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NEW METHOD FOR ANALYZING SMALL-SCALE FRACTURE-SPECIMEN DATA IN THE TRANSITION ZONE\*

**MASTER**

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Among the problems related to the use of small specimens for measuring fracture toughness, those concerning size effects and data scatter are perennial. Figure 1 shows an early case encountered by the HSST Program. These data are from Compact Specimens of three different sizes, for an irradiated A508 Class 2 forging steel.<sup>1</sup> Later on, as shown in Fig. 2, substantial size effects and data scatter were encountered in the material characterization and experimental phases of HSST Thermal Shock Experiment TSE-5A. The line labeled  $K_{IC}$  was drawn as a lower bound to the small specimen data before the test, but the actual test data, indicated by the solid triangles, fell below the line.<sup>2</sup>

Although the results shown in Fig. 2 were postulated to be statistical in nature, due to randomly dispersed brittle zones, metallographic examination failed to locate any such atypical regions.<sup>3</sup> In addition, a statistical approach to the problem of size effects and data scatter would be likely to require more specimens than are available in a surveillance capsule. Consequently, an attempt was made to find a suitable method for adjusting individual small specimen fracture toughness values for size effects in the transition range of temperature. The method selected was one already proposed by Irwin.<sup>4</sup> As illustrated in Fig. 3, taken from a study by Corten and Sailors,<sup>5</sup> Irwin's  $\beta_{IC}$  equation recognizes an interaction between toughness and size. If either toughness increases or size decreases, the ratio  $K_C/K_{IC}$  will increase. This interaction magnifies the scatter

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inherent in plane strain  $K_{IC}$  values. Although the more common application of the  $\beta_{IC}$  formula is the estimation of  $K_{IC}$  values from known values of  $B$  and  $K_{IC}$ , the original application<sup>4</sup> was the one considered here, i.e., the estimation of  $K_{IC}$  from measured values of  $B$  and  $K_C$ . So the new aspect of the application described here is mainly the use of small specimen test data, analyzed inelastically.

An algebraic development of the  $\beta_{IC}$  adjustment equation is described in Figs. 4 thru 6, and trial results, for both static and dynamic data, are shown in Figs. 7 thru 14. In Figs. 7 thru 14, the open points are the original small specimen toughness values, the closed points of the same shape are the corresponding  $\beta_{IC}$  adjusted values, and the solid triangles are valid or large specimen test data. The appropriate ASME Code  $K_{IC}$  or  $K_{IR}$  curves are shown for comparison. The original test data were obtained from References 2, and 6 thru 8.

The above results<sup>9</sup> are not without apparent contradiction, however.<sup>10</sup> Figure 15 shows that maximum load toughness values for A533-B steel plate show little data scatter or size effects. And as shown in Fig. 16, the same is true, with respect to data scatter, for the cylinder plate of HSST vessel V-9. However, Fig. 17 shows that the weld metal in vessels V-8 and V-9 develops considerable data scatter and size effects.<sup>11</sup> These observations concerning differences in the degree and scatter between plate, forgings and weld metal appear to be common, although unexplained.

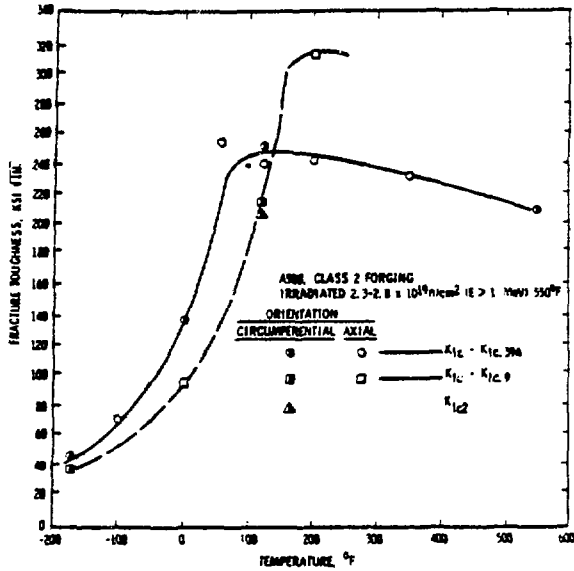
The question of the presence or absence of size effects also appears to involve differences between plate and forgings, at least for static data. It also involves the definition of the toughness measurement point. Figure 10 shows definite size effects in static data at the point of cleavage instability, for A533-B steel, and their removal by applying the  $\beta_{IC}$  formula. But Fig. 18 shows no appreciable size effects in the maximum load toughness values calculated for the same specimens,<sup>10</sup> although the final toughness values are higher in Fig. 18 than in Fig. 10. Fig. 19 shows the predictable results of applying the  $\beta_{IC}$  adjustment to the A533-B plate maximum load data shown in Fig. 15, in which no appreciable size effects were evident. This size effect enigma is probably due in large part to the fact, illustrated in Fig. 20, that the maximum load point is often not the point of onset of

