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MODELLING OF BURNING AND EXTINCTION CHARACTERISTICS OF A POLYMER DIFFUSION FLAME AND COMPARISON WITH EXPERIMENT

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INTRODUCTION

The importance and contribution of the inherrent flammability of polymeric materials to problems of fire safety is well recognized. This study was undertaken to provide a better understanding and interpretation of previous experimental investigations of polymer flammability. The structure of opposed flow diffusion flames have been measured by several investigators using both gaseous 1,2 and solid 3,4,5 fuels. The opposed flow diffusion flame is a convenient geometry for the study of the flammability of polymers because it allows both steady, diffusion controlled burning and extinction to be observed under well-controlled laboratory conditions. Conveniently available experimental parameters include fuel composition, oxidizer composition, and oxidizer blowing rate. Reported experiments generally have not included the variation of pressure or temperature. Radiation effects, which are important in fires, remain to be well-characterized in laboratory studies of opposed flow diffusion flames where radiation generally is of minor importance.

Theoretical descriptions of opposed flow diffusion flames have been provided by several investigators⁶⁻¹⁰. In most cases the flame sheet approximation of Burke and Schumann¹¹ has been invoked, generally to allow a simplified description and analytical solution. The flame structure in these models is simplified to a surface which is the locus of stoichiometric composition where an infinitely rapid reaction occurs. Experimentally we have observed that the flame zone is not thin, especially near extinction conditions, with oxidizer penetrating to the fuel side of the flame and fuel or intermediate, partially reacted fuel (primarily carbon monoxide) penetrating to the oxidizer side⁵.

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A previous study⁵ of the steady-state burning and extinction characteristics of high purity poly(ethylene) and mixtures of nitrogen and oxygen provides experimental data on the structure of the reaction zone in terms of composition and temperature, and steady burning rate data. Fundamental data on polymer pyrolysis product properties, reaction mechanisms, and reaction rates are not available and had to be approximated from earlier experiments or chosen to provide a fit between the model and experimental observations.

FORMULATION

Reacting flow in a stagnation point boundary layer was modelled with the objective of examining its properties near extinction, Fig. 1. Axi-symmetric geometry, laminar-flow and steady-state combustion were considered to conform to an existing diffusion flame burner⁵. Variation of fluid properties with both temperature and composition was included. The specific heats of the three species considered, fuel, oxygen and products were chosen to be those of ethylene, oxygen and nitrogen respectively, and approximated as cubic functions of temperature. The effective viscosity was obtained from the individual, temperaturedependent viscosities using Wilke's mixture rule¹². The Prandtl number was fixed at 0.7, which is a good approximation for the reactants and products considered. The Lewis number was assumed to be 1.

Finite reaction rate chemistry was incorporated with a one-step reaction and Arrhenius rate expression. Three species were considered: fuel, oxygen and products. The reaction rate was of order one with respect to both fuel and oxygen.

The boundary conditions for the N_2/O_2 flow were an ambient

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temperature of 294K and pressure of 1 atm. The polymer burning rate was calculated from the enthalpy gradient at the fuel surface and the effective latent heat. The latent heat was extracted from the mass transfer B number, previously reported by Holve and Sawyer⁸ based on their experimental observation of the combustion poly(ethylene) in an opposed flow diffusion flame, $B = 3.48Y_{0} \approx -0.18$, where $Y_{0} \approx$ is the oxygen mass fraction in the ambient N_2/O_2 flow. The fuel surface temperature was specified at 800K, also based on experimental observation⁸. The oxygen concentration at the fuel surface was set at zero. The fuel surface mass fraction, Y_{fw} , results from species and energy balances, $Y_{fw} = (B - Y_0 \infty/s)/(1 + B)$ (Eq. 1), where s is the stoichiometric ratio, mass fuel consumed per mass of oxygen⁷.

RESULTS AND DISCUSSION

Solutions to the boundary layer equations were found by using $GENMIX^{13}$, which solves the finite difference equations in implicit form. The flame structure was found for both finite and infinite reaction rate cases for an oxygen mass fraction, $Y_0\infty$, of 0.257 as shown in Figs. 2 and 3. If only a solid curve appears on the profiles, the finite and infinite reaction-rate curves are superimposed for this graph size. For an infinite reaction rate, the oxygen and fuel concentrations go to zero at the maximum temperature location which will be designated the flame. When the reaction rate is finite, some of the fuel and oxygen survive and are transferred through the flame. Transfer of molecular oxygen through the flame was also found experimentally⁵ as shown by the experimental points in Fig. 2.

Calculated temperature profiles for finite and infinite reaction rates and a measured temperature profile are presented in Fig. 3. The maximum

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Figure 2. Calculated species profiles for finite (---) and infinite (---) reaction rates. Measured oxygen profile (Δ).



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Figure 3. Calculated temperature profiles for finite (----) and infinite (---) reaction rates. Measured temperature profile (□). Calculated velocity profile.

temperature for the infinite rate case was adjusted to be the adiabatic equilibrium temperature by choice a multiplicative constant in the product specific heat. The maximum temperature for the infinite rate case is higher than the finite rate case. When the reaction rate is finite, heat transfer from the flame has time to occur, and the flame cools. Both calculated temperatures were higher than the measured temperatures taken in the opposed flow apparatus⁵. A thermocouple bead of 0.18mm diameter was used in the measurements. Its characteristic measurement volume resulted in some smoothing of the peak in the temperature profile. Additionally, the temperatures were reported without correction for radiation losses.

