Near-wall modelling in Eulerian-Eulerian simulations

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Abstract

The near-wall region in turbulent Eulerian-Eulerian (E-E) simulations has hitherto received little to no attention. A standard approach to modelling this region is through the employment of single-phase wall-functions in the fluid-phase and it is unclear whether such an approach is capable of capturing the turbulent fluid-particle interaction in the near-wall region. In order to both investigate and alleviate E-E models reliance on single-phase wall-functions we propose an E-E elliptic relaxation model to account for the near-wall non-homogeneity which arises in wall-bounded flows. The proposed model is derived within an E-E framework and enables the full resolution of the boundary layer and arbitrary wall sensitivity. The model is then compared against the conventional $k_f - \varepsilon_f$ turbulence model with standard single-phase wall-functions. Additionally, the modelling is compared against a low-Re number turbulence model. The elliptic relaxation model is implemented within the open-source CFD toolbox OpenFOAM, applied to a vertical downward-facing channel and validated against the benchmark experimental data of Kulick et al.

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[20]. Model results show marked improvements over the conventional turbulence model across mean flow and turbulence statistics predictions. The use of conventional single-phase wall functions were shown to negatively impede on the prediction of the velocity covariance coupling term and as a result the particle fluctuation energy. Moreover, this also lead to an underestimation of the near-wall volume fraction accumulation. Finally, the elliptic relaxation model, E-E model and accompanying validation cases are made open-source.

Keywords: v2-f, Boundary-layer, E-E, Near-wall

1 1. Introduction

Many researchers have investigated turbulent wall-bounded fluid-particle flow 2 through experimentation [4, 11, 15, 20, 23, 49]. Owing to their turbulent nature such 3 flows exhibit complex physical behaviour giving rise to turbulence modulation which 4 can be caused by: particle-particle, fluid-particle and/or particle-wall interactions. 5 Thus, research has been ongoing to understand and model these phenomena; most 6 of which are common in engineering processes, e.g. pneumatic conveyance and coal 7 particle combustion. The aforementioned experimental studies provide invaluable 8 physical insights and validation data for the development of predictive models. One 9 notable study is that of Kulick et al. [20] which has received considerable attention 10 from researchers developing, predominantly Euler-Lagrange (E-L), models [19, 25, 42, 11 47, 51, 52, 54]. This study is particularly attractive as there are several particle classes 12 giving rise to various particle-fluid and particle-wall interactions which contribute to 13 turbulence modulation. 14

Having identified the aspects of physical behaviour which are significant in these 15 flows, researchers can investigate them separately in a reductionist approach. We 16 now highlight some studies that contribute to the understanding of particle behaviour 17 within the case of Kulick et al. [20]; starting with the so-called feedback-force of the 18 particle phase on the fluid turbulence in the flow. Vreman [51] recently examined the 19 effect of the mean feedback-force and how it is exacerbated by wall roughness. An 20 increase in wall roughness enhances turbulence attenuation i.e. a reduction in fluid-21 phase velocity fluctuations. This explains the over prediction of the mean particle 22 velocities seen in previous studies Kubik and Kleiser [19], Wang and Squires [52], 23

²⁴ Yamamoto et al. ⁵⁴ as smooth walls were simulated.

Another phenomenon that has been investigated is turbophoresis, which refers to 25 the tendency of particles in the flow to migrate towards regions of lower turbulence. 26 The turbophoresis force is responsible for particles drifting from regions of high tur-27 bulence intensity to low turbulence intensity [27, 35], which often results in particles 28 accumulating in the near-wall region characterised by low-speed streaks [32, 34]. 29 This accumulation in the near-wall region is referred to as deposition and has been 30 researched numerically by [24, 27, 28, 30]. One of the first models for particle deposi-31 tion by Young and Leeming [55] showed that the turbophoretic velocity depends on 32 the gradient of wall-normal fluctuating velocities and provided one of the first physi-33 cal basis for explaining the turbophoresis force. Strömgren et al. [47] investigated the 34 effect of the turbophoresis force within an Eulerian-Eulerian (E-E) framework and 35 found that even for small volume fractions, $\alpha_p = 2 \times 10^{-4}$, two-way coupling effects 36 are non-negligible and the near-wall region may require special attention. This is due 37 to the accumulation of particles in the near-wall region i.e viscous sub layer, leading 38 to higher volume fractions in which two-way coupling effects become more relevant 39 14. 40

In turbulent single-phase simulations the near-wall region is typically modelled. Wall functions are applied to turbulence quantities, $\varepsilon_f \& \nu_{ft}$ with a zero gradient condition given to k_f , in order to avoid the computational overhead of detailed resolution of the flow in the near-wall region. Such wall functions are based on the lawof-the-wall, which is that the dimensionless velocity, u^+ varies through some function expressed generically as, $u^+ = f_{log}(y^+)$. The function f_{log} is logarithmic representing the outer log-law region of the turbulent boundary layer. This corresponds to the
constant-stress layer in which the turbulent shear stress is proportional to the friction velocity [45]. In turbulent quantity terms this means that the production and
dissipation of turbulent kinetic energy are equal.

The law-of-the-wall is assumed to be universal and is found through dimensional 51 reasoning, this then leads to a description of the near-wall region through dimen-52 sionless variables i.e. velocity and wall-normal coordinates. The dimensionless wall-53 normal coordinate is defined as $y^+ = y u_\tau / \nu_f$ and the log-law is applicable in the 54 range of $30 < y^+ < 300$, this then gives a universal relation that can be applied to 55 turbulent wall-bounded flows. This criterion places a requirement on the first com-56 putational cell i.e. the distance of the cell centre must be further than $y^+ > 30$. As 57 can already be deduced, the calculation of y^+ depends on the friction velocity, which 58 is not known *a priori*. Hence, this quantity is estimated prior to calculation using 59 standard skin friction relations and informs mesh generation. This approach then 60 sacrifices near-wall resolution for a computationally cheaper simulation. An impor-61 tant assumption about the nature of the law-of-the-wall has been made throughout 62 i.e. its universal nature. This is in fact not true as it has been shown experimen-63 tally that the boundary layer is affected by adverse pressure gradients and geometric 64 changes [16]. 65

The viability of single-phase wall functions applied to multiphase simulations has attracted some interest from researchers. A theoretical study by Rizk and Elghobashi [40] showed that increasing volume fraction can adversely effect the mean profile prediction. It was found that with increasing particle volume fraction the log layer broke

down resulting in an overestimation. Interestingly, a similar relationship between an 70 increasing mass loading and a reduction in the mean log-layer momentum was re-71 cently found experimentally by Saber et al. [44]. Benyahia et al. [2] included the 72 effect of the particle phase directly into the wall function. An additional term that 73 contains the drag and velocity fluctuation covariance is introduced in the log-law re-74 lation. This formulation allows the presence of the particles to influence the velocity 75 profile, although when extended to more complex geometries the short-comings of 76 single-phase wall functions remain. 77

Attempts to circumvent the reliance on single-phase wall functions have been 78 made by several authors [3, 6, 40, 57] in which a low-Re number turbulence model 79 is used. This allows the transport equations to be integrated up to the wall. This 80 approach has proven fruitful for numerous authors as without the use of wall func-81 tions, the presence of the particles within the boundary layer can exert their influence 82 [40, 57]. The low-Re turbulence model uses a damping function and a near wall cor-83 rection of Kolmogorov scaling [31]. The damping of the viscosity can be somewhat 84 arbitrary and validated on relatively simple flow leading to a range of different mod-85 els [5, 21, 41, 46] with an extensive summary found in Patel et al. [31]. The damping 86 functions used in Patel et al. [31] are often non-linear and can lead to numerical 87 stiffness further complicating their application. 88

⁸⁹ Durbin [8] proposes another way of accounting for wall-induced non-homogeneity. ⁹⁰ The quantity $\overline{v_f^2}$, which represents the turbulence-stress normal to streamlines, is in-⁹¹ troduced. This quantity is derived from the exact Reynolds-stress transport equation ⁹² and contains a source term that accounts for the redistribution of turbulence kinetic energy. This inclusion explicitly accounts for the wall-induced non-homogeneity and enables the wall-normal component to be dampened. The energy redistribution is governed by an *elliptic relaxation* equation f, that is free of geometric dependence or arbitrary fitting. The $\overline{v_f^2} - f$ elliptic relaxation model has been validated across various challenging single-phase flows [1, 7, 26, 29, 48] highlighting the benefit of such a modelling technique.

