# L-header Testing, Evaluation and Design Methodology 

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# L-HEADER TESTING, EVALUATION AND DESIGN METHODOLOGY <br> N. R. Elhajj ${ }^{1}$ and R. A. LaBoube ${ }^{2}$ 


#### Abstract

Cold-formed steel L-shaped headers have gained popularity over the past few years due to their simplicity and cost effectiveness. While cold-formed steel conventional headers are widely used and can be designed for most applications, often it is necessary to reduce cost by using less material and labor hours. The L-shaped header provides both, a fast and economical solution to safely transfer applied loads to other structural elements in a building. As the name suggests, the main components of an L-shaped header is a piece of cold-formed steel formed into a shape resembling the letter L. An L-header assembly consists of a cold-formed steel angle with one short leg lapping over the top track and one long leg extending down the side of the wall above openings. The current design equations in the AISI Specification do not provide a reasonable design values for Lheader assemblies. Testing of the assemblies was necessary to develop an easy to use design equations that can be used by designers. A total of 71 gravity tests and 38 uplift tests of L-header assemblies having variable sizes and thicknesses and spans were conducted at the NAHB Research Center. Results of the tests as well as a proposed design procedure is presented here.


## Introduction

The recent increase in the use of lighter material and more economical headers led to the development of the cold-formed steel L-shaped header. However, designers and engineers still do not have the proper tools to properly design and specify those headers. Design practice for header beams has been based on the AISI Specification (1996). However, application of the AISI design provisions often results in limited span capability for the header beam. Because the current design specification (Specification 1996) does not explicitly address L-shaped headers, a comprehensive study was initiated at the NAHB Research Center in 1998 and 1999. This study, funded by the American Iron and Steel Institute and USS POSCO specifically addressed the maximum span limitations of double shaped-L-headers. The testing and evaluation program was completed in two phases. The first phase looked at the most practical shapes and configurations of L-headers, while the second phase concentrated on the most economical sections.

This paper will summarize the findings of the bending tests, and a modified design methodology that has been developed. The modified design methodology is based on the flexural capacity of the L-sections alone.

## Experimental Approach

[^0]
## Test Specimens

The L-header test specimens utilized construction materials and methods appropriate for residential construction using cold-formed steel framing. All steel materials used in the tests conformed to the dimensional requirements of Table 1 and had a minimum specified tensile strength of 33 ksi , which was verified by tensile tests in accordance with ASTM A370. Base steel thickness was measured in accordance with ASTM A90. Mechanical properties were based on coupons cut from the center of the web from a sample of the test specimens.

Each L-header assembly consisted of a cold-formed steel angle with one short leg lapping over the top track and one long leg extending down the side of the wall above window or door openings. Each angle was fastened to the top track above an opening with \#8 (minimum) screws spaced 12 -inches on-center. "L" angles can be placed on one side or both sides of the steel track to form either a single or a double L-header. A detail of the built-up L-header assembly is shown in Figure 1. Header spans for the tests were selected to cover a range of the most common construction opening sizes (i.e. $3^{\prime}-0^{\prime \prime}, 6^{\prime}-0^{\prime \prime}, 8^{\prime}-0^{\prime \prime}, 12^{\prime}-$ $0^{\prime \prime}$, and $16^{\prime}-0^{\prime \prime}$ ). All screws used were \#8 or \#10 self-drilling with low profile heads or hex heads. All screws protruded through steel a minimum of $3 / 8$-inch with a minimum of three exposed threads.

Gravity Load Tests:
A total of 71 L-shaped header assemblies were constructed and tested in accordance with Table 2. Steel track sections had a minimum thickness of 33 mils. Eight L-header specimens were constructed and tested with 43 mil steel tracks to investigate the impact of the track thickness on the header's capacity. L-header assemblies with $3^{\prime}-0^{\prime \prime}$ clear spans were tested utilizing one point loading to simulate the worst case scenario when truss members (or floor joists) spaced at 24 inches on-center are bearing on the headers. L-header assemblies with $6^{\prime}-0^{\prime \prime}$ or greater clear spans were tested utilizing one-third point loading (two-point loading).

## Uplift Load Tests:

A total of 38 L-shaped header assemblies were constructed and tested for uplift loads in accordance with Table 3. Steel tracks had a minimum thickness of 33 mils. L-header assemblies with $3^{\prime}-0^{\prime \prime}$ clear spans were tested utilizing one point loading to simulate the worst case scenario when truss members, floor joists, or other structural members spaced at 24 inches on-center are bearing on the headers. L-header assemblies with 6'-0" or greater clear spans were tested utilizing one-third point loading (two-point loading).

## Test Apparatus

## Gravity Load Tests:

The header assemblies were tested using a universal-testing machine (UTM). The two-point or one-point loads were applied to the L-header specimen using heavy steel I-beam and 1.5 inch x 4 inch steel bearing plates as shown in Figure 1. Back-to-back steel studs (350S162-
54) were fastened to each end of the L-header. The load was applied at a load rate of $1 / 20$ inch per minute until failure. The ultimate load constituted failure of the header material (buckling, bearing or crippling) or failure of the screws (shear or pull out). Deflections at each header's midpoint were recorded during the full range of loads using linear variable differential transformers (LVDTs). Each header assembly was restrained against weak axis rotation and lateral movement. Rotation of the header was allowed in the plane of bending. Rollers were not used at the reactions because rotation was provided by the configuration of the end studs.

## Uplift Load Tests:

The header assemblies were tested using a universal testing machine. Two test set-ups were used:

1. A simply supported inverted L-header was subjected to a compressive load, applied through a 54 mil stud ( $350 \mathrm{~S} 162-54$ ) fastened to one side of the L-header with 6 No. 10 screws, to simulate an uplift condition, as shown in Figure 3. The load was applied at a load rate of $1 / 20$ inch per minute until failure.
2. The bottoms of the end studs (back-to-back studs) were fixed to prevent them from pulling out as shown in Figure 4. Each header assembly was subjected to an uplift (tension) load until failure. The ultimate load constituted failure of the header material (buckling or crippling) or failure of the screws (shear or pull out). The load (with a load rate of $1 / 20$ inch per minute) was applied at two-point or one-point location depending on the header span.

## Results

Tables 4 and 5 summarize the results of the gravity and uplift load tests.

## Gravity Load Tests

Web crippling and bending failure mode was observed in all 3-foot, 4 -foot, 6 -foot, and 8 foot header tests and bending failure mode for the 12 -foot and 16 -foot headers. During one test, 2-800L150-54 (3-foot), the cripple stud at the load points was removed and as expected the specimen failed at a lower load $(5,146 \mathrm{lb}$. compared with 6,386 for a standard header specimen) but with the same failure mode. This test confirmed that the typical practice of using a C-section web support fastened with screws to the track provides partial, not full, web crippling support.

## Uplift Load Tests:

All test specimens behaved consistently during the test and at failure. The header's webs between the point loads and between the point loads and the reactions showed severe buckling failure at peak loads. High vertical deflections (between 0.5 in . and 1.0 in .) were observed for all headers at peak loads.

Header specimen tested under both configurations (Figure 3 and 4) resulted in similar ultimate capacities.

