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Ulrich Grundmann

**The Code DYN3DR for Steady-State
and Transient Analyses
of Light Water Reactor Cores
with Cartesian Geometry**

Forschungszentrum Rossendorf e.V.

Postfach 51 01 19 · D-01314 Dresden

Bundesrepublik Deutschland

Telefon (0351) 260 3037

Telefax (0351) 260 2383

E-Mail grundman@fz-rossendorf.de

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U. Grundmann

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**Forschungszentrum Rossendorf e. V.
Institut für Sicherheitsforschung
Postfach 510119, D-01314 Dresden**

The code DYN3D/M2 was developed for 3-dimensional steady-state and transient analyses of reactor cores with hexagonal fuel assemblies. The neutron kinetics of the new version DYN3DR is based on a nodal method for the solution of the 3-dimensional 2-group neutron diffusion equation for Cartesian geometry. The thermal-hydraulic model FLOCAL simulating the two phase flow of coolant and the fuel rod behaviour is used in the two versions. The fundamentals for the solution of the neutron diffusion equations in DYN3DR are described. The 3-dimensional NEACRP benchmarks for rod ejections in LWR with quadratic fuel assemblies were calculated and the results were compared with the published solutions. The developed algorithm for neutron kinetics are suitable for using parallel processing. The behaviour of speed-up versus the number of processors is demonstrated for calculations of a static neutron flux distribution using a workstation with 4 processors.

Zusammenfassung

Der Code DYN3D/M2 wurde für 3-dimensionale stationäre und transiente Analysen von Reaktorkernen mit hexagonalen Brennelementen entwickelt. Die Neutronenkinetik der neuen Version DYN3DR basiert auf einer nodalen Methode zur Lösung der 3-dimensionalen 2-Gruppen-Neutronendiffusionsgleichung für kartesische Geometrie. Das thermohydraulische Modell FLOCAL zur Simulation der Zwei-Phasenströmung des Kühlmittels und des Brennstabverhaltens wird in beiden Versionen verwendet. Die Grundlagen zur Lösung der Neutronendiffusionsgleichungen in DYN3DR werden beschrieben. Die 3-dimensionalen NEACRP-Benchmarks für Regelstabauswürfe in LWR's mit quadratischen Brennelementen wurden berechnet und die Ergebnisse mit den veröffentlichten Lösungen verglichen. Der entwickelte Algorithmus für die Neutronenkinetik ist für Parallelrechner geeignet. Das Verhalten des Speedup-Faktors in Abhängigkeit von der Anzahl der Prozessoren wird für die Berechnung einer stationären Neutronenflußverteilung auf einer 4-Prozessor Workstation gezeigt.

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1. Introduction

The new version DYN3DR of the code DYN3D/M2 /1/, initially designed for hexagonal-z core analysis, has been developed for reactor cores with rectangular fuel assemblies. The neutronic part of the two code versions can be used for steady-state and transient calculations of reactor cores. The neutron kinetics in both versions is based on the solution of the two group neutron diffusion equation by nodal methods being different for the hexagonal /2/ and rectangular geometry. The thermal-hydraulic module FLOCAL /3,4/ simulating the two phase flow of coolant and the fuel rod behaviour is used in both versions.

The paper describes the fundamentals of the neutron kinetics of the rectangular option and demonstrates the high accuracy of the code by comparisons with rod ejection benchmarks for a PWR with rectangular fuel assemblies /8,9/.

All parts of the code are written in FORTRAN-77. Since neutron kinetics is very time consuming the developed algorithm's are suitable for using parallel processing. Special commands for utilizing the parallel capabilities on SUN workstations are included in the neutronic part of the code. The code is running on other computers too, because the parallelizing commands are interpreted as comments by usual compilers. The speedup of neutronic calculations is demonstrated for a steady state solution of diffusion equation.

2. Neutron Kinetic Model

The time-dependent neutron diffusion equations for a node n are used in the following form

$$\frac{1}{v_1^n} \frac{\partial \Phi_1^n(\mathbf{r}, t)}{\partial t} + \nabla \mathbf{J}_1^n(\mathbf{r}, t) + \Sigma_r^n(t) \Phi_1^n(\mathbf{r}, t) = \frac{1}{k_{eff}} \sum_{g'=1}^2 (1 - \beta_{g'}^n) \nu \Sigma_{f,g'}^n(t) \Phi_{g'}^n(\mathbf{r}, t) + \sum_{j=1}^M \lambda_j^n C_j^n(\mathbf{r}, t)$$

$$\frac{1}{v_2^n} \frac{\partial \Phi_2^n(\mathbf{r}, t)}{\partial t} + \nabla \mathbf{J}_2^n(\mathbf{r}, t) + \Sigma_a^n(t) \Phi_2^n(\mathbf{r}, t) = \Sigma_s^n(t) \Phi_1^n(\mathbf{r}, t) \quad (1)$$

$$\frac{dC_j^n(\mathbf{r}, t)}{dt} = \frac{1}{k_{eff}} \sum_{g'=1}^2 \beta_{g',j}^n \nu \Sigma_{f,g'}^n(t) \Phi_{g'}^n(\mathbf{r}, t) - \lambda_j^n C_j^n(\mathbf{r}, t) \quad j=1, \dots, M$$

$$\beta_g^n = \sum_{j=1}^M \beta_{g,j}^n$$

with Fick's law

$$\mathbf{J}_g^n(\mathbf{r}, t) = -D_g^n(t) \nabla \Phi_g^n(\mathbf{r}, t) \quad (2)$$

where

$\Phi_g^n(\mathbf{r}, t)$:	neutron flux of energy group g
$\mathbf{J}_g^n(\mathbf{r}, t)$:	net current of neutrons of group g
$\Sigma_r^n(t), \Sigma_a^n(t), \Sigma_s^n(t)$:	macroscopic removal, absorption and scattering cross sections
$\nu \Sigma_{f,g}^n(t)$:	macroscopic fission cross section multiplied by the number of fission neutrons
v_g^n :	mean group velocities of neutrons
$C_j^n(\mathbf{r}, t)$:	density of precursors of group j
$\beta_{g,j}^n$:	effective fractions of delayed neutrons of group j for a fission with an incident neutron of group g in the node n
λ_j :	decay constant for precursors of group j
M :	number of different precursor groups
k_{eff} :	eigenvalue of stationary state

The time integration over the step Δt is performed by using an implicit difference scheme and an exponential transformation technique

$$\Phi_g^n(\mathbf{r}, t') = \exp[\Omega^n(t' - t + \Delta t)] \Psi_g^n(\mathbf{r}, t') \quad (3)$$

$$t - \Delta t \leq t' \leq t$$

That yields the following approximation for the time derivative of neutron flux in the equations of prompt neutrons

$$\frac{d\Phi_g^n(\mathbf{r}, t)}{dt} \approx \frac{1}{\Delta t} [(1 + \Omega^n \Delta t) \Phi_g^n(\mathbf{r}, t) - \exp(\Omega^n \Delta t) \Phi_g^n(\mathbf{r}, t - \Delta t)] \quad (4)$$

The exponent Ω^n of the next time step can be determined from the logarithmic changes of fluxes within Δt by

$$\Omega^n = \frac{1}{\Delta t} \ln \frac{\sum_{g=1}^2 \Phi_g^n(\mathbf{r}, t)}{\sum_{g=1}^2 \Phi_g^n(\mathbf{r}, t - \Delta t)} \quad (5)$$

Integrating the equations for the precursors of delayed neutrons an exponential behaviour of neutron fluxes with exponent Ω^n within the time step Δt is assumed

$$C_j^n(\mathbf{r}, t) = C_j^n(\mathbf{r}, t - \Delta t) \exp(-\lambda_j \Delta t) + \quad (6)$$

$$\frac{1 - \exp[-(\lambda_j + \Omega^n) \Delta t]}{\lambda_j + \Omega^n} \frac{1}{k_{eff}} \sum_{g'=1}^2 \beta_{g',j}^n \nu \Sigma_{f,g'}^n(t) \Phi_{g'}^n(\mathbf{r}, t)$$

By inserting the expressions for time derivative of neutron flux (4) and the precursor densities (6) into the diffusion equations (1) the following inhomogeneous equation system is obtained:

$$-\nabla D_1^n(t) \nabla \Phi_1^n(\mathbf{r}, t) + \left[\Sigma_r^n(t) + \frac{1 + \Omega^n \Delta t}{v_1^n \Delta t} \right] \Phi_1^n(\mathbf{r}, t) = \frac{1}{k_{eff}} \sum_{g'=1}^2 \left(1 - \sum_{j=1}^M \beta_{g',j}^n \delta_j^n \right) v \Sigma_{f,g'}^n(t) \Phi_{g'}^n(\mathbf{r}, t) + \sum_{j=1}^M \lambda_j^n C_j^n(\mathbf{r}, t - \Delta t) \exp(-\lambda_j \Delta t) + \frac{e^{\Omega^n \Delta t}}{v_1 \Delta t} \Phi_1^n(\mathbf{r}, t - \Delta t) \quad (7)$$

$$-\nabla D_2^n(t) \nabla \Phi_2^n(\mathbf{r}, t) + \left[\Sigma_a^n(t) + \frac{1 + \Omega^n \Delta t}{v_2^n \Delta t} \right] \Phi_2^n(\mathbf{r}, t) = \Sigma_s^n(t) \Phi_1^n(\mathbf{r}, t) + \frac{e^{\Omega^n \Delta t}}{v_2 \Delta t} \Phi_2^n(\mathbf{r}, t - \Delta t)$$

with

$$\delta_j^n = \frac{\Omega^n + \lambda_j \exp[-(\lambda_j + \Omega^n) \Delta t]}{\lambda_j + \Omega^n}$$

The equation system can be written for both energy groups in a simpler form

$$-\nabla D_g^n(t) \nabla \Phi_g^n(\mathbf{r}, t) + \bar{\Sigma}_g^n(t) \Phi_g^n(\mathbf{r}, t) = S_g^n(\mathbf{r}, t) \quad (8)$$

with the source terms

$$S_1^n(\mathbf{r}, t) = \frac{1}{k_{eff}} \sum_{g'=1}^2 \left(1 - \sum_{j=1}^M \beta_{g',j}^n \delta_j^n \right) v \Sigma_{f,g'}^n(t) \Phi_{g'}^n(\mathbf{r}, t) + \sum_{j=1}^M \lambda_j^n C_j^n(\mathbf{r}, t - \Delta t) \exp(-\lambda_j \Delta t) + \frac{e^{\Omega^n \Delta t}}{v_1 \Delta t} \Phi_1^n(\mathbf{r}, t - \Delta t) \quad (9)$$

$$S_2^n(\mathbf{r}, t) = \Sigma_s^n(t) \Phi_1^n(\mathbf{r}, t) + \frac{e^{\Omega^n \Delta t}}{v_2 \Delta t} \Phi_2^n(\mathbf{r}, t - \Delta t)$$

and the modified cross sections

$$\bar{\Sigma}_1^n(t) = \left[\Sigma_r^n(t) + \frac{1 + \Omega^n \Delta t}{v_1^n \Delta t} \right] \quad (10)$$

$$\bar{\Sigma}_2^n(t) = \left[\Sigma_a^n(t) + \frac{1 + \Omega^n \Delta t}{v_2^n \Delta t} \right]$$

The term representing the modified fission source in (9) is iterated in the outer iteration process. The time t appears as a parameter in the inhomogeneous equation system only.

For the solution of the stationary equation prompt and delayed neutrons are in equilibrium. Thus a homogeneous equation system is obtained, because the source term consists in the fission source for the first group and in the scattering source for the second group only. Like the iteration in the transient calculation a fission source iteration process is applied also for the solution of static equations.

The node n being a rectangular box of volume

$$V^n = a_x^n \cdot a_y^n \cdot a_z^n \quad (12)$$

is considered when the fluxes are calculated. The used method is similar to nodal methods for rectangular and hexagonal geometry published in /5,6,7/. The index n of the box is omitted in the following. The equations are written without the parameter t for simplification also. Integration over the transverse directions y and z provides a one-dimensional equation in x direction.

$$-D_g \frac{d^2 \bar{\Phi}_g(x)}{dx^2} + \bar{\Sigma}_g \bar{\Phi}_g(x) = \bar{S}_g(x) - \bar{L}_g(x) \quad (13)$$

with

$$\bar{\Phi}_g(x) = \frac{1}{a_y} \frac{1}{a_z} \int_{-\frac{a_y}{2}}^{\frac{a_y}{2}} dy \int_{-\frac{a_z}{2}}^{\frac{a_z}{2}} dz \Phi_g(\mathbf{r}) \quad (14)$$

$$\bar{S}_g(x) = \frac{1}{a_y} \frac{1}{a_z} \int_{-\frac{a_y}{2}}^{\frac{a_y}{2}} dy \int_{-\frac{a_z}{2}}^{\frac{a_z}{2}} dz S_g(\mathbf{r})$$

with the transversal integrated leakage term

$$\bar{L}_g(x) = \frac{-D_g}{a_y a_z} \int_{-\frac{a_y}{2}}^{\frac{a_y}{2}} dy \int_{-\frac{a_z}{2}}^{\frac{a_z}{2}} dz \left(\frac{\partial^2 \Phi_g(\mathbf{r})}{\partial y^2} + \frac{\partial^2 \Phi_g(\mathbf{r})}{\partial z^2} \right) \quad (15)$$

In analogy the y - and z - directions are treated. The one-dimensional transverse-integrated neutron flux is expanded in polynomials up to the second order and exponential functions being solutions of the homogeneous part of equations. Previously, the space coordinate x is transformed by

$$x \rightarrow x/a_x \quad (16)$$

With

$$\begin{aligned} D_g &\rightarrow D_g/a_x \\ \bar{\Sigma}_g &\rightarrow \bar{\Sigma}_g \cdot a_x \\ \bar{S}_g &\rightarrow \bar{S}_g \cdot a_x \\ \bar{L}_g &\rightarrow \bar{L}_g \cdot a_x \end{aligned} \quad (17)$$

the one-dimensional equation (13) is dimensionless. The flux expansion has the following form

$$\bar{\Phi}_g(x) = \sum_{i=0}^2 c_{g,i} w_i(x) + a_{g,1} e^{\kappa_g x} + a_{g,2} e^{-\kappa_g x} \quad (18)$$

with the polynomials

$$\begin{aligned} w_0(x) &= 1 \\ w_1(x) &= 2\sqrt{3}x \\ w_2(x) &= \sqrt{5}\left(\frac{1}{2} - 6x^2\right) \end{aligned} \quad (19)$$

satisfying the relations

$$\int_{-\frac{1}{2}}^{\frac{1}{2}} dx \cdot w_i(x) \cdot w_j(x) = \delta_{ij} \quad (20)$$

δ_{ij} is Kronecker's symbol. The exponent κ_g is obtained from the homogeneous part of equation (13) by

$$\kappa_g = \frac{\bar{\Sigma}_g}{D_g} \quad (21)$$

Assuming that the source terms $\bar{S}_g(x)$ in equation (13) show a more smooth behaviour in the box the source is expanded in the polynomials only.

$$\bar{S}_g(x) \approx \sum_{i=0}^2 s_{g,i} w_i(x) \quad (22)$$

The transversal leakage $\bar{L}_g(x)$ is approximated in the same way from the current of the box and the adjacent boxes.

$$\bar{L}_g(x) \approx \sum_{i=0}^2 l_{g,i} w_i(x) \quad (23)$$

Inserting these expansions into the equation (13) and using the boundary conditions, for example the incoming partial currents at the outer surfaces of the boxes, yield the coefficients of flux expansion (18). Before the next box is calculated the outgoing partial currents are determined. The other directions are treated in an analogous way. After finishing the inner iteration, a new fission source is calculated by updating the polynomial coefficients of fluxes. The outer iteration is accelerated by a Chebychev extrapolation scheme.

The nodal powers calculated by means of the nodal fluxes are transferred to the FLOCAL code. The updated values of fuel temperatures, coolant temperatures and coolant densities are used for the determination of cross sections. An iteration between neutron kinetics and thermal-hydraulics is carried out. Several kinetic steps Δt_k are possible within one thermal hydraulic step Δt_{th} . The iterations over Δt_{th} are finished, when convergence within preset limits is reached.

3. Results of Benchmark Calculations

In order to assess the different codes developed for transient analysis of reactor cores the so-called NEACRP benchmarks were defined /8/. The PWR problems were calculated by DYN3DR and the obtained results were compared with the published reference solutions /9/. The thermal-hydraulic model for the transient calculations in /10/ was specified similar to the NEACRP benchmarks in /8/. The version of FLOCAL slightly modified for the calculations in /10/ is also used here.

The problems A1, A2, B1, B2, C1 and C2 consist in rod ejections of central or peripheral rods at hot zero power (HZP) and full power (FP) conditions.

Short description of the 6 cases:

A1: Ejection of the central rod at HZP

A2: Ejection of the central rod at FP

B1: Ejection of a peripheral rod in octant geometry at HZP

B2: Ejection of a peripheral rod in octant geometry at FP

C1: Ejection of one peripheral rod at HZP

C2: Ejection of one peripheral rod at FP

The nominal power of the reactor is 2775 MW. There are 157 fuel assemblies and 64 reflector elements each with a side length of 20 cm in the core (see fig. 3.1). The core including the axial reflector is divided into 18 layers of different heights as given in the specifications of the problems /8/. The part of a fuel element in a layer is described by one node. In all cases the ejection time of rods is 0.1 s. The comparisons of DYN3DR with other codes and the reference solution for some parameters can be seen in tables 3.1 - 3.9. The given results of the other codes published in /9/ show the range of results. The tables contain the value of the published reference solutions and the deviations of the results obtained by the different codes. The reference solution was generated by the PANTHER code using 4 nodes per assembly in one layer. The generation of a mathematically exact reference solution is still open.