unstable cleavage. It is hypothesized here that, although enough microscopically stable cleavage microcracking<sup>12, 13</sup> occurs to produce a temperature dependent maximum load toughness value, this value may not be a reliable  $K_{Ic}$  value because crack extension is predominantly by ductile tearing until the occurrence of unstable cleavage. This problem appears to be avoidable by using the point of onset of unstable cleavage as the toughness measurement point.

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Lower bound  $K_{1C-396}$  and  $K_{1C-9}$  fracture toughness of ASTM A508, class 2 forging irradiated to  $2.3-2.8 \times 10^{19} \text{ n/cm}^2$  ( $E > 1 \text{ MeV}$ ) at 550°F. Both circumferential and axial orientations are shown.

FIGURE 1

ILLUSTRATION OF THE APPLICATION OF THE IRWIN  $\beta_{1c}$  THICKNESS ADJUSTMENT EQUATION

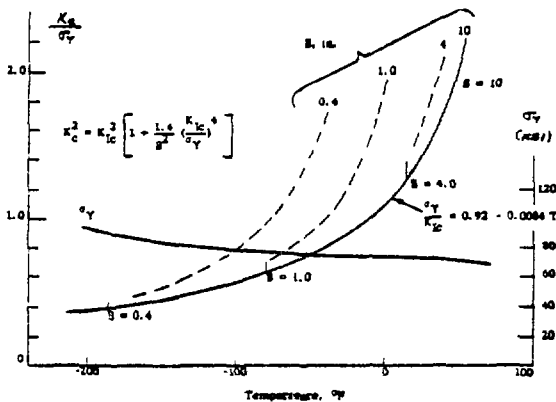


Illustration of "Thickness Fracture Mode Transition" Superimposed on "Fracture Micro-mechanism Transition" (Carpenter & Salinas)

FIGURE 3

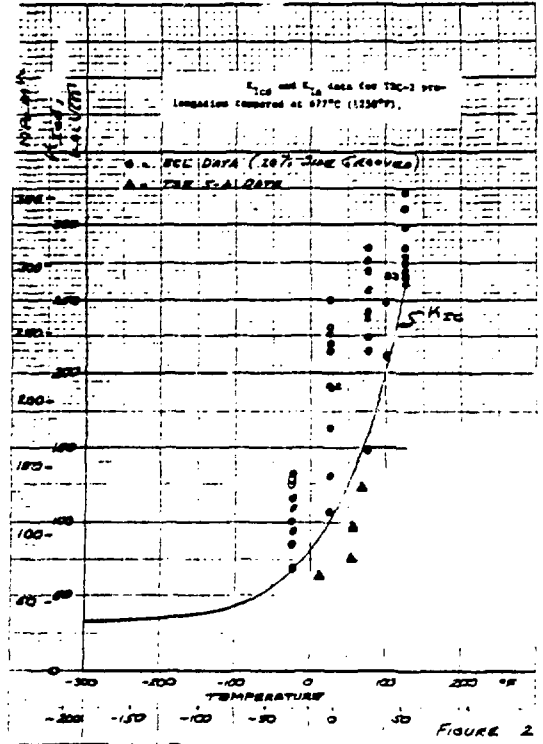


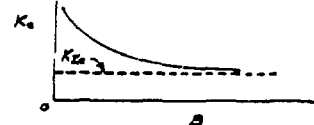
FIGURE 2

AN APPROACH TO THE DERIVATION AND APPLICATION OF IRWIN'S  $\beta_{1c}$  EQUATION FOR ESTIMATING CONSTRAINT EFFECTS

DEFINITIONS:

