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# RASD2019

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Recent Advances in Structural  
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Lyon, France

## **STRUCTURAL DAMPING BY LAYERS OF FIBROUS MEDIA ON A PERIODICALLY- CONSTRAINED VIBRATING PANEL**

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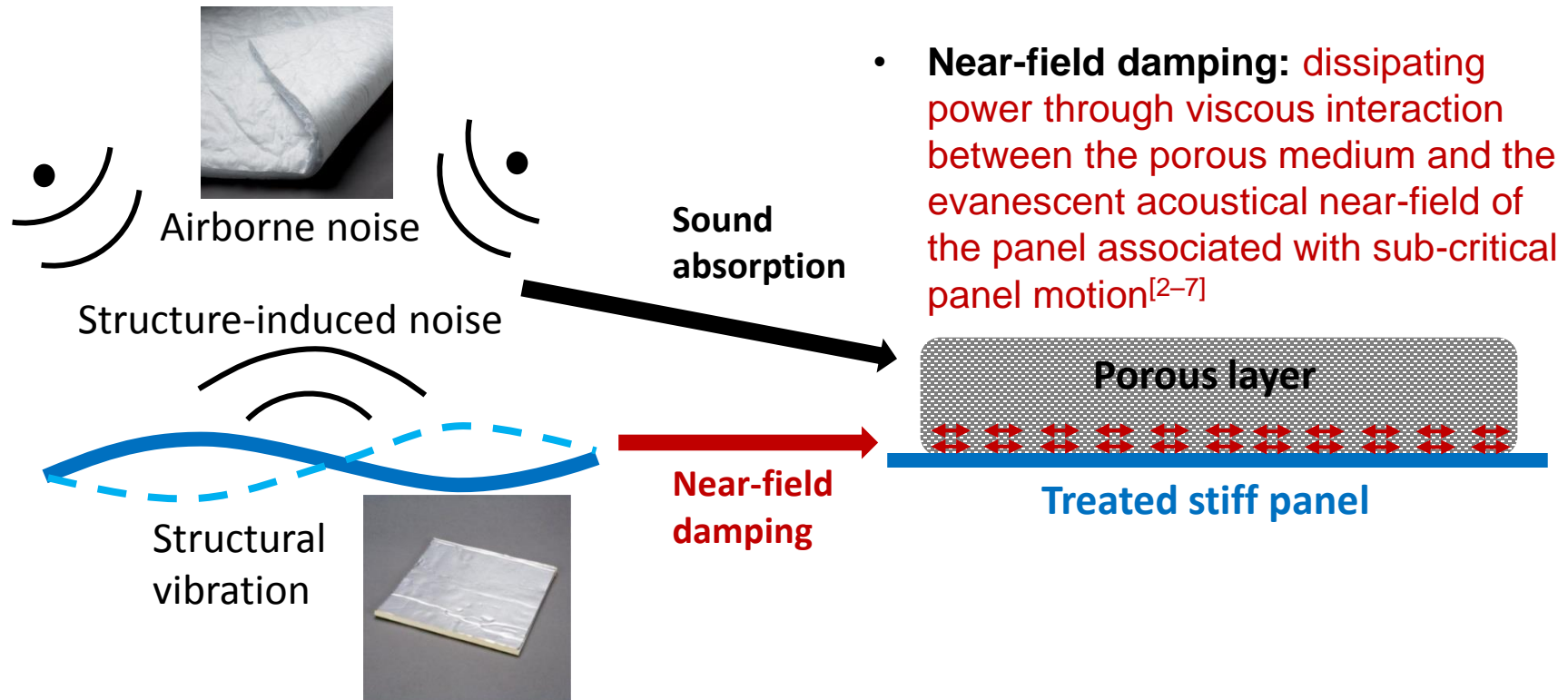
This presentation will be posted at Herrick E-Pubs: <http://docs.lib.purdue.edu/herrick/>  
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<https://iopscience.iop.org/article/10.1088/1742-6596/1264/1/012043/>



# Challenge

- **Advanced Noise Control Materials<sup>[1]</sup>**
  - **What's important about a noise control material?**
    - **Cost**
    - **Safety**
    - **Weight**
    - **Volume**
    - **Recyclability**
    - ...
    - ...
    - **Acoustical Performance**

# Objective: Multifunctionality



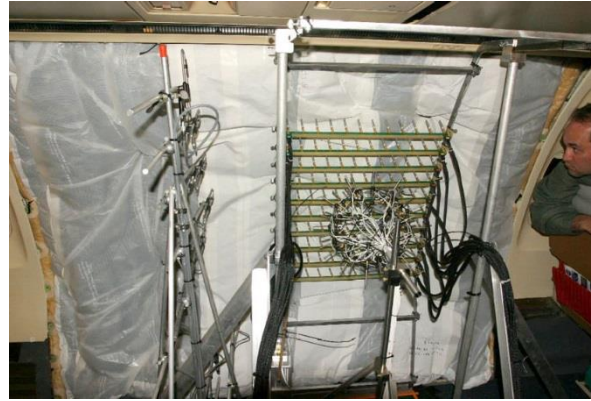
- **Objectives:** modeling, predicting and optimizing the near-field damping performance of conventional sound absorbing materials (fiber, foam, etc.), so that a properly-designed porous layer can achieve both structural damping and sound absorption at the same time → **save weight and cost**