Table 3.1 shows the critical boron concentration of the initial steady states. The nodal power peaking factor of steady state solutions is listed in table 3.2. Table 3.3 gives the values for the static reactivity of the ejected rod. The positive reactivity insertions lead to an excursion of nuclear power reduced by the negative feedback. The results for the time of power peak and the power peak value can be seen in tables 3.4 and 3.5, respectively. The values of the power at the end of calculation ($t = 5$ s) are shown in table 3.6. The core averaged Doppler temperatures at $t = 5$ s are contained in table 3.7. The fuel centerline temperatures calculated with FLOCAL by an extrapolation using the average temperatures of the fuel layers are somewhat higher than the reference in all 6 cases (see table 3.8). The coolant exit temperatures are about 0.6 K higher than the reference values. The differences can be explained by the used procedure for water properties because the same deviations were obtained for the steady state of the FP

of the FP cases. The coolant exit temperatures and the deviations are listed in table 3.9 for the final calculation time.

Fig 3.A1.1 shows the power peak in the case A1 compared to the reference solution, fig 3.A1.2 displays the power curves up to $t = 5$ s. The behaviour of averaged Doppler temperature can be seen in fig 3.A1.3. The maximum fuel centerline temperature $T_{F,C}$ is depicted in figure 3.A1.4. The larger deviation seems to be caused by the mentioned extrapolation method. The same method was also applied for calculating the surface fuel temperature $T_{F,S}$. The Doppler temperature T_D is given by

$$T_D = 0.3 \cdot T_{F,C} + 0.7 \cdot T_{F,S};$$

The figure 3.A1.5 shows the coolant exit temperatures, which are overestimated in DYN3DR in the static and transient results caused by the mentioned calculation of thermophysical properties. Similar notation is used for the curves of cases A2, B1, B2, C1, and C2. The most interesting cases C1 and C2 describing the ejection of a peripheral rod show significant changes of the radial flux shape. Fig. 3.C1.6 and 3.C2.6 show the asymmetric flux shape in axial layer 13 along the horizontal traverse (see fig 3.1) at the time of power maximum. A good agreement can be stated for the HZP case in fig. 3.C1.6. Fig. 3.C2.6 shows small deviations of about 5 % for the FP case in nodes near to the lower end of control rods.

As demonstrated by the tables, the results provided by DYN3DR are within the limits of the other codes. The time dependence of different quantities during the transient predicted by DYN3DR is close to the assumed reference solution in /9/.

4. Parallel Processing

First tests of the neutronic algorithm of DYN3DR were carried out by using the multiprocessor capabilities of a SUN-workstation SPARC 10/514 with a maximum of 4 processors. The code running on single processor machines could be translated by parallelizing compiler options without any changes of the source. That is possible since special commands for parallel DO-loops are interpreted as comments by other compilers.

Fig 4.1 shows the speedup of a 3 - dimensional steady state calculation in comparison to the theoretical curve. The algorithm gives identical results independent on the numbers of processors. The saving of CPU time for the static calculation is important for the parallelization of the transient calculation. The relatively small deviation to the theoretical curve in fig. 4.1 encourages to use a larger number of processors. The parallelization of the thermal-hydraulic code FLOCAL using parallel channel calculation is possible too.

5. Conclusions

The neutron kinetics of the rectangular code DYN3DR is briefly described. The results for the PWR rod ejection benchmarks are compared with the reference solution and published results of other codes. The good agreement with the reference solutions proves the code to be useful for transient calculations in PWR with rectangular fuel assemblies. The saving of CPU time by parallel processing is demonstrated for a steady state neutronic calculation. Further work on parallelization will be done for the transient neutronic calculation and the thermo-hydraulic code FLOCAL.

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Tables

Table 3.1: LWR CORE TRANSIENT BENCHMARKS
PWR: Reference Value of Critical Boron Concentration and Deviations in (%)

Case	A1	A2	B1	B2	C1	C2
Reference (ppm)	567.70	1160.60	1254.60	1189.40	1135.30	1160.60
OKAPI(s)	-1.215	0.974	-0.375	0.572	-0.484	0.974
OKAPI(a)	-1.215	0.974	-0.375	0.572	-0.484	0.974
BOREAS/TRAB	23.71	-0.250	0.486	-0.681	1.964	-0.250
CESAR	3.576	-2.292	0.988	-2.22	1.295	-2.292
PANBOX	-0.511	0.172	-0.072	0.00	-0.167	0.181
QUABOX/CUBBOX	0.458	0.327	1.124	0.336	1.092	0.310
QUANDRY- EN			0.510		0.502	
THYDE-NEU				5.936		
PRORIA	-41.695	3.739			-0.467	4.170
LWRSIM	3.329	1.086	0.319	0.832	0.573	1.086
SIMTRAN	0.194	0.207	0.279	0.193	0.317	0.207
ARROTTA					0.326	5.032
PANTHER	0.740	0.698	0.558	0.504	0.590	0.698
DYN3DR	0.856	0.263	0.138	0.236	0.020	0.252

Table 3.2: LWR CORE TRANSIENT BENCHMARKS
PWR: Steady State Solution: Reference Value of Nodal Power Peaking Factor and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (abs.)	2.874	2.221	1.932	2.109	2.187	2.221
OKAPI(s)	0.244	1.486	-0.259	0.711	-0.137	1.486
OKAPI(a)	0.244	1.486	-0.259	0.711	-0.137	1.486
BOREAS/TRAB	17.606	0.855	4.555	-0.427	2.423	0.855
CESAR	8.559	-1.846	8.178	-2.323	13.397	-1.846
PANBOX	-0.974	-0.630	-0.259	-0.759	-0.137	-0.675
QUABOX/CUBBOX	-1.113	-0.540	0.155	-0.142	0.503	-0.495
QUANDRY- EN			-0.104		-0.137	
LWRSIM	4.175	1.891	3.621	0.806	3.978	1.891
ARROTTA					-1.920	7.249
PANTHER	-0.974	-0.225	-0.311	-0.284	-0.183	-0.225
DYN3DR	-0.574	0.090	0.430	0.474	-0.274	0.135

Table 3.3: LWR CORE TRANSIENT BENCHMARKS
PWR: Reference Value of Reactivity Release and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (pcm)	822.00	90.00	831.00	99.00	958.0	78.00
CESAR	-4.015	-4.444	-2.407	8.081	-1.983	2.564
PANBOX	-1.460	-1.111	-0.963	2.020	-0.731	2.564
SIMTRAN	-1.460	0.000	0.000	2.020	0.313	3.846
PANTHER	-1.825	-3.333	0.120	1.010	0.731	1.282
DYN3DR	-1.749	-1.754	0.564	-2.92	-0.063	4.782

Table 3.4: LWR CORE TRANSIENT BENCHMARKS
PWR: Time of Power Maximum (Reference) and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (s)	0.560	0.100	0.520	0.120	0.270	0.100
OKAPI(s)	5.357	-20.000	-3.846	-16.667	-11.111	0.000
OKAPI(a)	3.571		-7.692		-11.111	
BOREAS/TRAB	12.500	110.000	-5.769	241.667	3.704	110.0
CESAR	37.500	0.000	17.308	-16.667	11.111	0.000
COCCINELLE					18.519	-20.0
PANBOX	7.143	0.000	0.000	-8.333	0.000	0.000
QUABOX/CUBBOX	5.357	-10.000	-19.231	-33.333	11.111	-10.0
QUANDRY- EN			-3.846		-3.704	
REFLA/TRAC	-73.214					
THYDE-NEU				575.000		
PRORIA	221.490	0.000				0.000
SIMTRAN	14.286	0.000	0.000	-8.333	0.000	0.000
ARROTTA					-3.704	-20.0
PANTHER	16.071	0.000	0.000	0.000	-3.704	20.0
DYN3DR	11.340	0.000	-0.960	-12.500	-1.852	-2.500

Table 3.5: LWR CORE TRANSIENT BENCHMARKS
PWR: Reference of Power Maximum (% of P/2775 MW) and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (%)	117.900	108.000	244.100	106.300	477.300	107.100
OKAPI(s)	-35.199	-0.370	2.417	-3.104	-36.141	-0.654
OKAPI(a)	-0.339		20.401		30.170	
BOREAS/TRAB	25.106	-0.185	3.154	-0.188	164.215	-0.093
CESAR	-46.141	1.019	-30.602	0.659	-31.636	1.401
COCCINELLE					-41.400	-0.093
PANBOX	-12.383	0.000	-1.680	0.282	-1.152	0.280
QUABOX/CUBBOX	12.299	0.833	125.604	1.035	130.254	1.774
QUANDRY- EN			0.901		10.245	
THYDE-NEU				-0.188		
PRORIA	-73.028	0.278				0.187
SIMTRAN	-28.329	0.278	-5.326	0.282	-7.668	0.560
ARROTTA					3.394	0.934
PANTHER	-23.749	-0.278	8.193	0.094	16.740	0.093
DYN3DR	-20.322	0.268	3.978	-0.039	-0.071	-1.666

Table 3.6: LWR CORE TRANSIENT BENCHMARKS
PWR: Reference of Final Power (% of P/2775 MW) and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (%)	19.600	103.500	32.000	103.800	14.600	103.000
OKAPI(s)	-1.020	-0.676	-4.688	-1.927	7.534	-0.485
OKAPI(a)	25.000		1.250		47.260	
BOREAS/TRAB	70.408	0.773	12.500	0.674	191.096	0.874
CESAR	-13.265	-0.097	-12.187	0.000	-13.014	0.000
COCCINELLE					1.370	0.097
PANBOX	0.510	0.097	1.563	0.289	2.740	0.194
QUABOX/CUBBOX	-0.510	0.676	3.750	0.771	4.110	1.068
QUANDRY- EN			-23.438		-17.123	
THYDE-NEU				0.963		
PRORIA	26.020	0.193				0.583
SIMTRAN	6.122	0.097	4.688	0.193	6.164	0.194
ARROTTA					19.178	0.000
PANTHER	-0.510	-0.097	1.563	0.096	2.740	0.097
DYN3DR	2.791	0.095	4.653	0.070	5.932	0.303

Table 3.7: LWR CORE TRANSIENT BENCHMARKS
Reference of Final Core Averaged Doppler Temperature and Deviations
in (%)

Case	A1	A2	B1	B2	C1	C2
Reference (°C)	324.300	554.600	349.900	552.000	315.900	553.500
OKAPI(s)	-0.278	-0.252	-0.343	-0.707	0.380	-0.199
OKAPI(a)	3.238		0.800		4.653	
BOREAS/TRAB	10.207	3.642	4.201	4.638	21.937	3.668
CESAR	-0.648	-7.339	-0.657	6.902	-0.443	7.371
COCCINELLE					0.222	0.452
PANBOX	0.463	1.100	0.915	1.159	0.538	1.102
QUABOX/CUBBOX			2.686	2.428		
QUANDRY- EN			0.714		0.823	
THYDE-NEU				9.601		
PRORIA	2.744	2.669				3.847
SIMTRAN	1.079	2.164	1.600	2.174	0.950	2.186
ARROTTA					-2.501	-23.035
PANTHER	-0.123	-0.072	0.372	-0.018	0.348	-0.018
DYN3DR	0.469	0.315	1.294	0.264	0.681	0.275

Table 3.8: LWR CORE TRANSIENT BENCHMARKS
Reference of Maximum Fuel Centerline Temperature and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (°C)	673.300	1691.800	559.800	1588.100	676.100	1733.500
OKAPI(s)	-2.866	-0.307	-1.304	-0.441	-0.666	-2.798
OKAPI(a)	19.768		2.697		33.279	
CESAR	-6.728	0.254	-5.359	-0.252	-2.721	-0.329
COCCINELLE					1.834	3.340
PANBOX	0.965	0.106	1.536	-0.025	2.588	0.317
REFLA/TRAC	-0.624		0.714			
THYDE-NEU				-0.617		
SIMTRAN	3.802	1.170	3.019	0.686	5.118	1.546
PANTHER	-1.233	-0.136	0.875	-0.164	3.062	-0.012
DYN3DR	1.604	0.669	3.530	0.929	4.788	1.603

Table 3.9: LWR CORE TRANSIENT BENCHMARKS
Reference of Final Coolant Outlet Temperature and Deviations (%)

Case	A1	A2	B1	B2	C1	C2
Reference (°C)	293.100	324.600	297.600	324.700	291.500	324.500
OKAPI(s)	-0.102	-0.031	-0.134	-0.154	0.069	0.000
OKAPI(a)	0.580		0.101		0.892	
BOREAS/TRAB	1.638	0.154	0.470	0.123	3.636	0.154
CESAR	-0.307	-0.031	-0.437	-0.031	-0.206	-0.031
COCCINELLE						
PANBOX	0.034	0.154	0.101	0.154	0.069	0.154
QUABOX/CUBBOX	-0.034	0.092	0.101	0.154	-0.034	0.062
QUANDRY- EN			-1.647		-0.755	
THYDE-NEU				0.400		
PRORIA	0.068	1.078				-1.079
SIMTRAN	0.068	-0.031	0.101	-0.031	0.069	-0.031
ARROTTA					0.343	0.000
PANTHER	-0.034	0.031	0.067	0.031	0.069	0.031
DYN3DR	0.060	-0.117	0.239	0.176	0.129	0.200