## Data Analysis

Tests indicated that the failure mode was flexure or combination of flexure and web crippling. The tested moment capacity, $\mathrm{M}_{\mathrm{t}}$, was determined and compared with the computed moment capacity as defined by Section C3.1.1(a) of the AISI Specification. The nominal moment capacity was computed using the following equation:

$$
\begin{equation*}
\mathrm{M}_{\mathrm{n}}=\mathrm{S}_{\mathrm{xc}} \mathrm{~F}_{\mathrm{y}} \tag{1}
\end{equation*}
$$

(a) Gravity Moment Capacity

A comparison of the $\mathrm{M}_{\mathrm{t}} / \mathrm{M}_{\mathrm{n}}$ yielded ratios that ranged from 0.78 to 1.67 with a mean of 1.11 (Table 6). Although a favorable mean value was achieved, the coefficient of variation was 0.21 .

Although seven tested header sections have $\mathrm{M}_{\mathrm{t}} / \mathrm{M}_{\mathrm{n}}$ ratios of 0.7 to 0.81 , the remainder of the data has a ratio of 0.9 or greater. These seven test specimens are all 10 inch Lheaders. Thus, the application of Eq. 1 appears questionable for the 10 inch L-header. Summary of the 10 inch L-header test results is given by Table 7. A review of the data indicates that the application of Eq. 1 is valid for test specimens having a span to leg dimension, L/leg, of 10 or greater. For the specimens having L/leg ratios less than 10 the $\mathrm{M}_{\mathrm{t}} / \mathrm{M}_{\mathrm{n}}$ ratio ranged from 0.70 to 0.81 with a mean of 0.77 (Table 8). Table 3 presents the ratio of $\mathrm{M}_{\mathrm{t}} /\left(0.8 \mathrm{M}_{\mathrm{n}}\right)$, which ranges from 0.89 to 1.02 with a mean of 0.97 .
(b) Uplift Moment Capacity

A comparison of the $\mathrm{M}_{\mathrm{t}} / \mathrm{M}_{\mathrm{n}}$ yielded ratios that ranged from 0.14 to 0.31 with a mean of 0.22 (Table 9). Further analysis of the moment ratio indicated that the behavior was influenced by the ratio of $L_{h} / t$. Therefore, uplift reduction factors, $R$, are given as a function of the $\mathrm{L}_{\mathrm{h}} / \mathrm{t}$ ratio. The $2-1000 \mathrm{~L} 150-54$ section resulted in unexplainable behavior and thus this section is singled out for a lower reduction factor. Applying the appropriate uplift reduction factor resulted in $\mathrm{M}_{\mathrm{t}} / \mathrm{M}_{\mathrm{n}}$ ratios that ranged from 0.82 to 1.30 with a coefficient of variation of 0.12 .

## Design Methodology

This design procedure is limited to double L-headers fabricated using cold-formed steel angles having the following limitations:

Minimum top flange width $=1.5 \mathrm{in}$.
Maximum vertical leg dimension $=10 \mathrm{in}$.
Minimum base material thickness $=0.033$ in.
Maximum design yield strength, $\mathrm{F}_{\mathrm{y}}=50 \mathrm{ksi}$
Cripple stud located at all load points

Minimum bearing length 1.5 in . at load points
Minimum No. 8 self-drilling screws installed per Fig. 1.
(a) Gravity Moment Capacity
(i) For a double L-header beam having vertical leg dimensions of less than 8 in . or less, the design shall be based on the flexural capacity of the L-sections alone. The nominal flexural strength, $\mathrm{M}_{\mathrm{n}}$, shall be calculated as follows:

$$
\begin{equation*}
M_{n}=S_{e c} F_{y} \tag{2}
\end{equation*}
$$

(ii) For a double L-header beam having vertical leg dimension greater than 8 in., and having a span-to-leg dimension ratio greater than or equal to10, design shall be based on the flexural capacity of the L-sections alone, Eq. 2. For an L-header beam having a $10-$ in. vertical leg dimension greater than 8 in . and having a span-to-vertical leg dimension ratio less than 10 in ., the nominal moment capacity shall be taken as 0.90 times Eq. 2.

## (b) Uplift Moment Capacity

For a double L-header beam, design shall be based on the flexural capacity of the Lsections alone. The nominal flexural strength, $\mathrm{M}_{\mathrm{n}}$, shall be calculated as follows:

$$
\begin{equation*}
\mathrm{M}_{\mathrm{n}}=\mathrm{R} \mathrm{M}_{\mathrm{ng}} \tag{3}
\end{equation*}
$$

Where

$$
\mathrm{M}_{\mathrm{ng}}=\text { gravity moment capacity determined by Eq. } 1
$$

$\mathrm{R}=0.25$ for $\mathrm{L}_{\mathrm{h}} / \mathrm{t} \leq 150$
$=0.20$ for $\mathrm{L}_{\mathrm{h}} / \mathrm{t} \leq 170$
$=$ use linear interpolation for $150<\mathrm{L}_{\mathrm{h}} / \mathrm{t}<170$
(c) Design Moment Capacity

The allowable stress design moment is determined as follows:

$$
\begin{equation*}
\mathrm{M}_{\mathrm{a}}=\mathrm{M}_{\mathrm{n}} / \Omega \tag{4}
\end{equation*}
$$

Where for gravity load moment capacity, $\Omega=1.67$ for members having vertical leg $\leq 8$ in., and $\Omega=2.26$ for members having vertical leg $>8 \mathrm{in}$. For uplift moment capacity, $\Omega$ $=2.0$.

The load and resistance factor design moment is determined as follows:

$$
\begin{equation*}
\mathrm{M}_{\mathrm{a}}=\phi \mathrm{M}_{\mathrm{n}} \tag{4}
\end{equation*}
$$

Where for gravity load moment capacity, $\phi=0.90$ for members having vertical leg $\leq 8$ in., and $\phi=0.71$ for members having vertical leg $>8 \mathrm{in}$. For uplift moment capacity, $\phi=$ 0.80 .

## Appendix - References

National Association of Home Builders (1998), "L-Shaped Header Testing, Evaluation, \& Design Guidance," NAHB Research Center, Upper Marlboro, MD

Specification for the Design of Cold-Formed Steel Structural Members (1996), American Iron and Steel Institute, Washington, D.C.