There are two closely linked issues with the current E-E modelling approaches: 99 the modelling of the near-wall region, through single-phase wall functions, and the 100 subsequent consequences of such an approach i.e. the prediction of turbulence mod-101 ulation and turbophoresis. In this work we seek to investigate this by carrying out 102 a side-by-side comparison of a conventional E-E simulation method with a newly-103 derived elliptic relaxation model in which the near-wall region has been resolved. 104 The main aim then is to reveal the consequences of modelling the near-wall region 105 whilst proposing new modelling to circumvent these consequences. 106

We begin at a recently proposed E-E model, namely the Reynolds-Averaged Two-107 Fluid Model of Fox [13]. This approach has proven particularly fruitful in modelling 108 high Re number flows due to the inclusion of particle inertia induced energy sep-109 aration (see Février et al. [12], Fox [13]) and has lead to a high level of validation 110 [38, 39]. The elliptic relaxation model of Durbin [8] is derived within the RA-TFM 111 framework and applied to the vertical downward facing channel of Kulick et al. [20]. 112 The elliptic relaxation model alleviates the use of wall functions and/or the use of 113 ad-hoc damping functions and their geometric dependency. To ascertain the con-114 sequences of a conventional E-E simulation, the RA-TFM with the solution of the 115

 $k_f - \varepsilon_f$ model, and is compared and contrasted against the newly proposed elliptic relaxation model, $\overline{v_f^2} - f$. Moreover, results are also compared against the low Re number model of Launder and Sharma [21] in order to facilitate a boundary layer resolved comparison.

¹²⁰ 2. Numerical model

We begin at the RA-TFM of Fox [13], and as we are interested in extending the fluid-phase turbulence modelling we present the fluid- and particle-phase governing equations as well as the particle-phase fluctuation energy equations in Table 1. We have neglected coupling through buoyancy due to the high density ratios simulated in this work and therefore, we begin at the fluid-phase turbulence equations.

The turbulent kinetic energy transport equation for the fluid-phase takes the form:

$$\frac{\partial(\alpha_f \rho_f k_f)}{\partial t} + \nabla \cdot (\alpha_f \rho_f k_f \mathbf{u}_f) = \nabla \cdot \left(\mu_t + \frac{\mu_{ft}}{\sigma_{fk}}\right) \nabla k_f + \alpha_f \rho_f \Pi_f - \alpha_f \rho_f \varepsilon_f$$
(1)
+2\beta(k_{fp} - k_f).

¹²⁸ The turbulent kinetic energy dissipation transport equation reads:

$$\frac{\partial(\alpha_f \rho_f \varepsilon_f)}{\partial t} + \nabla \cdot (\alpha_f \rho_f \varepsilon_f \mathbf{u}_f) = \nabla \cdot \left(\mu_t + \frac{\mu_{ft}}{\sigma_{fk}}\right) \nabla \varepsilon_f + \frac{\varepsilon_f}{k_f} \left[C_1 \alpha_f \rho_f \Pi_f - C_2 \alpha_f \rho_f \varepsilon_f\right] + 2C_3 \beta(\varepsilon_{fp} - \varepsilon_f).$$
(2)

¹²⁹ The first term on the RHS is the fluid-phase turbulent kinetic energy/dissipation ¹³⁰ flux. The second term, Π_f is the kinetic energy production due to mean shear with the third term being the turbulent kinetic energy dissipation. The remaining term is the coupling terms due to velocity covariance (see Table 9) and is a measure of how correlated the two phases are. This provides the primary coupling mechanism in this work as the two phases are only coupled through drag.

These two equations make up the conventional $k_f - \varepsilon_f$ turbulence model with model constants, largely taken from compressible turbulence modelling [43], found in Table 3. The complete set of equations that make up the RA-TFM are found in Table 1 and the equations associated with the low Re number model of Launder and Sharma [21] can be found in Table 2. Table 1: RA-TFM governing equations and the particle fluctuation energy equations.

$$\frac{\partial(\alpha_p \rho_p)}{\partial t} + \nabla \cdot (\alpha_p \rho_p \mathbf{u}_p) = 0$$
(3)

$$\frac{\partial(\alpha_f \rho_f)}{\partial t} + \nabla \cdot (\alpha_f \rho_f \mathbf{u}_f) = 0 \tag{4}$$

$$\frac{\partial(\alpha_p \rho_p \mathbf{u}_p)}{\partial t} + \nabla \cdot (\alpha_p \rho_p \mathbf{u}_p \mathbf{u}_p) = \nabla \cdot \left(2(\mu_p + \mu_{pt}) \overline{\mathbf{S}}_{\mathbf{p}} \right) + \beta \left[(\mathbf{u}_f - \mathbf{u}_p) - \frac{\nu_{ft}}{\mathrm{Sc}_{fs} \alpha_p \alpha_f} \nabla \alpha_p \right] - \nabla p_p - \alpha_p \nabla p_f + \alpha_p \rho_p \mathbf{g}$$
(5)

$$\frac{\partial(\alpha_f \rho_f \mathbf{u}_f)}{\partial t} + \nabla \cdot (\alpha_f \rho_f \mathbf{u}_f \mathbf{u}_f) = \nabla \cdot \left(2(\mu_f + \mu_{ft}) \overline{\mathbf{S}}_{\mathbf{f}} \right) + \beta \left[(\mathbf{u}_p - \mathbf{u}_f) + \frac{\nu_{ft}}{\mathrm{Sc}_{fs} \alpha_p \alpha_f} \nabla \alpha_p \right] -\alpha_f \nabla p_f + \alpha_f \rho_f \mathbf{g}$$
(6)

$$\frac{\partial(\alpha_p \rho_p k_p)}{\partial t} + \nabla \cdot (\alpha_p \rho_p k_p \mathbf{u}_p) = \nabla \cdot \left(\mu_p + \frac{\mu_{pt}}{\sigma_{pk}}\right) \nabla k_p + \alpha_p \rho_p \Pi_p - \alpha_p \rho_p \varepsilon_p + 2\beta(k_{fp} - k_p)$$
(7)

$$\frac{\partial(\alpha_p\rho_p\varepsilon_p)}{\partial t} + \nabla\cdot(\alpha_p\rho_p\varepsilon_p\mathbf{u}_p) = \nabla\cdot\left(\mu_p + \frac{\mu_{pt}}{\sigma_{p\varepsilon}}\right)\nabla\varepsilon_p + \frac{\varepsilon_p}{k_p}\left[C_1\alpha_p\rho_p\Pi_p - C_2\alpha_p\rho_p\varepsilon_p\right] + 2C_3\beta(\varepsilon_{fp} - \varepsilon_p)$$
(8)

$$\frac{3}{2} \Big[\frac{\partial (\alpha_p \rho_p \Theta_p)}{\partial t} + \nabla \cdot (\alpha_p \rho_p \Theta_p \mathbf{u}_p) \Big] = \nabla \cdot \Big(\kappa_{\Theta} + \frac{3\mu_{pt}}{2Pr_{pt}} \Big) \nabla \Theta_p + 2\mu_p \overline{\mathbf{S}}_{\mathbf{p}} : \overline{\mathbf{S}}_{\mathbf{p}} -p_p \nabla \cdot \mathbf{u}_p + \alpha_p \rho_p \varepsilon_p - 3\beta \Theta_p - \gamma$$

$$(9)$$

Table 2: Low Re number turbulence model of Launder and Sharma [21].

