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# **Design of Electric Machine Taking Into Account the Partial Discharges Phenomena for Future Hybrid Propelled Aircrafts**

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### **Abstract**

*Designing the electric motors to propel an aircraft represents a huge challenge. The constraint of weight requires such machines to be compact. Besides, the high specific power necessary (a few kW/kg, machine cooling included) leads to an increase of the voltage range of stator winding. Such characteristics increase considerably the risk of partial discharges (PD) in the stator winding. In this paper, the design of electric motors aimed to propel a future hybrid aircraft will be presented. It will be realized using 2 tools*.

## **Introduction**

The transition toward more electric aircraft started some years ago. In the 50s, more power has been required so that inboard voltage has been increased from 28V DC to 115/200 V AC at 400Hz frequency [1]. Until the 80s, flight controls used only hydraulic power. It is no more the case since the concept of "Fly by wire" [2] which introduced electric flight controls. Nowadays, in a vocation to become more ecofriendly and more economic, the aircraft industry is looking toward the "full electric aircraft".

Airbus A380 is equipped with the first electric actuator for reverse propulsion ETRAS, which was completely new back in 2005 [2]. Europe is one of the leader of such transition by continuing the Clean Sky program. In order to propel future hybrid aircrafts, 2 targets are aimed. In 2025, the program announces a 5 kW/kg electric motor and a 10 kW/kg in 2035 [3]. To increase specific power of electric machine, the rotational speed and the number of pole pairs (consequently electric frequency) should be increased. Therefore, it induces an increase of the voltage level. The on-board DC voltage bus is going to overpass 1kV. This increases the risk of PD in the machine fed by inverters (cf. Fig. 1).

This works represents a first collaboration between two studies: sizing machine and partial discharges phenomena. As a first level approach, some hypotheses are formulated. In that way, a star winding configuration with the neutral connected to the ground will be presented. This represents the worst configuration for the electrical stress between the conductors and the slot. In case of overshoot, only the first conductor will be affected. The winding class is chosen to 220°C and the winding temperature of the sizing point is 180°C. This temperature value does not influence the Paschen's criteria in the case of electric breakdown in air gaps of length in range of 10-200µm [4]. The slot is modelled with air in order to take into account the imperfection of the impregnation process [5]. By neglecting the winding capacitance effect and the voltage displacement between conductors, the conductor voltage varies linearly.

The first part introduces the machine sizing tool called 'Target Setting Tool (TST)' with an illustrative example of a high specific power electric machine. The second part focuses on taking into account the Partial Discharges (PD) phenomena, in the illustrative example, thanks to the second tool.



Fig. 1: Propulsion Chain of a Hybrid Aircraft [3]

### **I. Machine sizing tool: Target Setting Tool (TST)**

To reach each level of specific power of electric machine of hybrid aircraft, the machine sizing tool called "Target Setting Tool (TST)" is developed. It uses the loadability concepts [6] applied on the analytical model of ideal sinewave synchronous motor [7]. The loadability concepts characterize the performances and the technological level allowing to make tradeoffs on high specific power of electric machine [6]. The loadability concepts are mainly the electric, magnetic and thermal loads [6]:

- *the magnetic loading Bm*, indicates the magnetic flux density in airgap allowed by the soft and hard magnetic materials,
- *the electric loading Km*, indicates the surface current density in airgap allowed by the insulation materials and by the cooling technology,
- *the electromagnetic loading σt*, indicated the tangential stress in airgap allowed by the soft, hard magnetic materials, by the insulation material and by the cooling technologies.

$$
\sigma_t = \frac{B_m K_m}{2} \tag{1}
$$

The maximum electromagnetic torque (*Tem*) is given by:

$$
T_{em} = 2\sigma_t \pi R^2 L \tag{2}
$$

These loads are introduced as inputs to give main sizes, weight and performances of electric motor without to specify the stator winding configuration and the rotor structure. Indeed, this tool also needs very few data which presents a main interest for quickly sizing of several electric motor types. These inputs data are:

- the mechanical power *Pm*, the rotational speed *N*, the number of pole pairs *p*,
- the geometrical choices such as the form factor  $\lambda$ , the winding head coefficient k*wh* and fill factor *kfill* in the slot.

