1-end and the 2-end, which only reverses the signs of the coordinates. This affects the H matrix simply by reversing the signs of the off-diagonal elements.

### End Displacements

The equilibrium matrix, H, can also be used to relate the displacements at one end of the member to the displacements at the other end.

Let,  $\delta_1^*$ ,  $\delta_2^*$  equal rigid body end-displacements of a particular member. Since rigid body displacements by definition do not affect the equilibrium of the member, the total work resulting from the rigid body displacement is equal to zero. This can be expressed as follows,

$$p_1^t \delta_1^* + p_2^t \delta_2^* = 0, \quad (2-6)$$

where  $p_1^t$  and  $p_2^t$  are the transpose of the force matrix.

Eliminating  $p_1^t$  from equation (2-6) by using (2-5b) results in the following,

$$\text{if,} \quad p_1 = -Hp_2$$

$$\text{then,} \quad p_1^t = -p_2^t H^t.$$

Substituting into (2-6) then yields,

$$-p_2^t H^t \delta_1^* + p_2^t \delta_2^* = 0.$$

Finally,

$$\delta_2^* = H^t \delta_1^*. \quad (2-7)$$

If the rigid body displacements are now superimposed on the displacements in Figure 2.1 of  $\delta_1 = 0$  and  $\delta_2 = a$  the results are,

$$\delta_1 = \delta_1^* \quad (2-8)$$

$$\delta_2 = \delta_2^* + a$$

If the rigid body displacements are now eliminated from equations (2-7) and (2-8) the following results are obtained,

$$\delta_2^* = H^t \delta_1 \quad (2-9)$$

and,

$$\delta_2 = H^t \delta_1 + a$$

or,

$$a = \delta_2 - H^t \delta_1, \quad (2-9a)$$

where equation (2-9a) expresses a measure of the amount of elastic strain associated with any arbitrary end displacements  $\delta_1$  and  $\delta_2$ . Finally, if equation (2-9a) is substituted into equation (2-3) the equation for  $p_2$

becomes,

$$p_2 = -KH^t \delta_1 + K \delta_2.$$

Using equation (2-5b), ( $p_1 = -Hp_2$ ),  $p_1$  can now be defined as,

$$p_1 = +HKH^t \delta_1 - HK \delta_2.$$

This results in a set of equations in member coordinates,

$$p_1 = HKH^t \delta_1 - HK \delta_2 \tag{2-10}$$

$$, \quad p_2 = -KH^t \delta_1 + K \delta_2,$$

which are similar to equations (2-1) in system coordinates.

These equations can be written in matrix form as follows,

$$\begin{bmatrix} p_1 \\ p_2 \end{bmatrix} = \begin{bmatrix} K_{11} & -K_{12} \\ -K_{21} & K_{22} \end{bmatrix} \begin{bmatrix} \delta_1 \\ \delta_2 \end{bmatrix} \tag{2-11}$$

or,  $p = K \delta$

where,  $K_{11} = HKH^t$ ,  $K_{12} = HK$ ,  $K_{21} = KH^t$  and

$K_{22} = K$ , remembering that  $K =$  the inverse of  $F$ .

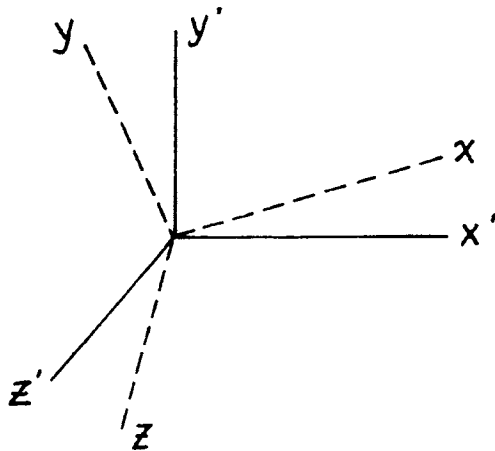
It can also be shown that if  $K$  is symmetric that  $K_{12} = K_{21}^t$ .

The symmetry of  $K$  can be verified by considering Maxwell's Law.



Because individual members of a structural framework can be connected in any orientation relative to the system coordinates, member forces and displacements must be expressed in terms of system coordinates before the conditions of compatibility and equilibrium can be applied in assembling the complete stiffness matrix of the structure.

It therefore becomes necessary to derive a transformation matrix to transform both  $p$  and  $\delta$  in member coordinates to  $p'$  and  $\delta'$  in system coordinates. This can be accomplished with a general three dimensional, rotation of axis transformation matrix of the type,



$$T = \begin{bmatrix} \lambda_{11} & \lambda_{12} & \lambda_{13} & 0 & 0 & 0 \\ \lambda_{21} & \lambda_{22} & \lambda_{23} & 0 & 0 & 0 \\ \lambda_{31} & \lambda_{32} & \lambda_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & \lambda_{11} & \lambda_{12} & \lambda_{13} \\ 0 & 0 & 0 & \lambda_{21} & \lambda_{22} & \lambda_{23} \\ 0 & 0 & 0 & \lambda_{31} & \lambda_{32} & \lambda_{33} \end{bmatrix} \quad (2-12)$$

where,  $\lambda_{ij}$  represents the cosines of the angles between the appropriate member and system axis. Since  $T$  is an orthogonal matrix it follows that  $T^t = T^{-1}$ . After  $T$  has been defined,  $p'$  and  $\delta'$  can be written as,

$$p' = Tp$$

$$\text{and, } \delta' = T\delta.$$

Using (2-13), the equations (2-11) can be re-written as,

$$p'_1 = TK_{11}T^t \delta'_1 - TK_{12}T^t \delta'_2$$

$$\text{and, } p'_2 = -TK_{21}T^t \delta'_1 + TK_{22}T^t \delta'_2$$

defining  $TK_{ij}T^t = K'_{ij}$  results in an expression of the form of Eq. (2-1)

$$p'_1 = K'_{11} \delta'_1 + K'_{12} \delta'_2$$

$$p'_2 = K'_{21} \delta'_1 + K'_{22} \delta'_2.$$

The general form of the four system stiffness matrices can now be written from equations (2-11) and (2-14) as follows,

$$\begin{aligned} K'_{11} &= THKH^tT^t \\ K'_{12} &= -THKT^t \\ K'_{21} &= -TKH^tT^t \\ K'_{22} &= TKT^t. \end{aligned}$$

It can be shown that  $K'_{12}{}^t$  and  $K'_{21}$  are equal thereby making  $K'$  symmetric. If  $K'$  is symmetric,

$$K'_{21} = K'_{12}{}^t$$

and,  $K'_{12}{}^t = -TKH^tT$ ,

( $K$  is symmetric,  $K = K^t$ ) and the symmetry is therefore verified.

### 2.3 Assembly of System Stiffness Matrix and Solution of Member Forces

The assembly of the system stiffness matrix from the member stiffness matrices makes no further reference to the elastic behavior of the material, but depends only on the way the members are connected together along with considerations of equilibrium and compatibility. Figure 2.3 illustrates a simple curvilinear grid which will be used to illustrate how the system stiffness matrix is assembled.

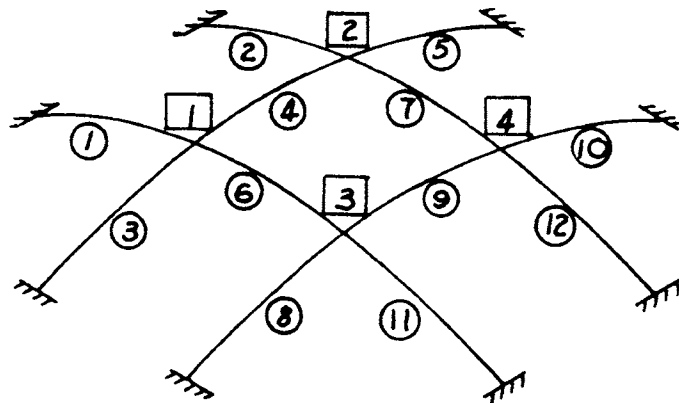


Figure 2.3 Simple curvilinear grid.

The principles involved are exactly the same for any space frame. Consider any member "I" connecting nodes J1 and J2, the member equations being,

$$\begin{aligned} p_{1I} &= (K'_{11})_I \delta'_{1I} + (K'_{12})_I \delta'_{2I} \\ p_{2I} &= (K'_{21})_I \delta'_{1I} + (K'_{22})_I \delta'_{2I} \end{aligned} \quad (2-16)$$

Considering compatibility,

$$\begin{aligned} \delta'_{1I} &= \delta'_{J1} \\ \delta'_{2I} &= \delta'_{J2} \end{aligned}$$

Equations (-16) can be re-written in terms of the displacements of nodes J1 and J2 as,

$$p_{1I} = (K'_{11})_I \delta'_{J1} + (K'_{12})_I \delta'_{J2}$$

$$\text{and, } p_{2I} = (K'_{21})_I \delta'_{J1} + (K'_{22})_I \delta'_{J2},$$

which expresses the forces and moments at each end of member "I" in terms of nodal displacements. From equilibrium considerations applied nodal loads can be equated in terms of member end loads at a given node as,

$$p_{J1} = p_{1I} + \text{the contributions of any other member at node "J1"},$$

$$\text{and, } p_{J2} = p_{2I} + \text{the contributions of any other member at node "J2"}.$$

The load-displacement equations can be written for nodes J1 and J2 as,

$$p_{J1} = (K'_{11})_I \delta'_{J1} + (K'_{12})_I \delta'_{J2} \\ + \text{contributions of other members,}$$

and

$$p_{J2} = (K'_{21})_I \delta'_{J1} + (K'_{22})_I \delta'_{J2} \\ + \text{contributions of other members.}$$

From the preceding equations the assembly scheme for system stiffness matrix can be deduced.

Consider for example the structure in Figure 2.3. Each node is capable of six degrees of freedom which produces a displacement vector  $\delta'$  having three translations and three rotations, a load vector  $p$ , having three forces and three moments, and a 6 X 6 nodal stiffness matrix,  $K'$ . The size of the system stiffness matrix necessary to describe the four node structure in Figure 2.3 equals six times the number of nodes square or 24 X 24.

As an example, the load-displacement equation will be written for node 1;

$$p_1 = [(K'_{22})_1 + (K'_{22})_3 + (K'_{11})_4 + (K'_{11})_6] \delta'_1 \\ + [(K'_{12})_4] \delta'_2 + [(K'_{12})_6] \delta'_3.$$

Similarly the equations for the other nodes are

$$p_2 = [(K'_{21})_4] \delta'_1 + [(K'_{22})_2 + (K'_{22})_4 + (K'_{22})_5 \\ + (K'_{11})_7] \delta'_2 + [(K'_{12})_7] \delta'_4,$$

$$p_3 = [(K'_{21})_6] \delta'_1 + [(K'_{22})_6 + (K'_{22})_8 + (K'_{11})_9 + (K'_{22})_{11}] \delta'_3 + [(K'_{12})_9] \delta'_4,$$

$$p_4 = [(K'_{21})_7] \delta'_2 + [(K'_{21})_9] \delta'_3 + [(K'_{22})_7 + (K'_{22})_9 + (K'_{22})_{10} + (K'_{22})_{12}] \delta'_4.$$

The preceding equations written in matrix form would appear as,

$$\begin{bmatrix} p_1 \\ p_2 \\ p_3 \\ p_4 \end{bmatrix} = \begin{bmatrix} K''_{11} & K''_{12} & K''_{13} & 0 \\ K''_{21} & K''_{22} & 0 & K''_{24} \\ K''_{31} & 0 & K''_{33} & K''_{34} \\ 0 & K''_{42} & K''_{34} & K''_{44} \end{bmatrix} \begin{bmatrix} \delta'_1 \\ \delta'_2 \\ \delta'_3 \\ \delta'_4 \end{bmatrix}$$

or as,

$$P = K'' \delta',$$

where  $K''$  is the system stiffness matrix. The contribution that a single member makes to system stiffness matrix can now be seen more easily from the preceding equations. Consider member  $\boxed{6}$ . The direction of member  $\boxed{6}$ , as well as all other members, is considered to be from the smaller node number to the larger node number. Member  $\boxed{6}$ , therefore, has its 1-end at node  $\textcircled{1}$  and its 2-end at node  $\textcircled{3}$ . The contribution of member  $\boxed{6}$  then to the system stiffness matrix will be as follows;

- (1)  $(K'_{11})_6$  will be added to  $K''_{11}$ , the direct stiffness of node ① ,
- (2)  $(K'_{22})_6$  will be added to  $K''_{33}$ , the direct stiffness of node ③ ,
- (3)  $(K'_{12})_6 = (K'_{21})_6^t$  will be added to the appropriate off diagonal stiffnesses  $K''_{13}$  and  $K''_{31}$ ,

which demonstrates the symmetry of the  $K''$  matrix. The systematic nature of the method of assembly of system stiffness matrix lends itself well to computer programming.

#### 2.4 Pinned-End Rigid Foundation Attachment

The discussion thus far has been concerned only with the fixed-end rigid foundation attachments with no reference to the pinned-end or other types of foundation attachments. The following discussion explains the modifications in formulation, necessary for the treatment of the non-fixed rigid foundation attachment. The more common case of the pinned-end rigid foundation attachment will be explained in detail.

The pinned-end case could be handled in a manner similar to what is done in slope deflection solutions, where modified end equations are developed by presolving the general slope equations of a member so as to eliminate the unknown displacements of the pinned-end. This reduces the number of unknowns which one has to

solve for and simplifies the solution. In the case of matrix methods, however, such as the equilibrium method where the high speed computer is being used to solve the problem it is desirable to preserve the general nature of the solution as much as possible so that the programming of the method of solution can be accomplished in a straightforward manner. The load-displacement equations (2-1) expressing the end-reactions of a member in terms of the end-displacements of that member are general in nature and are still equally valid even though some of the end displacements and end reactions may be zero, as is also the case in the general slope deflection equations. With this in mind the following treatment of the pinned-end foundation attachment can be seen more readily.

In the equilibrium method the "system stiffness matrix" is generated as though the pinned-end foundation attachment was like any other node in the structural system. Then the rows corresponding to zero joint displacements are altered. This is done simply by placing 1's on the leading diagonal of the system stiffness matrix and setting all other coefficients in the rows equal to zero. The 1's on the diagonal are necessary in order to prevent the equation solving routine from being upset by zeros on the diagonal. Symmetry of the system stiffness matrix can be maintained by placing zeros in the appropriate columns.



Similar adjustments in the system stiffness matrix can be made to handle other types of foundation attachments.

## 2.5 Intermediate Span Loads

The treatment of loads applied at points other than node points can be handled in a straightforward manner by the use of the principle of superposition. It is, therefore, necessary to assume linear behavior of the structure since the principle of superposition is only valid for structures which behave linearly.

The analysis would proceed in the following manner, the loaded span in question would first be considered fixed at each end and the forces and moments necessary to prevent any translation or rotation of the ends of the member would then be computed as in Figure 2.4a. Next, the sense of these forces and moments are reversed and applied to the ends of the loaded member, thus replacing the intermediate span loads with a system of "equivalent fixed-end forces and moments" as in Figure 2.4b. These fixed-end forces and moments are then combined with any other forces or moments applied directly to the nodes at the ends of the member, resulting in the final nodal load vector. It is necessary not to forget that before the nodal load vector can be incorporated in the solution of the structural system it must be transformed to system coordinates.

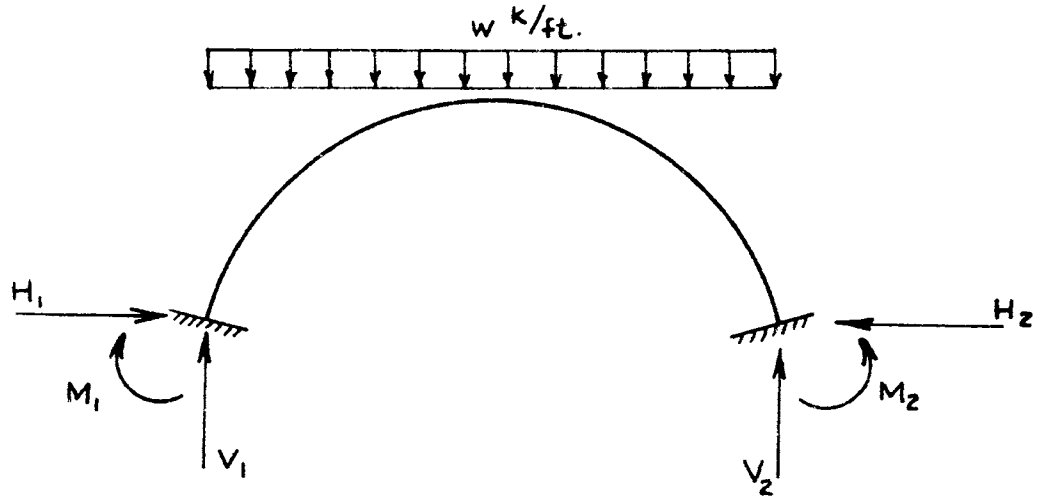
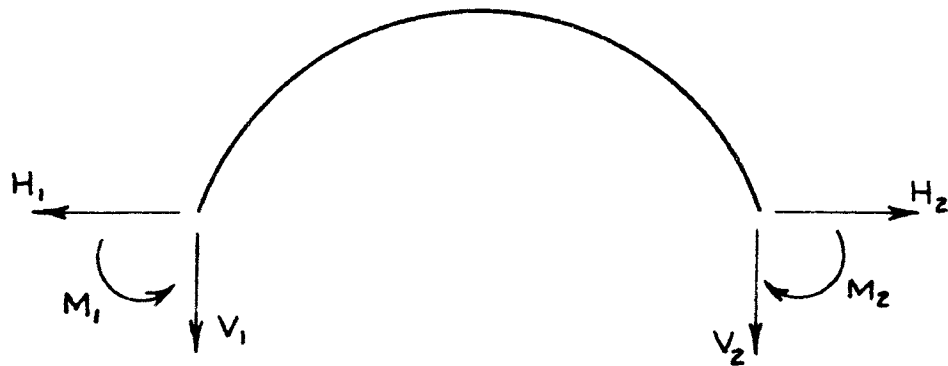


Figure 2.4a Fixed-end forces and moments on the loaded member.



2.4b Equivalent nodal loads

An additional assumption which can be made to simplify the solution of the fixed end forces and moments is that the intermediate span loads lie in the plane of the member, however, the nodal loads may be applied in any orientation.

## Chapter 3

## SEGMENTAL CIRCULAR MEMBER

## 3.1 Geometric Properties of a Segmental Circular Member

Although the stiffness method will be used to analyze the curvilinear system, the member flexibilities will be derived first and then the member stiffness matrix will be obtained by inverting the member flexibility matrix. The geometry is as follows,

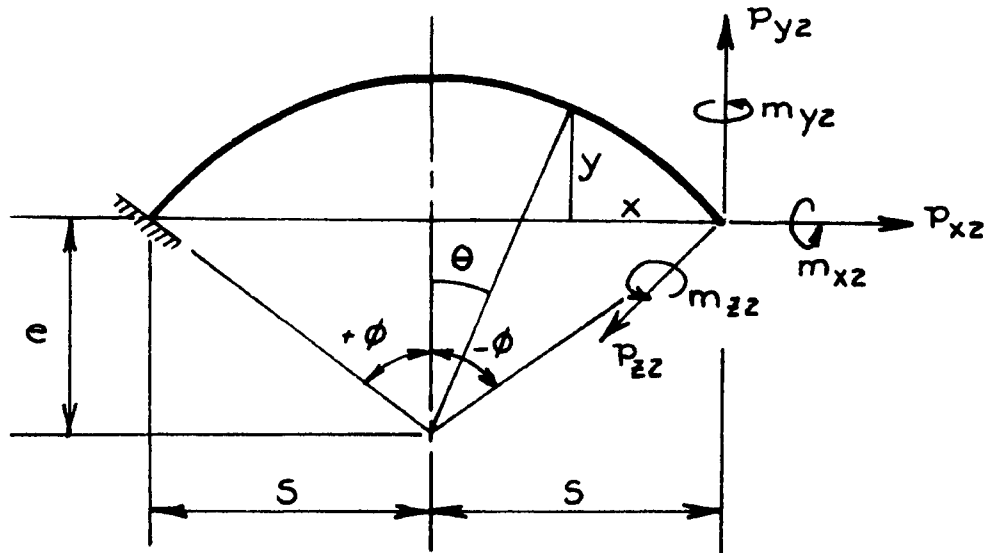


Figure 3.1 Geometry of the segmental circular member

$S$  = one-half member span

$R$  = radius of member

$e$  = vertical distance from focus to end of member

$\phi$  = one-half of the total angle subtended by the member

The variables  $x$  and  $y$  expressed in terms of geometric constants are,

$$\begin{aligned} x &= S - R \sin \Theta & S &= R \sin \emptyset \\ y &= R \cos \Theta - e & e &= R \cos \emptyset \\ ds &= R d\Theta \end{aligned}$$

### 3.2 Load-Displacement Equations Derivation

The load displacement equations will be derived using Castigliano's Second Theorem,

$$\frac{\partial U}{\partial p_i} = u_i \quad (3-1)$$

which says that the first derivative of the energy expression,  $U$ , with respect to  $p_i$  equals the corresponding displacement  $u_i$ . Energy due to bending, torsion and axial deformation will be considered in the following derivation, neglecting the effects of shear since members will generally be light and flexible.

The general expressions for moments at any point along the member about an  $x$ ,  $y$  and  $z$  axis are,

$$\text{about the "x" axis,} \quad M_x = -p_z y + m_x$$

$$\text{about the "y" axis,} \quad M_y = -p_z x + m_y$$

$$\text{about the "z" axis,} \quad M_z = p_x y + p_y x + m_z$$

In order to be able to write the general energy expression for the member, the forces and moments must be expressed with respect to the major axes of the cross-section at any point along the member. (Figure 3.2)

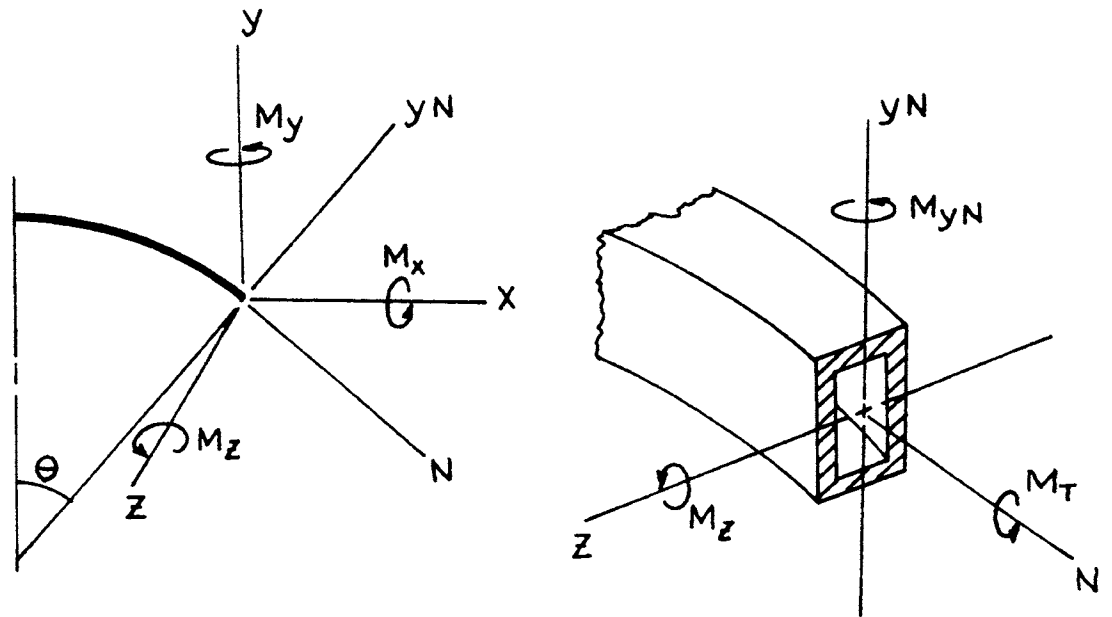


Figure 3.2 Forces and moments on major axes.

The moments and forces with respect to the axis in Figure 3.2, are,

a) Axial Force,

$$N = p_x \cos \theta - p_y \sin \theta$$

b) Torsion,

$$M_T = M_x \cos \theta - M_y \sin \theta$$

c) Moment about "z" axis,

$$M_z = p_x y + p_y x + m_z$$

d) Moment about "y<sub>N</sub>" axis,

$$M_{yN} = M_x \sin\theta + M_y \cos\theta .$$

The previous equations can now be written in terms of the end forces and moments, and the geometric constants as;

a)  $N = p_x \cos\theta - p_y \sin\theta$

b)  $M_T = p_z (S \sin\theta + e \cos\theta - R) + m_x \cos\theta - m_y \sin\theta$

c)  $M_z = p_x (R \cos\theta - e) + p_y (S - R \sin\theta) + m_z$

d)  $M_{yN} = p_z (e \sin\theta - S \cos\theta) + m_x \sin\theta + m_y \cos\theta .$

### 3.3 General Energy Expression

The general energy expression can now be written as,

$$U = \int \frac{N^2 ds}{2AE} + \int \frac{M_T^2 ds}{2GI_N} + \int \frac{M_z^2 ds}{2EI_z} + \int \frac{M_{yN}^2 ds}{2EI_{yN}}$$

The deflections corresponding to the six degrees of freedom are,

$$\delta_x = \frac{\partial U}{\partial p_x}, \quad \delta_y = \frac{\partial U}{\partial p_y}, \quad \delta_z = \frac{\partial U}{\partial p_z},$$

$$\theta_x = \frac{\partial U}{\partial m_x}, \quad \theta_y = \frac{\partial U}{\partial m_y}, \quad \theta_z = \frac{\partial U}{\partial m_z}.$$

The following Table 3.1 which tabulates the partial derivatives of  $N$ ,  $M_T$ ,  $M_z$  and  $M_{yN}$  with respect to the various coordinate forces and moments can be very helpful in setting up the various integrals.

TABLE 3.1

	$\partial N$	$\partial M_T$	$\partial M_z$	$\partial M_{yN}$
$\partial p_x$	$\cos\theta$	0	$(R \cos\theta - e)$	0
$\partial p_y$	$-\sin\theta$	0	$(S - R \sin\theta)$	0
$\partial p_z$	0	$(S \sin\theta + e \cos\theta - R)$	0	$(e \sin\theta - S \cos\theta)$
$\partial m_x$	0	$\cos\theta$	0	$\sin\theta$
$\partial m_y$	0	$-\sin\theta$	0	$\cos\theta$
$\partial m_z$	0	0	1	0

### 3.4 Deflection Calculations

The deflections will now be calculated assuming that members have constant cross sectional areas and that  $I_N$  is the equivalent polar moment of inertia for the section.

1) The deflection in the x-direction is,

$$\delta_x = \int \frac{N}{AE} \frac{\partial N}{\partial P_x} ds + \int M_z \frac{\partial M_z}{\partial P_x} \frac{ds}{EI_z},$$

$$\begin{aligned} \delta_x &= \frac{R}{AE} \int (p_x \cos^2 \theta - p_y \sin \theta \cos \theta) d\theta \\ &+ \frac{R}{EI_z} \int [p_x (R \cos \theta - e) + p_y (S - R \sin \theta) \\ &+ m_z] [R \cos \theta - e] d\theta \end{aligned}$$

after integrating and simplifying, the results are

$$\begin{aligned} \delta_x &= \frac{p_x R}{AE} \left( \theta + \frac{1}{2} \sin 2\theta \right) + \frac{p_x R^3}{EI_z} \left( \theta - \frac{3}{2} \sin 2\theta \right. \\ &+ \left. 2\theta \cos^2 \theta \right) + \frac{p_y R^3}{EI_z} (2 \sin^2 \theta - \theta \sin 2\theta) \\ &+ \frac{m_z R^2}{EI_z} (2 \sin \theta - 2\theta \cos \theta). \end{aligned}$$

2) The deflection in the y-direction is,

$$\delta_y = \int \frac{N}{AE} \frac{\partial N}{\partial P_y} ds + \int M_z \frac{\partial M_z}{\partial P_y} \frac{ds}{EI_z}$$



$$\begin{aligned} \delta_y &= \frac{R}{AE} \int (p_x \cos\theta - p_y \sin\theta) (-\sin\theta) d\theta \\ &+ \frac{R}{EI_z} \int \left[ p_x (R \cos\theta - e) + p_y (S - R \sin\theta) + m_z \right] \\ &\quad \left[ S - R \sin\theta \right] d\theta, \end{aligned}$$

after integrating and simplifying, the results are,

$$\begin{aligned} \delta_y &= \frac{p_y R}{AE} \left( \theta - \frac{1}{2} \sin 2\theta \right) + \frac{p_x R^3}{EI_z} \left( 2 \sin^2 \theta - \theta \sin 2\theta \right) \\ &+ \frac{p_y R^3}{EI_z} \left( 2\theta \sin^2 \theta + \theta - \frac{1}{2} \sin 2\theta \right) + \frac{m_z R^2}{EI_z} (2\theta \sin \theta). \end{aligned}$$

3) The deflection in the z-direction is,

$$\begin{aligned} \delta_z &= \int \frac{M_T \partial M_T}{\partial p_z} \frac{ds}{GI_N} + \int M_{yN} \frac{\partial M_{yN}}{\partial p_z} \frac{ds}{EI_{yN}}, \\ \delta_z &= \frac{R}{GI_N} \int \left[ p_z (S \sin\theta + e \cos\theta - R) + m_x \cos\theta \right. \\ &\quad \left. - m_y \sin\theta \right] \left[ S \sin\theta + e \cos\theta - R \right] d\theta \\ &+ \frac{R}{EI_{yN}} \int \left[ p_z (e \sin\theta - S \cos\theta) + m_x \sin\theta \right. \\ &\quad \left. + m_y \cos\theta \right] \left[ e \sin\theta - S \cos\theta \right] d\theta \end{aligned}$$

after integrating and simplifying, the results are,

$$\begin{aligned}
 \phi_z &= \frac{p_z R^3}{GI_N} \left( 3\phi + \frac{1}{4} \sin 4\phi - 2 \sin 2\phi \right) + \frac{m_x R^2}{GI_N} (\phi \cos \phi \\
 &\quad - 2 \sin \phi + \frac{1}{2} \cos \phi \sin 2\phi) - \frac{m_y R^2}{GI_N} (\phi \sin \phi \\
 &\quad - \frac{1}{2} \sin \phi \sin 2\phi) + \frac{p_z R^3}{EI_{yN}} \left( \phi - \frac{1}{4} \sin 4\phi \right) \\
 &\quad + \frac{m_x R^2}{EI_{yN}} (\phi \cos \phi - \frac{1}{2} \cos \phi \sin 2\phi) - \frac{m_y R^2}{EI_{yN}} (\phi \sin \phi \\
 &\quad + \frac{1}{2} \sin \phi \sin 2\phi).
 \end{aligned}$$

4) The rotation about the x-axis is,

$$\begin{aligned}
 \theta_x &= \int^{M_T} \frac{\partial M_T}{\partial m_x} \frac{ds}{GI_N} + \int^{M_{yN}} \frac{\partial M_{yN}}{\partial m_x} \frac{ds}{EI_{yN}}, \\
 \theta_x &= \frac{R}{GI_N} \int \left[ p_x (s \sin \theta + e \cos \theta - R) + m_x \cos \theta \right. \\
 &\quad \left. - m_y \sin \theta \right] \left[ \cos \theta \right] d\theta + \frac{R}{EI_{yN}} \int \left[ p_z (e \sin \theta \right. \\
 &\quad \left. - S \cos \theta) + m_x \sin \theta + m_y \cos \theta \right] \left[ \sin \theta \right] d\theta,
 \end{aligned}$$

after integrating and simplifying, the results are,

$$\begin{aligned}\Theta_x &= \frac{p_z R^2}{GI_N} (\varnothing \cos\varnothing - 2 \sin\varnothing + \frac{1}{2} \cos\varnothing \sin 2\varnothing) \\ &+ \frac{m_x R}{GI_N} (\varnothing + \frac{1}{2} \sin 2\varnothing) + \frac{p_z R^2}{EI_{yN}} (\varnothing \cos\varnothing - \frac{1}{2} \cos\varnothing \sin 2\varnothing) \\ &+ \frac{m_x R}{EI_{yN}} (\varnothing - \frac{1}{2} \sin 2\varnothing).\end{aligned}$$

5) The rotation about the y-axis is,

$$\begin{aligned}\Theta_y &= \int M_T \frac{\partial M_T}{\partial m_y} \frac{ds}{GI_N} + \int M_{yN} \frac{\partial M_{yN}}{\partial m_y} \frac{ds}{EI_{yN}}, \\ \Theta_y &= \frac{R}{GI_N} \int [p_z (S \sin\theta + e \cos\theta - R + m_x \cos\theta \\ &- m_y \sin\theta) [-\sin\theta] d\theta + \frac{R}{EI_{yN}} \int [p_z (e \sin\theta \\ &- s \cos\theta) + m_x \sin\theta + m_y \cos\theta] [\cos\theta] d\theta ,\end{aligned}$$

after integrating and simplifying, the results are,

$$\begin{aligned}\Theta_y &= \frac{-p_z R^2}{GI_N} (\varnothing \sin\varnothing - \frac{1}{2} \sin\varnothing \sin 2\varnothing) + \frac{m_y R}{GI_N} (\varnothing - \frac{1}{2} \sin 2\varnothing) \\ &- \frac{p_z R^2}{EI_{yN}} (\varnothing \sin\varnothing + \frac{1}{2} \sin\varnothing \sin 2\varnothing) + \frac{m_y R}{EI_{yN}} (\varnothing + \frac{1}{2} \sin 2\varnothing).\end{aligned}$$

6) The rotation about the z-axis is,

$$\Theta_z = \int M_z \frac{\partial M_z}{\partial m_z} \frac{ds}{EI_z}$$

$$\Theta_z = \frac{R}{EI_z} \int \left[ p_x (R \cos \theta - e) + p_y (S - R \sin \theta) + M_z \right] [1] d\theta ,$$

after integrating and simplifying, the results are,

$$\Theta_z = \frac{p_x R^2}{EI_z} (2 \sin \theta - 2\theta \cos \theta) + \frac{p_y R^2}{EI_z} (2\theta \sin \theta) - \frac{m_z R}{EI_z} (2\theta).$$

### 3.5 Flexibility Matrix

Now that the deflections and rotations of end 2 of the member have been found in terms of the six degrees of freedom, the flexibilities  $f_{ij}$  can be found, according to definition, by setting the  $j^{\text{th}}$  force equal to unity and computing the  $i^{\text{th}}$  displacement ( $i = 1, 2, \dots, 6$ ) when the other forces and moments are set equal to zero. Flexibility  $f_{ij}$  can also be found by,

$$f_{ij} = \frac{\partial^2 U}{\partial p_i \partial p_j} , \quad (3-2)$$

which is equivalent to the definition previously stated.

Now for simplicity let;

$$\begin{array}{llll}
 p_x = p_1 & m_x = p_4 & \delta_x = \delta_1 & \theta_x = \theta_4 \\
 p_y = p_2 & m_y = p_5 & \delta_y = \delta_2 & \theta_y = \theta_5 \\
 p_z = p_3 & m_z = p_6 & \delta_z = \delta_3 & \theta_z = \theta_6.
 \end{array}$$

The elements of the resulting flexibility matrix will then be;

$$f_{11} = \frac{R}{AE} \left( \theta + \frac{1}{2} \sin \theta \right) + \frac{R^3}{EI_z} \left( \theta - \frac{3}{2} \sin 2\theta + 2\theta \cos^2 \theta \right)$$

$$f_{22} = \frac{R}{AE} \left( \theta - \frac{1}{2} \sin \theta \right) + \frac{R^3}{EI_z} \left( \theta - \frac{1}{2} \sin 2\theta + 2\theta \sin^2 \theta \right)$$

$$f_{33} = \frac{R^3}{GI_N} \left( 3\theta - 2 \sin 2\theta + \frac{1}{4} \sin 4\theta \right) + \frac{R^3}{EI_{yN}} \left( \theta - \frac{1}{4} \sin 4\theta \right)$$

$$f_{44} = \frac{R}{GI_N} \left( \theta + \frac{1}{2} \sin 2\theta \right) + \frac{R}{EI_{yN}} \left( \theta - \frac{1}{2} \sin 2\theta \right)$$

$$f_{55} = \frac{R}{GI_N} \left( \theta - \frac{1}{2} \sin 2\theta \right) + \frac{R}{EI_{yN}} \left( \theta + \frac{1}{2} \sin 2\theta \right)$$

$$f_{66} = \frac{R}{EI_z} (2\theta),$$

which are the diagonal elements of the flexibility matrix.

The off diagonal elements of the flexibility matrix are;

$$f_{12} = f_{21} = \frac{R^3}{EI_z} (2 \sin^2 \theta - \theta \sin 2\theta)$$

$$f_{13} = f_{31} = f_{14} = f_{41} = f_{51} = f_{15} = 0$$

$$f_{61} = f_{16} = \frac{R^2}{EI_z} (2 \sin \theta - 2\theta \cos \theta)$$

$$f_{32} = f_{23} = f_{42} = f_{24} = f_{52} = f_{25} = 0$$

$$f_{26} = f_{62} = \frac{R^2}{EI_z} (2\theta \sin \theta)$$

$$f_{36} = f_{63} = f_{45} = f_{54} = f_{46} = f_{64} = f_{56} = f_{65} = 0$$

$$f_{34} = f_{43} = \frac{R^2}{GI_N} (\theta \cos \theta - 2 \sin \theta + \frac{1}{2} \cos \theta \sin 2\theta)$$

$$+ \frac{R^2}{EI_{yN}} (\theta \cos \theta - \frac{1}{2} \cos \theta \sin 2\theta)$$

$$f_{35} = f_{53} = - \frac{R^2}{GI_N} (\theta \sin \theta - \frac{1}{2} \sin \theta \sin 2\theta)$$

$$- \frac{R^2}{EI_{yN}} (\theta \sin \theta + \frac{1}{2} \sin \theta \sin 2\theta).$$

As can be seen a considerable number of the off diagonal elements of the flexibility matrix are equal to zero, which makes finding the inverse of the six by six much easier than if the matrix was completely full.

Finally, the assembled member flexibility matrix written in matrix form would be,

$$\begin{bmatrix} f_{11} & f_{12} & 0 & 0 & 0 & f_{16} \\ f_{21} & f_{22} & 0 & 0 & 0 & f_{26} \\ 0 & 0 & f_{33} & f_{34} & f_{35} & 0 \\ 0 & 0 & f_{43} & f_{44} & 0 & 0 \\ 0 & 0 & f_{53} & 0 & f_{55} & 0 \\ f_{61} & f_{62} & 0 & 0 & 0 & f_{66} \end{bmatrix},$$

and it is evident also that the flexibility matrix is symmetrical. The K matrix in equation (2-3) can now be computed by inverting F. This K corresponds to  $K_{22}$  in Equation (2-11). If the equilibrium matrix is known,  $K_{11}$ ,  $K_{12}$ ,  $K_{21}$  can be computed. These K's, however, are still in member coordinates and before they are transformed into system coordinates, the transformation matrix T must be derived.

### 3.6 Three Dimensional Axis Rotation Transformation Matrix

In order to keep the formulation of system stiffness matrix general in nature, the transformation matrix, T, will be derived for a member whose axis is skewed to all three system axes, as indicated in Figure 3.2.

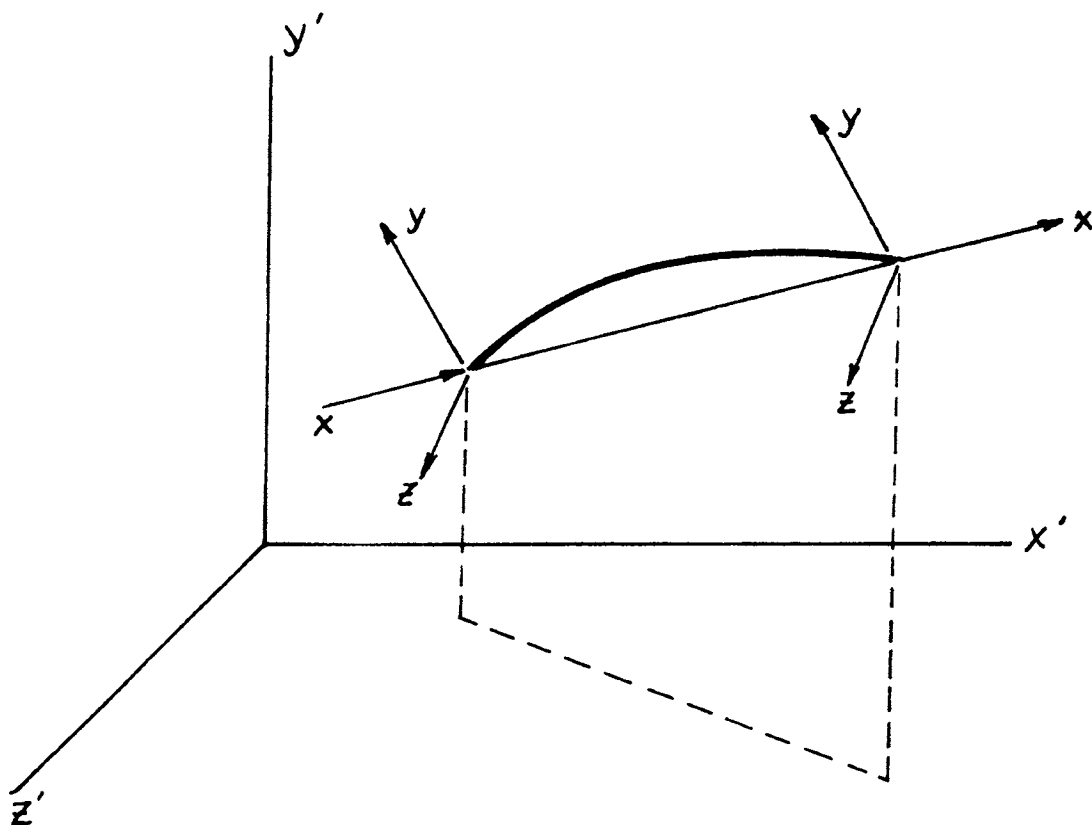


Figure 3.3 Typical member orientation

From equations (2.13) for example,

$$p' = Tp$$

and,

$$\delta' = T\delta$$

or in matrix form,

$$\begin{bmatrix} p'_x \\ p'_y \\ p'_z \end{bmatrix} = \begin{bmatrix} \lambda_{11} & \lambda_{12} & \lambda_{13} \\ \lambda_{21} & \lambda_{22} & \lambda_{23} \\ \lambda_{31} & \lambda_{32} & \lambda_{33} \end{bmatrix} \begin{bmatrix} p_x \\ p_y \\ p_z \end{bmatrix}$$



where  $\lambda_{11}$ ,  $\lambda_{12}$ , ...etc. are direction cosines. It can be shown that  $T$  is an orthogonal matrix thereby making  $T^{-1} = T^t$  so that

$$p = T^t p' \quad (3-1)$$

and,  $\delta = T^t \delta'$ .

The derivation of the transformation matrix  $T$  can now proceed in a systematic manner by considering  $T^t$  as the product of three individual rotation transformation matrices  $T_\alpha$ ,  $T_\beta$  and  $T_\gamma$  corresponding to the three rotations which must be performed required to transform from member to system coordinates as shown in Figure 3.3. In each row of  $T^t$  is located the direction cosines of the corresponding member axis.

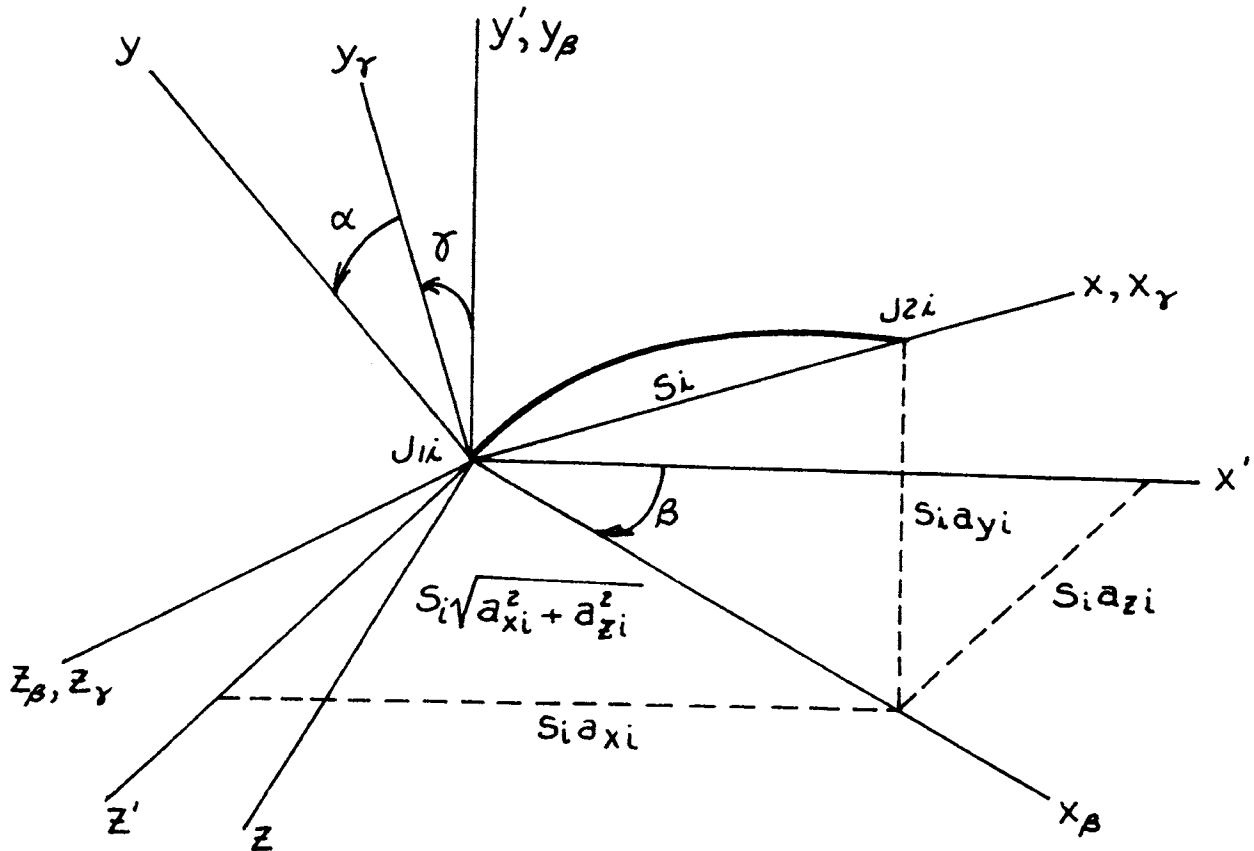


Figure 3.4 The three required rotations

Let  $a_x$ ,  $a_y$  and  $a_z$  equal the direction cosines of  $x$ .

Then,

$$a_x = \frac{x_{J2} - x_{J1}}{S} ; \quad a_y = \frac{y_{J2} - y_{J1}}{S} ; \quad a_z = \frac{z_{J2} - z_{J1}}{S}$$

where,

$$S = \sqrt{(x_{J2} - x_{J1})^2 + (y_{J2} - y_{J1})^2 + (z_{J2} - z_{J1})^2}$$

Row 1 of  $T^t$  equals  $a_x$ ,  $a_y$  and  $a_z$  which are the direction cosines of the x axis and can be found directly from the end coordinates of the member. The last two rows can be found by considering  $\beta$ ,  $\gamma$  and  $\alpha$  rotations. Consider first the  $\beta$  rotation and  $T_\beta$  where,

$$T_\beta = \begin{bmatrix} \cos\beta & 0 & \sin\beta \\ 0 & 1 & 0 \\ -\sin\beta & 0 & \cos\beta \end{bmatrix} \quad (3-2)$$

The elements  $\cos\beta$  and  $\sin\beta$  can be expressed in terms of  $a_x$ ,  $a_y$  and  $a_z$  as follows,

$$\cos\beta = \frac{a_x}{\sqrt{a_x^2 + a_z^2}} \quad \sin\beta = \frac{a_z}{\sqrt{a_x^2 + a_z^2}}$$

The system forces have now been transformed to the axis where,

$$p_\beta = T_\beta p'$$

Next the  $\beta$ -axis will be rotated to the  $\gamma$ -axis or i.e., the  $\beta$ -forces will be transformed to  $\gamma$  oriented forces.

$$T_\gamma = \begin{bmatrix} \cos\gamma & \sin\gamma & 0 \\ -\sin\gamma & \cos\gamma & 0 \\ 0 & 0 & 1 \end{bmatrix} \quad (3-3)$$

where,

$$\cos \gamma = \sqrt{a_x^2 + a_z^2}, \quad \sin \gamma = a_y,$$

so that the system forces have now been transformed to the  $\gamma$ -axis as,

$$p_\gamma = T_\gamma P_\beta$$

or,

$$p_\gamma = T_\gamma T_\beta P'$$

Finally, the  $\alpha$ -transformation, where  $\alpha$  is the rotation of the y-axis of the member with respect to  $Y_\gamma$ -axis about the member x-axis and is part of the input data.

$$T_\alpha = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \alpha & \sin \alpha \\ 0 & -\sin \alpha & \cos \alpha \end{bmatrix} \quad (3-4)$$

The member forces can now be expressed as,

$$p = T_\alpha P_\gamma$$

or, using Equations (3-2), (3-3) and (3-4),

$$p = T_\alpha T_\gamma T_\beta P'$$

where,

$$T^t = T_\alpha T_\gamma T_\beta$$

and,

$$p = T^t p'$$

or,

$$p' = T p .$$

In a similar manner the member stiffness matrix can be transformed to system stiffness matrix by,

$$K' = TKT^t.$$

Now if the appropriate terms for  $\cos\beta$ ,  $\sin\beta$ ,  $\cos\gamma$ ,  $\sin\gamma$  are substituted into the appropriate matrices and the indicated multiplications of  $T_\alpha$   $T_\gamma$   $T_\beta$  are carried out the final form of  $T^t$  will be as follows,

$$T^t = \begin{bmatrix} \lambda_{11} & \lambda_{21} & \lambda_{31} \\ \lambda_{12} & \lambda_{22} & \lambda_{32} \\ \lambda_{13} & \lambda_{23} & \lambda_{33} \end{bmatrix}$$

where,

$$\lambda_{11} = a_x$$

$$\lambda_{12} = \frac{-a_x a_y \cos\alpha - a_z \sin\alpha}{\sqrt{a_x^2 + a_z^2}}$$

$$\lambda_{13} = \frac{a_x a_y \sin\alpha - a_z \cos\alpha}{\sqrt{a_x^2 + a_z^2}}$$

$$\lambda_{21} = a_y$$

$$\lambda_{22} = \sqrt{a_x^2 + a_z^2} \cos\alpha$$

$$\lambda_{23} = -\sqrt{a_x^2 + a_z^2} \sin\alpha$$

$$\lambda_{31} = a_z$$

$$\lambda_{32} = \frac{-a_y a_z \cos\alpha + a_x \sin\alpha}{\sqrt{a_x^2 + a_z^2}}$$

$$\lambda_{33} = \frac{a_y a_z \sin\alpha + a_x \cos\alpha}{\sqrt{a_x^2 + a_z^2}} .$$

## Chapter 4

### APPLICATION OF ANALYSIS

In this chapter the information obtained from Chapters 2 and 3 will be applied to the analysis of "curvilinear space grids". The details of the computer program, which performs the necessary calculations, are discussed in the following chapter. The objective of the material in this chapter is to demonstrate the validity of the solution and to develop confidence in the analysis.

