

# Sliding Mode Controller Applied to Coupled Inductor Dual Boost Inverter

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## Abstract

A coupled inductor-dual boost-inverter (CIDBI) with differential structure has been presented to be applied to micro-inverter photovoltaic module system because of its turn ratio of high-voltage level. However, it is hard for CIDBI converter with conventional PI regulator to be designed stable and achieve good dynamic performance, given the fact that it is a high order system. In view of this situation, a sliding mode control (SMC) strategy is introduced in this paper, and two different sliding mode controllers (SMCs) are proposed and adopted in the left and right side of the two Boost sub-circuits respectively to implement corresponding regulation of voltage and current. The schemes of the SMCs have been elaborated in this paper including the establishment of the system variable structure model, the selection of the sliding surface, the determination of the control law, and the presentation of the reaching conditions and sliding domain. Finally, the mathematic analysis and the proposed SMC are verified by the experimental results.

**Key words:** Coupled inductor, Micro inverter, Photovoltaic (PV) power generation, Sliding mode control

## I. INTRODUCTION

Grid-connected micro-inverter is an effective way to solve the hot spot effect of photovoltaic modules and improve the efficiency of photovoltaic power generation. There are two main grid-connected power generation methods for low-voltage photovoltaic modules. 1) Single-stage inverters are connected to the grid by line transformer boost mode. 2) Two-stage inverter consist of front-stage DC-DC boost circuit and back-stage inverter. In the first method, when the voltage of photovoltaic module is low, the turn ratio of line transformer is large, and hence the power loss of transformer is large. In the second method, two-stage circuit is adopted, and the conversion efficiency of the system is not high. It can be seen that these two methods are not conducive to the efficient generation of small and medium power photovoltaic cells to the grid.

The CIDBI that proposed in work [1] can meet the

grid-connected, that is, to realize one-stage transformerless grid-connected.

The differential CIDBI, as shown in Fig.1 presented in work [1], is composed of the two coupled inductor boost circuits that generates an ac output voltage larger than dc input, depending on the instantaneous duty cycle and turn ratio of the coupled inductor as well, and uses only one power processing stage. Fig.1 shows that there is one DC source  $B_1$ ,  $L_1$  and  $L_2$  commonly share the same magnetic core, while  $L_3$  and  $L_4$  use another, so the two coupled inductors are constructed. One boost circuit consists of  $L_1$ ,  $L_2$ ,  $C_2$ , switches  $T_1$ ,  $T_2$  and their body diodes  $D_1$ ,  $D_2$ , the other boost circuit consists of  $L_3$ ,  $L_4$ ,  $C_3$ , switches  $T_3$ ,  $T_4$  and their body diodes  $D_3$ ,  $D_4$ . It is clear that the right-side circuit and the left one represent boost converters with coupled inductor. The work [1] tells us that CIDBI can be

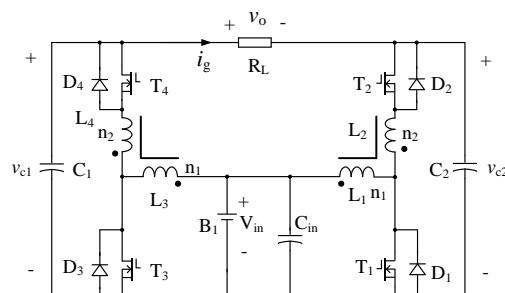


Fig. 1. Main circuit of CIDBI.

employed as micro inverter integrated with much lower voltage PV array and even achieve high energy conversion

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requirements of voltage boost of photovoltaic modules and AC



$v_{c2}^*$  can be achieved by controlling  $T_1$  and  $T_2$  in the right-side boost circuit, and the pulse control signals  $v_{dr1}$  and  $v_{dr2}$  should be complementary. In light of switching states, the right side of the converter can be equivalent to two sub-circuits as shown in the Fig.4. And the left side of the CIDBI can be considered as the current source. So the calculation formula of duty cycle ratio  $d_R(t)$  for the right side is as follows.

$$d_R(t) = \frac{v_{c2}^*(t) - V_{in}}{v_{c2}^*(t) + N \cdot V_{in}} = \frac{V_m^* \sin(\omega t + \pi) + V_{dc}^* - V_{in}}{V_m^* \sin(\omega t + \pi) + V_{dc}^* + N \cdot V_{in}} \quad (7)$$

Where,  $N$  is the turn ratio of the coupled inductor (seen in the Fig.1:  $n_2/n_1$ ).