- $K_{IIc}$  = PLANE STRAIN FRACTURE TOUGHNESS
- $K_{Ic}$  = NON PLANE STRAIN FRACTURE TOUGHNESS

EXPERIMENTAL OBSERVATIONS:



ANALYSIS

LET  $\beta_{1c} = \frac{(K_{IIc})^2}{B}$  AND  $\beta_c = \frac{(K_{Ic})^2}{B}$

ASSUME

$\beta_c = \beta_{1c} + c\beta_{1c}^3$

(SIMPLEST NON-LINEAR ODD POWER POLYNOMIAL)

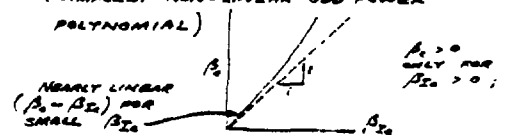


FIGURE 4

EVALUATION OF 'c' AT  $K_{Ic}$  LIMIT OF VALIDITY

$$A_1 = \beta_{Ic} = \frac{A}{\beta_{Ic}}$$

$$c = \frac{A_1}{\beta_{Ic}} - 1$$

FOR COMPLETE PLANE STRAIN

$$K_{Ic} = C \sigma_f \sqrt{a_0}$$

ORIGINAL CRACK SIZE  
PLANE STRAIN FRACTURE STRESS

TEST RESULT AT LIMIT OF VALIDITY

$$K_{Ic} = C \sigma_f \sqrt{a_0}$$

$$\frac{A_1}{\beta_{Ic}} = \frac{\sigma_f^2}{\sigma_c^2} \left( \frac{a}{a_0} \right)$$

ASSUME

$$\frac{\sigma_f}{\sigma_c} = 1.20 \text{ (STRAIN ENERGY RATIO)}$$

$$\frac{a}{a_0} = 1.02 \text{ (SEE ASTM E-399)}$$

THEN  $\frac{\beta_0}{\beta_{Ic}} = (1.20)(1.02) = 1.224$

ALSO,  $B = 2.3 \left( \frac{K_{Ic}}{\sigma_c} \right)^2$ , SO  $\beta_{Ic} = \frac{1}{2.5} = 0.4$

THEREFORE  $c = \frac{1.224 - 1}{0.16} = 1.4$

FIGURE 5

APPLICATION OF IRWIN'S  $\beta_{Ic}$  EQUATION

$$\beta_0 = \beta_{Ic} + 1.4 \beta_{Ic}^3$$

CALCULATION OF  $K_{Ic}$  WHEN  $K_{Ic}$  IS KNOWN :

$$\frac{K_c}{K_{Ic}} = \sqrt{1 + 1.4 \beta_{Ic}^2}$$

CALCULATION OF  $K_{Ic}$  WHEN  $K_c$  IS KNOWN :

$$\beta_{Ic}^3 + \frac{5}{7} \beta_{Ic} - \frac{5}{7} \beta_c = 0$$

$$\beta_c = \frac{\left( \frac{K_c}{\sigma_c} \right)^2}{B}; \text{ LET } m = \frac{5}{14} \beta_c$$