# Main Idea

- Periodic Structure in Practical Scenarios



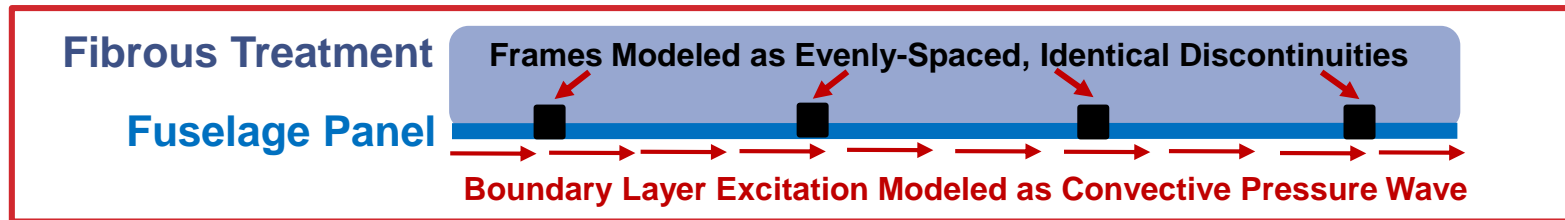
Nadeau et al., *J. of Aircraft* (1999)



Klos et al., *Noise-Con* (2005)



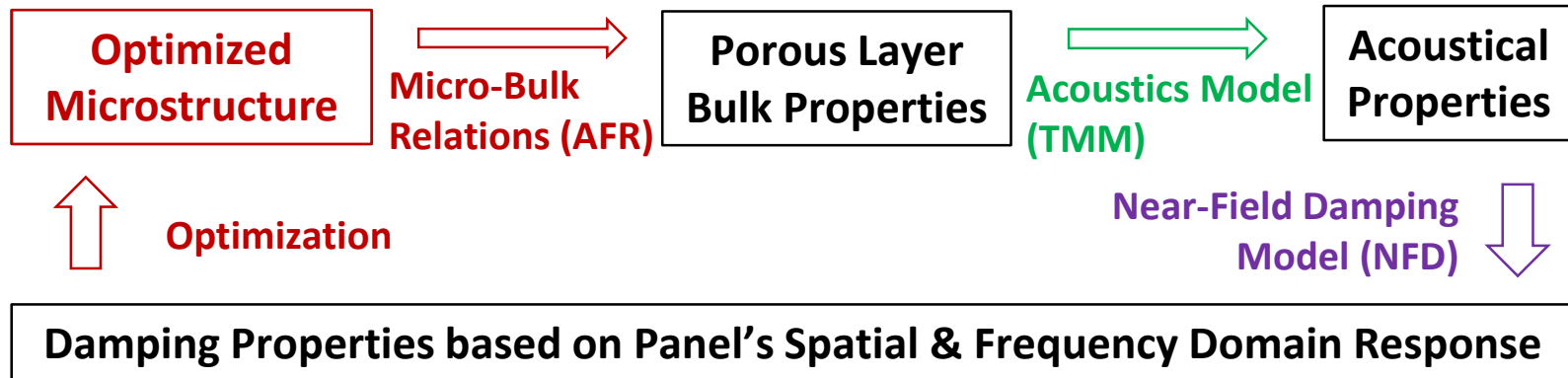
Room 3057, Herrick Labs



- Model the periodic structure, evaluate and optimize the damping of the treatment

# General Approach

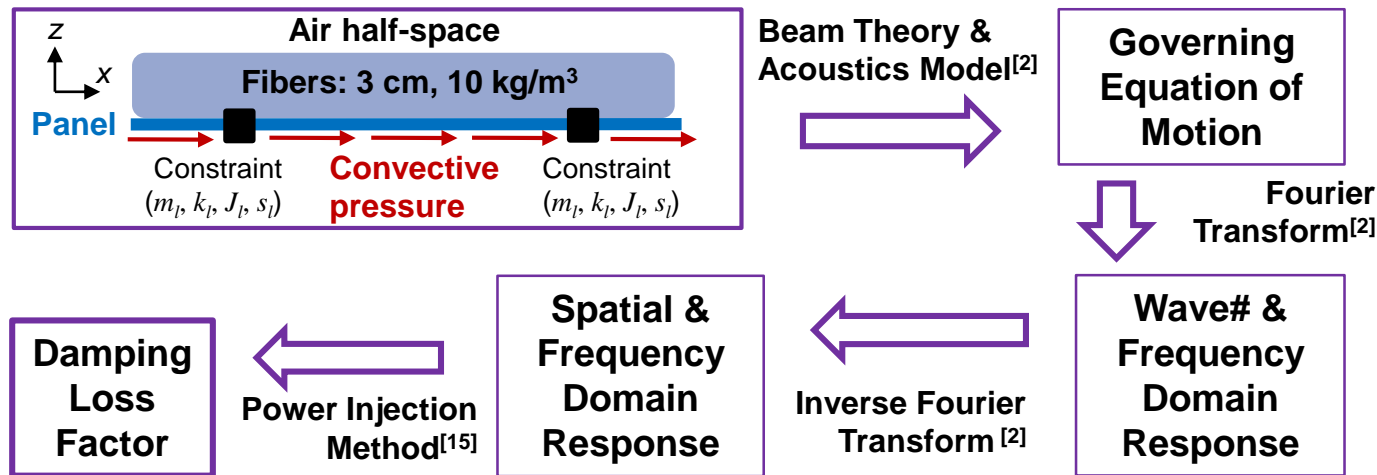
- Connecting damping material's properties and performance



- AFR:** micro-bulk relations for porous media made of fibers<sup>[8]</sup>
- TMM:** bulk-acoustical relations<sup>[9,10]</sup> including Johnson-Champoux-Allard (JCA) model<sup>[11]</sup>, Biot theory<sup>[11–15]</sup> and B.C.s interpretation<sup>[13,16]</sup>
- NFD:** acoustical-damping relations including Euler-Bernoulli beam theory, wavenumber-space Fourier transform<sup>[17]</sup> and power analysis<sup>[18]</sup>
- **TMM + NFD + AFR** provides an micro-damping model to maximize fibrous media's damping performance by optimizing their microstructures