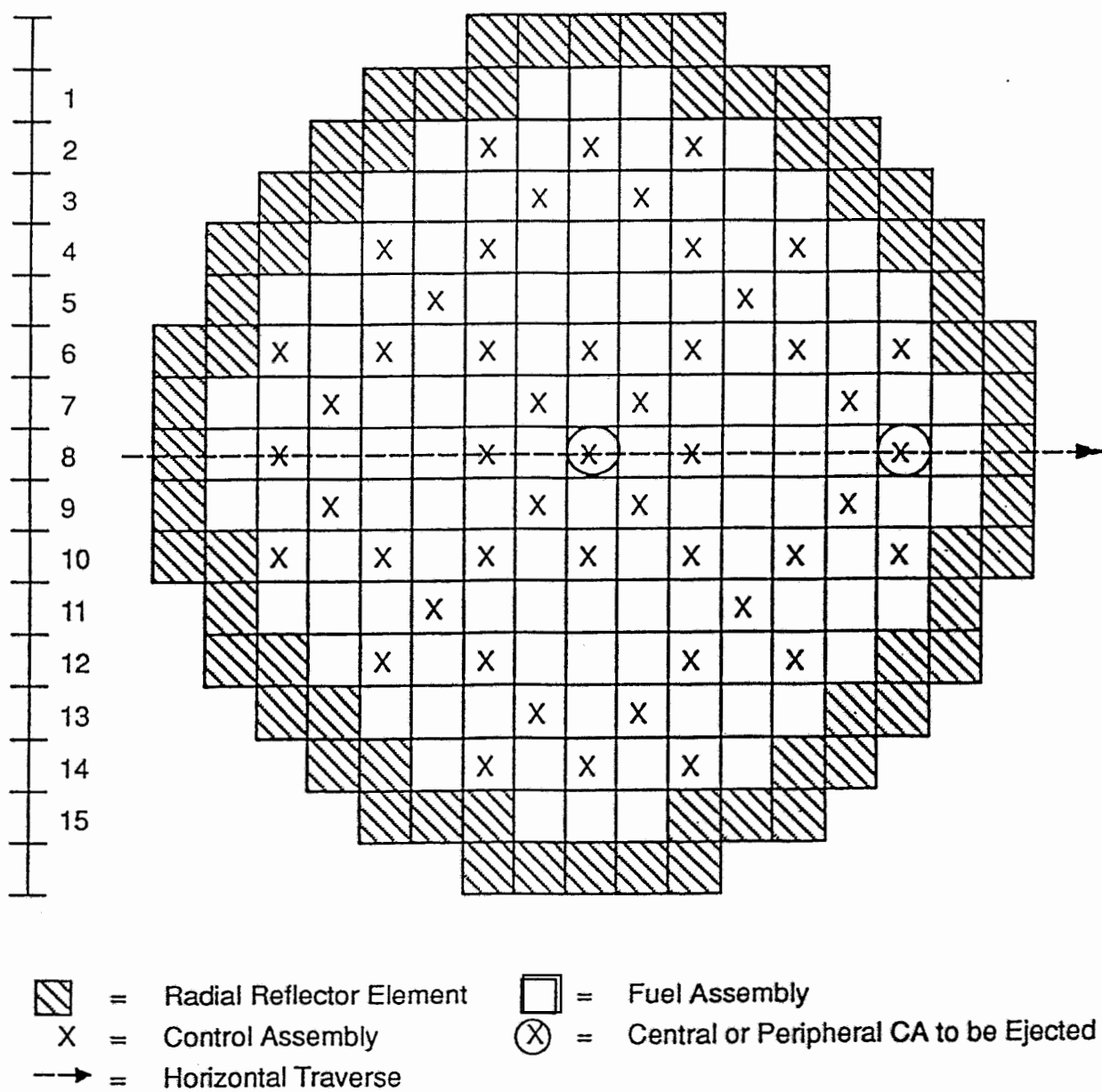


Fig. 3.1: Core Map of the PWR Benchmark Problems

PWR Case A1: Ejection of Central Rod at HZP

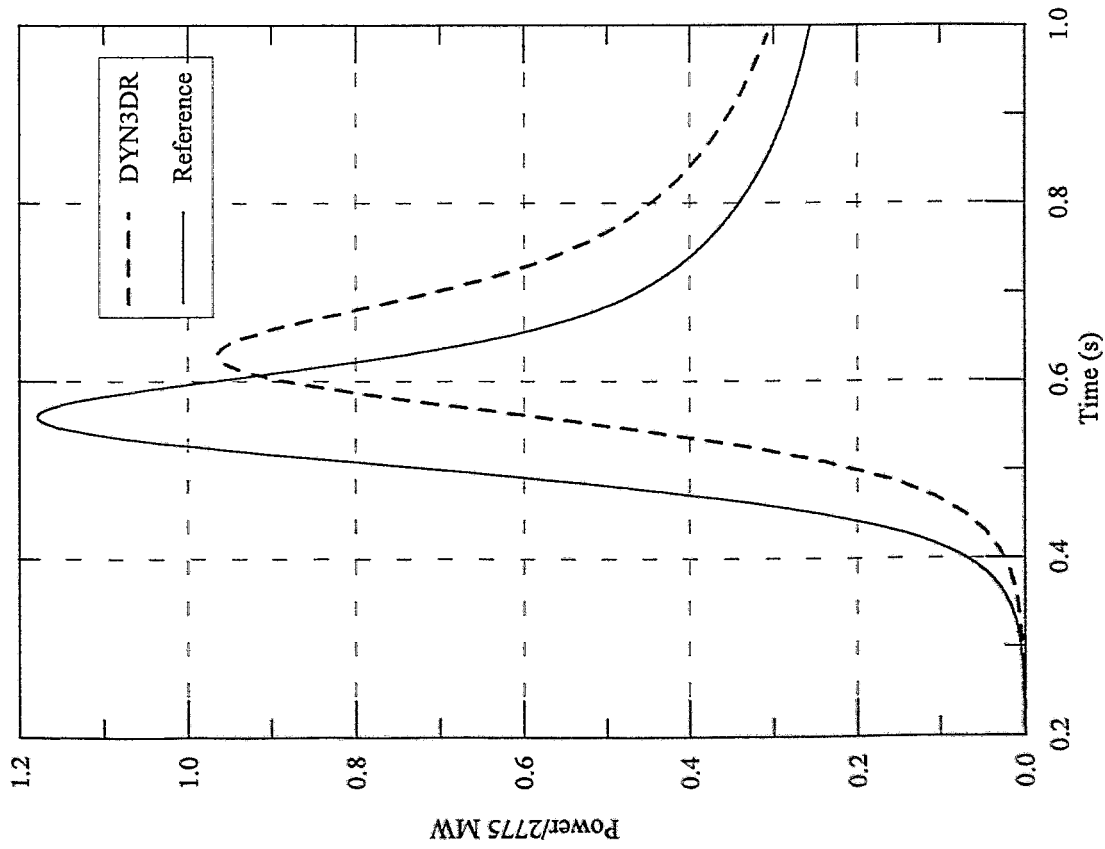


Fig. 3.A1.1: Nuclear Power

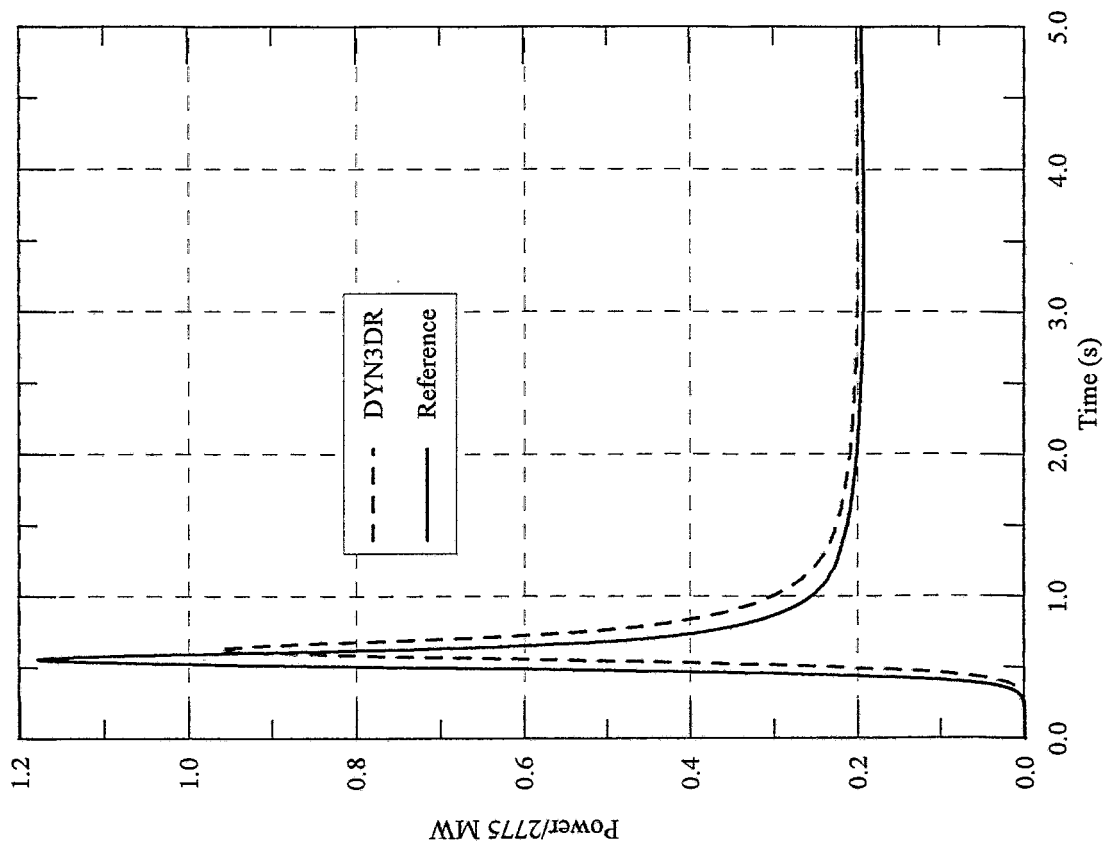


Fig. 3.A1.2: Nuclear Power

PWR Case A1: Ejection of Central Rod at HZP

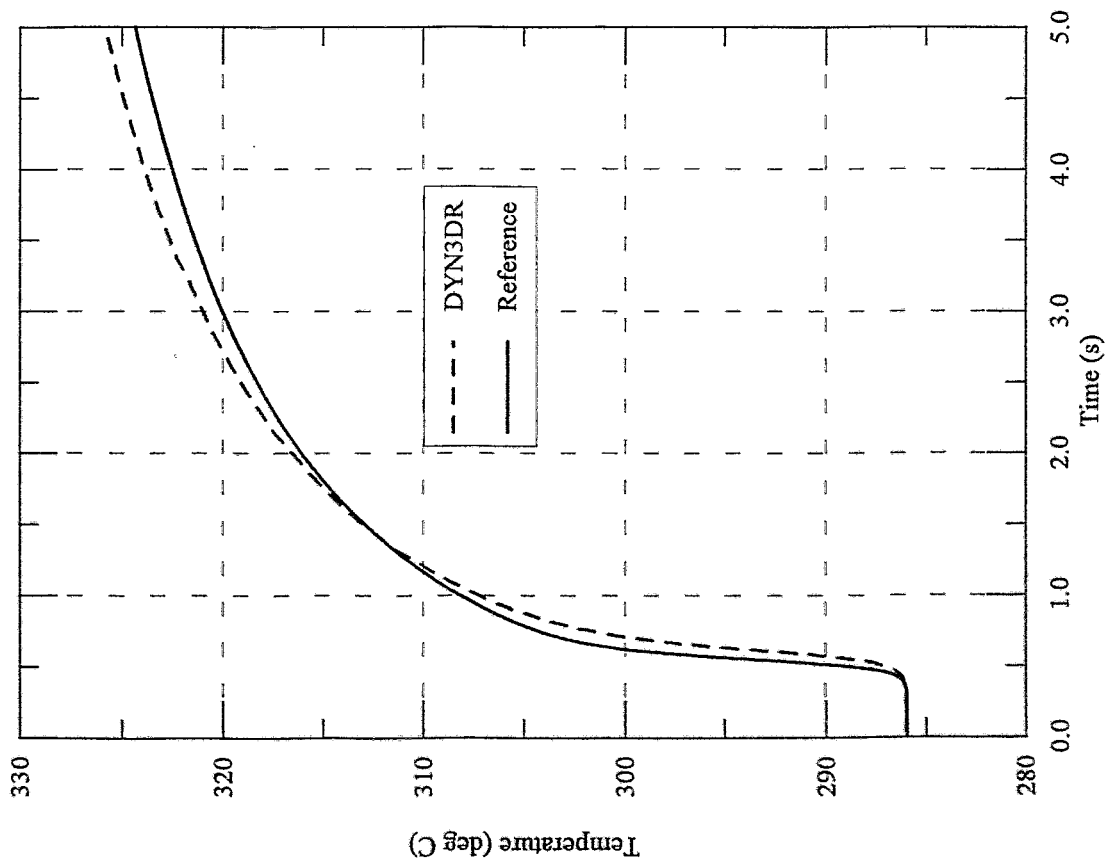


Fig. 3.A1.3: Core Averaged Doppler Temperature

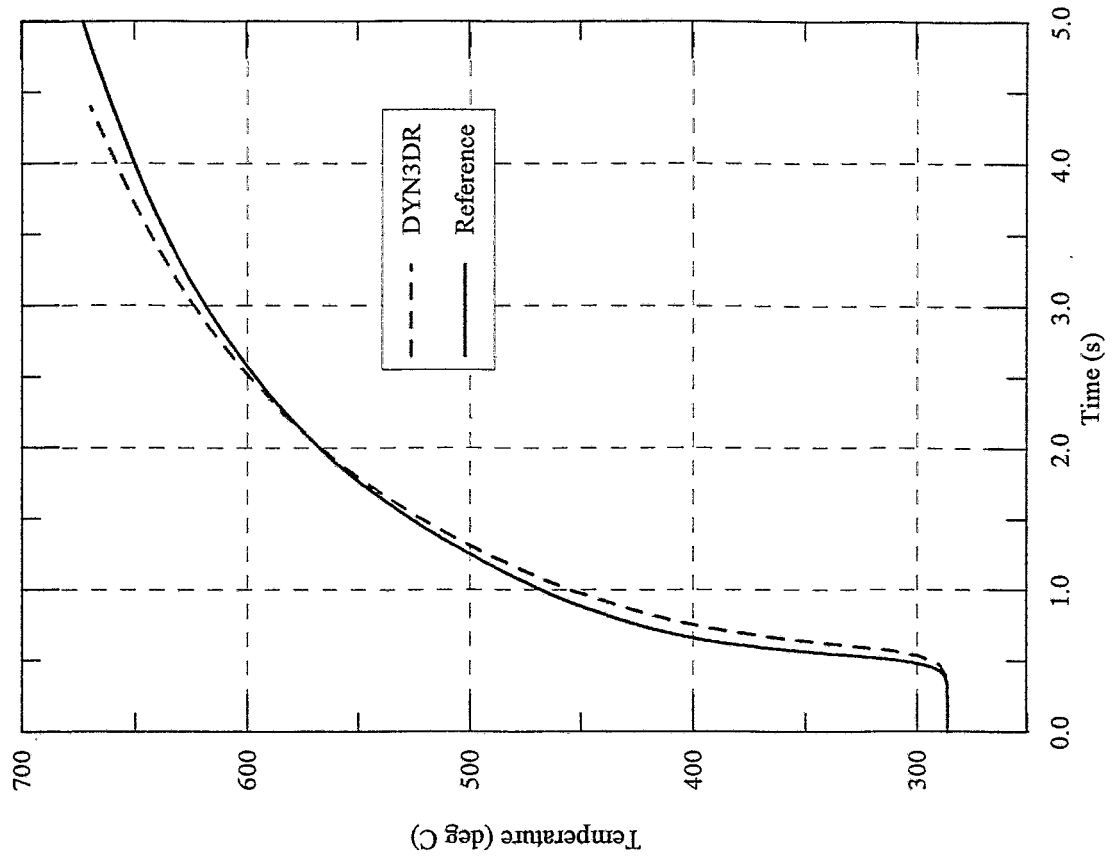


Fig. 3.A1.4: Maximum Fuel Centerline Temperature

PWR Case A1: Ejection of Central Rod at HZP

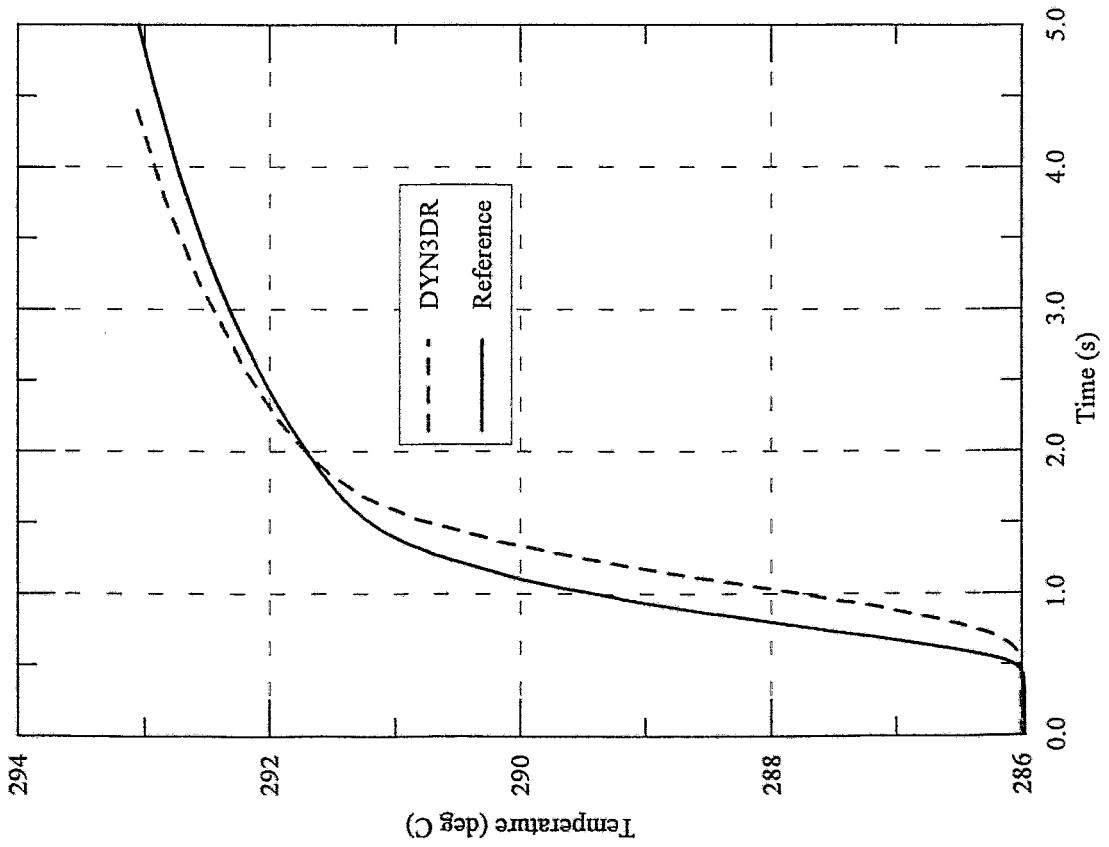


Fig. 3.A1.5: Coolant Exit Temperature