NOTE THAT  $\left( \frac{5}{7} \right)^3 = 0.0135$

$$A_1 = \sqrt{m^2 + 0.0135} + m$$

$$A_2 = \sqrt{m^2 + 0.0135} - m$$

$$\beta_{Ic} = A_1^{1/3} - A_2^{1/3}$$

$$K_{Ic} = K_c \sqrt{\frac{\beta_{Ic}}{\beta_c}}$$

FIGURE 6

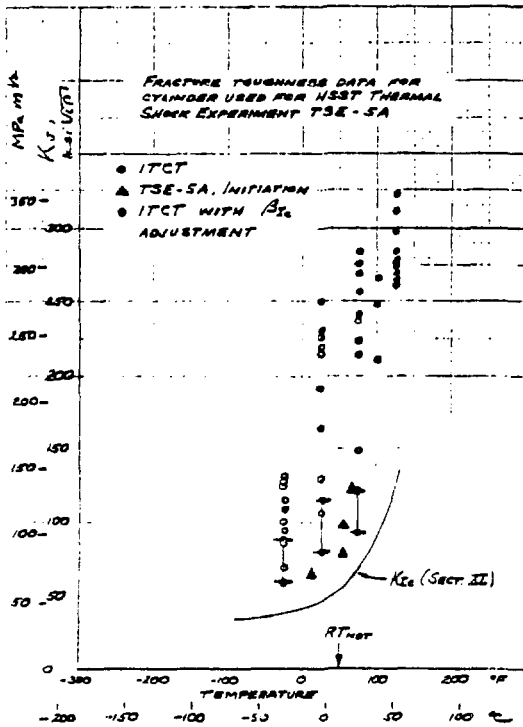
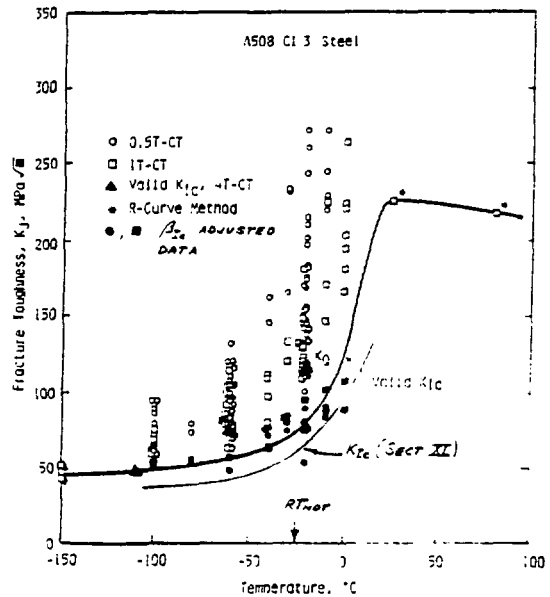
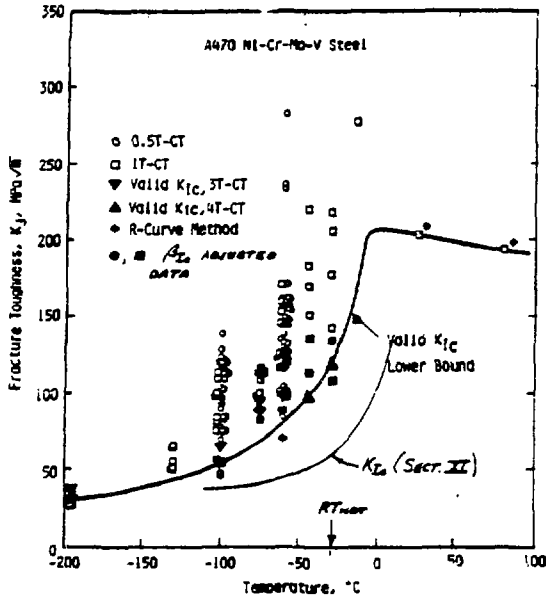


FIGURE 7



(DATA FROM IWABATE ET AL.)

FIGURE 8



$K_j$  for fracture calculated from  $J_0$  versus temperature relation for A470 Ni-Cr-Mo-V rotor steel; (DATA FROM INABATE ET AL.).

