

Modeling reactive magnetron sputtering: opportunities and challenges

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Abstract

The complexity of the reactive magnetron sputtering process is demonstrated by four simulation examples. The examples, commonly encountered during the application of this process for thin film deposition, are described by a numerical model for reactive sputter deposition. A short description of the current model precedes these case studies. In the first example, redeposition of sputtered atoms on the target is studied by its effect on the hysteresis behavior often observed during reactive sputtering. Secondly, the complexity of current-voltage characteristics during reactive magnetron sputtering is treated. The influence of substrate rotation and the pulsing of the discharge current illustrate the time dependence of the reactive sputtering process. As a conclusion, the two main challenges for a further improvement of the model are discussed.

Keywords: reactive magnetron sputtering, modeling

1. Introduction

Modeling of reactive magnetron sputtering is essential to get a full understanding of this process. This bold statement is based on the long experience of our team supported by many researchers in the thin film communities. Or stated by J.E. Greene in one of his review papers on the history of thin film deposition [1], *"Another important recent accomplishment in sputter deposition is the evolutionary development of very useful models, which are increasingly more accurate, of the highly complex reactive-sputtering process."*, and further in the same paper *"...their value is in allowing the researcher to pose "what if" questions before initiating experiments, thus greatly decreasing the number of iterations prior to achieving desired results."* Inspired by this latter statement, a few, rather complex, examples of simulations performed with the RSD model [2, 3], in combination with the Monte Carlo particle trajectory code SIMTRA [4, 5], will be discussed. These examples will not only demonstrate the opportunities for other researchers to apply the model to specific problems, but they will also show the complexity, and hence the challenges for future improvement in modeling reactive magnetron sputtering. The input files for the simulations presented in this paper are available on [6]. The paper starts with a short description of the RSD model, and consequently discusses the influence of redeposition, the behavior of I-V characteristics during reactive sputtering, substrate rotation, and the impact of current pulses on the reactive sputtering process.

23 **2. Reactive sputtering and the RSD model**

24 The key elements of reactive sputtering, and the RSD model are de-
25 scribed in this section. Reactive magnetron sputter deposition is a compli-
26 cated process which can puzzle the less experienced researcher. To assist the
27 understanding of this short description (and the remainder of the paper), the
28 interested reader can benefit from a tutorial paper on this topic [7]. Technical
29 details about the model implemented in the RSD software can be found in
30 previous published work [8, 2, 3].

31 The RSD model follows the tradition in modeling reactive sputtering to
32 describe this process based on the conservation of reactive species [9]. To
33 simplify the discussion, only diatomic gases such as oxygen and nitrogen
34 are considered. The molecular reactive gas enters the vacuum chamber at
35 a reactive gas flow equal to Q_{in} . The initial gas mainly reacts with the
36 deposited material to form the desired compound on the substrate. The
37 consumption rate of the reactive gas atoms in this process is described by
38 Q_c . The reactive gas reacts to a minor extent also at the target. Again a
39 consumption rate Q_t is defined. When reactive gas flow remains lower than a
40 given reactive flow, known as the first critical point $Q_{in,1}$, this description is
41 basically sufficient to explain the observed features of the process. Indeed, if
42 the deposition rate remains high, the reactive gas partial pressure is low, and
43 other deposition parameters hardly differ from the condition without reactive
44 gas addition. Therefore, this regime of the process is known as the metallic
45 mode. The deposited compound in this regime is typically substoichiometric.
46 At the first critical point, the reactive gas flow is balanced by the maximum
47 consumption rate, or the getter capacity, of the process. Further reactive

48 gas addition leads to important process changes: an increase of the reactive
49 gas partial pressure, and a drop in the deposition rate. The origin of these
50 changes is the interaction of the reactive gas with the target which leads
51 to compound formation. The lower sputter yield of the target under these
52 conditions reduces the gas getter capacity as less metal is deposited. Hence,
53 the reactive gas partial pressure increases until a new steady state condition
54 is reached where the target is fully covered by a compound layer. Under these
55 conditions, known as the compound or poisoned mode, a substantial fraction
56 of the reactive gas is consumed by the vacuum pump, or the consumption rate
57 Q_p of the pump is high. Due to the difference in sputter yield between the
58 metallic and the compound mode, it is not possible to return to the metallic
59 mode at the same reactive flow of the first critical point, and it is necessary
60 to reduce the flow to a lower value. The flow at which the process switches
61 back to the metallic mode is known as the second critical point $Q_{in,2}$. The
62 transition for metallic to poisoned mode (and reverse) occurs quite often
63 abrupt. So, when the reactive gas flow is set slightly higher (lower) than
64 the first (second) critical point a jump in the process parameters is observed
65 which explains the definition of these critical points. As the two critical points
66 do not coincide, a hysteresis is observed for the main process parameters as
67 a function of the reactive gas flow. The quantitative description of this
68 hysteresis is the main objective of the RSD model.