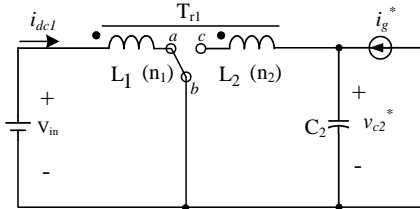
The average power of the output should be equal to that of the input according to the instantaneous power theory. So the equation (8) can be gotten.

$$V_m \cdot i_{dc1}^*(t) = C_2 \cdot v_{c2}^*(t) \cdot \frac{d(v_{c2}^*(t))}{dt} - v_{c2}^*(t) i_g^*(t) \quad (8)$$

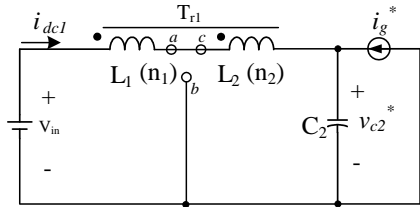
Substituting the expression (1), (2), (4) and (5) into the equation (8), the reference value  $i_{dc1}^*$  for dc input current can be expressed as (9).

$$i_{dc1}^*(t) = \frac{1}{V_{in}} \left[ \begin{array}{l} \frac{1}{2} V_m^* \cdot I_g^* (1 - \cos(2\omega t)) - V_{dc}^* \cdot I_g^* \sin(\omega t) \\ -\omega C_2 \cdot (V_{dc}^* \cdot V_m^* \cos(\omega t) - \frac{1}{2} (V_m^*)^2 \sin(2\omega t)) \end{array} \right] \quad (9)$$

The expression (9) describes the average input current  $i_{dc1}$  through the coupled inductor  $L_1/L_2$ . In fact, the input current contains the high-frequency ripple current that can be neglected because this ripple current has no effect on the effective value of current.



(a)  $T_1$  is on and  $T_2(D_2)$  off



(b)  $T_2(D_2)$  is on and  $T_1$  off

Fig.4. The two structures of the right-side converter.

The switching function can be defined in the expression(10), where,  $\gamma$  is the state variable.

$$\gamma = \begin{cases} 1 & \text{When } T_1 \text{ is ON and } T_2(D_2) \text{ off} \\ 0 & \text{When } T_1 \text{ is OFF and } T_2(D_2) \text{ on} \end{cases} \quad (10)$$

It is clear that the corresponding output energy will increase

when  $\gamma=1$  happens. In this state, the inductor  $L_1$  and the output capacitance  $C_2$  are decoupled and the current through  $L_1$  is increased linearly, i.e, the energy storage;  $i_g$  flows to  $C_2$  as shown in the Fig. 4(a). When  $\gamma=0$  occurs, the energy will be transferred to  $C_2$  by the coupled inductor  $L_1/L_2$  as shown in the Fig.4(b).

It can be seen from the Fig.4 that there are two error variables that are defined  $\varepsilon v_{c2}$  for output capacitance voltage and  $\varepsilon i_{dc1}$  for coupled inductor current respectively in the expression (11).

$$\begin{cases} \varepsilon v_{c2} = v_{c2} - v_{c2}^* \\ \varepsilon i_{dc1} = i_{dc1} - i_{dc1}^* \end{cases} \quad (11)$$

As a consequence, the state error equation of the right-side converter can be written as (12).

$$\dot{x} = Ax + B\gamma + D \quad (12)$$

Where,

$$x = [\varepsilon v_{c2}, \varepsilon i_{dc1}]^T \quad (13)$$

$$A = \begin{bmatrix} 0 & \frac{1}{C_2} \\ -\frac{1}{L_1 \cdot (1+N)} & 0 \end{bmatrix} \quad (14)$$

$$B = \begin{bmatrix} -\frac{i_{dc1}^*}{C_2} \\ \frac{v_{c2} + N \cdot V_m}{L_1 \cdot (1+N)} \end{bmatrix} \quad (15)$$

$$D = \begin{bmatrix} \frac{i_g^* + i_{dc1}^*}{C_2} \\ \frac{V_m - v_{c2}^*}{L_1 \cdot (1+N)} \end{bmatrix} \quad (16)$$

It is noted that the low-frequency reference functions, when compared to the switching frequency, can be considered as quasistatic, in such a way that the term of derived function for the reference is, in general, neglected in the equation (12).