## Appendix - Notation

$\mathrm{F}_{\mathrm{y}}=$ Design yield stress
$\mathrm{L}_{\mathrm{h}}=$ Vertical leg dimension of the angle
$\mathrm{M}_{\mathrm{a}}=$ Allowable design moment
$\mathrm{M}_{\mathrm{n}}=$ Nominal moment capacity at section being investigated.
$\mathrm{M}_{\mathrm{t}}=$ Tested moment capacity
$\mathrm{R}=$ Reduction factor
$S_{e c}=$ Elastic section modulus of the effective section calculated at $f=F_{y}$ in extreme compression fibers
$S_{x c}=$ Elastic section modulus of the effective section computed at $f=F_{y}$.
$\mathrm{t}=$ Base thickness of steel

TABLE 1 PHYSICAL AND MECHANICAL PROPERTIES OF TEST SPECIMENS ${ }^{1}$

| L-Header <br> Designation |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | L-Angle <br> Dimension <br> (inches) | Average <br> Yield <br> Point $^{3}$ <br> (psi) | Average <br> Tensile <br> Strength <br> 3 <br> (psi) | Average <br> Uncoated <br> Thickness <br> (inches) | Elongation $^{\text {(percent) }}$ |
| 600L150-33 | 6 | 34,900 | 40,100 | 0.0339 | 21.2 |
| 600L150-43 | 6 | 36,500 | 45,300 | 0.0459 | 22.7 |
| 600L150-54 | 6 | 51,500 | 60,600 | 0.0575 | 22.4 |
| 800L150-33 | 6 | 48,000 | 56,200 | 0.0342 | 23.5 |
| 800L150-43 | 6 | 37,500 | 43,700 | 0.0465 | 22.3 |
| 800L150-54 | 6 | 54,500 | 67,400 | 0.0550 | 23.5 |
| 800L150-68 | 6 | 53,450 | 62,890 | 0.0719 | 21.5 |
| 1000L150-33 | 6 | 46,550 | 54,500 | 0.0341 | 20.9 |
| 1000L150-43 | 6 | 50,500 | 59,100 | 0.0475 | 20.5 |
| 1000L150-54 | 6 | 51,800 | 63,800 | 0.0575 | 22.9 |
| 1000L150-68 | 6 | 53,450 | 62,890 | 0.0179 | 21.5 |

For SI: $1 \mathrm{in} .=25.4 \mathrm{~mm}, 1 \mathrm{psi}=0.0479 \mathrm{kN} / \mathrm{m}^{2}$
${ }^{1}$ All track members had 3-1/2" webs and $1-5 / 8$ " flanges.
${ }^{2}$ Header designation is as follows: The first number is the size of the long leg of the angle in $1 / 100$ inches (eg. 8 " leg is indicated as 800 ). The " L " represents an angle. The 3-digits indicate the size of the short leg of the angle in $1 / 100$ inches (eg. 1-1/2" leg is indicated as 150). The 2 -digits represent the thickness of the angle in mils.
${ }^{3}$ Average yield point and average tensile strength shown are based on actual average yield point and tensile strength from coupons cut from a sample of three specimens tested per ASTM A370.
${ }^{4}$ Average uncoated thickness shown is based on actual average uncoated thickness from a sample of three specimens tested per ASTM A90.
${ }^{5}$ Tested in accordance with ASTM A370 for a two-inch gauge length.

TABLE 2 DOUBLE L-SHAPED HEADER TEST SPECIMEN FOR SIMULATED GRAVITY LOADS

| No. <br> of <br> Tests | Header Desig- <br> nation | Angle <br> Size <br> (in.) | Angle <br> Thickness <br> (mils) | Header <br> Clear <br> Span | Track <br> Thickness <br> (mils) | Loading <br> Configuration |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2 | $2-600 \mathrm{~L} 150-33$ | $6^{\prime \prime}$ | 33 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-600 \mathrm{~L} 150-43$ | $6^{\prime \prime}$ | 43 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-600 \mathrm{~L} 150-54$ | $6^{\prime \prime}$ | 54 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-600 \mathrm{~L} 150-54$ | $6^{\prime \prime}$ | 54 | $6^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 2 point |
| 2 | $2-800 \mathrm{~L} 150-33$ | $8^{\prime \prime}$ | 33 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-800 \mathrm{~L} 150-43$ | $8^{\prime \prime}$ | 43 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-800 \mathrm{~L} 150-43$ | $8^{\prime \prime}$ | 43 | $6^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-800 \mathrm{~L} 150-43$ | $8^{\prime \prime}$ | 43 | $8^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-800 \mathrm{~L} 150-54$ | $8^{\prime \prime}$ | 54 | $3^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 1 point |
| 2 | $2-800 \mathrm{~L} 150-54$ | $8^{\prime \prime}$ | 54 | $6^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 2 point |
| 2 | $2-800 \mathrm{~L} 150-54$ | $8^{\prime \prime}$ | 54 | $8^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 2 point |
| 2 | $2-800 \mathrm{~L} 150-54$ | $8^{\prime \prime}$ | 54 | $1^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-800 \mathrm{~L} 150-54$ | $8^{\prime \prime}$ | 54 | $1^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-800 \mathrm{~L} 150-68$ | $8^{\prime \prime}$ | 68 | $12^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-800 \mathrm{~L} 150-68$ | $8^{\prime \prime}$ | 68 | $16^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-43$ | $10^{\prime \prime}$ | 43 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-1000 \mathrm{~L} 150-43$ | $10^{\prime \prime}$ | 43 | $6^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-43$ | $10^{\prime \prime}$ | 43 | $8^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-54$ | $10^{\prime \prime}$ | 54 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 2 | $2-1000 \mathrm{~L} 150-54$ | $10^{\prime \prime}$ | 54 | $6^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-54$ | $10^{\prime \prime}$ | 54 | $8^{\prime}-00^{\prime \prime}$ | $33 \& 43$ | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-54$ | $10^{\prime \prime}$ | 54 | $12^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-54$ | $10^{\prime \prime}$ | 54 | $16^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-68$ | $10^{\prime \prime}$ | 68 | $12^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 2 | $2-1000 \mathrm{~L} 150-68$ | $10^{\prime \prime}$ | 68 | $16^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 1 | $2-600 \mathrm{~L} 150-43^{\prime}$ | $6^{\prime \prime}$ | 43 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 1 | $2-800 \mathrm{~L} 150-43^{\mathrm{T}}$ | $8^{\prime \prime}$ | 43 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 1 | $2-800 \mathrm{~L} 150-54^{\prime}$ | $8^{\prime \prime}$ | 54 | $3^{\prime}-00^{\prime \prime}$ | 33 | 1 point |
| 1 | $2-800 \mathrm{~L} 150-54^{\prime}$ | $8^{\prime \prime}$ | 54 | $6^{\prime}-00^{\prime \prime}$ | 33 | 2 point |
| 1 | $2-1000 \mathrm{~L}-150-54^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | $6^{\prime}-0^{\prime \prime}$ | 33 | 2 point |

${ }^{1}$ No. 8 and No. 10 screws used for testing to investigate the impact of screw size on header's capacity.