#### 4.1 Procedure

Three different mathematical models were chosen to demonstrate the analysis. These models are circular framed domes having height to span ratios,  $(h/L)$ , of one-sixth, one-fourth and one-half respectively, with each one having a span of 72.0 inches.

Each of the structures were analyzed under the influence of symmetric and unsymmetric loads. In one case, all foundation attachments were fixed, while in the other case half of the foundation attachments were fixed and half were pinned. The details of the mathematical models can be seen clearly in the diagrams which follow in (4.2).

Finally, an experimental model was build and tested in order to provide the necessary correlation between the

analytical solution of the mathematical model and experimental results from the physical model.

#### 4.2 Mathematical Models

Each of the mathematical models are composed of twenty-four segmental circular members, whose topological arrangement describes a "lattice dome". All of the members lie on the arc of a great circle. The model dome has nine interior nodal points and twelve foundation attachments.

a.) Mathematical Model 1 - The computations involved in obtaining the geometric data will be shown in detail for this model as an example of the required hand calculations, and subsequently deleted from the remaining models as they are handled in the same manner.

1.) Basic Geometry:

$$\frac{h}{L} = \frac{1}{6}, \quad L = 72.0", \quad h = 12.0"$$

$$R = \frac{h}{2} \left[ 1 + \frac{1}{4\left(\frac{h}{L}\right)^2} \right]$$

$$R = 60.0"$$



- 2.) Plan View - Figures 4.1 and 4.2 illustrate the node and member numbering scheme used in the analysis of the model with fixed foundation attachments and combined fixed and pinned foundation attachments.

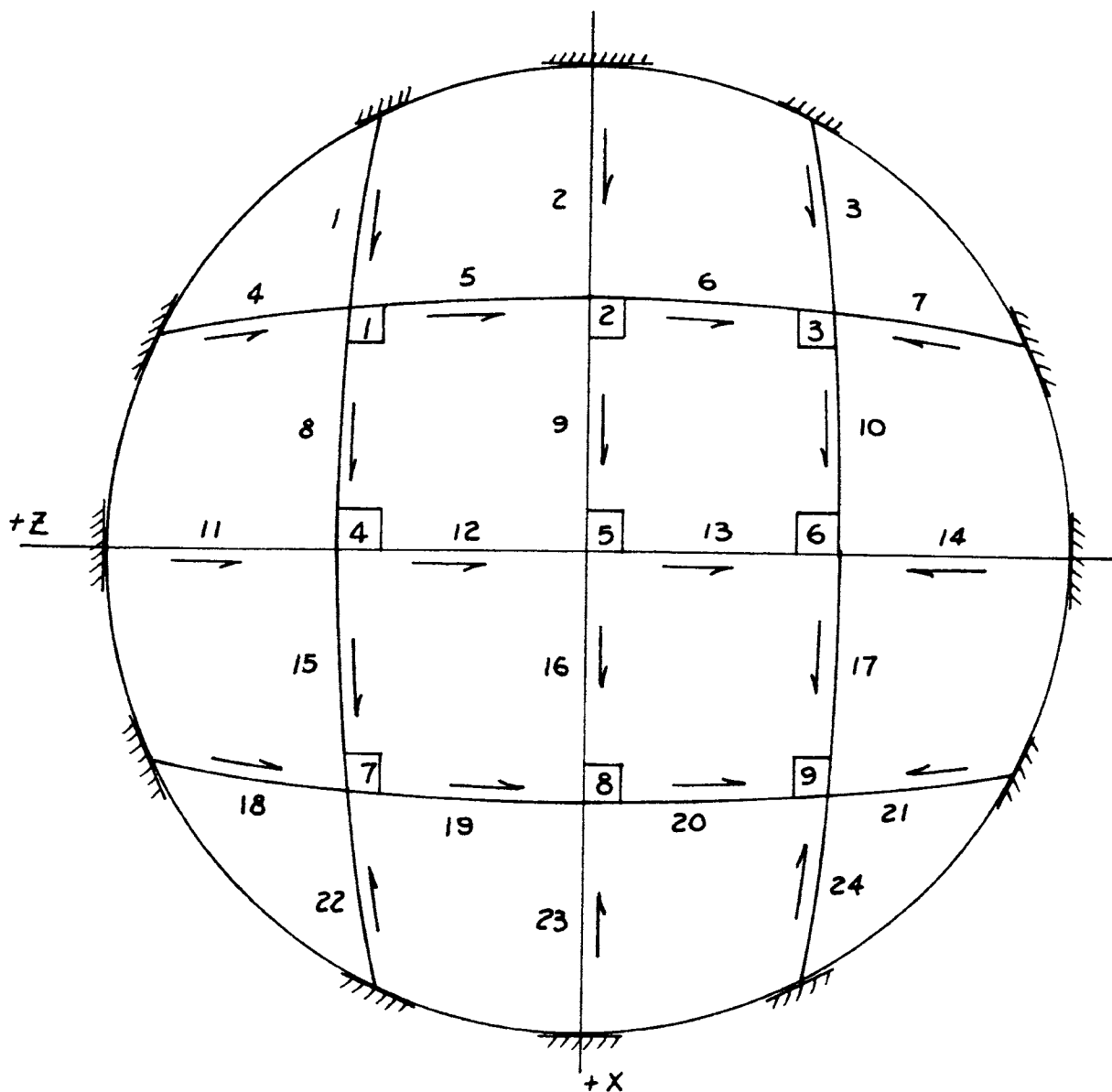


Figure 4.1 Mathematical Model-1 ( $\frac{h}{L} = \frac{1}{6}$ ), all foundation attachments fixed.

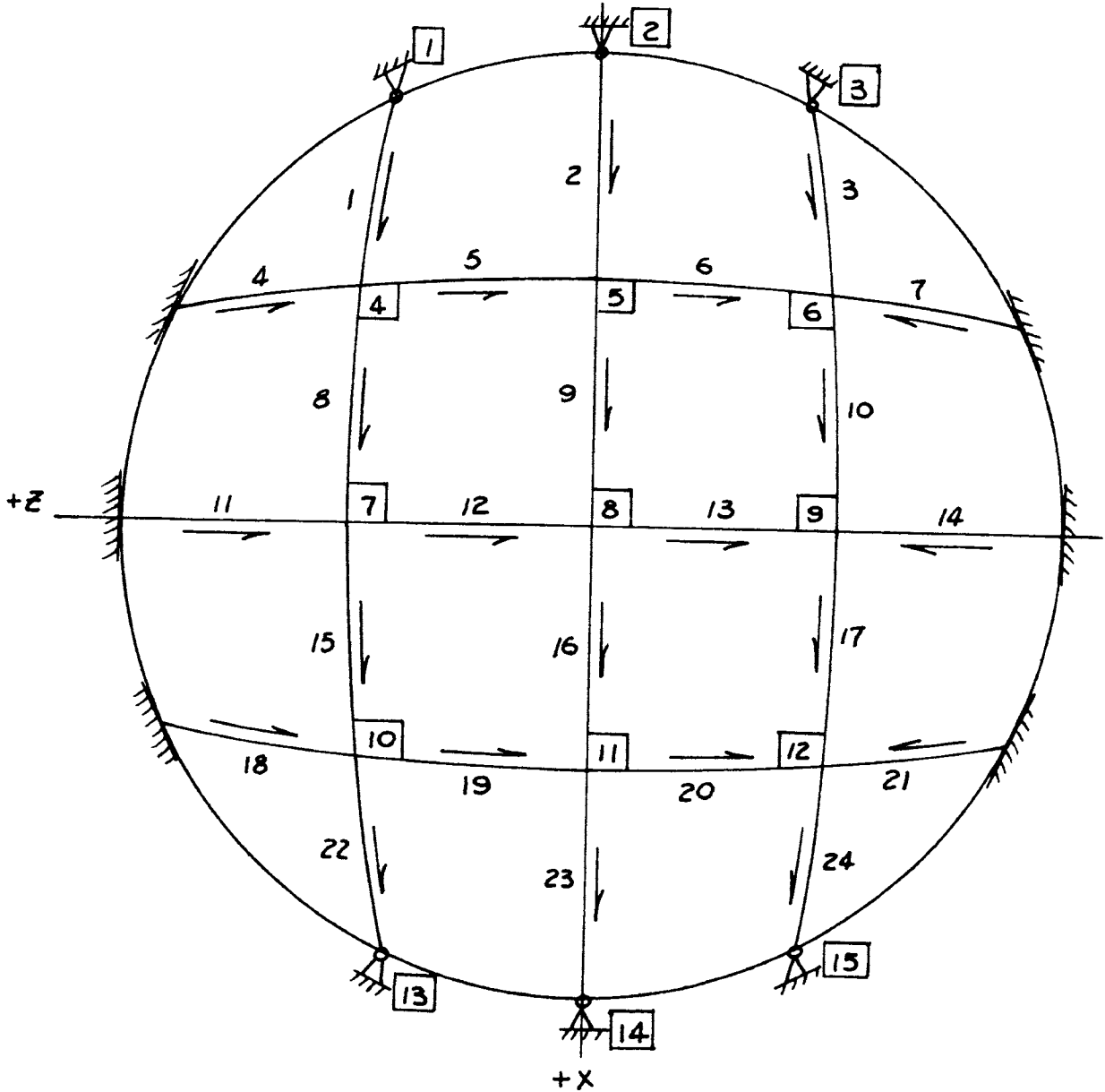


Figure 4.2 Mathematical Model-1  $\left(\frac{h}{L} = \frac{1}{6}\right)$ , with both pinned and fixed foundation attachments.

3.) Model profile:

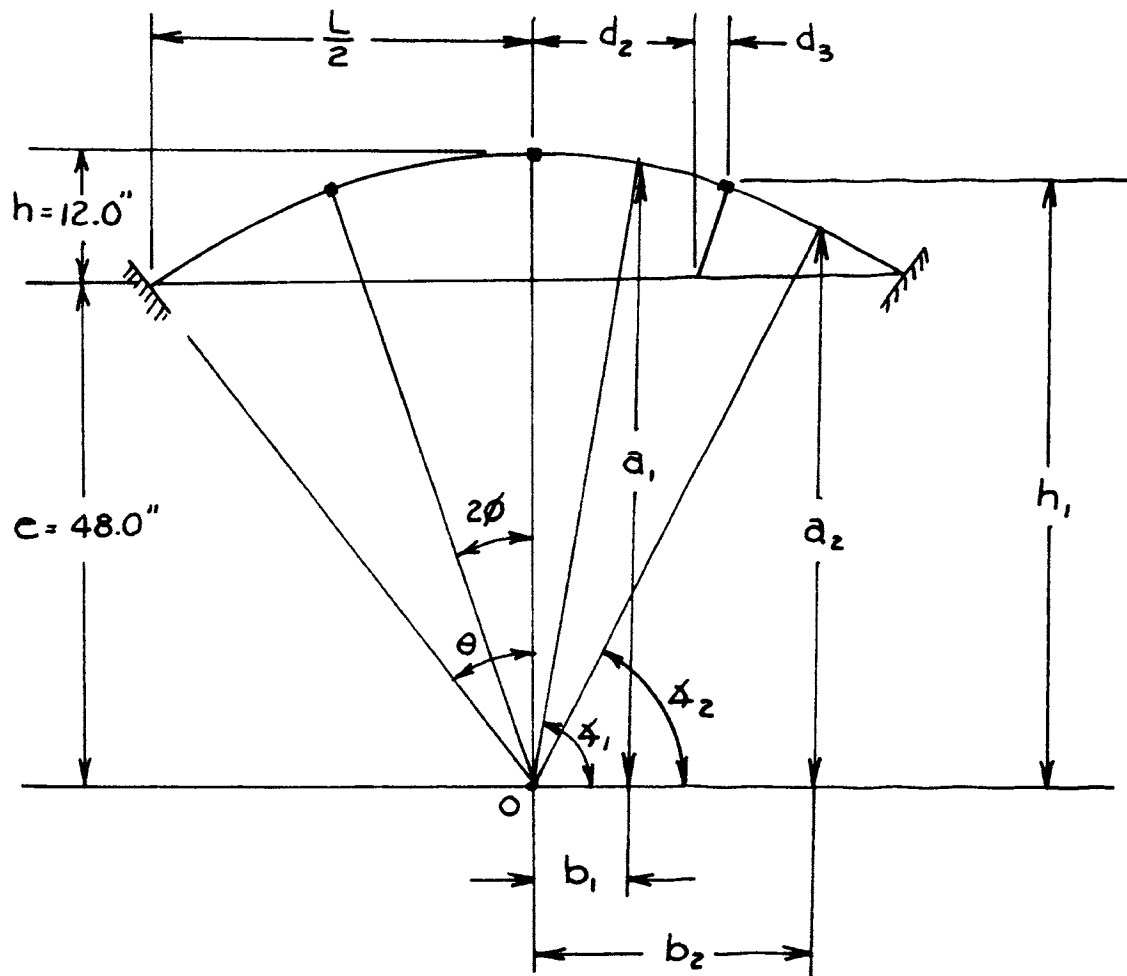


Figure 4.3 Basic coordinate dimensions

$$\theta = \sin^{-1} \frac{L/2}{R} = 36.87^\circ$$

$$h_1 = R \sin (90^\circ - 2\phi) = 56.92 \text{ in.}$$

$$d_1 = d_2 + d_3$$

$$d_1 = R \cos (90^\circ - 2\phi) = 18.97 \text{ in.}$$

$$d_3 = (h_1 - e) \frac{d_1}{h_1} = 2.97 \text{ in.}$$

$$d_2 = d_1 - d_3 = 16.0 \text{ in. (See Fig. 4.1)}$$

4.) General coordinates of nodes 1, 3, 7, 9

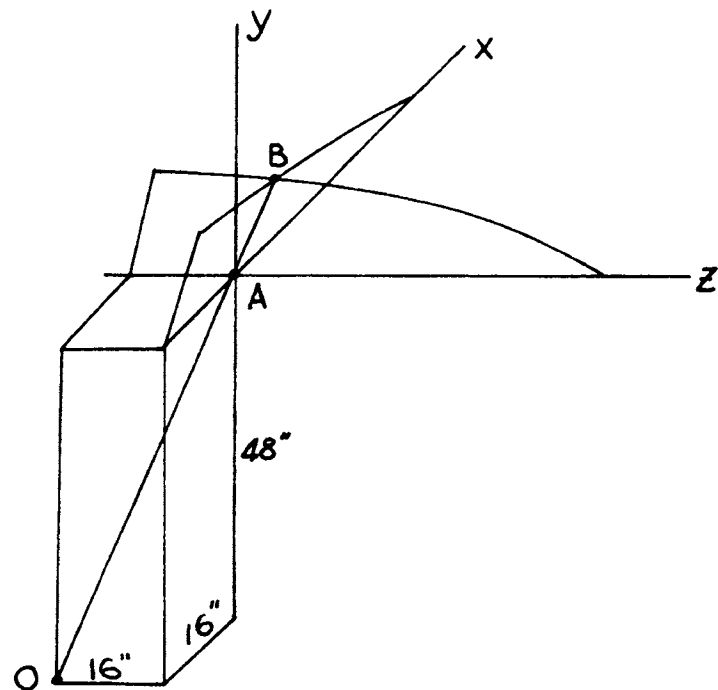


Figure 4.4 Geometry of 1, 3, 7, 9 node points.

$$OAB = 60.0 \text{ in.}$$

$$OA = \sqrt{2(16)^2 + (48)^2} = 53.066 \text{ in.}$$

$$AB = 60.000 - 53.066 = 6.934 \text{ in.}$$

$$\Delta x = \Delta z = 6.934 \frac{16}{53.066} = 2.09 \text{ in.}$$

$$y = 6.934 \frac{48}{53.066} = 6.27 \text{ in.}$$

The resulting coordinates of B without regard to sign are,

$$x = z = 16.0 + \Delta x = 18.09 \text{ in.}$$

$$y = 6.27 \text{ in.}$$

5.) General coordinates of nodes 2, 4, 6, and 8 are,

$$x = z = d_1 = 18.97 \text{ in.} \quad (\text{See Fig. 4.3})$$

$$y = h_1 - e = 8.92 \text{ in.}$$

6.) Member -  $\emptyset$  - Calculations

$$\emptyset_1 = \frac{\Theta}{4} = \frac{36.87}{4} = 9.2175^\circ$$

$$\emptyset_1 = 0.160877 \text{ rad.}$$

$\emptyset_2$  - Members 5, 6, 8, 10, 15, 17, 19, 20

$$\emptyset_2 = \sin^{-1} \frac{S_2}{60.0},$$

where,  $S_2$  = one half the chord length of the member.

Then,

$$2S_2 = \sqrt{(.88)^2 + (2.65)^2 + (18.091)^2} = 18.305 \text{ in.}$$

$$\emptyset_2 = \sin^{-1} \frac{18.305/2}{60} = 8.774^\circ$$

$$\emptyset_2 = 0.153135 \text{ rad.}$$

$\emptyset_3$  - Members 1, 3, 4, 7, 18, 21, 22, 24

$$\emptyset_3 = \sin^{-1} \frac{S_3}{60.0},$$

where,  $S_3$  equals one half the chord length of the member.

Then,

$$2S_3 = \sqrt{(14.16)^2 + (6.27)^2 + (2.09)^2} = 15.625 \text{ in.}$$

$$\phi_3 = \sin^{-1} \frac{15.625/2}{60.0} = 7.48^\circ$$

$$\phi_3 = 0.13055 \text{ rad.}$$

#### 7.) Member - $\alpha$ - computations

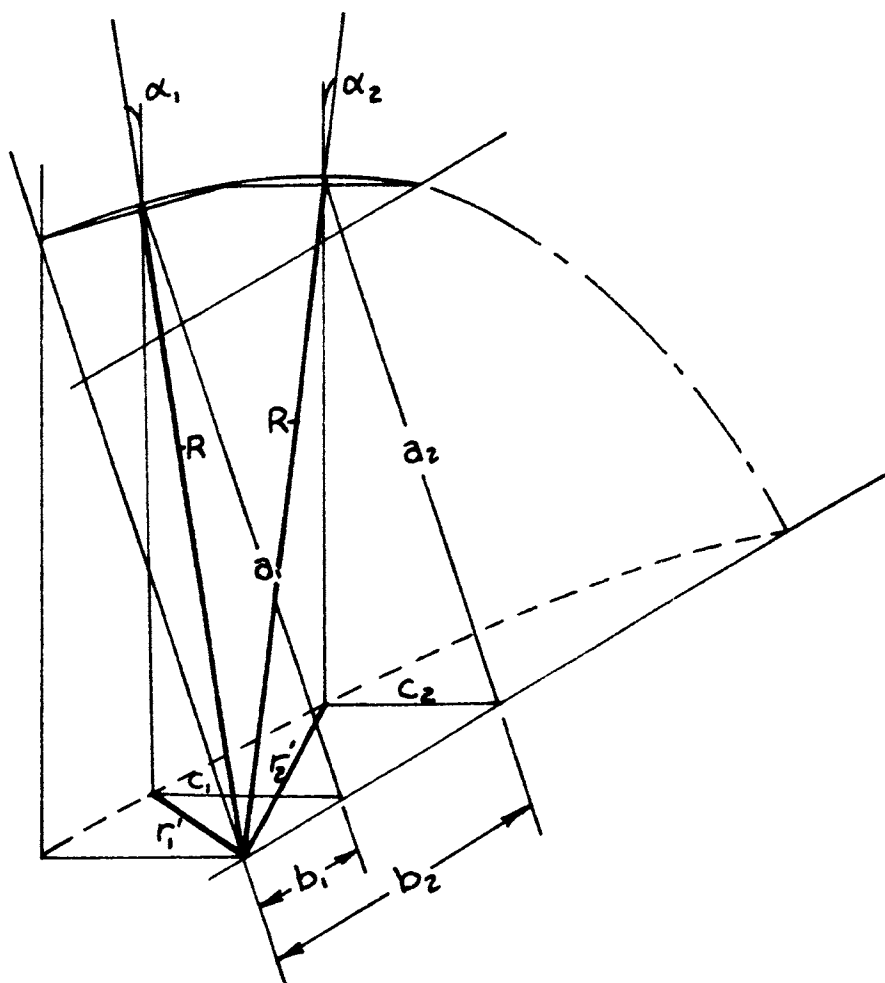


Figure 4.5 Required geometry for  $\alpha$ .

Computations continued,

$$\underline{\underline{\alpha_1 = 81.23^\circ:}} \quad (\text{See Fig. 4.3})$$

$$a_1 = 60 \sin (81.23^\circ) = 59.30 \text{ in.}$$

$$b_1 = 60 \cos (81.23^\circ) = 9.15$$

$$c_1 = 59.3 \sin (18.43^\circ) = 18.75$$

$$r'_1 = \sqrt{(9.15)^2 + (18.75)^2} = 20.864$$

$$\alpha_1 = \sin^{-1} \frac{20.864}{60.0} = 20.35^\circ$$

$$\underline{\underline{\alpha_1 = .35517 \text{ rad.}}}$$

$$\underline{\underline{\alpha_2 = 54.97^\circ:}}$$

$$a_2 = 60 \sin (54.97) = 49.13 \text{ in.}$$

$$b_2 = 60 \cos (54.97) = 34.44 \text{ in.}$$

$$c_2 = 49.13 \sin (18.43) = 15.53 \text{ in.}$$

$$r'_2 = \sqrt{(34.44)^2 + (15.53)^2} = 37.727 \text{ in.}$$

$$\alpha_2 = \sin^{-1} \frac{37.727}{60.0} = 38.96^\circ$$

$$\underline{\underline{\alpha_2 = .67998 \text{ rad.}}}$$

### 4.3 Example Computer Analysis

The computer solution of Mathematical Model-1, with fixed foundation attachments and symmetrically loaded is presented here to illustrate the computer output.

The input data, such as coordinates, member properties, nodal data and nodal loads is printed first. The input data is followed by the computed data such as the "Link" array, a sample flexibility, stiffness and transformation matrix, the nodal displacements and the member-end forces and moments in both system and member coordinates.

The computer solutions for the other mathematical models are presented in Appendix A.



NUMBER OF MEMBERS 9  
 NUMBER OF MEMBERS 24  
 NUMBER OF LOADING CONDITIONS 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER	X(1,1)	X(1,2)	MEMBER COORDINANTS X(1,1)	Y(1,1)	Z(1,1)	X(1,2)	Y(1,2)	Z(1,2)
1	-32.250000	-18.089996	0.0	6.270000	16.000000	18.089996	0.0	0.0
2	-32.250000	-18.089996	0.0	8.919999	0.0	0.0	0.0	0.0
3	-32.250000	-18.089996	0.0	6.270000	-16.000000	-18.089996	0.0	0.0
4	-16.000000	-18.089996	0.0	6.270000	32.250000	18.089996	0.0	0.0
5	-18.089996	-18.089996	6.270000	8.919999	18.089996	0.0	0.0	0.0
6	-18.089996	-18.089996	6.270000	8.919999	0.0	-18.089996	0.0	0.0
7	-16.000000	-18.089996	0.0	6.270000	-32.250000	-18.089996	0.0	0.0
8	-18.089996	-18.089996	6.270000	8.919999	18.089996	18.089996	0.0	0.0
9	-18.089996	-18.089996	6.270000	8.919999	12.000000	0.0	0.0	0.0
10	0.0	0.0	6.270000	8.919999	0.0	0.0	0.0	0.0
11	0.0	0.0	8.919999	12.000000	0.0	0.0	0.0	0.0
12	0.0	0.0	12.000000	8.919999	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	8.919999	-32.250000	-18.089996	0.0	0.0
14	0.0	18.089996	6.270000	6.270000	18.089996	18.089996	0.0	0.0
15	0.0	18.089996	12.000000	8.919999	0.0	0.0	0.0	0.0
16	0.0	18.089996	8.919999	6.270000	-18.089996	-18.089996	0.0	0.0
17	16.000000	18.089996	0.0	6.270000	32.250000	18.089996	0.0	0.0
18	18.089996	18.089996	6.270000	8.919999	18.089996	0.0	0.0	0.0
19	18.089996	18.089996	6.270000	8.919999	0.0	-18.089996	0.0	0.0
20	18.089996	18.089996	6.270000	8.919999	0.0	0.0	0.0	0.0
21	16.000000	18.089996	0.0	6.270000	-32.250000	-18.089996	0.0	0.0
22	32.250000	18.089996	0.0	6.270000	16.000000	18.089996	0.0	0.0
23	32.250000	18.089996	0.0	8.919999	0.0	0.0	0.0	0.0
24	32.250000	18.089996	0.0	6.270000	-16.000000	-18.089996	0.0	0.0

MEMBER	IX	IY	IZ	MEMBER PROPERTIES AREA	RADIUS	DHI	ALPHA
1	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
2	0.000326	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
3	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	-0.670000
4	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	-0.670000
5	0.000326	0.000326	0.000326	0.062500	60.000000	0.153135	-0.355170
6	0.000326	0.000326	0.000326	0.062500	60.000000	0.153135	-0.355170
7	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
8	0.000326	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
9	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
10	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
11	0.000326	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
12	0.000326	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
13	0.000326	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
14	0.000326	0.000326	0.000326	0.062500	60.000000	0.153135	-0.355170
15	0.000326	0.000326	0.000326	0.062500	60.000000	0.153135	-0.355170
16	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
17	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
18	0.000326	0.000326	0.000326	0.062500	60.000000	0.153135	-0.355170
19	0.000326	0.000326	0.000326	0.062500	60.000000	0.153135	-0.355170
20	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
21	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
22	0.000326	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
23	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000
24	0.000326	0.000326	0.000326	0.062500	60.000000	0.130550	0.670000

MEMBER END POINTS

MEMBER	1	2
1	1	2
2	1	3
3	1	4
4	2	3
5	2	4
6	3	4
7	1	5
8	2	5
9	3	5
10	4	5
11	1	6
12	2	6
13	3	6
14	4	6
15	5	6
16	1	7
17	2	7
18	3	7
19	4	7
20	5	7
21	1	8
22	2	8
23	3	8
24	4	8

MEMBER	1	2	3	4	5
1	4	-1	-4	6	0
2	4	-2	-6	6	0
3	4	-3	-6	2	13
4	4	-8	-11	12	15
5	4	-8	-12	13	16
6	4	-12	-12	14	17
7	4	-15	-12	14	22
8	4	-17	-12	28	23
9	4	-17	-20	21	24

NODE	DX	DY	APPLIED NODAL LOADS	MY	MZ
1	0.0	-10.000000	0.0	0.0	0.0
2	0.0	-10.000000	0.0	0.0	0.0
3	0.0	-10.000000	0.0	0.0	0.0
4	0.0	-10.000000	0.0	0.0	0.0
5	0.0	-10.000000	0.0	0.0	0.0
6	0.0	-10.000000	0.0	0.0	0.0
7	0.0	-10.000000	0.0	0.0	0.0
8	0.0	-10.000000	0.0	0.0	0.0
9	0.0	-10.000000	0.0	0.0	0.0

MEMBER 1

FLEXIBILITY MATRIX

0.00023270	0.00425726	0.0	0.0	0.0	0.00054506
0.00425726	0.13037485	0.0	0.0	0.0	0.01251159
0.0	0.0	0.13124073	-0.00106771	-0.01254572	0.0
0.0	0.0	-0.00106771	0.00236927	0.0	0.0
0.0	0.0	-0.01254572	0.0	0.00160621	0.0
0.00054506	0.01251159	0.0	0.0	0.0	0.00160184

STIFFNESS MATRIX

11175.05468750	0.06886506	0.0	0.0	0.0	-7205.85546875
0.06886506	30.62805176	0.0	0.0	0.0	-230.25207520
0.0	0.0	30.51765442	13.75273800	238.36636353	0.0
0.0	0.0	13.75273800	428.26928711	107.41940308	0.0
0.0	0.0	238.36636353	107.41940308	2484.40771484	0.0
-7205.85546875	-230.25207520	0.0	0.0	0.0	4944.98828125

TRANSFORMATION MATRIX

0.90615481	-0.40046787	0.13604748	0.0	0.0	0.0
0.40046787	0.71224648	-0.57594246	0.0	0.0	0.0
0.13604748	0.57594246	0.80609012	0.0	0.0	0.0
0.0	0.0	0.0	0.90615481	-0.40046787	0.13604748
0.0	0.0	0.0	0.40046787	0.71224648	-0.57594246
0.0	0.0	0.0	0.13604748	0.57594246	0.80609012

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODAL DISPLACEMENTS

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.001350	-0.004734	-0.001350	-0.000140	0.000000	-0.000140
2	-0.001350	-0.004734	0.000000	-0.000000	-0.000000	-0.000000
3	0.001350	-0.004734	0.001350	-0.000140	-0.000000	-0.000140
4	-0.000000	-0.004734	0.000000	-0.000000	0.000845	-0.000000
5	-0.000000	-0.004734	0.000000	-0.000000	-0.000000	-0.000000
6	-0.001350	-0.004734	-0.001350	-0.000140	0.000000	0.000140
7	0.001350	-0.004734	0.000000	-0.000000	-0.000000	-0.000000
8	-0.001350	-0.004734	0.001350	-0.000140	0.000000	0.000140

## MEMBER END FORCES AND MOMENTS

56

MEMBER	END	PPX		PPY		PPZ		MPX		MPY		MPZ	
		PX	PY	PZ	PX	PY	PZ	PX	PY	PZ	PX	PY	PZ
1	1		15.517446	7.001240	2.324160	-0.784251	3.107269	-3.710376					
	1	17.181229	0.112203	-0.047720	0.039862	0.388247	-4.887200						
	2		-15.517446	-7.001240	-2.324160	0.774175	-3.585968	5.553484					
2	1		16.367950	8.497645	-0.000007	0.000013	0.000026	-9.736763					
	1	18.442215	-0.065948	-0.000002	0.000024	0.000017	-9.736763						
	2		-16.367950	-8.497645	0.000007	-0.000034	0.000013	8.449424					
3	1		15.517225	7.001164	-2.324138	0.784242	-3.107175	-3.710217					
	1	17.181000	0.112226	0.047725	-0.039853	-0.388268	-4.887016						
	2		-15.517225	-7.001164	2.324138	-0.724180	3.585979	5.553543					
4	1		-2.324136	7.001158	-15.517267	-3.710324	-3.107235	-0.784243					
	1	17.181046	0.112203	0.047716	-0.039862	-0.388250	-4.887137						
	2		2.324136	-7.001158	15.517267	5.553438	3.585930	0.724165					
5	1		-0.620195	2.001202	-13.813441	-6.420940	-2.696871	0.143279					
	1	13.971318	-0.038348	0.040918	-0.223351	-0.292174	-6.956066						
	2		0.620195	-2.001201	13.813439	6.017086	1.760520	-0.260799					
6	1		0.620194	-2.001200	-13.813398	-6.017179	-1.760560	-0.260802					
	1	13.971276	0.038343	-0.040918	0.223355	0.455791	-6.254237						
	2		-0.620194	2.001200	13.813398	6.420956	2.696895	0.143291					
7	1		-2.324149	7.001185	15.517304	3.710232	3.107190	-0.784243					
	1	17.181001	0.112216	-0.047716	0.039858	0.388272	-4.887037						
	2		2.324149	-7.001185	-15.517304	-5.553582	-3.586001	0.724181					
8	1		13.913387	2.001194	0.620192	0.143279	2.696865	-6.420915					
	1	13.971276	-0.038347	-0.040918	0.223351	0.292177	-6.956070						
	2		-13.913387	-2.001193	-0.620192	-0.260800	-1.760509	6.017069					
9	1		15.127674	2.500037	-0.000007	0.000141	0.000038	-7.927845					
	1	15.332743	0.043326	-0.000007	0.000145	0.000015	-7.927845						
	2		-15.127697	-2.500037	0.000007	-0.000161	0.000088	8.760396					
10	1		13.913320	2.001156	-0.620184	-0.143184	-2.696865	-6.421007					
	1	13.971112	-0.038363	0.040914	-0.223253	-0.292160	-6.956128						
	2		-13.913328	-2.001156	0.620184	0.260694	1.760494	6.016863					
11	1		0.000002	8.497686	-16.368027	-9.736802	-0.000025	0.000013					
	1	18.442307	-0.065947	0.000002	-0.000024	-0.000016	-9.736802						
	2		0.000002	-8.497686	16.368027	8.449473	-0.000012	-0.000033					
12	1		0.000002	2.500028	-15.127552	-7.927817	-0.000038	0.000141					
	1	15.332600	0.043325	0.000007	-0.000145	-0.000015	-7.927817						
	2		-0.000002	-2.500028	15.127543	8.760363	-0.000088	-0.000161					
13	1		-0.000008	-2.499988	-15.127657	-8.760085	0.000094	-0.000155					
	1	15.332787	-0.043328	-0.000008	0.000148	0.000121	-8.760085						
	2		0.000008	2.499988	15.127557	7.927856	0.000049	0.000142					
14	1		0.000001	8.497681	16.367935	9.736562	0.000019	0.000014					
	1	18.442215	-0.066010	-0.000001	0.000021	0.000010	-9.736562						
	2		0.000001	-8.497681	-16.367935	-8.449933	0.000007	-0.000027					
15	1		0.000001	8.497681	16.367935	9.736562	0.000019	0.000014					
	1	18.442215	-0.066010	-0.000001	0.000021	0.000010	-9.736562						
	2		0.000001	-8.497681	-16.367935	-8.449933	0.000007	-0.000027					

#### 4.4 Experimental Model

For simplicity the experimental model was designed with four interior nodes and eight foundation attachments, while consisting of twelve members. The dimensions given for the experimental model are the true measured dimensions, as, there was some discrepancy between the original design dimensions and the finished model. This can be expected, however, from a model of this nature.

a.) Material and Member Properties - The model was made from one fourth inch square steel rods. The geometric properties of the members are as follows:

1.) Polar moment of inertia,  $I_x^*$ , is

$$I_x = \beta hb^3 ,$$

where 
$$\beta \approx \frac{1}{3} - 0.21 \frac{b}{h} \left( 1 - \frac{b^4}{12h^4} \right) ,$$

and since  $b = h$  for a square section it follows that

$$\beta \approx .1418 ,$$

and if  $h = 0.25$  in.,

$$I_x = .00055 \text{ in.}^4 .$$

\*(See appendix C, of reference 2) Here  $I_x$  is the equivalent polar moment of inertia and is the same as the previously defined,  $I_N$ .

2.) Moment of inertia about the y and z axis is

$$I_y = I_z = \frac{b^4}{12} = 0.000326 \text{ in.}^4.$$

3.) Cross-sectional area equals  $b^2$ , or

$$\text{Area} = 0.0625 \text{ in.}^2.$$

4.) Young's modulus, E, was determined from a tension test of the material. The results of this test are as follows:

$$\text{strain} = 0.00061 \text{ in./in.},$$

$$\text{load } p = 1200.0 \text{ lbs.}$$

therefore,

$$E = \frac{p}{\text{strain} \times \text{area}} = 31,475,000 \text{ psi}$$

5.) The shear modulus, G, was assumed as, 12,000,000 psi.

b.) Geometry and Numbering Scheme for the Model

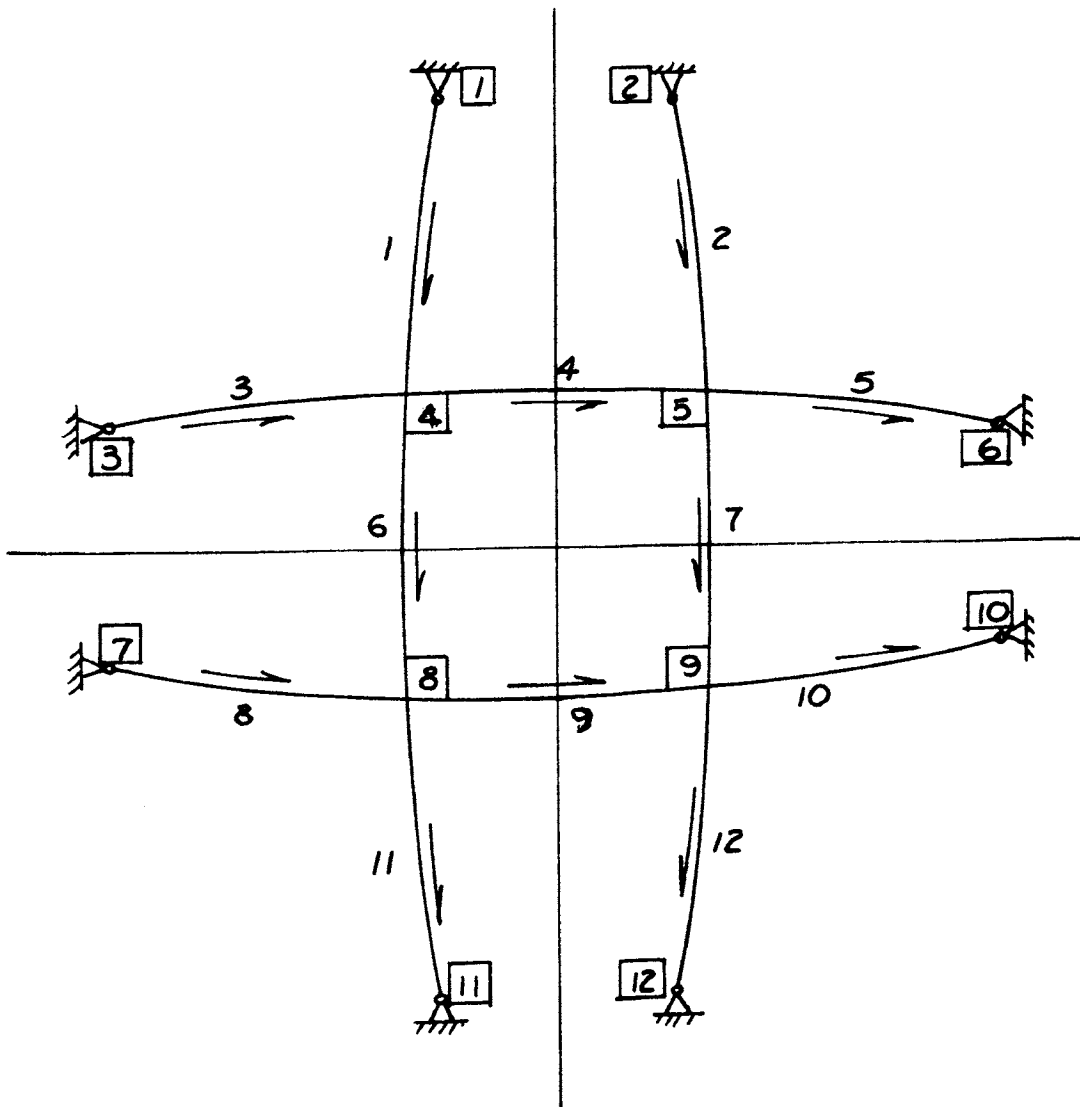


Figure 4.6 Experimental model with pinned foundation attachments.

The coordinates for each of the members are given in Table 4.1.

#### 4.5 Experimental Results

Three tests were run on the experimental model in which it was subjected to both symmetric and unsymmetric loads, for both pinned and fixed end foundation attachments. In each of the tests the strain in member-12 at 6.5 inches from node 4, and the vertical displacement at each of the nodes were measured. Because of the difficulty encountered in measuring the true vertical displacement of the node, these values are not too accurate. The following is a summary of the three tests.

##### a.) Test No. 1

Foundation attachments: all pinned

Load: 30.0 lbs. at each node

Strain in mem.-12:  $175.0 \times 10^{-6}$  in./in., ten.

Nodal displacement: 0.027 in. ave.

##### 2.) Theoretical calculations:

Member end forces on member-12 at End-1,  
in member coordinates are

$$p_x = 34.05 \text{ lbs.}$$

$$m_x = 0.0$$

$$p_y = -1.32 \text{ lbs.}$$

$$m_y = 1.08 \text{ in.-lbs.}$$

$$p_z = -0.071 \text{ lbs.}$$

$$m_z = -20.06 \text{ in.-lbs.,}$$

and at End-2,

$$p_x = -34.05 \text{ lbs.} \quad m_x = 0.0$$

$$p_y = 1.32 \text{ lbs.} \quad m_y = 0.0$$

$$p_z = 0.071 \text{ lbs.} \quad m_z = 0.0$$

Nodal displacements: 0.022 in. ave.

- 3.) Correlation of experimental stress in member-12, to the theoretical stress, neglecting  $p_z$  and  $m_y$ ;

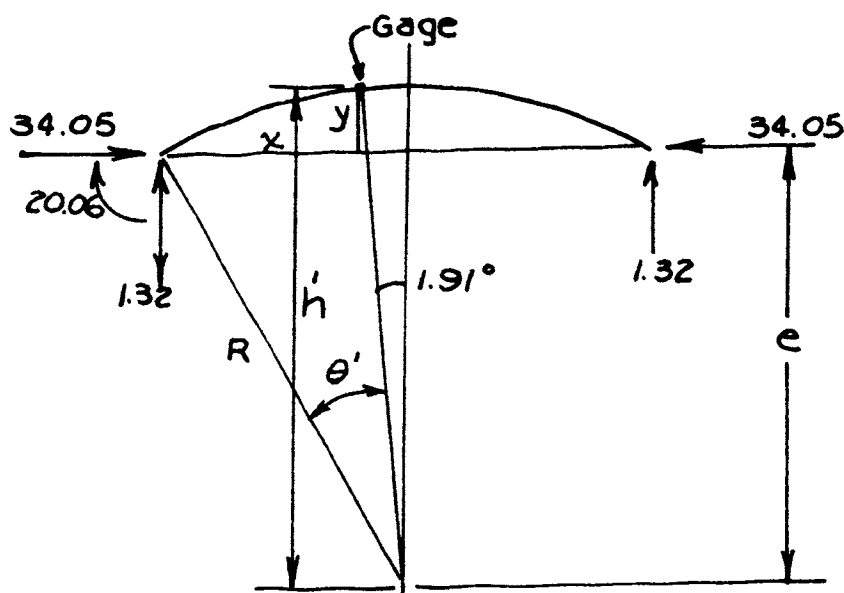


Figure 4.7 Free body diagram of Member-12.

$$\theta' = \frac{S}{R} = \frac{6.5}{35} = .1857 = 10.6^\circ$$

$$\phi - \theta' = 12.51 - 10.6 = 1.91^\circ$$

$$h' = 35 \cos(1.91^\circ) = 34.98 \text{ in.}$$



$$e = 35 \sin (77.49^\circ) = 34.17 = 0.81 \text{ in.}$$

$$x = R \sin (12.51^\circ) - \sin (1.91^\circ) = 6.41 \text{ in.}$$

The moment at the strain gage is,

$$M = 20.06 - 34.05 (.81) - 1.32 (6.41)$$

$$M = 15.98 \text{ in.-lb.}$$

$$\text{Axial} = 34.05 \cos (1.91^\circ) - 1.32 \sin (1.91^\circ)$$

$$\text{Axial} = 34.0 \#, \text{ compression.}$$

The resulting stress are,

$$\text{Axial stress} = \frac{34}{.0625} = 544 \text{ psi, C}$$

$$\text{Bending stress} = \frac{15.98 (.125)}{.000326} = 6127 \text{ psi, T}$$

$$\text{Total stress} = 6127 - 544 = 5583 \text{ psi, T}$$

4.) % - Deviation between experimental and theoretical stress:

$$\text{Exp. stress} = 175.0 \times 10^{-6} (31.475 \times 10^6)$$

$$" \quad " \quad = 5508 \text{ psi,}$$

Therefore,

$$\% \text{ Dev.} = \frac{5583 - 5508}{5508} \times 100 = 1.36\%$$

Table 4.1 Member Coordinates

Mem.	X(I,1)	Y(I,1)	Z(I,1)	X(I,2)	Y(I,2)	Z(I,2)
1	-20.97	0.0	5.87	-7.29	6.25	7.22
2	-20.97	0.0	-6.62	-7.23	6.25	-7.33
3	-6.06	0.0	20.97	-7.29	6.25	7.22
4	-7.29	6.25	7.22	-7.23	6.25	-7.33
5	-7.23	6.25	-7.33	-6.25	0.0	-20.97
6	-7.29	6.25	7.22	7.22	6.25	7.29
7	-7.23	6.25	-7.33	7.19	6.25	-7.19
8	6.06	0.0	20.97	7.22	6.25	7.29
9	7.22	6.25	7.29	7.19	6.25	-7.19
10	7.19	6.25	-7.19	6.12	0.0	-20.97
11	7.22	6.25	7.29	20.97	0.0	6.31
12	7.19	6.25	-7.19	20.97	0.0	-6.31

b.) Summary of Tests - A summary of the experimental results for Test No. 1, as well as Test No. 2 and Test No. 3 are given in Tables 4.2, 4.3 and 4.4 respectively.

Table 4.2 Summary of Test No. 1 - Pinned-End Foundation  
Symmetric Loading

- Experimental -				
	Nodal Displacements			
Load/node	1	2	3	4
30.0 lbs.	-0.027	-0.029	-0.027	-0.027
- Theoretical -				
30.0 lbs.	-0.021	-0.023	-0.022	-0.022

Mem.-12 - Strain =  $175.0 \times 10^{-6}$  in./in., T

Experimental Stress = 5508.0 psi., T

Computed Member-12 End Forces

End	$P_x$	$P_y$	$P_z$	$m_x$	$m_y$	$m_z$
1	34.05	-1.32	-0.07	0.0	1.08	-20.06
2	-34.05	1.32	0.07	0.0	0.0	0.0

Theoretical Stress = 5585.0 psi., T

$$\% \text{ Deviation} = \frac{5585 - 5508}{5508} \times 100 = 1.36\%$$

Table 4.3 Summary of Test No. 2 - Fixed-End Foundation  
Symmetric Loading

- Experimental -				
	Nodal Displacements			
Load/node	1	2	3	4
31.08 lbs.	-0.015	-0.017	-0.017	-0.014
- Theoretical -				
31.08 lbs.	-0.012	-0.013	-0.012	-0.012

Mem.-12 - Strain =  $135.0 \times 10^{-6}$  in./in., T

Experimental Stress = 4280.0 psi.

Computed Member-12 End Forces

End	$P_x$	$P_y$	$P_z$	$m_x$	$m_y$	$m_z$
1	38.32	-0.15	-0.11	0.69	0.85	-21.06
2	-38.32	0.15	0.11	-0.69	-0.80	18.76

Theoretical Stress = 3729.0 psi.

$$\% \text{ Deviation} = \frac{4280 - 3729}{4280} \times 100 = 12.9\%$$

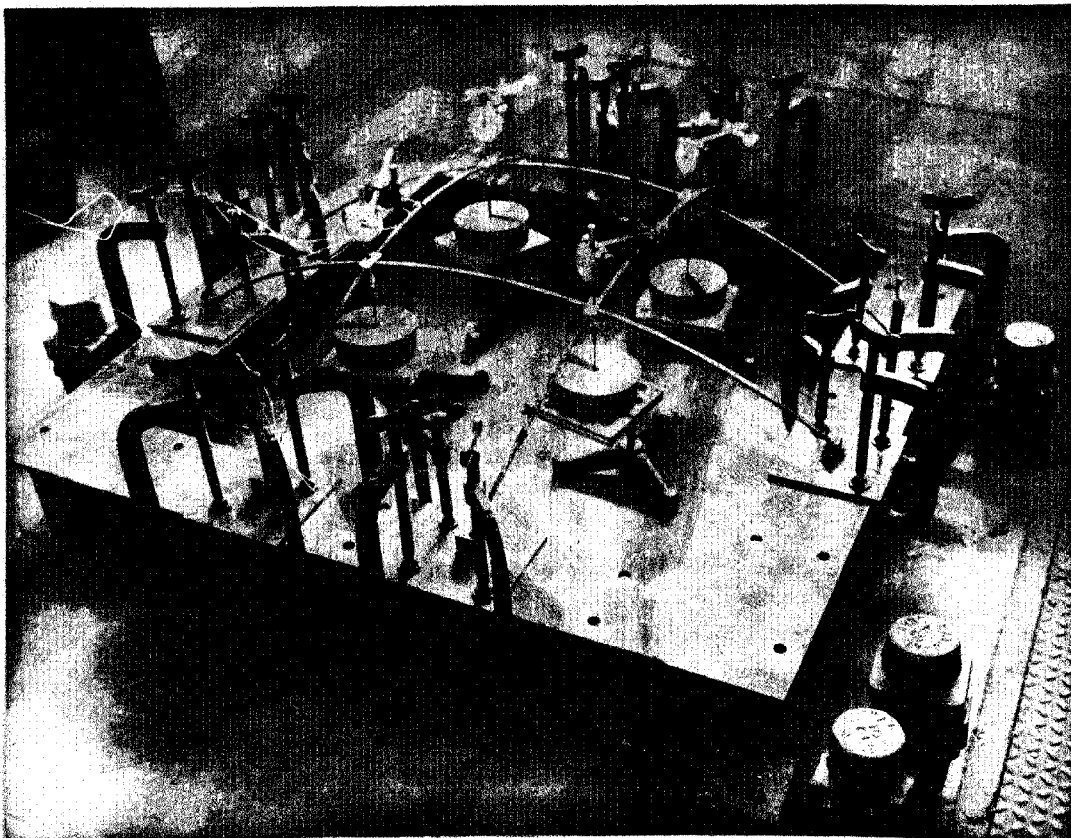
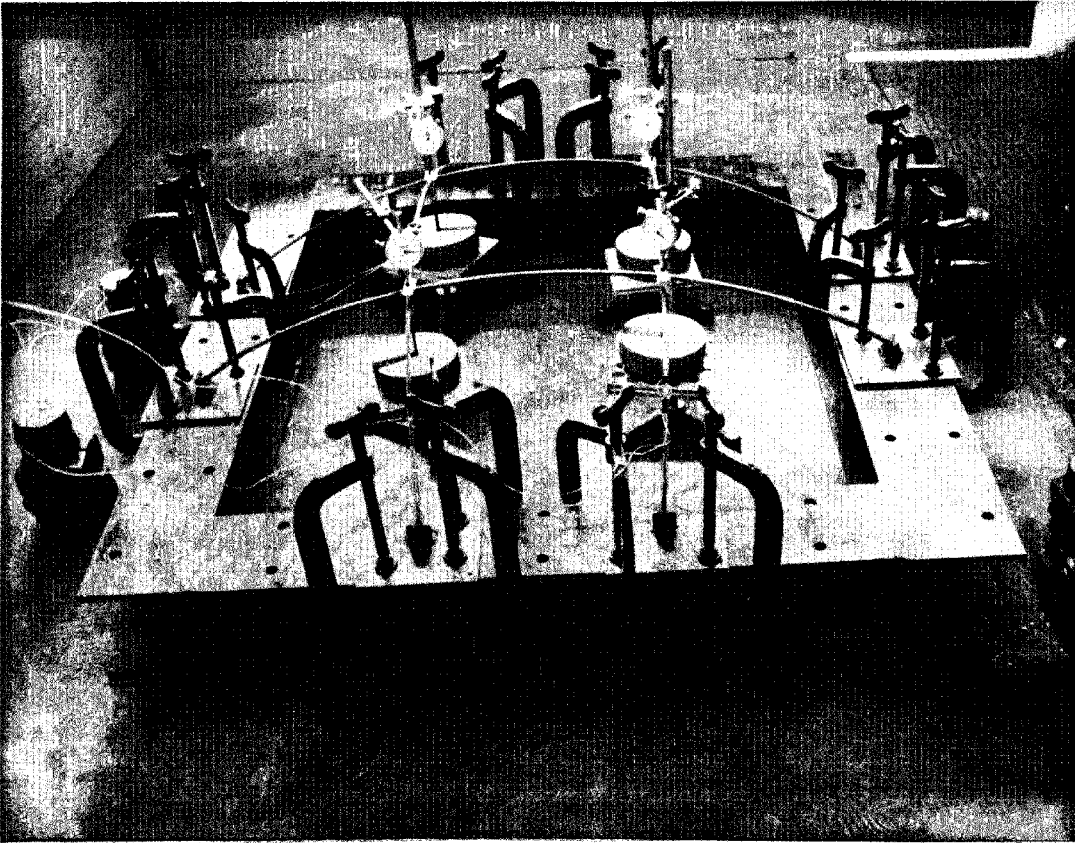


Figure 4.8 Photographs of Experimental Model

Table 4.4 Summary of Test No. 3 - Fixed-End Foundation  
 Unsymmetric Loading, Load Node 4 Only

- Experimental -				
	Nodal Displacements			
Load	1	2	3	4
31.08 lbs.	-0.0240	+0.030	+0.028	-0.051
-Theoretical -				
31.08 lbs.	-0.020	+0.024	+0.024	-0.041

Mem.-12 - Strain =  $55 \times 10^{-6}$  in./in.