In comparing infinite to finite reaction rate cases, the flame temperature drops and the flame moves closer to the surface, Fig. 3. The surface temperature gradient and thus the burning rate, however, is unchanged. The calculated burning rate is 29 g/m²s whereas the measured value is 8.3 g/m²s. This discrepancy could have resulted from the calculated flame temperatures being too high and a failure to consider surface radiation loss in the model. The peak temperature from measurement and the calculated peak temperature from the infinite rate case are at the same location which is a result of fitting rather than prediction. The peak temperature location in the measured profile was used to determine the velocity gradient at the boundary layer edge which is requisite for the calculation. The location of unity equivalence ratio, frequently used to denote the flame location, is the same for both the finite and infinite rate cases.

The calculated boundary layer thickness for temperature is less than the measured one, Fig. 3. The actual flame thickness is probably greater

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than that predicted by a one-step, finite rate reaction. The inclusion of complex flame chemistry would spread out the flame. For one-step chemistry, all the chemical reaction occurs very near the maximum temperature location. For an actual diffusion flame, fuel pyrolysis reactions occur near the surface, H_2^0 formation occurs near the peak temperatures, and CO_2 formation occurs on the oxidant side of the flame².

The velocity profiles for the finite and infinite reaction rate are identical on the graphic scale presented. The variable u/u_{max} is plotted since it is independent of radial location. The location of u_{max} is between the flame and fuel surface for the following reasons. Since the viscosity is temperature dependent, it is maximum at the flame. As indicated in Fig. 1, the streamlines pass through the flame convecting the flow into the lower viscosity region where the maximum velocity is attained.

Calculated flame stand-off distances for finite and infinite reaction rates and measured stand-off distances are shown in Fig. 4. All flame stand-off distances presented decrease with decreasing ambient oxygen mole fraction, $X_{0\infty}$. When $X_{0\infty}$ is lowered, the flame temperature is reduced, which decreases temperature gradient at the surface and thereby the fuel blowing rate. The fuel mass fraction at the surface is also decreased since the B number is lowered, Eq. 1. Both of these effects reduce the flame stand-off distance. It is crucial to include finite rate flame chemistry when predicting flame stand-off distances near extinction, Fig. 4. When $X_{0\infty}$ is increased, the flame condition moves out of the extinction regime and the flame stand-off distance predictions are the same for both finite and infinite rate cases. If the flame is near extinction, the net fluid motion through the flame and towards the surface becomes important.

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Figure 4. Flame stand-off distance calculated for finite (--) and infinite (--) reaction rate and measured by experiment (\blacksquare) .

The reactants are convected for significant distance before they are fully reacted and the maximum temperature is reached.

Overall reaction-rate parameters were extracted from extinction data. Extinction measurements in the opposed flow apparatus were performed for poly(ethylene) burning in $\mathrm{N_2/0_2}$ mixtures, Fig. 5. The values of the reaction-rate parameters were iterated until the calculated extinction curve approximated the least squares fit of the data. The resultant activation energy, ${\rm E}_{\Delta},$ and pre-exponential coefficient, K, are defined by the following Arrhenius expression, $\dot{m}_f = K p^2 Y_f Y_0 exp (-E_A/RT)$, where E_A is 31 ± 3 kcal/gmole, K is 5.4 x 10⁻² kg/(Pa)²m³s, \mathring{m}_f is the mass consumption rate of fuel per unit volume in kg/m 3 s, p is the pressure in Pa, \boldsymbol{Y}_{f} and \boldsymbol{Y}_{0} are the fuel and oxygen mass fractions, R is the gas constant and T is the temperature in K. These overall rate constants were found for flame temperatures less than 1850K as indicated by calculated temperatures. The extinction condition is very sensitive to the value of $X_0 \infty$. The flame is in a large activation energy regime where the reaction rate is mainly controlled by the flame temperature. A small change in $X_0\infty$ alters the flame temperature which significantly affects the reaction rate and thus the extinction condition.

The flame temperature at extinction increases slightly with nozzle velocity, Fig. 6. For higher nozzle velocities a faster reaction rate is required for the flame to survive. The reaction rate from the calculated activation energy is a strong function of temperature. Therefore, a slight increase in flame temperature was sufficient to accomodate the higher nozzle velocities. The calculated flame temperatures at extinction are lower than the infinite-rate temperatures. The infinite-rate temperatures

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Figure 5. Measured extinction conditions for PE burning in N2/O2 mixtures (Δ) and calculated extinction conditions ().



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are the adiabatic flame temperatures. At extinction, the calculations indicated about a 16% reduction in flame temperature from the infinite reaction rate value.

CONCLUSIONS

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- 1. The applicability of a boundary layer solution with a finite reaction rate to model the opposed flow diffusion flame combustion of poly(ethylene) has been demonstrated. Penetration of oxygen through the flame is predicted, in agreement with experimental observation.
- 2. The prediction of flame stand-off distance near extinction requires consideration of finite rate chemistry.
- 3. A global reaction rate for poly(ethylene) combustion in a diffusion flame was obtained through fitting extinction data to the model. The activation energy is 31 ± 3 kcal/gmole and the pre-exponential coefficient is 5.4 x 10^{-2} kg/(Pa)²m³s.
- 4. The maximum radial velocity occurs between the flame and the surface.

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