TABLE 3 L-SHAPED HEADER TEST SPECIMENS FOR SIMULATED UPLIFT LOADS

| $\begin{array}{\|c} \text { No. } \\ \text { of } \\ \text { Test } \end{array}$ | Header Designation | Angle Thickness (mils) | Header Clear Span | Track Thickness (mils) | Loading Configuration | Testing Configuration |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2 | 2-600L150-43 | 43 | (fieip) | 33 | 1 point | Fig. 3 |
| 2 | 2-600L150-54 ${ }^{1}$ | 54 | 3'-0" | 33 \& 43 | 1 point | Fig. 2 |
| 2 | 2-600L150-54 | 54 | 6'-0" | 33 | 2 point | Fig. 2 |
| 2 | 2-800L150-43 | 43 | 3'-0" | 33 | 1 point | Fig. 3 |
| 2 | 2-800L150-43 | 43 | 6'-0' | 33 | 2 point | Fig. 3 |
| 2 | 2-800L150-43 ${ }^{2}$ | 43 | 8'-0" | 33 | 2 point | Fig. 3 |
| 2 | 2-800L150-54 ${ }^{2}$ | 54 | 3'-0" | 33 | 1 point | Fig. 2 |
| 2 | 2-800L150-54 ${ }^{3}$ | 54 | 6'-0" | 33 \& 43 | 2 point | Fig. 3 |
| 2 | 2-800L150-54 | 54 | 8'-0" | 33 | 2 point | Fig. 3 |
| 2 | 2-800L150-43 | 54 | 12'-0" | 33 | 2 point | Fig. 3 |
| 2 | 2-1000L150-43 | 43 | 3'-0" | 33 | 1 point | Fig. 2 |
| 2 | 2-1000L150-43 | 43 | 6'-0" | 33 \& 43 | 2 point | Fig. 3 |
| 2 | 2-1000L150-43 | 43 | 8'0" | 33 | 2 point | Fig. 3 |
| 2 | 2-1000L150-54 ${ }^{2}$ | 54 | 3'-0" | 33 | 1 point | Fig. 3 |
| 2 | 2-1000L150-54 ${ }^{3}$ | 54 | 6'0" | 33 | 2 point | Fig. 3 |
| 2 | 2-1000L150-54 | 54 | 8'0" | 33 \& 43 | 2 point | Fig. 2 |
| 2 | 2-1000L150-54 | 54 | 12'-0" | 33 \& 43 | 2 point | Fig. 3 |

[^1]TABLE 4 TEST RESULTS - GRAVITY LOADS

| Test <br> No. | L-Header Designation | Angle Size (inches) | Angle Thickness <br> (mils) | Header Clear Span | Loading Configuration | Ultimate Load ${ }^{1}$ (lb.) | Load At <br> L/240 <br> Deflection <br> (lb.) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 2-600L150-43 | $6 "$ | 43 | 3'-0" | 1 point | 3,194 | 2,812 |
| 2 | 2-600L150-43 | 6 " | 43 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 3,282 | 3,073 |
| 3 | 2-600L150-54 | 6 " | 54 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 5,586 | 4,400 |
| 4 | 2-600L150-54 | 6 " | 54 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 4,878 | 4,066 |
| 5 | 2-600L150-54 | $6 "$ | 54 | 3'-0" | 1 point | 5,325 | 4,340 |
| 6 | 2-600L150-54 | 6 " | 54 | 6'-0" | 2 point | 5,652 | 4,142 |
| 7 | 2-600L150-54 | 6 " | 54 | 6'-0" | 2 point | 6,120 | 4,670 |
| 8 | 2-600L150-54 | 6 " | 54 | 6'-0" | 2 point | 7,112 | 5,140 |
| 9 | 2-600L150-54 | 6 " | 54 | 6'-0" | 2 point | 6,201 | 4708 |
| 10 | 2-800L150-33 | $8{ }^{\prime \prime}$ | 33 | $3^{\prime}-00^{\prime \prime}$ | 1 point | 4,302 | 4,277 |
| 11 | 2-800L150-33 | $8{ }^{\prime \prime}$ | 33 | $3^{\prime}-00^{\prime \prime}$ | 1 point | 4,634 | 4,235 |
| 12 | 2-800L150-43 | 8" | 43 | $3^{\prime}-00^{\prime \prime}$ | 1 point | 5,962 | 5,402 |
| 13 | 2-800L150-43 | 8' | 43 | $3^{\prime}-00{ }^{\prime \prime}$ | 1 point | 5,516 | 3,606 |
| 14 | 2-800L150-43 | 8' | 43 | 6'-00" | 2 point | 5.200 | 4,620 |
| 15 | 2-800L150-43 | 8" | 43 | 6'-00" | 2 point | 5,150 | 4,414 |
| 16 | 2-800L150-43 ${ }^{2}$ | 8' | 43 | $8^{\prime}-00^{\prime \prime}$ | 2 point | 5,136 | 4,375 |
| 17 | 2-800L150-43 | 8" | 43 | 8'-00" | 2 point | 5,452 | 4,400 |
| 22 | 2-800L150-54 | $8{ }^{\prime \prime}$ | 54 | 3'-00" | 1 point | 6,147 | 5,600 |
| 23 | 2-800L150-54 | 8" | 54 | $3^{\prime}-00^{\prime \prime}$ | 1 point | 6,624 | 6,150 |
| 24 | 2-800L150-54 | $8{ }^{\prime \prime}$ | 54 | 6'-00" | 2 point | 7,154 | 5,600 |
| 25 | 2-800L150-54 | $8{ }^{\prime \prime}$ | 54 | 6'-00" | 2 point | 7,230 | 5,680 |
| 26 | 2-800L150-54 | $8 "$ | 54 | $8^{\prime}-0^{\prime \prime}$ | 2 point | 7,244 | 5,818 |
| 27 | 2-800L150-54 ${ }^{2}$ | 8 " | 54 | $8^{\prime}-0^{\prime \prime}$ | 2 point | 7,252 | 5,990 |
| 28 | 2-800L150-54 | $8 "$ | 54 | 11'-6" | 2 point | 4,200 | 3,825 |
| 29 | 2-800L150-54 | 8" | 54 | 11'-6' | 2 point | 4,200 | 3,825 |
| 30 | 2-1000L150-43 | 10" | 43 | 3'-00" | 1 point | 6,014 | 5,200 |
| 31 | 2-1000L150-43 | $10^{\prime \prime}$ | 43 | 3'-00" | 1 point | 6,072 | 5,600 |
| 32 | 2-1000L150-43 | 10" | 43 | 6'-00' | 2 point | 6,986 | 6,150 |

TABLE 4 TESTED RESULTS - GRAVITY LOADS (continued)