$$\frac{\partial(\alpha_{f}\rho_{f}k_{f})}{\partial t} + \nabla \cdot (\alpha_{f}\rho_{f}k_{f}\mathbf{u}_{f}) = \nabla \cdot \left(\mu_{f} + \frac{\mu_{ft}}{\sigma_{fk}}\right) \nabla k_{f} + \alpha_{f}\rho_{f}\Pi_{f} - \alpha_{f}\rho_{f}\varepsilon_{f} \qquad (10)$$

$$+2\beta(k_{fp} - k_{f})$$

$$\frac{\partial(\alpha_{f}\rho_{f}\widetilde{\varepsilon})}{\partial t} + \nabla \cdot (\alpha_{f}\rho_{f}\widetilde{\varepsilon}\mathbf{u}_{f}) = \nabla \cdot \left(\mu_{f} + \frac{\mu_{ft}}{\sigma_{f\epsilon}}\right) \nabla\widetilde{\varepsilon} + C_{1}\alpha_{f}\rho_{f}\Pi_{f}\frac{\widetilde{\varepsilon}_{f}}{k_{f}} - C_{2}\alpha_{f}\rho_{f}f_{2}\frac{\widetilde{\varepsilon}^{2}}{k_{f}}$$

$$+E + 2C_{3}\beta(\varepsilon_{fp} - \widetilde{\varepsilon}_{f}) \qquad (11)$$

where

$$\mu_{ft} = c_{\mu} f_{\mu} \rho_f \frac{k_f^2}{\tilde{\varepsilon}}$$

$$\varepsilon_f = \tilde{\varepsilon} + D$$

$$f_{\mu} = \exp\left(\frac{-3.4}{(1 + R_T/50)^2}\right)$$

$$f_2 = 1 - 0.3 \exp(-R_T^2)$$

$$D = 2\mu_f (\nabla \sqrt{k_f})^2$$

$$E = 2\mu_f \mu_{ft} (\nabla^2 \mathbf{u}_f)^2$$

$$R_T = \frac{k_f^2}{\nu_f \tilde{\varepsilon}}$$

Table 3: $k_f - \varepsilon_f$ model constants.

C_1	C_2	C_3	β_k	β_{ε}	$C_{f\mu}$	$C_{p\mu}$	$\sigma_{f\epsilon}$	σ_{fk}
1.44	1.92	1	1	1	0.09	0.09	1.3	1

140 2.1. Fluid-phase elliptic relaxation model

We begin at the exact RA Reynolds stress transport equation for the fluid-phase and for the sake of brevity the derivation is presented in the Appendix. The equation then reads:

$$\frac{\partial \langle \alpha_{f} \rangle \langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f}}{\partial t} + \nabla \cdot \langle \alpha_{f} \rangle \langle \mathbf{u}_{f} \rangle_{f} \otimes \langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f} = -\nabla \cdot \langle \alpha_{f} \rangle \langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f} \\
- \underbrace{\langle \alpha_{f} \rangle (\langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{f} \cdot \nabla \langle \mathbf{u}_{f} \rangle_{f})}_{\text{Production}} + \frac{1}{\rho_{f}} \nabla \cdot \langle \overline{\boldsymbol{\sigma}}_{f} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle - \frac{1}{\rho_{f}} \nabla \langle p_{f} \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle}{\frac{1}{\rho_{f}} \langle \overline{\boldsymbol{\sigma}}_{f} \cdot \nabla \mathbf{u}_{f}^{\prime\prime\prime} \rangle} + \frac{1}{\rho_{f}} \nabla \langle \alpha_{f} \rangle \beta (\underbrace{\langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{p} - \langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{p}}_{\text{velocity correlations}}).$$
(12)

Firstly, as we intend to arrive at a transport equation for an 'imaginary' wall-normal stress component a caveat is worth mentioning. If one considers the production term in any classic eddy-viscosity model, the production term is proportional to the mean flow gradient - importantly in the stream-wise direction. This means that the turbulent kinetic energy is produced by the stream-wise mean flow gradients. Consequently, in the wall-normal direction the production term vanishes.

The velocity correlations which arise due to phase coupling are modelled analo-150 gously to those terms found in the $k_f - \varepsilon_f$ transport equations. We set the covariance 151 of the fluctuations $\langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{p}^{\prime\prime} \rangle_{p} = \overline{v_{fp}^{2}} = \beta_{v} \sqrt{\overline{v_{p}^{2} v_{f}^{2}}}$, where $\overline{v_{p}^{2}} = 2/3k_{p}$ owing to its defi-152 nition. The correlation factor, $\beta_v = 1$ along with the correlation factors found in the 153 transport equations for turbulent kinetic energy and dissipation (see Table 9) are all 154 set to 1. This is a crude first approximation and the correlation factor should depend 155 on both mass loading and Stokes number. This is out of the scope of this work but 156 only a weak dependency through the relatively low mass loadings is expected. 157

Following the approach used in classic eddy-viscosity turbulence models, the divergence terms appearing in the transport equation are closed by the eddy-viscosity approximation [33]. This reads as:

$$\nabla \cdot \left[\frac{\mu_{ft}}{\sigma_{fk}} \nabla \langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{f} \right] \approx -\nabla \cdot \langle \alpha_{f} \rangle \langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f} + \frac{1}{\rho_{f}} \nabla \cdot \langle \overline{\boldsymbol{\sigma}}_{f} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle - \frac{1}{\rho_{f}} \nabla \langle p_{f} \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle.$$

$$(13)$$

Finally, the terms left to close are the pressure strain and dissipation terms. These terms are explicitly modelled in the $\overline{v_f^2} - f$ model equations and are grouped into a source term denoted $k_f f$,

$$k_f f = \underbrace{\phi_{yy}}_{\text{pressure strain}} - \underbrace{\varepsilon_{yy}}_{\text{dissipation}} + \alpha_f \rho_f 6 \frac{v_f^2}{k_f} \varepsilon_f.$$
(14)

The source term effectively redistributes turbulence energy from the stream-wise 164 Reynolds stress component to the wall-normal component. This is intuitive as pre-165 viously discussed, when one considers a fully developed turbulent boundary layer 166 as the wall-normal Reynolds stress component's production is zero due to the mean 167 stream-wise flow gradient. This means that turbulence energy can only enter the 168 wall-normal component through redistribution. The original form of the source term 169 has been shown to overproduce in regions relatively far away from the wall and the 170 correction of Davidson et al. [7] is thus employed, this then reads 171

$$\overline{v_{f_{source}}^2} = \min\left\{k_f f, \ -\frac{1}{T} \left[(C_1 - 6)\overline{v_f^2} - \frac{2k_f}{3}(C_1 - 1) \right] + C_2 \Pi_f \right\}.$$
 (15)

¹⁷² Finally, setting the wall-normal component of the fluid-phase Reynolds stress

173 tensor $\langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{f}$ to $\overline{v_{f}^{2}}$ a transport equation can be written as:

$$\frac{\partial(\alpha_f \rho_f \overline{v_f^2})}{\partial t} + \nabla \cdot (\alpha_f \rho_f \overline{v_f^2} \mathbf{u}_f) = \nabla \cdot \left(\mu_f + \frac{\mu_{ft}}{\sigma_{fk}}\right) \nabla \overline{v_f^2} + \overline{v_f^2}_{source} - \alpha_f \rho_f 6 \frac{\overline{v_f^2}}{k_f} \varepsilon_f + 2\beta(\overline{v_{fp}^2} - \overline{v_f^2}).$$
(16)

The reader should note that the third term on the RHS is a sink term that is used to balance the source term $k_f f$. This is a modification proposed by Lien and Kalitzin [26] and ensures that the source term $k_f f \to 0$ as it approaches the wall.