# NFD Modeling

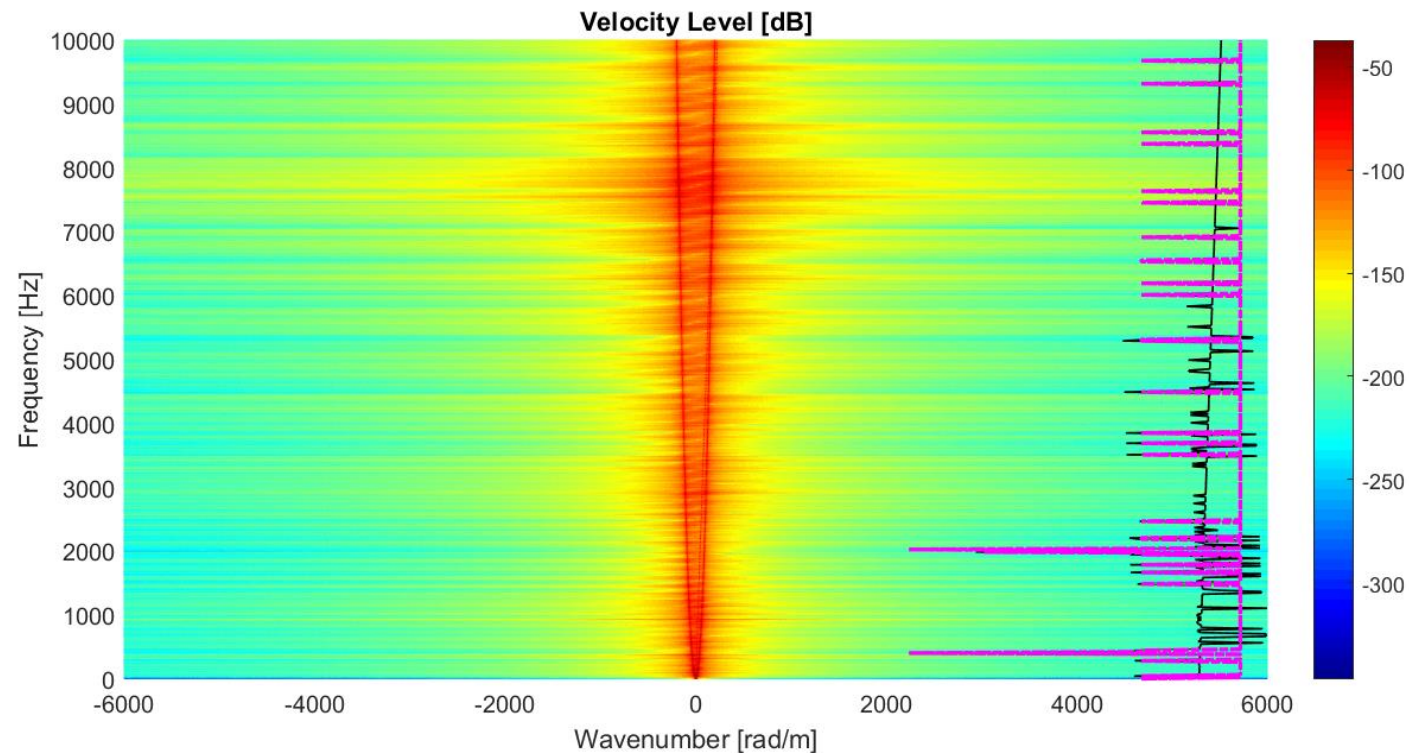
- Fibrous layer is assumed to be **limp**, 3 cm thick with bulk density 10 kg/m<sup>3</sup>
- Fiber solid material densities:  $\rho_{glass} = 2730 \text{ kg/m}^3$ ,  $\rho_{polymer} = 910 \text{ kg/m}^3$



**Governing Equation**

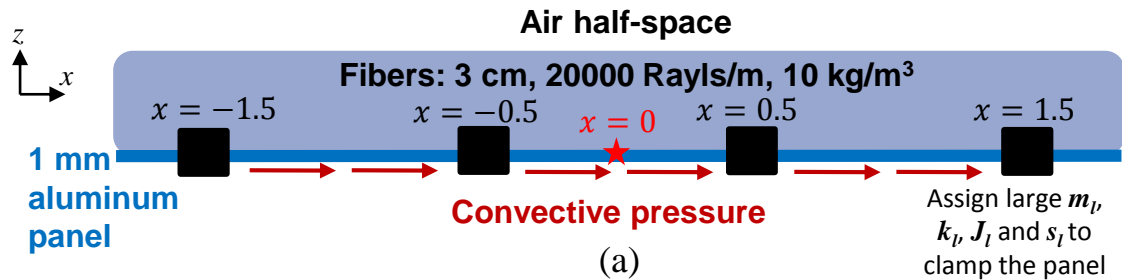
$$D \frac{\partial^4 w(x, t)}{\partial x^4} + m_s \frac{\partial^2 w(x, t)}{\partial t^2} = -p_1(x, t) + \underbrace{F e^{i\omega t} e^{-ik_v x}}_{\text{Convective pressure}} + \underbrace{\sum_{i=1}^{N_l} F_{l,i} \delta(x - x_{l,i}) + \sum_{i=1}^{N_l} M_{l,i} \delta(x - x_{l,i})}_{\text{Reaction forces due to constraints}}$$