Experimental stress = 1731 psi, T.

Computed Member-12 End Forces

End	$P_x$	$P_y$	$P_z$	$m_x$	$m_y$	$m_z$
1	29.45	1.55	-0.14	0.0	1.06	-4.4
2	-29.45	-1.55	0.14	0.0	1.08	27.87

Theoretical Stress = 1800 psi

$$\% \text{ Deviation} = \frac{1800 - 1731}{1731} \times 100 = 3.98\%$$

The computer solutions for the experimental model  
 is presented in Appendix B.

## Chapter 5

### COMPUTER PROGRAM FOR CURVILINEAR GRID SYSTEM

The computer program which performs the complex computations required by this or any matrix solution to a large structural system is an integral and extremely important part of the development. This chapter contains a detailed description of the development of the computer program including a complete documentation of the program including user instructions, flow diagrams, summary of program identifiers, and a complete listing of the program and subroutines.

#### 5.1 Introduction to the Program

Matrix Analysis of Curvilinear Systems (MACS), is written in Fortran IV for IBM System 360/40. The formulation of MACS is fundamentally an equilibrium-stiffness formulation. The program has been developed in two main parts; first the main line which consists of a general formulation of the equilibrium-stiffness method adaptable to the solution of any space frame, and second the subroutine package which consists of the necessary routines for performing the required repetitive calculations as well as those routines which enable MACS to be quite flexible as a space frame solver. The details of these routines will be discussed in (5.4).

In order for the engineer to use MACS it is only necessary that he provide MACS with the appropriate input data such as, Node data, geometry data, member properties and loading data as described in (5.2).

## 5.2 Input Data

The following is a detailed description of the necessary input data.

- a) System Indices - NN, NB, LL (Fixed pt. no.)
- NN - Number of nodes excluding fixed and foundation attachments.
- NB - Number of members in the structure.
- LL - Maximum number of members per node.
- b) Member Indices
- Node<sub>i1</sub> - Node number at 1-end of member I.
- Node<sub>i2</sub> - Node number at 2-end of member I.
- c) Member Coordinates
- X<sub>i1</sub>, Y<sub>i1</sub>, Z<sub>i1</sub> - Coordinates of the 1-end of member I in system coordinates.
- X<sub>i2</sub>, Y<sub>i2</sub>, Z<sub>i2</sub> - Coordinates of the 2-end of member I in system coordinates.
- d) Member Properties
- IX<sub>i</sub>, IY<sub>i</sub>, IZ<sub>i</sub> - Moments of inertia about the member X, Y and Z axis.
- Area<sub>i</sub> - Cross sectional area of member
- e) Geometric Properties
- R<sub>i</sub> - Radius of curvature of member



- $\phi_i$  - One half of central angle described by the member
- ALPH<sub>i</sub> - The rotation of the member about the member x-axis with respect to a vertical plane through its end points.

f) Elastic Properties

- E - Youngs Modulus
- G - Shear Modulus

g) Nodal Loads

- D<sub>j</sub> - Six element vector which describes  $p_x, p_y, p_z, m_x, m_y, m_z$ , the six possible components of the applied nodal load vector.

### 5.3 Output Data

The output data from MACS was designed to give the engineer all the information necessary for establishing the validity of the solution.

- a) Input Data - Before any structure can be analyzed correctly the correct input data must be used. This is a common source of error in an analysis of this type where large quantities of data are assembled, coded and finally key punched onto data cards. As a result it is quite easy for human error to be a significant factor. For this reason all of the input data is printed out in a convenient form to enable the engineer to verify that the data read by the computer is the correct data.

- b) Link Array<sup>6</sup> - The first computation performed by MACS computes,

Link (J, L) =  $\pm$  I, where I is the L<sup>th</sup> member to be attached to node J. The plus sign indicates that the 1-end of member I is at node J, the minus sign indicates that the 2-end of member I is at node J.

Link (J, L) provides MACS with the necessary member to node reference required for computing which members contribute to the stiffness of a particular node.

- c) Member Flexibility, Stiffness and Transformation Matrices - The flexibility, stiffness and transformation matrices for each member are printed out to provide the engineer with information necessary for developing a better understanding of the behavior of the system. The member stiffness matrix, for example, will show the contribution of each of the quantities such as axial, torsional and bending deformations to the overall member stiffness.
- d) System Stiffness Matrix - The system stiffness matrix furnishes additional information on the structural system as a whole, being the coefficients of the load-displacement equations for the structural system.

- e) Nodal Displacements - After the system stiffness matrix has been generated, the nodal loads and the system stiffness matrix are then solved simultaneously by SØLVE to obtain the nodal displacements. The six components of the displacement of each node are written out as follows,

$$\delta_x \quad \delta_y \quad \delta_z \quad \theta_x \quad \theta_y \quad \theta_z$$

- f) Member End Forces - The resultant force vector at each end of the member are first printed out in System Coordinates to facilitate checking the statics of the structure and then printed out in member coordinates which is more convenient for stress analysis.

#### 5.4 MACS Subroutine Package

The following discussion describes the subroutines required by MACS to perform the necessary repetitive calculations. These subroutines have been appropriately named STIFMA, MATRAN, MTXMUL, TRIPRØ AND SØLVE.

- a) STIFMA - The member stiffnesses are computed by calculating the member flexibilities from the equations derived in Chapter 2 and inverting the resulting flexibility matrix. STIFMA also computes the equilibrium matrix H, which is described in Chapter 2.

- b) MATRAN - The 3-axis rotation transformation matrix  $T$  is computed by MATRAN from the member end coordinates and ALPH which was described earlier in this chapter.
- c) MTXMUL - Matrix multiplication is performed by MTXMUL; given two matrices A and B. MTXMUL premultiplies B by A and places the resultant AB in the B array making efficient use of storage.
- d) TRIPRØ - Matrix Triple Product, premultiplies B by A and then post multiplies AB by the transpose of A to form  $ABA^t$ . The resulting array is then returned to the B array storage location which again makes efficient use of storage.
- e) SØLVE - This subroutine is a "simultaneous equation solver". SØLVE uses a Jordan Elimination and pivots on the largest element in the coefficient matrix<sup>8</sup>, thereby, reducing the roundoff error to a minimum and preserving the accuracy of the solution.

As was mentioned previously in (5.1) this subroutine package enables MACS to be quite flexible as a space frame solver. For example, if a space frame is composed of straight members instead of segmental circular members, the STIFMA which computes the stiffnesses of the circular members can be replaced by a new STIFMA which computes the

stiffness of a straight member. In fact, as long as one can derive the flexibility or stiffness matrix for a member, regardless of its shape, and substitute the appropriate STIFMA routine, MACS is capable of analyzing a system composed of these members. In addition, there is also the possibility that a structural system may be composed of both straight and curved members. In this case a stiffness subroutine could be included for each type of member and properly labeled for ready reference.

### 5.5 MACS User Instructions

It is impossible for any computer program to obtain the correct results unless the correct input data is supplied to the program. For this reason the engineer must be meticulous and systematic in preparing the input data for the computer program.

The data can be divided into two main parts, structure data and load data.

- a) Structure Data - It is advisable to begin the preparation of the structure data with a convenient sketch of the structural system. Next, the nodes and members can be numbered in any arbitrary fashion, except that fixed-end foundation attachments are labeled zero. Pinned-end foundation attachments are labeled the same as any other node.

The location of the system coordinate axes can now be placed at any convenient location. The data cards can now be listed in the order in which they appear in the data deck as follows:

- 1.) 1 Card - contains the system indices NB, NN and LL respectively in an I10 Format.
- 2.) 2NB/12 Cards - Node (I,1) and Node (I,2) are placed in a 12I6 Format. All values of (I,1) are read first followed immediately by the values of Node (I,2).
- 3.) NB Cards - The next NB cards contain the joint coordinates at each end of a member and ALPH (I) for each of the members. The order of the data is X(I,1), Y(I,1), Z(I,1) X(L,2), Y(I,2), Z(I,2) and ALPH (I) written in a 7E10.2 Format.
- 4.) NB Cards - The next NB cards contain the member properties IX(I), IY(I), IZ(I), AREA(I), R(I) and PHI(I) respectively for each member in a 6E10.2 Format.
- 5.) 1 Card - This card contains E and G in a 2E10.2 Format.

b) Load Data - The preceding data cards were concerned entirely with the structure data, next the load data will be read.

1.) NN Cards - The last NN cards of the data deck contain the nodal loads. There must be one card for every non-zero node with the six components of the load vector entered on the card in a 6E12.4 Format. If a node is not loaded a blank card can be used to represent a zero load vector.

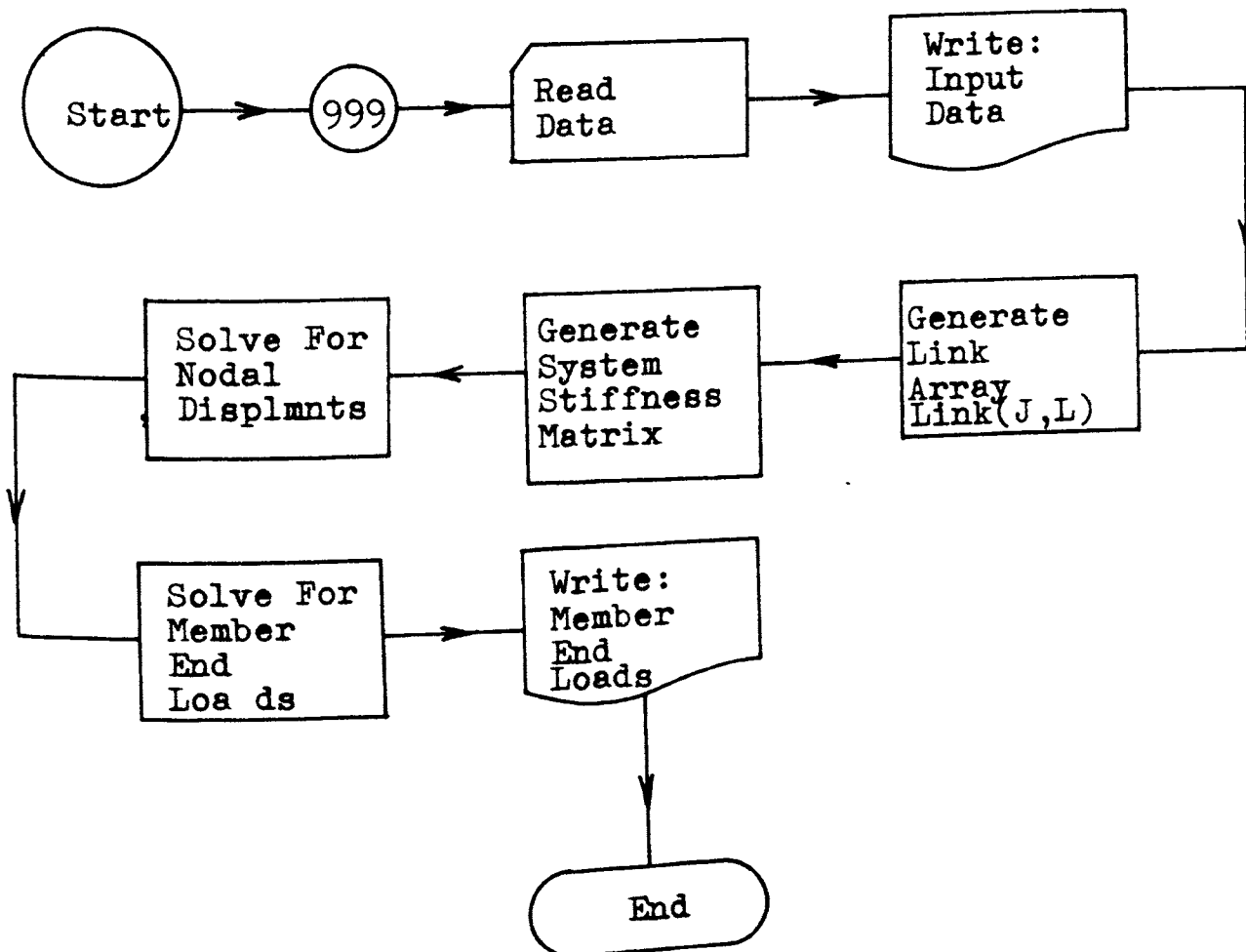
If a member is loaded between nodes the fixed end forces and moments must be computed and then their signs reversed and superimposed on the applied nodal loads as explained in Chapter 2.

## 5.6 Flow Diagrams and Computer Program

In the development of large, complex computer programs the flow diagram is a necessary and valuable tool. The flow diagram presents to the programmer, in the appropriate order, the various operations and decisions that the computer must make during the execution of the program.

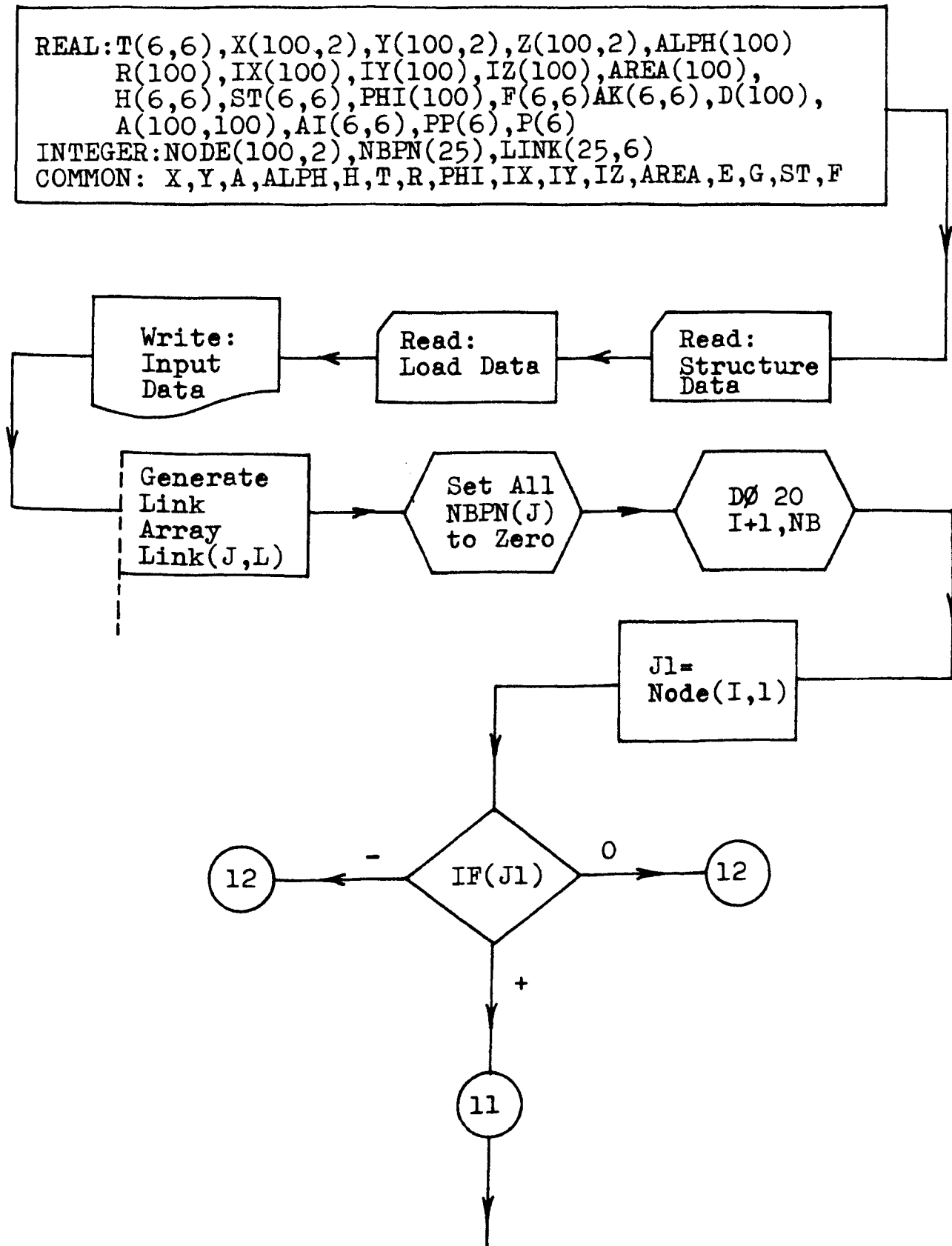
The first flow diagram is a block diagram and represents schematically the major operations performed by the computer.

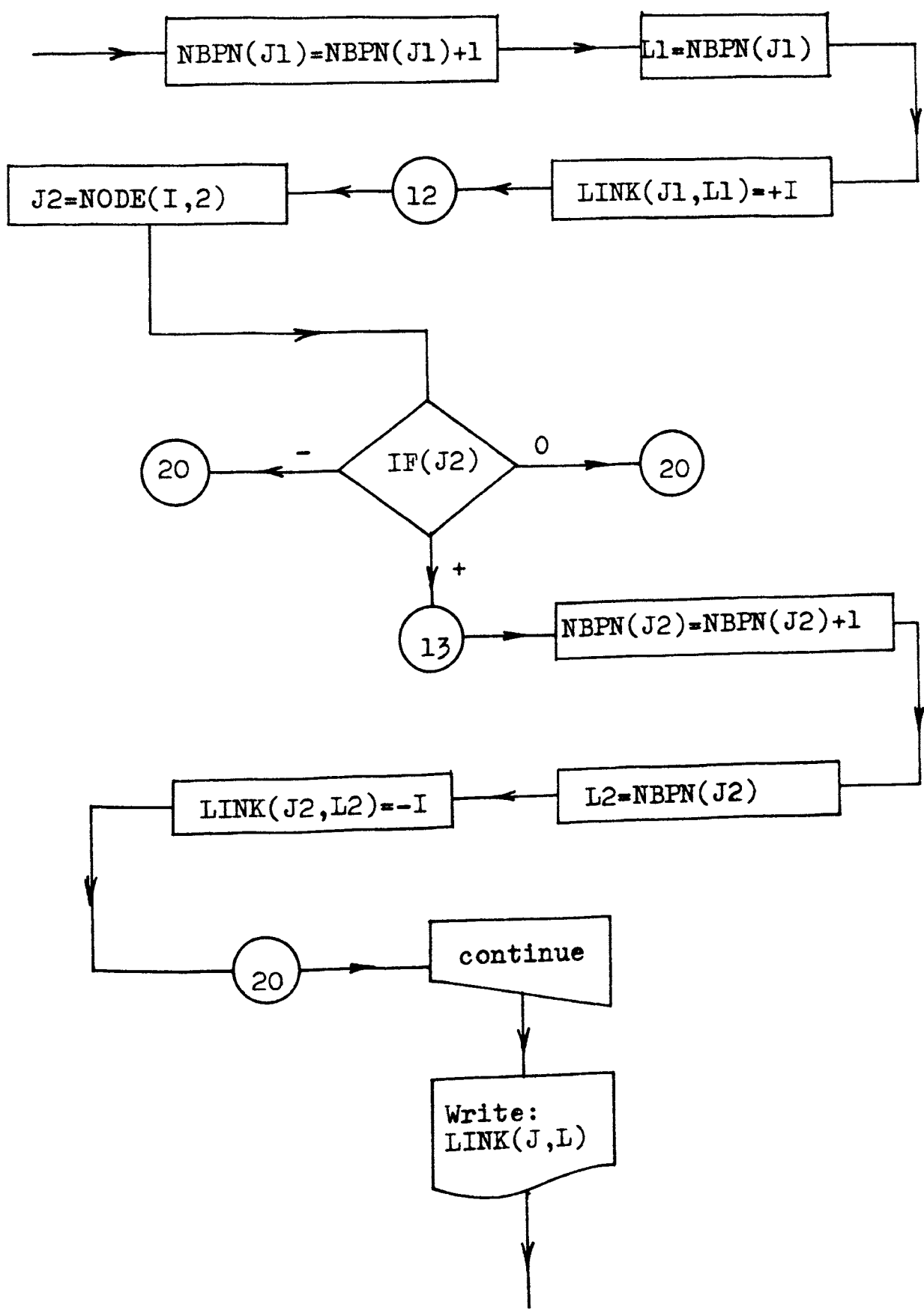
### a.) Block Diagram

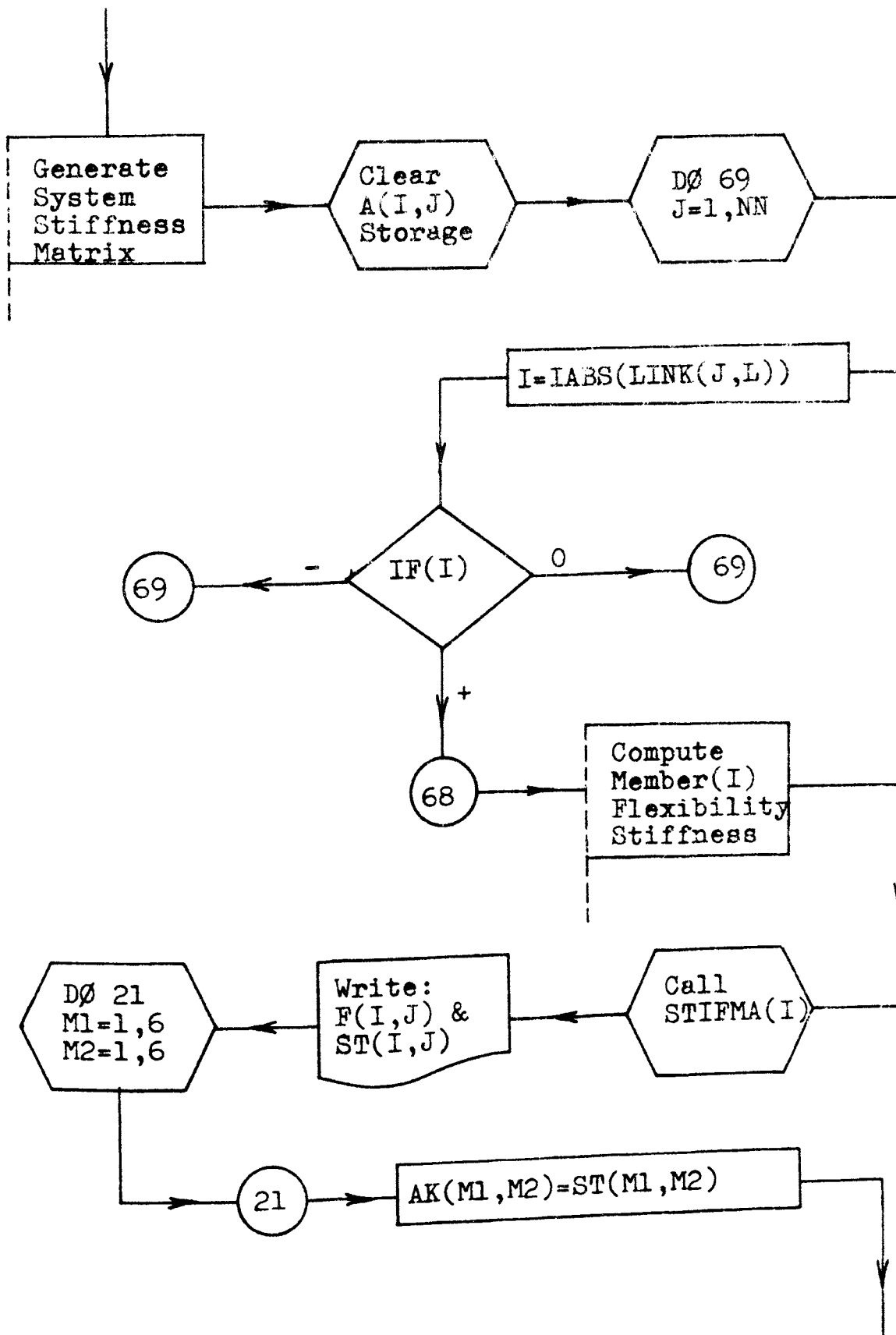


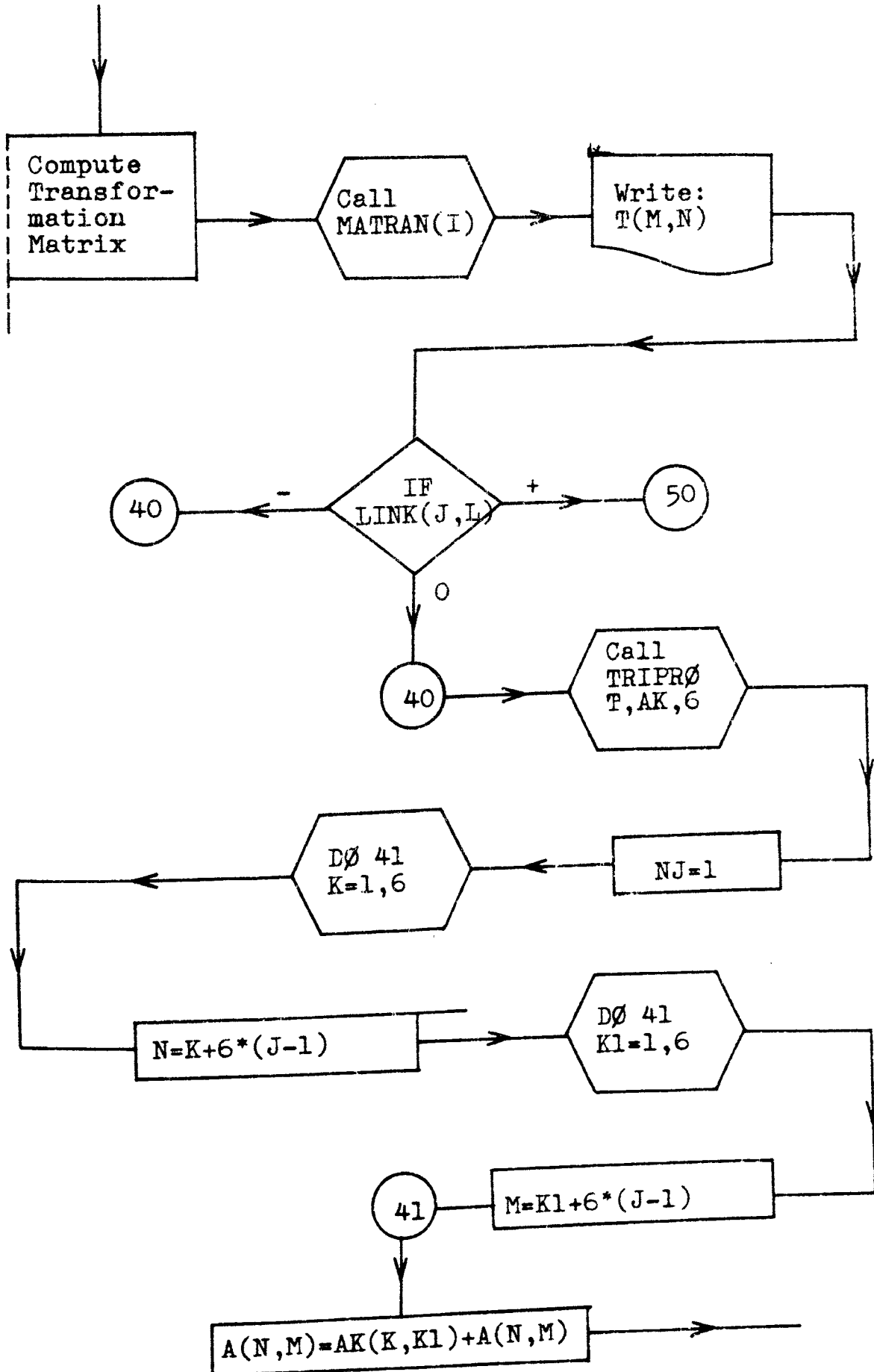


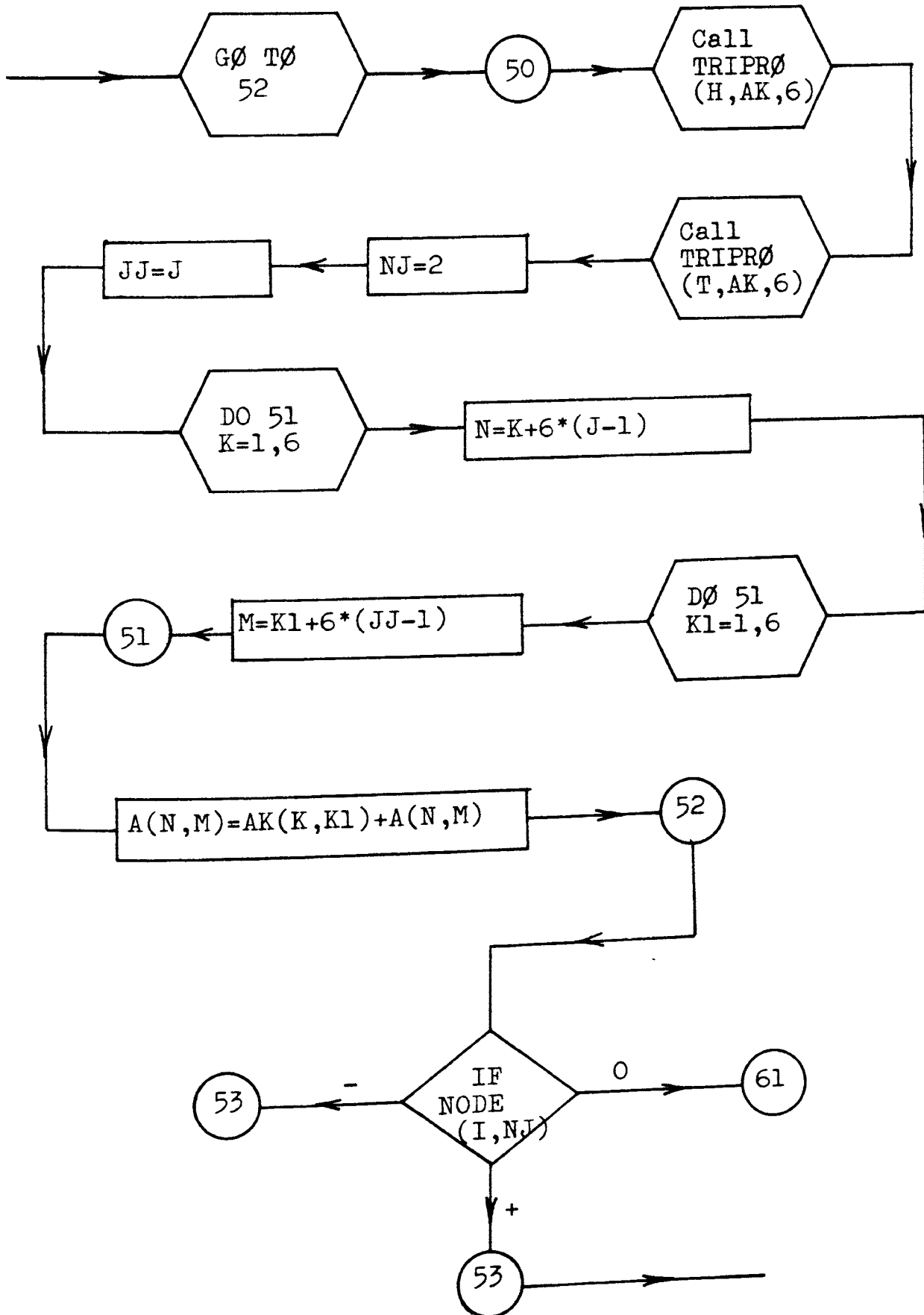
## b.) Main Line Flow Diagram.