PWR Case A2: Ejection of Central Rod at FP

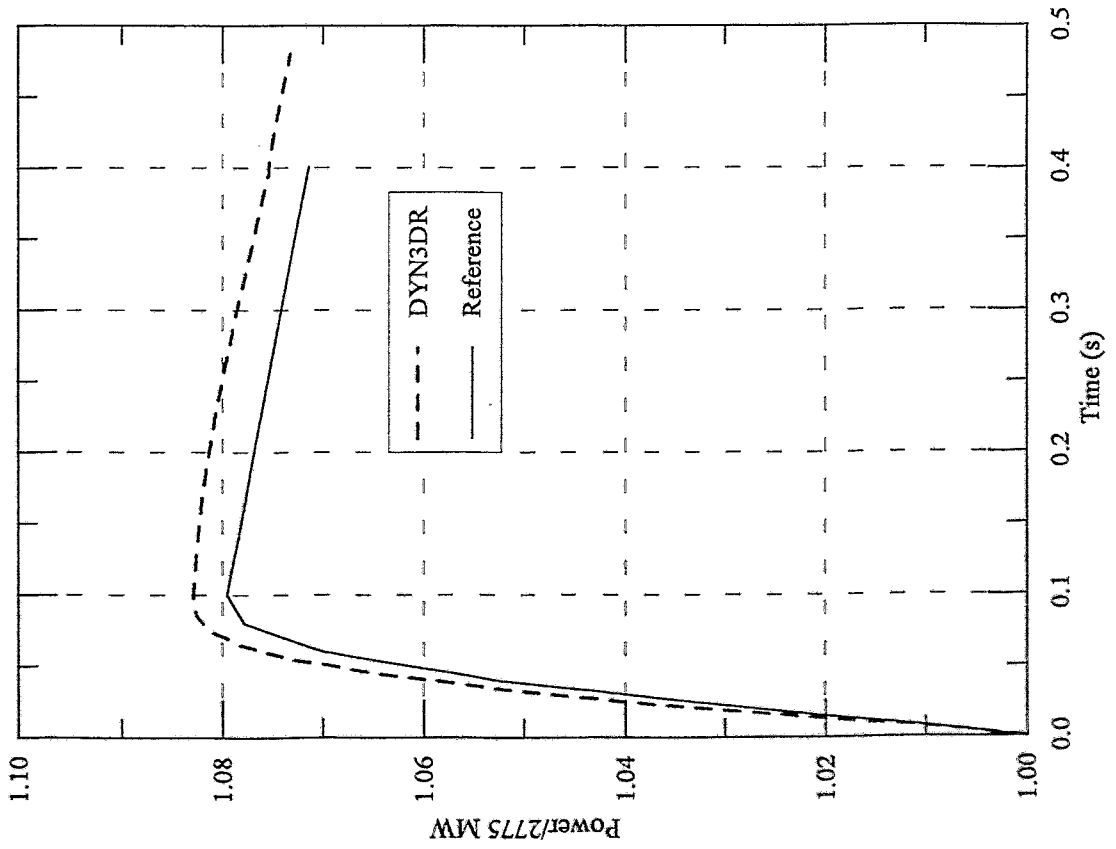


Fig 3.A2.1: Nuclear Power

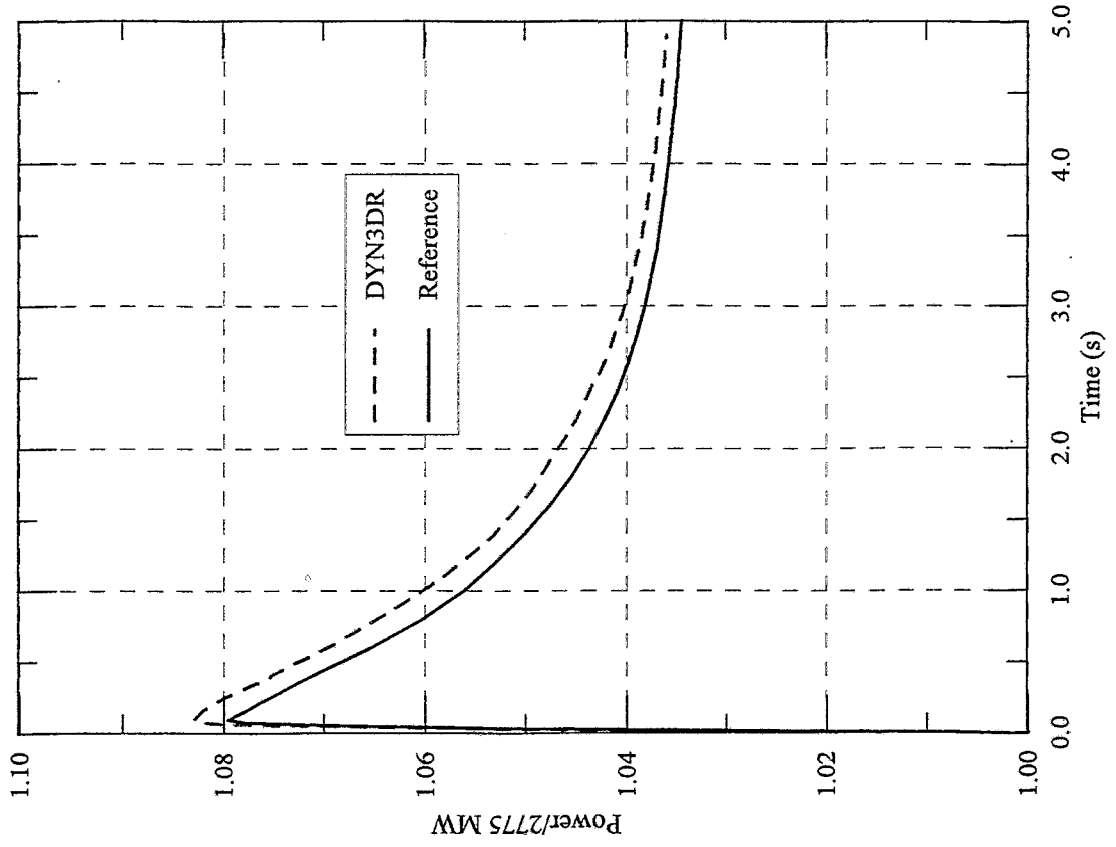


Fig 3.A2.2: Nuclear Power

PWR Case A2: Ejection of Central Rod at FP

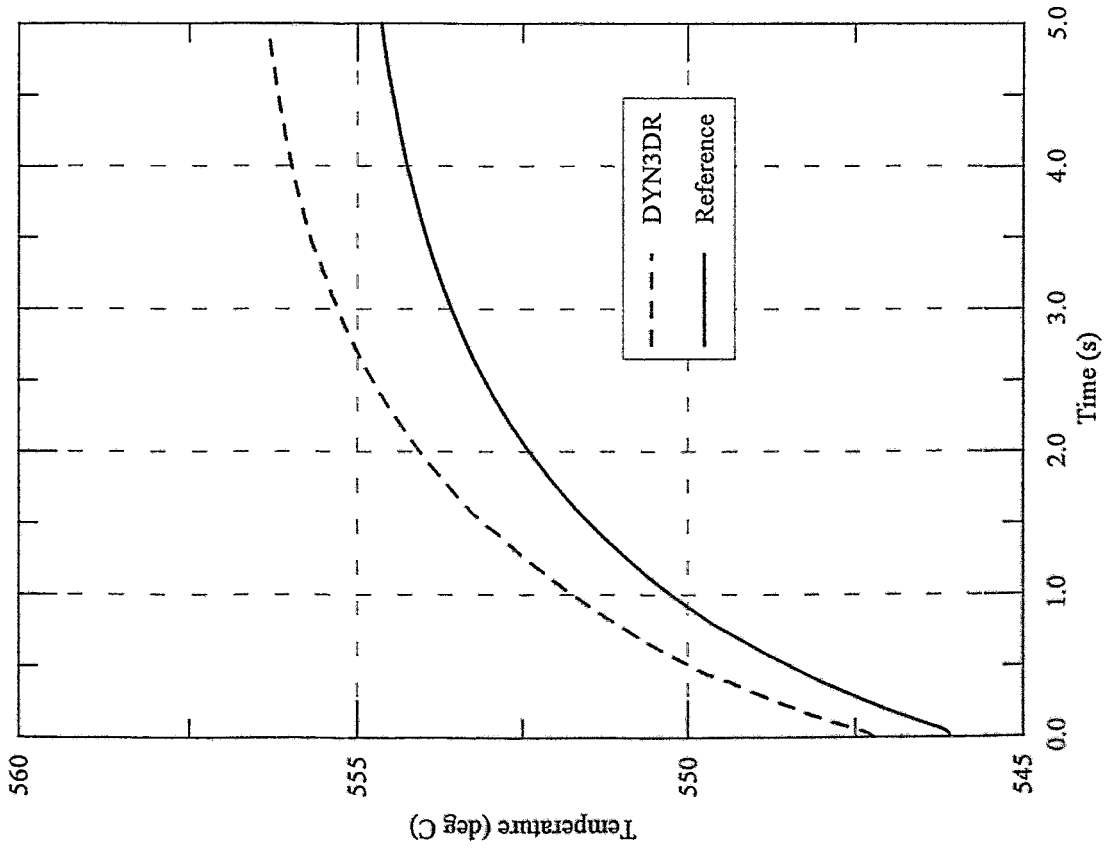


Fig. 3.A2.3: Core Averaged Doppler Temperature

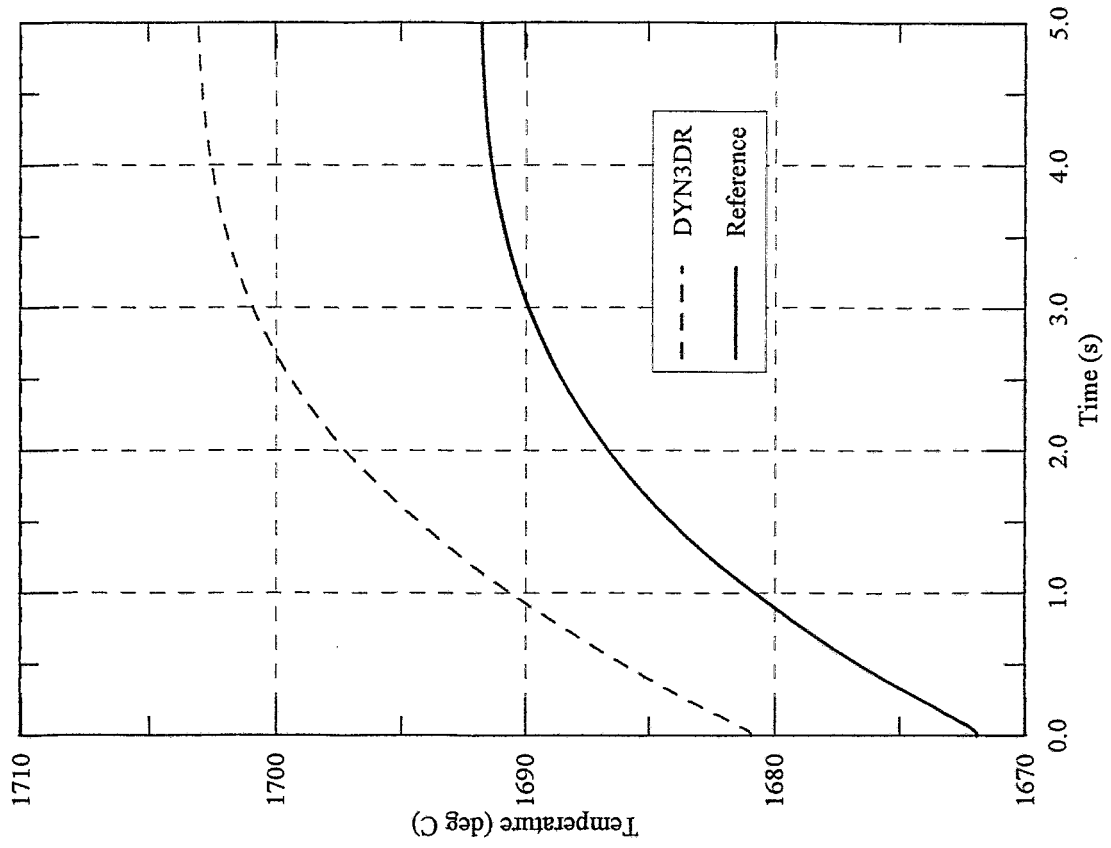


Fig. 3.A2.4: Maximum Fuel Centerline Temperature

PWR Case A2: Ejection of Central Rod at FP

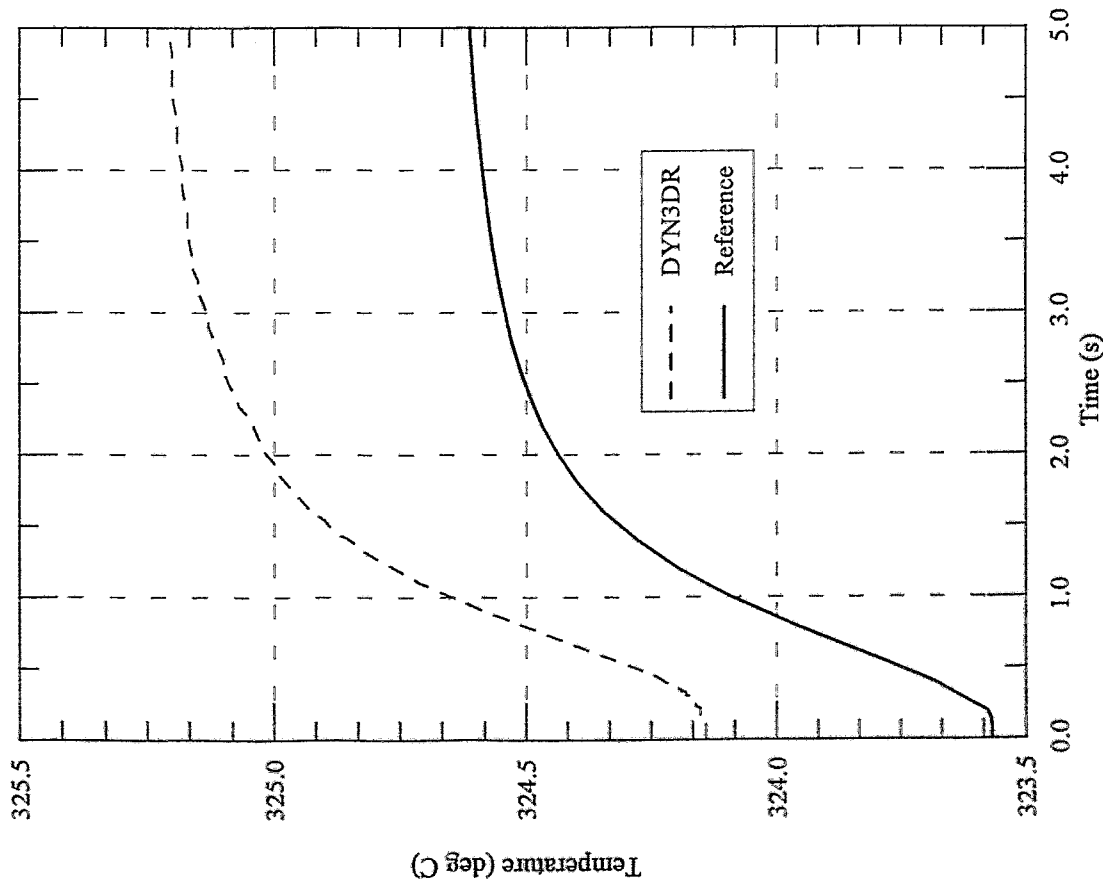


Fig. 3.A2.5: Coolant Exit Temperature

PWR Case B1: Ejection of a Peripheral Rod in Octant Geometry at HZP