69 So, essentially, the RSD model describes the balance between the gas
70 input, and the consumption rates of the pump, the substrate, and the target,

71

$$Q_{in} = Q_p + Q_c + Q_t \quad (1)$$

72 The easiest way to grasp the essence of the model is a short description of how
73 these rates are approached. Both the consumption rate by the vacuum pump
74 and the deposited material can easily be understood. The first is defined by
75 the reactive gas partial pressure, and the pumping speed of the deposition
76 set-up for the reactive gas. The consumption rate by the deposited material
77 is described as an incorporation process of the reactive gas molecules. The
78 efficiency of the process is defined by a material dependent incorporation, or
79 sticking, coefficient. The consumption rate is further defined by the flux of
80 the reactive gas molecules towards the deposited material, and the metal frac-
81 tion of the deposited material. The latter is defined by the deposition rate of
82 both metal and compound which depend on the target condition. To describe
83 the deposition rate at different locations in the deposition setup, the RSD
84 model can use the in-house build particle trajectory code SIMTRA. A short
85 description of this code will be given further in the text. Up to this point,
86 the RSD model builds on the contribution of several authors to the field.
87 The model distinguishes itself by the description of the consumption rate by
88 the target as a consequence of several target processes. Evidently sputtering
89 of metal and compound molecules is the first one. The chemisorption of re-
90 active gas molecules at the target surface is described in a similar fashion as
91 the chemical reaction between the reactive gas and the deposited material.
92 The implantation of the reactive gas into the target is treated in a distinctive
93 way for the RSD model. The chemical reaction of the implanted reactive gas
94 atoms is treated as a second order chemical reaction. By explicitly describing
95 this reaction, the model predicts a fraction of non-reacted gas atoms in the
96 target. Although no direct evidence can be given for their presence, there is

97 substantial experimental indirect evidence such as discharge voltage changes
98 [10, 11], time dependent sputter yields [12], and the observation of a second
99 hysteresis [13]. The latter is attributed to a change of the target state due
100 to the presence of these non-reacted atoms. Finally, the RSD model can
101 also account for the redeposition of sputtered atoms on the target. A non-
102 uniform current profile on the target, a deposition profile on the substrate,
103 and a dependence of the current-voltage discharge characteristic on the tar-
104 get condition, has made the model a versatile tool to investigate reactive
105 magnetron sputter deposition. As mentioned before, the deposition rate at
106 different locations in the deposition setup can be modeled with SIMTRA. It
107 is a test particle Monte Carlo code which simulates the trajectory of sput-
108 tered atoms from the target towards their final location. Test particles are
109 launched from the target with an energy and a direction randomly selected
110 from the nascent energy and angular distribution. The collisions with the
111 sputter gas atoms are described based on spherical symmetric interaction po-
112 tential. The test particle is followed until it arrives on any of the predefined
113 surfaces which describe the experimental set-up. More details on this code
114 can be found in [4, 5]. Both SIMTRA and RSD have been implemented in a
115 downloadable executable[14]. The focus of the current paper is to illustrate
116 the possibility of the RSD model with several interesting examples for the
117 thin film community.

118 **3. Example 1: Redeposition of sputtered atoms**

119 Atoms sputtered from the target travel through the gas phase towards
120 their final landing place. They can collide with the gas atoms, and recoil onto

121 the target. This process is known as redeposition. To study redeposition, the
 122 particle trajectory code SIMTRA is one of the possible tools. Figure 1 shows
 123 the fraction of the sputtered atoms that returns to the target as a function
 124 of the argon pressure in the vacuum chamber. The return probability for
 125 elements lighter than argon (Li, Al) is much larger than for heavier elements.
 126 This can be understood from binary collision physics. The deflection angle
 127 of the sputtered atoms increases smoothly as function of the ratio between
 128 the masses of the gas atom and the sputtered atom [15]. The influence of
 129 the argon pressure can be understood from the increased collision probability
 130 when the gas density between ejection and deposition position is increased.

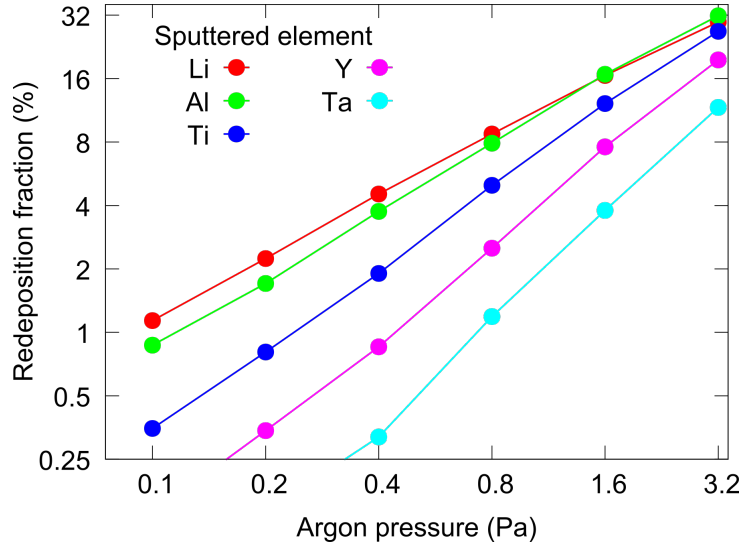


Figure 1: The simulated redeposition fraction on the target as a function of the argon pressure for different elements. Simulation conditions are a $30 \times 30 \times 30$ cm³ chamber with a two inch magnetron where the target surface is positioned in the center and 7.5 cm away of one of the chamber walls. A 2.5×2.5 cm² substrate was positioned parallel with the target surface at a mutual distance of 10 cm. The total number of simulated particles was set to 10^5 .