FIGURE 9

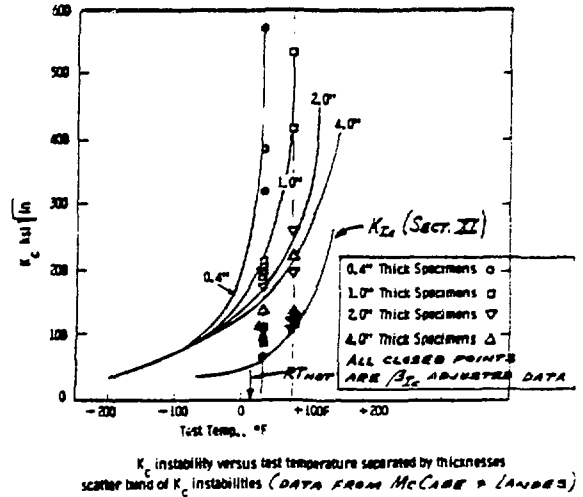
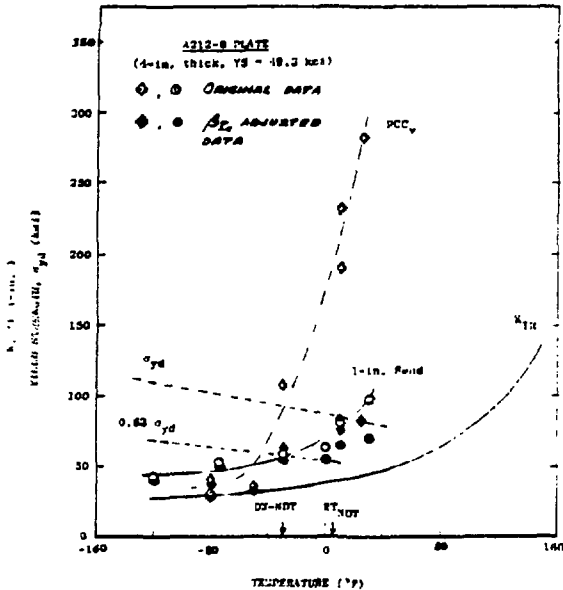
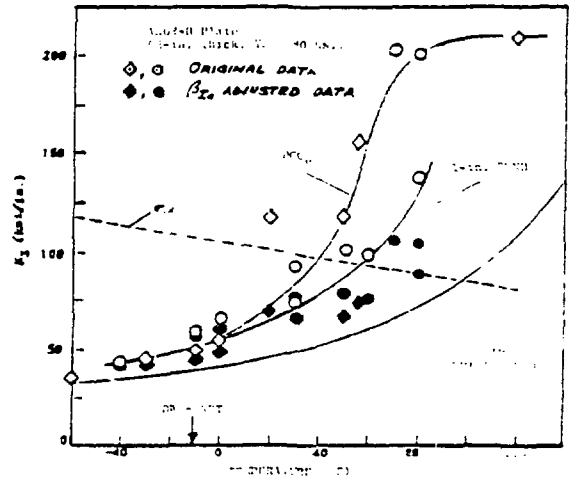


FIGURE 10



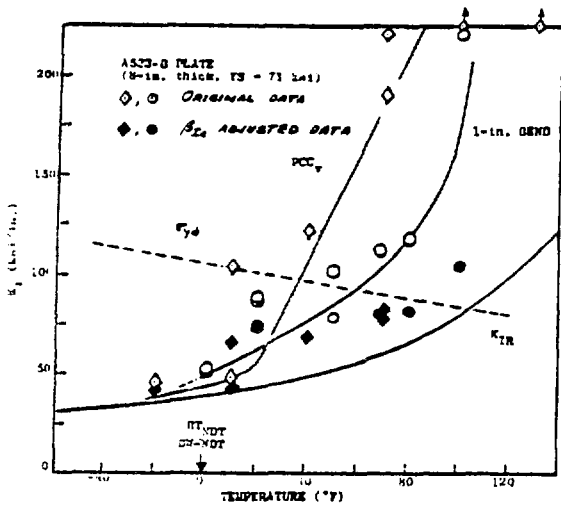
Comparison of dynamic fracture toughness trends for the A302-B reference plate. Data 1-in. bend and PCC specimens exhibited a  $K_{II} = 2.2 \times 10^{-10}$  (ksi√in.)<sup>2</sup>. The open symbols represent fracture before general yielding; the filled triangles were determined by the R-curve method. (DATA FROM LANDB).

FIGURE 11



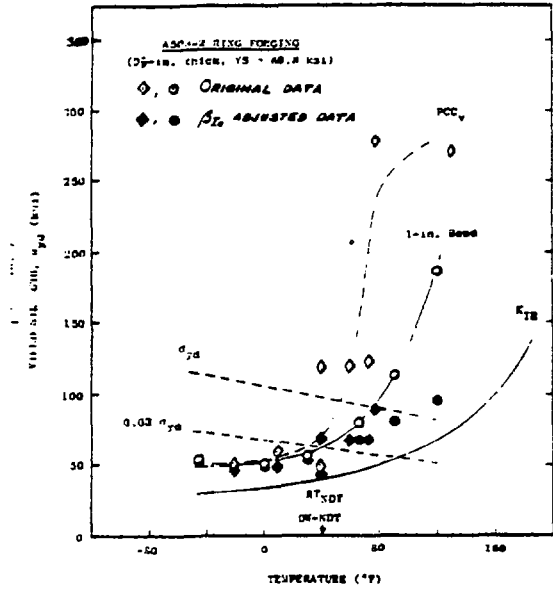
Comparison of dynamic fracture toughness trends for an A302-B plate. Data 1-in. bend and PCC specimens exhibited a  $K_{II} = 2.2 \times 10^{-10}$  (ksi√in.)<sup>2</sup>. Data from LanDB.