Equation 16 contains no sensitivity to the wall, this is introduced through a modified Helmholtz equation which forms an elliptic relaxation equation. The form of this equation accounts for anisotropy close to walls and is also independent of Reynolds number and y^+ value which reads:

$$L^{2} \frac{\partial^{2} f}{\partial x^{2}} - f = \underbrace{\frac{C_{1}}{T} \left(\frac{\overline{v_{f}^{2}}}{k_{f}} - \frac{2}{3} \right)}_{\phi_{yy,S}} - \underbrace{\frac{C_{2} \Pi_{f}}{k_{f}}}_{\phi_{yy,R}} - \frac{1}{T} \left(6 \frac{\overline{v_{f}^{2}}}{k_{f}} - \frac{2}{3} \right).$$
(17)

The terms $\phi_{yy,S}$ and $\phi_{yy,R}$ are the so-called slow and rapid pressure-strain terms [22, 33] with the final term being used to ensure far field behaviour i.e. that the elliptic relaxation function diminishes away from walls.

In the original formulation of this equation as given by Durbin [8] the boundary condition for f contains the wall distance to the fourth power in its denominator. This lead to computational stiffness and numerical oscillations in the near-wall region. This issue was resolved by Lien and Kalitzin [26] by introducing $6\frac{\overline{v_f}}{k_f}$ as a sink and source in the $k_f f$ source term, the $\overline{v_f}$ transport equation and the elliptical relaxation equation, f. This ensures that f tends to 0 at the wall enabling a Dirichlet boundary condition to be prescribed. The turbulent viscosity is calculated from the solution of the $\overline{v_f^2} - f$ model and a correction is employed to ensure the correct velocity scale is used as the wall is approached. The correction of Davidson et al. [7] is employed and the definition of the turbulent viscosity now reads

$$\nu_{ft} = \min\left\{C_{f\mu}k_f^2/\varepsilon_f, \ C_\mu \overline{v_f^2}T\right\},\tag{18}$$

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¹⁹⁵ where the turbulent time and length scales are defined as

$$T = \max\left(\frac{k_f}{\varepsilon_f}, 6\sqrt{\frac{\nu_f}{\varepsilon_f}}\right),\tag{19}$$

$$L = \max\left(\frac{k_f^{3/2}}{\varepsilon_f}, C_\eta \frac{\nu_f^{3/4}}{\varepsilon_f^{1/4}}\right).$$
(20)

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¹⁹⁷ Both time and length scales are limited in regions close to the wall. This is achieved ¹⁹⁸ by introducing a dependency on Kolmogorov scales which are only active in regions ¹⁹⁹ very close to the wall i.e. $y^+ < 5$. This ensures that a singularity is not introduced ²⁰⁰ into the solution matrix and that the scales collapse at the wall. Another modification ²⁰¹ close to the wall is to modify the "constant" $C_{\varepsilon 1}$ by damping it in the near-wall region ²⁰² by employing the following formulation

$$C_{\varepsilon 1} = 1.4 \left(1 + 0.05 \sqrt{k_f / \overline{v_f^2}} \right).$$
 (21)

To summarise, the $\overline{v_f^2} - f$ model equations can be found in Table 4 with the

turbulence modelling constants, taken from the original model [8], found in Table 8. The complete set of equations that make up the RA-TFM with the $\overline{v_f^2} - f$ model is then the equations found in Table 1 and the aforementioned equations.

Table 4: $\overline{v_f^2} - f$ model equations.

$$\frac{\partial(\alpha_f \rho_f k_f)}{\partial t} + \nabla \cdot (\alpha_f \rho_f k_f \mathbf{u}_f) = \nabla \cdot \left(\mu_f + \frac{\mu_{ft}}{\sigma_{fk}}\right) \nabla k_f + \alpha_f \rho_f \Pi_f - \alpha_f \rho_f \varepsilon_f + 2\beta(k_{fp} - k_f)$$
(22)

$$\frac{\partial(\alpha_f \rho_f \varepsilon_f)}{\partial t} + \nabla \cdot (\alpha_f \rho_f \varepsilon_f \mathbf{u}_f) = \nabla \cdot \left(\mu_f + \frac{\mu_{ft}}{\sigma_{f\epsilon}}\right) \nabla \varepsilon_f + \frac{\varepsilon_f}{k_f} \left[\frac{C_{\varepsilon_1} \alpha_f \rho_f \Pi_f - C_{\varepsilon_2} \alpha_f \rho_f \varepsilon_f}{T}\right] + 2C_3 \beta(\varepsilon_{fp} - \varepsilon_f)$$
(23)

$$\frac{\partial(\alpha_f \rho_f \overline{v_f^2})}{\partial t} + \nabla \cdot (\alpha_f \rho_f \overline{v_f^2} \mathbf{u}_f) = \nabla \cdot \left(\mu_f + \frac{\mu_{ft}}{\sigma_{fk}}\right) \nabla \overline{v_f^2} + \overline{v_f^2}_{source} - \alpha_f \rho_f 6 \frac{\overline{v_f^2}}{k_f} \varepsilon_f + 2\beta(\overline{v_{fp}^2} - \overline{v_f^2})$$
(24)

$$L^{2}\frac{\partial^{2} f}{\partial x^{2}} - f = \frac{C_{1}}{T} \left(\frac{\overline{v_{f}^{2}}}{k_{f}} - \frac{2}{3} \right) - C_{2}\frac{\Pi_{f}}{k_{f}} - \frac{1}{T} \left(6\frac{\overline{v_{f}^{2}}}{k_{f}} - \frac{2}{3} \right)$$
(25)

Table 5: $\overline{v_f^2} - f$ model parameters.

C_{μ}	C_1	C_2	C_L	C_{η}	$C_{\varepsilon 2}$	C_3	β_k	β_{ϵ}	β_v	$C_{f\mu}$	σ_k	$\sigma_{f\epsilon}$
0.22	1.4	0.3	0.23	70	1.9	1	1	1	1	0.09	1	1.3

Wall boundary conditions for ε_f can be found by a Taylor expansion around the no-slip condition at the wall [33] which reads as:

$$\varepsilon_f \to 2\nu_f \frac{k_f}{\mathrm{y}^2}$$
 (26)

For the remaining fluid-phase model variables the following boundary conditions 209 at the wall are prescribed, $\mathbf{u}_f = k_f = \overline{v_f^2} = f = 0$. For the particulate phase 210 a Neumann boundary condition is prescribed for the velocity and all turbulence 211 statistics. For the simulations with the $k_f - \varepsilon_f$ model the standard wall functions for 212 both turbulence statistics are employed. At the inlet the velocity of both the fluid-213 and particle-phase are set at 9.4ms^{-1} . A Neumann boundary condition is used for f 214 together with Dirichlet boundary conditions for all turbulent statistics. At the outlet 215 the a Dirichlet boundary condition for pressure is set whilst a Neumann boundary 216 condition is prescribed for all remaining variables. Both k_p and ε_p are initialised as 217 1/3rd of their fluid counterpart with $\Theta_p = 1.0 \ge 10^{-8} \text{m}^2 \text{s}^{-2}$. 218