In TST, the airgap flux density and the surface current density are assumed sinewaves. Therefore, more the sized motor is close to these assumptions more their performances are achieved [8]. The magnetic saturation and magnetic reaction of armature are not taking into account. This tool sizes electric motor at the maximum torque operating point. The main sizes of electric motor are: the external stator radius *Rout*, the inner stator radius *R*, the axial active length *L*, the stator yoke height *hy* and the slot height *hs*, as shown in Fig. 2.

The inner stator radius *R* and the axial active length *L* are given by the tangential stress in the airgap σt and the torque with a choice of the form factor  $\lambda$  [2]. The stator yoke height  $h<sub>v</sub>$  is given from the permitted airgap flux density  $B_m$  and yoke flux density  $B_v$ . The slot height is given by the permitted loads namely the surface current density *Arms*, current density *jrms*, airgap flux density  $B_m$  and tooth flux density  $B_t$  and also by fill factor *kfill* [8]. Therefore, the weight of electric motor can be deduced and then the specific power and torque densities.



Fig. 2: Main sizes of electric motor

#### **Sizing of a high specific power electric motor**

A high specific power electric motor is sized using TST and presented in [9]. Their mechanical specifications, the electric, the magnetic and the thermal loads fixed in order to obtain the main sizes, weight and the specific power and torque densities, are summarized in Table 1. A joint design of Halbach array PM and distributed stator winding are made in [9] to validate the performances of high specific power density of electric motor.

The three phase distributed winding is made in [9] with two slots per pole and per phase. The full pitch concentric winding configuration is shown in Fig. 3. To study the PDs phenomena in this winding configuration, a choice of 6 conductors per slot is made (Table 2). Conductor voltage is given:

$$
V_{condi} = \frac{L_i}{L_{tot}} V_{\text{max}} \tag{3}
$$

Where *Vcondi* is the voltage of *i th* conductor computed from the ground, *Li* is the length of conductor computed from the ground, *Ltot* is the total conductor length and *Vmax* is the amplitude of voltage per phase. Table 2 presents the voltages of each conductor. These voltages are calculated at the half axial active length of the stator (cf. Fig.3).

Parameters	Symbol	Unit	Value
<b>Mechanical specifications</b>			
Mechanical power	$P_m$	kW	900
Mechanical torque	Τ	N.m	430
Electric, Magnetic and thermal loads			
Rms current density	Irms	A.mm	11
Max airgap flux density	$B_m$	т	1
Max linear current density	$\mathsf{K}_m$	kA/m	76
Tooth flux density	$B_t$	Т	1.3
Stator yoke flux density	$B_{sy}$	т	1
<b>Geometric choices</b>			
Number of poles	2p	-	4
Fill factor	$k_{\text{fill}}$		0.5
Head winding coefficient	$k_{wh}$		1.4
Form factor	λ		0.5
<b>Main sizes</b>			
Stator outer radius	$R_{\mathit{out}}$	mm	136.2
Stator inner radius	R	mm	76.6
Active length machine	L	mm	306.5
Slot height	h <sub>s</sub>	mm	21.28
Weight, specific power and torque densities			
Weight of motor	$M_{mot}$	kg	152.5
Specific power density	$P_m/M_{mot}$	$kW.kg^{-1}$	5.9
Specific torque density	$T/M_{mot}$	Nm.kg <sup>-T</sup>	2.81

Table 1: Electric machine parameters [9]



Table 2: Conductor voltage



Fig. 3: *Stator winding configuration: concentric winding with 6 conductors per slot.* 