### B. Left-Side Circuit

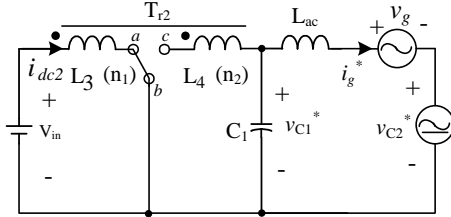
The left-side subsystem is looked as the current source and controlled to shape the grid current. Actually, the voltage of the  $L_{ac}$  can be regulated only if the voltage  $v_{c1}$  across the  $C_1$  is controlled, as the result that the grid current  $i_g$  is gotten. Additionally, since the inductor  $L_{ac}$  is designed to filter the high-frequency switching ripple, its steady-state low-frequency voltage is small, and also the voltage  $v_{c1}$  should be close to  $(v_{c2} + v_g)$ . So  $v_{c1}$  can be expressed as (17).

$$v_{c1}^*(t) \approx V_m^* \sin(\omega t) + V_{dc}^* \quad (17)$$

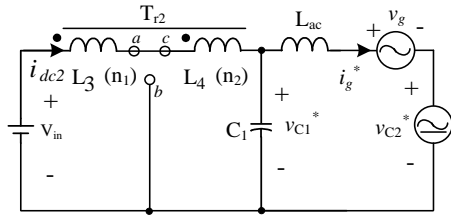
The ac component of the expression (17) is complementary to that of the expression (1), so the phase difference is  $180^\circ$ , besides the equations of duty cycle ratio  $d_1(t)$  and  $i_{dc2}^*$  are similar to (7) and (9), respectively, with  $\pi$  rad added to the arguments of sine or cosine function (or simply the terms in  $\sin(\omega t)$  and  $\cos(\omega t)$  change their signs, while the terms

involving  $2\omega t$  remain unchanged).

In order to follow  $i_g^*$ , the switches  $T_3$  and  $T_4$  in the left-side converter are controlled with some certain duty cycle ratio  $d_L$ , and their pulse control signals  $v_{dr3}$  and  $v_{dr4}$  should be complementary to each other. In light of the switching state, the left-side converter can be equivalent to two sub circuits as shown in the Fig.5, and the right-side converter is taken as voltage source in the Fig.5.



(a)  $T_3$  is on and  $T_4(D_4)$  off



(b)  $T_4(D_4)$  is on and  $T_3$  off

Fig.5. The two structures of the left-side converter.

The two states in the Fig.5 can be described by the switching function as (18).

$$\lambda = \begin{cases} 1 & \text{When } T_3 \text{ is on and } T_4(D_4) \text{ off} \\ 0 & \text{When } T_3 \text{ is off and } T_4(D_4) \text{ on} \end{cases} \quad (18)$$

Assuming that the error variable for filter  $L_{ac}$  is  $\varepsilon i_g$ , the error variable for filter  $C_1$  is  $\varepsilon v_{C1}$ , and the error variable for coupled inductor  $L_3/L_4$  is  $\varepsilon i_{dc2}$ . As are shown in the expression (19).

$$\begin{cases} \varepsilon i_g = i_g - i_g^* \\ \varepsilon v_{C1} = v_{C1} - v_{C1}^* \\ \varepsilon i_{dc2} = i_{dc2} - i_{dc2}^* \end{cases} \quad (19)$$

After defining the error variables, the state error equation (20) can be written.

$$\dot{y} = A'y + B'\lambda + D' \quad (20)$$

Where,

$$y = [\varepsilon i_g, \varepsilon v_{C1}, \varepsilon i_{dc2}]^T \quad (21)$$

$$A' = \begin{bmatrix} 0 & \frac{1}{L_{ac}} & 0 \\ -\frac{1}{C_1} & 0 & \frac{1}{C_1} \\ 0 & -\frac{1}{L_3 \cdot (1+N)} & 0 \end{bmatrix} \quad (22)$$

$$B' = \begin{bmatrix} 0 \\ -\frac{i_{dc2}^*}{C_1} \\ \frac{v_{C1} + N \cdot V_{in}}{L_3 \cdot (1+N)} \end{bmatrix} \quad (23)$$

$$D' = \begin{bmatrix} \frac{v_{C1}^* - v_{C2}^* - V_g}{L_{ac}} \\ \frac{i_{dc2}^* - i_g^*}{C_1} \\ \frac{V_{in} - v_{C1}^*}{L_3 \cdot (1+N)} \end{bmatrix} \quad (24)$$

The derived function for the reference is neglected in the state error equation (20) as in the previous section. Through the above analysis, the error state equation of CIDBI is obtained.

### III. CONTROL STRATEGY

The sliding mode takes place in the error state space, in which trajectory crossing the origin is designed. When the representative point of the system reaches the trajectory, it is confined close to it, and then directed toward the origin of the space (The origin represents the desired condition: null error) [10-16]. This trajectory is called sliding surface ( $\sigma$ ) which divides the space into two subspaces, with each one associated to a structure. One must ensure that the motion in the neighborhood of  $\sigma$  to be directed toward the surface, so that the representative point does not deviate from the specified trajectory [17-24]. This is known as the existence condition, mathematically expressed by (25).

$$\begin{cases} \dot{\sigma} < 0, & \text{When } \sigma > 0 \\ \dot{\sigma} > 0, & \text{When } \sigma < 0 \end{cases} \quad (25)$$

The practical sliding mode does not occur exactly on the sliding surface all the time, since this would require an infinite switching frequency. In fact, it happens very close to it, giving rise to what is called a quasi-sliding mode. It is a simple way to implement the quasi-sliding mode by hysteretic comparison between the actual state and the desired trajectory [25-27].