The RA-TFM and the recently derived $\overline{v_f^2} - f$ turbulence model is implemented 219 into the open-source toolbox OpenFOAM [53] and is denoted as ratfmFoam [36, 37] 220 which is made available for public use. To handle the pressure-velocity coupling the 221 Pressure Implicit with Splitting Operators (PISO) algorithm [10, 17] is used. The 222 volume fraction is solved using Multi-dimensional Universal Limiter with Explicit 223 Solution (MULES) [56] which is a flux-corrected transport algorithm which ensures 224 robustness, stability and convergence. Time derivative terms are discretised using the 225 first order accurate implicit Euler, gradients are discretised using the Gauss-Green 226 scheme, convective terms are discretised using the first-order upwind scheme Finally, 227 Laplacian schemes are discretised with the second order accurate central differencing 228 scheme. 229

230 2.2. Simulation cases

Table 6: Table of simulated cases

Case	Material	$d_p \; [\mu \mathrm{m}]$	$\rho_p [\mathrm{kg} \mathrm{m}^{-3}]$	Mass loading, ϕ	St
1	glass	50	2500	2%	0.57
2	copper	70	8800	10%	3

The cases used throughout are based on two experiments from Kulick et al. [20] 231 which include separately both glass and copper particles, the details of which can 232 be found in Table 6. For both cases the channel half-width is H = 0.02m with 233 a corresponding length of 5.2m and a wall friction velocity $u_{\tau} = 0.49 \text{ms}^{-1}$. The 234 viscosity of gas is $\nu_f = 15.11 \text{ x } 10^{-5} \text{m}^2 \text{s}^{-1}$ with a density of $\rho_f = 1.2 \text{kg m}^{-3}$ The flow 235 is orientated vertically with a uniform body force of gravity acting in the direction 236 of the flow $(g = 9.8 \text{m s}^{-2})$, this configuration resulted in a centerline velocity of 237 $U_{cl} = 10.5 \text{ms}^{-1}$. The mass loading is defined as $\phi = \frac{\alpha_p \rho_p}{\alpha_f \rho_f}$, and assuming uniform 238 velocity at the inlet. 239

Table 7: Properties of each mesh, f_x , f_y refer to mesh stretching with Mesh 1 [$f_x = 1.1$, $f_y = 1.1$] and Mesh 2 [$f_x = 1.2$, $f_y = 1.2$].

Mesh	$\Delta \mathbf{x}_{min}, \Delta \mathbf{x}_{max}[\mathbf{m}]$	$\Delta y_{min}, \Delta y_{max}[m]$	Mesh size	Comp time
1	$1.2 \times 10^{-3}, 0.02$	$1.2 \times 10^{-5}, 1.2 \times 10^{-3}$	202,761	32 hrs
//	//	//	//	20 hrs
2	$7 \times 10^{-4}, 9 \times 10^{-4}$	$7 \times 10^{-4}, 9 \times 10^{-4}$	66,481	4 hrs

Owing to the different modelling approaches used throughout two different meshes are employed and are detailed in Table 7. Mesh 1 is associated with the $\overline{v_f^2} - f$ model and the low Re Number model of Launder and Sharma [21] and is resolved to y⁺ < 1 ensuring that the resolution of the boundary layer is captured. Mesh 2 is associated with the $k_f - \varepsilon_f$ model and is resolved up to $y^+ > 30$ ensuring that the wall functions can be applied across the correct section of the boundary layer (i.e. log-layer). The final column refers to the computational time spent for a typical run consisting of 30 seconds of real flow time. For ease of reference the $\overline{v_f^2} - f$ formulation will hereafter be referred to as V2F, the low Re number formulation as LE and the $k_f - \varepsilon_f$ formulation as KE.

250 3. Results and Discussion





Figure 1: C1 - Mean fluid velocity profile.

Figure 2: C2 - Mean fluid velocity profile.

Figures 1 & 2 show the mean fluid velocity profiles for each case. It is evident from both plots that the prediction of both V2F & KE models are in good agreement with the experimental data of Kulick et al. [20]. For both C1 & C2 the mean fluid velocity profile remains unchanged, behaviour that is consistent with the experimental observations. Moreover, the experimental uncertainty was reported by the authors to be $\approx 2\%$ and it can be seen that across both profiles the numerical prediction lies well within this range.

This behaviour is not apparent in the predictions from the LE model as there is an underestimation of the fluid velocity. It is interesting to observe that the LE and KE predictions are similar outside the range of $10 < y^+ < 100$ across both plots. Over the transition region i.e. buffer layer to log-layer, the damping function tends

to over-predict the turbulence viscosity. The 'kink' is not reproduced leading to a 263 flattening of the velocity profile but despite this the overall prediction is satisfactory. 264 When comparing both the V2F & KE model predictions there is only a small dis-265 crepancy between each result. This disparity is at its most obvious across the viscous 266 and buffer layer i.e. $y^+ < 20$ in Fig. 1. Owing to the wall function the turbulence 267 statistics are integrated to the wall, with a presumed log-layer relationship, from the 268 first computational cell at $y^+ \approx 30$. This resulted in an over-prediction of turbulence 269 viscosity which is felt as an under-prediction in the mean velocity profile. This trend 270 is seen across the profile for both plots as the KE consistently under-predicts the 271 mean velocity profile in comparison with V2F although this difference is small. 272

273 3.2. Mean particle stream-wise velocity profiles



Figure 3: C1 - Mean particle velocity profile.

Figure 4: C2 - Mean particle velocity profile.

The mean particle velocity profiles are shown in Figs. 3 & 4. Focusing on the former it can be seen that V2F, LE & KE models accurately predict the trend seen

in the experimental observations. The trend is characterised by a flatter profile as 276 the particles approach the wall. As the particle velocities need not be zero at the 277 wall unlike in the fluid-phase, a large slip value exists. The particles deviate from the 278 fluid-phase velocities at around $y^+ < 100$ and maintain their momentum, leading to 279 a flattening of the profile as the wall is approached. 280

In Fig. 3 the profile predicted by the V2F model is in good agreement with 281 the experimental data. This is also true for the KE and LE models up until the 282 near-wall region is approached. Over the range $y^+ < 100$ the KE prediction deviates 283 from the experimental results as the momentum is over-predicted. The contrary 284 is true for the LE model in which the particles remain correlated with the carrier 285 flow up until $y^+ \approx 50$ and then begin to deviate resulting in a under-estimation 286 of the particle velocities. The cause of this behaviour is attributed to the particle 287 fluctuation energy calculation. For the KE model this results is an underestimation 288 of the energy exchange and for the LE model an overestimation, this behaviour will 289 be discussed further in Sec. 3.5. 290

Looking at Fig. 4 it can be seen that there is an over-estimation in the mean 291 particle velocities across all three models. This discrepancy was also predicted in the 292 E-L results of Yamamoto et al. [54] and Wang and Squires [52]. A recent study by 293 Vreman [51] suggests that this global reduction in the particle velocities is due to 294 the so-called "non-uniform feedback force" which is exacerbated by wall roughness. 295 This results in an additional drag force exerted on the particles leading to increased 296 turbulence attenuation. 297

298

This additional force would result in a much flatter profile as shown in Vreman

²⁹⁹ [51] and lead to results that closely align with the experimental data in Fig. 4. As ³⁰⁰ wall roughness has not been modelled in this study, and similar results have been ³⁰¹ reported by other researchers using higher resolution methods i.e. E-L [54, 52], it is ³⁰² plausible to conclude that this is the source of the overestimation. It is instructive ³⁰³ to note that despite this, the qualitative behaviour of the profile is captured by the ³⁰⁴ numerical models resulting in a comparable trend across the profile.