## **II. Tool to detect Partial Discharges (PDs) in stator winding**

#### *A. Paschen's theory*

In the literature, the Paschen's theory [10] remains the main criteria used to predict PDs in numerical models. This theory requires some strong hypothesis. One considers an air gap between 2 parallel metallic electrodes. A voltage drop is applied between these electrodes so that a uniform electric field is formed. The temperature and humidity conditions are the normal ones:  $T=20^{\circ}$ C and  $H_a=11g/m^3$  [4]. An electron initially present in the air gap will be accelerated by the applied electric field. In its way to reach one of the electrodes (the anode) it will collide with some neutral particles present in the gap. These interactions result in more electrons present in the air gap. This electronic avalanche is supposed to be the only ionization mechanism in the air gap. The ions, generated by the collision with electrons, will travel toward the opposite electrode (the cathode). They will bomb it and may release others electrons. If this secondary emission releases an electron that replaces the initial electron which started the electronic avalanche, the ionization in the air gap is self-maintained.

In this case, the breakdown voltage of the air gap is linked to the pressure and the length of the gap between the electrodes by the following expression:

$$
V_B = \frac{Bpd}{C + \ln(pd)}
$$
(4)

$$
C = \ln\left(\frac{A}{\ln(1 + \frac{I}{\gamma})}\right)
$$
 (5)

With *p* the pressure in the gap, *d* the length of the gap between the electrodes, A=11.40 Pa<sup>-1</sup>.m<sup>-1</sup> and *B*=277.40 V. Pa<sup>-1</sup>.m<sup>-1</sup> [4]. The coefficient  $\gamma$  is the secondary electron emission coefficient. It is the number of electrons leaving the cathode over the number of impacting ions. It equals 0.01 [5] in the case of plane metallic electrodes. This coefficient is highly dependents of the geometry of the electrodes, the uniformity of the electric field and the composition of the gas in the gap.

## *B. Partial Discharges (PDs) study of high specific power electric motor*

The TST tool gives the information on the winding so that the electromagnetic torque and specific power constrains are reached (cf. part 1). The winding is made of 6 conductors. Each conductor must have a copper cross section of 16.9mm² in order to be compliant with the current density constraint. The filling factor is 0.5. This factor is linked to the air section and insulation section (enamel & slot insulation) by the following equation:

$$
k_{\text{full}} = \frac{S_{\text{cu}}}{S_{\text{shot}}} = \frac{S_{\text{cu}}}{S_{\text{cu}} + S_{\text{e}} + S_{\text{my}} + S_{\text{air}}}
$$
(6)

With *SCu* the total copper section, *SSlot* the slot section, *Se* the total enamel section, *Smy* the total slot insulation section (made of Mylar [11]) and *Sair* the air section. In this example, the slot geometry (i. e. S<sub>Slot</sub>) and the copper cross section (*SCu*) are fixed. The wires are covered in PI enamel of *e* equals min grade 2 thickness [12] (Figure 4). The slot insulation is made of Mylar of *emy* thickness. The electric relative permittivity of these polymers is taken equal to  $\varepsilon_r = 3.5$ .

The voltage repartition over the turns is also given by TST. In the following, *Vi* refers to the voltage over the *i th* copper wire. The choice is made to use 2 wires in parallel per conductor. To take into account the rising time of the voltage due to the power converter, an overshoot of 33% of the voltage is taken into account on the first turn.

$$
V_1=1.33\sqrt{*}V_{cond1}
$$



Table 3: Initialization parameters



Fig. 4: (a) Slot dimensions (b) enamel covered wire

The winding is organized so that the electric constrain between conductors and conductors/slot is reduced (Fig. 5). In this configuration, the electrical stress between conductors and the slot (left side and end slot) are similar. Hence, only the conductors/end slot interfaces will be studied as conductor/slot interfaces. The slot and winding parameters are entered in MatLab (Fig. 6). An Ansys 2D model of the slot with the winding is then automatically built, meshed and solved. The slot is modelled filled with air at pressure 1 bar so that the simulated results are close to the ones obtained experimentally with an epoxy filled slot [5]. The voltage values and position of nodes along the wires perimeters and slot contour are sent back to MatLab for post processing. The *Vsimu*(p*d=d*) curved

obtained are compared to the experimental Paschen curve obtained by Dakin and *al* [13]. If there is some PD risk, the tool updates the winding configuration and the problem is solved again.