#### A. The Implementation of Right-Side SMC

The sliding mode surface  $\sigma_R$  is defined by the expression (26), which can be considered as the weight of error. If taking  $S_1$  and  $S_2$  is constant, the sliding mode surface is a straight line on the plane of  $\varepsilon_{v_{C2}} \times \varepsilon_{i_{dc1}}$ , and the slope is decided by  $\alpha_R = S_1/S_2$ . Among the expression (25), Units:  $S_2$  is ohms;  $S_1$  is dimensionless; and then  $\alpha_R$  is Siemens.

$$\sigma_R = S_1 \cdot \varepsilon_{v_{C2}} + S_2 \cdot \varepsilon_{i_{dc1}} \quad (26)$$

Assuming that  $S = [S_1, S_2]^T$ , the expression (25) can be expressed by (27).

$$\sigma_R = S^T x \quad (27)$$

Since the error variables are defined by(12), whenever the representative point lies below the switching line  $\sigma_R$ , the energy in the elements  $C_2$  and  $L_1$  is little, and thus  $\gamma$  is set to 1, conversely, the energy is surplus when the representative point lies above the switching line  $\sigma_R$ , and  $\gamma$  is set to 0. So the expression (28) can be gotten.

$$\gamma = \begin{cases} 1 & \text{When } \sigma_R < 0 \\ 0 & \text{When } \sigma_R > 0 \end{cases} \quad (28)$$

Due to that the error variables is very small and approaches to zero by comparison with the reference values, so the existing condition (25) for sliding mode surface on the right side can be written by (29) and (30).

$$\dot{\sigma}_R = S^T D < 0, \quad \sigma_R > 0 \quad (29)$$

$$\dot{\sigma}_R = S^T B + S^T D > 0, \quad \sigma_R < 0 \quad (30)$$

Substituting the matrix B and D into (29) and (30) respectively, the expression (31) and (32) can be gotten.

$$S_1 \cdot \frac{i_g^* + i_{dc1}^*}{C_2} + S_2 \cdot \frac{V_{in} - v_{C2}^*}{L_1 \cdot (1+N)} < 0 \quad (31)$$

$$S_1 \cdot \frac{i_g^* + i_{dc1}^* - i_{dc1}}{C_2} + S_2 \cdot \frac{v_{C2} - v_{C2}^* + (1+N) \cdot V_{in}}{L_1 \cdot (1+N)} > 0 \quad (32)$$

It can be concluded from (31) and (32) that the condition is  $S_1$  and  $S_2$  choose non-negative values if there is existence of the sliding mode surface as presented in (26).

If the actual system adopts the fixed hysteresis width  $2\delta$  in the Fig.6, the control rules can be rewritten into (33). And the phase plane trajectory will oscillate on both sides of the sliding mode surface ( $\sigma_R=0$ ), as shown in the Fig.7.

$$\gamma = \begin{cases} 1 & \text{When } \sigma_R < -\delta \text{ or } (\dot{\sigma}_R > 0 \text{ and } |\sigma_R| < \delta) \\ 0 & \text{When } \sigma_R > +\delta \text{ or } (\dot{\sigma}_R < 0 \text{ and } |\sigma_R| < \delta) \end{cases} \quad (33)$$

It can be seen from Fig.7 that the  $\sigma_R$  must increase from  $-\delta$  to  $+\delta$  during the  $\Delta t_1$  ( $\dot{\sigma}_R > 0$ ), while the  $\sigma_R$  must decrease from  $+\delta$  to  $-\delta$  during the  $\Delta t_2$  ( $\dot{\sigma}_R < 0$ ). Assuming that state trajectory remains the same near  $\sigma_R=0$ , the formula (34) for switching frequency can be deduced from Fig.7.

$$f_s = \frac{1}{\Delta t_1 + \Delta t_2} \quad (34)$$

Where,  $\Delta t_1$  corresponds to the switching state  $\gamma=1$ , and  $\Delta t_2$  to  $\gamma=0$ . And  $\Delta t_1$  and  $\Delta t_2$  can be solved by (35) and (36) respectively.

$$\Delta t_1 = \frac{2\delta}{\left( S_1 \cdot \frac{i_g^* + i_{dc1}^* - i_{dc1}}{C_2} + S_2 \cdot \frac{v_{C2} - v_{C2}^* + (1+N) \cdot V_{in}}{L_1 \cdot (1+N)} \right)} \quad (35)$$

$$\Delta t_2 = \frac{-2\delta}{\left( S_1 \cdot \frac{i_g^* + i_{dc1}^*}{C_2} + S_2 \cdot \frac{V_{in} - v_{C2}^*}{L_1 \cdot (1+N)} \right)} \quad (36)$$