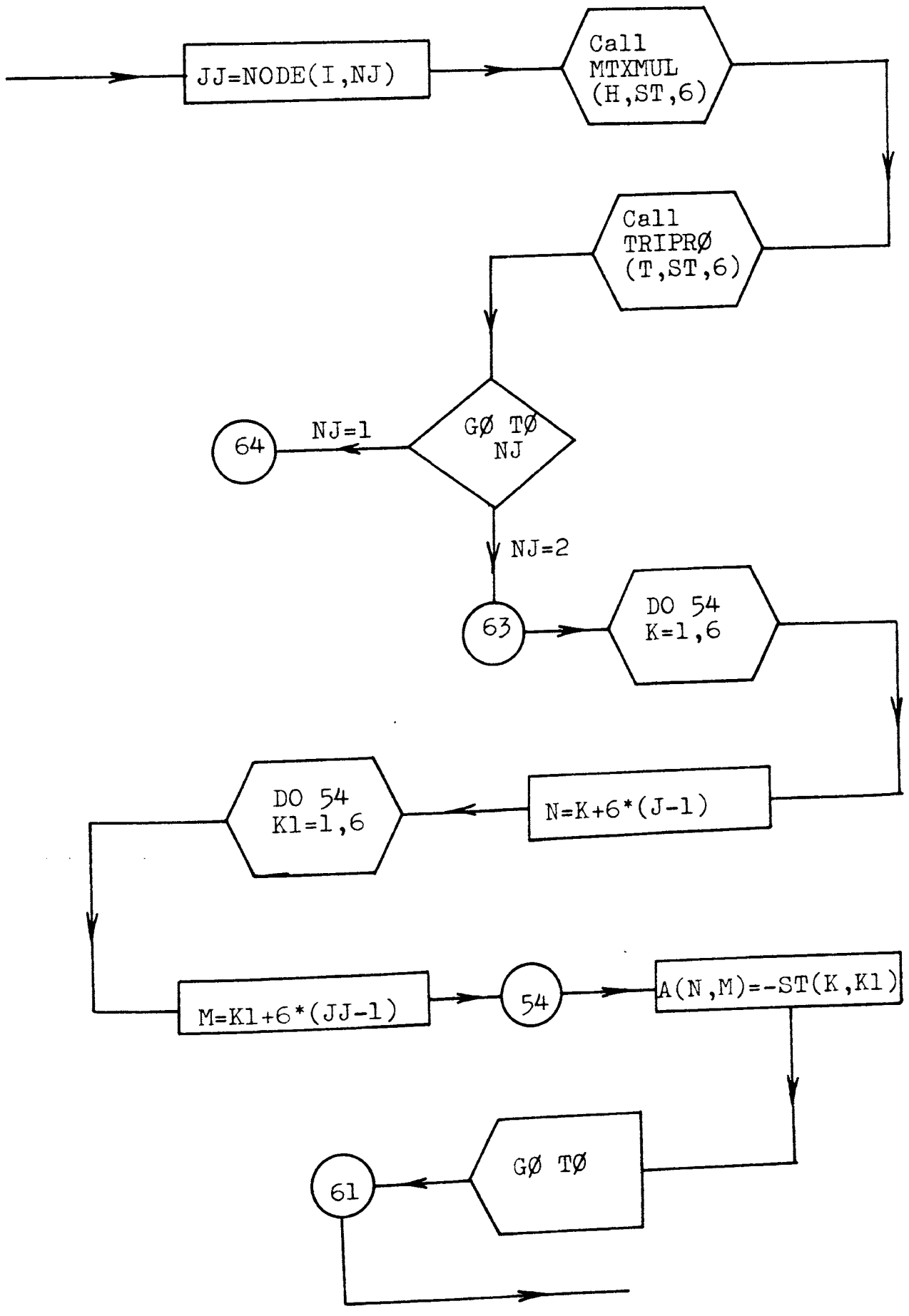


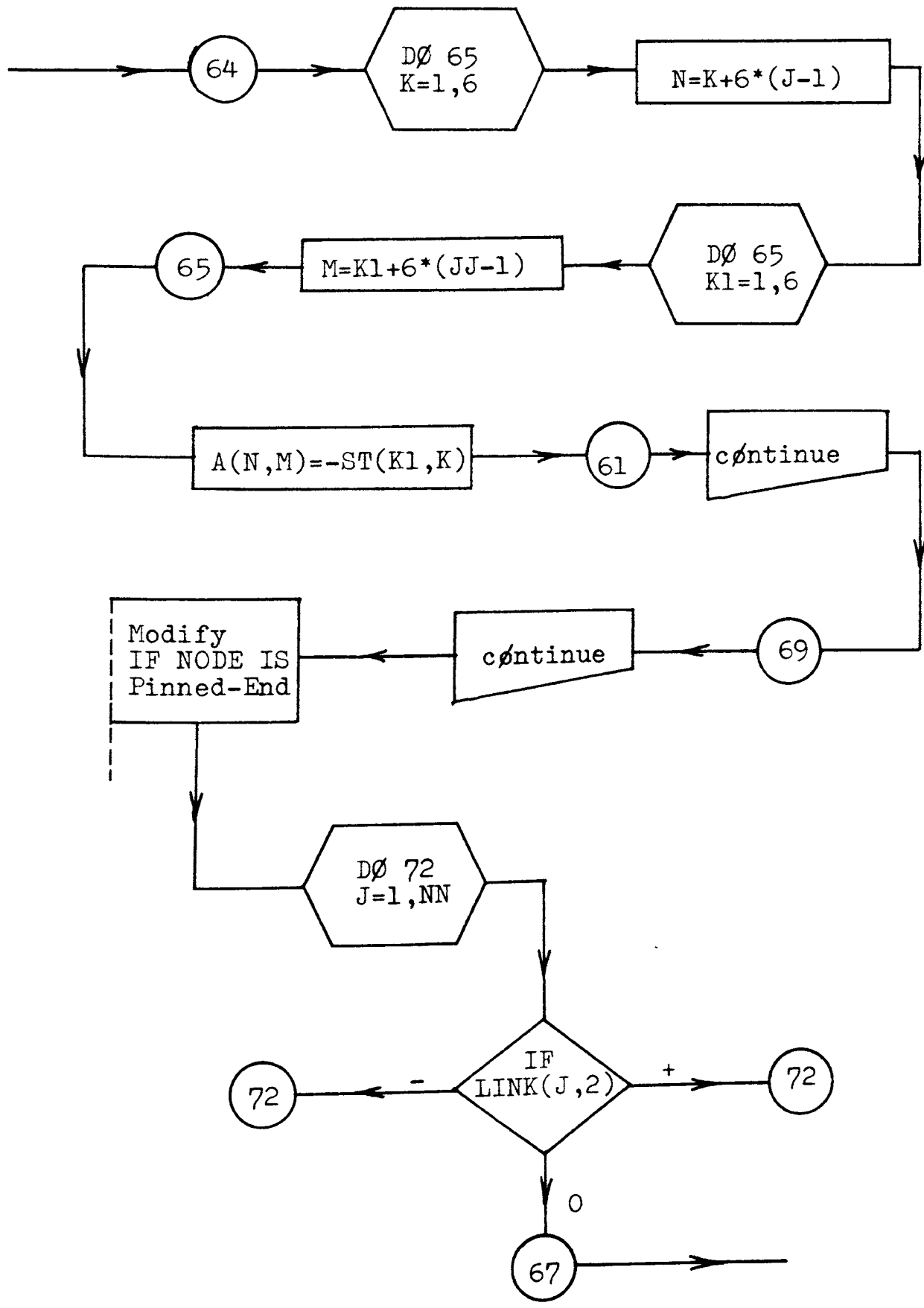


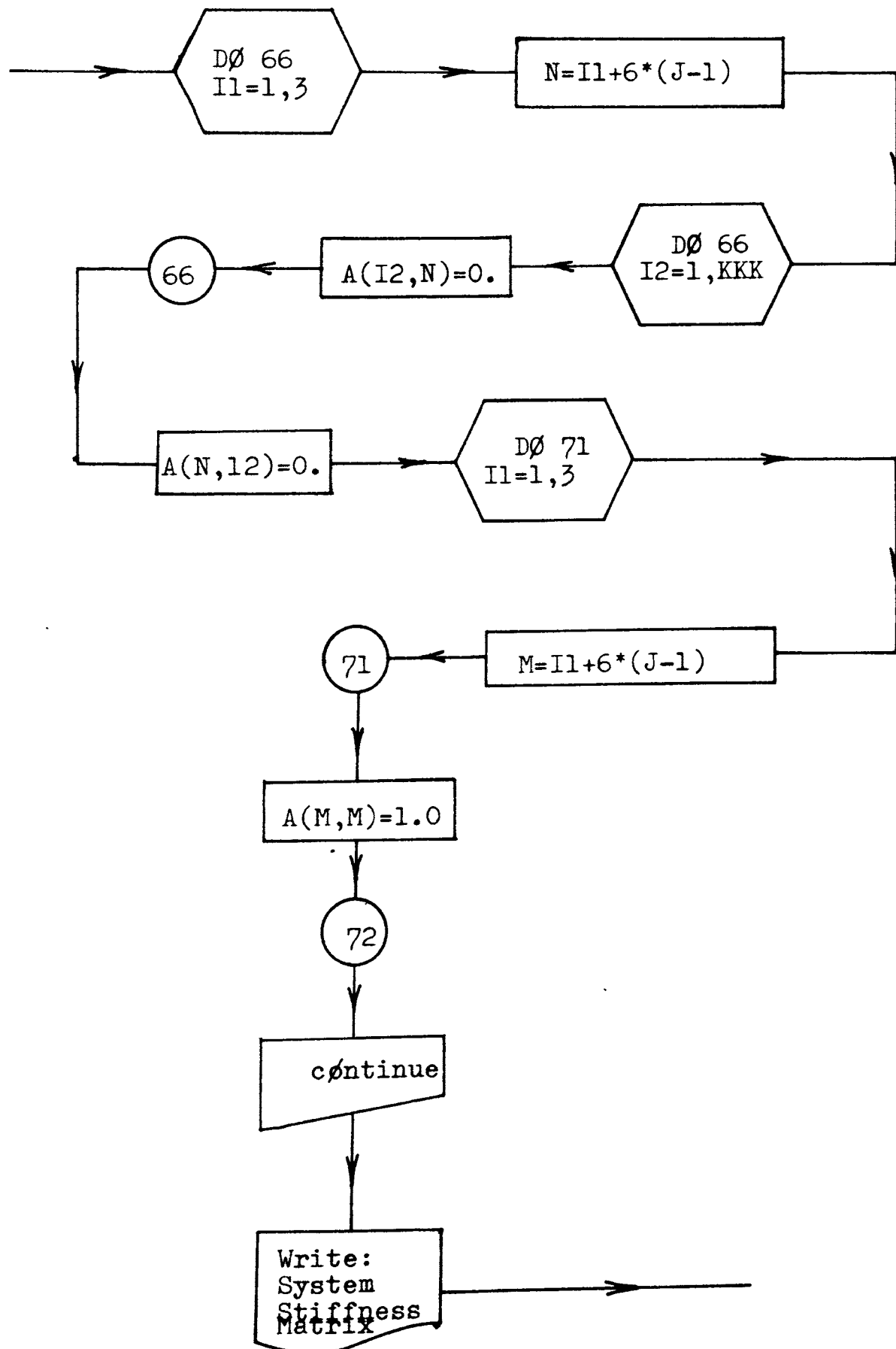




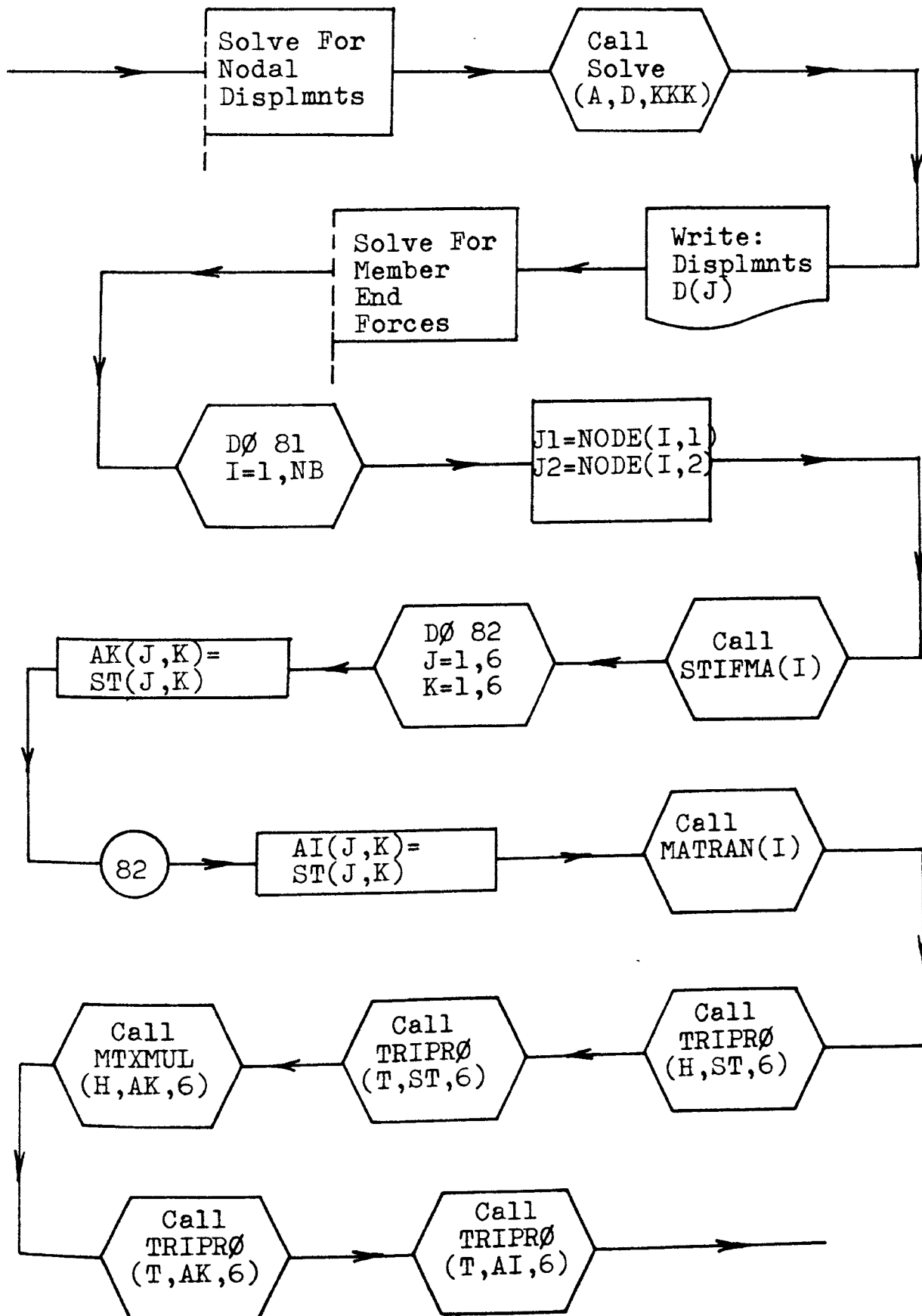


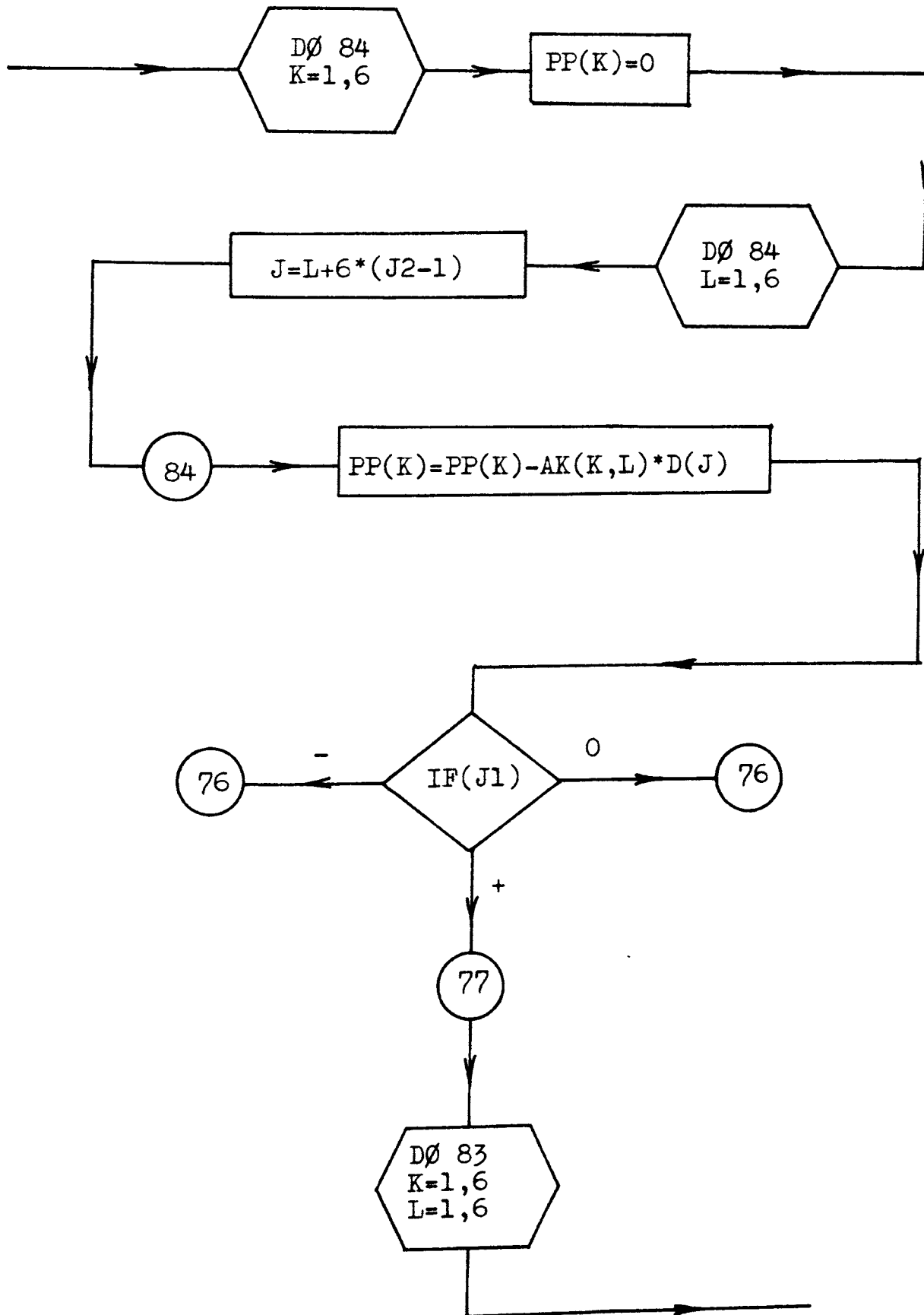


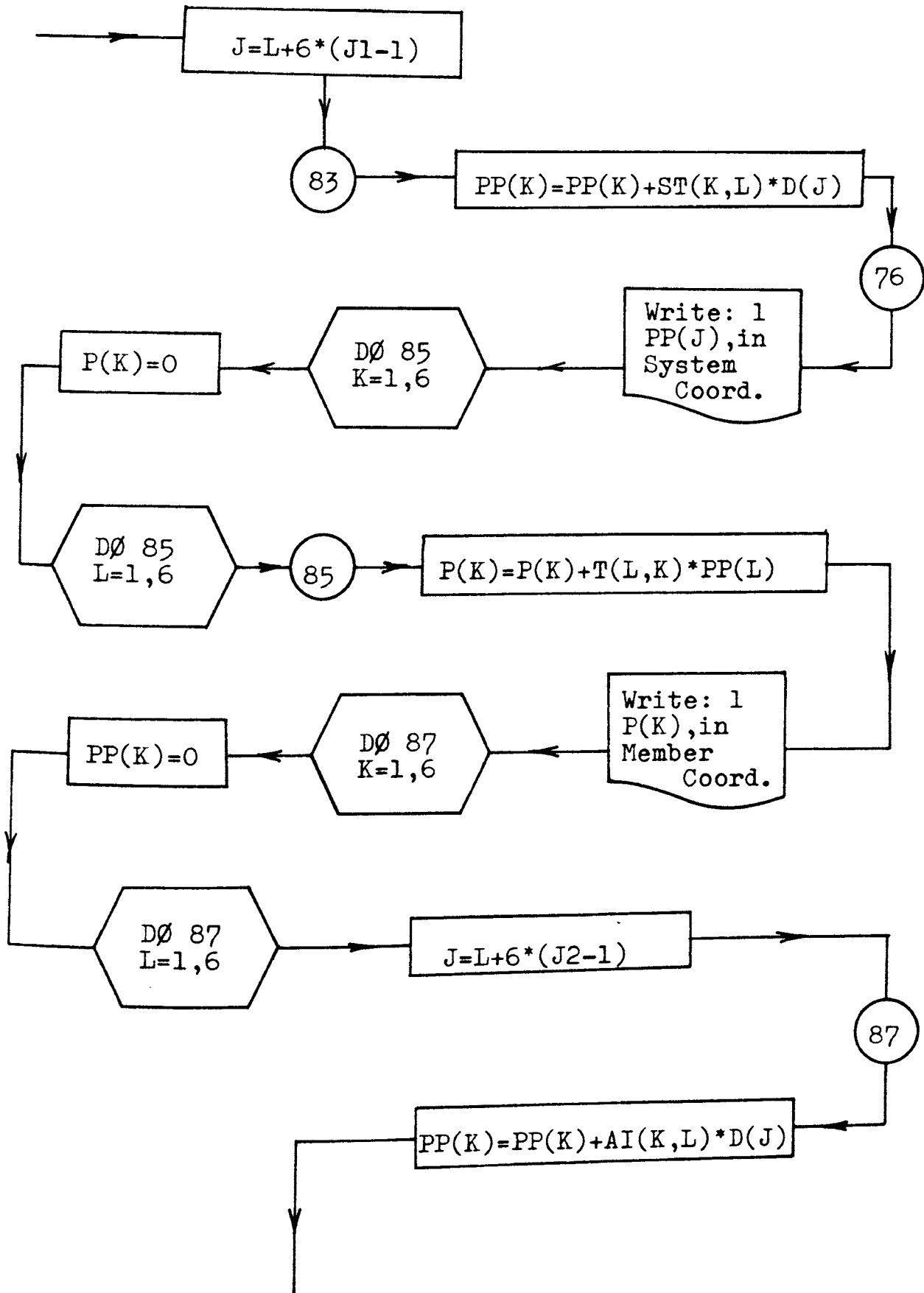


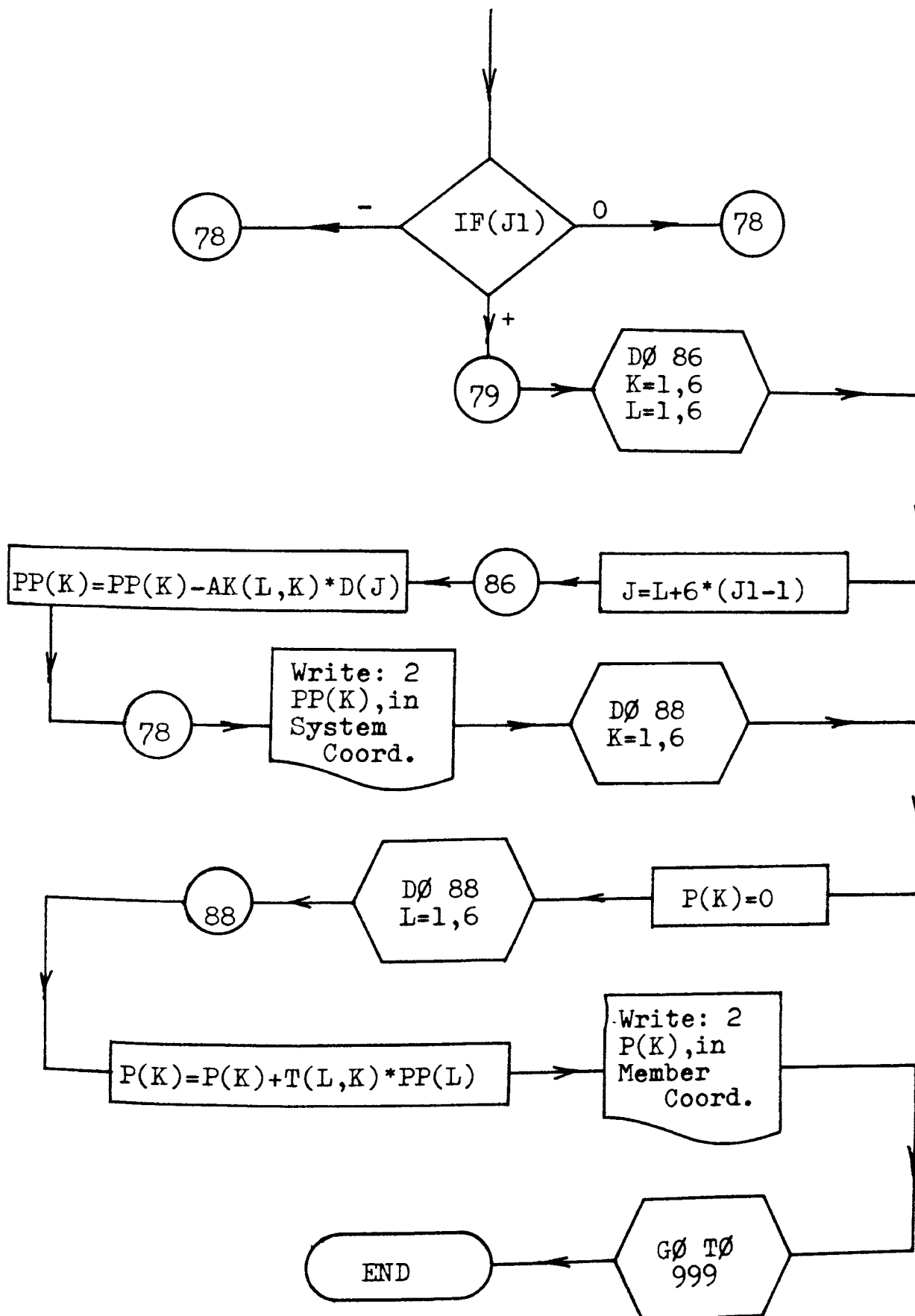




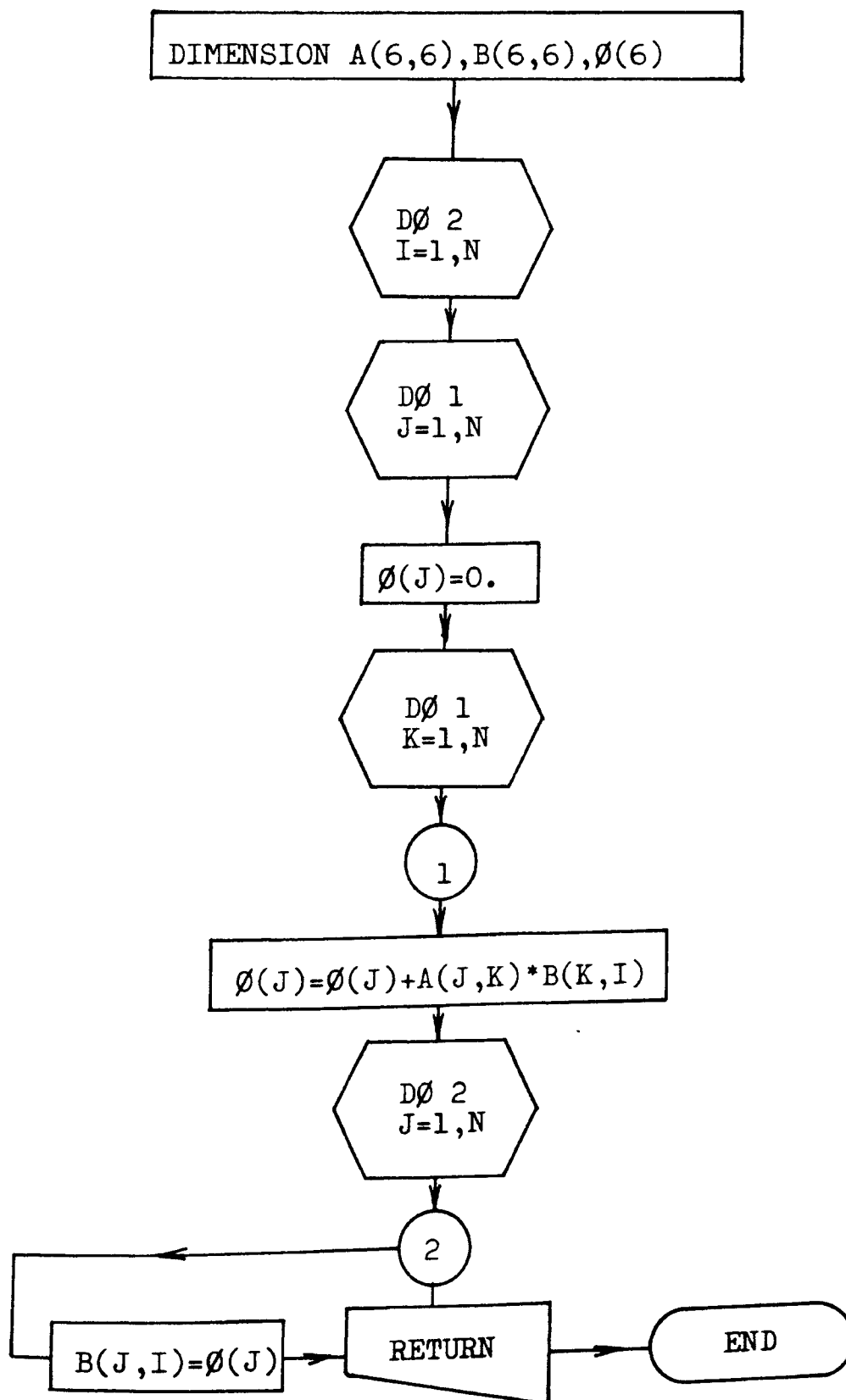




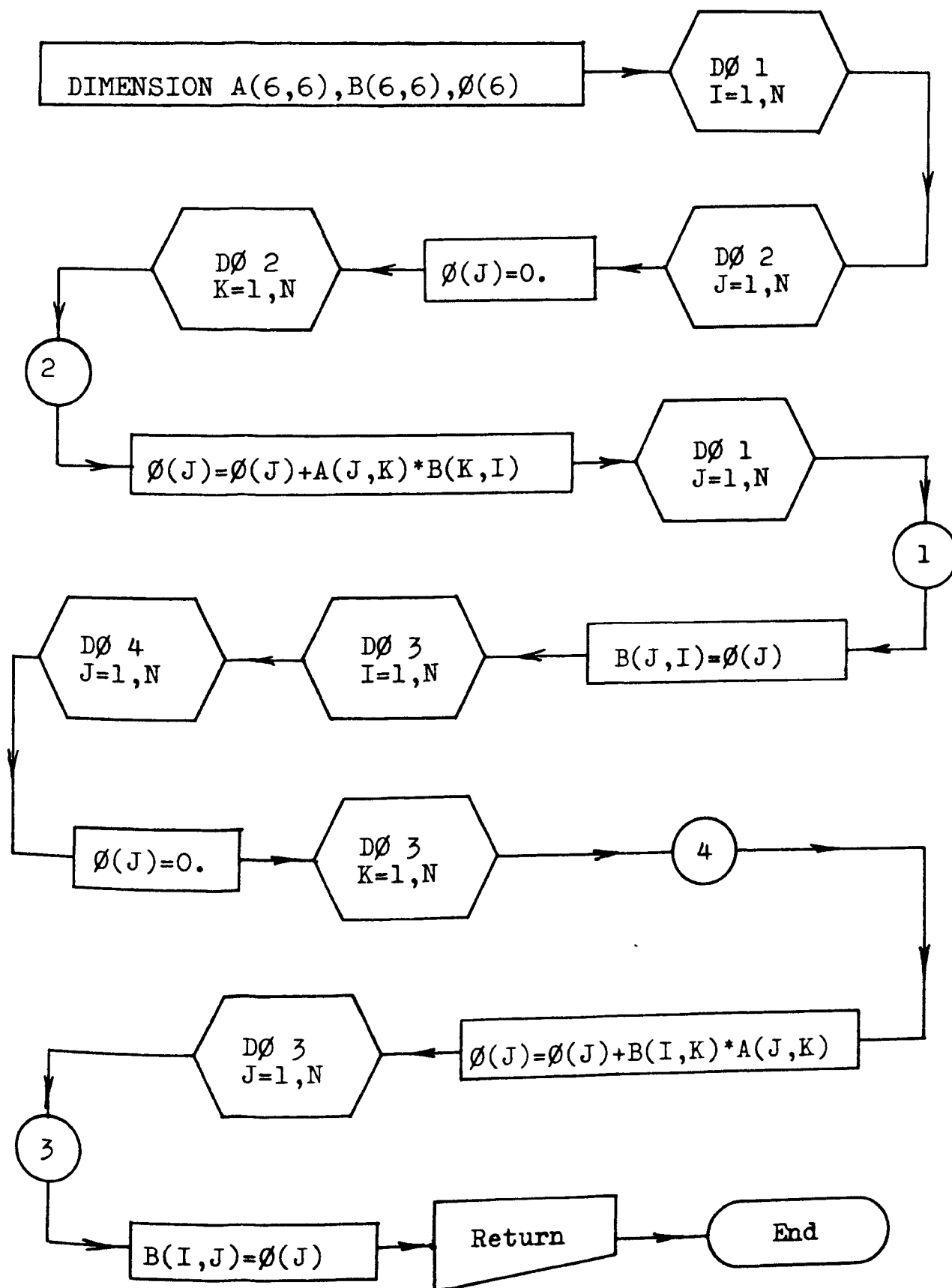




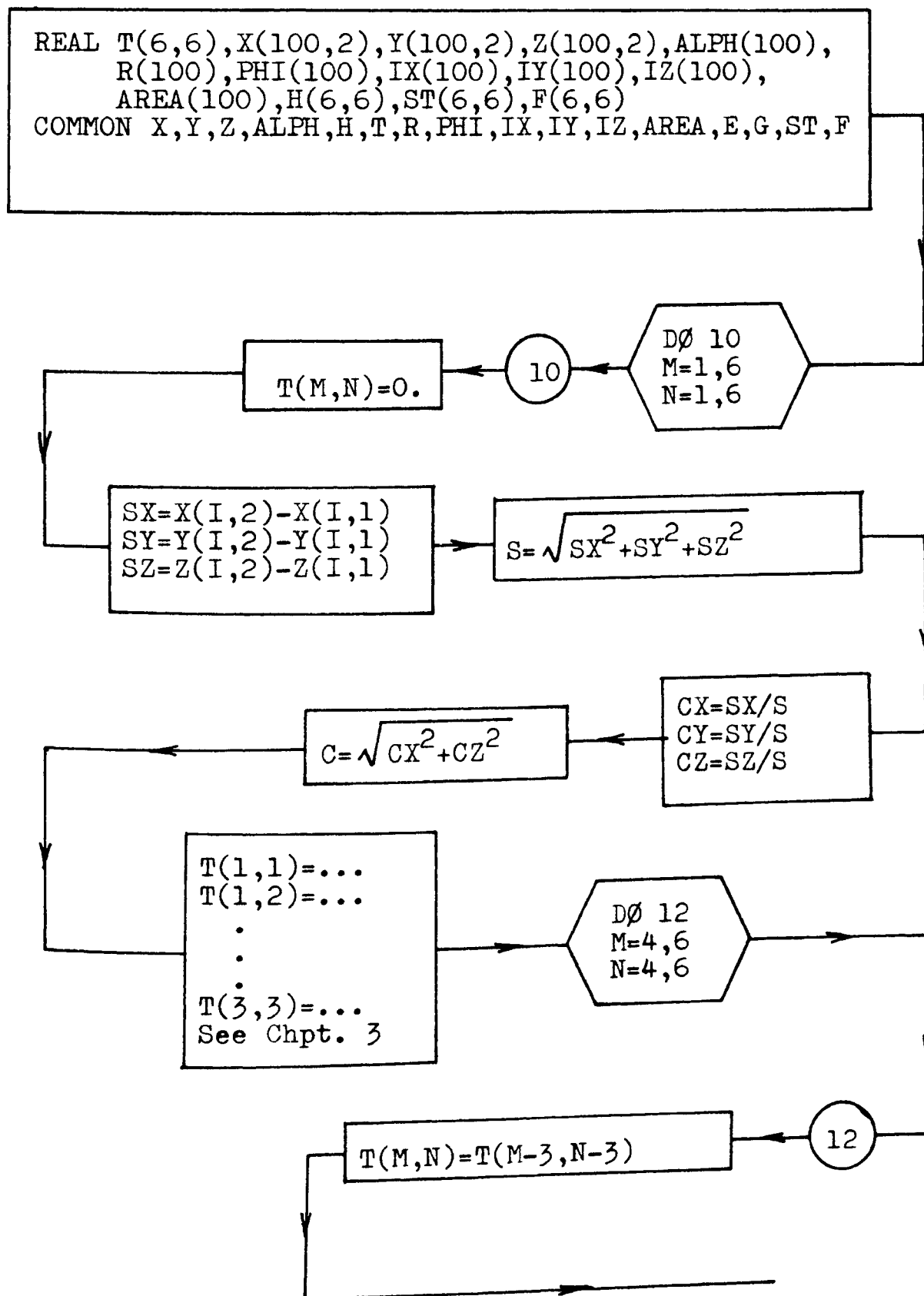
## c.) Subroutine MTXMUL Flow Diagram

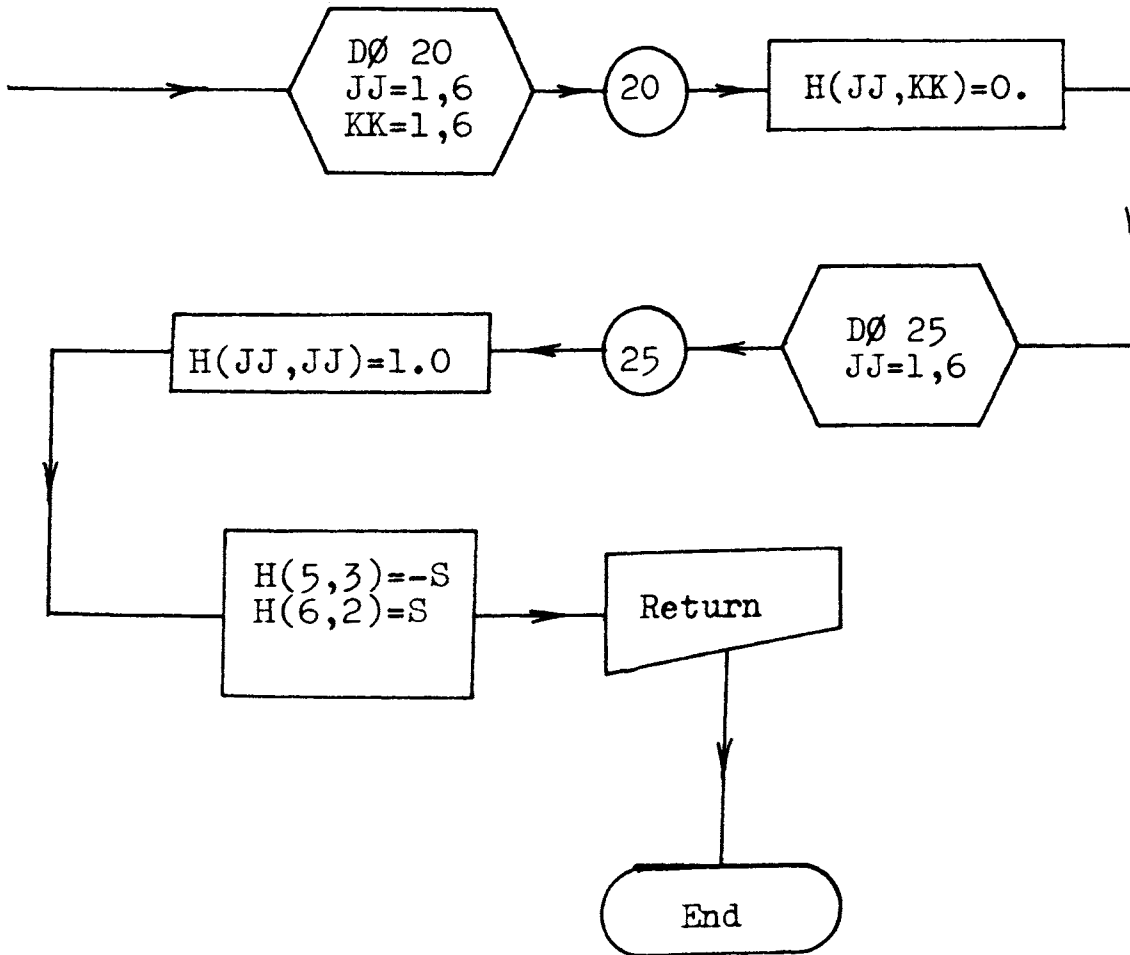


## d.) Subroutine TRIPRØ Flow Diagram



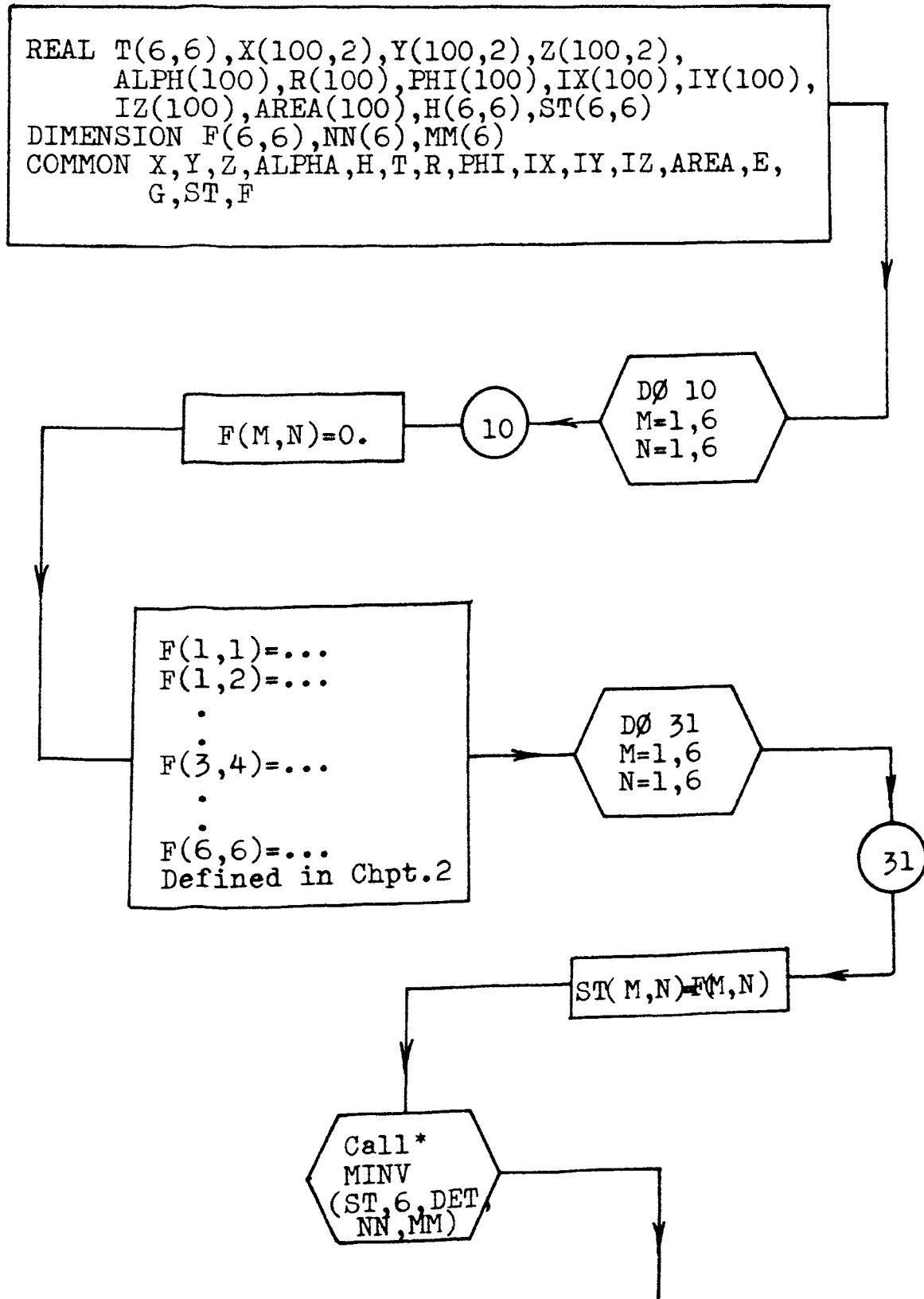
## e.) Subroutine MATRAN (I)



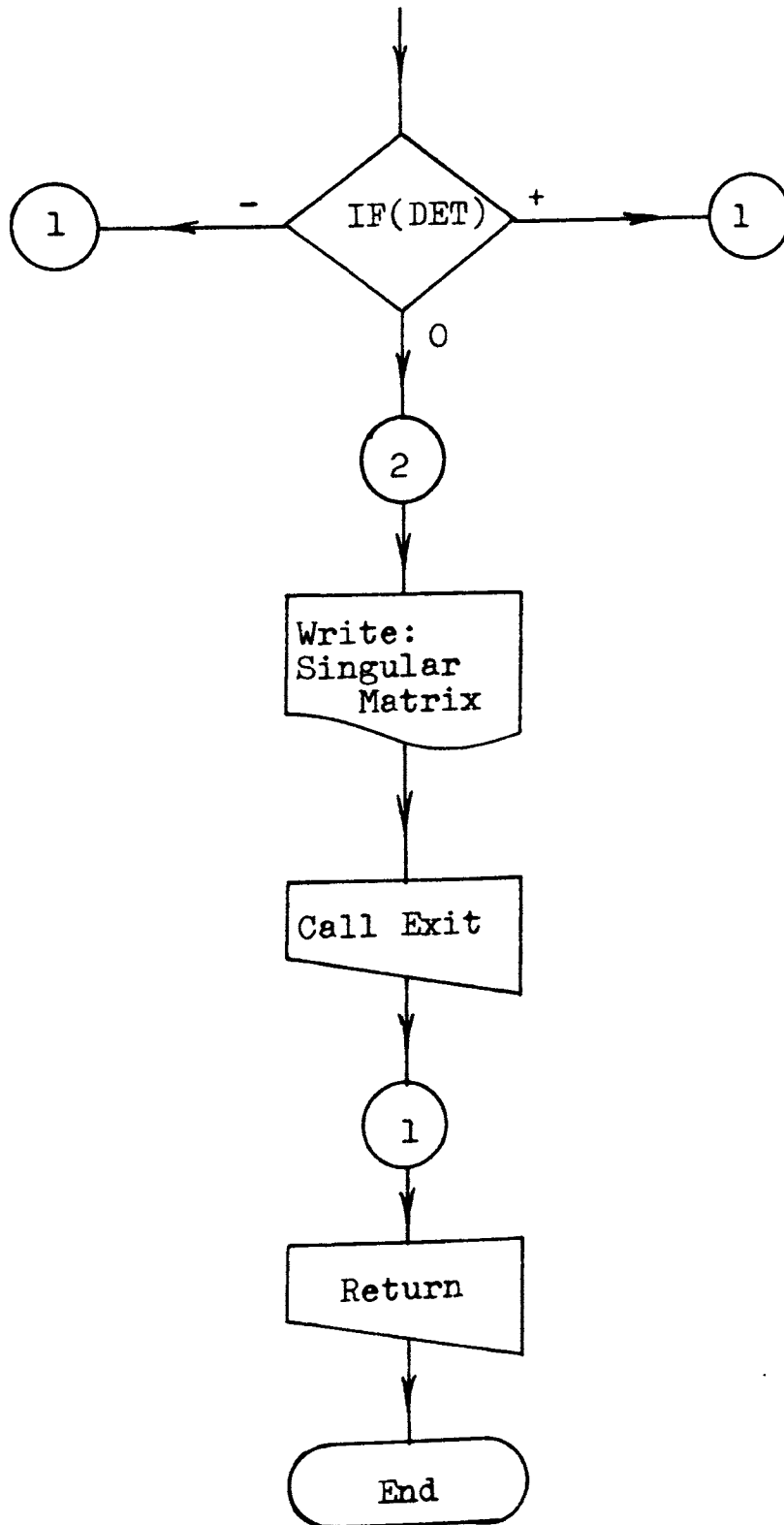




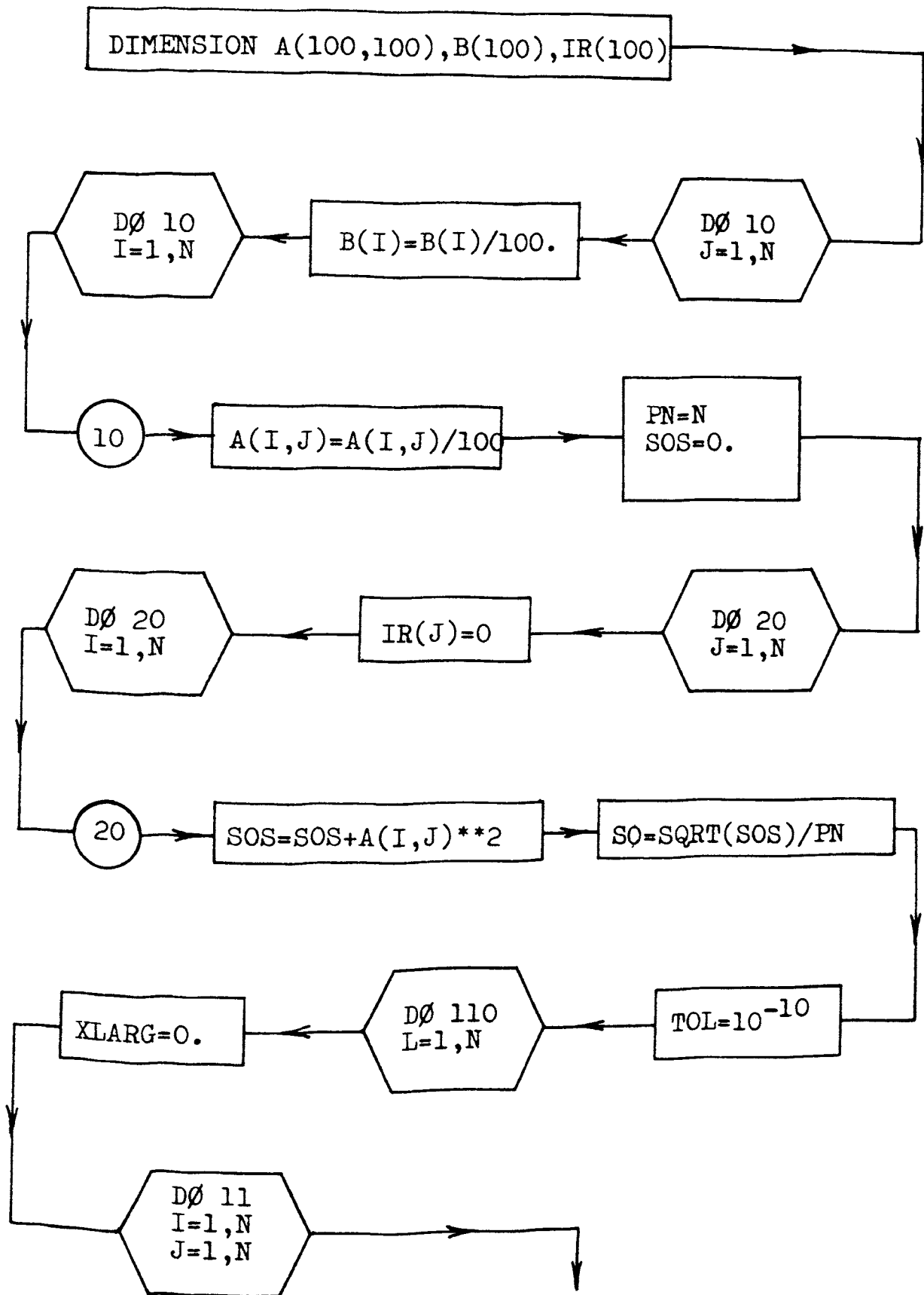
## f.) Subroutine STIFMA (I)

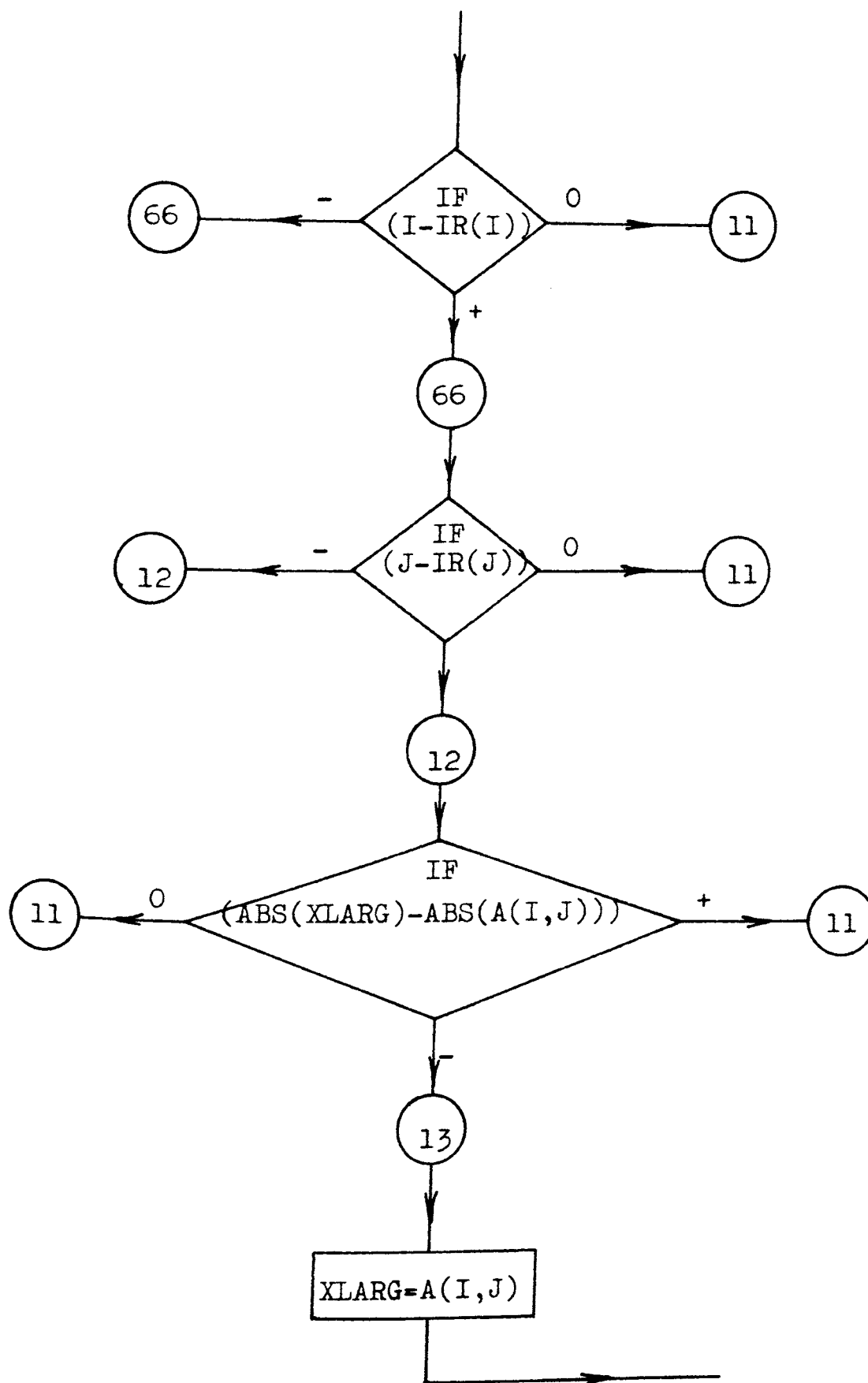


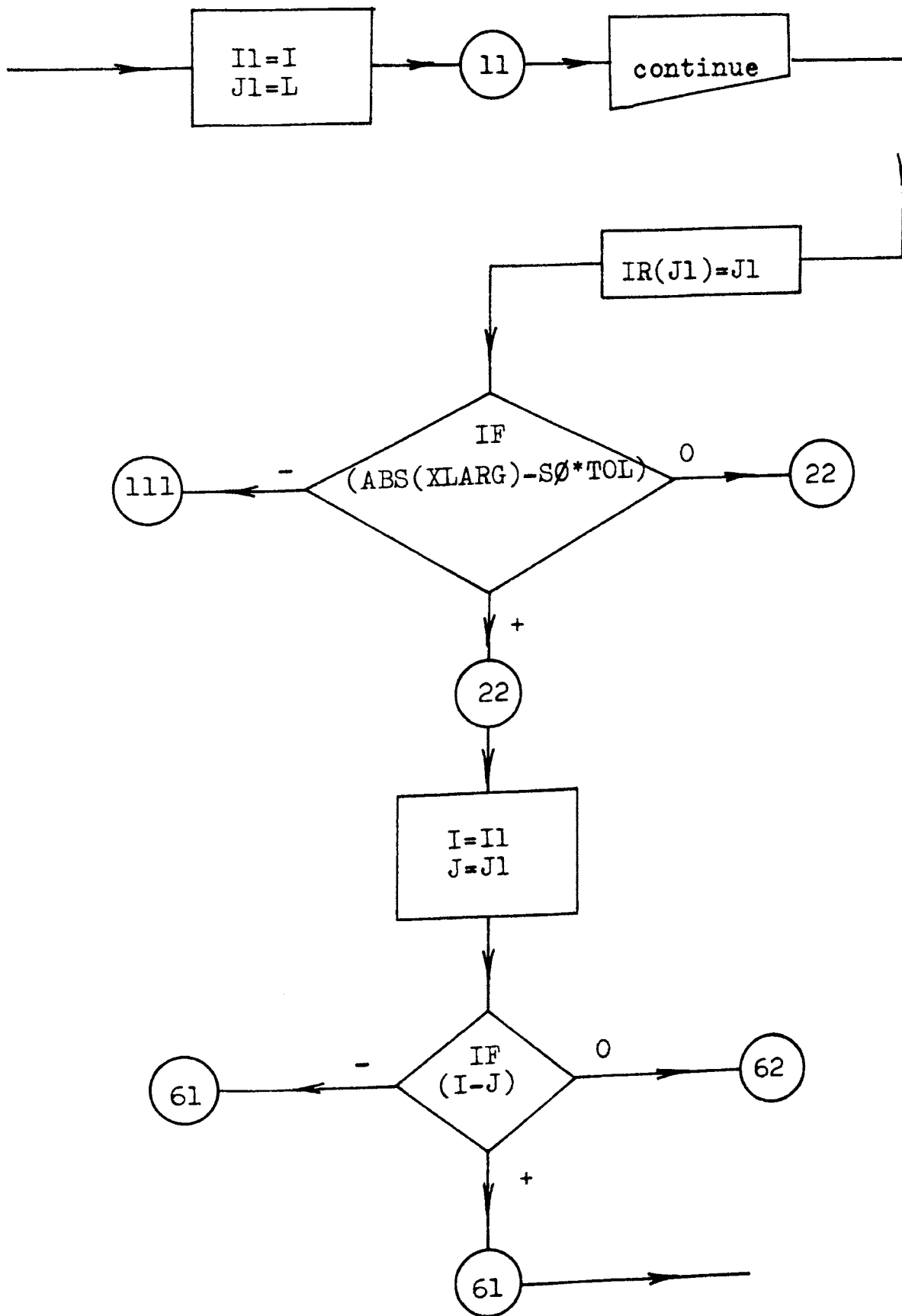
\*Machine loaded subroutine

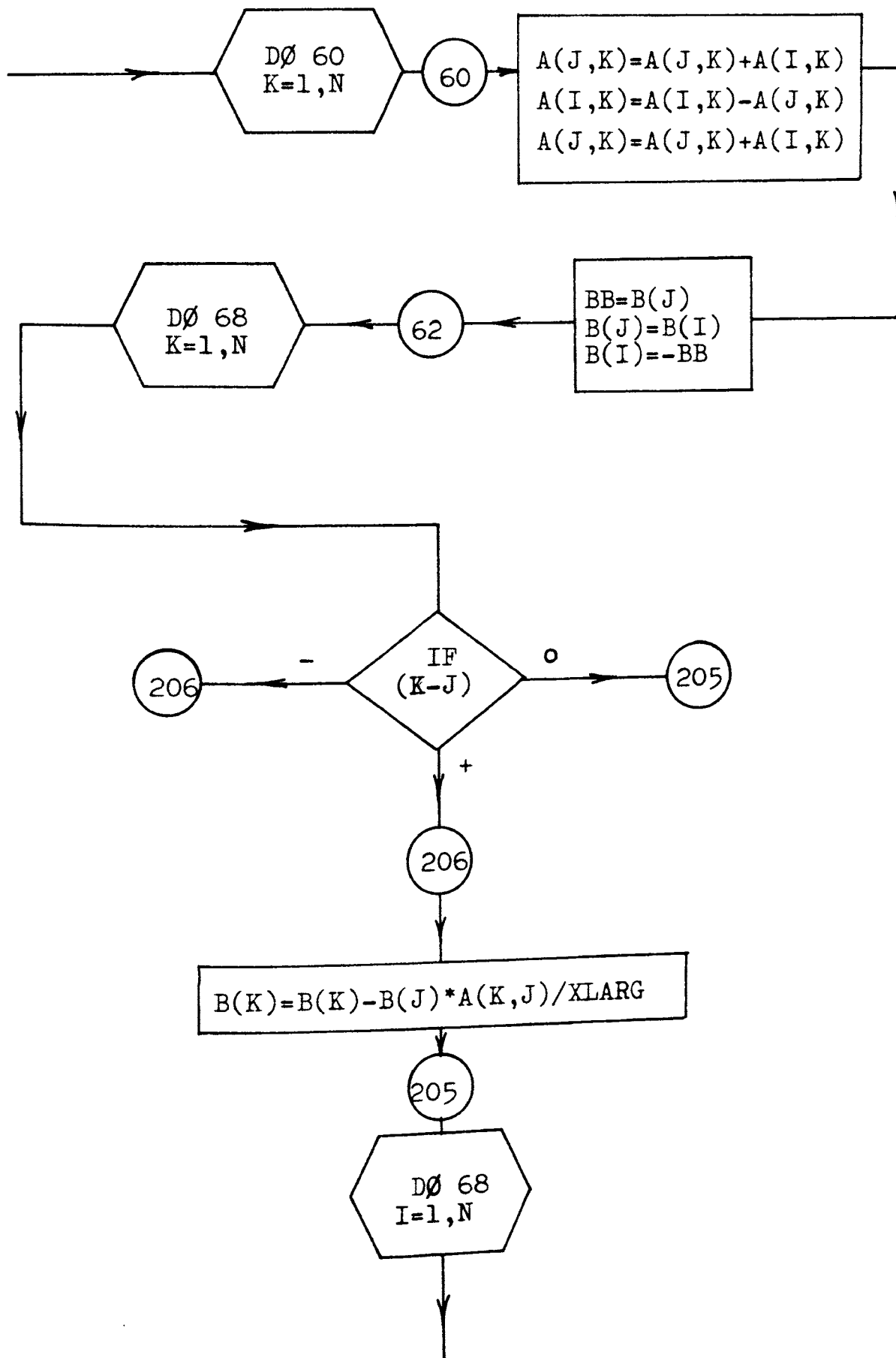


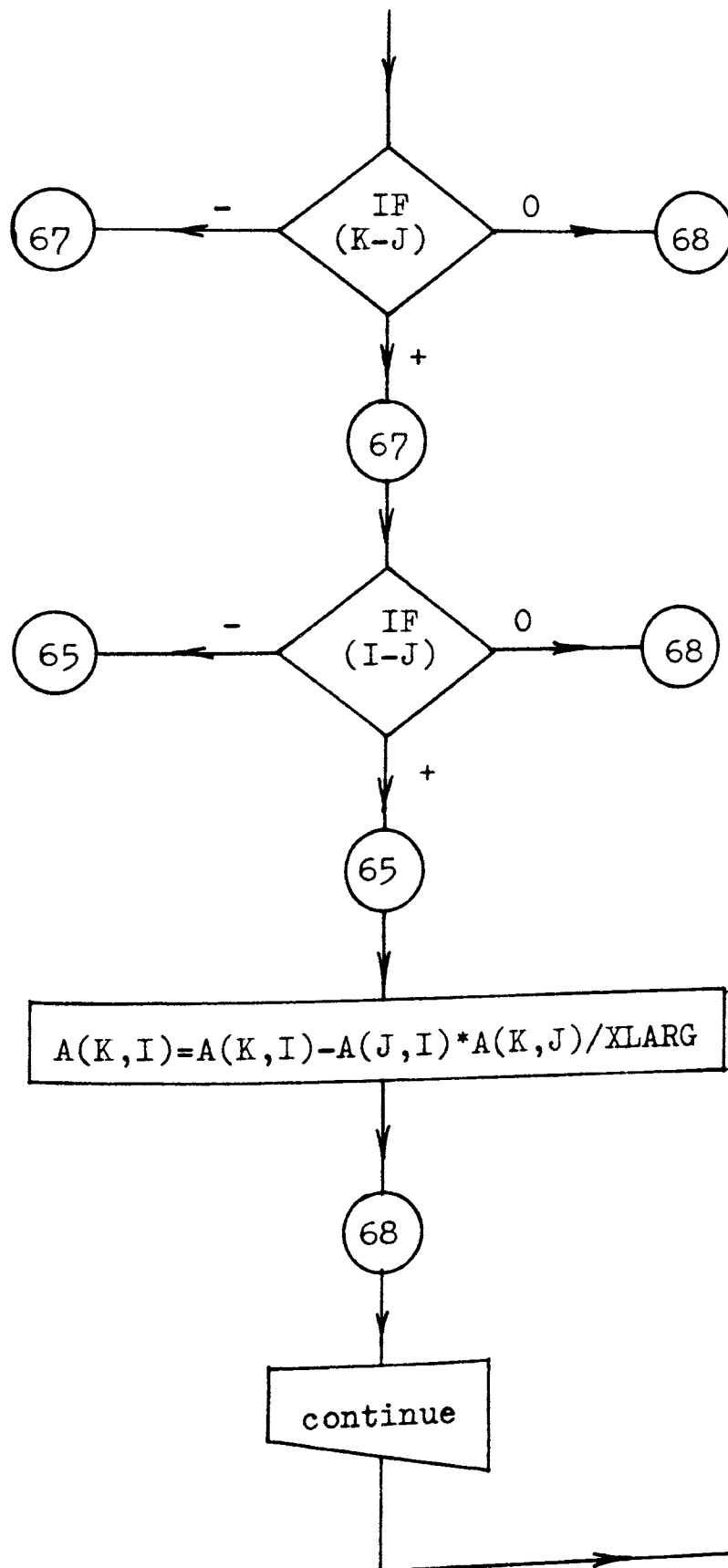
g.) Subroutine SOLVE Flow Diagram



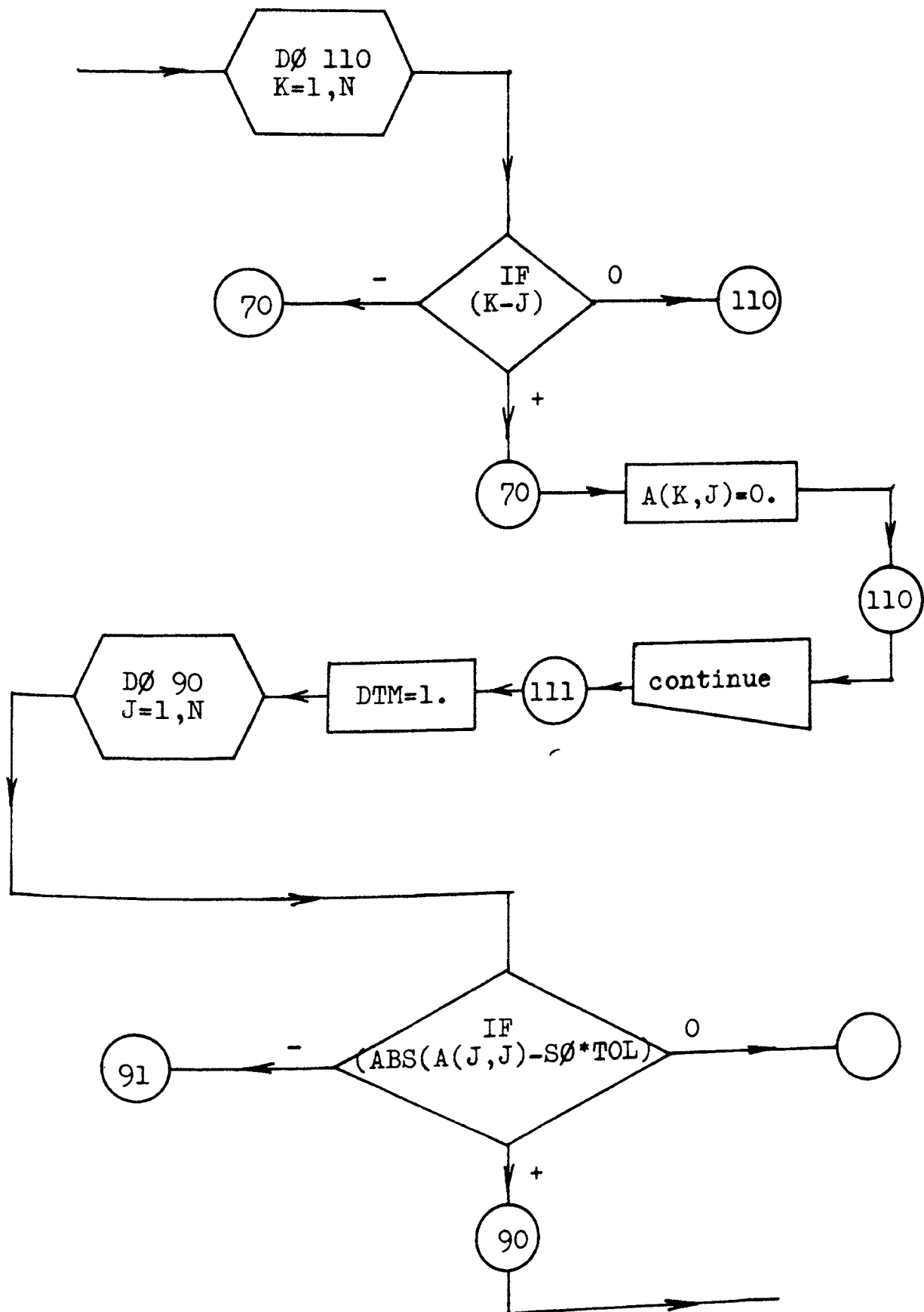








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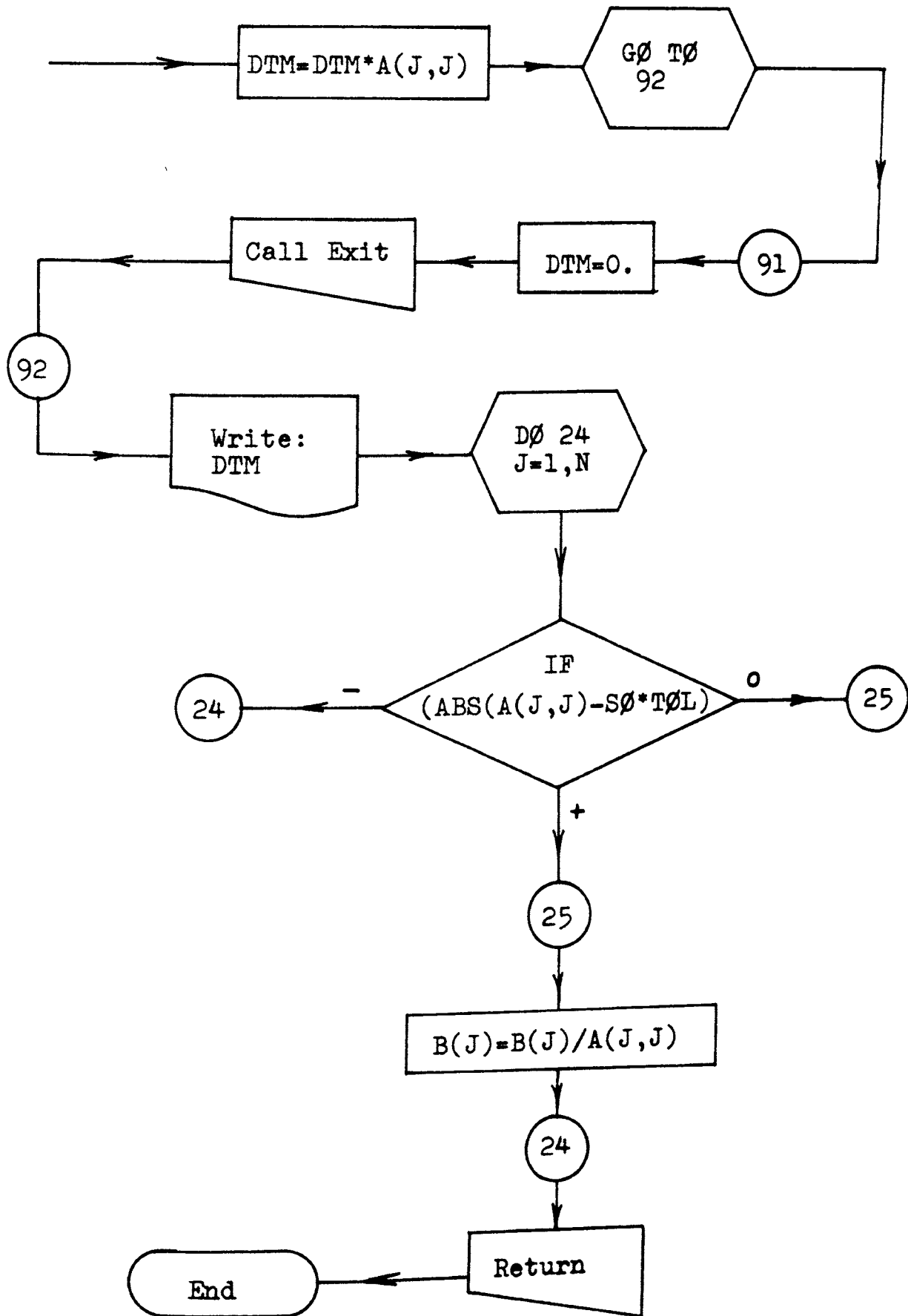