| Test <br> No. | L-Header <br> Designation | Angle Size (inches) | Angle Thickness (mils) | Header Clear Span | Loading Configuration | Ultimate Load ${ }^{1}$ (lb.) | Load At L/240 Deflection (lb.) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 33 | 2-1000L150-43 | 10 " | 43 | 6'-00" | 2 point | 7,344 | 5,990 |
| 34 | 2-1000L150-43 | 10 " | 43 | 8'-00" | 2 point | 4,130 | 3,922 |
| 35 | 2-1000L150-43 | $10^{\prime \prime}$ | 43 | 8'-00" | 2 point | 4.138 | 4,126 |
| 36 | 2-1000L150-33 | 10" | 33 | 3'-0"' | 1 point | 4,494 | 4,339 |
| 37 | 2-1000L150-33 | 10" | 33 | $3^{\prime}-0$ "' | 1 point | 4,232 | 4,067 |
| 38 | 2-1000L150-33 ${ }^{2}$ | 10" | 33 | 6'-0" | 2 point | 4,742 | 4,492 |
| 39 | 2-1000L150-33 ${ }^{2}$ | 10" | 33 | $6^{\prime}-0^{\prime \prime}$ | 2 point | 3,822 | 3,590 |
| 40 | 2-600L150-33 | 6 " | 33 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 2,238 | 1,811 |
| 41 | 2-600L150-33 | $6 "$ | 33 | $3^{\prime}-0$ "' | 1 point | 2,314 | 1,968 |
| 42 | 2-1000L150-54 | 10" | 54 | $3^{\prime}-0$ "' | 1 point | 7,484 | 5,600 |
| 43 | 2-1000L150-54 | 10" | 54 | $3^{\prime}-0$ " | 1 point | 7,830 | 6,500 |
| 44 | 2-1000L150-54 | 10" | 54 | $6^{\prime}-0^{\prime \prime}$ | 2 point | 8,616 | 7,170 |
| 45 | 2-1000L150-54 | 10" | 54 | 6'-0"' | 2 point | 7,722 | 7,160 |
| 46 | 2-1000L150-54 | 10" | 54 | $8^{\prime}-0{ }^{\prime \prime}$ | 2 point | 6,646 | 5,558 |
| 47 | 2-1000L150-54 | $10^{\prime \prime}$ | 54 | $8^{\prime}-0^{\prime \prime}$ | 2 point | 6,400 | 6,090 |
| 48 | 2-1000L150-54 | 10" | 54 | $8^{\prime}-0^{\prime \prime}$ | 2 point | 6,706 | 6,656 |
| 49 | 2-800L150-54 | 8" | 54 | $16^{\prime}-0^{\prime \prime}$ | 2 point | 3,248 | 1,450 |
| 50 | 2-800L150-54 | 8" | 54 | $16^{\prime}-0^{\prime \prime}$ | 2 point | 3,136 | 1,485 |
| 51 | 2-800L150-68 | 8" | 68 | 12'-0' | 2 point | 7,711 | 2,800 |
| 52 | 2-800L150-68 | 8" | 68 | 12'-0' | 2 point | 6,721 | 2,550 |
| 53 | 2-800L150-68 | 8" | 68 | $16^{\prime}-0^{\prime \prime}$ | 2 point | 5,578 | 1,700 |
| 54 | 2-800L150-68 | 8" | 68 | 16'-0' | 2 point | 5,850 | 1,550 |
| 55 | 2-1000L150-54 | 10" | 54 | 12'-0' | 2 point | 6,458 | 5,426 |
| 56 | 2-1000L150-54 | 10" | 54 | 12'-0'' | 2 point | 6,542 | 5,510 |
| 57 | 2-1000L150-54 | 10" | 54 | 16'-0' | 2 point | 5,428 | 2,110 |
| 58 | 2-1000L150-54 | 10" | 54 | 16'-0' | 2 point | 5,449 | 2,200 |
| 59 | 2-1000L150-68 | 10" | 68 | $12^{\prime}-0^{\prime \prime}$ | 2 point | 9,043 | 4,200 |
| 60 | 2-1000L150-68 | 10 " | 68 | 12'-0" | 2 point | 8,890 | 3,850 |
| 61 | 2-1000L150-68 | 10" | 68 | 16'-0'' | 2 point | 7,232 | 2,450 |
| 62 | 2-1000L150-68 | 10" | 68 | 16'-0' | 2 point | 6,857 | 2,600 |
| 63 | $2-800 \mathrm{~L} 150-43^{3}$ | 8 " | 54 | $3^{\prime}-0$ " | 1 point | 6,800 | 6,500 |

TABLE 4 TESTED RESULTS - GRAVITY LOADS (continued)

| Test <br> No. | L-Header <br> Designation | Angle <br> Size <br> (inches) | Angle <br> Thickness <br> (mils) | Header <br> Clear <br> Span | Loading <br> Config- <br> uration | Ultimate <br> Load <br> (lb.) | Load At <br> L/240 <br> Deflection <br> (lb.) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 64 | $2-600 \mathrm{~L} 150-43^{3}$ | $3^{\prime \prime}$ | 43 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 3,190 | 2,898 |
| 65 | $2-1000 \mathrm{~L} 150-54^{3}$ | $6^{\prime \prime}$ | 54 | $6^{\prime}-0^{\prime \prime}$ | 2 point | 7,812 | 7,090 |
| 66 | $2-800 \mathrm{~L} 150-54^{3}$ | $8^{\prime \prime}$ | 54 | $6^{\prime}-0^{\prime \prime}$ | 2 point | 6,940 | 5,712 |
| 67 | $2-800 \mathrm{~L} 150-43^{3}$ | $8^{\prime \prime}$ | 43 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 4,351 | 4,100 |
| 68 | $2-800 \mathrm{~L} 150-54^{4}$ | $8^{\prime \prime}$ | 54 | $6^{\prime}-0^{\prime \prime}$ | 2 point | 7,418 | 6,200 |
| 69 | $2-600 \mathrm{~L} 150-43^{4}$ | $6^{\prime \prime}$ | 43 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 3,586 | 3,000 |
| 70 | $2-800 \mathrm{~L} 150-54^{4}$ | $8^{\prime \prime}$ | 54 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 5,146 | - |
| 71 | $2-800 \mathrm{~L} 150-54^{5}$ | $8^{\prime \prime}$ | 54 | $3^{\prime}-0^{\prime \prime}$ | 1 point | 8,305 | - |

Notes to Table 4:
${ }^{1}$ Ultimate load is the total vertical load applied to the header.
${ }^{2}$ Ends of header near supports buckled prematurely. Tested values varied by more than 15 percent.
Specimens were unstable and hence additional tests were not performed.
${ }^{3}$ Screw size was changed to investigate the impact of screw size on header's capacity.
${ }^{4}$ Tested with 33 mil (20 gauge) end studs.
${ }^{5}$ Tested with no cripple stud at load point.
${ }^{4}$ Tested with 6 " wide bearing plate at load point.

TABLE 5 TESTED RESULTS - UPLIFT LOADS

| Test <br> No. | L-Header Designation | Header Clear Span | Angle Size (inches) | Angle Thickness (mils) | Loading Configur ation | Ultimate Load (lb.) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 2-L600150-43 | 3'-0" | 6 " | 43 | 1 point | 1,375 |
| 2 | 2-L600150-43 | 3'-0" | 6 " | 43 | 1 point | 1,260 |
| 3 | 2-L600150-54 | $3^{\prime}-0^{\prime \prime}$ | 6 " | 54 | 1 point | 1,800 |
| 4 | 2-L600150-54 | $3^{\prime}-0^{\prime \prime}$ | $6 "$ | 54 | 1 point | 1,701 |
| 5 | 2-L600150-54 | 6'-0" | 6 " | 54 | 2 point | 1,188 |
| 6 | 2-L600150-54 | 6'-0" | 6 " | 54 | 2 point | 1,256 |
| 7 | 2-L800150-43 | 3'-0" | 8" | 43 | 1 point | 1,305 |
| 8 | 2-L800150-43 | 3'-0" | 8" | 43 | 1 point | 1,280 |
| 9 | 2-L800150-43 | 6'-0" | 8" | 43 | 2 point | 1,004 |
| 10 | 2-L800150-43 | 6'-0" | 8" | 43 | 2 point | 936 |
| 11 | 2-L800150-43 | $8^{\prime}-0^{\prime \prime}$ | 8" | 43 | 2 point | 623 |
| 12 | 2-L800150-43 | $8^{\prime}-0$ " | 8" | 43 | 2 point | 697 |
| 17 | 2-L800150-54 | 3'-0" | 8" | 54 | 1 point | 2,530 |
| 18 | 2-L800150-54 | 3'-0" | 8" | 54 | 1 point | 2,608 |
| 19 | 2-L800150-54 | 6'-0" | 8" | 54 | 2 point | 1,697 |
| 20 | 2-L800150-54 | 6'-0" | 8" | 54 | 2 point | 1,876 |
| 21 | 2-L800150-54 | $8^{\prime}-0^{\prime \prime}$ | 8" | 54 | 2 point | 1,500 |
| 22 | 2-L800150-54 | $8^{\prime}-0^{\prime \prime}$ | 8" | 54 | 2 point | 1,436 |
| 23 | 2-L800150-54 | $12^{\prime}-0^{\prime \prime}$ | 8" | 54 | 2 point | 1,138 |
| 24 | 2-L800150-54 | 12'-0" | 8" | 54 | 2 point | 1,202 |
| 25 | 2-L1000150-43 | $3^{\prime}-0^{\prime \prime}$ | 10" | 43 | 1 point | 2,066 |
| 26 | 2-L1000150-43 | 3'-0" | $10^{\prime \prime}$ | 43 | 1 point | 2,134 |
| 27 | 2-L1000150-43 | 6'-0" | 10" | 43 | 2 point | 1,681 |
| 28 | 2-L1000150-43 | 6'-0" | $10^{\prime \prime}$ | 43 | 2 point | 1,791 |
| 29 | 2-L1000150-43 | $8^{\prime}-0^{\prime \prime}$ | 10" | 43 | 2 point | 987 |
| 30 | 2-L1000150-43 | $8^{\prime}-0^{\prime \prime}$ | 10" | 43 | 2 point | 934 |
| 31 | 2-L1000150-54 | $3^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 1 point | 2,935 |
| 32 | 2-L1000150-54 | 3'-0" | 10" | 54 | 1 point | 2,830 |