Substituting (35) and (36) into (34), the maximum switching frequency can be gotten as (37) at the positive peak value of

$v_{C2}$ , if assuming  $i_g^*=0$  and the corresponding  $i_{dc1}^*=0$  in the state of light load.

$$f_{smax} = \frac{S_2 \cdot V_{in}}{L_1 \cdot 2\delta} \cdot \frac{v_{C2}^* - V_{in}}{v_{C2}^* + N \cdot V_{in}} \quad (37)$$

The duty cycle ratio is defined as follows:

$$d_R(t) = \frac{\Delta t_1}{\Delta t_1 + \Delta t_2} \quad (38)$$

Substituting (34) and (35) into (38), the steady switching frequency can be gotten as (39), depending on the steady operation point such as duty cycle, control parameters and the value of grid current.

$$f_{SR}(t) = [d_R(t) / (2\delta)] \cdot [S_2 \cdot V_{in} / L_1 + S_1 \cdot i_g^*(t) / C_2] \quad (39)$$

When  $i_g$  reaches negative peak value, the variation of  $i_{dc1}$  and  $v_{C2}$  is the largest, at this time, the duty cycle is the largest and the switching frequency the highest. As a consequence,  $L_1$  and  $C_2$  can be designed as (40) in order to ensure that all the magnetic components are unsaturated within the scope of the switching frequency.

$$L_1(t) \geq \frac{V_{in} \cdot d_{Rmax}}{\Delta I_{dc1max} \cdot f_{SRmin}} \quad \text{and} \quad C_2 \geq \frac{I_g^* \cdot d_{Rmax}}{\Delta V_{C2max} \cdot f_{SRmin}} \quad (40)$$

### B. The Implementation of Left-Side SMC

Since the left side is the third order system, the sliding mode surface  $\sigma_L$  can be chosen by (41).

$$\sigma_L = S_3 \cdot \varepsilon i_g + S_4 \cdot \varepsilon v_{C1} + S_5 \cdot \varepsilon i_{dc2} \quad (41)$$

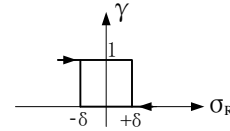


Fig.6. Switching function  $\gamma$ .

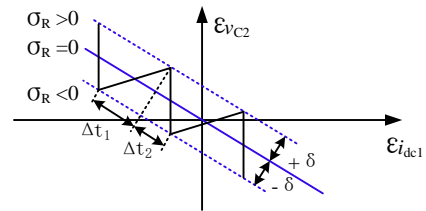


Fig.7. The waveform of  $\sigma_R$ .

Let  $S_L = [S_3, S_4, S_5]^T$ , the expression (41) can be written by (42).

$$\sigma_L = S_L^T y \quad (42)$$

Since the error variables are defined by (19), whenever the representative point lies below the switching line  $\sigma_L$ , the energy in the elements  $C_1$  and  $L_3$  is little, and thus  $\lambda$  is set to 1, conversely, the energy is surplus when the representative point lies above the switching line  $\sigma_L$ , and  $\lambda$  are set to 0. So the expression (43) can be gotten.

$$\lambda = \begin{cases} 1 & \text{When } \sigma_L < 0 \\ 0 & \text{When } \sigma_L > 0 \end{cases} \quad (43)$$

Due to that the error variables is very small and approaches to zero by comparing with the reference values, so the existing condition (25) for sliding mode surface on the left side can be written by (44) and (45).

$$\dot{\sigma}_L = S^T D^> < 0, \quad \sigma_L > 0 \quad (44)$$

$$\dot{\sigma}_L = S_L^T B^> + S_L^T D^> < 0, \quad \sigma_L < 0 \quad (45)$$

Substituting the matrix B^> and D^> into (44) and (45) respectively, the expression (46) and (47) can be gotten.

$$S_3 \frac{v_{C1}^* - v_{C2}^* - V_g}{L_{ac}} + S_4 \frac{i_{dc2}^* - i_g^*}{C_1} + S_5 \frac{V_m - v_{C1}^*}{L_3 \cdot (1+N)} < 0 \quad (46)$$

$$S_3 \frac{v_{C1}^* - v_{C2}^* - V_g}{L_{ac}} + S_4 \frac{i_{dc2}^* - i_{dc2} - i_g^*}{C_1} + S_5 \frac{V_m(1+N) + v_{C1} - v_{C1}^*}{L_3 \cdot (1+N)} > 0 \quad (47)$$

If the expression (46) and (47) are established, the existing condition for the sliding mode will be met. And the sufficient condition for the existence of the sliding mode surface defined by (41) is that S<sub>3</sub>, S<sub>4</sub> and S<sub>5</sub> choose nonnegative values.