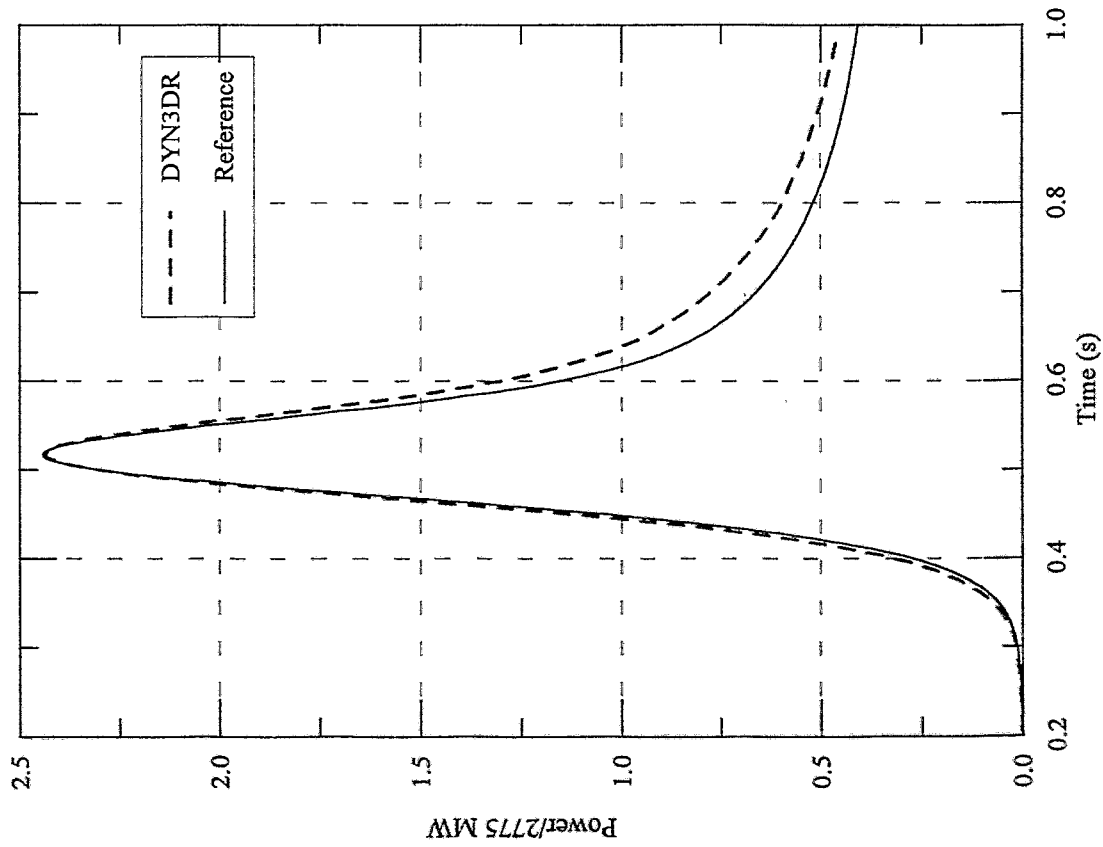


Fig. 3.B1.1: Nuclear Power

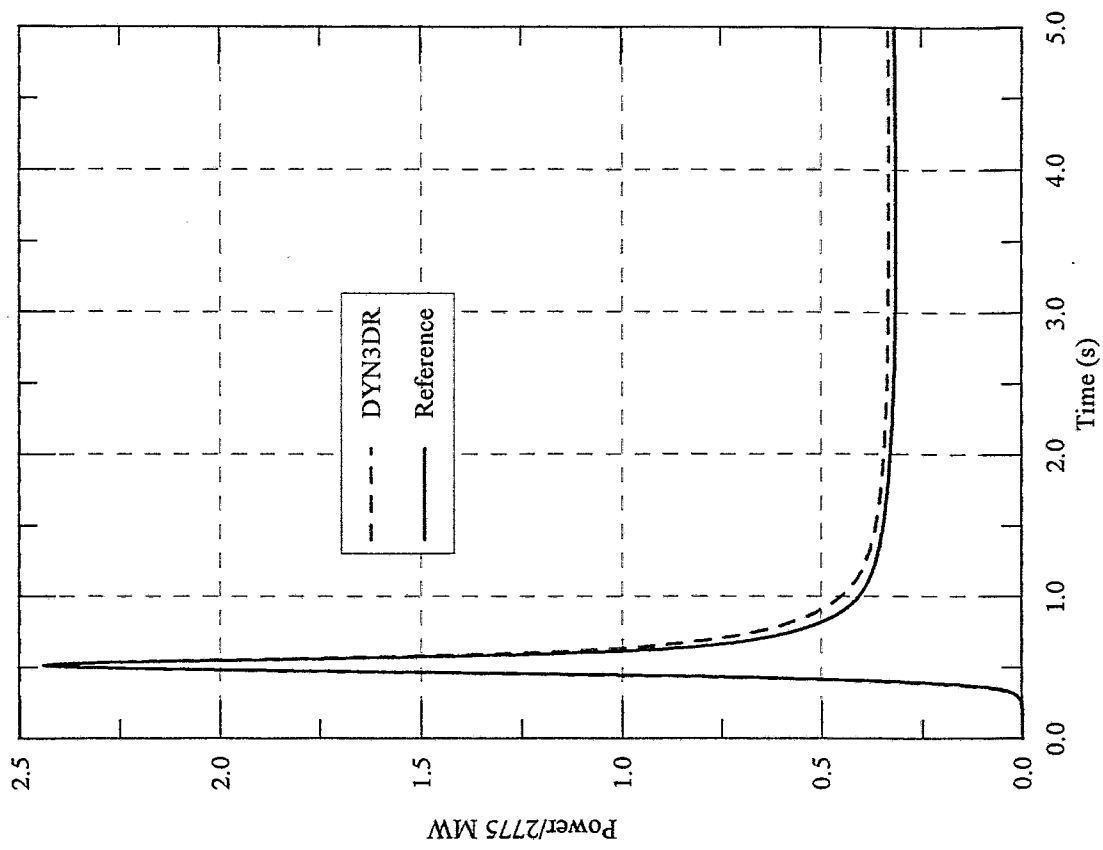


Fig. 3.B1.2: Nuclear Power

PWR Case B1: Ejection of a Peripheral Rod in Octant Geometry at HZP

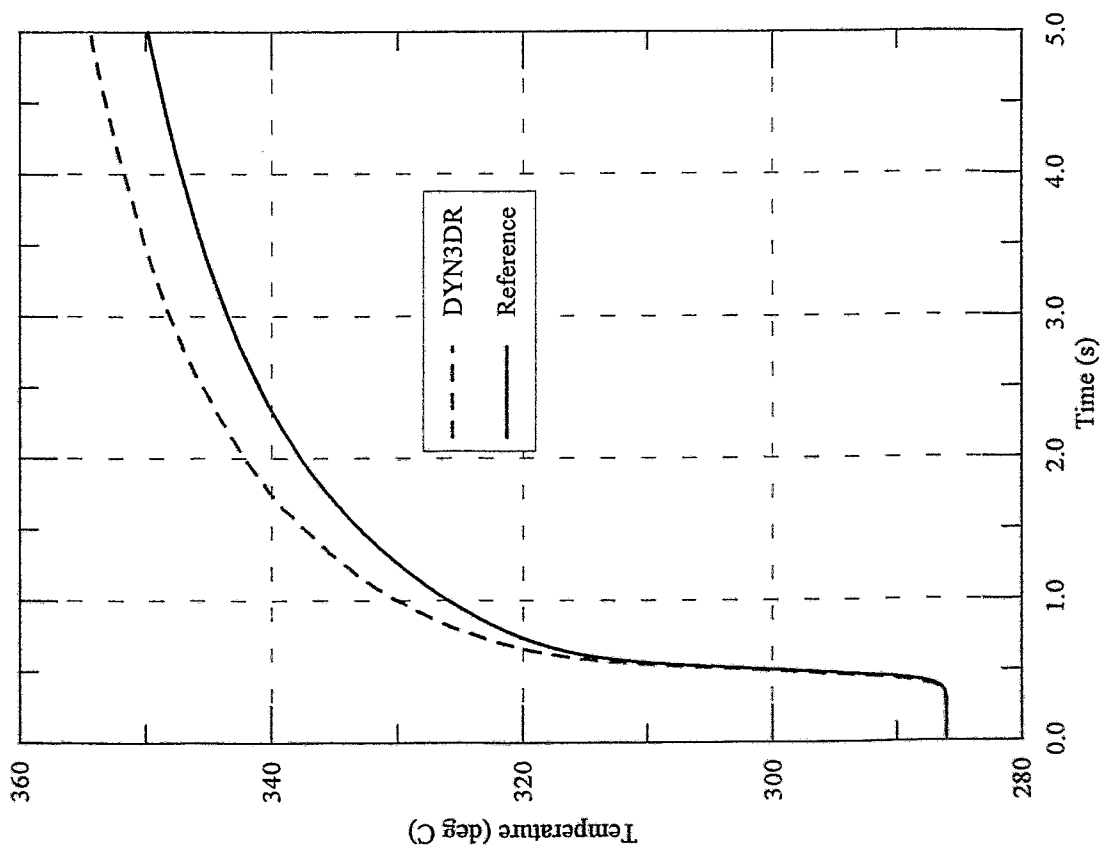


Fig. 3.B1.3: Core Averaged Doppler Temperature

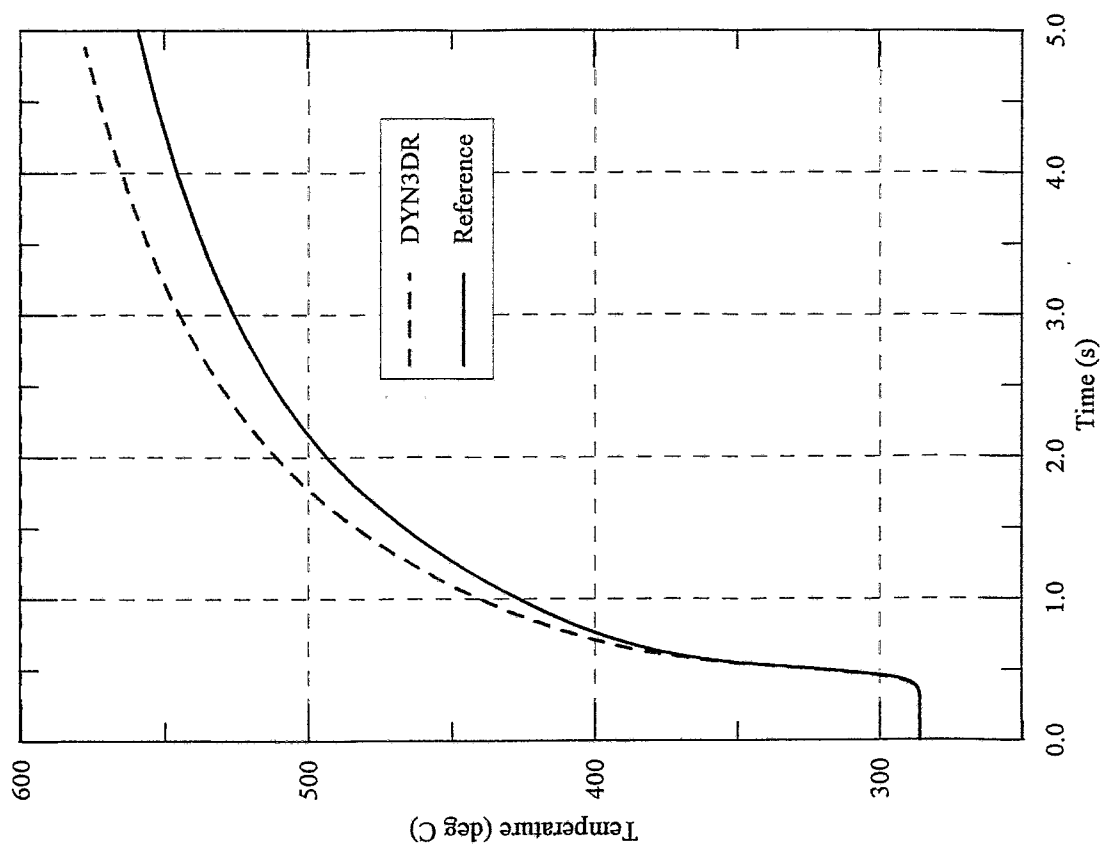


Fig. 3.B1.4: Maximum Fuel Centerline Temperature

PWR Case B1: Ejection of a Peripheral Rod in Octant Geometry at HZP

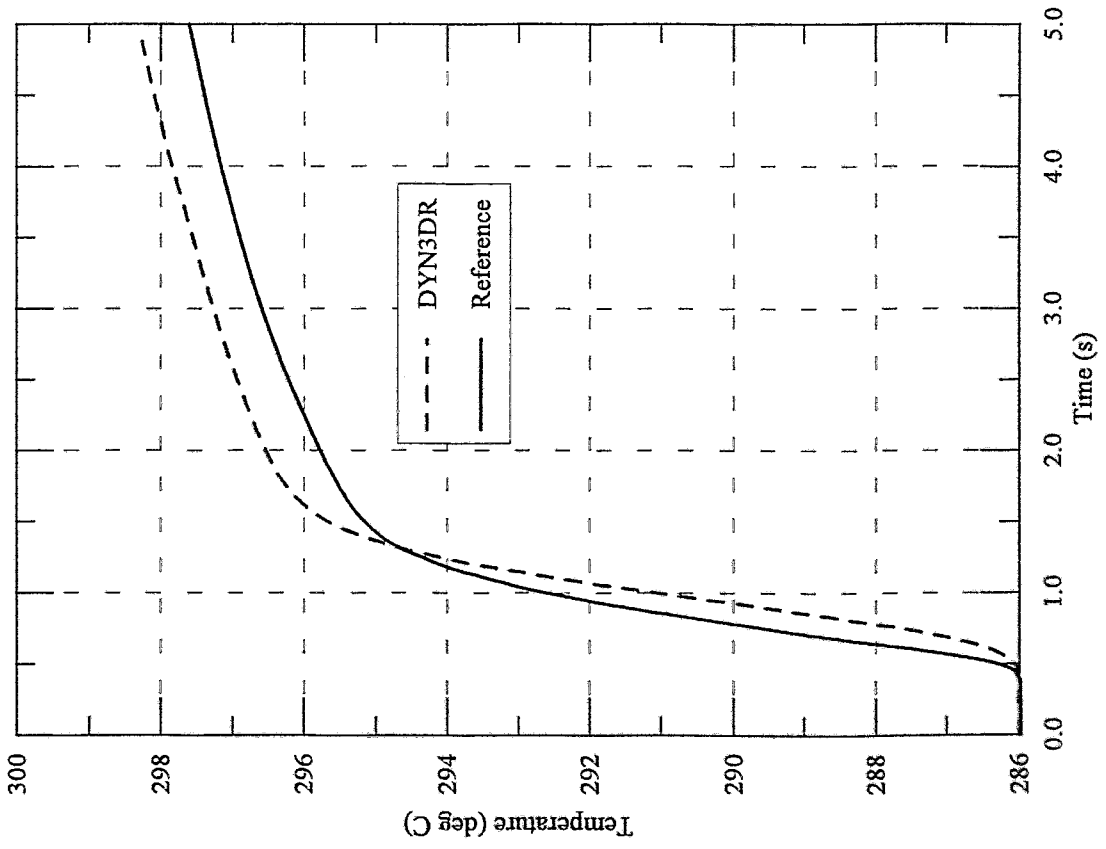


Fig. 3.B1.5: Coolant Exit Temperature

PWR Case B2: Ejection of a Peripheral Rod in Octant Geometry at FP