131 At higher pressures, and especially for light elements, the redeposition
132 fraction can become substantial which explains the necessity to include re-
133 deposition, next to chemisorption, reactive ion implantation, and sputtering,
134 as the fourth target process in the description of the target condition during
135 reactive sputtering [16]. Figure 2 shows the simulated influence of the rede-
136 position fraction on one of the typical studied process curves, i.e. the reactive
137 gas partial pressure as a function of the reactive gas flow. The simulation
138 shows that the first critical point, i.e. the transition point from the low oxy-
139 gen pressure to the high oxygen pressure regime, shifts towards lower oxygen
140 flows while the second critical point remains unaffected. This behavior could
141 have multiple explanations. When the redeposition fraction is increased, less
142 material is deposited on the different surfaces of the vacuum chamber. This
143 reduces the getter capacity, and hence less oxygen needs to be introduced to
144 switch from metallic mode to poisoned mode. Once in poisoned mode, the gas
145 consumption rate Q_c is a negligible part of the total gas consumption rate,
146 and hence the lower deposition rate does not influence the second critical
147 point as observed from the simulations. This explanation is consistent with
148 the result of an experimental study on facing target sputtering [17] which
149 shows a clear shift of the first critical point while a minor shift of the second
150 critical point. An alternative explanation could be based on the lowering of
151 the effective erosion rate of the target, i.e. the erosion rate due to sputter
152 bombardment reduced by the growth rate by redeposition. A change of only
153 the effective erosion rate can easily be implemented in the simulations by
154 changing the effective sputter yield of both the compound and metal.

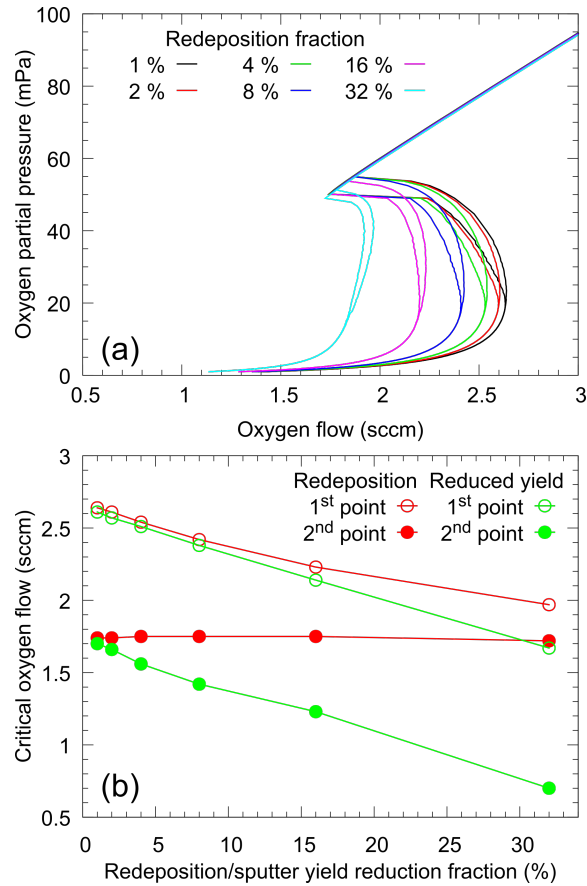


Figure 2: (a) The simulated oxygen partial pressure as a function of the oxygen flow for different levels of redeposition. The same chamber configuration as used for Figure 1 was implemented in SIMTRA, and the output was used in the RSD software. An Al target was used. The same simulated redeposition profile at 0.4 Pa was used, to allow a straightforward interpretation of the influence of the redeposition fraction on the different hystereses. All other parameters remained fixed. The smoother transition at the first critical point as compared to the second critical point is mainly to the gradual oxidation of both substrate and target. (b) The critical oxygen flow as a function of the redeposition fraction (red markers), or the sputter yield reduction fraction (green markers). The first critical point is indicated with open markers while the second critical point is represented by closed markers. The simulation conditions were identical apart from the shown dependency.