At the end of the post processing stage on MatLab, the tool returns the results presented on Fig. 7. One can notice the PD risk zones. These are the zones where the *Vsimu*(p*d=d*) curves are above the Paschen curve.





Fig. 5: (a) Initialization - Voltage map [V], (b) Electric field norm [V/m]





 $10^{-1}$ 



The tool then updates the enamel thickness to *e* equals max grade 3 [12].



Table 4: Step 0 to Step 1 Parameters

Considering the following system equation:

$$
\begin{cases}\nS_{e,step1} + S_{my,step1} + S_{air,step1} + S_{cu} = S_{slot} \\
S_{e,step1} > S_{e,step0} \\
S_{my,step1} = S_{my,step0} \\
S_{air,step1} < S_{air,step0}\n\end{cases} \tag{7}
$$



Fig. 8: Step 1 - *Vsimu*(p*d=d*) curves vs experimental Paschen curve (*e* max grade 3 [12])

At the end of step 1, there may be no more PD risk between conductors. However, PD risk remains for conductor/slot interfaces. In step 2, *e* is kept equals to max grade 3 and the Mylar thickness is increased to its maximum thickness allowed by the slot section (*emy*=480µm [11]).



Considering the following system equation:

$$
\begin{cases}\nS_{e, step 2} + S_{my, step 2} + S_{air, step 2} + S_{cu} = S_{slot} \\
S_{e, step 2} = S_{e, step 1} \\
S_{my, step 2} > S_{my, step 1} \\
S_{air, step 2} < S_{air, step 1}\n\end{cases} \tag{8}
$$

Where *Smy* is increasing while *Sair* is decreasing. *Scu*, *Se* and *SSlot* remain constant.



Fig. 9: *Step 2 - Vsimu(pd=d) curves vs experimental Paschen curve (emy=480µm [11])*

After 2 iterations, the tool probably suppresses the risk of PD in conductor/conductor interfaces and conductor/slot interfaces (under the hypothesis of a 33% overshoot on *Vcond1*). However, overshoot due to the inverter can reach twice the nominal voltage (i. e 100% overshoot on *Vcond1*) [5]. A simulation has been done in this case (Figs. 10 and 11).

PD will probably occur between conductors. Hence the total slot surface is used, the filling factor has to be decrease. This case illustrates the interest of the





Fig. 10: Voltage map with  $V_1=2V_{cond1}$ ,  $e_{mv}=480 \mu m$  and *e=85.5µm*<br>**Vsimu(d)** conductors/conductors interfaces



Fig. 11: *Vsimu(pd=d)* conductor/conductor interfaces for overshoot 100% (ie. V<sub>1</sub>=2V<sub>cond1</sub>)

#### *C. Discussion and perspectives*

The problem is that in a stator slot, the interfaces wire/wire and wire/slot is not metallic plane/plane electrodes. Indeed, the insulation coating creates a dielectric barrier discharge (DBD) configuration. Moreover, considering round wires, the geometry of the wire/wire and wire/slot interfaces are respectively cylinder/cylinder and cylinder/plane. Due to these facts, one cannot consider a uniform electric field on these interfaces. The secondary electron emission coefficient *γ* also changes due to the geometry and the polymer coating. In the literature, there is no way to determinate this coefficient by an analytical formulation due to the complexity of the mechanisms introduced by the polymer coating.

The author [14] quantify the error on field line length on a wire/wire configuration. He considered 2 round wires on which a voltage drop was applied (Fig. 12). He realized a study on the error on the length of the electric field line compared to an assumed straight field lines. For *d*=1mm the error is about 20%. In the illustrative example, it is assumed that the straight field lines hypothesis is valid for distances between wires and from the slot edge lower than 1mm.