# IDFT Sampling Parameters

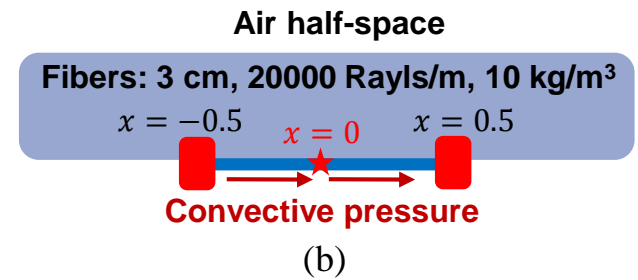




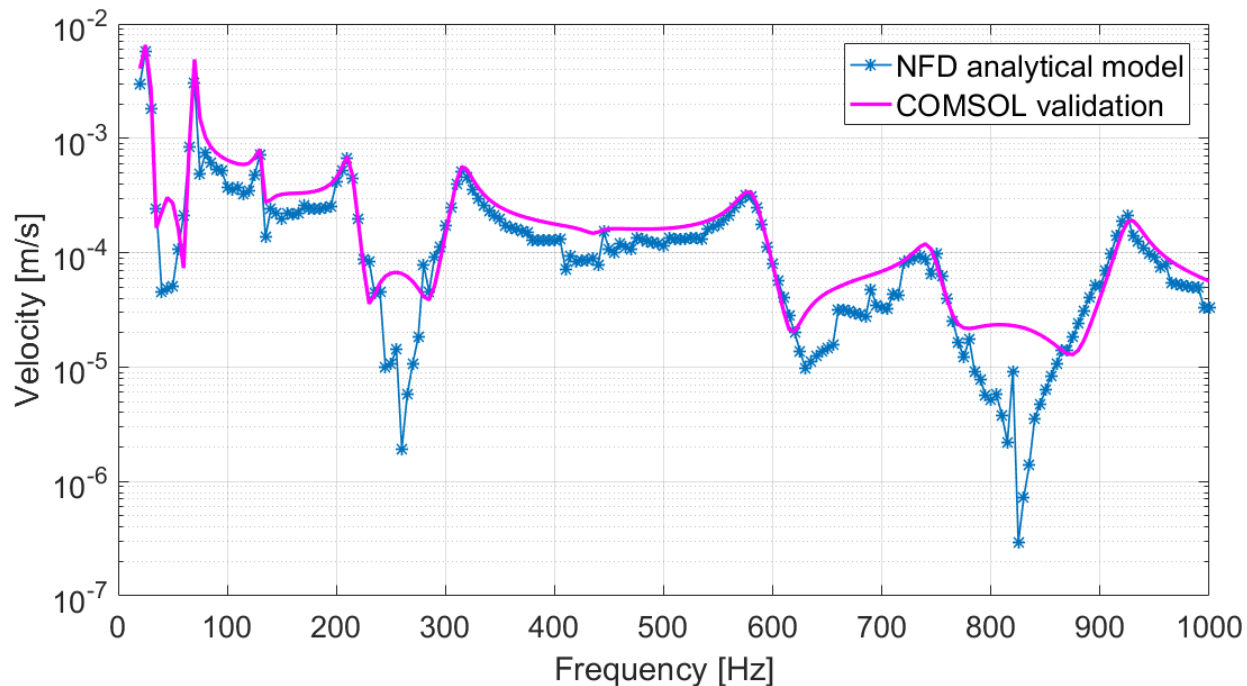
# Model Validation



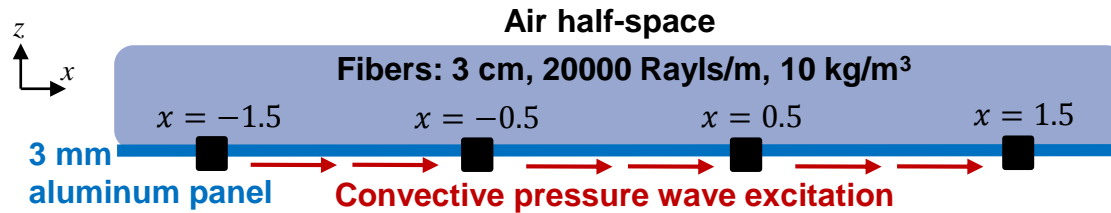
NFD model



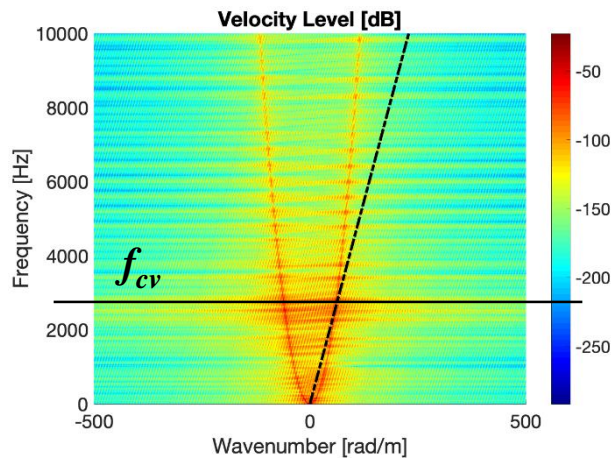
COMSOL finite element model



# Fuselage Structure Wavenumber Response

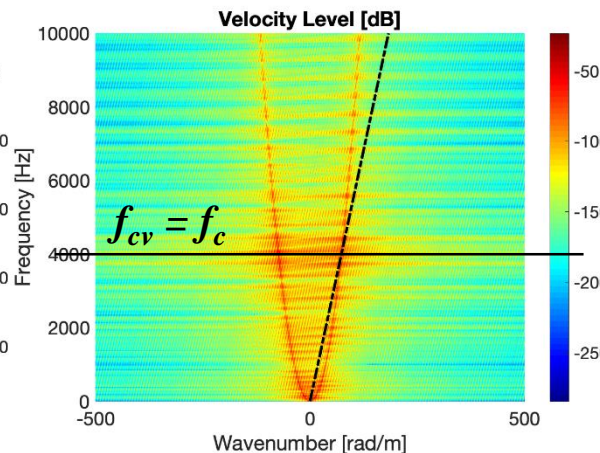


Identical aluminum constraints with size 5 cm x 5 cm ( $m_l = 6.75$  kg/m,  $k_l = 2.66 \times 10$  kg/(s<sup>2</sup>·m),  $J_l = 0.0028$  kg·m,  $s_l = 1.11 \times 10$  kg·m/(s<sup>2</sup>)) are evenly distributed along the panel with separation of 1 m



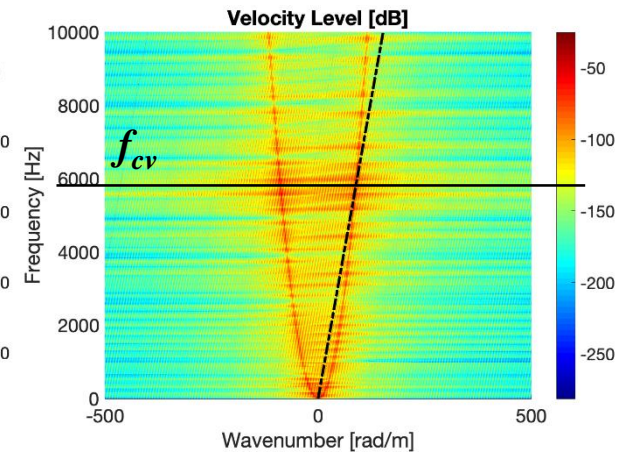
(a)

Ma=0.8  
(Subsonic)



(b)

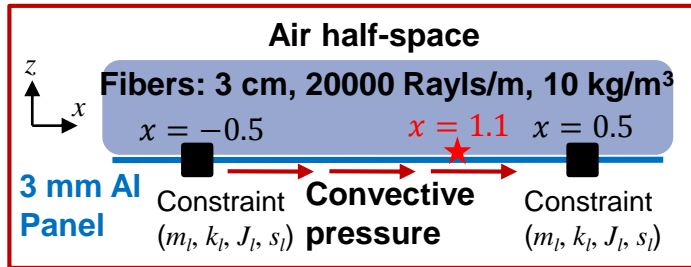
Ma=1.0  
(Sonic)



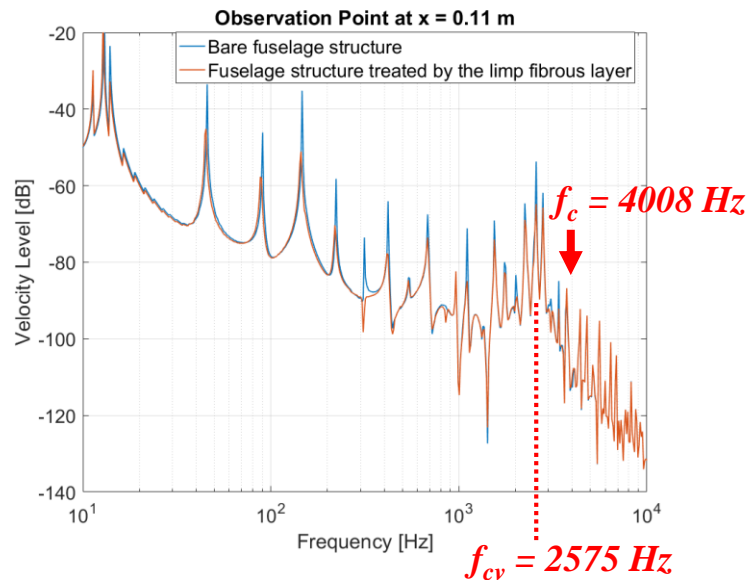
(c)

Ma=1.2  
(Supersonic)

# Fuselage Structure Velocity Response Spectrum

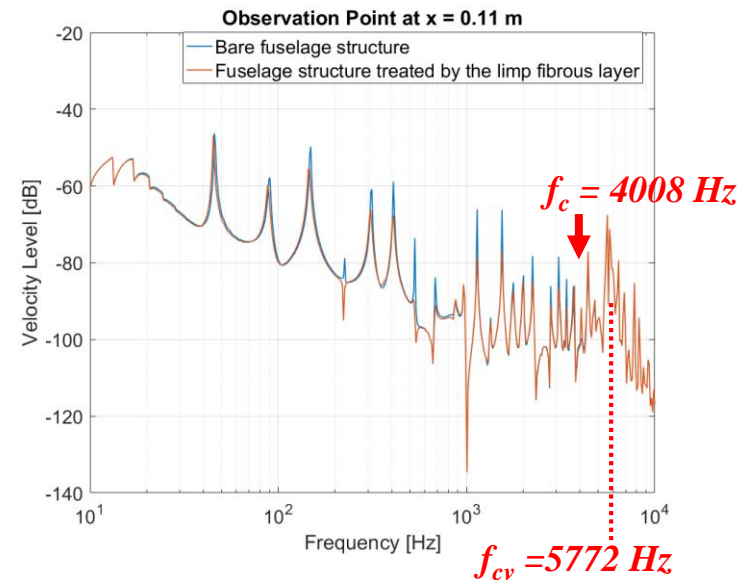


- Observation at  $x = 0.11$  m
- Vibration peaks below  $f_c$  were reduced by 5–15 dB by the fibrous layer