155 This will affect the getter capacity, and results in a similar shift of the
156 first critical point as shown in [Figure 2b](#). However, a clear difference for
157 the second critical point is observed. When the effective sputter yield is
158 reduced, the second critical point shifts in a similar manner as the first critical
159 point towards lower oxygen flows. The behavior of the second critical point
160 under these conditions can be explained as follows. The second critical point
161 is mainly defined by the chemical reaction of the implanted oxygen in the
162 target. When the effective sputter yield is reduced, the reaction time of the
163 implanted species becomes longer which results in a higher target oxidation
164 state. To return to the metallic mode, the oxygen fraction in the discharge
165 must be reduced to a lower value, or stated differently the oxygen flow at
166 the second critical point must be lower. This effect has no influence when
167 the redeposition fraction is increased. Indeed, in the current version of the
168 RSD code, as compared to the version used in [\[17\]](#), compound sputtering is
169 described as an atomistic process while the congruent aspect of sputtering
170 has been preserved. When the reactive atoms are redeposited on the target,
171 their reaction probability is low as the target surface at the second critical
172 point is fully poisoned. The redeposited metal atoms will only influence
173 to a minor extend the erosion rate as their sputter yield is high. Indeed,
174 the redeposited metal is incorporated at the surface as a non-reacted metal.
175 The rather low oxygen sticking probability of 0.1 used in the simulations
176 results in a lower probability for the reaction of the redeposited metal by
177 chemisorption as compared to the probability to be re-sputtered with a high
178 yield as non-reacted metal.

179 In summary, based on these simulations it is clear that redeposition can

180 be an important effect during magnetron sputter deposition, and its impact
181 on the first critical point can be understood from a reduced deposition rate
182 that lowers the getter capacity of the process.

183 **4. Example 2: I-V characteristics**

184 A now less common approach to study reactive magnetron sputtering
185 is sweeping the discharge current while the reactive gas flow is maintained.
186 Different process parameters, such as the reactive gas partial pressure or
187 the discharge voltage, as function of the discharge current can be measured.
188 When the discharge voltage is tracked, the obtained plot is known as an I-V
189 characteristic which presents the applied discharge current as a function of
190 the measured discharge voltage. [Figure 3](#) depicts two possible shapes of a I-V
191 characteristic. The top figure (a) represents a simulated I-V characteristic
192 for Ti measured in a mixture of O₂/Ar, while the bottom figure (b) shows
193 the same information for Al. The simulation of this kind of I-V character-
194 istic requires experimental input of the I-V characteristics in metallic and
195 poisoned mode. When this information is accessible, the RSD model permits
196 to calculate the entire characteristic based on the assumption that the ion
197 induced electron yield is a weighted function of the electron yield of the oxide
198 and the metal. More details on this approach can be found in Strijckmans
199 et al. [7], and in Depla et al. [18] In this context, it is important to mention
200 that in the Ti/O₂ case the simulated curve is an approximation to illustrate
201 the different behavior between Ti and Al. Although, the general trend is
202 correctly simulated, the RSD model does not account for the presence of
203 different titanium oxides.