Fig. 12: assumed straight field lines (left) & error on field line length (right): *U*=900V, *e*=30µm [14]

The author [5] modified the Paschen's curve. He used the same geometrical configuration than author [14]. He identified the breakdown voltage experimentally. Then he updated the secondary electron emission coefficient  $\gamma$  so that the modified Paschen curve intersects the modelled *V*(p*d=d*) curve at the coordinates obtained experimentally (Fig.13). This *V*(p*d=d*) curve is the simulated voltage drop along the assumed straight field line.



Fig. 13: Modified Paschen's curve for given enamel round wires [5]

#### **III. Interaction between the 2 tools**

For bigger overshoot on  $V_1$  (cf. §II.2. B) there may be PD risk remaining after the 2 steps. However, considering the given filling factor, the slot insulation has been increased to its maximum. The machine designer must now reconsider is design by reducing this filling factor or changing the number of turns as shown in Fig.14.



Fig. 14: *Interaction between the two tools*

## **Conclusion**

Partial discharges (PD) phenomena have to be taken into account in the design process of high specific power machine. For now, the PD tool uses the Paschen's criterion to conclude on the PD risk for a given winding. However, this criterion was not formulated for enamel covered round wire. Experimental work is under progress in order to establish a more representative criterion. It will lead to a better understanding of the physic of PD, in presence of polymer coating, in a stator slot. The final objective is to integrate these tools in a global optimization process.

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#### **References**

- [1] , X. Roboam, O. Langlois, H. Piquet, B. Morin, C. Turpin, Hybrid power generation system for aircraft electrical emergency network ',*IEEE IET Electrical Systems Transportation* , vol 1, 2011. .
- [2] F. Barruel, A. Caisley, N. Retiere, J.L. Schanen. 'Stability Approach for Vehicles DC Power Network: Application to Aircraft On-board System',*IEEE 36th Power Electronics Specialists Conference*,2005 .
- [3] Hybrid Aircraft Academic reSearch on Thermal & Electrical Components and Systems (HASTECS), Project selected from CFP in Cleansky II (H2020) framework.
- [4] E. Sili, J.P. Cambronne and F. Koliatene, "Temperature Dependence of electrical Breakdown Mechanism on the Left of the Paschen

Minimum", *IEEE Trans.Plas.Sci*.vol.39, Issue 11, 2011.

- [5] S. Duchesne, G. Parent, J. Moenecley, *et al, '*Prediction of PDIV in motor coils using finite element method. In: Dielectrics (ICD)', *2016 IEEE International Conference on Dielectrics*, pp. 638- 641, 2016.
- [6] J. Pyrhönen, T. Jokinen, V. Hrbovcovà, 'Design of Rotating Electrical Machine's. John Wiley & Sons, pp. 281-300, 2008.
- [7] G. R. Slemon, 'On the design of High-Performance Surface-Mounted PM Motors', *IEEE Transactions on Industry Applications*, vol. 30, n°1, January/February 1994.
- [8] Y. Lefevre, S. El. Aabid, J.F. Llibre, C. Henaux, S. Touhami, 'Performance assessment tool based on loadability concepts', *The 18th Symposium on Applied Electromagnetics and Mechanics*, Sept 3- 6 2017, France.
- [9] S. Touhami, Y. Lefevre, J.F. Llibre, 'Joint Design of Halbach Segment Array and Distributed Stator Winding', *IEEE International Conference Electrical Machine* ICEM'2018, Sept 3-6 2018, Greece.
- [10] F. Paschen, 'Annalen der Physik', vol 37, pp.69- 96, 1889.
- [11] Synflex Elektro GmbH, website: http://www.synflex.de.
- [12] Enamelled Copper Wires Sona Datasheet, website: http://www.universalmetalproducts.com/
- [13] T. W. Dakin et *al*, "Breakdown of gases in uniform fields Paschen curves for nitrogen, air and sulfur hexafluoride ", ELECTRA N°32, pp 61-81.
- [14] L. Benmamas, P. Teste, G. Krebs, E. Odic, F. Vangraefschepe, T. Hamiti, "Contribution to partial discharge analysis in inverter-fed motor windings for automotive application", *Electrical Insulation Conference (EIC) 2017 IEEE*, pp. 348-351, 2017.