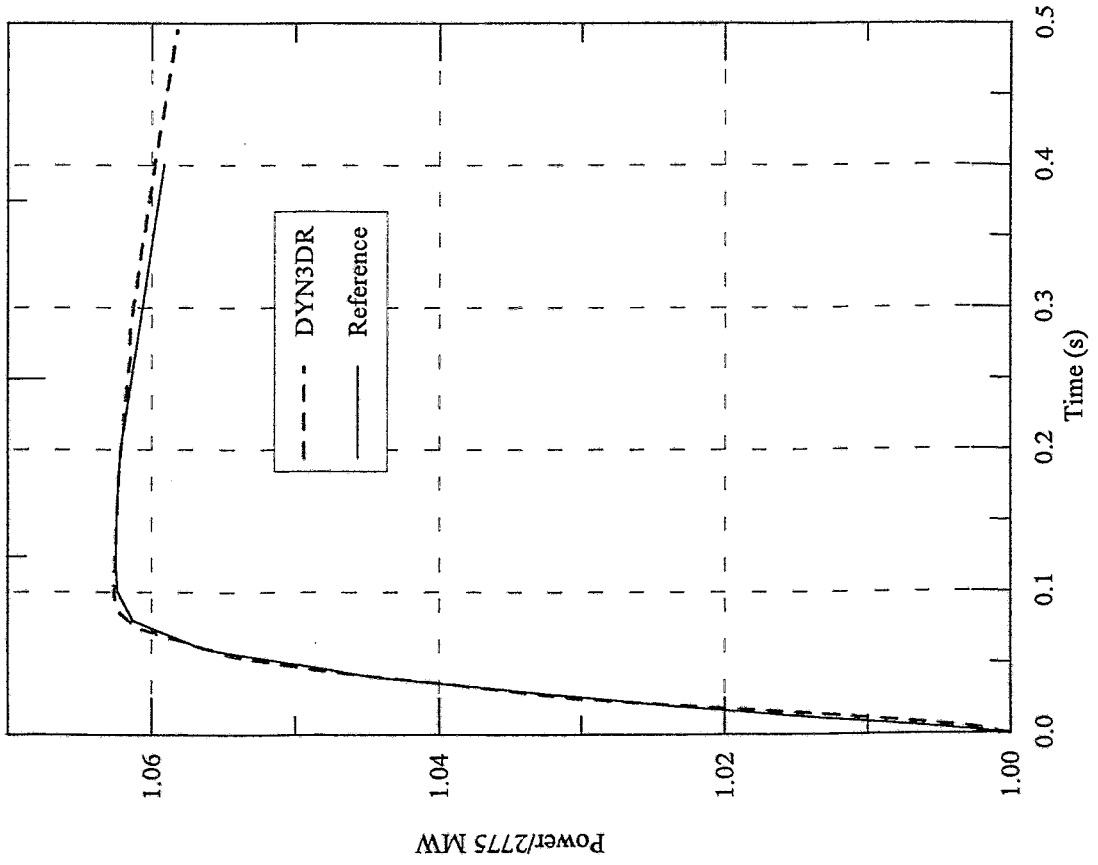


Fig. 3.B2.1: Nuclear Power

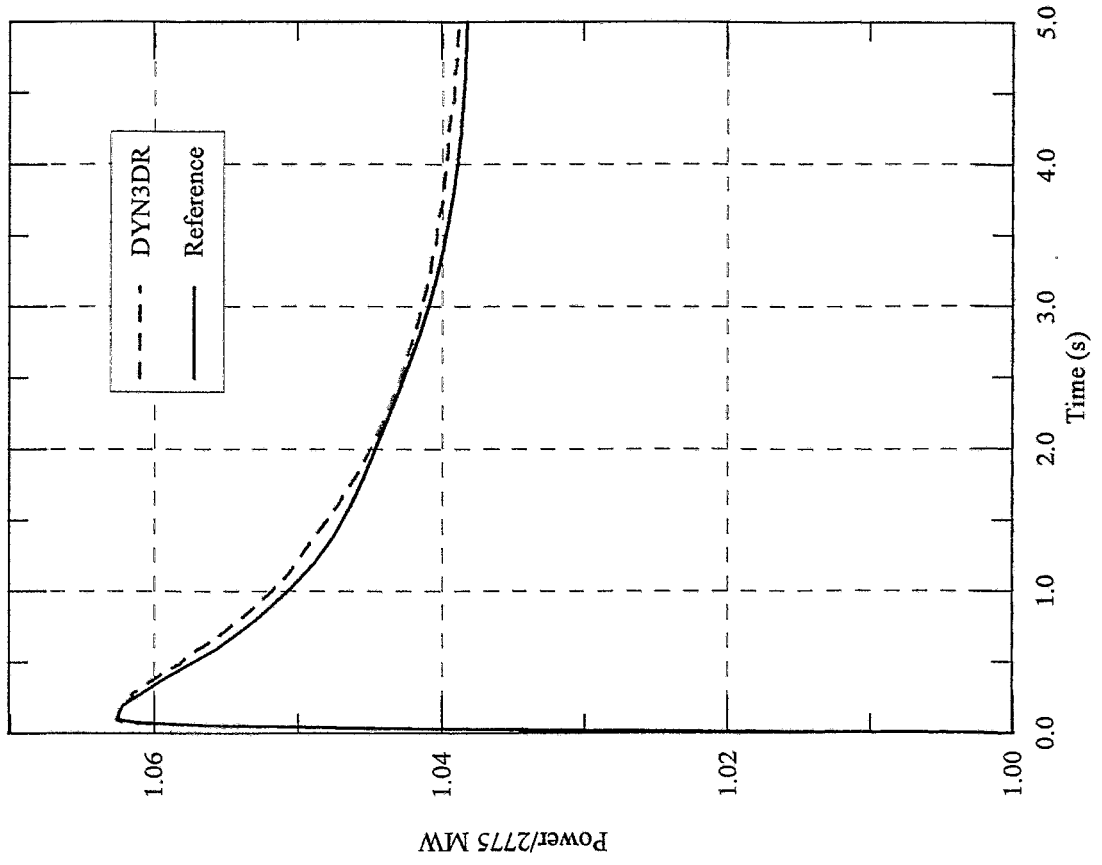


Fig. 3.B2.2: Nuclear Power

PWR Case B2: Ejection of a Peripheral Rod in Octant Geometry at FP

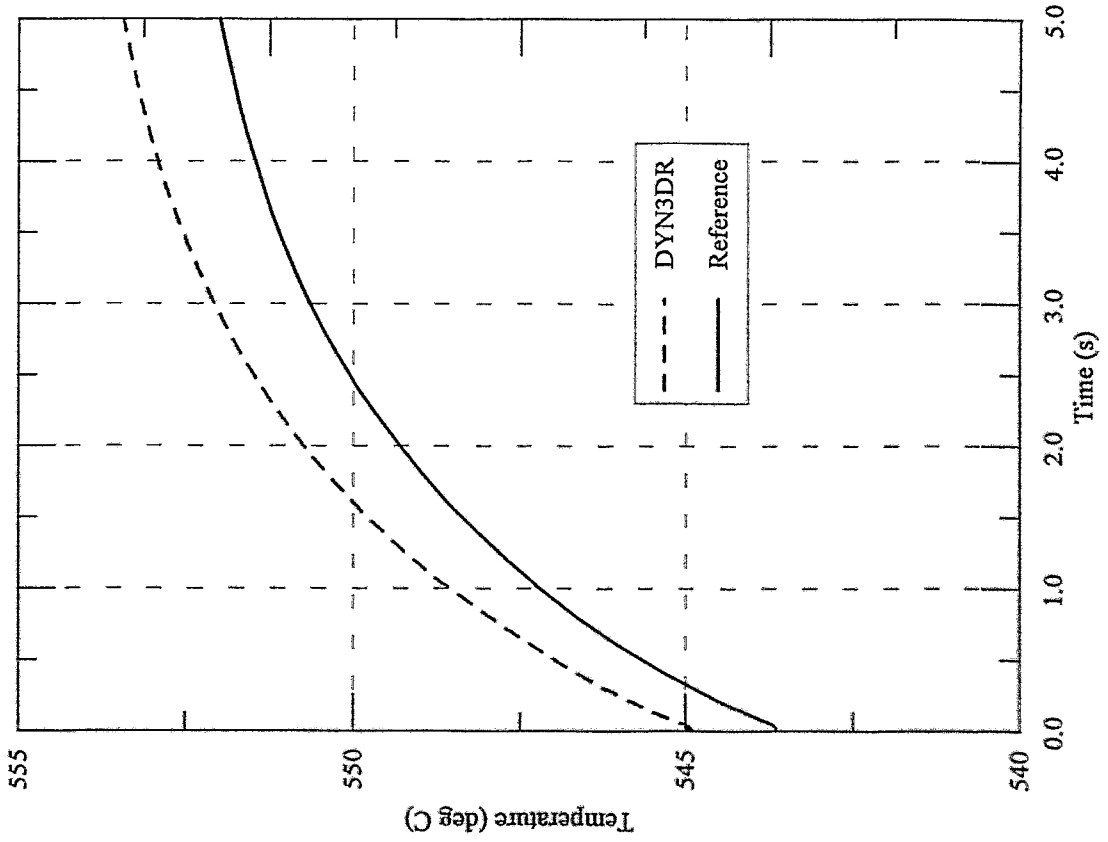


Fig. 3.B2.3: Core Averaged Doppler Temperature

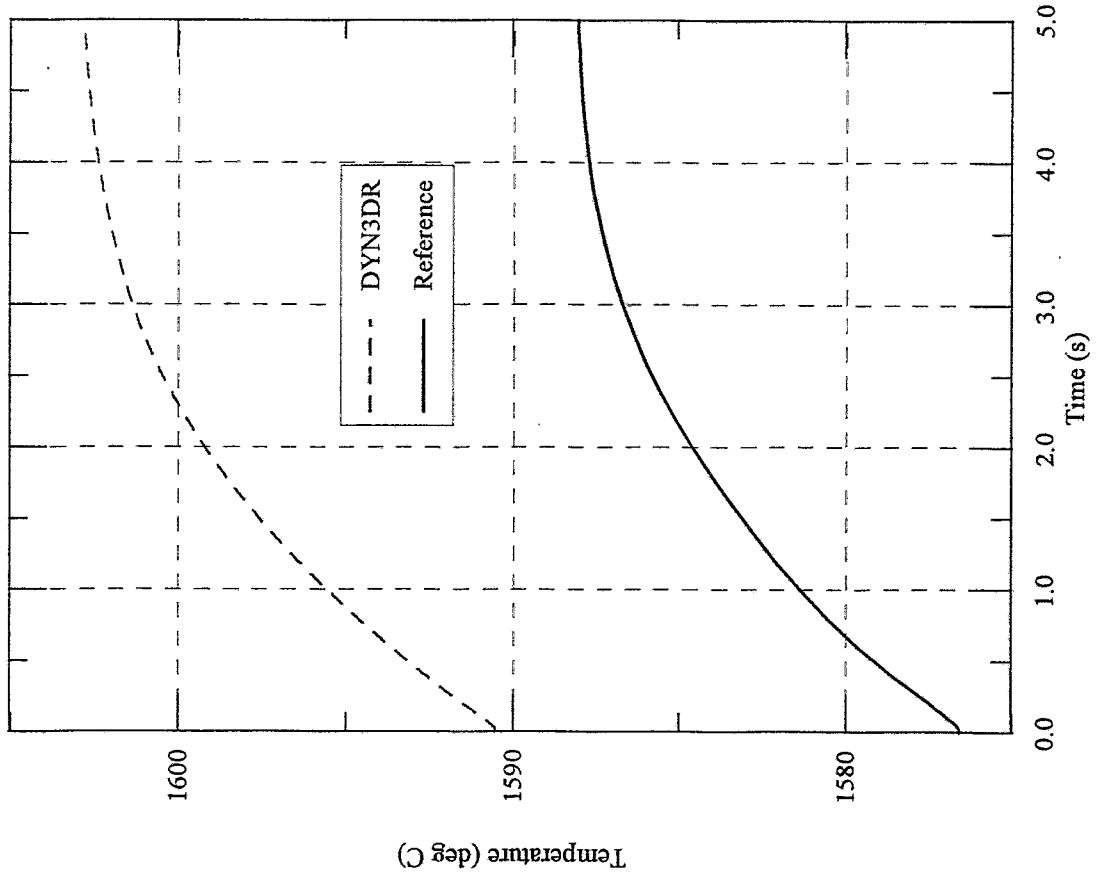


Fig. 3.B2.4: Maximum Fuel Centerline Temperature

PWR Case B2: Ejection of a Peripheral Rod in Octant Geometry at FP

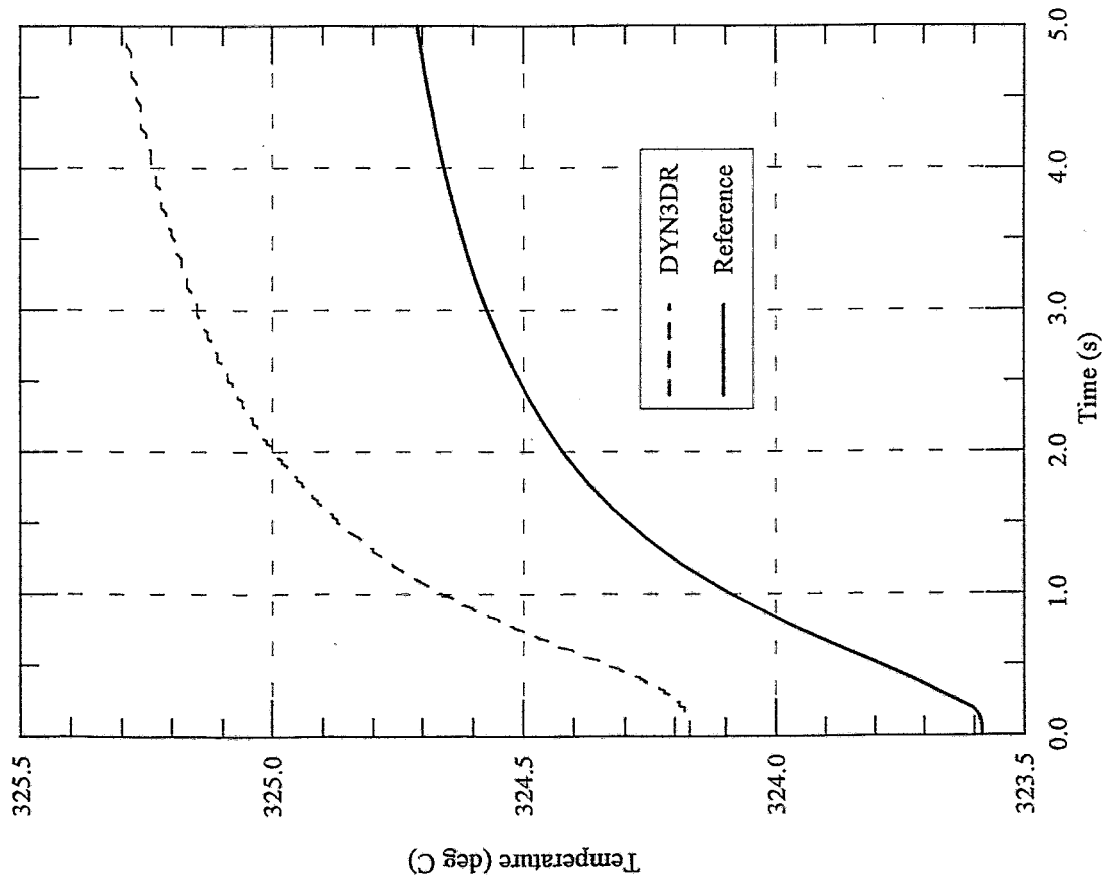


Fig. 3.B2.5: Coolant Exit Temperature

PWR Case C1: Ejection of a Peripheral Rod at HZP

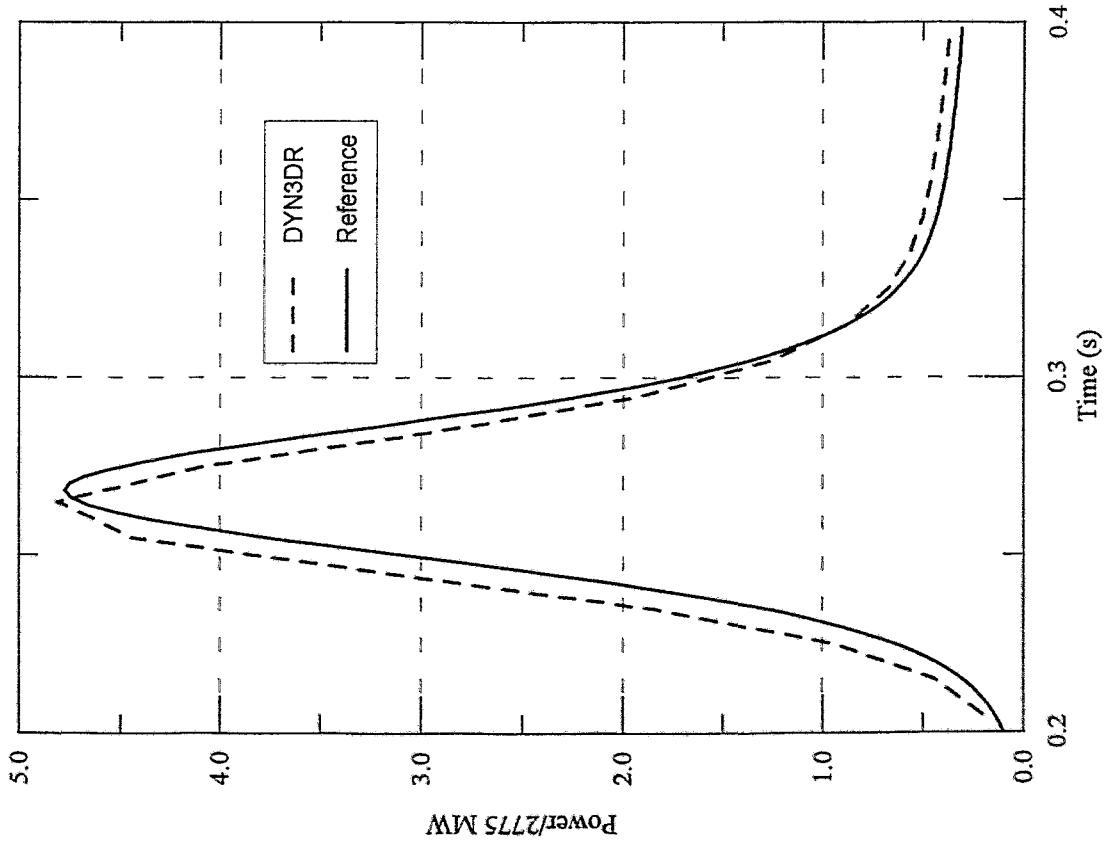