TABLE 5 TESTED RESULTS - UPLIFT LOADS (continued)

| Test <br> No. | L-Header Size | Header <br> Clear <br> Span | Angle <br> Size <br> (inches) | Angle <br> Thickness <br> (mils) | Loading <br> Configuration | Ultimate <br> Load <br> (lb.) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 33 | 2-L1000150-54 | $6^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 2 point | 1,801 |
| 34 | 2-L1000150-54 | $6^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 2 point | 1,831 |
| 35 | 2-L1000150-54 | $8^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 2 point | 1,167 |
| 36 | 2-L1000150-54 | $8^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 2 point | 1,180 |
| 37 | 2-L1000150-54 | $12^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 2 point | 788 |
| 38 | 2-L1000150-54 | $12^{\prime}-0^{\prime \prime}$ | $10^{\prime \prime}$ | 54 | 2 point | 832 |

TABLE 6 DOUBLE L-ANGLE TEST RESULTS

| $\begin{array}{\|l\|} \hline \text { Test } \\ \text { No. } \end{array}$ | Leg in. | Thickness in. | $\begin{aligned} & \mathrm{L} \\ & \text { in. } \end{aligned}$ | Leg / t | L / Leg | $\begin{gathered} \hline \mathbf{P} \\ \text { kips } \end{gathered}$ | $\begin{gathered} \mathbf{M}_{\mathbf{t}} \\ \text { kip-in. } \end{gathered}$ | $\begin{gathered} \mathbf{M}_{\mathrm{n}} \\ \text { kip-in. } \end{gathered}$ | $M_{t} / M_{n}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 6 | 0.0459 | 36.000 | 129.719 | 6.000 | 3.194 | 28.746 | 31.040 | 0.926 |
| 2 | 6 | 0.0459 | 36.000 | 129.719 | 6.000 | 3.282 | 29.538 | 31.040 | 0.952 |
| 3 | 6 | 0.0575 | 36.000 | 103.348 | 6.000 | 5.586 | 50.274 | 55.600 | 0.904 |
| 6 | 6 | 0.0575 | 72.000 | 103.348 | 12.000 | 5.652 | 67.824 | 55.600 | 1.220 |
| 7 | 6 | 0.0575 | 72.000 | 103.348 | 12.000 | 6.120 | 73.440 | 55.600 | 1.321 |
| 8 | 6 | 0.0575 | 72.000 | 103.348 | 12.000 | 7.112 | 85.344 | 55.600 | 1.535 |
| 9 | 6 | 0.0575 | 72.000 | 103.348 | 12.000 | 6.201 | 74.412 | 55.600 | 1.338 |
| 10 | 8 | 0.0342 | 36.000 | 232.918 | 4.500 | 4.302 | 38.718 | 34.260 | 1.130 |
| 11 | 8 | 0.0342 | 36.000 | 232.918 | 4.500 | 4.634 | 41.706 | 34.260 | 1.217 |
| 12 | 8 | 0.0465 | 36.000 | 171.043 | 4.500 | 5.962 | 53.658 | 52.070 | 1.030 |
| 13 | 8 | 0.0465 | 36.000 | 171.043 | 4.500 | 5.516 | 49.644 | 52.070 | 0.953 |
| 14 | 8 | 0.0465 | 72.000 | 171.043 | 9.000 | 5.200 | 62.400 | 52.070 | 1.198 |
| 15 | 8 | 0.0465 | 72.000 | 171.043 | 9.000 | 5.150 | 61.800 | 52.070 | 1.187 |
| 16 | 8 | 0.0465 | 96.000 | 171.043 | 12.000 | 5.136 | 82.176 | 52.070 | 1.578 |
| 17 | 8 | 0.0465 | 96.000 | 171.043 | 12.000 | 5.452 | 87.232 | 52.070 | 1.675 |
| 24 | 8 | 0.055 | 72.000 | 144.455 | 9.000 | 7.154 | 85.848 | 93.150 | 0.922 |
| 25 | 8 | 0.055 | 72.000 | 144.455 | 9.000 | 7.230 | 86.760 | 93.150 | 0.931 |
| 26 | 8 | 0.055 | 96.000 | 144.455 | 12.000 | 7.244 | 115.904 | 93.150 | 1.244 |
| 27 | 8 | 0.055 | 96.000 | 144.455 | 12.000 | 7.252 | 116.032 | 93.150 | 1.246 |
| 28 | 8 | 0.055 | 138.000 | 144.455 | 17.250 | 4.200 | 96.600 | 93.150 | 1.037 |
| 29 | 8 | 0.055 | 138.000 | 144.455 | 17.250 | 4.200 | 96.600 | 93.150 | 1.037 |
| 32 | 10 | 0.0475 | 72.000 | 209.526 | 7.200 | 6.986 | 83.832 | 82.950 | 1.011 |
| 33 | 10 | 0.0475 | 72.000 | 209.526 | 7.200 | 7.344 | 88.128 | 82.950 | 1.062 |
| 34 | 10 | 0.0475 | 96.000 | 209.526 | 9.600 | 4.130 | 66.080 | 82.950 | 0.797 |
| 35 | 10 | 0.0475 | 96.000 | 209.526 | 9.600 | 4.138 | 66.208 | 82.950 | 0.798 |
| 36 | 10 | 0.0341 | 36.000 | 292.255 | 3.600 | 4.494 | 40.446 | 40.960 | 0.987 |
| 37 | 10 | 0.0341 | 36.000 | 292.255 | 3.600 | 4.232 | 38.088 | 40.960 | 0.930 |
| 38 | 10 | 0.0341 | 72.000 | 292.255 | 7.200 | 4.742 | 56.904 | 40.960 | 1.389 |
| 39 | 10 | 0.0341 | 72.000 | 292.255 | 7.200 | 3.822 | 45.864 | 40.960 | 1.120 |
| 40 | 6 | 0.0339 | 36.000 | 175.991 | 6.000 | 2.238 | 20.142 | 20.300 | 0.992 |
| 41 | 6 | 0.0339 | 36.000 | 175.991 | 6.000 | 2.314 | 20.826 | 20.300 | 1.026 |
| 44 | 10 | 0.0575 | 72.000 | 172.913 | 7.200 | 8.616 | 103.392 | 131.780 | 0.785 |
| 45 | 10 | 0.0575 | 72.000 | 172.913 | 7.200 | 7.722 | 92.664 | 131.780 | 0.703 |
| 46 | 10 | 0.0575 | 96.000 | 172.913 | 9.600 | 6.646 | 106.336 | 131.780 | 0.807 |
| 47 | 10 | 0.0575 | 96.000 | 172.913 | 9.600 | 6.400 | 102.400 | 131.780 | 0.777 |
| 48 | 10 | 0.0575 | 96.000 | 172.913 | 9.600 | 6.706 | 107.296 | 131.780 | 0.814 |
| 49 | 8 | 0.055 | 192.000 | 144.455 | 24.000 | 3.248 | 103.936 | 93.150 | 1.116 |
| 50 | 8 | 0.055 | 192.000 | 144.455 | 24.000 | 3.136 | 100.352 | 93.150 | 1.077 |
| 51 | 8 | 0.0719 | 144.000 | 110.266 | 18.000 | 7.711 | 185.064 | 121.830 | 1.519 |
| 52 | 8 | 0.0719 | 144.000 | 110.266 | 18.000 | 6.721 | 161.304 | 121.830 | 1.324 |
| 53 | 8 | 0.0719 | 192.000 | 110.266 | 24.000 | 5.578 | 178.496 | 121.830 | 1.465 |
| 54 | 8 | 0.0719 | 192.000 | 110.266 | 24.000 | 5.850 | 187.200 | 121.830 | 1.537 |
| 55 | 10 | 0.0575 | 144.000 | 172.913 | 14.400 | 6.458 | 154.992 | 131.780 | 1.176 |
| 56 | 10 | 0.0575 | 144.000 | 172.913 | 14.400 | 6.542 | 157.008 | 131.780 | 1.191 |
| 57 | 10 | 0.0575 | 192.000 | 172.913 | 19.200 | 5.428 | 173.696 | 131.780 | 1.318 |
| 58 | 10 | 0.0575 | 192.000 | 172.913 | 19.200 | 5.449 | 174.368 | 131.780 | 1.323 |
| 59 | 10 | 0.0719 | 144.000 | 138.082 | 14.400 | 9.043 | 217.032 | 177.270 | 1.224 |
| 60 | 10 | 0.0719 | 144.000 | 138.082 | 14.400 | 8.890 | 213.360 | 177.270 | 1.204 |
| 61 | 10 | 0.0719 | 192.000 | 138.082 | 19.200 | 7.232 | 231.424 | 177.270 | 1.305 |
| 62 | 10 | 0.0719 | 192.000 | 138.082 | 19.200 | 6.857 | 219.424 | 177.270 | 1.238 |
| 64 | 6 | 0.0459 | 36.000 | 129.719 | 6.000 | 3.190 | 28.710 | 31.040 | 0.925 |
| 65 | 10 | 0.0575 | 72.000 | 172.913 | 7.200 | 7.812 | 93.744 | 131.780 | 0.711 |
| 66 | 8 | 0.055 | 72.000 | 144.455 | 9.000 | 6.940 | 83.280 | 93.150 | 0.894 |
| 68 | 8 | 0.055 | 72.000 | 144.455 | 9.000 | 7.418 | 89.016 | 93.150 | 0.956 |
| 69 | 6 | 0.0459 | 36.000 | 129.719 | 6.000 | 3.586 | 32.274 | 31.040 | 1.040 |
|  |  |  |  |  |  |  |  | Mean | 1.111 |
|  |  |  |  |  |  |  |  | Std. Dev. | 0.233 |
|  |  |  |  |  |  |  |  | COV | 0.210 |