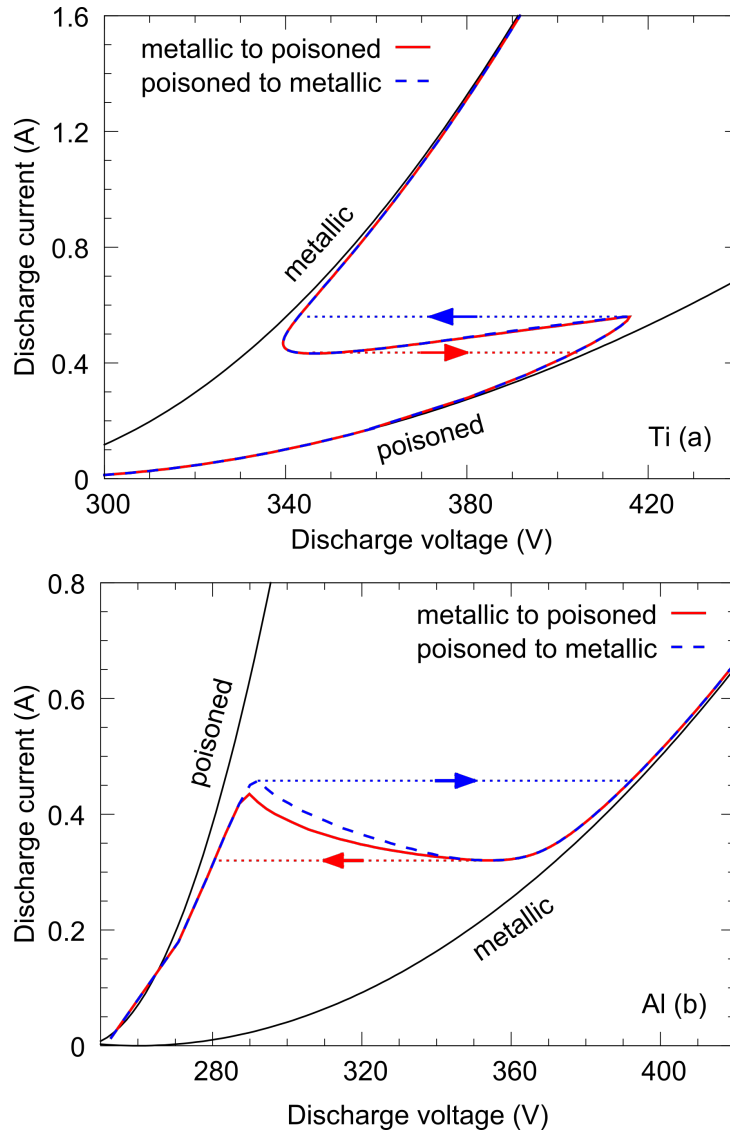


Figure 3: Simulated I-V characteristics for (a) Ti and (b) Al in a O_2/Ar atmosphere. The oxygen flow was fixed at 1.5 sccm. To describe the I-V characteristic in the transition zone, the I-V characteristics of a metal target and a fully poisoned target (black lines) are combined. This combination is based on a weighted average of the electron yields which depends on the target surface fractions. In the case of Al the I-V characteristic is doubled valued in the transition region, while this is not the case for Ti. The main reason for this difference is the larger ratio between the metal and the oxide sputter yield in the case of Al.

204 At low discharge current(power), the target is in poisoned mode. When
205 the current is increased, the target will abruptly change from the poisoned
206 state to the metal state as indicated by the dashed lines. The main difference
207 between Al and Ti is the direction of the transition. In the case of Al, the
208 discharge voltage increases when the target changes to the metallic state,
209 while for Ti the opposite occurs. This has an important effect on the process
210 control. With the available power supplies for DC magnetron sputtering, it
211 is possible to stabilize the discharge on current, voltage or power. The simu-
212 lations shows that for both Al and Ti it is impossible to reach the transition
213 region between the metallic and the poisoned state when the discharge is
214 current controlled. The target condition will abruptly change as indicated
215 by the dashed lines. In the case of Al, the simulated I-V characteristics is a
216 double valued function of the discharge voltage, in contrast to a triple val-
217 ued function for the Ti case, which permits to perform voltage-controlled
218 experiments. The implementation of the I-V characteristic of a magnetron
219 discharge in metallic and poisoned mode in the RSD code permits to describe
220 the behavior of the discharge current and voltage at fixed oxygen and argon
221 flows. Also the change of the oxygen partial pressure can be calculated under
222 these conditions. This permits to compare the simulation results with exper-
223 iments that are not influenced by long term effects. Indeed, in a "classical"
224 hysteresis experiment as shown in [Figure 2a](#), the oxygen flow is stepwise in-
225 creased, and the deposition parameters are registered. As the stabilization
226 time after an oxygen flow change is in the order of minutes, the measurement
227 of a full process curve requires quite some time. Long term effects such as
228 target erosion can in this way influence the measurement, and hence its in-

229 terpretation. By scanning I-V characteristics at different, but non-sequential
 230 flows, this problem can be circumvented, and permits to investigate reactive
 231 magnetron sputter deposition in an alternative way. . An example of this
 232 kind of experiments is presented in Figure 4. This kind of measurements
 233 (and simulations) have shown to be essential to understand the poisoning
 234 behavior of the target [13].

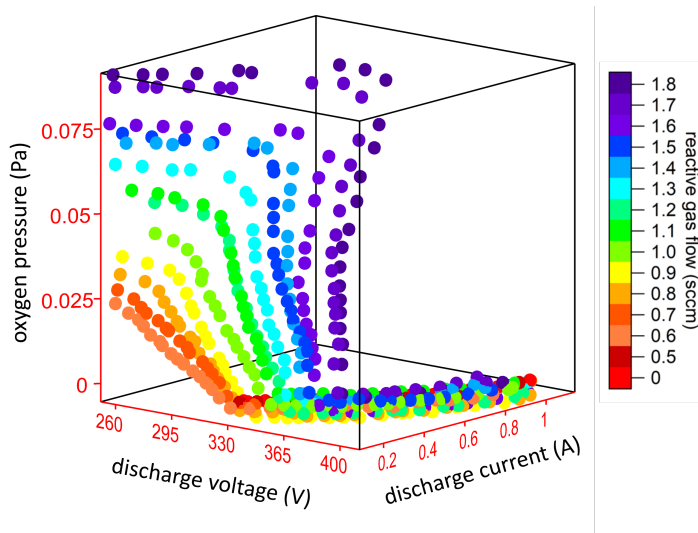


Figure 4: Measurement of the current, voltage, pressure and flow space during reactive magnetron sputtering of Al in a O_2/Ar mixture. Experimental conditions are a 2 inch target at an argon pressure 0.4 Pa. More experimental details can be found in Schelfhout et al. [13]

235 5. Example 3: Sample rotation

236 The gas distribution of both the reactive and the sputter gas are impor-
 237 tant to improve film uniformity, especially for large area coaters [19]. The
 238 reactive gas distribution can affect in a complex way the film properties [20]

239 and target poisoning [21]. The influence of the reactive gas distribution can
240 only be modeled with more advanced codes that include the gas dynamics
241 [22].