Table 5.1 Table of Symbols

<u>Program Symbol</u>	<u>Problem Symbol</u>	<u>Definition</u>
<u>Main Program</u>		
NN		Number of nodes
NB		Number of members
LL		Maximum number of members per node
NLC		Number of the loading condition
NODE		Name for node numbers
X, Y, Z		Geometric Coordinates
ALPH		Rotation of members about member x-axis
IX, IY, IZ	$I_N, I_{YN}, I_Z$	Principle moments of inertia
AREA	A	Cross-Sectional Area of member
R	R	Radius of curvature of member
PHI	$\emptyset$	One-half central angle of member
E	E	Modulus of elasticity

## Table of Symbols (continued)

<u>Program Symbol</u>	<u>Problem Symbol</u>	<u>Definition</u>
<u>Main Program</u>		
G	G	Shear modulus
KKK		Dimension of system matrix
D	,P	Used both as a label for nodal displacements and applied nodal loads
P	P	Member and forces and moments in member coordinates
PP	P'	Member end forces and moments in system coordinates
NPPN		Counter used to count number of members attached to a node
J1		Node number at end-1 of a member
J2		Node number at end-2 of a member
L1		The L <sup>th</sup> member attached to node
LINK		Name of member to node reference

## Table of Symbols (continued)

<u>Program Symbol</u>	<u>Problem Symbol</u>	<u>Definition</u>
<u>Main Program</u>		
A	$K''$	System stiffness matrix
ST,AK	$K=F^{-1}$	Member stiffness matrix in member coordinates
F	F	Member flexibility matrix
T	T	Transformation matrix
H	H	Equilibrium matrix
NJ		Member and reference
PP		Member end forces in system coordinates
P		Member end forces in member coordinates
<u>MTXMUL</u>		
A		A general six by six matrix
B		A general six by six matrix
Ø		An operating vector used to store rows of AB back into the used columns of B

## Table of Symbols (continued)

<u>Program Symbol</u>	<u>Problem Symbol</u>	<u>Definition</u>
<u>TRIPRØ</u>		
A	T	A general six by six transformation matrix
B		A general six by six matrix
Ø		An operating vector used to store rows of $ABA^t$ back into the used columns of B
<u>MATRAN</u>		
SX		The difference in X-coordinates of the 1-end and 2-end of the member
SY		The difference in Y-coordinates of the 1-end and 2-end of the member
SZ		The difference in Z-coordinates of the 1-end and 2-end of the member
S		Square root of the sum of the squares of SX, SY and SZ
CX	$a_x$	CX, CY, CZ are the direction cosines of the member X-axis
CY	$a_y$	
CZ	$a_z$	

## Table of Symbols (continued)

<u>Program Symbol</u>	<u>Problem Symbol</u>	<u>Definition</u>
C		Square root of the sum of the squares of CX and CZ
T	T	Transformation matrix
H	H	Equilibrium matrix
<u>SOLVE</u>		
A	K''	System stiffness matrix
B	P'	Applied nodal loads
IR		Operating vector
N		Size of system stiffness matrix which equals six times the number of nodes
SOS		Sum of the squares of the elements in the system stiffness matrix
SO		Square root of the sum of the squares.
TOL		Tolerance
XLARG		Largest element in system matrix at beginning of each pivot
DTM		Determinant

```

//DS JOB OFX039, FENTON D L          * 07/08/67 FORT 0003 0040 0000      15235-
REAL T(6,6), X(100,2), Y(100,2), Z(100,2), ALPH(100), P(100), PHI(100),
* IY(100), IZ(100), AREA(100), H(6,5), ST(6,6), IX(100), F(6,6)
REAL AK(6,6), D(100), A(100,100), AI(6,6), PP(6), P(6)
INTEGER NODE(100,2), NRPN(25), LINK(25,6)
COMMON X, Y, Z, ALPH, H, T, P, PHI, IX, IY, IZ, AREA, F, G, ST, F
0000 READ(1,100)NR,NN,LL,NLC
100  FORMAT(5I10)
    WRITE(3,201)NN,NR,NLC,LL
201  FORMAT(11 NUMBER OF NODES'I3/' NUMBER OF MEMBERS'I3/' NUMBER OF
*LOADING CONDITION' I3/' MAXIMUM NUMBER OF MEMBERS PER NODE'I3)
    READ(1,200)(NODE(I,1),I=1,NR),(NODE(I,2),I=1,NR)
    DO 3 I=1,NR
      3 READ(1,777)(X(I,J),Y(I,J),Z(I,J),J=1,2),ALPH(I)
    DO 4 I=1,NR
      4 READ(1,777)IX(I),IY(I),IZ(I),AREA(I),P(I),PHI(I)
      READ(1,777)F,G
777  FORMAT(7F10.2)
    KKK=6*NN
    READ(1,241)(D(IKK),IKK=1,KKK)
241  FORMAT(6F12.4)
    WRITE(3,202)
202  FORMAT(//45X'MEMBER COORDINANTS'/5X'MEMBER-I'4X'X(I,1)'9X'X(I,2)'
*9X'Y(I,1)'9X'Y(I,2)'9X'Z(I,1)'9X'Z(I,2)'/)
203  FORMAT(110,7F15.6)
    DO 204 I=1,NR
204  WRITE(3,203)I,X(I,1),X(I,2),Y(I,1),Y(I,2),Z(I,1),Z(I,2)
    WRITE(3,205)
205  FORMAT(11'54X'MEMBER PROPERTIES'/5X'MEMBER-I'8X'IX'13X'IY'13X'IZ'
*11X'AREA'10X'RADII'S'12X'PHI'11X'ALPHA'/)
    DO 206 I=1,NR
206  WRITE(3,203)I,IX(I),IY(I),IZ(I),AREA(I),P(I),PHI(I),ALPH(I)
    WRITE(3,207)
207  FORMAT(11'46X'APPLIED NODAL LOADS'/6X'NODE-I'7X'PX'12X'PY'13X'PZ'13X
*13X'MY'13X'MZ'/)
    DO 208 I=1,NN
      K=6*I
208  WRITE(3,203)I,D(K-5),D(K-4),D(K-3),D(K-2),D(K-1),D(K)
    WRITE(3,300)
    WRITE(3,500)
    DO 8 J=1,NN
      DO 9 L=1,LL
      8 LINK(L,1)=0
      DO 9 J=1,NN
      9 NRPN(J)=0
      DO 10 I=1,NR
10  WRITE(3,600)I,NODE(I,1),NODE(I,2)
    THE FOLLOWING 16 STATEMENTS DETERMINE WHICH MEMBERS IN A
    STRUCTURAL NETWORK ARE INCIDENT ON A GIVEN NODE. INPUT,
    NODE(I,1)=J1 AND NODE(I,2)=J2, WHERE J1 AND J2 ARE THE NODES
    AT THE NEAR AND FAR ENDS OF THE MEMBER RESPECTIVELY.
    DO 20 I=1,NR
      J1=NODE(I,1)
      IF(J1)12,12,11

```

```

11 NRODN(J1)=NRODN(J1)+1
   I1=NRODN(J1)
   I LINK(J1,I1)-I
12 J2=NRODN(I,2)
   IF(I,J)20,20,13
13 NRODN(J2)=NRODN(J2)+1
   I2=NRODN(J2)
   I LINK(J2,I2)-I
20 CONTINUE
   WRITE(3,330)
   WRITE(3,340)
   WRITE(3,350)
   DO 30 J=1,MM
30 WRITE(3,300)(I,NRODN(I),I LINK(I,I),I=1,4)
203 FORMAT(12I6)
300 FORMAT(14I1)
500 FORMAT(5Y,10MEMBR,2Y,10NODE(I,1)*,2Y,10NODE(I,2)*//)
500 FORMAT(5Y,13,4Y,13,8Y,13)
700 FORMAT(5Y,10NODE(I,2Y,10NODE(I,14Y,11INKI)
900 FORMAT(22Y,11,5Y,12,5Y,13,5Y,14,5Y,15,5Y,16)//)
203 FORMAT(5Y,13,2Y,13,4Y,13,3I6)
C CLEAR SYSTEM STIFFNESS (A) STORAGE LOCATIONS
DO 60 J=1,KKK
DO 60 I=1,KKK
62 A(I,I)=0.
C ****GENERATE SYSTEM STIFFNESS MATRIX****
DO 60 J=1,MM
DO 61 I=1,I1
I=IABS(I LINK(I,I))
IF(I)40,40,60
C COMPUTE MEMBER(I) STIFFNESS MATRIX
60 CALL STEEMAT(I)
WRITE(3,370)I
203 FORMAT(11I10Y,1MEMBR,13//50Y*1E1X*1011TY MATRIX//)
DO 66 M1=1,6
DO 67 I=1,440)IF(M1,M1,M=1,6)
WRITE(3,210)
210 FORMAT(//50Y*1E1X*1011TY MATRIX//)
WRITE(3,440)((ST(K1,K2),K2=1,6),K1=1,6)
440 FORMAT(6F10,8)
DO 21 M1=1,6
DO 21 M2=1,6
21 AK(M1,M2)=ST(M),M21
C COMPUTE MEMBER(I) AXIS ROTATION TRANSFORMATION MATRIX
CALL MATRAN(I)
WRITE(3,211)
211 FORMAT(//50Y*1E1X*1011TY MATRIX//)
DO 15 M=1,6
15 WRITE(3,440)(T(M,M),M=1,6)
C WHICH END OF MEMBER(I) IS AT NODE(I)
IF(I LINK(I,I))40,50,50
IF END=2 OF MEMBER(I) IS AT NODE(I) COMPUTE, K*(2,2)= T*K(2,2)*T'
C DIRECT STIFFNESS AND ADDS IT TO THE APPROPRIATE 6X6 LOCATION ON
C THE SYSTEM MATRIX DIAGONAL.

```



```

40 CALL TRIPRO(T,AK,6)
   NJ=1
   DO 41 K=1,6
     N=K+6*(J-1)
     DO 41 K1=1,6
       M=K1+6*(J-1)
41 A(N,M)=AK(K,K1)+A(N,M)
   GO TO 52
C   IF END-1 OF MEMBER(I) IS AT NODE(J) COMPUTE K'(1,1)=THKHIT' AND
C   ADDS IT TO THE APPROPRIATE 6X6 LOCATION ON THE SYSTEM MATRIX
C   DIAGONAL.
50 CALL TRIPRO(H,AK,6)
   CALL TRIPRO(T,AK,6)
   NJ=2
   JJ=J
   DO 51 K=1,6
     N=K+6*(J-1)
     DO 51 K1=1,6
       M=K1+6*(JJ-1)
51 A(N,M)=AK(K,K1)+A(N,M)
C   THE FOLLOWING 16 STATEMENTS LOCATE THE END OF MEMBER(I) NOT
C   INCIDENT ON NODE(J) AND COMPUTE THE APPROPRIATE OFF DIAGONAL 6X6
C   STIFFNESS MATRIX.
52 IF(NODE(I,NJ))53,61,53
53 JJ=NODE(I,NJ)
   CALL MTXMUL(H,ST,6)
   CALL TRIPRO(T,ST,6)
   GO TO(64,63),NJ
C   COMPUTE -THKT'
63 DO 54 K=1,6
   N=K+6*(J-1)
   DO 54 K1=1,6
     M=K1+6*(JJ-1)
54 A(N,M)=-ST(K,K1)
   GO TO 61
C   COMPUTE -TKHIT'
64 DO 65 K=1,6
   N=K+6*(J-1)
   DO 65 K1=1,6
     M=K1+6*(JJ-1)
65 A(N,M)=-ST(K1,K)
61 CONTINUE
60 CONTINUE
   DO 72 J=1,NN
     IF(I INK(J,2))72,67,72
67 DO 66 I1=1,3
     N=I1+6*(J-1)
     DO 66 I2=1,3
70 A(I2,N)=0.
66 A(N,I2)=0.
     DO 71 I1=1,3
       M=I1+6*(J-1)
71 A(M,M)=1.0
72 CONTINUE

```

```

WRITE(3,212)
212 FORMAT('1137X SYSTEM STIFFNESS MATRIX'//)
DO 213 I=1,KKK
213 WRITE(3,214)I,(A(I,J),J=1,KKK)
214 FORMAT(/'  ROW'15/(6E17.7))
C   ** SOLVE FOR NODAL DISPLACEMENTS **
CALL SOLVE(A,D,KKK)
WRITE(3,216)
216 FORMAT(40X'NODAL DISPLACEMENTS'//6X'NODE'17X'DELTA-X'18X'DELTA-Y'18X
*10X'DELTA-Z'18X'THETA-X'18X'THETA-Y'18X'THETA-Z'//)
DO 217 I=1,NN
J=6*I
217 WRITE(3,203)I,D(J-5),D(J-4),D(J-3),D(J-2),D(J-1),D(J)
C   THE REMAINDER OF THE PROGRAM COMPUTES THE MEMBER END REACTIONS
C   FIRST IN SYSTEM COORDINATES AND THEN TRANSFORMS THEM TO MEMBER
C   COORDINATES.
WRITE(3,218)
218 FORMAT('1137X MEMBER END FORCES AND MOMENTS'//34X'DPX'12X'DPY'12X
*10D7112X'DPZ'12X'DMX'12X'DMY'12X'DMZ'// MEMB. END'//20X'DX'13X'DY'13X'DZ'
*13X'MX'13X'MY'13X'MZ'//)
DO 81 I=1,NB
J1=NODE(I,1)
J2=NODE(I,2)
CALL STIFMA(I)
DO 82 J=1,6
DO 82 K=1,6
AK(J,K)=ST(J,K)
82 AT(J,K)=ST(J,K)
CALL MATRAN(I)
CALL TRIPRO(H,ST,6)
CALL TRIPRO(T,ST,6)
CALL MTXMUL(H,AK,6)
CALL TRIPRO(T,AK,6)
CALL TOTPRO(T,AT,6)
DO 84 K=1,6
DP(K)=0.
DO 84 L=1,6
J=1+6*(J2-1)
84 DP(K)=DP(K)-AK(K,L)*D(J)
IF(J)76,76,77
77 DO 83 K=1,6
DO 83 L=1,6
J=1+6*(J1-1)
83 DP(K)=DP(K)+ST(K,L)*D(J)
76 WRITE(3,186)(DP(J),J=1,6)
186 FORMAT(10X2H11',14X6F15.6/)
C   FORCES AT END-1 OF MEMBER (I) IN MEMBER COORDINATES.
DO 85 K=1,6
P(K)=0.
DO 85 L=1,6
85 P(K)=P(K)+T(L,K)*DP(L)
WRITE(3,187)(P(J),J=1,6),T
187 FORMAT(10X'116F15.6/15)
DO 87 K=1,6

```

```

      PP(K)=0.
      DO 87 L=1,6
        J=L+6*(J2-1)
      87 PP(K)=PP(K)+A1(K,L)*D(J)
        IF(J)78,78,79
      79 DO 86 K=1,6
        DO 86 L=1,6
          J=L+6*(J1-1)
      86 PP(K)=PP(K)-AK(L,K)*D(J)
      78 WRITE(3,190)(PP(J),J=1,6)
      190 FORMAT(10X2H2',14X6F15.6/)
C     FORCES AT END-2 OF MEMBER(I) IN MEMBER COORDINATES.
      DO 88 K=1,6
        P(K)=0.
        DO 88 L=1,6
      88 P(K)=P(K)+T(L,K)*PP(L)
      81 WRITE(3,188)(P(J),J=1,6)
      188 FORMAT(10X'2'6F15.6//)
      GO TO 9999
      END
      SUBROUTINE MTXMUL(A,R,N)
      DIMENSION A(6,6),R(6,6),D(6)
      DO 2 I=1,N
        DO 1 J=1,N
          D(J)=0.
          DO 1 K=1,N
      1 D(J)=D(J)+A(J,K)*R(K,I)
          DO 2 J=1,N
      2 R(J,I)=D(J)
        RETURN
      END
      SUBROUTINE TRIPRO(A,R,N)
      DIMENSION A(6,6),R(6,6),D(6)
      DO 1 I=1,N
        DO 2 J=1,N
          D(J)=0.
          DO 2 K=1,N
      2 D(J)=D(J)+A(J,K)*R(K,I)
          DO 1 J=1,N
      1 R(J,I)=D(J)
          DO 3 I=1,N
            DO 4 J=1,N
              D(J)=0.
              DO 4 K=1,N
      4 D(J)=D(J)+R(I,K)*A(J,K)
            DO 3 J=1,N
      3 R(I,J)=D(J)
          RETURN
        END
      SUBROUTINE MATPAN(I)
      REAL T(6,6),X(100,2),Y(100,2),Z(100,2),ALPH(100),R(100),PHI(100),
      *IX(100),IZ(100),AREA(100),H(6,6),ST(6,6),IX(100),F(6,6)
      COMMON X,Y,Z,ALPH,H,T,R,PHI,IX,IY,IZ,AREA,E,G,ST,F
      USING THE COORDINATES AT EACH OF A MEMBER THIS PROGRAM COMPUTES

```

```

C THE THREE DIMENSIONAL ROTATION TRANSFORMATION MATRIX WHICH
C TRANSFORMS ELEMENT COORDINATES TO SYSTEM COORDINATES.
DO 10 M=1,6
DO 10 N=1,6
10 T(M,N)=0.0
SX=X(I,2)-X(I,1)
SY=Y(I,2)-Y(I,1)
SZ=Z(I,2)-Z(I,1)
S=SQRT(SX**2+SY**2+SZ**2)
CX=SX/S
CY=SY/S
CZ=SZ/S
C=SQRT(CX**2+CZ**2)
T(1,1)=CX
T(1,2)=(-CX*CY*COS(ALPH(I))-CZ*SIN(ALPH(I)))/C
T(1,3)=(CX*CY*SIN(ALPH(I))-CZ*COS(ALPH(I)))/C
T(2,1)=CY
T(2,2)=C*COS(ALPH(I))
T(2,3)=-C*SIN(ALPH(I))
T(3,1)=CZ
T(3,2)=(-CY*CZ*COS(ALPH(I))+CX*SIN(ALPH(I)))/C
T(3,3)=(CY*CZ*SIN(ALPH(I))+CX*COS(ALPH(I)))/C
DO 12 M=4,6
DO 12 N=4,6
12 T(M,N)=T(M-3,N-3)
DO 20 JJ=1,6
DO 20 KK=1,6
20 H(JJ,KK)=0.0
DO 25 JJ=1,6
25 H(JJ,JJ)=1.0
H(5,3)=-S
H(6,2)=S
RETURN
END
SUBROUTINE STIFMA(I)
REAL T(6,6),X(100,2),Y(100,2),Z(100,2),ALPH(100),P(100),PHI(100),
*TY(100),TZ(100),AREA(100),H(6,6),ST(6,6),IX(100)
DIMENSION F(6,6),NN(6),MM(6)
COMMON X,Y,Z,ALPH,H,T,P,PHI,IX,IY,IZ,AREA,F,G,ST,F
DO 10 M=1,6
DO 10 N=1,6
10 F(M,N)=0.0
F(1,1)=(R(I)/AREA(I)/F)*(PHI(I)+.5*SIN(2.*PHI(I)))
F(1,1)=F(1,1)+(R(I)**3/F/TZ(I))*(PHI(I)-1.5*SIN(2.*PHI(I)))
F(1,1)=F(1,1)+(R(I)**3/F/TZ(I))*(2.*PHI(I)*(COS(PHI(I)))**2)
F(2,2)=(R(I)/AREA(I)/F)*(PHI(I)-.5*SIN(2.*PHI(I)))
F(2,2)=F(2,2)+(R(I)**3/F/TZ(I))*(PHI(I)-.5*SIN(2.*PHI(I)))
F(2,2)=F(2,2)+(R(I)**3/F/TZ(I))*(2.*PHI(I)*(SIN(PHI(I)))**2)
F(3,3)=(R(I)**3/G/IX(I))*(3.*PHI(I)-2.*SIN(2.*PHI(I)))
F(3,3)=F(3,3)+(R(I)**3/G/IX(I))*(.25*SIN(4.*PHI(I)))
F(3,3)=F(3,3)+(R(I)**3/F/IY(I))*(PHI(I)-.25*SIN(4.*PHI(I)))
F(4,4)=(R(I)/G/IX(I))*(PHI(I)+.5*SIN(2.*PHI(I)))
F(4,4)=F(4,4)+(R(I)/F/IY(I))*(PHI(I)-.5*SIN(2.*PHI(I)))
F(5,5)=(R(I)/G/IX(I))*(PHI(I)-.5*SIN(2.*PHI(I)))

```

```

F(5,5)=F(5,5)+(R(I)/E/IV(I))*(PHI(I)+.5*SIN(2.*PHI(I)))
F(6,6)=(R(I)/E/IV(I))*2.*PHI(I)
F(1,2)=(R(I)**3/E/IV(I))*(2.*(SIN(PHI(I)))**2)
F(1,2)=F(1,2)+(R(I)**3/E/IV(I))*(-PHI(I)*SIN(2.*PHI(I)))
F(2,1)=F(1,2)
F(1,6)=(R(I)**2/E/IV(I))*(2.*SIN(PHI(I))-2.*PHI(I)*COS(PHI(I)))
F(6,1)=F(1,6)
F(2,6)=(R(I)**2/E/IV(I))*(2.*PHI(I)*SIN(PHI(I)))
F(6,2)=F(2,6)
F(3,4)=(R(I)**2/G/IX(I))*(PHI(I)*COS(PHI(I))-2.*SIN(PHI(I)))
F(3,4)=F(3,4)+(R(I)**2/G/IX(I))*(.5*COS(PHI(I))*SIN(2*PHI(I)))
F(3,4)=F(3,4)+(R(I)**2/E/IV(I))*(PHI(I)*COS(PHI(I)))
F(3,4)=F(3,4)+(R(I)**2/E/IV(I))*(-.5*COS(PHI(I))*SIN(2.*PHI(I)))
F(4,3)=F(3,4)
F(3,5)=(-R(I)**2/G/IX(I))*(PHI(I)*SIN(PHI(I))-2.*SIN(PHI(I)))
**SIN(2.*PHI(I))
F(3,5)=F(3,5)+(-R(I)**2/E/IV(I))*(PHI(I)*SIN(PHI(I)))
F(3,5)=F(3,5)+(-R(I)**2/E/IV(I))*(.5*SIN(PHI(I))*SIN(2.*PHI(I)))
F(5,3)=F(3,5)
DO 31 M=1,6
DO 31 N=1,6
31 ST(M,N)=F(M,N)
CALL MINV(ST,6,DET,NN,MM)
IF (DET) 1,2,1
2 WRITE(3,550)
550 FORMAT('SINGULAR MATRIX')
CALL EXIT
1 RETURN
END
SUBROUTINE SOLVE(A,R,N)
C MATRIX INVERSION BY JORDAN ELIMINATION
C PERFORMING A COMPLETE PIVOT
DIMENSION A(100,100),R(100),IR(100)
DO 10 I=1,N
R(I)=R(I)/100.
DO 10 J=1,N
10 A(I,J)=A(I,J)/100.
PN=N
SOS=0.
DO 20 J=1,N
IR(J)=0
DO 20 I=1,N
20 SOS=SOS+A(I,J)**2
SQ=SQRT(SOS)/PN
TOL=1.E-10
DO 110 L=1,N
XLARG=0.
DO 11 I=1,N
DO 11 J=1,N
IF (I-IR(I)) 66,11,66
66 IF (J-IR(J)) 12,11,12
12 IF (ABS(XLARG)-ABS(A(I,J))) 13,11,11
13 XLARG=A(I,J)
I=I

```

```

      J1=J
11 CONTINUE
      IP(J1)=J1
      IF (ABS(XLARG)-SQ*TO1)1111,22,22
22 I=I1
      J=J1
      IF (I-J)61,62,61
61 DO 60 K=1,N
      A(J,K)=A(J,K)+A(I,K)
      A(I,K)=A(I,K)-A(J,K)
60 A(J,K)=A(J,K)+A(I,K)
      RR=R(J)
      R(J)=R(I)
      R(I)=-RR
62 DO 69 K=1,N
      IF (K-J)206,205,206
206 R(K)=R(K)-R(J)*A(K,J)/XLARG
205 DO 69 I=1,N
      IF (K-J)67,68,67
67 IF (I-J)65,68,65
65 A(K,I)=A(K,I)-A(J,I)*A(K,J)/XLARG
68 CONTINUE
      DO 110 K=1,N
      IF (K-J)70,110,70
70 A(K,I)=0.
110 CONTINUE
111 DTM=1.0
      DO 90 J=1,N
      IF (ABS(A(J,J))-SQ*TO1)91,90,90
90 DTM=DTM*A(J,J)
      GO TO 92
91 DTM=0.
      CALL EXIT
92 WRITE(3,215)DTM
215 FORMAT(11 ' DETERMINANT OF SYSTEM STIFFNESS MATRIX =',E17.7)
      DO 24 J=1,N
      IF (ABS(A(J,J))-SQ*TO1)24,25,25
25 R(J)=R(J)/A(J,J)
24 CONTINUE
      RETURN
      END

```

## Chapter 6

## CONCLUSIONS AND FUTURE INVESTIGATIONS

## 6.1 Conclusions

A method of analysis for space frameworks composed of segmental circular members has been developed in this paper along with the computer program to perform all the computations required by the analysis. This analysis is fundamental to a number of noteworthy investigations that could be made in the future, which are discussed in (6.2).

An analysis of this nature would be highly impractical without the aid of the computer, however, with the computer, the analysis of structures such as those described in this paper can be analyzed in a relatively short time. Since analysis described in this paper reflects more closely the true behavior of the structure, a more efficient and economical design of the "curvilinear space grid" should be expected.

An experimental model was built and tested in order that the theoretical solution could be correlated to the observed and measured physical results.

Because of the difficulty encountered in measuring the true vertical displacement of the node points, the correlation between the computed vertical nodal displacement and the measured vertical nodal displacement is

rather poor, however, the measured stress at a given point in a member correlates very closely with the stress in that member calculated from the member-end loads computed by the theoretical analysis. It can be concluded from these results that the theoretical analysis does compute the member end loads properly and if the nodal displacements had been measured more accurately the correlation between them would have been much closer.

The analysis of curved structures other than circular, could be made by approximating the curves with circular segments, which would increase the number of node points. This procedure, however, is suggested only in the case where the derivation of the actual flexibility is not feasible, since the addition of nodal points increases the number of equilibrium equations which must be solved and reduces the efficiency of analysis.

## 6.2 Topics for Further Investigation

- a.) Temperature Affect - The affect of temperature on the structure can be included in the analysis with little difficulty. If the change in temperature is equal to "t", the change in length of a member from its unstressed length and consequently the change in internal stress can be computed if the coefficient of thermal expansion for the material is known. The internal forces resulting from this



change in length can then be included in the member load-displacement equations in general as,

$$p_1 = K_{11} \delta_1 + K_{12} \delta_2 + p_{t1} \quad (6-1)$$

$$p_2 = K_{21} \delta_1 + K_{22} \delta_2 + p_{t2},$$

where  $p_{t1}$  and  $p_{t2}$  are the member end-loads resulting from the thermal expansion when the end-displacements are prevented.

The load-displacement equations for the structure can be assembled as was explained in Chapter 2 if equations (6-1) are first transformed into system coordinates as follows,

$$p'_1 = K'_{11} \delta'_1 + K'_{12} \delta'_2 + p'_{t1},$$

and 
$$p'_2 = K'_{21} \delta'_1 + K'_{22} \delta'_2 + p'_{t2}.$$

It is often desired to study the affects of temperature independent of any externally applied loads. This can be done in a straightforward manner from the following equation;

$$0 = K' \delta' + p'_t. \quad (6-2)$$

The results of equation (6-2) can then be superimposed on the normal solution.

- b.) Foundation Movements - The effects of foundation movements on the curvilinear space grid can easily be taken into account in the analysis by modifying the appropriate displacements in the load-displacement equations.
- c.) Optimum Design - It would also be interesting to see an optimum design study made for the lattice dome. Since the method of analysis described in this paper neglects only the shear energy in the general energy derivation, more efficient use is being made of the structural framework and the engineer is able to predict its behavior more accurately. It is therefore possible to study the effect of the grid spacing on the size of the material and arrive at a more efficient, economical design for the lattice dome.
- d.) Dynamic Loading - The problem of dynamic loading on the curvilinear space grid can now be studied. The generation of the system stiffness matrix is as fundamental a step to the formulation of the equations of motion for the dynamic system as it was to the formulation of the equilibrium equations for the static system. The loading vector is a function of time and the mass of the structure can be treated as a series of lumped masses and rotational inertias

concentrated at the node points. The problem can finally be reduced to the general eigenvalue problem.

- e.) Computational Problems - A discussion of the computational problems involved in analysis of this type are discussed in reference (3). There is, however, additional work needed in developing methods by which very large structural systems can be handled effectively such as by the substructure method, where the larger systems are divided into smaller units which can be handled more efficiently.<sup>3</sup>

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## VITA

The author, David L. Fenton, was born on March 20, 1941 in Murphysboro, Illinois. He received his primary and secondary education in the Dupou, Illinois Public School System graduating from high school in May, 1959.

He entered the University of Missouri at Rolla in September, 1959 and graduated in May, 1963 with a B.S. Degree in Civil Engineering. While employed as a stress analyst for McDonnell Aircraft Corporation he began graduate study at St. Louis University night school in September, 1963 and continued until February, 1964.

In September 1964 he entered the graduate school of the University of Missouri at Rolla. He received an M.S. Degree in Civil Engineering in January, 1966 and continued work towards the Doctor of Philosophy Degree in Civil Engineering. He has held a National Science Foundation Engineering Traineeship during the period from September, 1964 to August, 1967.

He is married to the former Janice A. Bieller and they have two children, David Scott and Susan Lynn.

APPENDIX A

COMPUTER SOLUTIONS FOR THE MATHEMATICAL MODEL

NUMBER OF NODES 9  
 NUMBER OF MEMBERS 24  
 NUMBER OF LOADING CONDITION 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	MEMBER COORDINANTS		MEMBER COORDINANTS		MEMBER COORDINANTS	
	X(I,1)	X(I,2)	Y(I,1)	Y(I,2)	Z(I,1)	Z(I,2)
1	-36.000000	-20.784988	0.0	20.784988	0.0	20.784988
2	-36.000000	-25.455994	0.0	25.455994	0.0	0.0
3	-36.000000	-20.784988	0.0	20.784988	0.0	-20.784988
4	0.0	-20.784988	0.0	20.784988	36.000000	0.0
5	-20.784988	-25.455994	20.784988	25.455994	20.784988	0.0
6	-25.455994	-20.784988	25.455994	20.784988	0.0	-20.784988
7	0.0	-20.784988	0.0	20.784988	-36.000000	0.0
8	-20.784988	0.0	20.784988	25.455994	20.784988	25.455994
9	-25.455994	0.0	25.455994	36.000000	0.0	0.0
10	-20.784988	0.0	20.784988	25.455994	-20.784988	-25.455994
11	0.0	0.0	0.0	25.455994	36.000000	25.455994
12	0.0	0.0	25.455994	36.000000	25.455994	0.0
13	0.0	0.0	36.000000	0.0	0.0	-25.455994
14	0.0	0.0	0.0	25.455994	-36.000000	-25.455994
15	0.0	20.784988	25.455994	20.784988	25.455994	20.784988
16	0.0	25.455994	36.000000	25.455994	0.0	0.0
17	0.0	20.784988	25.455994	20.784988	-25.455994	-20.784988
18	0.0	20.784988	0.0	20.784988	36.000000	20.784988
19	20.784988	25.455994	20.784988	25.455994	20.784988	0.0
20	25.455994	20.784988	25.455994	20.784988	0.0	-20.784988
21	0.0	20.784988	0.0	20.784988	-36.000000	-20.784988
22	36.000000	20.784988	0.0	20.784988	0.0	20.784988
23	36.000000	25.455994	0.0	25.455994	0.0	0.0
24	36.000000	20.784988	0.0	20.784988	0.0	-20.784988

MEMBER-I	MEMBER PROPERTIES				RADIUS	PHI	ALPHA
	IX	IY	IZ	AREA			
1	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	1.287000
2	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
3	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	-1.287000
4	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	-1.287000
5	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	-0.831423
6	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	-0.831423
7	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	1.287000
8	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.931423
9	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	-0.831423
10	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
11	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
12	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
13	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
14	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
15	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.831423
16	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
17	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	-0.831423
18	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	1.287000
19	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	-1.287000
20	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
21	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	-1.287000
22	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	0.0
23	0.000550	0.000326	0.000326	0.062500	36.000000	0.302700	1.287000
24	0.000550	0.000326	0.000326	0.062500	36.000000	0.477670	1.287000

MEMB	NODE (1,1)	NODE (1,2)
1	0	1
2	0	2
3	0	3
4	0	1
5	1	2
6	2	3
7	0	4
8	1	5
9	2	6
10	3	7
11	4	8
12	4	5
13	4	6
14	0	7
15	4	7
16	5	8
17	0	7
18	7	8
19	8	0
20	0	7
21	4	0
22	0	7
23	0	8
24	0	8

NODE	NODN	LINK					
		1	2	3	4	5	6
1	4	-1	-4	5	8		
2	4	-2	-6	7	0		
3	4	-3	-6	-7	10		
4	4	-8	-11	12	15		
5	4	-6	-12	13	16		
6	4	-10	-13	-14	-22		
7	4	-15	-18	19	27		
8	4	-16	-18	20	-33		
9	4	-17	-20	-21	-24		

NODE	APPLIED NODAL LOADS					
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	-40.000000	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.1562867E 61

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.055598	0.016225	-0.058159	-0.000824	-0.001359	-0.001427
2	0.076931	-0.051099	-0.072508	-0.009165	-0.001020	-0.005183
3	-0.078971	-0.292373	-0.078972	0.013052	0.000000	-0.019052
4	0.439066	0.015378	0.074752	0.002409	-0.002674	-0.000036
5	0.349578	0.043357	0.049578	-0.002420	0.000000	-0.002420
6	-0.072508	-0.051099	0.076829	0.005183	0.001020	0.009165
7	0.033001	-0.009746	0.033000	0.002105	-0.000000	-0.002105
8	0.026793	0.015078	0.039066	0.000036	0.002674	-0.002609
9	-0.058159	0.016225	0.055597	0.001427	0.001359	0.000824

MEMBER	END	MEMBER END FORCES AND MOMENTS					
		PPX	PPY	PPZ	MPX	MPY	MPZ
1	1*	-0.100601	0.018927	0.277201	3.033314	-2.267633	1.572212
1	2*	0.139715	0.153456	0.957667	-2.490432	3.113770	-0.005183
2	1*	0.100601	-0.018927	-0.277201	-2.336890	-4.060739	2.805744
2	2*	-0.139715	-0.153456	-0.957659	-2.588771	3.868771	-0.009165
3	1*	1.543047	3.993134	-0.145719	0.935995	0.376390	-4.593690
3	2*	4.279674	0.102487	-0.145719	0.705923	-0.720713	-4.593680
4	1*	-1.543047	-3.993134	0.145719	-4.645629	1.160076	2.417776
4	2*	-4.279674	-0.102487	0.145719	-0.705924	4.735763	7.417776
5	1*	12.369619	16.554779	-15.044730	21.413632	-61.914749	-44.037939
5	2*	-25.529633	-0.862700	0.951612	-0.728577	-13.375275	-18.358932
6	1*	-12.369619	-16.554779	15.044730	9.772490	33.718430	30.711914
6	2*	-25.529633	0.862700	-0.951612	0.728546	-18.121643	49.805054
7	1*	0.729789	-0.836330	0.702132	3.565194	4.674881	2.133312
7	2*	-1.300581	0.129121	-0.063611	-0.286561	1.392727	6.094994
8	1*	0.729789	-0.836330	-0.702132	-1.696170	-1.047219	0.762112
8	2*	1.300581	-0.129121	0.063611	0.286553	0.723141	-1.954406
9	1*	0.509947	-0.654749	1.445124	1.368699	-0.592475	1.444912
9	2*	-1.626452	-0.344078	-0.110757	-1.800809	-0.894636	0.529341
10	1*	0.509947	-0.654749	-1.445142	-0.729461	-3.234150	-1.267126
10	2*	1.626466	0.344063	0.110746	1.800805	3.311615	-8.034474
11	1*	0.417646	3.445484	2.264769	21.767334	-9.237518	0.000000
11	2*	-2.809364	1.298932	2.755906	-1.633978	-23.712666	8.340817
12	1*	0.417646	3.445453	-2.264860	30.268677	-9.875811	2.327216
12	2*	-2.809437	-1.298953	-2.755914	1.633979	-36.391953	10.988937
13	1*	-15.044777	16.554810	12.369680	44.033274	61.914933	-21.013632
13	2*	25.529709	-0.862727	-0.951594	0.728548	-13.352314	-18.352314
14	1*	15.044777	-16.554810	-12.369680	-39.711700	-33.719332	-0.722327
14	2*	-25.529709	0.862727	0.951596	-0.728553	18.121429	49.804871
15	1*	0.111362	-0.162628	-0.465691	-2.007455	6.681580	-3.817731
15	2*	-0.028438	-0.461748	-0.204200	1.235548	2.612046	-6.856617
16	1*	0.111362	-0.162627	0.465705	0.591839	6.517731	-1.302438
16	2*	0.028456	0.461753	0.204210	1.235541	1.841456	-4.713515
17	1*	1.641373	-0.106963	-0.965406	-8.893685	11.311256	-15.333076
17	2*	1.475502	-0.726937	-0.965406	-3.888159	13.853685	-15.333076
18	1*	-1.641383	0.106953	0.965406	-1.285555	13.264131	-4.694220
18	2*	-1.475515	0.726931	0.965406	3.888164	12.744421	-4.694280



10	1*		-2.264221	-3.445461	-0.410675	-9.329819	9.875580	-39.269562
	1	-2.807835	-1.299064	-2.755915	1.634019	36.392441	-19.989731	
	2*		2.264282	3.445455	0.410681	-8.682240	9.237925	-21.766739
11	1*		-0.096559	0.341775	-0.126125	-1.300026	0.970115	1.424407
	1	0.364025	0.014264	-0.096559	0.351185	1.687223	-1.300026	
	2*		0.096559	-0.341775	0.126125	1.693009	0.048007	1.033606
12	1*		0.072367	0.107008	-0.748080	-2.245688	-1.805944	-1.255922
	1	0.732087	-0.187410	0.072367	0.469269	-2.148999	-2.245688	
	2*		-0.072367	-0.107004	0.748075	-2.918265	-0.036328	0.492885
13	1*		0.965409	0.106982	-1.641309	4.696555	-13.264152	1.285570
	1	1.475435	0.726930	0.965409	3.888168	-12.746473	4.696555	
	2*		-0.965409	-0.106974	1.641296	15.332772	-11.311292	8.893700
14	1*		-0.145716	3.993148	1.543066	4.593898	-0.376377	-0.936025
	1	4.279694	0.102473	0.145716	-0.705923	0.720746	-4.593898	
	2*		0.145716	-3.993148	-1.543066	-7.417620	-1.160054	4.445371
15	1*		-0.057155	0.072206	0.156277	-0.039003	-2.759240	1.604104
	1	-0.103362	0.137803	0.056574	0.210228	-0.722257	3.101973	
	2*		0.057142	-0.071996	-0.156268	-0.354634	-0.221834	-0.374632
16	1*		0.748101	-0.107026	-0.072370	-0.492881	0.036358	2.018026
	1	0.732114	0.187401	-0.072370	-0.469278	-0.155023	2.918026	
	2*		-0.748091	0.107017	0.072370	1.255952	1.805894	2.245742
17	1*		-1.445042	0.654774	-0.508866	0.757240	3.234162	8.228302
	1	-1.626384	0.344132	0.110762	1.800812	-3.310483	8.034377	
	2*		1.445043	-0.654786	0.508879	-1.448779	0.592723	-1.368682
18	1*		-0.127284	-0.072203	0.023370	-0.613709	0.482272	1.023332
	1	-0.132797	-0.047158	-0.028905	-0.551570	0.740519	-0.883144	
	2*		0.127284	0.072203	-0.023379	0.003047	0.822317	0.023371
19	1*		-0.156328	-0.072066	0.057467	0.374863	0.221547	0.356431
	1	-0.103684	-0.137783	-0.056562	-0.210238	0.511588	0.097102	
	2*		0.156328	0.072073	-0.057465	-1.604377	2.759485	0.038291
20	1*		0.465777	0.162693	-0.111044	1.382535	-4.512956	-0.591795
	1	-0.028774	0.461753	0.204215	1.235541	-1.941524	4.213935	
	2*		-0.465796	-0.162711	0.111072	2.517452	-5.691447	2.537440
21	1*		0.272227	0.018945	-0.100587	-1.572144	2.267593	-3.033355
	1	0.130749	0.210360	-0.153461	-0.257667	2.400488	3.113719	
	2*		-0.272227	-0.018945	0.100587	-0.936797	4.041059	-2.336020
22	1*		-0.023353	-0.072046	-0.122257	-1.020348	-0.492405	0.513044
	1	-0.132751	-0.047153	0.028904	0.551578	-0.740503	-0.893333	
	2*		0.023353	0.072046	0.122257	-0.023246	-0.822312	-0.033048
23	1*		-0.126134	0.341783	-0.096559	-1.424390	-0.970122	1.300112
	1	0.364035	0.014260	0.096559	-0.351197	-1.687211	-1.300122	
	2*		0.126134	-0.341783	0.096559	-1.933406	-0.047294	-1.602291
24	1*		0.702182	-0.935393	0.720849	-2.139356	-4.475132	-3.665300
	1	-1.300681	0.129125	0.063612	0.286555	-1.382272	6.395330	
	2*		-0.702192	0.935393	-0.720849	-0.262191	1.049001	1.606320
25	1*		-0.063612	-0.935393	0.720849	-2.139356	-4.475132	-3.665300
	1	1.300681	-0.129125	-0.063612	-0.286543	-0.272153	-1.049001	
	2*		0.063612	0.935393	-0.720849	0.262153	1.049001	1.606320

NODE	PX	PY	APPLIED NODAL LOADS		MY	MZ
			PZ	MX		
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	-20.000000	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	-20.000000	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	-20.000000	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	-20.000000	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.1562867E-51

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	-0.041300	-0.036021	0.041300	-0.004453	-0.000000	-0.004453
2	0.079362	-0.089245	-0.000000	-0.000000	0.000000	-0.007350
3	-0.041300	-0.036021	-0.041300	0.004453	0.000000	-0.004453
4	0.000000	-0.089245	-0.079361	-0.007350	-0.000000	-0.000000
5	-0.000000	-0.116130	-0.000000	-0.000000	-0.000000	-0.000000
6	-0.000000	-0.089245	0.079361	0.007350	-0.000000	-0.000000
7	0.041300	-0.036021	0.041300	-0.004453	-0.000000	-0.004453
8	-0.079361	-0.089245	0.000000	0.000000	0.000000	0.007350
9	0.041300	-0.036021	-0.041300	0.004453	0.000000	0.004453

MEMB.	END	MEMBER END FORCES AND MOMENTS						
		DPY	DPX	DPZ	DPY	DPX	DPZ	
1	1	0.820550	0.950127	0.523771	-3.071856	6.346787	-3.629516	
1	2	-0.212025	-0.186927	0.204003	2.538904	-7.506739		
1	3	-0.820550	-0.950127	-0.523771	-3.711434	2.739515	-0.492309	
1	4	0.212025	0.186927	-0.204003	3.663120	-2.850664		
2	1	7.641934	18.209774	0.000001	-0.000018	-0.000005	-33.066376	
2	2	0.000000	0.000000	0.000000	-0.000018	-33.066376		
2	3	-7.641934	-18.209774	-0.000001	0.000015	0.000000	33.083917	
2	4	0.000000	0.000000	0.000000	-0.000017	33.083917		
3	1	0.000000	0.000000	-0.000000	3.071875	-6.346756	-3.629477	
3	2	-0.000000	-0.000000	-0.000000	-2.538932	7.506692		
3	3	0.000000	0.000000	0.000000	3.711457	-2.739536	-0.492303	
3	4	-0.000000	-0.000000	-0.000000	-3.663147	2.850677		
4	1	0.000000	0.000000	-0.000000	3.627344	-6.346710	-3.071858	
4	2	-0.000000	-0.000000	0.000000	-2.538927	7.506640		
4	3	0.000000	0.000000	0.000000	3.492376	-2.739510	-3.711456	
4	4	-0.000000	-0.000000	-0.000000	-3.663146	2.850700		
5	1	0.000000	0.000000	-1.000000	1.000000	0.000000	0.000000	
5	2	0.000000	0.000000	-1.000000	1.000000	0.000000	0.000000	
5	3	0.000000	0.000000	1.000000	-1.000000	0.000000	0.000000	
5	4	0.000000	0.000000	1.000000	-1.000000	0.000000	0.000000	
6	1	0.000000	0.000000	-1.000000	1.000000	0.000000	0.000000	
6	2	0.000000	0.000000	-1.000000	1.000000	0.000000	0.000000	
6	3	0.000000	0.000000	1.000000	-1.000000	0.000000	0.000000	
6	4	0.000000	0.000000	1.000000	-1.000000	0.000000	0.000000	
7	1	0.000000	0.000000	-0.000000	0.000000	0.000000	0.000000	
7	2	0.000000	0.000000	-0.000000	0.000000	0.000000	0.000000	
7	3	0.000000	0.000000	0.000000	0.000000	0.000000	0.000000	
7	4	0.000000	0.000000	0.000000	0.000000	0.000000	0.000000	
8	1	0.000000	0.000000	-1.000000	1.000000	0.000000	0.000000	
8	2	0.000000	0.000000	-1.000000	1.000000	0.000000	0.000000	
8	3	0.000000	0.000000	1.000000	-1.000000	0.000000	0.000000	
8	4	0.000000	0.000000	1.000000	-1.000000	0.000000	0.000000	