(a)

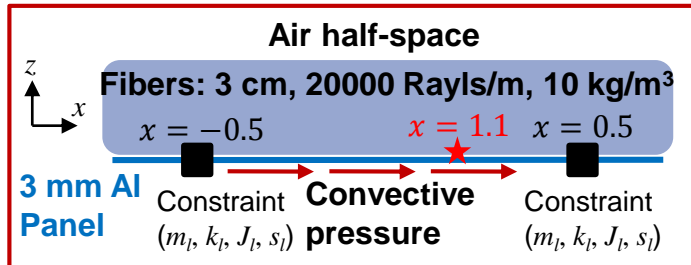
Ma=0.8



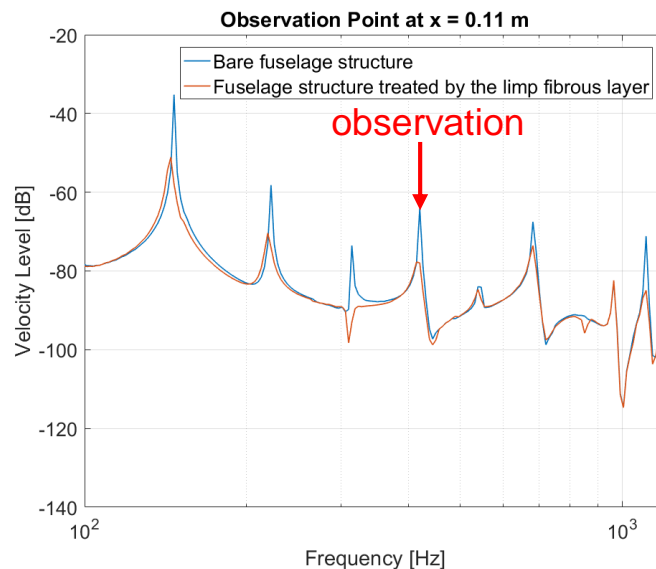
(b)

Ma=1.2

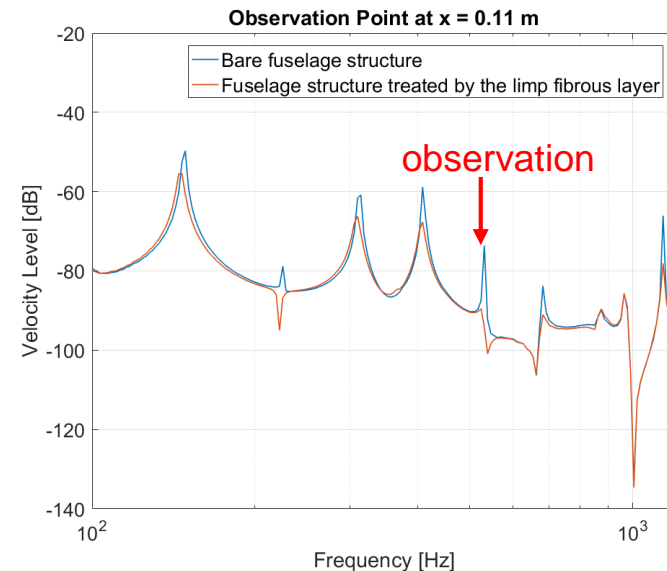
# Fuselage Structure Velocity Response Spectrum



- Zoom into **100–1000 Hz** to see the peak reduction
- Choose the peak at certain frequency as an observation point