305 3.3. Fluid stream-wise turbulence intensity



Figure 5: C1 - Fluid stream-wise turbulence intensity profile.

Figure 6: C2 - Fluid stream-wise turbulence intensity profile.

Figures 5 & 6 show the fluid-phase turbulence intensity for each case. When comparing the V2F & KE model across both cases it is apparent that there is a clear difference between the two. The V2F model is capable of predicting a strong peak at $y^+ \approx 20$ and then dissipating off into the core of the channel. This is, of course, not seen in the KE model result as the first computational cell is placed at $y^+ > 30$. This then omits the presence of the peak and results in a near constant value of $u_{f_{rms}}$ as the wall is approached. In the core of the flow, over the region (y⁺ > 70), there is better agreement with the experimental data as the transport terms begin to dominate.

The LE model performs well in comparison with the experimental data. Resolving 315 the near wall region enables the peak to be predicted although it is not as pronounced 316 or concentrated as the peak predicted by the V2F model. There is a "spreading" of 317 the turbulent kinetic energy across the range, $10 < y^+ < 100$, smearing the transition 318 region. This is suspected to be the cause of the degradation of the mean velocity 319 profile in Figs. 1 & 2. This could be improved through tweaking of the damping 320 function but as highlighted in the introduction this is an inherit shortcoming of the 321 approach and there are several different ad-hoc solutions to dampening the viscosity 322 but the underlying issues remain. 323

The KE models dependency on the wall function results in a deterioration of the 324 turbulence intensity prediction. This will be shown to have important consequences 325 when predicting the particle fluctuation energy behaviour. The V2F model shows 326 excellent agreement across both plots with the under prediction being confined to the 327 turbulence peak and dissipation towards the wall. It has been suggested [9, 8] that 328 the $\overline{v_f^2} - f$ model performs best at high Re number. In this work a relatively low Re 329 number of 14,000 is simulated which could be the cause of the under-prediction. This 330 could be improved with a manipulation of the turbulence constant i.e. C_2 although 331 this remains out of the scope of this study. 332

Kulick et al. [20] reports turbulence attenuation in C2. As discussed in Sec. 3.2

this is due to the lack of wall roughness modelled in this work. Across both Figs. the behaviour is similar with the velocity covariance terms contributing little to the prediction. This finding is also consistent with those of Yamamoto et al. [54], Wang and Squires [52] in which negligible attenuation was reported.

338 3.4. Fluid wall-normal turbulence intensity





Figure 7: C1 - Fluid wall-normal turbulence intensity profile.

Figure 8: C2 - Fluid wall-normal turbulence intensity profile.

Figures 7 & 8 show the fluctuating wall-normal component. This component is explicitly modelled in the V2F model as $\overline{v_f^2}$ and is crucial in enabling the resolution of the boundary layer. As it can be seen from Fig. 7 the distribution is in good agreement with the experimental predictions. The V2F model shows the correct dampening of the wall-normal component through the elliptic relaxation equation and enables a strong turbulence production peak as seen in Sec. 3.3. For C2 the wall-normal intensity was also attenuated in the same way the stream-wise intensity ³⁴⁶ was. As previously discussed no attenuation was reported in these results.



347 3.5. Particle fluctuation energy

Figure 9: C1 - Particle fluctuation energy profile.



Figure 10: C2 - Particle fluctuation energy profile.

In the RA-TFM we explicitly account for two contributions to the particle fluctuation energy [12], $\kappa_p = 3\Theta_p + k_p$ where Θ_p represents the small-scale kinetic collisional energy i.e. uncorrelated energy and k_p represents the large-scale turbulent kinetic energy i.e. correlated energy. Broadly speaking Θ_p is relevant at high *St* number and high mass loading, and k_p is relevant at low *St* number and low mass loading. This distinction has already proven crucial in the literature [12, 18, 38, 39, 50].

Figures 9 & 10 show the particle fluctuation energy for each case. As is evident from both plots the V2F model outperforms the KE model. This is a direct consequence of the poor prediction in the fluid turbulent kinetic energy. Owing to the relatively low *St* number in the core of the flow the particles are tightly correlated therefore they are governed by the velocity covariance term which arises due to coupling through drag. The fluctuation energy distribution is dominated by k_p up until the near-wall region is approached - this is confirmed by comparing the distribution with that of Fig. 5. For C2 this is not strictly true as the *St* number is larger in the core of the flow resulting in a contribution acting across the half-width of the channel, this can be seen by comparing the two figures.

In the near-wall region the St number increases dramatically. This ensures that the particles become uncorrelated with the main carrier flow and Θ_p is produced in the region y⁺ < 10. Additionally, an energy cascade exists in which the largescale particle turbulent kinetic energy dissipation, ε_p appears in the Θ_p transport equation through a source term. The particle turbulence kinetic energy dissipation is then highest in the near-wall region thus contributing to the loss of correlation with the carrier flow.

The LE model overestimates the fluctuation energy in C1 and the profile begins 371 to flatten out as the wall is approach. As shown in $\S3.2$, the particle velocity profile 372 was under predicted which is in line with the behaviour of the particle fluctuation 373 energy. This was caused by an overproduction of Θ_p as the energy transfer was 374 overproduced, this resulted in an excessively large value of Θ_p in the near-wall region. 375 It is not obvious why this occurred as the velocity profile predicted for C2 is in good 376 agreement with the experimental data. It can be speculated that the source of the 377 error is the velocity covariance term as we set the correlation factor to 1 or the 378 kinetic theory constitutive equations as we have employed standard expressions from 379 the literature, although this is far from conclusive. 380





Figure 11: C1 - Volume fraction distribution normalised by mean values. The E-L results of Yamamoto et al. [54] displaying the normalised particle number density function.

Figure 12: C2 - Volume fraction distribution normalised by mean values. The E-L results of Yamamoto et al. [54] displaying the normalised particle number density function.

Figures 11 & 12 show the volume fraction distribution for both cases. Additionally, the E-L results of Yamamoto et al. [54] have been displayed for qualitative understanding. Across both plots the predictions found herein are at odds with the E-L results. Some similarities can be drawn e.g. an accumulation in the near-wall region in C2 but in general it is difficult to draw any conclusions from the data. The volume fraction is a difficult statistic to predict in E-L and E-E simulations so this result is not unexpected.

It is clear from both plots that the V2F model predicts an accumulation of particles in the near-wall. The particles tend to drift across the channel width and reside in the near-wall region - characteristic behaviour of turbophoresis. As the force is determined by the fluctuating wall normal component, of which is explicitly modelled in the V2F model and coupled to the particle-phase correlated energy, the particles
are able to drift down the gradients of turbulent kinetic energy.