Just like the right side, the hysteresis comparison control is employed to avoid excessive switching frequency. Here, assuming that the hysteresis width is 2ζ, the phase plane trajectory will oscillate on both sides of the sliding mode surface(σ<sub>L</sub>=0)

Defining the on time of switch T<sub>3</sub> for Δt<sub>1</sub> that corresponds to the switching state λ=1, and the off time for Δt<sub>2</sub> that corresponds to the switching state λ=0, Δt<sub>1</sub> and Δt<sub>2</sub> can be solved by (48) and (49) respectively.

$$\Delta t_1 = \frac{2\zeta}{\left( S_3 \frac{v_{C1}^* - v_{C2}^* - V_g}{L_{ac}} + S_4 \frac{i_{dc2}^* - i_{dc2} - i_g^*}{C_1} + S_5 \frac{V_m(1+N) + v_{C1} - v_{C1}^*}{L_3 \cdot (1+N)} \right)} \quad (48)$$

$$\Delta t_2 = \frac{-2\zeta}{\left( S_3 \frac{v_{C1}^* - v_{C2}^* - V_g}{L_{ac}} + S_4 \frac{i_{dc2}^* - i_g^*}{C_1} + S_5 \frac{V_m - v_{C1}^*}{L_3 \cdot (1+N)} \right)} \quad (49)$$

Just the same with the right-side converter, the maximum switching frequency happens at the light load. Assuming that i<sub>g</sub><sup>\*</sup>=0 and the corresponding i<sub>dc2</sub><sup>\*</sup>=0, and noticing that the coefficient of terms S<sub>3</sub> is zero and the terms of S<sub>4</sub> and S<sub>5</sub> like (35) and (36), the expression (50) can be gotten.

$$f_{smax} = \frac{S_4 \cdot V_m}{L_3 \cdot 2\zeta} \cdot \frac{v_{C1}^* - V_m}{v_{C1}^* + N \cdot V_m} \quad (50)$$

It can be seen that the last two terms are the same with the

right side in the existing condition (46) and (47) for sliding mode surface of the left-side converter, i.e., S<sub>4</sub>=S<sub>1</sub>, S<sub>5</sub>=S<sub>2</sub>. So only the value of S<sub>3</sub> needs to be confirmed. Since the CIDBI is used as grid connected power generation, v<sub>C1</sub><sup>\*</sup> - v<sub>C2</sub><sup>\*</sup> - V<sub>g</sub> > 0 in (46) must be established and this difference value should equal to v<sub>Lac</sub> of L<sub>ac</sub>, thus S<sub>3</sub>>0; as a consequence, the upper limit of S<sub>3</sub> can be solved by (46) under the condition of the maximum of |v<sub>Lac</sub>|.

The steady state switching frequency can be solved by (51) only if Δt<sub>1</sub>+Δt<sub>2</sub> is calculated during the period of hysteresis which is like the right side.

$$f_{SL}(t) = [d_L(t) / (2\zeta)] \cdot [S_5 \cdot V_m / L_3 - S_4 \cdot i_g^*(t) / C_1] \quad (51)$$

When i<sub>g</sub> reaches the positive peak value, the variation of i<sub>dc2</sub> and v<sub>C1</sub> is the largest, and hence, the values of L<sub>3</sub> and C<sub>1</sub> can be chosen by (52).

$$L_3(t) \geq \frac{V_m \cdot d_{Lmax}}{\Delta i_{dc2max} \cdot f_{SLmin}} \quad \text{and} \quad C_1 \geq \frac{I_g^* \cdot d_{Lmax}}{\Delta V_{C2max} \cdot f_{SLmin}} \quad (52)$$

The application of such high-pass filters is preferred to the use of a calculated reference for the secondary variables. Since the controller seeks to nullify the weighted sum of errors, and not an error in particular. Thus, if one of those references is determined incorrectly, this would result in tracking imperfections in the output variables. This technique is well suited to dc-dc converter. In the present application, the low-frequency ac evolutions (up to doubled mains frequency) can be considered as quasi static state, therefore the derivative of these variables can be thought of as zero. In the feedback, the low-frequency components are filtered by high-pass filter, it can be considered that the low-frequency components have tracked the reference value in the steady state, and only high-frequency components including high-frequency components and even the spike need to be regulated.