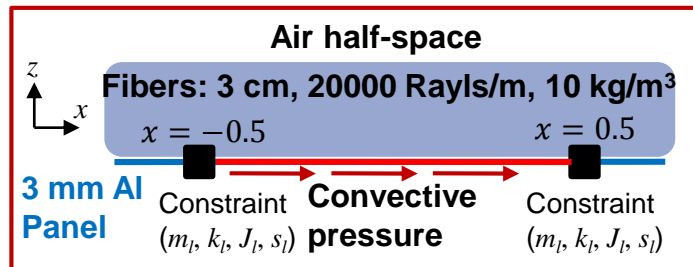


(a)  
 $Ma=0.8$

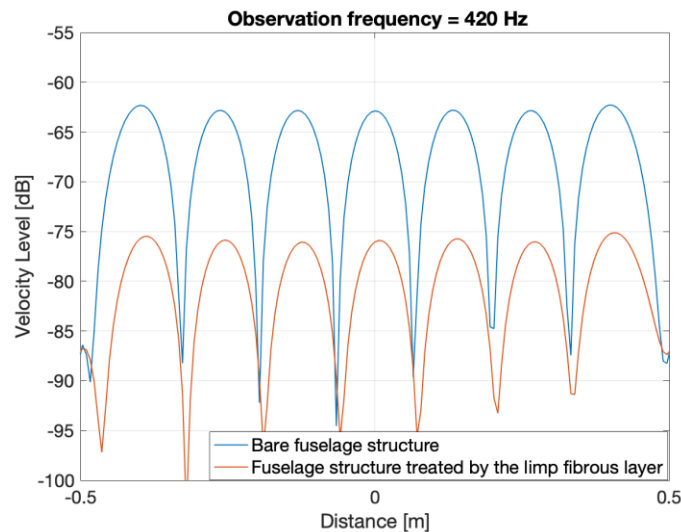


(b)  
 $Ma=1.2$

# Fuselage Structure Spatial Response

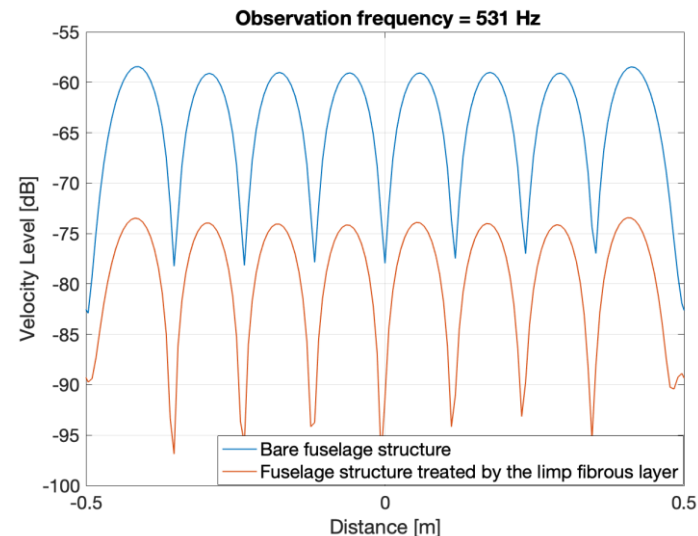


- About 15 dB reduction of vibration can be achieved at modal peak frequency  $f = 420$  Hz (Ma=0.8) or  $f = 531$  Hz (Ma=1.2)



(a)

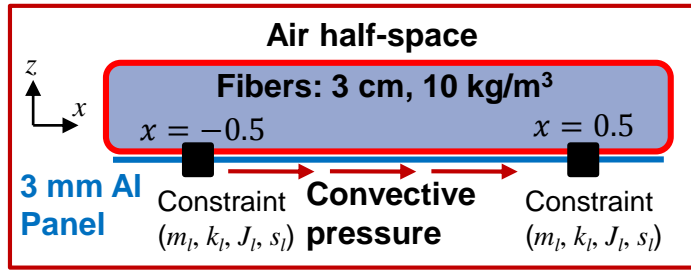
Ma=0.8



(b)

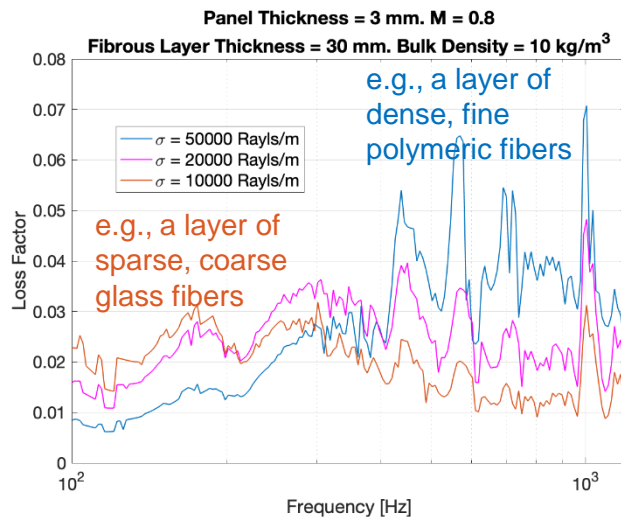
Ma=1.2

# Fuselage Treatment Damping Loss Factor



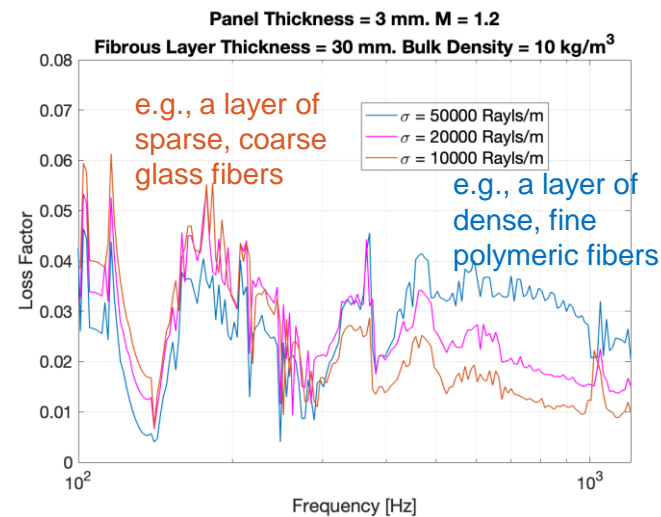
Power injection method<sup>[16]</sup>

- Fibrous layer with lower airflow resistivity is better at reducing lower frequency vibration
- Idea:** optimal bulk/micro structure can be found to achieve optimal damping at certain frequency for a given structure



(a)

$Ma = 0.8$



(b)

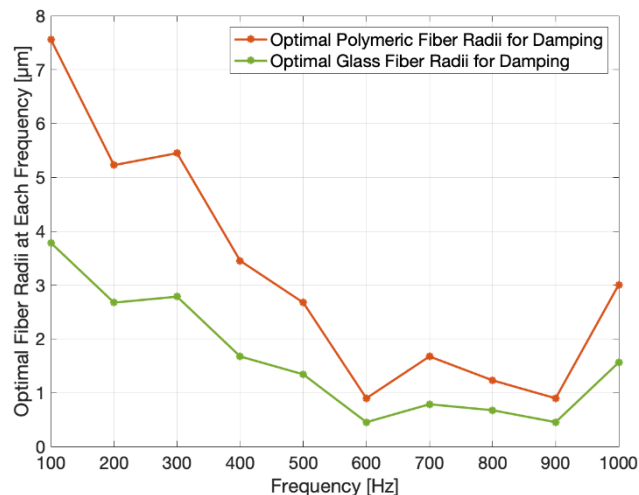
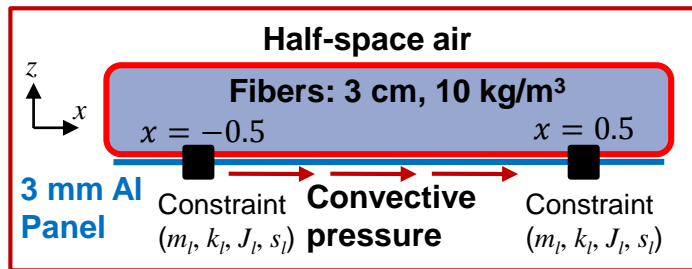
$Ma = 1.2$