The KE model predictions reveal a slightly different picture. In C1 an accumula-395 tion of particles in the near-wall region is seen but the sharp peak is not replicated, 396 instead these particles are found in the main core of the flow. The prediction for C2 397 reveals a breakdown in the volume fraction distribution in comparison to the V2F 398 model. The particles are nearly uniformly distributed with a higher concentration 399 in the main core of the flow. Due to the higher St number in C2 the particles are 400 less correlated with the carrier flow, therefore in order to migrate towards the wall a 401 larger dispersion is required. As the wall-normal component has not been explicitly 402 modelled the particles can not overcome the turbulent kinetic energy gradient and 403 remain in the main core of flow. Moreover, this can be a symptom of the mesh 404 resolution as the wall function constraint ensures the near-wall region can not be 405 resolved. The LE model results show no accumulation for C1 but do so for C2. The 406 second result corroborates the findings from the V2F model in that the resolution of 407 the boundary layer can lead to an accumulation of particles in the boundary layer. 408

409 4. Conclusions

This work has proposed a generic approach for accounting for near-wall induced non-homogeneity in Eulerian-Eulerian simulations. An E-E elliptic relaxation model, namely the $\overline{v_f^2} - f$ model, has been derived with in a Reynolds-Averaged Two-Fluid model framework and applied to a downward-facing vertical channel. Predictions are validated against the benchmark experimental data of Kulick et al. [20] and compared against the conventional $k_f - \varepsilon_f$ turbulence model. From this work the following conclusions can be drawn:



420 2. The new modelling has been validated against benchmark experimental data 421 with differing mass loading and Stokes number as well as being corroborated 422 with Euler-Lagrange results;

3. The elliptic relaxation model has shown a high level of validation, in line with
those from Euler-Lagrange, offering a viable way of achieving accurate results
at a lower computational cost;

- 426 4. The use of single-phase wall functions in E-E simulations can result in an
 427 under-prediction of the velocity covariance coupling term which impedes on
 428 the particle fluctuation energy prediction. This is expected to be exacerbated
 429 with increasing mass loading;
- 430

5. The elliptic relaxation model enabled the migration of particles towards the

⁴³¹ near-wall region, a result that was not replicated using the conventional tur-⁴³² bulence model;

6. The approach presented herein offers a novel way of accounting for the near-wall
region in E-E simulations.

435 5. Code repository

The source code of the ratfmFoam solver and the supplementary data used in this work can be downloaded from [36].

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443 7. Appendix

We begin at the fluid-phase momentum equation derived from a collisional Boltzmann equation like the one presented in Fox [13]. Here we couple the phases through drag and include a body force due to gravity. This results in the equation,

$$\frac{\partial(\alpha_f \mathbf{u}_f)}{\partial t} + \nabla \cdot (\alpha_f \mathbf{u}_f \otimes \mathbf{u}_f + \alpha_f \overline{\mathbf{P}}_f) = \frac{\alpha_p \rho_p}{\rho_f} \mathcal{A} + \alpha_f \mathbf{g}$$
(27)

⁴⁴⁷ Taking the Reynolds-Average (RA) of Eq. 27 gives:

$$\frac{\partial \langle \alpha_f \rangle \langle \mathbf{u}_f \rangle_f}{\partial t} + \nabla \cdot \left(\langle \alpha_f \rangle \langle \mathbf{u}_f \rangle_f \otimes \langle \mathbf{u}_f \rangle_f + \langle \alpha_f \rangle \langle \mathbf{u}_f' \otimes \mathbf{u}_f' \rangle_f + \langle \alpha_f \rangle \langle \overline{\mathbf{P}}_f \rangle \right) = \frac{\langle \alpha_p \rangle \rho_p}{\rho_f} \langle \mathcal{A} \rangle_f + \langle \alpha_f \rangle \mathbf{g}$$
(28)

where $\langle \alpha_f \rangle$ represents the RA fluid-phase volume fraction and $\langle \mathbf{u}_f \rangle_f = \langle \alpha_f \mathbf{u}_f \rangle / \langle \alpha_f \rangle$ is the Phase-Averaged (PA) fluid-phase velocity. Now grouping the stress terms as, $\langle \overline{\mathcal{P}}_f \rangle_f = \langle \overline{\mathbf{P}}_f \rangle_f + \langle \mathbf{u}_f'' \otimes \mathbf{u}_f'' \rangle_f$ and multiplying through by the PA fluid-phase velocity one arrives at:

$$\frac{\partial \langle \alpha_f \rangle \langle \mathbf{u}_f \rangle_f \otimes \langle \mathbf{u}_f \rangle_f}{\partial t} + \nabla \cdot \left(\langle \alpha_f \rangle \langle \mathbf{u}_f \rangle_f \otimes \langle \mathbf{u}_f \rangle_f \otimes \langle \mathbf{u}_f \rangle_f + \langle \alpha_f \rangle \langle \mathbf{u}_f \rangle_f \otimes \langle \overline{\mathcal{P}}_f \rangle_f \right)
= \frac{\langle \alpha_p \rangle \rho_p}{\rho_f} \langle \mathbf{u}_f \rangle_f \otimes \langle \mathcal{A} \rangle_f + \langle \alpha_f \rangle \langle \mathbf{u}_f \rangle_f \otimes \mathbf{g}$$
(29)

As we want to derive an equation for the Reynolds stress tensor we now find the transport equation for the fluid-phase velocity tensor product. Note this is prior to Reynolds-Averaging. Beginning at Eq. 27 we can multiply through by the fluid-phase velocity, which reads as:

$$\frac{\partial(\alpha_f \mathbf{u}_f \otimes \mathbf{u}_f)}{\partial t} + \nabla \cdot (\alpha_f \mathbf{u}_f \otimes \mathbf{u}_f \otimes \mathbf{u}_f + \alpha_f \mathbf{u}_f \otimes \overline{\mathbf{P}}_f) = \frac{\alpha_p \rho_p}{\rho_f} \mathbf{u}_f \otimes \mathcal{A} + \alpha_f \mathbf{u}_f \otimes \mathbf{g}$$
(30)

⁴⁵⁶ Now invoking the relation for the fluid-phase pressure-stress tensor,

$$\overline{\mathbf{P}}_{f} = \frac{1}{\rho_{f}\alpha_{f}}(p_{f}\mathbf{I} - \overline{\boldsymbol{\sigma}}_{f}), \qquad \langle \overline{\mathbf{P}}_{f} \rangle = \frac{1}{\rho_{f}\alpha_{f}}(\langle p_{f} \rangle \mathbf{I} - \langle \overline{\boldsymbol{\sigma}}_{f} \rangle)$$
(31)

457 and the momentum coupling,

$$\mathcal{A} = \frac{1}{\tau_p} (\mathbf{u}_p - \mathbf{u}_f), \qquad \langle \mathcal{A} \rangle_f = \frac{1}{\tau_p} (\langle \mathbf{u}_p \rangle_f - \langle \mathbf{u}_f \rangle_f)$$
(32)

and then subtracting Eq. 29 from the RA of Eq. 30, the transport equation for the fluid-phase Reynolds-Stress can be written as

$$\frac{\partial \langle \alpha_{f} \rangle \langle \mathbf{u}_{f}^{\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{f}}{\partial t} + \nabla \cdot \langle \alpha_{f} \rangle \langle \mathbf{u}_{f} \rangle_{f} \otimes \langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f} = -\nabla \cdot \langle \alpha_{f} \rangle \langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f} \\
- \underbrace{\langle \alpha_{f} \rangle (\langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{f} \cdot \nabla \langle \mathbf{u}_{f} \rangle_{f})}_{\text{Production}} + \frac{1}{\rho_{f}} \nabla \cdot \langle \overline{\boldsymbol{\sigma}}_{f} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle - \frac{1}{\rho_{f}} \nabla \langle p_{f} \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle}{\frac{1}{\rho_{f}} \langle \overline{\boldsymbol{\sigma}}_{f} \cdot \nabla \mathbf{u}_{f}^{\prime\prime\prime} \rangle} + \langle \alpha_{f} \rangle \beta (\underbrace{\langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime} \rangle_{p} - \langle \mathbf{u}_{f}^{\prime\prime\prime\prime} \otimes \mathbf{u}_{f}^{\prime\prime\prime\prime} \rangle_{p}}_{\text{velocity correlations}} \tag{33}$$

where the fluid-phase velocity fluctuations are defined as $\mathbf{u}_f'' = \mathbf{u}_f - \langle \mathbf{u}_f \rangle_f$.