	1*		1.820171	0.850050	-1.523337	-3.226437	9.651086	0.977324
10	1	2.242989	1.058309	-0.452937	-1.217192	6.655619	7.663540	
	2*		-1.820161	-0.850057	1.523345	0.081434	13.509062	8.188224
	2	-2.242982	-1.058321	0.452938	1.217192	3.222987	15.416857	
	1*		-0.000001	18.299805	-7.543853	-33.066483	0.000005	0.000019
11	1	19.793716	0.033278	-0.000001	-0.000003	0.000019	-33.066483	
	2*		0.000001	-18.299805	7.543853	33.983978	0.000010	0.000015
	2	-19.793716	-0.033278	0.000001	0.000003	0.000018	33.983978	
	1*		0.000005	0.000087	-4.497155	-33.821411	-0.000058	-0.000022
12	1	4.154974	-1.720876	0.000005	-0.000002	-0.000062	-33.821411	
	2*		-0.000005	-0.000009	4.497165	-13.594635	-0.000075	-0.000033
	2	-4.154888	1.720868	-0.000005	0.000007	-0.000082	-13.594635	
	1*		0.000000	-0.000057	-4.497087	13.594836	0.000013	-0.000015
13	1	4.154800	1.720877	0.000000	0.000008	0.000018	13.594836	
	2*		-0.000000	0.000057	4.497071	33.821167	-0.000020	0.000017
	2	-4.154786	-1.720871	-0.000000	-0.000008	-0.000025	33.821167	
	1*		-0.000001	18.299835	7.543866	33.066589	0.000010	0.000013
14	1	19.793747	0.033278	0.000001	0.000014	-0.000008	-33.066589	
	2*		0.000001	-18.299835	-7.543866	-33.983948	-0.000018	0.000007
	2	-19.793747	-0.033278	-0.000001	-0.000014	-0.000013	33.983948	
	1*		1.820295	-0.850069	-1.523369	-0.081422	13.509340	-8.188356
15	1	2.243118	-1.058305	-0.452946	-1.217212	3.223083	-15.417150	
	2*		-1.820235	0.850069	1.523369	3.226445	9.651039	-0.977305
	2	-2.243061	1.058323	0.452946	1.217205	6.655622	-7.663494	
	1*		4.497111	-0.000074	-0.000001	-0.000022	0.000005	13.594778
16	1	4.154820	1.720871	-0.000001	-0.000022	-0.000004	13.594778	
	2*		-4.497066	0.000074	0.000001	0.000033	0.000023	33.821213
	2	-4.154815	-1.720865	0.000001	0.000022	0.000034	33.821213	
	1*		1.820137	-0.850057	1.523334	0.081452	-13.508987	-8.188297
17	1	2.243014	-1.058303	0.452930	1.217177	-3.222883	-15.416854	
	2*		-1.820152	0.850057	-1.523334	-3.226361	-9.650845	-0.977324
	2	-2.242921	1.058314	-0.452930	-1.217171	-6.655437	-7.663367	
	1*		0.850141	-0.820570	-3.428567	6.346868	3.071829	
18	1	1.820096	-0.812027	-0.186930	0.294902	2.638939	-7.506836	
	2*		-0.850141	0.820570	-0.492775	-7.738522	3.711464	
	2	-1.820096	0.812027	0.186930	-0.294925	3.449162	-2.850651	
	1*		1.820136	-0.850039	-1.820152	0.073374	-0.651072	-3.226429
19	1	2.242967	-1.058325	0.452946	1.217198	-6.655567	-7.663561	
	2*		-1.820136	-0.850027	1.820152	9.182234	-13.509074	0.001432
	2	-2.242922	1.058304	-0.452946	-1.217180	-3.222982	15.416866	
	1*		-1.523356	-0.250074	-1.220357	-8.189372	13.509235	0.001428
20	1	0.242176	-1.058280	-0.452940	-1.217191	3.222985	-15.417177	
	2*		1.523366	0.250076	1.220246	-0.372207	9.650799	-3.226395
	2	-0.242003	1.058215	0.452930	1.217170	6.655438	-7.663347	
	1*		0.850140	-0.820535	-3.428355	-6.346622	3.071833	
21	1	1.820092	-0.812021	-0.186922	-0.294910	2.638917	-7.506864	
	2*		-0.850140	0.820535	-0.492375	-7.738502	3.711468	
	2	-1.820092	0.812021	-0.186922	0.294914	-3.449143	-2.850780	
	1*		0.820560	-0.850088	0.523238	-3.071888	-4.346738	3.029460
22	1	1.820093	-0.812020	-0.186920	-0.294928	-2.818940	-7.506868	
	2*		0.850054	-0.850098	-0.523239	-3.211482	-2.732673	0.492411
	2	-1.820093	0.812020	-0.186920	0.294903	-3.449153	-2.850854	
	1*		-7.544351	19.299789	-0.000002	-0.000035	-0.000011	33.066513
23	1	19.793761	0.033274	0.000002	0.000004	-0.000032	-33.066513	
	2*		7.544351	-19.299789	0.000002	-0.000017	-0.000011	-33.066587
	2	-19.793761	-0.033274	-0.000002	-0.000004	-0.000020	33.066587	
	1*		-0.000000	0.250168	-0.523313	3.711957	6.346917	3.628633
24	1	1.820094	-0.812024	-0.186924	0.294910	2.638884	-7.506811	
	2*		0.850063	-0.850168	0.523313	3.711394	2.732387	0.492210
	2	-1.820094	0.812024	0.186924	-0.294916	-3.449198	-2.850498	

NUMBER OF NODES 9  
 NUMBER OF MEMBERS 24  
 NUMBER OF LOADING CONDITION 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	X(I,1)		X(I,2)		MEMBER COORDINANTS		Z(I,1)	Z(I,2)
	Y(I,1)	Y(I,2)	Y(I,1)	Y(I,2)				
1	-33.369995	-18.369995	0.0	9.740000	13.500000	18.369995	0.0	0.0
2	-36.000000	-20.129990	0.0	13.250000	18.369995	18.369995	0.0	0.0
3	-33.369995	-18.369995	0.0	9.740000	-13.500000	-18.369995	0.0	0.0
4	-18.369995	-20.129990	9.740000	13.250000	18.369995	0.0	0.0	0.0
5	-18.369995	-20.129990	9.740000	13.250000	18.369995	0.0	0.0	0.0
6	-20.129990	-18.369995	13.250000	9.740000	0.0	-18.369995	0.0	0.0
7	-13.500000	-18.369995	0.0	9.740000	9.740000	-33.369995	-18.369995	20.129990
8	-18.369995	0.0	0.0	9.740000	13.250000	18.369995	20.129990	0.0
9	-20.129990	0.0	13.250000	18.000000	0.0	0.0	0.0	0.0
10	-18.369995	0.0	9.740000	13.250000	18.369995	-20.129990	-20.129990	0.0
11	0.0	0.0	0.0	13.250000	13.250000	36.000000	20.129990	0.0
12	0.0	0.0	13.250000	18.000000	18.000000	20.129990	-20.129990	0.0
13	0.0	0.0	18.000000	13.250000	13.250000	0.0	-20.129990	0.0
14	0.0	0.0	0.0	13.250000	-36.000000	-20.129990	-20.129990	0.0
15	0.0	18.369995	13.250000	9.740000	20.129990	18.369995	18.369995	0.0
16	0.0	20.129990	18.000000	13.250000	0.0	0.0	0.0	0.0
17	0.0	18.369995	13.250000	9.740000	-20.129990	-18.369995	-18.369995	0.0
18	13.500000	18.369995	0.0	9.740000	33.369995	18.369995	18.369995	0.0
19	18.369995	20.129990	9.740000	13.250000	18.369995	0.0	0.0	0.0
20	20.129990	18.369995	13.250000	9.740000	0.0	-18.369995	-18.369995	0.0
21	13.500000	18.369995	0.0	9.740000	-33.369995	-18.369995	-18.369995	0.0
22	33.369995	18.369995	0.0	9.740000	13.500000	18.369995	18.369995	0.0
23	36.000000	20.129990	0.0	13.250000	0.0	0.0	0.0	0.0
24	33.369995	18.369995	0.0	9.740000	-13.500000	-18.369995	-18.369995	0.0

MEMBER-I	MEMBER PROPERTIES			RADIUS	PHI	ALPHA
	IX	IY	IZ			
1	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
2	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
3	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
4	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
5	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200
6	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
7	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
8	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
9	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200
10	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
11	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
12	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
13	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
14	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200
15	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
16	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
17	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200
18	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
19	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
20	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200
21	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
22	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430
23	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870
24	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430

MEMB NODE (I,1) NODE (I,2)

1	1	1
2	1	2
3	1	3
4	1	4
5	2	1
6	2	2
7	2	3
8	2	4
9	3	1
10	3	2
11	3	3
12	3	4
13	4	1
14	4	2
15	4	3
16	4	4
17	5	1
18	5	2
19	5	3
20	5	4
21	6	1
22	6	2
23	6	3
24	6	4

MEMB	MEMB	1	2	3	4	5	6
1	4	-1	-4	5	8		
2	4	-2	-5	4	0		
3	4	-4	-4	-7	10		
4	4	-8	-11	12	15		
5	4	-6	-12	13	16		
6	4	-10	-13	-14	-15		
7	4	-11	-19	19	-22		
8	4	-11	-19	-22	-22		
9	4	-17	-20	-21	-25		

NODE	APPLIED NODAL LOADS					
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	-4.000000	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	-0.063337	0.068161	0.063201	0.004961	0.001831	0.001690
2	-0.038445	0.047327	0.069505	-0.008413	-0.005350	0.000062
3	-0.111500	-0.132761	-0.111696	-0.004755	-0.000003	-0.006769
4	-0.041259	-0.025325	-0.020060	-0.001324	0.003668	-0.006492
5	-0.022588	-0.023661	-0.022578	-0.001791	0.000000	-0.001792
6	0.069806	0.047311	-0.038432	-0.000042	0.005347	0.008609
7	-0.041040	-0.045679	-0.041041	-0.003167	-0.000000	0.003167
8	-0.020072	-0.025340	-0.041266	0.004494	-0.003669	0.001324
9	0.063213	0.068175	-0.063347	-0.001693	-0.001830	-0.004963

MEMB.	END	MEMBER END FORCES AND MOMENTS						
		PX	PY	PZ	MX	MY	MPY	MPZ
1	1*		-1.766142	-3.023878	-1.949082	-3.710666	8.506456	-12.902464
1	1	-3.530247	-1.897206	0.000872	-1.924691	0.415458	-15.776762	
1	2*		1.766142	3.023878	1.949082	-0.547315	12.128900	-15.246690
1	2	3.530247	1.897206	-0.000872	1.924678	-0.431345	-19.390198	
2	1*		-1.580640	-2.349012	-0.933435	-4.144855	8.376666	-6.003715
2	1	-2.718810	-0.790136	-0.933435	2.154827	0.048185	-6.003715	
2	2*		1.580640	2.349012	0.933435	-8.223157	6.486975	-10.331923
2	2	2.718810	0.790136	0.933435	-2.154828	10.249280	-10.331923	
3	1*		2.812057	20.379120	-10.894502	-2.484541	3.562222	6.394213
3	1	35.268266	3.008282	-0.009400	-1.818328	0.176939	7.511764	
3	2*		-2.812057	-20.379120	10.894502	-4.382036	20.296663	33.143326
3	2	-35.268266	-3.008282	0.009400	1.818317	-0.002121	48.251222	
4	1*		-2.417501	5.756901	-10.452099	-16.522233	-12.924223	-2.064622
4	1	12.118405	-1.160170	0.176668	0.757106	-1.338868	-21.291512	
4	2*		2.417501	-5.756901	10.452099	1.070031	-1.714665	-1.026212
4	2	-12.118405	1.160170	-0.176668	-0.757094	-1.034605	-0.213924	
5	1*		-1.153997	2.187143	-12.203726	-3.284628	-1.134145	1.026122
5	1	13.135491	-0.277666	-0.060197	-0.586377	0.794665	-3.423212	
5	2*		1.153996	-2.187129	12.002726	-1.830854	-0.377272	-1.026326
5	2	-13.135482	0.277658	0.060203	0.586377	0.426422	-1.723174	
6	1*		-0.548379	-0.378388	-15.367086	2.026263	11.022255	0.026561
6	1	15.046972	3.141318	-0.552045	-2.212254	5.832592	12.266342	
6	2*		0.548379	0.378434	15.367085	-38.003122	26.112518	-3.036226
6	2	-15.046970	-3.141278	0.552067	2.212192	4.068692	46.745122	
7	1*		-10.893972	20.377975	26.812119	-6.386791	-3.560735	2.483222
7	1	35.267380	3.007253	0.009656	1.816352	-0.178679	7.602410	
7	2*		10.893972	-20.377975	-26.812119	-38.128728	-29.222569	4.383264
7	2	-35.267380	-3.007253	-0.007656	-1.816404	-0.001226	48.236875	
8	1*		-3.028272	0.545901	0.502679	2.762594	-0.220220	16.122222
8	1	-2.812281	1.336432	0.161755	-2.378883	-1.814104	17.692262	
8	2*		3.028272	-0.545899	-0.502681	-1.958875	-5.284264	0.554222
8	2	2.812291	-1.336431	-0.161758	-2.378882	-1.874922	7.413022	
9	1*		-2.186502	0.216810	1.522975	2.056959	-17.117844	2.025522
9	1	-2.078355	0.713186	1.522975	-1.929293	-17.132226	2.002562	
9	2*		2.186577	-0.216825	-1.522975	5.200912	-13.440222	4.949615
9	2	2.078338	-0.713192	-1.522975	1.929285	-14.470125	4.948615	

	1*		15.367833	0.377543	0.548442	3.517219	-26.119980	-39.010284
10	1	15.047538	-3.142228	0.558693	2.213904	-4.967407	-46.763779	
	2*		-15.367833	-0.377547	-0.548445	-0.927705	-11.001536	-7.993570
	2	-15.047538	3.142227	-0.558698	-2.213938	-5.527942	-12.261002	
	1*		0.463893	-0.322708	0.765434	3.767108	-4.515789	-2.013635
11	1	-0.794391	0.242846	0.463893	-1.348439	-4.756975	3.767108	
	2*		-0.463893	0.322709	-0.765434	1.253481	-2.846193	-4.132946
	2	0.794391	-0.242846	-0.463893	1.348442	-4.833611	-1.253481	
	1*		-0.474916	-0.216680	1.133637	-0.023618	6.242937	0.858182
12	1	-1.153099	0.049462	-0.474916	0.598504	6.273164	-0.023618	
	2*		0.474916	0.216686	-1.133657	1.046471	3.317116	1.397664
	2	1.153119	-0.049460	0.474915	-0.598500	-3.549441	1.046471	
	1*		-1.527936	-0.216686	2.186391	-4.847777	13.642657	-5.209234
13	1	-2.078188	-0.713020	-1.527936	1.928078	14.472296	-4.847777	
	2*		1.527936	0.216702	-2.186400	-9.999059	17.116685	-2.057459
	2	2.078193	0.713038	1.527936	-1.928082	17.129745	-9.999059	
	1*		-0.933450	-2.348261	-1.579997	6.002417	-8.325880	4.146371
14	1	-2.717843	-0.789971	0.933450	-2.153173	-0.048570	-6.002417	
	2*		0.933450	2.348261	1.579997	10.322649	-6.497979	0.221849
	2	2.717843	0.789971	-0.933450	2.153173	-10.249706	-10.322649	
	1*		-2.089021	0.439911	0.134381	0.729396	1.888396	-2.319915
15	1	-2.137665	2.004320	-0.077235	0.576775	0.496611	-2.915339	
	2*		2.089025	-0.439881	-0.134379	-0.426853	-0.689183	1.317389
	2	2.137664	-0.004297	0.077222	-0.576782	0.753707	3.015240	
	1*		-1.133899	0.216801	0.475173	-1.399177	-3.312893	1.366236
16	1	-1.153294	-0.249384	0.475173	-0.599424	-3.552112	-1.046398	
	2*		1.133830	-0.216816	-0.475173	-0.957889	-6.245216	0.337092
	2	1.153317	0.049374	-0.475173	0.599429	-6.271891	0.925996	
	1*		12.903961	-2.187189	1.154101	0.497491	0.325629	1.939662
17	1	13.135749	0.277606	0.069268	0.587741	-0.422948	1.291931	
	2*		-12.903931	2.187131	-1.154309	-0.599374	1.134613	3.294319
	2	-13.135699	-0.277652	-0.069293	-0.587747	-0.294242	3.423588	
	1*		0.033549	-0.439606	1.929324	8.098997	-5.875657	-2.035666
18	1	-1.776176	0.851123	0.051920	1.249863	-0.549772	1.156961	
	2*		-0.033549	0.439606	-1.929324	3.629765	-3.979919	1.316119
	2	1.776176	-0.851193	-0.051920	-1.249879	-0.421924	-5.219939	
	1*		-0.134279	-0.439789	2.388927	-3.058959	0.489324	1.226433
19	1	-2.137642	-0.004290	0.077264	-0.576697	-0.754655	-3.018547	
	2*		0.134278	0.439789	-2.388927	2.311421	-1.899513	2.329991
	2	2.137642	0.004290	-0.077265	0.576693	-0.626916	2.932399	
	1*		-0.592763	-0.545787	3.928986	-5.582534	5.288619	1.511327
20	1	-2.812805	-1.334591	-0.161879	2.389534	1.626434	7.419174	
	2*		0.592763	0.545757	-3.928971	-15.063644	3.272122	1.292119
	2	2.812804	1.334672	0.161893	2.389535	1.511699	-12.682124	
	1*		-1.949217	-3.924102	-1.766679	12.992535	-8.599336	2.711214
21	1	-3.630829	-1.892113	-0.999835	1.925934	-0.415936	-15.274334	
	2*		1.949217	3.924102	1.766679	15.244516	-12.128331	1.269374
	2	3.630829	1.892113	0.999835	-1.925929	0.439434	-13.202138	
	1*		1.912764	-0.439358	0.033719	3.432297	5.076399	-1.491979
22	1	-1.775537	0.851137	-0.052010	-1.249949	0.415936	12.459133	
	2*		-1.912764	0.439358	-0.033756	-3.929334	3.929334	-1.929334
	2	1.775537	-0.851137	0.052010	1.249945	-0.439912	5.292965	
	1*		0.765581	-0.322544	0.463279	2.912582	5.439391	1.299636
23	1	-0.794411	0.243050	-0.463272	-1.349279	4.767695	3.767695	
	2*		-0.765581	0.322544	-0.463279	4.136289	3.894297	1.894297
	2	0.794411	-0.243050	0.463279	-1.349279	4.936456	3.914297	
	1*		-10.452179	5.756692	-2.417382	3.065674	19.339393	-2.229393
24	1	12.118329	-1.163435	-0.176428	-0.758126	1.336734	-11.292592	
	2*		10.452179	-5.756692	2.417397	0.524231	1.215621	1.215621
	2	-12.118330	1.163435	0.176428	0.758140	-1.933976	-0.215621	

NODE	PX	PY	APPLIED NODAL LOADS PZ	MX	MY	MZ
1	0.0	-10.000000	0.0	0.0	0.0	0.0
2	0.0	-10.000000	0.0	0.0	0.0	0.0
3	0.0	-10.000000	0.0	0.0	0.0	0.0
4	0.0	-10.000000	0.0	0.0	0.0	0.0
5	0.0	-10.000000	0.0	0.0	0.0	0.0
6	0.0	-10.000000	0.0	0.0	0.0	0.0
7	0.0	-10.000000	0.0	0.0	0.0	0.0
8	0.0	-10.000000	0.0	0.0	0.0	0.0
9	0.0	-10.000000	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODAL DISPLACEMENTS

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.000404	-0.005442	-0.000404	-0.000988	0.000000	-0.000988
2	-0.000705	-0.008225	0.000000	0.000000	-0.000000	-0.001477
3	0.000404	-0.005443	0.000404	0.000989	0.000000	-0.000989
4	-0.000000	-0.008226	0.000705	-0.001477	0.000000	0.000000
5	0.000000	-0.019327	-0.000000	0.000000	-0.000000	0.000000
6	-0.000000	-0.008226	-0.000705	0.001477	-0.000000	0.000000
7	-0.003474	-0.005443	-0.000404	-0.000988	-0.000000	0.000989
8	0.000705	-0.008226	0.000000	0.000000	0.000000	0.001477
9	-0.000404	-0.005443	0.000404	0.000989	-0.000000	0.000989

MEMBER END FORCES AND MOMENTS

MEMB.	END	RPX	RPY	RPZ	MPX	MPY	MPZ
		PX	PY	PZ	MX	MY	MZ
1	1	0.000340	6.314215	3.091253	-0.017822	5.141374	-6.160755
1	2	12.068174	-0.084451	-0.042371	0.340245	0.374170	-8.060225
1	3	-0.000340	-6.314215	-3.091253	0.017822	-5.141374	6.160755
1	4	-12.068174	0.084451	0.042371	-0.340244	-0.411143	6.495201
2	1	12.019438	9.871697	0.000011	0.000067	-0.000038	-13.072557
2	2	16.552426	-0.124810	0.000011	0.000027	-0.000071	-13.072556
2	3	-12.019439	-9.871697	-0.000011	-0.000078	-0.000135	10.492106
2	4	-16.552426	0.124810	-0.000011	-0.000027	-0.000154	10.492136
3	1	0.000340	6.314222	-3.091204	0.017771	-5.141497	-6.160701
3	2	12.068201	-0.084440	0.042384	-0.340360	-0.374240	-8.060271
3	3	-0.000340	-6.314222	-3.091204	-0.017771	5.141341	6.160702
3	4	-12.068201	0.084440	0.042384	0.340359	-0.411226	6.495521
4	1	-3.091231	6.314174	-0.000274	-6.160711	-5.141342	-0.017812
4	2	12.068265	-0.084451	0.042373	-0.340249	-0.374170	-8.060271
4	3	3.091231	-6.314174	0.000274	6.160712	5.141322	0.017812
4	4	-12.068265	0.084451	-0.042373	0.340247	-0.411130	6.495252
5	1	-0.000920	1.314141	-7.376707	-5.606506	-3.106823	3.033214
5	2	7.467549	-0.117223	0.041054	-0.000101	-0.000650	-6.476133
5	3	0.000920	-1.314141	7.376707	5.606279	3.106746	-3.033161
5	4	-7.467547	0.117220	-0.041053	0.000102	0.000652	6.476210
6	1	0.000916	-1.314186	-7.376754	-5.606439	-3.106735	-3.033131
6	2	7.467514	-0.117175	0.041070	0.000091	0.000642	-6.476116
6	3	-0.000917	1.314186	7.376750	5.606474	3.106447	3.033132
6	4	-7.467500	-0.117176	0.041072	-0.000090	-0.000630	6.476243
7	1	-3.091237	6.314181	0.000377	-6.161016	-5.141413	-0.017720
7	2	12.068123	-0.084501	-0.042360	0.340272	0.374111	-8.061012
7	3	3.091237	-6.314181	-0.000377	-6.161355	-5.141649	0.017643
7	4	-12.068133	0.084501	0.042360	-0.340271	-0.411021	6.494457
8	1	7.376754	1.314136	0.000917	0.000212	0.000212	-6.476200
8	2	-7.376755	-1.314136	-0.000917	-0.000213	-0.000213	6.476200
8	3	-7.467505	0.117217	0.041054	-0.000103	0.000744	4.274232
9	1	10.800742	3.000001	0.000018	0.000040	-0.000224	-10.000000
9	2	11.086206	-0.047311	0.000019	-0.000223	-0.000223	-10.000000
9	3	-10.800735	-3.000000	-0.000018	0.000039	0.000224	10.000000
9	4	-11.086200	0.047311	-0.000019	-0.000040	0.000224	10.000000

	11		7.326942	1.314161	-0.608833	-0.033158	-3.184870	-5.640677
10	1	7.467695	-0.117227	0.041051	-0.099031	-0.096388	-6.476290	
	2		-7.326941	-1.314161	0.608833	0.209078	1.473704	4.064315
	2	-7.467694	0.117227	-0.041051	0.099031	-0.674788	4.274227	
	11		-0.000011	9.871700	-12.018441	-13.072544	0.000041	0.000069
	1	15.552430	-0.124810	-0.000011	-0.000027	0.000075	-13.072544	
	2		0.000011	-9.871700	12.018441	10.492209	0.000139	0.000081
	2	-15.552430	0.124810	0.000011	0.000027	-0.000159	10.492209	
	11		-0.000017	2.499994	-10.800718	-10.073741	0.000211	0.000049
12	1	11.086182	-0.047311	-0.000017	0.000002	0.000216	-10.073741	
	2		0.000017	-2.499992	10.800712	9.095231	0.000132	0.000033
	2	-11.086175	0.047312	0.000017	-0.000002	0.000136	9.095231	
	11		0.000016	-2.500009	-10.800703	-9.095321	-0.000124	0.000030
	1	11.086170	0.047296	0.000016	-0.000001	-0.000128	-9.095321	
	2		-0.000016	2.500008	10.800689	10.073504	-0.000200	0.000047
	2	-11.086157	-0.047292	-0.000016	0.000001	-0.000205	10.073504	
	11		-0.000012	9.871642	12.018409	13.072691	-0.000047	0.000022
14	1	15.552368	-0.124834	0.000012	0.000025	-0.000082	-13.072691	
	2		0.000012	-9.871642	-12.018409	-10.491845	-0.000144	0.000020
	2	-15.552369	0.124834	-0.000012	-0.000025	-0.000146	10.491845	
	11		7.326725	-1.314182	-0.608816	-0.209200	1.473370	-4.064382
	1	7.467484	0.117171	0.041078	-0.099041	-0.675133	-4.274127	
	2		-7.326731	1.314182	0.608816	0.233189	-3.184418	5.640640
	2	-7.467490	-0.117173	-0.041077	0.099041	-0.096520	4.275135	
	11		10.800701	-2.500006	-0.000017	0.000035	0.000139	-0.000012
	1	11.086167	0.047297	-0.000017	0.000002	0.000142	-0.000012	
	2		-10.800687	2.500006	0.000017	-0.000049	0.000213	10.073520
	2	-11.086154	-0.047294	0.000017	-0.000002	-0.000210	10.073520	
	11		7.326950	-1.314217	0.608837	0.209130	-1.473365	-4.064400
	1	7.467713	0.117180	-0.041074	0.099015	0.675156	-4.274134	
	2		-7.326956	1.314218	-0.608837	-0.233139	3.184424	5.640652
	2	-7.467719	-0.117179	0.041074	-0.099014	-0.676437	4.275235	
	11		3.091311	6.314350	-0.909535	-6.167757	5.141520	0.001221
	1	12.068374	-0.084442	-0.042385	0.340360	0.374250	-0.060813	
	2		-3.091311	-6.314359	0.909535	6.161195	-3.738850	-0.001221
	2	-12.068374	0.084442	0.042385	-0.340361	-0.411299	6.495536	
	11		0.608834	1.314164	-7.326049	-5.647666	3.184402	-0.001221
	1	7.467702	-0.117225	-0.041050	0.099033	-0.674727	-4.274221	
	2		-0.608836	-1.314164	7.326949	4.064337	-1.473372	0.000012
	2	-7.467702	0.117225	0.041050	-0.099033	0.674727	4.274252	
	11		-0.608837	-1.314220	-7.326049	-4.064477	1.473372	-0.000012
	1	7.467714	0.117177	0.041076	-0.099015	-0.675162	-4.274212	
	2		-0.608838	1.314220	7.326955	5.647902	-3.184424	0.000012
	2	-7.467720	-0.117178	-0.041075	0.099014	-0.675162	4.275139	
	11		3.091223	6.314321	-0.909566	-6.161133	5.141471	0.001221
	1	12.068374	-0.084404	-0.042375	-0.340382	-0.374185	-0.061127	
	2		-3.091223	-6.314321	0.909566	6.161586	-3.738855	-0.001221
	2	-12.068374	0.084404	0.042375	0.340382	-0.411164	6.494822	
	11		-0.084438	6.314221	3.091254	-0.017995	-5.141550	0.001221
	1	12.068208	-0.084502	0.042362	-0.340272	-0.374115	-0.061126	
	2		0.084438	-6.314221	-3.091254	0.276440	3.738822	-0.001221
	2	-12.068208	0.084502	-0.042362	0.340272	-0.411007	6.494702	
	11		-12.018426	9.871656	0.000012	0.000027	0.000075	0.000020
	1	15.552391	-0.124834	-0.000012	-0.000025	-0.000078	-13.072544	
	2		12.018426	-9.871656	-0.000012	0.000027	0.000075	0.000020
	2	-15.552391	0.124834	0.000012	0.000025	-0.000075	-13.072544	
	11		-0.084519	6.314289	-3.091272	0.017755	5.141455	0.001221
	1	12.068316	-0.084436	-0.042374	0.340383	0.374181	-0.061126	
	2		0.084519	-6.314289	3.091272	-0.017286	-3.738821	-0.001221
	2	-12.068316	0.084436	0.042374	-0.340382	0.411153	6.494333	



NODE	PX	PY	APPLIED NODAL LOADS			
			PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	-20.000000	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	-20.000000	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	-20.000000	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	-20.000000	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0

DEFINITION OF SYSTEM STIFFNESS MATRIX = 0.0

NODAL DISPLACEMENTS

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	-0.028709	0.021993	0.025608	-0.002299	-0.000000	-0.002299
2	0.384809	-0.072127	-0.000003	0.000003	0.000002	-0.001319
3	-0.025611	0.021987	-0.025609	0.002297	-0.002292	-0.002299
4	1.303003	-0.072127	-0.053598	0.001319	-0.000002	0.000003
5	-0.000000	0.136364	0.000002	-0.000000	0.000000	-0.000000
6	1.303003	-0.072134	0.053604	-0.001319	0.000002	0.000003
7	-0.028609	0.021987	0.025611	-0.002299	0.000002	-0.002297
8	-0.053606	-0.072135	-0.000003	0.000003	-0.000002	-0.001319
9	1.303003	0.021993	-0.025613	0.002297	-0.000000	0.002297

MEMBER END FORCES AND MOMENTS

MEMBER	EAO	RFX			RFX		
		RPX	RPY	RPZ	MPX	MPY	MPZ
1	1	0.0	0.0	0.0	0.0	0.0	0.0
1	2	0.0	0.0	0.0	0.0	0.0	0.0
2	1	0.0	0.0	0.0	0.0	0.0	0.0
2	2	0.0	0.0	0.0	0.0	0.0	0.0
3	1	0.0	0.0	0.0	0.0	0.0	0.0
3	2	0.0	0.0	0.0	0.0	0.0	0.0
4	1	0.0	0.0	0.0	0.0	0.0	0.0
4	2	0.0	0.0	0.0	0.0	0.0	0.0
5	1	0.0	0.0	0.0	0.0	0.0	0.0
5	2	0.0	0.0	0.0	0.0	0.0	0.0
6	1	0.0	0.0	0.0	0.0	0.0	0.0
6	2	0.0	0.0	0.0	0.0	0.0	0.0
7	1	0.0	0.0	0.0	0.0	0.0	0.0
7	2	0.0	0.0	0.0	0.0	0.0	0.0
8	1	0.0	0.0	0.0	0.0	0.0	0.0
8	2	0.0	0.0	0.0	0.0	0.0	0.0
9	1	0.0	0.0	0.0	0.0	0.0	0.0
9	2	0.0	0.0	0.0	0.0	0.0	0.0

		6.746107	7.682724	-1.983243	-0.765847	9.792044	7.490111
10	1	7.284179	1.921782	-0.512036	0.867929	4.973021	11.250444
	2	-6.746107	-2.682773	1.983244	-1.973747	14.767141	18.112961
	3	-7.284179	-1.921782	0.512037	-0.867917	4.645561	22.971924
11	1	-0.000041	14.635203	-15.423984	-3.697471	0.000871	-0.000627
	2	21.219528	1.349178	-0.000041	0.001039	0.000266	-3.697471
	3	-21.219528	-1.349178	0.000041	-0.001039	31.590591	-0.000227
	4						0.001164
12	1	0.000125	0.000014	-11.457994	-35.540527	-0.002698	0.000262
	2	11.151744	-2.631421	0.000195	-0.000875	-0.002566	-35.540527
	3	-0.000125	-0.000000	11.457968	-18.884857	-0.001228	-0.001188
	4	-11.151717	2.631428	-0.000195	0.000875	-0.001469	-18.884857
13	1	-0.000199	0.000032	-11.458273	18.887329	0.001262	-0.001202
	2	11.152137	2.631529	-0.000199	0.000880	0.001505	18.887329
	3	0.000199	-0.000032	11.458380	35.539154	0.002743	0.000257
	4	-11.152111	-2.631554	0.000199	-0.000880	0.002611	35.539154
14	1	-0.000043	14.635247	15.423875	3.696330	-0.000896	-0.000616
	2	11.111677	1.342277	0.000043	-0.001047	-0.000293	-3.696330
	3	0.000043	-14.635247	-15.423875	-3.696330	0.000709	0.001189
	4	-11.111677	-1.342277	-0.000043	0.001047	-0.000601	31.591476
15	1	4.744703	-2.682475	-1.983179	1.975329	14.770069	-18.113297
	2	7.284183	-1.921413	-0.512102	0.868949	4.648252	-22.973618
	3	-4.744704	2.682472	1.983183	0.244565	9.786782	-7.482021
	4	7.284183	1.921411	-0.512101	-0.868952	4.972010	-11.240817
16	1	11.458223	0.000033	0.000193	-0.001173	-0.001207	18.887329
	2	11.152137	2.631535	0.000193	-0.000864	-0.001445	18.887329
	3	-11.458223	-0.000033	-0.000193	0.000254	-0.002685	35.539307
	4	-11.152137	-2.631559	-0.000193	0.000864	-0.002555	35.539307
17	1	6.746425	-2.682429	1.983004	-1.974280	-14.770056	-18.114166
	2	7.284186	-1.921367	0.511962	-0.868008	-4.647614	-22.974365
	3	-6.746425	2.682425	-1.983008	-0.244512	-9.785856	-7.484486
	4	-7.284186	1.921364	-0.511968	0.868012	-4.973169	-11.242542
18	1	1.010426	2.491910	-5.782032	-10.231022	8.586987	1.233372
	2	1.133296	-0.356868	0.825978	3.484186	-13.379631	
	3	-1.010426	-2.491910	5.782032	-5.857308	4.276350	1.897720
	4	-1.133296	0.356869	-0.825980	0.570347	-7.430749	
19	1	1.010268	2.482700	-6.746045	7.490314	-9.792186	-0.265852
	2	1.133276	-0.356948	0.825977	-4.973051	11.250600	
	3	-1.010268	-2.482700	6.746045	-19.113397	-14.767147	-1.973742
	4	-1.133276	0.356949	-0.825975	4.645544	22.971954	
20	1	-1.921333	-2.682399	-6.745420	-18.114243	14.770019	-1.974246
	2	7.284525	-1.921533	-0.511974	0.867998	4.647539	-22.974436
	3	-1.921333	2.682395	6.745465	-7.484817	9.785791	-0.265121
	4	-7.284525	1.921533	0.511983	-0.867999	4.973139	-11.242539
21	1	1.010322	2.482323	-5.792104	13.233362	-8.595277	1.233163
	2	1.133257	-0.356972	0.826019	-0.484316	-13.377570	
	3	-1.010322	-2.482323	5.792104	-8.463580	-4.276014	1.897161
	4	-1.133257	0.356973	-0.826019	0.572261	-7.424979	
22	1	-1.921326	-2.682806	1.920043	-1.222130	-8.585572	10.227966
	2	7.284111	-1.921335	-0.512712	-3.484356	-13.376191	
	3	-1.921326	2.682806	-1.920043	1.222876	-4.273969	5.850520
	4	-7.284111	1.921335	0.512713	-0.570574	-7.422110	
23	1	-10.424039	14.635390	0.000041	-0.000029	0.000079	3.696457
	2	11.111677	1.342279	-0.000041	0.001040	0.001277	-3.696457
	3	-10.424039	-14.635390	-0.000041	0.000029	-0.000222	-31.591660
	4	-11.111677	-1.342279	0.000041	-0.001040	0.001260	31.591660
24	1	-0.000125	0.000014	-11.457973	1.223126	0.585475	10.229215
	2	7.284767	-1.921256	-0.512093	0.867633	4.484309	-13.377337
	3	-0.000125	-0.000014	11.457973	1.223126	-0.585475	5.852529
	4	-7.284767	1.921256	0.512093	-0.867635	0.570342	-7.424916

NUMBER OF NODES 9  
 NUMBER OF MEMBERS 24  
 NUMBER OF LOADING CONDITION 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-1	MEMBER COORDINANTS					
	X(I,1)	X(I,2)	Y(I,1)	Y(I,2)	Z(I,1)	Z(I,2)
1	-32.250000	-18.089996	0.0	6.270000	16.000000	18.089996
2	-36.000000	-18.969986	0.0	8.919999	0.0	0.0
3	-32.250000	-18.089996	0.0	6.270000	-16.000000	-18.089996
4	-16.000000	-18.089996	0.0	6.270000	32.250000	18.089996
5	-18.089996	-18.969986	6.270000	8.919999	18.089996	0.0
6	-18.969986	-18.089996	8.919999	6.270000	0.0	-18.089996
7	-16.000000	-18.089996	0.0	6.270000	-32.250000	-18.089996
8	-18.089996	0.0	6.270000	8.919999	18.089996	18.969986
9	-18.969986	0.0	8.919999	12.000000	0.0	0.0
10	-18.089996	0.0	6.270000	8.919999	-18.089996	-18.969986
11	0.0	0.0	0.0	8.919999	12.000000	18.969986
12	0.0	0.0	12.000000	8.919999	0.0	-18.969986
13	0.0	0.0	0.0	8.919999	-36.000000	-18.969986
14	0.0	18.089996	8.919999	6.270000	18.969986	18.089996
15	0.0	18.969986	12.000000	8.919999	0.0	0.0
16	0.0	18.089996	8.919999	6.270000	-18.969986	-18.089996
17	0.0	18.969986	0.0	6.270000	32.250000	18.089996
18	16.000000	18.969986	6.270000	8.919999	18.089996	0.0
19	18.969986	18.089996	8.919999	6.270000	0.0	-18.089996
20	16.000000	18.089996	0.0	6.270000	-32.250000	-18.089996
21	16.000000	18.089996	0.0	6.270000	16.000000	18.089996
22	32.250000	18.089996	0.0	8.919999	0.0	0.0
23	36.000000	18.089996	0.0	6.270000	-16.000000	-18.089996
24	32.250000	18.089996	0.0	6.270000	0.0	0.0

MEMBER-1	MEMBER PROPERTIES				RADIUS	PHI	ALPHA
	IX	IY	IZ	ARFA			
1	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	0.679990
2	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
3	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	-0.679990
4	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	-0.679990
5	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
6	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
7	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
8	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
9	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
10	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
11	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
12	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
13	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
14	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
15	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
16	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
17	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
18	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135	-0.356173
19	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	0.679990
20	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	0.679990
21	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
22	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	0.679990
23	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877	0.0
24	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550	0.679990

MEMBER NODE (1,1) NODE (1,2)

1	0	1
2	0	2
3	0	3
4	0	1
5	1	3
6	0	2
7	0	4
8	1	3
9	1	5
10	3	4
11	4	5
12	2	6
13	0	7
14	0	6
15	4	8
16	2	9
17	0	0
18	0	7
19	0	0
20	8	0
21	0	0
22	0	7
23	0	0
24	0	9

MEMBER	NODE	1	2	3	4	5	6
1	4	-1	-4	5	9		
2	4	-2	-5	7	9		
3	4	-3	-6	7	15		
4	4	-4	-7	13	16		
5	4	-5	-8	14	17		
6	4	-6	-9	15	17		
7	4	-7	-10	16	23		
8	4	-8	-11	17	23		
9	4	-9	-12	18	24		

NODE	APPLIED NODAL LOADS					
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	-40.000000	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODAL DISPLACEMENTS

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	-0.043077	0.084918	0.042683	0.006593	-0.001564	0.001892
2	-0.032136	0.062454	0.049762	-0.010722	-0.004098	-0.000083
3	0.025088	-0.071427	0.085068	0.003754	-0.000000	-0.003754
4	-0.026281	-0.032331	-0.016428	-0.001516	0.002731	-0.005685
5	-0.018377	-0.028110	-0.018377	0.002813	0.000001	-0.002816
6	0.049761	0.062453	-0.032137	0.000085	0.004097	0.010722
7	-0.024293	-0.049763	-0.024293	-0.003572	-0.000000	-0.003572
8	-0.016424	-0.032330	-0.026281	0.005685	-0.002730	0.001516
9	0.042683	0.084817	-0.043076	-0.001891	-0.001564	-0.006593

MEMBER END FORCES AND MOMENTS

MEMB.	END	PPX	PPY	PPZ	MPX	MPY	MPZ
		PX	PY	PZ	MX	MY	MZ
1	1'		-2.444512	-3.945823	-1.625709	-3.390611	7.864278
1	1	-4.015769	-2.768638	0.629530	-2.609644	-4.646996	-21.219345
1	2'		2.444512	3.945823	1.625709	1.444373	10.045988
1	2	4.015769	2.768638	-0.629530	2.609702	-5.187362	-22.039597
2	1'		-3.699088	-3.252299	-0.895260	-0.758547	8.552631
2	1	-4.785834	-1.164695	-0.895260	3.296360	7.928238	-8.684322
2	2'		3.699088	3.252299	0.895260	-7.277168	6.693660
2	2	4.785834	1.164695	0.895260	-3.296357	9.287841	-13.707257
3	1'		33.887909	20.520157	-7.623210	-1.801078	11.242842
3	1	39.960815	5.439025	1.063071	-1.168472	-8.716777	31.114532
3	2'		-33.887909	-20.520157	7.623210	-3.110323	25.874847
3	2	-39.960815	-5.439025	-1.063071	1.168599	-7.889267	53.868301
4	1'		-2.259562	8.028668	-21.497416	-20.899155	-12.130882
4	1	22.999084	-1.586038	-0.121348	0.776227	2.148938	-24.260910
4	2'		2.259562	-8.028668	21.492416	-0.168851	-0.792101
4	2	-22.999084	1.586038	0.121348	-0.776285	-0.254238	-0.520091
5	1'		-0.972454	3.103307	-23.565868	-4.371877	-2.323667
5	1	23.766327	-0.379839	0.043348	-0.680571	-0.580476	-4.901756
5	2'		0.972440	-3.103322	23.545593	-1.885569	-0.804529
5	2	-23.766052	0.379788	-0.043368	0.680567	-0.212858	-2.050617
6	1'		-0.473727	-0.517383	-25.835571	10.625197	8.845891
6	1	25.587723	3.603690	-0.453644	-1.624153	4.454154	13.015935
6	2'		0.473724	0.517383	25.835587	48.480087	21.554306
6	2	-25.587738	-3.603691	0.453644	1.624122	3.849292	52.947296
7	1'		-7.622706	20.518707	33.884537	-30.264282	-11.243473
7	1	39.957108	5.439045	-1.063102	1.164602	8.717318	31.116364
7	2'		7.622706	-20.518707	-33.884537	-47.814072	-25.873925
7	2	-39.957108	-5.439045	1.063102	-1.168595	7.889935	53.867371
8	1'		-3.731860	0.979253	0.431044	3.097985	-6.929807
8	1	-3.525689	1.624354	0.050116	2.987086	-0.319480	20.535202
8	2'		3.731865	-0.979253	-0.431031	-2.817456	-4.151538
8	2	3.525695	-1.624351	-0.050104	-2.987080	-0.597825	9.196767
9	1'		-4.246271	0.369142	1.390156	-1.514117	-14.735421
9	1	-4.132227	1.044891	1.390156	-3.856092	-14.302307	13.556859
9	2'		4.246210	-0.369173	-1.390156	5.795790	-11.635812
9	2	4.132162	-1.044911	-1.390156	3.856086	-12.414269	6.525080

10	1*		25.837341	0.517596	0.423813	2.442166	-21.554947	-48.480377
	1	25.589493	-3.603790	0.453792	1.623693	-3.849731	-52.947784	
	2*		-25.837387	-0.517583	-0.423786	-0.863574	-8.848263	-10.674490
	2	-25.589539	3.603800	-0.453762	-1.623695	-4.456727	-13.016091	
11	1*		0.392652	-1.012810	2.700112	5.664502	-4.124129	-0.023860
	1	-2.861804	0.355627	0.392652	-1.892407	-3.664398	5.664502	
	2*		-0.392652	1.012810	-2.700112	1.172794	-2.562739	-3.478595
	2	2.861804	-0.355627	-0.392652	1.892407	-3.884206	1.172794	
12	1*		-0.301367	-0.369205	3.157532	0.806022	4.117459	-0.259188
	1	-3.175890	0.141604	-0.301367	0.914728	4.022862	0.806022	
	2*		0.301367	0.369221	-3.157569	1.914948	1.599458	1.186398
	2	3.175828	-0.141583	0.301367	-0.914728	1.768919	1.914948	
13	1*		-1.390332	-0.369305	4.246817	-6.524196	11.633545	-5.795672
	1	-4.132740	-1.045139	-1.390332	3.856333	12.412013	-6.524196	
	2*		1.390332	0.369306	-4.246806	-13.561557	14.741059	1.513452
	2	4.132729	1.045138	1.390332	-3.856339	14.307980	-13.561557	
14	1*		-0.895388	-1.252616	-3.699192	8.685723	-8.553239	0.759118
	1	-4.786073	-1.164927	0.895388	-3.296136	-7.929041	-8.695723	
	2*		0.895388	1.252616	3.699192	13.710475	-6.695219	7.227733
	2	4.786073	1.164927	-0.895388	3.296134	-9.284484	-13.710475	
15	1*		-3.033691	0.335554	-0.026070	0.838656	2.597199	-4.552170
	1	-3.045512	-0.160149	-0.125585	0.671680	0.986200	-5.171779	
	2*		3.033691	-0.335554	0.026072	-0.474318	0.544094	2.582128
	2	3.045512	0.160150	0.125587	-0.671677	-1.312283	-2.239455	
16	1*		-3.156701	0.369057	0.301294	-1.186542	-1.597185	-1.915973
	1	-3.175046	-0.141616	0.301294	-0.915235	-1.766700	-1.915973	
	2*		3.156713	-0.369059	-0.301294	0.258558	-4.118353	-0.905676
	2	3.175057	0.141616	-0.301294	0.915236	-4.023685	-0.805676	
17	1*		23.548645	-3.103660	0.972688	0.715090	0.802632	1.884491
	1	23.769119	0.379855	-0.043238	0.681118	0.211516	2.048955	
	2*		-23.548630	3.103660	-0.972684	-0.561527	-2.323421	4.372601
	2	-23.769104	-0.379855	0.043241	-0.681113	0.579906	4.902340	
18	1*		0.057870	-0.335724	3.003764	9.940258	-4.440490	-2.145207
	1	-2.848842	0.997153	-0.168649	1.491659	1.708550	10.862059	
	2*		-0.057870	0.335724	-3.003764	4.136353	-2.656528	1.089386
	2	2.848842	-0.997153	0.168649	-1.491680	0.925995	4.717206	
19	1*		0.026032	-0.335534	3.033951	-2.592102	-0.543943	2.474387
	1	-3.045762	0.160194	0.125541	-0.671721	-1.312123	-2.239482	
	2*		-0.026032	0.335534	-3.033936	4.551731	-2.596893	-0.939458
	2	3.045753	-0.160194	-0.125547	0.671718	-0.995982	5.171229	
20	1*		-0.431113	-0.979301	3.730601	-8.287552	4.151939	2.017374
	1	-3.524435	-1.624230	-0.050170	-2.987056	0.598192	-2.196560	
	2*		0.431111	0.979301	-3.730605	-10.315491	6.930225	-3.080835
	2	3.524439	1.624229	0.050168	2.987060	0.319795	-20.534288	
21	1*		-1.625732	-3.945951	-2.444767	20.132660	-7.864102	3.390469
	1	-4.016055	-2.764641	-0.629549	2.609566	4.647142	-21.210254	
	2*		1.625732	3.945951	2.444767	-20.497837	-10.045939	-1.443747
	2	4.016055	2.764641	0.629549	-2.609621	5.187717	-22.039923	
22	1*		3.007913	-0.335320	0.058003	2.145107	4.440226	-0.939924
	1	-2.847800	0.997177	0.168658	-1.491631	-1.708595	10.861674	
	2*		-3.007913	0.335320	-0.058003	-1.080358	2.656598	-4.136535
	2	2.847800	-0.997177	-0.168658	1.491659	-0.925130	4.717486	
23	1*		2.609588	-1.012533	0.392669	0.024011	4.124160	-5.664021
	1	-2.861211	0.355622	-0.392669	1.892293	-3.664503	5.664021	
	2*		-2.609588	1.012533	-0.392669	3.478593	2.562987	-1.172914
	2	2.861211	-0.355629	0.392669	-1.892290	3.884624	1.172916	
24	1*		-21.491318	8.028116	-2.259430	3.143342	12.130899	20.899759
	1	22.997849	-1.586067	0.121410	-0.776112	-2.148803	-24.260544	
	2*		21.491318	-8.028116	2.259430	-0.530604	0.792422	2.169221
	2	-22.997849	1.586067	-0.121410	0.776161	0.254490	-0.520474	

IMC2171  
 TRACEBACK FOLLOWS- ROUTINE ISN SEQ. 14  
 TRACOM R200FD00  
 MAIN 00002848

ENTRY POINT= 50005070

NODE	APPLIED NODAL LOADS		APPLIED NODAL LOADS			
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	-20.000000	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	-20.000000	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	-20.000000	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	-20.000000	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	-0.016924	0.030124	0.016924	-0.001354	0.000000	-0.001354
2	-0.036924	-1.071642	0.000000	0.000000	0.000000	0.002922
3	-0.016924	-0.030124	-0.016924	0.001354	-0.000000	-0.001354
4	-0.036924	-1.071642	-0.016924	0.000000	-0.000000	0.000000
5	0.000000	0.127776	-0.000000	0.000000	0.000000	0.000000
6	-0.036924	-0.030124	0.016924	-0.001354	-0.000000	-0.001354
7	-0.016924	0.030124	0.016924	-0.001354	0.000000	0.001354
8	-0.036924	-0.030124	0.000000	0.000000	0.000000	0.000000
9	-0.016924	0.030124	-0.016924	0.001354	-0.000000	-0.001354

MEMBER END FORCES AND MOMENTS

MEMBER	END	MEMBER END FORCES AND MOMENTS						
		PX	PY	PZ	MX	MY	MZ	
1	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
1	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
2	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
2	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
3	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
3	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
4	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
4	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
5	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
5	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
6	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
6	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
7	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
7	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
8	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
8	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	
9	1	11.495365	3.495017	0.922210	-0.595716	7.384768	-12.199625	
9	2	-11.495365	-3.495017	-0.922210	0.595716	-7.384768	12.199625	