So it is feasible to collect indirect variables with high-pass filter. Of course, the introduction of the high-pass filter will increase the order of the system. Generally speaking, during transients, the response becomes more oscillatory than it would be using defined references for the secondary variables. It is desirable to have fast filters with high cutoff frequency, in order not to alter greatly the dynamics of the circuit. But on the other hand they must present some sensitivity to the resonant frequency of L<sub>1</sub> and C<sub>1</sub> or L<sub>2</sub> and C<sub>2</sub>, remembering that the resonant frequency in a switched circuit depends also on the duty cycle. One must observe that, for safety concerns, a current limitation must be implemented for the dc inductors, because the system with filters is not sensitive to the value of the current when the switch is on for a long period of time(this is especially true during start-up).

Also the selection of L<sub>ac</sub> is conditioned by the high-pass filters, especially by the filter for v<sub>C1</sub>, which must be sensitive to the resonant frequency between L<sub>ac</sub>, and the series

association of  $C_1$  and  $C_2$ .

#### IV. DESIGN EXAMPLE AND EXPERIMENTAL RESULTS

In order to verify the correctness of the above theoretical analysis, a two-channel experimental prototype is built as shown in Fig.8. and each channel is connected to a photovoltaic module. Here, the performance parameters of only one channel are given here. The input dc source is PV array whose open voltage is 44V and the maximum power point (MPP) voltage is 35V. The prototype specifications are listed in the Tab.I.

In order to regulate grid current  $i_g$  linearly with the duty cycle ranging from 0.3 to 0.75, the dc bias voltage  $V_{dc}$  choose 280V, and the turn ratio of the coupled inductor is 4. Assuming that the grid current is 1.43Arms at MPP, the reference current and voltage are as follows.

$$\begin{cases} i_g^*(t) = 1.43\sin(\omega t) \\ v_{c2}^*(t) = 162.6\sin(\omega t + \pi) + 280 \end{cases} \quad (59)$$

It is clear that  $V_{C2max}=442.6V$ .

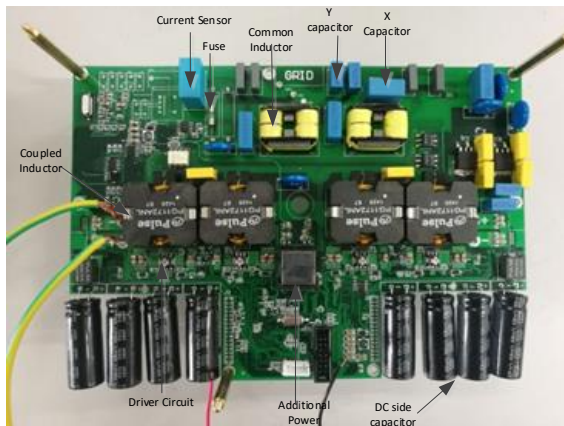


Fig.8. Experimental prototype of photovoltaic grid-connected micro-inverters based on CIDBI.

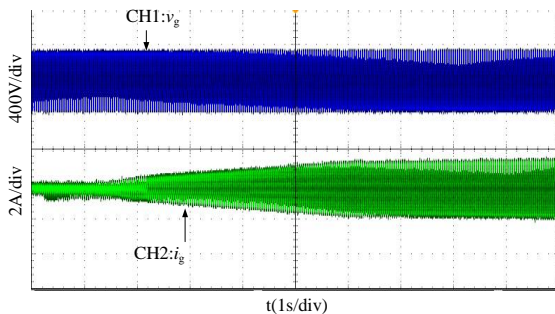


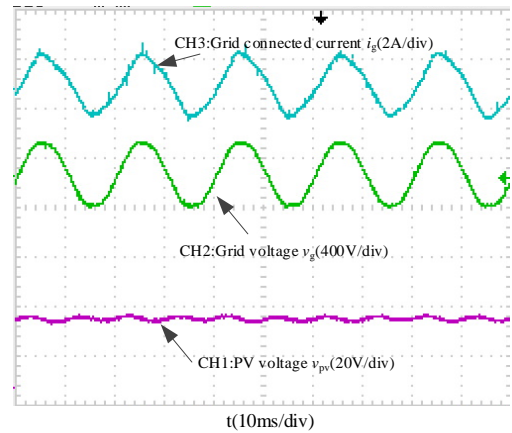
Fig.10. Start-up generation waveform of micro inverter.