Fig. 3.C1.1: Nuclear Power

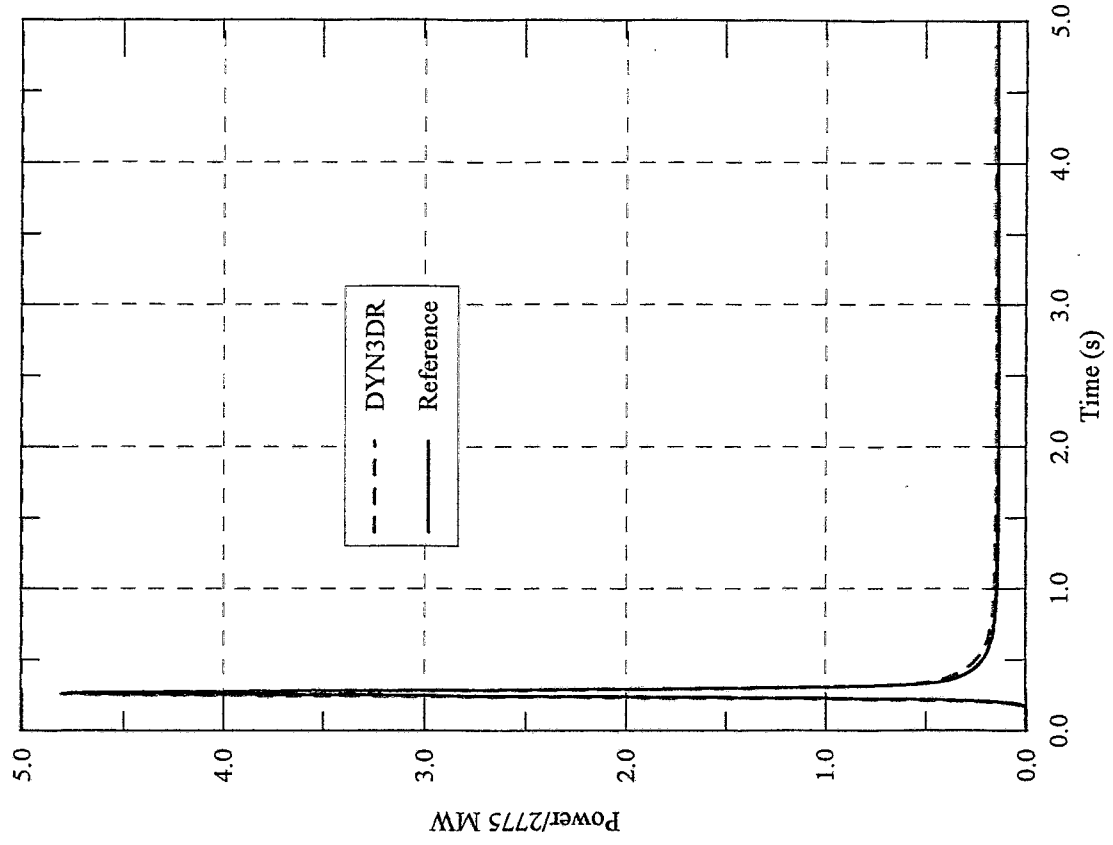


Fig. 3.C1.2: Nuclear Power

PWR Case C1: Ejection of a Peripheral Rod at HZP

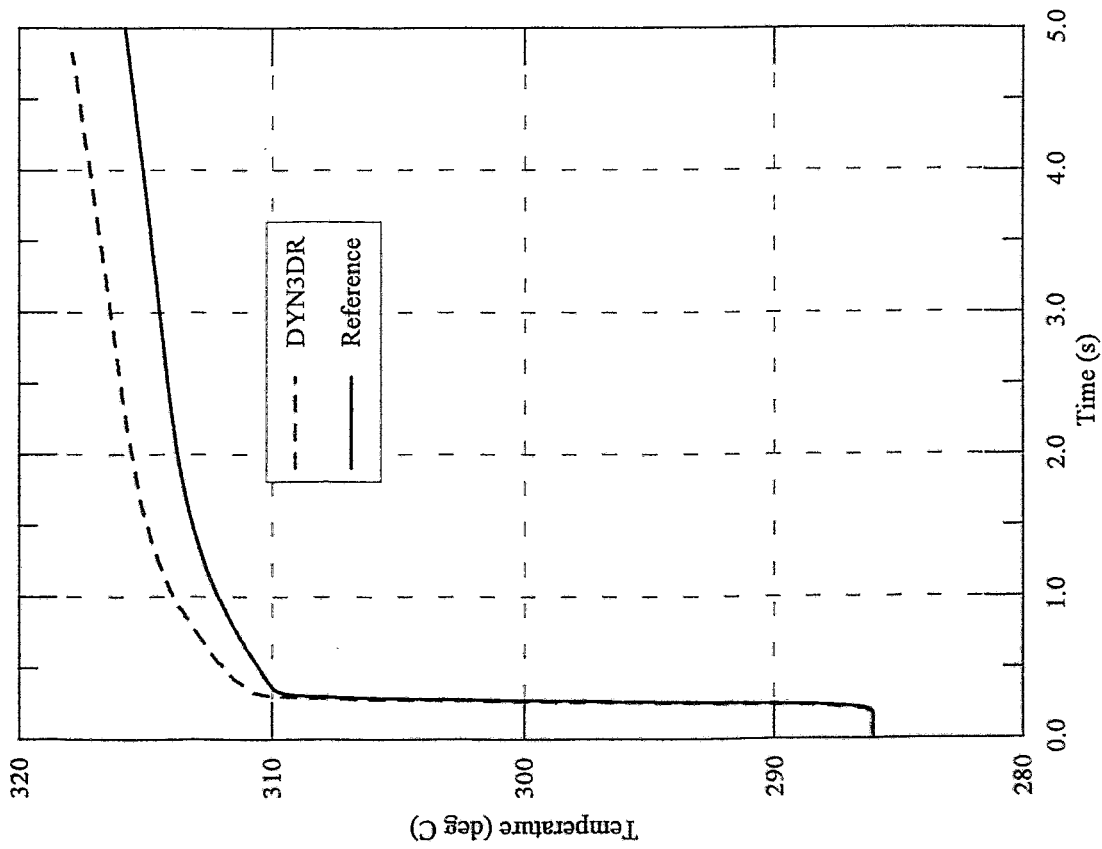


Fig. 3.C1.3: Core Averaged Doppler Temperature

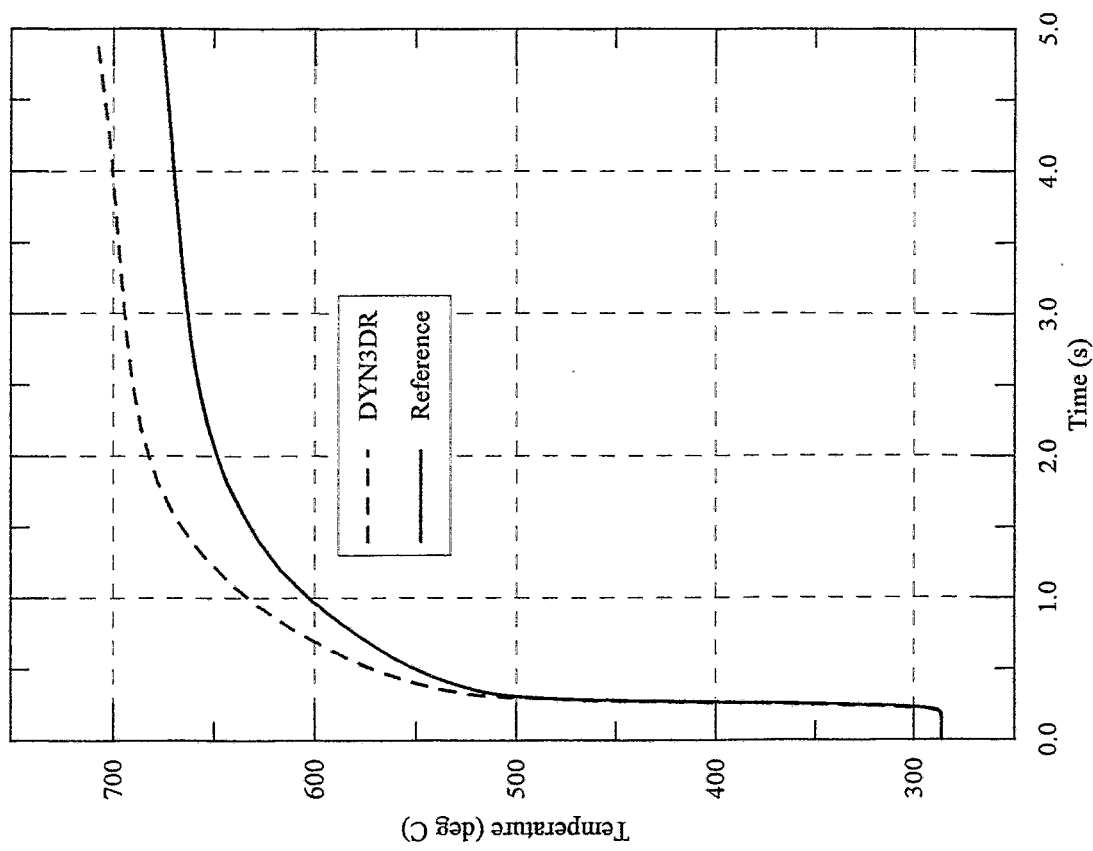


Fig. 3.C1.4: Maximum Fuel Centerline Temperature

PWR Case C1: Ejection of a Peripheral Rod at HZP

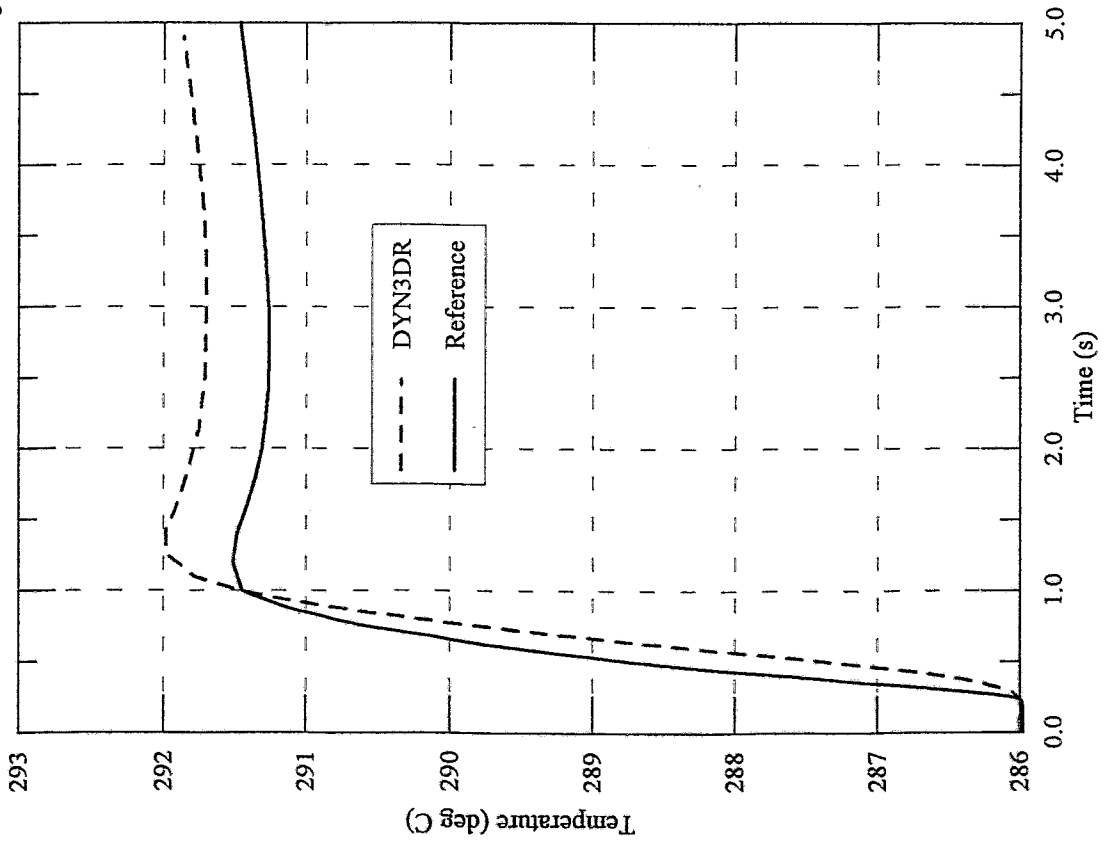


Fig. 3.C1.5: Coolant Exit Temperature

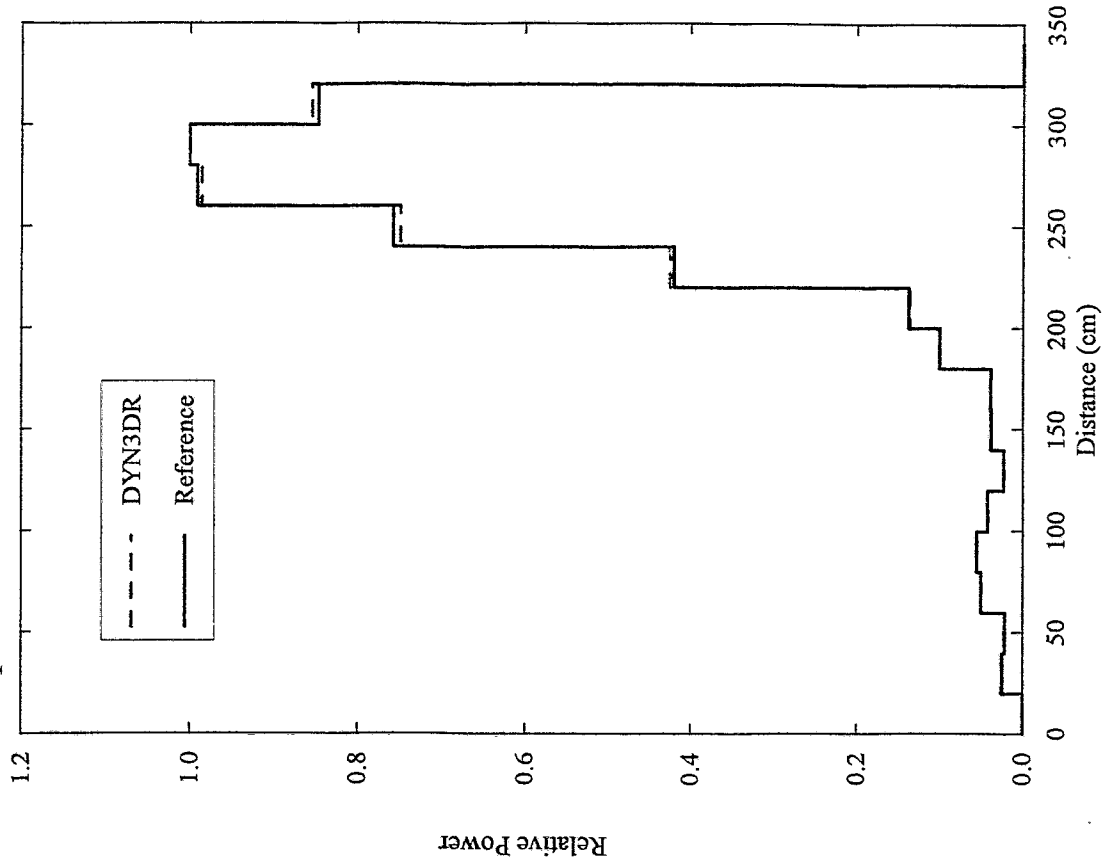


Fig. 3.C1.6: Radial Power Distribution at Time of Power Maximum in Axial Layer 13 (Along the Horizontal Traverse)