	1*		11.858961	3.485193	-1.587151	0.792241	7.019598	9.731461
10	1	12.301044	1.965959	-0.347116	1.331385	3.075422	11.548720	
	2*		-11.858963	-3.485193	1.587148	-1.031242	11.256245	21.889786
	2	-12.301045	-1.965858	0.347113	-1.331382	3.278195	24.435135	
	1*		-0.000004	13.029823	-21.050446	2.837913	0.000040	-0.000018
11	1	24.693039	1.775229	-0.000004	0.000034	0.000027	2.827913	
	2*		0.000004	-13.029823	21.050446	31.290863	0.000024	0.000051
	2	-24.693039	-1.775229	0.000004	-0.000034	0.000045	31.290863	
	1*		-0.000054	0.000103	-17.873749	-35.149750	0.000417	-0.000891
12	1	17.642731	-2.864401	-0.000054	0.000946	0.000769	-35.149750	
	2*		0.000054	-0.000075	17.873764	-19.900879	0.000612	0.001058
	2	-17.642731	2.864431	0.000054	-0.000946	0.000774	-19.900879	
	1*		0.000067	0.000386	-17.873169	19.998972	-0.000711	0.001108
13	1	17.642000	2.864791	0.000067	-0.000979	-0.000880	19.898972	
	2*		-0.000067	-0.000386	17.873169	35.158737	-0.000558	-0.000902
	2	-17.642000	-2.864791	-0.000067	0.000979	-0.000406	35.158737	
	1*		0.000005	13.030338	21.050400	-2.840647	0.000051	-0.000024
14	1	24.693222	1.775706	-0.000005	0.000003	0.000056	2.840647	
	2*		-0.000005	-13.030338	-21.050400	-17.297562	0.000041	-0.000024
	2	-24.693222	-1.775706	0.000005	-0.000003	0.000048	17.297562	
	1*		11.860391	-3.485593	-1.587187	1.029852	11.255795	-21.891113
15	1	12.302617	-1.066025	-0.347018	1.330136	3.277115	-24.436218	
	2*		-11.860391	3.485592	1.587198	-0.791048	7.019375	-9.732294
	2	-12.302617	1.066025	0.347019	-1.330133	3.075113	-11.549433	
	1*		17.873383	0.000366	-0.000075	0.001142	0.000792	19.898895
16	1	17.642303	2.864006	-0.000075	0.001000	0.000965	19.898895	
	2*		-17.873383	-0.000366	0.000075	-0.000910	0.000640	35.158866
	2	-17.642303	-2.864006	0.000075	-0.001000	0.000484	-35.158866	
	1*		11.860166	-3.485241	1.587136	-1.931192	-11.255818	-21.898992
17	1	12.301066	-1.066069	0.347086	-1.331361	-3.278164	-24.434235	
	2*		-11.860166	3.485240	-1.587137	0.792225	-7.019498	-9.731353
	2	-12.301066	1.066069	-0.347087	1.331359	-3.075369	-11.548586	
	1*		0.002035	3.484479	-11.193101	-12.799119	7.384395	0.595721
18	1	11.260000	-1.466098	0.259661	0.711269	-1.189372	-14.651275	
	2*		-0.002035	-3.484479	11.193101	9.033202	2.044457	0.002140
	2	-11.260000	1.466098	-0.259661	-0.711269	1.175399	-9.298931	
	1*		1.489207	3.485198	-11.958994	3.731645	-7.019631	0.792228
19	1	12.301000	1.066065	0.347128	-1.331377	-3.075438	11.548926	
	2*		-1.489207	-3.485213	11.958999	11.889965	-11.254457	-1.931244
	2	-12.301000	-1.066065	-0.347129	1.331376	-3.278239	24.435333	
	1*		-1.489216	-3.485279	-11.959271	-21.039236	11.255799	-1.931192
20	1	12.301000	-1.066079	-0.347090	1.331348	3.076064	-24.434494	
	2*		1.489216	3.485279	11.959371	-0.231253	7.019533	1.792210
	2	-12.301000	1.066079	0.347092	-1.331349	3.075492	-11.548598	
	1*		0.002035	3.484549	11.193199	12.799119	-7.384396	3.595701
21	1	11.260000	-1.466033	-0.259633	0.711299	-1.179317	-14.650768	
	2*		-0.002035	-3.484549	-11.193199	0.033202	2.044482	0.002011
	2	-11.260000	1.466033	0.259633	0.711299	-1.175736	-9.297500	
	1*		-11.193170	3.485073	0.002012	-1.020574	-7.384649	12.792432
22	1	11.260000	-1.466069	-0.259695	-0.711293	1.189334	-14.651655	
	2*		11.193170	-3.485079	-0.002012	-0.033213	-2.044442	0.002067
	2	-11.260000	1.466069	0.259695	0.711289	-1.175891	-9.298276	
	1*		-0.000004	13.030326	-0.000004	-1.000000	0.000004	-0.000004
23	1	24.693037	1.775797	0.000004	-0.000000	0.000000	2.840747	
	2*		0.000004	-13.030326	0.000004	0.000000	-0.000004	-0.000004
	2	-24.693037	-1.775797	-0.000004	0.000000	-0.000000	2.840747	
	1*		-11.193193	3.484592	-0.002083	3.506679	7.383780	12.792542
24	1	11.260000	-1.466033	0.259578	0.711212	-1.189148	-14.648542	
	2*		11.193193	-3.484592	0.002083	0.000000	2.044494	0.002011
	2	-11.260000	1.466033	-0.259578	-0.711212	1.175674	-9.297473	

EXECUTE  
 TRACE BACK (LINES) = 000000  
 PROGRAM = 00000000  
 MAIN = 00000000  
 ENTRY STATE = 00000000

MEMBER NUMBER 15  
 MEMBER NUMBER 24  
 MEMBER NUMBER 11  
 MEMBER NUMBER 12

MEMBER	Y(1,1)	Y(1,2)	MEMBER COORDINANTS Y(1,1)	Y(1,2)	Z(1,1)	Z(1,2)
1	0.0	0.0	0.0	20.784988	0.0	20.784988
2	0.0	0.0	0.0	25.455994	0.0	0.0
3	0.0	0.0	0.0	20.784988	0.0	-20.784988
4	0.0	0.0	0.0	20.784988	36.000000	20.784988
5	0.0	0.0	20.784988	25.455994	20.784988	0.0
6	0.0	0.0	25.455994	20.784988	0.0	-20.784988
7	0.0	0.0	0.0	20.784988	-36.000000	-20.784988
8	0.0	0.0	20.784988	25.455994	20.784988	25.455994
9	0.0	0.0	20.784988	25.455994	-20.784988	-25.455994
10	0.0	0.0	0.0	25.455994	36.000000	25.455994
11	0.0	0.0	25.455994	20.784988	0.0	-25.455994
12	0.0	0.0	0.0	20.784988	25.455994	0.0
13	0.0	0.0	20.784988	25.455994	-20.784988	-25.455994
14	0.0	0.0	0.0	20.784988	25.455994	20.784988
15	0.0	0.0	20.784988	25.455994	0.0	0.0
16	0.0	0.0	20.784988	25.455994	-20.784988	-20.784988
17	0.0	0.0	20.784988	25.455994	20.784988	20.784988
18	0.0	0.0	20.784988	25.455994	0.0	0.0
19	0.0	0.0	20.784988	25.455994	-20.784988	-20.784988
20	0.0	0.0	20.784988	25.455994	20.784988	20.784988
21	0.0	0.0	20.784988	25.455994	0.0	0.0
22	0.0	0.0	20.784988	25.455994	-20.784988	-20.784988
23	0.0	0.0	20.784988	25.455994	20.784988	20.784988
24	0.0	0.0	20.784988	25.455994	0.0	0.0

MEMBER	TY	TY	MEMBER PROPERTIES AREA	PHYS	PHI	ALPHA	
1	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	1.287000
2	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
3	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	-1.287000
4	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	-0.831423
5	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
6	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	1.287000
7	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	-0.831423
8	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	0.0
9	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
10	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	-0.831423
11	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
12	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	0.0
13	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	-0.831423
14	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	1.287000
15	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
16	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	-0.831423
17	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
18	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	1.287000
19	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
20	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	-0.831423
21	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
22	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	1.287000
23	0.000326	0.000326	0.000326	0.062500	36.000000	0.392700	0.0
24	0.000326	0.000326	0.000326	0.062500	36.000000	0.477470	-1.287000

MEMBER NODE (1,1) NODE(1,2)

1	1	2	4
2	1	3	5
3	1	4	6
4	2	4	6
5	2	5	7
6	2	6	8
7	3	6	8
8	3	7	9
9	3	8	10
10	4	8	11
11	4	9	12
12	4	10	13
13	5	10	14
14	5	11	15
15	5	12	16
16	6	12	17
17	6	13	18
18	6	14	19
19	7	14	20
20	7	15	21
21	7	16	22
22	8	16	23
23	8	17	24
24	8	18	25
25	9	18	26
26	9	19	27
27	9	20	28
28	10	20	29
29	10	21	30
30	10	22	31
31	11	22	32
32	11	23	33
33	11	24	34

MEMBER	NODE	LINK	1	2	3	4	5	6
1	1	1	0	0	0	0	0	0
2	1	2	0	0	0	0	0	0
3	1	3	0	0	0	0	0	0
4	2	4	-1	-4	6	0	0	0
5	2	5	-2	-5	6	0	0	0
6	2	6	-3	-6	12	15	0	0
7	3	7	-4	-7	12	15	0	0
8	3	8	-5	-8	18	18	0	0
9	3	9	-6	-9	18	18	0	0
10	4	10	-10	-13	-14	17	0	0
11	4	11	-15	-18	19	23	0	0
12	4	12	-16	-19	20	23	0	0
13	5	13	-20	-24	0	0	0	0
14	5	14	-21	-25	0	0	0	0
15	5	15	-22	-26	0	0	0	0
16	6	16	-23	-27	0	0	0	0
17	6	17	-24	-28	0	0	0	0



NODE	PX	PY	APPLIED NODAL LOADS PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	-40.000000	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.6754088E 39

NODAL DISPLACEMENTS

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.0	0.0	0.0	-0.005634	0.009215	-0.010434
3	0.0	0.0	0.0	-0.000655	0.005948	-0.003976
4	0.0	0.0	0.0	-0.002004	-0.004152	-0.001392
5	-0.158722	0.055776	-0.093064	-0.000074	0.000074	-0.002467
6	-0.010728	0.013363	-0.075798	-0.000222	0.005887	0.003423
7	-0.040652	0.002423	-0.078181	-0.002650	0.001957	0.004048
8	-0.122465	-0.087487	-0.143823	-0.007058	0.003064	-0.015069
9	-0.016548	0.051464	-0.191322	0.004063	0.001118	-0.002985
10	-0.053544	0.034495	-0.073972	-0.003862	-0.003254	0.000010
11	-0.108900	-0.428630	0.046474	-0.020330	-0.004191	0.021766
12	-0.048993	-0.053085	0.054137	0.012072	0.001784	0.007130
13	-0.078500	0.035088	0.048537	-0.000079	-0.007547	0.000317
14	0.0	0.0	0.0	0.018424	0.005641	-0.026546
15	0.0	0.0	0.0	-0.002435	-0.000408	-0.006370
16	0.0	0.0	0.0	0.004436	0.003921	0.005409

MEMBER END FORCES AND MOMENTS

MEMB.	END	PPX	PPY	PPZ	MPX	MPY	MPZ	
		PX	PY	PZ	MX	MY	MZ	
1	1		-1.232122	-2.142327	-2.204646	0.000039	-0.000245	0.000179
1	2	-3.206151	-0.321760	0.003122	-0.000024	0.000012	0.000004	
2	1	1.232121	2.142323	2.204642	-1.205363	7.934389	-6.296151	
2	2	3.206145	0.321759	-0.003122	0.000013	-0.000007	-0.000003	
3	1		-0.567262	-1.409019	0.010423	-0.000010	0.000007	
3	2	-1.518845	-0.015116	0.010423	0.000036	0.000002	-0.000002	
4	1	1.518841	0.015115	-0.010423	-0.000034	-0.000002	0.000002	
4	2		-0.744847	-1.259261	1.305142	0.000013	0.000005	
5	1	-1.052761	-0.175062	0.006876	0.000016	-0.000018	0.000078	
5	2	1.052759	0.174962	-0.006875	-0.000007	0.000007	-0.000032	
6	1		-2.172884	1.562993	-1.127028	-3.047172	-1.593038	
6	2	7.874001	0.274338	-0.395321	-2.989717	6.697121	-3.326223	
7	1		2.172884	-1.562993	1.127028	3.047854	11.228116	
7	2	-7.874001	-0.274338	0.395321	2.089677	-6.307440	11.404393	
8	1		0.473238	0.536191	-2.403889	-3.517875	-14.908717	
8	2	2.304452	-0.701203	0.698533	2.583114	-4.283018	-13.230125	
9	1		-0.473239	-0.536205	2.403869	3.433440	-6.156617	
9	2	-2.304436	0.701188	-0.698544	-2.583107	-6.250441	-2.063440	
10	1		0.324376	-0.470216	-2.143113	-0.568745	1.605123	
10	2	2.212624	0.115891	-0.100602	-0.066683	1.575238	0.767632	
11	1		-0.324407	0.470245	2.143106	0.806800	1.664200	
11	2	-2.212620	-0.115848	0.100602	0.066673	-0.618643	-1.761139	
12	1		-1.259295	1.297213	1.183397	6.665121	6.730654	
12	2	2.149271	-0.221609	0.005584	-0.914732	0.458049	-9.644215	
13	1		1.259295	-1.297213	-1.183397	-1.806348	-1.290929	
13	2	-2.149271	0.221609	-0.005584	0.914726	-0.642868	2.300251	
14	1		-3.879219	-1.115711	-0.227680	1.410549	-4.253512	
14	2	4.134661	0.199180	0.137367	1.742543	0.086741	7.310322	
15	1		3.879238	1.115742	0.227711	-0.532238	5.416380	
15	2	-4.134672	-0.199125	-0.137367	-1.742507	-3.982593	-11.165350	
16	1		-0.418147	-0.402718	-0.251366	-3.131734	4.660124	
16	2	-0.540430	-0.212049	-0.251366	-1.110032	5.503847	0.272751	
17	1		0.413152	0.402720	0.251366	0.481331	-1.734651	
17	2	0.540435	0.212048	0.251366	1.110034	-1.422114	-5.122457	

10	1	-1.767243	-0.014368	-0.061140	0.333039	0.790980	4.726430		
	2		1.679449	0.432070	-0.346071	-0.448097	-3.351555	-3.790487	
	3	1.767217	0.014359	0.061141	-0.333041	0.542393	-5.039484		
11	1		-0.194318	6.252525	-2.189238	-4.512263	0.357020	7.411133	
	2	6.614367	0.370099	-0.194318	-2.506224	6.983638	-4.512263		
	3		0.194318	-6.252525	2.189238	14.709961	1.691861	-2.464594	
	4	-6.614367	-0.370099	0.194318	2.506223	-1.629559	14.709961		
12	1		1.450416	0.324870	-3.385584	-23.487045	-16.929474	-13.838737	
	2	3.252201	-0.995442	1.450416	6.306886	-20.936600	-23.487045		
	3		-1.450416	-0.324890	3.385592	-3.940898	-19.992310	-1.454459	
	4	-3.252216	0.995427	-1.450416	-6.306886	19.027115	-3.940898		
13	1		0.266009	-0.705444	-2.107691	0.833989	-2.108016	-0.097741	
	2	2.217215	0.154817	0.266009	0.896991	-1.910154	0.833989		
	3		-0.266009	0.705413	2.107685	3.431840	-4.663514	2.902540	
	4	-2.217197	-0.154844	-0.266009	-0.896987	-5.419272	3.431840		
14	1		0.150464	1.584593	0.820566	6.665943	1.282538	-2.172465	
	2	1.777988	-0.151720	-0.150464	0.353567	2.497899	-6.665943		
	3		0.150464	-1.584593	-0.820566	-2.485449	0.303053	-1.657744	
	4	-1.777988	0.151720	0.150464	-0.353563	1.647877	2.485449		
15	1		-5.522141	4.812387	0.268787	9.308955	9.824526	27.484390	
	2	-6.351003	1.680551	-3.250528	0.881104	28.229385	11.869898		
	3		5.522217	-4.812345	-0.268746	11.914280	10.387884	46.742691	
	4	6.351059	-1.680471	3.250526	-0.881226	-42.663239	24.775131		
16	1		0.766305	0.627431	-1.528727	2.626074	20.361725	7.674747	
	2	0.467872	0.979019	-1.528727	-5.365779	19.816757	7.674747		
	3		-0.766299	-0.627427	1.528727	13.492826	18.553513	16.377029	
	4	-0.467869	0.979014	1.528727	5.365783	22.304842	16.377029		
17	1		-1.263977	0.447056	-0.941391	-0.498669	7.712433	2.547293	
	2	-1.501972	0.560369	-0.337314	-1.581523	3.169612	7.326010		
	3		1.263950	-0.447072	0.941406	2.807785	5.969400	0.832875	
	4	1.501953	-0.560398	0.337313	1.581527	4.187304	4.894090		
18	1		14.777126	16.999359	-12.215216	-41.049351	62.199356	22.305252	
	2	25.560778	-0.743957	-1.478292	-0.202210	21.836975	-77.054214		
	3		-14.777126	-16.999359	12.215216	45.822235	-33.140640	16.894196	
	4	-25.560778	0.743957	1.478292	0.202243	27.093018	57.430710		
19	1		1.453578	-4.417094	1.584736	-52.740540	-21.095917	-16.118027	
	2	-2.145001	-1.425500	4.193529	-0.452896	-55.118225	-21.159922		
	3		-1.453573	4.417111	-1.584640	-31.665329	-16.520091	-11.303876	
	4	2.144914	1.425544	-4.193537	0.452862	-36.122192	-0.917859		
20	1		0.735400	0.457269	0.405451	9.282838	-5.719547	1.256635	
	2	-0.661844	0.685081	0.181657	-2.627900	2.041363	10.566021		
	3		-0.735428	-0.457299	-0.405481	-1.672421	-7.671998	-0.657452	
	4	0.661884	-0.685111	-0.181654	2.627904	-6.003284	4.375328		
21	1		0.055140	-0.158724	-0.139066	-0.816839	2.455376	-2.733126	
	2	-0.128974	0.111074	-0.136427	-0.227461	2.588849	2.722603		
	3		-0.055140	0.158724	0.139066	0.341357	-1.274013	-1.711039	
	4	0.128974	-0.111074	0.136427	0.227448	-1.926678	0.952127		
22	1		7.800955	-13.772197	-13.531759	-4.996105	43.850632	-47.502989	
	2	20.731567	-1.205998	0.454573	-0.000031	-15.043879	-63.077011		
	3		-7.800969	13.772193	13.531742	-0.000179	-0.001265	0.001132	
	4	-20.731552	1.205987	-0.454579	-0.000002	0.000051	0.001209		
23	1		1.484360	-4.247066	-0.349324	8.990245	3.685341	-7.329816	
	2	4.491747	-0.255157	-0.349324	0.000007	2.623836	-7.029816		
	3		-1.484360	4.247059	0.349324	0.000031	0.000023	-2.002145	
	4	-4.491740	0.255155	0.349324	-0.000010	-0.000038	-0.000045		
24	1		-0.472502	0.745572	-0.674516	-1.474487	0.447469	1.538910	
	2	-1.108963	0.036402	0.055897	0.000007	-1.849809	1.138474		
	3		0.472501	-0.745569	0.674514	-0.000007	0.000032	0.000022	
	4	1.108960	-0.036402	-0.055896	-0.000009	0.000008	-0.000032		

IHC217F  
 TRACEBACK FOLLOWS- ROUTINE ISN REG. 14  
 IACOM 8200FNCO  
 MAIN 0000284P  
 ENTRY POINT= 50005000

NODE	APPLIED NODAL LOADS					
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	-40.000000	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.6754088E 39

NODAL DISPLACEMENTS

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.0	0.0	0.0	-0.004438	0.003920	-0.005411
2	0.0	0.0	0.0	0.002438	-0.000400	0.006372
3	0.0	0.0	0.0	-0.018429	0.059558	0.025565
4	-0.078613	0.036910	-0.048579	0.000078	-0.007487	-0.000718
5	0.040880	-0.053064	-0.054171	-0.012077	0.001787	-0.007130
6	-0.198081	-0.428638	-0.046506	0.073923	-0.004193	-0.021766
7	0.053607	0.034506	0.073923	0.003863	-0.003267	-0.000010
8	0.016542	0.051468	0.101349	-0.004064	0.001118	0.002984
9	-0.102527	-0.087505	0.143858	0.007059	0.003062	0.015070
10	0.040697	0.002422	0.078237	0.002551	0.001957	-0.004950
11	-0.010733	0.013364	0.052956	0.000221	0.005885	-0.003423
12	-0.159281	0.045780	0.003172	0.003824	0.000425	0.007468
13	0.0	0.0	0.0	0.002006	-0.004151	0.001389
14	0.0	0.0	0.0	0.000655	0.005551	0.003977
15	0.0	0.0	0.0	0.005639	0.009221	0.010439

MEMBER END FORCES AND MOMENTS

MEMB.	END	MEMBER END FORCES AND MOMENTS						
		PX	PY	PZ	MX	MY	MZ	
1	1*		-0.472051	-0.745210	-0.674093	0.000014	-0.000074	0.000000
1	2*	-1.108273	-0.034295	-0.055917	-0.000005	-0.000004	0.000004	-1.526844
2	1*		0.472051	0.745209	0.674090	1.478179	0.444796	-1.135176
2	2*	1.108270	0.034294	-0.055918	0.000004	-1.850805	-1.135176	0.000168
3	1*		1.483054	4.246929	-0.349334	-0.000055	0.000025	0.000168
3	2*	4.491102	0.255034	-0.349334	0.000002	0.000060	0.000168	7.027060
4	1*		1.483052	4.246927	-0.349334	-8.892600	1.683355	0.000168
4	2*	-4.491185	-0.255033	0.349334	-0.000006	9.625251	7.027069	7.027060
5	1*		7.795214	13.773074	-13.531968	0.000163	-0.001465	-1.000130
5	2*	20.730515	1.005673	0.454680	-0.000098	0.000044	-0.001891	47.514374
6	1*		-7.795208	-13.773055	13.531965	5.011091	43.846268	0.000168
6	2*	-20.730499	-1.005673	-0.454669	0.000061	-15.049174	63.077271	0.000168
7	1*		-0.054928	-0.159117	0.139395	0.818727	2.458030	0.000168
7	2*	-0.129505	0.111138	-0.136539	-0.228122	2.500897	2.726195	0.000168
8	1*		0.054928	0.159117	-0.139395	-0.342366	1.274919	0.000168
8	2*	0.129505	-0.111138	0.136539	0.228115	1.028327	0.953298	0.000168
9	1*		0.735975	-0.457291	0.406527	1.673833	-7.576162	0.000168
9	2*	-0.642997	-0.685166	0.182046	-2.626976	-6.007672	-4.377371	0.000168
10	1*		-0.735975	0.457276	-0.406530	-0.280033	-5.722392	0.000168
10	2*	0.642996	0.685156	-0.182056	2.626982	2.037427	-11.566085	0.000168
11	1*		1.453811	4.417094	1.586323	31.568198	-16.521835	11.304440
11	2*	-2.146464	1.424860	4.183681	-0.452409	-36.125488	0.919588	0.000168
12	1*		-1.453838	-4.417067	-1.586385	52.731476	-21.194752	16.118222
12	2*	2.146511	-1.424804	-4.183678	0.452498	-55.117880	21.152222	0.000168
13	1*		-14.775113	16.997787	12.213729	41.063248	62.193161	-20.302162
13	2*	25.566833	-0.743848	-1.478622	-0.204081	21.841675	-77.046127	0.000168
14	1*		14.775113	-16.997787	-12.213729	-45.823334	-33.134979	-16.882435
14	2*	-25.566833	0.743848	1.478622	0.204072	27.099335	52.426514	0.000168
15	1*		-1.263768	-0.446968	-0.941808	-2.810047	5.955952	-0.943626
15	2*	-1.501843	-0.567662	-0.337665	-1.582548	4.191443	-4.800416	0.000168
16	1*		1.263782	0.446998	0.941836	0.478691	7.714294	-2.546022
16	2*	1.501869	0.567696	0.337663	1.582535	3.172655	-7.320127	0.000168
17	1*		0.765459	-0.627217	-1.529228	-13.495960	18.540425	-16.374557
17	2*	0.466991	-0.872860	-1.529228	-5.366035	22.312225	-16.374557	0.000168
18	1*		-0.765472	0.627708	1.529228	-2.628229	20.367560	-2.675522
18	2*	-0.466997	0.872857	1.529228	5.366024	19.822983	-7.675522	0.000168

10	1'		-5.526794	-4.813293	0.268973	-11.919114	10.396805	-46.741486
	1	-6.355673	-1.679865	-3.251022	0.878271	42.669739	-24.767827	
	2'		5.526932	4.813259	-0.268941	-9.307311	9.826946	-27.487778
11	1'		-0.150593	1.585025	-0.820841	-6.668329	1.283508	2.174337
	1	1.778493	-0.151808	-0.150593	0.353741	2.500000	-6.668329	
	2'		0.150593	-1.585025	0.820841	2.485435	0.304341	1.659144
12	1'		0.266383	0.705784	-2.108379	-3.431993	-4.668794	-2.905354
	1	2.217980	-0.154767	0.266383	0.897566	-5.425227	-3.431993	
	2'		-0.266383	-0.705773	2.108375	-0.832590	-2.112740	0.096617
13	1'		1.450819	-0.325080	-3.386505	3.940749	-19.996994	1.455318
	1	3.253132	0.995601	1.450819	6.307845	-19.031784	3.940749	
	2'		-1.450819	0.325096	3.386495	23.490524	-16.935028	13.942131
14	1'		0.194513	6.253847	2.189587	4.512067	0.358154	-7.414121
	1	6.615723	0.370281	-0.194513	-2.506320	6.986833	-4.512067	
	2'		-0.194513	-6.253847	-2.189587	-14.714828	1.892791	2.462593
15	1'		-1.679699	0.432182	0.345951	0.448082	-3.350984	3.791692
	1	-1.767453	0.014275	-0.061302	0.332645	0.543594	5.039907	
	2'		1.679664	-0.432166	-0.345951	-0.045365	4.006431	-2.655100
16	1'		-0.419094	0.403023	-0.251645	-0.490533	1.741712	6.123472
	1	-0.541339	0.212002	-0.251645	-1.110468	1.425247	6.123472	
	2'		0.419019	-0.403026	0.251645	3.133876	4.664155	-0.282123
17	1'		-3.881577	1.116187	-0.928168	0.532098	5.416811	11.184505
	1	-4.137095	0.199025	0.137357	1.742393	-3.282709	11.654199	
	2'		3.881604	-1.116228	0.928225	-1.410290	-4.255327	-6.115310
18	1'		1.259159	1.297402	-1.183649	-6.658314	5.733206	2.079874
	1	2.149546	-0.221762	0.005517	-0.915338	0.459244	-9.447777	
	2'		-1.259159	-1.297402	1.183649	1.896157	-1.289203	-1.285031
19	1'		0.324642	0.470257	-2.143541	-0.805772	1.660836	0.261412
	1	2.213098	-0.115810	-0.100444	-0.066191	0.616970	-1.758134	
	2'		-0.324643	-0.470247	2.143541	0.568529	1.603016	4.185554
20	1'		0.473485	-0.536391	-2.404829	-3.435434	-6.158470	-0.555156
	1	2.305335	0.701533	0.698835	2.583836	-6.253266	2.063390	
	2'		-0.473531	0.536336	2.404806	3.519097	-14.015198	5.272222
21	1'		2.173549	1.563548	1.127316	3.047155	-1.522518	-1.533869
	1	2.865003	0.274496	-0.395391	-2.090044	6.488466	-2.226666	
	2'		-2.173549	-1.563548	-1.127316	-3.405279	11.232045	-4.145376
22	1'		-0.745410	1.259472	1.305398	-0.255371	-4.378395	2.679721
	1	-1.953245	0.175053	0.006900	0.000010	-0.228367	5.293321	
	2'		0.745420	-1.259423	-1.305397	0.000010	0.000012	-0.000017
23	1'		-0.567912	1.409674	0.010430	-0.265480	-0.110292	2.412724
	1	-1.519695	0.015198	0.010430	0.000035	-0.297401	0.418724	
	2'		0.567910	-1.409670	-0.010430	0.000038	0.000035	-0.000036
24	1'		-1.233493	2.143239	-2.205620	1.296339	2.937430	4.265000
	1	-1.207803	0.321925	0.001157	-0.000028	-0.124395	10.454911	
	2'		1.233492	-2.143237	2.205619	-0.000029	-2.000027	-0.000027



		0.087255	0.302844	-0.938466	-2.531481	9.256627	1.640639
10	1	0.368289	0.878999	-0.366585	-0.781427	5.728345	7.833314
	2	-0.087246	-0.392850	0.938474	-0.017185	9.841491	6.116443
	2	-0.368282	-0.879010	0.366586	0.781428	2.266905	11.336523
	1	0.000018	20.327377	-8.541472	-38.874161	0.000138	-0.000407
11	1	22.048706	-0.112499	0.000018	0.000283	-0.000324	-38.874161
	2	-0.000018	-20.327377	8.541472	38.875085	-0.000330	-0.000057
	2	-22.048706	0.112499	-0.000018	-0.000283	-0.000179	38.875085
	1	0.000315	1.113007	-6.663794	-35.811737	-0.003959	-0.002092
12	1	6.582482	-1.521792	0.000315	-0.000418	-0.004458	-35.811737
	2	-0.000315	-1.113018	6.663803	-6.118742	-0.004069	-0.001233
	2	-6.582496	1.521795	-0.000315	-0.000418	-0.004231	-6.118742
	1	-0.000315	-1.113026	-6.663864	6.118708	0.004066	-0.001235
13	1	6.582450	1.521792	-0.000315	-0.000415	0.004229	6.118708
	2	0.000315	1.113035	6.663852	35.811874	0.003950	-0.002095
	2	-6.582467	-1.521799	0.000315	0.000415	0.004447	35.811874
	1	0.000019	20.327332	-8.541446	38.874023	-0.000137	-0.000420
14	1	22.048653	-0.112492	-0.000019	-0.000287	0.000336	-38.874023
	2	-0.000019	-20.327332	8.541446	38.875089	0.000336	-0.000060
	2	-22.048653	0.112492	0.000019	0.000287	0.000194	38.875089
	1	0.006043	-0.392426	-0.939043	0.019273	9.845930	-6.114370
15	1	0.368007	-0.873341	-0.367139	-0.780823	2.271887	-11.338306
	2	-0.006043	0.392426	0.939043	2.533980	9.265638	-1.639239
	2	-0.368043	0.873355	0.367138	0.780821	5.735404	-7.839184
	1	2.493362	1.112850	0.000003	-0.000043	-0.000050	19.825021
16	1	1.066260	1.077566	0.000003	-0.000021	-0.000063	19.825027
	2	-0.493362	-1.112852	-0.000003	0.000003	0.000003	34.674193
	2	-1.066239	-1.077552	0.000003	0.000021	-0.000027	34.674193
	1	0.007269	-0.392494	0.939000	-0.019279	-0.946084	-6.114523
17	1	0.368303	-0.873327	0.367137	0.780820	-1.271896	-11.338524
	2	-0.007269	0.392494	-0.939008	-2.533965	-2.265262	-1.639352
	2	-0.368342	0.873343	-0.367137	-0.780810	-5.735432	-7.839470
	1	1.066336	1.061830	-1.503774	-6.391310	0.734442	3.297694
18	1	0.493371	-0.393322	-0.163312	0.594292	-11.960133	
	2	-1.066336	-1.061830	1.503774	6.391333	0.021326	3.111721
	2	-0.493387	0.393322	0.163312	-0.594298	4.199767	-0.825046
	1	1.066376	1.061888	-2.477204	1.224136	-6.249985	-2.892411
19	1	0.116533	0.077222	0.000217	1.739123	-0.240222	6.022172
	2	-0.077222	-0.116533	-0.000217	2.677221	11.240364	-11.822344
	2	-0.116556	-0.077222	0.000226	-1.739150	1.224265	16.358529
	1	-0.000217	-0.000206	-0.000206	-2.677222	-11.240364	11.822344
20	1	0.116713	-0.077222	-0.000206	-1.739149	-0.240222	-16.358526
	2	-0.000217	-0.000206	0.000206	2.677221	11.240364	11.822344
	2	-0.116736	0.077222	0.000206	1.739152	0.240222	16.358527
	1	1.066391	1.061872	1.003793	6.391310	-0.734442	3.297694
21	1	0.493376	-0.393322	0.163311	-0.594292	-11.960132	
	2	-1.066376	-1.061872	-1.003792	6.391332	0.021326	3.111721
	2	-0.493396	0.393322	0.163311	-0.594292	4.199767	-0.825046
	1	0.007269	-0.392492	0.939002	-0.019279	-0.946084	-6.114523
22	1	-0.167402	0.143922	0.124430	-0.333321	-0.333321	0.000000
	2	0.143922	-0.167402	-0.124430	0.333321	0.333321	0.000000
	2	0.167402	-0.143922	0.124430	-0.333321	-0.333321	0.000000
	1	0.000001	-16.413202	-0.000001	0.000001	0.000001	-38.874161
23	1	17.11213	-1.601200	-0.000001	0.000001	-38.874161	-0.000001
	2	-0.000001	16.413202	0.000001	-0.000001	0.000001	38.874161
	2	-17.11213	1.601200	0.000001	-0.000001	0.000001	-38.874161
	1	0.000001	0.000001	-0.000001	0.000001	0.000001	0.000001
24	1	-0.167471	0.143926	0.124443	-0.333326	-0.333326	0.000000
	2	0.143926	-0.167471	-0.124443	0.333326	0.333326	0.000000
	2	0.167471	-0.143926	0.124443	-0.333326	-0.333326	0.000000

NUMBER OF NODES 15  
 NUMBER OF MEMBERS 24  
 NUMBER OF LOADING CONDITION 2  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	MEMBER COORDINANTS		MEMBER COORDINANTS			
	X(I,1)	X(I,2)	Y(I,1)	Y(I,2)	Z(I,1)	Z(I,2)
1	-33.369995	-18.369995	0.0	9.740000	13.500000	18.369995
2	-36.000000	-20.129990	0.0	13.250000	0.0	0.0
3	-33.369995	-18.369995	0.0	9.740000	-13.500000	-18.369995
4	-13.500000	-18.369995	0.0	9.740000	13.369995	18.369995
5	-18.369995	-20.129990	9.740000	13.250000	18.369995	0.0
6	-20.129990	-18.369995	13.250000	9.740000	0.0	-18.369995
7	-13.500000	-18.369995	0.0	9.740000	-33.369995	-18.369995
8	-18.369995	0.0	9.740000	13.250000	18.369995	20.129990
9	-20.129990	0.0	13.250000	18.000000	0.0	0.0
10	-18.369995	0.0	9.740000	13.250000	-18.369995	-20.129990
11	0.0	0.0	0.0	13.250000	36.000000	20.129990
12	0.0	0.0	13.250000	18.000000	20.129990	0.0
13	0.0	0.0	18.000000	13.250000	0.0	-20.129990
14	0.0	0.0	0.0	0.0	-36.000000	-20.129990
15	0.0	18.369995	13.250000	9.740000	20.129990	18.369995
16	0.0	20.129990	18.000000	13.250000	0.0	0.0
17	0.0	18.369995	13.250000	9.740000	-20.129990	-18.369995
18	13.500000	18.369995	0.0	9.740000	33.369995	18.369995
19	18.369995	20.129990	9.740000	13.250000	18.369995	0.0
20	20.129990	-18.369995	13.250000	9.740000	0.0	-18.369995
21	13.500000	18.369995	0.0	9.740000	-33.369995	-18.369995
22	18.369995	33.369995	9.740000	0.0	18.369995	13.500000
23	20.129990	36.000000	13.250000	0.0	0.0	0.0
24	18.369995	33.369995	9.740000	0.0	-18.369995	-13.500000

MEMBER-I	MEMBER PROPERTIES				RADIUS	PHI	ALPHA
	IX	IY	IZ	AREA			
1	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	0.761490
2	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870	0.0
3	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	-0.761490
4	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	-0.761490
5	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	-0.505970
6	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	-0.505970
7	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	0.761490
8	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	0.505970
9	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870	0.0
10	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	-0.505970
11	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870	0.0
12	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870	0.0
13	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	0.505970
14	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	0.505970
15	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	0.761490
16	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	0.505970
17	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	0.761490
18	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	0.505970
19	0.000550	0.000326	0.000326	0.062500	45.000000	0.210200	0.505970
20	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	-0.761490
21	0.000550	0.000326	0.000326	0.062500	45.000000	0.231870	0.0
22	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	-0.761490
23	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	-0.761490
24	0.000550	0.000326	0.000326	0.062500	45.000000	0.207430	-0.761490

MEMBER I J K

1	1	2	3
2	2	3	4
3	3	4	5
4	4	5	6
5	5	6	7
6	6	7	8
7	7	8	9
8	8	9	10
9	9	10	11
10	10	11	12
11	11	12	13
12	12	13	14
13	13	14	15
14	14	15	16
15	15	16	17
16	16	17	18
17	17	18	19
18	18	19	20
19	19	20	21
20	20	21	22
21	21	22	23
22	22	23	24
23	23	24	25
24	24	25	26

NODE	MEMBER	LINK					
		1	2	3	4	5	6
1	1	1	0	0	0	0	0
2	1	2	0	0	0	0	0
3	1	3	0	0	0	0	0
4	1	4	-1	-4	5	6	7
5	2	4	-2	-5	6	7	8
6	2	5	-3	-6	7	8	9
7	2	6	-4	-7	8	9	10
8	2	7	-5	-8	9	10	11
9	2	8	-6	-9	10	11	12
10	2	9	-7	-10	11	12	13
11	2	10	-8	-11	12	13	14
12	2	11	-9	-12	13	14	15
13	2	12	-10	-13	14	15	16
14	2	13	-11	-14	15	16	17
15	2	14	-12	-15	16	17	18
16	2	15	-13	-16	17	18	19
17	2	16	-14	-17	18	19	20
18	2	17	-15	-18	19	20	21
19	2	18	-16	-19	20	21	22
20	2	19	-17	-20	21	22	23
21	2	20	-18	-21	22	23	24
22	2	21	-19	-22	23	24	25
23	2	22	-20	-23	24	25	26
24	2	23	-21	-24	25	26	27
25	2	24	-22	-25	26	27	28
26	2	25	-23	-26	27	28	29
27	2	26	-24	-27	28	29	30

NODE	PX	PY	APPLIED NODAL LOADS PZ	LOADS MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.5073614E 67

NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.0	0.0	0.0	0.006576	-0.002586	0.010161
2	0.0	0.0	0.0	-0.004542	-0.012898	0.003400
3	0.0	0.0	0.0	0.005852	-0.000714	-0.007455
4	-0.0088825	0.008096	0.008066	0.005951	0.003334	0.000784
5	-0.043170	0.048367	0.085477	-0.011205	-0.006980	-0.000602
6	0.136753	-0.152276	0.131245	0.005813	0.000951	-0.004698
7	-0.060974	-0.028105	-0.019541	-0.001174	0.004289	-0.005080
8	-0.026624	-0.021367	-0.020879	0.001576	-0.000157	-0.001588
9	0.091051	0.042268	-0.034825	0.000026	0.006010	0.009155
10	-0.065801	-0.062582	-0.058780	-0.003962	0.001319	0.002546
11	-0.025399	-0.027227	-0.056831	-0.006730	-0.005321	0.000861
12	0.002234	0.006584	-0.007954	-0.002072	0.000025	-0.003369
13	0.0	0.0	0.0	-0.004837	-0.002784	0.000801
14	0.0	0.0	0.0	0.003401	-0.008278	0.002889
15	0.0	0.0	0.0	-0.004065	-0.007016	-0.011159

MEMBER	END	PX	PY	PZ	MX	MY	MZ	
1	1		-3.769684	-3.473620	-1.902687	0.000253	-0.000486	-0.000275
	2	-6.375718	-0.997498	0.058928	-0.900123	-0.000593	0.000096	
	3		3.769684	3.473618	1.902686	-1.616362	10.184112	-15.389237
2	1	6.375716	0.997496	-0.058927	0.900126	-1.091418	-18.492325	
3	1		-2.537621	-2.729042	-0.396336	0.000055	-0.000059	-0.000356
	2	-3.696982	-0.468531	-0.396336	0.000005	-0.000081	-0.000356	
4	1		2.537613	2.729045	0.396336	-5.251510	6.280916	-0.686096
	2	3.696973	0.468531	0.396336	-0.000013	8.193995	-9.686096	
5	1		26.313859	19.634888	-10.377169	0.000486	-0.000850	-0.000549
	2	36.337906	2.557581	0.028209	0.000091	-0.000507	-0.000999	
6	1		-26.313858	-19.634888	10.377168	-5.453407	27.511169	38.225876
	2	-36.337906	-2.557581	-0.028210	-0.000115	-0.520401	47.408325	
7	1		-2.307817	6.026206	-11.227514	-10.265274	-16.572144	-5.952598
	2	12.856752	1.502920	0.333874	0.442278	-2.729108	-25.761810	
8	1		2.307817	-6.026206	11.227514	0.303367	-3.593675	-1.985448
	2	-12.856752	-1.502920	-0.333874	-0.442277	-3.460343	-2.095682	
9	1		-1.484395	2.291820	-13.485654	-1.720194	2.774769	1.347021
	2	13.755288	-0.154090	-0.306566	-0.630783	3.467402	-0.182524	
10	1		1.484393	-2.291821	13.485670	-3.516204	0.795439	-3.154127
	2	-13.755283	0.154094	0.306565	0.630815	2.292459	-2.714314	
11	1		-0.929581	-0.362033	-15.696216	6.530437	12.423164	0.325517
	2	15.330907	3.397881	-0.976307	-2.214621	8.417733	12.131426	
12	1		0.929581	0.362094	15.696214	-41.914474	31.112778	-6.193466
	2	-15.330916	-3.397827	0.976336	2.214595	8.043213	51.699295	
13	1		-11.322347	20.756302	26.841446	-9.592992	-6.947486	3.702203
	2	25.602448	3.654202	0.133982	1.847703	-1.290071	12.134844	
14	1		11.322347	-20.756302	-26.841446	-40.406097	-32.170020	6.487971
	2	-25.602448	-3.654202	-0.133982	-1.847712	-1.194248	51.893127	
15	1		-4.581454	0.262292	0.354015	3.032467	-9.364960	15.933122
	2	7.308077	1.347807	0.155615	2.708426	-1.242336	18.490097	
16	1		4.581455	-0.262288	-0.354018	-2.251474	-5.201298	4.965055
	2	-7.308079	-1.347805	-0.155612	-2.708429	-1.681124	6.827350	
17	1		-3.104267	-0.075094	1.813293	2.237558	-20.508881	9.516465
	2	3.038539	0.639846	1.813293	-2.532304	-20.474579	9.516465	
18	1		3.104254	0.075085	-1.813293	6.375579	-15.992710	3.717672
	2	-3.038526	-0.639843	-1.813293	2.532294	-17.029465	3.717672	



10	1*		14.065642	0.025189	0.763056	3.941886	-26.452484	-39.518356
	1	13.688166	-3.273151	0.588408	2.614664	-5.104645	-47.371719	
	2*		-14.065641	-0.025193	-0.763059	-1.219215	-12.320159	-9.388641
	2	-13.688165	3.273149	-0.588413	-2.614690	-5.948832	-14.113627	
11	1*		0.691235	0.523497	-0.226753	2.792375	-6.381434	-3.527856
	1	0.509570	0.256524	0.691235	-1.381771	-7.159554	2.792375	
	2*		-0.691235	-0.523497	0.226753	2.511161	-4.588459	-5.630998
	2	-0.509570	-0.256524	-0.691235	1.381772	-7.131119	2.511161	
12	1*		-0.715335	0.046862	-0.151646	-1.304646	8.943169	1.445222
	1	0.158355	0.010782	-0.715335	0.647288	9.036042	-1.304646	
	2*		0.715335	-0.046856	0.151628	-1.527449	5.456499	1.952611
	2	-0.158336	-0.010781	0.715335	-0.647285	-5.759091	-1.527449	
13	1*		-1.884956	-0.456676	0.825265	-5.269773	16.966095	-5.829310
	1	-0.698327	-0.633999	-1.884956	1.777081	17.851364	-5.269773	
	2*		1.884956	0.456692	-0.825275	-7.842919	20.978027	-3.124274
	2	0.698334	0.634017	1.884956	-1.777081	21.134811	-7.842919	
14	1*		-1.193804	-1.387649	-0.541168	6.364200	-10.428412	5.848132
	1	-1.304755	-0.718363	1.193804	-2.194364	-11.753184	-6.364200	
	2*		1.193804	1.387649	0.541168	-8.487003	-8.517257	0.969761
	2	1.304755	0.718363	-1.193804	2.194365	-12.927673	-8.487003	
15	1*		-3.172791	0.738409	0.278634	1.045334	0.847259	-0.780014
	1	-3.266800	0.101744	-0.085219	0.937013	0.582281	-1.096334	
	2*		3.172791	-0.738365	-0.278633	-0.723710	-0.381359	3.207912
	2	3.266777	-0.101709	0.085199	-0.937021	1.019740	3.007320	
16	1*		-1.934738	0.428409	0.836201	-2.632951	-6.429882	0.158514
	1	-1.081414	-0.027373	0.836201	-1.085894	-6.862704	0.158514	
	2*		1.934759	-0.428440	-0.836201	-1.438994	-10.402830	-0.724331
	2	1.081441	0.027347	-0.836201	1.085902	-10.432294	-0.724331	
17	1*		10.084177	-1.818010	1.046360	0.575659	-0.139565	3.545489
	1	11.179286	0.244058	0.123385	0.827503	-1.187927	2.175712	
	2*		-10.084146	1.817960	-1.046358	-1.048676	3.242158	2.613109
	2	-11.179246	-0.244103	-0.128412	-0.827515	-1.223576	2.406859	
18	1*		0.402625	0.268620	1.304331	10.762491	-8.860394	-3.406609
	1	-0.756188	1.243668	0.171972	1.003410	-1.657234	14.240145	
	2*		-0.402625	-0.268629	-1.304331	5.071143	-6.636247	-1.133051
	2	0.756188	-1.243668	-0.171972	-1.001293	-1.529653	8.812662	
19	1*		-0.273848	-0.273818	0.990803	-5.739258	4.741283	0.263587
	1	-1.046578	-0.152900	-0.111606	-0.591097	1.595636	-7.311392	
	2*		0.273847	0.273819	-0.990805	4.928359	-1.510693	-0.502349
	2	1.046580	0.152899	0.111605	0.591103	-0.501013	4.309938	
20	1*		-1.103673	-0.799845	2.027081	-5.359961	8.721959	3.295792
	1	-1.730514	-1.675388	-0.575335	-1.915083	4.259148	-8.884466	
	2*		1.103673	0.799800	-2.027080	-16.435028	15.072946	-3.273312
	2	1.730521	1.625349	0.575558	1.915082	5.261114	-21.650650	
21	1*		-2.067520	-2.541950	-0.511667	16.510031	-12.433234	5.033932
	1	-2.322069	-2.404210	0.202344	1.304288	-2.192992	-21.274222	
	2*		2.067500	2.541952	0.521667	17.038432	-15.069682	2.022830
	2	2.322069	2.404210	-0.202344	-1.304381	-1.553381	-22.295724	
22	1*		-0.203850	1.287120	0.523338	0.402220	0.374838	-3.373822
	1	-0.699416	-0.206350	-0.032623	0.000075	0.249966	-4.223422	
	2*		0.203860	-1.286966	-0.522977	-0.000070	0.000230	0.371418
	2	0.699187	0.206373	0.032606	-0.000019	-0.000457	-0.001230	
23	1*		-1.102350	0.947495	-0.107991	2.622382	3.143662	0.633639
	1	-1.453461	0.023822	-0.102792	-0.000002	4.001269	0.433533	
	2*		1.102379	-0.947510	0.107891	-0.000032	-0.000031	-0.000135
	2	1.453472	-0.023822	0.102791	0.000001	-0.000052	0.000135	
24	1*		7.815637	-5.290246	2.554481	0.444596	-3.294715	-1.278322
	1	0.778373	-0.086327	-0.059843	-0.000022	1.111114	-1.492156	
	2*		-7.815377	5.290047	-2.554388	-0.000104	0.000000	0.000135
	2	-0.778034	0.086209	0.059933	0.000048	-0.000273	0.000000	

NODE	APPLIED NODAL LOADS					
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	-20.000000	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	-20.000000	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	-20.000000	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	-20.000000	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.5073614E 67

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.0	0.0	0.0	-0.001365	-0.005138	0.003649
2	0.0	0.0	0.0	0.000003	0.000003	0.001003
3	0.0	0.0	0.0	0.001362	0.005136	-0.003651
4	-0.028400	0.010147	-0.025360	-0.002729	0.000925	-0.003647
5	-0.024304	-0.073555	-0.000003	0.000003	0.000002	0.000927
6	-0.028402	0.010151	-0.025361	0.002727	-0.000928	-0.003647
7	-0.000003	-0.078982	-0.057823	0.001403	-0.000002	0.000003
8	-0.000005	-0.142320	0.000003	-0.000000	0.000000	0.000001
9	-0.000004	-0.078983	0.057821	-0.001403	0.000002	0.000003
10	0.000000	0.010150	0.025364	-0.002729	-0.000923	0.003644
11	-0.054222	-0.073552	-0.000004	0.000003	-0.000002	-0.000928
12	0.028403	0.010157	-0.025366	-0.002728	0.000925	-0.003644
13	0.0	0.0	0.0	-0.001365	-0.005137	-0.003648
14	0.0	0.0	0.0	0.000003	-0.000003	-0.000924
15	0.0	0.0	0.0	0.001362	-0.005135	-0.003650

MEMBER	END	MEMBER END FORCES AND MOMENTS					
		DPY	DPX	DPZ	MPY	MPX	MPZ
1	1	1.845486	0.829156	0.303601	0.010338	-0.000077	-0.000322
1	2	-0.369333	-0.010080	-0.000005	-0.000243	-0.000209	-
1	3	-1.245822	-0.329156	-0.303601	-1.076106	4.440665	-5.566171
1	4	-0.000000	0.000000	0.000000	0.000000	0.000000	-
2	1	13.911349	13.183346	0.000030	-0.010007	0.000008	0.000007
2	2	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
2	3	-13.911349	-13.183346	-0.000030	0.010007	-0.000008	-0.000007
2	4	-1.463587	-0.000030	-0.000000	-0.000000	-0.000000	-0.000000
3	1	1.245822	0.329156	0.303601	1.076106	0.000000	0.000000
3	2	-0.369333	-0.010080	-0.000005	-0.000243	-0.000209	-0.000322
3	3	-1.245822	-0.329156	-0.303601	-1.076106	4.440665	-5.566171
3	4	-0.000000	0.000000	0.000000	0.000000	0.000000	-
4	1	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
4	2	-1.463587	-0.000030	-0.000000	-0.000000	-0.000000	-0.000000
4	3	13.911349	13.183346	0.000030	-0.010007	0.000008	0.000007
4	4	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
5	1	1.245822	0.329156	0.303601	1.076106	0.000000	0.000000
5	2	-0.369333	-0.010080	-0.000005	-0.000243	-0.000209	-0.000322
5	3	-1.245822	-0.329156	-0.303601	-1.076106	4.440665	-5.566171
5	4	-0.000000	0.000000	0.000000	0.000000	0.000000	-
6	1	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
6	2	-1.463587	-0.000030	-0.000000	-0.000000	-0.000000	-0.000000
6	3	13.911349	13.183346	0.000030	-0.010007	0.000008	0.000007
6	4	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
7	1	1.245822	0.329156	0.303601	1.076106	0.000000	0.000000
7	2	-0.369333	-0.010080	-0.000005	-0.000243	-0.000209	-0.000322
7	3	-1.245822	-0.329156	-0.303601	-1.076106	4.440665	-5.566171
7	4	-0.000000	0.000000	0.000000	0.000000	0.000000	-
8	1	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
8	2	-1.463587	-0.000030	-0.000000	-0.000000	-0.000000	-0.000000
8	3	13.911349	13.183346	0.000030	-0.010007	0.000008	0.000007
8	4	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
9	1	1.245822	0.329156	0.303601	1.076106	0.000000	0.000000
9	2	-0.369333	-0.010080	-0.000005	-0.000243	-0.000209	-0.000322
9	3	-1.245822	-0.329156	-0.303601	-1.076106	4.440665	-5.566171
9	4	-0.000000	0.000000	0.000000	0.000000	0.000000	-
10	1	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000
10	2	-1.463587	-0.000030	-0.000000	-0.000000	-0.000000	-0.000000
10	3	13.911349	13.183346	0.000030	-0.010007	0.000008	0.000007
10	4	1.463587	0.000030	0.000000	0.000000	0.000000	0.000000