TABLE 7 10" DOUBLE L-ANGLE RESULTS

| Test No. | $\begin{aligned} & \hline \text { Leg } \\ & \text { in. } \end{aligned}$ | Thickness in. | $\begin{aligned} & \mathrm{L} \\ & \text { in. } \end{aligned}$ | Leg/t | L/Leg | $\begin{gathered} P \\ \text { kips } \end{gathered}$ | $\begin{gathered} M_{t} \\ \text { kip-in. } \end{gathered}$ | $\begin{gathered} M_{c} \\ \text { kip-in. } \end{gathered}$ | $M_{t} / M_{c}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 32 | 10 | 0.0475 | 72.000 | 209.526 | 7.200 | 6.986 | 83.832 | 82.950 | 1.011 |
| 33 | 10 | 0.0475 | 72.000 | 209.526 | 7.200 | 7.344 | 88.128 | 82.950 | 1.062 |
| 34 | 10 | 0.0475 | 96.000 | 209.526 | 9.600 | 4.130 | 66.080 | 82.950 | 0.797 |
| 35 | 10 | 0.0475 | 96.000 | 209.526 | 9.600 | 4.138 | 66.208 | 82.950 | 0.798 |
| 36 | 10 | 0.0341 | 36.000 | 292.255 | 3.600 | 4.494 | 40.446 | 40.960 | 0.987 |
| 37 | 10 | 0.0341 | 36.000 | 292.255 | 3.600 | 4.232 | 38.088 | 40.960 | 0.930 |
| 38 | 10 | 0.0341 | 72.000 | 292.255 | 7.200 | 4.742 | 56.904 | 40.960 | 1.389 |
| 39 | 10 | 0.0341 | 72.000 | 292.255 | 7.200 | 3.822 | 45.864 | 40.960 | 1.120 |
| 44 | 10 | 0.0575 | 72.000 | 172.913 | 7.200 | 8.616 | 103.392 | 131.780 | 0.785 |
| 45 | 10 | 0.0575 | 72.000 | 172.913 | 7.200 | 7.722 | 92.664 | 131.780 | 0.703 |
| 46 | 10 | 0.0575 | 96.000 | 172.913 | 9.600 | 6.646 | 106.336 | 131.780 | 0.807 |
| 47 | 10 | 0.0575 | 96.000 | 172.913 | 9.600 | 6.400 | 102.400 | 131.780 | 0.777 |
| 48 | 10 | 0.0575 | 96.000 | 172.913 | 9.600 | 6.706 | 107.296 | 131.780 | 0.814 |
| 55 | 10 | 0.0575 | 144.000 | 172.913 | 14.400 | 6.458 | 154.992 | 131.780 | 1.176 |
| 56 | 10 | 0.0575 | 144.000 | 172.913 | 14.400 | 6.542 | 157.008 | 131.780 | 1.191 |
| 57 | 10 | 0.0575 | 192.000 | 172.913 | 19.200 | 5.428 | 173.696 | 131.780 | 1.318 |
| 58 | 10 | 0.0575 | 192.000 | 172.913 | 19.200 | 5.449 | 174.368 | 131.780 | 1.323 |
| 59 | 10 | 0.0719 | 144.000 | 138.082 | 14.400 | 9.043 | 217.032 | 177.270 | 1.224 |
| 60 | 10 | 0.0719 | 144.000 | 138.082 | 14.400 | 8.890 | 213.360 | 177.270 | 1.204 |
| 61 | 10 | 0.0719 | 192.000 | 138.082 | 19.200 | 7.232 | 231.424 | 177.270 | 1.305 |
| 62 | 10 | 0.0719 | 192.000 | 138.082 | 19.200 | 6.857 | 219.424 | 177.270 | 1.238 |
| 65 | 10 | 0.0575 | 72.000 | 172.913 | 7.200 | 7.812 | 93.744 | 131.780 | 0.711 |
|  |  |  |  |  |  |  |  | Mean | 1.013 |
|  |  |  |  |  |  |  |  | Std. | 0.313 |
|  |  |  |  |  |  |  |  | Dev. |  |