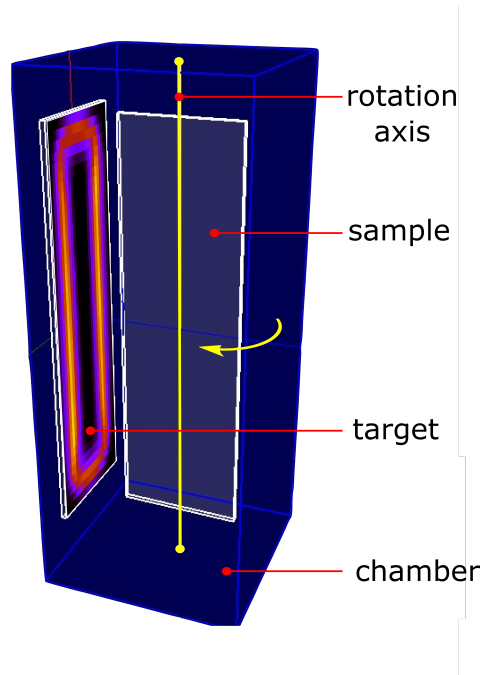


Figure 5: Overview of the implemented setup to investigate the influence of sample rotation on the hysteresis behavior. A large substrate ($40 \times 12 \text{ cm}^2$) is placed in a vacuum chamber with dimensions $20 \times 20 \times 50 \text{ cm}^3$ (dark blue). The target (Al) is a rectangular plate with the same size as the substrate. The racetrack of the rectangular target is an experimental measured erosion profile where the used resolution is $1 \times 0.5 \text{ cm}^2$. When the substrate is parallel to the target the distance between them is 10 cm. SIMTRA simulations were performed at 0.4 Pa argon. For the RSD simulations the discharge current was set at 2 A and the pumping speed equaled 55 l/s.

242 Due to the increased complexity, one often needs to find a compromise
243 to obtain reasonable simulation times, e.g. a less detailed description of the

244 target poisoning mechanisms. In the RSD code, the gas distribution is not
245 included as it is assumed that the oxygen pressure is uniform over the vacuum
246 chamber. The distribution of the sputtered atoms is however implemented
247 including the output of SIMTRA simulations. This permits to investigate
248 the influence of sample rotation as an alternative to optimize film uniformity.
249 When the substrate orientation affects in a major way the deposition profile,
250 it can be expected that the rotation speed of a substrate will influence the
251 reactive sputter process. This point will be discussed in more detail based
252 on the setup shown in [Figure 5](#). RSD simulations were performed for differ-
253 ent rotation speeds of the substrate. When the rotation speed is increased
254 the first critical point shifts towards higher oxygen flow, while the second
255 critical point remains unaffected ([Figure 6a](#)). To understand this behavior
256 it is instructive to follow the dynamic behavior of the oxygen partial pres-
257 sure ([Figure 6b](#)). When the substrate is parallel to the target, the oxygen
258 pressure is higher as compared to the situation when the substrate stands
259 perpendicular to the target as in [Figure 5](#). This indicates that less oxygen
260 is consumed during the parallel substrate orientation. The influence of the
261 substrate orientation lays in the blocking of the deposition on the chamber
262 walls which is low when the substrate is parallel with the target (*parallel* in
263 [Figure 6a](#)) and high when the substrate is perpendicular positioned towards
264 the target (*perpendicular* in [Figure 6a](#)). Or stated differently, a larger ef-
265 fective deposition area is obtained as substantial deposition on the chamber
266 walls is permitted when the substrate is in the perpendicular position.