# Glass Fibers vs. Polymeric Fibers

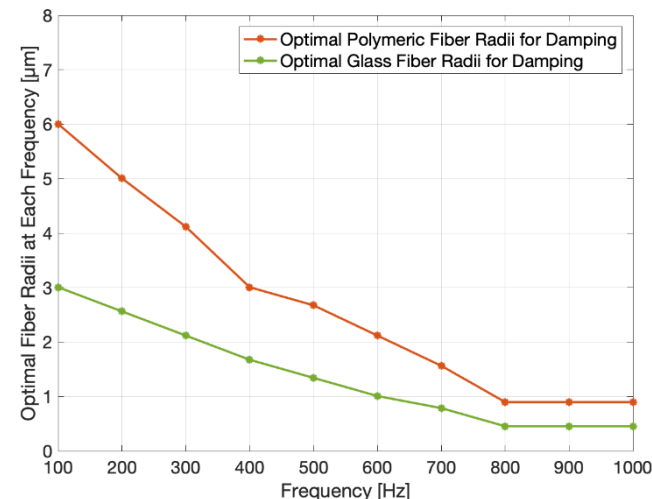
– finding optimal fiber radius

- Larger fiber is better at reducing lower frequency vibration
- Heavy (e.g., glass) fiber would need a smaller fiber size than light (e.g., polymeric) fiber to achieve optimal damping performance



(a)

Ma=0.8



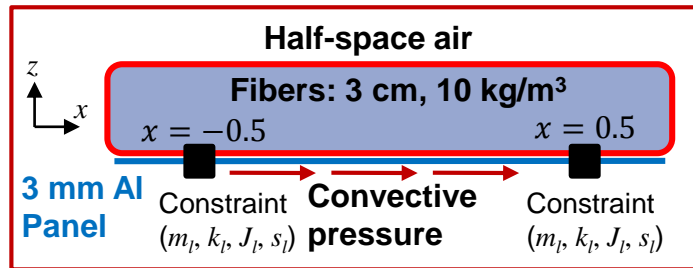
(b)

Ma=1.2

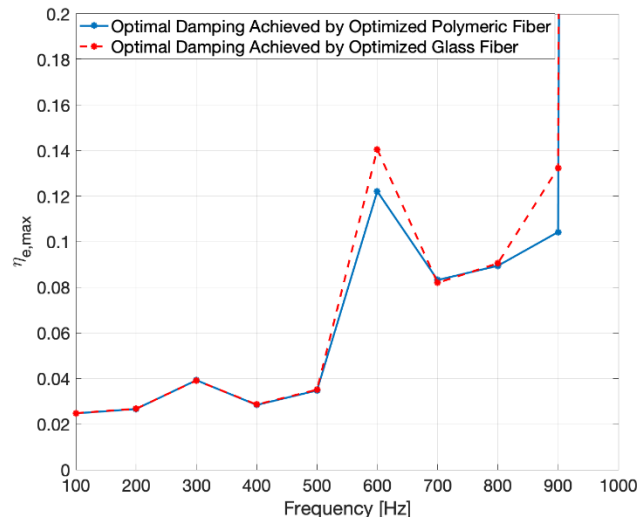
\* Fiber size was optimized to achieve the largest damping for certain frequency & panel of interests

# Glass Fibers vs. Polymeric Fibers

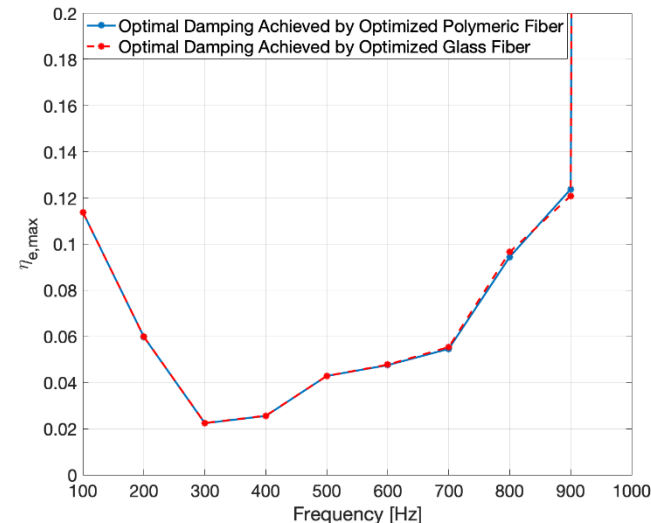
## – optimal damping loss factor



- Optimized fibrous layer made of either polymeric or glass can achieve equivalent optimal damping



(a)  
Ma=0.8



(b)  
Ma=1.2

\* Fiber size was optimized to achieve the largest damping for certain frequency & panel of interests



# CONCLUSIONS

## ❖ Lightweight fibrous damper parametric study on microscopic properties

- Relatively large fibers are effective at damping low frequency vibration
- Heavy (e.g., glass) fiber would need a smaller fiber size than light (e.g., polymeric) fiber to achieve equivalent optimal damping performance
- Significant levels of damping can be achieved by properly designed fibrous treatment → multifunctional (absorbing & damping) fibrous layer saves weight, space and cost

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