PWR Case C2: Ejection of a Peripheral Rod at FP

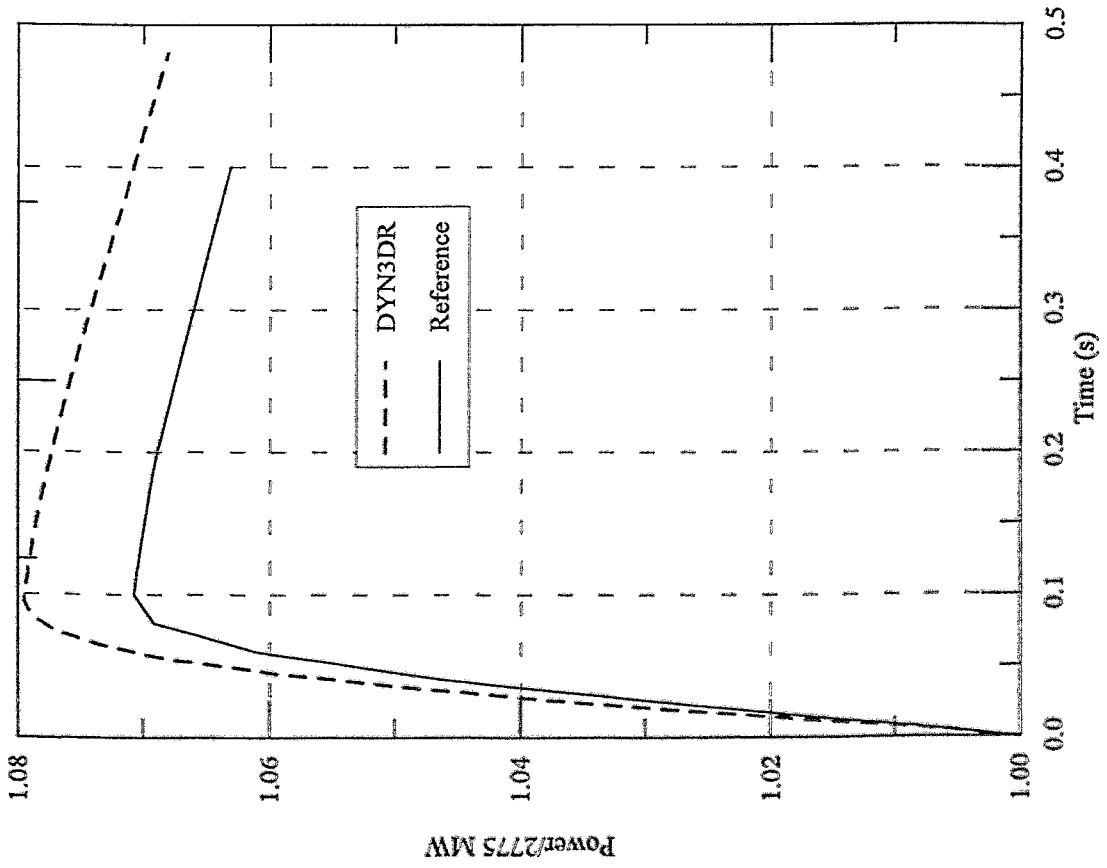


Fig. 3.C2.1: Nuclear Power

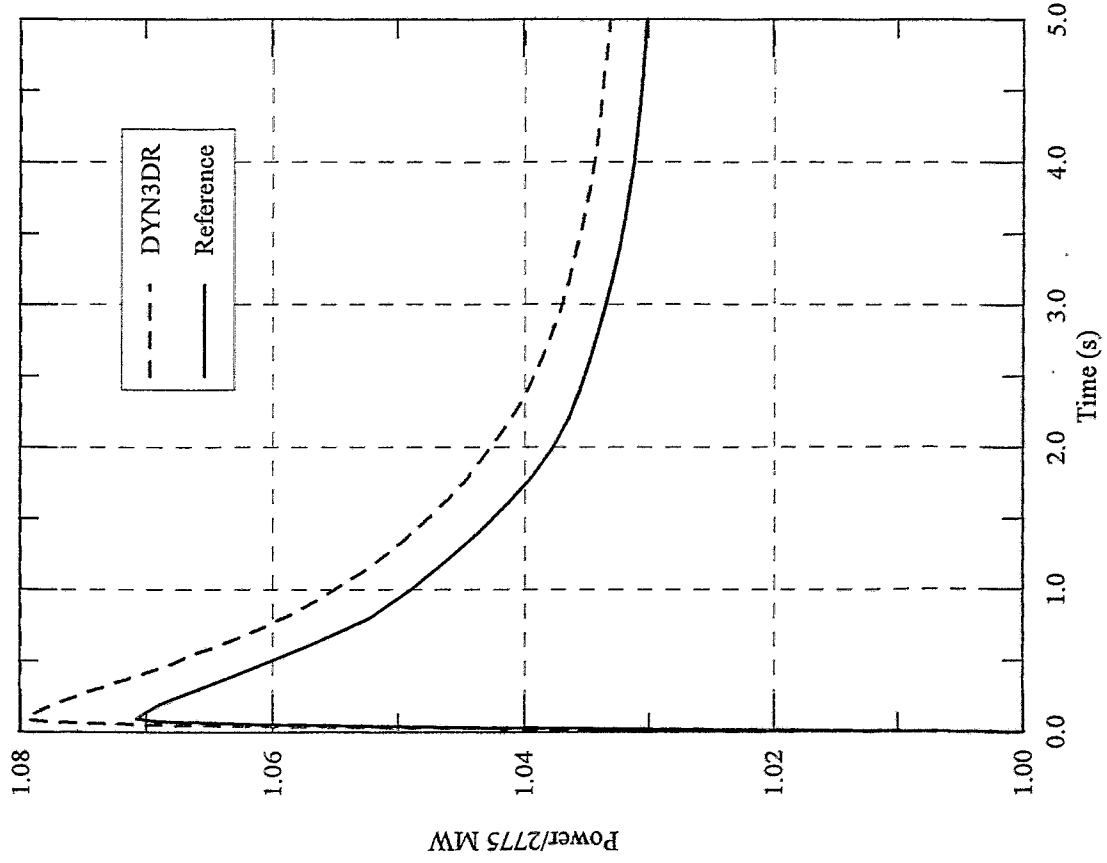


Fig. 3.C2.2: Nuclear Power

PWR Case C2: Ejection of a Peripheral Rod at FP

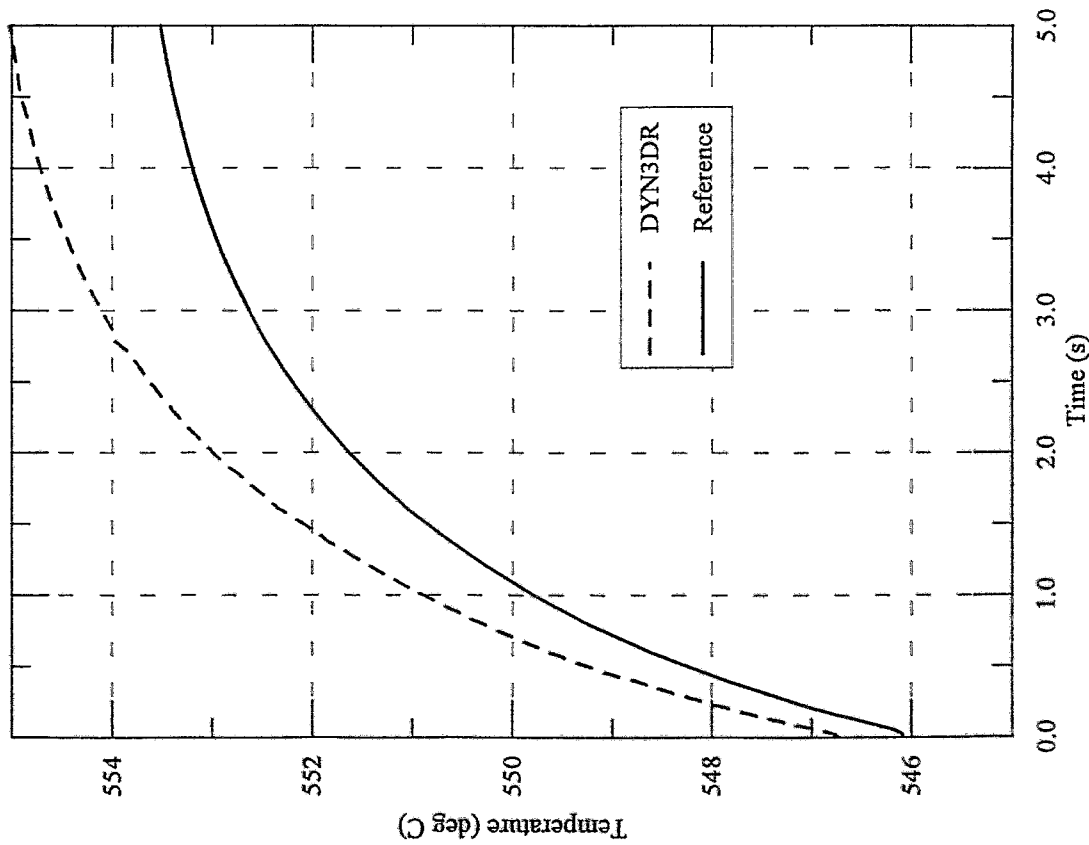


Fig. 3.C2.3: Core Averaged Doppler Temperature

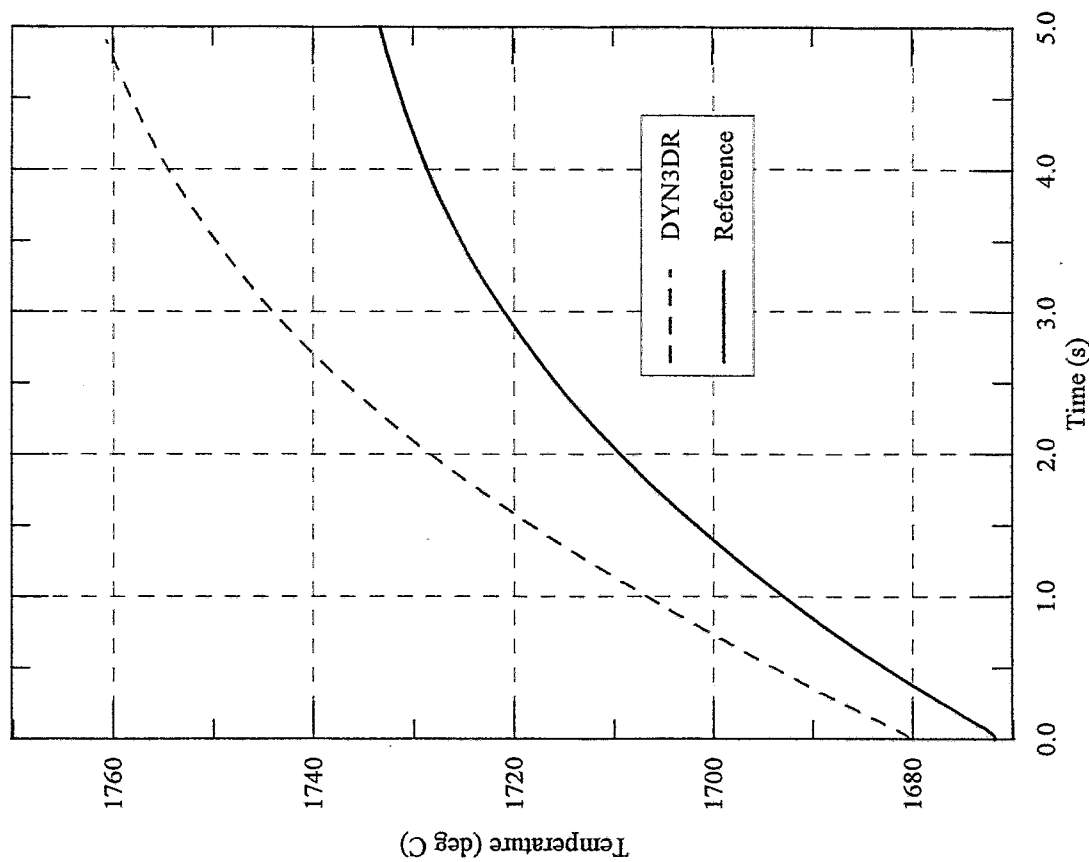


Fig. 3.C2.4: Maximum Fuel Centerline Temperature

PWR Case C2: Ejection of a Peripheral Rod at FP

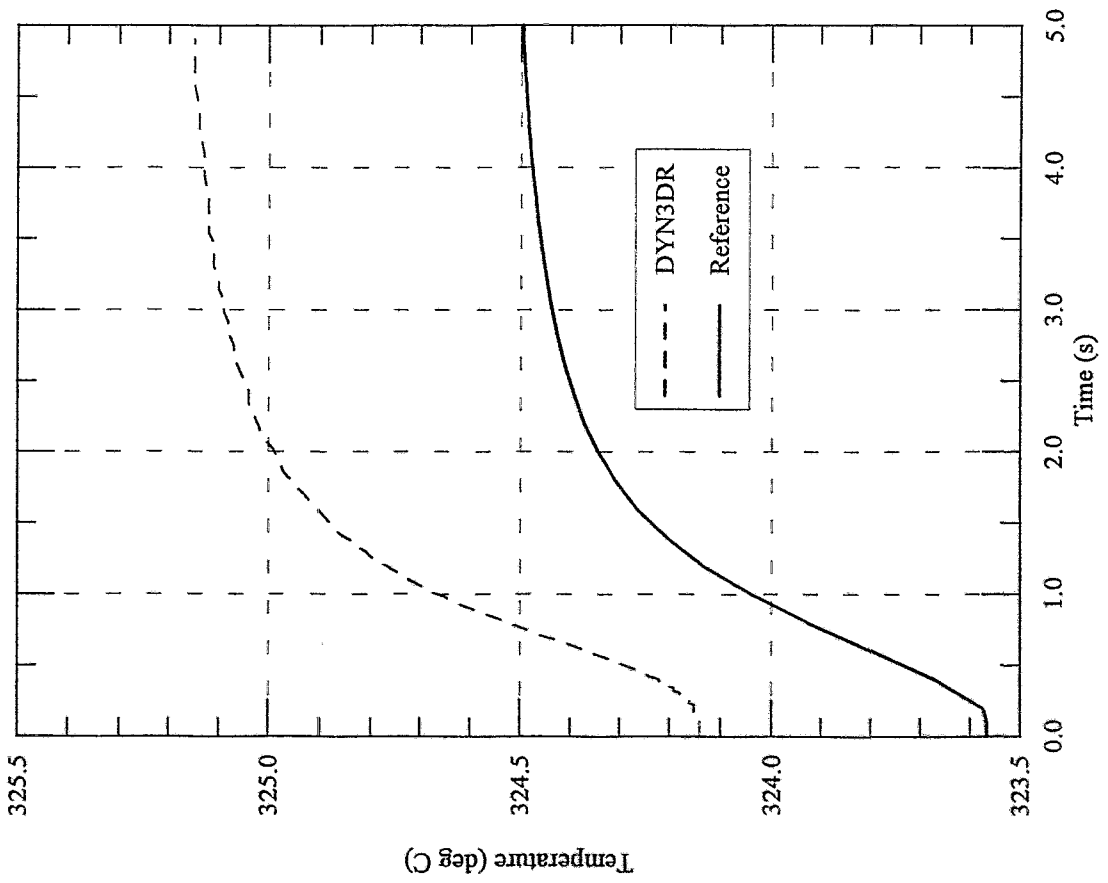


Fig. 3.C2.5: Coolant Exit Temperature

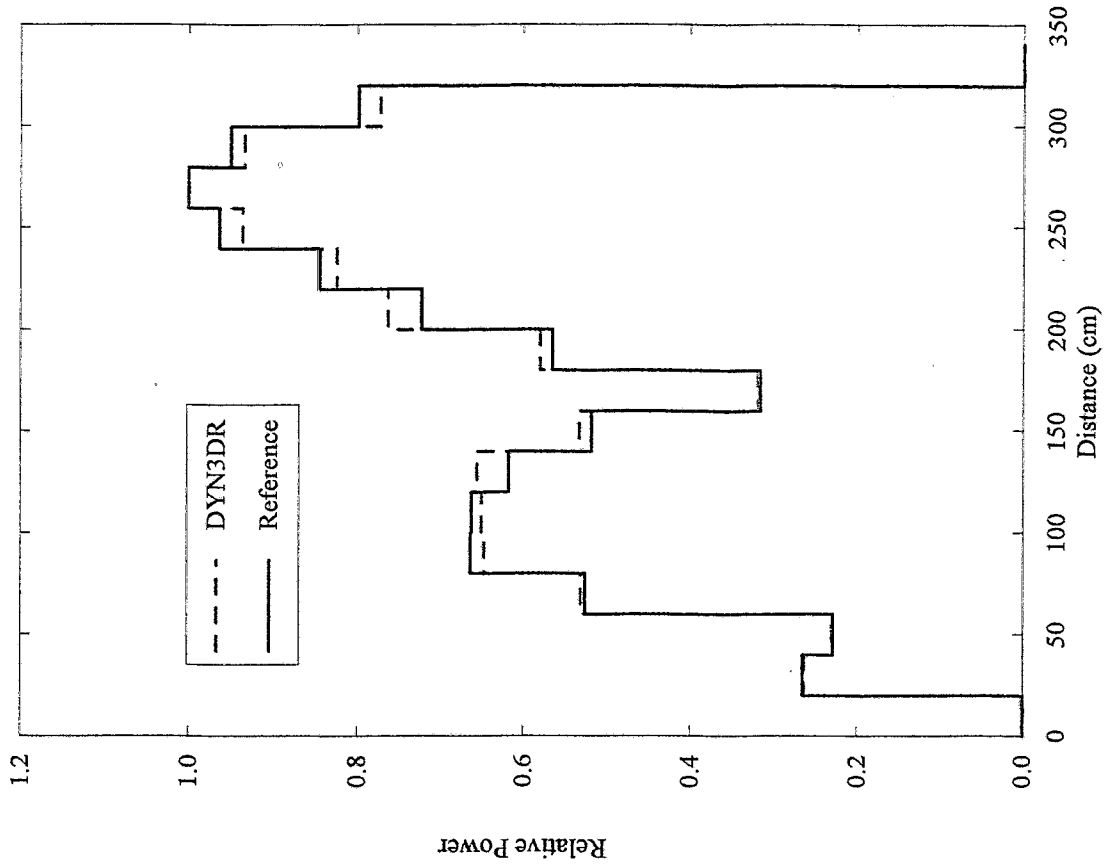


Fig. 3.C2.6: Radial Power Distribution at Time of Power Maximum in Axial Layer 13 (Along the Horizontal Traverse)

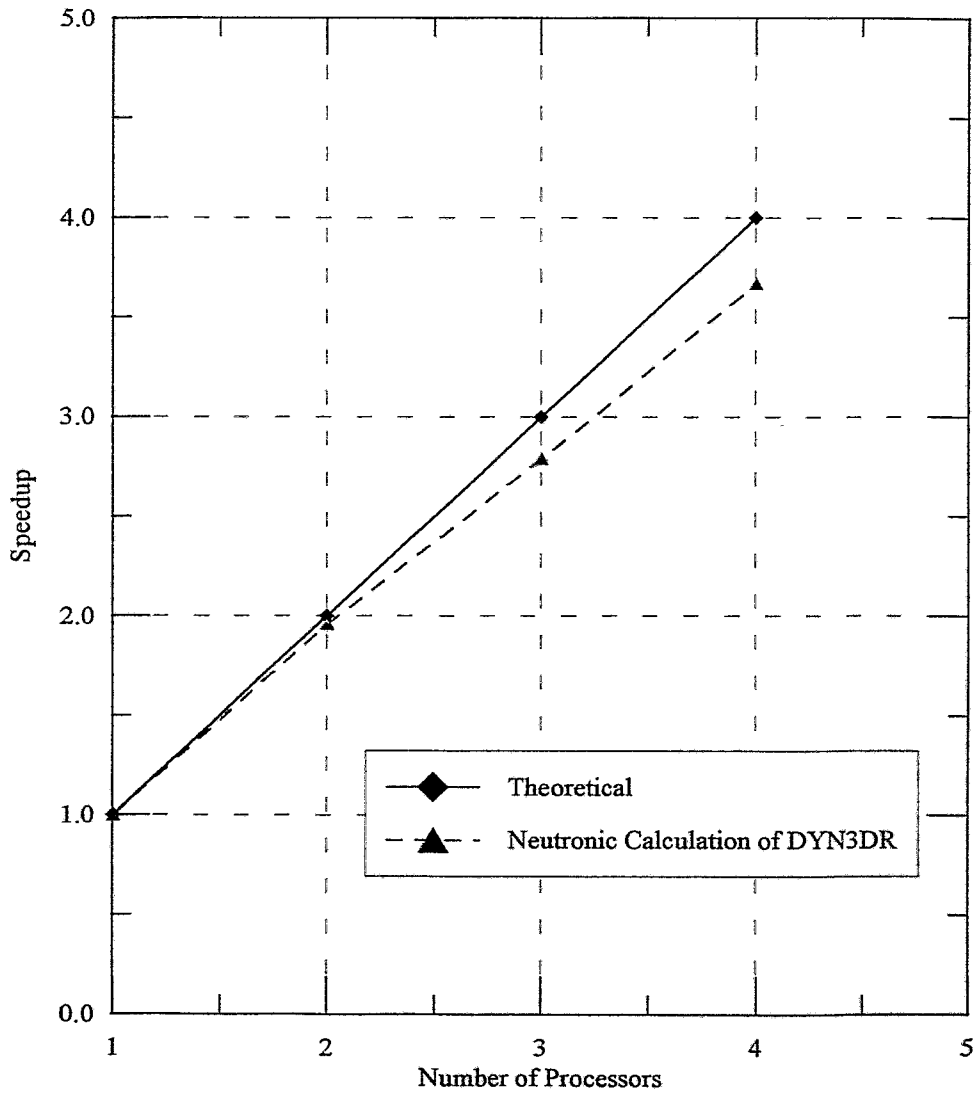


Fig. 4.1: Speedup of Steady State Neutronic Calculation in DYN3DR Using Different Number of Processors