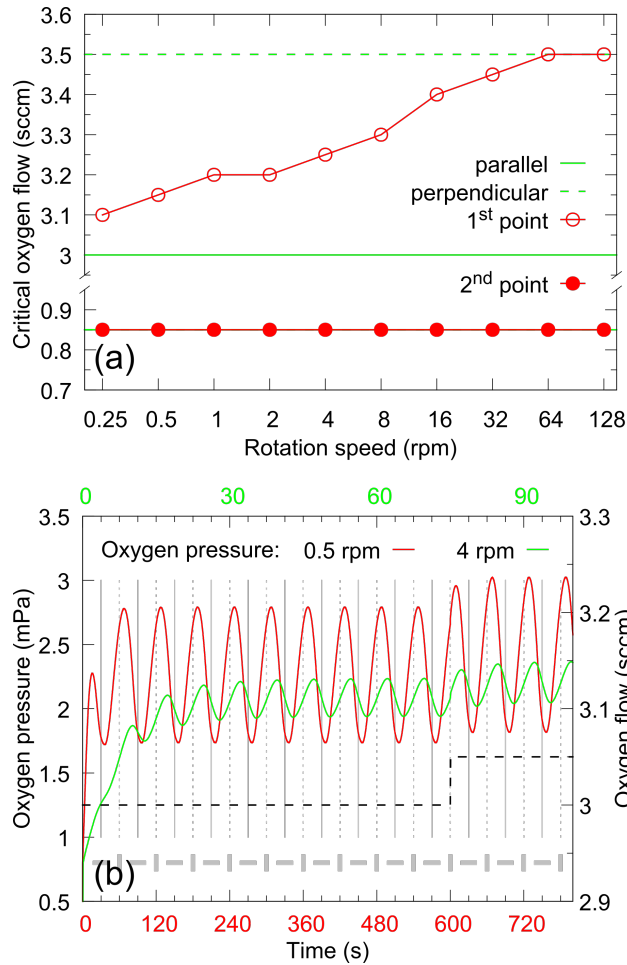


Figure 6: (a) The first (open markers) and second (closed markers) critical point as a function of the substrate rotation speed in rounds per minute (rpm). The first critical point calculated for two situations with a stationary substrate parallel to the target (full green line) and perpendicular to the target (dashed green line) are also indicated. (b) The time dependency of the oxygen pressure (left axis) when the substrate rotates at 0.5 rpm (lower axis) and 4 rpm (upper axis) in metallic mode. The value of the oxygen flow is indicated on the right axis. The grey vertical lines indicate when the substrate is parallel (dashed line) or perpendicular (full line) to the target. These positions of the substrate are indicated by the grey blocks in lower part of the figure.

267 From this reasoning it can also be understood that the influence of the
268 chamber size will be minor as in the simulation the wall surface area is already
269 sufficiently large to permit a sufficient spread of the sputtered material. The
270 influence of the chamber size will mainly influence the maximal value of the
271 first critical point. The lower first critical oxygen flows at low rotation speeds
272 is the result of larger fluctuations in the oxygen pressure. In this way the
273 critical pressure to induce the transition from metallic to poisoned mode is
274 easier accessed. By increasing the rotation speed, the average getter capacity
275 increases which requires higher oxygen flows to poison the process. The
276 above reasoning is confirmed by two simulations with a stationary substrate:
277 a parallel and a perpendicular substrate. The first critical point gradually
278 changes from the low value for the parallel configuration (full green line in
279 [Figure 6a](#)) to the high value for the perpendicular configuration (dashed green
280 line in [Figure 6a](#)). The simulations also show that the second critical point is
281 not affected by the sample rotation. Indeed, the return from poisoned mode
282 is solely defined by target processes.

283 **6. Example 4: Pulsing the discharge current**

284 In some cases, such as to avoid arcing, it can be beneficial to use a pulsed
285 current instead of a direct current. As target poisoning occurs at a time scale
286 of 0.1 to 10 seconds, it can be expected that at high frequencies no differences
287 in the hysteresis behavior is observed. This can indeed be observed in [Fig-
288 ure 7a](#). Above a frequency of 20 Hz the first and second critical point almost
289 coincide with the points simulated for the DC case (0 Hz). In the simula-
290 tion, the change of the discharge voltage with the changing discharge current

291 is included. This results in a minor increase of the average sputter yield
292 for the pulsing current simulations as compared to the DC case, which ex-
293 plains the small difference between the critical points at high current pulsing
294 frequencies. At low frequency, no hysteresis is observed as the two critical
295 points coincides. This is in agreement with the experiments by Billard et
296 al. [23, 24, 25] who demonstrated that hysteresis can be avoided by pulsing
297 the discharge current at low frequencies. The elimination of the hysteresis
298 can be understood as follows. If the oxygen flow is high enough, the target
299 surface oxidizes completely at low current, but is completely cleaned again
300 in the high current regime of the same cycle. At intermediate frequencies,
301 this is not possible anymore, and the hysteresis starts to widen. When the
302 target condition is followed in time at a frequency within this intermediate
303 regime, the complexity of the poisoning mechanism becomes visible. This
304 is illustrated in Figure 7b which shows the time dependency of the average
305 chemisorbed fraction $\bar{\theta}_c$ and the average oxide fraction $\bar{\theta}_r$ together with the
306 ion current (right axis). The chemisorbed fraction is defined as the target
307 compound fraction formed by chemisorption of oxygen molecules. Oxygen
308 is also implanted into the target. Due to the target erosion the implanted
309 oxygen travels towards the target surface, and as discussed before, it can
310 react with the target material during this journey to form compound. The
311 average target fraction covered by the oxide is given by $\bar{\theta}_r$. When the dis-
312 charge current increases, it is expected that the compound layer is sputtered,
313 and both fractions starts to decrease. This occurs however not immediately.
314 When the discharge current is zero, the implanted compound can further
315 react with the target material, slightly increasing the oxide fraction.