FIGURE 12



Comparison of dynamic fracture toughness trends for an A333-B Class 1 plate. Both 1-in. bend and PCC<sub>1</sub> specimens exhibited a K<sub>Ic</sub> of 2.50 x 10<sup>3</sup> ksi/in. sec.<sup>2</sup>; (Data from Loebe).

FIGURE 13



Comparison of dynamic fracture toughness trends for an A508 Class 2 forgings. Both 1-in. bend and PCC<sub>1</sub> specimens exhibited a K<sub>Ic</sub> of 2.5 x 10<sup>3</sup> ksi/in. sec.<sup>2</sup>; (Data from Loebe).

FIGURE 14

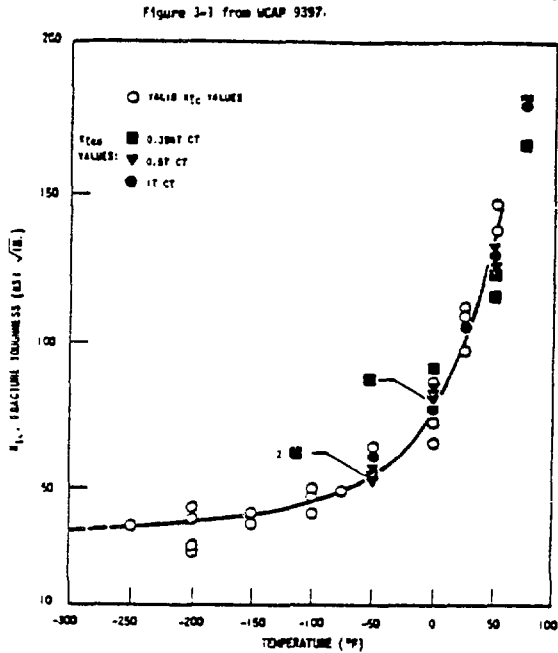


Figure C Comparison of Valid K<sub>Ic</sub> Results in the Lower Transition Region with Lower Bound (K<sub>L</sub>) Results Obtained for Charpy Thickness (0.25 in., 0.5 in. and 1 in.) Compact Specimens - A533 Grade B Class 1 Plate (MSST Plate 02)