TABLE 8 10" DOUBLE L-ANGLE WITH L/LEG < 10"

| Test <br> No. | Leg <br> in. | Thickness | Leg/t | L/Leg | $\mathbf{P}$ <br> kips | $\mathbf{L}$ <br> $\mathbf{i n}$. | $\mathbf{M}_{\mathbf{t}}$ <br> kip-in. | $\mathbf{M}_{\mathbf{c}}$ <br> kip-in. | $\mathbf{M}_{\mathbf{t}} / \mathbf{M}_{\mathbf{c}}$ | $\mathbf{M}_{\mathbf{t}} /\left(\mathbf{0 . 8} \mathbf{M}_{\mathbf{c}}\right)$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 34 | 10 | 0.0475 | 209.526 | 9.600 | 4.130 | 96.000 | 66.080 | 82.950 | 0.797 | 0.996 |
| 35 | 10 | 0.0475 | 209.526 | 9.600 | 4.138 | 96.000 | 66.208 | 82.950 | 0.798 | 0.998 |
| 45 | 10 | 0.0575 | 172.913 | 7.200 | 7.722 | 72.000 | 92.664 | 131.780 | 0.703 | 0.879 |
| 46 | 10 | 0.0575 | 172.913 | 9.600 | 6.646 | 96.000 | 106.336 | 131.780 | 0.807 | 1.009 |
| 47 | 10 | 0.0575 | 172.913 | 9.600 | 6.400 | 96.000 | 102.400 | 131.780 | 0.777 | 0.971 |
| 48 | 10 | 0.0575 | 172.913 | 9.600 | 6.706 | 96.000 | 107.296 | 131.780 | 0.814 | 1.018 |
| 65 | 10 | 0.0575 | 172.913 | 7.200 | 7.812 | 72.000 | 93.744 | 131.780 | 0.711 | 0.889 |
|  |  |  |  |  |  |  |  |  | Mean | 0.773 |
|  |  |  |  |  |  |  |  | Std. Dev. | 0.347 | 0.366 |
|  |  |  |  |  |  |  |  | 0.323 |  |  |

TABLE 9 DOUBLE L-ANGLE RESULTS - UPLIFT LOADS

| Test <br> No. | Leg <br> in. | Thickness <br> in. | $\mathbf{P}$ <br> kips | $\mathbf{L}$ <br> in. | $\mathbf{M}_{\mathbf{t}}$ <br> kip-in. | $\mathbf{M}_{\mathbf{n}}$ <br> kip-in. | $\mathbf{M}_{\mathbf{t}} / \mathbf{M}_{\mathbf{n}}$ | $\mathbf{M}_{\mathbf{t}} / \mathbf{R M}_{\mathbf{n}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5 | 6 | 0.0575 | 1.188 | 72 | 14.256 | 55.5 | 0.257 | 1.027 |
| 6 | 6 | 0.0575 | 1.256 | 72 | 15.072 | 55.5 | 0.272 | 1.086 |
| 9 | 8 | 0.0465 | 1.004 | 72 | 12.048 | 52.06 | 0.231 | 1.157 |
| 10 | 8 | 0.0465 | 0.936 | 72 | 11.232 | 52.06 | 0.215 | 1.079 |
| 11 | 8 | 0.0465 | 0.623 | 96 | 9.968 | 52.06 | 0.191 | 0.957 |
| 12 | 8 | 0.0465 | 0.697 | 96 | 11.152 | 52.06 | 0.214 | 1.071 |
| 19 | 8 | 0.055 | 1.697 | 72 | 20.364 | 93.15 | 0.219 | 0.874 |
| 20 | 8 | 0.055 | 1.876 | 72 | 22.512 | 93.15 | 0.242 | 0.967 |
| 21 | 8 | 0.055 | 1.5 | 96 | 24 | 93.15 | 0.258 | 1.031 |
| 22 | 8 | 0.055 | 1.436 | 96 | 22.976 | 93.15 | 0.247 | 0.987 |
| 23 | 8 | 0.055 | 1.138 | 144 | 27.312 | 93.15 | 0.293 | 1.173 |
| 24 | 8 | 0.055 | 1.202 | 144 | 28.848 | 93.15 | 0.310 | 1.239 |
| 27 | 10 | 0.0475 | 1.681 | 72 | 20.172 | 82.95 | 0.243 | 1.216 |
| 28 | 10 | 0.0475 | 1.791 | 72 | 21.492 | 82.95 | 0.260 | 1.295 |
| 29 | 10 | 0.0475 | 0.987 | 96 | 15.792 | 82.95 | 0.190 | 0.952 |
| 30 | 10 | 0.0475 | 0.934 | 96 | 14.944 | 82.95 | 0.180 | 0.901 |
| 33 | 10 | 0.0575 | 1.801 | 72 | 21.612 | 131.78 | 0.164 | 0.820 |
| 34 | 10 | 0.0575 | 1.831 | 72 | 21.972 | 131.78 | 0.167 | 0.834 |
| 35 | 10 | 0.0575 | 1.167 | 96 | 18.672 | 131.78 | 0.142 | 0.945 |
| 36 | 10 | 0.0575 | 1.18 | 96 | 18.88 | 131.78 | 0.143 | 0.955 |
| 37 | 10 | 0.0575 | 0.788 | 144 | 18.912 | 131.78 | 0.144 | 0.957 |
| 38 | 10 | 0.0575 | 0.832 | 144 | 19.968 | 131.78 | 0.152 | 1.010 |
|  |  |  |  |  |  | Mean | 0.215 | 1.024 |
|  |  |  |  |  |  | Std. Dev. | 0.049 | 0.126 |
|  |  |  |  |  |  | COV | 0.229 | 0.123 |



Figure 1- L-Header Configuration


Figure 2 - L-Header Test Apparatus - Gravity Load Test


Figure 3 - L-Header Test Apparatus - Uplift Load Test I


Figure 4 - L-Header Test Apparatus - Uplift Load Test II


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    ${ }^{2}$ Distinguished Teaching Prof., Dept. of Civil Eng., University of Missouri-Rolla

[^1]:    ${ }^{2}$ Tested with both 350T162-33 and 350T162-43 top tracks.
    ${ }^{2}$ Tested with both \#8 and \#10 screws.
    ${ }^{3}$ Tested with 350T162-33 and 33 mil king studs