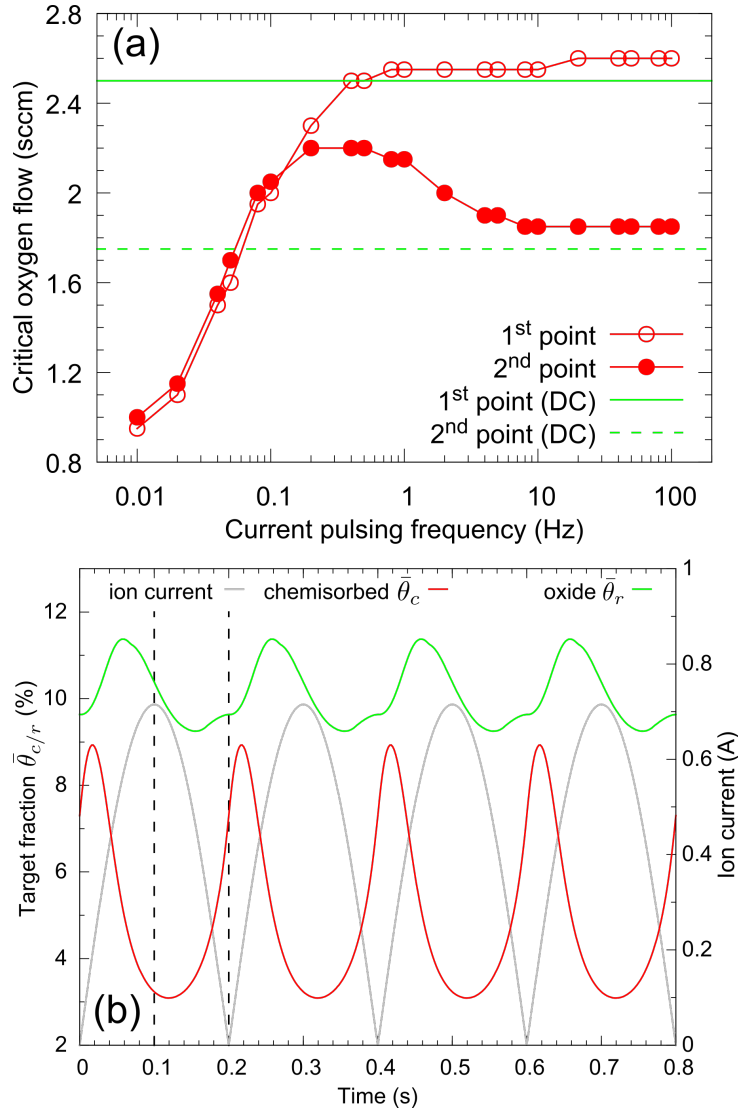


Figure 7: (a) The first (open markers) and second (closed markers) critical point as a function of the current pulsing frequency. The green lines indicate that first and second critical point for direct current (DC) conditions. (b) The time dependency of the average chemisorbed fraction $\bar{\theta}_c$ and the compound fraction $\bar{\theta}_r$. On the right hand axis the ion current to the target is shown. The current pulsing frequency was set to 5 Hz. The simulations were performed at an oxygen flow of 2.5 sccm, i.e. just before the first critical point. The simulation were performed for a two inch planar cylindrical aluminum target.

316 Also, oxide will be formed deeper in the target. When the discharge
317 current increases, the formed oxide travels towards the surface, and increases
318 the oxide fraction θ_r . As the reactive gas cannot chemisorb onto the oxide
319 fraction, its fraction will also decrease. The decrease will occur faster due to
320 sputtering. Nevertheless, due to the implantation, and the formation of an
321 oxide layer thicker than one monolayer, the target condition is not in phase
322 with the modulation of the discharge current. Although there is currently no
323 experimental evidence for this behavior, this behavior shows once more the
324 complexity of reactive magnetron sputtering.

325 7. Conclusion

326 Despite the improvements in understanding reactive sputtering by the
327 development of the RSD model, there is still a long way in going beyond
328 the "*what if*" questions. Two main obstacles need to be taken. The first is a
329 further mapping, and implementation of other target processes. For example,
330 diffusion of reactive species in the target seems to play a key role in the time
331 dependency of the poisoning process. The second obstacle is one of the
332 classical problems with modeling: finding reliable parameters. For a large
333 number of target/reactive gas combination there is a lack of fundamental
334 parameters such as incorporation coefficients and sputter yields. If the thin
335 film community wishes to obtain a predictive and quantitative understanding
336 of reactive sputtering, this information is a necessity. This sounds pessimistic,
337 but actually it is also an opportunity to keep *sputter deposition of thin films*
338 *a vibrant and active field* [1].

339 Acknowledgments

340 This paper is a collective effort by all current members of the research
341 group DRAFT who have used the RSD code to investigate a specific case
342 study. As their contribution is equal, their names are put in random order,
343 except for the first, second and last author. Special credits go to the second
344 author who have pushed the RSD model further to include more tools, and
345 to improve the description of the target processes. The authors also wish
346 to thank all previous team members who, by experiments and stimulating
347 ideas, enabled to seek for the limits of the RSD model.

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