			2.563194	1.744685	-1.549685	-0.202108	10.548908	8.028318
10	1	2.977766	1.687197	-0.549277	1.021251	5.358814	12.083651	
	2		-2.563193	-1.744684	1.549686	-2.166577	13.407743	15.025677
	3	-2.977765	-1.687197	0.549278	-1.021240	4.959042	19.611160	
	1		-0.000042	17.243790	-18.370392	-5.713165	0.000920	-0.000705
11	1	25.153107	1.463253	-0.000042	0.001130	0.000254	-5.713165	
	2		0.000042	-17.243790	18.370392	35.966478	-0.000253	0.001262
	3	-25.153107	-1.463253	0.000042	-0.001130	0.000615	35.966478	
	1		0.000224	0.732236	-15.271321	-40.299973	-0.003074	-0.000092
12	1	15.031312	-2.794537	0.000224	-0.000616	-0.003012	-40.299973	
	2		-0.000224	-0.732222	15.271311	-17.498795	-0.001440	-0.000973
	3	-15.031299	2.794546	-0.000224	0.000616	-0.001625	-17.498795	
	1		-0.000232	-0.732198	-15.271647	17.501511	0.001511	-0.000995
13	1	15.031621	2.794647	-0.000232	0.000622	0.001699	17.501511	
	2		0.000232	0.732198	15.271754	40.298798	0.003168	-0.000109
	3	-15.031725	-2.794672	0.000232	-0.000622	0.003108	40.298798	
	1		-0.000047	17.243820	18.370193	5.711428	-0.000965	-0.000681
14	1	25.152069	1.463403	0.000047	-0.001141	-0.000305	-5.711428	
	2		0.000047	-17.243820	-18.370193	-35.965820	0.000222	0.001301
	3	-25.152069	-1.463403	-0.000047	0.001141	-0.000664	35.965820	
	1		2.563405	-1.744665	-1.549714	2.168245	13.410933	-15.025157
15	1	2.977773	-1.687058	-0.549409	1.022227	4.962375	-19.612228	
	2		-2.563406	1.744662	1.549710	0.201071	10.545690	-8.022341
	3	-2.977765	1.687049	0.549415	-1.022222	5.358613	-12.076849	
	1		0.000049	0.732100	0.000234	-0.001312	-0.001519	22.266296
16	1	15.030299	2.771568	0.000049	-0.000933	-0.001731	22.266296	
	2		-0.000049	-0.732100	0.000234	0.000707	-0.003199	35.058838
	3	-15.030373	-2.771632	-0.000049	0.000933	-0.000056	35.058838	
	1		2.563061	-1.744611	1.549505	-2.167085	-13.410897	-15.026303
17	1	2.977422	-1.687135	0.549241	-1.021216	-4.961568	-19.613220	
	2		-2.563022	1.744607	-1.549510	-0.201594	-10.544530	-8.025287
	3	-2.977461	1.687120	-0.549267	1.021219	5.358359	-12.078826	
	1		1.563069	3.259423	-7.701447	-11.091574	19.227633	1.464444
18	1	0.201294	-1.153221	-0.143174	1.277954	1.169702	-15.050815	
	2		-1.563065	-3.259423	7.701446	-4.555291	3.923882	2.648171
	3	-0.201334	1.153281	0.143174	-1.277956	1.484123	-4.279031	
	1		3.273444	3.242500	-0.248600	5.403251	-10.031623	-0.188443
19	1	15.032410	1.096233	0.610730	-1.191470	-4.124852	2.551765	
	2		-3.273444	-3.242500	0.248600	19.252827	-15.982183	-2.436402
	3	-15.032427	-1.096234	-0.610730	1.191456	-5.247537	24.379990	
	1		-0.273185	-3.242246	-0.248051	-19.053375	15.984887	-2.437126
20	1	15.031822	-1.096086	-0.610698	1.191412	5.248733	-24.381775	
	2		0.273180	3.242242	0.248007	-5.248326	19.055224	-2.437524
	3	-15.031874	1.096046	0.610633	-1.191617	6.121319	-2.564086	
	1		1.153224	3.258766	7.701396	11.092540	-19.226375	1.464083
21	1	0.201512	-1.153226	-0.143145	-1.277913	-1.169703	-15.048450	
	2		-1.153224	-3.258765	7.701396	4.555291	3.923122	2.647538
	3	-0.201512	1.153227	0.143165	1.277906	-1.556131	-4.284582	
	1		1.242232	-0.242263	-0.243632	-1.275039	-4.427234	5.562837
22	1	15.030602	3.259049	0.012087	0.000035	-2.243228	2.148217	
	2		-1.242232	0.242262	0.243628	1.275019	-0.000223	0.000532
	3	-15.030611	-3.259042	-0.012072	-0.000036	2.243217	0.000643	
	1		13.412832	-14.123760	-0.000022	5.233425	0.000538	-10.185974
23	1	13.412830	-14.463929	-0.000032	0.000001	0.001652	-32.185974	
	2		-13.412844	13.133222	0.000022	0.000000	0.000022	-0.000224
	3	-13.412833	1.563089	0.000022	-0.000001	0.001321	-0.000224	
	1		1.045266	-0.827766	0.303442	1.276621	4.430259	5.565214
24	1	0.0000747	0.394001	-0.010055	-0.000031	0.353433	7.191025	
	2		-1.045189	0.827766	-0.303412	-0.122231	0.000182	0.001645
	3	-0.000065	-0.393919	0.010052	0.000056	-0.353272	0.000156	

NUMBER OF NODES 15  
 NUMBER OF MEMBERS 24  
 NUMBER OF LOADING CONDITION 2  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	MEMBER COORDINANTS		MEMBER COORDINANTS		MEMBER COORDINANTS	
	X(I,1)	X(I,2)	Y(I,1)	Y(I,2)	Z(I,1)	Z(I,2)
1	-32.250000	-18.089996	0.0	6.270000	16.000000	18.089996
2	-36.000000	-18.969986	0.0	8.919999	0.0	0.0
3	-32.250000	-18.089996	0.0	6.270000	-16.000000	-18.089996
4	-16.000000	-18.089996	0.0	6.270000	32.250000	18.089996
5	-18.969986	-18.969986	6.270000	8.919999	18.089996	0.0
6	-18.969986	-18.089996	8.919999	6.270000	0.0	-18.089996
7	-16.000000	-18.089996	0.0	6.270000	-32.250000	-18.089996
8	-18.089996	0.0	6.270000	8.919999	18.089996	18.969986
9	-18.969986	0.0	8.919999	12.000000	0.0	0.0
10	-18.089996	0.0	6.270000	8.919999	-18.089996	-18.969986
11	0.0	0.0	0.0	8.919999	36.000000	18.969986
12	0.0	0.0	8.919999	12.000000	18.969986	0.0
13	0.0	0.0	12.000000	8.919999	0.0	-18.969986
14	0.0	18.089996	8.919999	6.270000	18.969986	18.089996
15	0.0	18.969986	6.270000	8.919999	0.0	-18.969986
16	0.0	18.089996	0.0	6.270000	32.250000	18.089996
17	0.0	18.969986	6.270000	8.919999	18.089996	0.0
18	16.000000	18.089996	8.919999	6.270000	-18.089996	-18.969986
19	18.089996	18.969986	6.270000	8.919999	0.0	0.0
20	18.969986	18.089996	0.0	6.270000	-32.250000	-18.089996
21	16.000000	18.089996	8.919999	6.270000	18.089996	18.969986
22	18.089996	32.250000	6.270000	8.919999	0.0	0.0
23	18.969986	36.000000	8.919999	0.0	-18.089996	-16.000000
24	18.089996	32.250000	6.270000	0.0	0.0	0.0

MEMBER-I	MEMBER PROPERTIES			RADIUS	PHI	ALPHA
	IX	IY	IZ			
1	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
2	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
3	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
4	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
5	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
6	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
7	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
8	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
9	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
10	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
11	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
12	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
13	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
14	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
15	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
16	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
17	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
18	0.000550	0.000326	0.000326	0.062500	60.000000	0.153135
19	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
20	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
21	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
22	0.000550	0.000326	0.000326	0.062500	60.000000	0.160877
23	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550
24	0.000550	0.000326	0.000326	0.062500	60.000000	0.130550

MEMBER	NODE (I,1)	NODE (I,2)
1	1	4
2	2	5
3	3	6
4	4	5
5	5	6
6	4	5
7	5	6
8	6	7
9	7	8
10	8	9
11	9	10
12	10	11
13	11	12
14	12	13
15	13	14
16	14	15
17	15	16
18	16	17
19	17	18
20	18	19
21	19	20
22	20	21
23	21	22
24	22	23

MEMBER	NODE	LINK					
		1	2	3	4	5	6
1	1	1	0	0	0	0	0
2	1	2	0	0	0	0	0
3	1	3	0	0	0	0	0
4	4	-1	-4	5	6	0	0
5	4	-2	-5	6	5	0	0
6	4	-3	-6	5	4	0	0
7	4	-4	-7	6	4	0	0
8	4	-5	-8	7	5	0	0
9	4	-6	-9	8	6	0	0
10	4	-7	-10	9	7	0	0
11	4	-8	-11	10	8	0	0
12	4	-9	-12	11	9	0	0
13	4	-10	-13	12	10	0	0
14	4	-11	-14	13	11	0	0
15	4	-12	-15	14	12	0	0
16	4	-13	-16	15	13	0	0
17	4	-14	-17	16	14	0	0
18	4	-15	-18	17	15	0	0
19	4	-16	-19	18	16	0	0
20	4	-17	-20	19	17	0	0
21	4	-18	-21	20	18	0	0
22	4	-19	-22	21	19	0	0
23	4	-20	-23	22	20	0	0
24	4	-21	-24	23	21	0	0

NODE	PX	PY	APPLIED NODAL LOADS PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	-20.000000	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	-20.000000	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	-20.000000	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	-20.000000	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODAL DISPLACEMENTS						
NODE	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.0	0.0	0.0	-0.001178	-0.003630	0.005247
2	0.0	0.0	0.0	0.000000	0.000000	-0.002066
3	0.0	0.0	0.0	0.001178	0.003630	0.005247
4	-0.019556	0.029857	0.017311	-0.001646	0.000524	-0.003012
5	-0.038413	-0.073163	0.000000	0.000000	0.000000	0.003433
6	-0.019555	0.029855	-0.017310	0.001646	-0.000524	-0.003012
7	-0.000000	-0.079756	-0.039954	0.003260	0.000000	0.000000
8	0.000003	0.140372	-0.000001	-0.000003	0.000000	0.000003
9	0.000000	-0.079750	0.039953	-0.003264	0.000000	0.000000
10	0.019554	0.029852	0.017309	-0.001646	-0.000523	0.003010
11	-0.038408	-0.073154	0.000000	0.000000	0.000000	-0.003436
12	0.019552	0.029851	-0.017308	0.001646	0.000523	0.003009
13	0.0	0.0	0.0	-0.001177	0.003631	-0.005249
14	0.0	0.0	0.0	0.000000	-0.000000	0.002065
15	0.0	0.0	0.0	0.001177	-0.003631	-0.005249

MEMBER END FORCES AND MOMENTS							
MEMB.	END	PX	PPX PY	PPY PZ	PPZ MX	MPX MY	MPY MZ
1	1*		6.414230	2.376843	0.775358	0.000120	-0.000577
1	1	6.869677	-0.428816	0.128724	-0.000035	-0.000091	0.000977
1	2*		-6.414230	-2.376848	-0.775360	-0.106511	2.427423
1	2	-6.869678	0.428811	-0.128723	0.000030	-2.010393	-6.700938
2	1*		20.396713	12.456002	0.000001	0.000001	-0.000003
2	1	23.847687	1.570237	0.000001	-0.000000	-0.000003	-0.001247
2	2*		-20.396713	-12.455998	-0.000001	0.000000	-0.000016
2	2	-23.847687	-1.570233	-0.000001	0.000000	-0.000018	30.189926
3	1*		6.413498	2.376454	-0.775234	-0.000111	0.000554
3	1	6.868839	-0.428870	-0.128749	0.000042	0.000096	0.000814
3	2*		-6.413498	-2.376459	0.775236	0.106511	-2.427374
3	2	-6.868842	0.428867	0.128748	-0.000037	-2.010305	-6.700654
4	1*		-0.962855	3.908360	-12.713599	-13.168047	-8.553738
4	1	12.764388	-1.552371	-0.186787	-1.417705	1.386847	-15.572110
4	2*		0.962855	-3.908360	12.713599	-8.066319	-3.338361
4	2	-12.764388	1.552371	0.186787	1.417691	1.530615	-8.678862
5	1*		-1.764144	3.614352	-12.788038	9.120300	8.270018
5	1	13.246457	2.004331	-0.473251	2.055943	4.365232	11.402997
5	2*		1.764140	-3.614367	12.788030	-22.375885	12.389114
5	2	-13.246450	-2.004345	0.473243	-2.055943	4.297020	-25.285477
6	1*		1.764136	-3.614450	-12.788450	-22.375153	-12.388640
6	1	13.246887	-2.004354	0.473212	-2.055858	-4.296812	-25.284637
6	2*		-1.764137	3.614449	12.788459	22.375294	12.389026
6	2	-13.246887	2.004354	-0.473213	2.055856	-4.365160	-11.402950
7	1*		-0.962899	3.908433	12.713453	13.167350	8.553323
7	1	12.764291	-1.552236	0.186761	1.417583	-1.386720	-15.573117
7	2*		0.962899	-3.908433	-12.713453	8.065355	3.337047
7	2	-12.764291	1.552236	-0.186761	-1.417571	-1.530425	-8.678824
8	1*		7.218300	2.672415	1.347970	-0.948276	-7.359927
8	1	7.585526	1.837928	0.380260	-1.535222	-3.369890	11.658109
8	2*		-7.218300	-2.672415	-1.347975	2.168694	-10.672935
8	2	-7.585533	-1.837925	-0.380257	1.535224	-3.590325	21.994116
9	1*		16.863495	-0.316916	0.000071	-0.000973	-0.007579
9	1	16.594727	-3.015415	0.000071	-0.001053	-0.000416	-35.800720
9	2*		-16.863449	0.316958	-0.000071	0.001192	0.007720
9	2	-16.594681	3.015451	-0.000071	0.001053	-0.000953	-22.151749

	1*		7.216971	2.672085	-1.347951	0.949527	7.360441	9.723665
10	1	7.584164	1.837814	-0.380341	1.536551	3.370300	11.657937	
	2*		-7.216996	-2.672085	1.347947	-2.170156	10.673151	19.490005
	2	-7.584188	-1.837810	0.380338	-1.536547	3.591393	21.987407	
	1*		-0.000000	14.973040	-24.327194	2.635213	0.000051	-0.000124
11	1	28.497345	1.976242	-0.000000	0.000134	-0.000012	2.635213	
	2*		0.000000	-14.973040	24.327194	35.358109	-0.000045	0.000127
	2	-28.497345	-1.976242	0.000000	-0.000134	0.000019	35.358109	
	1*		-0.000088	0.316885	-21.627853	-39.693695	0.000667	-0.000847
12	1	21.399094	-3.153358	-0.000088	0.000943	0.000523	-39.693695	
	2*		0.000088	-0.316843	21.627869	-20.909607	0.000993	0.001116
	2	-21.399094	3.153402	0.000088	-0.000943	0.001159	-20.909607	
	1*		0.000087	-0.316319	-21.627167	20.907593	-0.000988	0.001113
13	1	21.398315	3.153807	0.000087	-0.000940	-0.001154	20.907593	
	2*		-0.000087	0.316319	21.627167	39.703842	-0.000661	-0.000845
	2	-21.398315	-3.153807	-0.000087	0.000940	-0.000517	39.703842	
	1*		-0.000001	14.973688	24.327225	-2.638282	-0.000056	-0.000125
14	1	28.497681	1.976802	0.000001	-0.000136	0.000008	2.638282	
	2*		0.000001	-14.973688	-24.327225	-35.365646	0.000041	0.000132
	2	-28.497681	-1.976802	-0.000001	0.000136	-0.000025	35.365646	
	1*		7.219108	-2.672731	-1.348034	2.168503	10.672476	-19.491516
15	1	7.586371	-1.838118	-0.380207	1.535083	3.589998	-21.983597	
	2*		-7.219108	2.672730	1.348035	-0.948152	7.360659	-9.726625
	2	-7.586371	1.838117	0.380208	-1.535081	3.369702	-11.660798	
	1*		16.862793	0.317490	-0.000071	0.001188	0.000769	22.149872
16	1	16.593948	3.015870	-0.000071	0.001049	0.000950	22.149872	
	2*		-16.862793	-0.317490	0.000071	-0.000970	0.000573	35.810684
	2	-16.593948	-3.015870	0.000071	-0.001049	0.000411	35.810684	
	1*		7.217083	-2.672293	1.347979	-2.169954	-10.672629	-19.489349
17	1	7.584306	-1.837999	0.380296	-1.536390	-3.591101	-21.981613	
	2*		-7.217084	2.672293	-1.347980	0.949385	-7.361288	-9.726664
	2	-7.584307	1.838000	-0.380298	1.536389	-3.370087	-11.661044	
	1*		0.962949	3.908484	-12.213516	-13.166635	9.552585	0.280122
18	1	12.764375	-1.552195	0.186780	1.416759	-1.386557	-15.577410	
	2*		-0.962949	-3.908484	12.213516	-8.064511	3.337667	1.851063
	2	-12.764375	1.552195	-0.186780	-1.416741	-1.530503	-8.674858	
	1*		1.764014	3.614139	-12.788005	9.118540	-8.269420	1.312907
19	1	13.246387	2.004093	0.473203	-2.056368	-4.364355	11.400818	
	2*		-1.764011	-3.614139	12.787995	22.373596	-12.388311	-2.807117
	2	-13.246377	-2.004093	-0.473200	2.056370	-4.297120	25.283051	
	1*		-1.763966	-3.614096	-12.787572	-22.372467	12.387708	-2.806980
20	1	13.245951	-2.004104	-0.473170	2.056269	4.296924	-25.281799	
	2*		1.763967	3.614096	12.787572	-9.118999	8.269450	1.312823
	2	-13.245951	2.004104	0.473171	-2.056268	4.364240	-11.401259	
	1*		0.963056	3.908768	12.213729	13.166097	-8.552268	0.280223
21	1	12.764696	-1.552018	-0.186732	-1.416614	1.386433	-15.576806	
	2*		-0.963056	-3.908768	12.213729	-8.063417	-3.337125	1.850864
	2	-12.764696	1.552018	0.186732	1.416605	1.530289	-8.673678	
	1*		6.415092	-2.376989	-0.775359	-0.106752	-2.429104	6.563810
22	1	6.870515	0.429056	-0.128757	0.000034	2.011040	6.704569	
	2*		-6.415198	2.377052	0.775373	0.000107	0.000484	-0.000607
	2	-6.870640	-0.429046	0.128747	-0.000016	0.000038	-0.000783	
	1*		20.396296	-12.456316	-0.000001	0.000009	0.000018	-30.199341
23	1	23.847458	-1.577133	-0.000001	-0.000001	0.000020	-30.199341	
	2*		-20.396271	12.456341	0.000001	-0.000001	-0.000004	0.001414
	2	-23.847443	1.570742	0.000001	0.000001	-0.000004	0.001414	
	1*		6.414659	-2.376788	0.775298	0.106725	2.428953	6.563425
24	1	6.870033	0.429061	-0.128766	-0.000040	-2.010976	6.704225	
	2*		-6.414765	2.376851	-0.775312	-0.000048	-0.000307	-0.000290
	2	-6.870157	-0.429050	-0.128755	0.000041	-0.000071	-0.000417	

IHC2171  
 TRACKBACK FOLLOWS- ROUTINE ISN RFG. 14  
 JRCOM R200FDC0  
 MAIN 00002848

ENTRY POINT= 50005020

NODE	APPLIED NODAL LOADS					
	PX	PY	PZ	MX	MY	MZ
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	-40.000000	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.0

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.0	0.0	0.0	0.008642	-0.001862	0.014070
2	0.0	0.0	0.0	-0.010961	-0.012649	0.005225
3	0.0	0.0	0.0	0.005178	-0.005230	-0.018277
4	-0.066522	0.118756	0.060233	0.008793	0.003384	0.000448
5	-0.029103	0.069149	0.066235	-0.015488	-0.005841	-0.000987
6	0.114802	-0.212704	0.107008	0.005287	0.001728	-0.000886
7	-0.042379	-0.038829	-0.019660	-0.001638	0.003528	-0.007318
8	-0.022584	-0.029071	-0.020897	0.003202	-0.000990	-0.002208
9	0.070531	0.068911	-0.035547	0.000812	0.004895	0.012411
10	-0.043993	-0.080460	-0.040197	-0.005447	0.001143	0.003215
11	-0.022822	-0.039540	-0.041035	0.003604	-0.004433	0.000970
12	0.068947	0.172864	-0.083257	-0.003071	-0.000183	-0.005643
13	0.0	0.0	0.0	-0.005695	-0.001477	0.000465
14	0.0	0.0	0.0	0.007324	-0.007757	-0.003883
15	0.0	0.0	0.0	-0.004306	-0.006046	-0.013970

MEMBER	END	MEMBER END FORCES AND MOMENTS						
		PPX	PPY	PPZ	MPX	MPY	MPZ	
		PX	PY	PZ	MX	MY	MZ	
1	1	-4.773315	-3.579529	-1.318680	0.000348	-0.001587	0.002029	
1	2	-5.037089	-1.399140	0.349231	-0.000050	0.002597		
1	3	4.773345	3.579529	1.318676	-0.788355	8.696993	-23.749094	
1	4	5.938016	1.398126	-0.349230	0.000093	-5.451365	-21.841843	
1	5	-4.767577	-3.214030	-0.426699	0.000013	-0.000031	0.001312	
2	1	-4.716367	-0.634100	-0.426699	-0.000002	-0.000033	0.001312	
2	2	4.759567	3.214018	0.426699	-3.806175	7.266719	-12.102650	
2	3	5.716364	0.634098	-0.426699	8.203174	-12.102650		
3	1	36.046036	19.160843	-7.060614	0.000417	-0.002136	-0.002655	
3	2	3.282272	0.440081	-0.000112	-0.000136	-0.003419		
3	3	-36.046036	-19.160843	7.060607	-4.225394	74.640045	45.294434	
3	4	-41.295776	-3.282256	-0.440081	0.000122	-6.848835	51.279000	
4	1	-1.852825	7.718354	-22.239838	-27.009209	-16.848114	-3.322517	
4	2	23.497457	-2.747852	-0.073890	-0.091881	2.210124	-31.933335	
4	3	1.852825	-7.718354	22.239838	-3.137664	-3.396043	-1.141057	
4	4	-23.497457	2.740852	0.073890	0.091813	-1.067335	-4.640525	
5	1	-1.233211	3.257465	-23.931244	0.050918	1.648178	3.643331	
5	2	24.182053	-0.206736	-0.150261	-0.400593	1.620221	0.617271	
5	3	1.233196	-3.257481	23.931213	-4.541579	-0.398525	-0.242796	
5	4	-24.182022	0.206720	0.150263	0.400577	1.160705	-4.402455	
6	1	-1.008451	-0.174896	-26.179565	10.464497	12.533992	-0.835307	
6	2	25.850005	4.173110	-0.883187	-0.476281	8.120737	14.331723	
6	3	1.008448	0.174895	26.179581	56.567150	28.766567	-1.099022	
6	4	-25.850021	-4.173112	0.883193	0.476779	8.044713	62.055374	
7	1	-8.450715	21.535355	33.751022	-38.817139	-17.997335	3.118874	
7	2	8.493944	-0.999346	-0.105568	8.717378	41.938112		
7	3	40.354797	6.493944	-21.535355	-33.751022	-54.493561	-31.134523	
7	4	-40.354797	-6.493944	0.999346	0.105508	6.892439	62.655524	
8	1	-5.393247	0.884357	0.376129	3.876795	-4.967721	21.365129	
8	2	5.179070	1.770164	0.023378	3.846212	0.292426	22.356627	
8	3	5.393225	-0.884371	-0.376114	-3.657512	-4.608171	0.031651	
8	4	-5.179047	-1.770160	-0.023352	-3.846203	-0.630810	13.044971	
9	1	-4.992771	0.219953	1.813953	-2.321373	-12.492249	13.249057	
9	2	-4.991014	1.016947	-5.400914	-18.779907	13.249057		
9	3	4.990714	-0.219998	-1.813953	7.908345	-15.007956	4.276211	
9	4	4.990950	-1.016981	-1.813953	5.400907	-16.081390	6.276211	

	1*		26.584991	0.516701	0.507666	3.169601	-22.253540	-49.162628
10	1	26.324234	-3.747191	0.528679	2.274275	-4.376266	-53.832199	
	2*		-26.585022	-0.518671	-0.507638	-1.369552	-10.325038	-11.940347
	2	-26.324265	3.747214	-0.528642	-2.274291	-5.301225	-14.757328	
	1*		0.642986	-0.578029	2.085026	6.299972	-6.442738	-0.728144
11	1	-2.115202	0.455383	0.642986	-2.344328	-6.045097	6.299972	
	2*		-0.642986	0.578029	-2.085026	2.455194	-4.507311	-5.007281
	2	2.115202	-0.455383	-0.642986	2.344326	-6.316081	2.455194	
	1*		-0.526411	-0.305105	2.452864	-0.293638	6.632126	-0.272168
12	1	-2.470057	0.091942	-0.526411	1.331535	6.502786	-0.293638	
	2*		0.526411	0.305122	-2.452915	2.060197	3.353888	1.893513
	2	2.470110	-0.091934	0.526411	-1.331533	-3.613998	2.060197	
	1*		-1.849249	-0.541700	3.639183	-7.314548	15.672317	-6.789939
13	1	-3.505321	-1.117026	-1.849248	4.189494	16.557755	-7.314548	
	2*		1.849248	0.541702	-3.639171	-14.170905	19.389931	1.096333
	2	3.505318	1.117026	1.848248	-4.189494	18.962616	-14.170905	
	1*		-1.238339	-3.099888	-3.174291	10.039996	-11.585178	1.906051
14	1	-4.250229	-1.273182	1.238339	-3.686916	-11.147024	-10.039996	
	2*		1.238339	3.099888	3.174291	14.437375	-10.503772	0.139935
	2	4.250229	1.273182	-1.238339	3.686911	-12.659657	-14.437375	
	1*		-4.212677	0.611032	0.008518	1.426015	2.473985	-3.754078
15	1	-4.252266	-0.072546	-0.182320	1.300811	1.244094	-6.383097	
	2*		4.212662	-0.611032	-0.008515	-0.980898	1.078641	3.644971
	2	4.252231	0.072545	0.182323	-1.300813	2.093102	3.055453	
	1*		-3.662637	0.456247	0.628317	-2.553224	-4.018289	-1.380281
16	1	-3.695128	-0.137725	0.628317	-1.975617	-4.321473	-1.380281	
	2*		3.669449	-0.456249	-0.628317	0.718691	-7.902875	-1.266589
	2	3.695167	0.137725	-0.628317	1.975620	-7.683576	-1.266589	
	1*		23.487091	-3.123865	0.974602	1.103935	0.442264	1.209833
17	1	23.711304	0.251072	-0.049684	1.109232	-0.004900	1.761475	
	2*		-23.487091	3.123865	-0.974598	-0.237415	2.592582	4.220637
	2	-23.711304	-0.251074	0.049688	-1.109215	0.898871	4.664440	
	1*		0.467701	0.286367	3.092228	15.744151	-8.073024	-2.628220
18	1	-2.624572	1.711922	-0.208612	1.248525	2.273523	17.698412	
	2*		-0.467701	0.286367	-3.092228	7.695243	-5.012175	0.224333
	2	2.624572	-1.711922	0.208612	-1.248576	0.884439	9.050134	
	1*		-0.002171	-0.320294	2.680505	-6.779153	2.225617	0.252622
19	1	-2.709273	0.015907	0.326950	-0.745412	-0.120294	-7.134499	
	2*		0.002155	0.320309	-2.680497	6.822243	-2.953221	-0.855027
	2	2.709256	-0.015925	-0.324969	0.745411	-0.545722	7.422662	
	1*		-0.054318	-1.421861	3.521440	-9.399330	3.305252	1.296011
20	1	-1.228402	-2.181867	-0.392010	-2.380728	3.399536	-11.339464	
	2*		0.054316	1.421861	-3.521445	-25.654449	13.002515	-1.078925
	2	3.228406	2.181866	0.392008	2.380725	3.873223	-78.607132	
	1*		-2.171917	-4.554594	-1.395789	29.736222	-13.691168	3.936512
21	1	-3.374522	-3.042522	-0.685042	1.924523	5.279541	-31.529622	
	2*		2.171917	4.554594	1.385789	27.090258	-16.165594	2.262323
	2	3.374522	3.042522	0.685042	-1.924533	6.422254	-22.231209	
	1*		-3.657150	1.299814	-0.419273	0.066123	1.207824	-4.188567
22	1	-3.886912	-0.305513	0.003655	-0.000055	-1.462814	-4.771229	
	2*		3.657640	-1.288243	-0.419254	-0.000279	-0.000347	2.202634
	2	3.887359	0.305529	-0.003616	0.000019	-0.000243	2.000522	
	1*		-2.902530	1.487093	-0.208403	1.859955	3.549111	3.225349
23	1	-3.172521	0.015992	-0.208403	-0.000002	4.226684	2.225349	
	2*		2.902497	-1.487076	0.208403	-0.000001	-0.000012	3.002647
	2	3.172554	-0.014922	0.208403	0.000005	-0.000011	3.000647	
	1*		20.369884	-9.100273	3.113073	-0.492737	-1.525192	-1.202634
24	1	22.517882	-0.122384	0.039232	-0.000082	-0.592440	-1.015525	
	2*		-20.361374	9.100486	-3.113136	-0.000251	-0.001232	-1.202634
	2	-22.518402	0.122376	-0.039225	0.000010	-0.000272	-0.001425	



APPENDIX B

COMPUTER SOLUTIONS FOR THE EXPERIMENTAL MODEL

NUMBER OF NODES 4  
 NUMBER OF MEMBERS 12  
 NUMBER OF LOADING CONDITION 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	X(I,1)	Y(I,2)	MEMBER COORDINANTS		Z(I,1)	Z(I,2)
			Y(I,1)	Y(I,2)		
1	-20.260000	-7.200000	0.0	6.250000	5.870000	7.210000
2	-20.260000	-7.200000	0.0	6.250000	-6.420000	-7.330000
3	-20.260000	-7.200000	0.0	6.250000	20.969986	7.219999
4	-20.260000	-7.200000	6.250000	6.250000	7.210000	-7.330000
5	-6.250000	-7.200000	0.0	6.250000	-20.969986	-7.330000
6	-7.200000	-7.219999	6.250000	6.250000	7.219999	7.290000
7	-7.200000	-7.190000	6.250000	6.250000	-7.330000	-7.190000
8	-7.200000	-7.210000	0.0	6.250000	20.969986	7.290000
9	7.210000	-7.190000	6.250000	6.250000	7.290000	-7.190000
10	6.250000	-7.190000	0.0	6.250000	-20.969986	-7.190000
11	20.969986	-7.210000	0.0	6.250000	6.299999	7.290000
12	20.969986	-7.190000	0.0	6.250000	-6.399999	-7.190000

MEMBER-I	IX	IY	MEMBER PROPERTIES		RADIUS	PHI	ALPHA
			J7	AREA			
1	0.000000	0.000000	0.000000	0.062500	35.000000	0.217400	0.471900
2	0.000000	0.000000	0.000000	0.062500	35.000000	0.217600	-0.470900
3	0.000000	0.000000	0.000000	0.062500	35.000000	0.218100	-0.472900
4	0.000000	0.000000	0.000000	0.062500	35.000000	0.209400	-0.298100
5	0.000000	0.000000	0.000000	0.062500	35.000000	0.216500	-0.470900
6	0.000000	0.000000	0.000000	0.062500	35.000000	0.208800	-0.208900
7	0.000000	0.000000	0.000000	0.062500	35.000000	0.207500	-0.208900
8	0.000000	0.000000	0.000000	0.062500	35.000000	0.217100	0.471300
9	0.000000	0.000000	0.000000	0.062500	35.000000	0.208400	0.208100
10	0.000000	0.000000	0.000000	0.062500	35.000000	0.218400	0.471900
11	0.000000	0.000000	0.000000	0.062500	35.000000	0.218000	0.472200
12	0.000000	0.000000	0.000000	0.062500	35.000000	0.218300	-0.471300

MEMB	NODE(I,1)	NODE(I,2)
1	1	4
2	1	5
3	1	6
4	4	5
5	4	6
6	4	7
7	5	6
8	5	7
9	6	7
10	10	11
11	10	12
12	11	12

NODE	NODE	LINK	1	2	3	4
1	1	1	0	0	0	0
2	1	2	0	0	0	0
3	1	3	0	0	0	0
4	4	4	-1	0	0	0
5	4	5	-2	0	0	0
6	4	6	-1	0	0	0
7	5	7	0	0	0	11
8	5	8	-1	0	0	12
9	6	9	0	0	0	0
10	10	10	-1	0	0	0
11	10	11	-1	0	0	0
12	11	12	0	0	0	0

NODE	PX	PY	APPLIED NODAL LOADS		PX	PY
			FX	FY		
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	-10.000000	0.0	0.0	0.0	0.0
5	0.0	-10.000000	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	-10.000000	0.0	0.0	0.0	0.0
8	0.0	-10.000000	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.1251125E 25

NODE	DELTA-X	DELTA-Y	THETA-X	THETA-Y	THETA-X	THETA-Y
1	0.0	0.0	0.0	-0.003161	-0.005700	0.004113
2	0.0	0.0	0.0	-0.003464	-0.005733	0.004314
3	0.0	0.0	0.0	-0.004403	-0.006756	-0.003033
4	-0.001715	-0.001139	-0.002114	-0.004048	-0.003737	-0.004940
5	-0.002066	-0.001295	-0.002476	-0.004889	-0.004369	-0.005885
6	0.0	0.0	0.0	-0.005982	-0.005719	0.002091
7	0.0	0.0	0.0	-0.006273	-0.006594	0.002889
8	0.000000	-0.002559	-0.001123	-0.004974	-0.003744	0.004987
9	0.001891	-0.002295	-0.001537	-0.004977	-0.003744	0.004987
10	0.0	0.0	0.0	-0.004207	-0.005077	0.003034
11	0.0	0.0	0.0	-0.004963	-0.005916	-0.004074
12	0.0	0.0	0.0	-0.004960	-0.005909	-0.004074

MEMBER END FORCES AND MOMENTS

MEMBR.	END	PX	DDX			DDY			MPX	MPY	MPZ
			PY	PZ	MX	PY	PZ	MY			
1	1*		29.547409	14.820759	3.426393	-0.000006	-0.000122	0.000153			
	1	33.208252	1.283884	-0.084844	-0.000042	-0.000033	0.000148				
	2*		-29.547394	-14.820743	-3.426389	1.406838	-6.983936	18.077484			
2	2	-33.208221	-1.283883	0.084844	0.000034	1.281693	19.388306				
	1*		30.225372	15.090419	-2.110114	0.000051	0.000097	0.000066			
	1	33.822800	1.326509	-0.060938	0.000085	0.000039	0.000074				
3	2*		-30.225357	-15.090420	2.110113	-2.473993	7.532850	18.435089			
	2	-33.822782	-1.326514	0.060943	-0.000072	-0.021525	20.046616				
	1*		-3.262359	15.211133	-30.453650	0.000063	0.000065	-0.000025			
4	1	34.170944	1.334428	0.081670	0.000044	0.000016	0.000082				
	2*		3.262358	-15.211149	30.453690	18.818359	7.399334	1.679987			
	2	-34.170975	-1.334429	-0.081674	-0.000036	-1.237991	20.252655				
5	1*		0.109624	0.088730	-27.192734	-20.314774	-7.844772	-0.113202			
	1	27.192732	0.089040	-0.080654	0.029429	1.413719	-20.464478				
	2*		-0.109623	-0.088730	27.192749	20.441747	7.844740	0.113225			
6	2	-27.192947	-0.089050	0.080655	-0.029428	-1.404182	20.596268				
	1*		2.654429	-14.822336	-29.609843	-18.070663	-7.188804	1.995709			
	1	33.224701	-1.299245	-0.071053	-0.000043	1.048718	-19.520950				
7	2*		-2.654429	14.822351	29.609848	0.000040	-0.000036	-0.000033			
	2	-33.224685	1.299265	0.071061	0.000047	-0.000033	0.000025				
	1*		26.175522	0.023201	0.165042	0.089537	2.429336	-19.644257			
8	1	28.175995	0.023235	0.033199	-0.085233	-1.687425	-19.721191				
	2*		-26.175541	-0.023200	-0.165042	-0.091161	-2.991729	19.980927			
	2	-28.175974	-0.023234	-0.033199	0.085233	1.217031	20.167175				
9	1*		27.689634	-0.089531	0.307172	0.102865	-3.225393	-20.544510			
	1	27.689292	-0.016314	0.035821	-0.096592	1.105335	-20.766769				
	2*		-27.689618	0.089532	-0.307172	-0.101668	3.671268	20.421478			
10	2	-27.689291	0.016315	-0.035821	0.096595	-1.421999	20.531494				
	1*		3.126793	15.297081	-30.458954	0.000094	-0.000140	0.000031			
	1	34.106701	1.341101	-0.075043	-0.000079	-0.000052	0.000139				
11	2*		-3.126779	-15.297067	30.458968	18.759794	-7.661978	-1.809360			
	2	-34.106640	-1.341102	0.075044	0.000066	1.137497	20.231247				
	1*		-0.101870	-0.004136	-27.672219	-20.459035	3.211113	-0.018884			
12	1	27.672255	-0.013248	-0.042721	0.061683	-1.126122	-20.875655				
	2*		0.101870	0.004136	27.672218	20.598145	-2.566222	0.013900			
	2	-27.672255	0.013249	0.042721	-0.061684	1.244726	20.483338				
13	1*		-2.942702	-14.866484	-29.847410	-10.313156	2.212411	-1.897828			
	1	33.441010	-1.204142	0.071047	0.000051	-1.227325	-19.783433				
	2*		2.942700	14.866485	29.847590	0.000021	0.000055	0.000030			
14	2	-33.440972	1.204179	-0.071053	-0.000051	1.310342	-0.000000				
	1*		20.404053	-14.695574	-2.620989	1.089295	2.222728	-18.152625			
	1	32.047152	-1.206309	0.073227	0.000033	-1.108870	-19.406504				
15	2*		-20.404075	14.695573	2.620988	-0.000045	0.000010	0.000032			
	2	-32.046991	1.206329	-0.073232	-0.000047	0.000004	0.000000				
	1*		30.427505	-15.146171	2.481998	-2.183323	-7.372574	-18.542725			
16	1	34.053499	-1.323231	-0.071095	-0.000040	1.077253	-20.063919				
	2*		-30.427475	15.146165	-2.481998	0.000047	0.000019	0.000065			
	2	-34.053452	1.323230	0.071096	0.000039	0.000005	0.000022				

NUMBER OF NODES 4  
 NUMBER OF MEMBERS 12  
 NUMBER OF LOADING CONDITION 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	MEMBER COORDINANTS					
	X(I,1)	X(I,2)	Y(I,1)	Y(I,2)	Z(I,1)	Z(I,2)
1	-20.969986	-7.290000	0.0	6.250000	5.870000	-7.219999
2	-20.969986	-7.290000	0.0	6.250000	-6.420000	-7.330000
3	-6.059999	-7.290000	0.0	6.250000	20.969986	-7.219999
4	-7.290000	-7.290000	6.250000	6.250000	7.219999	-7.330000
5	-6.250000	-7.290000	0.0	6.250000	-20.969986	-7.330000
6	-7.290000	7.219999	6.250000	6.250000	7.219999	-7.290000
7	-7.290000	7.190000	6.250000	6.250000	-7.330000	-7.190000
8	6.059999	7.219999	0.0	6.250000	20.969986	-7.290000
9	7.219999	7.190000	6.250000	6.250000	7.290000	-7.190000
10	6.120000	7.190000	0.0	6.250000	-20.969986	-7.190000
11	20.969986	7.219999	0.0	6.250000	6.309999	-7.290000
12	20.969986	7.190000	0.0	6.250000	6.309999	-7.190000

MEMBER-I	MEMBER PROPERTIES						
	IV	IV	IV	AREA	RADIUS	PHI	ALPHA
1	0.000550	0.000326	0.000326	0.062500	35.000000	0.217400	0.471900
2	0.000550	0.000326	0.000326	0.062500	35.000000	0.218100	-0.472000
3	0.000550	0.000326	0.000326	0.062500	35.000000	0.209400	-0.208100
4	0.000550	0.000326	0.000326	0.062500	35.000000	0.216500	-0.470900
5	0.000550	0.000326	0.000326	0.062500	35.000000	0.208800	0.208800
6	0.000550	0.000326	0.000326	0.062500	35.000000	0.207500	-0.208000
7	0.000550	0.000326	0.000326	0.062500	35.000000	0.217100	-0.471300
8	0.000550	0.000326	0.000326	0.062500	35.000000	0.208400	0.208100
9	0.000550	0.000326	0.000326	0.062500	35.000000	0.218400	-0.471800
10	0.000550	0.000326	0.000326	0.062500	35.000000	0.218000	-0.472200
11	0.000550	0.000326	0.000326	0.062500	35.000000	0.218000	-0.471300
12	0.000550	0.000326	0.000326	0.062500	35.000000	0.218000	-0.471300

MEMBER NODE (1,1) NODE (1,2)

1	1	2
2	1	2
3	1	3
4	1	4
5	1	3
6	1	4
7	2	3
8	2	4
9	2	3
10	2	4
11	3	4
12	3	4

NODE	NODI	LINK			
		1	2	3	4
1	4	-1	-2	6	5
2	4	-2	-6	0	7
3	4	-6	0	-10	-11
4	4	-7	-0	-10	-12

NODE	PX	PY	APPLIED NODAL LOADS		MY	MZ
			PZ	MX		
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	-31.079997	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.5138284E 45

NODE	NODAL DISPLACEMENTS					
	DELTA-Y	DELTA-Y	DELTA-Z	THETA-Y	THETA-Y	THETA-Z
1	0.011238	-0.012021	-0.011062	0.000210	-0.000000	0.000320
2	-0.014957	0.024725	-0.011620	0.000715	-0.000599	-0.001570
3	0.011612	0.024132	0.015374	-0.001548	0.000000	0.000252
4	0.018060	0.047532	0.012202	0.000275	0.000000	0.000275

MEMBER END FORCES AND MOMENTS

MEMB.	END	PPX			PPY			PPZ			MPX			MPY			MPZ			
		PX	PY	PZ	PX	PY	PZ	PX	PY	PZ	MX	MY	MZ	MX	MY	MZ	MX	MY	MZ	
1	1*		-5.515539	-1.434613	-0.042377	0.814041	-4.629140	9.732392												
	1	-5.594247	1.088785	0.004890	-0.306765	-0.034807	10.801790													
	2*		5.515539	1.434613	0.042377	0.456616	-2.238565	5.114753												
2	1*		20.453768	7.807212	-0.404281	2.804404	-9.064337	-20.389435												
	1	21.936005	-1.536638	-0.045197	-0.241083	0.397881	-22.484192													
	2*		-20.453768	-7.807212	0.404281	0.311910	0.076400	-0.787744												
3	1*		-0.037906	-1.244091	5.097093	9.472454	4.509384	0.860342												
	1	-5.134943	1.077912	-0.001060	0.310339	0.019323	10.521663													
	2*		0.037906	1.244091	-5.097093	5.278820	2.281328	0.906780												
4	1*		1.112828	-1.937521	6.226292	-6.621561	-7.217366	0.431269												
	1	-6.221649	-1.543877	0.837498	-0.458871	-5.694010	-7.968023													
	2*		-1.112828	1.937522	-6.226295	-12.839427	-9.347838	-0.511522												
5	1*		1.027711	-4.245849	-6.543375	4.235605	2.295510	0.092007												
	1	-7.767864	-1.248722	0.011646	0.761589	0.019309	-4.757917													
	2*		-1.027711	4.245849	6.543375	-12.782243	5.310082	-2.354277												
6	1*		-6.666543	-1.341434	-1.171288	0.486334	7.174512	-6.452651												
	1	-6.672105	-1.548498	-0.836133	0.455199	5.679981	-7.802499													
	2*		6.666543	1.341434	1.171288	-0.392333	7.353883	-13.011796												
7	1*		22.683900	2.223970	-0.721108	-0.154595	4.011444	3.443595												
	1	22.685822	2.270850	-0.459711	-0.119021	3.164188	4.417145													
	2*		-22.683924	-2.223977	0.721107	-0.156765	2.564094	28.406250												
8	1*		1.081690	7.399890	-20.970990	-20.355901	0.129882	2.144254												
	1	21.767044	-1.543877	0.031059	0.272710	-0.309131	-22.407980													
	2*		-1.081690	-7.399870	20.970990	-0.888747	0.114656	0.277729												
9	1*		0.883060	2.222319	-22.574646	3.460682	-3.093535	-0.078200												
	1	22.572756	2.364475	0.450634	0.070616	-2.151041	4.406922													
	2*		-0.883060	-2.222319	22.574646	28.518539	-9.470277	0.011531												
10	1*		2.585798	13.174564	25.545502	3.283307	-7.529344	8.895763												
	1	20.815842	1.536832	0.131516	0.001348	-1.016471	-4.116302													
	2*		-2.585799	-13.174564	-25.545502	-25.170959	10.566409	-2.830448												
11	1*		6.408544	-4.190204	-1.932918	0.030142	-2.359070	4.188418												
	1	-7.700929	-1.244441	0.001535	-0.730319	-0.110651	-4.750267													
	2*		6.408544	4.190204	1.932918	-2.379398	-5.475005	12.812292												
12	1*		-26.144653	13.450445	-2.248552	0.082389	2.682357	3.511650												
	1	20.450577	1.548500	-0.140720	0.009040	1.056914	-4.401453													
	2*		26.144653	-13.450445	2.248552	-3.191445	-10.660041	-25.578912												

INCP171  
 TRACKBACK FOLLOWS- ROUTINE ISN REC. 14  
 FROM 83001000  
 MAIN 00002868  
 ENTRY POINT= 50005020

MEMBER END FORCES AND MOMENTS

MEMB.	END	DPX	DPY	DPZ	MPX	MPY	MPZ
		PX	PY	PZ	MX	MY	MZ
1	1	37.416824	-0.145364	-0.113842	0.700360	0.879639	-20.398636
	2	-37.416824	0.145364	0.113842	-0.700353	0.839569	18.203812
2	1	38.044250	-0.124330	0.098112	-0.652696	-0.783352	-20.585129
	2	-38.044250	0.124330	-0.098112	0.652692	-0.699449	18.706772
3	1	38.219894	-0.145473	0.113871	-0.697050	-0.866794	-20.982666
	2	-38.219894	0.145473	-0.113871	0.697051	-0.858893	18.778473
4	1	31.710342	-0.135034	0.012717	-31.710052	-18.198761	-3.353313
	2	-31.710342	0.135034	-0.012717	31.710068	18.383774	3.291184
5	1	37.105302	-0.099630	-0.109169	0.692812	0.857931	-19.730377
	2	-37.105302	0.099630	0.109169	-0.692815	-0.783550	18.232590
6	1	30.872543	-0.018751	0.012026	0.006864	-0.629138	-17.977661
	2	-30.872543	0.018749	-0.012025	-0.006864	0.441590	18.249695
7	1	32.485492	-0.007023	0.015462	-0.044187	0.135516	-3.544415
	2	-32.485492	0.007023	-0.015462	0.044188	-0.134879	3.302598
8	1	30.264481	-0.123743	-0.111391	0.688816	0.862418	-20.630814
	2	-30.264481	0.123743	0.111391	-0.688813	-0.818002	18.744435
9	1	32.310077	-0.094655	0.004050	-32.318285	-19.492529	3.483242
	2	-32.310077	0.094655	-0.004050	32.318985	18.551239	-3.227092
10	1	37.433456	-0.136603	0.105104	-0.715139	-0.839666	-20.514709
	2	-37.433456	0.136603	-0.105104	0.715142	-0.754738	18.442825
11	1	37.108246	-0.121718	0.106713	-0.688466	-0.937911	-20.184442
	2	-37.108246	0.121718	-0.106713	0.688461	0.777366	18.322754
12	1	30.321762	-0.151800	-0.108905	0.686358	0.854047	-21.061615
	2	-30.321762	0.151800	0.108905	-0.686355	-0.786713	18.761078

JHC2171  
 TRACEBACK FOLLOWS- POINTING ISM REC. 14  
 FROM 8200EBC0  
 MAIN 00002848

ENTRY POINT= 50005020

NUMBER OF NODES 4  
 NUMBER OF MEMBERS 12  
 NUMBER OF LOADING CONDITION 1  
 MAXIMUM NUMBER OF MEMBERS PER NODE 4

MEMBER-I	MEMBER COORDINANTS					
	X(I,1)	X(I,2)	Y(I,1)	Y(I,2)	Z(I,1)	Z(I,2)
1	-20.969986	-7.290000	0.0	6.250000	5.870000	7.219999
2	-20.969986	-7.230000	0.0	6.250000	-6.620000	-7.230000
3	-20.969986	-7.290000	0.0	6.250000	20.969986	7.219999
4	-7.290000	-7.230000	6.250000	6.250000	7.219999	-7.230000
5	-4.250000	-7.230000	0.0	6.250000	-20.969986	-7.230000
6	-7.290000	7.219999	6.250000	6.250000	7.219999	7.290000
7	-7.230000	7.190000	6.250000	6.250000	-7.230000	-7.190000
8	6.059999	7.219999	0.0	6.250000	20.969986	7.290000
9	7.219999	7.190000	6.250000	6.250000	7.290000	-7.190000
10	6.120000	7.190000	0.0	6.250000	-20.969986	-7.190000
11	20.969986	7.219999	0.0	6.250000	6.309999	7.290000
12	20.969986	7.190000	0.0	6.250000	-6.309999	-7.190000

MEMBER-I	MEMBER PROPERTIES				RADIUS	PHI	ALPHA
	IX	IY	IZ	AREA			
1	0.000550	0.000326	0.000326	0.062500	35.000000	0.217400	0.471900
2	0.000550	0.000326	0.000326	0.062500	35.000000	0.218100	-0.472900
4	0.000550	0.000326	0.000326	0.062500	35.000000	0.209400	-0.208100
5	0.000550	0.000326	0.000326	0.062500	35.000000	0.216500	0.470900
6	0.000550	0.000326	0.000326	0.062500	35.000000	0.208800	0.208400
7	0.000550	0.000326	0.000326	0.062500	35.000000	0.207500	-0.208900
8	0.000550	0.000326	0.000326	0.062500	35.000000	0.217100	0.471300
9	0.000550	0.000326	0.000326	0.062500	35.000000	0.208400	0.208100
10	0.000550	0.000326	0.000326	0.062500	35.000000	0.218400	-0.471900
11	0.000550	0.000326	0.000326	0.062500	35.000000	0.218000	-0.472200
12	0.000550	0.000326	0.000326	0.062500	35.000000	0.218300	0.471300

MEMB NODE(I,1) NODE(I,2)

1	2	1
2	1	2
3	1	3
4	2	2
5	1	3
6	2	4
7	1	3
8	2	4
9	1	3
10	2	4
11	1	3
12	2	4

MEMB NODE(I,1) NODE(I,2)

1	4	-1	-2	4	4
2	4	-2	-4	0	7
3	4	-4	-8	0	-11
4	4	-7	-10	-10	-13

NODE	PX	PY	APPLIED NODAL LOADS		MX	MY
			PZ	MZ		
1	0.0	-31.079987	0.0	0.0	0.0	0.0
2	0.0	-31.079987	0.0	0.0	0.0	0.0
3	0.0	-31.079987	0.0	0.0	0.0	0.0
4	0.0	-31.079987	0.0	0.0	0.0	0.0

DETERMINANT OF SYSTEM STIFFNESS MATRIX = 0.813824E 45

NODE	NODAL DISPLACEMENTS					
	DELTA-X	DELTA-Y	DELTA-Z	THETA-X	THETA-Y	THETA-Z
1	0.000241	-0.011792	-0.000283	-0.001801	-0.000013	-0.001978
2	0.000510	-0.012570	-0.000792	-0.001757	-0.000039	-0.001907
3	0.000670	-0.013338	-0.001466	-0.001819	0.000027	0.001787
4	-0.000361	-0.013776	0.000348	0.001741	0.000043	0.001884