FIGURE 15

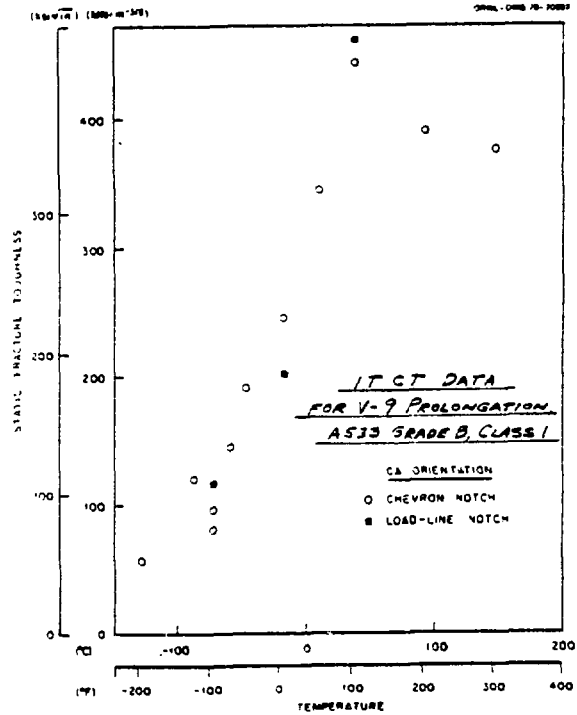


FIGURE 16



STATIC FRACTURE TOUGHNESS OF V-8 & V-9

PROLONGATION WELD METAL

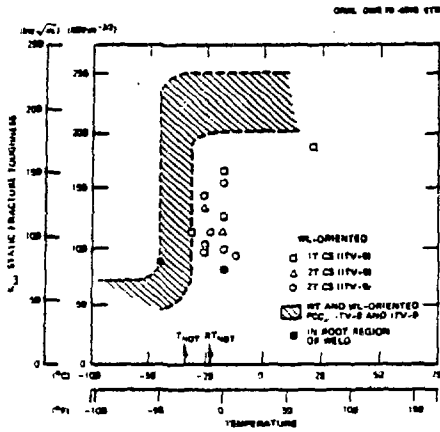


FIGURE 17

MAXIMUM LOAD K\_Ic VALUES

FOR A533 GRADE B, CLASS 1, PLATE

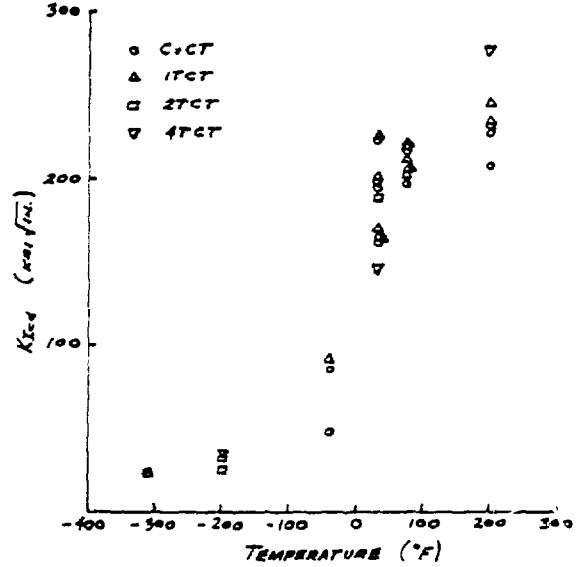


FIGURE 18

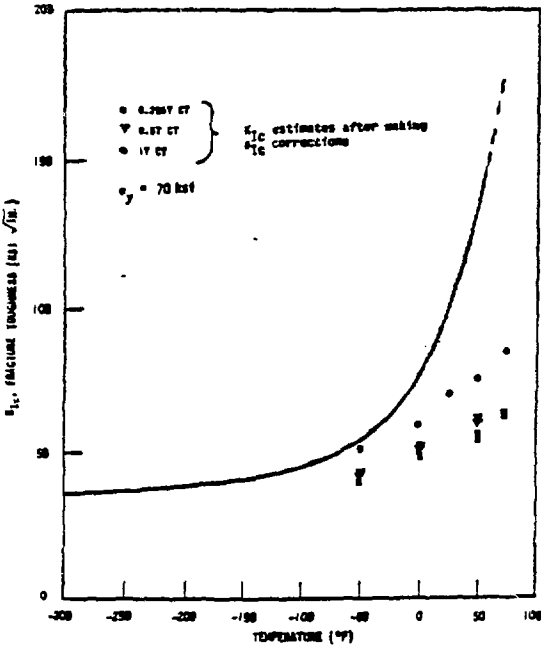


Figure 19 Comparison of  $K_{Ic}$  estimates using  $S_{Ic}$  correction with valid  $K_{Ic}$  values<sup>(1)</sup> (HSST Plate 02)

FIGURE 19

EXAMPLE OF A LOAD-DISPLACEMENT CURVE FOR A SPECIMEN IN WHICH UNSTABLE CLEAVAGE OCCURRED AFTER THE MAXIMUM LOAD POINT

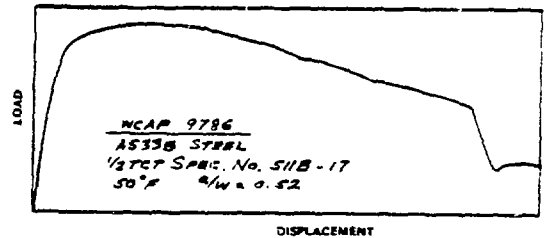


FIGURE 20