461 Nomenclature

U_{cl}	centreline velocity, $[ms^{-1}]$
C_D	drag coefficient, $[-]$
g	gravity, $[ms^{-2}]$
n	unit vector normal to the wall, $[-]$
Re_p	particle Reynolds number, $[-]$
d_p	particle diameter, [m]
\mathbf{u}_i	velocity, $[ms^{-1}]$
\mathbf{u}_p''	particle velocity fluctuation w.r.t PA velocity, $[ms^{-1}]$
$\mathbf{u}_{f}^{'''}$	fluid velocity fluctuation w.r.t PA velocity, $[ms^{-1}]$
p_i	pressure, [Pa]
g_0	radial distribution coefficient, $[-]$
t	time, [s]
k_i	turbulent kinetic energy, $[m^2s^{-2}]$

462 Greek letters

$lpha_i$	volume fraction, $[-]$
$\alpha_{p,max}$	maximum particle volume fraction, $[-]$
eta	momentum exchange coefficient, $[kgm^{-3}s^{-1}]$
$\Delta \mathbf{x}$	length of the cell in the x direction, [m]
Δy	length of the cell in the y direction, [m]
ε_i	turbulent kinetic energy dissipation, $[m^2s^{-3}]$
Θ_p	granular temperature, $[m^2s^{-2}]$
κ_p	particle fluctuation energy, $[m^2s^{-2}]$
$\kappa_{\Theta s}$	diffusion coefficient for granular energy, $[kgm^{-1}s^{-1}]$
μ_i	shear viscosity, $[kgm^{-1}s^{-1}]$
$\mu_{i,t}$	turbulent shear viscosity, $[kgm^{-1}s^{-1}]$
$ u_i$	kinematic viscosity, $[m^2s^{-1}]$
$ u_{i,t}$	turbulent kinematic viscosity, $[m^2 s^{-1}]$
$ ho_i$	density, $[kgm^{-3}]$
$\overline{oldsymbol{\sigma}}_{f}$	fluid phase stress tensor, $[kgm^{-1}s^{-2}]$
$\overline{oldsymbol{\sigma}}_p$	particle phase stress tensor, $[kgm^{-1}s^{-2}]$
$ au_p$	particle relaxation time, [s]
$ au_{f}$	characteristic flow time, [s]
•	

463 Subscripts

- f fluid
- i general index
- p particle
- x x direction
- y y direction
- yy wall normal component
- z z direction

464 Superscripts

PA particle velocity fluctuationPA fluid velocity fluctuation

465 Special notation

 $\begin{array}{ll} \langle \cdot \rangle & & \mbox{Reynolds averaging operator} \\ \langle \cdot \rangle_i & & \mbox{phase averaging operator associated with phase i} \end{array}$

Table 8: Model characteristics & turbulence variables.

$$\begin{split} \beta &= \frac{\rho_p \alpha_p}{\tau_d} = \frac{3}{4} \frac{\alpha_p \alpha_f \rho_f \mathbf{u}_r}{d_p} C_d \\ C_d &= \begin{cases} \frac{24}{Re_p} \Big[1 + 0.15 Re_p^{0.287} \Big] & \text{if } Re_p < 1000 \\ 0.44 & \text{if } Re_p \ge 1000 \end{cases} \\ \kappa_p &= k_p + 3/2\Theta_p \\ u_{prms} &= \sqrt{(2/3)\kappa_p} \\ u_{frms} &= \sqrt{(2/3)\kappa_p} \\ u_{frms} &= \sqrt{(2/3)k_f} \\ \tau_p &= \frac{\rho_p d_p^2}{18\rho_f \nu_f} \\ \tau_f &= \frac{k_f}{\varepsilon_f} \\ St &= \tau_d / \tau_f \\ e &= 0.9 \\ \Pi_p &= 2\nu_{pt} \overline{\mathbf{S}}_{\mathbf{p}} : \overline{\mathbf{S}}_{\mathbf{p}} + \frac{2}{3}k_p \nabla \cdot \mathbf{u}_p \\ \Pi_f &= 2\nu_{ft} \overline{\mathbf{S}}_{\mathbf{f}} : \overline{\mathbf{S}}_{\mathbf{f}} + \frac{2}{3}k_f \nabla \cdot \mathbf{u}_f \end{split}$$

$$\begin{split} \kappa_{p} &= k_{p} + 1.5\Theta_{p} \\ \mu_{f} &= \rho_{f}\nu_{f} \\ \mu_{ft} &= \alpha_{f}\rho_{f}\nu_{ft} = \alpha_{f}\rho_{f}C_{f\mu}\frac{k_{f}^{2}}{\varepsilon_{f}} \\ \mu_{p} &= \alpha_{p}\rho_{p}\nu_{p} = \frac{2\mu_{p_{dil}}}{(1+e)g_{0}} \left[1 + \frac{4}{5}(1+e)g_{0}\alpha_{p}\right]^{2} + \frac{4}{5}\alpha_{p}^{2}\rho_{p}d_{p}g_{0}(1+e)\left(\frac{\Theta_{p}}{\pi}\right)^{1/2} \\ \mu_{p_{dil}} &= \frac{5\sqrt{\pi}}{96}\rho_{p}d_{p}\Theta_{p}^{1/2} \\ \mu_{pt} &= \alpha_{p}\rho_{p}\nu_{pt} = \alpha_{p}\rho_{p}C_{p\mu}\frac{k_{p}^{2}}{\varepsilon_{p}} \\ p_{p} &= \rho_{p}\alpha_{p}\Theta_{p} + 2(1+e)\rho_{p}\alpha_{p}^{2}g_{0}\Theta_{p} \\ \gamma &= \frac{12(1-e^{2})g_{o}}{\sqrt{\pi}d_{p}}\alpha_{p}^{2}\rho_{p}\Theta_{p}^{3/2} \\ \kappa_{\Theta} &= \frac{2}{(1+e)g_{0}} \left[1 + \frac{6}{5}(1+e)g_{0}\alpha_{p}\right]^{2}\kappa_{\Theta,dil} + 2\alpha_{p}^{2}\rho_{p}d_{p}g_{0}(1+e)\left(\frac{\Theta_{p}}{\pi}\right)^{\frac{1}{2}} \\ \kappa_{\Theta,dil} &= \frac{75}{384}\sqrt{\pi}\rho_{p}d_{p}\Theta_{p}^{1/2} \\ g_{0} &= \left[1 - \left(\frac{\alpha_{p}}{\alpha_{p,max}}\right)^{\frac{1}{3}}\right]^{-1} \\ \overline{\mathbf{S}}_{\mathbf{p}} &= \frac{1}{2}[\nabla\mathbf{u}_{p} + (\nabla\mathbf{u}_{p})^{T}] - \frac{1}{3}\nabla\cdot\mathbf{u}_{p}\mathbf{I} \\ \overline{\mathbf{S}}_{\mathbf{f}} &= \frac{1}{2}[\nabla\mathbf{u}_{f} + (\nabla\mathbf{u}_{f})^{T}] - \frac{1}{3}\nabla\cdot\mathbf{u}_{f}\mathbf{I} \\ k_{fp} &= \beta_{k}\sqrt{k_{f}k_{p}} \\ \varepsilon_{fp} &= \beta_{\varepsilon}\sqrt{\varepsilon_{f}\varepsilon_{p}} \end{split}$$

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