TABLE I

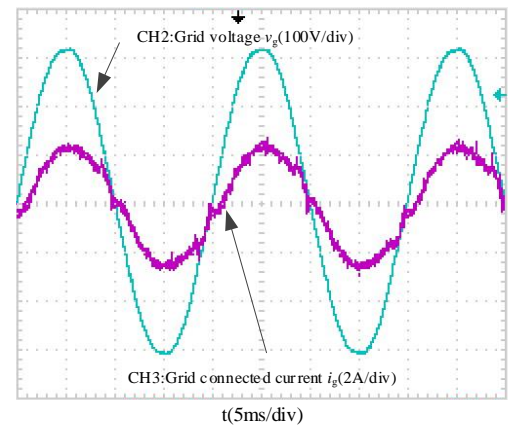
THE PROTOTYPE SPECIFICATIONS

Output power: $P_o$	480W
Grid maximum voltage: $v_g$	230Vrms
MPP voltage: $V_{in}$	35V
Grid frequency: $f_0$	50Hz
Maximum switching frequency: $f_{smax}$	130kHz

DC capacitors  $C_1$  choose  $6 \times 2200\mu F$ ,  $T_1$  and  $T_3$  choose IPB072N15N3G from Infineon (150V/93A @100°C),  $R_{dson}=7.2m\Omega$ ;  $T_2$  and  $T_4$  choose SPB17N80C3 from Infineon (800V/17A @100°). Main experimental waveforms are given in the Fig.9, here,  $v_g$  is utility grid voltage,  $i_g$  is grid connected current, and  $v_{pv}$  is the PV voltage. The Fig.9(a) shows the steady operation waveforms with PI controller in reference [1] and Fig.9(b) presents the waveforms based on SMC in this paper, and the grid connected current waveforms  $i_g$ (CH3) in the Fig.9(b) is clearly much better than that in the Fig.9(a). When the output power reaches to 221W, the efficiency of CIDBI is greater than 95.8%, the total harmonic distortion of  $i_g$  is equal to 1.52%, and the power factor is not less than 0.99. It is clear that the presented sliding mode controller can be applied to CIDBI and transfer energy from dc side to mains grid with high quality.



(a) With PI controller in reference [1]



(b) With SMC in this paper

Fig.9. The main steady operation waveforms in the grid-connected generation.

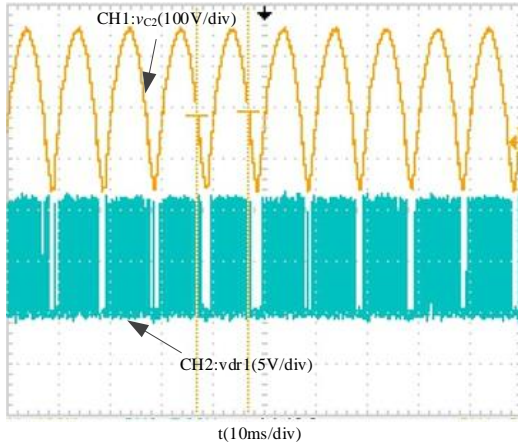


Fig.11. Driving waveform of T1 & voltage on C<sub>2</sub> in right-side circuit.

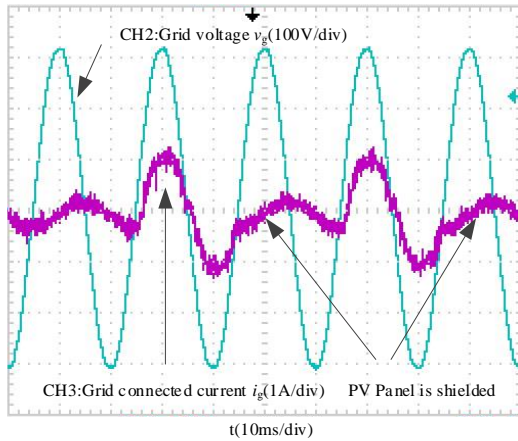


Fig.12. Occlusion experiment of photovoltaic module.

Fig.10 shows the waveforms of grid voltage and grid-connected current when the prototype starts. As can be seen from the figure, when the prototype starts to grid-connected power generation, the current  $i_g$  begins to increase gradually. After a period of time, the current reaches its maximum value and constant. Channel 1 of Fig.11 shows the voltage waveform on capacitor C<sub>2</sub> in the right circuit of CIDBI, and channel 2 is the driving waveform of switch T<sub>1</sub>. Channel 3 of Fig.12 shows the grid-connected current waveform ( $i_g$ ) when the photovoltaic panel is shielded or not, and channel 2 of Fig.12 is grid voltage. From the above experimental waveforms, using the two sliding mode controllers proposed in this paper can not only obtain the stability of micro-inverters based on CIDBI architecture, but also obtain fast dynamic response speed.

## V. CONCLUSIONS

The CIDBI can be used to generate an ac output voltage larger than the dc input one and transfer the energy from PV array or dc energy storage device to the grid with only one power processing stage. Aiming at the problem that it is difficult to design parameters of traditional regulators for high-order systems of CIDBI circuits, the different sliding-mode

controllers are proposed respectively on the right-side converter and the left-side one with different switching scheme in this paper, and the SMCs can track the time-varying reference functions very well. These functions are only defined for the two output variables; other secondary signals are obtained after high-pass filtering. The use of such SMCs can achieve overall performance of the system remarkably.

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